

Overview of Electrified Aircraft Propulsion at NASA

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My NASA story ...



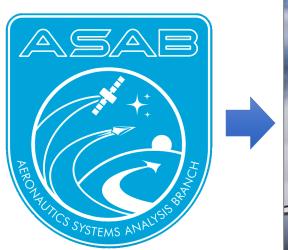














Advanced Air Transport Technology (AATT) Project

NASA's vision for advanced fixed wing transport aircraft is revolutionary energy efficiency and environmental compatibility. The overarching goal of the AATT Project is to explore and develop technologies and concepts to enable this vision.





The Sustainability Challenge

Revolutionizing
Aviation through EAP

Systems-Level EAP
Assessment





The Aviation Carbon Reduction Challenge

- By 2050, an estimated 10 billion passengers will fly each year a distance of 22 trillion revenue passenger kilometers.
- With today's fleet and operational efficiency, this activity would require over 620 megatonnes (Mt) of fuel and generate close to 2000 Mt of CO₂.
- Imagine enabling the same level of demand while reducing net CO₂ emissions to zero by 2050.



Meeting the challenge is the opportunity for the United States to lead the world in innovation and reductions in CO₂ aviation emissions, and to maintain economic competitiveness in a critical export industry (\$6 trillion-plus market over the next 20 years).

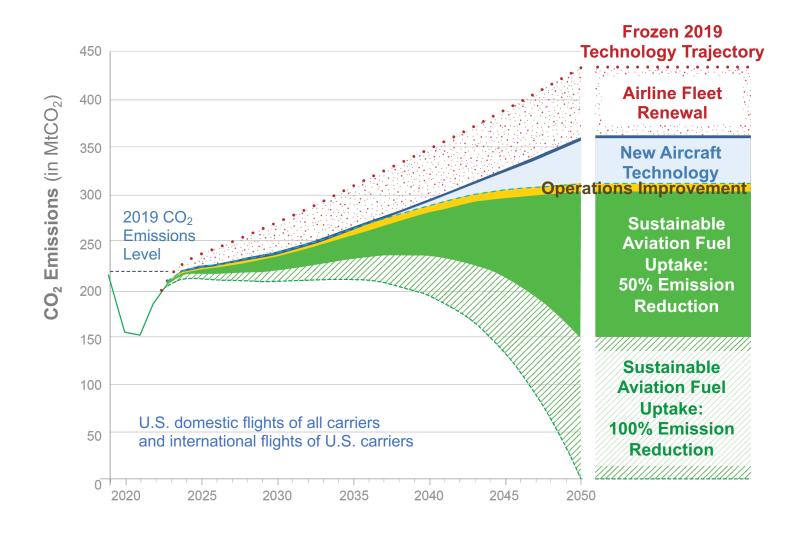
U.S. Aviation Climate Action Plan

Global Context for Sustainable Aviation

U.S. aviation goal is to achieve net-zero greenhouse gas emissions by 2050.

U.S. Aviation Climate Action Plan is aligned with

- U.S. economy-wide goal
- **International Civil Aviation Organization**
- Air Transport Action Group



US Government Sustainability Strategies



FAA

Net-Zero GHG emissions from aviation by 2050

Assumption of SAF-dominated aviation fuels landscape

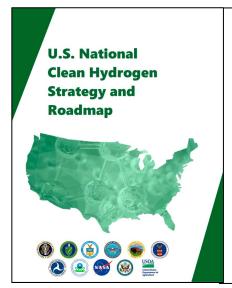


DOE/DOT/FAA/USDA/EPA

References US Aviation Climate Action Plan goals

Establishes national goals for SAF production of:

- 3B gal/year by 2030
- 35B gal/year by 2050

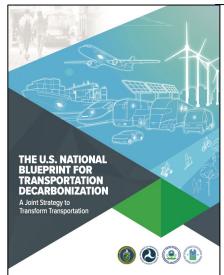


DOC/DOD/DOE/DOI/DOL/ DOT/EPA/NASA/USDA+

National clean H2 production and infrastructure strategy

National H2 production goals:

- 10MMT/year by 2030
- 20MMT/year by 2040
- 35MMT/year by 2050



DOE/DOT/EPA/DOH

Transportation sector wide decarbonization strategy

Three strategic pillars:

