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ST-23

SPACE NUCLEAR PROPULSION (SNP)

# MATERIAL PROPERTY HANDBOOK

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## I REVISION AND HISTORY PAGE

| Document Revision | Revision Date | Description                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 |
|-------------------|---------------|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| 0                 | 08/05/2020    | Initial SNP Material Property Databook                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      |
| 1                 | 08/02/2021    | <ul style="list-style-type: none"> <li>• Inclusion of Section 8 Composites and Structures</li> <li>• Format change</li> </ul>                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               |
| 2                 | 04/26/2023    | <ul style="list-style-type: none"> <li>• Update to Section 3.1.2 Mo-30W</li> <li>• Inclusion of Section 3.1.3 Titanium-Zirconium-Molybdenum (TZM)</li> <li>• Inclusion of Section 3.3.1 Rhenium (Re)</li> <li>• Update to elastic properties of Section 6.2.2 Beryllium (Be)</li> <li>• Switch from SBU to CUI designation</li> <li>• Minor technical fixes</li> <li>• Minor formatting changes</li> </ul>                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  |
| 2.1               | 08/25/2023    | Units fixed in room temperature tables – Young’s modulus and Shear modulus values are in [GPa], not [MPa]                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   |
| 3                 | 01/25/2024    | <ul style="list-style-type: none"> <li>• Inclusion of Section 6.1.3</li> <li>• Inclusion of Section IV Preface</li> <li>• Inclusion of Section V Introduction</li> <li>• Inclusion of Section VI Front Matter References</li> <li>• Phase diagrams in Sections 6.1.1 Uranium Nitride (UN) and 6.1.2 Uranium Carbide (UC) were replaced for readability</li> <li>• Removed the incorrect reference to “S-basis tubing” in figure captions of Section 2.1.1 Inconel 718 (IN718)</li> <li>• Altered Figure 6.2.1-3 to include phase information and an explanation of the inflection point</li> <li>• Added a trend for the specific heat of ZrH<sub>x</sub> below 300 K in Figure 6.2.1-8</li> <li>• Formatting fixes to various material property tables</li> <li>• All front matter is numbered with Roman numerals and is now included in Section II Table of Contents</li> <li>• Updated Section I Revision and History Page to include more details in the Description column</li> </ul> |
| 3.1               | 08/05/2024    | <ul style="list-style-type: none"> <li>• Switch from CUI//SP-EXPT to Uncontrolled Unclassified Information (UII) designation</li> <li>• Updated the title page and added an acknowledgement</li> <li>• Phrasing changes to Section IV Preface and Section V Introduction</li> <li>• Moved Section 6.1.3 to a new, controlled annex of the handbook</li> <li>• Removed stress rupture data from Section 2.1.1</li> </ul>                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     |

## II TABLE OF CONTENTS

|            |                                                                                       |             |
|------------|---------------------------------------------------------------------------------------|-------------|
| <b>I</b>   | <b>Revision and History Page .....</b>                                                | <b>ii</b>   |
| <b>II</b>  | <b>Table of Contents .....</b>                                                        | <b>iii</b>  |
| <b>III</b> | <b>Property List .....</b>                                                            | <b>v</b>    |
| <b>IV</b>  | <b>Preface .....</b>                                                                  | <b>xiii</b> |
| <b>V</b>   | <b>Introduction .....</b>                                                             | <b>xv</b>   |
| V.1        | Applicable Documents .....                                                            | xv          |
| V.2        | Overview of Material Properties .....                                                 | xvi         |
| V.2.1      | Density .....                                                                         | xvi         |
| V.2.2      | Melting Point .....                                                                   | xvii        |
| V.2.3      | Specific Heat Capacity .....                                                          | xviii       |
| V.2.4      | Thermal Conductivity .....                                                            | xix         |
| V.2.5      | Mean Coefficient of Thermal Expansion .....                                           | xx          |
| V.2.6      | Electrical Resistivity .....                                                          | xxi         |
| V.2.7      | Young's Modulus .....                                                                 | xxii        |
| V.2.8      | Shear Modulus .....                                                                   | xxiii       |
| V.2.9      | Poisson's Ratio .....                                                                 | xxiv        |
| V.3        | Handbook Development Methodology .....                                                | xxv         |
| V.3.1      | Data Generation and Literature Searches .....                                         | xxv         |
| V.3.2      | The Curve Fitting Process .....                                                       | xxvi        |
| V.3.3      | The Use of Models .....                                                               | xxx         |
| V.4        | Using the Material Property Handbook .....                                            | xxxix       |
| V.4.1      | Definitions .....                                                                     | xxxix       |
| V.4.2      | Navigation .....                                                                      | xxxiv       |
| V.4.3      | How to Use Material Property Plots .....                                              | xxxv        |
| V.4.4      | How to Use Material Property, Room Temperature, Composition, and Summary Tables ..... | xxxvi       |
| V.4.5      | How to Use Fit Equations .....                                                        | xxxix       |
| <b>VI</b>  | <b>Front Matter References .....</b>                                                  | <b>xli</b>  |
| <b>1</b>   | <b>Light Metal Alloys .....</b>                                                       | <b>1</b>    |
| 1.1        | Aluminum Alloys .....                                                                 | 1           |
| 1.1.1      | Aluminum 7075 (Al7075) .....                                                          | 2           |
| 1.2        | Titanium Alloys .....                                                                 | 17          |
| 1.2.1      | Titanium – 6Al – 4V (Ti64) .....                                                      | 18          |
| <b>2</b>   | <b>Nickel – Base Alloys .....</b>                                                     | <b>33</b>   |
| 2.1        | Precipitation Hardened Alloys .....                                                   | 33          |
| 2.1.1      | Inconel 718 (IN718) .....                                                             | 34          |
| <b>3</b>   | <b>Refractory Metals and Alloys .....</b>                                             | <b>59</b>   |
| 3.1        | Molybdenum and Molybdenum Alloys .....                                                | 59          |
| 3.1.1      | Molybdenum (Mo) .....                                                                 | 60          |
| 3.1.2      | Molybdenum – 30 wt% Tungsten (Mo-30W) .....                                           | 83          |
| 3.1.3      | Titanium-Zirconium-Molybdenum (TZM) .....                                             | 113         |
| 3.2        | Tungsten and Tungsten Alloys .....                                                    | 136         |
| 3.2.1      | Tungsten (W) .....                                                                    | 137         |
| 3.3        | Rhenium and Rhenium Alloys .....                                                      | 159         |
| 3.3.1      | Rhenium (Re) .....                                                                    | 160         |
| <b>4</b>   | <b>Other Nonferrous Metals and Alloys .....</b>                                       | <b>182</b>  |
| 4.1        | Copper Alloys .....                                                                   | 182         |
| 4.1.1      | GRCop84 .....                                                                         | 183         |

|          |                                             |            |
|----------|---------------------------------------------|------------|
| 4.2      | Zirconium Alloys .....                      | 202        |
| 4.2.1    | Zircaloy .....                              | 203        |
| <b>5</b> | <b>Ferrous Alloys .....</b>                 | <b>228</b> |
| 5.1      | Austenitic Stainless Steels .....           | 228        |
| 5.1.1    | Stainless Steel 304 (SS304) .....           | 229        |
| 5.1.2    | Stainless Steel 347 (SS347) .....           | 251        |
| <b>6</b> | <b>Nuclear Materials .....</b>              | <b>276</b> |
| 6.1      | Nuclear Fuel Materials .....                | 276        |
| 6.1.1    | Uranium Nitride (UN) .....                  | 277        |
| 6.1.2    | Uranium Carbide (UC) .....                  | 309        |
| 6.1.3    | <i>See Annex</i> .....                      | 337        |
| 6.2      | Neutron Control Materials .....             | 338        |
| 6.2.1    | Zirconium Hydride (ZrH <sub>x</sub> ) ..... | 339        |
| 6.2.2    | Beryllium (Be) .....                        | 365        |
| <b>7</b> | <b>Refractory Ceramics .....</b>            | <b>392</b> |
| 7.1      | Oxides .....                                | 392        |
| 7.1.1    | Beryllium Oxide (BeO) .....                 | 393        |
| 7.1.2    | Zirconia (Yttria Stabilized) (YSZ) .....    | 411        |
| 7.2      | Carbides .....                              | 433        |
| 7.2.1    | Boron Carbide (B <sub>4</sub> C) .....      | 434        |
| 7.2.2    | Zirconium Carbide (ZrC) .....               | 458        |
| 7.3      | Nitrides .....                              | 482        |
| 7.3.1    | Zirconium Nitride (ZrN) .....               | 483        |
| 7.3.2    | Hafnium Nitride (HfN) .....                 | 500        |
| <b>8</b> | <b>Composites and Structures .....</b>      | <b>515</b> |
| 8.1      | Carbide Based .....                         | 515        |
| 8.1.1    | Crystalline SiC/SiC Composite .....         | 516        |
| 8.1.2    | ZrC-C Composite .....                       | 529        |
| 8.1.3    | High Porosity ZrC .....                     | 542        |

## III PROPERTY LIST

|          |                                                             |           |
|----------|-------------------------------------------------------------|-----------|
| <b>1</b> | <b>Light Metal Alloys.....</b>                              | <b>1</b>  |
| 1.1      | Aluminum Alloys.....                                        | 1         |
| 1.1.1    | Aluminum 7075 (Al7075).....                                 | 2         |
|          | Room Temperature Properties.....                            | 2         |
|          | Composition.....                                            | 2         |
|          | Density with Temperature.....                               | 3         |
|          | Thermal Conductivity with Temperature.....                  | 5         |
|          | Thermal Expansion with Temperature.....                     | 7         |
|          | Mean Coefficient of Thermal Expansion with Temperature..... | 9         |
|          | Specific Heat with Temperature.....                         | 11        |
|          | Elastic Modulus with Temperature.....                       | 13        |
|          | Tabulated Property Data.....                                | 15        |
|          | References.....                                             | 16        |
| 1.2      | Titanium Alloys.....                                        | 17        |
| 1.2.1    | Titanium – 6Al – 4V (Ti64).....                             | 18        |
|          | Room Temperature Properties.....                            | 18        |
|          | Composition.....                                            | 18        |
|          | Density with Temperature.....                               | 19        |
|          | Thermal Conductivity with Temperature.....                  | 21        |
|          | Thermal Expansion with Temperature.....                     | 23        |
|          | Mean Coefficient of Thermal Expansion with Temperature..... | 25        |
|          | Specific Heat with Temperature.....                         | 27        |
|          | Elastic Modulus with Temperature.....                       | 29        |
|          | Tabulated Property Data.....                                | 31        |
|          | References.....                                             | 32        |
| <b>2</b> | <b>Nickel – Base Alloys.....</b>                            | <b>33</b> |
| 2.1      | Precipitation Hardened Alloys.....                          | 33        |
| 2.1.1    | Inconel 718 (IN718).....                                    | 34        |
|          | Room Temperature Properties.....                            | 34        |
|          | Composition.....                                            | 34        |
|          | Density with Temperature.....                               | 35        |
|          | Thermal Conductivity with Temperature.....                  | 37        |
|          | Thermal Expansion with Temperature.....                     | 39        |
|          | Mean Coefficient of Thermal Expansion with Temperature..... | 41        |
|          | Specific Heat with Temperature.....                         | 43        |
|          | Electrical Resistivity with Temperature.....                | 45        |
|          | Young's Modulus with Temperature.....                       | 47        |
|          | Shear Modulus with Temperature.....                         | 49        |
|          | Poisson's Ratio with Temperature.....                       | 51        |
|          | Yield Strength with Temperature.....                        | 53        |
|          | Tensile Strength with Temperature.....                      | 55        |
|          | Tabulated Property Data.....                                | 57        |
|          | References.....                                             | 58        |
| <b>3</b> | <b>Refractory Metals and Alloys.....</b>                    | <b>59</b> |
| 3.1      | Molybdenum and Molybdenum Alloys.....                       | 59        |
| 3.1.1    | Molybdenum (Mo).....                                        | 60        |
|          | Room Temperature Properties.....                            | 60        |

|                                                              |     |
|--------------------------------------------------------------|-----|
| Density with Temperature .....                               | 61  |
| Thermal Conductivity with Temperature .....                  | 63  |
| Thermal Expansion with Temperature .....                     | 65  |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 67  |
| Specific Heat with Temperature .....                         | 69  |
| Electrical Resistivity with Temperature .....                | 71  |
| Young's Modulus with Temperature .....                       | 73  |
| Shear Modulus with Temperature .....                         | 75  |
| Poisson's Ratio with Temperature .....                       | 77  |
| Tabulated Property Data .....                                | 79  |
| References .....                                             | 80  |
| 3.1.2 Molybdenum – 30 wt% Tungsten (Mo-30W) .....            | 83  |
| Room Temperature Properties .....                            | 83  |
| Molybdenum - Tungsten Phase Diagram .....                    | 83  |
| Composition .....                                            | 83  |
| Density with Temperature .....                               | 84  |
| Thermal Conductivity with Temperature .....                  | 86  |
| Thermal Diffusivity with Temperature .....                   | 88  |
| Specific Heat with Temperature .....                         | 90  |
| Thermal Expansion with Temperature .....                     | 92  |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 94  |
| Electrical Resistivity with Temperature .....                | 96  |
| Young's Modulus with Temperature .....                       | 98  |
| Shear Modulus with Temperature .....                         | 101 |
| Poisson's Ratio with Temperature .....                       | 104 |
| Yield Strength with Temperature .....                        | 107 |
| Tensile Strength with Temperature .....                      | 108 |
| Tabulated Property Data .....                                | 110 |
| References .....                                             | 111 |
| 3.1.3 Titanium-Zirconium-Molybdenum (TZM) .....              | 113 |
| Room Temperature Properties .....                            | 113 |
| Composition .....                                            | 113 |
| Density with Temperature .....                               | 114 |
| Thermal Conductivity with Temperature .....                  | 116 |
| Thermal Expansion with Temperature .....                     | 118 |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 120 |
| Specific Heat with Temperature .....                         | 122 |
| Electrical Resistivity with Temperature .....                | 124 |
| Young's Modulus with Temperature .....                       | 126 |
| Yield Strength with Temperature .....                        | 128 |
| Tensile Strength with Temperature .....                      | 130 |
| Additional Application Notes .....                           | 132 |
| Tabulated Property Data .....                                | 133 |
| References .....                                             | 134 |
| 3.2 Tungsten and Tungsten Alloys .....                       | 136 |
| 3.2.1 Tungsten (W) .....                                     | 137 |
| Room Temperature Properties .....                            | 137 |
| Density with Temperature .....                               | 138 |
| Thermal Conductivity with Temperature .....                  | 140 |
| Thermal Expansion with Temperature .....                     | 142 |

|                                                                         |            |
|-------------------------------------------------------------------------|------------|
| Mean Coefficient of Thermal Expansion with Temperature .....            | 144        |
| Specific Heat with Temperature .....                                    | 146        |
| Electrical Resistivity with Temperature .....                           | 148        |
| Young's Modulus with Temperature .....                                  | 150        |
| Shear Modulus with Temperature .....                                    | 152        |
| Poisson's Ratio with Temperature .....                                  | 154        |
| Tabulated Property Data .....                                           | 156        |
| References .....                                                        | 157        |
| <b>3.3 Rhenium and Rhenium Alloys .....</b>                             | <b>159</b> |
| 3.3.1 Rhenium (Re) .....                                                | 160        |
| Room Temperature Properties .....                                       | 160        |
| Density with Temperature .....                                          | 161        |
| Thermal Conductivity with Temperature .....                             | 163        |
| Thermal Expansion with Temperature .....                                | 165        |
| Mean Coefficient of Thermal Expansion with Temperature .....            | 167        |
| Specific Heat with Temperature .....                                    | 169        |
| Electrical Resistivity with Temperature .....                           | 171        |
| Young's Modulus with Temperature .....                                  | 173        |
| Shear Modulus with Temperature .....                                    | 175        |
| Poisson's Ratio with Temperature .....                                  | 177        |
| Tabulated Property Data .....                                           | 179        |
| References .....                                                        | 180        |
| <b>4 Other Nonferrous Metals and Alloys .....</b>                       | <b>182</b> |
| 4.1 Copper Alloys .....                                                 | 182        |
| 4.1.1 GRCop84 .....                                                     | 183        |
| Room Temperature Properties .....                                       | 183        |
| Composition .....                                                       | 183        |
| Density with Temperature .....                                          | 184        |
| Thermal Conductivity with Temperature .....                             | 186        |
| Thermal Expansion with Temperature .....                                | 188        |
| Mean Coefficient of Thermal Expansion with Temperature .....            | 190        |
| Specific Heat with Temperature .....                                    | 192        |
| Electrical Resistivity with Temperature .....                           | 194        |
| Yield Strength with Temperature .....                                   | 196        |
| Tensile Strength with Temperature .....                                 | 198        |
| Tabulated Property Data .....                                           | 200        |
| References .....                                                        | 201        |
| 4.2 Zirconium Alloys .....                                              | 202        |
| 4.2.1 Zircaloy .....                                                    | 203        |
| Room Temperature Properties .....                                       | 203        |
| Composition .....                                                       | 203        |
| Density with Temperature .....                                          | 204        |
| Electrical Resistivity with Temperature .....                           | 206        |
| Thermal Conductivity with Temperature .....                             | 208        |
| Axial Thermal Expansion with Temperature .....                          | 210        |
| Diametral Thermal Expansion with Temperature .....                      | 212        |
| Axial Coefficient of Thermal Expansion (CTE) with Temperature .....     | 214        |
| Diametral Coefficient of Thermal Expansion (CTE) with Temperature ..... | 216        |
| Specific Heat with Temperature .....                                    | 218        |
| Youngs Modulus with Temperature .....                                   | 220        |

|                                                              |            |
|--------------------------------------------------------------|------------|
| Shear Modulus with Temperature .....                         | 222        |
| Poisson's Ratio with Temperature .....                       | 224        |
| Tabulated Property Data .....                                | 226        |
| References .....                                             | 227        |
| <b>5 Ferrous Alloys .....</b>                                | <b>228</b> |
| 5.1 Austenitic Stainless Steels .....                        | 228        |
| 5.1.1 Stainless Steel 304 (SS304) .....                      | 229        |
| Room Temperature Properties .....                            | 229        |
| Composition .....                                            | 229        |
| Density with Temperature .....                               | 230        |
| Thermal Conductivity with Temperature .....                  | 232        |
| Thermal Expansion with Temperature .....                     | 234        |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 236        |
| Specific Heat with Temperature .....                         | 238        |
| Electrical Resistivity with Temperature .....                | 240        |
| Young's Modulus with Temperature .....                       | 242        |
| Shear Modulus with Temperature .....                         | 244        |
| Poisson's Ratio with Temperature .....                       | 246        |
| Tabulated Property Data .....                                | 248        |
| References .....                                             | 249        |
| 5.1.2 Stainless Steel 347 (SS347) .....                      | 251        |
| Room Temperature Properties .....                            | 251        |
| Composition .....                                            | 251        |
| Density with Temperature .....                               | 252        |
| Thermal Conductivity with Temperature .....                  | 254        |
| Thermal Expansion with Temperature .....                     | 256        |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 258        |
| Specific Heat with Temperature .....                         | 260        |
| Electrical Resistivity with Temperature .....                | 262        |
| Young's Modulus with Temperature .....                       | 264        |
| Shear Modulus with Temperature .....                         | 266        |
| Poisson's Ratio with Temperature .....                       | 268        |
| Yield Strength with Temperature .....                        | 270        |
| Tensile Strength with Temperature .....                      | 272        |
| Tabulated Property Data .....                                | 274        |
| References .....                                             | 275        |
| <b>6 Nuclear Materials .....</b>                             | <b>276</b> |
| 6.1 Nuclear Fuel Materials .....                             | 276        |
| 6.1.1 Uranium Nitride (UN) .....                             | 277        |
| Room Temperature Properties .....                            | 277        |
| Uranium – Nitrogen Phase Diagram .....                       | 277        |
| Density with Temperature .....                               | 278        |
| Thermal Conductivity with Temperature .....                  | 280        |
| Thermal Conductivity with Porosity .....                     | 282        |
| Thermal Expansion with Temperature .....                     | 284        |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 286        |
| Specific Heat with Temperature .....                         | 288        |
| Electrical Resistivity with Temperature .....                | 290        |
| Young's Modulus with Porosity .....                          | 292        |
| Young's Modulus with Temperature .....                       | 294        |

|                                                              |     |
|--------------------------------------------------------------|-----|
| Shear Modulus with Porosity .....                            | 296 |
| Shear Modulus with Temperature .....                         | 298 |
| Poisson's Ratio with Porosity .....                          | 300 |
| Poisson's Ratio with Temperature .....                       | 302 |
| Tabulated Property Data .....                                | 304 |
| References .....                                             | 305 |
| 6.1.2 Uranium Carbide (UC).....                              | 309 |
| Room Temperature Properties.....                             | 309 |
| Uranium – Carbon Phase Diagram .....                         | 309 |
| Density with Temperature .....                               | 310 |
| Thermal Conductivity with Temperature.....                   | 312 |
| Thermal Expansion with Temperature.....                      | 314 |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 316 |
| Specific Heat with Temperature .....                         | 318 |
| Electrical Resistivity with Temperature.....                 | 320 |
| Young's Modulus with Porosity .....                          | 322 |
| Young's Modulus with Temperature.....                        | 324 |
| Shear Modulus with Porosity.....                             | 326 |
| Shear Modulus with Temperature .....                         | 328 |
| Poisson's Ratio with Porosity .....                          | 330 |
| Poisson's Ratio with Temperature.....                        | 332 |
| Tabulated Property Data .....                                | 334 |
| References .....                                             | 335 |
| 6.1.3 See Annex.....                                         | 337 |
| 6.2 Neutron Control Materials.....                           | 338 |
| 6.2.1 Zirconium Hydride (ZrH <sub>x</sub> ).....             | 339 |
| ZrH <sub>2</sub> Room Temperature Properties.....            | 339 |
| Zirconium – Hydrogen Phase Diagram .....                     | 340 |
| Density with H/Zr Ratio .....                                | 341 |
| Density with Temperature .....                               | 343 |
| Thermal Conductivity with Temperature.....                   | 345 |
| Thermal Expansion with Temperature.....                      | 347 |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 349 |
| Thermal Diffusivity with Temperature .....                   | 351 |
| Specific Heat with Temperature .....                         | 353 |
| Electrical Resistivity with H/Zr Ratio .....                 | 355 |
| Electrical Resistivity with Temperature.....                 | 357 |
| Young's Modulus with H/Zr Ratio .....                        | 359 |
| Shear Modulus with H/Zr Ratio .....                          | 360 |
| Poisson's Ratio with H/Zr Ratio.....                         | 361 |
| Tabulated Property Data .....                                | 362 |
| References .....                                             | 363 |
| 6.2.2 Beryllium (Be).....                                    | 365 |
| Room Temperature Properties.....                             | 365 |
| Density with Temperature .....                               | 366 |
| Thermal Conductivity with Temperature.....                   | 368 |
| Thermal Expansion with Temperature.....                      | 370 |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 372 |
| Specific Heat with Temperature .....                         | 374 |
| Electrical Resistivity with Temperature.....                 | 376 |

|                                                              |            |
|--------------------------------------------------------------|------------|
| Young's Modulus with Temperature .....                       | 378        |
| Shear Modulus with Temperature .....                         | 380        |
| Poisson's Ratio with Temperature .....                       | 382        |
| Yield Strength with Temperature .....                        | 384        |
| Tensile Strength with Temperature .....                      | 386        |
| Additional Application Notes.....                            | 388        |
| Tabulated Property Data .....                                | 389        |
| References .....                                             | 390        |
| <b>7 Refractory Ceramics .....</b>                           | <b>392</b> |
| 7.1 Oxides .....                                             | 392        |
| 7.1.1 Beryllium Oxide (BeO).....                             | 393        |
| Room Temperature Properties.....                             | 393        |
| Density with Temperature .....                               | 394        |
| Thermal Conductivity with Temperature.....                   | 396        |
| Thermal Expansion with Temperature.....                      | 398        |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 400        |
| Specific Heat with Temperature .....                         | 402        |
| Young's Modulus with Temperature .....                       | 404        |
| Compressive Strength with Temperature .....                  | 406        |
| Tabulated Property Data .....                                | 408        |
| References .....                                             | 409        |
| 7.1.2 Zirconia (Yttria Stabilized) (YSZ) .....               | 411        |
| Room Temperature Properties.....                             | 411        |
| Zirconia - Yttria Phase Diagram .....                        | 411        |
| Composition.....                                             | 411        |
| Density with Temperature .....                               | 412        |
| Thermal Conductivity with Temperature.....                   | 414        |
| Thermal Expansion with Temperature.....                      | 416        |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 418        |
| Specific Heat with Temperature .....                         | 420        |
| Electrical Resistivity with Temperature.....                 | 422        |
| Young's Modulus with Temperature .....                       | 424        |
| Shear Modulus with Temperature .....                         | 426        |
| Poisson's Ratio with Temperature.....                        | 428        |
| Tabulated Property Data .....                                | 430        |
| References .....                                             | 431        |
| 7.2 Carbides.....                                            | 433        |
| 7.2.1 Boron Carbide (B <sub>4</sub> C).....                  | 434        |
| Room Temperature Properties.....                             | 434        |
| Boron – Carbon Phase Diagram .....                           | 434        |
| Density with Temperature .....                               | 435        |
| Thermal Conductivity with Temperature.....                   | 437        |
| Thermal Expansion with Temperature.....                      | 439        |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 441        |
| Specific Heat with Temperature .....                         | 443        |
| Young's Modulus with Porosity .....                          | 445        |
| Young's Modulus with Temperature .....                       | 447        |
| Modulus of Rupture with Porosity.....                        | 449        |
| Modulus of Rupture with Temperature .....                    | 451        |
| Fracture Toughness with Temperature .....                    | 453        |

|                                                              |            |
|--------------------------------------------------------------|------------|
| Tabulated Property Data .....                                | 455        |
| References .....                                             | 456        |
| 7.2.2    Zirconium Carbide (ZrC) .....                       | 458        |
| Room Temperature Properties .....                            | 458        |
| Zirconium – Carbon Phase Diagram .....                       | 458        |
| Density with Temperature .....                               | 459        |
| Thermal Conductivity with Temperature .....                  | 461        |
| Thermal Expansion with Temperature .....                     | 463        |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 465        |
| Specific Heat with Temperature .....                         | 467        |
| Electrical Resistivity with Temperature .....                | 469        |
| Young's Modulus with Temperature .....                       | 471        |
| Shear Modulus with Temperature .....                         | 473        |
| Poisson's Ratio with Temperature .....                       | 475        |
| Hardness with Temperature .....                              | 477        |
| Tabulated Property Data .....                                | 479        |
| References .....                                             | 480        |
| 7.3    Nitrides .....                                        | 482        |
| 7.3.1    Zirconium Nitride (ZrN) .....                       | 483        |
| Room Temperature Properties .....                            | 483        |
| Zirconium – Nitrogen Phase Diagram .....                     | 483        |
| Density with Temperature .....                               | 484        |
| Thermal Conductivity with Temperature .....                  | 486        |
| Thermal Expansion with Temperature .....                     | 488        |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 490        |
| Specific Heat with Temperature .....                         | 492        |
| Electrical Resistivity with Temperature .....                | 494        |
| Tabulated Property Data .....                                | 496        |
| References .....                                             | 497        |
| 7.3.2    Hafnium Nitride (HfN) .....                         | 500        |
| Room Temperature Properties .....                            | 500        |
| Hafnium – Nitrogen Phase Diagram .....                       | 500        |
| Density with Temperature .....                               | 501        |
| Thermal Conductivity with Temperature .....                  | 503        |
| Thermal Expansion with Temperature .....                     | 505        |
| Mean Coefficient of Thermal Expansion with Temperature ..... | 507        |
| Specific Heat with Temperature .....                         | 509        |
| Electrical Resistivity with Temperature .....                | 511        |
| Tabulated Property Data .....                                | 513        |
| References .....                                             | 514        |
| <b>8    Composites and Structures .....</b>                  | <b>515</b> |
| 8.1    Carbide Based .....                                   | 515        |
| 8.1.1    Crystalline SiC/SiC Composite .....                 | 516        |
| Room Temperature Properties .....                            | 516        |
| Thermal Conductivity with Temperature .....                  | 517        |
| Coefficient of Thermal Expansion with Temperature .....      | 519        |
| Specific Heat with Temperature .....                         | 521        |
| Young's Modulus with Temperature .....                       | 523        |
| Additional Application Notes .....                           | 525        |
| References .....                                             | 527        |

|       |                                                             |     |
|-------|-------------------------------------------------------------|-----|
| 8.1.2 | ZrC-C Composite .....                                       | 529 |
|       | Room Temperature Properties.....                            | 529 |
|       | Thermal Conductivity with Temperature.....                  | 530 |
|       | Coefficient of Thermal Expansion with Temperature .....     | 532 |
|       | Specific Heat with Temperature .....                        | 534 |
|       | Young's Modulus with Temperature and Carbon + Porosity..... | 536 |
|       | RT Young's Modulus with % Carbon + Porosity .....           | 538 |
|       | References .....                                            | 540 |
| 8.1.3 | High Porosity ZrC.....                                      | 542 |
|       | Room Temperature Properties.....                            | 542 |
|       | Thermal Conductivity with Temperature.....                  | 543 |
|       | Coefficient of Thermal Expansion with Temperature .....     | 545 |
|       | Specific Heat with Temperature .....                        | 547 |
|       | Young's Modulus with Temperature and Porosity .....         | 549 |
|       | References .....                                            | 551 |

## IV PREFACE

The National Aeronautics and Space Administration's (NASA's) Space Nuclear Propulsion (SNP) project aims to demonstrate the feasibility of nuclear thermal propulsion (NTP) as a thrust mechanism for spacecraft. An initial Demonstration Rocket for Agile Cislunar Operations (DRACO) launch will serve as a test of the technology in Earth's orbit [1]. Future work may involve the development of an NTP engine and spacecraft design with the capability to transport astronauts from Earth to Mars and back safely, though NTP has other potential uses too.

A large team of experts is required to accomplish these lofty goals. While NASA's Marshall Space Flight Center (MSFC) leads engine and nuclear reactor work, the Department of Defense's (DOD's) Defense Advanced Research Projects Agency (DARPA) controls development of the rest of the spacecraft for DRACO [1]. There are many contributors to this project beyond MSFC and DARPA, including other NASA centers, Department of Energy (DOE) national laboratories, researchers at universities, and industry design teams, among others. A list of contributors to the project can be found on the [SNP SharePoint site](#) [2].

Several materials under active consideration by the SNP project, including materials in this handbook, will operate in extreme environments should they ever be incorporated into spacecraft. While any rocket must contend with cryogenic temperatures, vacuum pressures, and the cosmic radiation imposed by outer space, NTP brings its own suite of challenges to the table. Since the specific impulse of an NTP-propelled spacecraft is dependent on nuclear reactor temperatures, engineers are incentivized to design a reactor that functions at the highest temperatures possible. Simultaneously, other components must have a low temperature due to their environments or for operational reasons. Radiation builds on the dangers of the wide-ranging temperatures and vacuum exposure. While cosmic radiation imposed by space is always an external threat, the reactors aboard NTP rockets simultaneously produce radiation from within, albeit for a limited duration.

For any engineering project, designers must ensure that materials – and the components that use them – can function in their anticipated operating environments without cracking, improperly deforming, or otherwise compromising the integrity of the structure created.

To predict whether a material will function properly in its application environment, one can look to a material's standardized properties. Various properties are important for SNP modelers and designers, such as physical, thermal, and mechanical properties. SNP-HDBK-0008, referred to as “the handbook” going forward, was created to consolidate data on solid materials of interest to SNP across a wide range of temperatures, thereby improving access to the data throughout the project. Simultaneously, the handbook serves to standardize values across the project, which aids in modeling and review activities.

Mechanical properties often exhibit a greater dependency on a material's composition, processing methods, grain size, and geometry than thermophysical properties do. Since this handbook is primarily built from historical literature data with limited traceability, thermophysical properties are provided more frequently than mechanical properties. For mechanical properties to be better understood, experiments should be devised for key materials while keeping geometries, manufacturing methods, and other factors as consistent as possible.

Peer reviews by experts are a key aspect of the scientific method, and they help to locate and address technical issues. Therefore, beginning with Revision 3 of this handbook, the SNP Materials Characterization Working Group (MCWG) community examined new additions to the handbook and provided their feedback. With each potential change to the handbook, the MCWG board was given an opportunity for review and the authority to prevent alterations. Changes range from handbook formatting improvements to material and material property additions. For more details on the SNP MCWG, the MCWG board, and the SNP-HDBK-0008 review process, please refer to the MCWG Charter [3].

All readers are recommended to read Section V.4 *Using the Material Property Handbook* for techniques to make the most of this handbook. Other areas of Section V *Introduction* are useful for gaining a basic understanding of material properties and learning how this handbook was created. In addition, technical experience and engineering judgement will aid readers in using SNP material property data correctly. When using the data from this handbook in their own work, readers should think about the real-world situations that

the data represents. Errors are constantly tracked down and removed from the handbook, but readers' additional diligence will help them maintain high quality designs and models.

Most of the trends in this handbook are created through the average value curve fitting process illustrated in Section V.3.2. This quality of data may be acceptable for spaceflight applications, but it is dependent on both the individual material property and the project's risk posture. NASA-STD-6016, Standard Materials and Processes Requirements for Spacecraft, directs designers to use the Metallic Materials Properties Development and Standardization Handbook (MMPDS) for determining data accuracy requirements or tolerance levels [4, 5]. Unless a project is willing to take on a considerable amount of risk, the materials used in spaceflight systems must be well understood.

Of course, the two NTP systems envisioned by the SNP project have different risk profiles – one is intended to prove that humanity can build upon the successes of NTP seen in the 1960s, and this mission has a relatively short duration. The second mission will pose a greater challenge due to many factors. Increased mission duration and the need to one day support crew on the spacecraft expand on and introduce new engineering constraints. These constraints result in new problems for the materials science community to solve.

Beginning with Revision 3.1, this handbook is no longer subject to export control restrictions. Instead, an SNP-HDBK-0008 Annex document will contain all data evaluated as Controlled Unclassified Information (CUI). Material sections located in the Annex are denoted in Section II *Table of Contents* by the marking “*See Annex*”.

It is hoped that the following handbook helps address readers' material property data needs. If readers have any questions or comments on the SNP Material Property Handbook, then they may contact the SNP MCWG coordinator, Peter Karkos ([peter.d.karkos@nasa.gov](mailto:peter.d.karkos@nasa.gov)). All feedback is welcome.

## V INTRODUCTION

This section serves to heighten readers' understanding of this handbook. Those who wish to learn more about material properties, data generation, the analytical methods used to create trends, and methods for using the handbook to its full potential should read on.

The material property handbook is divided into broad sections primarily based on material composition. Dividing the handbook by material use in parts and components was contemplated, but rarely pursued, due to overlapping use cases for the various materials and the export control requirements that would be imposed. At present, the primary categories of materials for SNP applications are as follows: light metal alloys, nickel-base alloys, refractory metals and alloys, other nonferrous metals and alloys, ferrous alloys, nuclear materials, refractory ceramics, and composites and structures.

The broad, primary sections are broken down once again by composition. From there, individual material chapters are listed within the resulting subsections. For example, light metal alloys are one primary section, consisting of aluminum and titanium alloy secondary sections. In this example, aluminum 7075 (Al7075) and titanium – 6Al – 4V (Ti64) are the material sections (or chapters) within these two secondary subsections, respectively. This style of section division was inspired not only by several modern, digitized handbooks in MSFC's Materials and Processes Technical Information System (MAPTIS) [6], but also by Rover / Nuclear Engine for Rocket Vehicle Application (NERVA) documentation from the 1960s [7]. NERVA achieved many important milestones in the name of space nuclear propulsion [8], so the project is a regular topic of discussion for this handbook's design team.

Across the handbook, a focus is placed on thermophysical properties, but mechanical properties appear as well. Thermophysical properties plotted include density, thermal conductivity, linear thermal expansion, the mean coefficient of linear thermal expansion, thermal diffusivity, specific heat capacity (specific heat), and electrical resistivity. Mechanical properties like Young's modulus (elastic modulus), shear modulus (modulus of rigidity), Poisson's ratio, offset yield strength, and tensile strength are included for several materials, though a wide data spread is often observed for the reasons mentioned in Section IV *Preface*.

### V.1 Applicable Documents

The documents listed in Table V-1 contain requirements and guidance that can aid in the use of this handbook. Note that this table is not all-inclusive. Further, the sources that material property data was extracted from can be found separately in the *References* subsection of each material section. A list of commonly used data sources is provided in Table V-2 of Section V.3.1. The most recent versions of government standards, as well as many commercial standards, can be found at <https://standards.nasa.gov>.

Table V-1: Applicable Documents and References.

| Reference                          | Title                                                                               |
|------------------------------------|-------------------------------------------------------------------------------------|
| <a href="#">MAPTIS</a> [6]         | Materials and Processes Technical Information System                                |
| <a href="#">MMPDS</a> [5]          | The Metallic Materials Properties Development and Standardization Handbook          |
| <a href="#">NASA-STD-6016</a> [4]  | Standard Materials and Processes Requirements for Spacecraft                        |
| <a href="#">SNPP MDB</a> [9]       | Space Nuclear Power and Propulsion Materials Database; unavailable – in development |
| <a href="#">SNP-PLAN-0010</a> [10] | Nuclear Thermal Propulsion Fuel and Moderator Development Plan                      |

## V.2 Overview of Material Properties

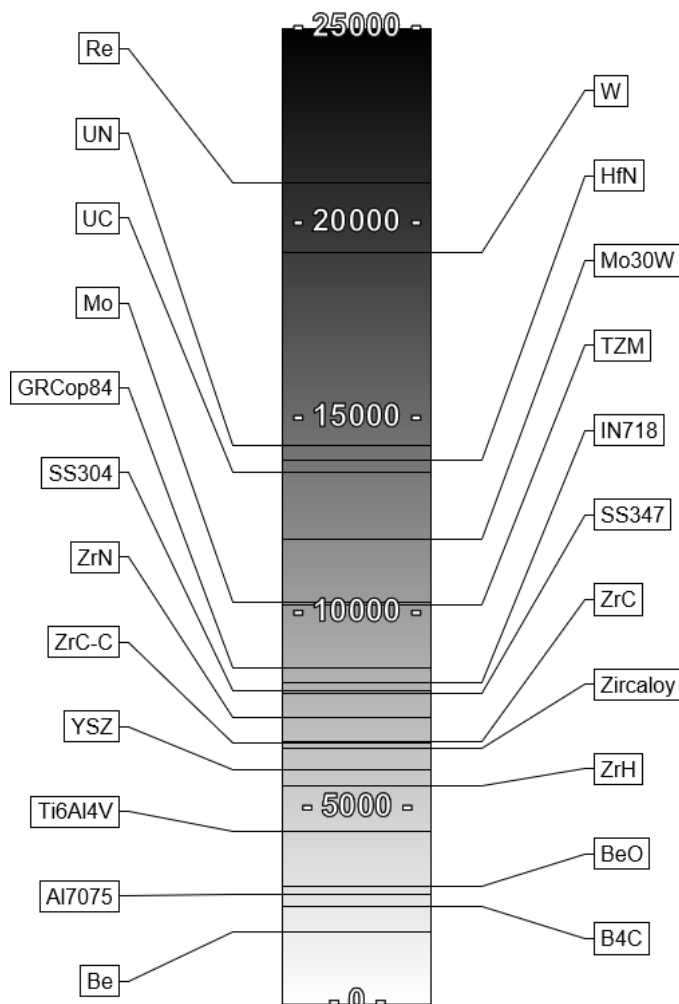


Figure V-1: Room temperature densities of select materials in kg/m<sup>3</sup>.

### V.2.1 Density

Materials' density, or mass per unit volume, is calculated for most materials using the general formula of Equation (V-1).

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3 \quad (V-1)$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\rho_{RT} = \text{Room Temperature Density [kg/m}^3\text{]}$$

$$dL/L_0(T) = \text{Linear Thermal Expansion [\%]}$$

$$T = \text{Temperature [K]}$$

In these equations,  $dL/L_0(T)$  represents the material's corresponding linear thermal expansion trend, which can be found in the same section as all the other material's properties. Section V.2.5 discusses thermal expansion in greater detail.

Density is an incredibly important property for spaceflight applications since it is used to evaluate hardware weights. The greater the weight of a spacecraft, the more thrust is required during launch. Additionally, for a given thrust, a rocket with greater mass will experience less acceleration than a rocket with a lesser mass. Therefore, using materials with low densities is a high priority for spaceflight projects.

To the left, Figure V-1 illustrates the densities of several SNP materials at room temperature (RT), in units of kg/m<sup>3</sup>. While common aerospace materials –

like aluminum alloy 7075 – will often be found at the lower end of the density scale, refractory metals occupy the higher tiers. Despite their many beneficial properties, materials like tungsten and rhenium are used sparingly in aerospace applications due to their high densities.

Density is often treated as a constant for a given material, but it does change with temperature. As temperatures rise, most materials expand, leading to a slight drop in density. Densities within this handbook will often be described as “theoretical densities” (TD), which correspond to the theoretical maximum density of a material, and “bulk densities”, which are the actual real-world densities of samples.

## V.2.2 Melting Point

Material melting points are also an important property, noting the temperature – or temperature range – where a material transitions from a solid to a liquid. See Figure V-2 for a visual representation of several materials' melting points. But this property has additional uses too. Materials regularly become weaker and more ductile as temperatures climb, so their effectiveness is reduced well before their melting point is reached.

Melting points are dependent on pressure and impurity concentrations in a material. Therefore, when a high accuracy melting point is needed, one should perform testing at the desired pressure and with the material processing method of interest. Unless otherwise specified, melting points are provided at atmospheric pressure.

Some materials lack clearly defined melting points. Or a material could be considered for use at various stoichiometries, as is the case with zirconium hydride ( $ZrH_x$ ). When possible, phase diagrams have been provided for these materials. Phase diagrams illustrate the different states a material will attain at different temperature and pressure combinations. Though harder to work with than single value melting points, phase diagrams better explain the hazards associated with changing environmental conditions. An example phase diagram is provided in Figure V-3.

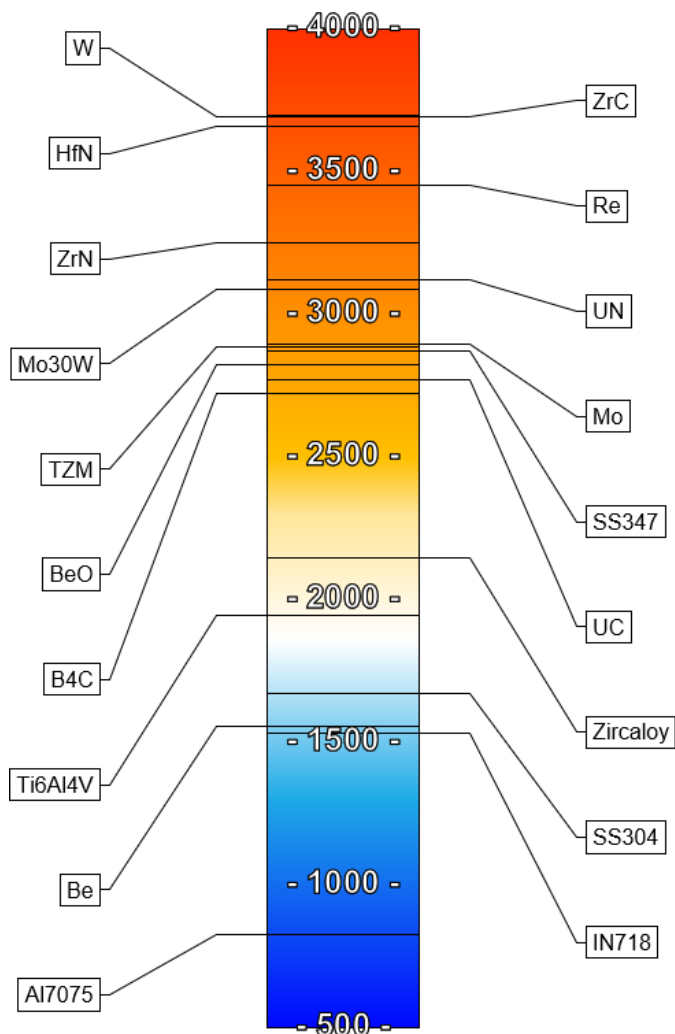


Figure V-2: Melting points of select materials in K.

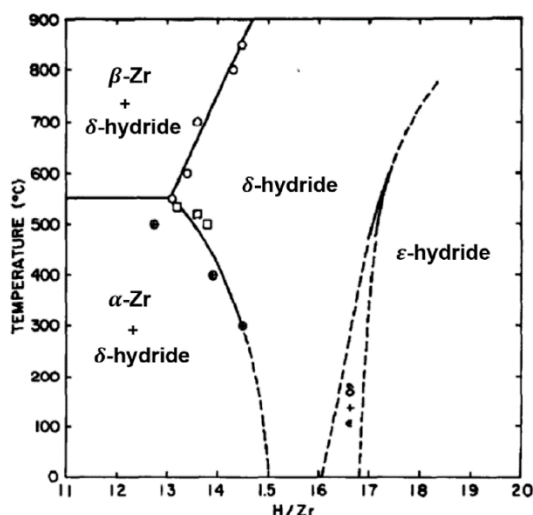


Figure V-3: Example phase diagram for zirconium hydride. See Section 6.2.1 for details.

For nuclear thermal propulsion to be function at the desired efficiency, certain components will be exposed to very high operating temperatures. Fuel and thermal insulator materials are expected to experience particularly high temperatures. Melting points can aid in quickly identifying ill-suited materials for different tasks, though other properties must be tested to understand the details of a material's performance at temperature.

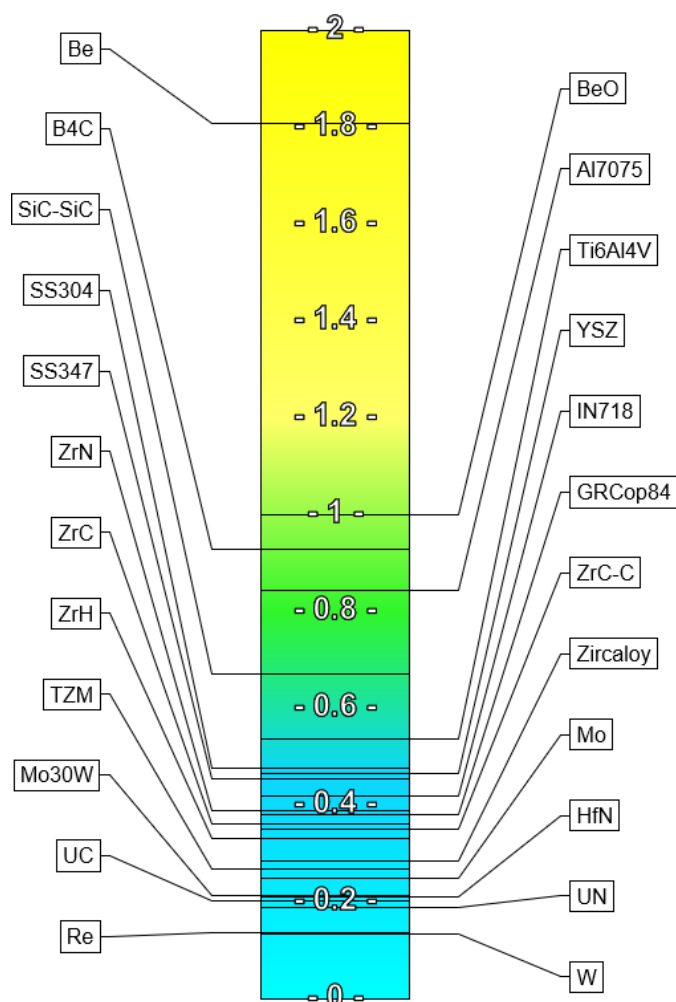


Figure V-4: Room temperature specific heat capacities of select materials in J/(g-K).

### V.2.3 Specific Heat Capacity

A material's specific heat capacity is a measure of its ability to store energy without experiencing a change in temperature. Within this handbook, that ability is often stated in terms of joules required to increase a gram of a material by one Kelvin. Figure V-4 contains the RT specific heat values for several materials.

Knowing the specific heat of a material is important when questioning the amount of energy to provide to or extract from a component to alter its temperature by a desired amount.

Heat transfer calculations are dependent on specific heat, including heat conduction, convection, and radiation scenarios. A material's ability to expand or contract is tied to its specific heat, as are its phase changes.

For most materials, specific heat increases with temperature, though the relationship is not a simple one. The Shomate style of equation employed by NIST, an example of which is provided in Table V-3 of Section V.3.2, typically provides good correlations for specific heat. This property benefits from small uncertainties during testing.

This handbook only concerns solid materials; therefore, only specific heat capacities at constant pressure are provided.

## V.2.4 Thermal Conductivity

Thermal conductivity represents the rate at which thermal energy passes through a material. It is often a calculated property; Equation (V-2) is used throughout this handbook to determine a material's thermal conductivity when its density, specific heat, and thermal diffusivity are known. The thermal conductivity of several materials at RT is provided in Figure V-5.

Metals are extremely good thermal and electrical conductors, with copper and its alloys having exceptional conductivity. Conversely, nonmetals are poor conductors, with conductivity values sometimes orders of magnitude lower than metals. Conductivity of metals is tied to the mobility of free electrons, and above cryogenic temperatures, it is roughly proportional to the product of electrical conductivity and the metal's absolute temperature [11]. Non-metals, on the other hand, lack a "sea" of valence electrons [12], and their conductivity is due to lattice vibrations and phonons [11].

In solid-fuel nuclear thermal propulsion, it is important that the fuel be sufficiently thermally conductive to avoid building up too much heat – the fuel must remain solid and stable to function correctly. Besides, the heat transferred out of nuclear fuel is needed to energize the propellant and create thrust for the rocket.

$$k(T) = \alpha(T)\rho(T)C_p(T) \quad (V-2)$$

$k(T)$  = Thermal Conductivity

$\alpha(T)$  = Thermal Diffusivity

$\rho(T)$  = Density

$C_p(T)$  = Specific Heat

$T$  = Temperature

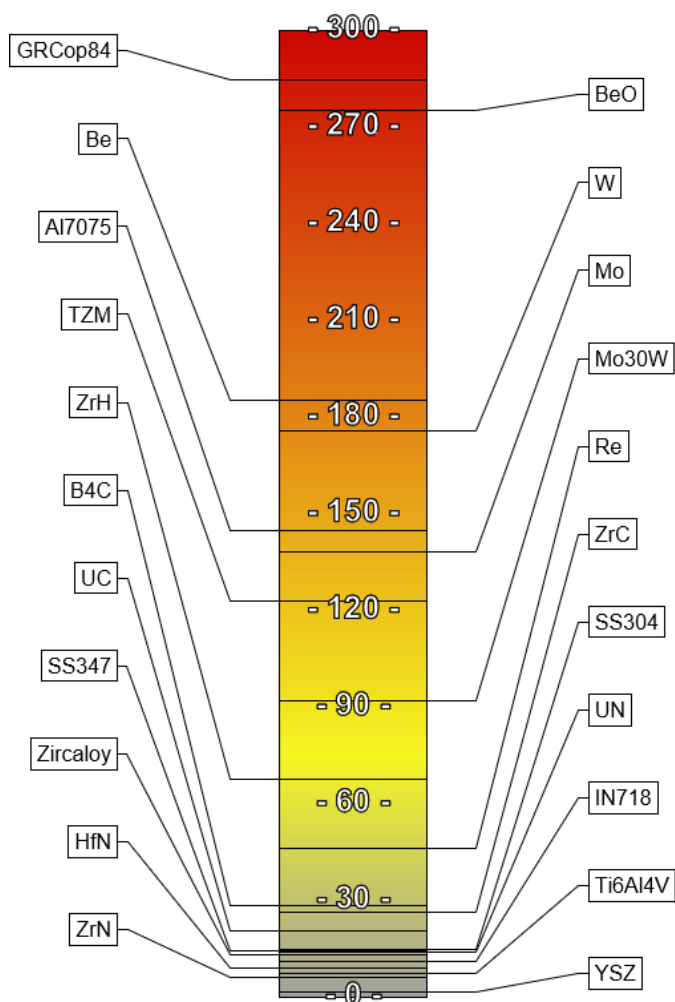


Figure V-5: Room temperature thermal conductivity for select materials in W/(m-K).

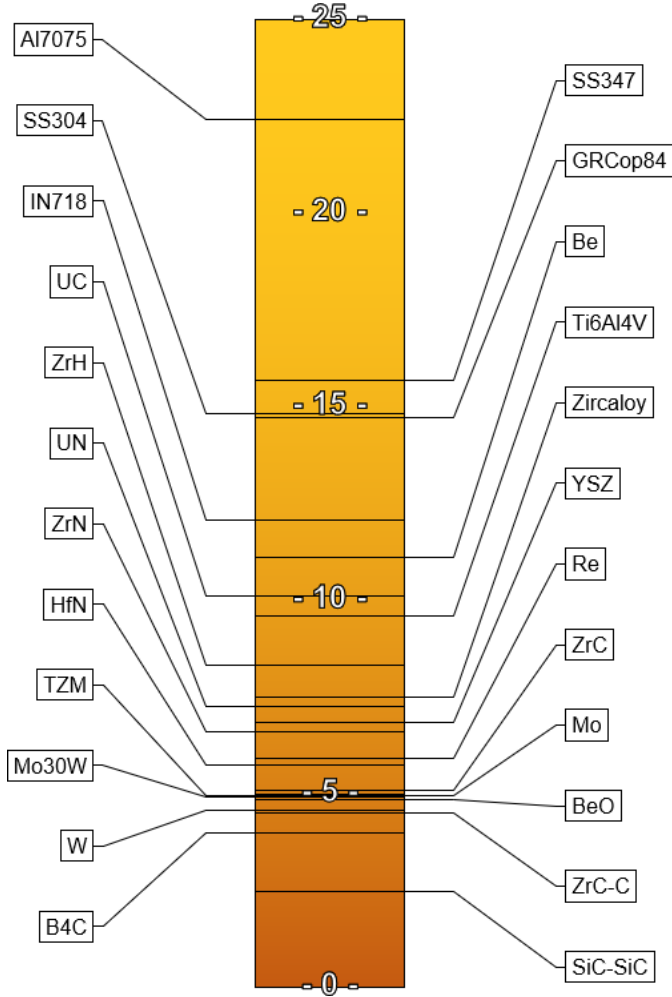


Figure V-6: Room temperature mean coefficients of linear thermal expansion for select materials in μm/(m·K).

temperature rises slightly. With the second term, one considers how a material expands due to a change in temperature from a fixed reference temperature to a new temperature it attains [13]. Unless otherwise specified, all the CTE data in this handbook is mean CTE data with RT as the reference temperature. The mean CTE reporting format was chosen due to its usefulness in computing thermal strains – no integration is required to perform these calculations.

Mean CTE values, like the ones shown in Figure V-6, are calculated from linear thermal expansion with the expression shown in Equation (V-3). For an example of a linear thermal expansion trend, see Figure V-7.

$$\bar{\alpha}(T) = \frac{\frac{dL}{L_0}(T)/100}{T - T_{ref}} * 10^6 \quad (V-3)$$

$\bar{\alpha}(T)$  = Mean Coefficient of Thermal Expansion [μm/(m · K)]

$\frac{dL}{L_0}(T)$  = Linear Thermal Expansion [%]

$T$  = Temperature [K]

$T_{ref}$  = Reference Temperature [K], usually RT

## V.2.5 Mean Coefficient of Thermal Expansion

When a material changes temperature, it will shrink or expand. In some systems – where temperature changes are minimal – these size fluctuations may be insignificant. But for nuclear systems, the opposite is true. Here, the large temperature changes experienced by materials make thermal expansion effects noteworthy. The transient periods of NTP have the potential for rapid temperature changes, compounding the issue. It is important that materials in contact can expand or contract without causing cracking.

Linear thermal expansion – or the one-dimensional change in a material when heated – is often measured through dilatometry. This data can be represented by the change in a material's length divided by the original length of the material, expressed as a percentage.

Coefficients of thermal expansion (CTEs) can be calculated from linear thermal expansion data. There are two types – the instantaneous CTE and mean (secant) CTE. The first represents how a material expands at a particular temperature when the

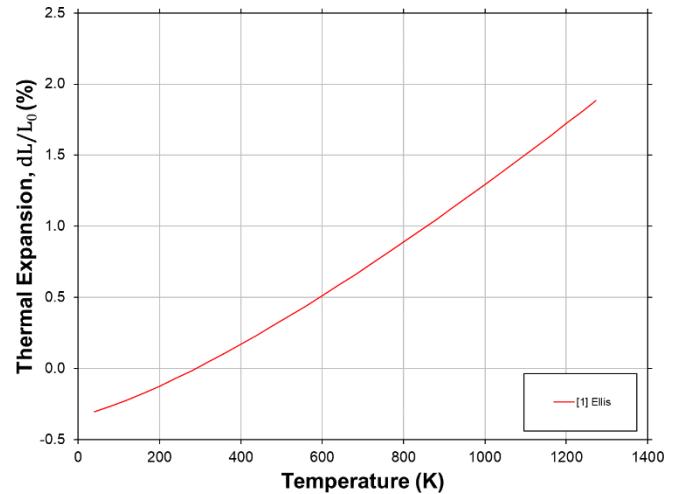


Figure V-7: Example GRCop84 thermal expansion plot. Note the symbol and units of the y-axis. See Section 4.1.1 for references and details on GRCop84.

## V.2.6 Electrical Resistivity

As discussed along with thermal conductivity in Section V.1.4, electrical conductivity reflects the mobility of electrons within a material [14]. Electrical resistivity is the inverse of electrical conductivity, demonstrating how electrically insulating a material can be. A plot of the RT electrical resistivity for multiple materials is shown in Figure V-8.

Various quantum mechanics models can be combined to understand electron motion and electron energy levels. However, this information is not required for the use of electrical resistivity at the macroscopic level – as a material property.

Ultimately, resistivity increases with atomic displacements, crystal lattice imperfections, and foreign atom substitution [14]. A material's composition, temperature, and radiation exposure affect these events, thereby influencing electrical resistivity.

Besides its occasional usefulness in the estimation of thermal conductivity, electrical resistivity aids in the design of systems from an electrical standpoint. Various components are used in spacecraft, each with its own electricity (or protection from electricity) requirements. For example, since the 1970's, spacecraft surface charging became an issue for the spaceflight industry [15]. Projects have experienced part degradation, part failure, mission redesign, and mission loss due to charging. The number of modern charging-related issues has dropped due to improved design practices and material choices [16].

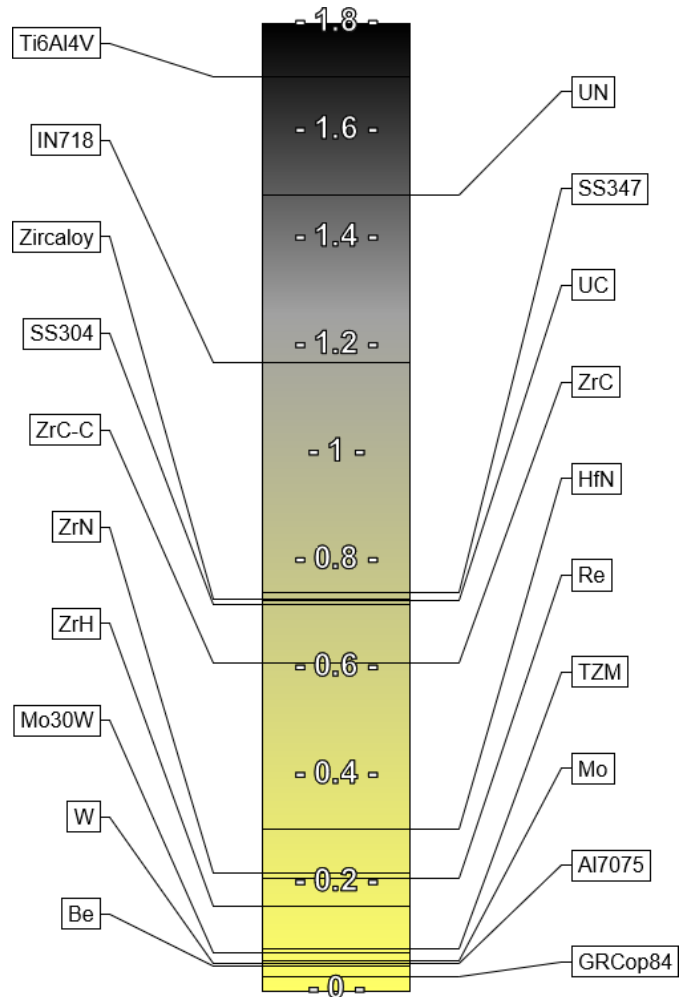


Figure V-8: Room temperature electrical resistivity for select materials in  $\mu\Omega\cdot m$ .

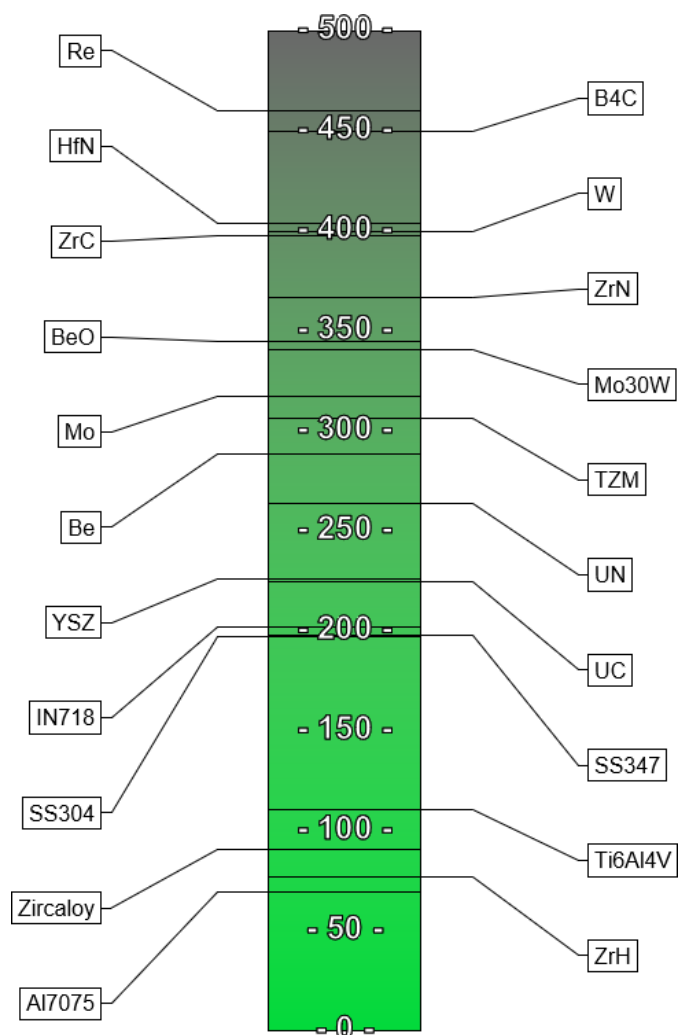


Figure V-9: Room temperature Young's modulus for select materials in GPa.

## V.2.7 Young's Modulus

The modulus of elasticity, also known as Young's modulus, is important for determining how a material will elastically deform when exposed to a force. Other elastic properties like the shear modulus and Poisson's ratio, which are described in Sections V.1.8 and V.1.9 respectively, are also involved in elastic behavior. Figure V-9 contains Young's modulus values for various materials at RT.

When a material is exposed to a tensile or compressive force, the Young's modulus recognizes the deformation the material should experience *parallel* to the applied force. Some tests – tensile and compressive tests – do exactly this, applying a force and marking down stress-strain curves. The initial, relatively constant slope of these curves is the Young's modulus. From a theoretical perspective, the tensile Young's modulus and compressive variant should be equal, though residual stresses can cause a discrepancy [17].

The *static modulus* is the material property being measured by tensile and compressive testing in the elastic region. Unfortunately, a wide uncertainty band is often observed for static modulus results.

*Dynamic modulus* testing, however, provides data of greater consistency, and the property involves vibration testing. While the static modulus arises from a sample deforming under an increasing stress, the dynamic modulus is found through the application of an oscillatory stress. Ultimately, dynamic modulus testing

can be faster, lower cost, and nondestructive in comparison to static modulus testing [18].

Note that static and dynamic moduli values are often similar but not identical. Researchers have proposed different theories on the difference between the moduli. Some state that static loading provides isothermal elastic constants, while dynamic loading yields adiabatic elastic constants. Alternatively, the varying atomic displacements due to the different magnitudes of strain could explain the discrepancy [18]. Another researcher noted:

*"At temperatures where time dependent deformation is absent the static and dynamic moduli are essentially equal with the dynamic value in some materials being slightly lower than the static value. At temperatures where time dependent deformation would occur in static tests the dynamic modulus is useful where vibratory loads accompany creep deformation"* [17].

When materials are known to have significantly different static and dynamic moduli values, equations can be used to convert between the two moduli [19].

## V.2.8 Shear Modulus

Dividing shear stress by shear strain yields the modulus of rigidity, or the shear modulus. This property represents a material's elastic shear stiffness.

The deformation of a solid resulting from a force parallel to one of its surfaces is characterized by the shear modulus. See Figure V-10 for the shear modulus values of several materials at RT.

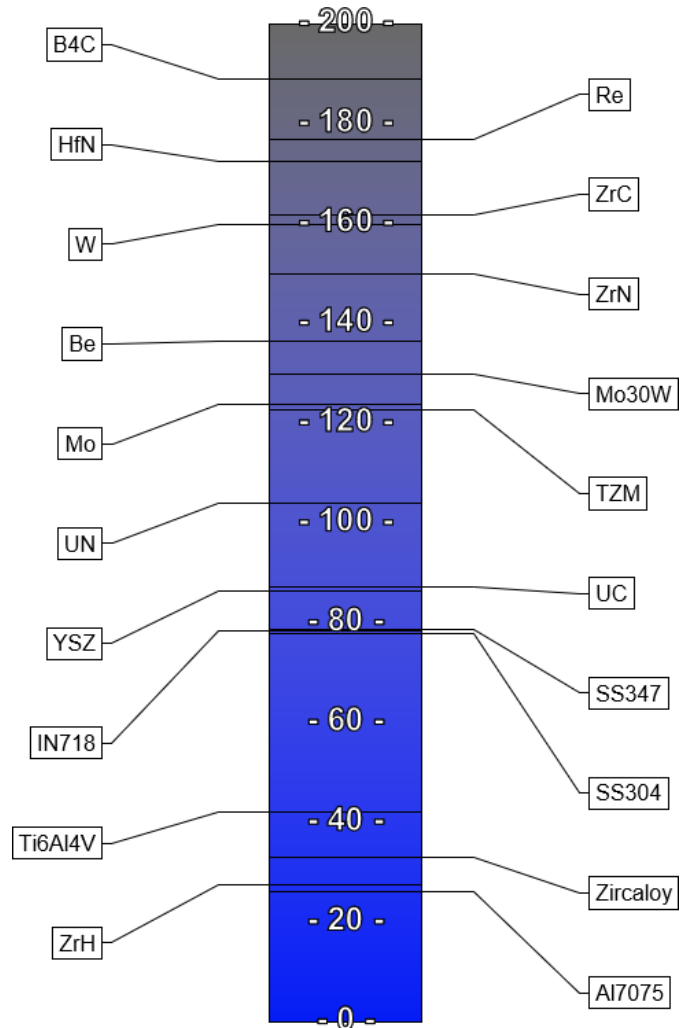


Figure V-10: Room temperature shear modulus for select materials in GPa.

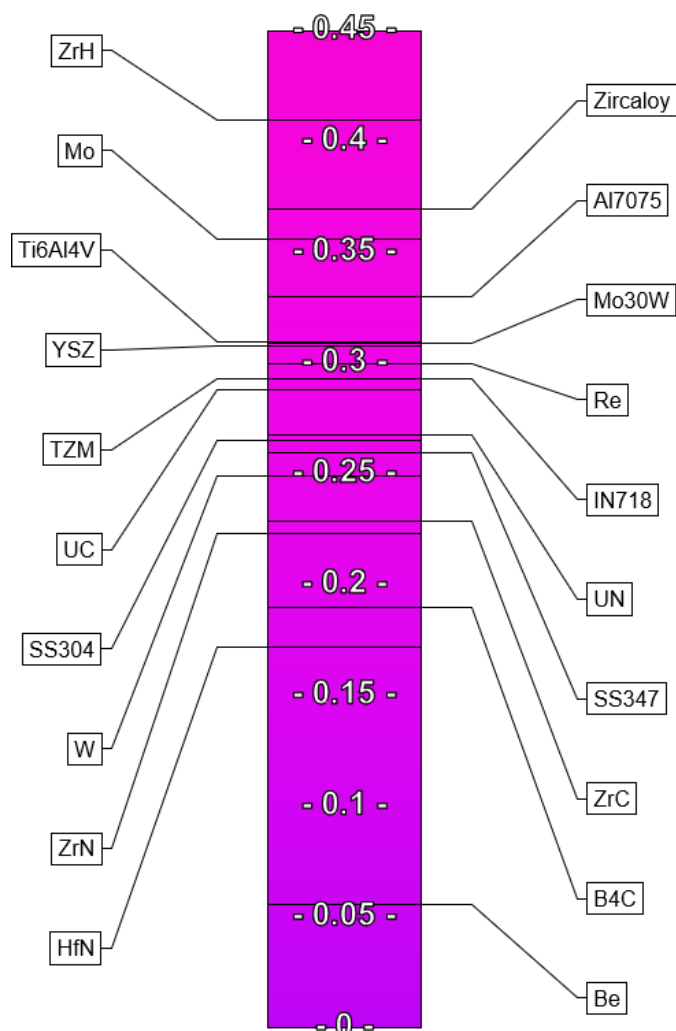


Figure V-11: Room temperature Poisson's ratio [-] for select materials.

## V.2.9 Poisson's Ratio

Young's modulus, shear modulus, and Poisson's ratio are related properties. They have a simple relationship within isotropic materials, as demonstrated in Equation (V-4). This equation is regularly used in this handbook, since literature data on Poisson's ratio is rarer than data on Young's modulus and shear modulus. However, on rare occasions when Poisson's ratio data is more readily available than shear modulus data, Equation (V-4) is rearranged to calculate the shear modulus.

$$\nu(T) = \frac{E(T)}{2G(T)} - 1 \quad (V-4)$$

$\nu(T)$  = Poisson's ratio [-]

$E(T)$  = Young's modulus [GPa]

$G(T)$  = Shear modulus [GPa]

$T$  = Temperature [K]

Poisson's ratio represents deformation of a material in perpendicular directions to an applied load. For example, for a rod lengthening due to a tensile force, the change in the rod's cross-sectional area can be predicted via the Poisson's ratio of the material. In Figure V-11, the RT Poisson's ratio values for multiple materials are provided.

Most materials have a positive Poisson's ratio, meaning that the material's cross section shrinks as its length expands, and vice versa. Some materials have

a negative Poisson's ratio, and their cross sections increase with increasing length. Note that this requires a density change, assuming the material's mass is held constant as its volume fluctuates. Finally, materials of near-zero Poisson's ratio experience no cross-sectional area change when their lengths are altered.

Isotropic material relationships are used in the handbook. But to have their elastic behavior characterized in a way that captures their full, real-world behavior, anisotropic materials require tensor expressions of their elastic constants. Anisotropic material testing and data acquisition is substantially more involved than isotropic material testing; as a result, tensor expressions are outside of the current scope of this handbook. These tensors are also more difficult for modelers and designers to work with, even if they improve the accuracy of results.

### V.3 Handbook Development Methodology

Here, the process used to modify this handbook is outlined. Modelers and designers should consult the information to follow when details on data generation, curve fitting, or the models used in the handbook are needed. This section largely exists for traceability purposes. The major strengths and limitations of this handbook's development can be discerned from the following explanations.

#### V.3.1 Data Generation and Literature Searches

The data used in this handbook stems from two sources – SNP material property testing campaigns and historical literature.

Details on SNP testing campaigns vary and are held within different planning documents. The Fuel and Moderator Development Plan (FMDP) [10] currently covers the bulk of SNP material testing work. Please also see SNP-PLAN-0021, Space Nuclear Propulsion Task 1.10 Materials & Process Mapping Documentation Recommendations [20], which aids readers in understanding the intent of the FMDP and applying the document in their own testing efforts. Data originating from SNP testing campaigns is considered “internal” data, whereas data acquired from other programs and projects is considered “external”.

Literature searches vary in duration – from a couple days to about a week – but are intended to be exhaustive. The searches typically include most databases in MAPTIS, with particular emphasis placed on MMPDS, Aerospace Structural Metals Handbook, and Material Universe. The results of past NASA, DOE, DOD, and industry aerospace efforts are examined during each literature search. Table V-2 is an incomplete list of important sources of material property data. Note that databases of academic journals and internet searches are also utilized when seeking historical literature data. The table is sorted alphabetically.

Numerous reports and handbooks from key projects have also been added to the table. Contact the SNP Material Property Handbook team for an index document noting the materials and properties included within past nuclear projects' data books. An avid reader will find that a tremendous amount of information on nuclear systems has been gathered by the references of Table V-2.

Table V-2: Frequently used handbooks, databases, and services for literature searches.

| Title                                                                          | Description                                                                                                                                                                                                       |
|--------------------------------------------------------------------------------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| Aerospace Structural Metals Handbook (ASMH)                                    | Contains data on metals and alloys from AMS, federal, and industry sources, with a focus on physical properties, mechanical properties, and factors limiting the load carrying capacity of each material          |
| AFCI Materials Handbook – Materials Data for Particle Accelerator Applications | Provides data on materials applicable to nuclear transmutation and Gen IV fast neutron spectrum systems, originally created by various DOE national labs to support the Accelerator Production of Tritium Project |
| Comprehensive Nuclear Materials                                                | Set of volumes encompassing various nuclear fuel systems, spent fuel processes, modeling nuclear systems, radiation effects in materials, and other material properties                                           |
| Defense Technical Information Center – Technical Reports                       | Contains technical reports funded by the DOD                                                                                                                                                                      |
| DOE Data Explorer (DDE)                                                        | Tool for locating data records resulting from DOE funding                                                                                                                                                         |
| DOE Office of Scientific and Technical Information – OSTI.GOV                  | Collection of research and technology information funded by the DOE                                                                                                                                               |
| DOE Public Access Gateway for Energy and Science (DOE PAGES)                   | Tool for locating peer-reviewed publications founded upon DOE research                                                                                                                                            |

# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

Table V-2: Frequently used handbooks, databases, and services for literature searches.

| Title                                                                 | Description                                                                                                                                                                                                |
|-----------------------------------------------------------------------|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| IAEA Thermo-Physical Materials Properties Database (IAEA THERPRO)     | Digital database of thermophysical materials data resulting from IAEA coordinated research projects                                                                                                        |
| Materials and Processes Technical Information System (MAPTIS)         | Houses many databases, which in turn provide physical, mechanical, and environmental properties of materials                                                                                               |
| MaterialUniverse                                                      | Database of material properties, environmental data, and costs, as well as information on many manufacturing processes – produced by ANSYS Granta                                                          |
| Metallic Materials Properties Development and Standardization (MMPDS) | Source of design data for many aerospace materials, as well as for different material geometries (fasteners, etc.)                                                                                         |
| NASA Glenn Library                                                    | Provides reference books in a physical and virtual collection                                                                                                                                              |
| NASA Technical Reports Server (NTRS)                                  | A repository of scientific and technical information created or funded by NASA                                                                                                                             |
| NERVA Program – Materials Properties Data Book                        | An Aerojet Nuclear Systems Company and Westinghouse Astronuclear Laboratory publication which was used to provide basic design criteria – including property data – for materials within the NERVA Program |
| NIST Chemistry WebBook                                                | Source of chemical and physical property data, as maintained by the NIST Standard Reference Data Program                                                                                                   |
| SNAP Technology Handbook                                              | Handbook including data and descriptions regarding liquid metals, hydride fuels and claddings, and refractory fuels and claddings for use in the SNAP Program                                              |
| SP-100 Materials Handbook                                             | Compilation of data on Re, BeO, Nb-1Zr, PWC-11, Li, and UN for use in the SP-100 Program                                                                                                                   |
| SP-100 PCA Materials Properties Database                              | Compilation of data on SiGe/GaP, GS-526, Al <sub>2</sub> O <sub>3</sub> , W, graphite, Nb, and Cu – originally intended for inclusion in the SP-100 Materials Handbook                                     |
| Thermophysical Properties of Matter – The TPRC Data Series            | Set of volumes on the thermophysical properties of over 5,000 materials; experimental data and trends are both provided                                                                                    |

Once data is gathered through a comprehensive literature search and from new testing, it is housed in “Excel Database” files. Important details about the data, such as the material manufacturing method and specimen geometry are also stored here. Data is plotted to quickly assess its variance. Outlier datasets – or individual data points – are investigated to determine why discrepancies exist. When issues with experimental methods or conditions are identified, the data may be discarded, adjusted through a correction factor, or marked with a note on its irregularity, whichever is most applicable. Note that data that is “discarded” is merely hidden from view in plots and is held separate from trend creation – it can still be found in the Excel Database if the data is later determined to be useful. After the initial vetting of the data, correlations can be developed.

## V.3.2 The Curve Fitting Process

Matrix Laboratory (MATLAB) and the MATLAB curve fitting tool [21] were used to develop material property trends. These trends are empirical and based on test data, rather than scientific theory. For chapters introduced in SNP-HDBK-0008 Rev. 2 and onwards, including the rhenium and TZM sections, note that an open-source code [22] was used to create polynomial curve fits for the thermal expansion property. This code allowed for trends to demonstrate zero thermal expansion at RT, which previously needed to be accounted for manually.

To develop a new trend, the data gathered through the process of Section V.3.1 is copied from Excel into MATLAB. When all datasets are approximately the same size – or when the data shows little variation when

plotted – the data can be pooled together into a single matrix in MATLAB. From there, a trendline encapsulating all the data can be created.

However, sometimes datasets vary notably and have different numbers of data points. In these cases, pooling the data would lead to some datasets having greater influence over the trend than other datasets. If all else is held equal – such as equipment being of similar quality and testing procedures being equivalent – then the datasets should have equal influence on the eventual trendline. The issue can be visualized in Figure V-12A. To avoid this issue, techniques like overfitting, underfitting, and extrapolating can be employed. See Figure V-12B for an example of mitigating the problem posed by differing amounts of data.

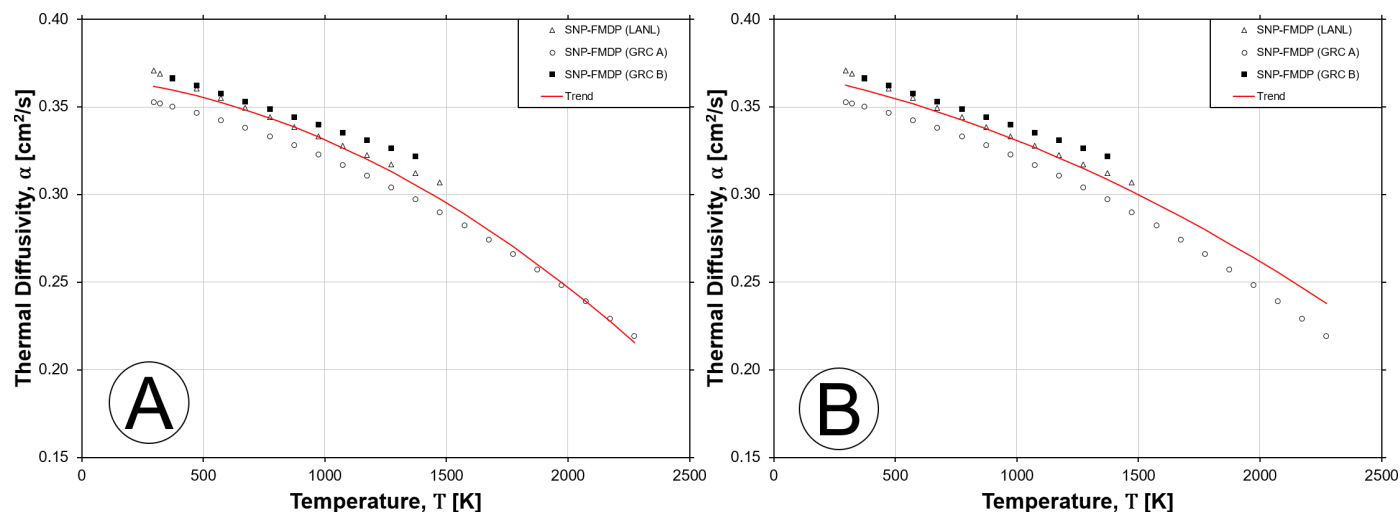


Figure V-12: Example of generating curve fits when datasets have varying numbers of data points or different maximum or minimum independent variable values. This Mo30W thermal diffusivity data and its references can be found in Section 3.1.2. In (A), the data is pooled together, while in (B), overfitting is used to produce a more realistic trend. Note that the high temperature data of SNP-FMDP (GRC A) pulls the trend downwards, while we would expect – if the other two datasets were to continue to higher temperatures – that the resulting trend would have its values slightly elevated at these temperatures.

For overfitting and underfitting, a separate trend is first created for each individual dataset. These trends are then evaluated for the same series of independent variable values – or the same set of temperatures. Previously large datasets have now been condensed into smaller datasets, whereas originally small datasets end up with more values than they started out with.

An extra level of complexity is introduced when the datasets have wildly different maximum and minimum independent variable values. If a trend is generated for pooled data, it will favor areas with greater numbers of data points – even if that causes the trend to skew away from a realistic trajectory. Again, see Figure V-12A. To show caution, the trend can be cut off early. But the SNP project requires trends that reach extremes. Thus, trends can be extrapolated to higher and lower independent variable values. Since trends exist for individual datasets, those trends may be evaluated at values outside of their preexisting maximums and minimums. Within plots, this may be referred to as either “extrapolation” or as an “extended temperature range”. By contrast, a “valid temperature range” involves no extrapolation.

When datasets exhibit consistent patterns, this turns out well. But sometimes a dataset’s individual trend will exhibit abnormalities at its extremes. If this were extrapolated and incorporated into the trendline of the material property as a whole, it could lead to an unrealistic trend shape. See Figure V-13 for an example.

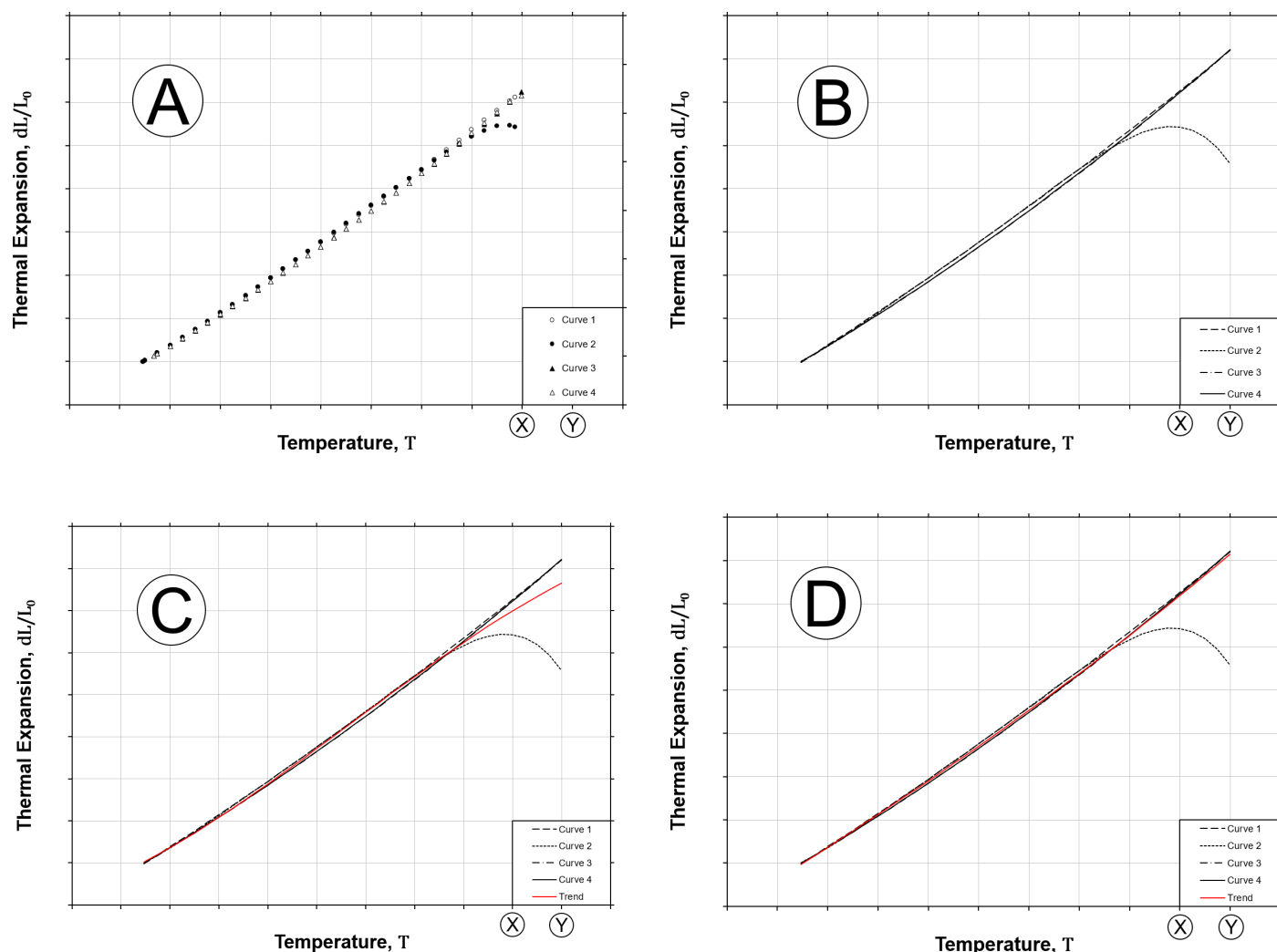


Figure V-13: Exemplifying a risk of extrapolation. (A) Four reduced datasets are shown, each of which ends near temperature X. (B) Trends are created for the individual datasets, and the trends are extrapolated to temperature Y. Note how low Curve 2 drops. (C) Using each of the individual trendlines, a generalized trend is created for the material. This new trend is lower than expected at temperature Y. (D) If the extrapolated portion of Curve 2 is removed from the trend generation, a more realistic trend is obtained.

Again, NTP conditions expose materials to a wide range of temperatures. Sometimes a single trend cannot capture the changing patterns in experimental material property data. These shifting patterns are a regular occurrence in this handbook, but phase changes in materials provide the most notable examples – please see Figure V-14. This figure illustrates the specific heat of zircaloy. Due to the material's phase change at around 1200 K, one can see a significant spike in specific heat. In situations like this one, piecewise trend equations are used in place of single trends. Reusing data close to where the piecewise trends meet allows for the piecewise equations to intersect. However, the slopes of the piecewise trends are not guaranteed to be equal at the intersection point. Caution is advised for modelers who work with models that incorporate material property trends *and* require differentiability.

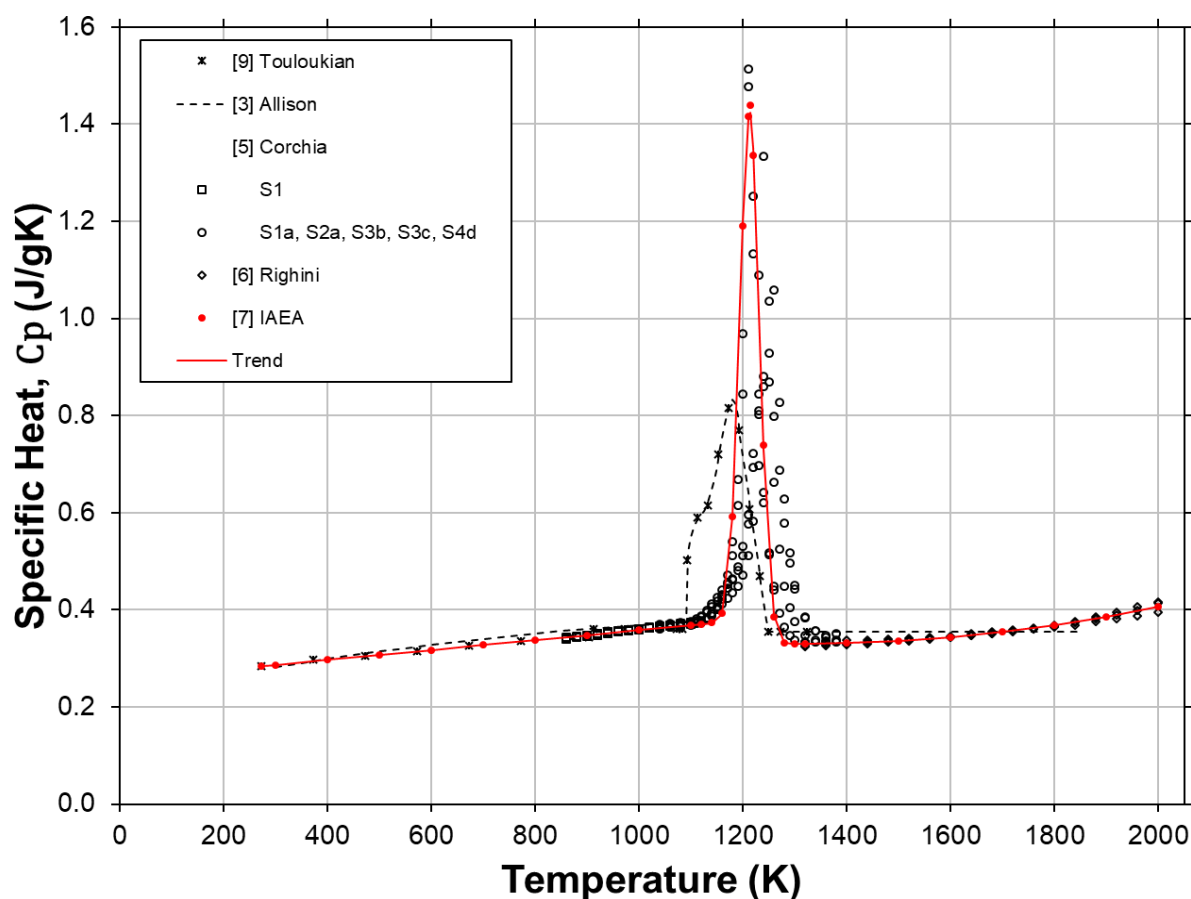


Figure V-14: Impact of a phase change on a material's properties. This plot captures specific heat data on zircaloy. See Section 4.2.1 for the data and its references. The large spike in specific heat around 1200 K is atypical, and it is due to a phase change.

Several types of curve fits are used for trend development. Since the trends are experimental rather than theoretical, there are no real-world physics or thermodynamics driving the use of one fit type over any other. MATLAB's curve fitting tool automatically calculates the sum of squared errors, an  $R^2$  value, and other statistical measures for each curve fit. To be incorporated in the handbook, a given fit must surpass the other fits tested when these statistical measures are taken into consideration.

The primary fit types used to develop this handbook are listed in Table V-3. Equations in the table serve as generic examples – actual fit equations elsewhere in the handbook have different numbers of terms. For example, some polynomial fits of material properties may be linear, quadratic, or quartic – forms that vary from the cubic shown in the table. When developing a curve fit for a material property and the different curve fit options have approximately equivalent merit based on statistical measures, the simplest of the acceptable curve fits is chosen. That is, curve fits with fewer coefficients are given preference over those with many coefficients. Similarly, curve fits that involve the basic addition and multiplication of terms are prioritized over those with complex numerator and denominator terms. It is hoped that the extra simplicity benefits the modelers and designers who translate trend equations into their own work.

Table V-3: Categories of curve fits.

| Generic Equation                                                                                                                                                              | Fit Name / Description                  |
|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-----------------------------------------|
| $y(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$                                          | Polynomial                              |
| $y(T) = \left[ A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 \right] / \left[ B0 + B1 \cdot \left(\frac{T}{1000}\right) \right]$         | Rational                                |
| $y(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3 + A_{2/} / \left(\frac{T}{1000}\right)^2$ | Shomate                                 |
| $y(T) = AN \cdot \left(\frac{T}{1000}\right)^N + A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$                                          | Power fit with a polynomial addition    |
| $y(T) = AN \cdot \left(\frac{T}{1000}\right)^N / \left[ A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 \right]$                           | Power fit divided by polynomial         |
| $y(T) = \frac{AN}{\left(\frac{T}{1000}\right)^N} + A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$                                        | Polynomial fit with a rational addition |

Most trends in this handbook represent a curve fit of average values. That is, they pass through the “center” of clusters of data points, rather than being offset to indicate maximums, minimums, or other kinds of upper and lower bounds. Fits of average values are labeled “typical basis” fits, according to MMPDS [5] – and NASA often utilizes the MMPDS method of basis designation [4]. Unless otherwise specified, assume the trend for each plot is a typical basis trend. Even when multiple trends are provided – such as “minimum” and “high” trends – readers should continue to make the typical basis assumption until explicitly told otherwise. As stated in Table 9.2.4 of MMPDS-2023, typical basis values are often suitable for thermophysical properties [5].

The design basis associated with a trend is an indicator of the trend’s reliability. Typical basis trends have no statistical significance – that is, the risk associated with using the trend is unknown. Other basis designations include S-basis, B-basis, and A-basis, listed in order of increasing reliability. NASA strongly prefers A-basis values for use in the design of “primary structures”, or components intended to sustain significant applied loads and lead to catastrophic hazards when they fail. Redundant structures, on the other hand, are often designed with B-basis values [4]. But note that both A- and B-basis trend generation require a substantial amount of data, depending on the material property of interest. According to Table 9.2.4 of MMPDS-2023, applying an A- or B-basis designation to certain mechanical properties can require hundreds of samples [5].

This handbook contains material property data that spans vast temperature regimes. With current cost and schedule constraints, acquiring A- and B-basis trends for all materials of interest for SNP applications is impractical. It is advised that future testing and literature search campaigns be used to determine A- and/or B-basis values for material properties and materials when designs or models require the heightened level of fidelity. However, even small-scale test series can be used to strengthen the accuracy of typical basis trends. Table 9.2.3 in MMPDS-2023 provides a list of industry testing standards that accounts for each of the material properties in the database [5].

### V.3.3 The Use of Models

On occasion, theoretical equations are provided alongside the experimental data and trends within property plots. These models can serve multiple functions. At times when data is limited – or the quality of datasets is unknown – models can be used to provide clarity and increase one’s confidence that experimental data or trends are valid. Alternatively, the reverse situation may occur, and a theoretical model may be of unknown reliability, while experimental data has well-defined accuracy. In these cases, comparing the experimental and theoretical trends can provide insights for those who wish to use models – or extrapolate those models to extreme temperatures, different materials, etc. – in the future.

A commonly used model methodology is the rule of mixtures (ROM). This weighted mean serves to predict material properties for composite materials and alloys based on their constituent materials.

Unless explicitly stated otherwise, designers should only use the experimental trends provided in this handbook. See Sections V.4.3, V.4.4, and V.4.5 for guidance on using the three distinct forms of data provided.

## V.4 Using the Material Property Handbook

### V.4.1 Definitions

Table V-4 defines the acronyms, abbreviations, and several technical terms used in this handbook, while Table V-5 includes a list of common symbols. Materials with their own sections in the handbook have already been defined in Section III, and they are not included in the table. However, materials referenced in the text that do not have corresponding sections in the handbook are still shown.

The symbols' definitions are also provided alongside each fit equation to aid in readers' understanding of the equations. Note that Greek letters within phase diagrams refer to different phases of materials, rather than to the definitions provided in Table V-5.

