

An Open-Source Numerical Model for Mitigating Refractory Alloy Hot Cracking Susceptibility

Jeffrey W. Sowards, Andrew O'Connor, Fredrick N. Michael, Carly J. Romnes,
Fernando Reyes Tirado, Omar Mireles

Marshall Space Flight Center



- Problem & Motivation
- Background on the Model and Algorithm
- Algorithm Verification vs Past Aluminum Alloy Studies
- Algorithm Verification vs Past Refractory Alloys Weldability Data
- Extrapolations to Refractory-Interstitial (O,C,N) Binary Alloys
- Extending the Approach for Development of Hot Cracking Susceptibility Equations

Background on Refractory Metals



- Refractory metals and alloys are used for service in extreme high temperature environments:
 - Reaction Control System (RCS) thrusters
 - Space Nuclear Propulsion (SNP) clad and structure
 - Hypergolic / green propulsion chambers and catalyst
 - Electric propulsion grids
 - Power conversion system heat pipes and regenerators
 - Hypersonic wing leading edges
- Refractory metals are desirable due to:
 - High melt temperature (T_m)
 - Retain strength and hardness at elevated temperature
 - Corrosion and wear resistant (outside of propulsion)
- Aerospace refractory metal parts tend to be:
 - Thin-walled geometries (converging-diverging nozzles)
 - Relatively simple geometries
 - High buy-to-fly ratio (20:1 to 50:1)
 - Low production rate



Apollo CSM RCS using C103.
Courtesy Aerojet-Rocketdyne



TZM alloy heat pipe.
Courtesy Advanced Cooling Technologies.



Green propulsion Re thruster.

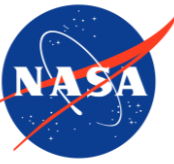
Base	Name	Composition (wt%)
Nb	Nb	Nb
	Nb-1Zr	Nb-1Zr
	C103	Nb-10Hf-1Ti
	C129Y	Nb-10Hf-10W-0.1Y
	Cb752	Nb-10W-2.5Zr
	C3009	Nb-30Hf-10W
Mo	WC3015	Nb-28Hf-13W-5Ti-2Ta-1Zr
	FS85	Nb-28Ta-10W-1Zr
Mo	Mo	Mo
	Mo-21Re	Mo-21Re
	Mo-41Re	Mo-41Re
	Mo-44Re	Mo-44Re
	Mo-47.5Re	Mo-47.5Re
W	TZM	Mo-0.5Ti-0.08-Zr-0.2C
W	W	W
	W-25Re	W-25Re
Ta	Ta	Ta
	Ta-10W	Ta-10W
Ir	Ir	
	DOP26	Ir-0.3W-0.006Th-0.005Al
Re	Re	Re

Traditional Refractory Alloys



X-51A hypersonic test vehicle. Courtesy USAF.

Problem and Goal: Fabricating Refractory Alloys



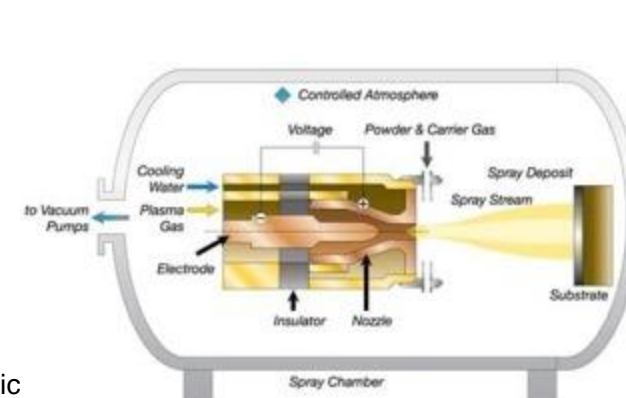
- Typically exhibit poor weldability. Existing alloys were design 60+ years ago and never optimized to be weldable and printable.
 - Thermal shock (thermal stress builds due to extreme high melting point)
 - Brittleness at room temperature (due to shift of ductile to brittle transition)
 - Solidification cracking (due to segregation of alloying elements and wide solidification temperature ranges induced by alloying)
- Traditional refractory manufacture is difficult and expensive:
 - Bar, plate, tube, sheet stocks and sizes limited (constrains design)
 - Powder feedstock are angular and not usually alloyed
 - High feedstock cost
 - Relatively difficult to form/machine (fracture prone)
 - Heat treatment requires specialized facilities (O, C, N sensitive)
 - Joining options limited (Usually electron beam welded)
 - Inspection options limited
- Alloys designed for traditional manufacture:
 - Powder metallurgy (CIP, HIP, deposition)
 - Forging
 - Wire and/or plunge EDM
 - W (\$100/kg) or Mo (\$80/kg) alloyed with 25-47.5 wt% Re (\$2.76k/kg) to improve ductility
- **Goal. Develop new refractory alloys using a CALPHAD approach, optimized for printability with L-PBF L-DED and weldability by reducing solidification cracking susceptibility**



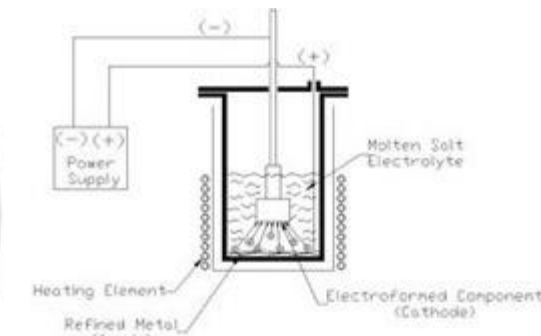
C103 forged bar stock.
Courtesy ATI.



Hot Isostatic Press (HIP) process [1].



Vacuum Plasma Spray (VPS) process [2].



Electro Deposition / Forming process
[3].

[1] <https://www.malvernpanalytical.com/en/industries/advanced-manufacturing/powder-metallurgy/isostatic>

[2] <https://www.neodynamiki.gr/>

[3] <https://plasmapro.com/processes/>

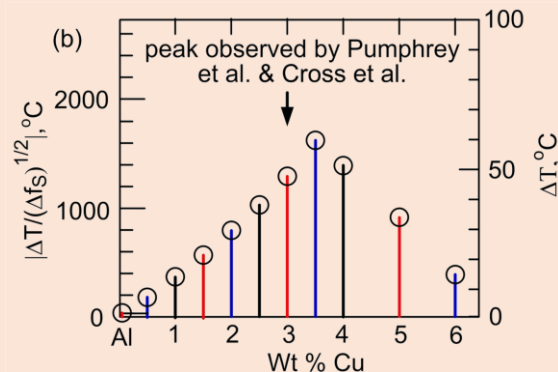
Model: Kou's Solidification Cracking Criterion



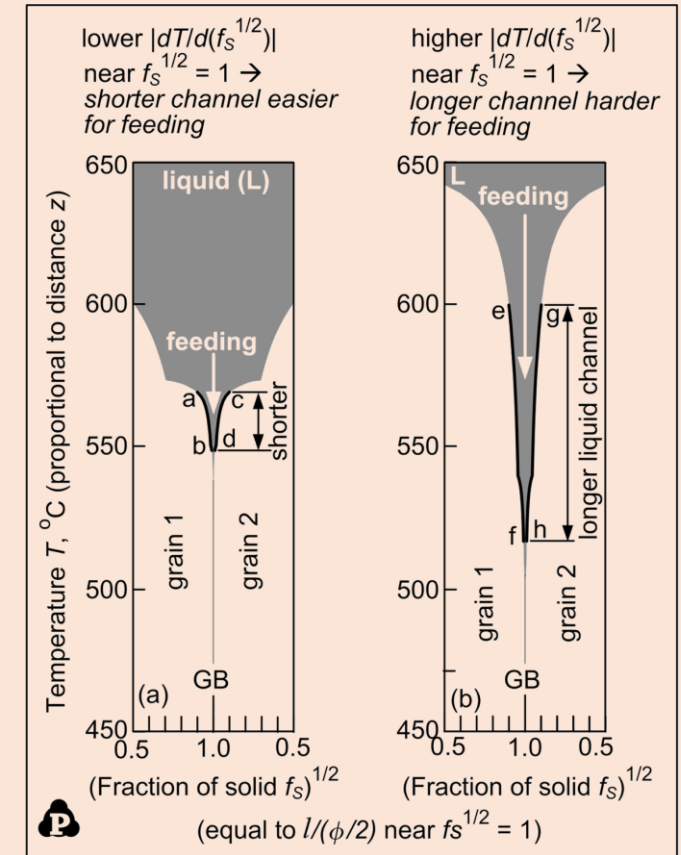
Kou's Cracking Criterion [4]:
$$\left\{ \underset{\text{(separation)}}{V_{\text{local}}} > \underset{\text{(growth)}}{\phi \sqrt{1-\beta} \frac{d\sqrt{f_s}}{dT} \frac{dT}{dt}} + \underset{\text{(feeding)}}{\phi \frac{d}{dz} [(1 - \sqrt{1-\beta} \sqrt{f_s}) v_z]} \right\} \sqrt{f_s} \rightarrow 1$$

- Considers a balance between grain boundary separation (cracking), lateral growth of grains, and liquid feeding between dendrites
 - v is velocity, ϕ is dendrite diameter, β is shrinkage, T is temperature, f_s is fraction solid
- Crack susceptibility increases as $|dT/d(f_s^{1/2})|$ increases near $f_s^{1/2} = 1$.
 - $f_s^{1/2}$ significance is similarity to dimensionless radius of dendrite
 - Steepness of solidification path near terminal solidification results in higher index: suggesting increased crack susceptibility due to slower transverse growth rate and longer passageway for feeding
- Criterion does not predict occurrence but rather susceptibility.
- The Scheil equation is used to predict the solidification path of an alloy, i.e., the plot of f_s vs T and usefully couples to this criterion for evaluating influence of composition.