- 1. Community redesign
- 2. Improved efficiency
- 3. Transition to clean fuels

Aviation Pillars for a Sustainable Future

Global Aviation GOAL: net-zero carbon emissions by 2050





NASA = Supporting Role



NASA = Primary Role

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NASA Sustainable Aviation Strategy

2008-2013

2014-2019

2020-2025

2025-2030

2030+

Subsonic Concept/Technology Studies **Electrified Aircraft Propulsion** Transonic Truss-Braced Wing **Blended Wing Body**

> Environmentally Responsible Aviation **Project**

Flight Demonstrator Studies

Advanced Composites Project

SUSTAINABLE FLIGHT NATIONAL PARTNERSHIP

Sustainable Flight National Partnership to mature and integrate key technologies for next-generation subsonic airliners (2030s)

TODAY

ACCELERATING TOWARD NET-ZERO CARBON

Cast a wide net for zero-emission concepts and technologies

Select and develop promising concepts in partnership with universities, industry

Create a credible mission. architecture, and technologies for beyond next-generation subsonic transports for 2050 horizon

POWERING AVIATION TO NET-ZERO CARBON AND BEYOND

Investment in innovation today paves the way to a net-zero carbon and beyond aviation future.

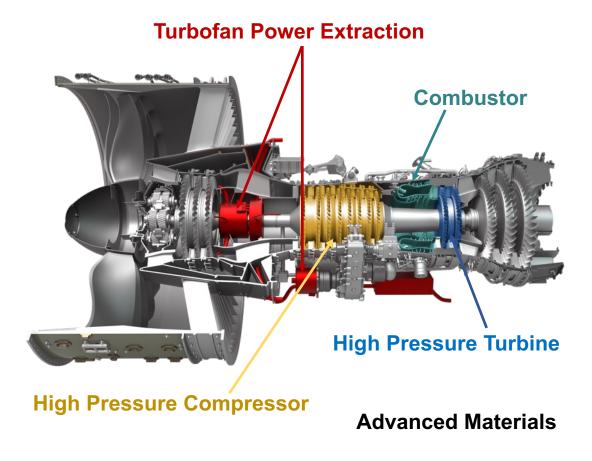
Sustainable Flight National Partnership

Ensure U.S. industry is the first to establish the new "S Curve" for the next 50 years of airliners



Hybrid Thermally Efficient Core

Accelerate development and demonstration of advanced turbine engine technologies



Scope

 Develop and demonstrate in integrated ground tests engine core technologies to Increase thermal efficiency, reduce engine core size and facilitate hybridization

Benefit

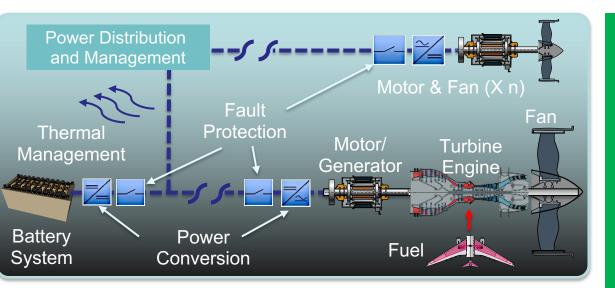
- Achieve 5-10% fuel burn reduction versus 2020 best in class
- Achieve up to 20% power extraction (4 times current state of the art) at altitude to optimize propulsion system performance and enable hybridization

Approach

Partner with industry to mature and demonstrate promising technologies

Focused Technologies for Electrified Aircraft Propulsion

Retire barrier technical and integration risks for megawatt-class electrified aircraft propulsion systems





Scope

- Address critical challenges for electrified aircraft propulsion by maturing and reducing risk for Electrified Aircraft Propulsion (EAP) technology, focused on:
 - Mass and weight reduction
 - Electrical losses
 - Reliability

- EMI, power quality, dynamic stability
- Limits on DC voltage levels
- System design and integration

Benefit

- Accelerate U.S. industry readiness to transition to EAP-based commercial transport aircraft.
- Reduce key risks for a range of future applications and help enable new standards that are needed for EAP-based aircraft certification

Approach

- Conduct technology-focused integrated ground tests
- Partner with industry on testing of electrified propulsion architectures and component technologies
- Leverage prior electric aircraft propulsion advances (TRL ~4)

NASA Electrified Powertrain Flight Demonstration

Electrified Powertrain Flight Demonstration (EPFD) project:

 Partnership with U.S. industry to establish and demonstrate integrated megawatt-class powertrain systems.