Table V-4: Acronyms, Abbreviations, and Terminology.

| Term                        | Definition                                                                                                                                                                            |
|-----------------------------|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| AM                          | Additive manufacturing <i>or</i> additively manufactured                                                                                                                              |
| As – [manufacturing method] | For example, “As – wrought”; indicates that measurements were taken on material samples in the same condition that they were manufactured in, without further treatment or processing |
| at%                         | Atomic percent                                                                                                                                                                        |
| basis (A-, B-, S-, typical) | A designation for data and trends that indicates statistical significance; NASA uses the MMPDS method of basis determination; see Section V.3.2 for details                           |
| bcc                         | Body-centered cubic crystal structure                                                                                                                                                 |
| BP                          | Boiling point                                                                                                                                                                         |
| cercer                      | Composite material consisting of ceramic particles in a ceramic matrix                                                                                                                |
| cermet                      | Composite material consisting of ceramic particles in a metallic matrix                                                                                                               |
| correlation                 | See “trend”                                                                                                                                                                           |
| CTE                         | Mean coefficient of thermal expansion <sup>a</sup>                                                                                                                                    |
| CUI                         | Controlled unclassified information                                                                                                                                                   |
| Curve [number]              | For example, “Curve 2”; used to designate between different datasets in plots; for more details on a given curve, please see its associated reference                                 |
| curve fit                   | See “trend”                                                                                                                                                                           |
| CVI                         | Chemical vapor infiltration <i>or</i> chemically vapor-infiltrated                                                                                                                    |

Table V-4: Acronyms, Abbreviations, and Terminology.

| Term                       | Definition                                                                                                                                         |
|----------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------|
| DARPA                      | Defense Advanced Research Projects Agency                                                                                                          |
| DFT                        | Density functional theory                                                                                                                          |
| DOD                        | Department of Defense                                                                                                                              |
| DOE                        | Department of Energy                                                                                                                               |
| DRACO                      | Demonstration Rocket for Agile Cislunar Operations                                                                                                 |
| EAR                        | Export Administration Regulations                                                                                                                  |
| extended temperature range | A temperature range for which a trend is supported by limited data; see “extrapolation”                                                            |
| external data              | Material property data produced by programs and projects other than the SNP project                                                                |
| extrapolation              | Method for predicting material property values outside the ranges for which data exists; see Section V.3.2 for details                             |
| fcc                        | Face-centered cubic crystal structure                                                                                                              |
| FMDP                       | Fuel and Moderator Development Plan                                                                                                                |
| FSP                        | Fission surface power                                                                                                                              |
| GRC                        | NASA John H. Glenn Research Center                                                                                                                 |
| hcp                        | Hexagonal close packed crystal structure                                                                                                           |
| high (trend)               | One of multiple trends – this trend expresses higher values than the others; it should be treated as a typical basis trend unless otherwise stated |
| HNLS                       | Hi-Nicalon Type S SiC fibers by Nippon Carbon Co., Tokyo, Japan                                                                                    |
| INL                        | Idaho National Laboratory                                                                                                                          |
| internal data              | Material property data produced by the SNP project                                                                                                 |
| ITAR                       | International Traffic in Arms Regulations                                                                                                          |
| LANL                       | Los Alamos National Laboratory                                                                                                                     |
| LaRC                       | NASA Langley Research Center                                                                                                                       |
| MAPTIS                     | Materials and Processes Technical Information Center                                                                                               |
| MATLAB                     | Matrix Laboratory; a programming language by MathWorks                                                                                             |
| MCWG                       | Materials Characterization Working Group                                                                                                           |
| minimum (trend)            | One of multiple trends – this trend expresses lower values than the others; it should be treated as a typical basis trend unless otherwise stated  |
| MMPDS                      | Metallic Materials Properties Development and Standardization                                                                                      |
| mol%                       | Mole percent, equivalent to at%                                                                                                                    |
| MP                         | Melting point                                                                                                                                      |
| MSFC                       | NASA George C. Marshall Space Flight Center                                                                                                        |

Table V-4: Acronyms, Abbreviations, and Terminology.

| Term                    | Definition                                                                                                                                                                                                                                              |
|-------------------------|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| MST                     | Missouri University of Science and Technology                                                                                                                                                                                                           |
| MW                      | Molecular weight                                                                                                                                                                                                                                        |
| NASA                    | The National Aeronautics and Space Administration                                                                                                                                                                                                       |
| NEP                     | Nuclear electric propulsion                                                                                                                                                                                                                             |
| NERVA                   | Nuclear Engine for Rocket Vehicle Application                                                                                                                                                                                                           |
| NTP                     | Nuclear thermal propulsion                                                                                                                                                                                                                              |
| ORNL                    | Oak Ridge National Laboratory                                                                                                                                                                                                                           |
| PIP                     | Polymer impregnation and pyrolysis                                                                                                                                                                                                                      |
| PLS                     | Proportional limit stress                                                                                                                                                                                                                               |
| PyC                     | Pyrolytic carbon                                                                                                                                                                                                                                        |
| ROM                     | Rule of mixtures; a model for predicting material property values; see Section V.3.3 for details                                                                                                                                                        |
| RT                      | Room temperature                                                                                                                                                                                                                                        |
| SA3                     | Tyranno 3 SiC fibers by Ube Industries, Ltd., Ube, Japan                                                                                                                                                                                                |
| SI units                | International System of Units                                                                                                                                                                                                                           |
| SNP                     | Space nuclear propulsion                                                                                                                                                                                                                                |
| SNPP DB                 | Space Nuclear Power and Propulsion Materials Database                                                                                                                                                                                                   |
| SPS                     | Spark plasma sintering <i>or</i> spark plasma sintered                                                                                                                                                                                                  |
| SSC                     | Solid solution carbide                                                                                                                                                                                                                                  |
| TD                      | Theoretical density; materials with zero porosity                                                                                                                                                                                                       |
| temp.                   | Temperature                                                                                                                                                                                                                                             |
| trend                   | The general course of a material's given material property data; the values composing the trend should be used by modelers and designers; see V.3.2 for background on how trends are formed and V.4.3 to V.4.5 for a discussion on making use of trends |
| trendline               | The red line in a material property plot that represents the trend of experimental data                                                                                                                                                                 |
| valid temperature range | A temperature range for which a trend is supported by sufficient data; see "extrapolation"                                                                                                                                                              |
| vol%                    | Percent by volume                                                                                                                                                                                                                                       |
| w/o                     | see "wt%"                                                                                                                                                                                                                                               |
| wt%                     | Percent by weight                                                                                                                                                                                                                                       |

<sup>a</sup> Also known as the average coefficient of thermal expansion or the linear expansion coefficient

Table V-5: Symbols.

| Symbol         | Definition                                                                                                                                          |
|----------------|-----------------------------------------------------------------------------------------------------------------------------------------------------|
| $\rho$         | Density – assume TD unless otherwise stated                                                                                                         |
| $\rho_{bulk}$  | Bulk density, or the density of real-world, porous material                                                                                         |
| $\rho_e$       | Electrical resistivity                                                                                                                              |
| $P$            | Fractional porosity                                                                                                                                 |
| $K_{ic}$       | Fracture toughness                                                                                                                                  |
| $dL/L_0$       | Linear thermal expansion                                                                                                                            |
| $\bar{\alpha}$ | Mean coefficient of thermal expansion <sup>a</sup>                                                                                                  |
| $M$            | Modulus of rupture                                                                                                                                  |
| $\nu$          | Poisson's ratio                                                                                                                                     |
| $x$            | Ratio of one material to another material                                                                                                           |
| $T_{ref}$      | Reference temperature                                                                                                                               |
| $\rho_{RT}$    | Room temperature density                                                                                                                            |
| $G$            | Shear modulus                                                                                                                                       |
| $C_p$          | Specific heat                                                                                                                                       |
| $T$            | Temperature                                                                                                                                         |
| $k$            | Thermal conductivity for perfectly dense material                                                                                                   |
| $\lambda$      | See “ $k$ ”                                                                                                                                         |
| $k_p$          | Thermal conductivity for porous material; an equation relating $k$ and $k_p$ is often provided for readers who wish to perform porosity conversions |
| $\alpha$       | Thermal diffusivity                                                                                                                                 |
| $\delta$       | See “ $\alpha$ ”                                                                                                                                    |
| $UTS$          | Ultimate tensile strength                                                                                                                           |
| $TS$           | See “ $UTS$ ”                                                                                                                                       |
| $YS$           | Yield strength                                                                                                                                      |
| $E$            | Young's modulus <sup>b</sup>                                                                                                                        |

<sup>a</sup> Also known as the average coefficient of thermal expansion or the linear expansion coefficient

<sup>b</sup> Also known as the modulus of elasticity

### V.4.2 Navigation

Following this introductory section of the handbook, readers will find sections on materials and their properties. A tiered structure is used to categorize materials, and the organization resembles the format used by the NERVA Program in their own Materials Properties Data Book [7]. The Table of Contents provides a good

visualization of the high-level structure, while Figure V-15 expands the breakdown to show the individual subsections within a given material's section.

Given the size of the handbook, readers are encouraged to use built-in mechanisms to quickly navigate the document. First, the Table of Contents and the Property List are composed of links; in the Portable Document Format (PDF) version of the handbook, one can click on the links to jump to sections and subsections of interest.

However, readers may need to jump between areas of the handbook – and constantly returning to the Table of Contents takes time. Therefore, making use of Word's Navigation Pane, Adobe Reader's (or Acrobat's) Bookmarks side panel, or browser-based equivalents (i.e., the document outline in Google Chrome) is highly recommended. In each case, a sidebar appears that allows readers to switch between the different materials and material properties at will.

Each material section begins with room temperature and single-value properties, as well as any applicable composition or phase information. From there, material properties are each typically given two sheets each to relay data through plots, tables, and equations. Additional application notes sheets follow the material properties in cases where paragraph-style explanations would assist readers in making use of the data. Next comes a summary table of the most common thermophysical properties within the handbook. Finally, the material section ends with a reference list. All data sources used to develop the handbook section can be found within the list.

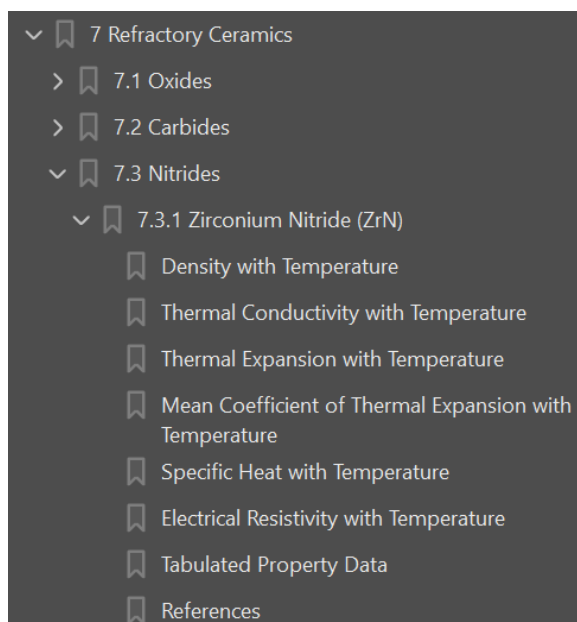


Figure V-15: Material section breakdown for zirconium nitride. Most subsections correspond with material properties, but the Tabulated Property Data and References subsections are exceptions.

### V.4.3 How to Use Material Property Plots

Within each material's section, a set of material property trends are provided. For each property, a plot of trendlines and data points is first to appear. An example can be seen in Figure V-16. Individual data points are indicated by black data markers and are either hollow or filled. The dashed line from reference [1] (of Section 2.1.1) indicates a trend, though since it is black, it was not generated by the SNP Material Property Handbook team. Those trends appear red instead. Sometimes data or trends from references will be shown in a plot but were not used in trend formulation. Theoretical models are occasionally provided, and these likewise do not contribute to experimental, red trends. The application notes for a property, which are almost always found on

the page following the plot, provide clarity on how a trend was generated. Additionally, the reference numbers listed in each plot, plot caption, and application note correspond with the sources listed in the References subsection at the end of each material's section. Figure V-15 in Section V.4.2 is a screenshot showing the generic breakdown of a material section.

Modelers and designers are advised to glance over the plots of relevance to their work before extracting information from the handbook. Visualizing the data in this way can serve as a helpful check – both that the reader has interpreted the information correctly, and that the handbook author did not err when adding data into the handbook. Plots are also useful for providing quick, rough estimates of property values at key temperatures of interest.

Readers are not advised to take data from the plots directly, though it is possible. Web plot digitizer [23] exemplifies an online tool that enables the conversion of trendlines and data markers into a series of tabulated values. Please obtain approval to use external software from the appropriate authority at your organization before utilizing browser-based tools like these. Each page following a property plot contains tabulated data and a fit equation – and these two tools are superior methods for incorporating material property values into models and designs than using the material property plots directly.

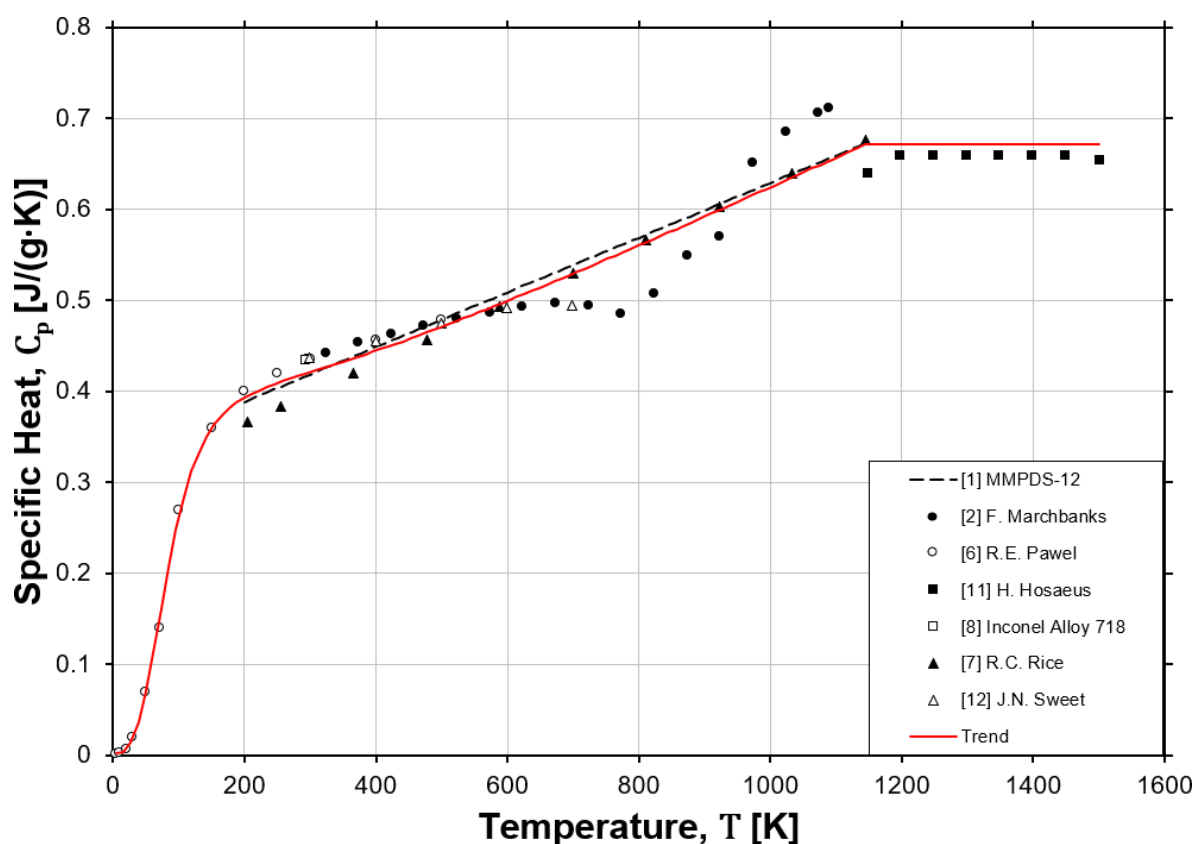


Figure V-16: Example of a material property plot. This plot captures specific heat data on IN718 from references [2, 6-8, 11, 12] of Section 2.1.1 as black data points, shows a red trendline of the data, and compares that trend with a trend found in reference [1] of Section 2.1.1.

## V.4.4 How to Use Material Property, Room Temperature, Composition, and Summary Tables

Tables are used in several ways within the handbook. The primary use is as a material property table, or the tabular counterpart of the red trends provided in material property plots. Table V-6 is one such table. These tables serve to capture the values of a property trend at different intervals. The tables typically begin at the trend's minimum defined temperature and extend up to the maximum temperature. When using these tables,

note that the intervals between temperature values are not uniform, particularly near the extremes of the trend. For that reason, copying and pasting data from these tables is recommended. Double checking that values have been transferred correctly is also useful for verification.

Temperatures are listed in both Kelvin and degrees Fahrenheit within these tables. Likewise, both metric and English units are provided for the corresponding material property. To aid in differentiating between the unit systems, English units are always surrounded by parentheses.

Table V-6: Example of a material property table. This table contains values for the beryllium electrical resistivity trend with temperature.

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                           | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                           |
|-------------------|------------|-------------------------------------|---------------------------|-------------------|------------|-------------------------------------|---------------------------|
| K                 | ( °F )     | $\mu\Omega\text{-m}$                | ( $\mu\Omega\text{-in}$ ) | K                 | ( °F )     | $\mu\Omega\text{-m}$                | ( $\mu\Omega\text{-in}$ ) |
| 79                | ( -317.5 ) | 0.010                               | ( 0.39 )                  | 800               | ( 980.3 )  | 0.190                               | ( 7.48 )                  |
| 100               | ( -279.7 ) | 0.012                               | ( 0.47 )                  | 850               | ( 1070.3 ) | 0.207                               | ( 8.14 )                  |
| 150               | ( -189.7 ) | 0.018                               | ( 0.70 )                  | 900               | ( 1160.3 ) | 0.224                               | ( 8.82 )                  |
| 200               | ( -99.7 )  | 0.026                               | ( 1.02 )                  | 950               | ( 1250.3 ) | 0.242                               | ( 9.53 )                  |
| 250               | ( -9.7 )   | 0.035                               | ( 1.39 )                  | 1000              | ( 1340.3 ) | 0.261                               | ( 10.28 )                 |
| 300               | ( 80.3 )   | 0.046                               | ( 1.82 )                  | 1050              | ( 1430.3 ) | 0.281                               | ( 11.08 )                 |
| 350               | ( 170.3 )  | 0.058                               | ( 2.29 )                  | 1100              | ( 1520.3 ) | 0.303                               | ( 11.94 )                 |
| 400               | ( 260.3 )  | 0.071                               | ( 2.79 )                  | 1150              | ( 1610.3 ) | 0.327                               | ( 12.87 )                 |
| 450               | ( 350.3 )  | 0.084                               | ( 3.32 )                  | 1200              | ( 1700.3 ) | 0.353                               | ( 13.89 )                 |
| 500               | ( 440.3 )  | 0.098                               | ( 3.88 )                  | 1250              | ( 1790.3 ) | 0.381                               | ( 15.01 )                 |
| 550               | ( 530.3 )  | 0.113                               | ( 4.44 )                  | 1300              | ( 1880.3 ) | 0.413                               | ( 16.25 )                 |
| 600               | ( 620.3 )  | 0.128                               | ( 5.03 )                  | 1329              | ( 1932.5 ) | 0.433                               | ( 17.03 )                 |
| 650               | ( 710.3 )  | 0.143                               | ( 5.62 )                  |                   |            |                                     |                           |
| 700               | ( 800.3 )  | 0.158                               | ( 6.23 )                  |                   |            |                                     |                           |
| 750               | ( 890.3 )  | 0.174                               | ( 6.85 )                  |                   |            |                                     |                           |

Another commonly used table type is the room temperature properties table, as illustrated by Table V-7. As stated in the table's name, these tables contain data at RT. For the purposes of this handbook, RT corresponds with temperatures from 288 to 298 K, or 59 to 77 °F. Metric units are used in this table type, and International System of Units (SI units) are preferred.

Property values at RT are often derived directly from the material property trends stemming from other subsections in the handbook. When property values at RT cannot be provided, values may be taken at different temperatures, and notes are used to call out the discrepancy. Alternatively, values at RT can be found by extrapolating material property trends. But see Section V.3.2 to learn more about the dangers of extrapolation.

## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

Table V-7: Example of a room temperature properties table. This table contains single-point values for several properties of zirconium carbide. Note the units in square brackets and the melting point tolerance.

|                                           |         |
|-------------------------------------------|---------|
| Molar Mass, [g/mol]                       | 103.23  |
| Theoretical Density, [kg/m <sup>3</sup> ] | 6,730   |
| Melting Point, [K]                        | 3693±30 |
| Specific Heat, [J/(g-K)]                  | 0.363   |
| Thermal Conductivity, [W/(m-K)]           | 26.3    |
| Linear expansion coefficient, [μm/(m-K)]  | 5.10    |
| Electrical resistivity, [μΩ-m]            | 0.61    |
| Young's Modulus, [GPa]                    | 397.9   |
| Shear Modulus, [GPa]                      | 161.8   |
| Poisson's Ratio, [-]                      | 0.229   |

Not all properties in the RT properties table are truly “room temperature properties”. For example, a material’s melting point is not taken at RT – it is evaluated at the temperature for which a solid material experiences a phase change to become liquid. Regardless, these single-value properties are still presented in these RT properties tables for ease of organization.

Following the RT properties tables are composition tables, such as Table V-8. These tables show the elemental constituents within a material. Values are provided in wt% unless otherwise specified. Minimums and maximums compositions for each element are regularly listed. Dashes reflect compositions of zero wt%, while the word “balance” reveals the element making up the remainder of the material once all the other elemental composition percentages are considered. When different grades – or designations – of materials are under consideration, composition tables can reflect the varying compositions. Unless stated otherwise, the material property values in the rest of a material section correspond with the composition(s) provided in the composition table.

Table V-8: Example composition table. Here, the elemental breakdown for SS304 is provided in wt%. Dashes are used to indicate zero or unavailable compositions.

| Grade       |      | Fe      | Cr   | Ni   | C    | Mn  | Si   | P     | S     | N    |
|-------------|------|---------|------|------|------|-----|------|-------|-------|------|
| <b>304</b>  | Min. | Balance | 18.0 | 8.0  | -    | -   | -    | -     | -     | -    |
|             | Max. |         | 20.0 | 10.5 | 0.08 | 2.0 | 0.75 | 0.045 | 0.030 | 0.10 |
| <b>304L</b> | Min. | Balance | 18.0 | 8.0  | -    | -   | -    | -     | -     | -    |
|             | Max. |         | 20.0 | 12   | 0.03 | 2.0 | 0.75 | 0.045 | 0.030 | 0.10 |
| <b>304H</b> | Min. | Balance | 18.0 | 8.0  | 0.04 | -   | -    | -     | -     | -    |
|             | Max. |         | 20.0 | 10.5 | 0.10 | 2.0 | 0.75 | 0.045 | 0.030 | -    |

At the end of each material section, immediately before the References subsection, comes the Tabulated Property Data subsection. These subsections consist entirely of material property summary tables. See Table V-9 for an example. Much like the material property tables, these summary tables are founded on the red trends of various material property plots. The unique aspect of these summary tables is the fact that material property values for several different properties are collected in a single location. Those who use the handbook regularly and require rapid access to data on several properties for a material may find these tables useful.

# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

Additionally, since the summary tables are built from trends, the tables can also be used to determine the range of temperatures for which a given trend exists.

Table V-9: Example of a material property summary table. This example corresponds with the property values for TZM. Note that the temperatures in the leftmost column correspond with the property values in all the other columns.

| Temp.<br>(T)<br>K | Physical and Electrical                    |                                                                   | Thermal                                   |                                           |                                                          |                                          | Elastic                          |                                |                                 |
|-------------------|--------------------------------------------|-------------------------------------------------------------------|-------------------------------------------|-------------------------------------------|----------------------------------------------------------|------------------------------------------|----------------------------------|--------------------------------|---------------------------------|
|                   | Density<br>( $\rho$ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( $\rho_e$ )<br>$\mu\Omega\text{-m}$ | Thermal<br>Conductivity<br>(k)<br>W/(m-K) | Thermal<br>Expansion<br>( $dL/L_0$ )<br>% | Mean<br>CTE<br>( $\bar{\alpha}$ )<br>10 <sup>-6</sup> /K | Specific<br>Heat<br>( $C_p$ )<br>J/(g-K) | Young's<br>Modulus<br>(E)<br>GPa | Shear<br>Modulus<br>(G)<br>GPa | Poisson's<br>Ratio<br>( $\nu$ ) |
| 200               | -                                          | -                                                                 | -                                         | -                                         | -                                                        | -                                        | 314.04                           | -                              | -                               |
| 293               | 10220                                      | 0.079                                                             | 123.22                                    | 0.000                                     | 4.97                                                     | 0.269                                    | 306.31                           | 122.7                          | 0.293                           |
| 300               | 10219                                      | 0.081                                                             | 122.94                                    | 0.004                                     | 4.99                                                     | 0.268                                    | 305.71                           | -                              | -                               |
| 400               | 10201                                      | 0.106                                                             | 119.40                                    | 0.062                                     | 5.18                                                     | 0.265                                    | 297.09                           | -                              | -                               |
| 500               | 10184                                      | 0.132                                                             | 116.55                                    | 0.119                                     | 5.32                                                     | 0.265                                    | 288.17                           | -                              | -                               |
| 600               | 10166                                      | 0.157                                                             | 114.20                                    | 0.176                                     | 5.43                                                     | 0.267                                    | 278.95                           | -                              | -                               |
| 700               | 10149                                      | 0.182                                                             | 112.22                                    | 0.233                                     | 5.51                                                     | 0.269                                    | 269.43                           | -                              | -                               |
| 800               | 10131                                      | 0.207                                                             | 110.46                                    | 0.291                                     | 5.58                                                     | 0.272                                    | 259.62                           | -                              | -                               |
| 900               | 10114                                      | 0.233                                                             | 108.80                                    | 0.349                                     | 5.65                                                     | 0.276                                    | 249.51                           | -                              | -                               |
| 1000              | 10096                                      | 0.258                                                             | 107.16                                    | 0.409                                     | 5.72                                                     | 0.281                                    | 239.10                           | -                              | -                               |
| 1100              | 10077                                      | 0.283                                                             | 105.44                                    | 0.470                                     | 5.79                                                     | 0.287                                    | 228.39                           | -                              | -                               |
| 1200              | 10058                                      | -                                                                 | 103.57                                    | 0.534                                     | 5.86                                                     | 0.293                                    | 217.39                           | -                              | -                               |
| 1300              | 10038                                      | -                                                                 | 101.52                                    | 0.599                                     | 5.95                                                     | 0.301                                    | 206.09                           | -                              | -                               |
| 1400              | 10018                                      | -                                                                 | 99.25                                     | 0.668                                     | 6.03                                                     | 0.309                                    | 194.49                           | -                              | -                               |
| 1500              | 9997                                       | -                                                                 | 96.74                                     | 0.739                                     | 6.13                                                     | 0.317                                    | 182.60                           | -                              | -                               |
| 1600              | 9975                                       | -                                                                 | 94.00                                     | 0.814                                     | 6.24                                                     | 0.327                                    | 170.40                           | -                              | -                               |
| 1700              | 9951                                       | -                                                                 | 91.06                                     | 0.892                                     | 6.36                                                     | 0.337                                    | 157.91                           | -                              | -                               |
| 1800              | 9927                                       | -                                                                 | 87.94                                     | 0.975                                     | 6.49                                                     | 0.348                                    | 145.13                           | -                              | -                               |
| 1900              | 9901                                       | -                                                                 | 84.71                                     | 1.062                                     | 6.63                                                     | 0.360                                    | 132.04                           | -                              | -                               |
| 2000              | 9874                                       | -                                                                 | 81.43                                     | 1.155                                     | 6.78                                                     | 0.372                                    | 118.66                           | -                              | -                               |
| 2100              | 9846                                       | -                                                                 | 78.20                                     | 1.252                                     | 6.94                                                     | 0.386                                    | 104.98                           | -                              | -                               |
| 2200              | 9816                                       | -                                                                 | 75.13                                     | 1.355                                     | 7.11                                                     | 0.399                                    | 91.00                            | -                              | -                               |
| 2300              | 9784                                       | -                                                                 | 72.34                                     | 1.464                                     | 7.30                                                     | 0.414                                    | 76.73                            | -                              | -                               |
| 2500              | 9716                                       | -                                                                 | 68.19                                     | 1.702                                     | 7.71                                                     | 0.446                                    | 47.29                            | -                              | -                               |
| 2700              | 9640                                       | -                                                                 | -                                         | 1.968                                     | 8.16                                                     | 0.480                                    | -                                | -                              | -                               |

## V.4.5 How to Use Fit Equations

Fit equations are equivalent to the red trends of material property plots – but they take the form of mathematical expressions. Various symbols are used in the equations. Symbols representing material properties are defined in Table V-5 of Section V.4.1, and they are often functions of temperature, as exemplified in Figure V-17 below. Most of the remaining symbols in fit equations are constants. The symbols for these constants are composed of a letter at a minimum, though they often include a number as well, as in the case of “A0”, “B\_2”, etc. At present, the symbol composition is multifaceted. Sometimes the letters of coefficients are used to differentiate between multiple piecewise equations, while in other situations, the letters might distinguish between the numerator and denominator of a single equation. Readers are advised to pay close attention to the full expression provided in the upper left corner of each fit equation listing.

Below the mathematical expression of the fit equation, brief definitions of the independent and dependent variables are given. For example, in Figure V-17,  $T$  corresponds with temperature, the independent variable, and  $G(T)$  is shear modulus, the dependent variable. Units are provided for these variables, and fit equations *must* be used with matching units. Various conversions are required to use the fit equations with units they were not built around.

Ranges of the independent variables for which the fit equation can be used are also listed, though these ranges can be stated in either the left or right half of the fit equation figure, as space allows. Using the expression outside of such a range may be permissible, though the act is an extrapolation, and it should be documented as such.

Constants are defined in the right half of the fit equation figure. The units associated with the constants vary greatly and are not necessary for modelers and designers – for that reason, they are not included. Please contact the SNP Material Property Handbook team if the units for fit equation constants are desired and cannot be calculated by the reader. When multiple trends are used to characterize a material property, coefficients can be given multiple values to define the different equations. For example, in Figure V-17, coefficients are given different values depending on whether the reader wishes to use the “minimum” or the “high” trend for tungsten’s shear modulus. In either case, the reader only needs to take the appropriate values from the right half of the fit equation figure and substitute them into the mathematical expression of the upper left part of the fit equation figure. While Figure V-17 applies for two distinct curves with the same temperature range, the process for using fit equations is identical for scenarios involving piecewise equations with multiple temperature ranges.

Note that temperatures are regularly divided by a thousand, as shown in Figure V-17. This adjustment is a form of scaling that helps manage the magnitudes of constants. In addition, this scaling aids MATLAB – the curve fitting software described in Section V.3.2 – in avoiding rounding errors [24]. Within MATLAB’s curve fitting tool, one can detect the threat of rounding errors by the presence of an “equation is badly conditioned” warning.

| Fit Equations:                                                                                                                       | Constants: |                |             |
|--------------------------------------------------------------------------------------------------------------------------------------|------------|----------------|-------------|
| $G(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$ | Curve:     | <u>Minimum</u> | <u>High</u> |
| $G(T) = \text{Shear Modulus [GPa]}$                                                                                                  | A0 =       | 158.4          | 163.5       |
| $T = \text{Temperature [K]}$                                                                                                         | A1 =       | -16.99         | -11.98      |
| T Range [K]: $273 \leq T \leq 2873$                                                                                                  | A2 =       | 6.101          | -3.414      |
|                                                                                                                                      | A3 =       | -3.92          | 0           |

Figure V-17: Example of a fit equation, including the definition of constants. This equation applies to the shear modulus of tungsten.

## VI FRONT MATTER REFERENCES

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## **1 Light Metal Alloys**

### **1.1 Aluminum Alloys**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075 (Al7075)

Revision 0: 08-05-2020

General

### Room Temperature Properties

|                                          |            |
|------------------------------------------|------------|
| Density, [kg/m <sup>3</sup> ]            | 2,800      |
| Melting Point, [K]                       | 745 to 910 |
| Specific Heat, [J/(g-K)]                 | 0.845      |
| Thermal Conductivity, [W/(m-K)]          | 144.7      |
| Linear expansion coefficient, [μm/(m-K)] | 22.44      |
| Electrical resistivity, [μΩ-m]           | 0.05       |
| Young's Modulus, [GPa]                   | 69.4       |
| Shear Modulus, [GPa]                     | 26.1       |
| Poisson's Ratio, [-]                     | 0.330      |

### Composition

Table 1.1.1-1: Typical Composition ranges for Aluminum 7075 alloy (percent by weight).

| Grade  |      | Al   | Zn  | Cr   | Cu  | Fe   | Mg  | Mn   | Si  | Ti  | Other |
|--------|------|------|-----|------|-----|------|-----|------|-----|-----|-------|
| Al7075 | Min. | 87.1 | 5.1 | 0.18 | 1.2 |      | 2.1 |      |     |     |       |
|        | Max. | 91.4 | 6.1 | 0.28 | 2.0 | 0.50 | 2.9 | 0.30 | 0.4 | 0.2 | 0.15  |



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 2: 04-26-2023

Density with Temperature

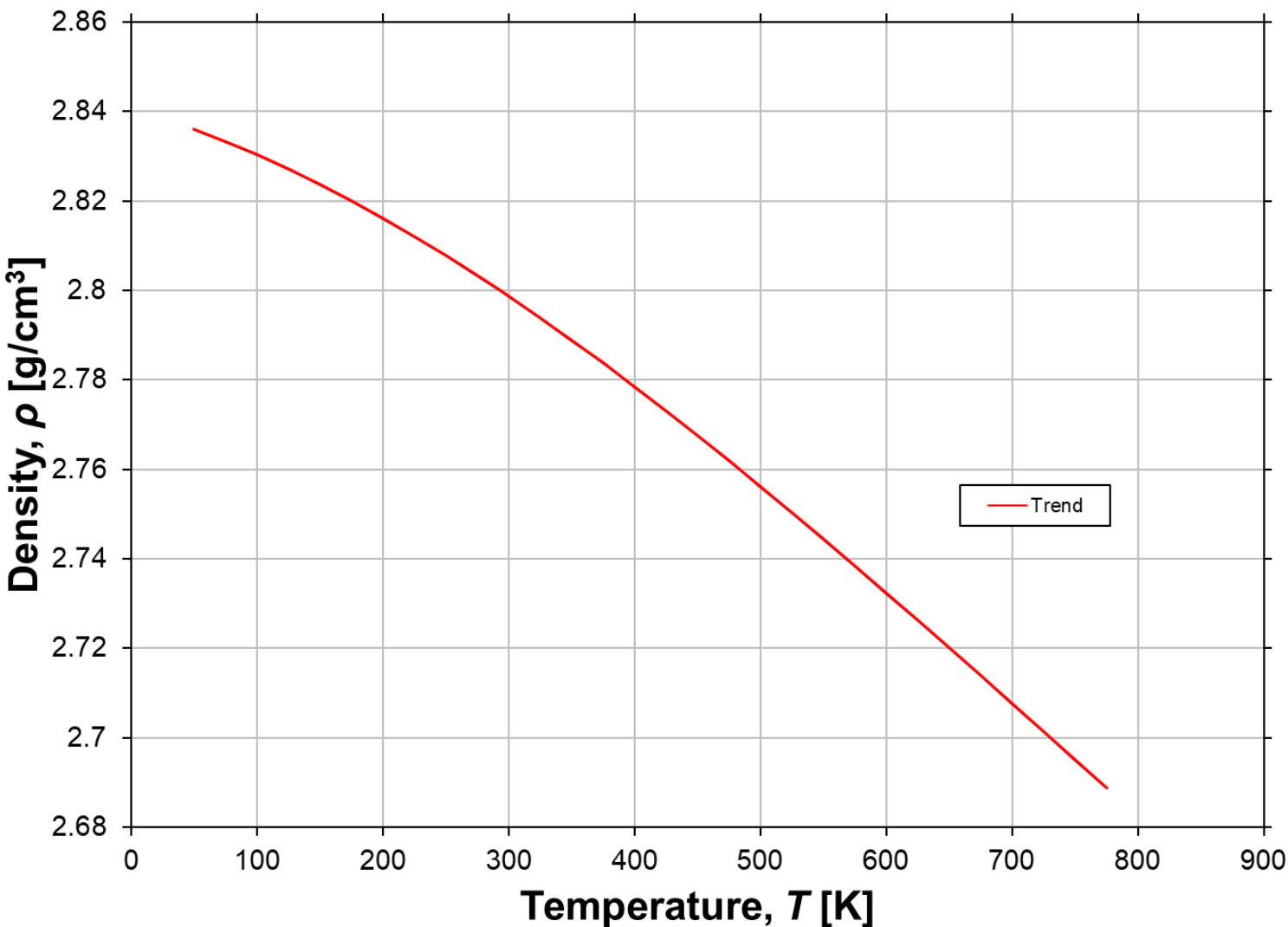


Figure 1.1.1-1: Density versus Temperature for Aluminum 7075. Calculated from fitted trend of Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Density with Temperature

## 100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |           | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|-----------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )    | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 50                | ( -369.7 ) | 2836              | ( 177.1 )              | 425               | ( 305.3 ) | 2773              | ( 173.1 )              |
| 75                | ( -324.7 ) | 2833              | ( 176.9 )              | 450               | ( 350.3 ) | 2768              | ( 172.8 )              |
| 100               | ( -279.7 ) | 2830              | ( 176.7 )              | 475               | ( 395.3 ) | 2762              | ( 172.4 )              |
| 125               | ( -234.7 ) | 2827              | ( 176.5 )              | 500               | ( 440.3 ) | 2756              | ( 172.1 )              |
| 150               | ( -189.7 ) | 2824              | ( 176.3 )              | 525               | ( 485.3 ) | 2750              | ( 171.7 )              |
| 175               | ( -144.7 ) | 2820              | ( 176.1 )              | 550               | ( 530.3 ) | 2744              | ( 171.3 )              |
| 200               | ( -99.7 )  | 2816              | ( 175.8 )              | 575               | ( 575.3 ) | 2738              | ( 171.0 )              |
| 225               | ( -54.7 )  | 2812              | ( 175.6 )              | 600               | ( 620.3 ) | 2732              | ( 170.6 )              |
| 250               | ( -9.7 )   | 2808              | ( 175.3 )              | 625               | ( 665.3 ) | 2726              | ( 170.2 )              |
| 275               | ( 35.3 )   | 2803              | ( 175.0 )              | 650               | ( 710.3 ) | 2720              | ( 169.8 )              |
| 300               | ( 80.3 )   | 2799              | ( 174.7 )              | 675               | ( 755.3 ) | 2714              | ( 169.4 )              |
| 325               | ( 125.3 )  | 2794              | ( 174.4 )              | 700               | ( 800.3 ) | 2708              | ( 169.0 )              |
| 350               | ( 170.3 )  | 2789              | ( 174.1 )              | 725               | ( 845.3 ) | 2701              | ( 168.6 )              |
| 375               | ( 215.3 )  | 2784              | ( 173.8 )              | 750               | ( 890.3 ) | 2695              | ( 168.3 )              |
| 400               | ( 260.3 )  | 2778              | ( 173.5 )              | 775               | ( 935.3 ) | 2689              | ( 167.9 )              |

**Application Notes:** Density is calculated as a function of thermal expansion as seen in the equation below to approximate property trend with respect to temperature.

### Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 2800 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $50 \leq T \leq 775$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

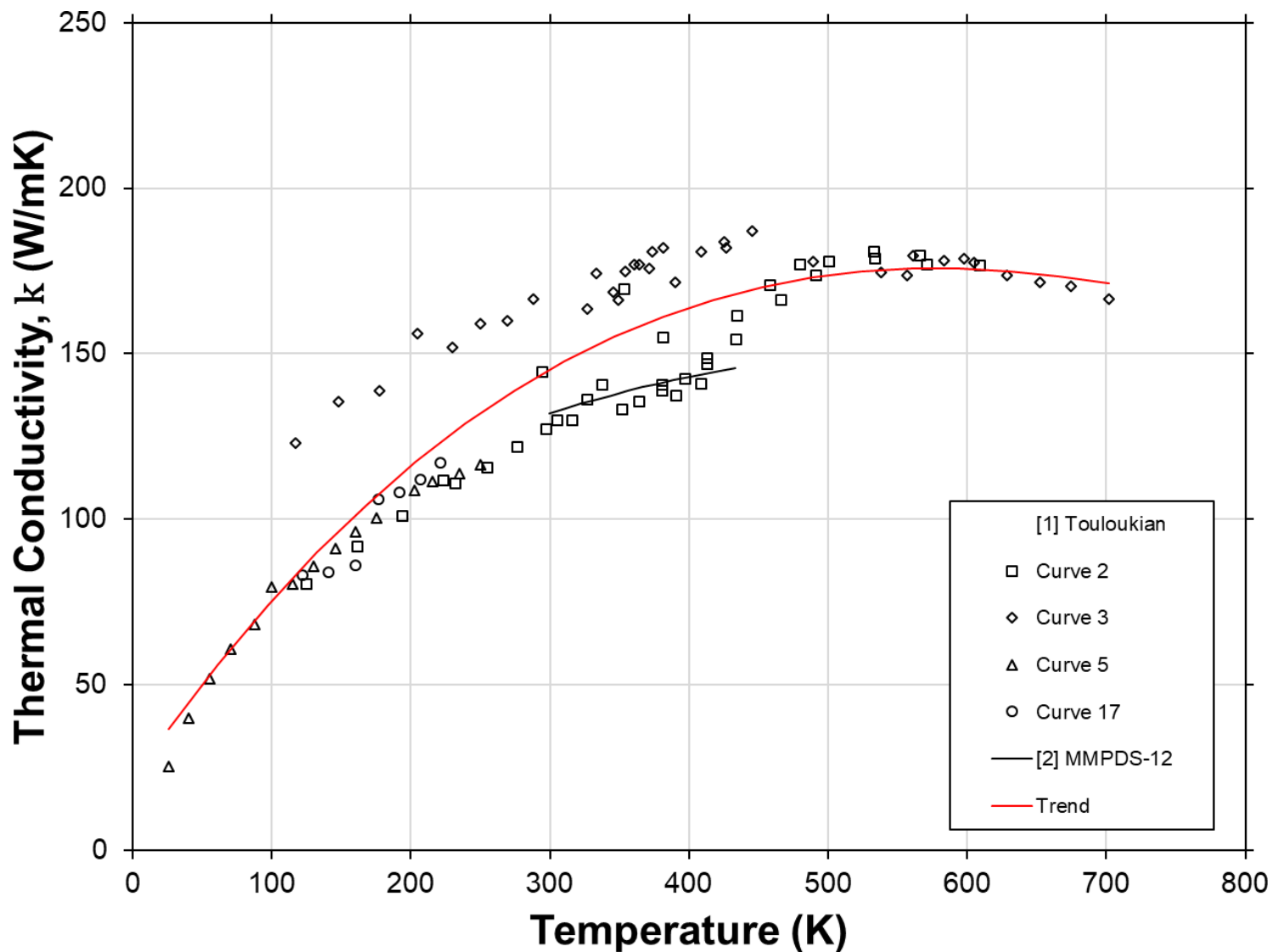


Figure 1.1.1-2: Thermal Conductivity versus Temperature for Aluminum 7075, compared to trend from MMPDS-12, Al 7075, T-6, A Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |           | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|-----------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )    | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 25                | ( -414.7 ) | 36.17                      | ( 250.93 )                          | 375               | ( 215.3 ) | 160.17                     | ( 1111.31 )                         |
| 50                | ( -369.7 ) | 50.03                      | ( 347.12 )                          | 400               | ( 260.3 ) | 163.97                     | ( 1137.65 )                         |
| 75                | ( -324.7 ) | 63.04                      | ( 437.40 )                          | 425               | ( 305.3 ) | 167.20                     | ( 1160.06 )                         |
| 100               | ( -279.7 ) | 75.22                      | ( 521.91 )                          | 450               | ( 350.3 ) | 169.89                     | ( 1178.70 )                         |
| 125               | ( -234.7 ) | 86.59                      | ( 600.80 )                          | 475               | ( 395.3 ) | 172.05                     | ( 1193.69 )                         |
| 150               | ( -189.7 ) | 97.17                      | ( 674.19 )                          | 500               | ( 440.3 ) | 173.71                     | ( 1205.18 )                         |
| 175               | ( -144.7 ) | 106.98                     | ( 742.25 )                          | 525               | ( 485.3 ) | 174.88                     | ( 1213.32 )                         |
| 200               | ( -99.7 )  | 116.04                     | ( 805.10 )                          | 550               | ( 530.3 ) | 175.59                     | ( 1218.24 )                         |
| 225               | ( -54.7 )  | 124.37                     | ( 862.89 )                          | 575               | ( 575.3 ) | 175.85                     | ( 1220.09 )                         |
| 250               | ( -9.7 )   | 131.99                     | ( 915.76 )                          | 600               | ( 620.3 ) | 175.70                     | ( 1219.01 )                         |
| 275               | ( 35.3 )   | 138.92                     | ( 963.85 )                          | 625               | ( 665.3 ) | 175.14                     | ( 1215.13 )                         |
| 300               | ( 80.3 )   | 145.19                     | ( 1007.31 )                         | 650               | ( 710.3 ) | 174.20                     | ( 1208.61 )                         |
| 325               | ( 125.3 )  | 150.80                     | ( 1046.28 )                         | 675               | ( 755.3 ) | 172.90                     | ( 1199.59 )                         |
| 350               | ( 170.3 )  | 155.79                     | ( 1080.90 )                         | 702               | ( 803.9 ) | 171.11                     | ( 1187.19 )                         |

**Application Notes:** Data for thermal conductivity is collected from reference [1], and compared against data trend from reference [2]. Data is fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$k(T) = \text{Thermal Conductivity } [W/(m \cdot K)]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 25 \leq T \leq 702$$

$$A0 = 21.43$$

$$A1 = 607.2$$

$$A2 = -714.4$$

$$A3 = 218.2$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Thermal Expansion with Temperature

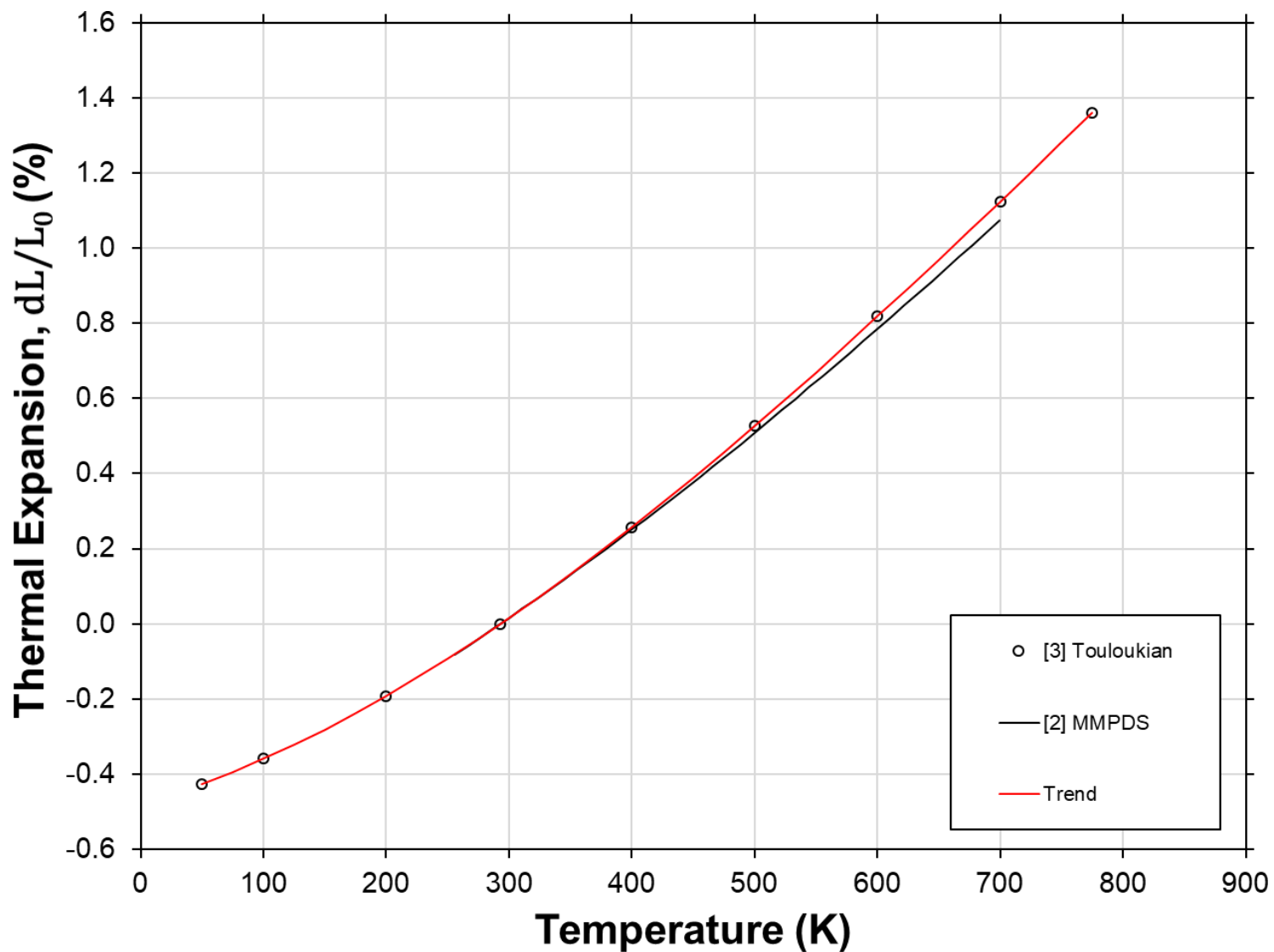


Figure 1.1.1-3: Thermal Expansion versus Temperature for Aluminum 7075, compared against trend from MMPDS -12, Al 7075, T-6, A Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |           | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|-----------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )    | %                                       |
| 50                | ( -369.7 ) | -0.425                                  | 425               | ( 305.3 ) | 0.322                                   |
| 75                | ( -324.7 ) | -0.393                                  | 450               | ( 350.3 ) | 0.389                                   |
| 100               | ( -279.7 ) | -0.359                                  | 475               | ( 395.3 ) | 0.457                                   |
| 125               | ( -234.7 ) | -0.321                                  | 500               | ( 440.3 ) | 0.527                                   |
| 150               | ( -189.7 ) | -0.281                                  | 525               | ( 485.3 ) | 0.598                                   |
| 175               | ( -144.7 ) | -0.238                                  | 550               | ( 530.3 ) | 0.670                                   |
| 200               | ( -99.7 )  | -0.192                                  | 575               | ( 575.3 ) | 0.744                                   |
| 225               | ( -54.7 )  | -0.144                                  | 600               | ( 620.3 ) | 0.818                                   |
| 250               | ( -9.7 )   | -0.093                                  | 625               | ( 665.3 ) | 0.893                                   |
| 275               | ( 35.3 )   | -0.040                                  | 650               | ( 710.3 ) | 0.970                                   |
| 300               | ( 80.3 )   | 0.016                                   | 675               | ( 755.3 ) | 1.046                                   |
| 325               | ( 125.3 )  | 0.073                                   | 700               | ( 800.3 ) | 1.124                                   |
| 350               | ( 170.3 )  | 0.133                                   | 725               | ( 845.3 ) | 1.202                                   |
| 375               | ( 215.3 )  | 0.194                                   | 750               | ( 890.3 ) | 1.280                                   |
| 400               | ( 260.3 )  | 0.257                                   | 775               | ( 935.3 ) | 1.359                                   |

**Application Notes:** Data for thermal expansion is collected from reference [3] and compared to data trend from reference [2]. Data is fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

T Range [K]: 50 ≤ T ≤ 775

$$A0 = -0.4785$$

$$A1 = 0.9386$$

$$A2 = 2.684$$

$$A3 = -1.078$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

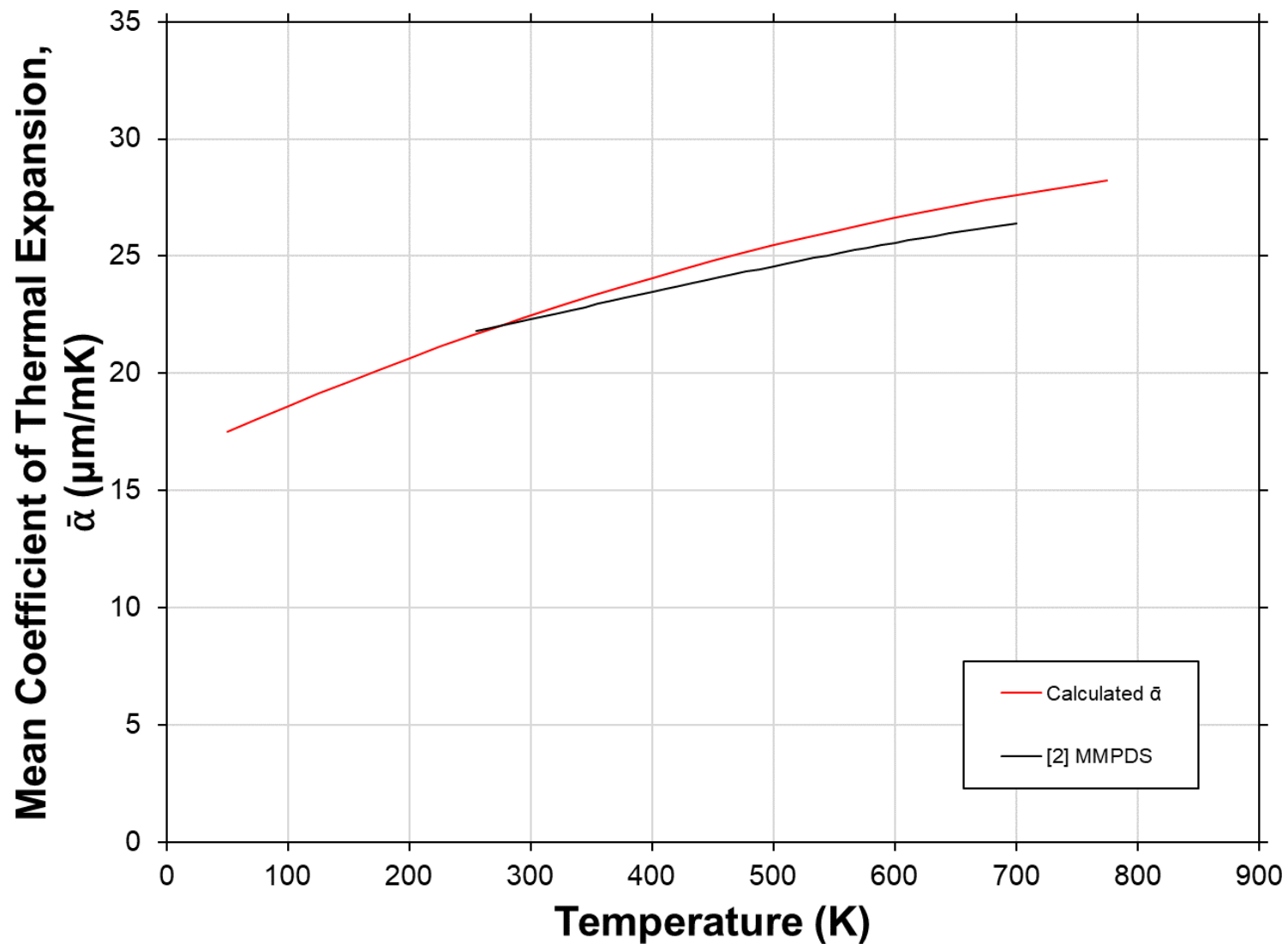


Figure 1.1.1-4: Mean Coefficient of Thermal Expansion versus Temperature for Aluminum 7075. Calculated from fitted trend of the Thermal Expansion data, and compared against trend from MMPDS -12, Al 7075, T-6, A Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |           | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|-----------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )    | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 50                | ( -369.7 ) | 17.495                                | ( 9.720 )                                           | 425               | ( 305.3 ) | 24.439                                | ( 13.577 )                                          |
| 75                | ( -324.7 ) | 18.057                                | ( 10.031 )                                          | 450               | ( 350.3 ) | 24.793                                | ( 13.774 )                                          |
| 100               | ( -279.7 ) | 18.603                                | ( 10.335 )                                          | 475               | ( 395.3 ) | 25.134                                | ( 13.963 )                                          |
| 125               | ( -234.7 ) | 19.136                                | ( 10.631 )                                          | 500               | ( 440.3 ) | 25.461                                | ( 14.145 )                                          |
| 150               | ( -189.7 ) | 19.654                                | ( 10.919 )                                          | 525               | ( 485.3 ) | 25.775                                | ( 14.320 )                                          |
| 175               | ( -144.7 ) | 20.158                                | ( 11.199 )                                          | 550               | ( 530.3 ) | 26.076                                | ( 14.487 )                                          |
| 200               | ( -99.7 )  | 20.648                                | ( 11.471 )                                          | 575               | ( 575.3 ) | 26.365                                | ( 14.647 )                                          |
| 225               | ( -54.7 )  | 21.124                                | ( 11.736 )                                          | 600               | ( 620.3 ) | 26.640                                | ( 14.800 )                                          |
| 250               | ( -9.7 )   | 21.587                                | ( 11.993 )                                          | 625               | ( 665.3 ) | 26.903                                | ( 14.946 )                                          |
| 275               | ( 35.3 )   | 22.035                                | ( 12.242 )                                          | 650               | ( 710.3 ) | 27.153                                | ( 15.085 )                                          |
| 300               | ( 80.3 )   | 22.470                                | ( 12.483 )                                          | 675               | ( 755.3 ) | 27.390                                | ( 15.217 )                                          |
| 325               | ( 125.3 )  | 22.891                                | ( 12.717 )                                          | 700               | ( 800.3 ) | 27.615                                | ( 15.341 )                                          |
| 350               | ( 170.3 )  | 23.298                                | ( 12.943 )                                          | 725               | ( 845.3 ) | 27.827                                | ( 15.459 )                                          |
| 375               | ( 215.3 )  | 23.692                                | ( 13.162 )                                          | 750               | ( 890.3 ) | 28.026                                | ( 15.570 )                                          |
| 400               | ( 260.3 )  | 24.072                                | ( 13.374 )                                          | 775               | ( 935.3 ) | 28.214                                | ( 15.674 )                                          |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated from the trend for thermal expansion. Calculated data is then fitted with the equation below to approximate property trend with respect to temperature and compared against data trend from reference [2].

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $50 \leq T \leq 775$

A0 = 16.33

A1 = 23.89

A2 = -11.65

A3 = 0.7872



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Specific Heat with Temperature

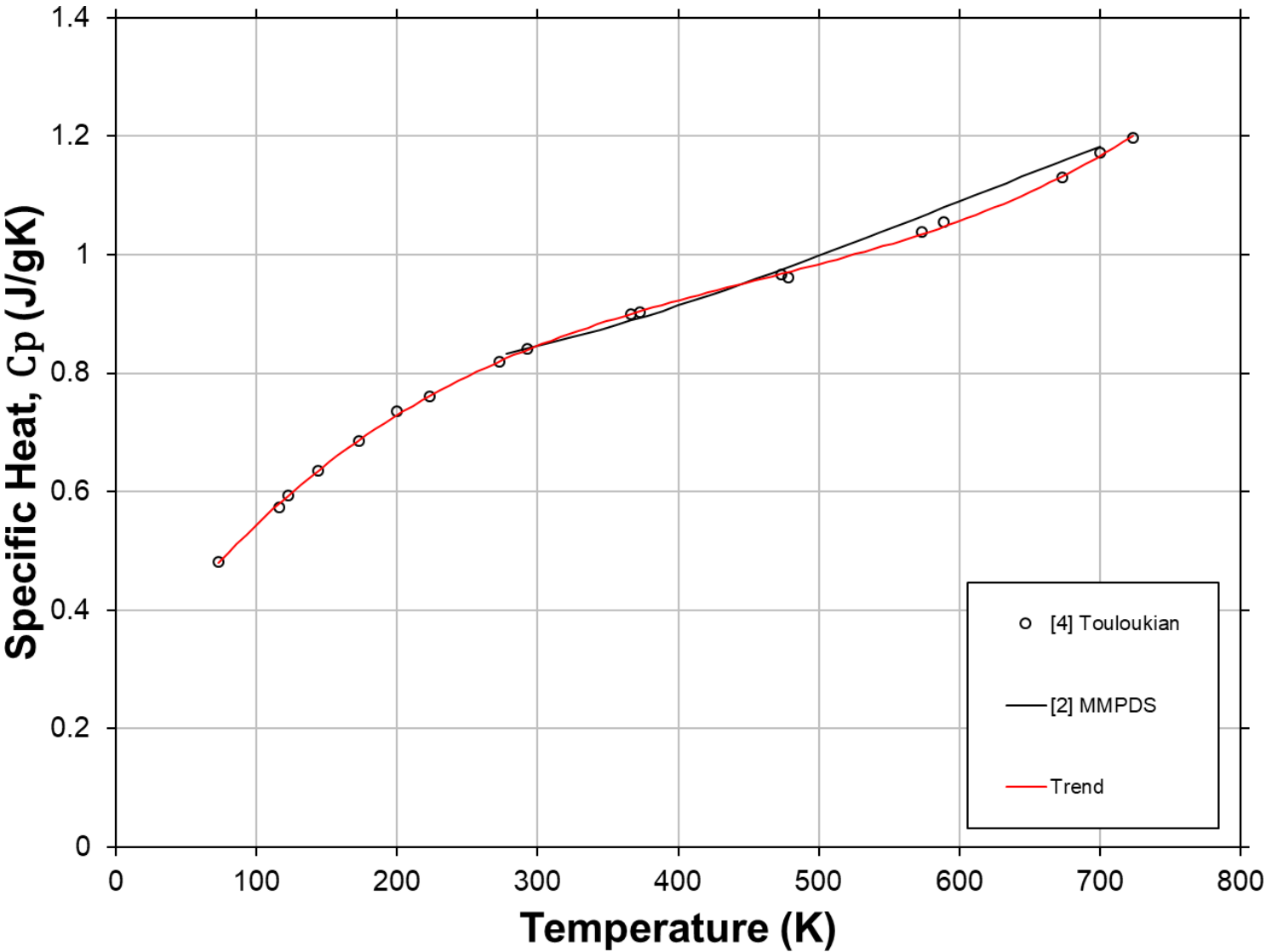


Figure 1.1.1-5: Specific Heat versus Temperature for Aluminum 7075, compared against data trend from MMPDS-12, Al 7075, T-6, A Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |           | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|-----------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )    | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 73                | ( -328.3 ) | 0.479                            | ( 0.115 )       | 425               | ( 305.3 ) | 0.938                            | ( 0.224 )       |
| 100               | ( -279.7 ) | 0.544                            | ( 0.130 )       | 450               | ( 350.3 ) | 0.954                            | ( 0.228 )       |
| 125               | ( -234.7 ) | 0.598                            | ( 0.143 )       | 475               | ( 395.3 ) | 0.969                            | ( 0.232 )       |
| 150               | ( -189.7 ) | 0.647                            | ( 0.155 )       | 500               | ( 440.3 ) | 0.984                            | ( 0.235 )       |
| 175               | ( -144.7 ) | 0.690                            | ( 0.165 )       | 525               | ( 485.3 ) | 1.000                            | ( 0.239 )       |
| 200               | ( -99.7 )  | 0.729                            | ( 0.174 )       | 550               | ( 530.3 ) | 1.017                            | ( 0.243 )       |
| 225               | ( -54.7 )  | 0.764                            | ( 0.183 )       | 575               | ( 575.3 ) | 1.036                            | ( 0.248 )       |
| 250               | ( -9.7 )   | 0.795                            | ( 0.190 )       | 600               | ( 620.3 ) | 1.057                            | ( 0.253 )       |
| 275               | ( 35.3 )   | 0.822                            | ( 0.196 )       | 625               | ( 665.3 ) | 1.080                            | ( 0.258 )       |
| 300               | ( 80.3 )   | 0.847                            | ( 0.202 )       | 650               | ( 710.3 ) | 1.106                            | ( 0.264 )       |
| 325               | ( 125.3 )  | 0.868                            | ( 0.208 )       | 675               | ( 755.3 ) | 1.135                            | ( 0.271 )       |
| 350               | ( 170.3 )  | 0.888                            | ( 0.212 )       | 700               | ( 800.3 ) | 1.167                            | ( 0.279 )       |
| 375               | ( 215.3 )  | 0.906                            | ( 0.217 )       | 723               | ( 841.7 ) | 1.201                            | ( 0.287 )       |
| 400               | ( 260.3 )  | 0.923                            | ( 0.221 )       |                   |           |                                  |                 |

**Application Notes:** Data for specific heat is collected from reference [4], and fitted with the equation below to approximate property trend with respect to temperature and compared against data trend from reference [2].

## Fit Equation:

$$C_p(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$C_p(T) = \text{Specific Heat [J / (g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

$$T. \text{ Range [K]: } \underline{73 \leq T \leq 723}$$

$$A0 = 0.2656$$

$$A1 = 3.343$$

$$A2 = -6.003$$

$$A3 = 4.382$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Elastic Modulus with Temperature

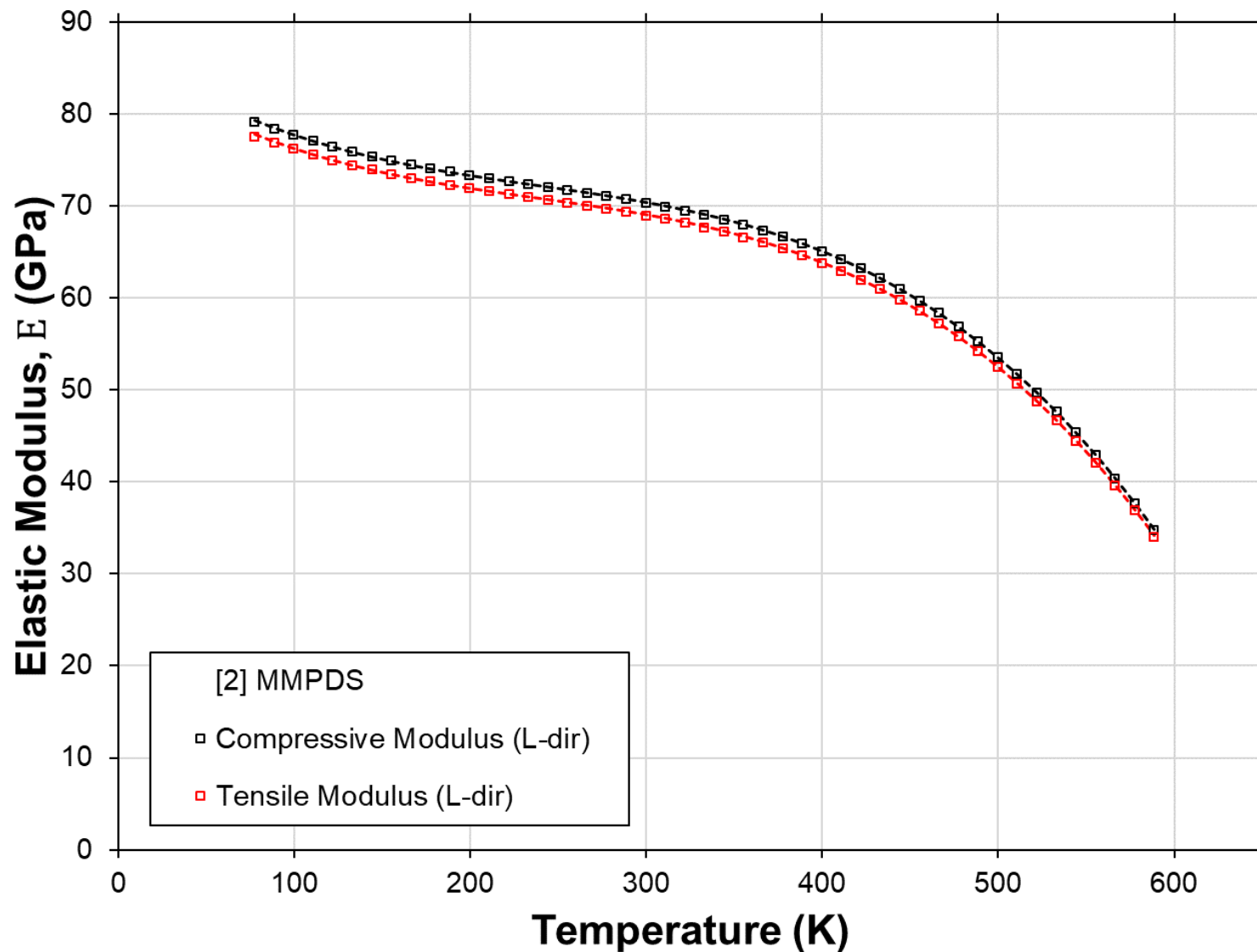


Figure 1.1.1-6: Compressive and Tensile Modulus versus Temperature for Aluminum 7075, T-6, A Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Elastic Modulus with Temperature

## 100% Theoretical Density Tensile Modulus

| Temperature ( T ) |            | Young's Modulus ( E ) |           | Temperature ( T ) |           | Young's Modulus ( E ) |          |
|-------------------|------------|-----------------------|-----------|-------------------|-----------|-----------------------|----------|
| K                 | ( °F )     | GPa                   | ( Msi )   | K                 | ( °F )    | GPa                   | ( Msi )  |
| 75                | ( -324.7 ) | 77.93                 | ( 11.31 ) | 380               | ( 224.3 ) | 65.21                 | ( 9.46 ) |
| 100               | ( -279.7 ) | 76.18                 | ( 11.05 ) | 400               | ( 260.3 ) | 63.80                 | ( 9.26 ) |
| 120               | ( -243.7 ) | 75.01                 | ( 10.88 ) | 420               | ( 296.3 ) | 62.14                 | ( 9.02 ) |
| 140               | ( -207.7 ) | 74.03                 | ( 10.74 ) | 440               | ( 332.3 ) | 60.20                 | ( 8.74 ) |
| 160               | ( -171.7 ) | 73.20                 | ( 10.62 ) | 460               | ( 368.3 ) | 57.96                 | ( 8.41 ) |
| 180               | ( -135.7 ) | 72.49                 | ( 10.52 ) | 480               | ( 404.3 ) | 55.37                 | ( 8.03 ) |
| 200               | ( -99.7 )  | 71.86                 | ( 10.43 ) | 500               | ( 440.3 ) | 52.42                 | ( 7.61 ) |
| 220               | ( -63.7 )  | 71.30                 | ( 10.35 ) | 520               | ( 476.3 ) | 49.07                 | ( 7.12 ) |
| 240               | ( -27.7 )  | 70.76                 | ( 10.27 ) | 540               | ( 512.3 ) | 45.29                 | ( 6.57 ) |
| 260               | ( 8.3 )    | 70.22                 | ( 10.19 ) | 560               | ( 548.3 ) | 41.05                 | ( 5.96 ) |
| 280               | ( 44.3 )   | 69.64                 | ( 10.11 ) | 580               | ( 584.3 ) | 36.32                 | ( 5.27 ) |
| 300               | ( 80.3 )   | 69.01                 | ( 10.01 ) | 600               | ( 620.3 ) | 31.07                 | ( 4.51 ) |
| 320               | ( 116.3 )  | 68.27                 | ( 9.91 )  |                   |           |                       |          |
| 340               | ( 152.3 )  | 67.42                 | ( 9.78 )  |                   |           |                       |          |
| 360               | ( 188.3 )  | 66.41                 | ( 9.64 )  |                   |           |                       |          |

**Application Notes:** The data for Young's Modulus (tensile and compressive) is collected from reference [2], and fitted with the equation below to approximate property trend with respect to temperature.

### Fit Equations:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

### Constants:

T Range [K]:

$75 \leq T \leq 600$

Compressive

Tensile

|      |        |        |
|------|--------|--------|
| A0 = | 87.47  | 85.76  |
| A1 = | -137.6 | -134.9 |
| A2 = | 463.2  | 454.1  |
| A3 = | -648   | -635.3 |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 2830                                  |                                                         | 75.22                                       | -0.359                                             | 18.60                                     | 0.544                                          | 76.18                              |                                  |                             |
| 150                 | 2824                                  |                                                         | 97.17                                       | -0.281                                             | 19.65                                     | 0.647                                          | 73.60                              |                                  |                             |
| 200                 | 2816                                  |                                                         | 116.04                                      | -0.192                                             | 20.65                                     | 0.729                                          | 71.86                              |                                  |                             |
| 250                 | 2808                                  |                                                         | 131.99                                      | -0.093                                             | 21.59                                     | 0.795                                          | 70.49                              |                                  |                             |
| 293                 | 2800                                  |                                                         | 143.50                                      | 0.000                                              | 22.35                                     | 0.840                                          | 69.24                              |                                  |                             |
| 300                 | 2799                                  |                                                         | 145.19                                      | 0.016                                              | 22.47                                     | 0.847                                          | 69.01                              |                                  |                             |
| 325                 | 2794                                  |                                                         | 150.80                                      | 0.073                                              | 22.89                                     | 0.868                                          | 68.07                              |                                  |                             |
| 350                 | 2789                                  |                                                         | 155.79                                      | 0.133                                              | 23.30                                     | 0.888                                          | 66.93                              |                                  |                             |
| 375                 | 2784                                  |                                                         | 160.17                                      | 0.194                                              | 23.69                                     | 0.906                                          | 65.53                              |                                  |                             |
| 400                 | 2778                                  |                                                         | 163.97                                      | 0.257                                              | 24.07                                     | 0.923                                          | 63.80                              |                                  |                             |
| 425                 | 2773                                  |                                                         | 167.20                                      | 0.322                                              | 24.44                                     | 0.938                                          | 61.68                              |                                  |                             |
| 450                 | 2768                                  |                                                         | 169.89                                      | 0.389                                              | 24.79                                     | 0.954                                          | 59.12                              |                                  |                             |
| 475                 | 2762                                  |                                                         | 172.05                                      | 0.457                                              | 25.13                                     | 0.969                                          | 56.05                              |                                  |                             |
| 500                 | 2756                                  |                                                         | 173.71                                      | 0.527                                              | 25.46                                     | 0.984                                          | 52.42                              |                                  |                             |
| 525                 | 2750                                  |                                                         | 174.88                                      | 0.598                                              | 25.78                                     | 1.000                                          | 48.17                              |                                  |                             |
| 550                 | 2744                                  |                                                         | 175.59                                      | 0.670                                              | 26.08                                     | 1.017                                          | 43.23                              |                                  |                             |
| 575                 | 2738                                  |                                                         | 175.85                                      | 0.744                                              | 26.36                                     | 1.036                                          | 37.55                              |                                  |                             |
| 600                 | 2732                                  |                                                         | 175.70                                      | 0.818                                              | 26.64                                     | 1.057                                          | 31.07                              |                                  |                             |
| 625                 | 2726                                  |                                                         | 175.14                                      | 0.893                                              | 26.90                                     | 1.080                                          | -                                  |                                  |                             |
| 650                 | 2720                                  |                                                         | 174.20                                      | 0.970                                              | 27.15                                     | 1.106                                          | -                                  |                                  |                             |
| 675                 | 2714                                  |                                                         | 172.90                                      | 1.046                                              | 27.39                                     | 1.135                                          | -                                  |                                  |                             |
| 700                 | 2708                                  |                                                         | 171.26                                      | 1.124                                              | 27.61                                     | 1.167                                          | -                                  |                                  |                             |
| 725                 | 2701                                  |                                                         | -                                           | 1.202                                              | 27.83                                     | -                                              | -                                  |                                  |                             |
| 750                 | 2695                                  |                                                         | -                                           | 1.280                                              | 28.03                                     | -                                              | -                                  |                                  |                             |
| 775                 | 2689                                  |                                                         | -                                           | 1.359                                              | 28.21                                     | -                                              | -                                  |                                  |                             |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.1 Aluminum Alloys

1.1.1 Aluminum 7075

**Revision 0: 08-05-2020**

**References**

- [1] Y.S. Touloukian, R.W. Powell, C.Y. Ho, P.G. Klemens, Thermal Conductivity - Metallic Elements and Alloys, Thermophysical Properties of Matter - The TPRC Data Series, Vol. 1, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1970.
- [2] MMPDS-12. (Accessed July 2019).
- [3] Y.S. Touloukian, R.K. Kirby, R.E. Taylor, P.D. Desai, Thermal Expansion - Metallic Elements and Alloys, Thermophysical Properties of Matter - the TPRC Data Series, Vol. 12, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, web, 1975.
- [4] Y.S. Touloukian, E.H. Buyco, Specific Heat - Nonmetallic Solids, Thermophysical Properties of Matter - the TPRC Data Series, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1970.

## **1 Light Metal Alloys**

### **1.2 Titanium Alloys**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V (Ti64)

Revision 0: 08-05-2020

General

### Room Temperature Properties

|                                          |       |
|------------------------------------------|-------|
| Density, [kg/m <sup>3</sup> ]            | 4,430 |
| Melting Point, [K]                       | 1947  |
| Specific Heat, [J/(g-K)]                 | 0.538 |
| Thermal Conductivity, [W/(m-K)]          | 7.3*  |
| Linear expansion coefficient, [μm/(m-K)] | 9.59  |
| Electrical resistivity, [μΩ-m]           | 1.70  |
| Young's Modulus, [GPa]                   | 110.5 |
| Shear Modulus, [GPa]                     | 42.2  |
| Poisson's Ratio, [-]                     | 0.310 |

\*at 311 K

### Composition

Table 1.2.1-1: Typical Composition ranges for grade 5 Ti-6Al-4V alloy (percent by weight) [1].

| Grade     |      | Ti   | Al   | V   | Fe   | O    | C    | N    | H     | Other |
|-----------|------|------|------|-----|------|------|------|------|-------|-------|
| Ti-6Al-4V | Min. | 87.6 | 5.5  | 3.5 |      |      |      |      |       |       |
|           | Max. | 91.0 | 6.75 | 4.5 | 0.40 | 0.20 | 0.08 | 0.05 | 0.015 | 0.40  |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Density with Temperature

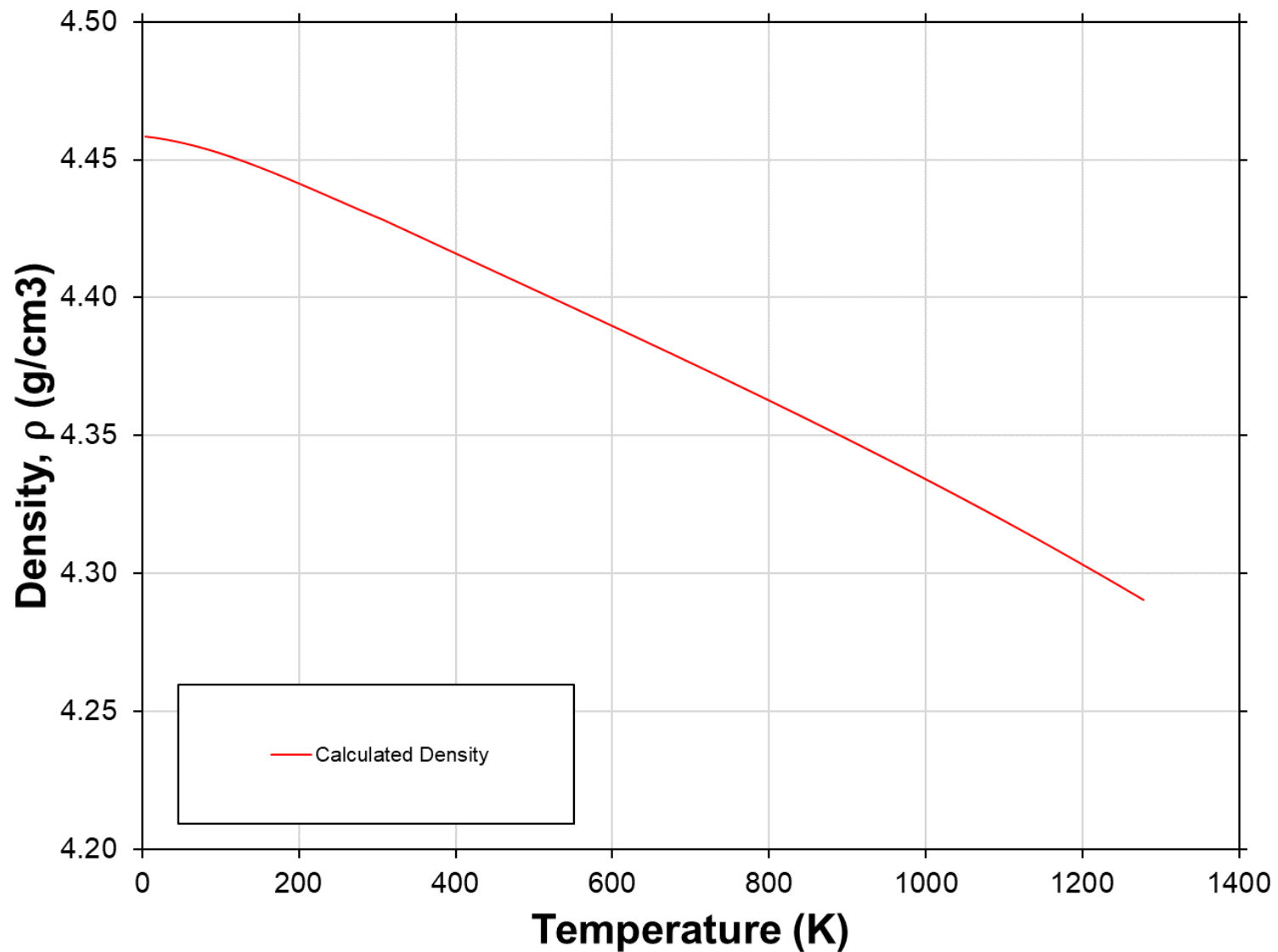


Figure 1.2.1-1: Density versus Temperature for Ti-6Al-4V. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Density with Temperature

## 100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 5                 | ( -450.7 ) | 4459              | ( 278.3 )              | 650               | ( 710.3 )  | 4383              | ( 273.6 )              |
| 25                | ( -414.7 ) | 4458              | ( 278.3 )              | 700               | ( 800.3 )  | 4376              | ( 273.2 )              |
| 50                | ( -369.7 ) | 4456              | ( 278.2 )              | 750               | ( 890.3 )  | 4370              | ( 272.8 )              |
| 75                | ( -324.7 ) | 4455              | ( 278.1 )              | 800               | ( 980.3 )  | 4363              | ( 272.4 )              |
| 100               | ( -279.7 ) | 4452              | ( 278.0 )              | 850               | ( 1070.3 ) | 4356              | ( 271.9 )              |
| 150               | ( -189.7 ) | 4447              | ( 277.6 )              | 900               | ( 1160.3 ) | 4349              | ( 271.5 )              |
| 200               | ( -99.7 )  | 4442              | ( 277.3 )              | 950               | ( 1250.3 ) | 4341              | ( 271.0 )              |
| 250               | ( -9.7 )   | 4435              | ( 276.9 )              | 1000              | ( 1340.3 ) | 4334              | ( 270.6 )              |
| 300               | ( 80.3 )   | 4429              | ( 276.5 )              | 1050              | ( 1430.3 ) | 4327              | ( 270.1 )              |
| 350               | ( 170.3 )  | 4423              | ( 276.1 )              | 1100              | ( 1520.3 ) | 4319              | ( 269.6 )              |
| 400               | ( 260.3 )  | 4416              | ( 275.7 )              | 1150              | ( 1610.3 ) | 4311              | ( 269.1 )              |
| 450               | ( 350.3 )  | 4410              | ( 275.3 )              | 1200              | ( 1700.3 ) | 4303              | ( 268.6 )              |
| 500               | ( 440.3 )  | 4403              | ( 274.9 )              | 1250              | ( 1790.3 ) | 4295              | ( 268.1 )              |
| 550               | ( 530.3 )  | 4396              | ( 274.5 )              | 1278              | ( 1840.7 ) | 4290              | ( 267.8 )              |
| 600               | ( 620.3 )  | 4390              | ( 274.1 )              |                   |            |                   |                        |

**Application Notes:** Density is calculated as a function of Thermal Expansion as seen in the equation below to approximate property trend with respect to temperature.

### Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 4,430 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $4 \leq T \leq 1278$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

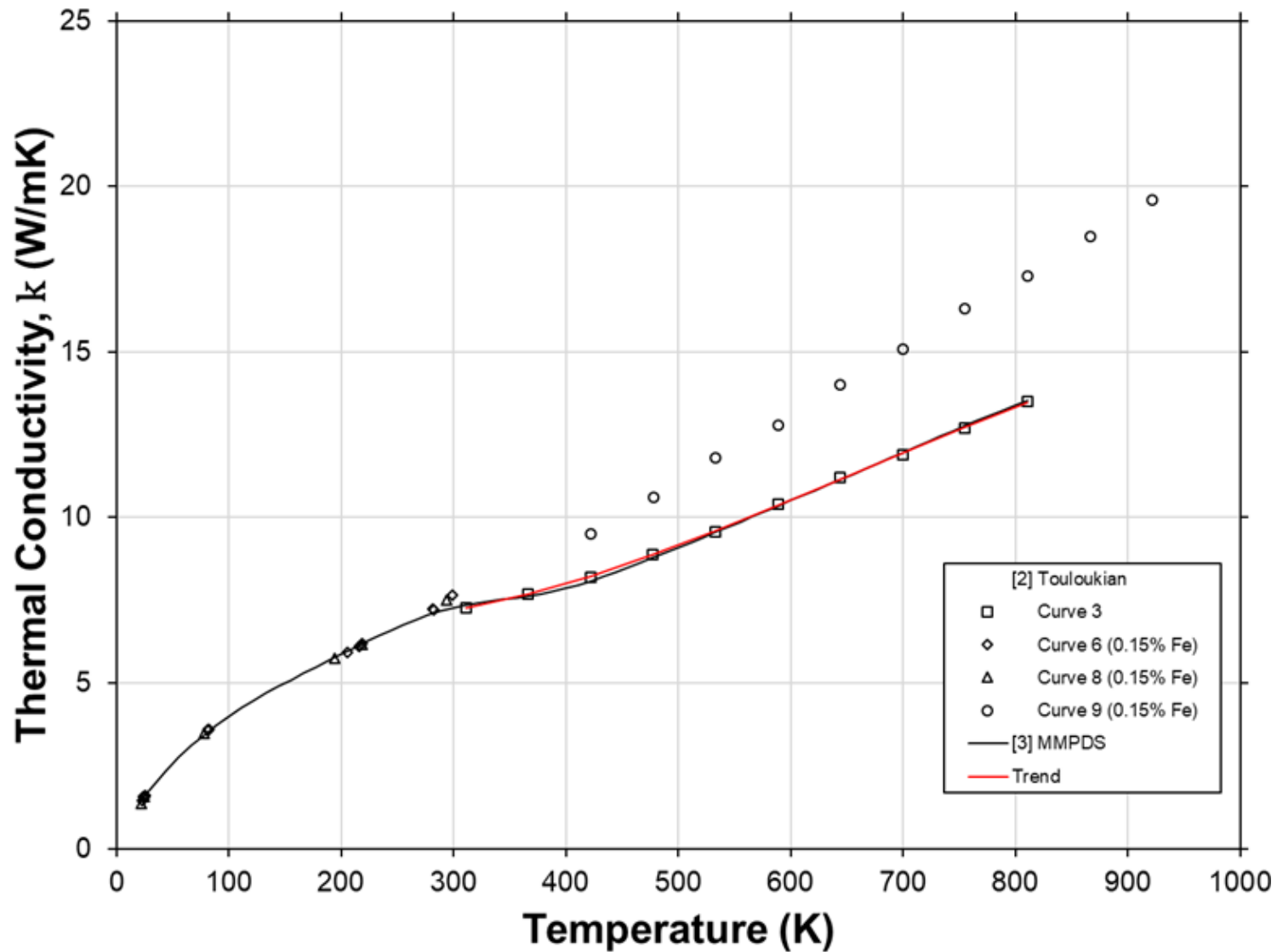


Figure 1.2.1-2: Thermal Conductivity versus Temperature for Ti-6Al-4V with comparison to MMPDS-12, solution treated and aged, S-Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|-----------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )    | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 311               | ( 100.1 ) | 7.27                       | ( 50.47 )                           | 580               | ( 584.3 )  | 10.24                      | ( 71.05 )                           |
| 320               | ( 116.3 ) | 7.33                       | ( 50.88 )                           | 600               | ( 620.3 )  | 10.52                      | ( 73.01 )                           |
| 340               | ( 152.3 ) | 7.48                       | ( 51.88 )                           | 620               | ( 656.3 )  | 10.81                      | ( 74.99 )                           |
| 360               | ( 188.3 ) | 7.64                       | ( 53.00 )                           | 640               | ( 692.3 )  | 11.10                      | ( 76.99 )                           |
| 380               | ( 224.3 ) | 7.82                       | ( 54.24 )                           | 660               | ( 728.3 )  | 11.39                      | ( 78.99 )                           |
| 400               | ( 260.3 ) | 8.01                       | ( 55.57 )                           | 680               | ( 764.3 )  | 11.67                      | ( 80.99 )                           |
| 420               | ( 296.3 ) | 8.22                       | ( 57.01 )                           | 700               | ( 800.3 )  | 11.96                      | ( 82.98 )                           |
| 440               | ( 332.3 ) | 8.44                       | ( 58.53 )                           | 720               | ( 836.3 )  | 12.25                      | ( 84.96 )                           |
| 460               | ( 368.3 ) | 8.67                       | ( 60.13 )                           | 740               | ( 872.3 )  | 12.53                      | ( 86.91 )                           |
| 480               | ( 404.3 ) | 8.91                       | ( 61.81 )                           | 760               | ( 908.3 )  | 12.80                      | ( 88.83 )                           |
| 500               | ( 440.3 ) | 9.16                       | ( 63.55 )                           | 780               | ( 944.3 )  | 13.07                      | ( 90.71 )                           |
| 520               | ( 476.3 ) | 9.42                       | ( 65.36 )                           | 800               | ( 980.3 )  | 13.34                      | ( 92.54 )                           |
| 540               | ( 512.3 ) | 9.69                       | ( 67.21 )                           | 811               | ( 1000.1 ) | 13.48                      | ( 93.52 )                           |
| 560               | ( 548.3 ) | 9.96                       | ( 69.11 )                           |                   |            |                            |                                     |

**Application Notes:** Data for thermal conductivity is collected from reference [2] and fitted with the equation below to approximate the property trend with respect to temperature. Fitted trend is compared against reference [3].

## Fit Equation:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$k(T)$  = Thermal Conductivity (W/m · K)

$T$  = Temperature [K]

## Constants:

T Range [K] 311 ≤ T ≤ 811

A0 = 8.254

A1 = -14.85

A2 = 44.69

A3 = -22.73



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Thermal Expansion with Temperature

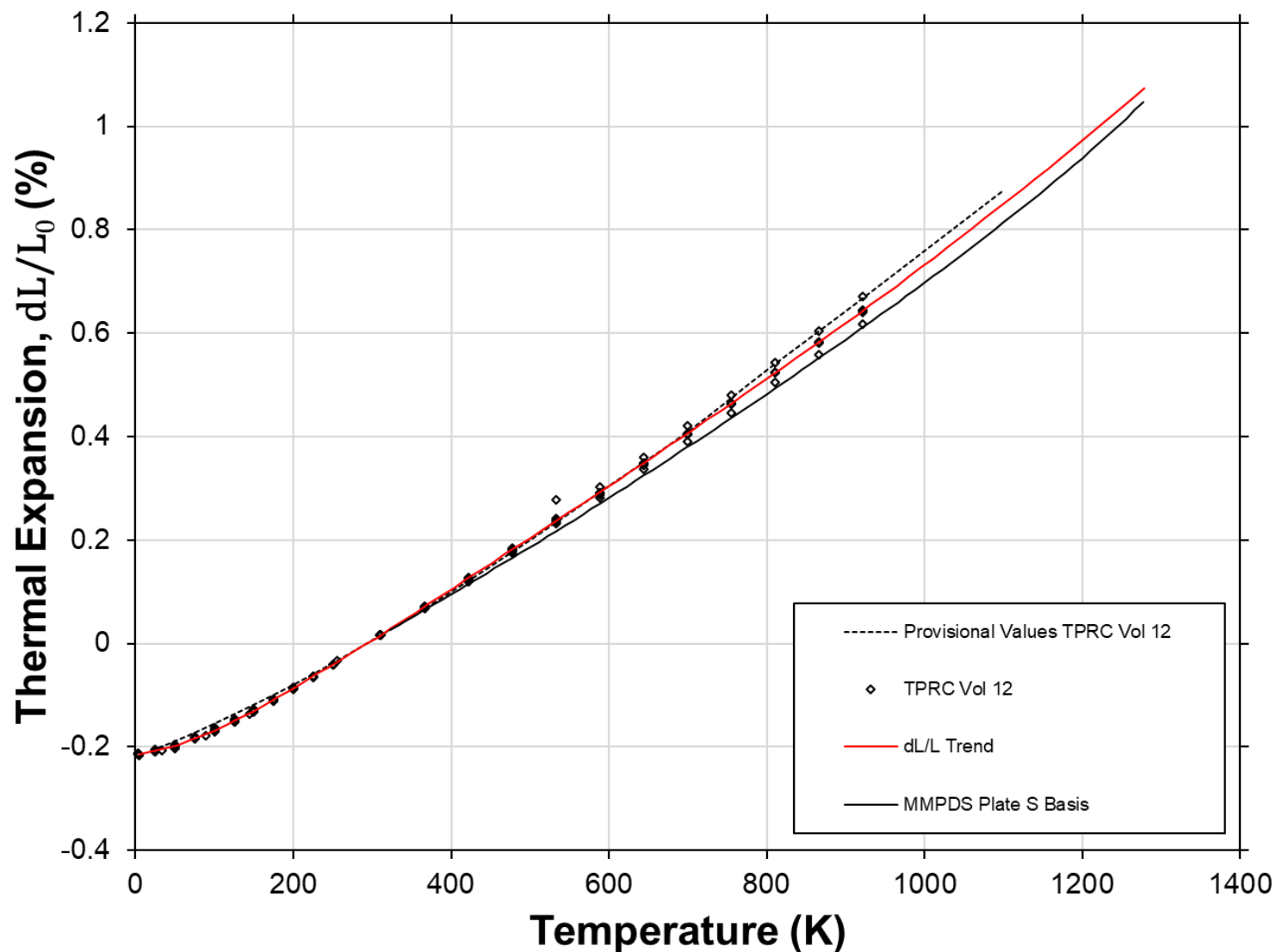


Figure 1.2.1-3: Thermal Expansion versus Temperature for Ti-6Al-4V. Calculated from fitted trend of Thermal Expansion with comparison against MMPDS-12, solution treated and aged, S-Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 5                 | ( -450.7 ) | -0.214                                  | 650               | ( 710.3 )  | 0.355                                   |
| 25                | ( -414.7 ) | -0.208                                  | 700               | ( 800.3 )  | 0.407                                   |
| 50                | ( -369.7 ) | -0.198                                  | 750               | ( 890.3 )  | 0.459                                   |
| 75                | ( -324.7 ) | -0.184                                  | 800               | ( 980.3 )  | 0.511                                   |
| 100               | ( -279.7 ) | -0.168                                  | 850               | ( 1070.3 ) | 0.565                                   |
| 150               | ( -189.7 ) | -0.130                                  | 900               | ( 1160.3 ) | 0.619                                   |
| 200               | ( -99.7 )  | -0.087                                  | 950               | ( 1250.3 ) | 0.675                                   |
| 250               | ( -9.7 )   | -0.040                                  | 1000              | ( 1340.3 ) | 0.732                                   |
| 300               | ( 80.3 )   | 0.005                                   | 1050              | ( 1430.3 ) | 0.790                                   |
| 350               | ( 170.3 )  | 0.055                                   | 1100              | ( 1520.3 ) | 0.849                                   |
| 400               | ( 260.3 )  | 0.105                                   | 1150              | ( 1610.3 ) | 0.910                                   |
| 450               | ( 350.3 )  | 0.154                                   | 1200              | ( 1700.3 ) | 0.973                                   |
| 500               | ( 440.3 )  | 0.204                                   | 1250              | ( 1790.3 ) | 1.037                                   |
| 550               | ( 530.3 )  | 0.254                                   | 1278              | ( 1840.7 ) | 1.074                                   |
| 600               | ( 620.3 )  | 0.304                                   |                   |            |                                         |

**Application Notes:** Data for thermal expansion is collected from reference [4] and fitted with the equation below to approximate property trend with respect to temperature. Fitted trend is compared against trend from reference [3].

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

| T Range [K]: | $2 \leq T < 293$ | $293 \leq T \leq 1278$ |
|--------------|------------------|------------------------|
| A0 =         | -0.2151          | -0.3017                |
| A1 =         | 0.2126           | 1.06                   |
| A2 =         | 2.949            | -0.1675                |
| A3 =         | -4.007           | 0.1411                 |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

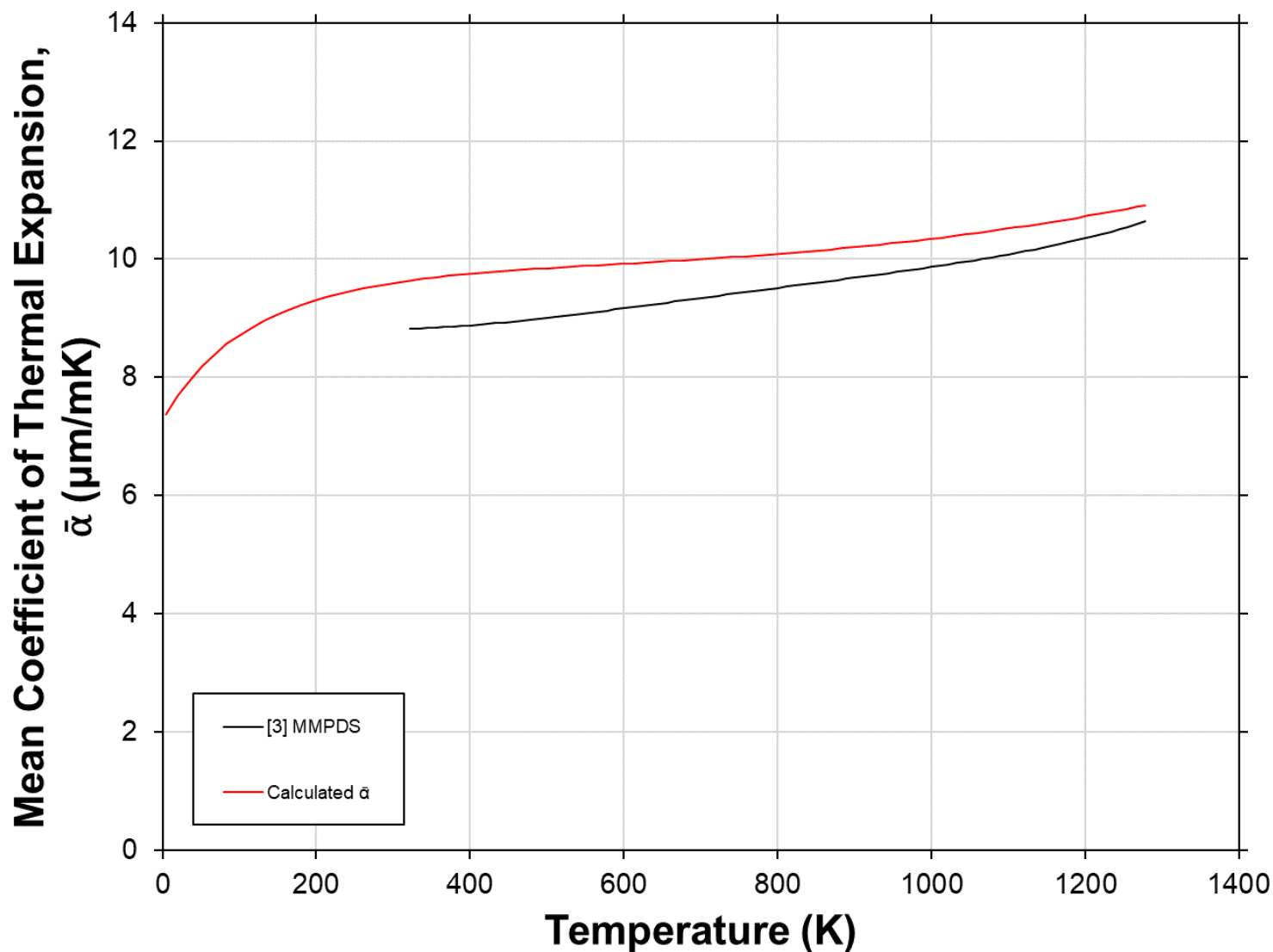


Figure 1.2.1-4: Mean Coefficient of Thermal Expansion versus Temperature for Ti-6Al-4V. Calculated from fitted trend of the Thermal Expansion data, with comparison to MMPDS-12, solution treated and aged, S-Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 5                 | ( -450.7 ) | 7.394                                 | ( 4.108 )                                           | 650               | ( 710.3 )  | 9.959                                 | ( 5.533 )                                           |
| 25                | ( -414.7 ) | 7.782                                 | ( 4.324 )                                           | 700               | ( 800.3 )  | 9.999                                 | ( 5.555 )                                           |
| 50                | ( -369.7 ) | 8.168                                 | ( 4.538 )                                           | 750               | ( 890.3 )  | 10.042                                | ( 5.579 )                                           |
| 75                | ( -324.7 ) | 8.472                                 | ( 4.707 )                                           | 800               | ( 980.3 )  | 10.090                                | ( 5.606 )                                           |
| 100               | ( -279.7 ) | 8.714                                 | ( 4.841 )                                           | 850               | ( 1070.3 ) | 10.144                                | ( 5.636 )                                           |
| 150               | ( -189.7 ) | 9.070                                 | ( 5.039 )                                           | 900               | ( 1160.3 ) | 10.204                                | ( 5.669 )                                           |
| 200               | ( -99.7 )  | 9.311                                 | ( 5.173 )                                           | 950               | ( 1250.3 ) | 10.271                                | ( 5.706 )                                           |
| 250               | ( -9.7 )   | 9.479                                 | ( 5.266 )                                           | 1000              | ( 1340.3 ) | 10.346                                | ( 5.748 )                                           |
| 300               | ( 80.3 )   | 9.598                                 | ( 5.332 )                                           | 1050              | ( 1430.3 ) | 10.428                                | ( 5.794 )                                           |
| 350               | ( 170.3 )  | 9.685                                 | ( 5.381 )                                           | 1100              | ( 1520.3 ) | 10.519                                | ( 5.844 )                                           |
| 400               | ( 260.3 )  | 9.751                                 | ( 5.417 )                                           | 1150              | ( 1610.3 ) | 10.618                                | ( 5.899 )                                           |
| 450               | ( 350.3 )  | 9.803                                 | ( 5.446 )                                           | 1200              | ( 1700.3 ) | 10.727                                | ( 5.959 )                                           |
| 500               | ( 440.3 )  | 9.847                                 | ( 5.470 )                                           | 1250              | ( 1790.3 ) | 10.844                                | ( 6.024 )                                           |
| 550               | ( 530.3 )  | 9.885                                 | ( 5.492 )                                           | 1278              | ( 1840.7 ) | 10.914                                | ( 6.063 )                                           |
| 600               | ( 620.3 )  | 9.922                                 | ( 5.512 )                                           |                   |            |                                       |                                                     |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated as a function of thermal expansion. Calculated data is fitted with the equation below to approximate property trend with respect to temperature. Fitted trend is compared against trend from reference [3].