Composition Influence



Geometrical Significance



[4] Kou. Acta Mat 88 (2015): 366-374
<https://doi.org/10.1016/j.actamat.2015.01.034>

Process Flow and Algorithm to Compute Crack Susceptibility



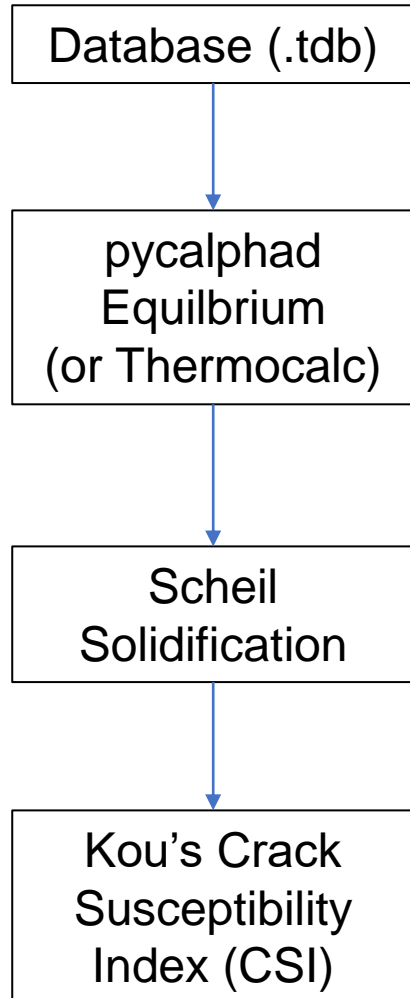
In this work, we numerically implement calculation of Kou's CSI in a Jupyter Notebook with python scripting.

[5] de Walle et al. *Calphad* 61 (2018): 173-178
<https://avdwgroup.engin.berkeley.edu/>

[6] Otis & Liu. *J. Open Res. Soft.* 5 (2017): 1
<https://pycalphad.org/>

[7] Bocklund, et al. (2020).
<https://github.com/pycalphad/scheil>

[4] Kou. *Acta Mat* 88 (2015): 366-374
<https://doi.org/10.1016/j.actamat.2015.01.034>



Jupyter Notebook Flow for pycalphad (Python 3)

```
import Dependencies #pycalphad and math packages
Variables = database, elements, phases
Conditions = start_temp, temp_step, filter #Scheil setup
Scheil = T vs fs plot #Calculate
fsnew = sqrt(fs) #take
PowerSmooth = savgol.(T,fsnew) #Savitsky-Golay power smoothing
derivative = abs.gradient(power_smooth) / gradient(fsnew)
Max_value = max(derivative) #between 0.9 and 0.99 fsnew

#Iterate for multiple elements to generate e.g., ternary:
for i in x_element
    for j in y_element
        Perform above

#Log data
#Perform postprocessing and plotting
```

Complete code examples are available in a report online. Plans to post on GitHub.

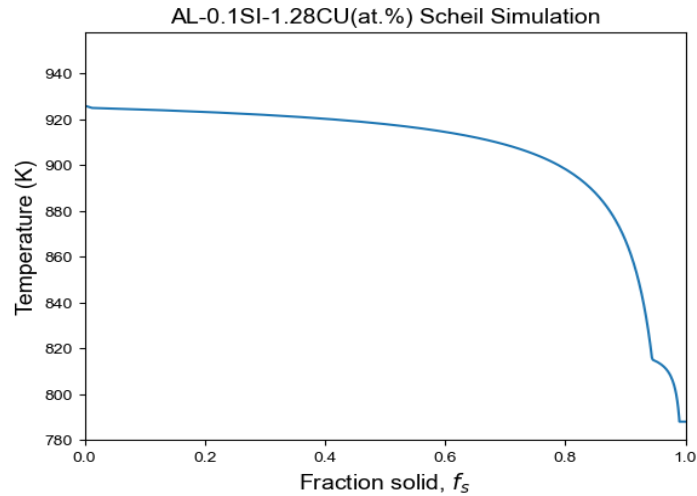
[8] Michael & Sowards (2023) NASA-TM-20230002218.

<https://ntrs.nasa.gov/citations/20230002218>

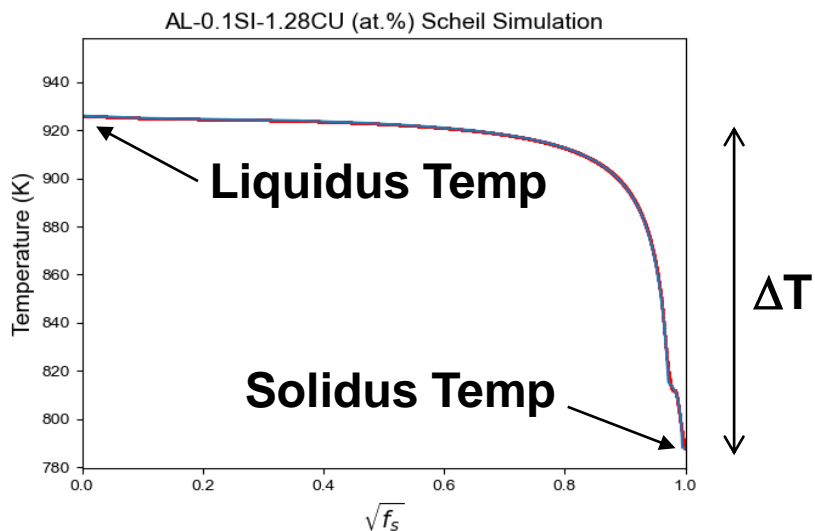
Example: Jupyter Notebook Output and CSI Calculation