EPFD Goals:

- Accelerate US industry technology readiness and competitiveness
- Facilitate new aviation industry S Curve for electrification
- 2030-2035 Entry Into Service: Next generation thin haul, regional and Single-Aisle markets

Regulations and Standards will play a large role:

 NASA is partnering with industry to identify the regulatory and standards gaps that may exist for the highest priority electric technologies and gather data to support future regulations and standards development.



Technology Development and Barrier Challenges

Technical Performance	Technical Performance Description	Target Performance
KPP/TPM-1	Total Power level of the Integrated MW-Class Powertrain System	1.5MW
KPP/TPM-2	Power level of individual electrical components	1MW
KPP/TPM-3	Operating Voltage of the Integrated MW-Class Powertrain System	1000V
KPP/TPM-4	Altitude Capability of the Integrated MW- Class Powertrain System	30,000 ft.
KPP/TPM-5	Specific Power of the Integrated MW-Class Powertrain System	1.25 kW/kg
KPP/TPM-6	End to End loss of the Integrated MW-Class Powertrain System	20%
KPP-7	Mission Fuel Burn/Energy Reduction	4%

Integrated Ground Testing

- Control systems will play a vital role in ensuring the realization of electric powertrain technologies
 - Hardware-In-the-Loop (HIL) testing is a means to bridge the gap between conceptual simulations and full-scale system testing

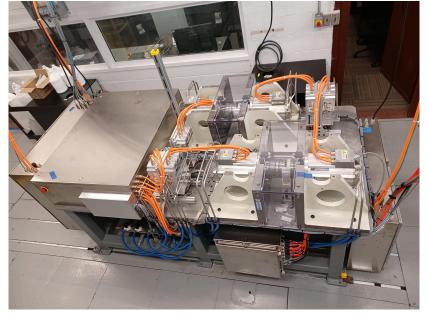


NASA Electric Aircraft Testbed (NEAT) facility

- Test integrated powertrains in MW-class
- Simulated Altitude







Hybrid Performance Emulation Rig (HyPER)

<100 kW facility with energy storage

EPFD Industry Partners

NASA selected two U.S. companies to that will rapidly mature Electrified Aircraft Propulsion technologies through ground and flight demonstrations.

GE Aviation, \$179 million.

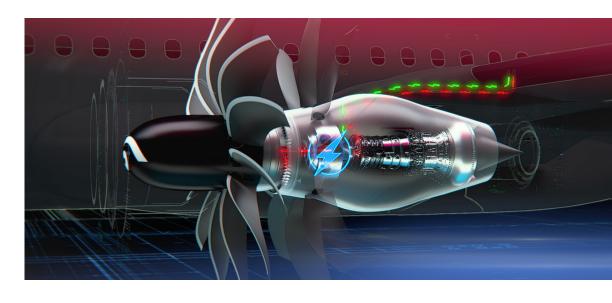




- Partnered with Boeing to modify Saab 340B powered by GE CT7-9B turboprop engines
- Megawatt-class and multi-kilovolt hybridelectric propulsion system tested in simulated altitude conditions at NASA NEAT facility

Future Engine Designs

- Hybrid electric compatible with Sustainable Aviation Fuel and hydrogen
- Demonstrator informs future GE engine product designs with hybrid electric capability



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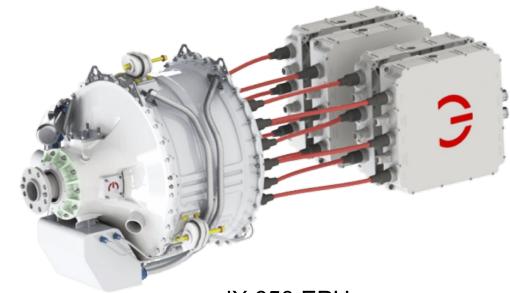
magniX, \$74.3 million.