## Fit Equation:

$$\bar{\alpha}(T) = \left[ A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right] / \left[ A_0 + \left( \frac{T}{1000} \right) \right]$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $4 \leq T \leq 1278$

$A_0 = 1.477$

$A_1 = 11.91$

$A_2 = -3.152$

$A_3 = 2.209$

$A_0 = 0.2028$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Specific Heat with Temperature

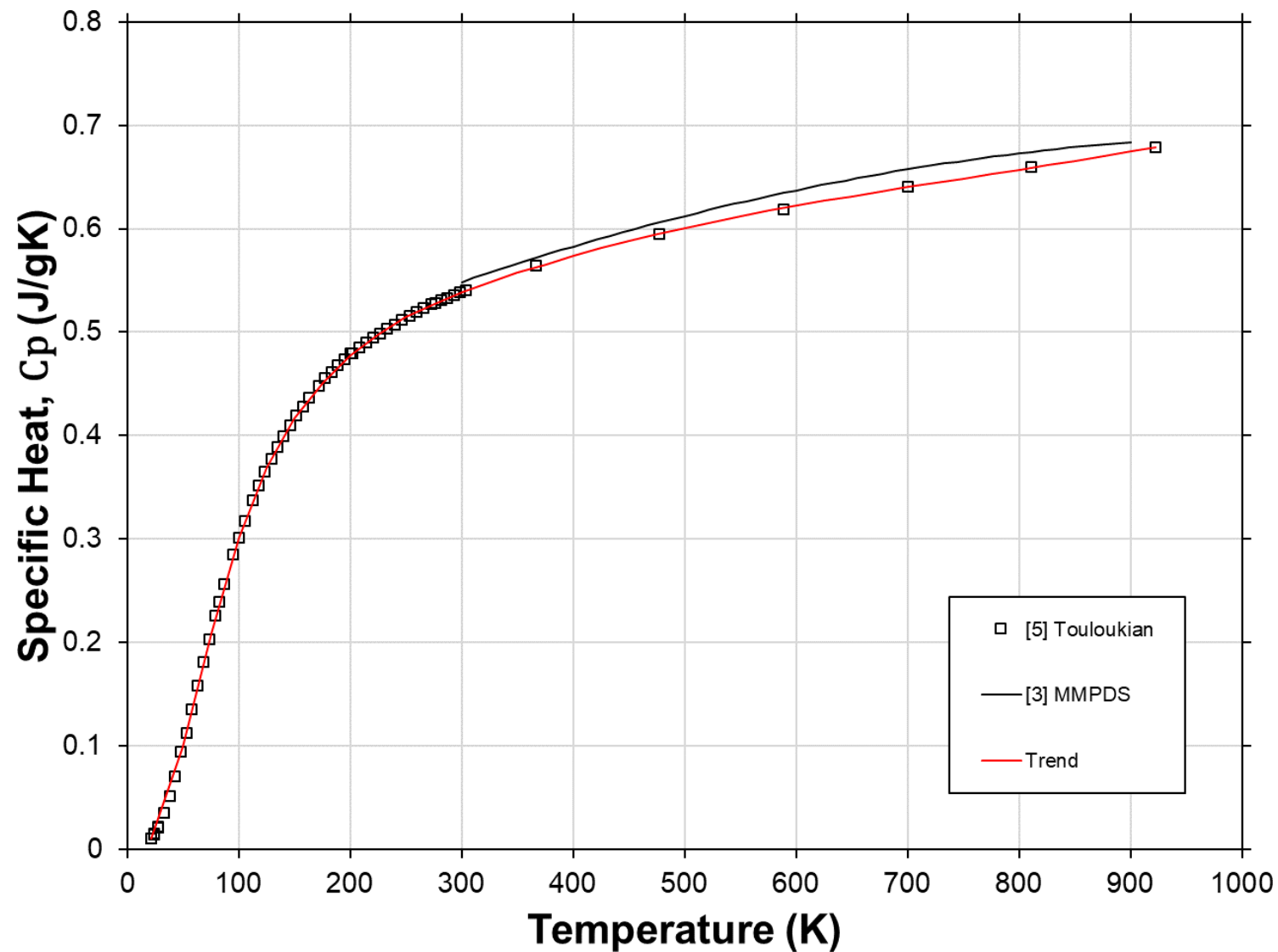


Figure 1.2.1-5: Specific Heat verses Temperature for Ti-6Al-4V with comparison to MMPDS-12, solution treated and aged, S-Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 21                | ( -421.9 ) | 0.011                            | ( 0.003 )       | 350               | ( 170.3 )  | 0.557                            | ( 0.133 )       |
| 50                | ( -369.7 ) | 0.099                            | ( 0.024 )       | 400               | ( 260.3 )  | 0.574                            | ( 0.137 )       |
| 75                | ( -324.7 ) | 0.209                            | ( 0.050 )       | 450               | ( 350.3 )  | 0.588                            | ( 0.141 )       |
| 100               | ( -279.7 ) | 0.301                            | ( 0.072 )       | 500               | ( 440.3 )  | 0.601                            | ( 0.144 )       |
| 125               | ( -234.7 ) | 0.368                            | ( 0.088 )       | 550               | ( 530.3 )  | 0.612                            | ( 0.146 )       |
| 150               | ( -189.7 ) | 0.416                            | ( 0.100 )       | 600               | ( 620.3 )  | 0.622                            | ( 0.149 )       |
| 175               | ( -144.7 ) | 0.451                            | ( 0.108 )       | 650               | ( 710.3 )  | 0.631                            | ( 0.151 )       |
| 200               | ( -99.7 )  | 0.478                            | ( 0.114 )       | 700               | ( 800.3 )  | 0.640                            | ( 0.153 )       |
| 225               | ( -54.7 )  | 0.498                            | ( 0.119 )       | 750               | ( 890.3 )  | 0.648                            | ( 0.155 )       |
| 250               | ( -9.7 )   | 0.514                            | ( 0.123 )       | 800               | ( 980.3 )  | 0.657                            | ( 0.157 )       |
| 275               | ( 35.3 )   | 0.528                            | ( 0.126 )       | 850               | ( 1070.3 ) | 0.666                            | ( 0.159 )       |
| 300               | ( 80.3 )   | 0.538                            | ( 0.129 )       | 900               | ( 1160.3 ) | 0.675                            | ( 0.161 )       |
| 325               | ( 125.3 )  | 0.548                            | ( 0.131 )       | 922               | ( 1199.9 ) | 0.679                            | ( 0.162 )       |

**Application Notes:** Data for specific heat is collected from reference [5] and fitted with the equations below to approximate property trend with respect to temperature. Fitted trend is compared against trend from reference [3].

## Fit Equation:

For temperature range:  $21 \leq T < 293$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right]$$

For temperature range:  $293 \leq T \leq 922$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_2 / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

T. Range [K]:  $21 \leq T < 293$

$$N = 2.982$$

$$A_0 = 1126$$

$$A_1 = -0.415$$

$$A_2 = 124.9$$

$$A_3 = 1688$$

$293 \leq T \leq 922$

$$B_0 = 0.3615$$

$$B_1 = 0.8142$$

$$B_2 = -0.8631$$

$$B_2 = 0.3836$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Elastic Modulus with Temperature

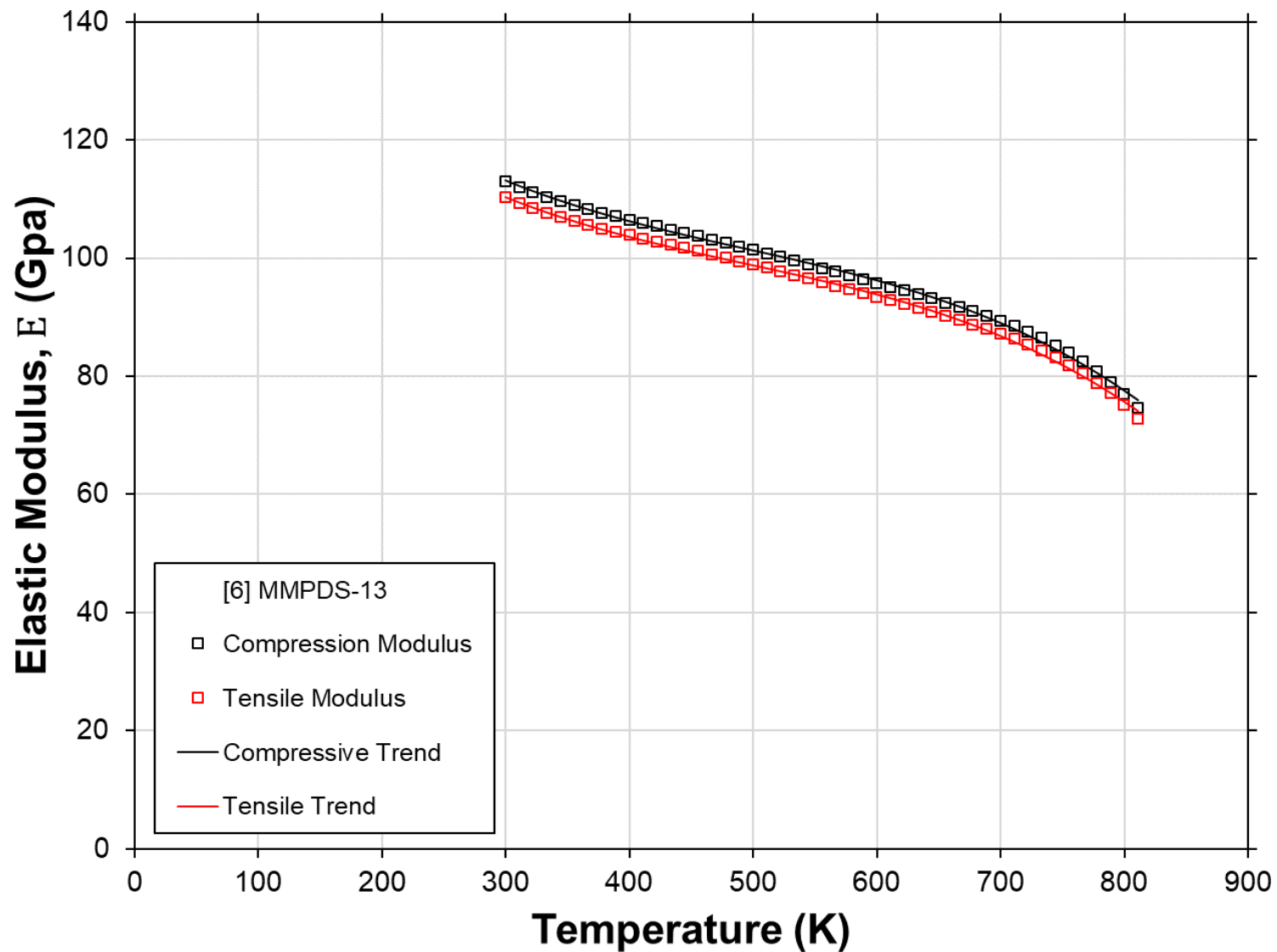


Figure 1.2.1-6: Tensile and Compressive Youngs Modulus vs Temperature, adapted from MMPDS-13, solution treated and aged, S-Basis.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 0: 08-05-2020

Elastic Modulus with Temperature

## 100% Theoretical Density Tensile Modulus

| Temperature ( T ) |           | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|-----------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )    | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 298               | ( 76.7 )  | 110.50                | ( 16.03 ) | 600               | ( 620.3 )  | 93.85                 | ( 13.62 ) |
| 320               | ( 116.3 ) | 108.74                | ( 15.78 ) | 620               | ( 656.3 )  | 92.67                 | ( 13.45 ) |
| 340               | ( 152.3 ) | 107.29                | ( 15.57 ) | 640               | ( 692.3 )  | 91.39                 | ( 13.26 ) |
| 360               | ( 188.3 ) | 105.97                | ( 15.38 ) | 660               | ( 728.3 )  | 90.00                 | ( 13.06 ) |
| 380               | ( 224.3 ) | 104.75                | ( 15.20 ) | 680               | ( 764.3 )  | 88.47                 | ( 12.84 ) |
| 400               | ( 260.3 ) | 103.62                | ( 15.03 ) | 700               | ( 800.3 )  | 86.80                 | ( 12.59 ) |
| 420               | ( 296.3 ) | 102.56                | ( 14.88 ) | 720               | ( 836.3 )  | 84.96                 | ( 12.33 ) |
| 440               | ( 332.3 ) | 101.57                | ( 14.74 ) | 740               | ( 872.3 )  | 82.94                 | ( 12.03 ) |
| 460               | ( 368.3 ) | 100.62                | ( 14.60 ) | 760               | ( 908.3 )  | 80.72                 | ( 11.71 ) |
| 480               | ( 404.3 ) | 99.70                 | ( 14.47 ) | 780               | ( 944.3 )  | 78.29                 | ( 11.36 ) |
| 500               | ( 440.3 ) | 98.79                 | ( 14.33 ) | 800               | ( 980.3 )  | 75.64                 | ( 10.97 ) |
| 520               | ( 476.3 ) | 97.87                 | ( 14.20 ) | 811               | ( 1000.1 ) | 74.07                 | ( 10.75 ) |
| 540               | ( 512.3 ) | 96.94                 | ( 14.07 ) |                   |            |                       |           |
| 560               | ( 548.3 ) | 95.96                 | ( 13.92 ) |                   |            |                       |           |
| 580               | ( 584.3 ) | 94.94                 | ( 13.78 ) |                   |            |                       |           |

**Application Notes:** Data for Young's Modulus (tensile and compressive) is adapted from reference [6] and fitted with the equation below to approximate property trend with respect to temperature.

### Fit Equations:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Elastic Modulus [GPa]

$T$  = Temperature [K]

### Constants:

T Range [K]:  $298 \leq T \leq 811$

Tensile                      Compressive

A0 =                      161.8                      165.9

A1 =                      -289.8                      -297.2

A2 =                      494.1                      506.8

A3 =                      -333.1                      -341.7



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

Revision 2: 04-26-2023

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                          |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( α )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 4452                                  |                                                         | -                                           | -0.168                                             | 8.71                                     | 0.301                                          | -                                  |                                  |                             |
| 150                 | 4447                                  |                                                         | -                                           | -0.130                                             | 9.07                                     | 0.416                                          | -                                  |                                  |                             |
| 200                 | 4442                                  |                                                         | -                                           | -0.087                                             | 9.31                                     | 0.478                                          | -                                  |                                  |                             |
| 250                 | 4435                                  |                                                         | -                                           | -0.040                                             | 9.48                                     | 0.514                                          | -                                  |                                  |                             |
| 295                 | 4430                                  |                                                         | -                                           | 0.000                                              | 9.59                                     | 0.536                                          | -                                  |                                  |                             |
| 300                 | 4429                                  |                                                         | -                                           | 0.005                                              | 9.60                                     | 0.538                                          | 110.34                             |                                  |                             |
| 350                 | 4423                                  |                                                         | 7.56                                        | 0.055                                              | 9.69                                     | 0.557                                          | 106.62                             |                                  |                             |
| 400                 | 4416                                  |                                                         | 8.01                                        | 0.105                                              | 9.75                                     | 0.574                                          | 103.62                             |                                  |                             |
| 450                 | 4410                                  |                                                         | 8.55                                        | 0.154                                              | 9.80                                     | 0.588                                          | 101.09                             |                                  |                             |
| 500                 | 4403                                  |                                                         | 9.16                                        | 0.204                                              | 9.85                                     | 0.601                                          | 98.79                              |                                  |                             |
| 550                 | 4396                                  |                                                         | 9.82                                        | 0.254                                              | 9.89                                     | 0.612                                          | 96.46                              |                                  |                             |
| 600                 | 4390                                  |                                                         | 10.52                                       | 0.304                                              | 9.92                                     | 0.622                                          | 93.85                              |                                  |                             |
| 650                 | 4383                                  |                                                         | 11.24                                       | 0.355                                              | 9.96                                     | 0.631                                          | 90.71                              |                                  |                             |
| 700                 | 4376                                  |                                                         | 11.96                                       | 0.407                                              | 10.00                                    | 0.640                                          | 86.80                              |                                  |                             |
| 750                 | 4370                                  |                                                         | 12.67                                       | 0.459                                              | 10.04                                    | 0.648                                          | 81.85                              |                                  |                             |
| 800                 | 4363                                  |                                                         | 13.34                                       | 0.511                                              | 10.09                                    | 0.657                                          | 75.64                              |                                  |                             |
| 850                 | 4356                                  |                                                         | -                                           | 0.565                                              | 10.14                                    | 0.666                                          | -                                  |                                  |                             |
| 900                 | 4349                                  |                                                         | -                                           | 0.619                                              | 10.20                                    | 0.675                                          | -                                  |                                  |                             |
| 950                 | 4341                                  |                                                         | -                                           | 0.675                                              | 10.27                                    | -                                              | -                                  |                                  |                             |
| 1000                | 4334                                  |                                                         | -                                           | 0.732                                              | 10.35                                    | -                                              | -                                  |                                  |                             |
| 1050                | 4327                                  |                                                         | -                                           | 0.790                                              | 10.43                                    | -                                              | -                                  |                                  |                             |
| 1100                | 4319                                  |                                                         | -                                           | 0.849                                              | 10.52                                    | -                                              | -                                  |                                  |                             |
| 1150                | 4311                                  |                                                         | -                                           | 0.910                                              | 10.62                                    | -                                              | -                                  |                                  |                             |
| 1200                | 4303                                  |                                                         | -                                           | 0.973                                              | 10.73                                    | -                                              | -                                  |                                  |                             |
| 1250                | 4295                                  |                                                         | -                                           | 1.037                                              | 10.84                                    | -                                              | -                                  |                                  |                             |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

1 Light Metal Alloys

1.2 Titanium Alloys

1.2.1 Titanium – 6Al – 4V

**Revision 0: 08-05-2020**

**References**

[1] Material Universe. (Accessed July 2019).

[2] Y.S. Touloukian, R.W. Powell, C.Y. Ho, P.G. Klemens, Thermal Conductivity - Metallic Elements and Alloys, Thermophysical Properties of Matter - The TPRC Data Series, Vol. 1, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1970.

[3] MMPDS-12. (Accessed July 2019).

[4] Y.S. Touloukian, R.K. Kirby, R.E. Taylor, P.D. Desai, Thermal Expansion - Metallic Elements and Alloys, Thermophysical Properties of Matter - the TPRC Data Series, Vol. 12, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, web, 1975.

[5] Y.S. Touloukian, R.K. Kirby, R.E. Taylor, P.D. Desai, Specific Heat - Metallic Elements and Alloys, Thermophysical Properties of Matter - The TPRC Data Series, Vol. 4, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1971.

[6] MMPDS-13. (Accessed 06/19/2020).

## **2 Nickel – Base Alloys**

### **2.1 Precipitation Hardened Alloys**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718 (IN718)

Revision 2.1: 08-25-2023

General

## Room Temperature Properties

|                                           |       |
|-------------------------------------------|-------|
| Theoretical Density, [kg/m <sup>3</sup> ] | 8,220 |
| Melting Point, [K]                        | 1533  |
| Specific Heat, [J/(g-K)]                  | 0.420 |
| Thermal Conductivity, [W/(m-K)]           | 11.0  |
| Linear expansion coefficient, [μm/(m-K)]  | 12.09 |
| Electrical resistivity, [μΩ-m]            | 1.17  |
| Young's Modulus, [GPa]                    | 202.3 |
| Shear Modulus, [GPa]                      | 78.4  |
| Poisson's Ratio, [-]                      | 0.293 |

## Composition

Table 2.1.1-1: Typical Composition ranges for Inconel 718 (percent by weight) [1].

| Grade |      | C    | Mn    | P     | S     | Si   | Cr    | Ni   | Mo      | Nb   |
|-------|------|------|-------|-------|-------|------|-------|------|---------|------|
| IN718 | Min. | -    | -     | -     | -     | -    | 17.0  | 50.0 | 2.80    | 2.40 |
|       | Max. | 0.08 | 0.35  | 0.015 | 0.015 | 0.35 | 21.0  | 55.5 | 3.30    | 2.80 |
|       |      | Ta   | Nb+Ta | Ti    | Al    | Co   | B     | Cu   | Fe      |      |
|       | Min. | 2.40 | 4.75  | 0.65  | 0.20  | -    | -     | -    | Balance |      |
|       | Max. | 2.80 | 5.50  | 1.15  | 0.80  | 1.0  | 0.006 | 0.30 |         |      |
|       |      |      |       |       |       |      |       |      |         |      |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Density with Temperature

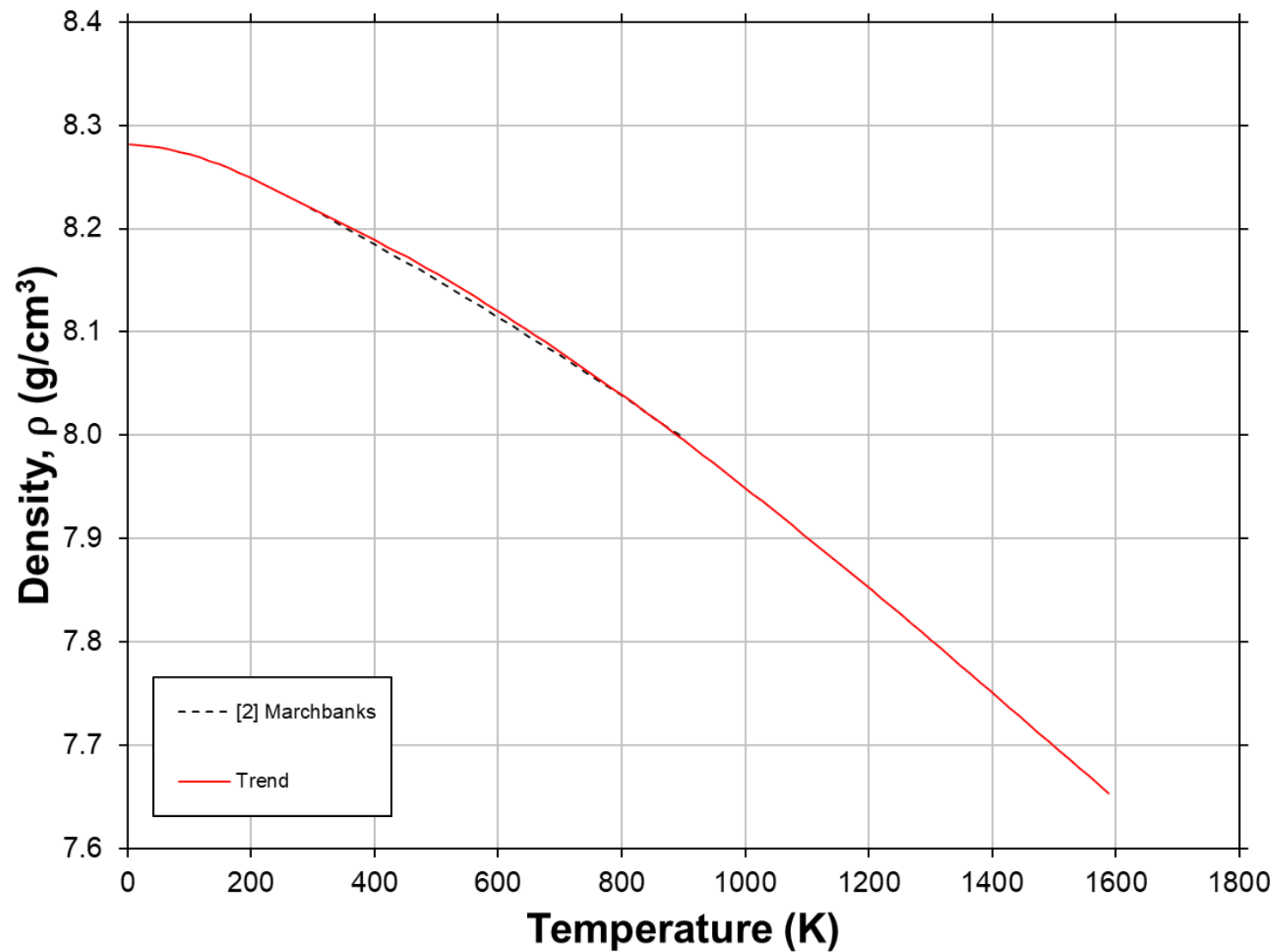


Figure 2.1.1-1: Density versus Temperature for Inconel 718. Calculated from fitted trend of the Thermal Expansion data with comparison to Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Density with Temperature

## 100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 5                 | ( -450.7 ) | 8281              | ( 517.0 )              | 700               | ( 800.3 )  | 8081              | ( 504.5 )              |
| 25                | ( -414.7 ) | 8281              | ( 517.0 )              | 750               | ( 890.3 )  | 8060              | ( 503.2 )              |
| 50                | ( -369.7 ) | 8279              | ( 516.8 )              | 800               | ( 980.3 )  | 8039              | ( 501.9 )              |
| 100               | ( -279.7 ) | 8272              | ( 516.4 )              | 850               | ( 1070.3 ) | 8017              | ( 500.5 )              |
| 150               | ( -189.7 ) | 8262              | ( 515.8 )              | 900               | ( 1160.3 ) | 7995              | ( 499.1 )              |
| 200               | ( -99.7 )  | 8249              | ( 515.0 )              | 950               | ( 1250.3 ) | 7972              | ( 497.7 )              |
| 250               | ( -9.7 )   | 8234              | ( 514.1 )              | 1000              | ( 1340.3 ) | 7949              | ( 496.3 )              |
| 300               | ( 80.3 )   | 8219              | ( 513.1 )              | 1050              | ( 1430.3 ) | 7926              | ( 494.8 )              |
| 350               | ( 170.3 )  | 8204              | ( 512.2 )              | 1100              | ( 1520.3 ) | 7901              | ( 493.3 )              |
| 400               | ( 260.3 )  | 8189              | ( 511.2 )              | 1150              | ( 1610.3 ) | 7877              | ( 491.8 )              |
| 450               | ( 350.3 )  | 8173              | ( 510.2 )              | 1200              | ( 1700.3 ) | 7852              | ( 490.2 )              |
| 500               | ( 440.3 )  | 8156              | ( 509.2 )              | 1300              | ( 1880.3 ) | 7802              | ( 487.1 )              |
| 550               | ( 530.3 )  | 8138              | ( 508.1 )              | 1400              | ( 2060.3 ) | 7751              | ( 483.9 )              |
| 600               | ( 620.3 )  | 8120              | ( 506.9 )              | 1500              | ( 2240.3 ) | 7699              | ( 480.7 )              |
| 650               | ( 710.3 )  | 8101              | ( 505.7 )              | 1588              | ( 2398.7 ) | 7653              | ( 477.8 )              |

**Application Notes:** Density trend is calculated as a function of thermal expansion, shown in the equation below, and compared against trend from reference [2].

### Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 8,220 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $4 \leq T \leq 1588$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

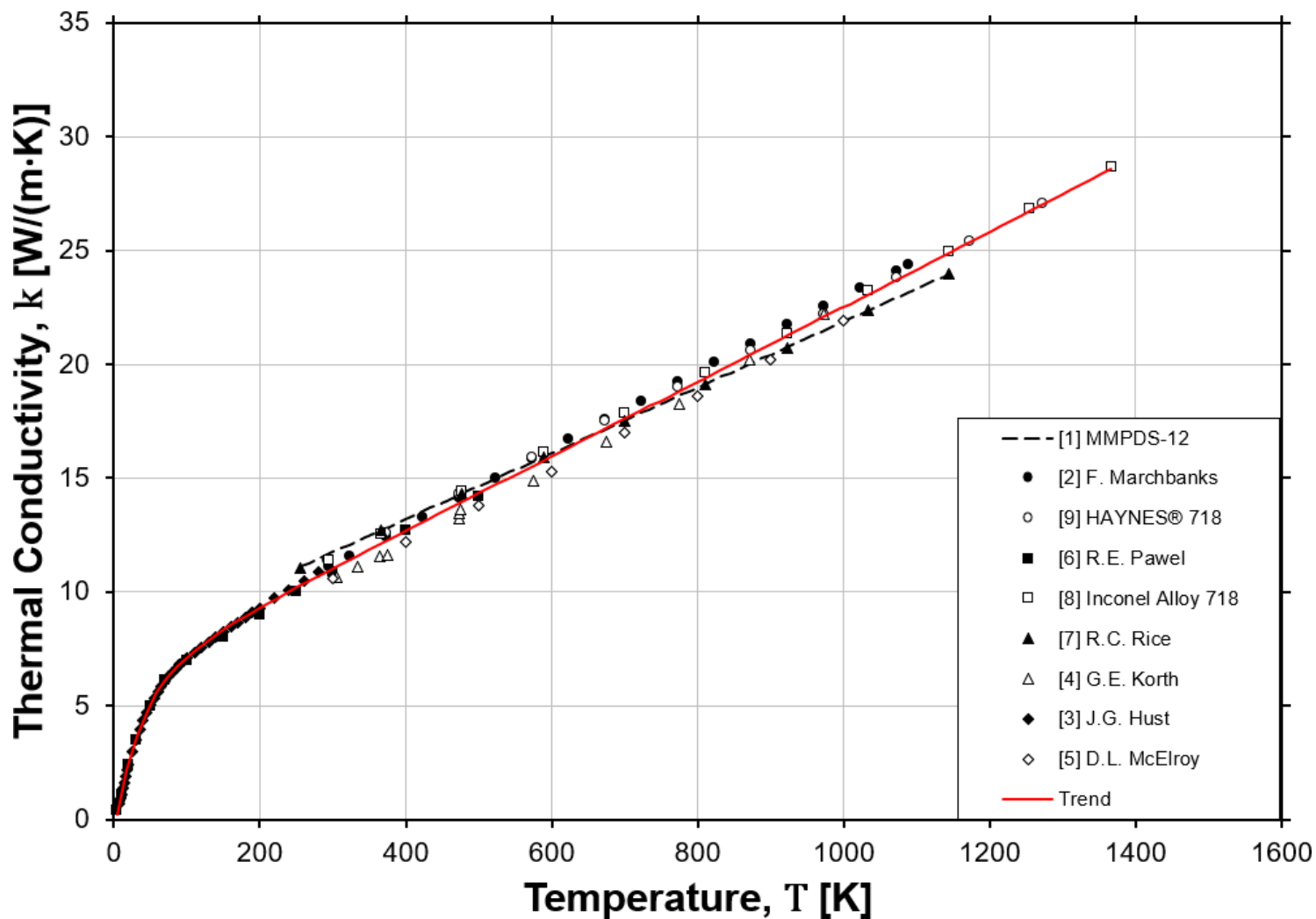


Figure 2.1.1-2: Thermal Conductivity versus Temperature of Inconel 718 with comparison to MMPDS-12.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 5                 | ( -450.7 ) | 0.38                       | ( 2.65 )                            | 700               | ( 800.3 )  | 17.60                      | ( 122.13 )                          |
| 25                | ( -414.7 ) | 2.96                       | ( 20.55 )                           | 750               | ( 890.3 )  | 18.42                      | ( 127.81 )                          |
| 50                | ( -369.7 ) | 5.04                       | ( 34.95 )                           | 800               | ( 980.3 )  | 19.24                      | ( 133.50 )                          |
| 100               | ( -279.7 ) | 7.10                       | ( 49.23 )                           | 850               | ( 1070.3 ) | 20.06                      | ( 139.19 )                          |
| 150               | ( -189.7 ) | 8.32                       | ( 57.72 )                           | 900               | ( 1160.3 ) | 20.88                      | ( 144.88 )                          |
| 200               | ( -99.7 )  | 9.29                       | ( 64.47 )                           | 950               | ( 1250.3 ) | 21.70                      | ( 150.58 )                          |
| 250               | ( -9.7 )   | 10.18                      | ( 70.61 )                           | 1000              | ( 1340.3 ) | 22.53                      | ( 156.29 )                          |
| 300               | ( 80.3 )   | 11.03                      | ( 76.51 )                           | 1050              | ( 1430.3 ) | 23.35                      | ( 162.00 )                          |
| 350               | ( 170.3 )  | 11.86                      | ( 82.29 )                           | 1100              | ( 1520.3 ) | 24.17                      | ( 167.72 )                          |
| 400               | ( 260.3 )  | 12.69                      | ( 88.03 )                           | 1150              | ( 1610.3 ) | 25.00                      | ( 173.44 )                          |
| 450               | ( 350.3 )  | 13.51                      | ( 93.73 )                           | 1200              | ( 1700.3 ) | 25.82                      | ( 179.17 )                          |
| 500               | ( 440.3 )  | 14.33                      | ( 99.42 )                           | 1250              | ( 1790.3 ) | 26.65                      | ( 184.91 )                          |
| 550               | ( 530.3 )  | 15.15                      | ( 105.10 )                          | 1300              | ( 1880.3 ) | 27.48                      | ( 190.65 )                          |
| 600               | ( 620.3 )  | 15.97                      | ( 110.77 )                          | 1350              | ( 1970.3 ) | 28.31                      | ( 196.40 )                          |
| 650               | ( 710.3 )  | 16.78                      | ( 116.45 )                          | 1367              | ( 2000.9 ) | 28.59                      | ( 198.35 )                          |

**Application Notes:** Data for thermal conductivity is collected from references [1-9] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equation:

$$k(T) = A_0 + A_1 \cdot \left(\frac{T}{1000}\right) + A_2 \cdot \left(\frac{T}{1000}\right)^2 + A_3 \cdot \left(\frac{T}{1000}\right)^3 + A_4 \cdot \left(\frac{T}{1000}\right)^4 + A_2 / \left(\frac{T}{1000}\right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

| T Range [K]: | $4 \leq T < 115$ | $115 \leq T \leq 1367$ |
|--------------|------------------|------------------------|
| A0 =         | -4.293E-01       | 6.290E+00              |
| A1 =         | 1.699E+02        | 1.605E+01              |
| A2 =         | -1.546E+03       | 1.944E-01              |
| A3 =         | 7.385E+03        | 0                      |
| A4 =         | -1.390E+04       | 0                      |
| A_2 =        | 0                | -8.603E-03             |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Thermal Expansion with Temperature

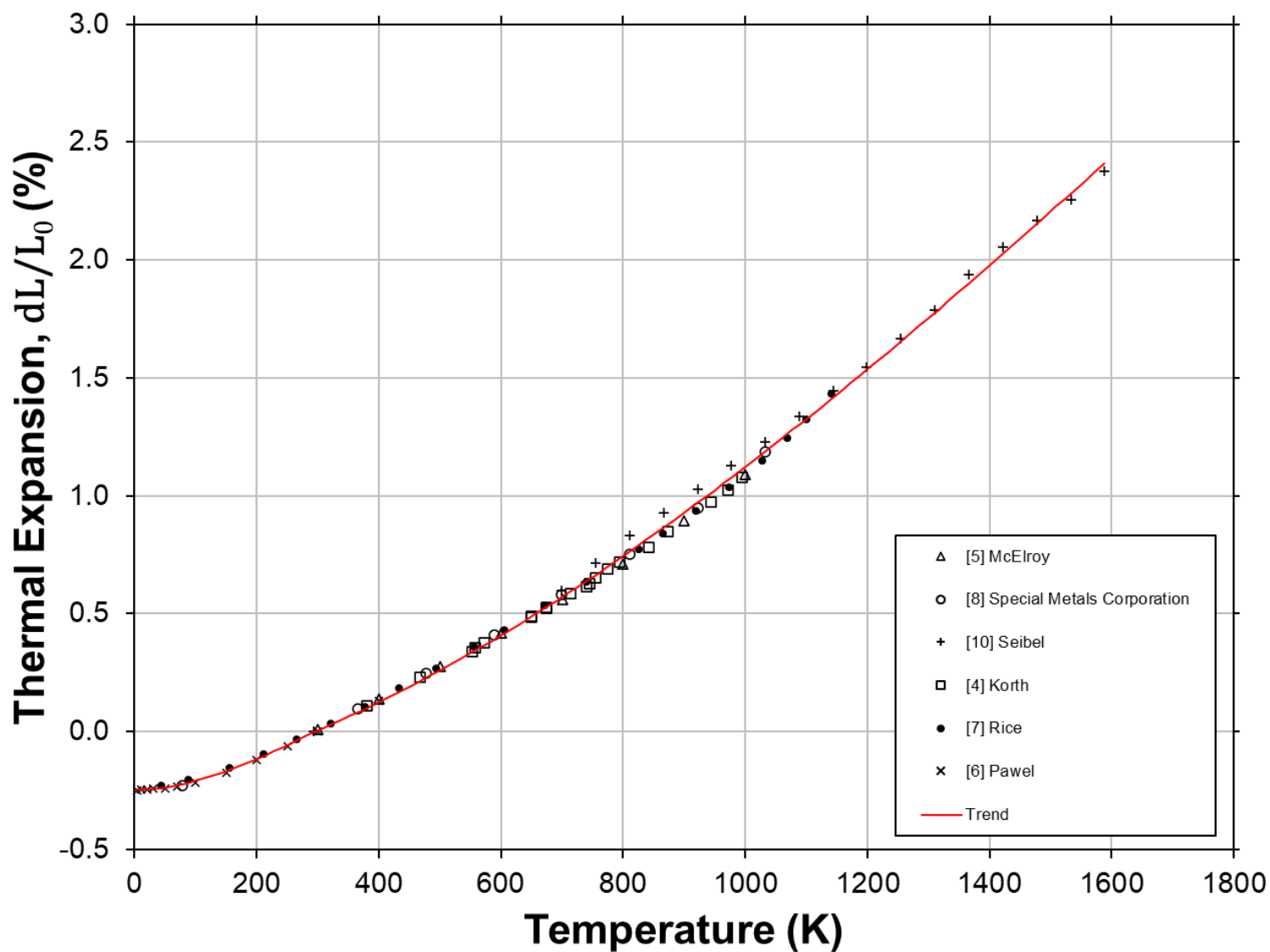


Figure 2.1.1-3: Thermal Expansion versus Temperature of Inconel 718.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 5                 | ( -450.7 ) | -0.247                                  | 700               | ( 800.3 )  | 0.571                                   |
| 25                | ( -414.7 ) | -0.244                                  | 750               | ( 890.3 )  | 0.656                                   |
| 50                | ( -369.7 ) | -0.237                                  | 800               | ( 980.3 )  | 0.744                                   |
| 100               | ( -279.7 ) | -0.209                                  | 850               | ( 1070.3 ) | 0.835                                   |
| 150               | ( -189.7 ) | -0.168                                  | 900               | ( 1160.3 ) | 0.929                                   |
| 200               | ( -99.7 )  | -0.116                                  | 950               | ( 1250.3 ) | 1.025                                   |
| 250               | ( -9.7 )   | -0.057                                  | 1000              | ( 1340.3 ) | 1.123                                   |
| 300               | ( 80.3 )   | 0.006                                   | 1050              | ( 1430.3 ) | 1.223                                   |
| 350               | ( 170.3 )  | 0.064                                   | 1100              | ( 1520.3 ) | 1.326                                   |
| 400               | ( 260.3 )  | 0.125                                   | 1150              | ( 1610.3 ) | 1.431                                   |
| 450               | ( 350.3 )  | 0.191                                   | 1200              | ( 1700.3 ) | 1.537                                   |
| 500               | ( 440.3 )  | 0.260                                   | 1300              | ( 1880.3 ) | 1.754                                   |
| 550               | ( 530.3 )  | 0.333                                   | 1400              | ( 2060.3 ) | 1.978                                   |
| 600               | ( 620.3 )  | 0.409                                   | 1500              | ( 2240.3 ) | 2.206                                   |
| 650               | ( 710.3 )  | 0.488                                   | 1588              | ( 2398.7 ) | 2.410                                   |

**Application Notes:** Data for thermal expansion is collected from [4-8, 10] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

| T Range [K]: | <u>4 &lt; T ≤ 294</u> | <u>294 &lt; T ≤ 1588</u> |
|--------------|-----------------------|--------------------------|
| A0 =         | -2.477E-01            | -2.537E-01               |
| A1 =         | 3.602E-02             | 5.934E-01                |
| A2 =         | 3.907E+00             | 9.545E-01                |
| A3 =         | -4.015E+00            | -1.713E-01               |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

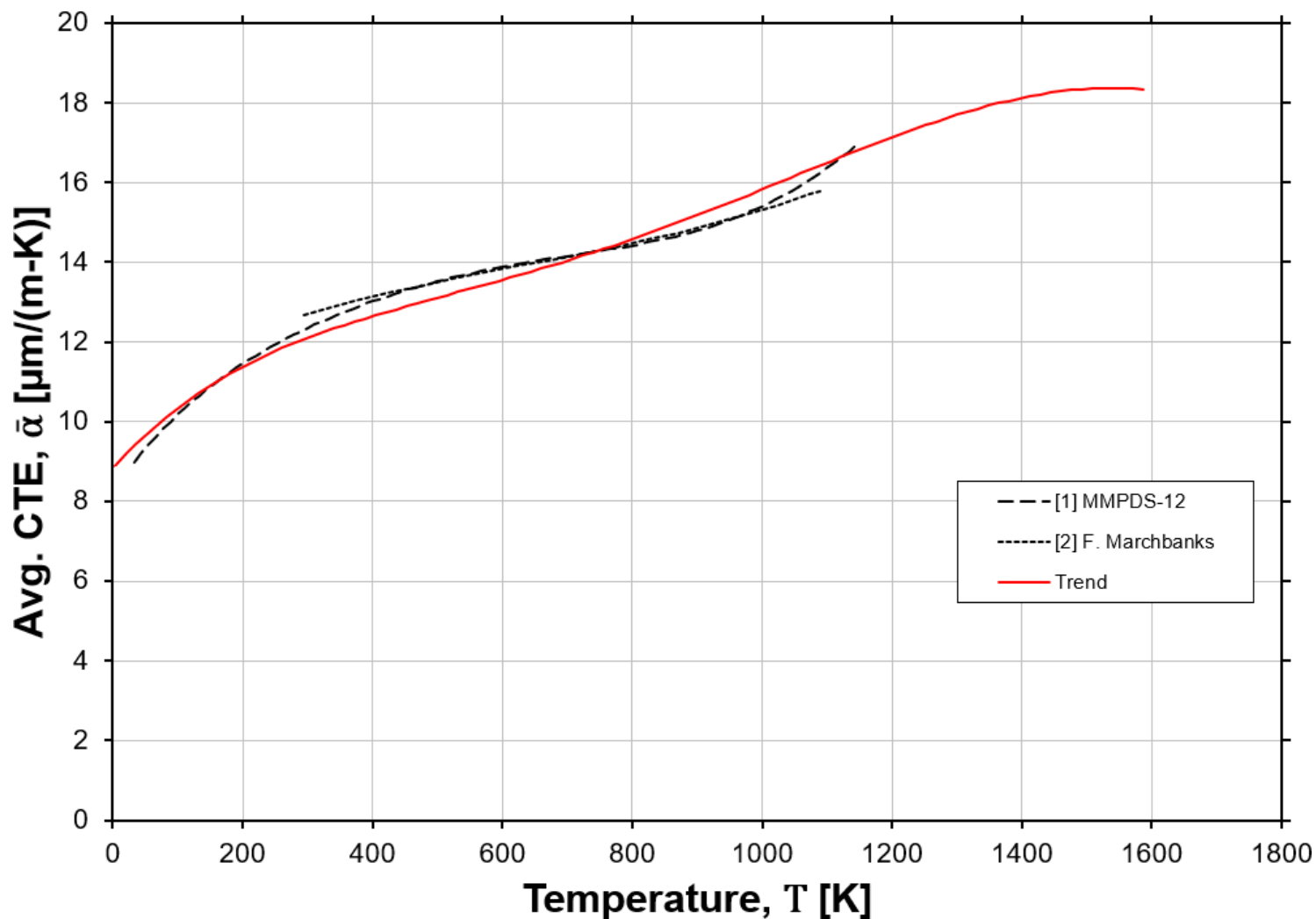


Figure 2.1.1-4: Mean Coefficient of Thermal Expansion versus Temperature of Inconel 718, with comparison to MMPDS-12 and Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 2: 04-26-2023

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 5                 | ( -450.7 ) | 8.904                                 | ( 4.946 )                                           | 700               | ( 800.3 )  | 14.040                                | ( 7.800 )                                           |
| 25                | ( -414.7 ) | 9.249                                 | ( 5.138 )                                           | 750               | ( 890.3 )  | 14.302                                | ( 7.946 )                                           |
| 50                | ( -369.7 ) | 9.646                                 | ( 5.359 )                                           | 800               | ( 980.3 )  | 14.580                                | ( 8.100 )                                           |
| 100               | ( -279.7 ) | 10.336                                | ( 5.742 )                                           | 850               | ( 1070.3 ) | 14.874                                | ( 8.264 )                                           |
| 150               | ( -189.7 ) | 10.906                                | ( 6.059 )                                           | 900               | ( 1160.3 ) | 15.182                                | ( 8.435 )                                           |
| 200               | ( -99.7 )  | 11.379                                | ( 6.321 )                                           | 950               | ( 1250.3 ) | 15.502                                | ( 8.612 )                                           |
| 250               | ( -9.7 )   | 11.773                                | ( 6.540 )                                           | 1000              | ( 1340.3 ) | 15.831                                | ( 8.795 )                                           |
| 300               | ( 80.3 )   | 12.106                                | ( 6.726 )                                           | 1050              | ( 1430.3 ) | 16.164                                | ( 8.980 )                                           |
| 350               | ( 170.3 )  | 12.394                                | ( 6.885 )                                           | 1100              | ( 1520.3 ) | 16.496                                | ( 9.165 )                                           |
| 400               | ( 260.3 )  | 12.649                                | ( 7.027 )                                           | 1150              | ( 1610.3 ) | 16.822                                | ( 9.346 )                                           |
| 450               | ( 350.3 )  | 12.884                                | ( 7.158 )                                           | 1200              | ( 1700.3 ) | 17.136                                | ( 9.520 )                                           |
| 500               | ( 440.3 )  | 13.109                                | ( 7.283 )                                           | 1300              | ( 1880.3 ) | 17.699                                | ( 9.833 )                                           |
| 550               | ( 530.3 )  | 13.331                                | ( 7.406 )                                           | 1400              | ( 2060.3 ) | 18.125                                | ( 10.070 )                                          |
| 600               | ( 620.3 )  | 13.557                                | ( 7.532 )                                           | 1500              | ( 2240.3 ) | 18.352                                | ( 10.195 )                                          |
| 650               | ( 710.3 )  | 13.792                                | ( 7.662 )                                           | 1588              | ( 2398.7 ) | 18.334                                | ( 10.186 )                                          |

**Application Notes:** Mean coefficient of thermal expansion is calculated as a function of thermal expansion. Calculated data has been fitted with the equation below to approximate property trend with respect to temperature, and compared to references [1, 2].

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3 + A4 \cdot \left(\frac{T}{1000}\right)^4 + A5 \cdot \left(\frac{T}{1000}\right)^5$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $4 < T \leq 1588$

A0 = 8.813

A1 = 18.27

A2 = -33.91

A3 = 36.66

A4 = -16.21

A5 = 2.208



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Specific Heat with Temperature

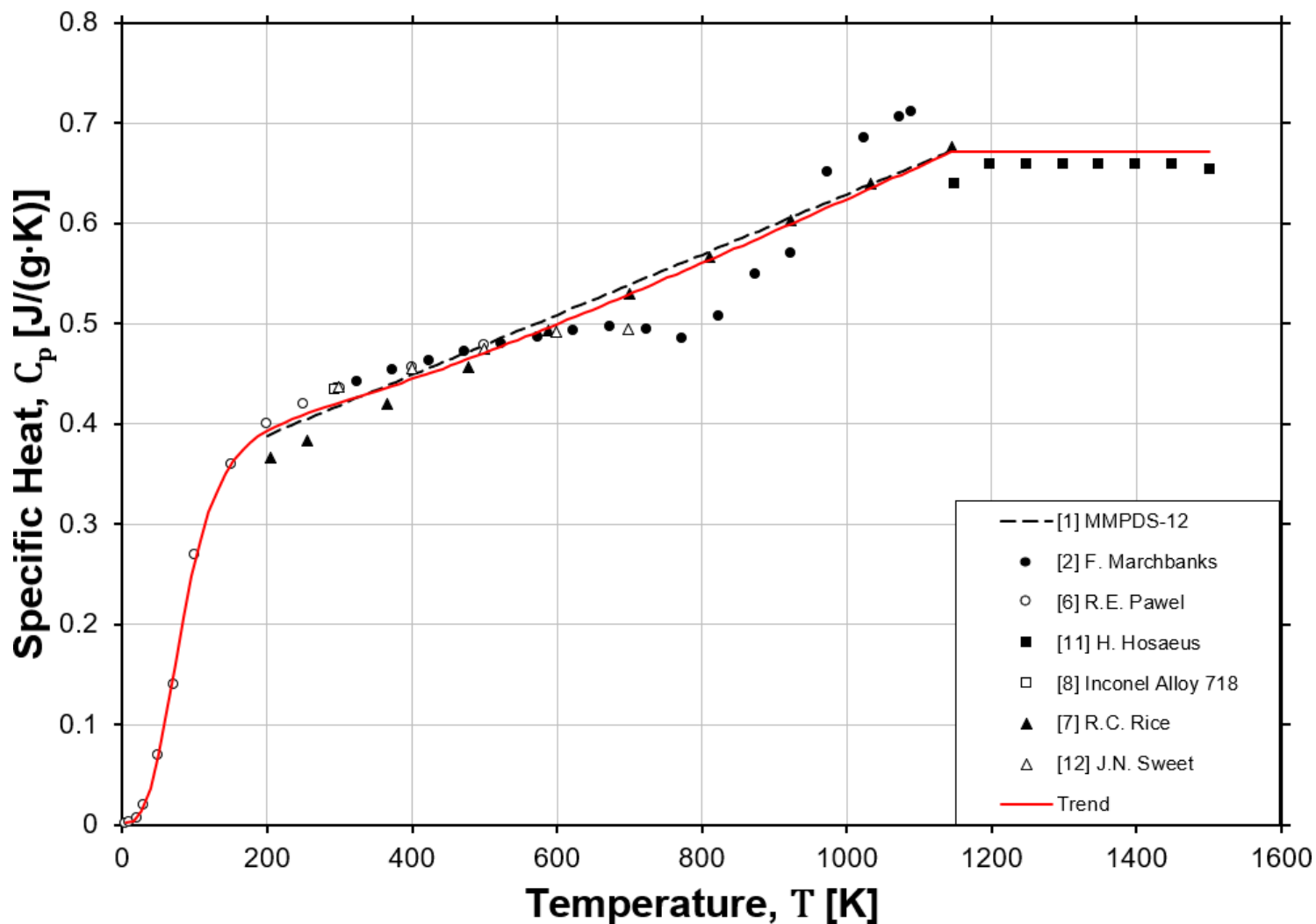


Figure 2.1.1-5: Specific Heat versus Temperature of Inconel 718 with comparison to MMPDS-12.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 5                 | ( -450.7 ) | 0.002                            | ( 0.001 )       | 700               | ( 800.3 )  | 0.529                            | ( 0.127 )       |
| 25                | ( -414.7 ) | 0.011                            | ( 0.003 )       | 750               | ( 890.3 )  | 0.545                            | ( 0.130 )       |
| 50                | ( -369.7 ) | 0.069                            | ( 0.017 )       | 800               | ( 980.3 )  | 0.560                            | ( 0.134 )       |
| 100               | ( -279.7 ) | 0.261                            | ( 0.062 )       | 850               | ( 1070.3 ) | 0.576                            | ( 0.138 )       |
| 150               | ( -189.7 ) | 0.360                            | ( 0.086 )       | 900               | ( 1160.3 ) | 0.592                            | ( 0.142 )       |
| 200               | ( -99.7 )  | 0.393                            | ( 0.094 )       | 950               | ( 1250.3 ) | 0.608                            | ( 0.145 )       |
| 250               | ( -9.7 )   | 0.409                            | ( 0.098 )       | 1000              | ( 1340.3 ) | 0.624                            | ( 0.149 )       |
| 300               | ( 80.3 )   | 0.421                            | ( 0.101 )       | 1050              | ( 1430.3 ) | 0.641                            | ( 0.153 )       |
| 350               | ( 170.3 )  | 0.432                            | ( 0.103 )       | 1100              | ( 1520.3 ) | 0.657                            | ( 0.157 )       |
| 400               | ( 260.3 )  | 0.445                            | ( 0.106 )       | 1150              | ( 1610.3 ) | 0.671                            | ( 0.160 )       |
| 450               | ( 350.3 )  | 0.458                            | ( 0.109 )       | 1200              | ( 1700.3 ) | 0.671                            | ( 0.160 )       |
| 500               | ( 440.3 )  | 0.471                            | ( 0.113 )       | 1300              | ( 1880.3 ) | 0.671                            | ( 0.160 )       |
| 550               | ( 530.3 )  | 0.485                            | ( 0.116 )       | 1400              | ( 2060.3 ) | 0.671                            | ( 0.160 )       |
| 600               | ( 620.3 )  | 0.500                            | ( 0.119 )       | 1500              | ( 2240.3 ) | 0.671                            | ( 0.160 )       |
| 650               | ( 710.3 )  | 0.514                            | ( 0.123 )       |                   |            |                                  |                 |

**Application Notes:** Data for specific heat is collected from references [2, 6-8, 11, 12] and fitted with the equation below to approximate the property trend with respect to temperature. Trend is compared to reference [1].

## Fit Equation:

$$C_p(T) = \left[ A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 + A3 \cdot \left( \frac{T}{1000} \right)^3 \right] / \left[ B0 + B1 \cdot \left( \frac{T}{1000} \right) + \left( \frac{T}{1000} \right)^2 \right]$$

$C_p(T)$  = Specific Heat [J/(g · K)]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $4 \leq T \leq 1150$

A0 = 3.356E-05

A1 = -3.836E-03

A2 = 2.230E-01

A3 = 3.441E-01

B0 = 9.237E-03

B1 = -1.071E-01

$C_p(T) = \frac{1150 \leq T \leq 1500}{0.6715}$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

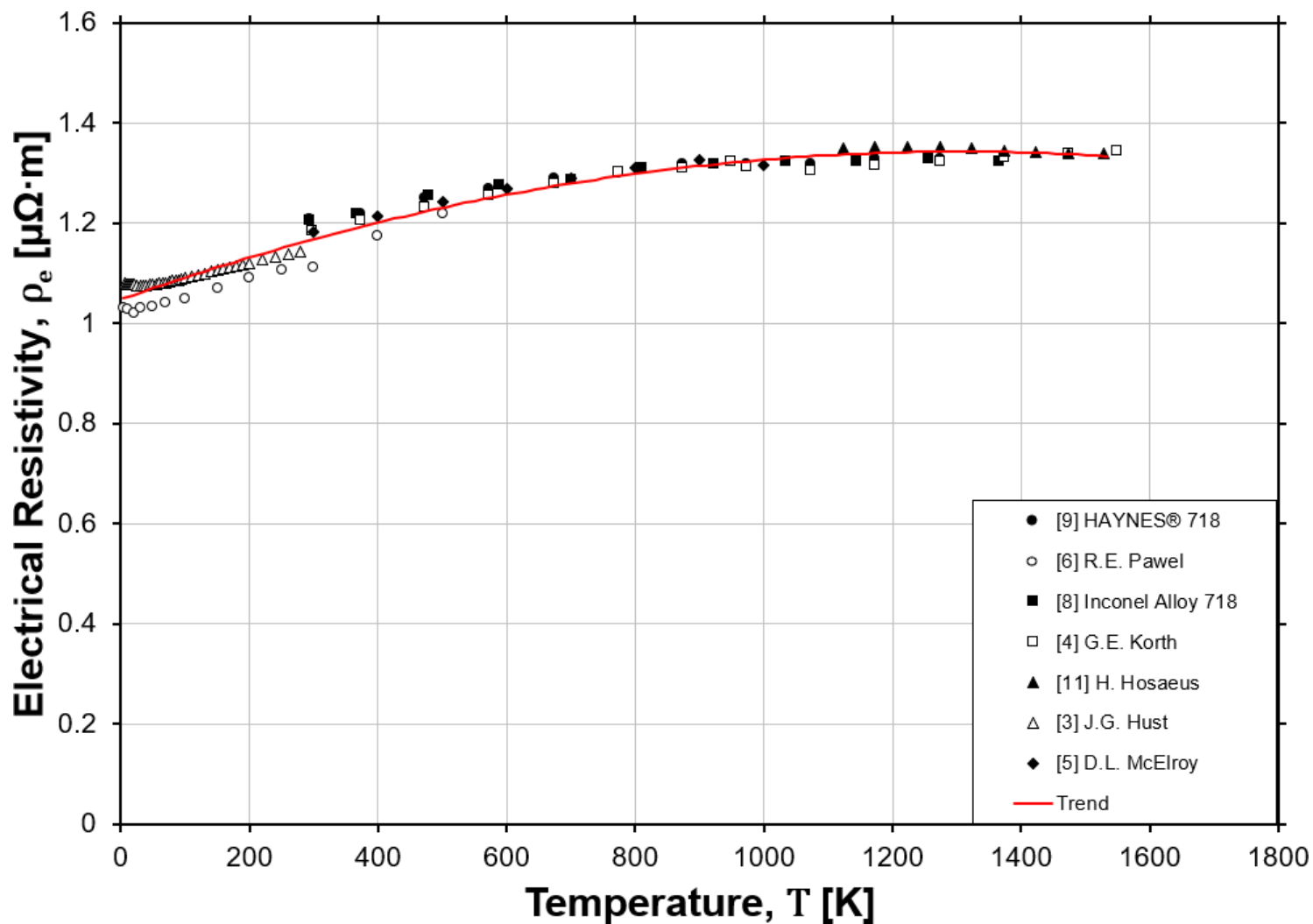


Figure 2.1.1-6: Electrical Resistivity versus Temperature of Inconel 718.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 5                 | ( -450.7 ) | 1.050                               | ( 41.35 )               | 700               | ( 800.3 )  | 1.280                               | ( 50.39 )               |
| 25                | ( -414.7 ) | 1.059                               | ( 41.70 )               | 750               | ( 890.3 )  | 1.290                               | ( 50.78 )               |
| 50                | ( -369.7 ) | 1.070                               | ( 42.13 )               | 800               | ( 980.3 )  | 1.299                               | ( 51.15 )               |
| 100               | ( -279.7 ) | 1.092                               | ( 42.97 )               | 850               | ( 1070.3 ) | 1.307                               | ( 51.47 )               |
| 150               | ( -189.7 ) | 1.112                               | ( 43.78 )               | 900               | ( 1160.3 ) | 1.315                               | ( 51.77 )               |
| 200               | ( -99.7 )  | 1.132                               | ( 44.55 )               | 950               | ( 1250.3 ) | 1.321                               | ( 52.03 )               |
| 250               | ( -9.7 )   | 1.150                               | ( 45.29 )               | 1000              | ( 1340.3 ) | 1.327                               | ( 52.25 )               |
| 300               | ( 80.3 )   | 1.168                               | ( 45.99 )               | 1050              | ( 1430.3 ) | 1.332                               | ( 52.44 )               |
| 350               | ( 170.3 )  | 1.185                               | ( 46.66 )               | 1100              | ( 1520.3 ) | 1.336                               | ( 52.60 )               |
| 400               | ( 260.3 )  | 1.201                               | ( 47.30 )               | 1150              | ( 1610.3 ) | 1.339                               | ( 52.72 )               |
| 450               | ( 350.3 )  | 1.217                               | ( 47.90 )               | 1200              | ( 1700.3 ) | 1.341                               | ( 52.81 )               |
| 500               | ( 440.3 )  | 1.231                               | ( 48.46 )               | 1300              | ( 1880.3 ) | 1.343                               | ( 52.89 )               |
| 550               | ( 530.3 )  | 1.245                               | ( 49.00 )               | 1400              | ( 2060.3 ) | 1.342                               | ( 52.82 )               |
| 600               | ( 620.3 )  | 1.257                               | ( 49.49 )               | 1500              | ( 2240.3 ) | 1.337                               | ( 52.63 )               |
| 650               | ( 710.3 )  | 1.269                               | ( 49.96 )               | 1548              | ( 2326.7 ) | 1.333                               | ( 52.48 )               |

**Application Notes:** Data for electrical resistivity is collected from references [3-6, 8, 9, 11] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$\rho_e(T)$  = Electrical Resistivity ( $\mu\Omega \cdot m$ )

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $4 \leq T \leq 1548$

A0 = 1.048

A1 = 0.4527

A2 = -0.1735



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Young's Modulus with Temperature

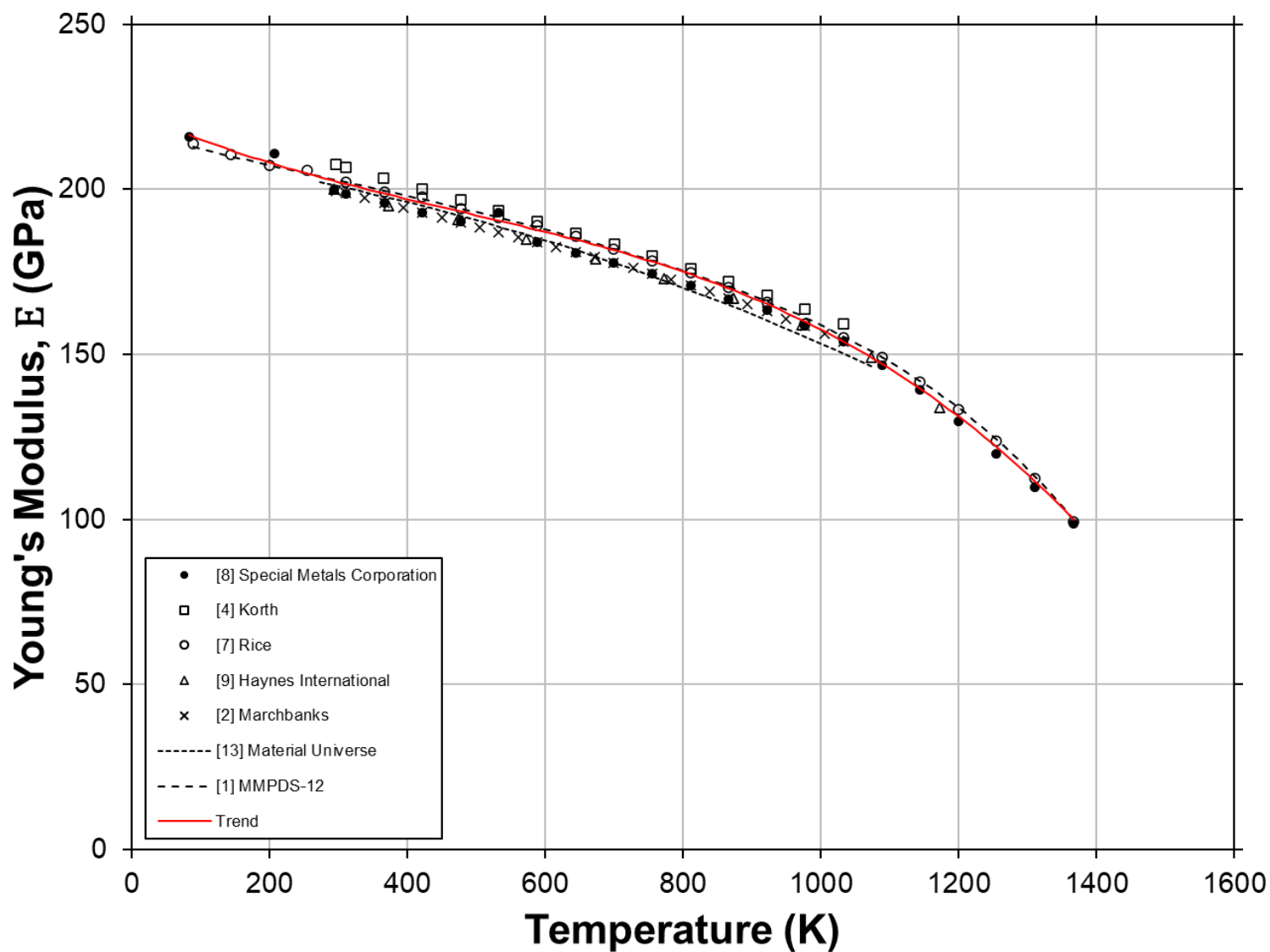


Figure 2.1.1-7: Young's Modulus versus Temperature of Inconel 718 with comparison to MMPDS-12, solution-treated and aged.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 84                | ( -308.5 ) | 216.46                | ( 31.41 ) | 800               | ( 980.3 )  | 175.03                | ( 25.40 ) |
| 100               | ( -279.7 ) | 215.19                | ( 31.22 ) | 850               | ( 1070.3 ) | 171.29                | ( 24.85 ) |
| 150               | ( -189.7 ) | 211.50                | ( 30.69 ) | 900               | ( 1160.3 ) | 167.15                | ( 24.25 ) |
| 200               | ( -99.7 )  | 208.14                | ( 30.20 ) | 950               | ( 1250.3 ) | 162.56                | ( 23.59 ) |
| 250               | ( -9.7 )   | 205.06                | ( 29.75 ) | 1000              | ( 1340.3 ) | 157.48                | ( 22.85 ) |
| 300               | ( 80.3 )   | 202.22                | ( 29.34 ) | 1050              | ( 1430.3 ) | 151.85                | ( 22.03 ) |
| 350               | ( 170.3 )  | 199.55                | ( 28.96 ) | 1100              | ( 1520.3 ) | 145.61                | ( 21.13 ) |
| 400               | ( 260.3 )  | 197.02                | ( 28.59 ) | 1150              | ( 1610.3 ) | 138.71                | ( 20.13 ) |
| 450               | ( 350.3 )  | 194.55                | ( 28.23 ) | 1200              | ( 1700.3 ) | 131.11                | ( 19.02 ) |
| 500               | ( 440.3 )  | 192.11                | ( 27.88 ) | 1250              | ( 1790.3 ) | 122.75                | ( 17.81 ) |
| 550               | ( 530.3 )  | 189.64                | ( 27.52 ) | 1300              | ( 1880.3 ) | 113.57                | ( 16.48 ) |
| 600               | ( 620.3 )  | 187.09                | ( 27.15 ) | 1350              | ( 1970.3 ) | 103.53                | ( 15.02 ) |
| 650               | ( 710.3 )  | 184.41                | ( 26.76 ) | 1367              | ( 2000.9 ) | 99.91                 | ( 14.50 ) |
| 700               | ( 800.3 )  | 181.54                | ( 26.34 ) |                   |            |                       |           |
| 750               | ( 890.3 )  | 178.43                | ( 25.89 ) |                   |            |                       |           |

**Application Notes:** Data for Young's Modulus is collected from references [1, 2, 4, 7-9, 13] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equations:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]: 84 < T ≤ 1367

A0 = 223.8

A1 = -95.22

A2 = 98.49

A3 = -69.59



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Shear Modulus with Temperature

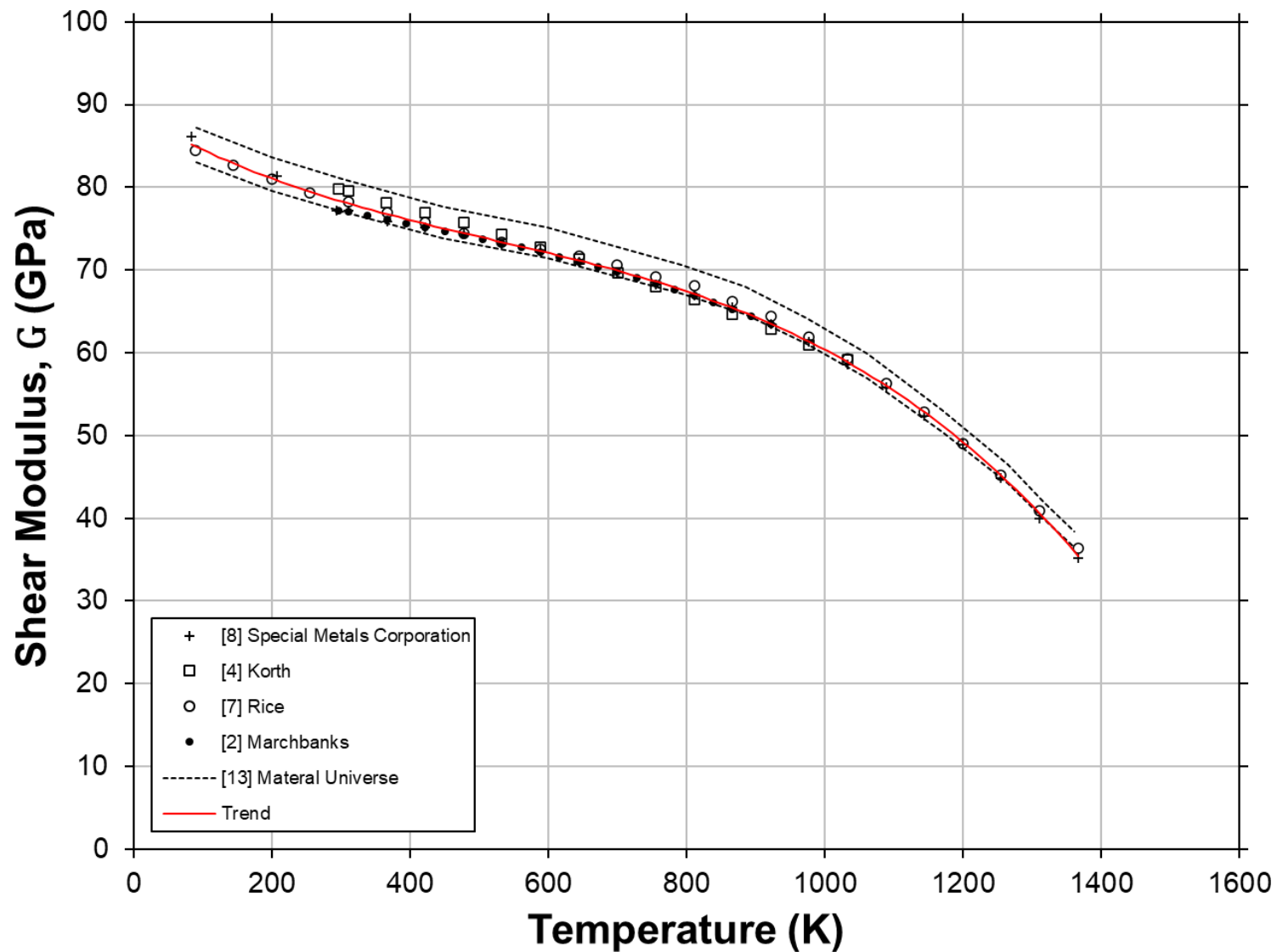


Figure 2.1.1-8: Shear Modulus versus Temperature of Inconel 718 with comparison to Material Universe High and Low trends for Inconel 718.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( G ) |           | Temperature ( T ) |            | Shear Modulus ( G ) |          |
|-------------------|------------|---------------------|-----------|-------------------|------------|---------------------|----------|
| K                 | ( °F )     | GPa                 | ( Msi )   | K                 | ( °F )     | GPa                 | ( Msi )  |
| 84                | ( -308.5 ) | 85.19               | ( 12.36 ) | 800               | ( 980.3 )  | 67.41               | ( 9.78 ) |
| 100               | ( -279.7 ) | 84.55               | ( 12.27 ) | 850               | ( 1070.3 ) | 65.94               | ( 9.57 ) |
| 150               | ( -189.7 ) | 82.72               | ( 12.00 ) | 900               | ( 1160.3 ) | 64.29               | ( 9.33 ) |
| 200               | ( -99.7 )  | 81.08               | ( 11.77 ) | 950               | ( 1250.3 ) | 62.44               | ( 9.06 ) |
| 250               | ( -9.7 )   | 79.63               | ( 11.55 ) | 1000              | ( 1340.3 ) | 60.36               | ( 8.76 ) |
| 300               | ( 80.3 )   | 78.32               | ( 11.36 ) | 1050              | ( 1430.3 ) | 58.02               | ( 8.42 ) |
| 350               | ( 170.3 )  | 77.13               | ( 11.19 ) | 1100              | ( 1520.3 ) | 55.39               | ( 8.04 ) |
| 400               | ( 260.3 )  | 76.03               | ( 11.03 ) | 1150              | ( 1610.3 ) | 52.45               | ( 7.61 ) |
| 450               | ( 350.3 )  | 75.01               | ( 10.88 ) | 1200              | ( 1700.3 ) | 49.17               | ( 7.14 ) |
| 500               | ( 440.3 )  | 74.02               | ( 10.74 ) | 1250              | ( 1790.3 ) | 45.53               | ( 6.61 ) |
| 550               | ( 530.3 )  | 73.04               | ( 10.60 ) | 1300              | ( 1880.3 ) | 41.49               | ( 6.02 ) |
| 600               | ( 620.3 )  | 72.05               | ( 10.46 ) | 1350              | ( 1970.3 ) | 37.02               | ( 5.37 ) |
| 650               | ( 710.3 )  | 71.02               | ( 10.31 ) | 1367              | ( 2000.9 ) | 35.41               | ( 5.14 ) |
| 700               | ( 800.3 )  | 69.92               | ( 10.15 ) |                   |            |                     |          |
| 750               | ( 890.3 )  | 68.73               | ( 9.97 )  |                   |            |                     |          |

**Application Notes:** Data for shear modulus is collected from references [2, 4, 7, 8, 13] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$G(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $84 \leq T \leq 1367$

A0 = 88.94

A1 = -49.19

A2 = 56.78

A3 = -36.17



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

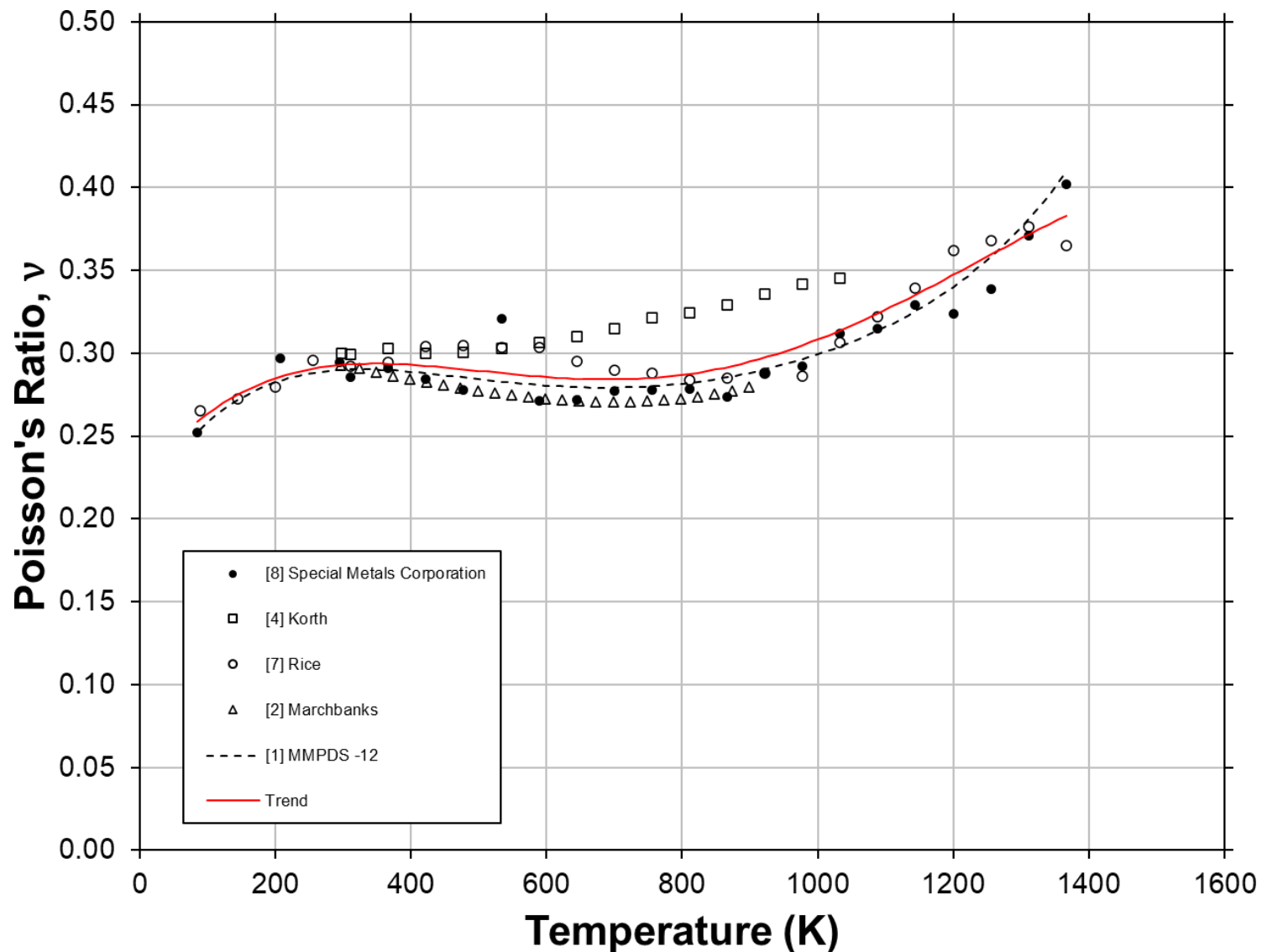


Figure 2.1.1-9: Poisson's Ratio versus Temperature of Inconel 718 with comparison to MMPDS-12, solution-treated and aged.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|------------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )     |                           | K                 | ( °F )     |                           |
| 84                | ( -308.5 ) | 0.259                     | 800               | ( 980.3 )  | 0.287                     |
| 100               | ( -279.7 ) | 0.264                     | 850               | ( 1070.3 ) | 0.290                     |
| 150               | ( -189.7 ) | 0.276                     | 900               | ( 1160.3 ) | 0.295                     |
| 200               | ( -99.7 )  | 0.285                     | 950               | ( 1250.3 ) | 0.301                     |
| 250               | ( -9.7 )   | 0.290                     | 1000              | ( 1340.3 ) | 0.308                     |
| 300               | ( 80.3 )   | 0.293                     | 1050              | ( 1430.3 ) | 0.317                     |
| 350               | ( 170.3 )  | 0.294                     | 1100              | ( 1520.3 ) | 0.326                     |
| 400               | ( 260.3 )  | 0.293                     | 1150              | ( 1610.3 ) | 0.336                     |
| 450               | ( 350.3 )  | 0.291                     | 1200              | ( 1700.3 ) | 0.347                     |
| 500               | ( 440.3 )  | 0.289                     | 1250              | ( 1790.3 ) | 0.358                     |
| 550               | ( 530.3 )  | 0.287                     | 1300              | ( 1880.3 ) | 0.369                     |
| 600               | ( 620.3 )  | 0.286                     | 1350              | ( 1970.3 ) | 0.380                     |
| 650               | ( 710.3 )  | 0.284                     | 1367              | ( 2000.9 ) | 0.383                     |
| 700               | ( 800.3 )  | 0.284                     |                   |            |                           |
| 750               | ( 890.3 )  | 0.285                     |                   |            |                           |

**Application Notes:** Data for Poisson's Ratio is collected from references [1, 2, 4, 7, 8] and fitted with the equation below to approximate the property trend with respect to temperature. Fitted trend is compared against trend from reference [1].

## Fit Equations:

$$\nu(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3 + A4 \cdot \left(\frac{T}{1000}\right)^4$$

$\nu(T)$  = Poisson's Ratio

$T$  = Temperature [K]

## Constants:

T Range [K]: 84 < T ≤ 1367

A0 = 0.2236  
A1 = 0.5184  
A2 = -1.286  
A3 = 1.197  
A4 = -0.3448



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Yield Strength with Temperature

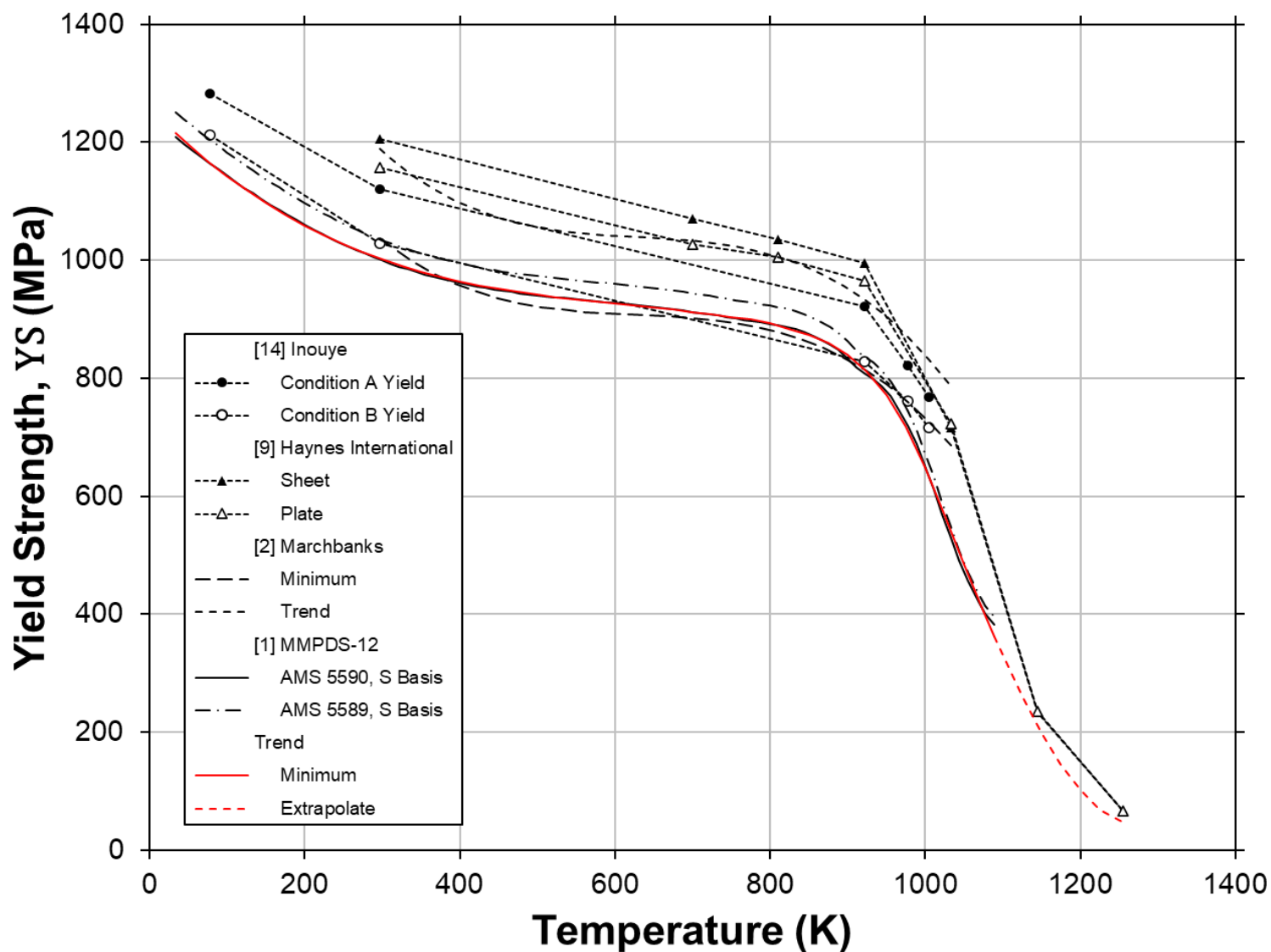


Figure 2.1.1-10: Yield Strength versus Temperature of Inconel 718, with comparison to MMPDS-12, solution-treated and aged wrought product.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Yield Strength with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Yield Strength ( YS ) |            | Temperature ( T ) |            | Yield Strength ( YS ) |            |
|-------------------|------------|-----------------------|------------|-------------------|------------|-----------------------|------------|
| K                 | ( °F )     | MPa                   | ( Ksi )    | K                 | ( °F )     | MPa                   | ( Ksi )    |
| 33                | ( -400.3 ) | 1215.69               | ( 176.40 ) | 650               | ( 710.3 )  | 919.46                | ( 133.41 ) |
| 50                | ( -369.7 ) | 1195.48               | ( 173.46 ) | 700               | ( 800.3 )  | 912.39                | ( 132.39 ) |
| 75                | ( -324.7 ) | 1167.61               | ( 169.42 ) | 750               | ( 890.3 )  | 904.14                | ( 131.19 ) |
| 100               | ( -279.7 ) | 1141.88               | ( 165.69 ) | 800               | ( 980.3 )  | 893.94                | ( 129.71 ) |
| 150               | ( -189.7 ) | 1096.45               | ( 159.09 ) | 850               | ( 1070.3 ) | 873.87                | ( 126.80 ) |
| 200               | ( -99.7 )  | 1058.38               | ( 153.57 ) | 900               | ( 1160.3 ) | 840.52                | ( 121.96 ) |
| 250               | ( -9.7 )   | 1026.92               | ( 149.01 ) | 950               | ( 1250.3 ) | 772.05                | ( 112.02 ) |
| 300               | ( 80.3 )   | 1001.28               | ( 145.29 ) | 1000              | ( 1340.3 ) | 650.00                | ( 94.32 )  |
| 350               | ( 170.3 )  | 980.68                | ( 142.30 ) | 1050              | ( 1430.3 ) | 484.62                | ( 70.32 )  |
| 400               | ( 260.3 )  | 964.35                | ( 139.93 ) | 1100              | ( 1520.3 ) | 330.22                | ( 47.91 )  |
| 450               | ( 350.3 )  | 951.51                | ( 138.06 ) | 1150              | ( 1610.3 ) | 198.46                | ( 28.80 )  |
| 500               | ( 440.3 )  | 941.38                | ( 136.59 ) | 1200              | ( 1700.3 ) | 100.96                | ( 14.65 )  |
| 550               | ( 530.3 )  | 933.17                | ( 135.40 ) | 1256              | ( 1801.1 ) | 46.82                 | ( 6.79 )   |
| 600               | ( 620.3 )  | 926.13                | ( 134.38 ) |                   |            |                       |            |

**Application Notes:** Data for yield strength is collected from references [1, 2, 9, 14] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equations:

$$YS(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$YS(T)$  = Yield Strength [MPa]

$T$  = Temperature [K]

**Valid Temperature Range:**  $33 \leq T \leq 1089$

**Extended Temperature Range:**  $1089 < T < 1256$

## Constants:

| T Range [K]: | <u><math>33 &lt; T \leq 800</math></u> | <u><math>800 &lt; T \leq 1000</math></u> | <u><math>1000 &lt; T \leq 1256</math></u> |
|--------------|----------------------------------------|------------------------------------------|-------------------------------------------|
| A0 =         | 1.258E+3                               | 1.396E+4                                 | -1.166E+4                                 |
| A1 =         | -1.345E+3                              | -4.814E+4                                | 4.348E+4                                  |
| A2 =         | 1.942E+3                               | 5.945E+4                                 | -4.669E+4                                 |
| A3 =         | -1.037E+3                              | -2.462E+4                                | 1.552E+4                                  |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Tensile Strength with Temperature

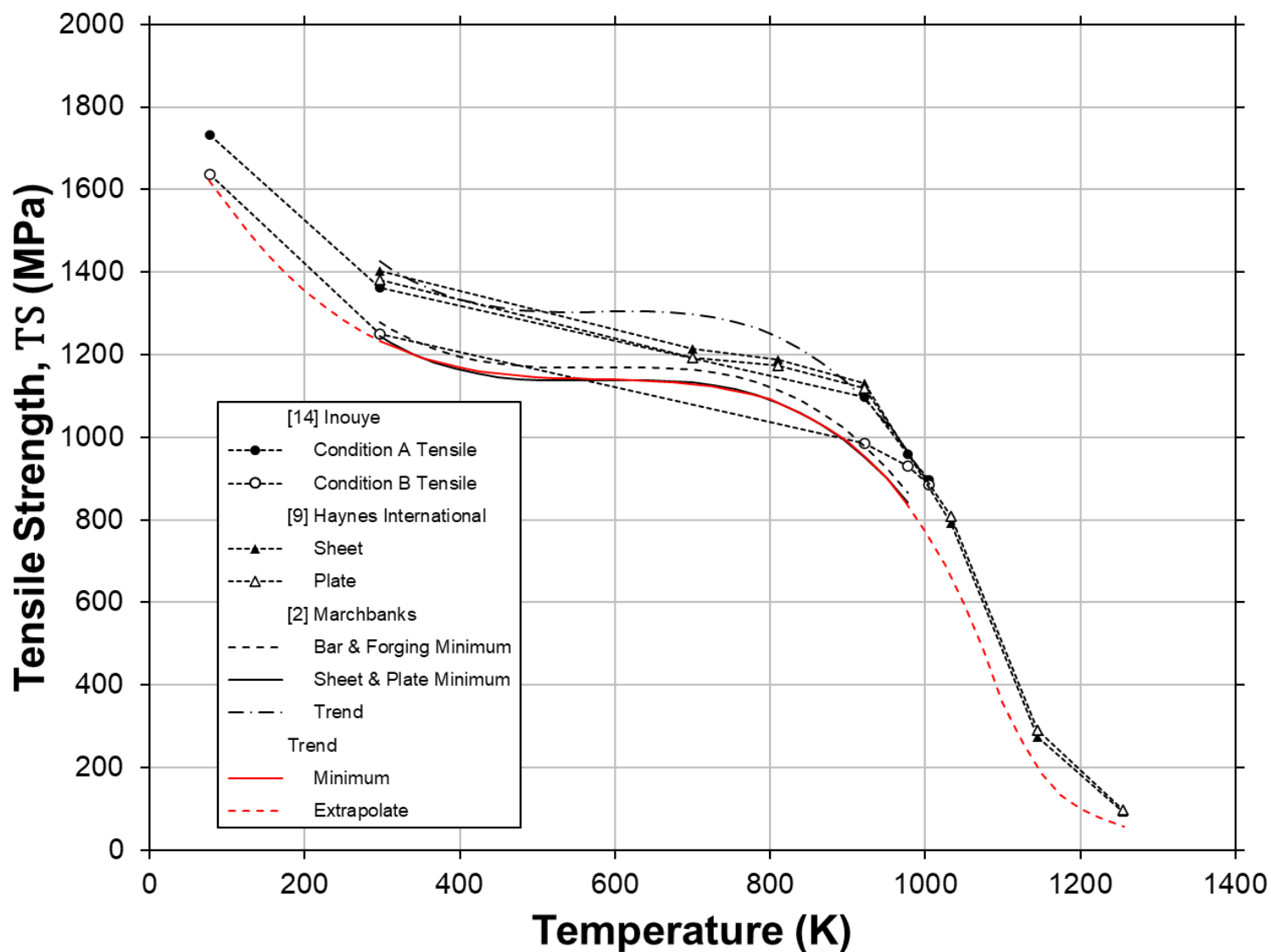


Figure 2.1.1-11: Tensile Strength versus Temperature of Inconel 718.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Tensile Strength with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Tensile Strength ( TS ) |            | Temperature ( T ) |            | Tensile Strength ( TS ) |            |
|-------------------|------------|-------------------------|------------|-------------------|------------|-------------------------|------------|
| K                 | ( °F )     | MPa                     | ( Ksi )    | K                 | ( °F )     | MPa                     | ( Ksi )    |
| 77                | ( -321.1 ) | 1622.12                 | ( 235.37 ) | 700               | ( 800.3 )  | 1128.75                 | ( 163.78 ) |
| 100               | ( -279.7 ) | 1560.80                 | ( 226.47 ) | 750               | ( 890.3 )  | 1115.00                 | ( 161.79 ) |
| 150               | ( -189.7 ) | 1446.16                 | ( 209.84 ) | 800               | ( 980.3 )  | 1091.99                 | ( 158.45 ) |
| 200               | ( -99.7 )  | 1354.89                 | ( 196.59 ) | 850               | ( 1070.3 ) | 1048.28                 | ( 152.11 ) |
| 250               | ( -9.7 )   | 1284.25                 | ( 186.34 ) | 900               | ( 1160.3 ) | 989.09                  | ( 143.52 ) |
| 300               | ( 80.3 )   | 1231.53                 | ( 178.70 ) | 950               | ( 1250.3 ) | 901.24                  | ( 130.77 ) |
| 350               | ( 170.3 )  | 1194.02                 | ( 173.25 ) | 1000              | ( 1340.3 ) | 774.00                  | ( 112.31 ) |
| 400               | ( 260.3 )  | 1168.98                 | ( 169.62 ) | 1050              | ( 1430.3 ) | 596.66                  | ( 86.58 )  |
| 450               | ( 350.3 )  | 1153.72                 | ( 167.40 ) | 1100              | ( 1520.3 ) | 358.51                  | ( 52.02 )  |
| 500               | ( 440.3 )  | 1145.50                 | ( 166.21 ) | 1150              | ( 1610.3 ) | 185.90                  | ( 26.97 )  |
| 550               | ( 530.3 )  | 1141.61                 | ( 165.65 ) | 1200              | ( 1700.3 ) | 100.48                  | ( 14.58 )  |
| 600               | ( 620.3 )  | 1139.34                 | ( 165.32 ) | 1256              | ( 1801.1 ) | 56.70                   | ( 8.23 )   |
| 650               | ( 710.3 )  | 1135.95                 | ( 164.83 ) |                   |            |                         |            |

**Application Notes:** Data for tensile strength is collected from references [2, 9, 14] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equations:

$$TS(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$TS(T)$  = Tensile Strength [MPa]

$T$  = Temperature [K]