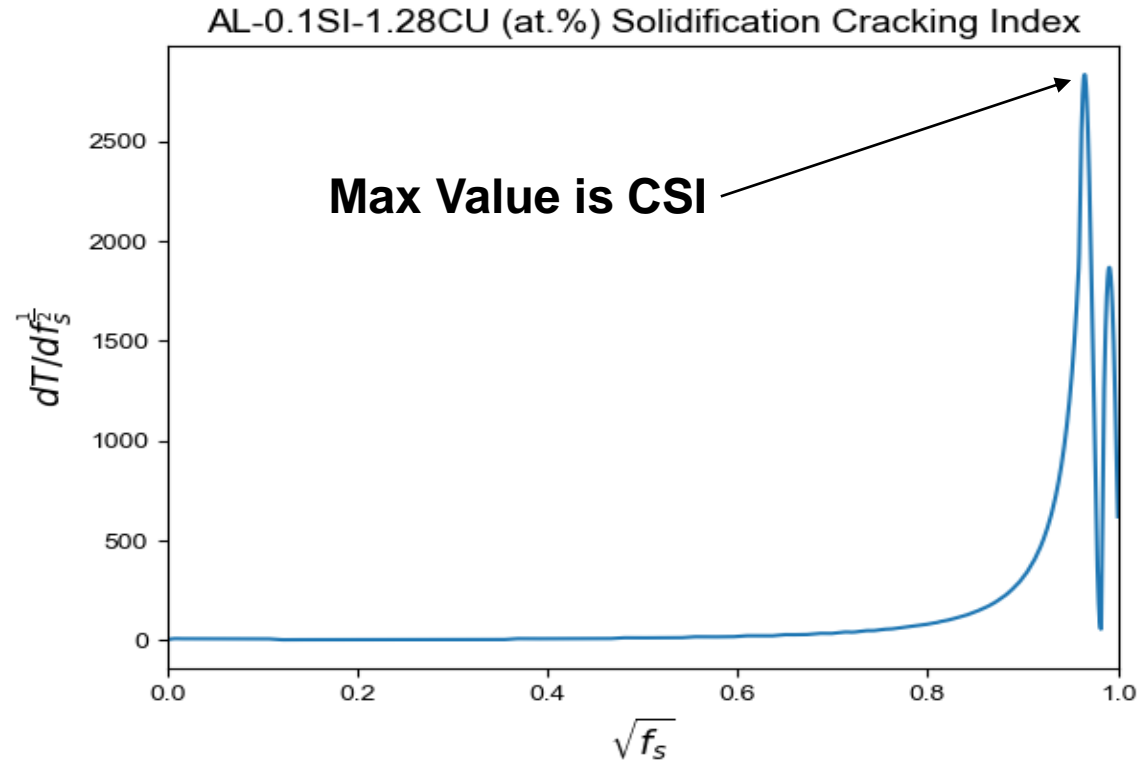
1. Compute Scheil Solidification Path



2. Perform Best Fit to $f_s^{1/2}$ -T Plot (optional)



3. Compute Derivative of $f_s^{1/2}$ -T Best Fit Line

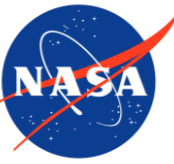


4. Find Max CSI and Log Results

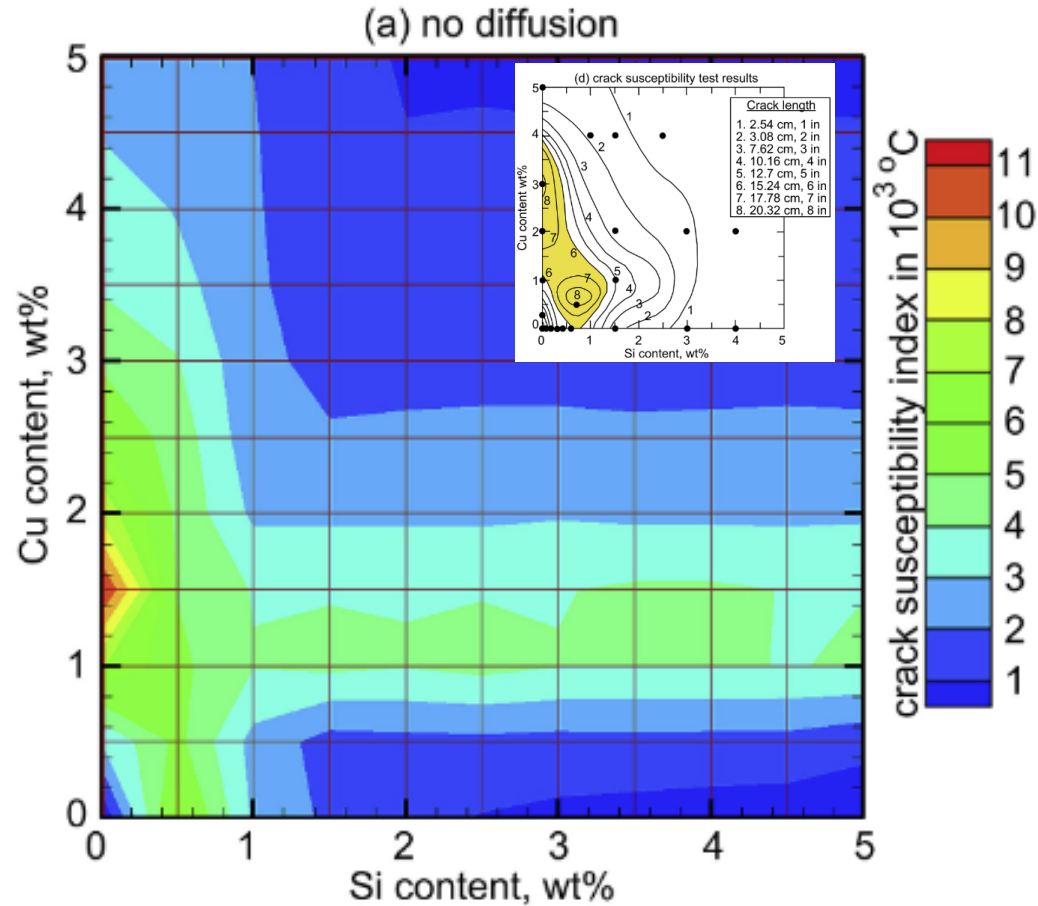
Jupyter Notebook Output

```
Run # = 18 Total Run time = 166.1 seconds
Composition = {W_SI: 0.001, W_CU: 0.029724137931034483}
Max CSI = 2832.9 K, Max CSI with Filter = 2832.9 K, Solidus Temperature = 788.0 K
```

Algorithm Verification in Al-Si-Cu Ternary with Open-source Software

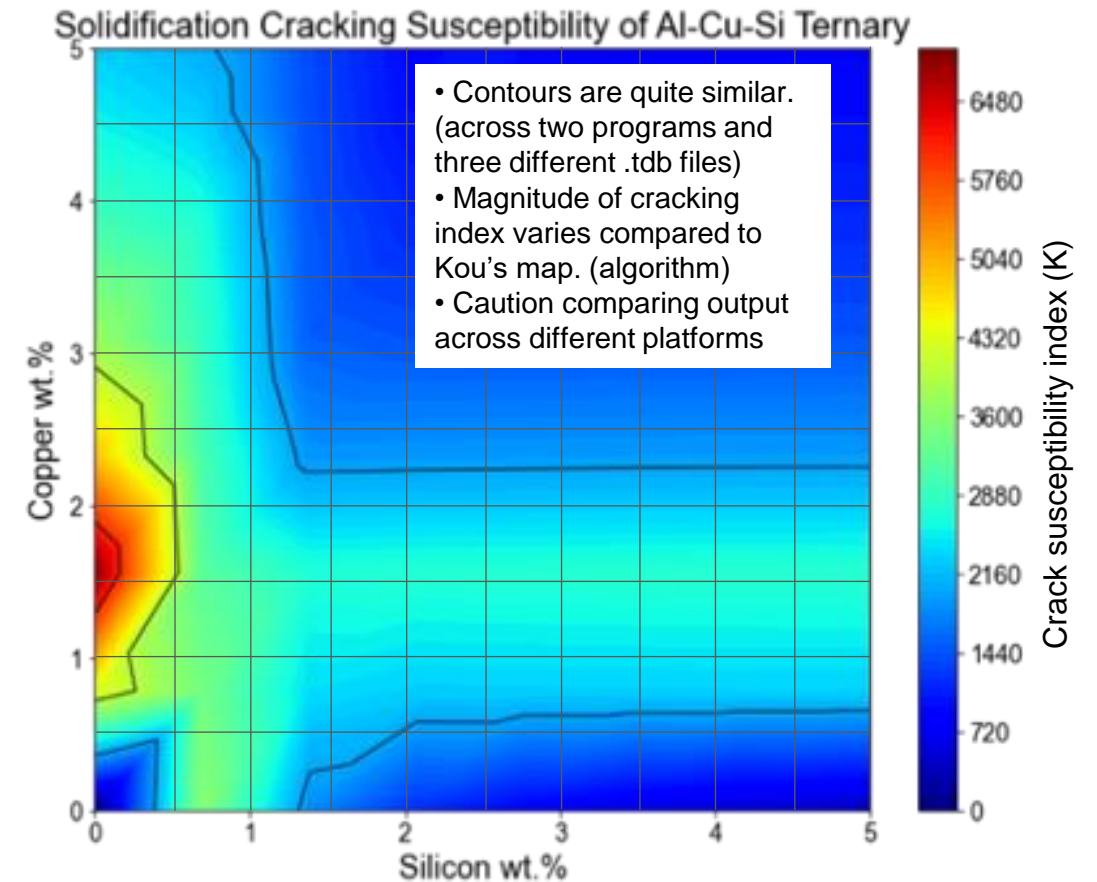


Liu and Kou's Cracking Index Map [9] produced with Pandat + Pan aluminum database. Solidification with no diffusion (Scheil).



[9] Liu and Kou. *Acta Mat.* 125, 15 (2017): 513-523.
<https://doi.org/10.1016/j.actamat.2016.12.028>

Kou's Cracking Index Map produced with open-source pycalphad + COST507.tdb Solidification with no diffusion (Scheil).



Two open-source TDB were tested producing similar map results:

[10] Ansara et al. (1998) COST 507.

[11] Hallstedt et al. *Calphad* 53 (2016): 25-38.

1. Algorithm Verification with Refractory Alloy Vareststraint Data



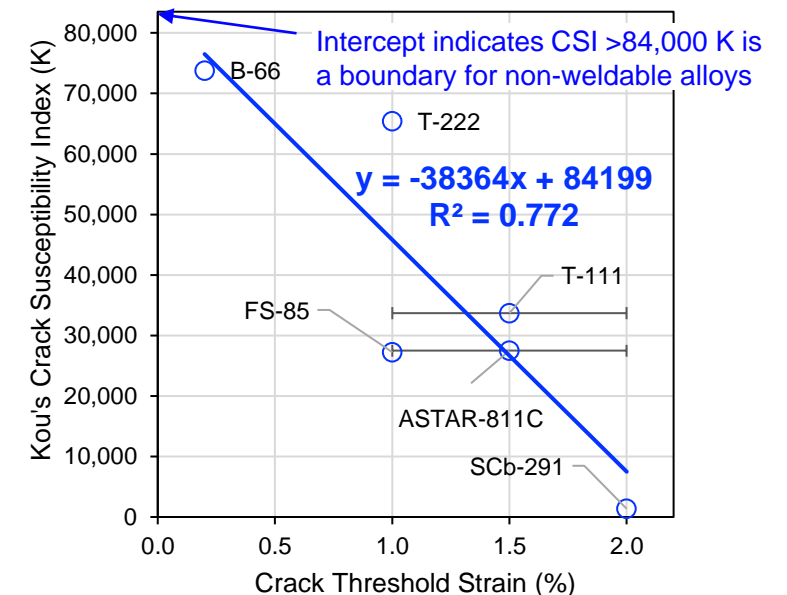
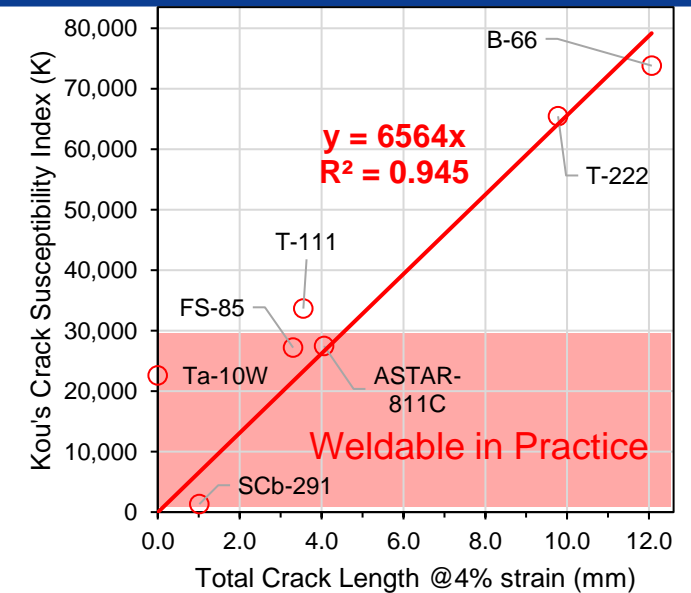
- Lessman and Gold [12] published refractory metal Vareststraint testing of seven refractory alloys subject to GTA welding in inert vacuum.