- Aims to demonstrate electric propulsion technology to power a hybrid De Havilland Canada Dash 7 aircraft, with first flight planned for mid-2020's
- The retrofitted aircraft will be powered by two PT6 engines and two magni650 electric propulsion units (EPUs)

Electric Propulsion Unit

Advances the regional turboprop market



magniX 650 EPU

Long-Term Transport Technology and Innovation

Generational studies to inform future technology investments

2040 - 2050 2008-2019 2020-2029 2030 - 2039 Subsonic Fixed Wing — N+2 Studies, ERA for the 2020s Impact N+3 Advanced Concept Studies SFNP for 2030s Impact Concept Studies and Technology Development Needed for 2040s Impact **Opportunities to Define Future Aviation Systems and Concepts** Advanced Concept Studies for 2040+ EIS **Net-Zero Emissions Concepts** Promising Technologies and Architectures Support Aviation Community with Hydrogen NASA-unique Contributions 100% ALTERNATIVE

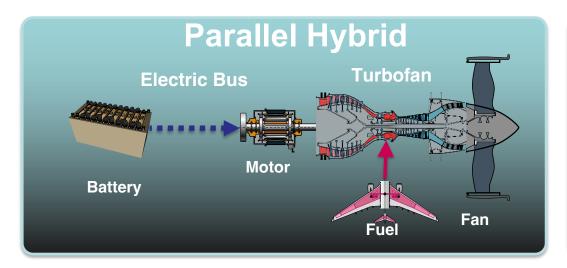
The Sustainability Challenge

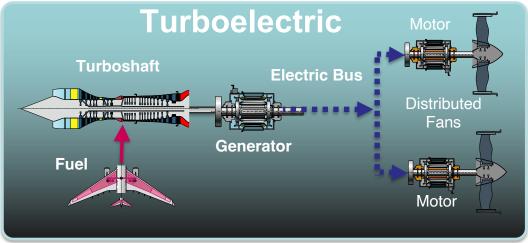
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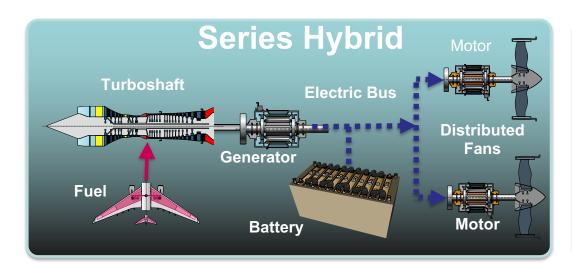
Systems-Level EAP Assessment

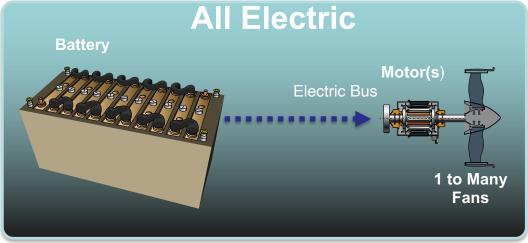


Fundamental EAP Architectures



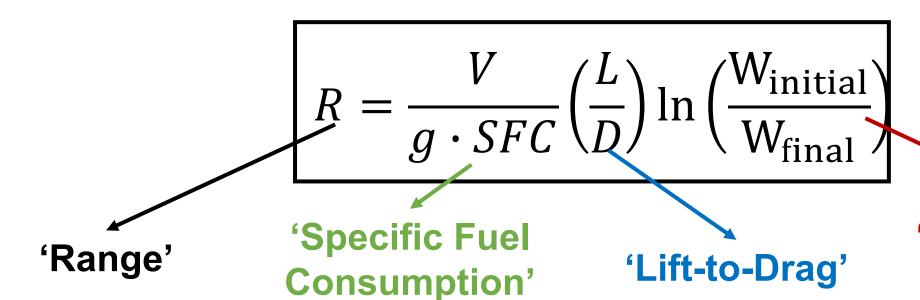






Fundamental Aircraft Performance

The Breguet Range Equation



How far an airplane can fly

How efficiently an airplane turns fuel into thrust

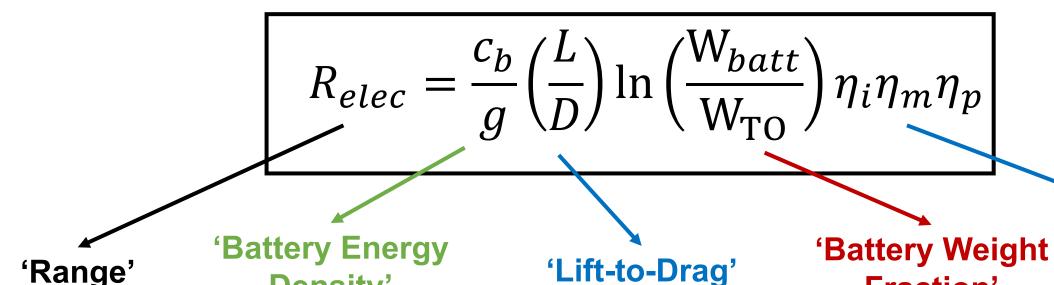
How efficiently an airplane generates lift