**Valid Temperature Range:**  $297 \leq T \leq 978$

**Extended Temperature Range:**  $77 < T < 297$  and  $978 < T < 1256$

## Constants:

| T Range [K]: | <u><math>77 &lt; T \leq 800</math></u> | <u><math>800 &lt; T \leq 1100</math></u> | <u><math>1100 &lt; T \leq 1256</math></u> |
|--------------|----------------------------------------|------------------------------------------|-------------------------------------------|
| A0 =         | 1.871E+3                               | 8.054E+3                                 | 1.396E+5                                  |
| A1 =         | -3.696E+3                              | -2.584E+4                                | -3.356E+5                                 |
| A2 =         | 6.302E+3                               | 3.285E+4                                 | 2.696E+5                                  |
| A3 =         | -3.624E+3                              | -1.429E+4                                | -7.234E+4                                 |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 8272                                  | 1.092                                                   | 7.10                                        | -0.209                                             | 10.34                                     | 0.261                                          | 215.19                             | 84.55                            | 0.264                       |
| 150                 | 8262                                  | 1.112                                                   | 8.32                                        | -0.168                                             | 10.91                                     | 0.360                                          | 211.50                             | 82.72                            | 0.276                       |
| 200                 | 8249                                  | 1.132                                                   | 9.29                                        | -0.116                                             | 11.38                                     | 0.393                                          | 208.14                             | 81.08                            | 0.285                       |
| 250                 | 8234                                  | 1.150                                                   | 10.18                                       | -0.057                                             | 11.77                                     | 0.409                                          | 205.06                             | 79.63                            | 0.290                       |
| 295                 | 8220                                  | 1.166                                                   | 10.94                                       | 0.000                                              | 12.07                                     | 0.420                                          | 202.49                             | 78.44                            | 0.293                       |
| 300                 | 8219                                  | 1.168                                                   | 11.03                                       | 0.006                                              | 12.11                                     | 0.421                                          | 202.22                             | 78.32                            | 0.293                       |
| 350                 | 8204                                  | 1.185                                                   | 11.86                                       | 0.064                                              | 12.39                                     | 0.432                                          | 199.55                             | 77.13                            | 0.294                       |
| 400                 | 8189                                  | 1.201                                                   | 12.69                                       | 0.125                                              | 12.65                                     | 0.445                                          | 197.02                             | 76.03                            | 0.293                       |
| 450                 | 8173                                  | 1.217                                                   | 13.51                                       | 0.191                                              | 12.88                                     | 0.458                                          | 194.55                             | 75.01                            | 0.291                       |
| 500                 | 8156                                  | 1.231                                                   | 14.33                                       | 0.260                                              | 13.11                                     | 0.471                                          | 192.11                             | 74.02                            | 0.289                       |
| 550                 | 8138                                  | 1.245                                                   | 15.15                                       | 0.333                                              | 13.33                                     | 0.485                                          | 189.64                             | 73.04                            | 0.287                       |
| 600                 | 8120                                  | 1.257                                                   | 15.97                                       | 0.409                                              | 13.56                                     | 0.500                                          | 187.09                             | 72.05                            | 0.286                       |
| 650                 | 8101                                  | 1.269                                                   | 16.78                                       | 0.488                                              | 13.79                                     | 0.514                                          | 184.41                             | 71.02                            | 0.284                       |
| 700                 | 8081                                  | 1.280                                                   | 17.60                                       | 0.571                                              | 14.04                                     | 0.529                                          | 181.54                             | 69.92                            | 0.284                       |
| 750                 | 8060                                  | 1.290                                                   | 18.42                                       | 0.656                                              | 14.30                                     | 0.545                                          | 178.43                             | 68.73                            | 0.285                       |
| 800                 | 8039                                  | 1.299                                                   | 19.24                                       | 0.744                                              | 14.58                                     | 0.560                                          | 175.03                             | 67.41                            | 0.287                       |
| 850                 | 8017                                  | 1.307                                                   | 20.06                                       | 0.835                                              | 14.87                                     | 0.576                                          | 171.29                             | 65.94                            | 0.290                       |
| 900                 | 7995                                  | 1.315                                                   | 20.88                                       | 0.929                                              | 15.18                                     | 0.592                                          | 167.15                             | 64.29                            | 0.295                       |
| 1000                | 7949                                  | 1.327                                                   | 22.53                                       | 1.123                                              | 15.83                                     | 0.624                                          | 157.48                             | 60.36                            | 0.308                       |
| 1100                | 7901                                  | 1.336                                                   | 24.17                                       | 1.326                                              | 16.50                                     | 0.657                                          | 145.61                             | 55.39                            | 0.326                       |
| 1200                | 7852                                  | 1.341                                                   | 25.82                                       | 1.537                                              | 17.14                                     | 0.671                                          | 131.11                             | 49.17                            | 0.347                       |
| 1300                | 7802                                  | 1.343                                                   | 27.48                                       | 1.754                                              | 17.70                                     | 0.671                                          | 113.57                             | 41.49                            | 0.369                       |
| 1400                | 7751                                  | 1.342                                                   | -                                           | 1.978                                              | 18.13                                     | 0.671                                          | -                                  | -                                | -                           |
| 1500                | 7699                                  | 1.337                                                   | -                                           | 2.206                                              | 18.35                                     | 0.671                                          | -                                  | -                                | -                           |
| 1550                | 7673                                  | -                                                       | -                                           | 2.321                                              | 18.37                                     | -                                              | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

2 Nickel – Base Alloys

2.1 Precipitation Hardened Alloys

2.1.1 Inconel 718

**Revision 0: 08-05-2020**

### References

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### **3 Refractory Metals and Alloys**

#### **3.1 Molybdenum and Molybdenum Alloys**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 2.1: 08-25-2023

General

### Room Temperature Properties

|                                           |        |
|-------------------------------------------|--------|
| Atomic Mass, [amu]                        | 95.95  |
| Theoretical Density, [kg/m <sup>3</sup> ] | 10,280 |
| Melting Point, [K]                        | 2896   |
| Boiling Point, [K]                        | 4912   |
| Specific Heat, [J/(g-K)]                  | 0.249  |
| Heat of Fusion, [kJ/mol]                  | 36     |
| Heat of Vaporization, [kJ/mol]            | 600    |
| Thermal Conductivity, [W/(m-K)]           | 138.1  |
| Linear expansion coefficient, [μm/(m-K)]  | 4.96   |
| Electrical resistivity, [μΩ-m]            | 0.055  |
| Young's Modulus, [GPa]                    | 317.6  |
| Shear Modulus, [GPa]                      | 123.9  |
| Poisson's Ratio, [-]                      | 0.356  |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Density with Temperature

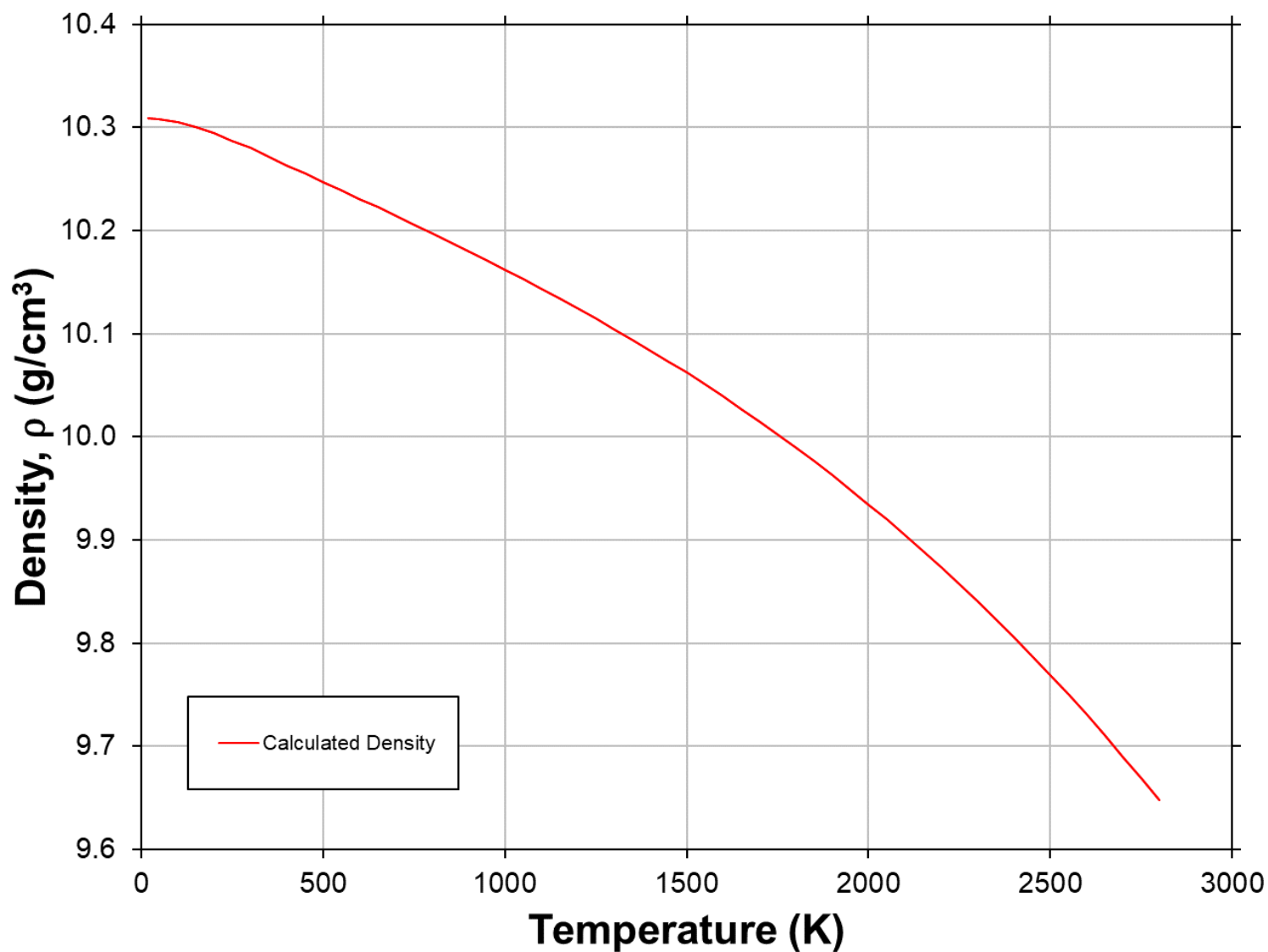


Figure 3.1.1-1: Density versus Temperature for Molybdenum. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 20                | ( -423.7 ) | 10309             | ( 643.6 )              | 1400              | ( 2060.3 ) | 10084             | ( 629.5 )              |
| 60                | ( -351.7 ) | 10308             | ( 643.5 )              | 1500              | ( 2240.3 ) | 10062             | ( 628.2 )              |
| 100               | ( -279.7 ) | 10306             | ( 643.4 )              | 1600              | ( 2420.3 ) | 10039             | ( 626.7 )              |
| 200               | ( -99.7 )  | 10294             | ( 642.7 )              | 1700              | ( 2600.3 ) | 10015             | ( 625.2 )              |
| 300               | ( 80.3 )   | 10280             | ( 641.8 )              | 1800              | ( 2780.3 ) | 9990              | ( 623.7 )              |
| 400               | ( 260.3 )  | 10264             | ( 640.8 )              | 1900              | ( 2960.3 ) | 9963              | ( 622.0 )              |
| 500               | ( 440.3 )  | 10247             | ( 639.7 )              | 2000              | ( 3140.3 ) | 9935              | ( 620.2 )              |
| 600               | ( 620.3 )  | 10231             | ( 638.7 )              | 2100              | ( 3320.3 ) | 9905              | ( 618.4 )              |
| 700               | ( 800.3 )  | 10214             | ( 637.7 )              | 2200              | ( 3500.3 ) | 9874              | ( 616.4 )              |
| 800               | ( 980.3 )  | 10197             | ( 636.6 )              | 2300              | ( 3680.3 ) | 9841              | ( 614.4 )              |
| 900               | ( 1160.3 ) | 10180             | ( 635.5 )              | 2400              | ( 3860.3 ) | 9806              | ( 612.2 )              |
| 1000              | ( 1340.3 ) | 10162             | ( 634.4 )              | 2500              | ( 4040.3 ) | 9769              | ( 609.9 )              |
| 1100              | ( 1520.3 ) | 10144             | ( 633.3 )              | 2600              | ( 4220.3 ) | 9731              | ( 607.5 )              |
| 1200              | ( 1700.3 ) | 10125             | ( 632.1 )              | 2700              | ( 4400.3 ) | 9690              | ( 605.0 )              |
| 1300              | ( 1880.3 ) | 10105             | ( 630.8 )              | 2800              | ( 4580.3 ) | 9648              | ( 602.3 )              |

**Application Notes:** Density trend is calculated as a function of thermal expansion, as seen in equation below.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 10,280 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $20 \leq T \leq 2800$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

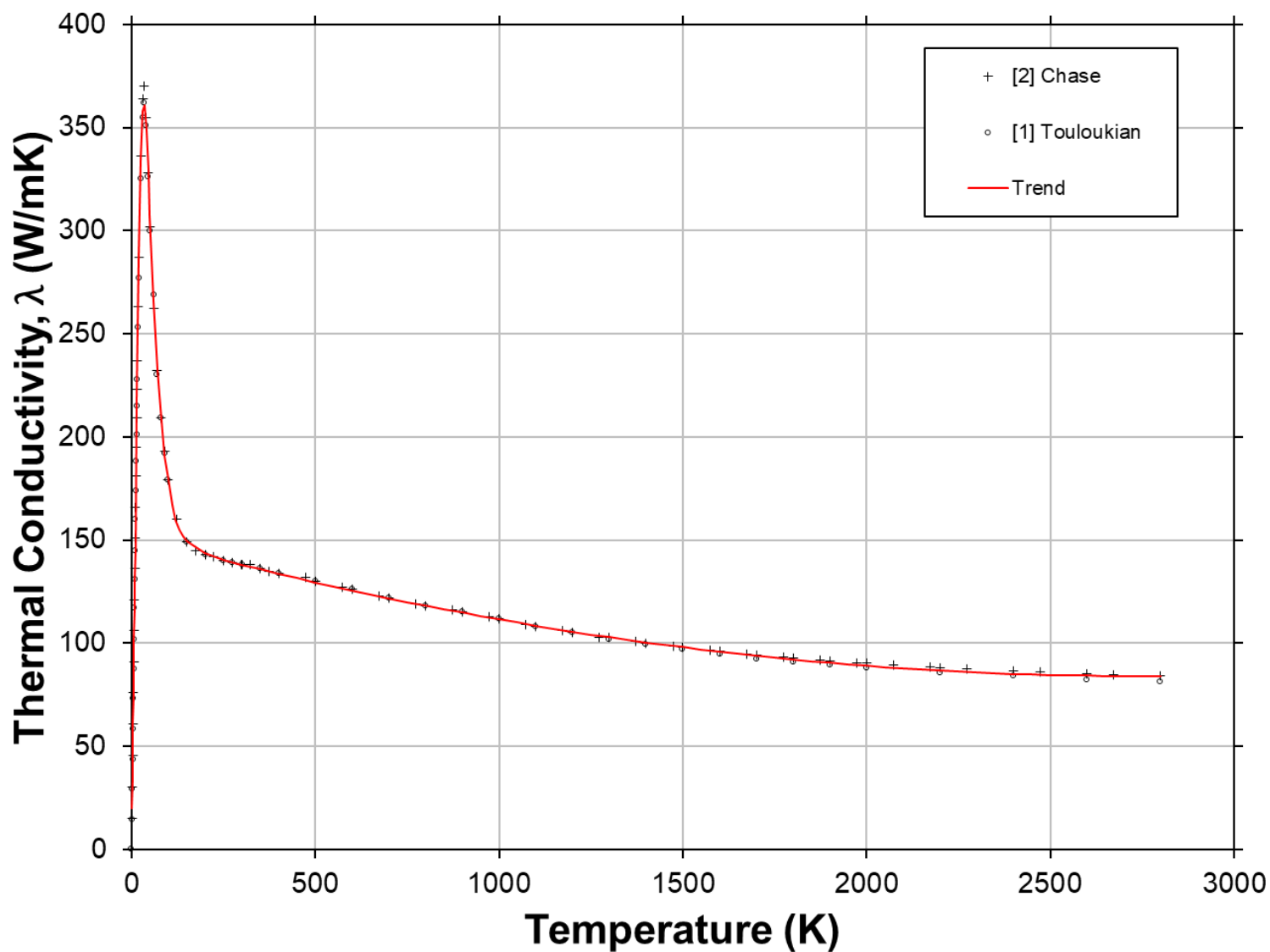


Figure 3.1.1-2: Thermal Conductivity versus Temperature of Molybdenum.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 100               | ( -279.7 ) | 179.19                     | ( 1243.26 )                         | 1500              | ( 2240.3 ) | 98.06                      | ( 680.38 )                          |
| 200               | ( -99.7 )  | 143.58                     | ( 996.16 )                          | 1600              | ( 2420.3 ) | 95.91                      | ( 665.44 )                          |
| 300               | ( 80.3 )   | 138.03                     | ( 957.65 )                          | 1700              | ( 2600.3 ) | 93.94                      | ( 651.74 )                          |
| 400               | ( 260.3 )  | 133.59                     | ( 926.87 )                          | 1800              | ( 2780.3 ) | 92.14                      | ( 639.28 )                          |
| 500               | ( 440.3 )  | 129.44                     | ( 898.07 )                          | 1900              | ( 2960.3 ) | 90.52                      | ( 628.06 )                          |
| 600               | ( 620.3 )  | 125.49                     | ( 870.67 )                          | 2000              | ( 3140.3 ) | 89.08                      | ( 618.07 )                          |
| 700               | ( 800.3 )  | 121.73                     | ( 844.55 )                          | 2100              | ( 3320.3 ) | 87.82                      | ( 609.33 )                          |
| 800               | ( 980.3 )  | 118.14                     | ( 819.68 )                          | 2200              | ( 3500.3 ) | 86.74                      | ( 601.83 )                          |
| 900               | ( 1160.3 ) | 114.74                     | ( 796.06 )                          | 2300              | ( 3680.3 ) | 85.84                      | ( 595.56 )                          |
| 1000              | ( 1340.3 ) | 111.51                     | ( 773.68 )                          | 2400              | ( 3860.3 ) | 85.11                      | ( 590.53 )                          |
| 1100              | ( 1520.3 ) | 108.46                     | ( 752.54 )                          | 2500              | ( 4040.3 ) | 84.57                      | ( 586.75 )                          |
| 1200              | ( 1700.3 ) | 105.60                     | ( 732.64 )                          | 2600              | ( 4220.3 ) | 84.20                      | ( 584.20 )                          |
| 1300              | ( 1880.3 ) | 102.91                     | ( 713.98 )                          | 2700              | ( 4400.3 ) | 84.01                      | ( 582.89 )                          |
| 1400              | ( 2060.3 ) | 100.40                     | ( 696.56 )                          | 2800              | ( 4580.3 ) | 84.00                      | ( 582.82 )                          |

**Application Notes:** Data for thermal conductivity is collected from reference [1, 2] and fitted with the equations below to approximate the property trend with respect to temperature.

## Fit Equation:

For temperature range:  $1 \leq T < 100$

$$k(T) = \left( A0 \cdot \left( \frac{T}{1000} \right)^N \right) / \left( 1 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 + A3 \cdot \left( \frac{T}{1000} \right)^3 \right)$$

For temperature range:  $100 \leq T \leq 2801$

$$k(T) = B0 + B1 \cdot \left( \frac{T}{1000} \right) + B2 \cdot \left( \frac{T}{1000} \right)^2 + B3 / \left( \frac{T}{1000} \right)^N$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]:

$1 \leq T < 100$

$100 \leq T \leq 2801$

N = 7.361E-01

B0 = 1.518E+02

A0 = 3.075E+03

B1 = -4.922E+01

A1 = -3.869E+01

B2 = 8.931

A2 = 1.029E+03

B3 = 6.874E-04

A3 = -4.274E+03

N = 4.671



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

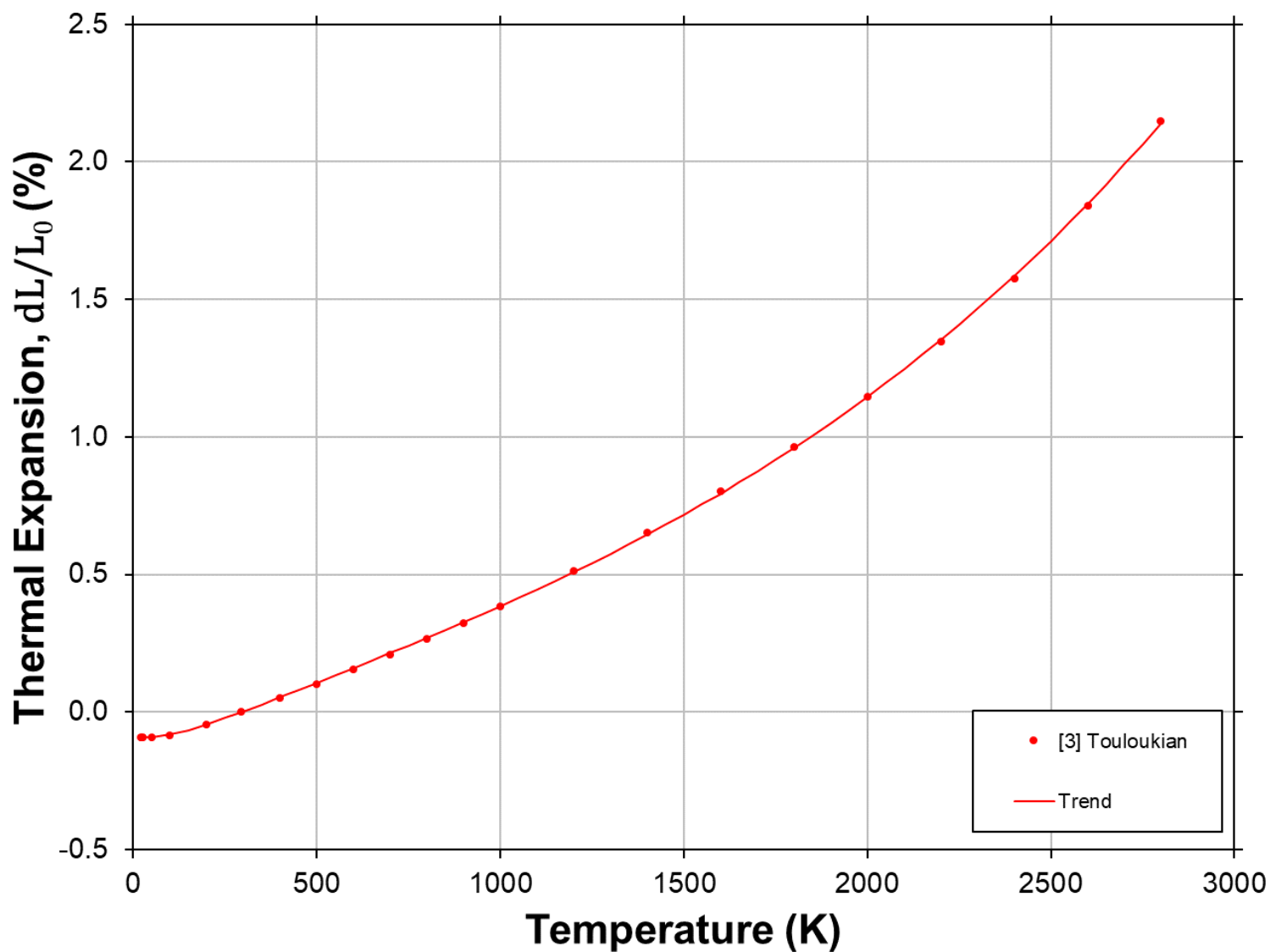


Figure 3.1.1-3: Thermal Expansion versus Temperature of Molybdenum.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 20                | ( -423.7 ) | -0.092                                  | 1400              | ( 2060.3 ) | 0.644                                   |
| 60                | ( -351.7 ) | -0.090                                  | 1500              | ( 2240.3 ) | 0.717                                   |
| 100               | ( -279.7 ) | -0.083                                  | 1600              | ( 2420.3 ) | 0.793                                   |
| 200               | ( -99.7 )  | -0.046                                  | 1700              | ( 2600.3 ) | 0.874                                   |
| 300               | ( 80.3 )   | 0.000                                   | 1800              | ( 2780.3 ) | 0.959                                   |
| 400               | ( 260.3 )  | 0.053                                   | 1900              | ( 2960.3 ) | 1.050                                   |
| 500               | ( 440.3 )  | 0.106                                   | 2000              | ( 3140.3 ) | 1.145                                   |
| 600               | ( 620.3 )  | 0.160                                   | 2100              | ( 3320.3 ) | 1.246                                   |
| 700               | ( 800.3 )  | 0.214                                   | 2200              | ( 3500.3 ) | 1.353                                   |
| 800               | ( 980.3 )  | 0.269                                   | 2300              | ( 3680.3 ) | 1.466                                   |
| 900               | ( 1160.3 ) | 0.326                                   | 2400              | ( 3860.3 ) | 1.586                                   |
| 1000              | ( 1340.3 ) | 0.385                                   | 2500              | ( 4040.3 ) | 1.713                                   |
| 1100              | ( 1520.3 ) | 0.445                                   | 2600              | ( 4220.3 ) | 1.847                                   |
| 1200              | ( 1700.3 ) | 0.509                                   | 2700              | ( 4400.3 ) | 1.989                                   |
| 1300              | ( 1880.3 ) | 0.575                                   | 2800              | ( 4580.3 ) | 2.139                                   |

**Application Notes:** Data for thermal expansion is collected from reference [3] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

| T Range [K]: | $20 \leq T \leq 294$ | $294 < T \leq 2800$ |
|--------------|----------------------|---------------------|
| A0 =         | -9.113E-02           | -1.612E-01          |
| A1 =         | -1.096E-01           | 5.529E-01           |
| A2 =         | 2.178E+00            | -6.426E-02          |
| A3 =         | -2.532E+00           | 5.720E-02           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

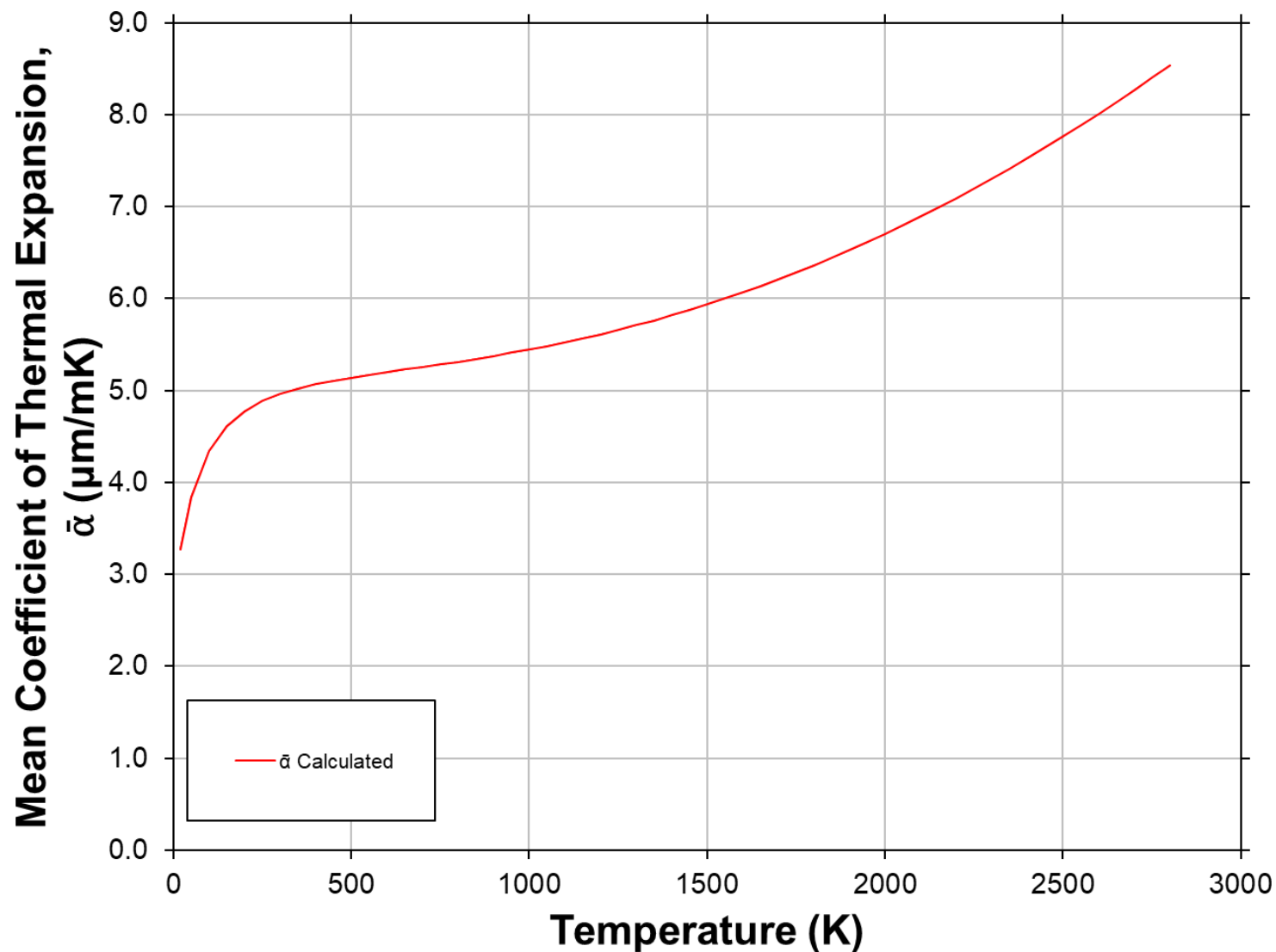


Figure 3.1.1-4: Mean Coefficient of Thermal Expansion versus Temperature of Molybdenum. Calculated from fitted trend of Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                   | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                   |
|-------------------|------------|---------------------------------------|---------------------------------------------------|-------------------|------------|---------------------------------------|---------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^\circ\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^\circ\text{F})$ ) |
| 20                | ( -423.7 ) | 3.270                                 | ( 1.817 )                                         | 1400              | ( 2060.3 ) | 5.819                                 | ( 3.233 )                                         |
| 60                | ( -351.7 ) | 3.971                                 | ( 2.206 )                                         | 1500              | ( 2240.3 ) | 5.938                                 | ( 3.299 )                                         |
| 100               | ( -279.7 ) | 4.342                                 | ( 2.412 )                                         | 1600              | ( 2420.3 ) | 6.068                                 | ( 3.371 )                                         |
| 200               | ( -99.7 )  | 4.776                                 | ( 2.653 )                                         | 1700              | ( 2600.3 ) | 6.210                                 | ( 3.450 )                                         |
| 300               | ( 80.3 )   | 4.964                                 | ( 2.758 )                                         | 1800              | ( 2780.3 ) | 6.363                                 | ( 3.535 )                                         |
| 400               | ( 260.3 )  | 5.069                                 | ( 2.816 )                                         | 1900              | ( 2960.3 ) | 6.527                                 | ( 3.626 )                                         |
| 500               | ( 440.3 )  | 5.141                                 | ( 2.856 )                                         | 2000              | ( 3140.3 ) | 6.703                                 | ( 3.724 )                                         |
| 600               | ( 620.3 )  | 5.199                                 | ( 2.889 )                                         | 2100              | ( 3320.3 ) | 6.891                                 | ( 3.829 )                                         |
| 700               | ( 800.3 )  | 5.255                                 | ( 2.919 )                                         | 2200              | ( 3500.3 ) | 7.091                                 | ( 3.940 )                                         |
| 800               | ( 980.3 )  | 5.313                                 | ( 2.952 )                                         | 2300              | ( 3680.3 ) | 7.303                                 | ( 4.057 )                                         |
| 900               | ( 1160.3 ) | 5.376                                 | ( 2.986 )                                         | 2400              | ( 3860.3 ) | 7.527                                 | ( 4.181 )                                         |
| 1000              | ( 1340.3 ) | 5.446                                 | ( 3.025 )                                         | 2500              | ( 4040.3 ) | 7.762                                 | ( 4.312 )                                         |
| 1100              | ( 1520.3 ) | 5.524                                 | ( 3.069 )                                         | 2600              | ( 4220.3 ) | 8.010                                 | ( 4.450 )                                         |
| 1200              | ( 1700.3 ) | 5.612                                 | ( 3.118 )                                         | 2700              | ( 4400.3 ) | 8.269                                 | ( 4.594 )                                         |
| 1300              | ( 1880.3 ) | 5.710                                 | ( 3.172 )                                         | 2800              | ( 4580.3 ) | 8.541                                 | ( 4.745 )                                         |

**Application Notes:** Mean coefficient of thermal expansion is calculated as a function of the thermal expansion trend. The calculated data has been fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = \left( A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right) / \left( A_0 + \left( \frac{T}{1000} \right) \right)$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $20 \leq T \leq 2800$

A0 = 1.975E-01

A1 = 5.685E+00

A2 = -6.415E-01

A3 = 6.136E-01

A\_0 = 7.509E-02



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Specific Heat with Temperature

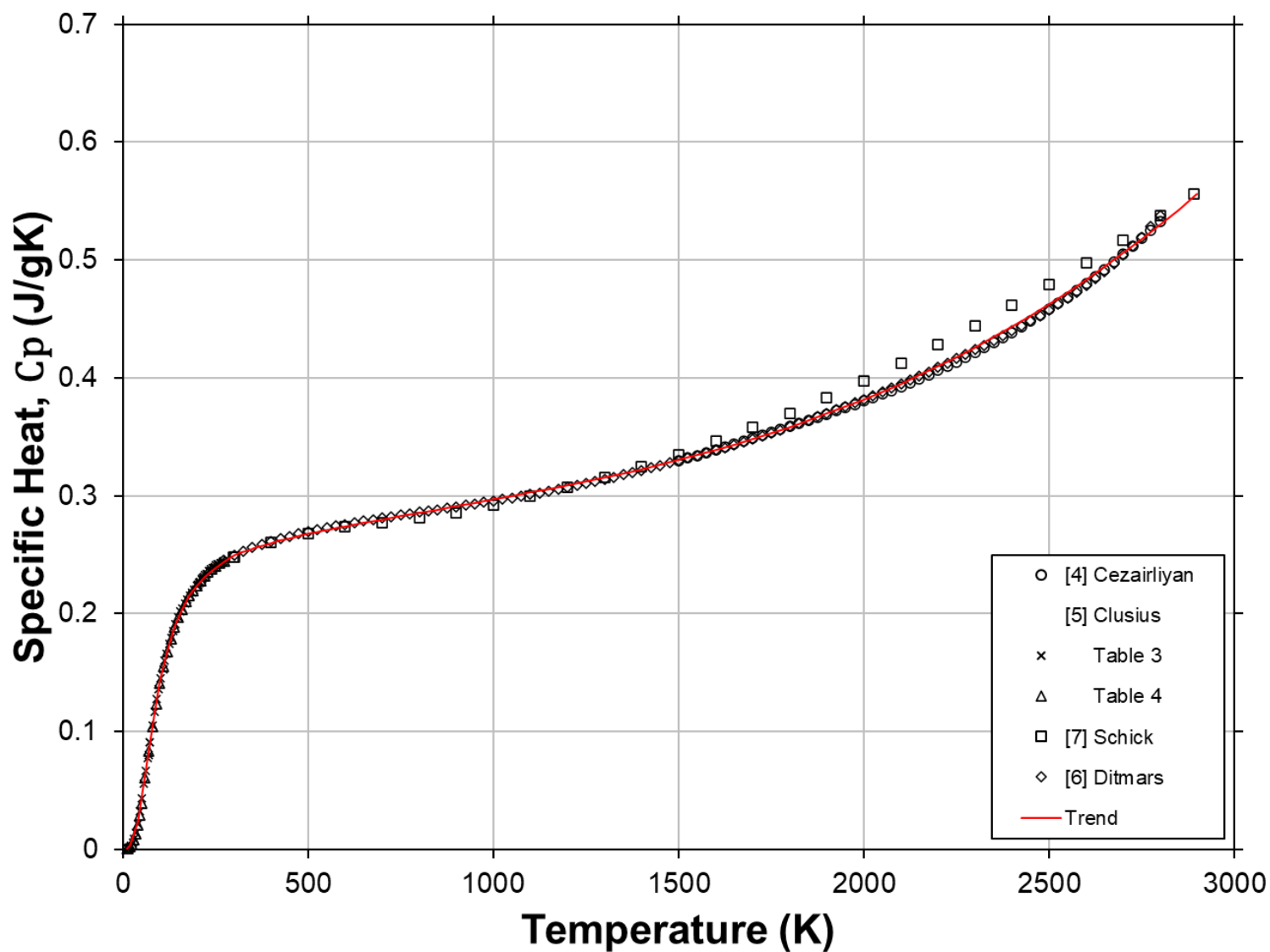


Figure 3.1.1-5: Specific Heat versus Temperature of Molybdenum.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 20                | ( -423.7 ) | 0.003                            | ( 0.001 )       | 1100              | ( 1520.3 ) | 0.303                            | ( 0.072 )       |
| 60                | ( -351.7 ) | 0.062                            | ( 0.015 )       | 1200              | ( 1700.3 ) | 0.309                            | ( 0.074 )       |
| 100               | ( -279.7 ) | 0.142                            | ( 0.034 )       | 1300              | ( 1880.3 ) | 0.315                            | ( 0.075 )       |
| 200               | ( -99.7 )  | 0.224                            | ( 0.053 )       | 1400              | ( 2060.3 ) | 0.323                            | ( 0.077 )       |
| 300               | ( 80.3 )   | 0.249                            | ( 0.060 )       | 1500              | ( 2240.3 ) | 0.330                            | ( 0.079 )       |
| 400               | ( 260.3 )  | 0.260                            | ( 0.062 )       | 1600              | ( 2420.3 ) | 0.339                            | ( 0.081 )       |
| 500               | ( 440.3 )  | 0.268                            | ( 0.064 )       | 1800              | ( 2780.3 ) | 0.358                            | ( 0.086 )       |
| 600               | ( 620.3 )  | 0.274                            | ( 0.066 )       | 2000              | ( 3140.3 ) | 0.382                            | ( 0.091 )       |
| 700               | ( 800.3 )  | 0.280                            | ( 0.067 )       | 2200              | ( 3500.3 ) | 0.410                            | ( 0.098 )       |
| 800               | ( 980.3 )  | 0.286                            | ( 0.068 )       | 2400              | ( 3860.3 ) | 0.444                            | ( 0.106 )       |
| 900               | ( 1160.3 ) | 0.291                            | ( 0.070 )       | 2600              | ( 4220.3 ) | 0.483                            | ( 0.116 )       |
| 1000              | ( 1340.3 ) | 0.297                            | ( 0.071 )       | 2800              | ( 4580.3 ) | 0.530                            | ( 0.127 )       |
| 1100              | ( 1520.3 ) | 0.303                            | ( 0.072 )       | 2900              | ( 4760.3 ) | 0.556                            | ( 0.133 )       |

**Application Notes:** Data for specific heat is collected from references [4-7] and fitted with the equations below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $10 < T \leq 293$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right]$$

For temperature range:  $293 < T \leq 2900$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_3 \cdot \left( \frac{T}{1000} \right)^3 + B_{-2} / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$T$  = Temperature [K]

## Constants:

| T. Range [K]: | $10 < T < 293$ | $293 < T < 2900$             |
|---------------|----------------|------------------------------|
| N =           | 2.964E+00      | B0 = 2.422E-01               |
| A0 =          | 3.230E+02      | B1 = 7.138E-02               |
| A1 =          | -5.704E+00     | B2 = -3.057E-02              |
| A2 =          | 1.002E+02      | B3 = 1.493E-02               |
| A3 =          | 1.046E+03      | B <sub>-2</sub> = -1.082E-03 |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

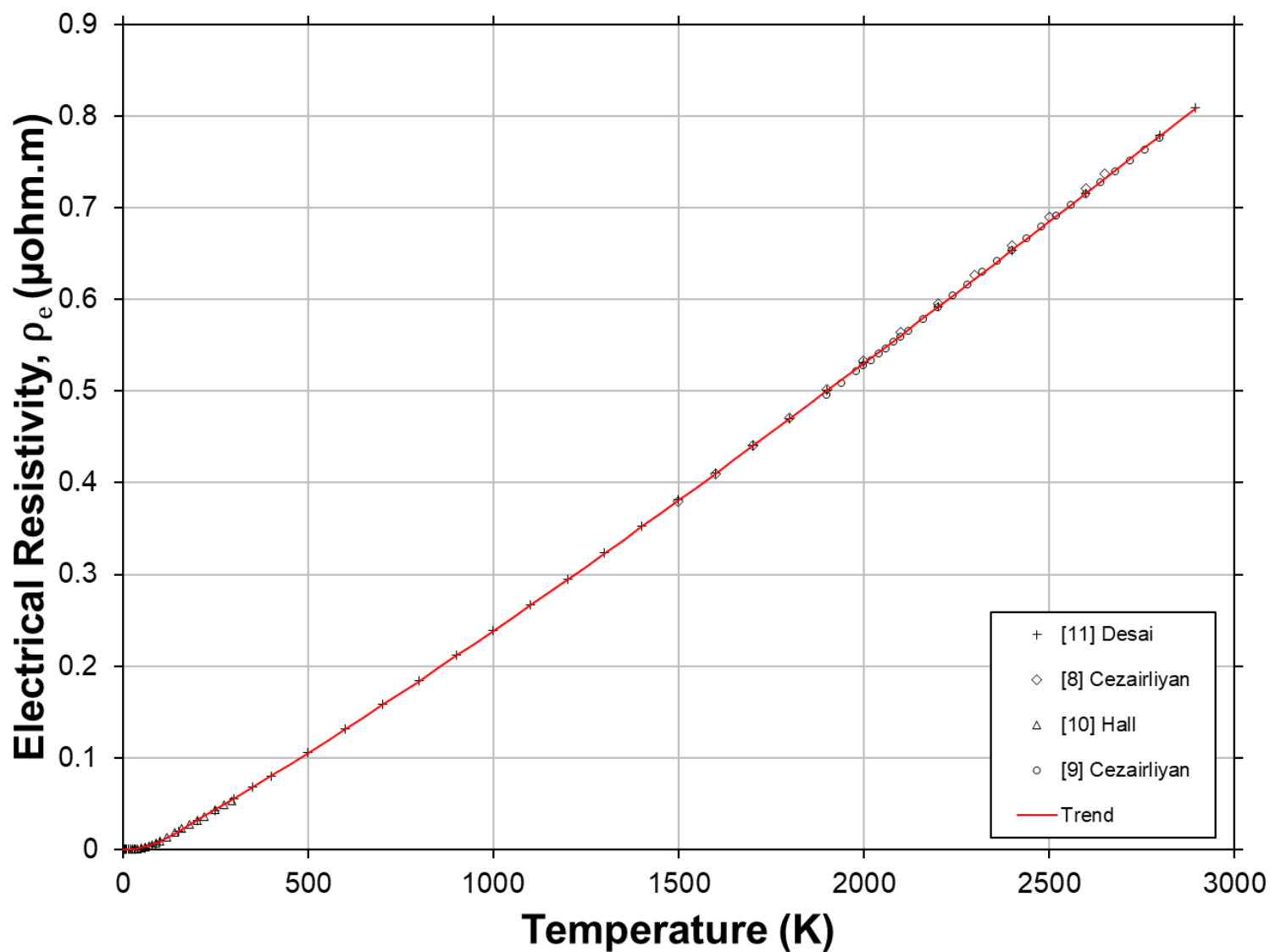


Figure 3.1.1-6: Electrical Resistivity versus Temperature for Molybdenum.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 20                | ( -423.7 ) | 0.000                               | ( -0.01 )               | 1500              | ( 2240.3 ) | 0.381                               | ( 14.99 )               |
| 100               | ( -279.7 ) | 0.009                               | ( 0.34 )                | 1600              | ( 2420.3 ) | 0.410                               | ( 16.15 )               |
| 200               | ( -99.7 )  | 0.032                               | ( 1.25 )                | 1700              | ( 2600.3 ) | 0.440                               | ( 17.32 )               |
| 300               | ( 80.3 )   | 0.055                               | ( 2.18 )                | 1800              | ( 2780.3 ) | 0.470                               | ( 18.49 )               |
| 400               | ( 260.3 )  | 0.080                               | ( 3.15 )                | 1900              | ( 2960.3 ) | 0.500                               | ( 19.68 )               |
| 500               | ( 440.3 )  | 0.105                               | ( 4.15 )                | 2000              | ( 3140.3 ) | 0.530                               | ( 20.88 )               |
| 600               | ( 620.3 )  | 0.131                               | ( 5.16 )                | 2100              | ( 3320.3 ) | 0.561                               | ( 22.08 )               |
| 700               | ( 800.3 )  | 0.157                               | ( 6.19 )                | 2200              | ( 3500.3 ) | 0.592                               | ( 23.29 )               |
| 800               | ( 980.3 )  | 0.184                               | ( 7.24 )                | 2300              | ( 3680.3 ) | 0.622                               | ( 24.51 )               |
| 900               | ( 1160.3 ) | 0.211                               | ( 8.31 )                | 2400              | ( 3860.3 ) | 0.654                               | ( 25.73 )               |
| 1000              | ( 1340.3 ) | 0.238                               | ( 9.39 )                | 2500              | ( 4040.3 ) | 0.685                               | ( 26.96 )               |
| 1100              | ( 1520.3 ) | 0.266                               | ( 10.48 )               | 2600              | ( 4220.3 ) | 0.716                               | ( 28.19 )               |
| 1200              | ( 1700.3 ) | 0.294                               | ( 11.59 )               | 2700              | ( 4400.3 ) | 0.747                               | ( 29.42 )               |
| 1300              | ( 1880.3 ) | 0.323                               | ( 12.71 )               | 2800              | ( 4580.3 ) | 0.779                               | ( 30.66 )               |
| 1400              | ( 2060.3 ) | 0.352                               | ( 13.84 )               | 2895              | ( 4751.3 ) | 0.809                               | ( 31.84 )               |

**Application Notes:** Data for electrical resistivity is collected from references [8-11] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

| T. Range [K]: | $1 \leq T \leq 293$ | $293 < T \leq 2895$ |
|---------------|---------------------|---------------------|
| A0 =          | -1.023E-04          | -1.602E-02          |
| A1 =          | -3.404E-02          | 2.298E-01           |
| A2 =          | 1.472E+00           | 2.758E-02           |
| A3 =          | -2.526E+00          | -2.954E-03          |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Young's Modulus with Temperature

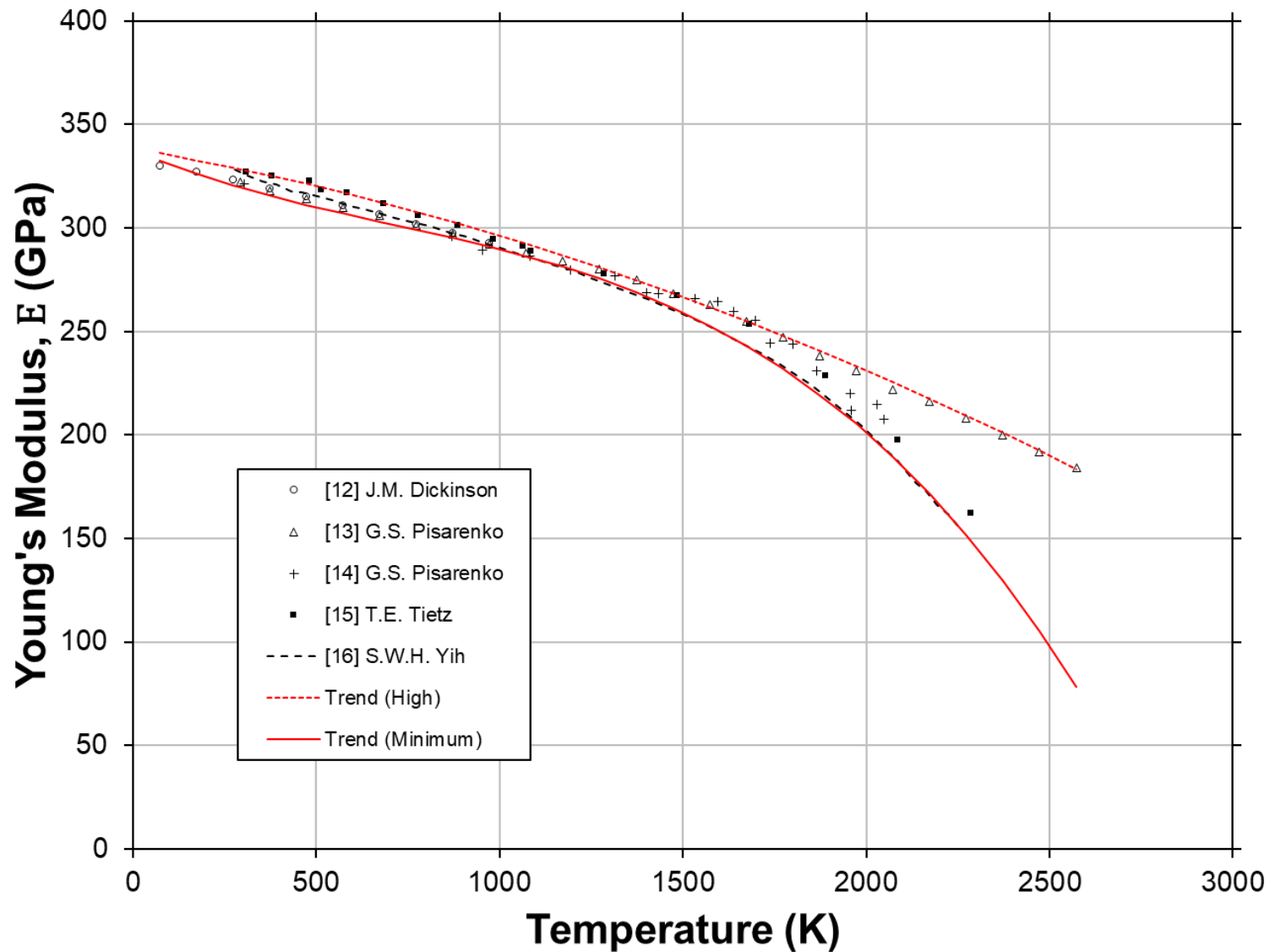


Figure 3.1.1-7: Young's modulus versus temperature for molybdenum.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Young's Modulus with Temperature

Minimum Young's Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   |
| 73                | ( -328.3 ) | 332.54                | ( 48.25 ) |
| 200               | ( -99.7 )  | 324.65                | ( 47.11 ) |
| 293               | ( 67.7 )   | 319.61                | ( 46.38 ) |
| 500               | ( 440.3 )  | 310.05                | ( 44.99 ) |
| 700               | ( 800.3 )  | 302.01                | ( 43.82 ) |
| 900               | ( 1160.3 ) | 294.02                | ( 42.66 ) |
| 1100              | ( 1520.3 ) | 284.97                | ( 41.35 ) |
| 1300              | ( 1880.3 ) | 273.75                | ( 39.72 ) |
| 1500              | ( 2240.3 ) | 259.22                | ( 37.61 ) |
| 1700              | ( 2600.3 ) | 240.29                | ( 34.87 ) |
| 1900              | ( 2960.3 ) | 215.83                | ( 31.32 ) |
| 2100              | ( 3320.3 ) | 184.73                | ( 26.80 ) |
| 2300              | ( 3680.3 ) | 145.87                | ( 21.17 ) |
| 2500              | ( 4040.3 ) | 98.13                 | ( 14.24 ) |
| 2573              | ( 4171.7 ) | 78.28                 | ( 11.36 ) |

High Young's Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   |
| 73                | ( -328.3 ) | 336.09                | ( 48.77 ) |
| 200               | ( -99.7 )  | 331.79                | ( 48.14 ) |
| 293               | ( 67.7 )   | 328.40                | ( 47.65 ) |
| 500               | ( 440.3 )  | 320.15                | ( 46.45 ) |
| 700               | ( 800.3 )  | 311.25                | ( 45.16 ) |
| 900               | ( 1160.3 ) | 301.43                | ( 43.74 ) |
| 1100              | ( 1520.3 ) | 290.70                | ( 42.18 ) |
| 1300              | ( 1880.3 ) | 279.05                | ( 40.49 ) |
| 1500              | ( 2240.3 ) | 266.49                | ( 38.67 ) |
| 1700              | ( 2600.3 ) | 253.01                | ( 36.71 ) |
| 1900              | ( 2960.3 ) | 238.62                | ( 34.62 ) |
| 2100              | ( 3320.3 ) | 223.31                | ( 32.40 ) |
| 2300              | ( 3680.3 ) | 207.09                | ( 30.05 ) |
| 2500              | ( 4040.3 ) | 189.95                | ( 27.56 ) |
| 2573              | ( 4171.7 ) | 183.47                | ( 26.62 ) |

**Application Notes:** Data for Young's modulus is collected from references [12-16] and fitted with the equation below to approximate the property trend with respect to temperature. Since two distinct trends were observed, a minimum and a high equation were generated.

## Fit Equations:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

T Range [K]:  $73 \leq T \leq 2573$

## Constants:

| Curve: | <u>Minimum</u> | <u>High</u> |
|--------|----------------|-------------|
| A0 =   | 337.7          | 338.4       |
| A1 =   | -74.24         | -30.78      |
| A2 =   | 49.49          | -11.44      |
| A3 =   | -23.25         | 0           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Shear Modulus with Temperature

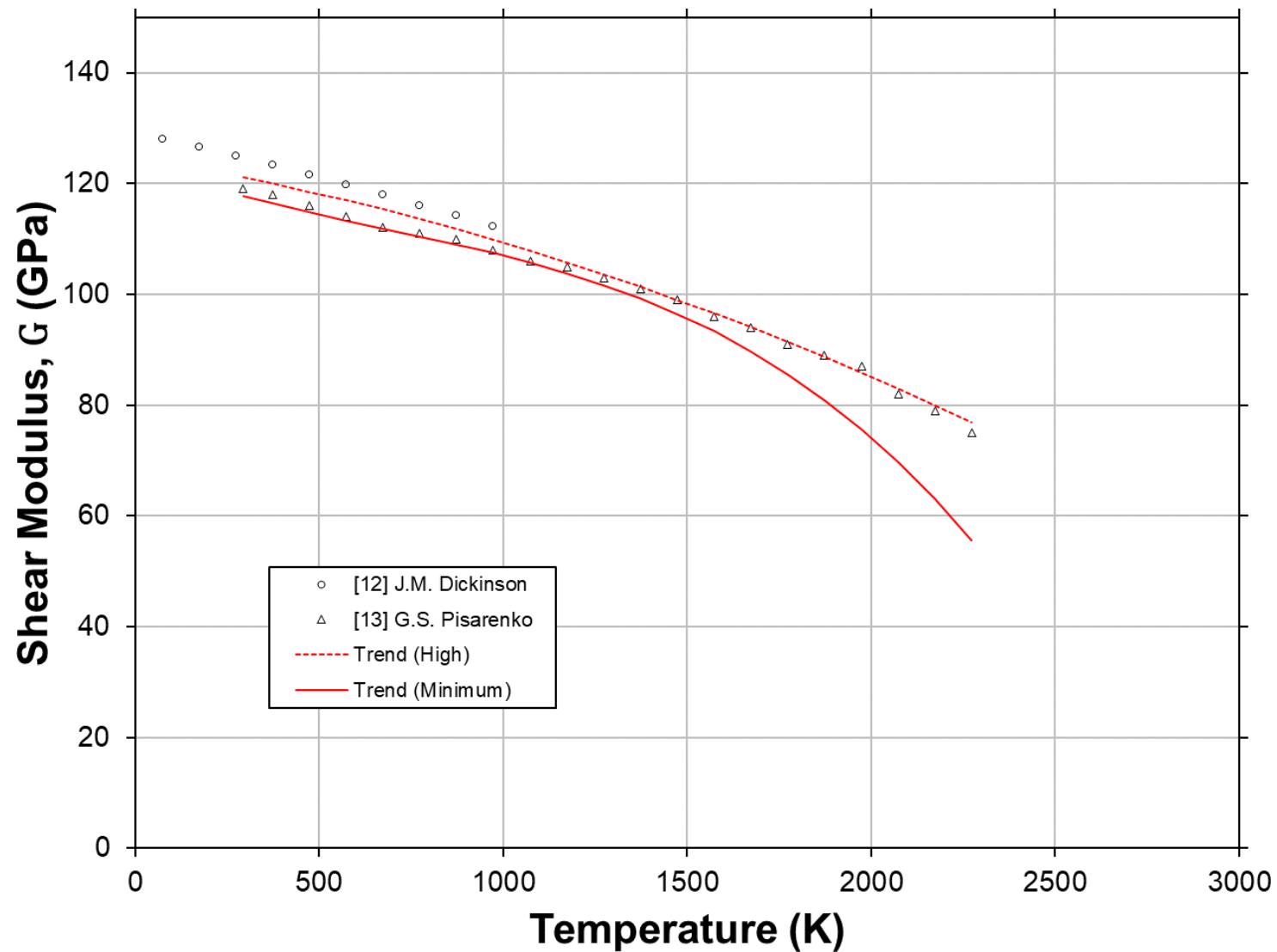


Figure 3.1.1-8: Shear modulus versus temperature for molybdenum. Trends were calculated from the fitted Young's modulus and Poisson's ratio trends.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Shear Modulus with Temperature

Minimum Young's Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( G ) |           |
|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   |
| 293               | ( 67.7 )   | 117.80              | ( 17.09 ) |
| 300               | ( 80.3 )   | 117.68              | ( 17.07 ) |
| 450               | ( 350.3 )  | 115.16              | ( 16.71 ) |
| 600               | ( 620.3 )  | 112.92              | ( 16.39 ) |
| 750               | ( 890.3 )  | 110.79              | ( 16.08 ) |
| 900               | ( 1160.3 ) | 108.58              | ( 15.76 ) |
| 1050              | ( 1430.3 ) | 106.13              | ( 15.40 ) |
| 1200              | ( 1700.3 ) | 103.28              | ( 14.99 ) |
| 1350              | ( 1970.3 ) | 99.84               | ( 14.49 ) |
| 1500              | ( 2240.3 ) | 95.65               | ( 13.88 ) |
| 1650              | ( 2510.3 ) | 90.54               | ( 13.14 ) |
| 1800              | ( 2780.3 ) | 84.33               | ( 12.24 ) |
| 1950              | ( 3050.3 ) | 76.86               | ( 11.15 ) |
| 2100              | ( 3320.3 ) | 67.95               | ( 9.86 )  |
| 2273              | ( 3631.7 ) | 55.67               | ( 8.08 )  |

High Young's Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( G ) |           |
|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   |
| 293               | ( 67.7 )   | 121.08              | ( 17.57 ) |
| 300               | ( 80.3 )   | 120.98              | ( 17.55 ) |
| 450               | ( 350.3 )  | 118.87              | ( 17.25 ) |
| 600               | ( 620.3 )  | 116.55              | ( 16.91 ) |
| 750               | ( 890.3 )  | 114.03              | ( 16.55 ) |
| 900               | ( 1160.3 ) | 111.30              | ( 16.15 ) |
| 1050              | ( 1430.3 ) | 108.37              | ( 15.72 ) |
| 1200              | ( 1700.3 ) | 105.23              | ( 15.27 ) |
| 1350              | ( 1970.3 ) | 101.89              | ( 14.78 ) |
| 1500              | ( 2240.3 ) | 98.35               | ( 14.27 ) |
| 1650              | ( 2510.3 ) | 94.60               | ( 13.73 ) |
| 1800              | ( 2780.3 ) | 90.65               | ( 13.15 ) |
| 1950              | ( 3050.3 ) | 86.49               | ( 12.55 ) |
| 2100              | ( 3320.3 ) | 82.13               | ( 11.92 ) |
| 2273              | ( 3631.7 ) | 76.84               | ( 11.15 ) |

**Application Notes:** Shear modulus is calculated as a function of Young's modulus and Poisson's ratio to approximate the property trend with respect to temperature. Since a minimum and a high trend were found for Young's modulus, this resulted in two shear modulus curves. Equivalent equations can be found below for readers' convenience. Trends are compared with [12, 13].

## Fit Equations:

$$G(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

T Range [K]:  $293 \leq T \leq 2273$

## Constants:

| Curve: | <u>Minimum</u> | <u>High</u> |
|--------|----------------|-------------|
| A0 =   | 124.2          | 124.6       |
| A1 =   | -26.22         | -10.69      |
| A2 =   | 17.44          | -4.541      |
| A3 =   | -8.433         | 0           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

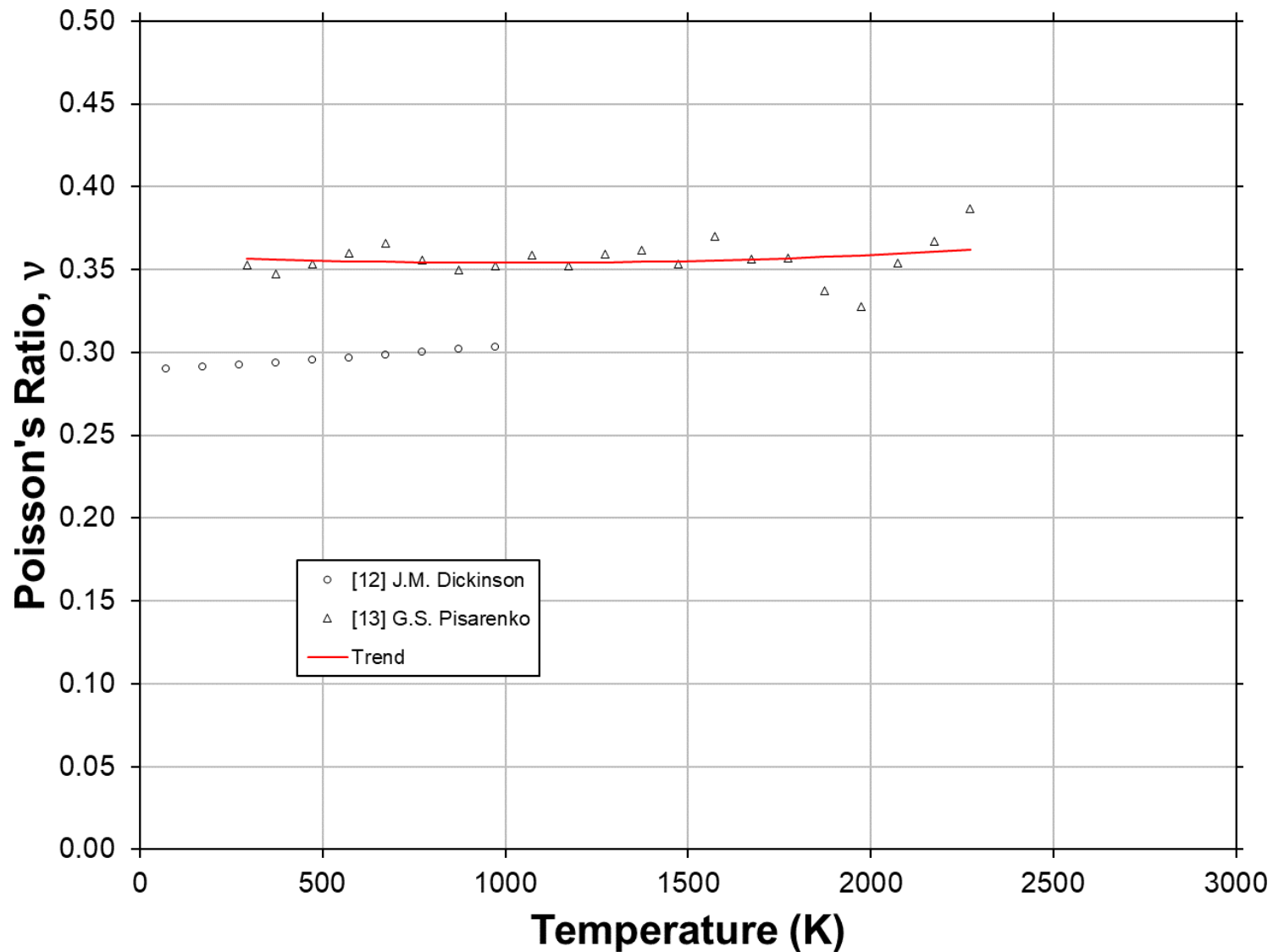


Figure 3.1.1-9: Poisson's ratio versus temperature for molybdenum.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Poisson's Ratio ( ν ) | Temperature ( T ) |            | Poisson's Ratio ( ν ) |
|-------------------|------------|-----------------------|-------------------|------------|-----------------------|
| K                 | ( °F )     |                       | K                 | ( °F )     |                       |
| 293               | ( 67.7 )   | 0.357                 | 1100              | ( 1520.3 ) | 0.354                 |
| 300               | ( 80.3 )   | 0.356                 | 1200              | ( 1700.3 ) | 0.354                 |
| 350               | ( 170.3 )  | 0.356                 | 1300              | ( 1880.3 ) | 0.354                 |
| 400               | ( 260.3 )  | 0.356                 | 1400              | ( 2060.3 ) | 0.355                 |
| 450               | ( 350.3 )  | 0.356                 | 1500              | ( 2240.3 ) | 0.355                 |
| 500               | ( 440.3 )  | 0.355                 | 1600              | ( 2420.3 ) | 0.356                 |
| 550               | ( 530.3 )  | 0.355                 | 1700              | ( 2600.3 ) | 0.356                 |
| 600               | ( 620.3 )  | 0.355                 | 1800              | ( 2780.3 ) | 0.357                 |
| 650               | ( 710.3 )  | 0.355                 | 1900              | ( 2960.3 ) | 0.358                 |
| 700               | ( 800.3 )  | 0.354                 | 2000              | ( 3140.3 ) | 0.359                 |
| 800               | ( 980.3 )  | 0.354                 | 2100              | ( 3320.3 ) | 0.360                 |
| 900               | ( 1160.3 ) | 0.354                 | 2200              | ( 3500.3 ) | 0.361                 |
| 1000              | ( 1340.3 ) | 0.354                 | 2273              | ( 3631.7 ) | 0.362                 |

**Application Notes:** Data for Poisson's ratio was collected from [12, 13] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equations:

$$\nu(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$\nu(T)$  = Poisson's Ratio

$T$  = Temperature [K]

## Constants:

T Range [K]: 293 ≤ T ≤ 2273

A0 = 0.359

A1 = -9.966E-3

A2 = 4.934E-3



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 10306                                 | 0.009                                                   | 179.19                                      | -0.083                                             | 4.34                                      | 0.142                                          | 325.62                             | 127.61                           | 0.358                       |
| 200                 | 10294                                 | 0.032                                                   | 143.58                                      | -0.046                                             | 4.78                                      | 0.224                                          | 321.55                             | 125.72                           | 0.357                       |
| 300                 | 10280                                 | 0.055                                                   | 138.03                                      | 0.000                                              | 4.96                                      | 0.249                                          | 317.49                             | 123.84                           | 0.356                       |
| 400                 | 10264                                 | 0.080                                                   | 133.59                                      | 0.053                                              | 5.07                                      | 0.260                                          | 313.41                             | 121.95                           | 0.356                       |
| 500                 | 10247                                 | 0.105                                                   | 129.44                                      | 0.106                                              | 5.14                                      | 0.268                                          | 309.28                             | 120.05                           | 0.355                       |
| 600                 | 10231                                 | 0.131                                                   | 125.49                                      | 0.160                                              | 5.20                                      | 0.274                                          | 305.08                             | 118.13                           | 0.355                       |
| 700                 | 10214                                 | 0.157                                                   | 121.73                                      | 0.214                                              | 5.26                                      | 0.280                                          | 300.81                             | 116.18                           | 0.354                       |
| 800                 | 10197                                 | 0.184                                                   | 118.14                                      | 0.269                                              | 5.31                                      | 0.286                                          | 296.42                             | 114.19                           | 0.354                       |
| 900                 | 10180                                 | 0.211                                                   | 114.74                                      | 0.326                                              | 5.38                                      | 0.291                                          | 291.91                             | 112.16                           | 0.354                       |
| 1000                | 10162                                 | 0.238                                                   | 111.51                                      | 0.385                                              | 5.45                                      | 0.297                                          | 287.25                             | 110.07                           | 0.354                       |
| 1100                | 10144                                 | 0.266                                                   | 108.46                                      | 0.445                                              | 5.52                                      | 0.303                                          | 282.42                             | 107.93                           | 0.354                       |
| 1200                | 10125                                 | 0.294                                                   | 105.60                                      | 0.509                                              | 5.61                                      | 0.309                                          | 277.40                             | 105.71                           | 0.354                       |
| 1300                | 10105                                 | 0.323                                                   | 102.91                                      | 0.575                                              | 5.71                                      | 0.315                                          | 272.16                             | 103.42                           | 0.354                       |
| 1400                | 10084                                 | 0.352                                                   | 100.40                                      | 0.644                                              | 5.82                                      | 0.323                                          | 266.70                             | 101.05                           | 0.355                       |
| 1500                | 10062                                 | 0.381                                                   | 98.06                                       | 0.717                                              | 5.94                                      | 0.330                                          | 260.98                             | 98.59                            | 0.355                       |
| 1600                | 10039                                 | 0.410                                                   | 95.91                                       | 0.793                                              | 6.07                                      | 0.339                                          | 254.98                             | 96.03                            | 0.356                       |
| 1700                | 10015                                 | 0.440                                                   | 93.94                                       | 0.874                                              | 6.21                                      | 0.348                                          | 248.69                             | 93.36                            | 0.356                       |
| 1800                | 9990                                  | 0.470                                                   | 92.14                                       | 0.959                                              | 6.36                                      | 0.358                                          | 242.08                             | 90.58                            | 0.357                       |
| 1900                | 9963                                  | 0.500                                                   | 90.52                                       | 1.050                                              | 6.53                                      | 0.370                                          | 235.13                             | 87.68                            | 0.358                       |
| 2000                | 9935                                  | 0.530                                                   | 89.08                                       | 1.145                                              | 6.70                                      | 0.382                                          | 227.82                             | 84.64                            | 0.359                       |
| 2100                | 9905                                  | 0.561                                                   | 87.82                                       | 1.246                                              | 6.89                                      | 0.395                                          | 220.14                             | 81.47                            | 0.360                       |
| 2200                | 9874                                  | 0.592                                                   | 86.74                                       | 1.353                                              | 7.09                                      | 0.410                                          | 212.05                             | 78.16                            | 0.361                       |
| 2400                | 9806                                  | 0.654                                                   | 85.11                                       | 1.586                                              | 7.53                                      | 0.444                                          | 194.58                             | -                                | -                           |
| 2600                | 9731                                  | 0.716                                                   | 84.20                                       | 1.847                                              | 8.01                                      | 0.483                                          | -                                  | -                                | -                           |
| 2800                | 9648                                  | 0.779                                                   | 84.00                                       | 2.139                                              | 8.54                                      | 0.530                                          | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

**Revision 0: 08-05-2020**

**References**

- [1] Y.S. Touloukian, R.W. Powell, C.Y. Ho, P.G. Klemens, Thermal Conductivity - Metallic Elements and Alloys, Thermophysical Properties of Matter - The TPRC Data Series, Vol. 1, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1970.
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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.1 Molybdenum (Mo)

**Revision 0: 08-05-2020**

**References**

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### **3 Refractory Metals and Alloys**

#### **3.1 Molybdenum and Molybdenum Alloys**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Molybdenum – 30 wt% Tungsten (Mo-30W)

Revision 2: 04-26-2023

General

## Room Temperature Properties

|                                          |        |
|------------------------------------------|--------|
| Density, [kg/m <sup>3</sup> ]            | 11,900 |
| Melting Point, [K]                       | 3,090  |
| Specific Heat, [J/(g-K)]                 | 0.214  |
| Thermal Conductivity, [W/(m-K)]          | 92.2   |
| Thermal Diffusivity [cm <sup>2</sup> /s] | 0.363  |
| Linear expansion coefficient, [μm/(m-K)] | 4.92   |
| Electrical resistivity, [μΩ-m]           | 0.070  |
| Young's Modulus, [GPa]                   | 340.7  |
| Shear Modulus, [GPa]                     | 130.0  |
| Poisson's Ratio, [-]                     | 0.309  |

## Molybdenum - Tungsten Phase Diagram

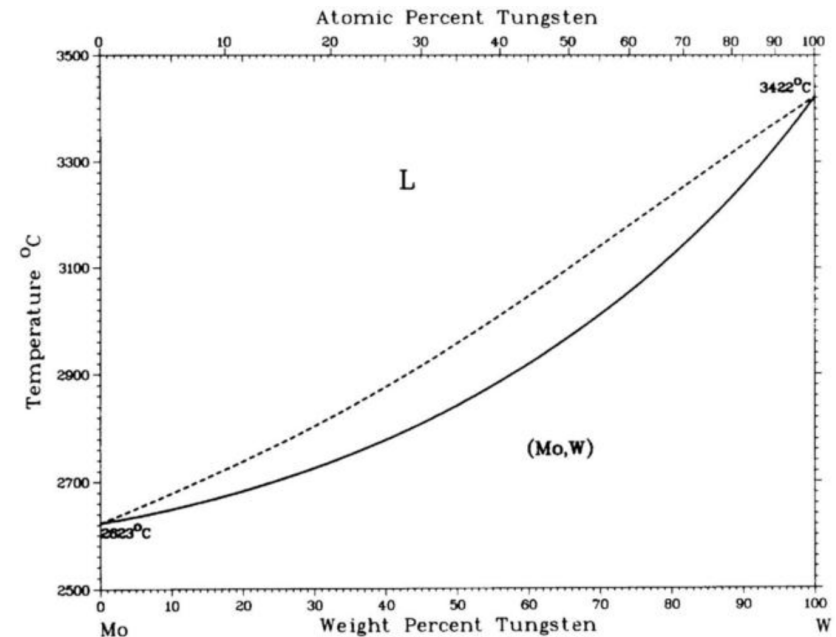


Figure 3.1.2-1: Molybdenum – Tungsten Phase Diagram [1].

## Composition

Table 3.1.2-1: Typical Composition ranges for Mo-30W (percent by weight) [2].

| Grade  |      | Mo   | W  | C   | Fe  | Si  | N    | Ni   |
|--------|------|------|----|-----|-----|-----|------|------|
| Mo-30W | Min. | 66.9 | 27 | -   | -   | -   | -    | -    |
|        | Max. | -    | 33 | .03 | .01 | .01 | .002 | .002 |



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 1: 08-05-2020

Density with Temperature

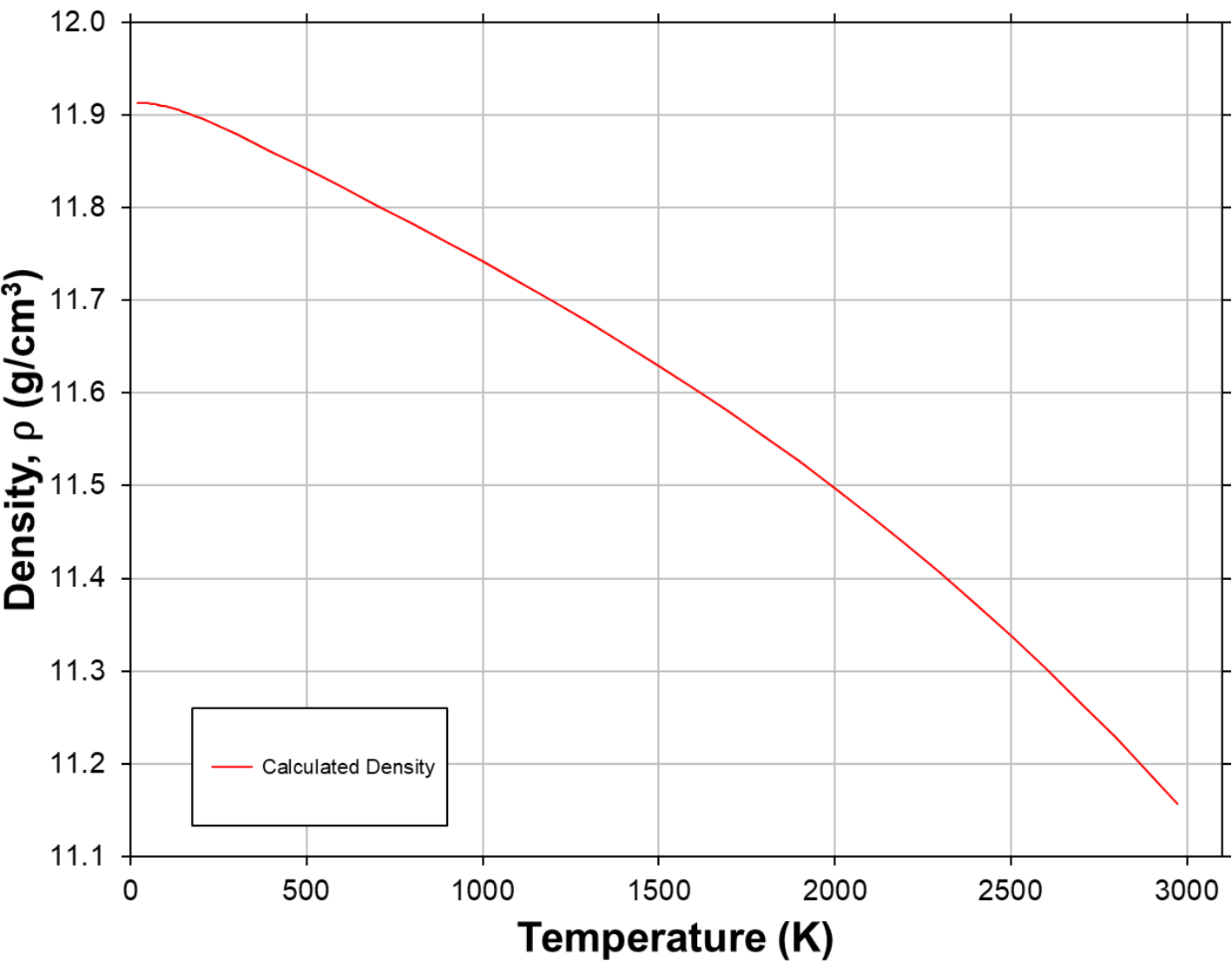


Figure 3.1.2-2: Density versus Temperature for Mo-30W. Calculated from Thermal Expansion trend.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 1: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 50                | ( -369.7 ) | 11912             | ( 743.7 )              | 1500              | ( 2240.3 ) | 11630             | ( 726.0 )              |
| 100               | ( -279.7 ) | 11909             | ( 743.5 )              | 1600              | ( 2420.3 ) | 11605             | ( 724.5 )              |
| 200               | ( -99.7 )  | 11896             | ( 742.7 )              | 1700              | ( 2600.3 ) | 11580             | ( 722.9 )              |
| 300               | ( 80.3 )   | 11880             | ( 741.6 )              | 1800              | ( 2780.3 ) | 11553             | ( 721.3 )              |
| 400               | ( 260.3 )  | 11861             | ( 740.5 )              | 1900              | ( 2960.3 ) | 11526             | ( 719.6 )              |
| 500               | ( 440.3 )  | 11841             | ( 739.3 )              | 2000              | ( 3140.3 ) | 11498             | ( 717.8 )              |
| 600               | ( 620.3 )  | 11822             | ( 738.0 )              | 2100              | ( 3320.3 ) | 11468             | ( 716.0 )              |
| 700               | ( 800.3 )  | 11802             | ( 736.8 )              | 2200              | ( 3500.3 ) | 11438             | ( 714.0 )              |
| 800               | ( 980.3 )  | 11783             | ( 735.6 )              | 2300              | ( 3680.3 ) | 11406             | ( 712.1 )              |
| 900               | ( 1160.3 ) | 11762             | ( 734.3 )              | 2400              | ( 3860.3 ) | 11373             | ( 710.0 )              |
| 1000              | ( 1340.3 ) | 11742             | ( 733.0 )              | 2500              | ( 4040.3 ) | 11338             | ( 707.9 )              |
| 1100              | ( 1520.3 ) | 11720             | ( 731.7 )              | 2600              | ( 4220.3 ) | 11303             | ( 705.6 )              |
| 1200              | ( 1700.3 ) | 11699             | ( 730.3 )              | 2700              | ( 4400.3 ) | 11266             | ( 703.3 )              |
| 1300              | ( 1880.3 ) | 11676             | ( 729.0 )              | 2800              | ( 4580.3 ) | 11227             | ( 700.9 )              |
| 1400              | ( 2060.3 ) | 11653             | ( 727.5 )              | 2900              | ( 4760.3 ) | 11187             | ( 698.4 )              |

**Application Notes:** Density trend with respect to temperature is calculated as a function of thermal expansion as shown in the equation below.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\rho_{RT} (\text{Room Temperature Density}) = 11,900 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $20 \leq T \leq 2974$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Thermal Conductivity with Temperature

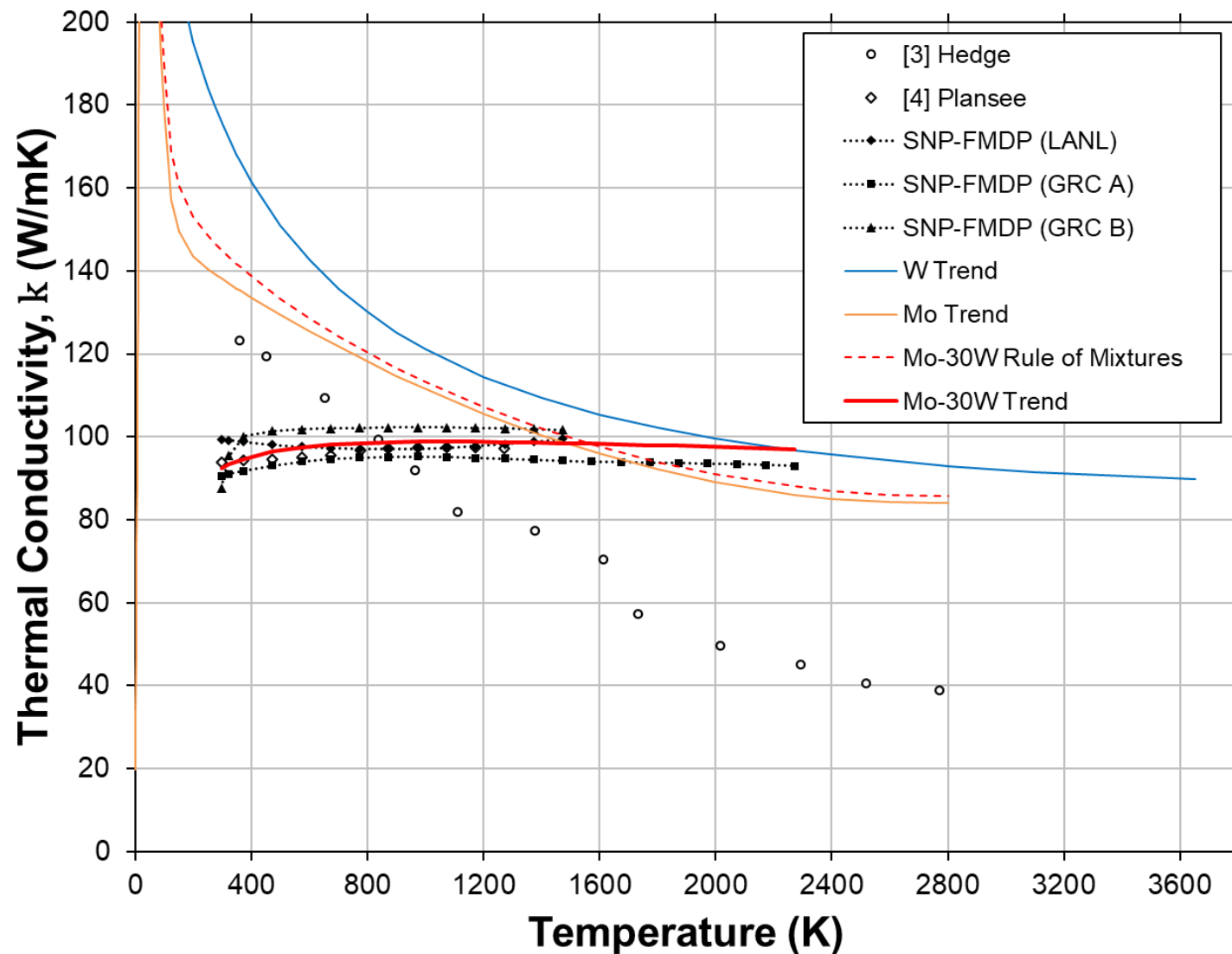


Figure 3.1.2-3: Thermal Conductivity versus Temperature of Mo-30W. Displaying fitted trend of the data with comparison to Molybdenum, Tungsten and Rule of Mixtures (70%Mo, 30%W).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 293               | ( 67.7 )   | 92.25                      | ( 640.01 )                          | 1300              | ( 1880.3 ) | 98.66                      | ( 684.50 )                          |
| 300               | ( 80.3 )   | 92.53                      | ( 641.98 )                          | 1400              | ( 2060.3 ) | 98.53                      | ( 683.60 )                          |
| 400               | ( 260.3 )  | 95.25                      | ( 660.84 )                          | 1500              | ( 2240.3 ) | 98.39                      | ( 682.61 )                          |
| 500               | ( 440.3 )  | 96.74                      | ( 671.21 )                          | 1600              | ( 2420.3 ) | 98.24                      | ( 681.59 )                          |
| 600               | ( 620.3 )  | 97.66                      | ( 677.56 )                          | 1700              | ( 2600.3 ) | 98.09                      | ( 680.55 )                          |
| 700               | ( 800.3 )  | 98.23                      | ( 681.55 )                          | 1800              | ( 2780.3 ) | 97.94                      | ( 679.49 )                          |
| 800               | ( 980.3 )  | 98.58                      | ( 683.97 )                          | 1900              | ( 2960.3 ) | 97.77                      | ( 678.36 )                          |
| 900               | ( 1160.3 ) | 98.77                      | ( 685.28 )                          | 2000              | ( 3140.3 ) | 97.59                      | ( 677.12 )                          |
| 1000              | ( 1340.3 ) | 98.84                      | ( 685.79 )                          | 2100              | ( 3320.3 ) | 97.38                      | ( 675.65 )                          |
| 1100              | ( 1520.3 ) | 98.83                      | ( 685.72 )                          | 2200              | ( 3500.3 ) | 97.12                      | ( 673.82 )                          |
| 1200              | ( 1700.3 ) | 98.77                      | ( 685.24 )                          | 2273              | ( 3631.7 ) | 96.88                      | ( 672.18 )                          |

**Application Notes:** Data from reference [3] was excluded from the Mo30W Trend Fit. All other internal data, and external data from reference [4], was used to generate the fit.

## Calculated Equation:

$$k(T) = \alpha(T) \cdot \rho(T) \cdot C_p(T)$$

$$k(T) = \text{Thermal Conductivity [W / (m · K)]}$$

$$\alpha(T) = \text{Thermal Diffusivity (m}^2\text{/s)}$$

$$\rho(T) = \text{Density (g/m}^3\text{)}$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Fit Equation:

$$k(T) = A_0 + A_1 \cdot \left(\frac{T}{1000}\right) + A_2 \cdot \left(\frac{T}{1000}\right)^2 + A_{-1} / \left(\frac{T}{1000}\right)$$

## Fit Equation Constants:

$$T \text{ Range [K]: } \underline{293 \leq T < 2273}$$

$$A_0 = 106$$

$$A_1 = -3.693$$

$$A_2 = 0.1844$$

$$A_{-1} = -3.732$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Thermal Diffusivity with Temperature

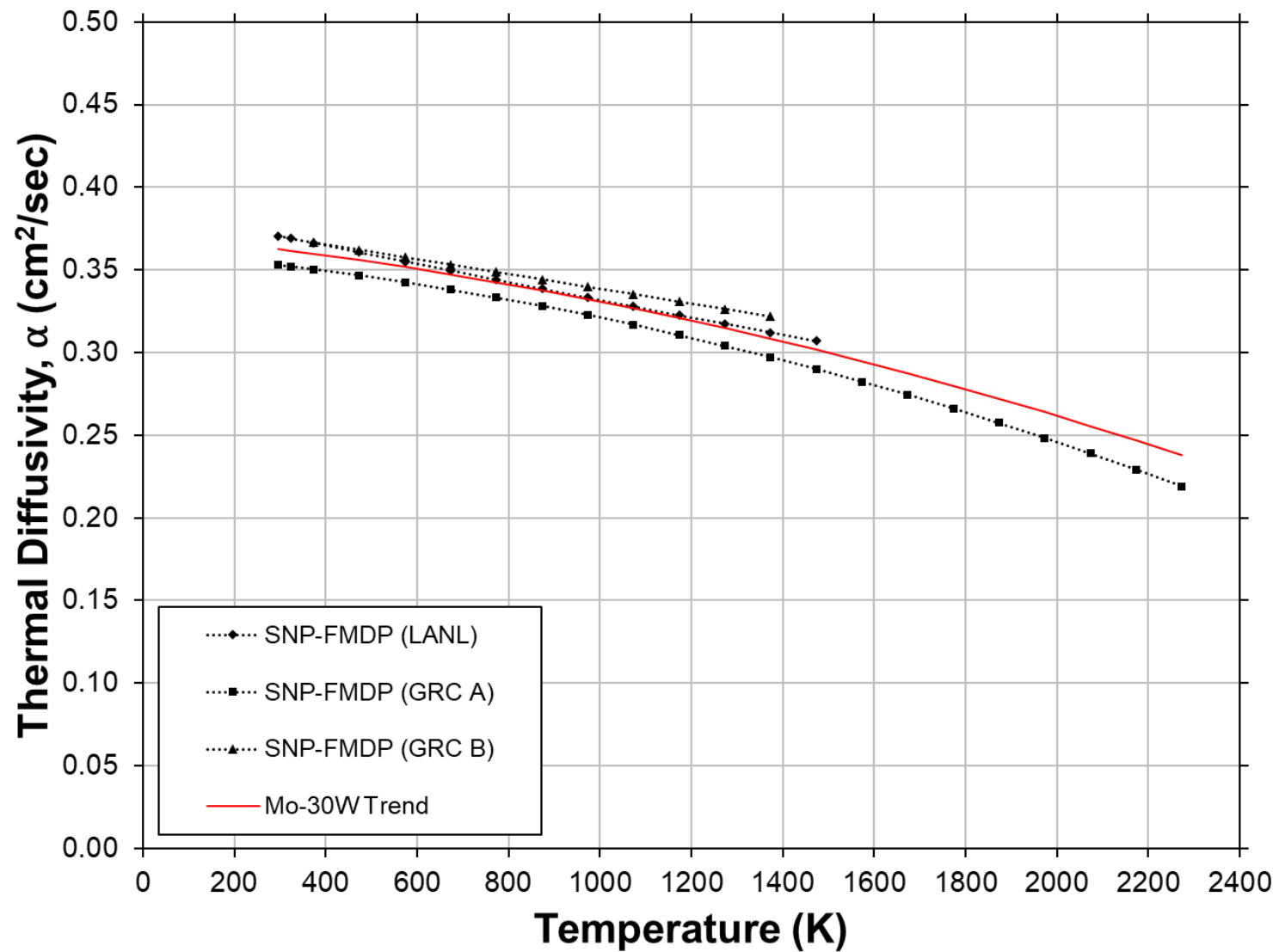


Figure 3.1.2-4: Thermal Diffusivity versus Temperature of Mo-30W.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Thermal Diffusivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Diffusivity ( $\alpha$ ) |                           | Temperature ( T ) |            | Thermal Diffusivity ( $\alpha$ ) |                           |
|-------------------|------------|----------------------------------|---------------------------|-------------------|------------|----------------------------------|---------------------------|
| K                 | ( °F )     | cm <sup>2</sup> /s               | ( in. <sup>2</sup> /sec ) | K                 | ( °F )     | cm <sup>2</sup> /s               | ( in. <sup>2</sup> /sec ) |
| 293               | ( 67.7 )   | 0.363                            | ( 0.056 )                 | 1300              | ( 1880.3 ) | 0.313                            | ( 0.049 )                 |
| 300               | ( 80.3 )   | 0.362                            | ( 0.056 )                 | 1400              | ( 2060.3 ) | 0.307                            | ( 0.048 )                 |
| 400               | ( 260.3 )  | 0.359                            | ( 0.056 )                 | 1500              | ( 2240.3 ) | 0.300                            | ( 0.046 )                 |
| 500               | ( 440.3 )  | 0.355                            | ( 0.055 )                 | 1600              | ( 2420.3 ) | 0.293                            | ( 0.045 )                 |
| 600               | ( 620.3 )  | 0.351                            | ( 0.054 )                 | 1700              | ( 2600.3 ) | 0.286                            | ( 0.044 )                 |
| 700               | ( 800.3 )  | 0.346                            | ( 0.054 )                 | 1800              | ( 2780.3 ) | 0.278                            | ( 0.043 )                 |
| 800               | ( 980.3 )  | 0.341                            | ( 0.053 )                 | 1900              | ( 2960.3 ) | 0.270                            | ( 0.042 )                 |
| 900               | ( 1160.3 ) | 0.336                            | ( 0.052 )                 | 2000              | ( 3140.3 ) | 0.262                            | ( 0.041 )                 |
| 1000              | ( 1340.3 ) | 0.331                            | ( 0.051 )                 | 2100              | ( 3320.3 ) | 0.253                            | ( 0.039 )                 |
| 1100              | ( 1520.3 ) | 0.325                            | ( 0.050 )                 | 2200              | ( 3500.3 ) | 0.244                            | ( 0.038 )                 |
| 1200              | ( 1700.3 ) | 0.319                            | ( 0.050 )                 | 2273              | ( 3631.7 ) | 0.238                            | ( 0.037 )                 |

**Application Notes:** Data for thermal diffusivity was fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\alpha(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$\alpha(T)$  = Thermal Diffusivity (cm<sup>2</sup>/s)

$T$  = Temperature [K]

## Constants:

T Range [K]:  $293 \leq T \leq 2273$

A0 = 0.3714

A1 = -0.02609

A2 = -0.01437



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Specific Heat with Temperature

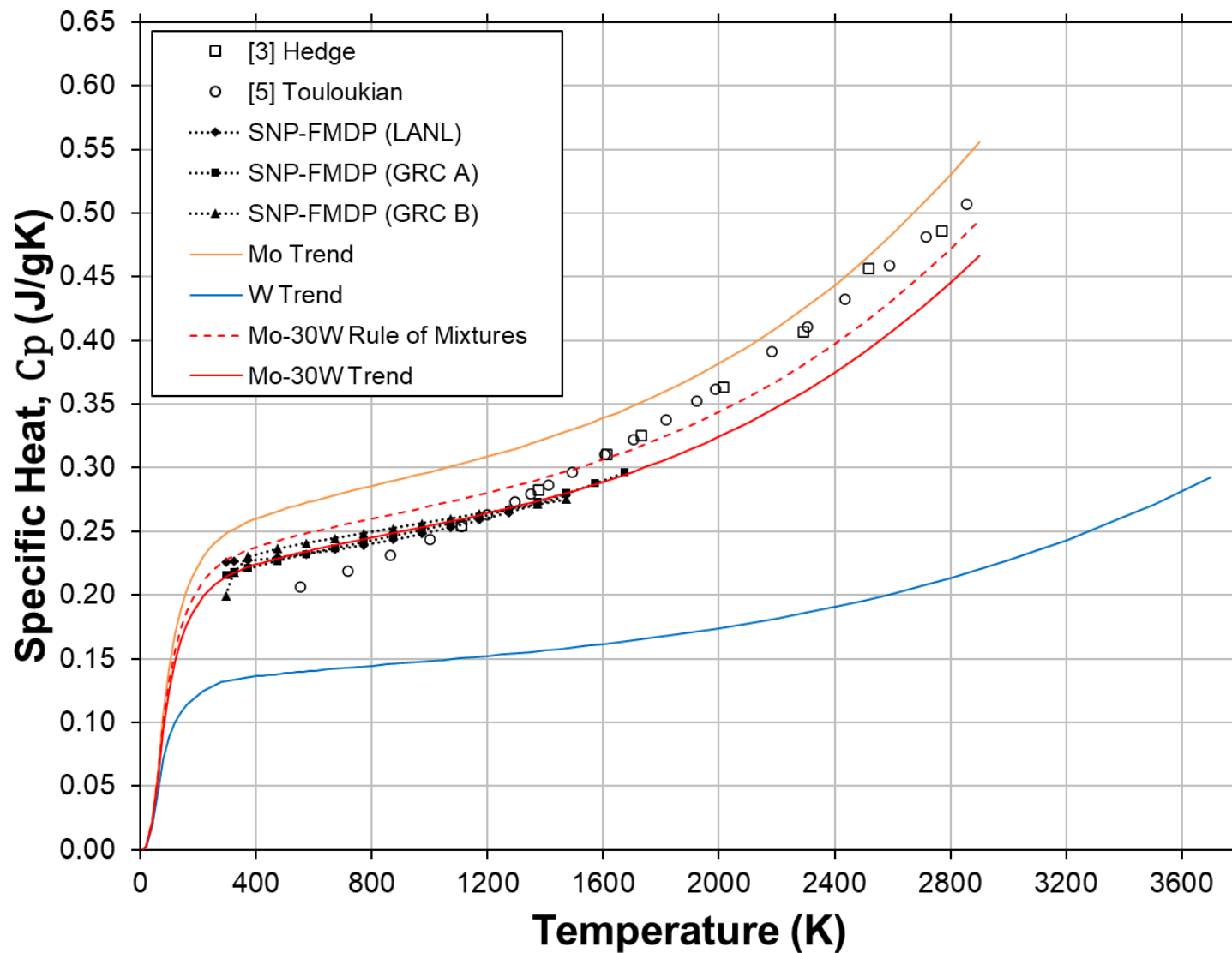


Figure 3.1.2-5: Specific Heat versus Temperature of Mo-30W. Displaying fitted trend of the data with comparison to Molybdenum, Tungsten and Rule of Mixtures (70%Mo, 30%W).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 40                | ( -387.7 ) | 0.021                            | ( 0.005 )       | 1100              | ( 1520.3 ) | 0.259                            | ( 0.062 )       |
| 60                | ( -351.7 ) | 0.056                            | ( 0.013 )       | 1200              | ( 1700.3 ) | 0.264                            | ( 0.063 )       |
| 80                | ( -315.7 ) | 0.093                            | ( 0.022 )       | 1300              | ( 1880.3 ) | 0.270                            | ( 0.064 )       |
| 100               | ( -279.7 ) | 0.124                            | ( 0.030 )       | 1400              | ( 2060.3 ) | 0.276                            | ( 0.066 )       |
| 200               | ( -99.7 )  | 0.194                            | ( 0.046 )       | 1500              | ( 2240.3 ) | 0.282                            | ( 0.067 )       |
| 300               | ( 80.3 )   | 0.215                            | ( 0.051 )       | 1600              | ( 2420.3 ) | 0.289                            | ( 0.069 )       |
| 400               | ( 260.3 )  | 0.224                            | ( 0.054 )       | 1800              | ( 2780.3 ) | 0.305                            | ( 0.073 )       |
| 500               | ( 440.3 )  | 0.230                            | ( 0.055 )       | 2000              | ( 3140.3 ) | 0.324                            | ( 0.078 )       |
| 600               | ( 620.3 )  | 0.236                            | ( 0.056 )       | 2200              | ( 3500.3 ) | 0.347                            | ( 0.083 )       |
| 700               | ( 800.3 )  | 0.240                            | ( 0.057 )       | 2400              | ( 3860.3 ) | 0.375                            | ( 0.090 )       |
| 800               | ( 980.3 )  | 0.245                            | ( 0.059 )       | 2600              | ( 4220.3 ) | 0.407                            | ( 0.097 )       |
| 900               | ( 1160.3 ) | 0.250                            | ( 0.060 )       | 2800              | ( 4580.3 ) | 0.445                            | ( 0.106 )       |
| 1000              | ( 1340.3 ) | 0.254                            | ( 0.061 )       | 2900              | ( 4760.3 ) | 0.467                            | ( 0.112 )       |

**Application Notes:** Data from references [3, 5] was excluded from the Mo30W trend fit. Data was extrapolated based on rule-of-mixtures.