Refractory Alloy Compositions:

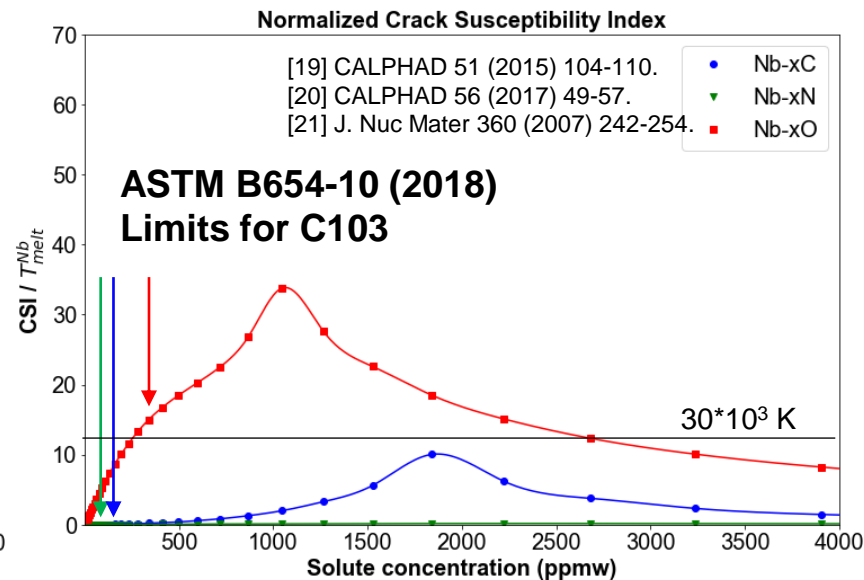
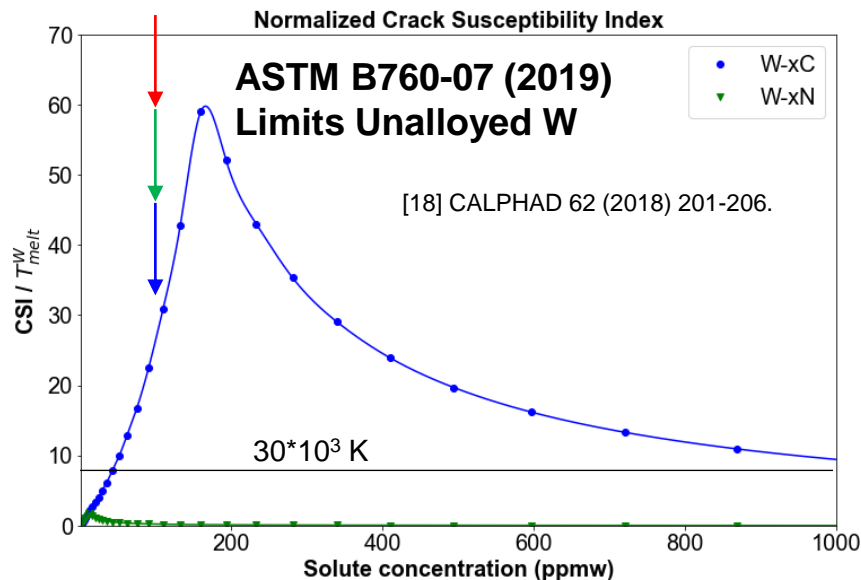
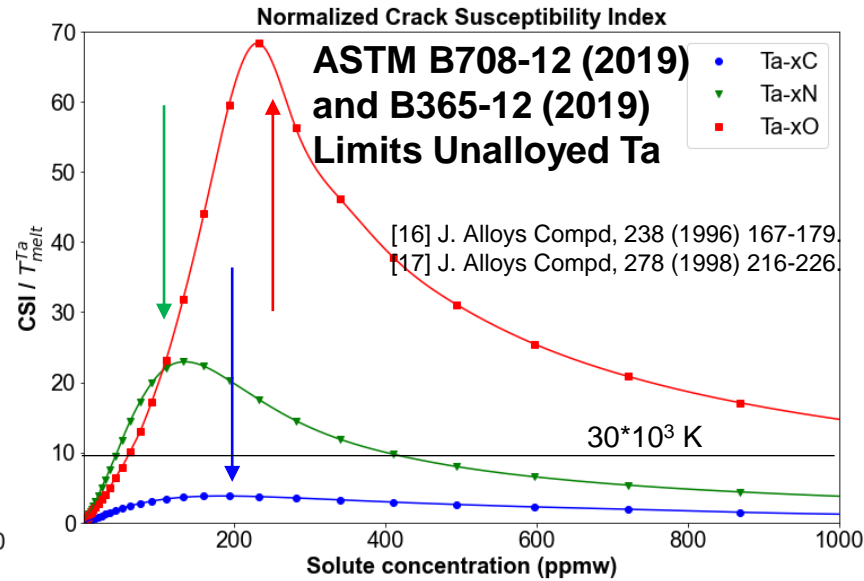
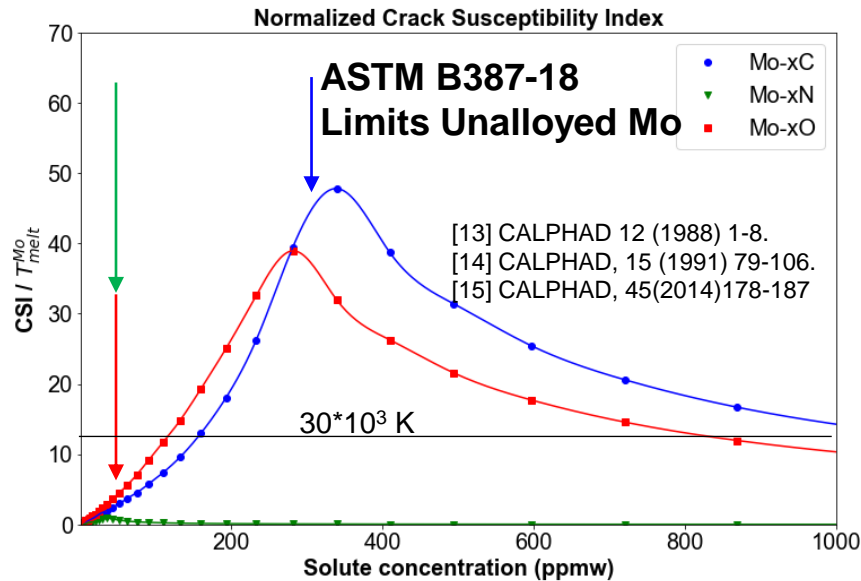
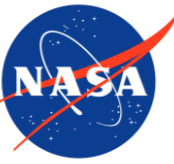
Alloy	Nominal Composition	Ta	Nb	W	Hf	Mo	Re	V	Zr	C ppm	O ppm	N ppm	C ppm	O ppm	N ppm
T-111	Ta-8W-2Hf	balance	-	8.2	2.0	-	-	-	-	40	80	12	33	40	12
ASTAR-811C	Ta-8W-1Re-0.7Hf-0.025C	balance	-	8.1	0.9	-	1.4	-	-	300	70	10	210	5	5
FS-85	Nb-27Ta-10W-1Zr	28.1	balance	10.6	-	-	-	-	0.94	20	90	60	32	53	47
T-222	Ta-9.6W-2.4Hf-0.01C	balance	-	9.2	2.55	-	-	-	-	115	50	20	119	17	11
B-66	Nb-5Mo-5V-1Zr	-	balance	-	-	5.17	-	4.89	1	95	110	63	37	120	70
Ta-10W	Ta-10W	balance	-	9.9	-	-	-	-	-	50	40	20	5	10	10
SCb-291	Nb-10W-10Ta	9.83	balance	10.0	-	-	-	-	-	20	110	40	22	101	20

- Themocalc (TCHEA6.tdb) was used to calculate Scheil solidification paths of those seven alloys and subsequent CSI. Oxygen was **not** in the database.
- CSI shows good correlation to Ta- and Nb-based refractory alloy Vareststraint test data.
- Refractory alloys with $CSI < 30 \cdot 10^3$ K are weldable in practice.
- Refractory alloys with $CSI > 80 \cdot 10^3$ K would likely crack at all augmented strains.

[12] Lessman and Gold. *Welding J.* (1971): 1s – 8s.