'Mission Weight Fraction'

The takeoff weight divided by the landing weight

Understanding EAP Impact on Aircraft Performance

The [Electric] Breguet Range Equation



'Range' How far an

airplane

can fly

How efficiently a battery stores energy

Density'

How efficiently an airplane generates lift

Fraction'

The battery weight divided by the takeoff weight

'Powertrain Efficiency'

Inverter, motor, propulsive efficiencies Electric Machine Power as a Strategic Parametric **Electric Machine on Aircraft 1 MW** Non-cryogenic 100 kW Superconducting 3 MW 10 MW 30 MW 9 Seat 0.5 MW Total Propulsive Power 19 Seat 2 MW Total Propulsive Power **Total Propulsive Power – Total required** vehicle propulsive power 50 Seat Turboprop 12 MW Total Propulsive Power **Electric Machine Power – Power** provided by discrete elements in powertrain (e.g. motors, inverters) 50 Seat Jet 12 MW Total Propulsive Power Bars represent range of electric machines relevant to each class 150 Seat 22 MW Total Propulsive Power of vehicle 300 Seat 60 MW Total Propulsive Power PS-01758-1115

Aeronautics Systems Analysis for EAP

- Making sound investment decisions requires an understanding of both vision vehicle opportunities and challenges
- While EAP offers significant promise for vehicle-level emissions reductions, the realized benefits are sensitive to vehicle-integration, technology, and sizing assumptions
- Numerous studies have been conducted internally and with external partners to elucidate the challenges and implications on the vehicle system
- Results from a few recent studies will be summarized here, emphasizing the important of parametrics, metrics of interest, and assessment of cost and benefits

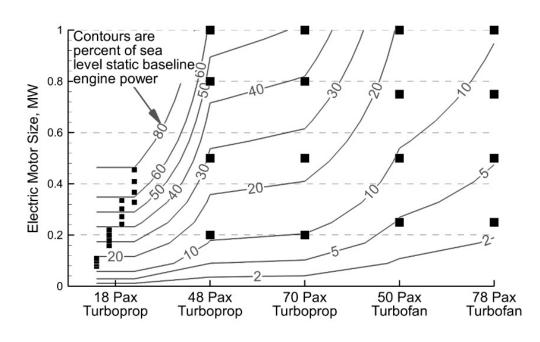
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Parallel Hybrid Design Space Exploration Study (Marien et al.)