## Fit Equation:

For temperature range:  $10 < T \leq 293$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right]$$

For temperature range:  $293 < T \leq 2900$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_3 \cdot \left( \frac{T}{1000} \right)^3 + B_{-2} / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$T$  = Temperature [K]

## Constants:

| T. Range [K]: | $10 \leq T \leq 293$ | $293 < T \leq 2900$          |
|---------------|----------------------|------------------------------|
| N =           | 2.967E+00            | B0 = 2.091E-01               |
| A0 =          | 3.002E+02            | B1 = 5.885E-02               |
| A1 =          | -5.805E+00           | B2 = -2.481E-02              |
| A2 =          | 1.054E+02            | B3 = 1.212E-02               |
| A3 =          | 1.129E+03            | B <sub>-2</sub> = -8.867E-04 |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Thermal Expansion with Temperature

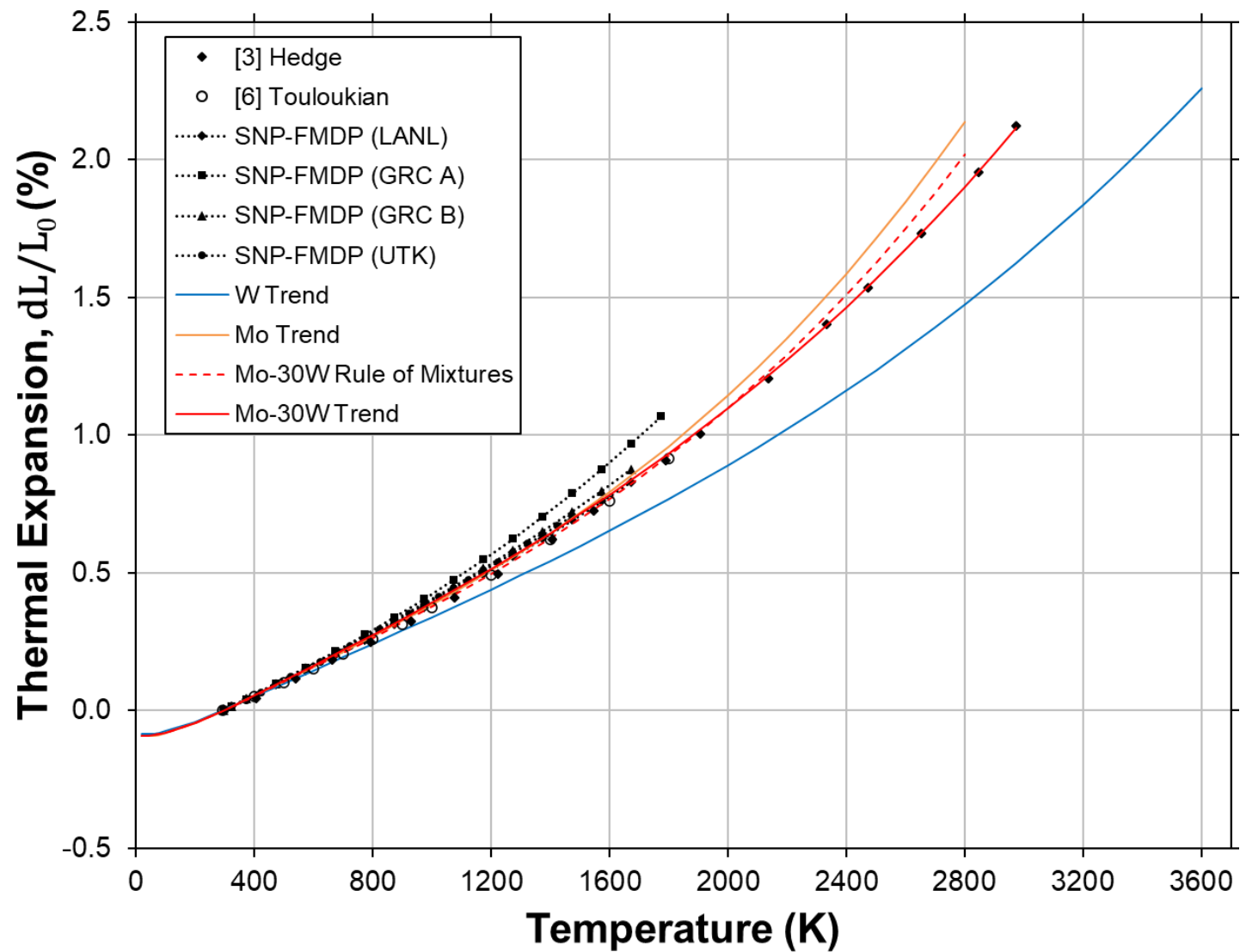


Figure 3.1.2-6: Thermal Expansion versus Temperature of Mo-30W. Displaying fitted trend of the data with comparison to Molybdenum, Tungsten and Rule of Mixtures (70%Mo, 30%W).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 50                | ( -369.7 ) | -0.090                                  | 1500              | ( 2240.3 ) | 0.713                                   |
| 100               | ( -279.7 ) | -0.081                                  | 1600              | ( 2420.3 ) | 0.784                                   |
| 200               | ( -99.7 )  | -0.045                                  | 1700              | ( 2600.3 ) | 0.857                                   |
| 300               | ( 80.3 )   | 0.001                                   | 1800              | ( 2780.3 ) | 0.934                                   |
| 400               | ( 260.3 )  | 0.054                                   | 1900              | ( 2960.3 ) | 1.014                                   |
| 500               | ( 440.3 )  | 0.109                                   | 2000              | ( 3140.3 ) | 1.097                                   |
| 600               | ( 620.3 )  | 0.163                                   | 2100              | ( 3320.3 ) | 1.183                                   |
| 700               | ( 800.3 )  | 0.219                                   | 2200              | ( 3500.3 ) | 1.273                                   |
| 800               | ( 980.3 )  | 0.275                                   | 2300              | ( 3680.3 ) | 1.367                                   |
| 900               | ( 1160.3 ) | 0.332                                   | 2400              | ( 3860.3 ) | 1.465                                   |
| 1000              | ( 1340.3 ) | 0.391                                   | 2500              | ( 4040.3 ) | 1.567                                   |
| 1100              | ( 1520.3 ) | 0.452                                   | 2600              | ( 4220.3 ) | 1.674                                   |
| 1200              | ( 1700.3 ) | 0.514                                   | 2700              | ( 4400.3 ) | 1.785                                   |
| 1300              | ( 1880.3 ) | 0.578                                   | 2800              | ( 4580.3 ) | 1.901                                   |
| 1400              | ( 2060.3 ) | 0.644                                   | 2900              | ( 4760.3 ) | 2.023                                   |

**Application Notes:** Data from reference SNP-FMDP (GRC A) was excluded from the Mo30W trend fit. All other internal data, and external data from [3, 6], was used to generate the fit. Data below room temperature is based on rule-of-mixtures.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion} [\%]$$

$$T = \text{Temperature} [K]$$

## Constants:

| T Range [K]: | $20 \leq T < 294$ | $294 \leq T \leq 2974$ |
|--------------|-------------------|------------------------|
| A0 =         | -0.09111          | -0.1599                |
| A1 =         | -0.07375          | 0.5386                 |
| A2 =         | 1.935             | -0.01953               |
| A3 =         | -2.121            | 0.03218                |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Coefficient of Thermal Expansion with Temperature

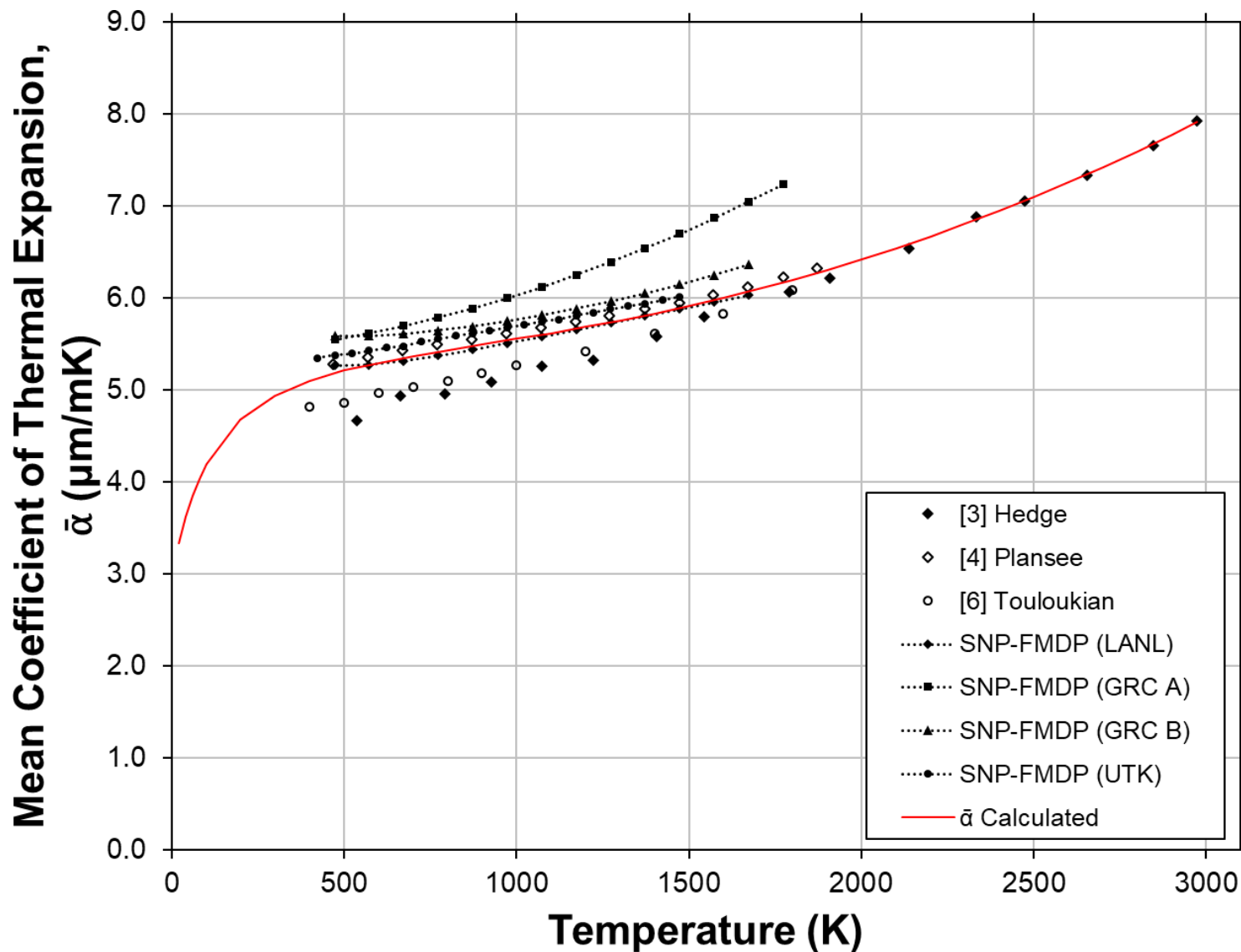


Figure 3.1.2-7: Mean Coefficient of Thermal Expansion versus Temperature of Mo-30W. Calculated from trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 50                | ( -369.7 ) | 3.741                                 | ( 2.078 )                                           | 1500              | ( 2240.3 ) | 5.912                                 | ( 3.284 )                                           |
| 100               | ( -279.7 ) | 4.187                                 | ( 2.326 )                                           | 1600              | ( 2420.3 ) | 5.999                                 | ( 3.333 )                                           |
| 200               | ( -99.7 )  | 4.676                                 | ( 2.598 )                                           | 1700              | ( 2600.3 ) | 6.093                                 | ( 3.385 )                                           |
| 300               | ( 80.3 )   | 4.936                                 | ( 2.742 )                                           | 1800              | ( 2780.3 ) | 6.194                                 | ( 3.441 )                                           |
| 400               | ( 260.3 )  | 5.096                                 | ( 2.831 )                                           | 1900              | ( 2960.3 ) | 6.301                                 | ( 3.501 )                                           |
| 500               | ( 440.3 )  | 5.208                                 | ( 2.893 )                                           | 2000              | ( 3140.3 ) | 6.415                                 | ( 3.564 )                                           |
| 600               | ( 620.3 )  | 5.293                                 | ( 2.941 )                                           | 2100              | ( 3320.3 ) | 6.536                                 | ( 3.631 )                                           |
| 700               | ( 800.3 )  | 5.364                                 | ( 2.980 )                                           | 2200              | ( 3500.3 ) | 6.665                                 | ( 3.703 )                                           |
| 800               | ( 980.3 )  | 5.429                                 | ( 3.016 )                                           | 2300              | ( 3680.3 ) | 6.800                                 | ( 3.778 )                                           |
| 900               | ( 1160.3 ) | 5.490                                 | ( 3.050 )                                           | 2400              | ( 3860.3 ) | 6.943                                 | ( 3.857 )                                           |
| 1000              | ( 1340.3 ) | 5.552                                 | ( 3.084 )                                           | 2500              | ( 4040.3 ) | 7.094                                 | ( 3.941 )                                           |
| 1100              | ( 1520.3 ) | 5.616                                 | ( 3.120 )                                           | 2600              | ( 4220.3 ) | 7.251                                 | ( 4.029 )                                           |
| 1200              | ( 1700.3 ) | 5.683                                 | ( 3.157 )                                           | 2700              | ( 4400.3 ) | 7.417                                 | ( 4.120 )                                           |
| 1300              | ( 1880.3 ) | 5.754                                 | ( 3.197 )                                           | 2800              | ( 4580.3 ) | 7.590                                 | ( 4.217 )                                           |
| 1400              | ( 2060.3 ) | 5.830                                 | ( 3.239 )                                           | 2900              | ( 4760.3 ) | 7.770                                 | ( 4.317 )                                           |

**Application Notes:** Data from reference SNP-FMDP (GRC A) was excluded from the Mo30W trend fit. Data below room temperature is based on rule-of-mixtures. The trend is compared against data from references [3, 4, 6].

## Fit Equation:

$$\bar{\alpha}(T) = \left( A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 + A3 \cdot \left( \frac{T}{1000} \right)^3 \right) / \left( \left( \frac{T}{1000} \right) + B0 \right)$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T Range [K]:

$$20 \leq T \leq 2974$$

|      |         |      |        |
|------|---------|------|--------|
| A0 = | 0.4278  | A3 = | 0.3975 |
| A1 = | 5.979   | B0 = | 0.144  |
| A2 = | -0.4529 |      |        |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Electrical Resistivity with Temperature

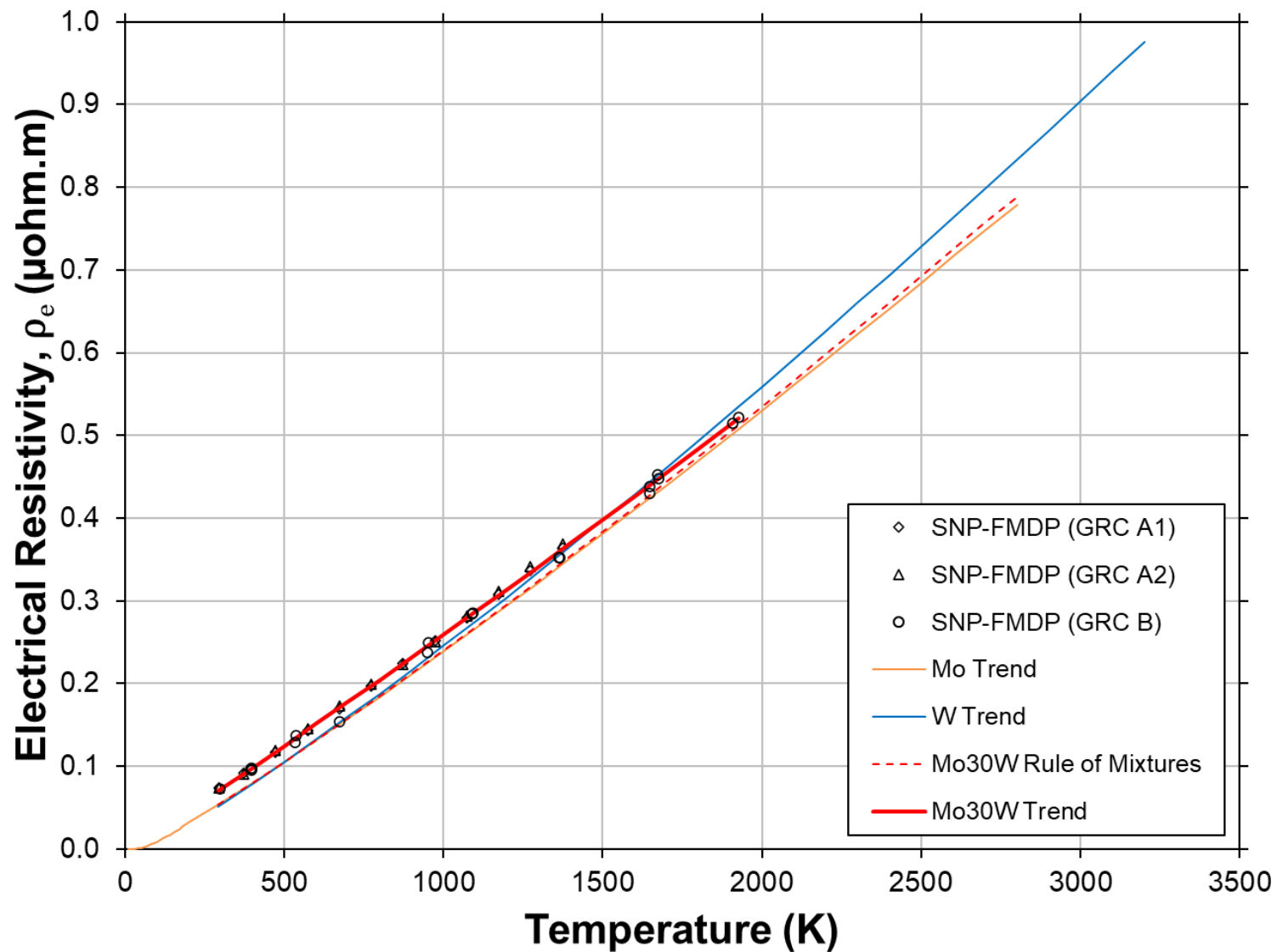


Figure 3.1.2-8: Electrical Resistivity versus Temperature for Mo-30W.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                           | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                           |
|-------------------|------------|-------------------------------------|---------------------------|-------------------|------------|-------------------------------------|---------------------------|
| K                 | ( °F )     | $\mu\Omega\text{-m}$                | ( $\mu\Omega\text{-in}$ ) | K                 | ( °F )     | $\mu\Omega\text{-m}$                | ( $\mu\Omega\text{-in}$ ) |
| 293               | ( 67.7 )   | 0.0702                              | ( 0.004 )                 | 1200              | ( 1700.3 ) | 0.3139                              | ( 0.020 )                 |
| 400               | ( 260.3 )  | 0.0981                              | ( 0.006 )                 | 1300              | ( 1880.3 ) | 0.3418                              | ( 0.021 )                 |
| 500               | ( 440.3 )  | 0.1244                              | ( 0.008 )                 | 1400              | ( 2060.3 ) | 0.3698                              | ( 0.023 )                 |
| 600               | ( 620.3 )  | 0.1509                              | ( 0.009 )                 | 1500              | ( 2240.3 ) | 0.3980                              | ( 0.025 )                 |
| 700               | ( 800.3 )  | 0.1776                              | ( 0.011 )                 | 1600              | ( 2420.3 ) | 0.4264                              | ( 0.027 )                 |
| 800               | ( 980.3 )  | 0.2045                              | ( 0.013 )                 | 1700              | ( 2600.3 ) | 0.4550                              | ( 0.028 )                 |
| 900               | ( 1160.3 ) | 0.2316                              | ( 0.014 )                 | 1800              | ( 2780.3 ) | 0.4838                              | ( 0.030 )                 |
| 1000              | ( 1340.3 ) | 0.2588                              | ( 0.016 )                 | 1900              | ( 2960.3 ) | 0.5128                              | ( 0.032 )                 |
| 1100              | ( 1520.3 ) | 0.2863                              | ( 0.018 )                 | 1926              | ( 3007.1 ) | 0.5203                              | ( 0.032 )                 |

**Application Notes:** Data for electrical resistivity was fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]: 293 ≤ T < 1926

A0 = -5.182E-03

A1 = 2.545E-01

A2 = 9.527E-03



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Young's Modulus with Temperature

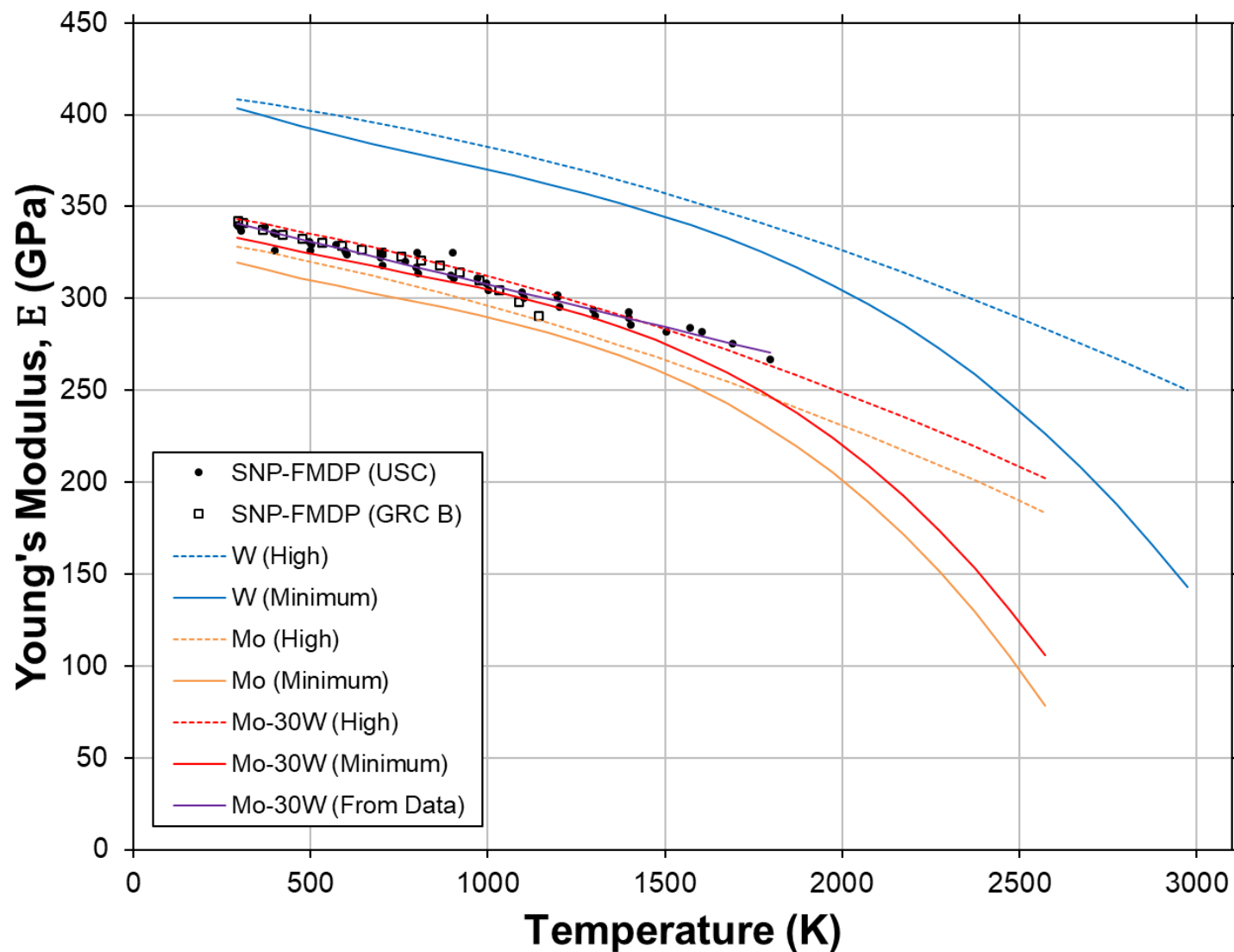


Figure 3.1.2-9: Young's modulus versus temperature for Mo-30W, with comparison to molybdenum and tungsten trends. Mo-30W high and minimum trends were calculated using the rule of mixtures, while the "from data" curve was fit using data alone.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Young's Modulus with Temperature

Minimum Young's Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   |
| 293               | ( 67.7 )   | 333.26                | ( 48.36 ) |
| 300               | ( 80.3 )   | 332.93                | ( 48.31 ) |
| 400               | ( 260.3 )  | 328.47                | ( 47.66 ) |
| 500               | ( 440.3 )  | 324.35                | ( 47.06 ) |
| 700               | ( 800.3 )  | 316.69                | ( 45.95 ) |
| 900               | ( 1160.3 ) | 308.94                | ( 44.83 ) |
| 1100              | ( 1520.3 ) | 300.11                | ( 43.55 ) |
| 1300              | ( 1880.3 ) | 289.20                | ( 41.96 ) |
| 1500              | ( 2240.3 ) | 275.22                | ( 39.93 ) |
| 1700              | ( 2600.3 ) | 257.17                | ( 37.32 ) |
| 1900              | ( 2960.3 ) | 234.05                | ( 33.96 ) |
| 2100              | ( 3320.3 ) | 204.87                | ( 29.73 ) |
| 2300              | ( 3680.3 ) | 168.62                | ( 24.47 ) |
| 2500              | ( 4040.3 ) | 124.32                | ( 18.04 ) |
| 2573              | ( 4171.7 ) | 105.94                | ( 15.37 ) |

High Young's Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   |
| 293               | ( 67.7 )   | 343.29                | ( 49.81 ) |
| 300               | ( 80.3 )   | 343.04                | ( 49.77 ) |
| 400               | ( 260.3 )  | 339.32                | ( 49.24 ) |
| 500               | ( 440.3 )  | 335.38                | ( 48.66 ) |
| 700               | ( 800.3 )  | 326.81                | ( 47.42 ) |
| 900               | ( 1160.3 ) | 317.33                | ( 46.04 ) |
| 1100              | ( 1520.3 ) | 306.92                | ( 44.53 ) |
| 1300              | ( 1880.3 ) | 295.61                | ( 42.89 ) |
| 1500              | ( 2240.3 ) | 283.37                | ( 41.12 ) |
| 1700              | ( 2600.3 ) | 270.22                | ( 39.21 ) |
| 1900              | ( 2960.3 ) | 256.16                | ( 37.17 ) |
| 2100              | ( 3320.3 ) | 241.17                | ( 34.99 ) |
| 2300              | ( 3680.3 ) | 225.28                | ( 32.69 ) |
| 2500              | ( 4040.3 ) | 208.46                | ( 30.25 ) |
| 2573              | ( 4171.7 ) | 202.10                | ( 29.32 ) |

**Application Notes:** Young's modulus minimum and high curves for Mo-30W were calculated from corresponding molybdenum and tungsten curves using the rule of mixtures and a tungsten volume fraction of 0.186. Equivalent equations can be found below for readers' convenience.

## Fit Equations:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$E(T) = \text{Young's Modulus [GPa]}$$

$$T = \text{Temperature [K]}$$

$$T \text{ Range [K]: } 293 \leq T \leq 2573$$

## Constants:

| Curve: | <u>Minimum</u> | <u>High</u> |
|--------|----------------|-------------|
| A0 =   | 349.7          | 352.8       |
| A1 =   | -66.79         | -29.11      |
| A2 =   | 42.58          | -11.45      |
| A3 =   | -20.77         | 0           |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Young's Modulus with Temperature

## Data-Based Young's Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 293               | ( 67.7 )   | 340.72                | ( 49.44 ) | 1000              | ( 1340.3 ) | 307.72                | ( 44.65 ) |
| 300               | ( 80.3 )   | 340.40                | ( 49.39 ) | 1050              | ( 1430.3 ) | 305.39                | ( 44.31 ) |
| 350               | ( 170.3 )  | 338.06                | ( 49.05 ) | 1100              | ( 1520.3 ) | 303.05                | ( 43.97 ) |
| 400               | ( 260.3 )  | 335.73                | ( 48.71 ) | 1150              | ( 1610.3 ) | 300.72                | ( 43.63 ) |
| 450               | ( 350.3 )  | 333.39                | ( 48.38 ) | 1200              | ( 1700.3 ) | 298.38                | ( 43.30 ) |
| 500               | ( 440.3 )  | 331.06                | ( 48.04 ) | 1250              | ( 1790.3 ) | 296.05                | ( 42.96 ) |
| 550               | ( 530.3 )  | 328.73                | ( 47.70 ) | 1300              | ( 1880.3 ) | 293.72                | ( 42.62 ) |
| 600               | ( 620.3 )  | 326.39                | ( 47.36 ) | 1350              | ( 1970.3 ) | 291.38                | ( 42.28 ) |
| 650               | ( 710.3 )  | 324.06                | ( 47.02 ) | 1400              | ( 2060.3 ) | 289.05                | ( 41.94 ) |
| 700               | ( 800.3 )  | 321.72                | ( 46.68 ) | 1450              | ( 2150.3 ) | 286.71                | ( 41.60 ) |
| 750               | ( 890.3 )  | 319.39                | ( 46.34 ) | 1500              | ( 2240.3 ) | 284.38                | ( 41.26 ) |
| 800               | ( 980.3 )  | 317.06                | ( 46.00 ) | 1550              | ( 2330.3 ) | 282.05                | ( 40.92 ) |
| 850               | ( 1070.3 ) | 314.72                | ( 45.67 ) | 1600              | ( 2420.3 ) | 279.71                | ( 40.59 ) |
| 900               | ( 1160.3 ) | 312.39                | ( 45.33 ) | 1700              | ( 2600.3 ) | 275.04                | ( 39.91 ) |
| 950               | ( 1250.3 ) | 310.05                | ( 44.99 ) | 1797              | ( 2774.9 ) | 270.52                | ( 39.25 ) |

**Application Notes:** The “from data” equation for Young’s modulus was fitted using SNP-FDP (USC) and SNP-FDP (SR) data, and it approximates the property trend with respect to temperature.

### Fit Equations:

$$E(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right)$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

T Range [K]:  $293 \leq T \leq 1797$

### Constants:

Curve: From Data

A0 = 354.4

A1 = -46.68



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 0: 08-05-2020

Shear Modulus with Temperature

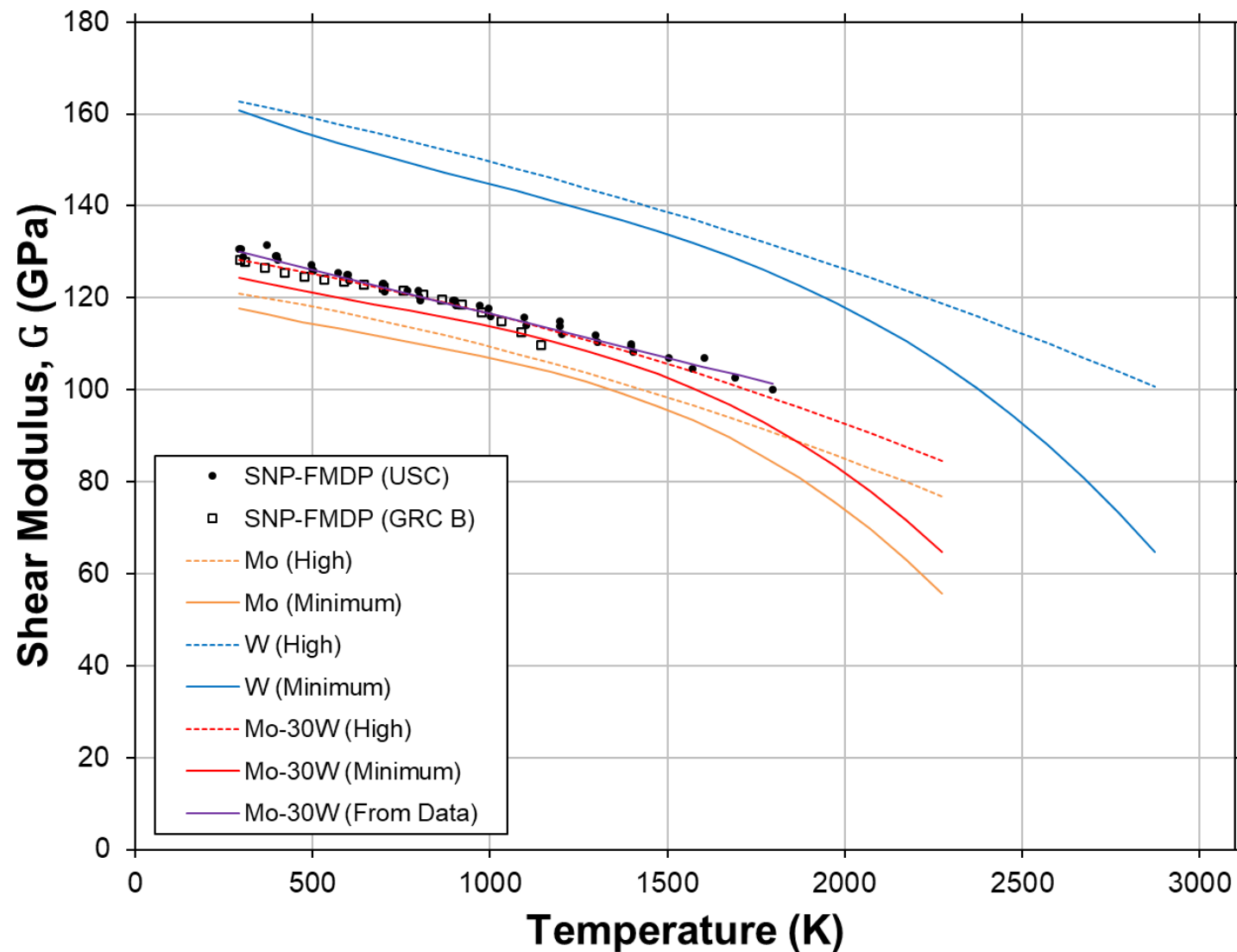


Figure 3.1.2-10: Shear modulus versus temperature for Mo-30W, with comparison to molybdenum and tungsten trends. Mo-30W high and minimum trends were calculated using the rule of mixtures, while the “from data” curve was fit using data alone.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Shear Modulus with Temperature

Minimum Shear Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( E ) |           |
|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   |
| 293               | ( 67.7 )   | 124.46              | ( 18.06 ) |
| 300               | ( 80.3 )   | 124.34              | ( 18.04 ) |
| 450               | ( 350.3 )  | 121.91              | ( 17.69 ) |
| 600               | ( 620.3 )  | 119.70              | ( 17.37 ) |
| 750               | ( 890.3 )  | 117.56              | ( 17.06 ) |
| 900               | ( 1160.3 ) | 115.35              | ( 16.74 ) |
| 1050              | ( 1430.3 ) | 112.90              | ( 16.38 ) |
| 1200              | ( 1700.3 ) | 110.07              | ( 15.97 ) |
| 1350              | ( 1970.3 ) | 106.69              | ( 15.48 ) |
| 1500              | ( 2240.3 ) | 102.62              | ( 14.89 ) |
| 1650              | ( 2510.3 ) | 97.70               | ( 14.18 ) |
| 1800              | ( 2780.3 ) | 91.78               | ( 13.32 ) |
| 1950              | ( 3050.3 ) | 84.71               | ( 12.29 ) |
| 2100              | ( 3320.3 ) | 76.32               | ( 11.07 ) |
| 2273              | ( 3631.7 ) | 64.83               | ( 9.41 )  |

High Shear Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( E ) |           |
|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   |
| 293               | ( 67.7 )   | 128.23              | ( 18.61 ) |
| 300               | ( 80.3 )   | 128.13              | ( 18.59 ) |
| 450               | ( 350.3 )  | 126.00              | ( 18.28 ) |
| 600               | ( 620.3 )  | 123.68              | ( 17.95 ) |
| 750               | ( 890.3 )  | 121.17              | ( 17.58 ) |
| 900               | ( 1160.3 ) | 118.45              | ( 17.19 ) |
| 1050              | ( 1430.3 ) | 115.55              | ( 16.77 ) |
| 1200              | ( 1700.3 ) | 112.45              | ( 16.32 ) |
| 1350              | ( 1970.3 ) | 109.15              | ( 15.84 ) |
| 1500              | ( 2240.3 ) | 105.66              | ( 15.33 ) |
| 1650              | ( 2510.3 ) | 101.97              | ( 14.80 ) |
| 1800              | ( 2780.3 ) | 98.09               | ( 14.23 ) |
| 1950              | ( 3050.3 ) | 94.01               | ( 13.64 ) |
| 2100              | ( 3320.3 ) | 89.74               | ( 13.02 ) |
| 2273              | ( 3631.7 ) | 84.57               | ( 12.27 ) |

**Application Notes:** Shear modulus minimum and high curves for Mo-30W were calculated from corresponding molybdenum and tungsten curves using the rule of mixtures and a tungsten volume fraction of 0.186. Equivalent equations can be found below for readers' convenience.

## Fit Equations:

$$G(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

T Range [K]:  $293 \leq T \leq 2273$

## Constants:

| Curve: | <u>Minimum</u> | <u>High</u> |
|--------|----------------|-------------|
| A0 =   | 130.5          | 131.8       |
| A1 =   | -24.43         | -10.93      |
| A2 =   | 15.26          | -4.332      |
| A3 =   | -7.577         | 0           |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Shear Modulus with Temperature

## Data-Based Shear Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( G ) |           | Temperature ( T ) |            | Shear Modulus ( G ) |           |
|-------------------|------------|---------------------|-----------|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   | K                 | ( °F )     | GPa                 | ( Msi )   |
| 293               | ( 67.7 )   | 130.01              | ( 18.86 ) | 1000              | ( 1340.3 ) | 116.53              | ( 16.91 ) |
| 300               | ( 80.3 )   | 129.88              | ( 18.85 ) | 1050              | ( 1430.3 ) | 115.58              | ( 16.77 ) |
| 350               | ( 170.3 )  | 128.93              | ( 18.71 ) | 1100              | ( 1520.3 ) | 114.62              | ( 16.63 ) |
| 400               | ( 260.3 )  | 127.97              | ( 18.57 ) | 1150              | ( 1610.3 ) | 113.67              | ( 16.49 ) |
| 450               | ( 350.3 )  | 127.02              | ( 18.43 ) | 1200              | ( 1700.3 ) | 112.72              | ( 16.36 ) |
| 500               | ( 440.3 )  | 126.07              | ( 18.29 ) | 1250              | ( 1790.3 ) | 111.76              | ( 16.22 ) |
| 550               | ( 530.3 )  | 125.11              | ( 18.15 ) | 1300              | ( 1880.3 ) | 110.81              | ( 16.08 ) |
| 600               | ( 620.3 )  | 124.16              | ( 18.02 ) | 1350              | ( 1970.3 ) | 109.86              | ( 15.94 ) |
| 650               | ( 710.3 )  | 123.20              | ( 17.88 ) | 1400              | ( 2060.3 ) | 108.90              | ( 15.80 ) |
| 700               | ( 800.3 )  | 122.25              | ( 17.74 ) | 1450              | ( 2150.3 ) | 107.95              | ( 15.66 ) |
| 750               | ( 890.3 )  | 121.30              | ( 17.60 ) | 1500              | ( 2240.3 ) | 107.00              | ( 15.52 ) |
| 800               | ( 980.3 )  | 120.34              | ( 17.46 ) | 1550              | ( 2330.3 ) | 106.04              | ( 15.39 ) |
| 850               | ( 1070.3 ) | 119.39              | ( 17.32 ) | 1600              | ( 2420.3 ) | 105.09              | ( 15.25 ) |
| 900               | ( 1160.3 ) | 118.44              | ( 17.19 ) | 1700              | ( 2600.3 ) | 103.18              | ( 14.97 ) |
| 950               | ( 1250.3 ) | 117.48              | ( 17.05 ) | 1797              | ( 2774.9 ) | 101.33              | ( 14.70 ) |

**Application Notes:** The “from data” equation for shear modulus was fitted SNP-FDP (USC) and SNP-FDP (SR) data, and it approximates the property trend with respect to temperature.

### Fit Equations:

$$G(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right)$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

T Range [K]:  $293 \leq T \leq 1797$

### Constants:

Curve: From Data

A0 = 135.6

A1 = -19.07



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Poisson's Ratio with Temperature

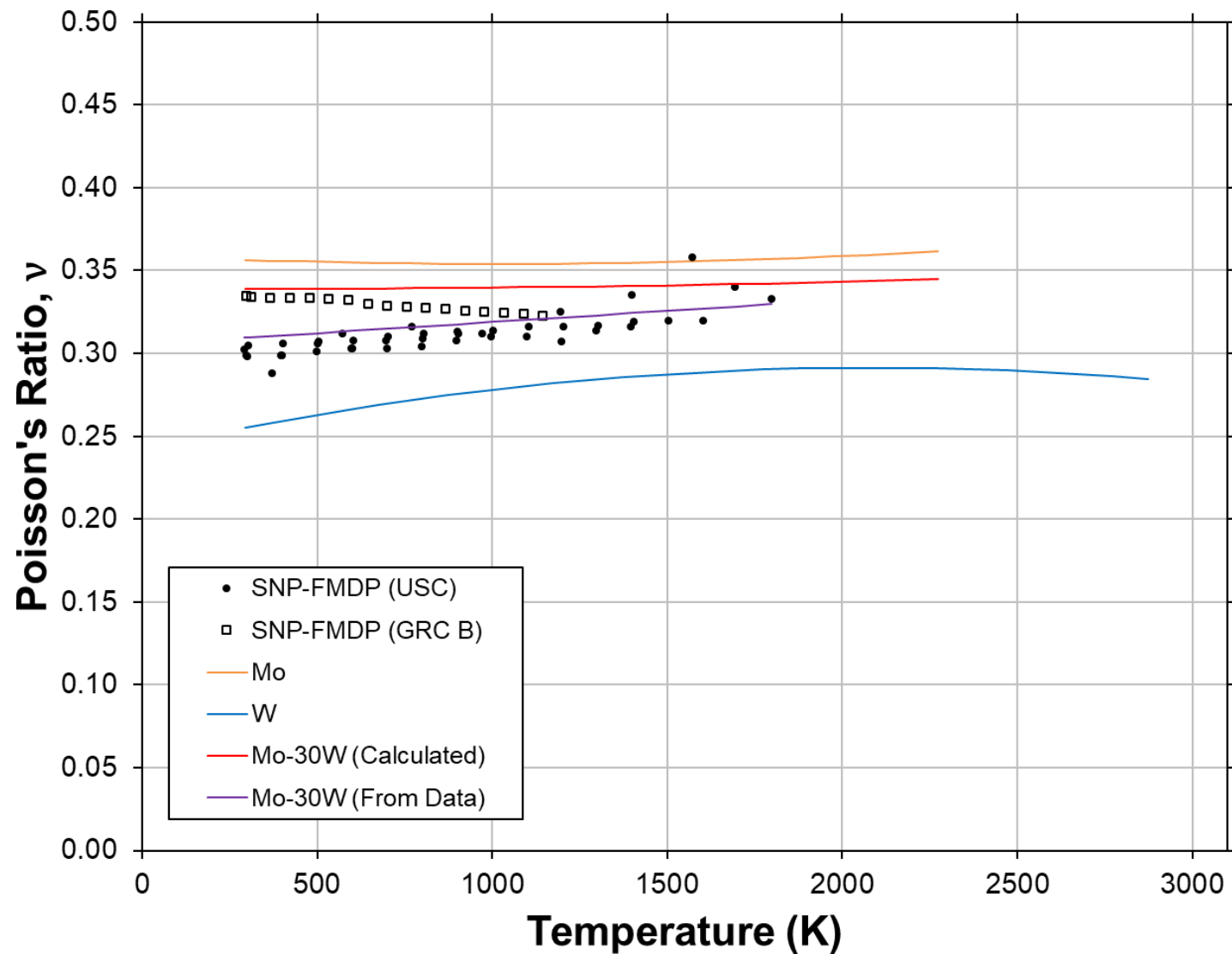


Figure 3.1.2-11: Poisson's ratio versus temperature for Mo-30W, with comparison to molybdenum and tungsten trends. The "calculated" curve was developed from the fitted Young's modulus (high) and shear modulus (high) curves, while the "from data" curve was fit using data alone.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Poisson's Ratio with Temperature

Rule of Mixtures-Based Poisson's Ratio at 100% Theoretical Density

| Temperature ( T ) |            | Poisson's Ratio ( ν ) | Temperature ( T ) |            | Poisson's Ratio ( ν ) |
|-------------------|------------|-----------------------|-------------------|------------|-----------------------|
| K                 | ( °F )     |                       | K                 | ( °F )     |                       |
| 293               | ( 67.7 )   | 0.339                 | 1000              | ( 1340.3 ) | 0.340                 |
| 300               | ( 80.3 )   | 0.339                 | 1050              | ( 1430.3 ) | 0.340                 |
| 350               | ( 170.3 )  | 0.339                 | 1100              | ( 1520.3 ) | 0.340                 |
| 400               | ( 260.3 )  | 0.339                 | 1200              | ( 1700.3 ) | 0.340                 |
| 450               | ( 350.3 )  | 0.339                 | 1300              | ( 1880.3 ) | 0.340                 |
| 500               | ( 440.3 )  | 0.339                 | 1400              | ( 2060.3 ) | 0.341                 |
| 550               | ( 530.3 )  | 0.339                 | 1500              | ( 2240.3 ) | 0.341                 |
| 600               | ( 620.3 )  | 0.339                 | 1600              | ( 2420.3 ) | 0.341                 |
| 650               | ( 710.3 )  | 0.339                 | 1700              | ( 2600.3 ) | 0.342                 |
| 700               | ( 800.3 )  | 0.339                 | 1800              | ( 2780.3 ) | 0.342                 |
| 750               | ( 890.3 )  | 0.339                 | 1900              | ( 2960.3 ) | 0.343                 |
| 800               | ( 980.3 )  | 0.339                 | 2000              | ( 3140.3 ) | 0.343                 |
| 850               | ( 1070.3 ) | 0.339                 | 2100              | ( 3320.3 ) | 0.344                 |
| 900               | ( 1160.3 ) | 0.339                 | 2200              | ( 3500.3 ) | 0.344                 |
| 950               | ( 1250.3 ) | 0.340                 | 2273              | ( 3631.7 ) | 0.345                 |

**Application Notes:** The Poisson's ratio (calculated) curve for Mo-30W was calculated from corresponding molybdenum and tungsten curves using the rule of mixtures and a tungsten volume fraction of 0.186. An equivalent equation can be found below for readers' convenience.

## Fit Equations:

$$\nu(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\nu(T)$  = Poisson's Ratio

$T$  = Temperature [K]

## Constants:

T Range [K]: 293 ≤ T ≤ 2273

A0 = 0.3383

A1 = 1.164E-3

A2 = -1.842E-4

A3 = 4.13E-4



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Poisson's Ratio with Temperature

Data-Based Poisson's Ratio at 100% Theoretical Density

| Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|------------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )     |                           | K                 | ( °F )     |                           |
| 293               | ( 67.7 )   | 0.309                     | 1000              | ( 1340.3 ) | 0.319                     |
| 300               | ( 80.3 )   | 0.310                     | 1050              | ( 1430.3 ) | 0.320                     |
| 350               | ( 170.3 )  | 0.310                     | 1100              | ( 1520.3 ) | 0.320                     |
| 400               | ( 260.3 )  | 0.311                     | 1150              | ( 1610.3 ) | 0.321                     |
| 450               | ( 350.3 )  | 0.312                     | 1200              | ( 1700.3 ) | 0.322                     |
| 500               | ( 440.3 )  | 0.312                     | 1250              | ( 1790.3 ) | 0.322                     |
| 550               | ( 530.3 )  | 0.313                     | 1300              | ( 1880.3 ) | 0.323                     |
| 600               | ( 620.3 )  | 0.314                     | 1350              | ( 1970.3 ) | 0.324                     |
| 650               | ( 710.3 )  | 0.314                     | 1400              | ( 2060.3 ) | 0.324                     |
| 700               | ( 800.3 )  | 0.315                     | 1450              | ( 2150.3 ) | 0.325                     |
| 750               | ( 890.3 )  | 0.316                     | 1500              | ( 2240.3 ) | 0.326                     |
| 800               | ( 980.3 )  | 0.316                     | 1550              | ( 2330.3 ) | 0.326                     |
| 850               | ( 1070.3 ) | 0.317                     | 1600              | ( 2420.3 ) | 0.327                     |
| 900               | ( 1160.3 ) | 0.318                     | 1700              | ( 2600.3 ) | 0.328                     |
| 950               | ( 1250.3 ) | 0.318                     | 1797              | ( 2774.9 ) | 0.330                     |

**Application Notes:** The “from data” equation for Poisson's ratio was fitted using SNP-FMDP (USC) and SNP-FMDP (GRC B) data, and it approximates the property trend with respect to temperature.

## Fit Equations:

$$\nu(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right)$$

$\nu(T)$  = Poisson's Ratio

$T$  = Temperature [K]

## Constants:

T Range [K]: 293 ≤ T ≤ 1797

A0 = 0.3055

A1 = 0.01346



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Yield Strength with Temperature

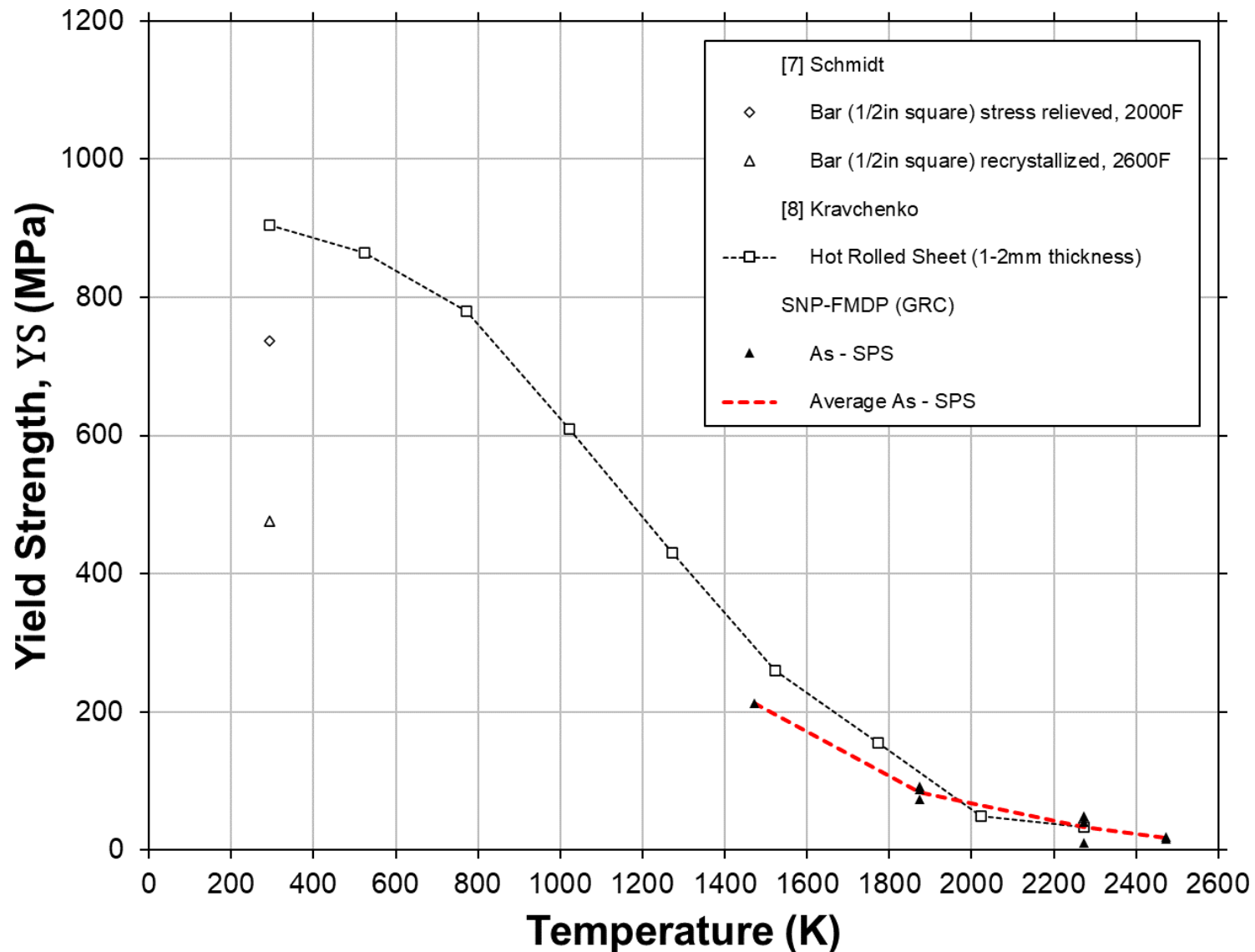


Figure 3.1.2-12: Yield Strength versus Temperature for Mo-30W. Displaying fitted trend for 1-2 mm thick hot rolled sheet.



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Tensile Strength with Temperature

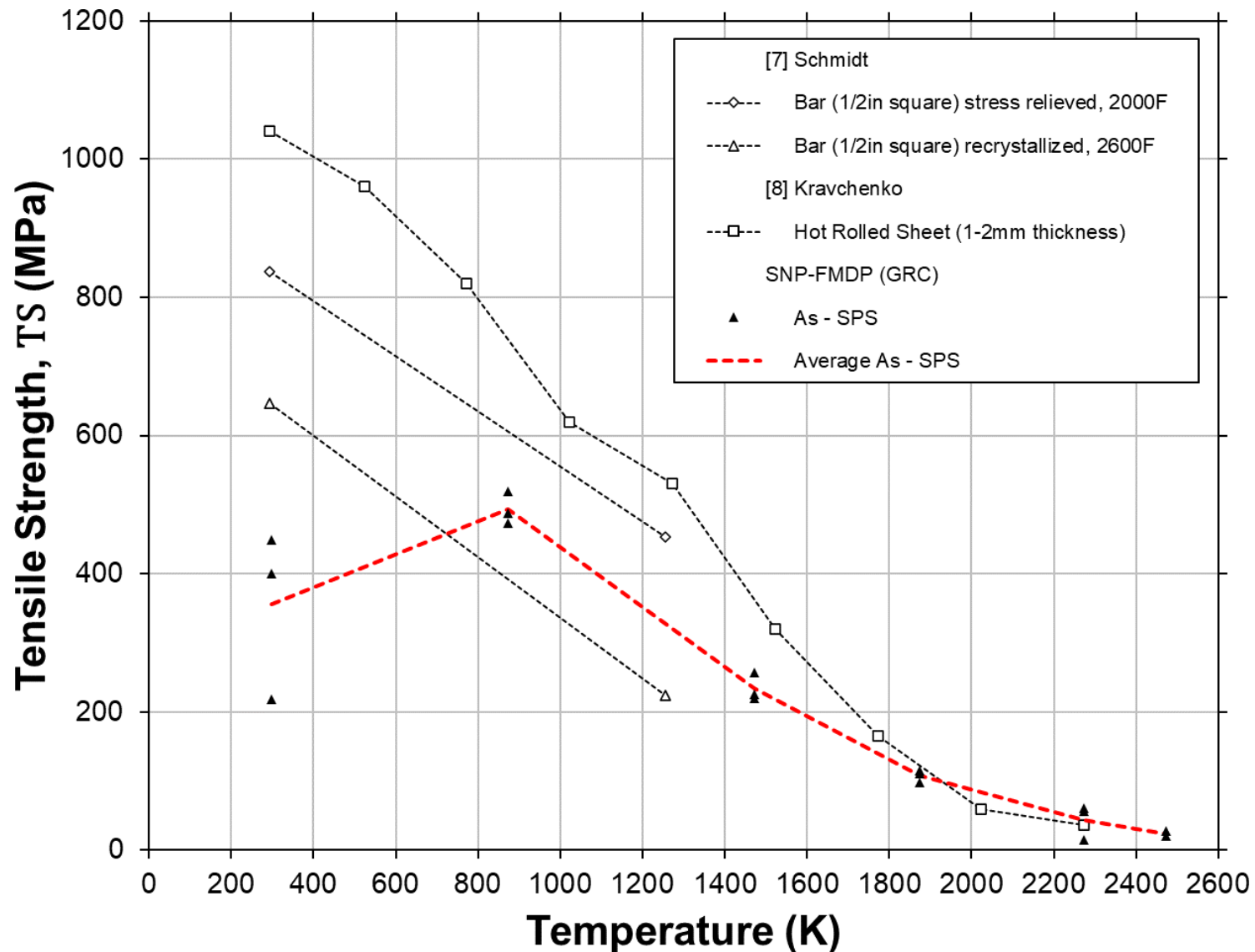


Figure 3.1.2-13: Tensile Strength versus Temperature for Mo-30W. Displaying fitted trend for 1-2 mm thick hot rolled sheet.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Tensile Strength with Temperature

| Temperature ( T )                                |          | Yield Strength ( YS ) |           | Tensile Strength ( TS ) |           | Elongation | Reduction in Area |
|--------------------------------------------------|----------|-----------------------|-----------|-------------------------|-----------|------------|-------------------|
| K                                                | ( F )    | MPa                   | ( ksi )   | MPa                     | ( ksi )   | %          | ( RA )<br>%       |
| Bar (1/2in square) stress relieved, 1366 K       |          |                       |           |                         |           |            |                   |
| 293                                              | ( 68 )   | 737                   | ( 106.9 ) | 838                     | ( 121.5 ) | 26         | 40                |
| 1255                                             | ( 1800 ) | -                     | -         | 453                     | ( 65.7 )  | 25         | 77                |
| Bar (1/2in square) recrystallized, 1700 K        |          |                       |           |                         |           |            |                   |
| 293                                              | ( 68 )   | 476                   | ( 69.1 )  | 647                     | ( 93.8 )  | 12         | 10                |
| 1255                                             | ( 1800 ) | -                     | -         | 223                     | ( 32.4 )  | 83         | 90                |
| Hot Rolled Sheet (1-2mm thickness)               |          |                       |           |                         |           |            |                   |
| 293                                              | ( 68 )   | 905                   | ( 131.2 ) | 1040                    | ( 150.8 ) | 9          | -                 |
| 523                                              | ( 482 )  | 865                   | ( 125.4 ) | 960                     | ( 139.2 ) | 6.5        | -                 |
| 773                                              | ( 932 )  | 780                   | ( 113.1 ) | 820                     | ( 118.9 ) | 6          | -                 |
| 1023                                             | ( 1382 ) | 610                   | ( 88.5 )  | 620                     | ( 89.9 )  | 8.5        | -                 |
| 1273                                             | ( 1832 ) | 430                   | ( 62.4 )  | 530                     | ( 76.9 )  | 8.5        | -                 |
| 1523                                             | ( 2282 ) | 260                   | ( 37.7 )  | 320                     | ( 46.4 )  | 9          | -                 |
| 1773                                             | ( 2732 ) | 155                   | ( 22.5 )  | 165                     | ( 23.9 )  | 17         | -                 |
| 2023                                             | ( 3182 ) | 50                    | ( 7.3 )   | 60                      | ( 8.7 )   | 18         | -                 |
| 2273                                             | ( 3632 ) | 34                    | ( 4.9 )   | 37                      | ( 5.4 )   | 14         | -                 |
| As - Spark Plasma Sintered (As - SPS) (SNP-FMDP) |          |                       |           |                         |           |            |                   |
| 297                                              | ( 75 )   | -                     | -         | 356                     | ( 51.6 )  | 0.6        | -                 |
| 873                                              | ( 1112 ) | -                     | -         | 494                     | ( 71.6 )  | 9.1        | -                 |
| 1473                                             | ( 2192 ) | 212                   | ( 30.8 )  | 234                     | ( 34.0 )  | 0.8        | -                 |
| 1873                                             | ( 2912 ) | 84                    | ( 12.2 )  | 108                     | ( 15.6 )  | 2.6        | -                 |
| 2273                                             | ( 3632 ) | 33                    | ( 4.8 )   | 44                      | ( 6.4 )   | 4.5        | -                 |
| 2473                                             | ( 3992 ) | 18                    | ( 2.5 )   | 24                      | ( 3.5 )   | 6.9        | -                 |

**Application Notes:** Yield and tensile strength data was used from [7, 8] and SNP-FMDP (GRC) to develop the strength table and plots above.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

Revision 2: 04-26-2023

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                          |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( α )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 11909                                 | -                                                       | -                                           | -0.081                                             | 4.19                                     | 0.124                                          | -                                  | -                                | -                           |
| 200                 | 11896                                 | -                                                       | -                                           | -0.045                                             | 4.68                                     | 0.194                                          | -                                  | -                                | -                           |
| 300                 | 11880                                 | 0.072                                                   | 92.53                                       | 0.001                                              | 4.94                                     | 0.215                                          | 340.40                             | 129.88                           | 0.31                        |
| 400                 | 11861                                 | 0.098                                                   | 95.25                                       | 0.054                                              | 5.10                                     | 0.224                                          | 335.73                             | 127.97                           | 0.31                        |
| 500                 | 11841                                 | 0.124                                                   | 96.74                                       | 0.109                                              | 5.21                                     | 0.230                                          | 331.06                             | 126.07                           | 0.31                        |
| 600                 | 11822                                 | 0.151                                                   | 97.66                                       | 0.163                                              | 5.29                                     | 0.236                                          | 326.39                             | 124.16                           | 0.31                        |
| 700                 | 11802                                 | 0.178                                                   | 98.23                                       | 0.219                                              | 5.36                                     | 0.240                                          | 321.72                             | 122.25                           | 0.31                        |
| 800                 | 11783                                 | 0.205                                                   | 98.58                                       | 0.275                                              | 5.43                                     | 0.245                                          | 317.06                             | 120.34                           | 0.32                        |
| 900                 | 11762                                 | 0.232                                                   | 98.77                                       | 0.332                                              | 5.49                                     | 0.250                                          | 312.39                             | 118.44                           | 0.32                        |
| 1000                | 11742                                 | 0.259                                                   | 98.84                                       | 0.391                                              | 5.55                                     | 0.254                                          | 307.72                             | 116.53                           | 0.32                        |
| 1100                | 11720                                 | 0.286                                                   | 98.83                                       | 0.452                                              | 5.62                                     | 0.259                                          | 303.05                             | 114.62                           | 0.32                        |
| 1200                | 11699                                 | 0.314                                                   | 98.77                                       | 0.514                                              | 5.68                                     | 0.264                                          | 298.38                             | 112.72                           | 0.32                        |
| 1300                | 11676                                 | 0.342                                                   | 98.66                                       | 0.578                                              | 5.75                                     | 0.270                                          | 293.72                             | 110.81                           | 0.32                        |
| 1400                | 11653                                 | 0.370                                                   | 98.53                                       | 0.644                                              | 5.83                                     | 0.276                                          | 289.05                             | 108.90                           | 0.32                        |
| 1500                | 11630                                 | 0.398                                                   | 98.39                                       | 0.713                                              | 5.91                                     | 0.282                                          | 284.38                             | 107.00                           | 0.33                        |
| 1600                | 11605                                 | 0.426                                                   | 98.24                                       | 0.784                                              | 6.00                                     | 0.289                                          | 279.71                             | 105.09                           | 0.33                        |
| 1700                | 11580                                 | 0.455                                                   | 98.09                                       | 0.857                                              | 6.09                                     | 0.297                                          | 275.04                             | 103.18                           | 0.33                        |
| 1800                | 11553                                 | 0.484                                                   | 97.94                                       | 0.934                                              | 6.19                                     | 0.305                                          | 270.38                             | 101.27                           | 0.33                        |
| 1900                | 11526                                 | 0.513                                                   | 97.77                                       | 1.014                                              | 6.30                                     | 0.314                                          | -                                  | -                                | -                           |
| 2000                | 11498                                 | -                                                       | 97.59                                       | 1.097                                              | 6.42                                     | 0.324                                          | -                                  | -                                | -                           |
| 2200                | 11438                                 | -                                                       | 97.12                                       | 1.273                                              | 6.66                                     | 0.347                                          | -                                  | -                                | -                           |
| 2400                | 11373                                 | -                                                       | -                                           | 1.465                                              | 6.94                                     | 0.375                                          | -                                  | -                                | -                           |
| 2600                | 11303                                 | -                                                       | -                                           | 1.674                                              | 7.25                                     | 0.407                                          | -                                  | -                                | -                           |
| 2800                | 11227                                 | -                                                       | -                                           | 1.901                                              | 7.59                                     | 0.445                                          | -                                  | -                                | -                           |
| 2900                | 11187                                 | -                                                       | -                                           | 2.023                                              | 7.77                                     | 0.467                                          | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.2 Mo-30W

**Revision 2: 04-26-2023**

**References**

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### **3 Refractory Metals and Alloys**

#### **3.1 Molybdenum and Molybdenum Alloys**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 Titanium-Zirconium-Molybdenum (TZM)

Revision 2: 04-26-2023

General

## Room Temperature Properties

|                                                  |                   |
|--------------------------------------------------|-------------------|
| Density, [kg/m <sup>3</sup> ]                    | 10,220            |
| Melting Point, [K]                               | 2,883 – 2,894 [1] |
| Specific Heat, [J/(g-K)]                         | 0.269             |
| Thermal Conductivity, [W/(m-K)]                  | 123.2             |
| Average linear expansion coefficient, [μm/(m-K)] | 4.97              |
| Electrical resistivity, [μΩ-m]                   | 0.079             |
| Young's Modulus, [GPa]                           | 306.3             |
| Shear Modulus, [GPa]                             | 122.7 [1]         |
| Poisson's Ratio, [-]                             | 0.293 [1]         |

## Composition

Table 3.1.3-1: Typical Composition ranges for molybdenum alloy 363, TZM (percent by weight) [1].

| Grade |      | Mo   | Ti   | Zr   | C     | Si   | Fe   | O     | Ni    | N     |
|-------|------|------|------|------|-------|------|------|-------|-------|-------|
| TZM   | Min. | 99.3 | 0.4  | 0.06 | 0.010 | -    | -    | -     | -     | -     |
|       | Max. | 99.5 | 0.55 | 0.12 | 0.040 | 0.01 | 0.01 | 0.003 | 0.002 | 0.002 |



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Density with Temperature

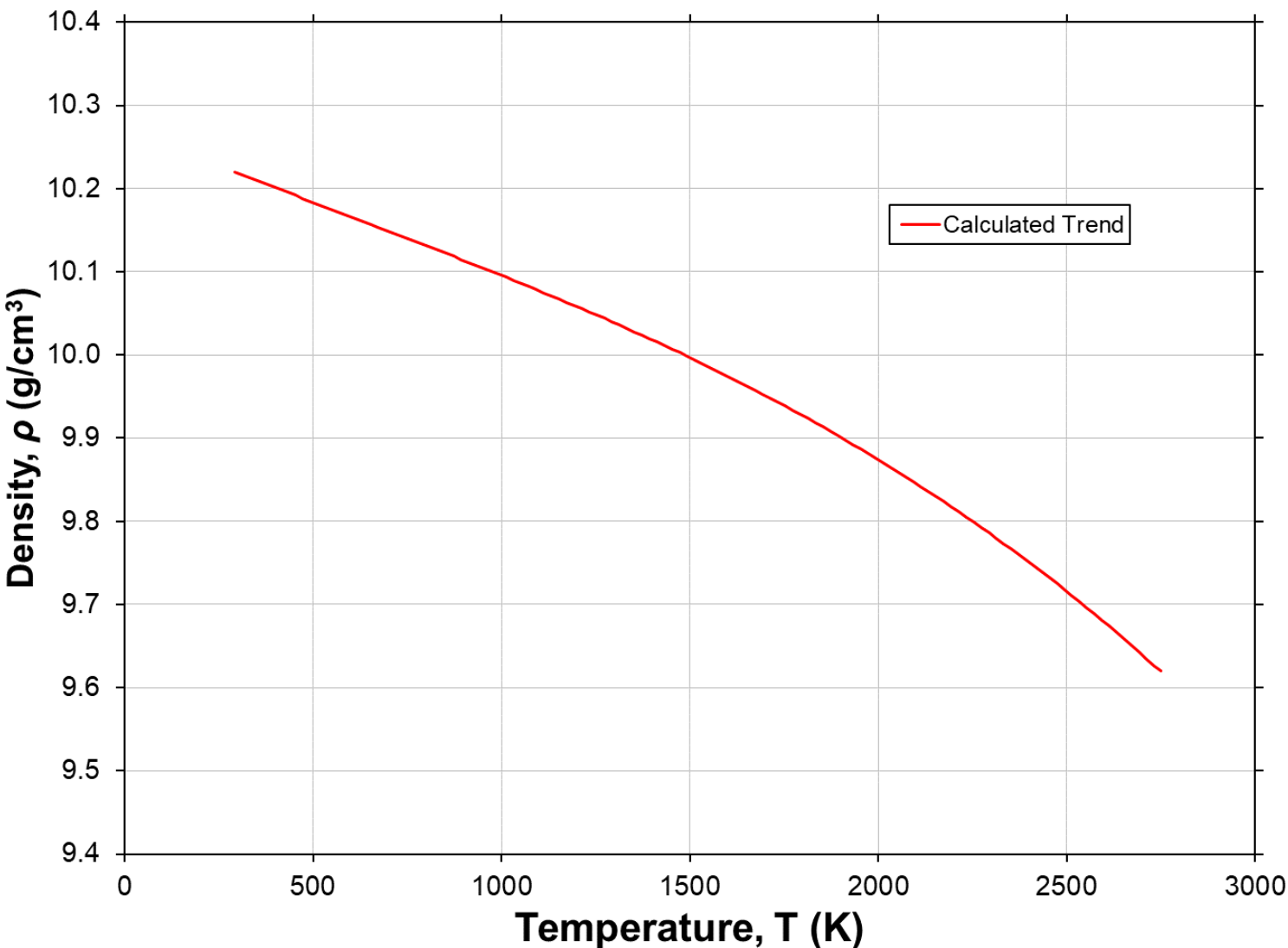


Figure 3.1.3-1: Density versus temperature for TZM. Calculated from fitted trend of the thermal expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 293               | ( 67.7 )   | 10220             | ( 638.0 )              | 1400              | ( 2060.3 ) | 10018             | ( 625.4 )              |
| 300               | ( 80.3 )   | 10219             | ( 638.0 )              | 1500              | ( 2240.3 ) | 9997              | ( 624.1 )              |
| 350               | ( 170.3 )  | 10210             | ( 637.4 )              | 1600              | ( 2420.3 ) | 9975              | ( 622.7 )              |
| 400               | ( 260.3 )  | 10201             | ( 636.8 )              | 1700              | ( 2600.3 ) | 9951              | ( 621.3 )              |
| 450               | ( 350.3 )  | 10192             | ( 636.3 )              | 1800              | ( 2780.3 ) | 9927              | ( 619.7 )              |
| 500               | ( 440.3 )  | 10184             | ( 635.8 )              | 1900              | ( 2960.3 ) | 9901              | ( 618.1 )              |
| 550               | ( 530.3 )  | 10175             | ( 635.2 )              | 2000              | ( 3140.3 ) | 9874              | ( 616.4 )              |
| 600               | ( 620.3 )  | 10166             | ( 634.7 )              | 2100              | ( 3320.3 ) | 9846              | ( 614.7 )              |
| 700               | ( 800.3 )  | 10149             | ( 633.6 )              | 2200              | ( 3500.3 ) | 9816              | ( 612.8 )              |
| 800               | ( 980.3 )  | 10131             | ( 632.5 )              | 2300              | ( 3680.3 ) | 9784              | ( 610.8 )              |
| 900               | ( 1160.3 ) | 10114             | ( 631.4 )              | 2400              | ( 3860.3 ) | 9751              | ( 608.7 )              |
| 1000              | ( 1340.3 ) | 10096             | ( 630.3 )              | 2500              | ( 4040.3 ) | 9716              | ( 606.5 )              |
| 1100              | ( 1520.3 ) | 10077             | ( 629.1 )              | 2600              | ( 4220.3 ) | 9679              | ( 604.2 )              |
| 1200              | ( 1700.3 ) | 10058             | ( 627.9 )              | 2700              | ( 4400.3 ) | 9640              | ( 601.8 )              |
| 1300              | ( 1880.3 ) | 10038             | ( 626.7 )              | 2749              | ( 4488.5 ) | 9620              | ( 600.6 )              |

**Application Notes:** Density is calculated as a function of thermal expansion, as seen in the equation below, to approximate property trend with respect to temperature.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 10,220 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $293 \leq T \leq 2749$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Thermal Conductivity with Temperature

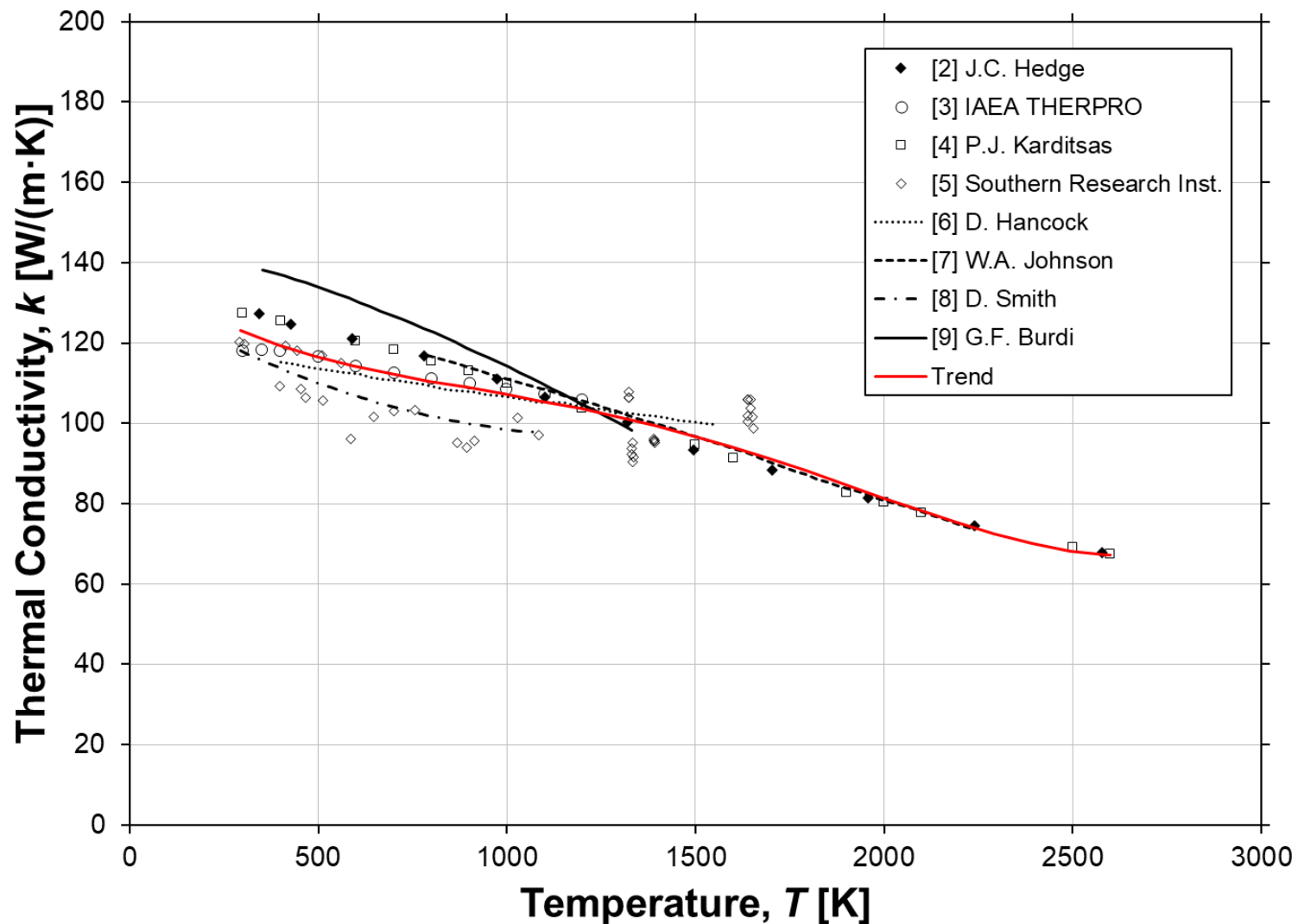


Figure 3.1.3-2: Thermal conductivity versus temperature for TZM.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                 | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                 |
|-------------------|------------|----------------------------|---------------------------------|-------------------|------------|----------------------------|---------------------------------|
| K                 | ( °F )     | W/(m-K)                    | (Btu-in/ft <sup>2</sup> -hr-°F) | K                 | ( °F )     | W/(m-K)                    | (Btu-in/ft <sup>2</sup> -hr-°F) |
| 293               | ( 67.7 )   | 123.22                     | ( 854.91 )                      | 1200              | ( 1700.3 ) | 103.57                     | ( 718.60 )                      |
| 300               | ( 80.3 )   | 122.94                     | ( 852.98 )                      | 1300              | ( 1880.3 ) | 101.52                     | ( 704.36 )                      |
| 350               | ( 170.3 )  | 121.08                     | ( 840.04 )                      | 1400              | ( 2060.3 ) | 99.25                      | ( 688.59 )                      |
| 400               | ( 260.3 )  | 119.40                     | ( 828.43 )                      | 1500              | ( 2240.3 ) | 96.74                      | ( 671.20 )                      |
| 450               | ( 350.3 )  | 117.90                     | ( 818.01 )                      | 1600              | ( 2420.3 ) | 94.00                      | ( 652.21 )                      |
| 500               | ( 440.3 )  | 116.55                     | ( 808.61 )                      | 1700              | ( 2600.3 ) | 91.06                      | ( 631.77 )                      |
| 550               | ( 530.3 )  | 115.32                     | ( 800.10 )                      | 1800              | ( 2780.3 ) | 87.94                      | ( 610.14 )                      |
| 600               | ( 620.3 )  | 114.20                     | ( 792.34 )                      | 1900              | ( 2960.3 ) | 84.71                      | ( 587.71 )                      |
| 650               | ( 710.3 )  | 113.17                     | ( 785.20 )                      | 2000              | ( 3140.3 ) | 81.43                      | ( 564.98 )                      |
| 700               | ( 800.3 )  | 112.22                     | ( 778.57 )                      | 2100              | ( 3320.3 ) | 78.20                      | ( 542.59 )                      |
| 750               | ( 890.3 )  | 111.32                     | ( 772.32 )                      | 2200              | ( 3500.3 ) | 75.13                      | ( 521.28 )                      |
| 800               | ( 980.3 )  | 110.46                     | ( 766.35 )                      | 2300              | ( 3680.3 ) | 72.34                      | ( 501.92 )                      |
| 900               | ( 1160.3 ) | 108.80                     | ( 754.89 )                      | 2400              | ( 3860.3 ) | 69.97                      | ( 485.49 )                      |
| 1000              | ( 1340.3 ) | 107.16                     | ( 743.47 )                      | 2500              | ( 4040.3 ) | 68.19                      | ( 473.10 )                      |
| 1100              | ( 1520.3 ) | 105.44                     | ( 731.52 )                      | 2600              | ( 4220.3 ) | 67.16                      | ( 465.98 )                      |

**Application Notes:** Data for thermal conductivity is collected from references [2-9] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3 + A4 \cdot \left(\frac{T}{1000}\right)^4$$

$k(T)$  = Thermal Conductivity [W/(m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]: 293 ≤ T < 2600

A0 = 139.6

A1 = -74.98

A2 = 76.56

A3 = -41.24

A4 = 7.217



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Thermal Expansion with Temperature

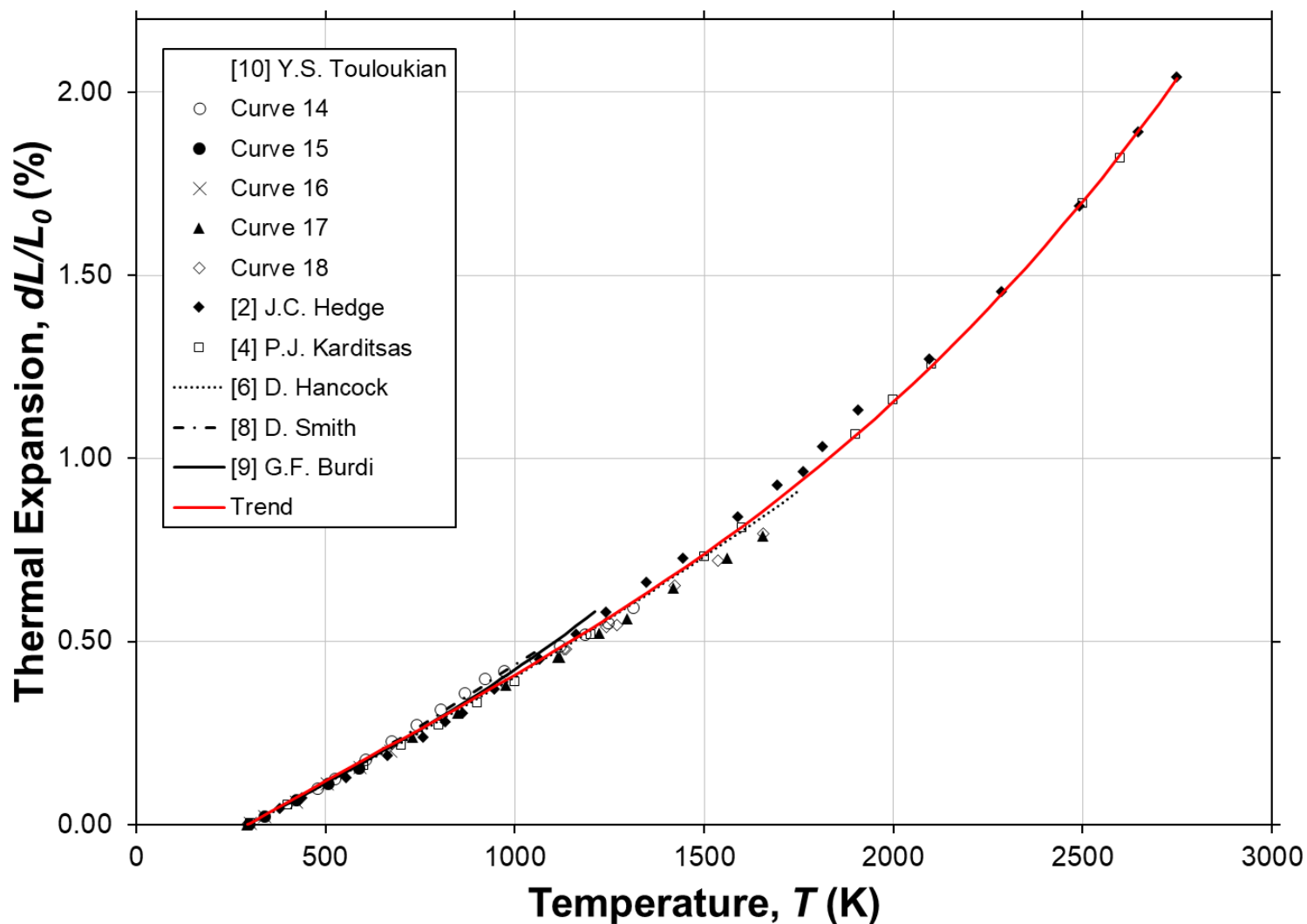


Figure 3.1.3-3: Thermal expansion versus temperature for TZM.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 293               | ( 67.7 )   | 0.000                                   | 1400              | ( 2060.3 ) | 0.668                                   |
| 300               | ( 80.3 )   | 0.004                                   | 1500              | ( 2240.3 ) | 0.739                                   |
| 350               | ( 170.3 )  | 0.033                                   | 1600              | ( 2420.3 ) | 0.814                                   |
| 400               | ( 260.3 )  | 0.062                                   | 1700              | ( 2600.3 ) | 0.892                                   |
| 450               | ( 350.3 )  | 0.091                                   | 1800              | ( 2780.3 ) | 0.975                                   |
| 500               | ( 440.3 )  | 0.119                                   | 1900              | ( 2960.3 ) | 1.062                                   |
| 550               | ( 530.3 )  | 0.148                                   | 2000              | ( 3140.3 ) | 1.155                                   |
| 600               | ( 620.3 )  | 0.176                                   | 2100              | ( 3320.3 ) | 1.252                                   |
| 700               | ( 800.3 )  | 0.233                                   | 2200              | ( 3500.3 ) | 1.355                                   |
| 800               | ( 980.3 )  | 0.291                                   | 2300              | ( 3680.3 ) | 1.464                                   |
| 900               | ( 1160.3 ) | 0.349                                   | 2400              | ( 3860.3 ) | 1.579                                   |
| 1000              | ( 1340.3 ) | 0.409                                   | 2500              | ( 4040.3 ) | 1.702                                   |
| 1100              | ( 1520.3 ) | 0.470                                   | 2600              | ( 4220.3 ) | 1.831                                   |
| 1200              | ( 1700.3 ) | 0.534                                   | 2700              | ( 4400.3 ) | 1.968                                   |
| 1300              | ( 1880.3 ) | 0.599                                   | 2749              | ( 4488.5 ) | 2.037                                   |

**Application Notes:** Data for thermal expansion is collected from references [2, 4, 6, 8-10] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 293 \leq T \leq 2749$$

$$A0 = -0.1767$$

$$A1 = 0.6282$$

$$A2 = -0.1035$$

$$A3 = 0.0611$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Coefficient of Thermal Expansion with Temperature

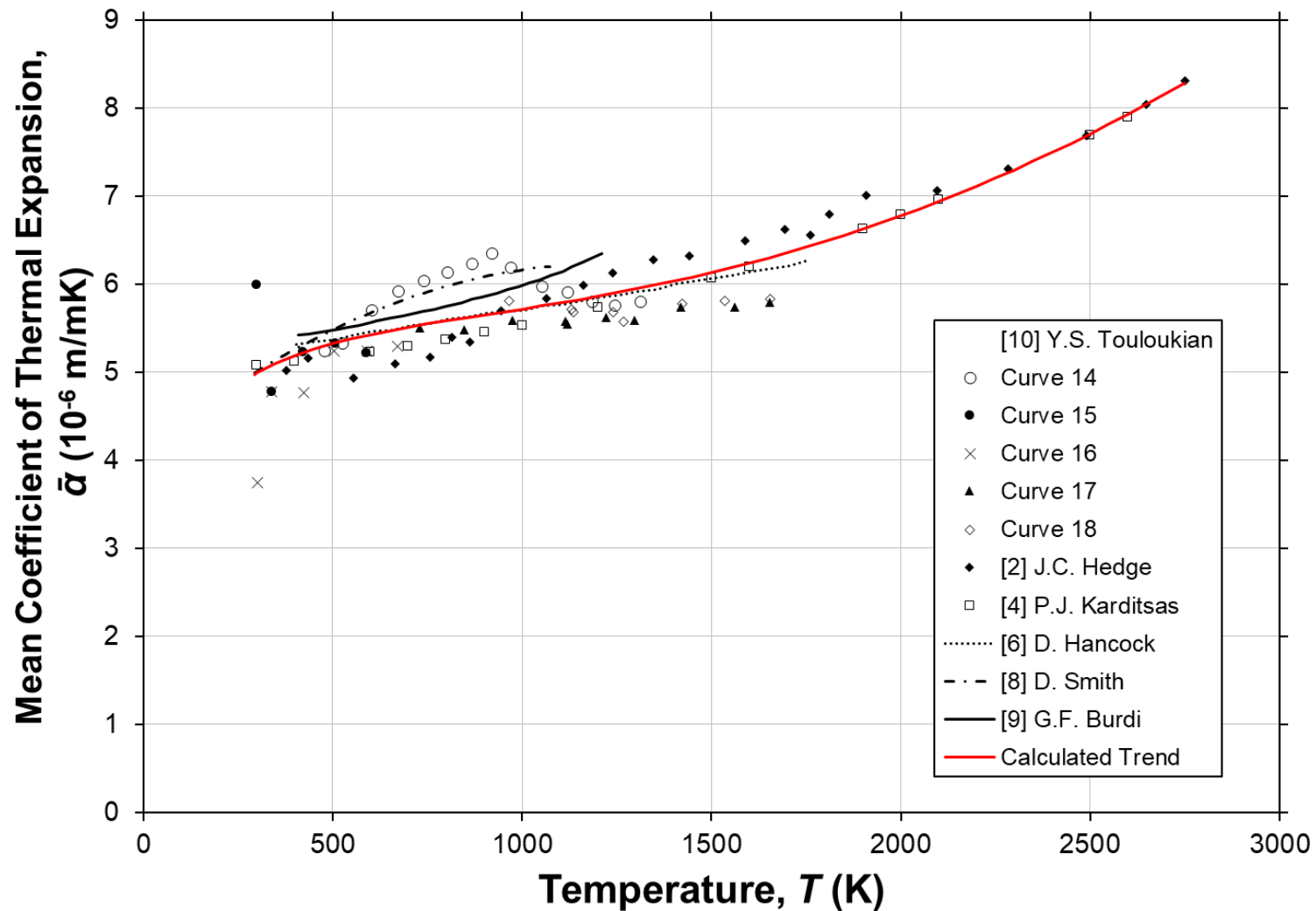


Figure 3.1.3-4: Mean coefficient of thermal expansion versus temperature for TZM, as calculated from fitted thermal expansion trend.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 293               | ( 67.7 )   | 4.971                                 | ( 2.762 )                                           | 1400              | ( 2060.3 ) | 6.035                                 | ( 3.353 )                                           |
| 300               | ( 80.3 )   | 4.988                                 | ( 2.771 )                                           | 1500              | ( 2240.3 ) | 6.133                                 | ( 3.407 )                                           |
| 350               | ( 170.3 )  | 5.096                                 | ( 2.831 )                                           | 1600              | ( 2420.3 ) | 6.241                                 | ( 3.467 )                                           |
| 400               | ( 260.3 )  | 5.185                                 | ( 2.880 )                                           | 1700              | ( 2600.3 ) | 6.358                                 | ( 3.532 )                                           |
| 450               | ( 350.3 )  | 5.259                                 | ( 2.922 )                                           | 1800              | ( 2780.3 ) | 6.487                                 | ( 3.604 )                                           |
| 500               | ( 440.3 )  | 5.322                                 | ( 2.957 )                                           | 1900              | ( 2960.3 ) | 6.626                                 | ( 3.681 )                                           |
| 550               | ( 530.3 )  | 5.377                                 | ( 2.987 )                                           | 2000              | ( 3140.3 ) | 6.776                                 | ( 3.765 )                                           |
| 600               | ( 620.3 )  | 5.426                                 | ( 3.014 )                                           | 2100              | ( 3320.3 ) | 6.938                                 | ( 3.855 )                                           |
| 700               | ( 800.3 )  | 5.509                                 | ( 3.060 )                                           | 2200              | ( 3500.3 ) | 7.112                                 | ( 3.951 )                                           |
| 800               | ( 980.3 )  | 5.582                                 | ( 3.101 )                                           | 2300              | ( 3680.3 ) | 7.298                                 | ( 4.054 )                                           |
| 900               | ( 1160.3 ) | 5.650                                 | ( 3.139 )                                           | 2400              | ( 3860.3 ) | 7.496                                 | ( 4.164 )                                           |
| 1000              | ( 1340.3 ) | 5.718                                 | ( 3.176 )                                           | 2500              | ( 4040.3 ) | 7.706                                 | ( 4.281 )                                           |
| 1100              | ( 1520.3 ) | 5.788                                 | ( 3.216 )                                           | 2600              | ( 4220.3 ) | 7.929                                 | ( 4.405 )                                           |
| 1200              | ( 1700.3 ) | 5.863                                 | ( 3.257 )                                           | 2700              | ( 4400.3 ) | 8.164                                 | ( 4.535 )                                           |
| 1300              | ( 1880.3 ) | 5.945                                 | ( 3.303 )                                           | 2749              | ( 4488.5 ) | 8.284                                 | ( 4.602 )                                           |

**Application Notes:** Trend for mean coefficient of thermal expansion is calculated from thermal expansion and shown in the equation below to approximate property trend with respect to temperature. The slope near room temperature was estimated using the trend for pure molybdenum.