2. Crack Susceptibility Index in Binary Refractory Mixtures



- Solidification cracking is strongly influenced by interstitial elements in practice
- CSI of C, N, O interstitial alloys
- CSI is normalized by T_{melt} for scaling
- Effect of interstitials as follows:

Effect of **Carbon** on CSI:
W > Mo > Ta > Nb

Effect of **Nitrogen** on CSI:
Ta > Mo > W > Nb

Effect of **Oxygen** on CSI (No W-O):
Ta > Mo > Nb

- Comparison to ASTM chemistry limits:

Mo: C Limit is near peak CSI

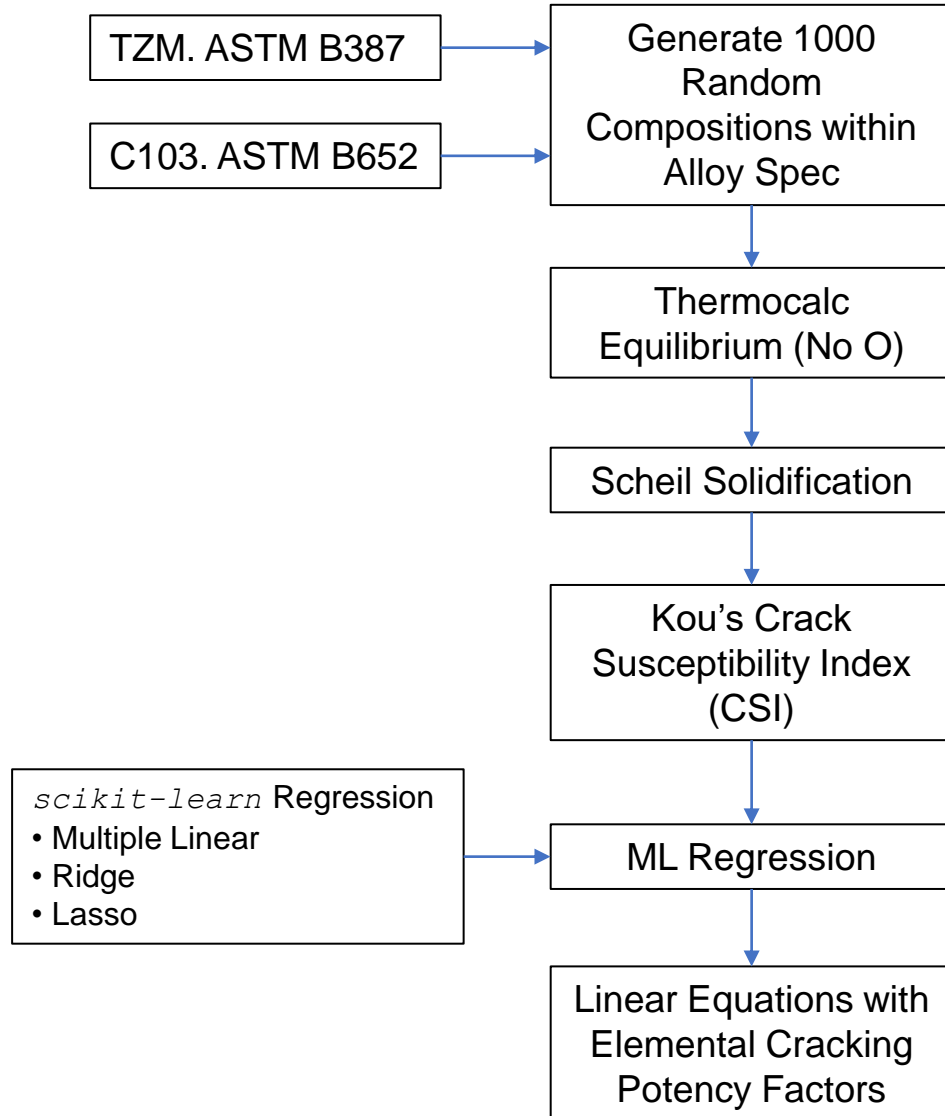
Ta: C, N, O limits are near peak CSI

W: C limit may be concern, O is unknown

Nb: O limit may be concern

- Additive powder recycling pickup of C and O especially will promote cracking.

3. Extending the Model: Chemistry-dependent Cracking



Many equations have been developed to relate solidification cracking to alloying elements through multiple linear regression [22].

[22] Matsuda (1990). Proc 1st US-Japan Symposium on Advances in Welding Metallurgy. 19-36.

Element	TZM Ingot - ASTM B387 (wt.%)
C	0.01 – 0.04
O*	0.003 max
N	0.002 max
Fe	0.01 max
Ti	0.4 – 0.55
Si	0.01 max
Ni	0.002 max
Zr	0.06 – 0.12
Mo	balance

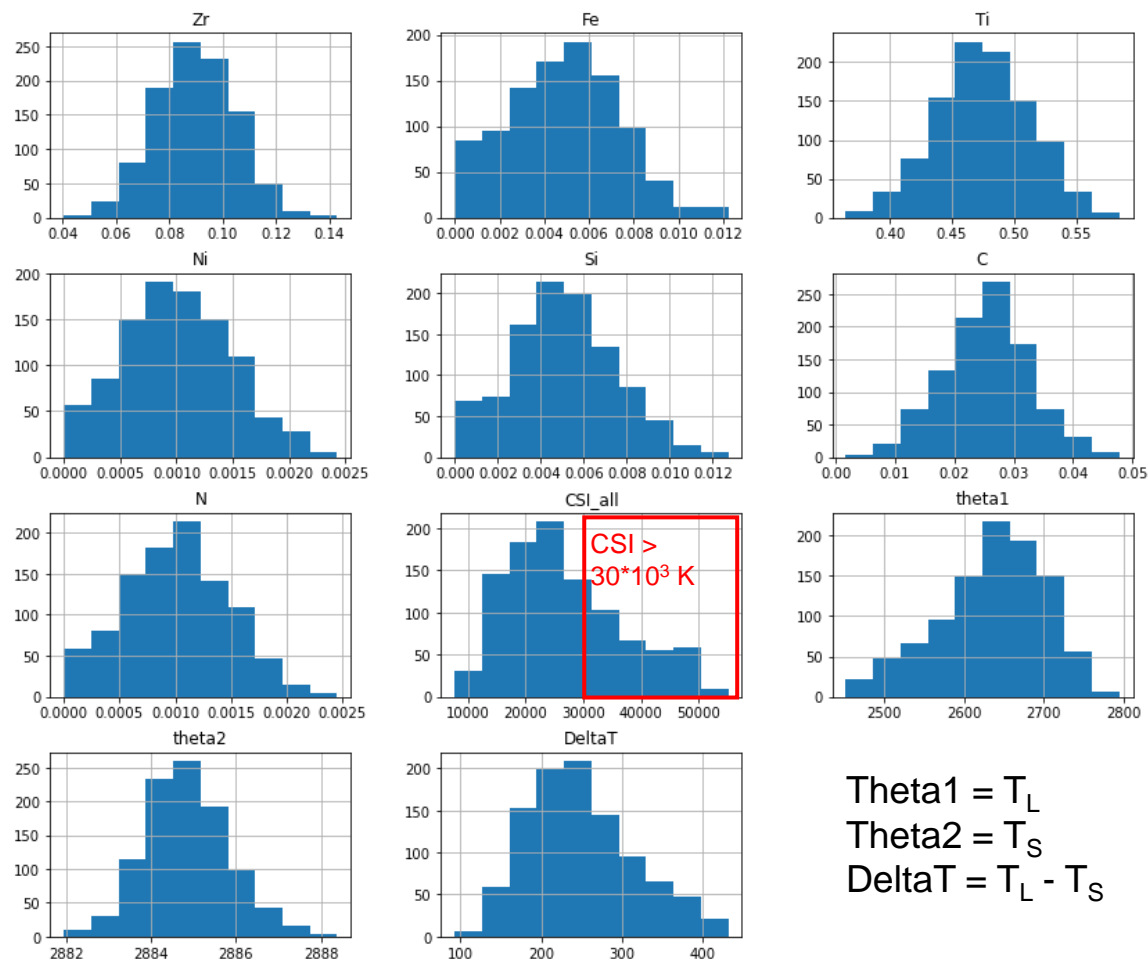
*O in powder metallurgy alloy is 0.05 max

Element	C103 Ingot - ASTM B652 (wt.%)
C	0.015 max
O	0.025 max
N	0.010 max
H	0.0015 max
Hf	9 – 11
Ti	0.7 – 1.3
Zr	0.700 max
W	0.500 max
Ta	0.500 max
Nb	balance