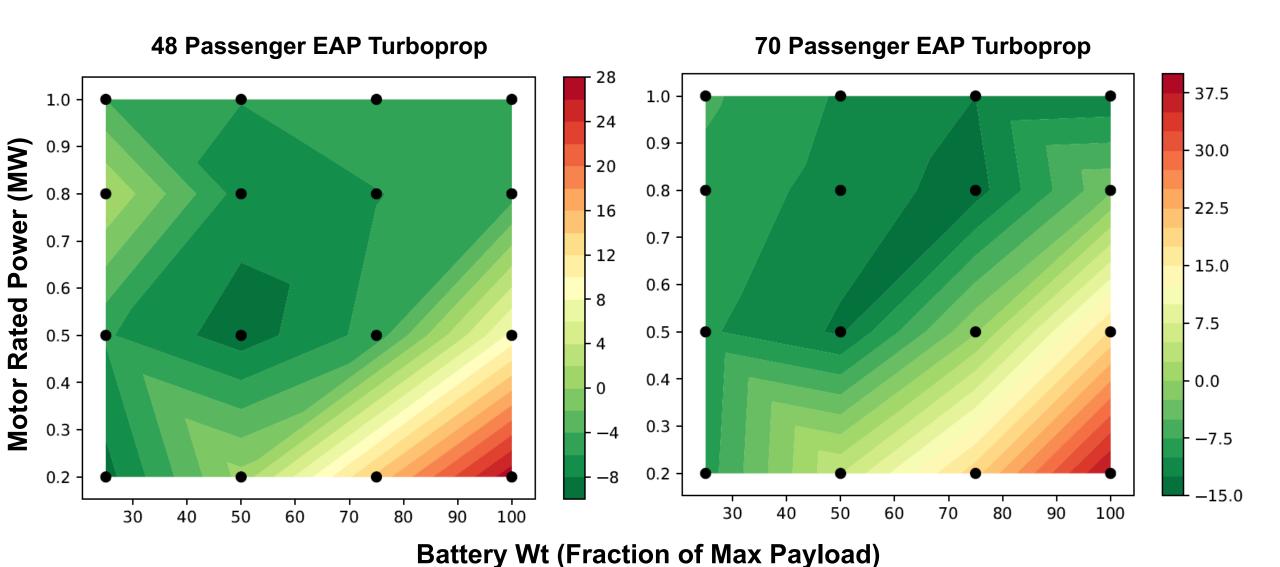
Aircraft	Passengers	Entry-Into- Service	Engine	Propulsion Architecture
Embraer EMB 110P	18	1968	PT6A-34	
ATR 42-500	48	1995	PW127E	Turboprop
ATR 72-600	70	2010	PW127F	
Embraer ERJ 145	50	1997	AE3007A1	Turbofan
Embraer E175	78	2002	CF34-8E	1 ul bolali

Reference Aircraft

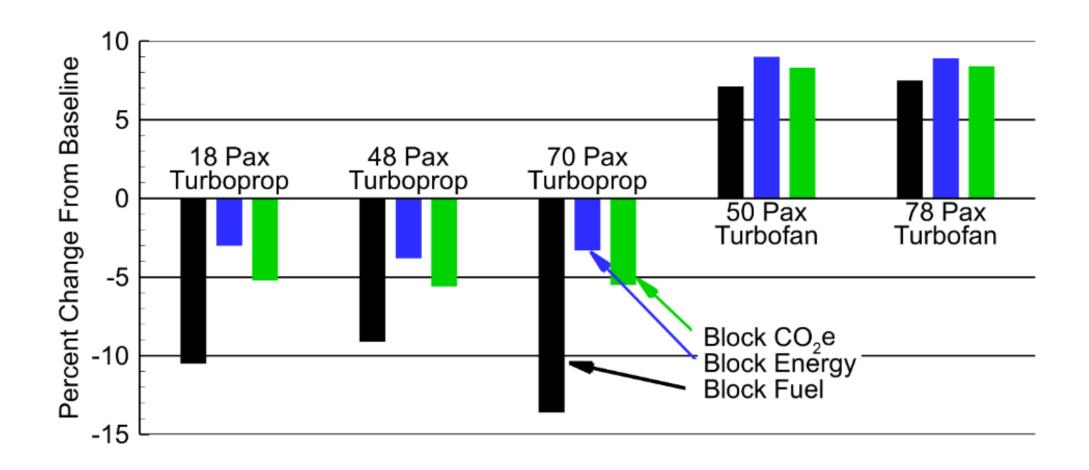




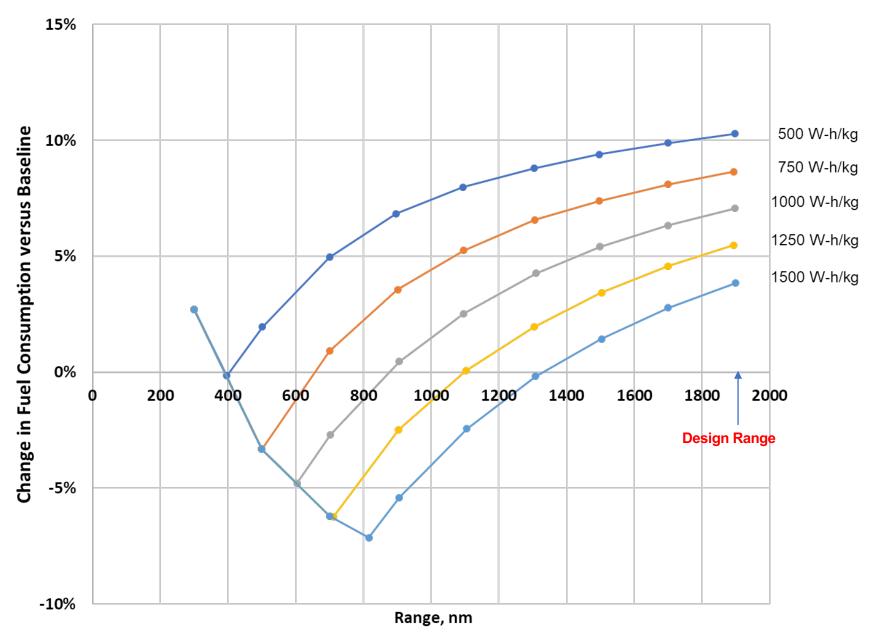
Parallel Hybrid Design Space Exploration Study (Marien et al.)