## Fit Equation:

$$\bar{\alpha}(T) = \left[ A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right] / \left[ \left( \frac{T}{1000} \right) + A_{-0} \right]$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $293 \leq T < 2749$

A0 = 0.9359

A1 = 6.897

A2 = -1.158

A3 = 0.6734

A<sub>-0</sub> = 0.2852

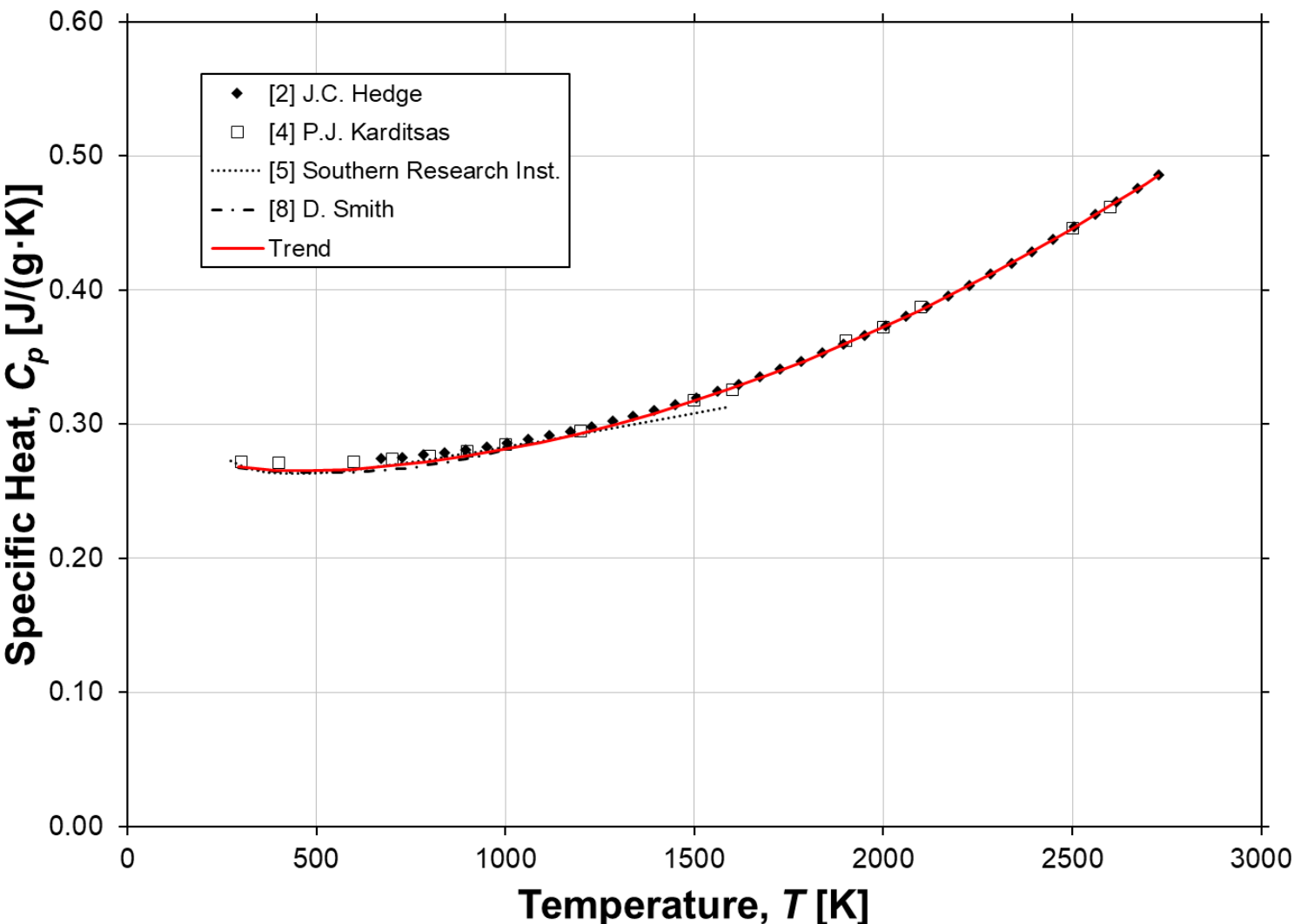


Figure 3.1.3-5: Specific heat versus temperature for TZM.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 293               | ( 67.7 )   | 0.269                            | ( 0.064 )       | 1400              | ( 2060.3 ) | 0.309                            | ( 0.074 )       |
| 300               | ( 80.3 )   | 0.268                            | ( 0.064 )       | 1500              | ( 2240.3 ) | 0.317                            | ( 0.076 )       |
| 350               | ( 170.3 )  | 0.266                            | ( 0.064 )       | 1600              | ( 2420.3 ) | 0.327                            | ( 0.078 )       |
| 400               | ( 260.3 )  | 0.265                            | ( 0.063 )       | 1700              | ( 2600.3 ) | 0.337                            | ( 0.081 )       |
| 450               | ( 350.3 )  | 0.265                            | ( 0.063 )       | 1800              | ( 2780.3 ) | 0.348                            | ( 0.083 )       |
| 500               | ( 440.3 )  | 0.265                            | ( 0.063 )       | 1900              | ( 2960.3 ) | 0.360                            | ( 0.086 )       |
| 550               | ( 530.3 )  | 0.266                            | ( 0.064 )       | 2000              | ( 3140.3 ) | 0.372                            | ( 0.089 )       |
| 600               | ( 620.3 )  | 0.267                            | ( 0.064 )       | 2100              | ( 3320.3 ) | 0.386                            | ( 0.092 )       |
| 700               | ( 800.3 )  | 0.269                            | ( 0.064 )       | 2200              | ( 3500.3 ) | 0.399                            | ( 0.095 )       |
| 800               | ( 980.3 )  | 0.272                            | ( 0.065 )       | 2300              | ( 3680.3 ) | 0.414                            | ( 0.099 )       |
| 900               | ( 1160.3 ) | 0.276                            | ( 0.066 )       | 2400              | ( 3860.3 ) | 0.430                            | ( 0.103 )       |
| 1000              | ( 1340.3 ) | 0.281                            | ( 0.067 )       | 2500              | ( 4040.3 ) | 0.446                            | ( 0.107 )       |
| 1100              | ( 1520.3 ) | 0.287                            | ( 0.069 )       | 2600              | ( 4220.3 ) | 0.463                            | ( 0.111 )       |
| 1200              | ( 1700.3 ) | 0.293                            | ( 0.070 )       | 2700              | ( 4400.3 ) | 0.480                            | ( 0.115 )       |
| 1300              | ( 1880.3 ) | 0.301                            | ( 0.072 )       | 2728              | ( 4450.7 ) | 0.485                            | ( 0.116 )       |

**Application Notes:** Data for specific heat is collected from references [2, 4, 5, 8] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$C_p(T) = A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_2 / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

$$T \text{ Range [K]: } 293 \leq T < 2728$$

$$A_0 = 0.2631$$

$$A_1 = -0.01961$$

$$A_2 = 0.03706$$

$$A_2 = 6.891\text{E-4}$$

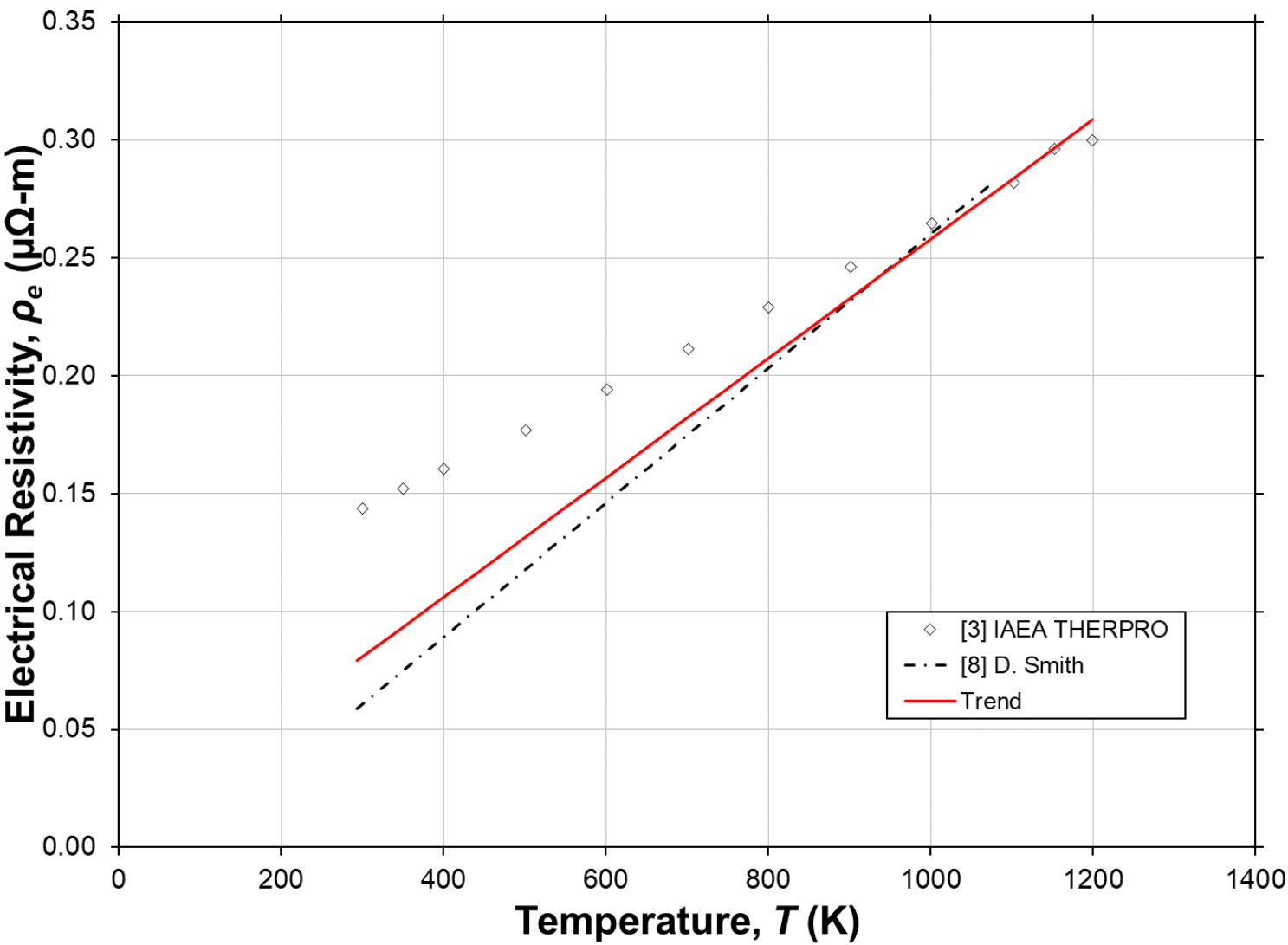


Figure 3.1.3-6: Electrical resistivity versus temperature for TZM.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|-----------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )    | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 293               | ( 67.7 )  | 0.079                               | ( 0.0049 )              | 580               | ( 584.3 )  | 0.152                               | ( 0.0095 )              |
| 300               | ( 80.3 )  | 0.081                               | ( 0.0050 )              | 600               | ( 620.3 )  | 0.157                               | ( 0.0098 )              |
| 320               | ( 116.3 ) | 0.086                               | ( 0.0054 )              | 620               | ( 656.3 )  | 0.162                               | ( 0.0101 )              |
| 340               | ( 152.3 ) | 0.091                               | ( 0.0057 )              | 650               | ( 710.3 )  | 0.169                               | ( 0.0106 )              |
| 360               | ( 188.3 ) | 0.096                               | ( 0.0060 )              | 700               | ( 800.3 )  | 0.182                               | ( 0.0114 )              |
| 380               | ( 224.3 ) | 0.101                               | ( 0.0063 )              | 750               | ( 890.3 )  | 0.195                               | ( 0.0122 )              |
| 400               | ( 260.3 ) | 0.106                               | ( 0.0066 )              | 800               | ( 980.3 )  | 0.207                               | ( 0.0130 )              |
| 420               | ( 296.3 ) | 0.111                               | ( 0.0069 )              | 850               | ( 1070.3 ) | 0.220                               | ( 0.0137 )              |
| 440               | ( 332.3 ) | 0.116                               | ( 0.0073 )              | 900               | ( 1160.3 ) | 0.233                               | ( 0.0145 )              |
| 460               | ( 368.3 ) | 0.121                               | ( 0.0076 )              | 950               | ( 1250.3 ) | 0.245                               | ( 0.0153 )              |
| 480               | ( 404.3 ) | 0.126                               | ( 0.0079 )              | 1000              | ( 1340.3 ) | 0.258                               | ( 0.0161 )              |
| 500               | ( 440.3 ) | 0.132                               | ( 0.0082 )              | 1050              | ( 1430.3 ) | 0.271                               | ( 0.0169 )              |
| 520               | ( 476.3 ) | 0.137                               | ( 0.0085 )              | 1100              | ( 1520.3 ) | 0.283                               | ( 0.0177 )              |
| 540               | ( 512.3 ) | 0.142                               | ( 0.0088 )              | 1150              | ( 1610.3 ) | 0.296                               | ( 0.0185 )              |
| 560               | ( 548.3 ) | 0.147                               | ( 0.0092 )              | 1199              | ( 1698.5 ) | 0.308                               | ( 0.0193 )              |

**Application Notes:** Data for electrical resistivity is collected from references [3, 8] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right)$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $293 \leq T < 1199$

A0 = 4.904E-3

A1 = 0.2532

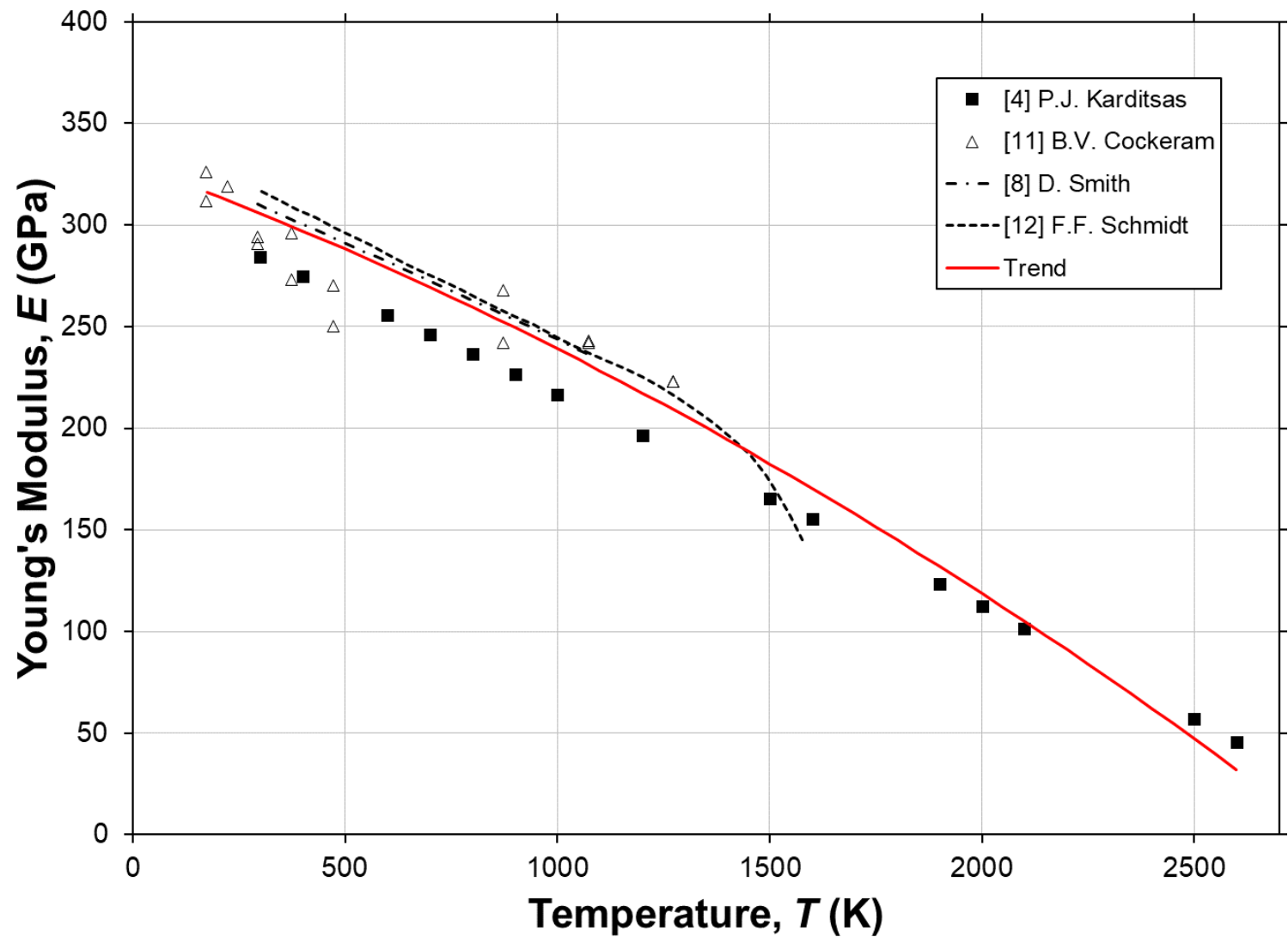


Figure 3.1.3-7: Young's modulus versus temperature for TZM.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |          | Temperature ( T ) |            | Young's Modulus ( E ) |          |
|-------------------|------------|-----------------------|----------|-------------------|------------|-----------------------|----------|
| K                 | ( °F )     | GPa                   | ( Msi )  | K                 | ( °F )     | GPa                   | ( Msi )  |
| 174               | ( -146.5 ) | 316                   | ( 45.9 ) | 1200              | ( 1700.3 ) | 217                   | ( 31.5 ) |
| 200               | ( -99.7 )  | 314                   | ( 45.6 ) | 1300              | ( 1880.3 ) | 206                   | ( 29.9 ) |
| 293               | ( 67.7 )   | 306                   | ( 44.4 ) | 1400              | ( 2060.3 ) | 194                   | ( 28.2 ) |
| 300               | ( 80.3 )   | 306                   | ( 44.4 ) | 1500              | ( 2240.3 ) | 183                   | ( 26.5 ) |
| 350               | ( 170.3 )  | 301                   | ( 43.7 ) | 1600              | ( 2420.3 ) | 170                   | ( 24.7 ) |
| 400               | ( 260.3 )  | 297                   | ( 43.1 ) | 1700              | ( 2600.3 ) | 158                   | ( 22.9 ) |
| 450               | ( 350.3 )  | 293                   | ( 42.5 ) | 1800              | ( 2780.3 ) | 145                   | ( 21.1 ) |
| 500               | ( 440.3 )  | 288                   | ( 41.8 ) | 1900              | ( 2960.3 ) | 132                   | ( 19.2 ) |
| 550               | ( 530.3 )  | 284                   | ( 41.1 ) | 2000              | ( 3140.3 ) | 119                   | ( 17.2 ) |
| 600               | ( 620.3 )  | 279                   | ( 40.5 ) | 2100              | ( 3320.3 ) | 105                   | ( 15.2 ) |
| 700               | ( 800.3 )  | 269                   | ( 39.1 ) | 2200              | ( 3500.3 ) | 91                    | ( 13.2 ) |
| 800               | ( 980.3 )  | 260                   | ( 37.7 ) | 2300              | ( 3680.3 ) | 77                    | ( 11.1 ) |
| 900               | ( 1160.3 ) | 250                   | ( 36.2 ) | 2400              | ( 3860.3 ) | 62                    | ( 9.0 )  |
| 1000              | ( 1340.3 ) | 239                   | ( 34.7 ) | 2500              | ( 4040.3 ) | 47                    | ( 6.9 )  |
| 1100              | ( 1520.3 ) | 228                   | ( 33.1 ) | 2600              | ( 4220.3 ) | 32                    | ( 4.7 )  |

**Application Notes:** Data for Young's modulus is collected from references [4, 8, 11, 12] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equations:

$$E(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$$E(T) = \text{Young's Modulus [GPa]}$$

$$T = \text{Temperature [K]}$$

## Constants:

$$T \text{ Range [K]: } 174 \leq T \leq 2600$$

$$A0 = 329.8$$

$$A1 = -75.83$$

$$A2 = -14.87$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Yield Strength with Temperature

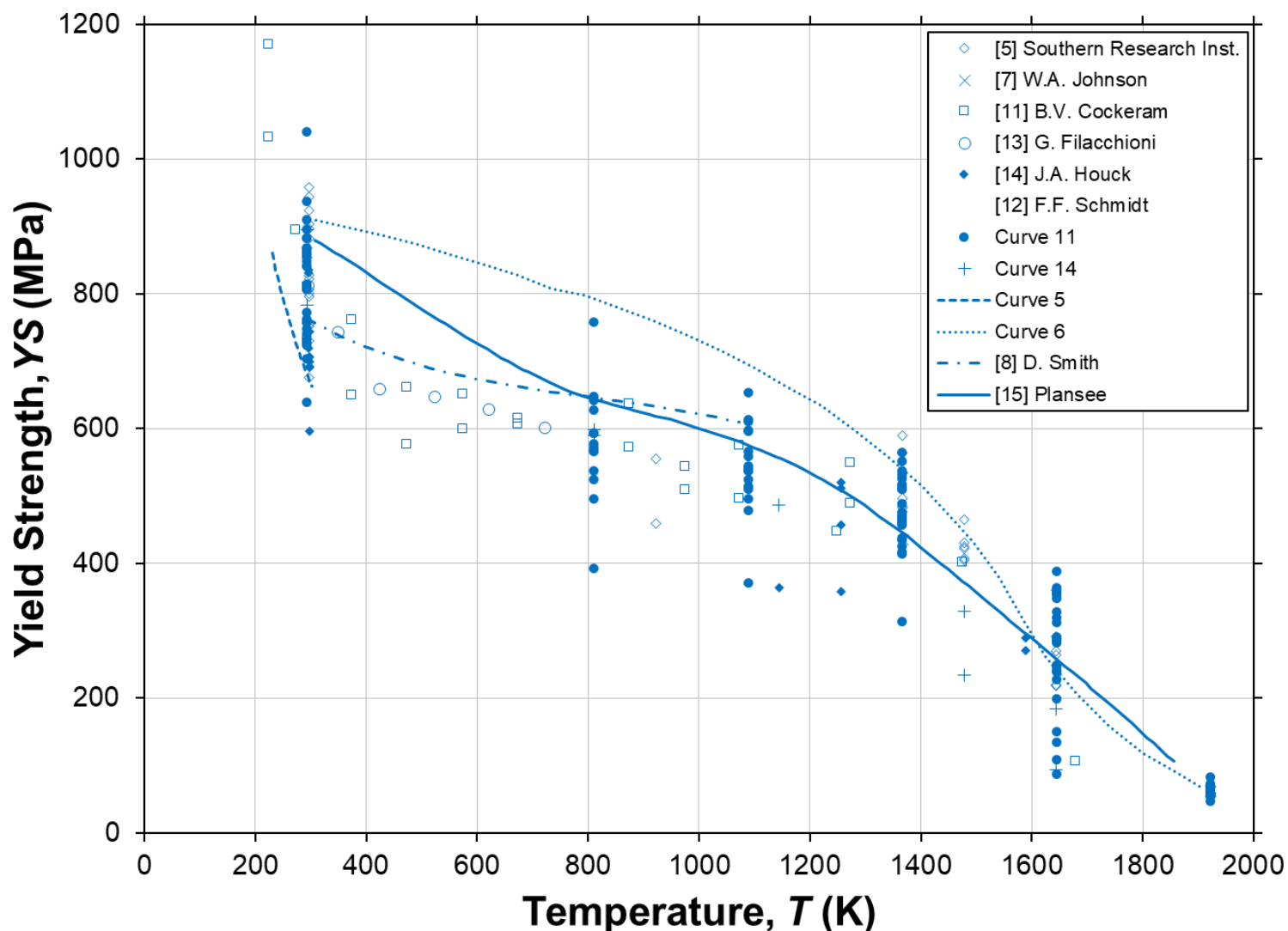


Figure 3.1.3-8: Yield strength versus temperature for stress relieved TZM. Due to the high variance in data, no trend is provided.



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Yield Strength with Temperature

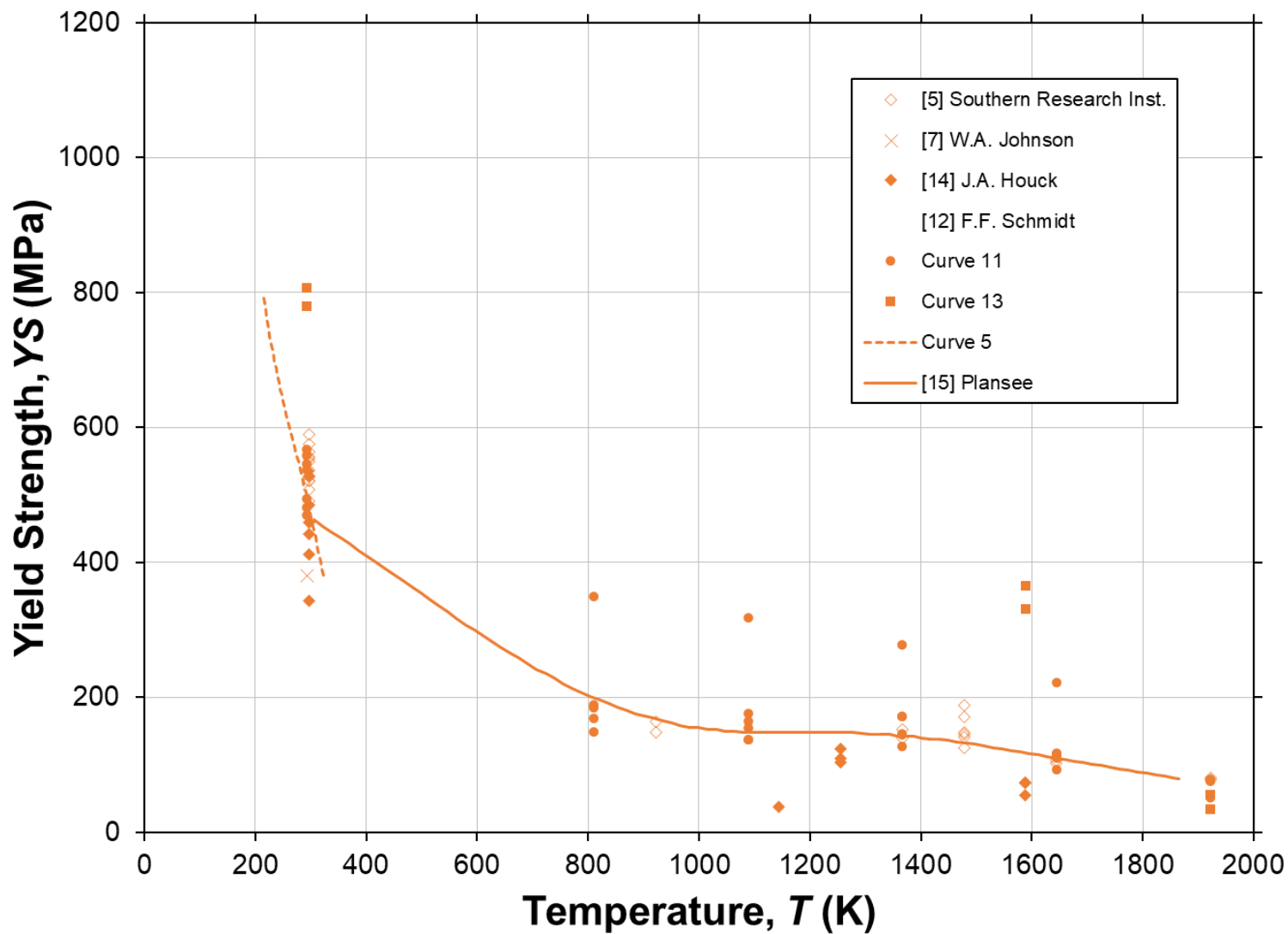


Figure 3.1.3-9: Yield strength versus temperature for recrystallized TZM. Due to the high variance in data, no trend is provided.



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Tensile Strength with Temperature

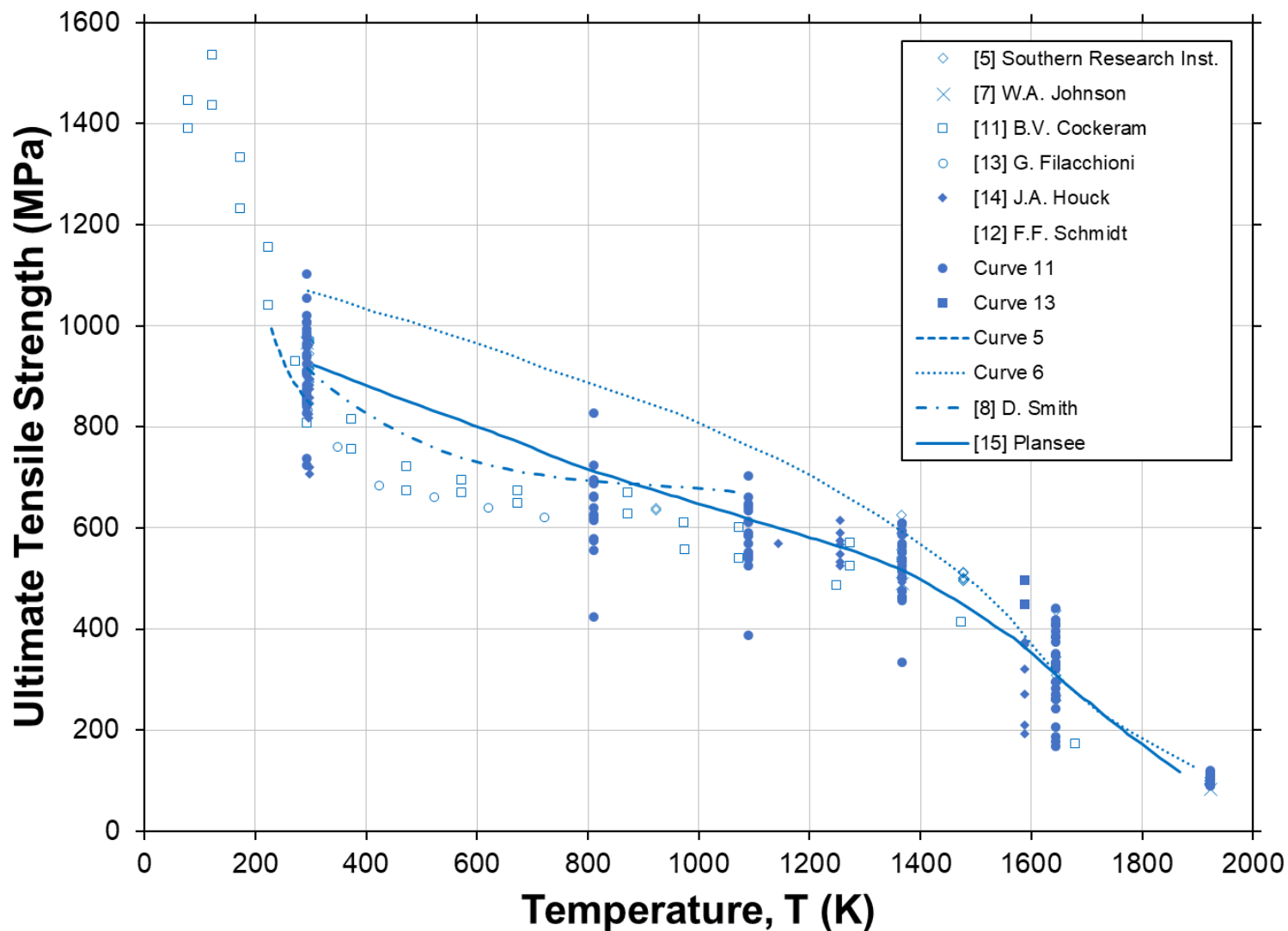


Figure 3.1.3-10: Tensile strength versus temperature for stress relieved TZM. Due to the high variance in data, no trend is provided.



### 3.1.3 TZM

### Tensile Strength with Temperature





## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Additional Application Notes

### Yield Strength Application Notes:

A material's yield strength is highly dependent on manufacturing procedures. A notable change in yield strength values occurs as a result of heat treatment. Stress relieved materials typically show higher yield strength at a given temperature than their otherwise-equivalent recrystallized forms. As a result, both stress relieved and recrystallized yield strength data have been included in this chapter. Stress relieved data is shown in blue, while recrystallized data is shown in orange. Data for stress relieved TZM is collected from references [5, 7, 8, 11-15]. Data for recrystallized TZM is collected from references [5, 7, 12, 14, 15]. No trends were generated for yield strength, due to the large fluctuations in yield strength at a given temperature.

### Tensile Strength Application Notes:

As with yield strength, a material's ultimate tensile strength (or tensile strength) is highly dependent on manufacturing procedures. A notable change in tensile strength values occurs as a result of heat treatment. Stress relieved materials typically show higher tensile strength at a given temperature than their otherwise-equivalent recrystallized forms. As a result, both stress relieved and recrystallized tensile strength data have been included in this chapter. Stress relieved data is shown in blue, while recrystallized data is shown in orange. Data for stress relieved TZM is collected from references [5, 7, 8, 11-15]. Data for recrystallized TZM is collected from references [5, 7, 12, 14, 15]. No trends were generated for tensile strength, due to the large fluctuations in tensile strength at a given temperature.

### A Note on [12] F.F. Schmidt:

Schmidt's report is a compilation of data from several other reports, none of which could be found before the creation of this TZM chapter. Schmidt typically separated data by references, so each reference was shown in its own separate table or plot. For simplicity, when translating Schmidt's data into this TZM chapter, all data is referred to as a "Curve," followed by the reference number it originated from in Schmidt's report. For those interested in verifying data, do not, for example, search for "Curve 11" in Schmidt's report. Instead look for a table or plot that cites "Reference 11." Also note that Schmidt repeats some data in their report.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

Revision 2: 04-26-2023

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                          |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( α )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 200                 | -                                     | -                                                       | -                                           | -                                                  | -                                        | -                                              | 314.04                             | -                                | -                           |
| 293                 | 10220                                 | 0.079                                                   | 123.22                                      | 0.000                                              | 4.97                                     | 0.269                                          | 306.31                             | 122.7                            | 0.293                       |
| 300                 | 10219                                 | 0.081                                                   | 122.94                                      | 0.004                                              | 4.99                                     | 0.268                                          | 305.71                             | -                                | -                           |
| 400                 | 10201                                 | 0.106                                                   | 119.40                                      | 0.062                                              | 5.18                                     | 0.265                                          | 297.09                             | -                                | -                           |
| 500                 | 10184                                 | 0.132                                                   | 116.55                                      | 0.119                                              | 5.32                                     | 0.265                                          | 288.17                             | -                                | -                           |
| 600                 | 10166                                 | 0.157                                                   | 114.20                                      | 0.176                                              | 5.43                                     | 0.267                                          | 278.95                             | -                                | -                           |
| 700                 | 10149                                 | 0.182                                                   | 112.22                                      | 0.233                                              | 5.51                                     | 0.269                                          | 269.43                             | -                                | -                           |
| 800                 | 10131                                 | 0.207                                                   | 110.46                                      | 0.291                                              | 5.58                                     | 0.272                                          | 259.62                             | -                                | -                           |
| 900                 | 10114                                 | 0.233                                                   | 108.80                                      | 0.349                                              | 5.65                                     | 0.276                                          | 249.51                             | -                                | -                           |
| 1000                | 10096                                 | 0.258                                                   | 107.16                                      | 0.409                                              | 5.72                                     | 0.281                                          | 239.10                             | -                                | -                           |
| 1100                | 10077                                 | 0.283                                                   | 105.44                                      | 0.470                                              | 5.79                                     | 0.287                                          | 228.39                             | -                                | -                           |
| 1200                | 10058                                 | -                                                       | 103.57                                      | 0.534                                              | 5.86                                     | 0.293                                          | 217.39                             | -                                | -                           |
| 1300                | 10038                                 | -                                                       | 101.52                                      | 0.599                                              | 5.95                                     | 0.301                                          | 206.09                             | -                                | -                           |
| 1400                | 10018                                 | -                                                       | 99.25                                       | 0.668                                              | 6.03                                     | 0.309                                          | 194.49                             | -                                | -                           |
| 1500                | 9997                                  | -                                                       | 96.74                                       | 0.739                                              | 6.13                                     | 0.317                                          | 182.60                             | -                                | -                           |
| 1600                | 9975                                  | -                                                       | 94.00                                       | 0.814                                              | 6.24                                     | 0.327                                          | 170.40                             | -                                | -                           |
| 1700                | 9951                                  | -                                                       | 91.06                                       | 0.892                                              | 6.36                                     | 0.337                                          | 157.91                             | -                                | -                           |
| 1800                | 9927                                  | -                                                       | 87.94                                       | 0.975                                              | 6.49                                     | 0.348                                          | 145.13                             | -                                | -                           |
| 1900                | 9901                                  | -                                                       | 84.71                                       | 1.062                                              | 6.63                                     | 0.360                                          | 132.04                             | -                                | -                           |
| 2000                | 9874                                  | -                                                       | 81.43                                       | 1.155                                              | 6.78                                     | 0.372                                          | 118.66                             | -                                | -                           |
| 2100                | 9846                                  | -                                                       | 78.20                                       | 1.252                                              | 6.94                                     | 0.386                                          | 104.98                             | -                                | -                           |
| 2200                | 9816                                  | -                                                       | 75.13                                       | 1.355                                              | 7.11                                     | 0.399                                          | 91.00                              | -                                | -                           |
| 2300                | 9784                                  | -                                                       | 72.34                                       | 1.464                                              | 7.30                                     | 0.414                                          | 76.73                              | -                                | -                           |
| 2500                | 9716                                  | -                                                       | 68.19                                       | 1.702                                              | 7.71                                     | 0.446                                          | 47.29                              | -                                | -                           |
| 2700                | 9640                                  | -                                                       | -                                           | 1.968                                              | 8.16                                     | 0.480                                          | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

**Revision 2: 04-26-2023**

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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.1 Molybdenum and Its Alloys

3.1.3 TZM

**Revision 2: 04-26-2023**

**References**

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### **3 Refractory Metals and Alloys**

#### **3.2 Tungsten and Tungsten Alloys**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 2.1: 08-25-2023

General

### Room Temperature Properties

|                                           |        |
|-------------------------------------------|--------|
| Atomic Mass, [amu]                        | 183.84 |
| Theoretical Density, [kg/m <sup>3</sup> ] | 19,250 |
| Melting Point, [K]                        | 3695   |
| Boiling Point, [K]                        | 5828   |
| Specific Heat, [J/(g-K)]                  | 0.133  |
| Heat of Fusion, [kJ/mol]                  | 35     |
| Heat of Vaporization, [kJ/mol]            | 800    |
| Thermal Conductivity, [W/(m-K)]           | 175.7  |
| Linear expansion coefficient, [μm/(m-K)]  | 4.58   |
| Electrical resistivity, [μΩ-m]            | 0.05   |
| Young's Modulus, [GPa]                    | 399.6  |
| Shear Modulus, [GPa]                      | 159.9  |
| Poisson's Ratio, [-]                      | 0.249  |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Density with Temperature

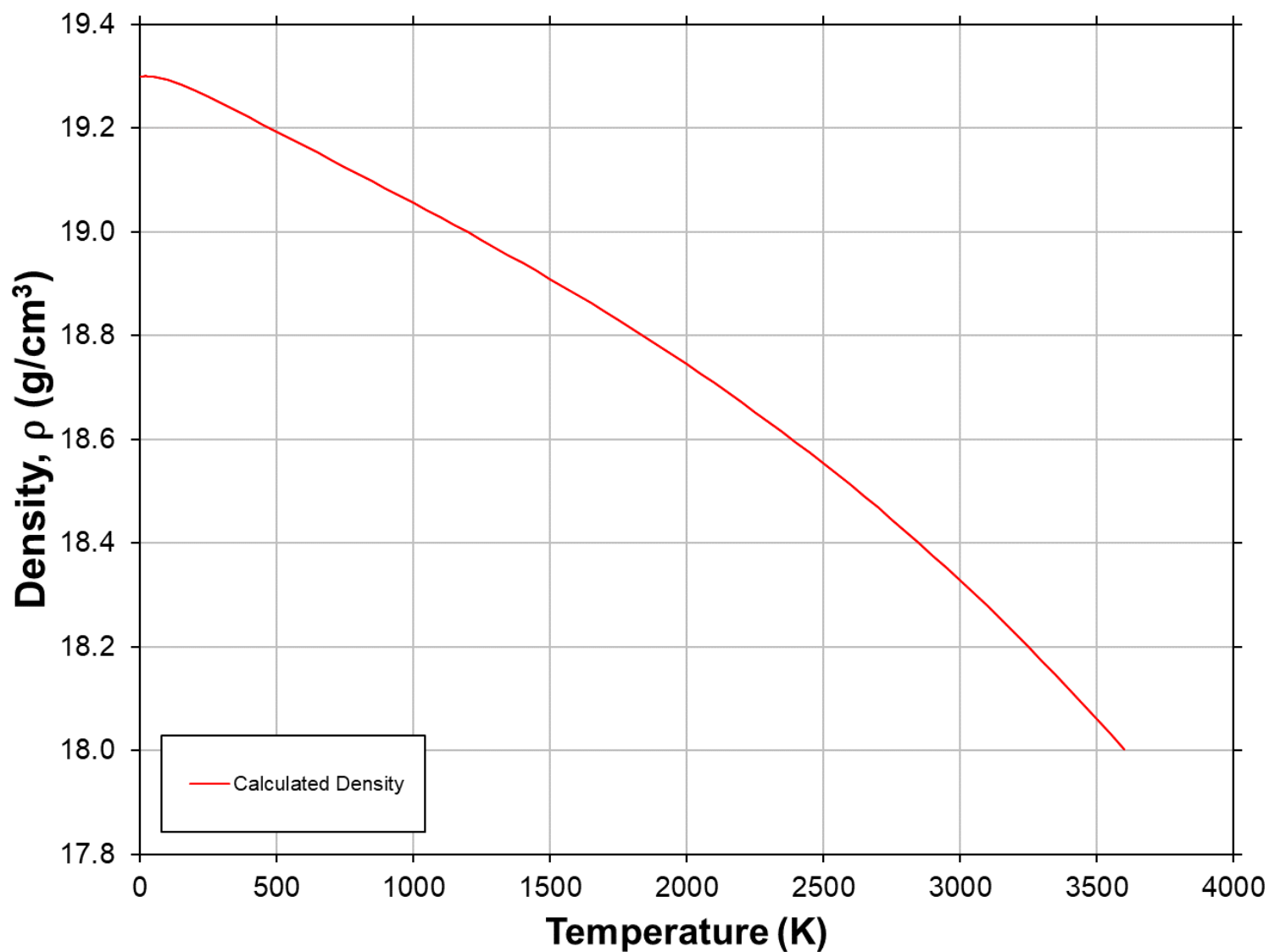


Figure 3.2.1-1: Density versus Temperature for Tungsten. Calculated from fitted trend of Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 20                | ( -423.7 ) | 19300             | ( 1204.9 )             | 1400              | ( 2060.3 ) | 18940             | ( 1182.4 )             |
| 100               | ( -279.7 ) | 19294             | ( 1204.5 )             | 1500              | ( 2240.3 ) | 18909             | ( 1180.5 )             |
| 200               | ( -99.7 )  | 19273             | ( 1203.2 )             | 1600              | ( 2420.3 ) | 18878             | ( 1178.6 )             |
| 300               | ( 80.3 )   | 19248             | ( 1201.6 )             | 1700              | ( 2600.3 ) | 18846             | ( 1176.6 )             |
| 400               | ( 260.3 )  | 19220             | ( 1199.9 )             | 1800              | ( 2780.3 ) | 18813             | ( 1174.5 )             |
| 500               | ( 440.3 )  | 19193             | ( 1198.2 )             | 1900              | ( 2960.3 ) | 18780             | ( 1172.4 )             |
| 600               | ( 620.3 )  | 19166             | ( 1196.5 )             | 2000              | ( 3140.3 ) | 18745             | ( 1170.2 )             |
| 700               | ( 800.3 )  | 19139             | ( 1194.8 )             | 2200              | ( 3500.3 ) | 18672             | ( 1165.7 )             |
| 800               | ( 980.3 )  | 19111             | ( 1193.1 )             | 2400              | ( 3860.3 ) | 18595             | ( 1160.9 )             |
| 900               | ( 1160.3 ) | 19084             | ( 1191.4 )             | 2600              | ( 4220.3 ) | 18512             | ( 1155.7 )             |
| 1000              | ( 1340.3 ) | 19056             | ( 1189.6 )             | 2800              | ( 4580.3 ) | 18423             | ( 1150.2 )             |
| 1100              | ( 1520.3 ) | 19027             | ( 1187.9 )             | 3000              | ( 4940.3 ) | 18328             | ( 1144.2 )             |
| 1200              | ( 1700.3 ) | 18999             | ( 1186.1 )             | 3200              | ( 5300.3 ) | 18227             | ( 1137.9 )             |
| 1300              | ( 1880.3 ) | 18970             | ( 1184.3 )             | 3400              | ( 5660.3 ) | 18118             | ( 1131.1 )             |
| 1400              | ( 2060.3 ) | 18940             | ( 1182.4 )             | 3600              | ( 6020.3 ) | 18002             | ( 1123.9 )             |

**Application Notes:** Density trend with respect to temperature is calculated as a function of the Thermal Expansion trend as seen the equation below to approximate property trend with respect to temperature.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 19,250 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $5 \leq T \leq 3600$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

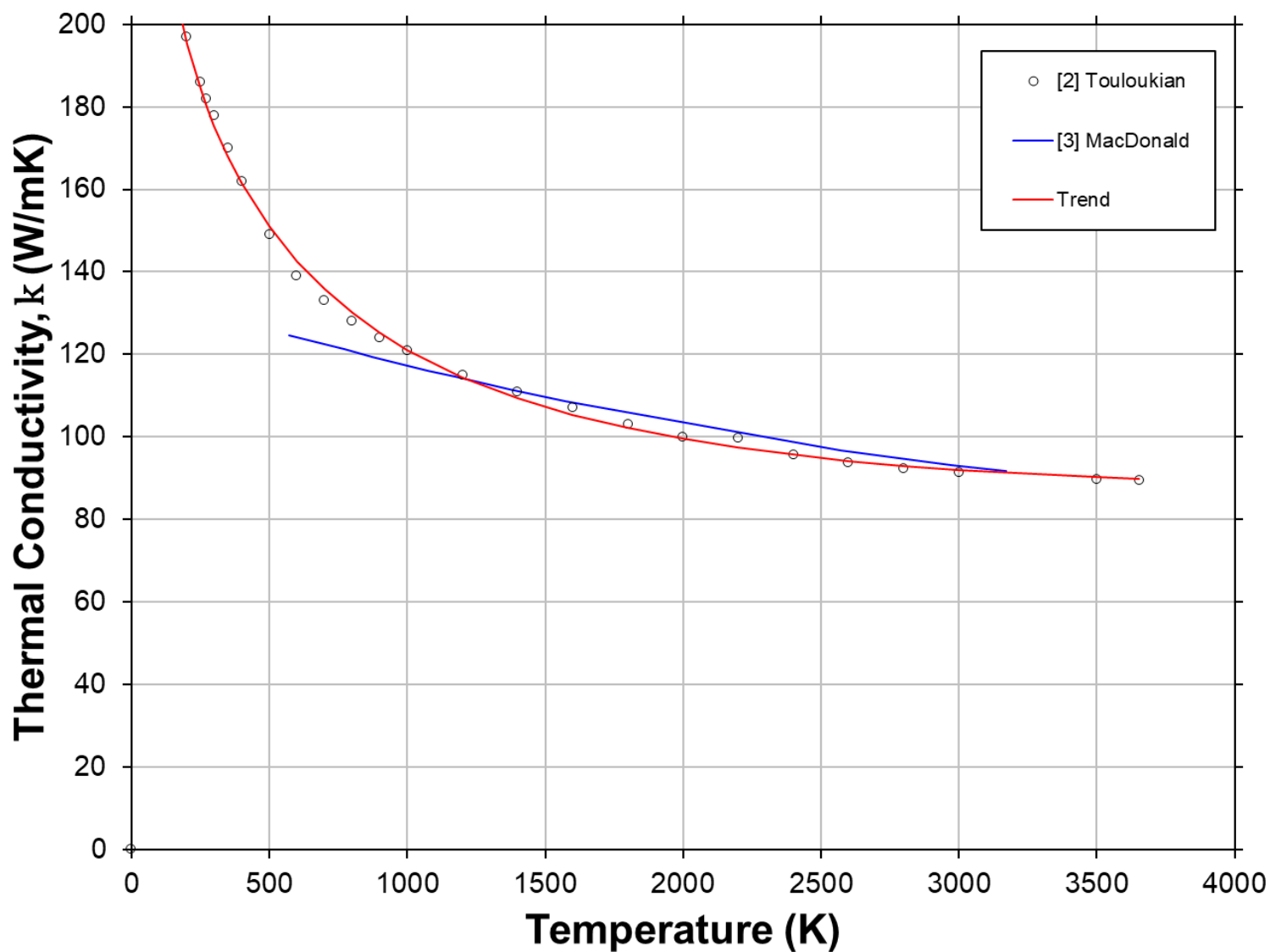


Figure 3.2.1-2: Thermal Conductivity versus Temperature of Tungsten with comparison to MacDonald (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

## 3 Refractory Metals and Alloys

## 3.2 Tungsten and Its Alloys

## 3.2.1 Tungsten (W)

Revision 0: 08-05-2020

### Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 55                | ( -360.7 ) | 349.03                     | ( 2421.58 )                         | 1300              | ( 1880.3 ) | 111.78                     | ( 775.53 )                          |
| 100               | ( -279.7 ) | 236.43                     | ( 1640.35 )                         | 1400              | ( 2060.3 ) | 109.39                     | ( 758.93 )                          |
| 200               | ( -99.7 )  | 195.24                     | ( 1354.57 )                         | 1600              | ( 2420.3 ) | 105.37                     | ( 731.09 )                          |
| 300               | ( 80.3 )   | 175.33                     | ( 1216.47 )                         | 1800              | ( 2780.3 ) | 102.16                     | ( 708.83 )                          |
| 400               | ( 260.3 )  | 161.49                     | ( 1120.46 )                         | 2000              | ( 3140.3 ) | 99.56                      | ( 690.78 )                          |
| 500               | ( 440.3 )  | 150.96                     | ( 1047.37 )                         | 2200              | ( 3500.3 ) | 97.43                      | ( 676.01 )                          |
| 600               | ( 620.3 )  | 142.59                     | ( 989.27 )                          | 2400              | ( 3860.3 ) | 95.68                      | ( 663.84 )                          |
| 700               | ( 800.3 )  | 135.75                     | ( 941.83 )                          | 2600              | ( 4220.3 ) | 94.23                      | ( 653.76 )                          |
| 800               | ( 980.3 )  | 130.06                     | ( 902.34 )                          | 2800              | ( 4580.3 ) | 93.02                      | ( 645.38 )                          |
| 900               | ( 1160.3 ) | 125.25                     | ( 868.97 )                          | 3000              | ( 4940.3 ) | 92.02                      | ( 638.42 )                          |
| 1000              | ( 1340.3 ) | 121.13                     | ( 840.44 )                          | 3200              | ( 5300.3 ) | 91.18                      | ( 632.65 )                          |
| 1100              | ( 1520.3 ) | 117.58                     | ( 815.80 )                          | 3400              | ( 5660.3 ) | 90.50                      | ( 627.88 )                          |
| 1200              | ( 1700.3 ) | 114.49                     | ( 794.35 )                          | 3600              | ( 6020.3 ) | 89.93                      | ( 623.96 )                          |
| 1300              | ( 1880.3 ) | 111.78                     | ( 775.53 )                          | 3653              | ( 6115.7 ) | 89.80                      | ( 623.05 )                          |

**Application Notes:** Data for thermal conductivity is collected from reference [2] and fitted with the equations below to approximate the property trend with respect to temperature. Fitted trend is compared to trend from reference [3].

### Fit Equation:

For temperature range:  $1 \leq T < 55$

$$k(T) = \left( A0 \cdot \left( \frac{T}{1000} \right)^N \right) / \left( 1 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 + A3 \cdot \left( \frac{T}{1000} \right)^3 \right)$$

For temperature range:  $55 \leq T \leq 3653$

$$k(T) = \left( B0 + B1 \cdot \left( \frac{T}{1000} \right) + B2 \cdot \left( \frac{T}{1000} \right)^2 + B3 \cdot \left( \frac{T}{1000} \right)^3 \right) / \left( C0 + C1 \cdot \left( \frac{T}{1000} \right) + \left( \frac{T}{1000} \right)^2 \right)$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

### Constants:

T Range

[K]:

$1 \leq T < 55$

$55 \leq T \leq 3653$

|      |            |      |            |      |            |
|------|------------|------|------------|------|------------|
| N =  | 8.740E-01  | B0 = | -3.679E+00 | B3 = | 2.867E+00  |
| A0 = | 7.348E+05  | B1 = | 1.181E+02  | C0 = | -2.052E-02 |
| A1 = | 2.544E+01  | B2 = | 5.879E+01  | C1 = | 4.741E-01  |
| A2 = | -8.304E+03 |      |            |      |            |
| A3 = | 1.180E+06  |      |            |      |            |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

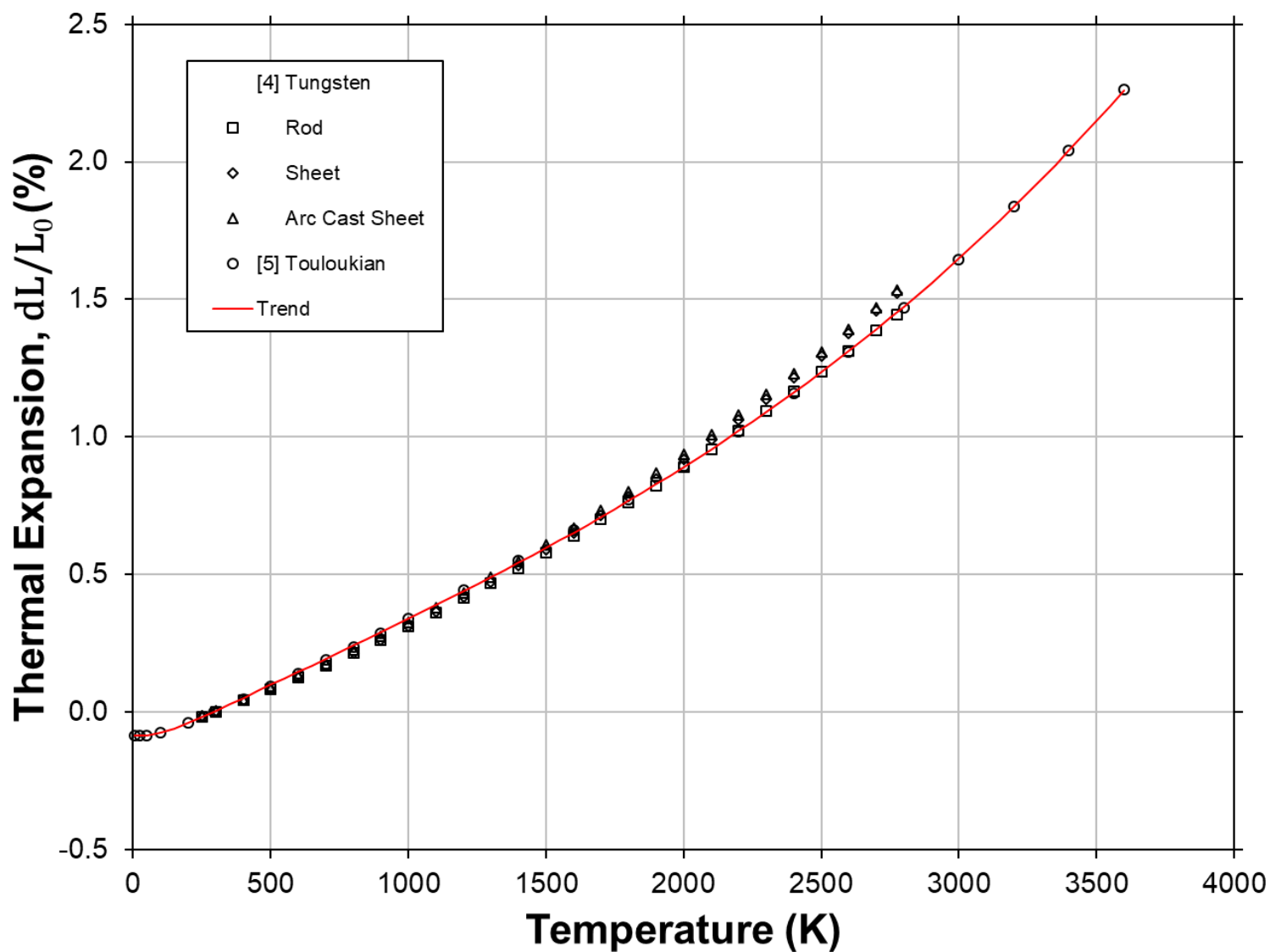


Figure 3.2.1-3: Thermal Expansion versus Temperature of Tungsten.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

## 3 Refractory Metals and Alloys

## 3.2 Tungsten and Its Alloys

## 3.2.1 Tungsten (W)

Revision 0: 08-05-2020

### Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 20                | ( -423.7 ) | -0.086                                  | 1400              | ( 2060.3 ) | 0.543                                   |
| 100               | ( -279.7 ) | -0.076                                  | 1500              | ( 2240.3 ) | 0.597                                   |
| 200               | ( -99.7 )  | -0.040                                  | 1600              | ( 2420.3 ) | 0.652                                   |
| 300               | ( 80.3 )   | 0.004                                   | 1700              | ( 2600.3 ) | 0.709                                   |
| 400               | ( 260.3 )  | 0.051                                   | 1800              | ( 2780.3 ) | 0.768                                   |
| 500               | ( 440.3 )  | 0.099                                   | 1900              | ( 2960.3 ) | 0.828                                   |
| 600               | ( 620.3 )  | 0.146                                   | 2000              | ( 3140.3 ) | 0.890                                   |
| 700               | ( 800.3 )  | 0.194                                   | 2200              | ( 3500.3 ) | 1.021                                   |
| 800               | ( 980.3 )  | 0.241                                   | 2400              | ( 3860.3 ) | 1.161                                   |
| 900               | ( 1160.3 ) | 0.290                                   | 2600              | ( 4220.3 ) | 1.312                                   |
| 1000              | ( 1340.3 ) | 0.339                                   | 2800              | ( 4580.3 ) | 1.474                                   |
| 1100              | ( 1520.3 ) | 0.388                                   | 3000              | ( 4940.3 ) | 1.649                                   |
| 1200              | ( 1700.3 ) | 0.439                                   | 3200              | ( 5300.3 ) | 1.837                                   |
| 1300              | ( 1880.3 ) | 0.490                                   | 3400              | ( 5660.3 ) | 2.040                                   |
| 1400              | ( 2060.3 ) | 0.543                                   | 3600              | ( 6020.3 ) | 2.259                                   |

**Application Notes:** Data for thermal expansion is collected from references [4, 5] and fitted with the equation below to approximate property trend with respect to temperature.

#### Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

#### Constants:

| T Range [K]: | $5 \leq T \leq 294$ | $294 < T \leq 3600$ |
|--------------|---------------------|---------------------|
| A0 =         | -8.529E-02          | -1.400E-01          |
| A1 =         | -9.915E-02          | 4.869E-01           |
| A2 =         | 2.257E+00           | -3.056E-02          |
| A3 =         | -3.157E+00          | 2.234E-02           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

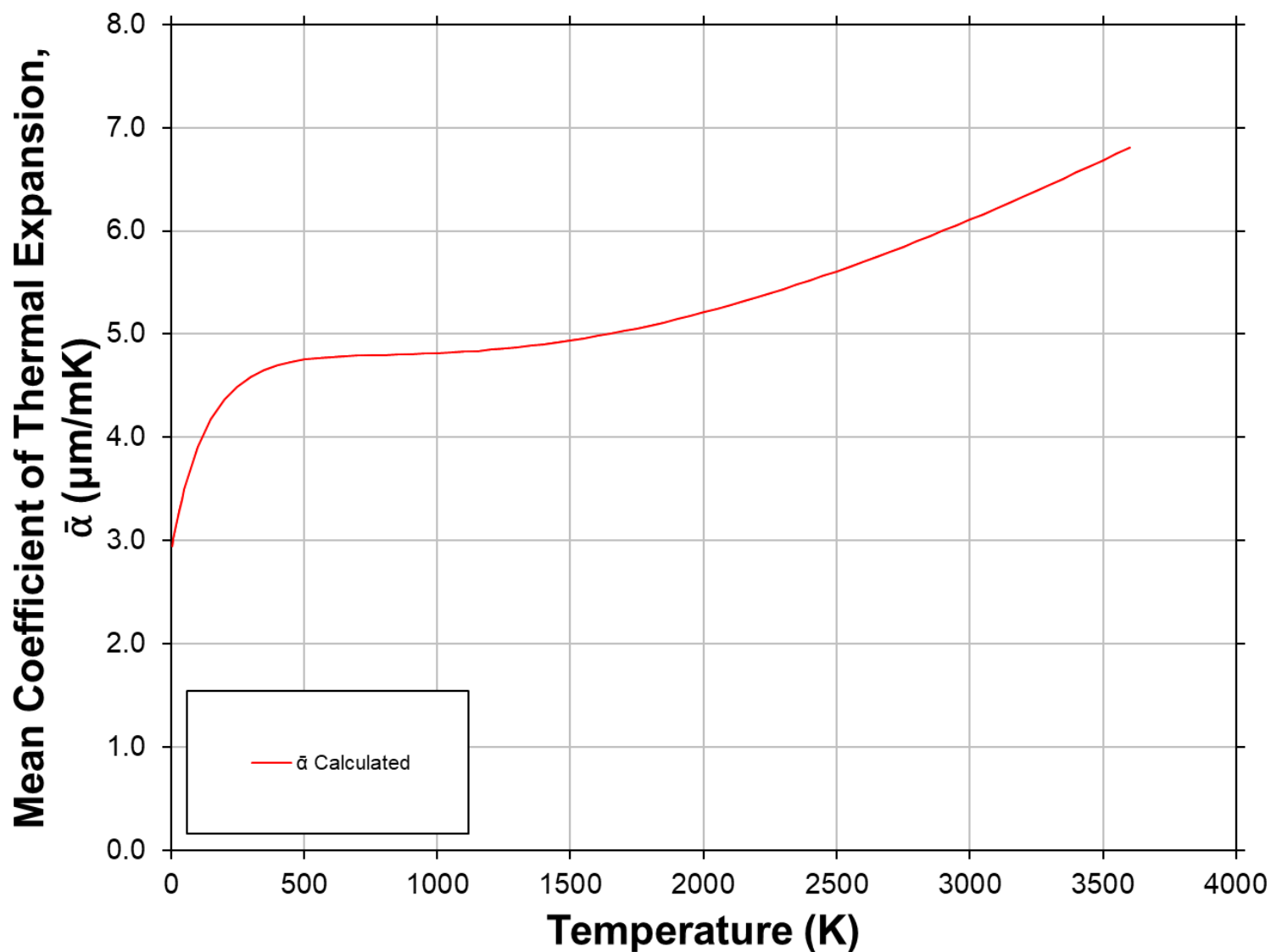


Figure 3.2.1-4: Mean Coefficient of Thermal Expansion versus Temperature of Tungsten. Calculated from fitted trend of Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 20                | ( -423.7 ) | 3.158                                 | ( 1.755 )                                           | 1400              | ( 2060.3 ) | 4.902                                 | ( 2.723 )                                           |
| 100               | ( -279.7 ) | 3.904                                 | ( 2.169 )                                           | 1500              | ( 2240.3 ) | 4.938                                 | ( 2.743 )                                           |
| 200               | ( -99.7 )  | 4.361                                 | ( 2.423 )                                           | 1600              | ( 2420.3 ) | 4.981                                 | ( 2.767 )                                           |
| 300               | ( 80.3 )   | 4.584                                 | ( 2.547 )                                           | 1700              | ( 2600.3 ) | 5.030                                 | ( 2.794 )                                           |
| 400               | ( 260.3 )  | 4.696                                 | ( 2.609 )                                           | 1800              | ( 2780.3 ) | 5.084                                 | ( 2.825 )                                           |
| 500               | ( 440.3 )  | 4.752                                 | ( 2.640 )                                           | 1900              | ( 2960.3 ) | 5.144                                 | ( 2.858 )                                           |
| 600               | ( 620.3 )  | 4.779                                 | ( 2.655 )                                           | 2000              | ( 3140.3 ) | 5.209                                 | ( 2.894 )                                           |
| 700               | ( 800.3 )  | 4.791                                 | ( 2.662 )                                           | 2200              | ( 3500.3 ) | 5.355                                 | ( 2.975 )                                           |
| 800               | ( 980.3 )  | 4.799                                 | ( 2.666 )                                           | 2400              | ( 3860.3 ) | 5.520                                 | ( 3.066 )                                           |
| 900               | ( 1160.3 ) | 4.806                                 | ( 2.670 )                                           | 2600              | ( 4220.3 ) | 5.701                                 | ( 3.167 )                                           |
| 1000              | ( 1340.3 ) | 4.815                                 | ( 2.675 )                                           | 2800              | ( 4580.3 ) | 5.897                                 | ( 3.276 )                                           |
| 1100              | ( 1520.3 ) | 4.828                                 | ( 2.682 )                                           | 3000              | ( 4940.3 ) | 6.107                                 | ( 3.393 )                                           |
| 1200              | ( 1700.3 ) | 4.847                                 | ( 2.693 )                                           | 3200              | ( 5300.3 ) | 6.330                                 | ( 3.517 )                                           |
| 1300              | ( 1880.3 ) | 4.871                                 | ( 2.706 )                                           | 3400              | ( 5660.3 ) | 6.564                                 | ( 3.647 )                                           |
| 1400              | ( 2060.3 ) | 4.902                                 | ( 2.723 )                                           | 3600              | ( 6020.3 ) | 6.808                                 | ( 3.782 )                                           |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated as a function thermal expansion data, then fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = \left( A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right) / \left( A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + \left( \frac{T}{1000} \right)^2 \right)$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

T = Temperature [K]

## Constants:

T. Range [K]:

$5 \leq T \leq 3600$

|      |           |       |           |
|------|-----------|-------|-----------|
| A0 = | 1.554E+00 | A3 =  | 1.971E+00 |
| A1 = | 1.581E+01 | A_0 = | 5.412E-01 |
| A2 = | 2.714E-01 | A_1 = | 2.531E+00 |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Specific Heat with Temperature

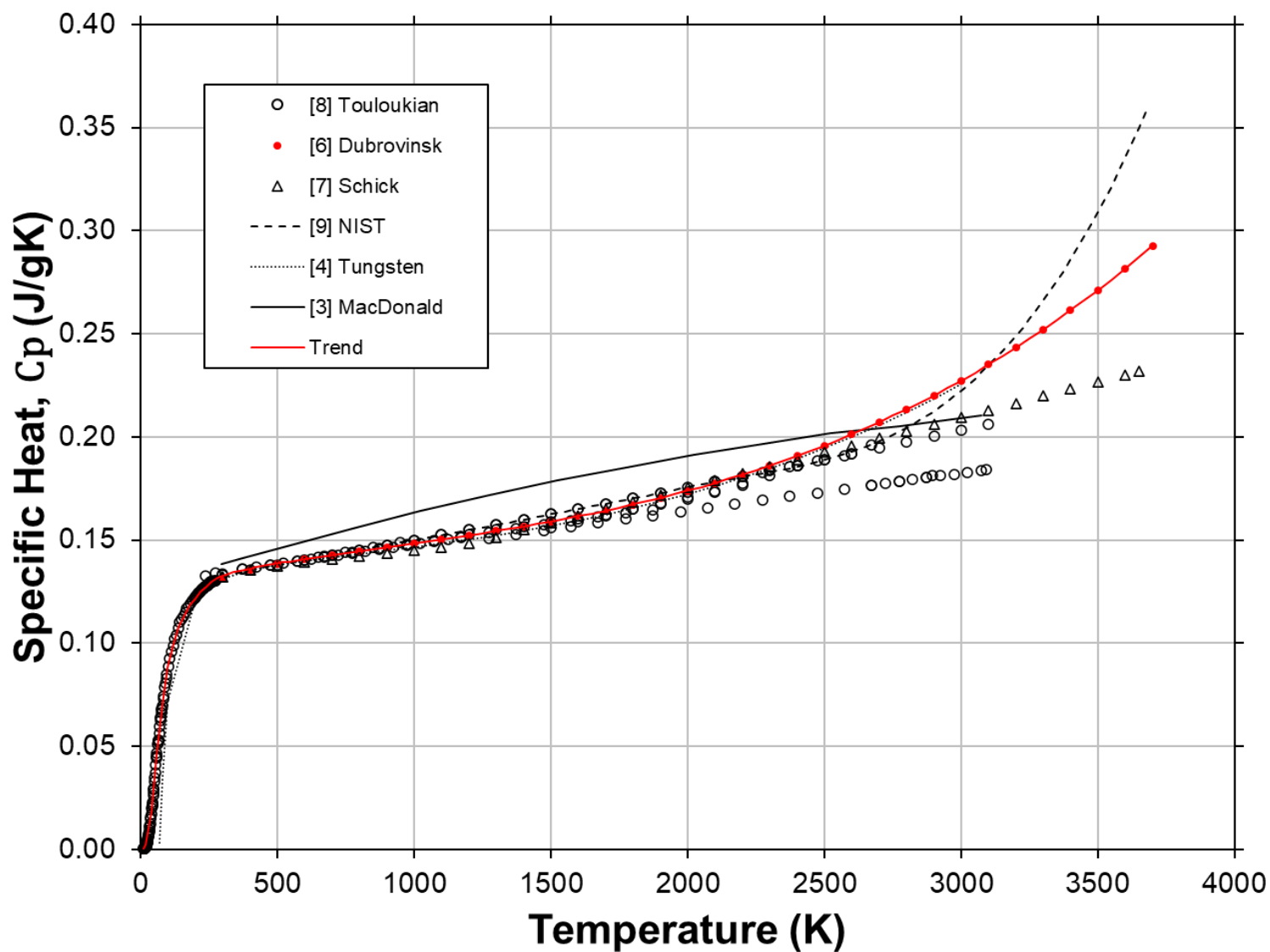


Figure 3.2.1-5: Specific Heat versus Temperature of Tungsten with comparison to reference.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 20                | ( -423.7 ) | 0.002                            | ( 0.001 )       | 1400              | ( 2060.3 ) | 0.157                            | ( 0.037 )       |
| 100               | ( -279.7 ) | 0.088                            | ( 0.021 )       | 1600              | ( 2420.3 ) | 0.162                            | ( 0.039 )       |
| 200               | ( -99.7 )  | 0.122                            | ( 0.029 )       | 1800              | ( 2780.3 ) | 0.167                            | ( 0.040 )       |
| 300               | ( 80.3 )   | 0.133                            | ( 0.032 )       | 2000              | ( 3140.3 ) | 0.174                            | ( 0.042 )       |
| 400               | ( 260.3 )  | 0.136                            | ( 0.033 )       | 2200              | ( 3500.3 ) | 0.182                            | ( 0.043 )       |
| 500               | ( 440.3 )  | 0.139                            | ( 0.033 )       | 2400              | ( 3860.3 ) | 0.191                            | ( 0.046 )       |
| 600               | ( 620.3 )  | 0.141                            | ( 0.034 )       | 2600              | ( 4220.3 ) | 0.201                            | ( 0.048 )       |
| 700               | ( 800.3 )  | 0.143                            | ( 0.034 )       | 2800              | ( 4580.3 ) | 0.213                            | ( 0.051 )       |
| 800               | ( 980.3 )  | 0.145                            | ( 0.035 )       | 3000              | ( 4940.3 ) | 0.227                            | ( 0.054 )       |
| 900               | ( 1160.3 ) | 0.146                            | ( 0.035 )       | 3200              | ( 5300.3 ) | 0.243                            | ( 0.058 )       |
| 1000              | ( 1340.3 ) | 0.148                            | ( 0.035 )       | 3400              | ( 5660.3 ) | 0.261                            | ( 0.062 )       |
| 1100              | ( 1520.3 ) | 0.150                            | ( 0.036 )       | 3600              | ( 6020.3 ) | 0.282                            | ( 0.067 )       |
| 1200              | ( 1700.3 ) | 0.152                            | ( 0.036 )       | 3700              | ( 6200.3 ) | 0.293                            | ( 0.070 )       |

**Application Notes:** Data for specific heat is collected from references [6-8] and fitted with the equations below to approximate property trend with respect to temperature. Fitted trend is compared against trends from references [3, 4, 9].

## Fit Equation:

For temperature range:  $11 \leq T < 293$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right]$$

For temperature range:  $293 \leq T \leq 3700$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_3 \cdot \left( \frac{T}{1000} \right)^3 + B_{-2} / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$T = \text{Temperature [K]}$

## Constants:

T. Range [K]:

$11 \leq T < 293$

$293 \leq T \leq 3700$

N = 3.030E+00

B<sub>0</sub> = 1.301E-01

A<sub>0</sub> = 3.103E+02

B<sub>1</sub> = 2.225E-02

A<sub>1</sub> = -8.815E+00

B<sub>2</sub> = -7.224E-03

A<sub>2</sub> = 1.295E+02

B<sub>3</sub> = 3.539E-03

A<sub>3</sub> = 1.874E+03

B<sub>-2</sub> = -3.061E-04



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

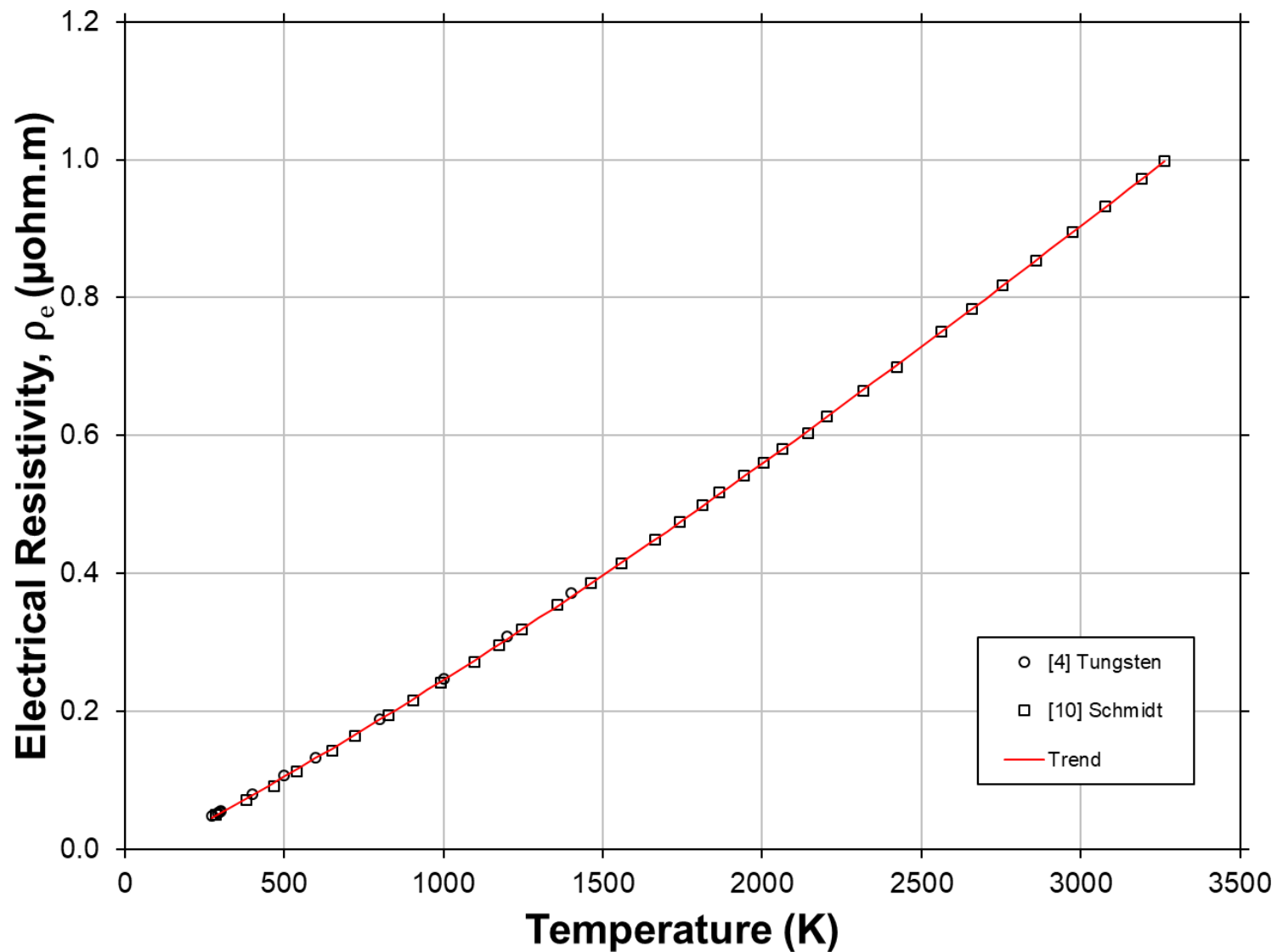


Figure 3.2.1-6: Electrical Resistivity versus Temperature for Tungsten.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

## 3 Refractory Metals and Alloys

## 3.2 Tungsten and Its Alloys

## 3.2.1 Tungsten (W)

Revision 0: 08-05-2020

### Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 273               | ( 31.7 )   | 0.046                               | ( 1.81 )                | 1700              | ( 2600.3 ) | 0.461                               | ( 18.15 )               |
| 300               | ( 80.3 )   | 0.053                               | ( 2.08 )                | 1800              | ( 2780.3 ) | 0.493                               | ( 19.42 )               |
| 400               | ( 260.3 )  | 0.079                               | ( 3.10 )                | 1900              | ( 2960.3 ) | 0.526                               | ( 20.71 )               |
| 500               | ( 440.3 )  | 0.105                               | ( 4.15 )                | 2000              | ( 3140.3 ) | 0.559                               | ( 22.01 )               |
| 600               | ( 620.3 )  | 0.132                               | ( 5.21 )                | 2100              | ( 3320.3 ) | 0.592                               | ( 23.32 )               |
| 700               | ( 800.3 )  | 0.160                               | ( 6.29 )                | 2200              | ( 3500.3 ) | 0.626                               | ( 24.65 )               |
| 800               | ( 980.3 )  | 0.188                               | ( 7.40 )                | 2300              | ( 3680.3 ) | 0.660                               | ( 25.99 )               |
| 900               | ( 1160.3 ) | 0.216                               | ( 8.52 )                | 2400              | ( 3860.3 ) | 0.694                               | ( 27.33 )               |
| 1000              | ( 1340.3 ) | 0.245                               | ( 9.66 )                | 2500              | ( 4040.3 ) | 0.729                               | ( 28.69 )               |
| 1100              | ( 1520.3 ) | 0.275                               | ( 10.83 )               | 2600              | ( 4220.3 ) | 0.763                               | ( 30.05 )               |
| 1200              | ( 1700.3 ) | 0.305                               | ( 12.01 )               | 2700              | ( 4400.3 ) | 0.798                               | ( 31.43 )               |
| 1300              | ( 1880.3 ) | 0.335                               | ( 13.20 )               | 2800              | ( 4580.3 ) | 0.833                               | ( 32.81 )               |
| 1400              | ( 2060.3 ) | 0.366                               | ( 14.42 )               | 3000              | ( 4940.3 ) | 0.904                               | ( 35.60 )               |
| 1500              | ( 2240.3 ) | 0.397                               | ( 15.64 )               | 3200              | ( 5300.3 ) | 0.976                               | ( 38.41 )               |
| 1600              | ( 2420.3 ) | 0.429                               | ( 16.89 )               | 3261              | ( 5410.1 ) | 0.998                               | ( 39.28 )               |

**Application Notes:** Data for electrical resistivity is collected from references [4, 10] and fitted with the equation below to approximate property trend with respect to temperature.

#### Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

#### Constants:

T. Range [K]:  $273 \leq T \leq 3261$

A0 = -2.122E-02

A1 = 2.381E-01

A2 = 3.116E-02

A3 = -2.566E-03



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Young's Modulus with Temperature

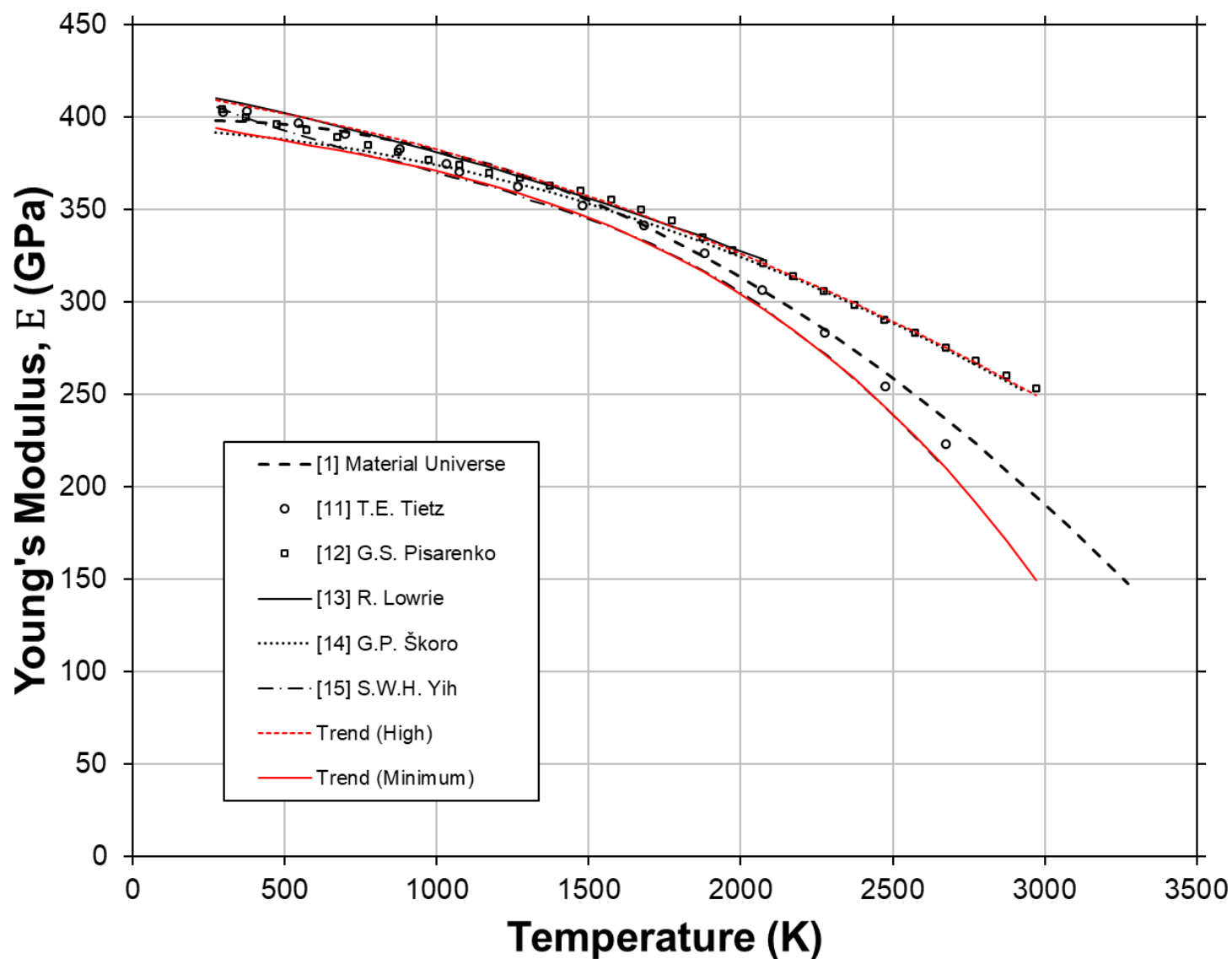


Figure 3.2.1-7: Young's modulus versus temperature for tungsten.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Young's Modulus with Temperature

Minimum Young's Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   |
| 273               | ( 31.7 )   | 393.79                | ( 57.14 ) |
| 300               | ( 80.3 )   | 392.99                | ( 57.02 ) |
| 500               | ( 440.3 )  | 387.16                | ( 56.18 ) |
| 700               | ( 800.3 )  | 381.12                | ( 55.30 ) |
| 900               | ( 1160.3 ) | 374.41                | ( 54.33 ) |
| 1100              | ( 1520.3 ) | 366.55                | ( 53.19 ) |
| 1300              | ( 1880.3 ) | 357.05                | ( 51.81 ) |
| 1500              | ( 2240.3 ) | 345.45                | ( 50.12 ) |
| 1700              | ( 2600.3 ) | 331.27                | ( 48.07 ) |
| 1900              | ( 2960.3 ) | 314.04                | ( 45.57 ) |
| 2100              | ( 3320.3 ) | 293.27                | ( 42.55 ) |
| 2300              | ( 3680.3 ) | 268.50                | ( 38.96 ) |
| 2500              | ( 4040.3 ) | 239.24                | ( 34.71 ) |
| 2700              | ( 4400.3 ) | 205.03                | ( 29.75 ) |
| 2973              | ( 4891.7 ) | 149.46                | ( 21.69 ) |

High Young's Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   |
| 273               | ( 31.7 )   | 408.89                | ( 59.33 ) |
| 300               | ( 80.3 )   | 408.13                | ( 59.22 ) |
| 500               | ( 440.3 )  | 401.93                | ( 58.32 ) |
| 700               | ( 800.3 )  | 394.81                | ( 57.29 ) |
| 900               | ( 1160.3 ) | 386.77                | ( 56.12 ) |
| 1100              | ( 1520.3 ) | 377.82                | ( 54.82 ) |
| 1300              | ( 1880.3 ) | 367.94                | ( 53.39 ) |
| 1500              | ( 2240.3 ) | 357.14                | ( 51.82 ) |
| 1700              | ( 2600.3 ) | 345.42                | ( 50.12 ) |
| 1900              | ( 2960.3 ) | 332.78                | ( 48.29 ) |
| 2100              | ( 3320.3 ) | 319.23                | ( 46.32 ) |
| 2300              | ( 3680.3 ) | 304.75                | ( 44.22 ) |
| 2500              | ( 4040.3 ) | 289.35                | ( 41.98 ) |
| 2700              | ( 4400.3 ) | 273.03                | ( 39.62 ) |
| 2973              | ( 4891.7 ) | 249.27                | ( 36.17 ) |

**Application Notes:** Data for Young's modulus is collected from references [1, 11-15] and fitted with the equation below to approximate the property trend with respect to temperature. Since two distinct trends were observed, a minimum and a high equation were generated.

## Fit Equations:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

T Range [K]:  $273 \leq T \leq 2973$

## Constants:

| Curve: | <u>Minimum</u> | <u>High</u> |
|--------|----------------|-------------|
| A0 =   | 402.4          | 415.7       |
| A1 =   | -34.17         | -21.79      |
| A2 =   | 12.33          | -11.5       |
| A3 =   | -9.907         | 0           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Shear Modulus with Temperature

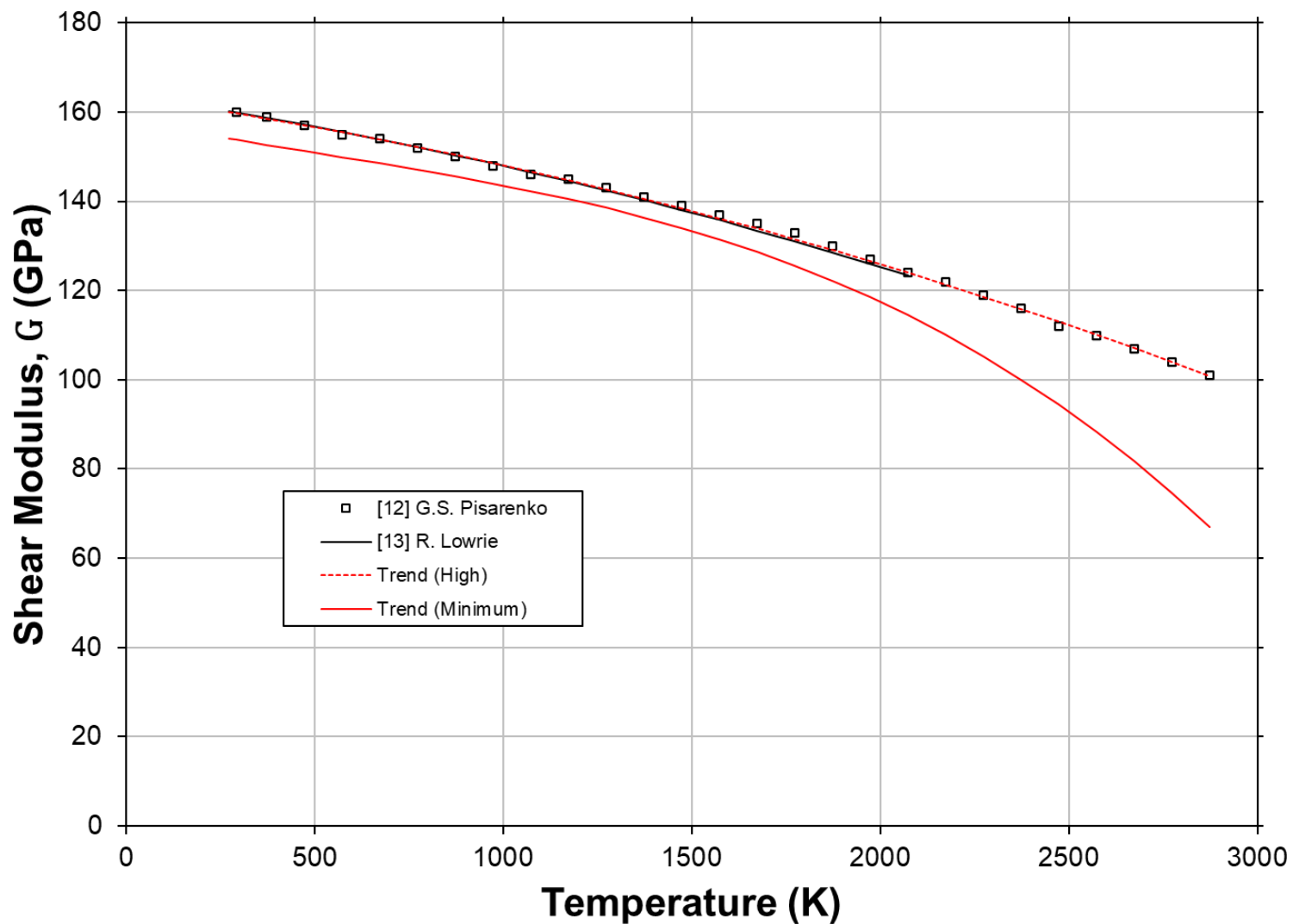


Figure 3.2.1-8: Shear modulus versus temperature for tungsten. The high trend was fitted using data, while the minimum trend was calculated from the Young's modulus (minimum) trend and the Poisson's ratio trend.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Shear Modulus with Temperature

Minimum Shear Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( E ) |           |
|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   |
| 273               | ( 31.7 )   | 154.14              | ( 22.37 ) |
| 300               | ( 80.3 )   | 153.75              | ( 22.31 ) |
| 500               | ( 440.3 )  | 150.94              | ( 21.90 ) |
| 700               | ( 800.3 )  | 148.15              | ( 21.50 ) |
| 900               | ( 1160.3 ) | 145.19              | ( 21.07 ) |
| 1100              | ( 1520.3 ) | 141.88              | ( 20.59 ) |
| 1300              | ( 1880.3 ) | 138.01              | ( 20.03 ) |
| 1500              | ( 2240.3 ) | 133.41              | ( 19.36 ) |
| 1700              | ( 2600.3 ) | 127.89              | ( 18.56 ) |
| 1900              | ( 2960.3 ) | 121.26              | ( 17.59 ) |
| 2100              | ( 3320.3 ) | 113.32              | ( 16.44 ) |
| 2300              | ( 3680.3 ) | 103.90              | ( 15.08 ) |
| 2500              | ( 4040.3 ) | 92.81               | ( 13.47 ) |
| 2700              | ( 4400.3 ) | 79.85               | ( 11.59 ) |
| 2873              | ( 4711.7 ) | 66.99               | ( 9.72 )  |

High Shear Modulus at 100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( E ) |           |
|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   |
| 273               | ( 31.7 )   | 159.98              | ( 23.21 ) |
| 300               | ( 80.3 )   | 159.60              | ( 23.16 ) |
| 500               | ( 440.3 )  | 156.66              | ( 22.73 ) |
| 700               | ( 800.3 )  | 153.44              | ( 22.26 ) |
| 900               | ( 1160.3 ) | 149.95              | ( 21.76 ) |
| 1100              | ( 1520.3 ) | 146.19              | ( 21.21 ) |
| 1300              | ( 1880.3 ) | 142.16              | ( 20.63 ) |
| 1500              | ( 2240.3 ) | 137.85              | ( 20.00 ) |
| 1700              | ( 2600.3 ) | 133.27              | ( 19.34 ) |
| 1900              | ( 2960.3 ) | 128.41              | ( 18.63 ) |
| 2100              | ( 3320.3 ) | 123.29              | ( 17.89 ) |
| 2300              | ( 3680.3 ) | 117.89              | ( 17.11 ) |
| 2500              | ( 4040.3 ) | 112.21              | ( 16.28 ) |
| 2700              | ( 4400.3 ) | 106.27              | ( 15.42 ) |
| 2873              | ( 4711.7 ) | 100.90              | ( 14.64 ) |

**Application Notes:** Data for shear modulus is collected from references [12, 13] and fitted with the equation below to approximate a high property trend with respect to temperature. The minimum trend is calculated as a function of Young's modulus (minimum) and Poisson's ratio. Equivalent fit equations are provided below for readers' convenience.

## Fit Equations:

$$G(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

T Range [K]:  $273 \leq T \leq 2873$

## Constants:

| Curve: | <u>Minimum</u> | <u>High</u> |
|--------|----------------|-------------|
| A0 =   | 158.4          | 163.5       |
| A1 =   | -16.99         | -11.98      |
| A2 =   | 6.101          | -3.414      |
| A3 =   | -3.92          | 0           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

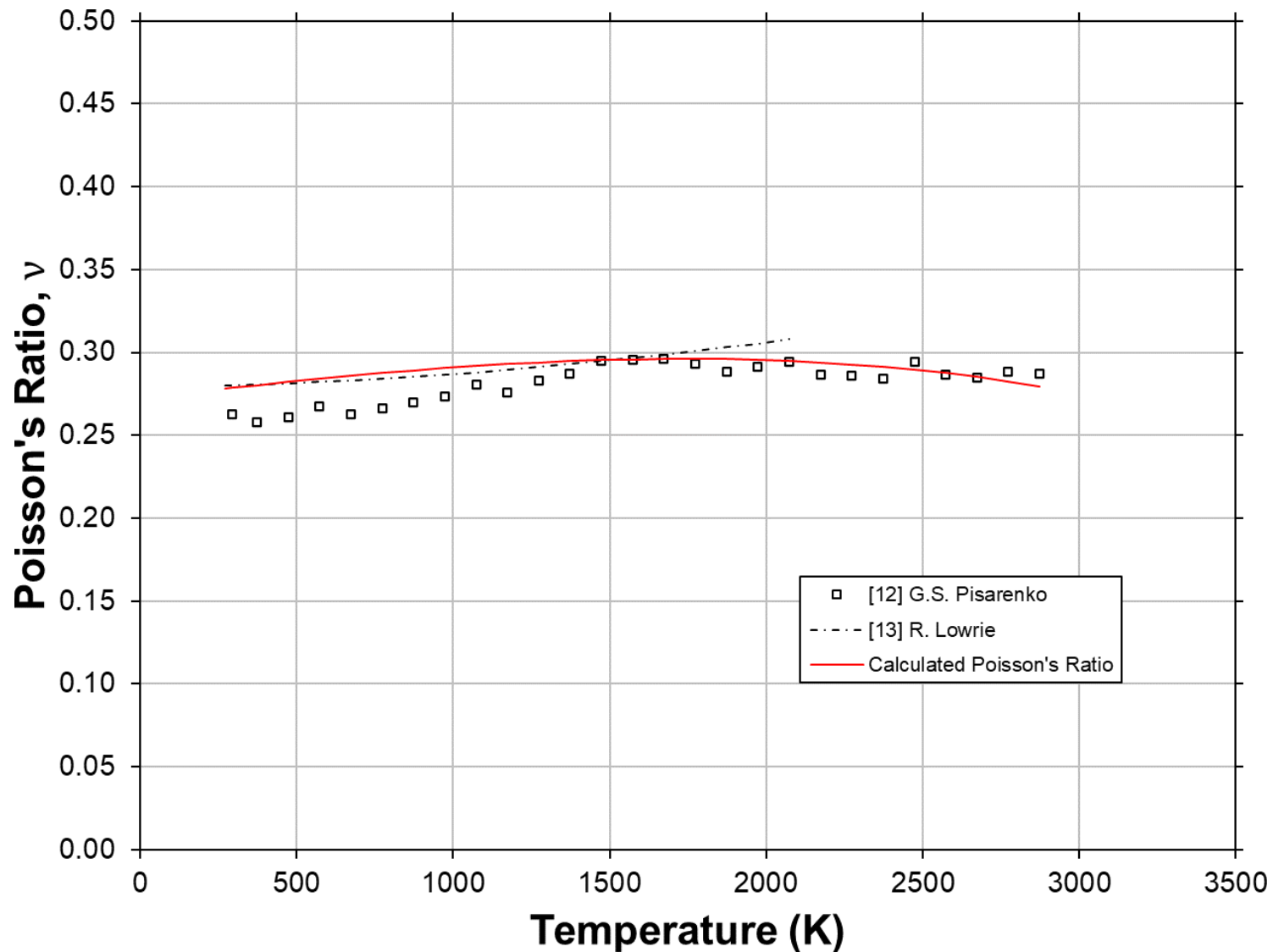


Figure 3.2.1-9: Poisson's ratio versus temperature for tungsten, calculated from fitted Young's modulus (high) and shear modulus (high) curves.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|------------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )     |                           | K                 | ( °F )     |                           |
| 273               | ( 31.7 )   | 0.278                     | 1000              | ( 1340.3 ) | 0.291                     |
| 300               | ( 80.3 )   | 0.279                     | 1100              | ( 1520.3 ) | 0.292                     |
| 350               | ( 170.3 )  | 0.280                     | 1200              | ( 1700.3 ) | 0.293                     |
| 400               | ( 260.3 )  | 0.281                     | 1300              | ( 1880.3 ) | 0.294                     |
| 450               | ( 350.3 )  | 0.282                     | 1400              | ( 2060.3 ) | 0.295                     |
| 500               | ( 440.3 )  | 0.283                     | 1500              | ( 2240.3 ) | 0.295                     |
| 550               | ( 530.3 )  | 0.284                     | 1600              | ( 2420.3 ) | 0.296                     |
| 600               | ( 620.3 )  | 0.285                     | 1700              | ( 2600.3 ) | 0.296                     |
| 650               | ( 710.3 )  | 0.286                     | 1800              | ( 2780.3 ) | 0.296                     |
| 700               | ( 800.3 )  | 0.287                     | 1900              | ( 2960.3 ) | 0.296                     |
| 750               | ( 890.3 )  | 0.287                     | 2000              | ( 3140.3 ) | 0.295                     |
| 800               | ( 980.3 )  | 0.288                     | 2200              | ( 3500.3 ) | 0.294                     |
| 850               | ( 1070.3 ) | 0.289                     | 2400              | ( 3860.3 ) | 0.291                     |
| 900               | ( 1160.3 ) | 0.290                     | 2600              | ( 4220.3 ) | 0.287                     |
| 950               | ( 1250.3 ) | 0.290                     | 2873              | ( 4711.7 ) | 0.279                     |

**Application Notes:** Poisson's ratio is calculated as a function of Young's modulus (high) and shear modulus (high) to approximate the property trend with respect to temperature and is compared against references [12, 13]. The trend was then fitted with an equivalent equation below for readers' convenience.