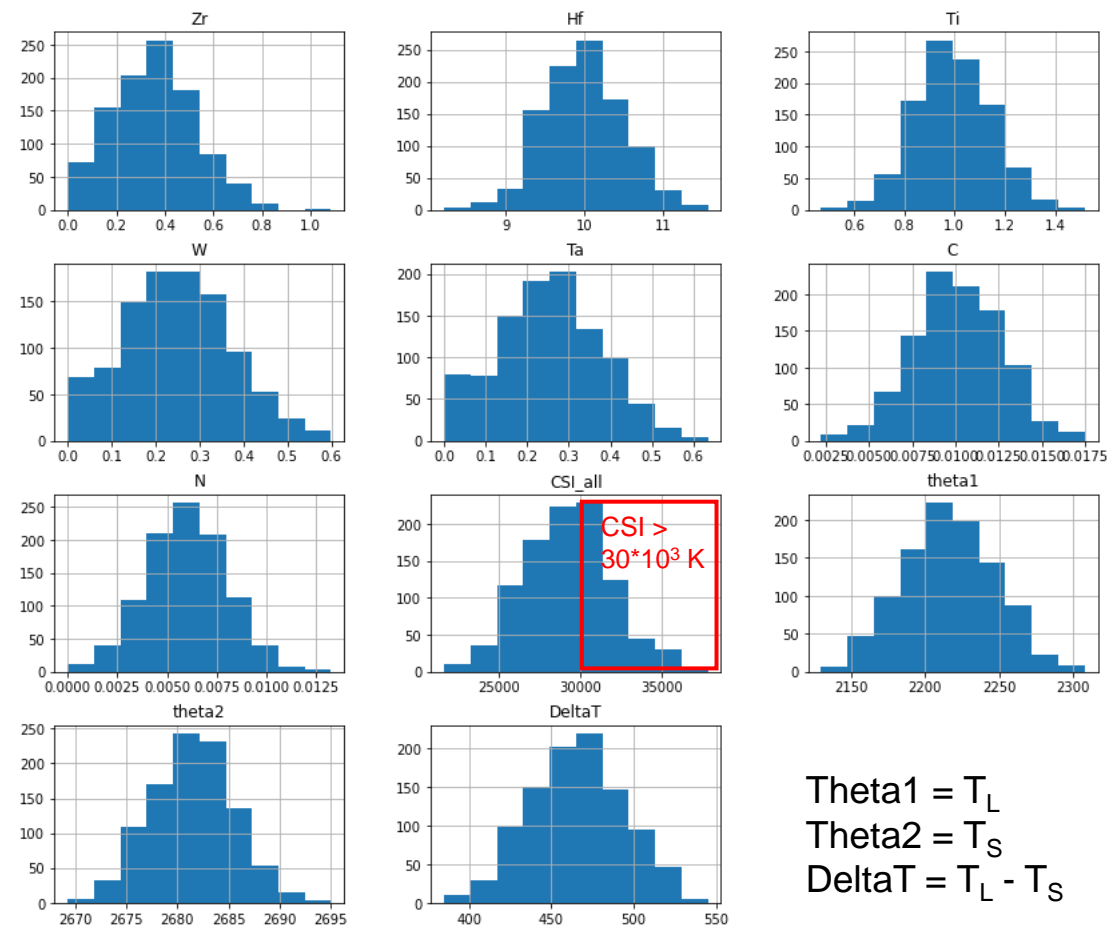
Input and Results



TZM Distribution $N = 1,000$

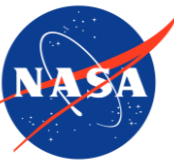


C103 Distribution $N = 1,000$

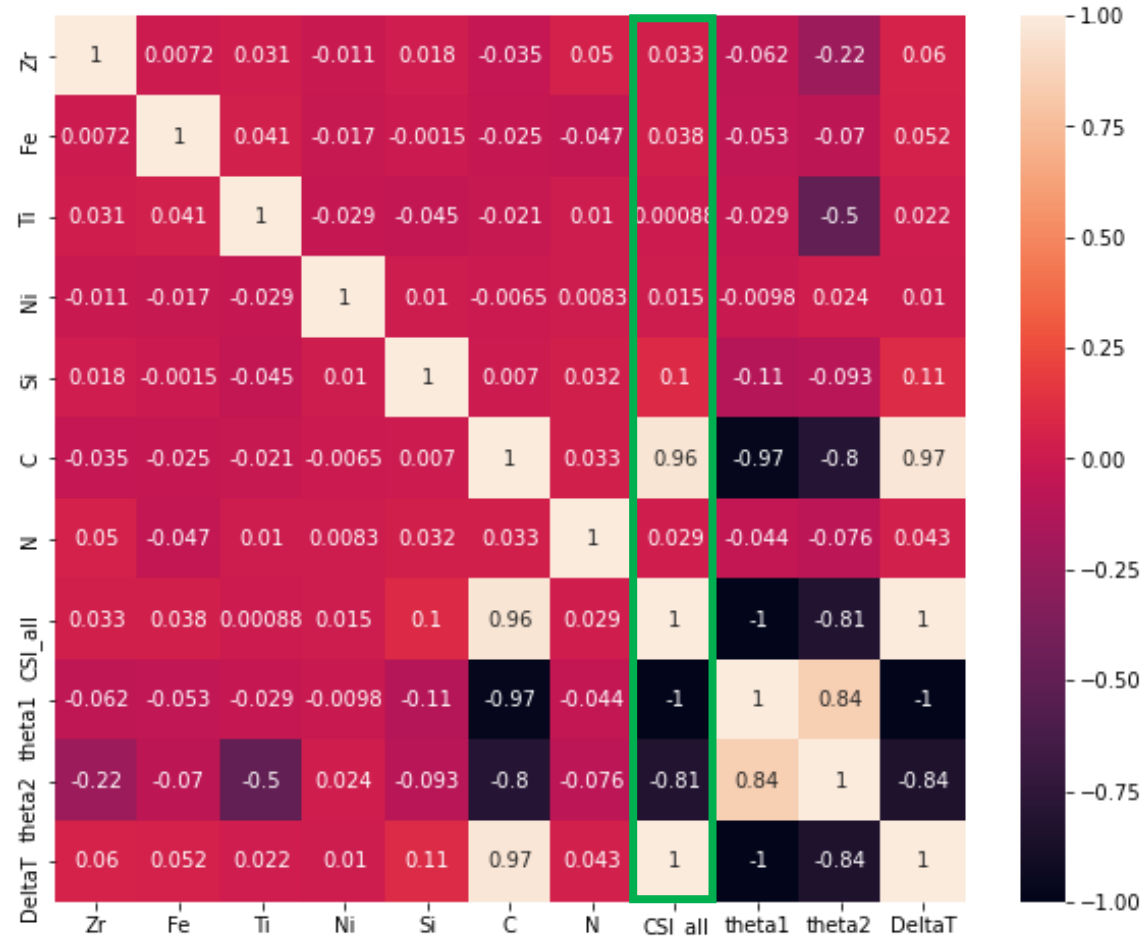


Negative values of composition are assumed zero. Data are normally distributed. A large portion of compositions produce a $CSI > 30 \times 10^3 \text{ K}$.

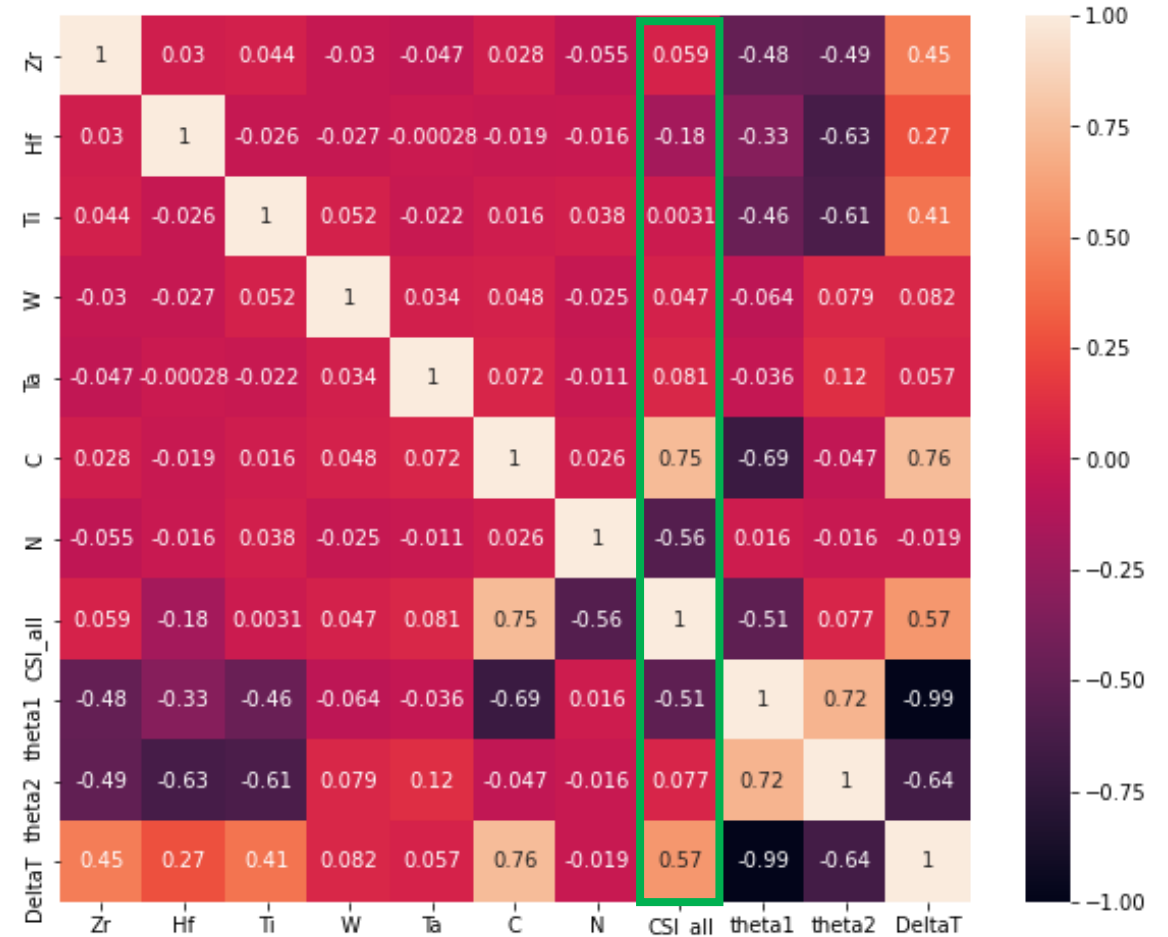
Interaction Matrix to Determine Correlations



TZM



C103



Multicollinearity (several independent variables are correlated) is not observed.