Summary of Block Fuel, Block Energy, Lifecycle CO2 Results



Relative Fuel Consumption as Function of Range (78 PAX Turbofan)



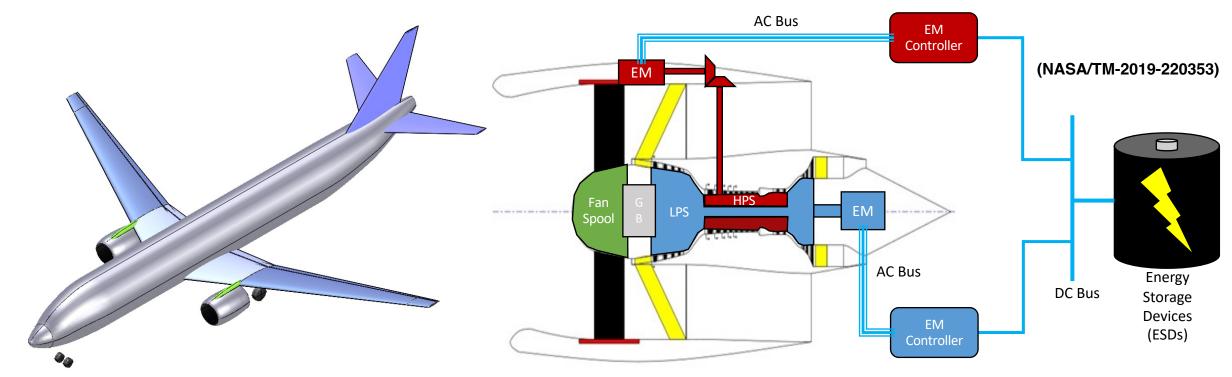
EAP Aircraft Cost Results Summary

	18 Passenger Turboprops		48 Passenger Turboprops		70 Passenger Turboprops		50 Passenger Turbofans		78 Passenger Turbofans	
	Baseline	EAP	Baseline	EAP	Baseline	EAP	Baseline	EAP	Baseline	EAP
Economic Mission Range (nm)	250	250	459	459	488	488	500	500	500	500
Empty Weight, less EAP hardware (lb)	6,412	6,817	21,617	22,765	26,135	28,585	23,876	25,553	43,844	46,916
EAP hardware (lb)		964		3,290		8,947		4,257		7,596
Block Fuel (lb)	484	440	1,834	1,673	1,949	1,698	2,212	2,185	3,645	3,524
Block Electricity (kW- h)		157		496		1,358		1,045		1,531
DOC+I (\$ /seat-mile)	0.491	0.531	0.199	0.218	0.081	0.097	0.131	0.150	0.112	0.130
% Change in DOC+I Compared to Baseline		8.1%		9.5%		19.8%		14.5%		16.1%

Despite reduced block fuel burn, added vehicle weight results in increased direct operating costs of parallel hybrid EAP vehicle concepts.

Mild Hybrid Electric Concept Study (Frederick et al.)

- Design exploration of 2035 mild hybrid electric single-aisle aircraft including e-Taxi and e-Climb
- Propulsion system architecture based on NASA Turbine Electrified Energy Management (TEEM)
- Alternative energy storage technologies considered batteries, supercapacitors, flywheels



https://arc.aiaa.org/doi/10.2514/6.2023-4226

EM – Electric Machine

GB – Gearbox

HPS – High Pressure Spool AC – Alternating Current

LPS – Low Pressure Spool

DC – Direct Current

Mild Hybrid Electric Concept Study (Frederick et al.)