## Fit Equation:

$$\nu(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\nu(T)$  = Poisson's Ratio

$T$  = Temperature [K]

## Constants:

T Range [K]: 273 ≤ T ≤ 2873

A0 = 0.2722

A1 = 0.02244

A2 = -2.077E-3

A3 = -1.682E-3



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 19294                                 | -                                                       | 236.43                                      | -0.076                                             | 3.90                                      | 0.088                                          | -                                  | -                                | -                           |
| 200                 | 19273                                 | -                                                       | 195.24                                      | -0.040                                             | 4.36                                      | 0.122                                          | -                                  | -                                | -                           |
| 300                 | 19248                                 | 0.053                                                   | 175.33                                      | 0.004                                              | 4.58                                      | 0.133                                          | 399.54                             | 159.87                           | 0.250                       |
| 400                 | 19220                                 | 0.079                                                   | 161.49                                      | 0.051                                              | 4.70                                      | 0.136                                          | 397.23                             | 158.37                           | 0.254                       |
| 500                 | 19193                                 | 0.105                                                   | 150.96                                      | 0.099                                              | 4.75                                      | 0.139                                          | 394.66                             | 156.80                           | 0.259                       |
| 600                 | 19166                                 | 0.132                                                   | 142.59                                      | 0.146                                              | 4.78                                      | 0.141                                          | 391.83                             | 155.16                           | 0.263                       |
| 700                 | 19139                                 | 0.160                                                   | 135.75                                      | 0.194                                              | 4.79                                      | 0.143                                          | 388.75                             | 153.45                           | 0.267                       |
| 800                 | 19111                                 | 0.188                                                   | 130.06                                      | 0.241                                              | 4.80                                      | 0.145                                          | 385.40                             | 151.68                           | 0.270                       |
| 900                 | 19084                                 | 0.216                                                   | 125.25                                      | 0.290                                              | 4.81                                      | 0.146                                          | 381.79                             | 149.85                           | 0.274                       |
| 1000                | 19056                                 | 0.245                                                   | 121.13                                      | 0.339                                              | 4.81                                      | 0.148                                          | 377.92                             | 147.95                           | 0.277                       |
| 1100                | 19027                                 | 0.275                                                   | 117.58                                      | 0.388                                              | 4.83                                      | 0.150                                          | 373.79                             | 145.98                           | 0.280                       |
| 1200                | 18999                                 | 0.305                                                   | 114.49                                      | 0.439                                              | 4.85                                      | 0.152                                          | 369.40                             | 143.95                           | 0.283                       |
| 1300                | 18970                                 | 0.335                                                   | 111.78                                      | 0.490                                              | 4.87                                      | 0.154                                          | 364.75                             | 141.86                           | 0.286                       |
| 1400                | 18940                                 | 0.366                                                   | 109.39                                      | 0.543                                              | 4.90                                      | 0.157                                          | 359.84                             | 139.69                           | 0.288                       |
| 1600                | 18878                                 | 0.429                                                   | 105.37                                      | 0.652                                              | 4.98                                      | 0.162                                          | 349.24                             | 135.17                           | 0.292                       |
| 1800                | 18813                                 | 0.493                                                   | 102.16                                      | 0.768                                              | 5.08                                      | 0.167                                          | 337.60                             | 130.39                           | 0.295                       |
| 2000                | 18745                                 | 0.559                                                   | 99.56                                       | 0.890                                              | 5.21                                      | 0.174                                          | 324.92                             | 125.34                           | 0.296                       |
| 2200                | 18672                                 | 0.626                                                   | 97.43                                       | 1.021                                              | 5.36                                      | 0.182                                          | 311.20                             | 120.03                           | 0.296                       |
| 2400                | 18595                                 | 0.694                                                   | 95.68                                       | 1.161                                              | 5.52                                      | 0.191                                          | 296.43                             | 114.46                           | 0.295                       |
| 2600                | 18512                                 | 0.763                                                   | 94.23                                       | 1.312                                              | 5.70                                      | 0.201                                          | 280.63                             | 108.63                           | 0.292                       |
| 2800                | 18423                                 | 0.833                                                   | 93.02                                       | 1.474                                              | 5.90                                      | 0.213                                          | 263.79                             | 102.53                           | 0.286                       |
| 3000                | 18328                                 | 0.904                                                   | 92.02                                       | 1.649                                              | 6.11                                      | 0.227                                          | 245.90                             | -                                | -                           |
| 3200                | 18227                                 | 0.976                                                   | 91.18                                       | 1.837                                              | 6.33                                      | 0.243                                          | 226.97                             | -                                | -                           |
| 3400                | 18118                                 | -                                                       | 90.50                                       | 2.040                                              | 6.56                                      | 0.261                                          | 207.01                             | -                                | -                           |
| 3600                | 18002                                 | -                                                       | 89.93                                       | 2.259                                              | 6.81                                      | 0.282                                          | 186.00                             | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

**Revision 0: 08-05-2020**

**References**

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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.2 Tungsten and Its Alloys

3.2.1 Tungsten (W)

**Revision 0: 08-05-2020**

**References**

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### **3 Refractory Metals and Alloys**

#### **3.3 Rhenium and Rhenium Alloys**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2.1: 08-25-2023

General

### Room Temperature Properties

|                                               |        |
|-----------------------------------------------|--------|
| Molar Mass, [g/mol]                           | 186.2  |
| Theoretical Density, [kg/m <sup>3</sup> ]     | 21,040 |
| Melting Point, [K]                            | 3,453  |
| Specific Heat, [J/(g-K)]                      | 0.138  |
| Thermal Conductivity, [W/(m-K)]               | 46.3   |
| Avg. Linear Expansion Coefficient, [μm/(m-K)] | 5.93   |
| Electrical resistivity, [μΩ-m]                | 0.209  |
| Young's Modulus, [GPa]                        | 460    |
| Shear Modulus, [GPa]                          | 177    |
| Poisson's Ratio, [-]                          | 0.300  |

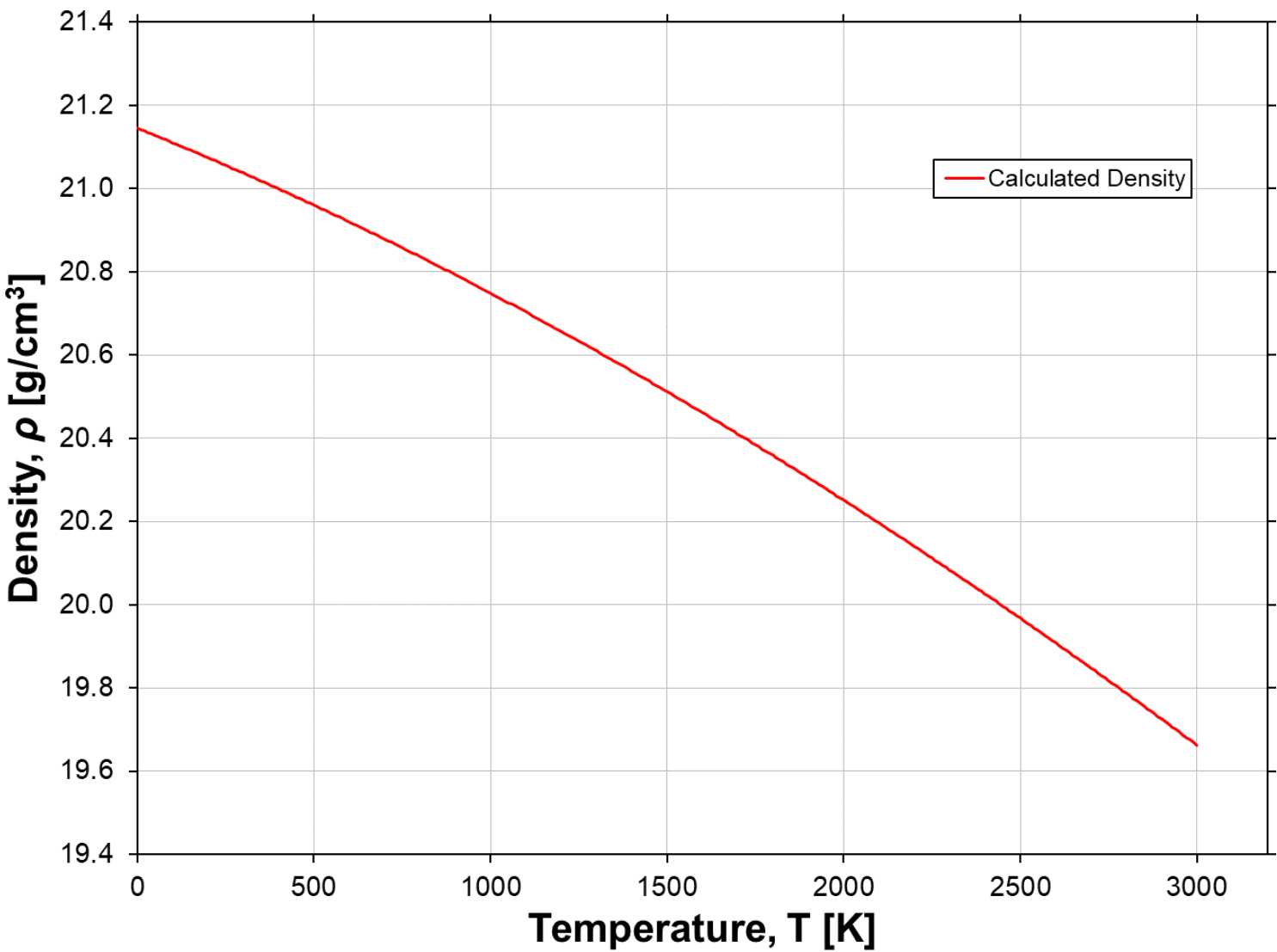


Figure 3.3.1-1: Density versus temperature for rhenium. Calculated from fitted trend of the thermal expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 100               | ( -279.7 ) | 21110             | ( 1317.9 )             | 1600              | ( 2420.3 ) | 20462             | ( 1277.4 )             |
| 200               | ( -99.7 )  | 21074             | ( 1315.7 )             | 1700              | ( 2600.3 ) | 20410             | ( 1274.2 )             |
| 300               | ( 80.3 )   | 21037             | ( 1313.4 )             | 1800              | ( 2780.3 ) | 20358             | ( 1271.0 )             |
| 400               | ( 260.3 )  | 20999             | ( 1311.0 )             | 1900              | ( 2960.3 ) | 20305             | ( 1267.6 )             |
| 500               | ( 440.3 )  | 20960             | ( 1308.5 )             | 2000              | ( 3140.3 ) | 20251             | ( 1264.3 )             |
| 600               | ( 620.3 )  | 20920             | ( 1306.0 )             | 2100              | ( 3320.3 ) | 20196             | ( 1260.8 )             |
| 700               | ( 800.3 )  | 20879             | ( 1303.5 )             | 2200              | ( 3500.3 ) | 20140             | ( 1257.3 )             |
| 800               | ( 980.3 )  | 20836             | ( 1300.8 )             | 2300              | ( 3680.3 ) | 20083             | ( 1253.8 )             |
| 900               | ( 1160.3 ) | 20793             | ( 1298.1 )             | 2400              | ( 3860.3 ) | 20026             | ( 1250.2 )             |
| 1000              | ( 1340.3 ) | 20749             | ( 1295.3 )             | 2500              | ( 4040.3 ) | 19967             | ( 1246.6 )             |
| 1100              | ( 1520.3 ) | 20703             | ( 1292.5 )             | 2600              | ( 4220.3 ) | 19908             | ( 1242.8 )             |
| 1200              | ( 1700.3 ) | 20657             | ( 1289.6 )             | 2700              | ( 4400.3 ) | 19848             | ( 1239.1 )             |
| 1300              | ( 1880.3 ) | 20610             | ( 1286.7 )             | 2800              | ( 4580.3 ) | 19787             | ( 1235.3 )             |
| 1400              | ( 2060.3 ) | 20561             | ( 1283.6 )             | 2900              | ( 4760.3 ) | 19725             | ( 1231.4 )             |
| 1500              | ( 2240.3 ) | 20512             | ( 1280.6 )             | 3000              | ( 4940.3 ) | 19662             | ( 1227.5 )             |

**Application Notes:** Density is calculated as a function of thermal expansion, as seen in the equation below, to approximate property trend with respect to temperature.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 21,040 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $0 \leq T \leq 3000$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Thermal Conductivity with Temperature

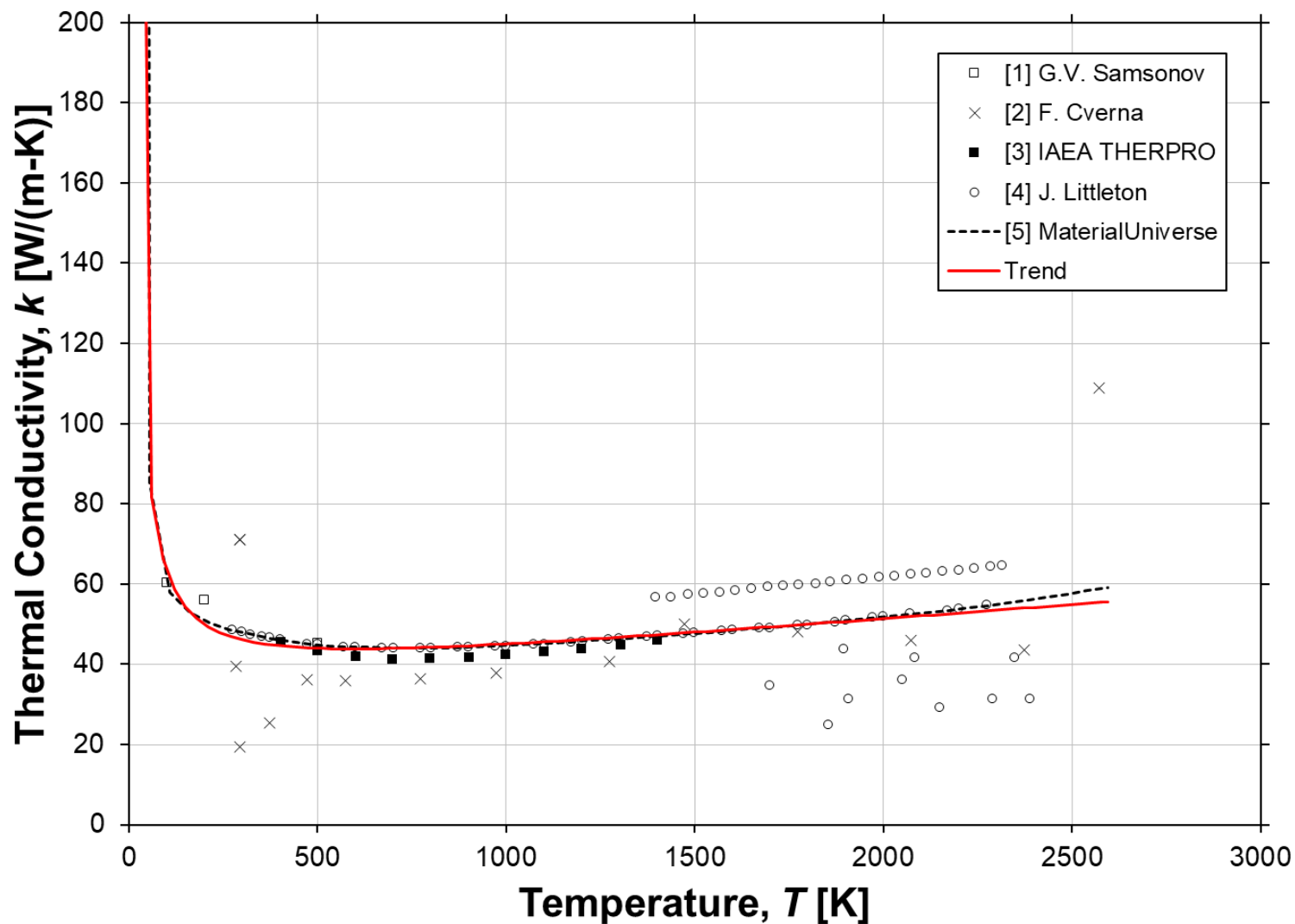


Figure 3.3.1-2: Thermal conductivity versus temperature for rhenium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m-K)                    | ((Btu-in)/(ft <sup>2</sup> -hr-°F)) | K                 | ( °F )     | W/(m-K)                    | ((Btu-in)/(ft <sup>2</sup> -hr-°F)) |
| 3                 | ( -454.3 ) | 883.92                     | ( 6132.76 )                         | 1200              | ( 1700.3 ) | 46.18                      | ( 320.42 )                          |
| 100               | ( -279.7 ) | 63.17                      | ( 438.27 )                          | 1300              | ( 1880.3 ) | 46.77                      | ( 324.50 )                          |
| 200               | ( -99.7 )  | 49.99                      | ( 346.83 )                          | 1400              | ( 2060.3 ) | 47.38                      | ( 328.76 )                          |
| 300               | ( 80.3 )   | 46.15                      | ( 320.18 )                          | 1500              | ( 2240.3 ) | 48.02                      | ( 333.15 )                          |
| 400               | ( 260.3 )  | 44.62                      | ( 309.60 )                          | 1600              | ( 2420.3 ) | 48.67                      | ( 337.66 )                          |
| 500               | ( 440.3 )  | 44.02                      | ( 305.40 )                          | 1700              | ( 2600.3 ) | 49.33                      | ( 342.26 )                          |
| 600               | ( 620.3 )  | 43.87                      | ( 304.38 )                          | 1800              | ( 2780.3 ) | 50.00                      | ( 346.93 )                          |
| 700               | ( 800.3 )  | 43.99                      | ( 305.17 )                          | 1900              | ( 2960.3 ) | 50.69                      | ( 351.67 )                          |
| 800               | ( 980.3 )  | 44.26                      | ( 307.10 )                          | 2000              | ( 3140.3 ) | 51.38                      | ( 356.46 )                          |
| 900               | ( 1160.3 ) | 44.65                      | ( 309.77 )                          | 2200              | ( 3500.3 ) | 52.78                      | ( 366.18 )                          |
| 1000              | ( 1340.3 ) | 45.11                      | ( 312.97 )                          | 2400              | ( 3860.3 ) | 54.20                      | ( 376.04 )                          |
| 1100              | ( 1520.3 ) | 45.63                      | ( 316.55 )                          | 2594              | ( 4209.5 ) | 55.59                      | ( 385.71 )                          |

**Application Notes:** Data for thermal conductivity is collected from references [1-5] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $3 \leq T < 55$

$$k(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^{N_A} \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $55 \leq T \leq 2594$

$$k(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 / \left[ \left( \frac{T}{1000} \right)^{N_B} \right]$$

$k(T)$  = Thermal Conductivity [W/(m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $3 \leq T < 55$        $55 \leq T \leq 2594$

A0 = 1323      B0 = 34.85

A1 = -115.8      B1 = 7.609

A2 = 5548      B\_2 = 2.65

N\_A = 0.1302      N\_B = 1.017



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Thermal Expansion with Temperature

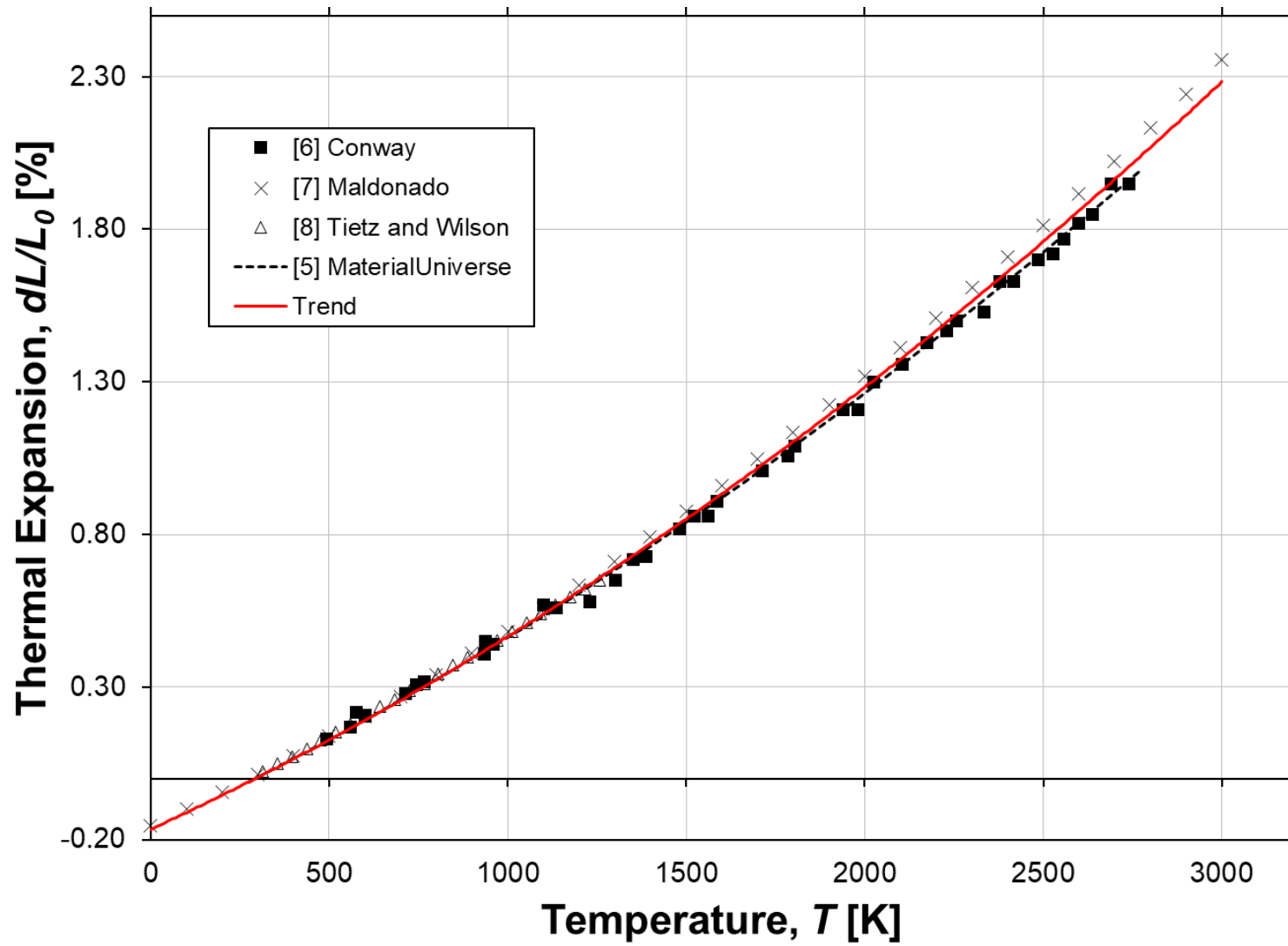


Figure 3.3.1-3: Thermal expansion versus temperature for rhenium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 100               | ( -279.7 ) | -0.111                                  | 1600              | ( 2420.3 ) | 0.933                                   |
| 200               | ( -99.7 )  | -0.054                                  | 1700              | ( 2600.3 ) | 1.018                                   |
| 300               | ( 80.3 )   | 0.004                                   | 1800              | ( 2780.3 ) | 1.104                                   |
| 400               | ( 260.3 )  | 0.065                                   | 1900              | ( 2960.3 ) | 1.192                                   |
| 500               | ( 440.3 )  | 0.127                                   | 2000              | ( 3140.3 ) | 1.282                                   |
| 600               | ( 620.3 )  | 0.191                                   | 2100              | ( 3320.3 ) | 1.374                                   |
| 700               | ( 800.3 )  | 0.257                                   | 2200              | ( 3500.3 ) | 1.468                                   |
| 800               | ( 980.3 )  | 0.325                                   | 2300              | ( 3680.3 ) | 1.563                                   |
| 900               | ( 1160.3 ) | 0.394                                   | 2400              | ( 3860.3 ) | 1.660                                   |
| 1000              | ( 1340.3 ) | 0.466                                   | 2500              | ( 4040.3 ) | 1.760                                   |
| 1100              | ( 1520.3 ) | 0.539                                   | 2600              | ( 4220.3 ) | 1.861                                   |
| 1200              | ( 1700.3 ) | 0.614                                   | 2700              | ( 4400.3 ) | 1.964                                   |
| 1300              | ( 1880.3 ) | 0.691                                   | 2800              | ( 4580.3 ) | 2.068                                   |
| 1400              | ( 2060.3 ) | 0.770                                   | 2900              | ( 4760.3 ) | 2.175                                   |
| 1500              | ( 2240.3 ) | 0.851                                   | 3000              | ( 4940.3 ) | 2.283                                   |

**Application Notes:** Data for thermal expansion is collected from references [5-8] and fitted with the equation below to approximate property trend with respect to temperature. Due to the limited data at low and extremely high temperatures, the trend provided is only recommended for use between 300 and 2700 K.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$$dL/L_0(T) = \text{Thermal Expansion} [\%]$$

$$T = \text{Temperature} [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 0 \leq T \leq 3000 \quad (\text{recommended: } 300 \leq T \leq 2700)$$

$$A0 = -0.1659$$

$$A1 = 0.5392$$

$$A2 = 0.0924$$

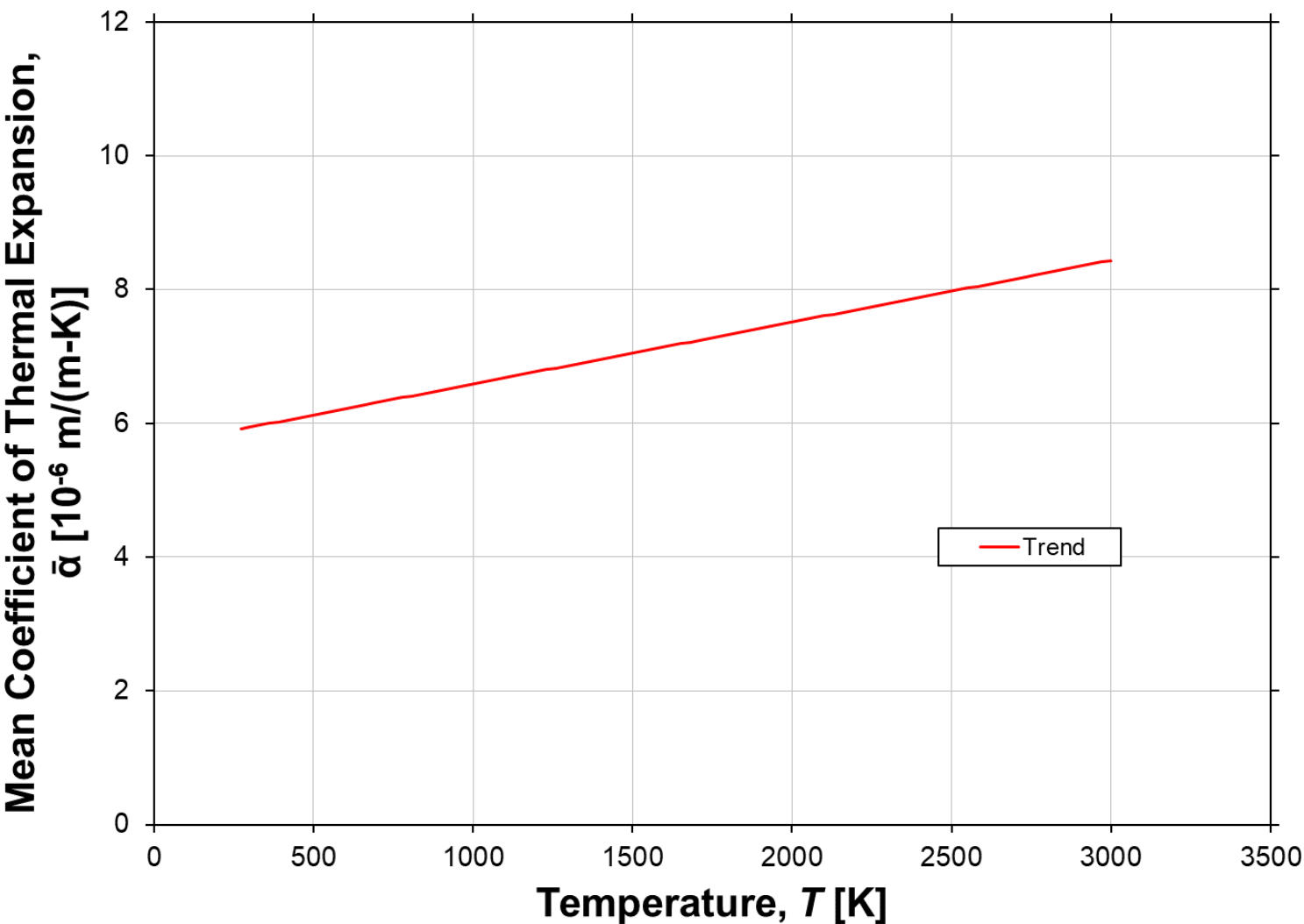


Figure 3.3.1-4: Mean coefficient of thermal expansion versus temperature for rhenium, as calculated from fitted thermal expansion trend.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 273               | ( 31.7 )   | 5.915                                 | ( 3.286 )                                           | 1600              | ( 2420.3 ) | 7.142                                 | ( 3.968 )                                           |
| 300               | ( 80.3 )   | 5.940                                 | ( 3.300 )                                           | 1700              | ( 2600.3 ) | 7.234                                 | ( 4.019 )                                           |
| 350               | ( 170.3 )  | 5.986                                 | ( 3.326 )                                           | 1800              | ( 2780.3 ) | 7.326                                 | ( 4.070 )                                           |
| 400               | ( 260.3 )  | 6.033                                 | ( 3.351 )                                           | 1900              | ( 2960.3 ) | 7.419                                 | ( 4.122 )                                           |
| 500               | ( 440.3 )  | 6.125                                 | ( 3.403 )                                           | 2000              | ( 3140.3 ) | 7.511                                 | ( 4.173 )                                           |
| 600               | ( 620.3 )  | 6.217                                 | ( 3.454 )                                           | 2100              | ( 3320.3 ) | 7.604                                 | ( 4.224 )                                           |
| 700               | ( 800.3 )  | 6.310                                 | ( 3.505 )                                           | 2200              | ( 3500.3 ) | 7.696                                 | ( 4.276 )                                           |
| 800               | ( 980.3 )  | 6.402                                 | ( 3.557 )                                           | 2300              | ( 3680.3 ) | 7.788                                 | ( 4.327 )                                           |
| 900               | ( 1160.3 ) | 6.495                                 | ( 3.608 )                                           | 2400              | ( 3860.3 ) | 7.881                                 | ( 4.378 )                                           |
| 1000              | ( 1340.3 ) | 6.587                                 | ( 3.660 )                                           | 2500              | ( 4040.3 ) | 7.973                                 | ( 4.430 )                                           |
| 1100              | ( 1520.3 ) | 6.680                                 | ( 3.711 )                                           | 2600              | ( 4220.3 ) | 8.066                                 | ( 4.481 )                                           |
| 1200              | ( 1700.3 ) | 6.772                                 | ( 3.762 )                                           | 2700              | ( 4400.3 ) | 8.158                                 | ( 4.532 )                                           |
| 1300              | ( 1880.3 ) | 6.864                                 | ( 3.814 )                                           | 2800              | ( 4580.3 ) | 8.250                                 | ( 4.584 )                                           |
| 1400              | ( 2060.3 ) | 6.957                                 | ( 3.865 )                                           | 2900              | ( 4760.3 ) | 8.343                                 | ( 4.635 )                                           |
| 1500              | ( 2240.3 ) | 7.049                                 | ( 3.916 )                                           | 3000              | ( 4940.3 ) | 8.435                                 | ( 4.686 )                                           |

**Application Notes:** Trend for mean coefficient of thermal expansion is calculated from thermal expansion and shown in the equation below to approximate property trend with respect to temperature. For reasons discussed in the thermal expansion section, the trend provided is only recommended for use between 300 and 2700 K.

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right)$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T Range [K]: 273 ≤ T ≤ 3000 (recommended: 300 ≤ T ≤ 2700)

A0 = 5.663

A1 = 0.9241



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Specific Heat with Temperature

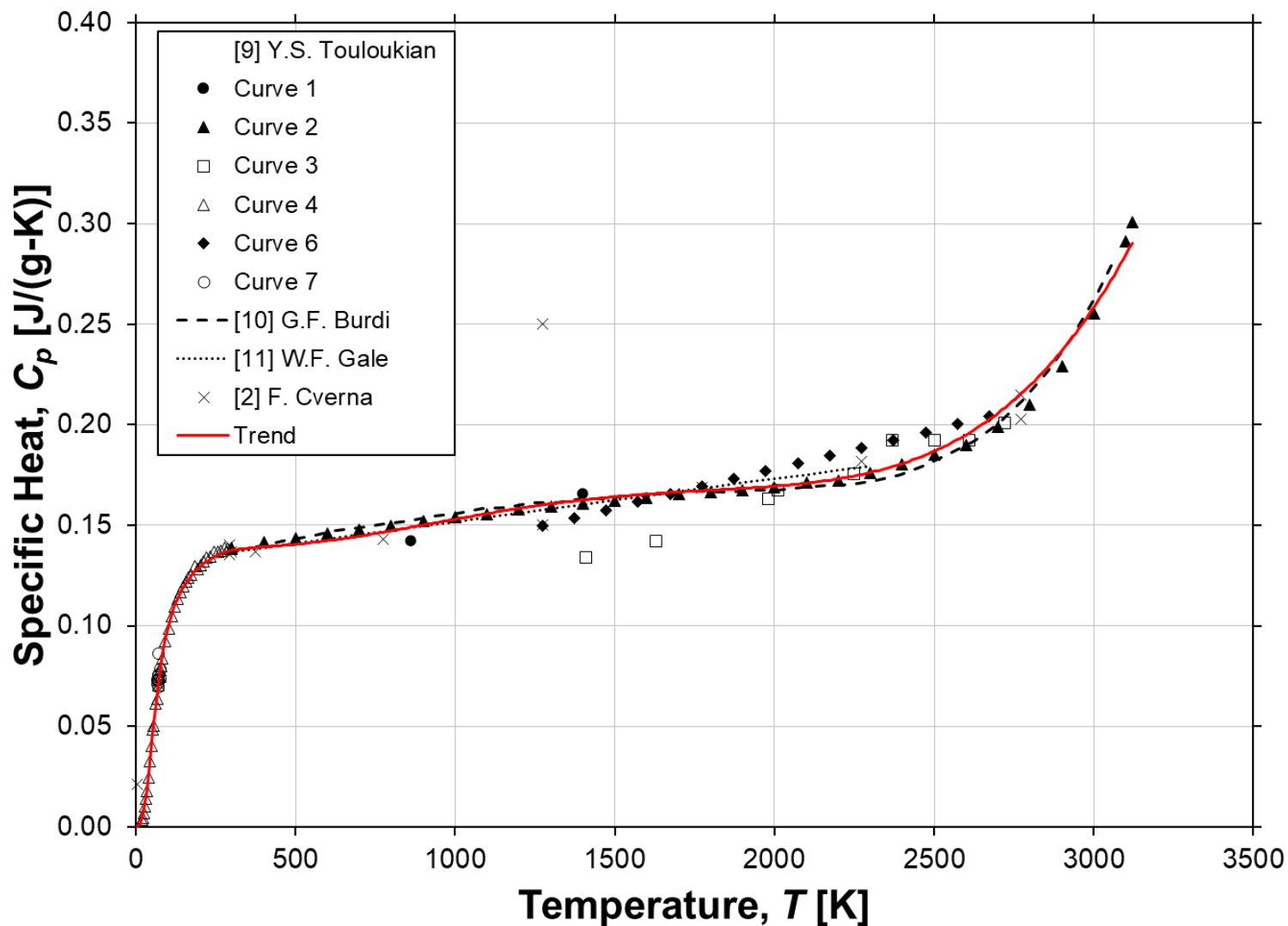


Figure 3.3.1-5: Specific heat versus temperature for rhenium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 3                 | ( -454.3 ) | 0.000                            | ( 0.000 )       | 1500              | ( 2240.3 ) | 0.164                            | ( 0.039 )       |
| 100               | ( -279.7 ) | 0.099                            | ( 0.024 )       | 1600              | ( 2420.3 ) | 0.165                            | ( 0.040 )       |
| 200               | ( -99.7 )  | 0.129                            | ( 0.031 )       | 1700              | ( 2600.3 ) | 0.166                            | ( 0.040 )       |
| 300               | ( 80.3 )   | 0.138                            | ( 0.033 )       | 1800              | ( 2780.3 ) | 0.167                            | ( 0.040 )       |
| 400               | ( 260.3 )  | 0.139                            | ( 0.033 )       | 1900              | ( 2960.3 ) | 0.168                            | ( 0.040 )       |
| 500               | ( 440.3 )  | 0.141                            | ( 0.034 )       | 2000              | ( 3140.3 ) | 0.169                            | ( 0.040 )       |
| 600               | ( 620.3 )  | 0.142                            | ( 0.034 )       | 2100              | ( 3320.3 ) | 0.171                            | ( 0.041 )       |
| 700               | ( 800.3 )  | 0.145                            | ( 0.035 )       | 2200              | ( 3500.3 ) | 0.173                            | ( 0.041 )       |
| 800               | ( 980.3 )  | 0.147                            | ( 0.035 )       | 2300              | ( 3680.3 ) | 0.176                            | ( 0.042 )       |
| 900               | ( 1160.3 ) | 0.150                            | ( 0.036 )       | 2400              | ( 3860.3 ) | 0.181                            | ( 0.043 )       |
| 1000              | ( 1340.3 ) | 0.153                            | ( 0.037 )       | 2500              | ( 4040.3 ) | 0.187                            | ( 0.045 )       |
| 1100              | ( 1520.3 ) | 0.156                            | ( 0.037 )       | 2600              | ( 4220.3 ) | 0.195                            | ( 0.047 )       |
| 1200              | ( 1700.3 ) | 0.158                            | ( 0.038 )       | 2800              | ( 4580.3 ) | 0.219                            | ( 0.052 )       |

**Application Notes:** Data for specific heat is collected from references [2, 9-11] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $3 \leq T < 293$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right]$$

For temperature range:  $293 \leq T \leq 3120$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_3 \cdot \left( \frac{T}{1000} \right)^3 + B_4 \cdot \left( \frac{T}{1000} \right)^4 + B_2 / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

T Range [K]:

$3 \leq T < 293$

$293 \leq T \leq 3120$

N = 2.816

B0 = 0.1626

A0 = 255.2

B1 = -0.101

A1 = -10.71

B2 = 0.1637

A2 = 244.5

B3 = -0.08759

A3 = 1576

B4 = 0.01593

B\_2 = -6.397E-4



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Electrical Resistivity with Temperature

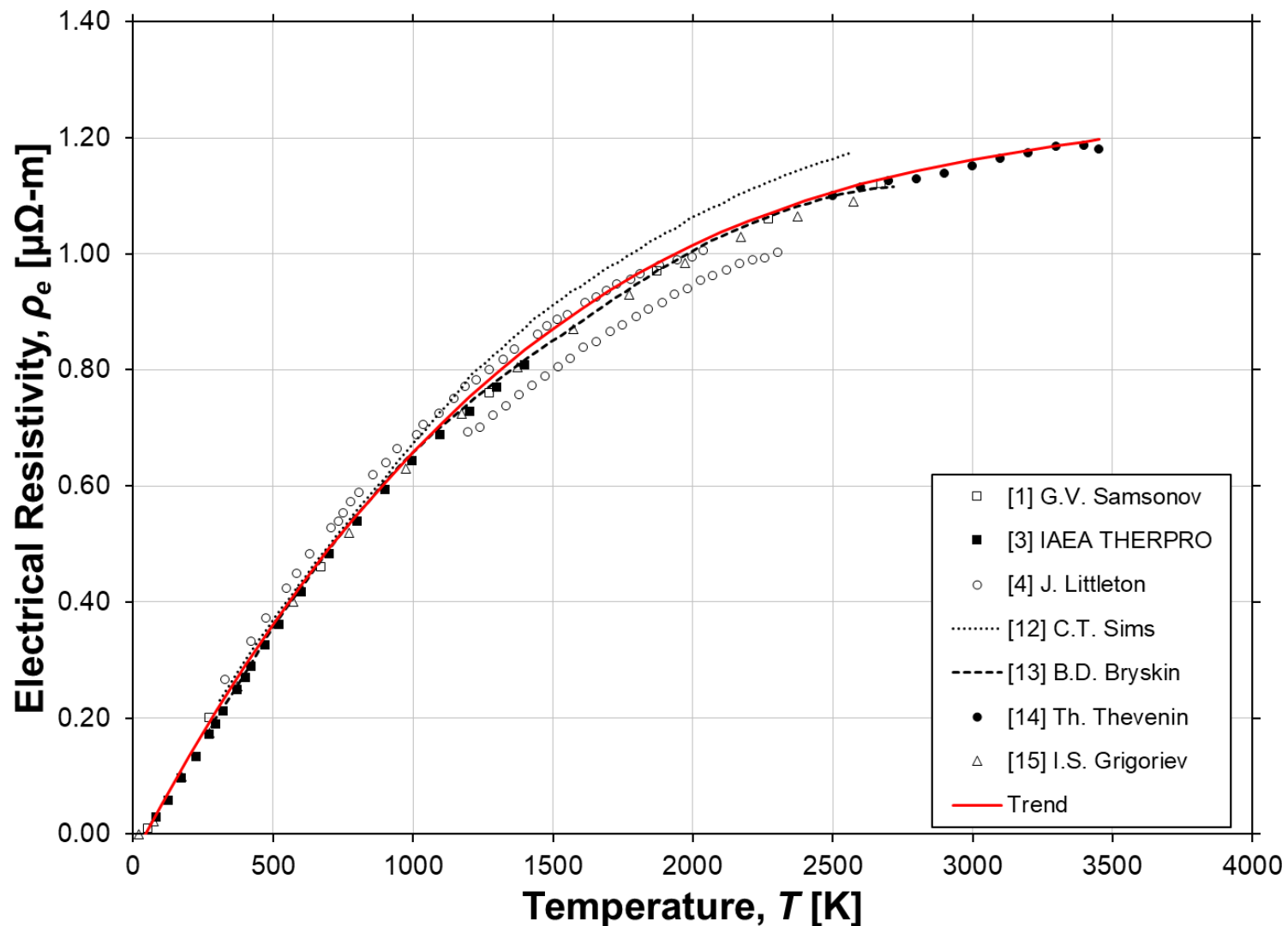


Figure 3.3.1-6: Electrical resistivity versus temperature for rhenium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 100               | ( -279.7 ) | 0.050                               | ( 1.966 )               | 1600              | ( 2420.3 ) | 0.905                               | ( 35.630 )              |
| 200               | ( -99.7 )  | 0.134                               | ( 5.289 )               | 1700              | ( 2600.3 ) | 0.936                               | ( 36.853 )              |
| 300               | ( 80.3 )   | 0.214                               | ( 8.434 )               | 1800              | ( 2780.3 ) | 0.965                               | ( 37.978 )              |
| 400               | ( 260.3 )  | 0.290                               | ( 11.408 )              | 1900              | ( 2960.3 ) | 0.991                               | ( 39.011 )              |
| 500               | ( 440.3 )  | 0.361                               | ( 14.215 )              | 2000              | ( 3140.3 ) | 1.015                               | ( 39.956 )              |
| 600               | ( 620.3 )  | 0.428                               | ( 16.862 )              | 2100              | ( 3320.3 ) | 1.037                               | ( 40.820 )              |
| 700               | ( 800.3 )  | 0.492                               | ( 19.353 )              | 2200              | ( 3500.3 ) | 1.057                               | ( 41.607 )              |
| 800               | ( 980.3 )  | 0.551                               | ( 21.693 )              | 2300              | ( 3680.3 ) | 1.075                               | ( 42.322 )              |
| 900               | ( 1160.3 ) | 0.607                               | ( 23.888 )              | 2400              | ( 3860.3 ) | 1.091                               | ( 42.972 )              |
| 1000              | ( 1340.3 ) | 0.659                               | ( 25.943 )              | 2600              | ( 4220.3 ) | 1.120                               | ( 44.093 )              |
| 1100              | ( 1520.3 ) | 0.708                               | ( 27.863 )              | 2800              | ( 4580.3 ) | 1.143                               | ( 45.013 )              |
| 1200              | ( 1700.3 ) | 0.753                               | ( 29.654 )              | 3000              | ( 4940.3 ) | 1.163                               | ( 45.774 )              |
| 1300              | ( 1880.3 ) | 0.796                               | ( 31.321 )              | 3200              | ( 5300.3 ) | 1.179                               | ( 46.418 )              |
| 1400              | ( 2060.3 ) | 0.835                               | ( 32.869 )              | 3400              | ( 5660.3 ) | 1.193                               | ( 46.988 )              |
| 1500              | ( 2240.3 ) | 0.871                               | ( 34.303 )              | 3453              | ( 5755.7 ) | 1.197                               | ( 47.132 )              |

**Application Notes:** Data for electrical resistivity is collected from references [1, 3, 4, 12-15] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $20 \leq T < 3453$

A0 = -0.03907

A1 = 0.9137

A2 = -0.238

A3 = 0.02232



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Young's Modulus with Temperature

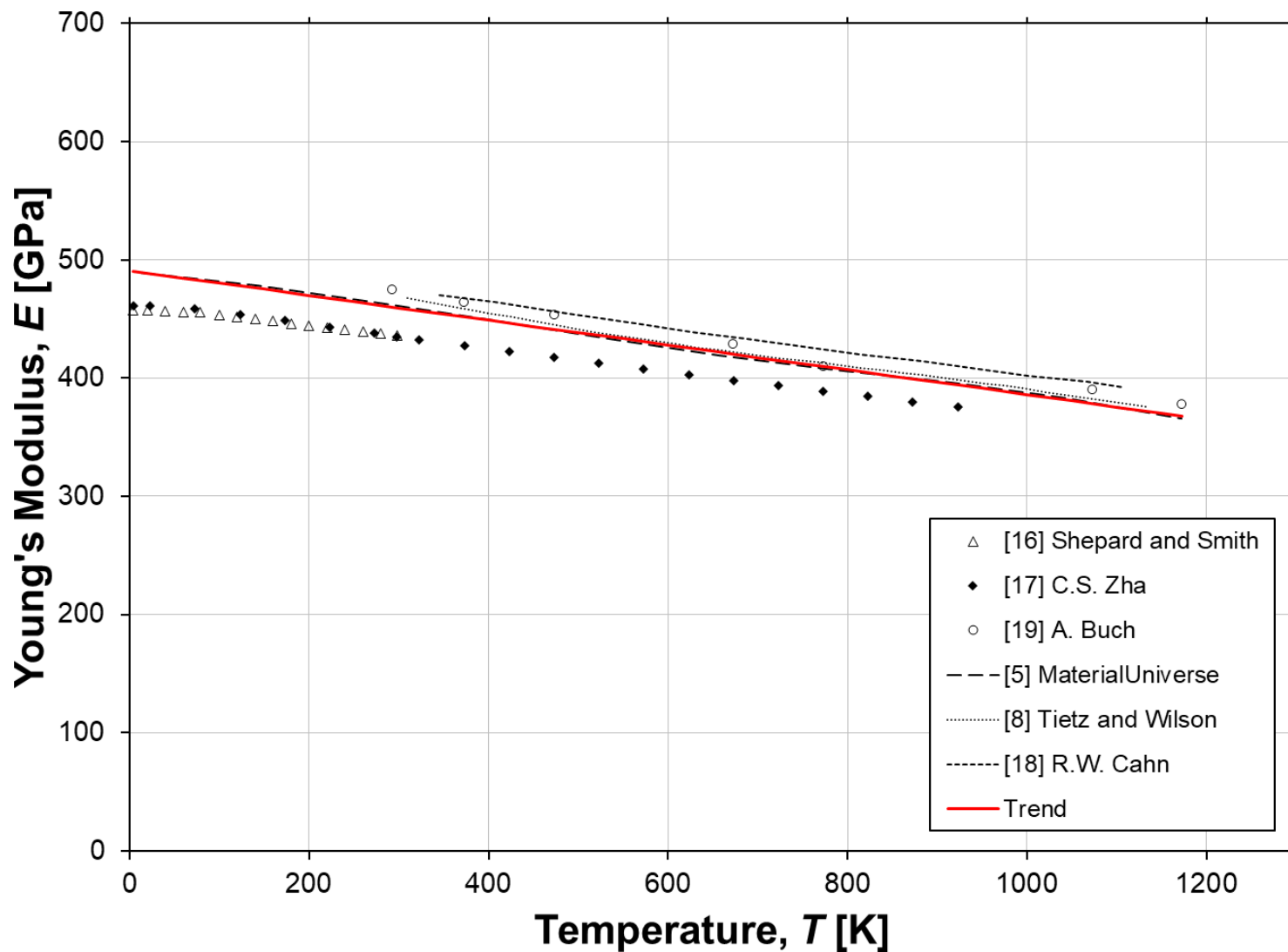


Figure 3.3.1-7: Young's modulus versus temperature for rhenium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 4                 | ( -452.5 ) | 490.48                | ( 71.17 ) | 700               | ( 800.3 )  | 417.47                | ( 60.57 ) |
| 50                | ( -369.7 ) | 485.66                | ( 70.47 ) | 750               | ( 890.3 )  | 412.23                | ( 59.81 ) |
| 100               | ( -279.7 ) | 480.41                | ( 69.71 ) | 800               | ( 980.3 )  | 406.98                | ( 59.05 ) |
| 150               | ( -189.7 ) | 475.17                | ( 68.95 ) | 850               | ( 1070.3 ) | 401.74                | ( 58.29 ) |
| 200               | ( -99.7 )  | 469.92                | ( 68.19 ) | 900               | ( 1160.3 ) | 396.49                | ( 57.53 ) |
| 250               | ( -9.7 )   | 464.68                | ( 67.42 ) | 950               | ( 1250.3 ) | 391.25                | ( 56.77 ) |
| 293               | ( 67.7 )   | 460.16                | ( 66.77 ) | 1000              | ( 1340.3 ) | 386.00                | ( 56.01 ) |
| 300               | ( 80.3 )   | 459.43                | ( 66.66 ) | 1050              | ( 1430.3 ) | 380.76                | ( 55.25 ) |
| 350               | ( 170.3 )  | 454.19                | ( 65.90 ) | 1100              | ( 1520.3 ) | 375.51                | ( 54.49 ) |
| 400               | ( 260.3 )  | 448.94                | ( 65.14 ) | 1150              | ( 1610.3 ) | 370.27                | ( 53.73 ) |
| 450               | ( 350.3 )  | 443.70                | ( 64.38 ) | 1173              | ( 1651.7 ) | 367.85                | ( 53.38 ) |
| 500               | ( 440.3 )  | 438.45                | ( 63.62 ) |                   |            |                       |           |
| 550               | ( 530.3 )  | 433.21                | ( 62.86 ) |                   |            |                       |           |
| 600               | ( 620.3 )  | 427.96                | ( 62.10 ) |                   |            |                       |           |
| 650               | ( 710.3 )  | 422.72                | ( 61.34 ) |                   |            |                       |           |

**Application Notes:** Data for Young's modulus is collected from references [5, 8, 16-19] and fitted with the equation below to approximate the property trend with respect to temperature. Note that the polycrystalline values displayed in the figure above for [16] and [17] were calculated from single crystal data using an average of Voigt and Reuss methods.

## Fit Equations:

$$E(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right)$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $4 \leq T \leq 1173$

A0 = 490.9

A1 = -104.9



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Shear Modulus with Temperature

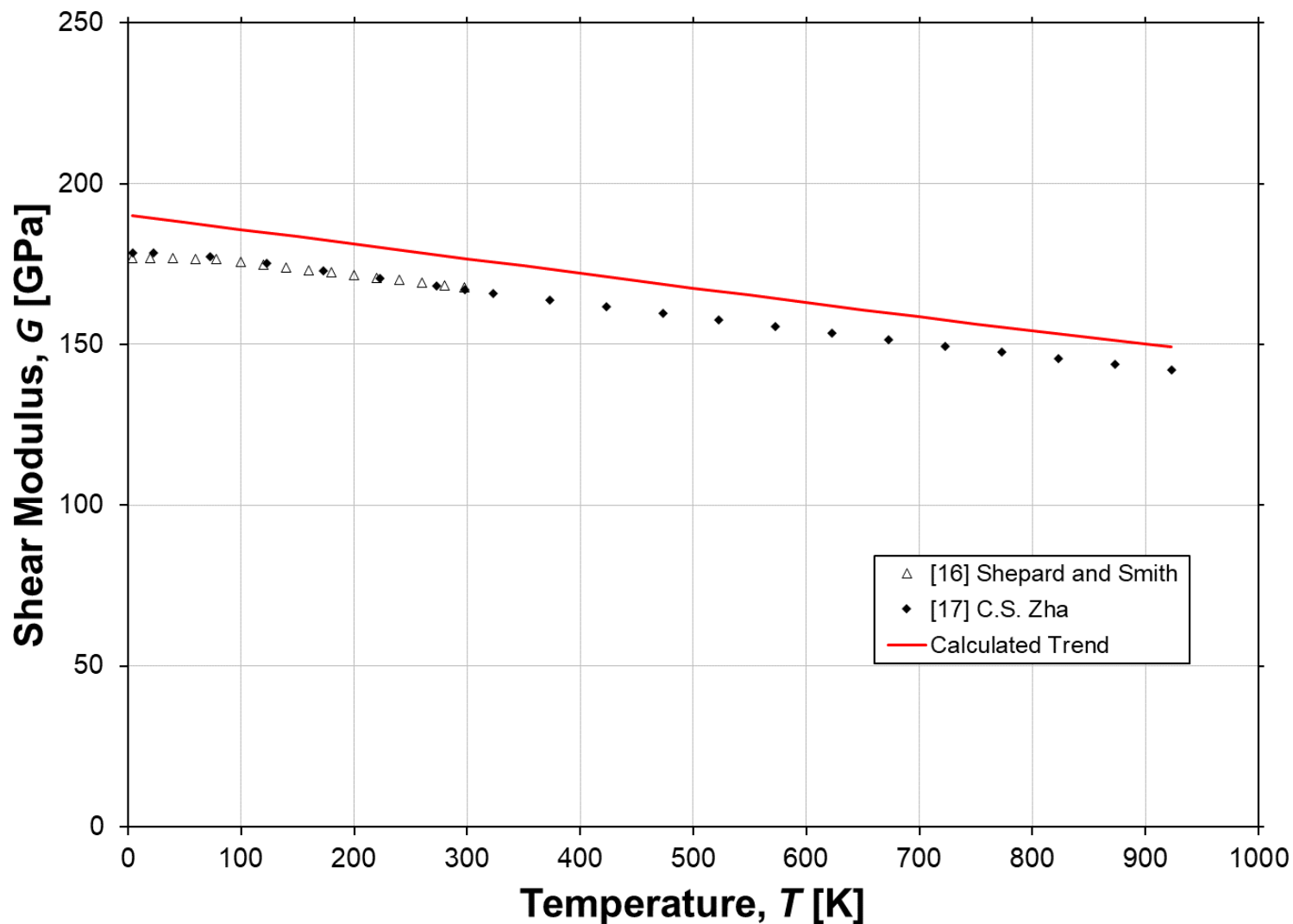


Figure 3.3.1-8: Shear modulus versus temperature for rhenium. Shear modulus is calculated from the Young's modulus and Poisson's ratio trends.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( G ) |           | Temperature ( T ) |            | Shear Modulus ( G ) |           |
|-------------------|------------|---------------------|-----------|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   | K                 | ( °F )     | GPa                 | ( Msi )   |
| 4                 | ( -452.5 ) | 189.93              | ( 27.56 ) | 420               | ( 296.3 )  | 171.19              | ( 24.84 ) |
| 30                | ( -405.7 ) | 188.79              | ( 27.39 ) | 450               | ( 350.3 )  | 169.82              | ( 24.64 ) |
| 60                | ( -351.7 ) | 187.48              | ( 27.20 ) | 480               | ( 404.3 )  | 168.45              | ( 24.44 ) |
| 90                | ( -297.7 ) | 186.15              | ( 27.01 ) | 510               | ( 458.3 )  | 167.09              | ( 24.24 ) |
| 120               | ( -243.7 ) | 184.81              | ( 26.82 ) | 540               | ( 512.3 )  | 165.73              | ( 24.05 ) |
| 150               | ( -189.7 ) | 183.47              | ( 26.62 ) | 570               | ( 566.3 )  | 164.38              | ( 23.85 ) |
| 180               | ( -135.7 ) | 182.12              | ( 26.43 ) | 600               | ( 620.3 )  | 163.03              | ( 23.66 ) |
| 210               | ( -81.7 )  | 180.77              | ( 26.23 ) | 630               | ( 674.3 )  | 161.69              | ( 23.46 ) |
| 240               | ( -27.7 )  | 179.41              | ( 26.03 ) | 660               | ( 728.3 )  | 160.35              | ( 23.27 ) |
| 270               | ( 26.3 )   | 178.04              | ( 25.83 ) | 700               | ( 800.3 )  | 158.59              | ( 23.01 ) |
| 293               | ( 67.7 )   | 176.99              | ( 25.68 ) | 750               | ( 890.3 )  | 156.41              | ( 22.70 ) |
| 300               | ( 80.3 )   | 176.67              | ( 25.64 ) | 800               | ( 980.3 )  | 154.26              | ( 22.38 ) |
| 330               | ( 134.3 )  | 175.30              | ( 25.44 ) | 850               | ( 1070.3 ) | 152.15              | ( 22.08 ) |
| 360               | ( 188.3 )  | 173.93              | ( 25.24 ) | 900               | ( 1160.3 ) | 150.08              | ( 21.78 ) |
| 390               | ( 242.3 )  | 172.56              | ( 25.04 ) | 923               | ( 1201.7 ) | 149.14              | ( 21.64 ) |

**Application Notes:** Data for shear modulus with respect to temperature is collected from references [16, 17]. The shear modulus trend was first calculated from the Young's modulus and Poisson's ratio trends. An equivalent equation was developed and listed below for readers' convenience. Note that the polycrystalline values displayed in the figure above for [16] and [17] were calculated from single crystal data using an average of Voigt and Reuss methods.

## Fit Equations:

$$G(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $4 \leq T \leq 923$

A0 = 190.1

A1 = -43.38

A2 = -6.272

A3 = 5.625



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Poisson's Ratio with Temperature

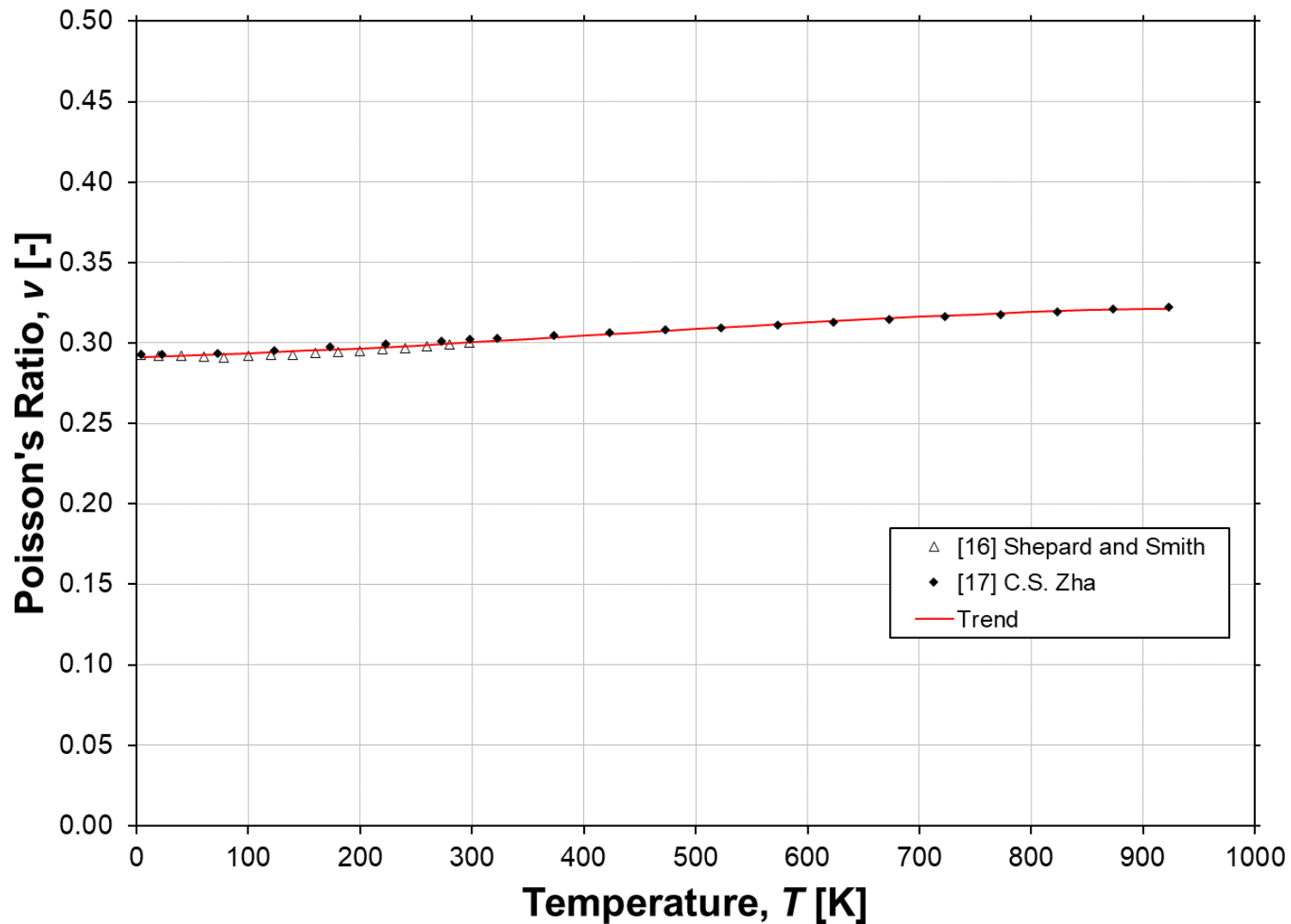


Figure 3.3.1-9: Poisson's ratio versus temperature for rhenium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Poisson's Ratio ( ν ) | Temperature ( T ) |            | Poisson's Ratio ( ν ) |
|-------------------|------------|-----------------------|-------------------|------------|-----------------------|
| K                 | ( °F )     |                       | K                 | ( °F )     |                       |
| 4                 | ( -452.5 ) | 0.291                 | 420               | ( 296.3 )  | 0.305                 |
| 30                | ( -405.7 ) | 0.292                 | 450               | ( 350.3 )  | 0.307                 |
| 60                | ( -351.7 ) | 0.293                 | 480               | ( 404.3 )  | 0.308                 |
| 90                | ( -297.7 ) | 0.293                 | 510               | ( 458.3 )  | 0.309                 |
| 120               | ( -243.7 ) | 0.294                 | 540               | ( 512.3 )  | 0.310                 |
| 150               | ( -189.7 ) | 0.295                 | 570               | ( 566.3 )  | 0.312                 |
| 180               | ( -135.7 ) | 0.296                 | 600               | ( 620.3 )  | 0.313                 |
| 210               | ( -81.7 )  | 0.297                 | 630               | ( 674.3 )  | 0.314                 |
| 240               | ( -27.7 )  | 0.298                 | 660               | ( 728.3 )  | 0.315                 |
| 270               | ( 26.3 )   | 0.299                 | 700               | ( 800.3 )  | 0.316                 |
| 293               | ( 67.7 )   | 0.300                 | 750               | ( 890.3 )  | 0.318                 |
| 300               | ( 80.3 )   | 0.300                 | 800               | ( 980.3 )  | 0.319                 |
| 330               | ( 134.3 )  | 0.302                 | 850               | ( 1070.3 ) | 0.320                 |
| 360               | ( 188.3 )  | 0.303                 | 900               | ( 1160.3 ) | 0.321                 |
| 390               | ( 242.3 )  | 0.304                 | 923               | ( 1201.7 ) | 0.321                 |

**Application Notes:** Data for Poisson's ratio is collected from references [16, 17] and fitted with the equation below to approximate the property trend with respect to temperature. Note that the polycrystalline values displayed in the figure above for [16] and [17] were calculated from single crystal data using an average of Voigt and Reuss methods.

## Fit Equations:

$$\nu(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\nu(T)$  = Poisson's Ratio

$T$  = Temperature [K]

## Constants:

T Range [K]:  $4 \leq T \leq 923$

A0 = 0.2914  
A1 = 0.01628  
A2 = 0.05884  
A3 = -0.04469



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

Revision 2: 04-26-2023

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 21110                                 | 0.050                                                   | 63.17                                       | -0.111                                             | -                                         | 0.099                                          | 480                                | 186                              | 0.294                       |
| 200                 | 21074                                 | 0.134                                                   | 49.99                                       | -0.054                                             | -                                         | 0.129                                          | 470                                | 181                              | 0.297                       |
| 300                 | 21037                                 | 0.214                                                   | 46.15                                       | 0.004                                              | 5.94                                      | 0.138                                          | 459                                | 177                              | 0.300                       |
| 400                 | 20999                                 | 0.290                                                   | 44.62                                       | 0.065                                              | 6.03                                      | 0.139                                          | 449                                | 172                              | 0.304                       |
| 500                 | 20960                                 | 0.361                                                   | 44.02                                       | 0.127                                              | 6.13                                      | 0.141                                          | 438                                | 168                              | 0.309                       |
| 600                 | 20920                                 | 0.428                                                   | 43.87                                       | 0.191                                              | 6.22                                      | 0.142                                          | 428                                | 163                              | 0.313                       |
| 700                 | 20879                                 | 0.492                                                   | 43.99                                       | 0.257                                              | 6.31                                      | 0.145                                          | 417                                | 159                              | 0.316                       |
| 800                 | 20836                                 | 0.551                                                   | 44.26                                       | 0.325                                              | 6.40                                      | 0.147                                          | 407                                | 154                              | 0.319                       |
| 900                 | 20793                                 | 0.607                                                   | 44.65                                       | 0.394                                              | 6.49                                      | 0.150                                          | 396                                | 150                              | 0.321                       |
| 1000                | 20749                                 | 0.659                                                   | 45.11                                       | 0.466                                              | 6.59                                      | 0.153                                          | 386                                | -                                | -                           |
| 1100                | 20703                                 | 0.708                                                   | 45.63                                       | 0.539                                              | 6.68                                      | 0.156                                          | 376                                | -                                | -                           |
| 1200                | 20657                                 | 0.753                                                   | 46.18                                       | 0.614                                              | 6.77                                      | 0.158                                          | -                                  | -                                | -                           |
| 1300                | 20610                                 | 0.796                                                   | 46.77                                       | 0.691                                              | 6.86                                      | 0.161                                          | -                                  | -                                | -                           |
| 1400                | 20561                                 | 0.835                                                   | 47.38                                       | 0.770                                              | 6.96                                      | 0.163                                          | -                                  | -                                | -                           |
| 1500                | 20512                                 | 0.871                                                   | 48.02                                       | 0.851                                              | 7.05                                      | 0.164                                          | -                                  | -                                | -                           |
| 1600                | 20462                                 | 0.905                                                   | 48.67                                       | 0.933                                              | 7.14                                      | 0.165                                          | -                                  | -                                | -                           |
| 1800                | 20358                                 | 0.965                                                   | 50.00                                       | 1.104                                              | 7.33                                      | 0.167                                          | -                                  | -                                | -                           |
| 2000                | 20251                                 | 1.015                                                   | 51.38                                       | 1.282                                              | 7.51                                      | 0.169                                          | -                                  | -                                | -                           |
| 2200                | 20140                                 | 1.057                                                   | 52.78                                       | 1.468                                              | 7.70                                      | 0.173                                          | -                                  | -                                | -                           |
| 2400                | 20026                                 | 1.091                                                   | 54.20                                       | 1.660                                              | 7.88                                      | 0.181                                          | -                                  | -                                | -                           |
| 2600                | 19908                                 | 1.120                                                   | -                                           | 1.861                                              | 8.07                                      | 0.195                                          | -                                  | -                                | -                           |
| 2800                | 19787                                 | 1.143                                                   | -                                           | 2.068                                              | 8.25                                      | 0.219                                          | -                                  | -                                | -                           |
| 3000                | 19662                                 | 1.163                                                   | -                                           | 2.283                                              | 8.44                                      | 0.258                                          | -                                  | -                                | -                           |
| 3200                | -                                     | 1.179                                                   | -                                           | -                                                  | -                                         | -                                              | -                                  | -                                | -                           |
| 3400                | -                                     | 1.193                                                   | -                                           | -                                                  | -                                         | -                                              | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

**Revision 2: 04-26-2023**

**References**

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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

3 Refractory Metals and Alloys

3.3 Rhenium and Rhenium Alloys

3.3.1 Rhenium (Re)

**Revision 2: 04-26-2023**

**References**

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## **4 Other Nonferrous Metals and Alloys**

### **4.1 Copper Alloys**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCop 84

Revision 0: 08-05-2020

General

### Room Temperature Properties

|                                           |       |
|-------------------------------------------|-------|
| Theoretical Density, [kg/m <sup>3</sup> ] | 8,620 |
| Specific Heat, [J/(g-K)]                  | 0.382 |
| Thermal Conductivity, [W/(m-K)]           | 284.6 |
| Linear expansion coefficient, [μm/(m-K)]  | 14.72 |
| Electrical resistivity, [μΩ-m]            | 0.025 |

### Composition

Table 4.1.1-1: Typical Composition ranges for GRCop84 alloy (percent by weight).

| Grade   |      | Cr   | Nb   | O       | Fe     | Other  | Cu      |
|---------|------|------|------|---------|--------|--------|---------|
| GRCop84 | Min. | 6.53 | 5.64 |         |        |        | Balance |
|         | Max. | 6.72 | 5.82 | 600 ppm | 20 ppm | 50 ppm |         |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCo 84

Revision 0: 08-05-2020

Density with Temperature

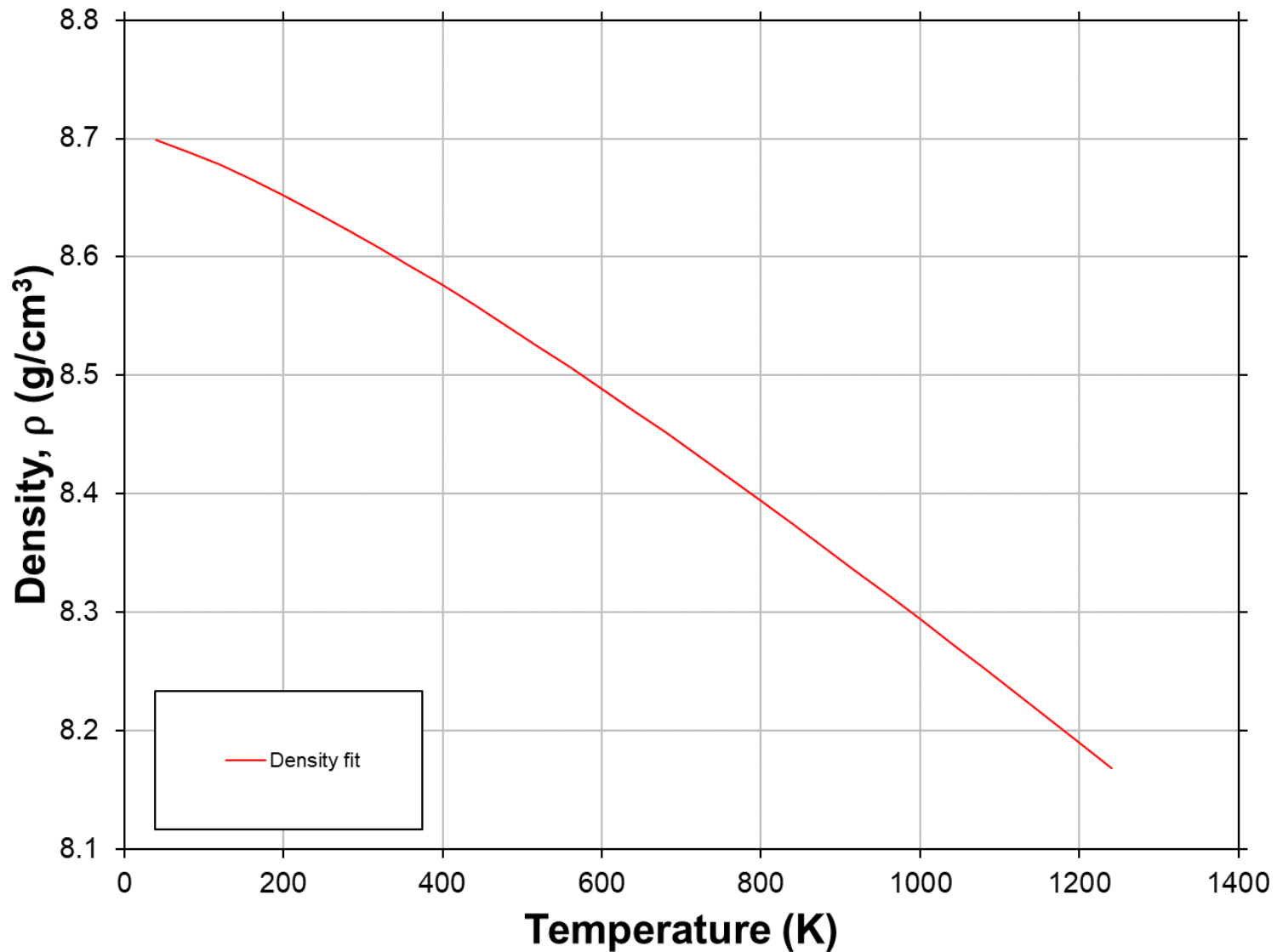


Figure 4.1.1-1: Density versus Temperature for GRCo84. Calculated from fitted trend of Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCo 84

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 70                | ( -333.7 ) | 8692              | ( 542.6 )              | 750               | ( 890.3 )  | 8418              | ( 525.6 )              |
| 100               | ( -279.7 ) | 8684              | ( 542.1 )              | 800               | ( 980.3 )  | 8394              | ( 524.0 )              |
| 150               | ( -189.7 ) | 8669              | ( 541.2 )              | 850               | ( 1070.3 ) | 8370              | ( 522.5 )              |
| 200               | ( -99.7 )  | 8652              | ( 540.1 )              | 900               | ( 1160.3 ) | 8345              | ( 521.0 )              |
| 250               | ( -9.7 )   | 8634              | ( 539.0 )              | 950               | ( 1250.3 ) | 8319              | ( 519.4 )              |
| 293               | ( 67.7 )   | 8618              | ( 538.0 )              | 1000              | ( 1340.3 ) | 8294              | ( 517.8 )              |
| 300               | ( 80.3 )   | 8616              | ( 537.9 )              | 1050              | ( 1430.3 ) | 8268              | ( 516.2 )              |
| 350               | ( 170.3 )  | 8596              | ( 536.7 )              | 1100              | ( 1520.3 ) | 8242              | ( 514.6 )              |
| 400               | ( 260.3 )  | 8576              | ( 535.4 )              | 1150              | ( 1610.3 ) | 8216              | ( 512.9 )              |
| 450               | ( 350.3 )  | 8555              | ( 534.1 )              | 1200              | ( 1700.3 ) | 8190              | ( 511.3 )              |
| 500               | ( 440.3 )  | 8533              | ( 532.7 )              | 1250              | ( 1790.3 ) | 8163              | ( 509.6 )              |
| 550               | ( 530.3 )  | 8511              | ( 531.4 )              | 1273              | ( 1831.7 ) | 8151              | ( 508.9 )              |
| 600               | ( 620.3 )  | 8489              | ( 530.0 )              |                   |            |                   |                        |
| 650               | ( 710.3 )  | 8466              | ( 528.5 )              |                   |            |                   |                        |
| 700               | ( 800.3 )  | 8442              | ( 527.0 )              |                   |            |                   |                        |

**Application Notes:** Density trend with respect to temperature is calculated as a function of thermal expansion as seen in the equation below.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 8,620 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $70 \leq T \leq 1273$

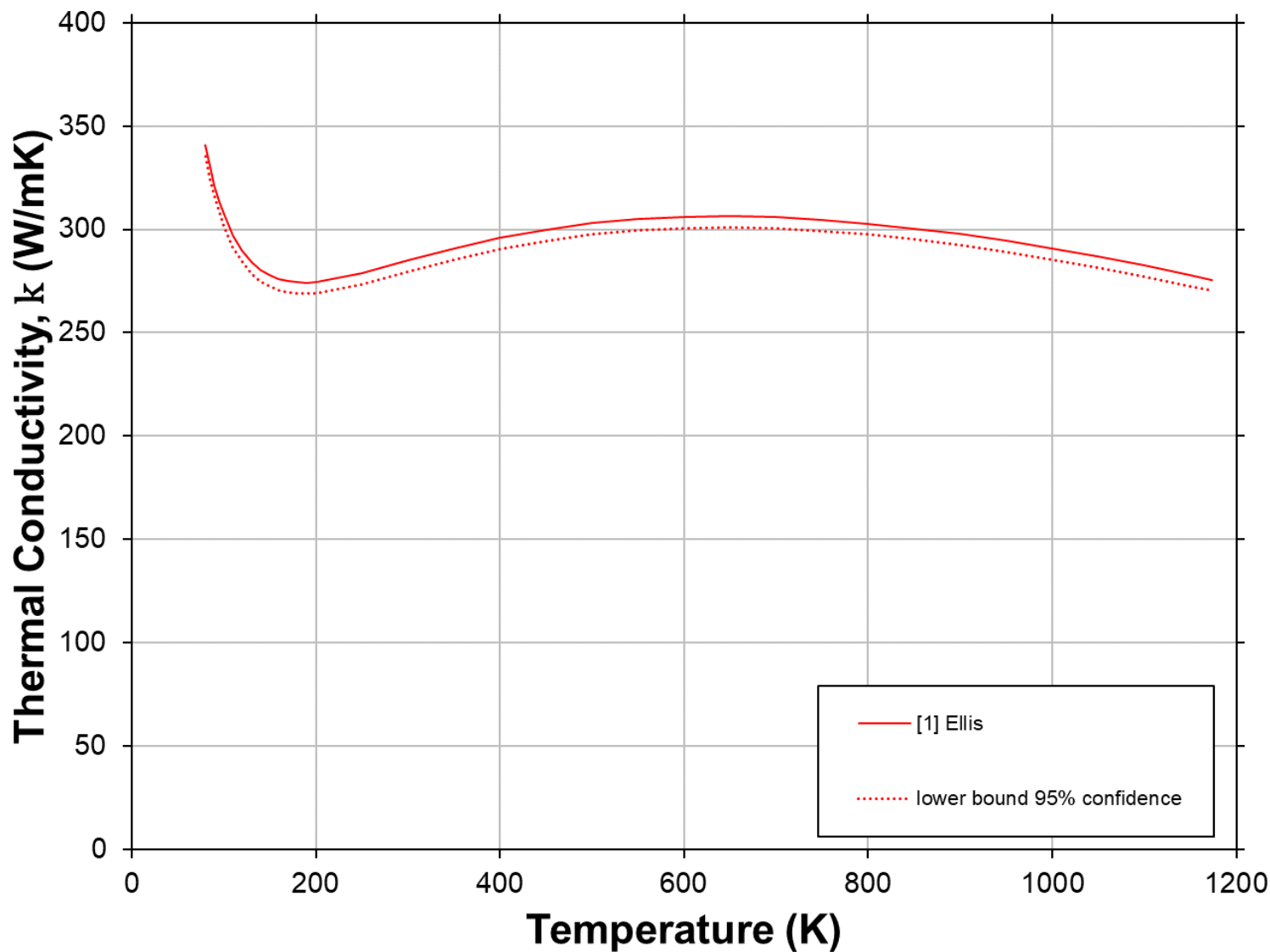


Figure 4.1.1-2: Thermal Conductivity versus Temperature of GRCop84 adapted from Ellis (2000).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCop 84

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 80                | ( -315.7 ) | 340.53                     | ( 2362.67 )                         | 450               | ( 350.3 )  | 299.88                     | ( 2080.61 )                         |
| 100               | ( -279.7 ) | 307.26                     | ( 2131.78 )                         | 475               | ( 395.3 )  | 301.52                     | ( 2091.99 )                         |
| 125               | ( -234.7 ) | 286.65                     | ( 1988.84 )                         | 500               | ( 440.3 )  | 302.90                     | ( 2101.58 )                         |
| 150               | ( -189.7 ) | 277.64                     | ( 1926.32 )                         | 525               | ( 485.3 )  | 304.03                     | ( 2109.42 )                         |
| 175               | ( -144.7 ) | 274.52                     | ( 1904.64 )                         | 550               | ( 530.3 )  | 304.92                     | ( 2115.58 )                         |
| 200               | ( -99.7 )  | 274.53                     | ( 1904.69 )                         | 575               | ( 575.3 )  | 305.58                     | ( 2120.14 )                         |
| 225               | ( -54.7 )  | 276.21                     | ( 1916.38 )                         | 600               | ( 620.3 )  | 306.01                     | ( 2123.16 )                         |
| 250               | ( -9.7 )   | 278.77                     | ( 1934.11 )                         | 625               | ( 665.3 )  | 306.24                     | ( 2124.73 )                         |
| 275               | ( 35.3 )   | 281.72                     | ( 1954.61 )                         | 650               | ( 710.3 )  | 306.27                     | ( 2124.92 )                         |
| 300               | ( 80.3 )   | 284.80                     | ( 1975.96 )                         | 675               | ( 755.3 )  | 306.11                     | ( 2123.80 )                         |
| 325               | ( 125.3 )  | 287.83                     | ( 1996.97 )                         | 700               | ( 800.3 )  | 305.77                     | ( 2121.45 )                         |
| 350               | ( 170.3 )  | 290.71                     | ( 2016.95 )                         | 800               | ( 980.3 )  | 302.80                     | ( 2100.89 )                         |
| 375               | ( 215.3 )  | 293.38                     | ( 2035.49 )                         | 900               | ( 1160.3 ) | 297.66                     | ( 2065.22 )                         |
| 400               | ( 260.3 )  | 295.81                     | ( 2052.34 )                         | 1000              | ( 1340.3 ) | 290.77                     | ( 2017.40 )                         |
| 425               | ( 305.3 )  | 297.98                     | ( 2067.40 )                         | 1173              | ( 1651.7 ) | 275.65                     | ( 1912.51 )                         |

**Application Notes:** Trend for thermal conductivity is collected from reference [1] presented in the equation below to approximate a fit of average property trend with respect to temperature.

## Fit Equation:

$$k(T) = A0 + A1 \cdot \ln(T) + A2 \cdot [\ln(T)]^2 + A3 \cdot [\ln(T)]^3$$

$$k(T) = \text{Thermal Conductivity } [W / (m \cdot K)]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 80 \leq T \leq 1173$$

$$A0 = 6893$$

$$A1 = -3466$$

$$A2 = 599.5$$

$$A3 = -34.18$$



|                                    |                   |                                    |
|------------------------------------|-------------------|------------------------------------|
| 4 Other Nonferrous Metals & Alloys | 4.1 Copper Alloys | 4.1.1 GRCop 84                     |
| Revision 0: 08-05-2020             |                   | Thermal Expansion with Temperature |

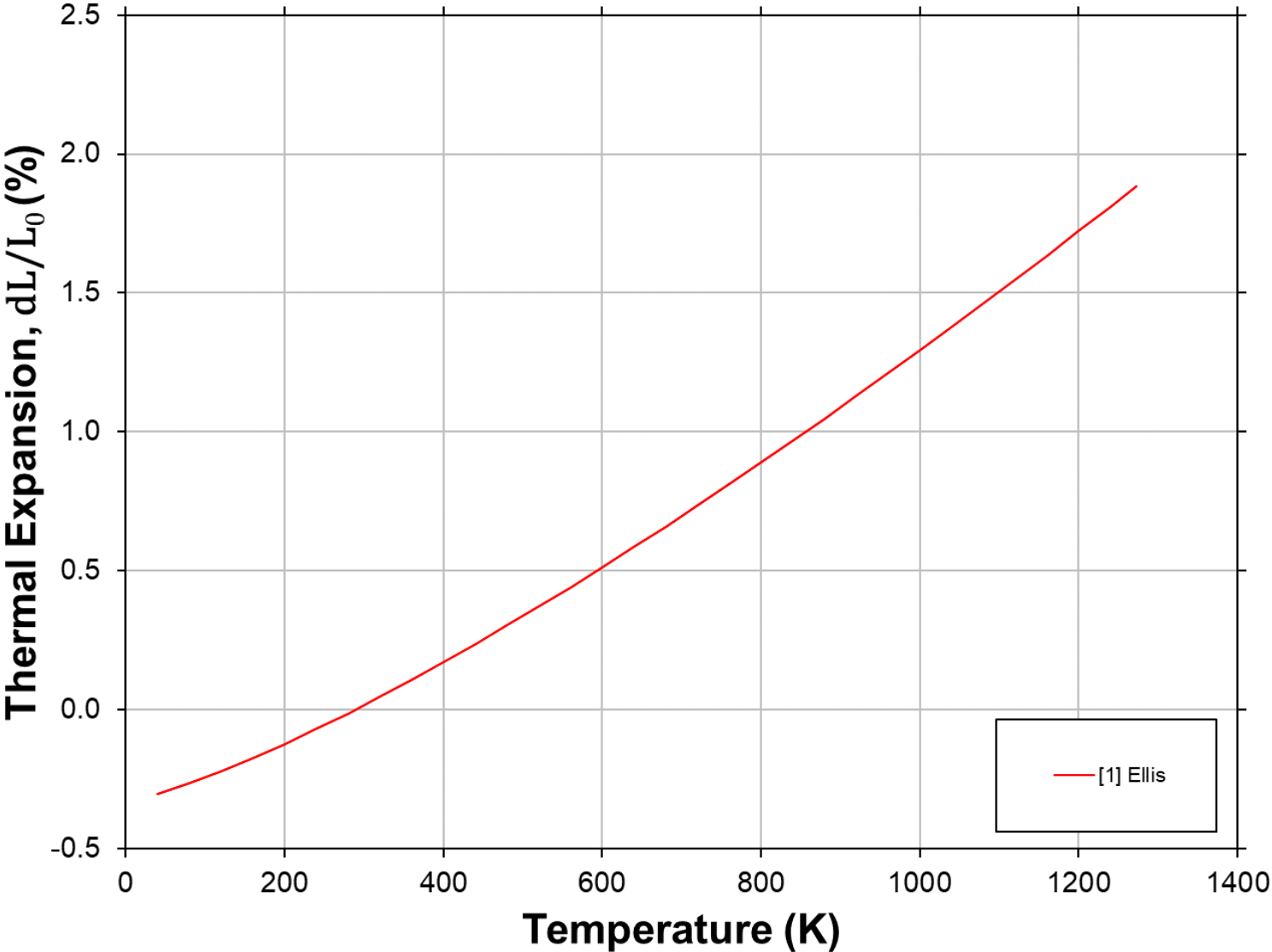


Figure 4.1.1-3: Thermal Expansion versus Temperature of GRCop84 adapted from Ellis (2000).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCo 84

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 70                | ( -333.7 ) | -0.275                                  | 450               | ( 350.3 )  | 0.253                                   |
| 100               | ( -279.7 ) | -0.245                                  | 500               | ( 440.3 )  | 0.337                                   |
| 125               | ( -234.7 ) | -0.217                                  | 550               | ( 530.3 )  | 0.424                                   |
| 150               | ( -189.7 ) | -0.187                                  | 600               | ( 620.3 )  | 0.513                                   |
| 175               | ( -144.7 ) | -0.156                                  | 650               | ( 710.3 )  | 0.604                                   |
| 200               | ( -99.7 )  | -0.124                                  | 700               | ( 800.3 )  | 0.697                                   |
| 225               | ( -54.7 )  | -0.090                                  | 750               | ( 890.3 )  | 0.792                                   |
| 250               | ( -9.7 )   | -0.056                                  | 800               | ( 980.3 )  | 0.889                                   |
| 275               | ( 35.3 )   | -0.020                                  | 850               | ( 1070.3 ) | 0.988                                   |
| 300               | ( 80.3 )   | 0.017                                   | 900               | ( 1160.3 ) | 1.088                                   |
| 325               | ( 125.3 )  | 0.054                                   | 950               | ( 1250.3 ) | 1.190                                   |
| 350               | ( 170.3 )  | 0.092                                   | 1000              | ( 1340.3 ) | 1.293                                   |
| 375               | ( 215.3 )  | 0.131                                   | 1100              | ( 1520.3 ) | 1.505                                   |
| 400               | ( 260.3 )  | 0.171                                   | 1200              | ( 1700.3 ) | 1.722                                   |
| 425               | ( 305.3 )  | 0.211                                   | 1273              | ( 1831.7 ) | 1.883                                   |

**Application Notes:** Trend for thermal expansion is collected from reference [1] and presented in the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot T^N$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 70 \leq T \leq 1273$$

$$A0 = -0.3287$$

$$A1 = 2.265E-04$$

$$N = 1.285$$



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCop 84

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

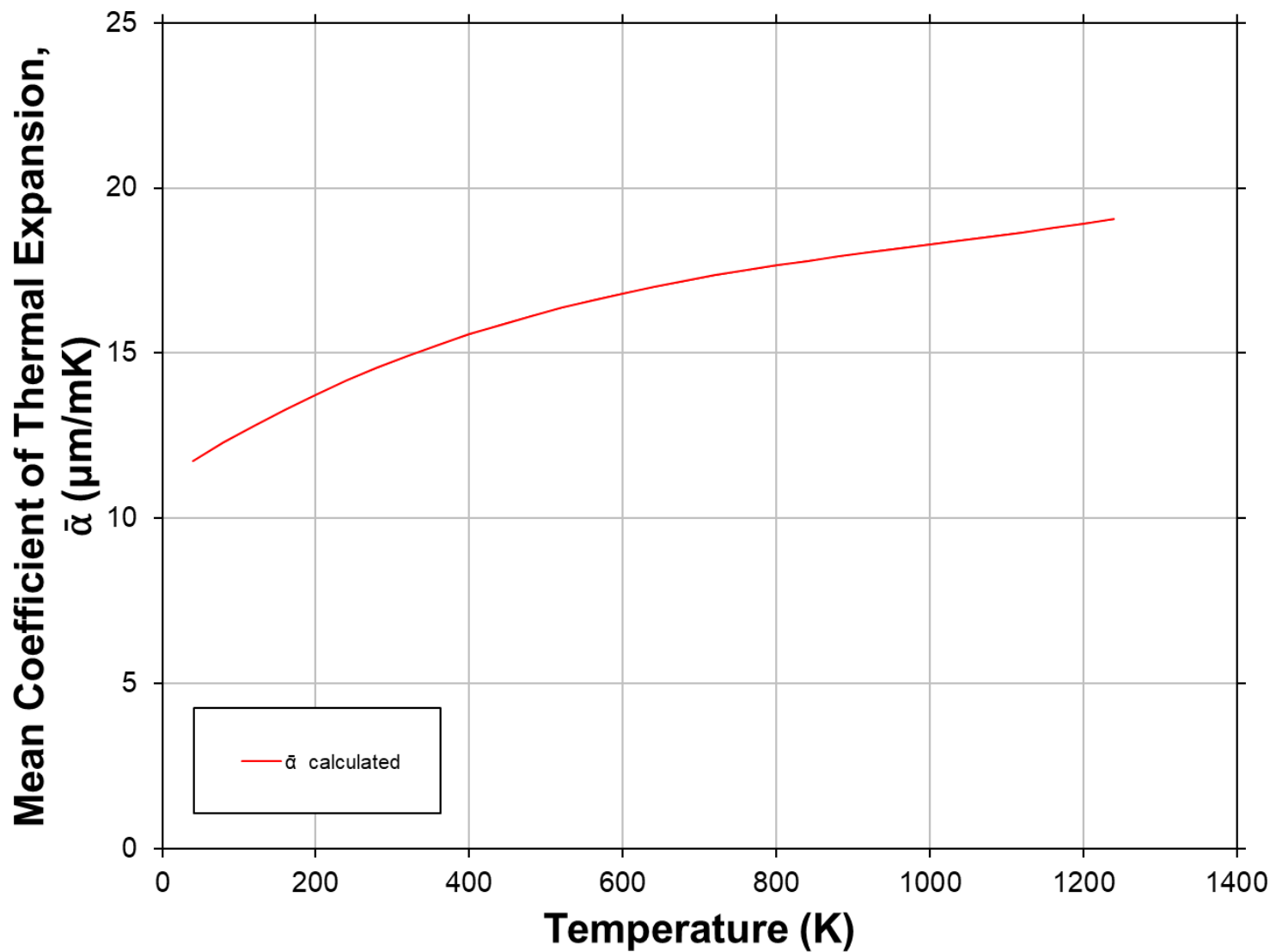


Figure 4.1.1-4: Mean Coefficient of Thermal Expansion versus Temperature of GRCop84. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCop 84

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 70                | ( -333.7 ) | 12.151                                | ( 6.751 )                                           | 450               | ( 350.3 )  | 15.928                                | ( 8.849 )                                           |
| 100               | ( -279.7 ) | 12.550                                | ( 6.972 )                                           | 500               | ( 440.3 )  | 16.252                                | ( 9.029 )                                           |
| 125               | ( -234.7 ) | 12.868                                | ( 7.149 )                                           | 550               | ( 530.3 )  | 16.546                                | ( 9.192 )                                           |
| 150               | ( -189.7 ) | 13.172                                | ( 7.318 )                                           | 600               | ( 620.3 )  | 16.811                                | ( 9.340 )                                           |
| 175               | ( -144.7 ) | 13.464                                | ( 7.480 )                                           | 650               | ( 710.3 )  | 17.052                                | ( 9.474 )                                           |
| 200               | ( -99.7 )  | 13.743                                | ( 7.635 )                                           | 700               | ( 800.3 )  | 17.272                                | ( 9.595 )                                           |
| 225               | ( -54.7 )  | 14.010                                | ( 7.783 )                                           | 750               | ( 890.3 )  | 17.472                                | ( 9.707 )                                           |
| 250               | ( -9.7 )   | 14.265                                | ( 7.925 )                                           | 800               | ( 980.3 )  | 17.657                                | ( 9.809 )                                           |
| 275               | ( 35.3 )   | 14.509                                | ( 8.060 )                                           | 850               | ( 1070.3 ) | 17.828                                | ( 9.905 )                                           |
| 300               | ( 80.3 )   | 14.742                                | ( 8.190 )                                           | 900               | ( 1160.3 ) | 17.990                                | ( 9.995 )                                           |
| 325               | ( 125.3 )  | 14.964                                | ( 8.313 )                                           | 950               | ( 1250.3 ) | 18.145                                | ( 10.081 )                                          |
| 350               | ( 170.3 )  | 15.176                                | ( 8.431 )                                           | 1000              | ( 1340.3 ) | 18.296                                | ( 10.164 )                                          |
| 375               | ( 215.3 )  | 15.378                                | ( 8.543 )                                           | 1100              | ( 1520.3 ) | 18.598                                | ( 10.332 )                                          |
| 400               | ( 260.3 )  | 15.570                                | ( 8.650 )                                           | 1200              | ( 1700.3 ) | 18.919                                | ( 10.511 )                                          |
| 425               | ( 305.3 )  | 15.753                                | ( 8.752 )                                           | 1273              | ( 1831.7 ) | 19.180                                | ( 10.655 )                                          |

**Application Notes:** Trend for mean coefficient of thermal expansion is calculated as a function of thermal expansion and shown in the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]  
 $T$  = Temperature [K]

## Constants:

T. Range [K]:  $70 \leq T \leq 1273$

A0 = 11.14  
A1 = 15.27  
A2 = -12.07  
A3 = 3.956



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

|                                    |                   |                                |
|------------------------------------|-------------------|--------------------------------|
| 4 Other Nonferrous Metals & Alloys | 4.1 Copper Alloys | 4.1.1 GRCop 84                 |
| Revision 0: 08-05-2020             |                   | Specific Heat with Temperature |

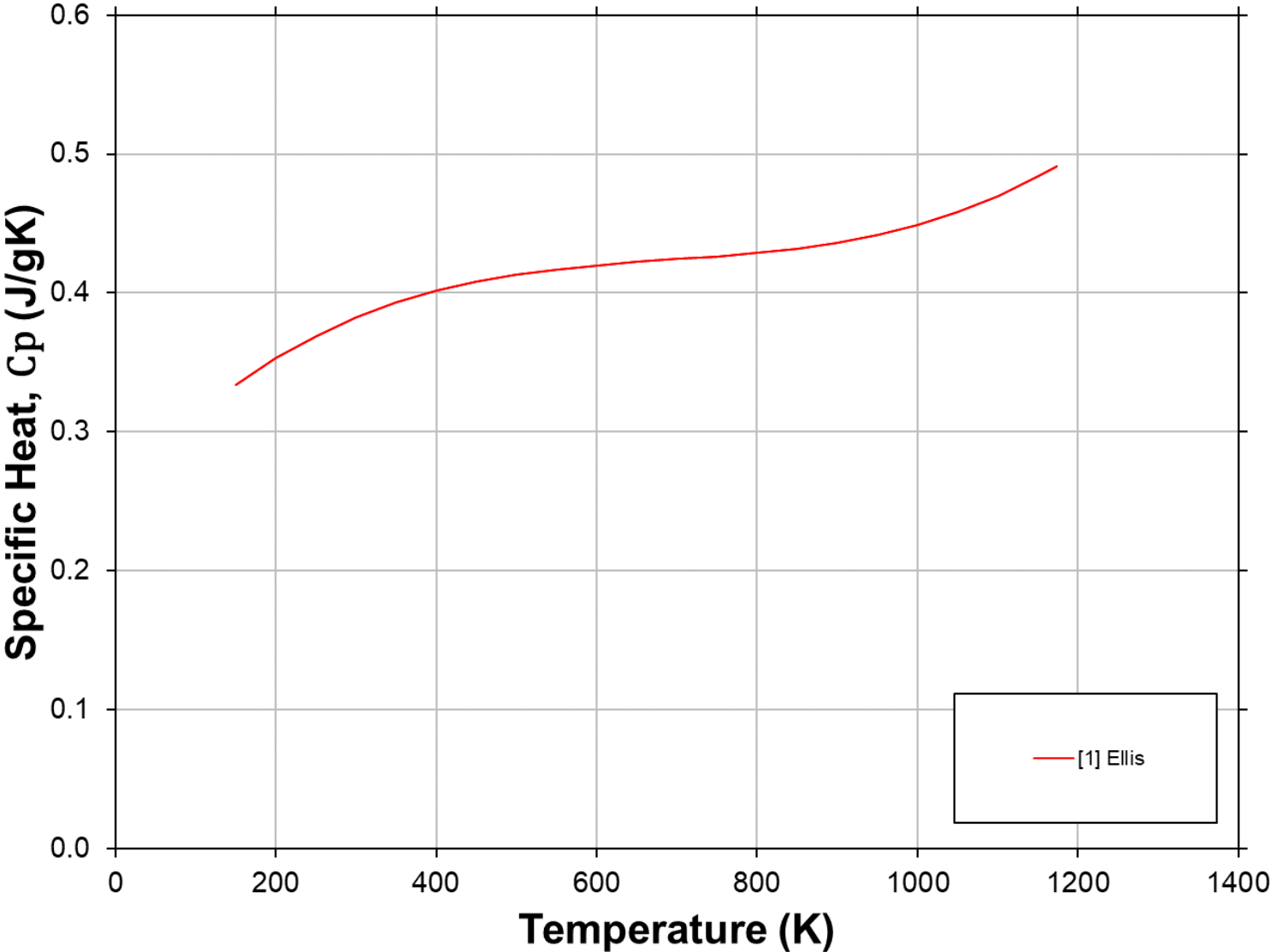


Figure 4.1.1-5: Specific Heat versus Temperature of GRCop84, adapted from Ellis (2000).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCo 84

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|-----------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )    | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 293               | ( 67.7 )  | 0.381                            | ( 0.091 )       | 700               | ( 800.3 )  | 0.424                            | ( 0.101 )       |
| 325               | ( 125.3 ) | 0.388                            | ( 0.093 )       | 725               | ( 845.3 )  | 0.425                            | ( 0.102 )       |
| 350               | ( 170.3 ) | 0.393                            | ( 0.094 )       | 750               | ( 890.3 )  | 0.426                            | ( 0.102 )       |
| 375               | ( 215.3 ) | 0.397                            | ( 0.095 )       | 775               | ( 935.3 )  | 0.428                            | ( 0.102 )       |
| 400               | ( 260.3 ) | 0.401                            | ( 0.096 )       | 800               | ( 980.3 )  | 0.429                            | ( 0.103 )       |
| 425               | ( 305.3 ) | 0.405                            | ( 0.097 )       | 825               | ( 1025.3 ) | 0.430                            | ( 0.103 )       |
| 450               | ( 350.3 ) | 0.408                            | ( 0.098 )       | 850               | ( 1070.3 ) | 0.432                            | ( 0.103 )       |
| 475               | ( 395.3 ) | 0.411                            | ( 0.098 )       | 875               | ( 1115.3 ) | 0.434                            | ( 0.104 )       |
| 500               | ( 440.3 ) | 0.413                            | ( 0.099 )       | 900               | ( 1160.3 ) | 0.436                            | ( 0.104 )       |
| 525               | ( 485.3 ) | 0.415                            | ( 0.099 )       | 925               | ( 1205.3 ) | 0.439                            | ( 0.105 )       |
| 550               | ( 530.3 ) | 0.417                            | ( 0.100 )       | 950               | ( 1250.3 ) | 0.442                            | ( 0.106 )       |
| 575               | ( 575.3 ) | 0.419                            | ( 0.100 )       | 975               | ( 1295.3 ) | 0.445                            | ( 0.106 )       |
| 600               | ( 620.3 ) | 0.420                            | ( 0.100 )       | 1000              | ( 1340.3 ) | 0.449                            | ( 0.107 )       |
| 625               | ( 665.3 ) | 0.421                            | ( 0.101 )       | 1100              | ( 1520.3 ) | 0.470                            | ( 0.112 )       |
| 650               | ( 710.3 ) | 0.422                            | ( 0.101 )       | 1173              | ( 1651.7 ) | 0.491                            | ( 0.117 )       |

**Application Notes:** Trend for specific heat is collected from reference [1] and presented here in the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$C_p(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

T. Range [K]: 293 ≤ T ≤ 1173

A0 = 0.2539

A1 = 0.6563

A2 = -0.8903

A3 = 0.4292



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

|                                    |                   |                                         |
|------------------------------------|-------------------|-----------------------------------------|
| 4 Other Nonferrous Metals & Alloys | 4.1 Copper Alloys | 4.1.1 GRCop 84                          |
| Revision 0: 08-05-2020             |                   | Electrical Resistivity with Temperature |