Regression Results and Best Fit Model



TZM

$CSI = \beta_0 + \beta_{Zr}X_{Zr} + \beta_{Fe}X_{Fe} + \beta_{Ti}X_{Ti} + \beta_{Ni}X_{Ni} + \beta_{Si}X_{Si} + \beta_CX_C + \beta_NX_N$
where X expressed in [wt.%]

Model	Linear	Ridge	Lasso
a	--	0.0001	0.0001
R ²	0.94774	0.94768	0.94774
b ₀	-17066.3	-16805.1	-17065.8
b _{Zr}	43924.2	43753.9	43922.9
b _{Fe}	246464	242320	246450
b _{Ti}	5732.26	5657.58	5731.96
b _{Ni}	465135	325775	464705
b _{Si}	385805	379454	385787
b _C	1334869	1332153	1334866
b _N	-156242	-105938	-155788

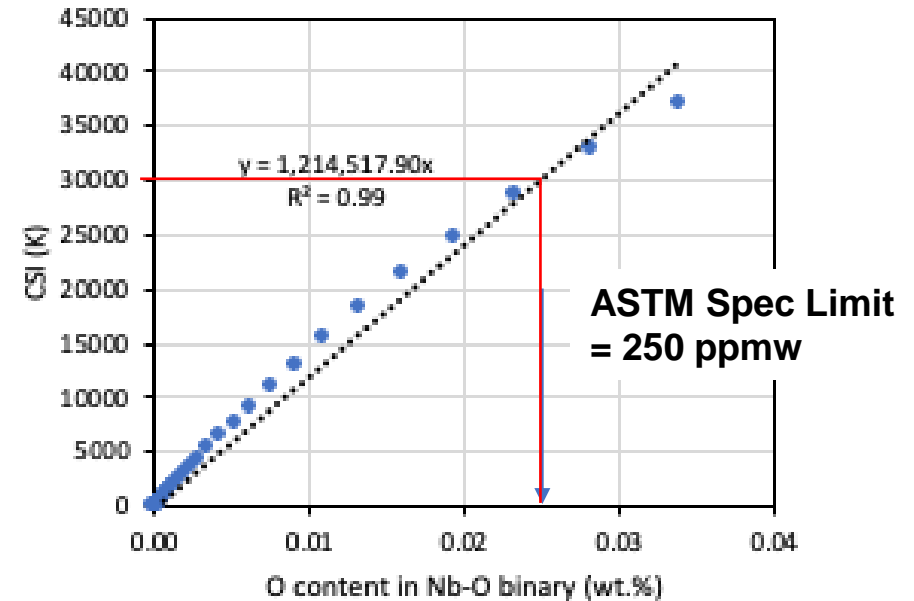
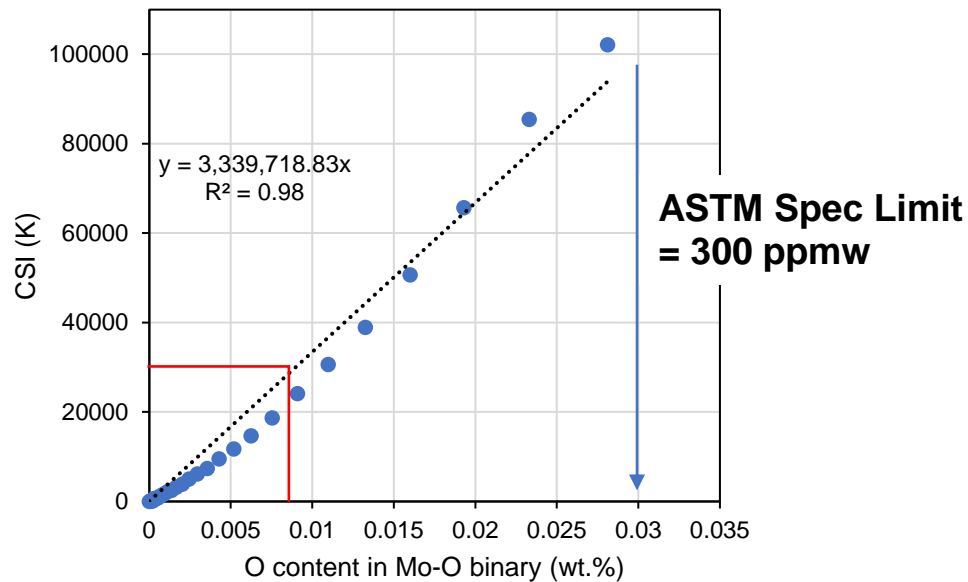
C103

$CSI = \beta_0 + \beta_{Zr}X_{Zr} + \beta_{Hf}X_{Hf} + \beta_{Ti}X_{Ti} + \beta_WX_W + \beta_{Ta}X_{Ta} + \beta_CX_C + \beta_NX_N$
where X expressed in [wt.%]

Model	Linear	Ridge	Lasso
a	--	0.0001	0.0001
R ²	0.92197	0.92163	0.92197
b ₀	34966.3	34977.1	34966.3
b _{Zr}	177.93	197.265	177.952
b _{Hf}	-960.084	-960.276	-960.083
b _{Ti}	160.762	153.071	160.747
b _W	-208.024	-186.839	-207.989
b _{Ta}	449.447	472.192	449.47
b _C	809597	796558	809580
b _N	-771158	-752294	-771133

All models produce excellent fits to data. As the alpha value $\rightarrow 0$, for Ridge and Lasso the coefficients approached ordinary Least Squares Regression model. Linear multiple regression is selected for further discussion.

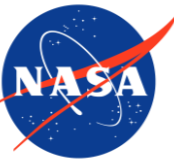
Discussion. Effect of Oxygen



Oxygen was not considered in the complex alloys due to lack of available thermodynamic data for higher order mixtures. The Mo-O and Nb-O binary systems above show that oxygen drastically increases CSI.

We develop a weight factor based on linear interpolation above revealing a weight factor of 3.34×10^6 K/[O] and 1.21×10^6 K/[O], for Oxygen in TZM and C103, respectively.

Simplified Linear Models of Elemental Potency on Cracking



Steps:	TZM	C103
1. View Raw CSI coefficients. X_i in [wt. %]	$\text{CSI} = -17,066 + 43,924 X_{\text{Zr}} + 246,464 X_{\text{Fe}} + 5,732 X_{\text{Ti}} + 465,135 X_{\text{Ni}} + 385,805 X_{\text{Si}} + 1,334,869 X_{\text{C}} - 156,242 X_{\text{N}}$	$\text{CSI} = 34,966 + 178 X_{\text{Zr}} - 960 X_{\text{Hf}} + 161 X_{\text{Ti}} - 208 X_{\text{W}} + 449 X_{\text{Ta}} + 809,597 X_{\text{C}} - 771,158 X_{\text{N}}$
2. Modify with estimated oxygen term based on binary calculation. X_i in [wt. %]	$\text{CSI} = -17,066 + 43,924 X_{\text{Zr}} + 246,464 X_{\text{Fe}} + 5,732 X_{\text{Ti}} + 465,135 X_{\text{Ni}} + 385,805 X_{\text{Si}} + 1,334,869 X_{\text{C}} + 3,339,718 X_{\text{O}} - 156,242 X_{\text{N}}$	$\text{CSI} = 34,966 + 178 X_{\text{Zr}} - 960 X_{\text{Hf}} + 161 X_{\text{Ti}} - 208 X_{\text{W}} + 449 X_{\text{Ta}} + 809,597 X_{\text{C}} + 1,214,518 X_{\text{O}} - 771,158 X_{\text{N}}$
3. Normalize coefficients by max coefficient (Oxygen in both cases) revealing model with relative potency the alloying elements have on hot cracking susceptibility (HCS)	$\text{HCS} = 0.013 \cdot \text{Zr} + 0.074 \cdot \text{Fe} + 0.002 \cdot \text{Ti} + 0.139 \cdot \text{Ni} + 0.116 \cdot \text{Si} + 0.4 \cdot \text{C} + \text{O} - 0.047 \cdot \text{N}$	$\text{HCS} = 0.667 \cdot \text{C} + \text{O} - 0.001 \cdot \text{Hf} - 0.635 \cdot \text{N}$

- Oxygen and Carbon strongly promote solidification crack susceptibility.
- Nitrogen apparently decreases crack susceptibility especially in C103.
- Fe, Ni, Si promote crack susceptibility in TZM, as do Zr and Ti to lesser extent.

For nominal TZM (Mo-0.475Ti-0.09Zr-0.025C), CSI exceeds $30 \cdot 10^3$ K at 23 ppmwt O!
 For nominal composition of C103 (Nb-10Hf-1Ti), CSI exceeds $30 \cdot 10^3$ K at 37 ppmwt O!