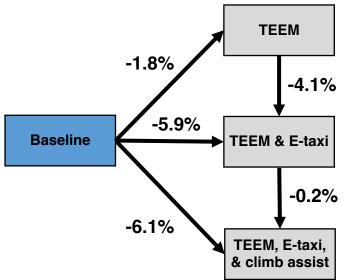
Energy storage system

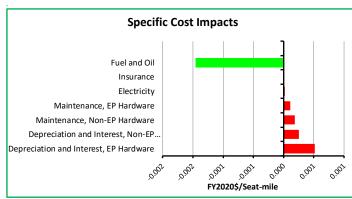
- Unusual requirements high power, low energy capacity
- Peak power to meet requirements for TEEM and acceleration phases of E-taxi
- Continuous power for electric climb assist and constant speed E-taxi
- Best system is a combination batteries for energy storage and supercapacitor for power output

	Specific Energy [W·h/kg]	Specific Power [W/kg]		
Configuration		Peak	Continuous	
Battery	350	700	350	
Supercapacitor	60	8,000	1,600	
Flywheel*	150	500	500	

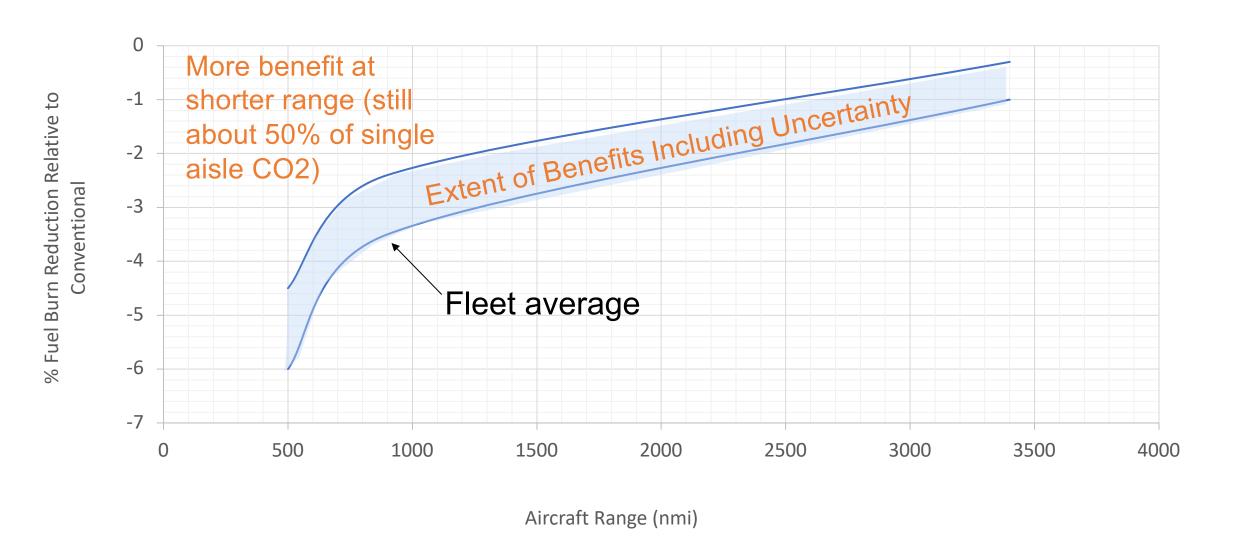
Integrated Performance Assessment

	Block fuel		
Configuration	Value [lbs]	wrt baseline	
Baseline	6,900	-	
TEEM* only	6,772	-1.8%	
E-taxi only	6,638	-3.8%	
TEEM* and E-taxi	6,491	-5.9%	
TEEM*, E-taxi, and 400kW climb assist	6,478	-6.1%	





Mild Hybrid Electric Concept Study for EIS 2030 (Gladin et al.)



Summary

- ARMD is making significant investment in EAP component and integrated system technologies and is leveraging expertise throughout industry and academia in this pursuit
- EAP is one critical element of the sustainable aviation solution; but it will take more
- EAP benefits at the integrated vehicle level are highly sensitive to a few key parametrics, including battery specific energy, motor specific power, and component efficiencies

Environmental sustainability may be the most significant aviation challenge we see in our lifetimes, and with your support, today's EAP research and technology development will help enable the revolutionary vehicle concepts needed to solve this problem.

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