Initial validation with Thermo-Calc[®] databases (TZM)

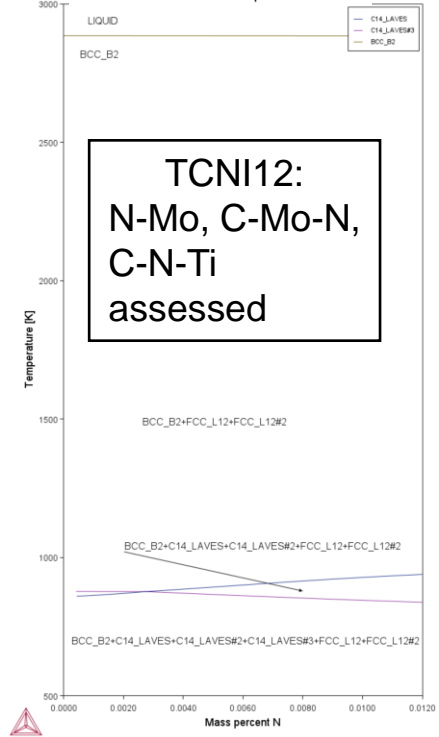
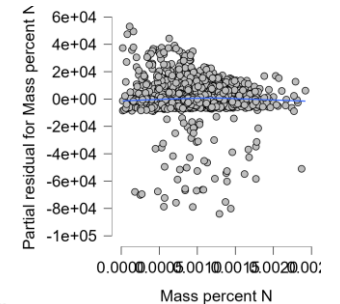
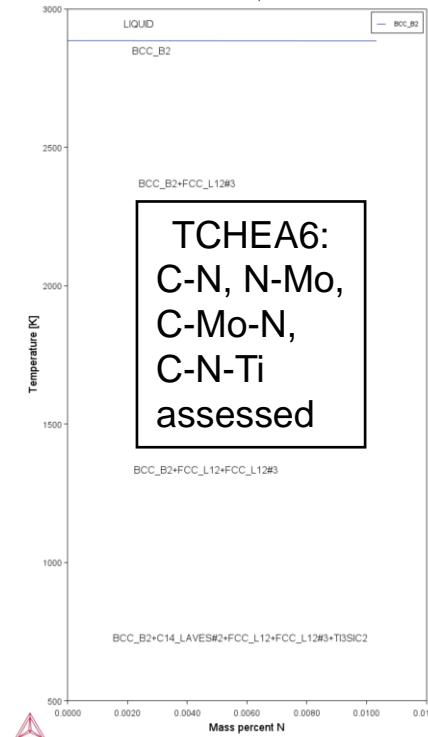
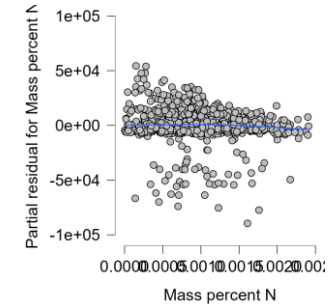


TZM	Database		
	TCHEA6	TCNI12	COST507
β_{O}	-13640	-15991	-17066
β_{Zr}	125768	71909	43924
β_{Fe}	571421	513975	246464
β_{Ti}	-6084	20005	5732
β_{Ni}	1128000	-5	465135
β_{Si}	870811	531083	385805
β_{C}	2130000	2049000	1334869
β_{N}	-4162300	-27	-156242
β_{O}	1400000	1400000	3339719

TZM	Database		
	TCHEA6	TCNI12	COST507
β_{O}			
β_{Zr}	0.089834	0.051364	0.013152
β_{Fe}	0.408158	0.367125	0.073798
β_{Ti}	-0.00435	0.014289	0.001716
β_{Ni}	0.805714		0.139274
β_{Si}	0.622008	0.379345	0.11552
β_{C}	1.521429	1.463571	0.399695
β_{N}	-2.97307		-0.04678
β_{O}	1	1	1

*Mo-O for TCHEA6 & TCNI12 calculations taken from TCNI12

Database	Zr	Fe	Ti	Ni	Si	C	O	N
TCNI12	0.051	0.367	0.014		0.379	1.46	1	
COST507	0.013	0.074	0.002	0.139	0.116	0.4	1	-0.047



Some agreement w.r.t HCS potency from C, N, and O;
other contributions of Fe, Ni, and Si also notable; TCNI12 closer to COST507

Initial validation with Thermo-Calc[®] databases (C103)

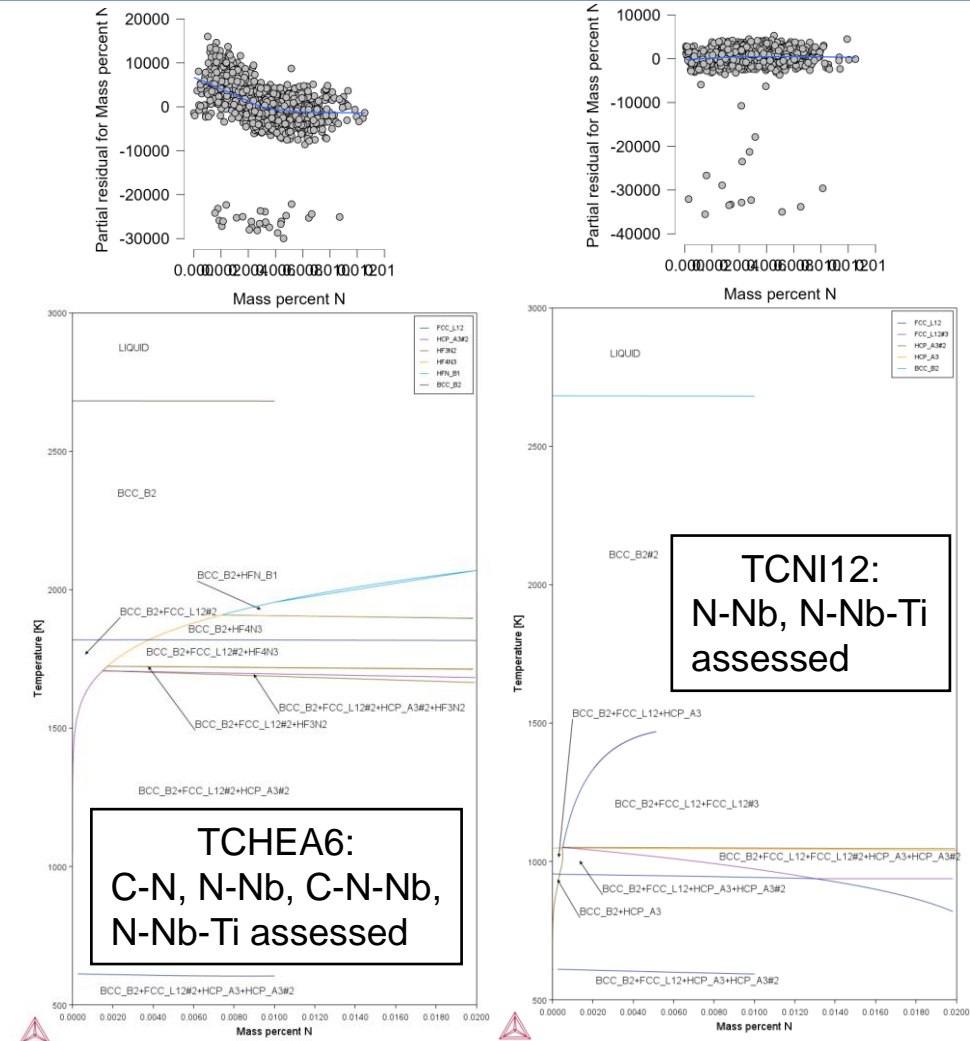


C103	Database		
	TCHEA6	TCNI12	COST507
β_0	71798.16	36715.06	34966.3
β_{Zr}	-433.77	5677.079	177.93
β_{Hf}	-3303.51	-1474.5	-960.084
β_{Ti}	-5747.47	391.995	160.762
β_W	-204.478	1285.546	-208.024
β_{Ta}	89.486	-1127.69	449.447
β_C	784781.5	1517000	809597
β_N	-775735	94119.62	-771158
β_O	1225000	1225000	1214518

C103	Database		
	TCHEA6	TCNI12	COST507
β_0			
β_{Zr}		0.004634	
β_{Hf}	-0.0027	-0.0012	-0.00079
β_{Ti}	-0.00469		
β_W		0.001049	
β_{Ta}		-0.00092	
β_C	0.640638	1.238367	0.666599
β_N	-0.63325	0.076832	-0.63495
β_O	1	1	1

*Nb-O for TCHEA6 & TCNI12 calculations taken from TCNI12

Database	C	O	Hf	Ti	N
TCHEA6	0.641	1	-0.003	-0.005	-0.633
COST507	0.667	1	-0.001		-0.635



Agreement w.r.t HCS potency from C and O; N potency agreement for TCHEA6 & COST507, more confident in TCHEA6 than TCNI12 as C-N-Nb assessed

1. A numerical approach was developed to calculate Kou's Solidification Crack Susceptibility Index (CSI) using open-source Python code with both an open-source and a commercial CALPHAD equilibrium solver.
 - The method was verified against previous calculations and aluminum alloy solidification cracking data.
2. The numerical approach was applied to refractory metals, which are inherently difficult to study from a weldability testing standpoint since welding is often done in vacuum.
 - Calculated CSI showed strong empirical correlation to vacuum Vareststraint testing of Ta- and Nb-alloys.
 - Correlations indicate that refractory alloys with $CSI < 30 \times 10^3 \text{ K}$ are weldable in practice.
3. Calculation of CSI for refractory-interstitial (O,C,N) binary systems revealed ASTM chemistry specs are not ideal for optimal weldability and AM printability.
4. This work revealed the effect of compositional variations on a series of refractory metals and showed the framework defined here will be useful in:
 - The development of new alloys that have improved weldability and AM printability
 - Placing compositional limits on existing alloys
 - Consideration of manufacturing process controls such as powder reuse during 3D printing