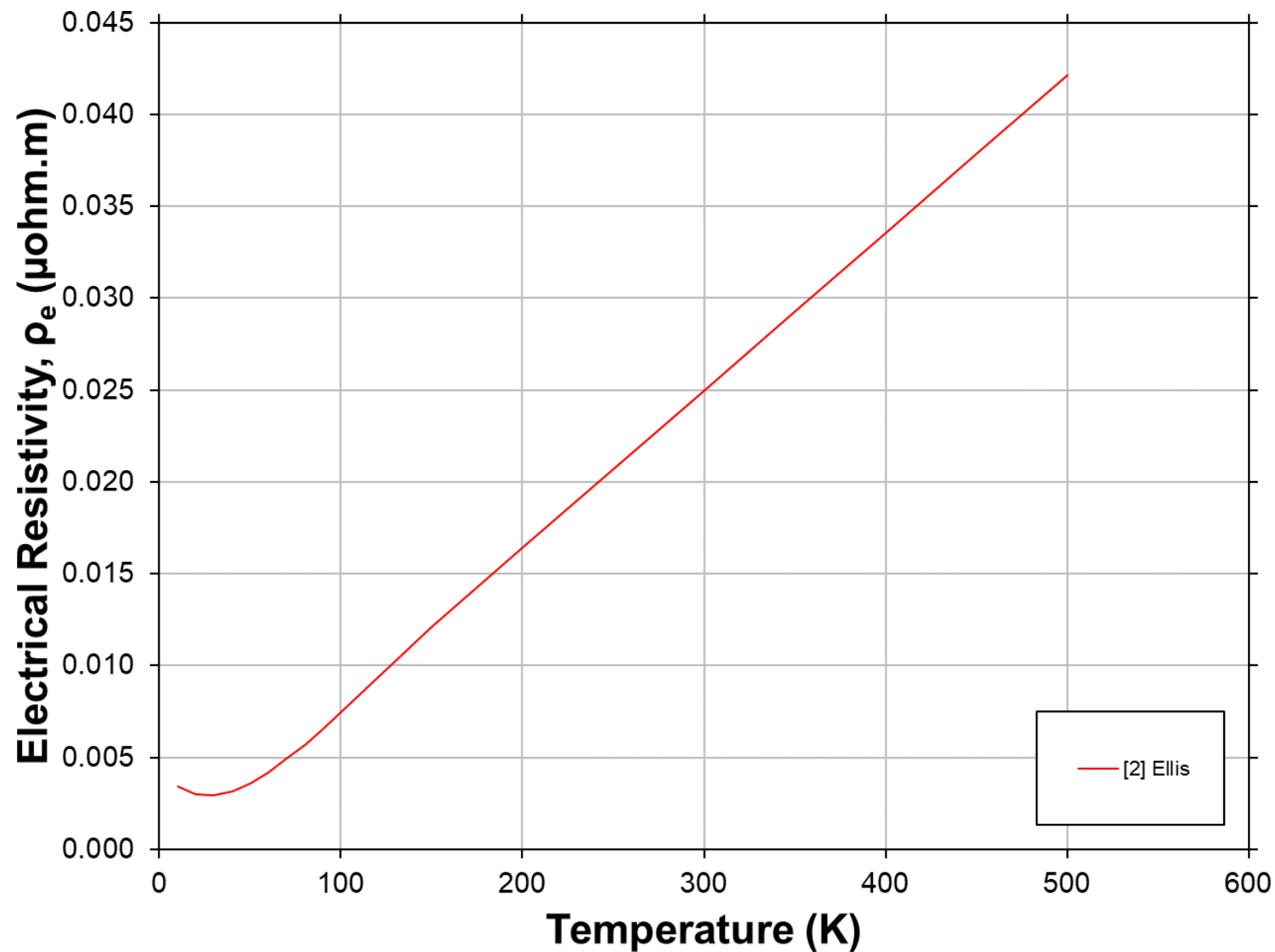


Figure 4.1.1-6: Electrical Resistivity versus Temperature of GRCop84, adapted from Ellis (2005).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCo 84

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |           | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|-----------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )    | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 10                | ( -441.7 ) | 0.003                               | ( 0.14 )                | 260               | ( 8.3 )   | 0.022                               | ( 0.85 )                |
| 20                | ( -423.7 ) | 0.003                               | ( 0.12 )                | 280               | ( 44.3 )  | 0.023                               | ( 0.92 )                |
| 40                | ( -387.7 ) | 0.003                               | ( 0.12 )                | 300               | ( 80.3 )  | 0.025                               | ( 0.98 )                |
| 60                | ( -351.7 ) | 0.004                               | ( 0.17 )                | 320               | ( 116.3 ) | 0.027                               | ( 1.05 )                |
| 80                | ( -315.7 ) | 0.006                               | ( 0.22 )                | 340               | ( 152.3 ) | 0.028                               | ( 1.12 )                |
| 100               | ( -279.7 ) | 0.007                               | ( 0.29 )                | 360               | ( 188.3 ) | 0.030                               | ( 1.19 )                |
| 120               | ( -243.7 ) | 0.009                               | ( 0.37 )                | 380               | ( 224.3 ) | 0.032                               | ( 1.25 )                |
| 140               | ( -207.7 ) | 0.011                               | ( 0.44 )                | 400               | ( 260.3 ) | 0.034                               | ( 1.32 )                |
| 160               | ( -171.7 ) | 0.013                               | ( 0.51 )                | 420               | ( 296.3 ) | 0.035                               | ( 1.39 )                |
| 180               | ( -135.7 ) | 0.015                               | ( 0.58 )                | 440               | ( 332.3 ) | 0.037                               | ( 1.46 )                |
| 200               | ( -99.7 )  | 0.016                               | ( 0.65 )                | 460               | ( 368.3 ) | 0.039                               | ( 1.52 )                |
| 220               | ( -63.7 )  | 0.018                               | ( 0.71 )                | 480               | ( 404.3 ) | 0.040                               | ( 1.59 )                |
| 240               | ( -27.7 )  | 0.020                               | ( 0.78 )                | 500               | ( 440.3 ) | 0.042                               | ( 1.66 )                |

**Application Notes:** Trend for electrical resistivity is collected from reference [2] and presented in the equations below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $10 \leq T < 150$

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3 + A4 \cdot \left(\frac{T}{1000}\right)^4 + A5 \cdot \left(\frac{T}{1000}\right)^5$$

For temperature range:  $150 < T \leq 500$

$$\rho_e(T) = B0 + B1 \cdot \left(\frac{T}{1000}\right)$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot cm$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $10 \leq T < 150$

A0= 0.442

A1= -12.7

A2= 326

A3= -2610

A4= 11200

A5= -19800

$150 \leq T \leq 500$

B0 = -0.0717

B1 = 8.57



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

|                                    |                   |                                 |
|------------------------------------|-------------------|---------------------------------|
| 4 Other Nonferrous Metals & Alloys | 4.1 Copper Alloys | 4.1.1 GRCop 84                  |
| Revision 0: 08-05-2020             |                   | Yield Strength with Temperature |

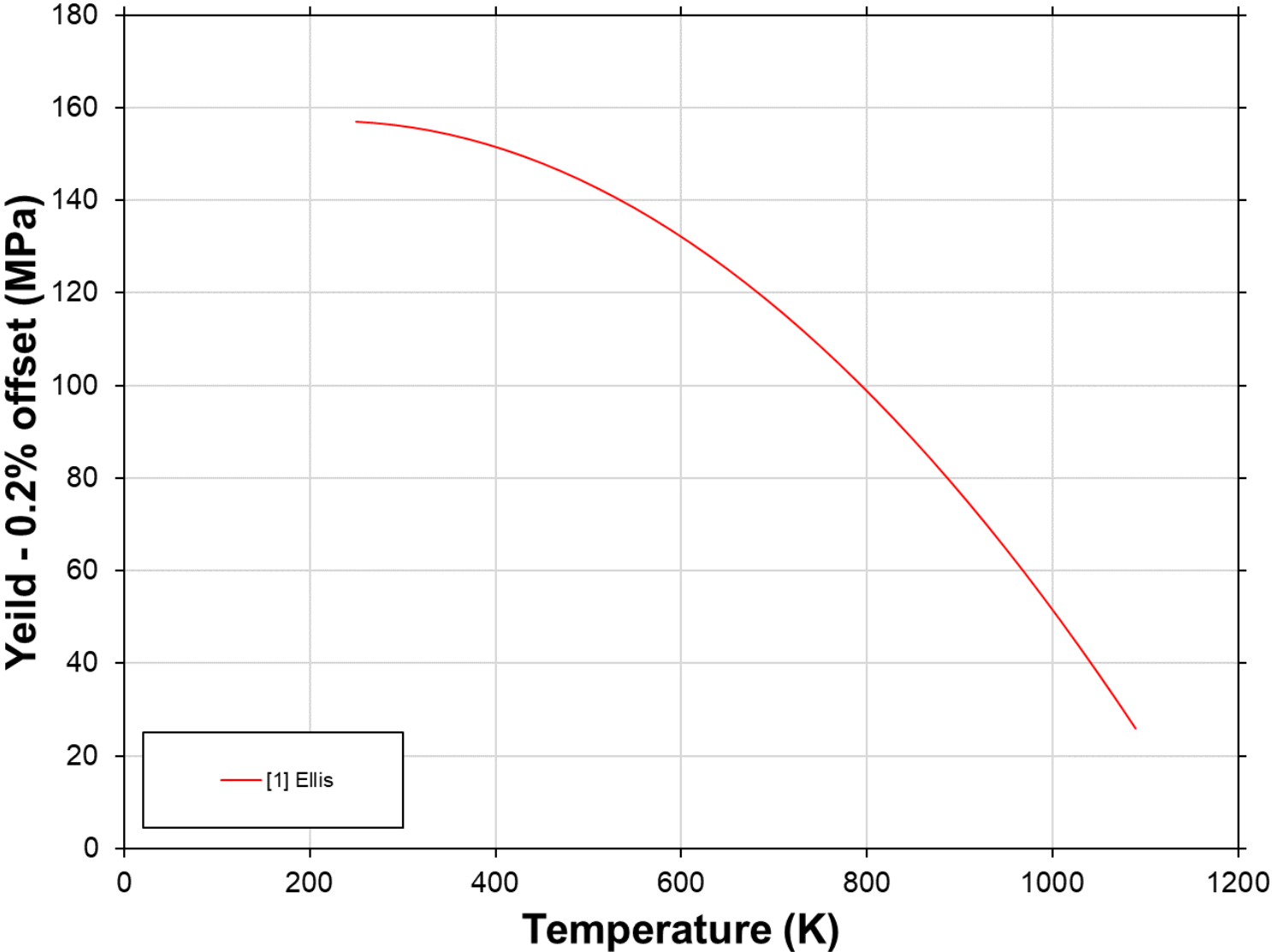


Figure 4.1.1-7: Yield versus Temperature of GRCop84, adapted from Ellis (2000).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCop 84

Revision 0: 08-05-2020

Yield Strength with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Yield Strength ( YS ) |           | Temperature ( T ) |            | Yield Strength ( YS ) |           |
|-------------------|-----------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )    | MPa                   | ( Ksi )   | K                 | ( °F )     | MPa                   | ( Ksi )   |
| 250               | ( -9.7 )  | 157.08                | ( 22.79 ) | 625               | ( 665.3 )  | 128.86                | ( 18.70 ) |
| 275               | ( 35.3 )  | 156.72                | ( 22.74 ) | 650               | ( 710.3 )  | 125.24                | ( 18.17 ) |
| 300               | ( 80.3 )  | 156.14                | ( 22.66 ) | 675               | ( 755.3 )  | 121.40                | ( 17.61 ) |
| 325               | ( 125.3 ) | 155.35                | ( 22.54 ) | 700               | ( 800.3 )  | 117.34                | ( 17.03 ) |
| 350               | ( 170.3 ) | 154.34                | ( 22.39 ) | 725               | ( 845.3 )  | 113.07                | ( 16.41 ) |
| 375               | ( 215.3 ) | 153.11                | ( 22.22 ) | 750               | ( 890.3 )  | 108.58                | ( 15.75 ) |
| 400               | ( 260.3 ) | 151.66                | ( 22.01 ) | 775               | ( 935.3 )  | 103.87                | ( 15.07 ) |
| 425               | ( 305.3 ) | 150.00                | ( 21.76 ) | 800               | ( 980.3 )  | 98.94                 | ( 14.36 ) |
| 450               | ( 350.3 ) | 148.12                | ( 21.49 ) | 825               | ( 1025.3 ) | 93.80                 | ( 13.61 ) |
| 475               | ( 395.3 ) | 146.02                | ( 21.19 ) | 850               | ( 1070.3 ) | 88.44                 | ( 12.83 ) |
| 500               | ( 440.3 ) | 143.70                | ( 20.85 ) | 900               | ( 1160.3 ) | 77.06                 | ( 11.18 ) |
| 525               | ( 485.3 ) | 141.17                | ( 20.48 ) | 950               | ( 1250.3 ) | 64.82                 | ( 9.40 )  |
| 550               | ( 530.3 ) | 138.42                | ( 20.08 ) | 1000              | ( 1340.3 ) | 51.70                 | ( 7.50 )  |
| 575               | ( 575.3 ) | 135.45                | ( 19.65 ) | 1050              | ( 1430.3 ) | 37.72                 | ( 5.47 )  |
| 600               | ( 620.3 ) | 132.26                | ( 19.19 ) | 1090              | ( 1502.3 ) | 25.90                 | ( 3.76 )  |

**Application Notes:** Trend for yield strength is found in reference [1] and is presented here in the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$YS(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$YS(T)$  = Yield Strength [MPa]

$T$  = Temperature [K]

## Constants:

T Range [K]: 250 < T < 1090

A0 = 148.7

A1 = 77.0

A2 = -174.0



|                                    |                   |                                   |
|------------------------------------|-------------------|-----------------------------------|
| 4 Other Nonferrous Metals & Alloys | 4.1 Copper Alloys | 4.1.1 GRCop 84                    |
| Revision 0: 08-05-2020             |                   | Tensile Strength with Temperature |

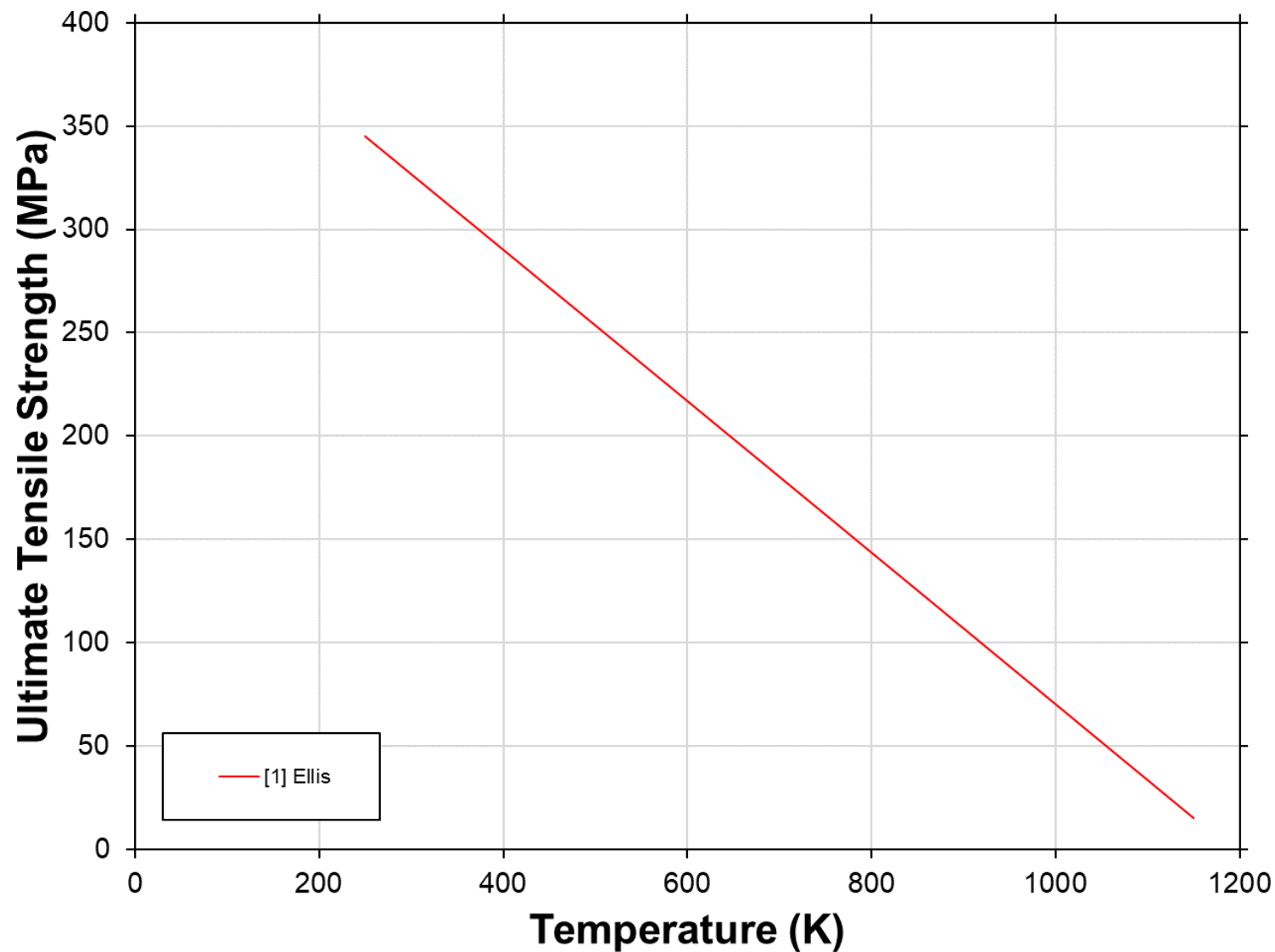


Figure 4.1.1-8: Tensile Strength versus Temperature of GRCop84, adapted from Ellis (2000).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCo 84

Revision 0: 08-05-2020

Tensile Strength with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Tensile Strength ( TS ) |           | Temperature ( T ) |            | Tensile Strength ( TS ) |           |
|-------------------|-----------|-------------------------|-----------|-------------------|------------|-------------------------|-----------|
| K                 | ( °F )    | MPa                     | ( Ksi )   | K                 | ( °F )     | MPa                     | ( Ksi )   |
| 250               | ( -9.7 )  | 344.98                  | ( 50.06 ) | 625               | ( 665.3 )  | 207.75                  | ( 30.14 ) |
| 275               | ( 35.3 )  | 335.84                  | ( 48.73 ) | 650               | ( 710.3 )  | 198.59                  | ( 28.82 ) |
| 300               | ( 80.3 )  | 326.70                  | ( 47.40 ) | 675               | ( 755.3 )  | 189.43                  | ( 27.49 ) |
| 325               | ( 125.3 ) | 317.56                  | ( 46.08 ) | 700               | ( 800.3 )  | 180.27                  | ( 26.16 ) |
| 350               | ( 170.3 ) | 308.42                  | ( 44.75 ) | 725               | ( 845.3 )  | 171.10                  | ( 24.83 ) |
| 375               | ( 215.3 ) | 299.27                  | ( 43.42 ) | 750               | ( 890.3 )  | 161.94                  | ( 23.50 ) |
| 400               | ( 260.3 ) | 290.13                  | ( 42.10 ) | 775               | ( 935.3 )  | 152.77                  | ( 22.17 ) |
| 425               | ( 305.3 ) | 280.98                  | ( 40.77 ) | 800               | ( 980.3 )  | 143.60                  | ( 20.84 ) |
| 450               | ( 350.3 ) | 271.83                  | ( 39.44 ) | 850               | ( 1070.3 ) | 125.26                  | ( 18.18 ) |
| 475               | ( 395.3 ) | 262.68                  | ( 38.11 ) | 900               | ( 1160.3 ) | 106.92                  | ( 15.51 ) |
| 500               | ( 440.3 ) | 253.53                  | ( 36.79 ) | 950               | ( 1250.3 ) | 88.57                   | ( 12.85 ) |
| 525               | ( 485.3 ) | 244.37                  | ( 35.46 ) | 1000              | ( 1340.3 ) | 70.21                   | ( 10.19 ) |
| 550               | ( 530.3 ) | 235.22                  | ( 34.13 ) | 1050              | ( 1430.3 ) | 51.85                   | ( 7.52 )  |
| 575               | ( 575.3 ) | 226.06                  | ( 32.80 ) | 1100              | ( 1520.3 ) | 33.48                   | ( 4.86 )  |
| 600               | ( 620.3 ) | 216.91                  | ( 31.47 ) | 1150              | ( 1610.3 ) | 15.11                   | ( 2.19 )  |

**Application Notes:** Trend for tensile strength is found in reference [1] and is presented in the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$TS(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$TS(T)$  = Tensile Strength [MPa]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $250 \leq T \leq 1150$

A0 = 436.3

A1 = -365.0

A2 = -1.09



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCo 84

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                          | Thermal                                     |                                                    |                                          |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|----------------------------------------------------------|---------------------------------------------|----------------------------------------------------|------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-cm | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( α )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 50                  | 8697                                  | 0.360                                                    | 462.30                                      | -0.294                                             | -                                        | -                                              | -                                  | -                                | -                           |
| 100                 | 8684                                  | 0.744                                                    | 307.26                                      | -0.245                                             | 12.55                                    | -                                              | -                                  | -                                | -                           |
| 150                 | 8669                                  | 1.214                                                    | 277.64                                      | -0.187                                             | 13.17                                    | -                                              | -                                  | -                                | -                           |
| 200                 | 8652                                  | 1.642                                                    | 274.53                                      | -0.124                                             | 13.74                                    | -                                              | -                                  | -                                | -                           |
| 250                 | 8634                                  | 2.071                                                    | 278.77                                      | -0.056                                             | 14.26                                    | -                                              | -                                  | -                                | -                           |
| 300                 | 8616                                  | 2.499                                                    | 284.80                                      | 0.017                                              | 14.74                                    | 0.382                                          | -                                  | -                                | -                           |
| 350                 | 8596                                  | 2.928                                                    | 290.71                                      | 0.092                                              | 15.18                                    | 0.393                                          | -                                  | -                                | -                           |
| 400                 | 8576                                  | 3.356                                                    | 295.81                                      | 0.171                                              | 15.57                                    | 0.401                                          | -                                  | -                                | -                           |
| 450                 | 8555                                  | 3.785                                                    | 299.88                                      | 0.253                                              | 15.93                                    | 0.408                                          | -                                  | -                                | -                           |
| 500                 | 8533                                  | 4.213                                                    | 302.90                                      | 0.337                                              | 16.25                                    | 0.413                                          | -                                  | -                                | -                           |
| 550                 | 8511                                  | -                                                        | 304.92                                      | 0.424                                              | 16.55                                    | 0.417                                          | -                                  | -                                | -                           |
| 600                 | 8489                                  | -                                                        | 306.01                                      | 0.513                                              | 16.81                                    | 0.420                                          | -                                  | -                                | -                           |
| 650                 | 8466                                  | -                                                        | 306.27                                      | 0.604                                              | 17.05                                    | 0.422                                          | -                                  | -                                | -                           |
| 700                 | 8442                                  | -                                                        | 305.77                                      | 0.697                                              | 17.27                                    | 0.424                                          | -                                  | -                                | -                           |
| 750                 | 8418                                  | -                                                        | 304.59                                      | 0.792                                              | 17.47                                    | 0.426                                          | -                                  | -                                | -                           |
| 800                 | 8394                                  | -                                                        | 302.80                                      | 0.889                                              | 17.66                                    | 0.429                                          | -                                  | -                                | -                           |
| 850                 | 8370                                  | -                                                        | 300.48                                      | 0.988                                              | 17.83                                    | 0.432                                          | -                                  | -                                | -                           |
| 900                 | 8345                                  | -                                                        | 297.66                                      | 1.088                                              | 17.99                                    | 0.436                                          | -                                  | -                                | -                           |
| 950                 | 8319                                  | -                                                        | 294.41                                      | 1.190                                              | 18.15                                    | 0.442                                          | -                                  | -                                | -                           |
| 1000                | 8294                                  | -                                                        | 290.77                                      | 1.293                                              | 18.30                                    | 0.449                                          | -                                  | -                                | -                           |
| 1050                | 8268                                  | -                                                        | 286.77                                      | 1.398                                              | 18.45                                    | 0.458                                          | -                                  | -                                | -                           |
| 1100                | 8242                                  | -                                                        | 282.46                                      | 1.505                                              | 18.60                                    | 0.470                                          | -                                  | -                                | -                           |
| 1150                | 8216                                  | -                                                        | 277.86                                      | 1.612                                              | 18.75                                    | 0.484                                          | -                                  | -                                | -                           |
| 1200                | 8190                                  | -                                                        | -                                           | 1.722                                              | 18.92                                    | -                                              | -                                  | -                                | -                           |
| 1250                | 8163                                  | -                                                        | -                                           | 1.832                                              | 19.09                                    | -                                              | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.1 Copper Alloys

4.1.1 GRCop 84

**Revision 0: 08-05-2020**

**References**

- [1] D.L. Ellis, D.J. Keller, M. Nathal, Thermophysical properties of GRCop-84, Case Western Reserve Univ.; Cleveland, OH United States 2000.
- [2] D.L. Ellis, GRCop-84: A high-temperature copper alloy for high-heat-flux applications, NASA Glenn Research Center; Cleveland, OH, United States 2005.

## **4 Other Nonferrous Metals and Alloys**

### **4.2 Zirconium Alloys**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

General

## Room Temperature Properties

|                                          |       |
|------------------------------------------|-------|
| Density, [kg/m <sup>3</sup> ]            | 6551  |
| Melting Point, [K]                       | 2150  |
| Specific Heat, [J/(g-K)]                 | 0.286 |
| Thermal Conductivity, [W/(m-K)]          | 13.4  |
| Linear expansion coefficient, [μm/(m-K)] | 7.5   |
| Electrical resistivity, [μΩ-m]           | 0.729 |
| Young's Modulus, [GPa]                   | 91    |
| Shear Modulus, [GPa]                     | 33.1  |
| Poisson's Ratio, [-]                     | 0.37  |

## Composition

Table 4.2.1-1: Typical Composition ranges for Zircaloy (percent by weight), adapted from [1].

| Grade      |                 | Sn  | Ni   | Fe   | Cr   | Zr    | Other      |
|------------|-----------------|-----|------|------|------|-------|------------|
| Zircaloy-2 | High Zr Content | 1.2 | 0.03 | 0.07 | 0.05 | 98.65 | O-1118ppm  |
|            | Low Zr Content  | 1.7 | 0.08 | 0.2  | 0.15 | 97.87 | C-113 ppm  |
| Zircaloy-4 | High Zr Content | 1.2 | -    | 0.18 | 0.07 | 98.55 | O-1220 ppm |
|            | Low Zr Content  | 1.7 | -    | 0.24 | 0.13 | 97.93 | C-113 ppm  |



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Density with Temperature

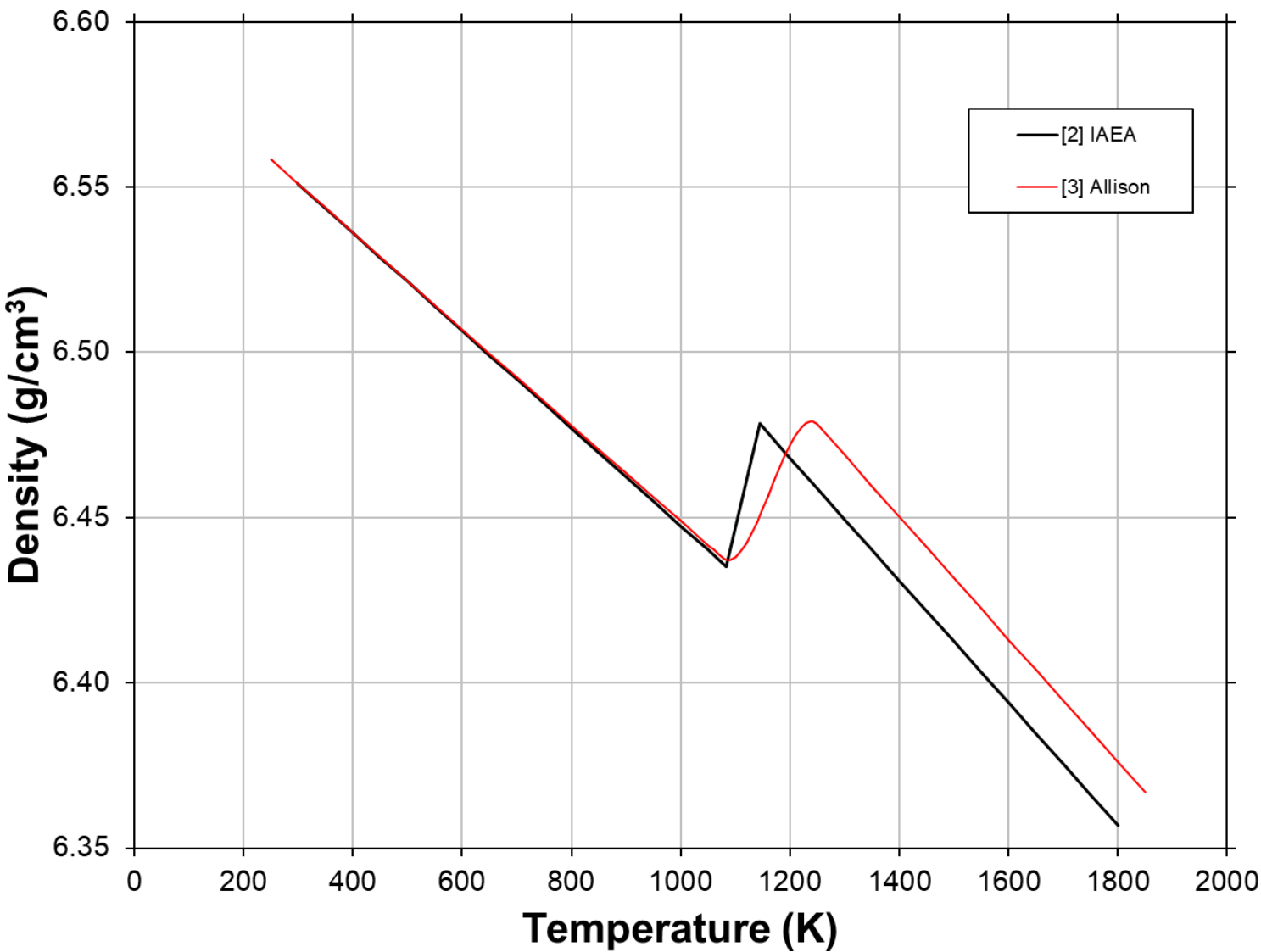


Figure 4.2.1-1: Density versus Temperature for Zircaloy.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 300               | ( 80.3 )   | 6551              | ( 409.0 )              | 1000              | ( 1340.3 ) | 6449              | ( 402.6 )              |
| 350               | ( 170.3 )  | 6544              | ( 408.5 )              | 1050              | ( 1430.3 ) | 6442              | ( 402.2 )              |
| 400               | ( 260.3 )  | 6536              | ( 408.1 )              | 1100              | ( 1520.3 ) | 6438              | ( 401.9 )              |
| 450               | ( 350.3 )  | 6529              | ( 407.6 )              | 1150              | ( 1610.3 ) | 6453              | ( 402.8 )              |
| 500               | ( 440.3 )  | 6522              | ( 407.1 )              | 1200              | ( 1700.3 ) | 6472              | ( 404.0 )              |
| 550               | ( 530.3 )  | 6514              | ( 406.7 )              | 1250              | ( 1790.3 ) | 6478              | ( 404.4 )              |
| 600               | ( 620.3 )  | 6507              | ( 406.2 )              | 1300              | ( 1880.3 ) | 6469              | ( 403.9 )              |
| 650               | ( 710.3 )  | 6500              | ( 405.8 )              | 1350              | ( 1970.3 ) | 6460              | ( 403.3 )              |
| 700               | ( 800.3 )  | 6492              | ( 405.3 )              | 1400              | ( 2060.3 ) | 6450              | ( 402.7 )              |
| 750               | ( 890.3 )  | 6485              | ( 404.9 )              | 1450              | ( 2150.3 ) | 6441              | ( 402.1 )              |
| 800               | ( 980.3 )  | 6478              | ( 404.4 )              | 1500              | ( 2240.3 ) | 6432              | ( 401.5 )              |
| 850               | ( 1070.3 ) | 6471              | ( 404.0 )              | 1600              | ( 2420.3 ) | 6413              | ( 400.4 )              |
| 900               | ( 1160.3 ) | 6463              | ( 403.5 )              | 1700              | ( 2600.3 ) | 6395              | ( 399.2 )              |
| 950               | ( 1250.3 ) | 6456              | ( 403.1 )              | 1800              | ( 2780.3 ) | 6376              | ( 398.1 )              |

**Application Notes:** Data for density is collected from reference [2]. The red curve is calculated from thermal expansion data collected from reference [3], utilizing the equation below to approximate density trend with respect to temperature.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 6551 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $0 \leq T \leq 1800$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

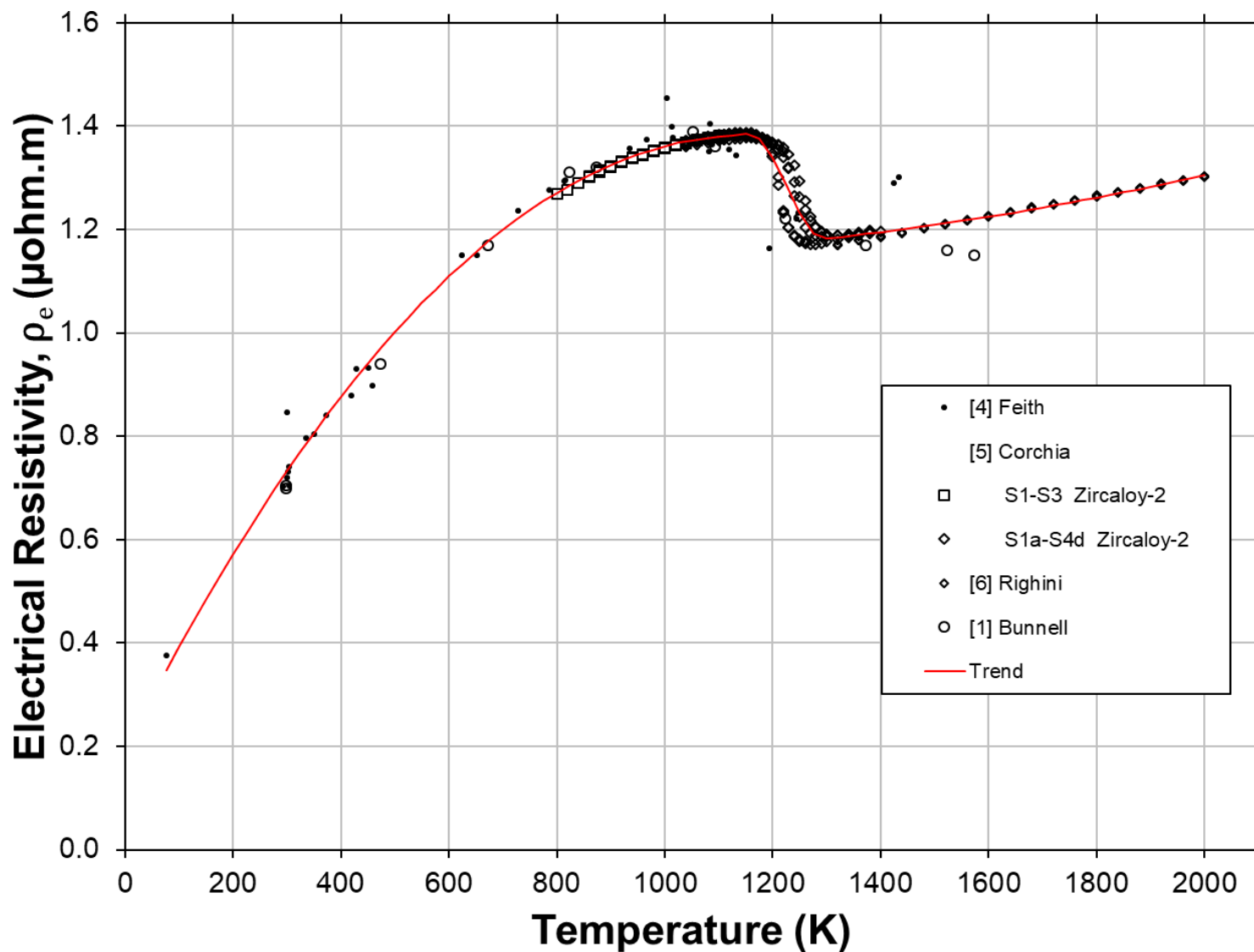


Figure 4.2.1-2: Electrical Resistivity versus Temperature of Zircaloy.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 77                | ( -321.1 ) | 0.348                               | ( 13.70 )               | 1100              | ( 1520.3 ) | 1.379                               | ( 54.30 )               |
| 100               | ( -279.7 ) | 0.392                               | ( 15.42 )               | 1200              | ( 1700.3 ) | 1.339                               | ( 52.73 )               |
| 200               | ( -99.7 )  | 0.571                               | ( 22.48 )               | 1300              | ( 1880.3 ) | 1.183                               | ( 46.56 )               |
| 300               | ( 80.3 )   | 0.732                               | ( 28.84 )               | 1400              | ( 2060.3 ) | 1.195                               | ( 47.04 )               |
| 400               | ( 260.3 )  | 0.876                               | ( 34.49 )               | 1500              | ( 2240.3 ) | 1.209                               | ( 47.59 )               |
| 500               | ( 440.3 )  | 1.002                               | ( 39.43 )               | 1600              | ( 2420.3 ) | 1.225                               | ( 48.21 )               |
| 600               | ( 620.3 )  | 1.109                               | ( 43.67 )               | 1700              | ( 2600.3 ) | 1.242                               | ( 48.90 )               |
| 700               | ( 800.3 )  | 1.199                               | ( 47.21 )               | 1800              | ( 2780.3 ) | 1.261                               | ( 49.66 )               |
| 800               | ( 980.3 )  | 1.271                               | ( 50.04 )               | 1900              | ( 2960.3 ) | 1.283                               | ( 50.49 )               |
| 900               | ( 1160.3 ) | 1.325                               | ( 52.16 )               | 2000              | ( 3140.3 ) | 1.305                               | ( 51.40 )               |
| 1000              | ( 1340.3 ) | 1.361                               | ( 53.58 )               |                   |            |                                     |                         |

**Application Notes:** Data for electrical resistivity is collected from references [1, 4-6] and fitted with the equations below to approximate the property trend with respect to temperature. Trend equations are shown for 77 to 1150 K, and 1300 to 2000 K. There is a phase change in this material between 1150 and 1300 K. An equation for this section is not given, however there is tabulated data for this range at the end of the chapter.

## Fit Equations:

For temperature range:  $77 \leq T \leq 1150$

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

For temperature range:  $1300 \leq T \leq 2000$

$$\rho_e(T) = C0 + C1 \cdot \left(\frac{T}{1000}\right) + C2 \cdot \left(\frac{T}{1000}\right)^2$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

Temperature  
Range [K]:

$77 \leq T \leq 1150$

$1300 \leq T \leq 2000$

A0 = 0.1944

C0 = 1.187

A1 = 2.062

C1 = -0.1196

A2 = -0.8954

C2 = 0.08941



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

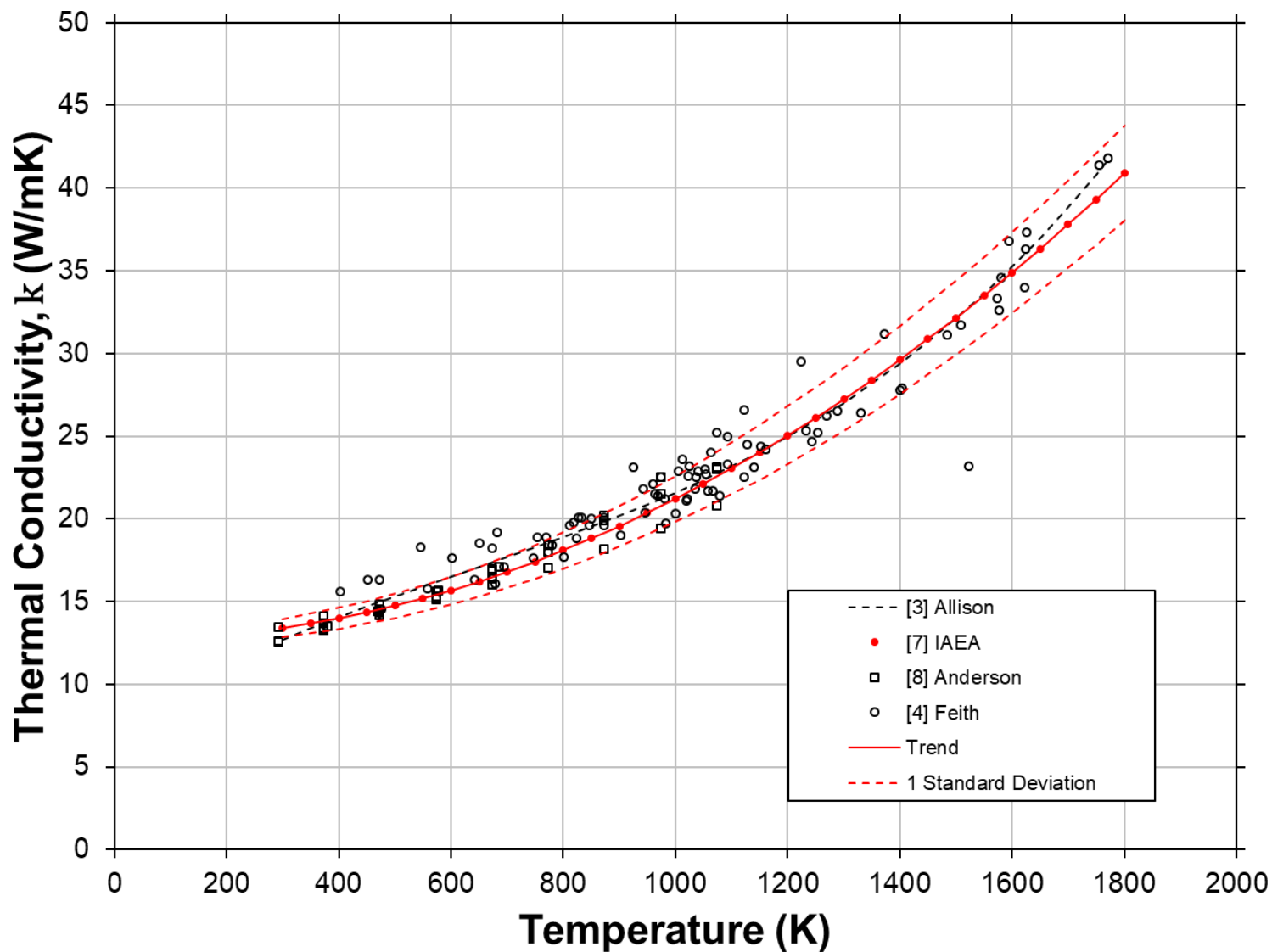


Figure 4.2.1-3: Thermal Conductivity versus Temperature of Zircaloy.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       |
|-------------------|------------|----------------------------|---------------------------------------|-------------------|------------|----------------------------|---------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) |
| 300               | ( 80.3 )   | 13.41                      | ( 93.06 )                             | 1000              | ( 1340.3 ) | 21.21                      | ( 147.12 )                            |
| 350               | ( 170.3 )  | 13.68                      | ( 94.89 )                             | 1050              | ( 1430.3 ) | 22.10                      | ( 153.32 )                            |
| 400               | ( 260.3 )  | 13.99                      | ( 97.04 )                             | 1100              | ( 1520.3 ) | 23.04                      | ( 159.83 )                            |
| 450               | ( 350.3 )  | 14.34                      | ( 99.50 )                             | 1150              | ( 1610.3 ) | 24.02                      | ( 166.66 )                            |
| 500               | ( 440.3 )  | 14.74                      | ( 102.27 )                            | 1200              | ( 1700.3 ) | 25.05                      | ( 173.79 )                            |
| 550               | ( 530.3 )  | 15.19                      | ( 105.36 )                            | 1250              | ( 1790.3 ) | 26.12                      | ( 181.24 )                            |
| 600               | ( 620.3 )  | 15.67                      | ( 108.75 )                            | 1300              | ( 1880.3 ) | 27.24                      | ( 188.99 )                            |
| 650               | ( 710.3 )  | 16.21                      | ( 112.46 )                            | 1350              | ( 1970.3 ) | 28.40                      | ( 197.06 )                            |
| 700               | ( 800.3 )  | 16.79                      | ( 116.47 )                            | 1400              | ( 2060.3 ) | 29.61                      | ( 205.44 )                            |
| 750               | ( 890.3 )  | 17.41                      | ( 120.80 )                            | 1450              | ( 2150.3 ) | 30.86                      | ( 214.13 )                            |
| 800               | ( 980.3 )  | 18.08                      | ( 125.45 )                            | 1500              | ( 2240.3 ) | 32.16                      | ( 223.14 )                            |
| 850               | ( 1070.3 ) | 18.79                      | ( 130.40 )                            | 1600              | ( 2420.3 ) | 34.89                      | ( 242.08 )                            |
| 900               | ( 1160.3 ) | 19.55                      | ( 135.66 )                            | 1700              | ( 2600.3 ) | 37.80                      | ( 262.26 )                            |
| 950               | ( 1250.3 ) | 20.36                      | ( 141.24 )                            | 1800              | ( 2780.3 ) | 40.89                      | ( 283.70 )                            |

**Application Notes:** Data for thermal conductivity is collected from references [3, 4, 7, 8] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equations:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]: 298 ≤ T < 1800

A0 = 12.767

A1 = -0.54348

A2 = 8.9818



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Axial Thermal Expansion with Temperature

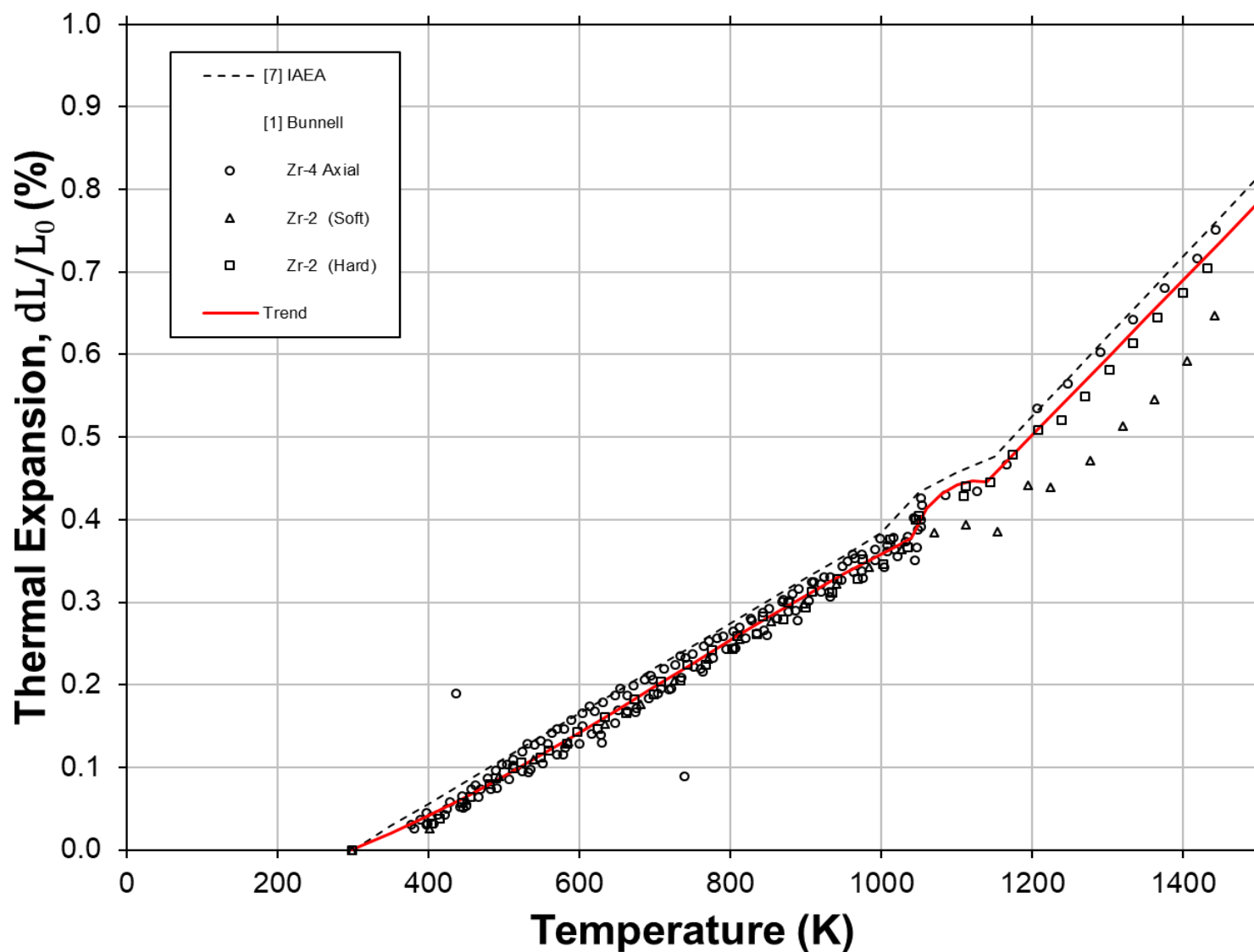


Figure 4.2.1-4: Axial Thermal Expansion versus Temperature of Zircaloy.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Axial Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Axial Thermal Expansion ( dL/L0 ) | Temperature ( T ) |            | Axial Thermal Expansion ( dL/L0 ) |
|-------------------|-----------|-----------------------------------|-------------------|------------|-----------------------------------|
| K                 | ( °F )    |                                   | K                 | ( °F )     |                                   |
| 300               | ( 80.3 )  | 0.001                             | 700               | ( 800.3 )  | 0.198                             |
| 325               | ( 125.3 ) | 0.010                             | 750               | ( 890.3 )  | 0.226                             |
| 350               | ( 170.3 ) | 0.020                             | 800               | ( 980.3 )  | 0.254                             |
| 375               | ( 215.3 ) | 0.031                             | 850               | ( 1070.3 ) | 0.282                             |
| 400               | ( 260.3 ) | 0.042                             | 900               | ( 1160.3 ) | 0.309                             |
| 425               | ( 305.3 ) | 0.053                             | 950               | ( 1250.3 ) | 0.334                             |
| 450               | ( 350.3 ) | 0.065                             | 1000              | ( 1340.3 ) | 0.359                             |
| 475               | ( 395.3 ) | 0.077                             | 1050              | ( 1430.3 ) | 0.382                             |
| 500               | ( 440.3 ) | 0.090                             | 1100              | ( 1520.3 ) | 0.442                             |
| 525               | ( 485.3 ) | 0.103                             | 1150              | ( 1610.3 ) | 0.456                             |
| 550               | ( 530.3 ) | 0.116                             | 1200              | ( 1700.3 ) | 0.503                             |
| 575               | ( 575.3 ) | 0.129                             | 1250              | ( 1790.3 ) | 0.550                             |
| 600               | ( 620.3 ) | 0.143                             | 1300              | ( 1880.3 ) | 0.597                             |
| 625               | ( 665.3 ) | 0.157                             | 1400              | ( 2060.3 ) | 0.691                             |
| 650               | ( 710.3 ) | 0.170                             | 1500              | ( 2240.3 ) | 0.785                             |

**Application Notes:** Data for thermal expansion in axial direction was collected from references [1, 7] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$dL/L_0(T)$  = Thermal Expansion [%]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $298 \leq T \leq 1050$   $1050 < T < 1140$   $1140 \leq T \leq 1500$

|      |           |        |        |
|------|-----------|--------|--------|
| A0 = | -0.0596   | -10.01 | -0.625 |
| A1 = | 0.0005036 | 18.61  | 0.9398 |
| A2 = | 0.7761    | -8.28  | 0      |
| A3 = | -0.3579   | 0      | 0      |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Diametral Thermal Expansion with Temperature

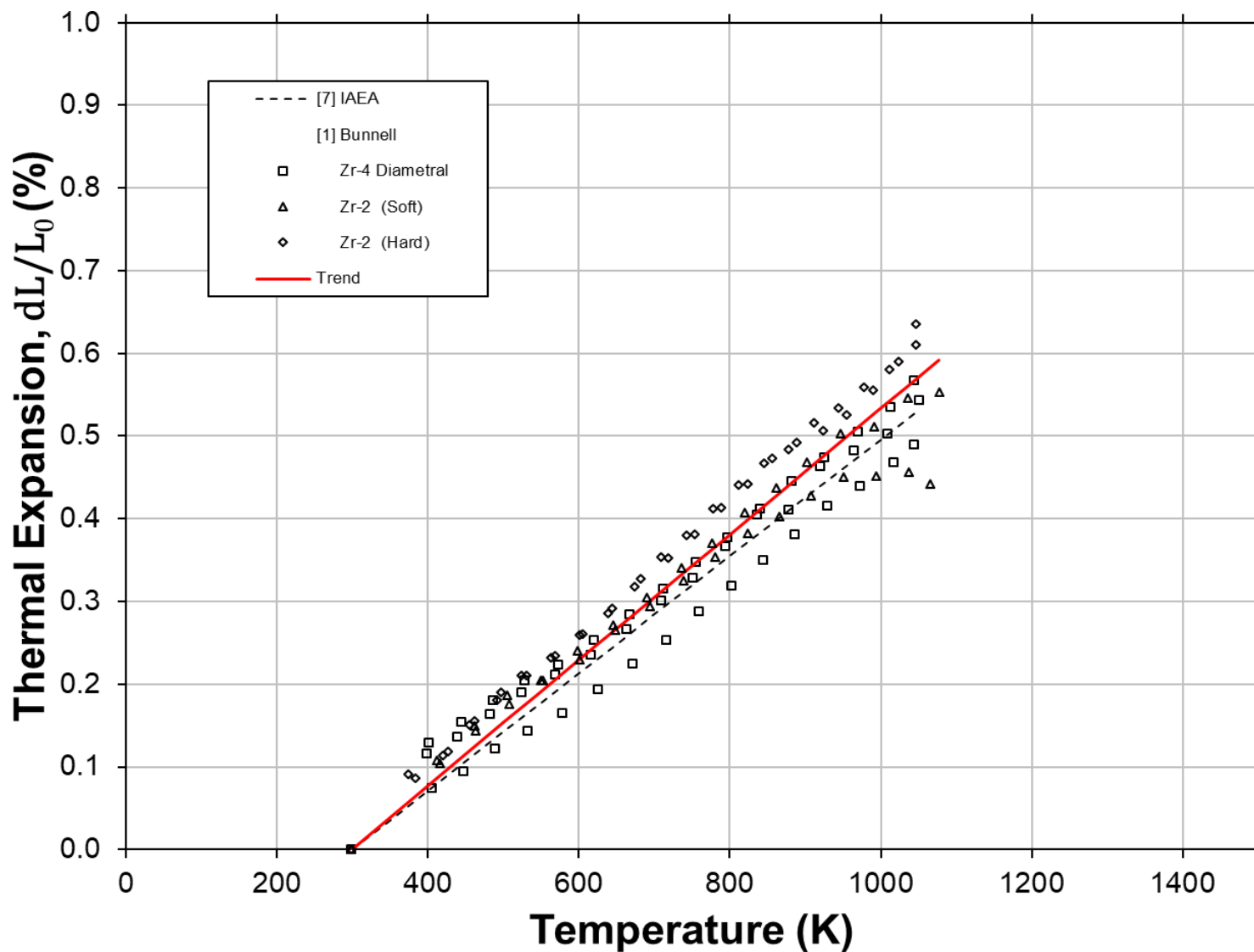


Figure 4.2.1-5: Diametral Thermal Expansion versus Temperature of Zircaloy.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Diametral Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Diametral Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Diametral Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|-----------|---------------------------------------------------|-------------------|------------|---------------------------------------------------|
| K                 | ( °F )    |                                                   | K                 | ( °F )     |                                                   |
| 300               | ( 80.3 )  | 0.002                                             | 675               | ( 755.3 )  | 0.286                                             |
| 325               | ( 125.3 ) | 0.021                                             | 700               | ( 800.3 )  | 0.305                                             |
| 350               | ( 170.3 ) | 0.039                                             | 725               | ( 845.3 )  | 0.324                                             |
| 375               | ( 215.3 ) | 0.058                                             | 750               | ( 890.3 )  | 0.343                                             |
| 400               | ( 260.3 ) | 0.077                                             | 775               | ( 935.3 )  | 0.362                                             |
| 425               | ( 305.3 ) | 0.096                                             | 800               | ( 980.3 )  | 0.381                                             |
| 450               | ( 350.3 ) | 0.115                                             | 825               | ( 1025.3 ) | 0.400                                             |
| 475               | ( 395.3 ) | 0.134                                             | 850               | ( 1070.3 ) | 0.419                                             |
| 500               | ( 440.3 ) | 0.153                                             | 875               | ( 1115.3 ) | 0.438                                             |
| 525               | ( 485.3 ) | 0.172                                             | 900               | ( 1160.3 ) | 0.457                                             |
| 550               | ( 530.3 ) | 0.191                                             | 925               | ( 1205.3 ) | 0.476                                             |
| 575               | ( 575.3 ) | 0.210                                             | 950               | ( 1250.3 ) | 0.495                                             |
| 600               | ( 620.3 ) | 0.229                                             | 975               | ( 1295.3 ) | 0.514                                             |
| 625               | ( 665.3 ) | 0.248                                             | 1000              | ( 1340.3 ) | 0.533                                             |
| 650               | ( 710.3 ) | 0.267                                             | 1077              | ( 1478.9 ) | 0.592                                             |

**Application Notes:** Data for thermal expansion in the diametral direction was collected from references [1, 7] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right)$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 298 < T \leq 1077$$

$$A0 = -0.2264$$

$$A1 = 0.7597$$



|                                    |                      |                            |
|------------------------------------|----------------------|----------------------------|
| 4 Other Nonferrous Metals & Alloys | 4.2 Zirconium Alloys | 4.2.1 Zircaloy             |
| Revision 2: 04-26-2023             |                      | Axial CTE with Temperature |

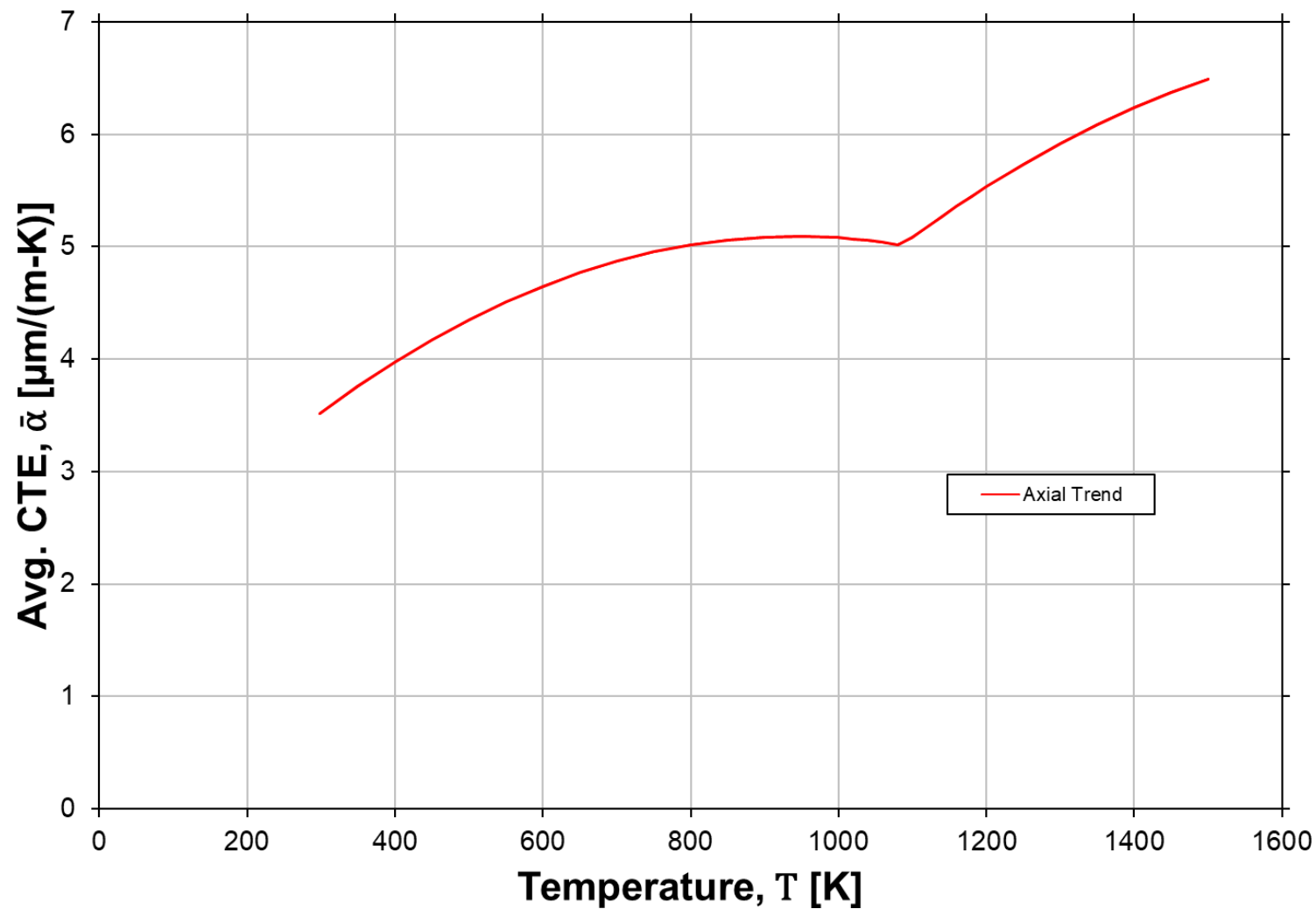


Figure 4.2.1-6: Axial Mean Coefficient of Thermal Expansion versus Temperature of Zircaloy. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 2: 04-26-2023

Axial CTE with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|-----------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )    | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 298               | ( 76.7 )  | 3.513                                 | ( 1.952 )                                           | 850               | ( 1070.3 ) | 5.061                                 | ( 2.812 )                                           |
| 300               | ( 80.3 )  | 3.523                                 | ( 1.957 )                                           | 900               | ( 1160.3 ) | 5.087                                 | ( 2.826 )                                           |
| 320               | ( 116.3 ) | 3.619                                 | ( 2.011 )                                           | 950               | ( 1250.3 ) | 5.093                                 | ( 2.829 )                                           |
| 340               | ( 152.3 ) | 3.713                                 | ( 2.063 )                                           | 1000              | ( 1340.3 ) | 5.080                                 | ( 2.822 )                                           |
| 360               | ( 188.3 ) | 3.803                                 | ( 2.113 )                                           | 1050              | ( 1430.3 ) | 5.048                                 | ( 2.804 )                                           |
| 380               | ( 224.3 ) | 3.890                                 | ( 2.161 )                                           | 1100              | ( 1520.3 ) | 5.083                                 | ( 2.824 )                                           |
| 400               | ( 260.3 ) | 3.975                                 | ( 2.208 )                                           | 1150              | ( 1610.3 ) | 5.317                                 | ( 2.954 )                                           |
| 450               | ( 350.3 ) | 4.172                                 | ( 2.318 )                                           | 1200              | ( 1700.3 ) | 5.535                                 | ( 3.075 )                                           |
| 500               | ( 440.3 ) | 4.350                                 | ( 2.417 )                                           | 1250              | ( 1790.3 ) | 5.736                                 | ( 3.187 )                                           |
| 550               | ( 530.3 ) | 4.509                                 | ( 2.505 )                                           | 1300              | ( 1880.3 ) | 5.921                                 | ( 3.289 )                                           |
| 600               | ( 620.3 ) | 4.649                                 | ( 2.583 )                                           | 1350              | ( 1970.3 ) | 6.088                                 | ( 3.382 )                                           |
| 650               | ( 710.3 ) | 4.769                                 | ( 2.650 )                                           | 1400              | ( 2060.3 ) | 6.239                                 | ( 3.466 )                                           |
| 700               | ( 800.3 ) | 4.871                                 | ( 2.706 )                                           | 1450              | ( 2150.3 ) | 6.374                                 | ( 3.541 )                                           |
| 750               | ( 890.3 ) | 4.954                                 | ( 2.752 )                                           | 1500              | ( 2240.3 ) | 6.492                                 | ( 3.606 )                                           |
| 800               | ( 980.3 ) | 5.017                                 | ( 2.787 )                                           |                   |            |                                       |                                                     |

**Application Notes:** Trend for axial mean coefficient of thermal expansion is calculated based on the axial thermal expansion trend, and it is fitted with the equations below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$\bar{\alpha}(T)$  = Mean Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

T = Temperature [K]

## Constants:

| Temperature Range [K]: | <u><math>298 \leq T \leq 1083</math></u> | <u><math>1083 &lt; T \leq 1500</math></u> |
|------------------------|------------------------------------------|-------------------------------------------|
| A0 =                   | 1.71                                     | -4.292                                    |
| A1 =                   | 7.189                                    | 12.19                                     |
| A2 =                   | -3.819                                   | -3.334                                    |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Diametral CTE with Temperature

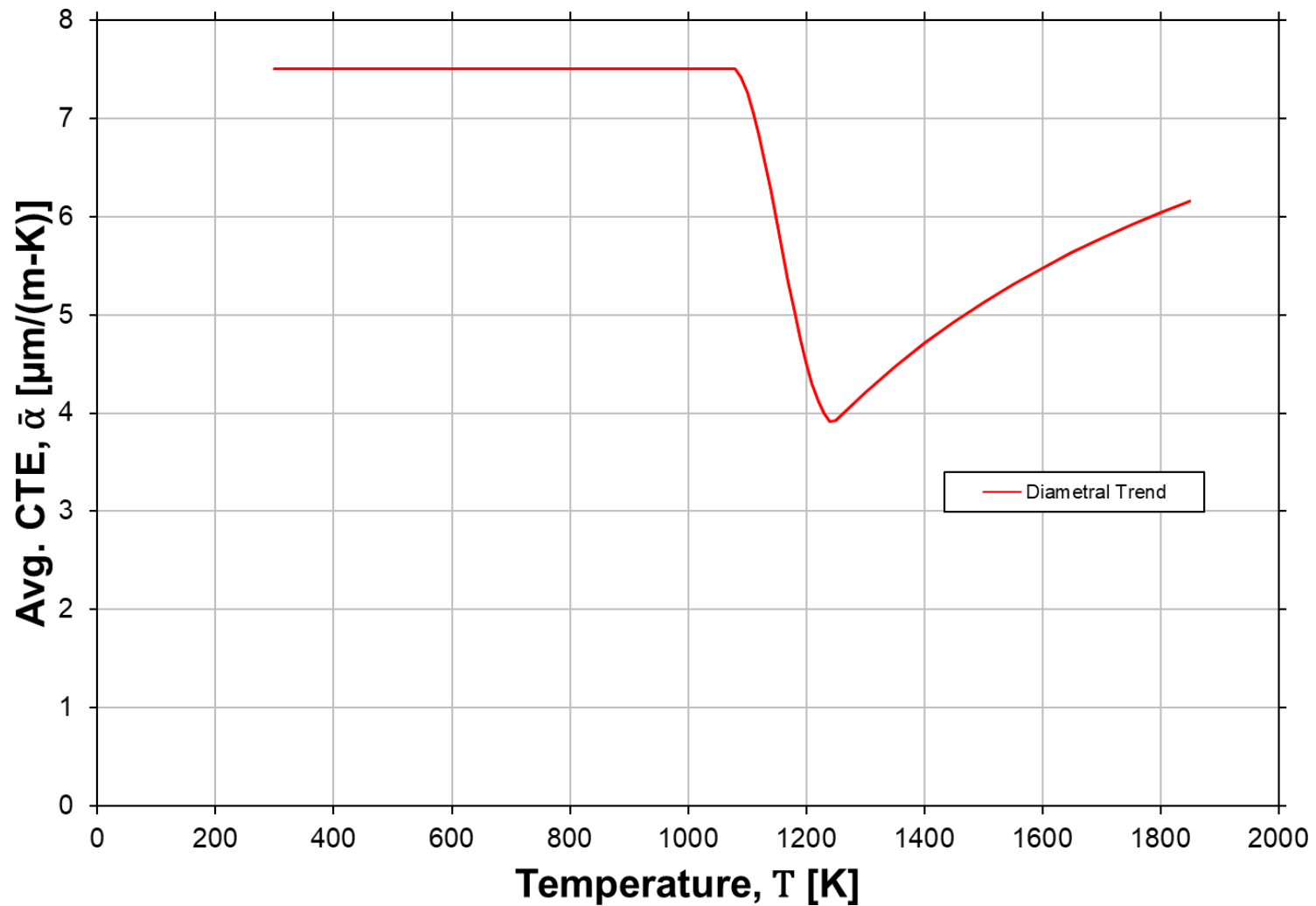


Figure 4.2.1-7: Diametral Mean Coefficient of Thermal Expansion versus Temperature of Zircaloy. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 2: 04-26-2023

Diametral CTE with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 300               | ( 80.3 )   | 7.500                                 | ( 4.167 )                                           | 1083              | ( 1489.7 ) | 7.500                                 | ( 4.167 )                                           |
| 350               | ( 170.3 )  | 7.500                                 | ( 4.167 )                                           | 1100              | ( 1520.3 ) | 7.266                                 | ( 4.037 )                                           |
| 400               | ( 260.3 )  | 7.500                                 | ( 4.167 )                                           | 1150              | ( 1610.3 ) | 5.953                                 | ( 3.307 )                                           |
| 450               | ( 350.3 )  | 7.500                                 | ( 4.167 )                                           | 1200              | ( 1700.3 ) | 4.507                                 | ( 2.504 )                                           |
| 500               | ( 440.3 )  | 7.500                                 | ( 4.167 )                                           | 1244              | ( 1779.5 ) | 3.894                                 | ( 2.163 )                                           |
| 550               | ( 530.3 )  | 7.500                                 | ( 4.167 )                                           | 1300              | ( 1880.3 ) | 4.210                                 | ( 2.339 )                                           |
| 600               | ( 620.3 )  | 7.500                                 | ( 4.167 )                                           | 1350              | ( 1970.3 ) | 4.471                                 | ( 2.484 )                                           |
| 650               | ( 710.3 )  | 7.500                                 | ( 4.167 )                                           | 1400              | ( 2060.3 ) | 4.709                                 | ( 2.616 )                                           |
| 700               | ( 800.3 )  | 7.500                                 | ( 4.167 )                                           | 1450              | ( 2150.3 ) | 4.926                                 | ( 2.737 )                                           |
| 750               | ( 890.3 )  | 7.500                                 | ( 4.167 )                                           | 1500              | ( 2240.3 ) | 5.125                                 | ( 2.847 )                                           |
| 800               | ( 980.3 )  | 7.500                                 | ( 4.167 )                                           | 1600              | ( 2420.3 ) | 5.477                                 | ( 3.043 )                                           |
| 850               | ( 1070.3 ) | 7.500                                 | ( 4.167 )                                           | 1700              | ( 2600.3 ) | 5.779                                 | ( 3.210 )                                           |
| 900               | ( 1160.3 ) | 7.500                                 | ( 4.167 )                                           | 1800              | ( 2780.3 ) | 6.040                                 | ( 3.356 )                                           |
| 1000              | ( 1340.3 ) | 7.500                                 | ( 4.167 )                                           | 1850              | ( 2870.3 ) | 6.158                                 | ( 3.421 )                                           |

**Application Notes:** Trend for diametral mean coefficient of thermal expansion is adapted from [3], and it is fitted with the equations below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $300 \leq T \leq 1083$

$$\bar{\alpha}(T) = A0$$

For temperature range:  $1083 \leq T \leq 1244$

$$\bar{\alpha}(T) = ((B0 + B1 * \cos(A * (\frac{T}{1000} + X0)))) / (\frac{T}{1000} - T_{Ref})$$

For temperature range:  $1244 \leq T \leq 1850$

$$\bar{\alpha}(T) = ((C0 + C1 * \frac{T}{1000})) / (\frac{T}{1000} - T_{Ref})$$

$$\bar{\alpha}(T) = \text{Mean Coefficient of Thermal Expansion } [\mu\text{m}/(\text{m} \cdot \text{K})]$$

$$T = \text{Temperature } [K]$$

## Constants:

T Range [K]:  $300 \leq T < 1083$      $1083 \leq T < 1244$      $1244 \leq T < 1850$

A0 = 0.75    B0 = 0.477428    C0 = -0.84  
 B1 = 0.109822    C1 = 0.97  
 $T_{Ref} = 0.3$      $T_{Ref} = 0.3$   
 A = 19.513  
 X0 = -1.083



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Specific Heat with Temperature

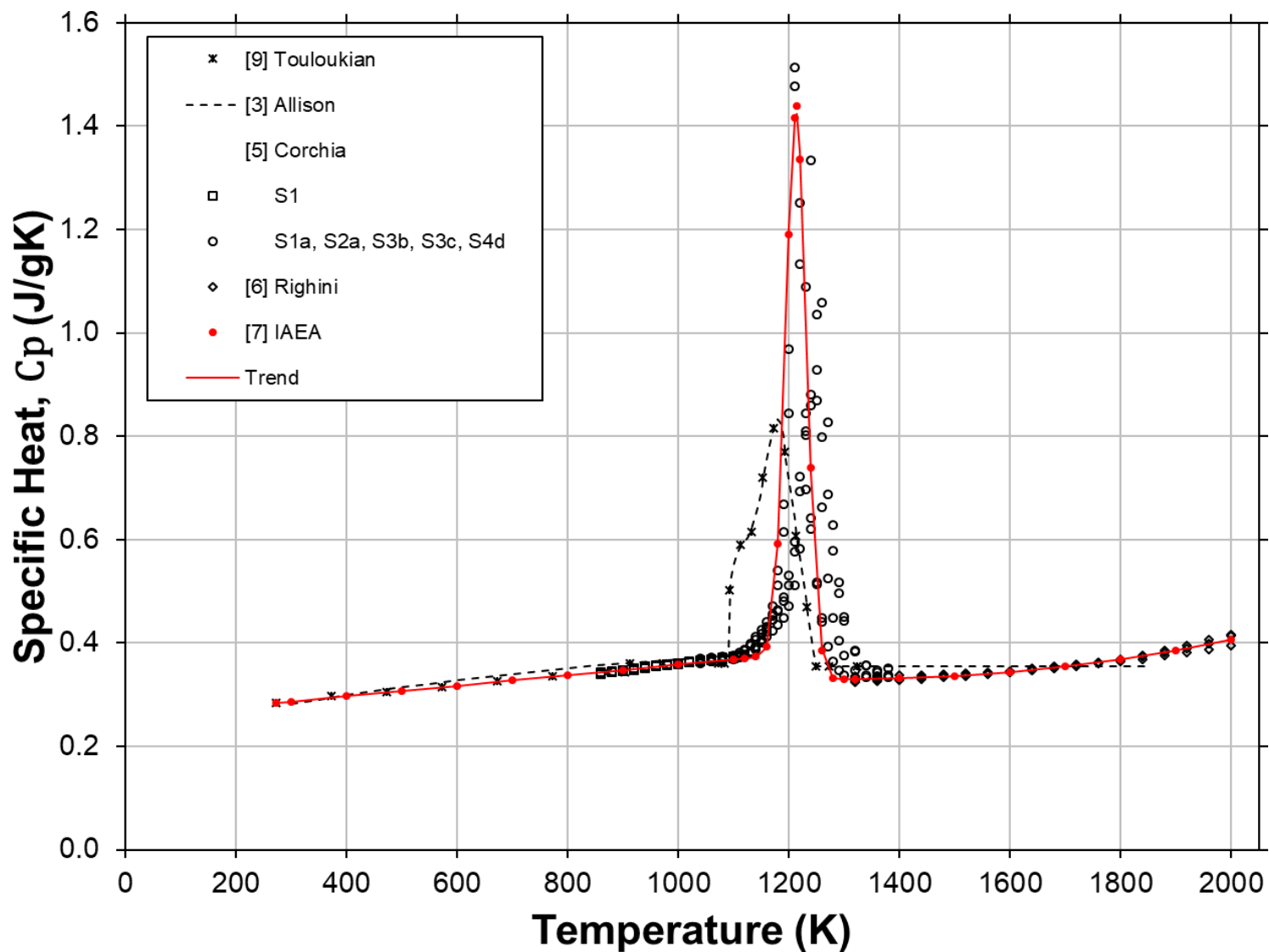


Figure 4.2.1-8: Specific Heat versus Temperature for Zircaloy.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 300               | ( 80.3 )   | 0.286                            | ( 0.068 )       | 1050              | ( 1430.3 ) | 0.363                            | ( 0.087 )       |
| 350               | ( 170.3 )  | 0.292                            | ( 0.070 )       | 1100              | ( 1520.3 ) | 0.368                            | ( 0.088 )       |
| 400               | ( 260.3 )  | 0.297                            | ( 0.071 )       | 1150              | ( 1610.3 ) | 0.375                            | ( 0.090 )       |
| 450               | ( 350.3 )  | 0.302                            | ( 0.072 )       | 1200              | ( 1700.3 ) | 1.198                            | ( 0.286 )       |
| 500               | ( 440.3 )  | 0.307                            | ( 0.073 )       | 1250              | ( 1790.3 ) | 0.502                            | ( 0.120 )       |
| 550               | ( 530.3 )  | 0.312                            | ( 0.075 )       | 1300              | ( 1880.3 ) | 0.330                            | ( 0.079 )       |
| 600               | ( 620.3 )  | 0.317                            | ( 0.076 )       | 1350              | ( 1970.3 ) | 0.330                            | ( 0.079 )       |
| 650               | ( 710.3 )  | 0.322                            | ( 0.077 )       | 1400              | ( 2060.3 ) | 0.332                            | ( 0.079 )       |
| 700               | ( 800.3 )  | 0.327                            | ( 0.078 )       | 1450              | ( 2150.3 ) | 0.333                            | ( 0.080 )       |
| 750               | ( 890.3 )  | 0.333                            | ( 0.079 )       | 1500              | ( 2240.3 ) | 0.336                            | ( 0.080 )       |
| 800               | ( 980.3 )  | 0.338                            | ( 0.081 )       | 1600              | ( 2420.3 ) | 0.344                            | ( 0.082 )       |
| 850               | ( 1070.3 ) | 0.343                            | ( 0.082 )       | 1700              | ( 2600.3 ) | 0.354                            | ( 0.085 )       |
| 900               | ( 1160.3 ) | 0.348                            | ( 0.083 )       | 1800              | ( 2780.3 ) | 0.368                            | ( 0.088 )       |
| 950               | ( 1250.3 ) | 0.353                            | ( 0.084 )       | 1900              | ( 2960.3 ) | 0.385                            | ( 0.092 )       |
| 1000              | ( 1340.3 ) | 0.358                            | ( 0.086 )       | 2000              | ( 3140.3 ) | 0.406                            | ( 0.097 )       |

**Application Notes:** Data for specific heat is collected from references [3, 5-7, 9] and fitted with the equations below to approximate the property trend with respect to temperature. Trend equations are shown for 273 to 1100 K, and 1320 to 2000 K. There is a phase change in this material between 1100 and 1320 K. An equation for this section is not given, however there is tabulated data for this range in the table.

## Fit Equation:

$$C_p(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

T = Temperature [K]

## Constants:

T Range [K]: 273 ≤ T ≤ 1100 1320 < T < 2000

|      |        |         |
|------|--------|---------|
| A0 = | 0.2557 | 0.5971  |
| A1 = | 0.1024 | -0.4088 |
| A2 = | 0      | 0.1565  |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Young's Modulus with Temperature

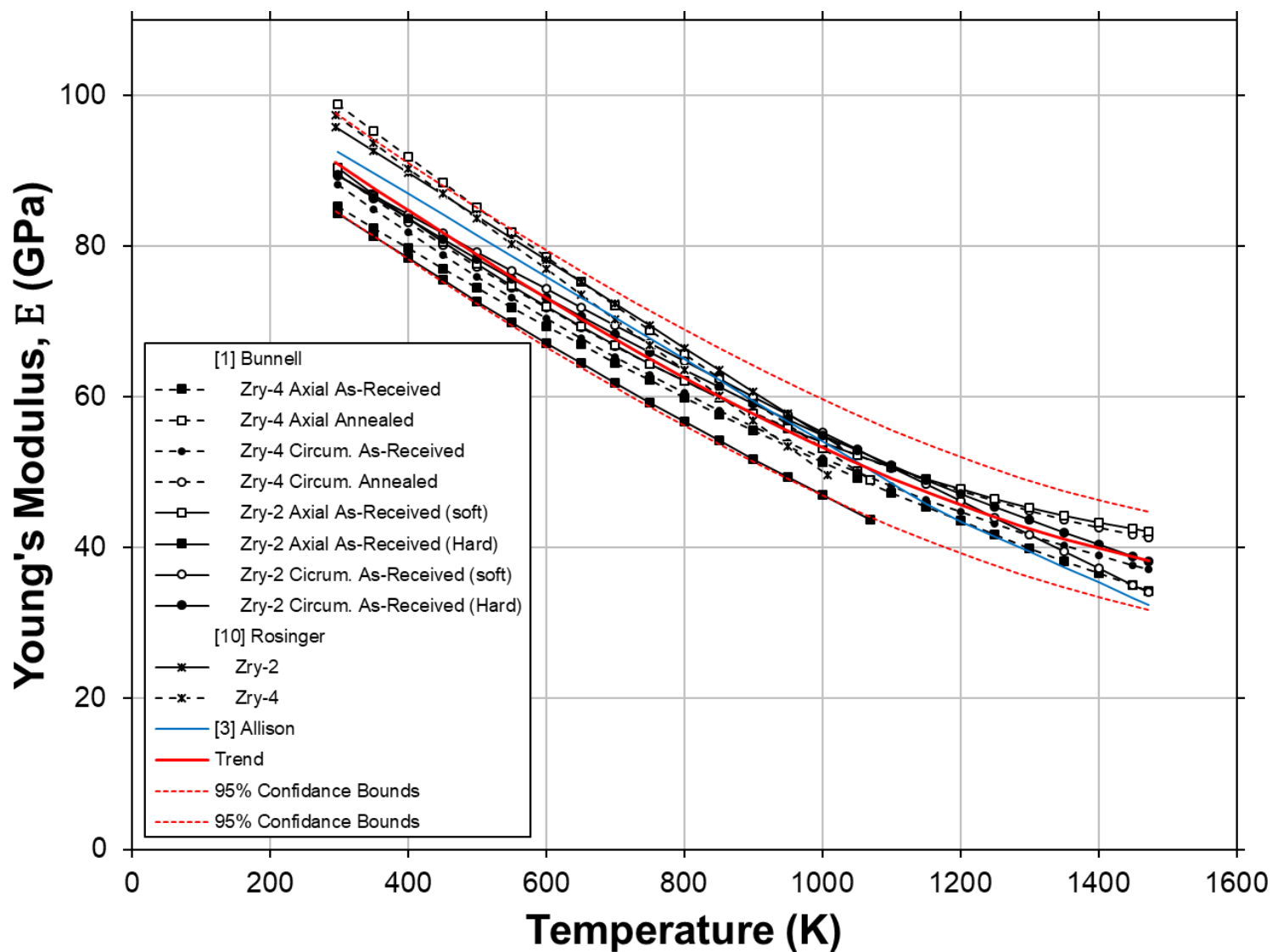


Figure 4.2.1-9: Young's Modulus versus Temperature of Zircaloy.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |          |
|-------------------|-----------|-----------------------|-----------|-------------------|------------|-----------------------|----------|
| K                 | ( °F )    | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )  |
| 300               | ( 80.3 )  | 90.83                 | ( 13.18 ) | 750               | ( 890.3 )  | 65.04                 | ( 9.44 ) |
| 325               | ( 125.3 ) | 89.28                 | ( 12.95 ) | 800               | ( 980.3 )  | 62.52                 | ( 9.07 ) |
| 350               | ( 170.3 ) | 87.74                 | ( 12.73 ) | 850               | ( 1070.3 ) | 60.08                 | ( 8.72 ) |
| 375               | ( 215.3 ) | 86.22                 | ( 12.51 ) | 900               | ( 1160.3 ) | 57.72                 | ( 8.38 ) |
| 400               | ( 260.3 ) | 84.70                 | ( 12.29 ) | 950               | ( 1250.3 ) | 55.46                 | ( 8.05 ) |
| 425               | ( 305.3 ) | 83.20                 | ( 12.07 ) | 1000              | ( 1340.3 ) | 53.29                 | ( 7.73 ) |
| 450               | ( 350.3 ) | 81.71                 | ( 11.86 ) | 1050              | ( 1430.3 ) | 51.22                 | ( 7.43 ) |
| 475               | ( 395.3 ) | 80.23                 | ( 11.64 ) | 1100              | ( 1520.3 ) | 49.26                 | ( 7.15 ) |
| 500               | ( 440.3 ) | 78.77                 | ( 11.43 ) | 1150              | ( 1610.3 ) | 47.41                 | ( 6.88 ) |
| 525               | ( 485.3 ) | 77.32                 | ( 11.22 ) | 1200              | ( 1700.3 ) | 45.66                 | ( 6.63 ) |
| 550               | ( 530.3 ) | 75.89                 | ( 11.01 ) | 1250              | ( 1790.3 ) | 44.04                 | ( 6.39 ) |
| 575               | ( 575.3 ) | 74.48                 | ( 10.81 ) | 1300              | ( 1880.3 ) | 42.53                 | ( 6.17 ) |
| 600               | ( 620.3 ) | 73.08                 | ( 10.60 ) | 1350              | ( 1970.3 ) | 41.15                 | ( 5.97 ) |
| 650               | ( 710.3 ) | 70.33                 | ( 10.20 ) | 1400              | ( 2060.3 ) | 39.89                 | ( 5.79 ) |
| 700               | ( 800.3 ) | 67.65                 | ( 9.82 )  | 1450              | ( 2150.3 ) | 38.78                 | ( 5.63 ) |

**Application Notes:** Data for Young's Modulus is collected from references [1, 3, 10] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equation:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]: 295 < T ≤ 1473

A0 = 110.1

A1 = -65.75

A2 = 3.43

A3 = 5.511



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Shear Modulus with Temperature

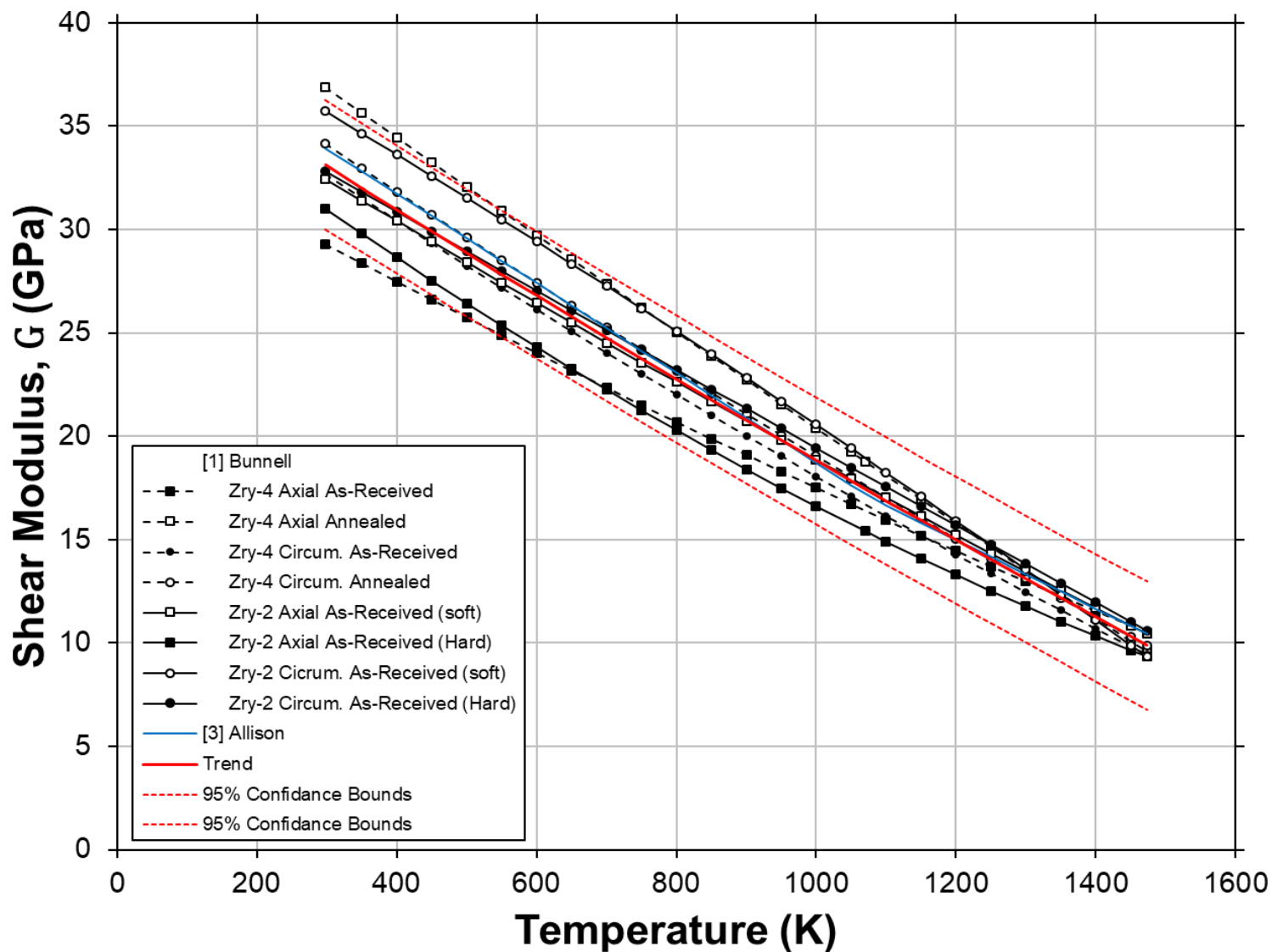


Figure 4.2.1-10: Shear Modulus versus Temperature of Zircaloy.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Shear Modulus ( G ) |          | Temperature ( T ) |            | Shear Modulus ( G ) |          |
|-------------------|-----------|---------------------|----------|-------------------|------------|---------------------|----------|
| K                 | ( °F )    | GPa                 | ( Msi )  | K                 | ( °F )     | GPa                 | ( Msi )  |
| 300               | ( 80.3 )  | 33.08               | ( 4.80 ) | 750               | ( 890.3 )  | 23.77               | ( 3.45 ) |
| 325               | ( 125.3 ) | 32.55               | ( 4.72 ) | 800               | ( 980.3 )  | 22.77               | ( 3.30 ) |
| 350               | ( 170.3 ) | 32.02               | ( 4.65 ) | 850               | ( 1070.3 ) | 21.77               | ( 3.16 ) |
| 375               | ( 215.3 ) | 31.49               | ( 4.57 ) | 900               | ( 1160.3 ) | 20.78               | ( 3.02 ) |
| 400               | ( 260.3 ) | 30.97               | ( 4.49 ) | 950               | ( 1250.3 ) | 19.80               | ( 2.87 ) |
| 425               | ( 305.3 ) | 30.44               | ( 4.42 ) | 1000              | ( 1340.3 ) | 18.82               | ( 2.73 ) |
| 450               | ( 350.3 ) | 29.92               | ( 4.34 ) | 1050              | ( 1430.3 ) | 17.85               | ( 2.59 ) |
| 475               | ( 395.3 ) | 29.40               | ( 4.27 ) | 1100              | ( 1520.3 ) | 16.88               | ( 2.45 ) |
| 500               | ( 440.3 ) | 28.88               | ( 4.19 ) | 1150              | ( 1610.3 ) | 15.92               | ( 2.31 ) |
| 525               | ( 485.3 ) | 28.36               | ( 4.12 ) | 1200              | ( 1700.3 ) | 14.97               | ( 2.17 ) |
| 550               | ( 530.3 ) | 27.84               | ( 4.04 ) | 1250              | ( 1790.3 ) | 14.03               | ( 2.04 ) |
| 575               | ( 575.3 ) | 27.33               | ( 3.97 ) | 1300              | ( 1880.3 ) | 13.09               | ( 1.90 ) |
| 600               | ( 620.3 ) | 26.82               | ( 3.89 ) | 1350              | ( 1970.3 ) | 12.15               | ( 1.76 ) |
| 650               | ( 710.3 ) | 25.79               | ( 3.74 ) | 1400              | ( 2060.3 ) | 11.23               | ( 1.63 ) |
| 700               | ( 800.3 ) | 24.78               | ( 3.60 ) | 1450              | ( 2150.3 ) | 10.31               | ( 1.50 ) |

**Application Notes:** Data for shear modulus is collected from references [1, 3] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$G(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]: 295 < T ≤ 1473

A0 = 39.57

A1 = -22.01

A2 = 1.249

A3 = 0.008872

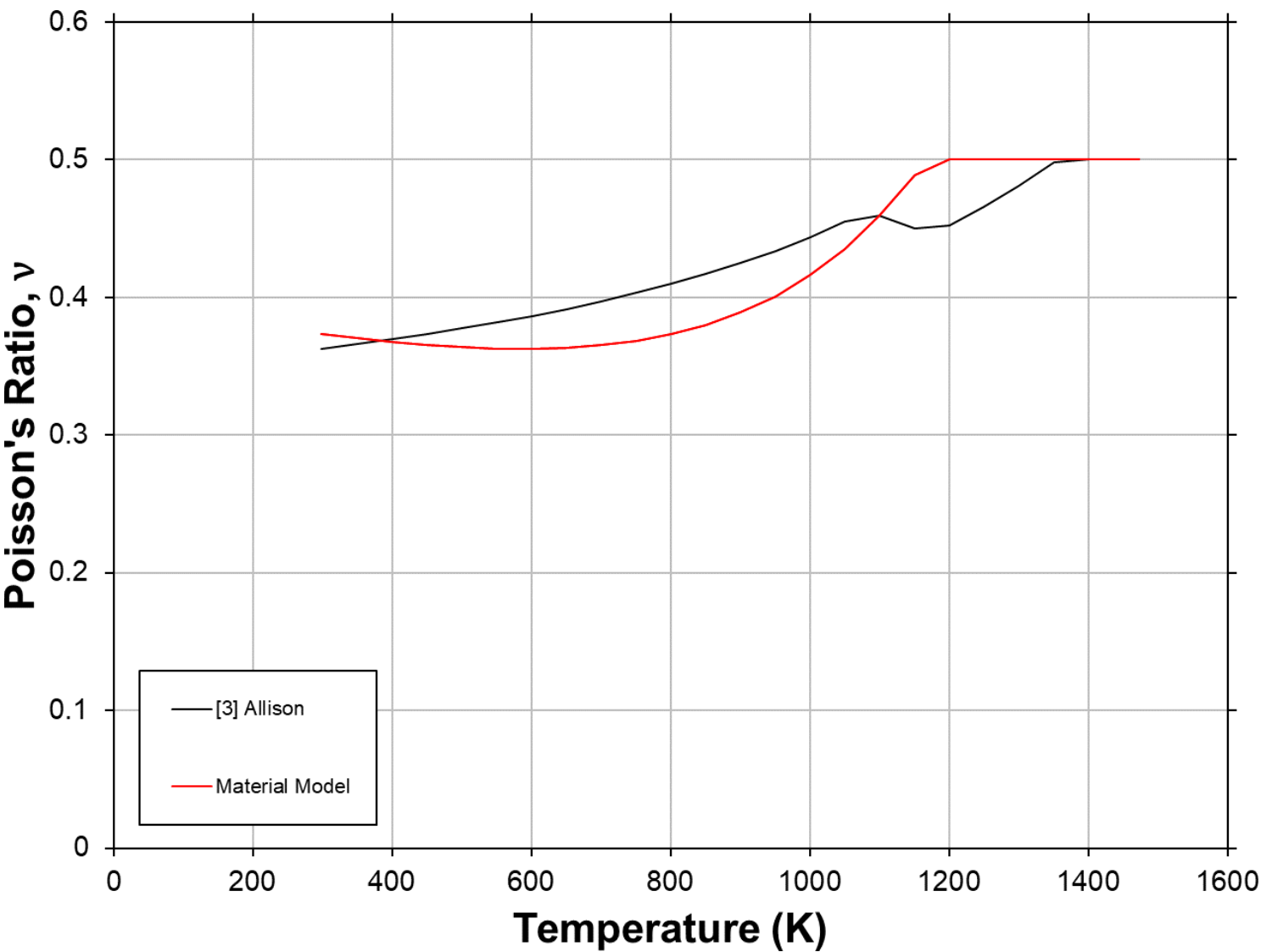


Figure 4.2.1-11: Poisson's Ratio versus Temperature of Zircaloy.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|-----------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )    |                           | K                 | ( °F )     |                           |
| 300               | ( 80.3 )  | 0.373                     | 750               | ( 890.3 )  | 0.368                     |
| 325               | ( 125.3 ) | 0.372                     | 800               | ( 980.3 )  | 0.373                     |
| 350               | ( 170.3 ) | 0.370                     | 850               | ( 1070.3 ) | 0.380                     |
| 375               | ( 215.3 ) | 0.369                     | 900               | ( 1160.3 ) | 0.389                     |
| 400               | ( 260.3 ) | 0.368                     | 950               | ( 1250.3 ) | 0.401                     |
| 425               | ( 305.3 ) | 0.367                     | 1000              | ( 1340.3 ) | 0.416                     |
| 450               | ( 350.3 ) | 0.365                     | 1050              | ( 1430.3 ) | 0.435                     |
| 475               | ( 395.3 ) | 0.365                     | 1100              | ( 1520.3 ) | 0.459                     |
| 500               | ( 440.3 ) | 0.364                     | 1150              | ( 1610.3 ) | 0.489                     |
| 525               | ( 485.3 ) | 0.363                     | 1200              | ( 1700.3 ) | 0.500                     |
| 550               | ( 530.3 ) | 0.363                     | 1250              | ( 1790.3 ) | 0.500                     |
| 575               | ( 575.3 ) | 0.363                     | 1300              | ( 1880.3 ) | 0.500                     |
| 600               | ( 620.3 ) | 0.363                     | 1350              | ( 1970.3 ) | 0.500                     |
| 650               | ( 710.3 ) | 0.363                     | 1400              | ( 2060.3 ) | 0.500                     |
| 700               | ( 800.3 ) | 0.365                     | 1450              | ( 2150.3 ) | 0.500                     |

**Application Notes:** Data for Poisson's Ratio is collected from reference [3]. Trend is calculated as a function of Young's and shear modulus as seen in the equation below to approximate the property trend with respect to temperature.

## Fit Equation:

$$\nu(T) = 0.5 \cdot (E(T)/G(T)) - 1$$

$$\nu(T) = \text{Poisson's Ratio}$$

$$T = \text{Temperature [K]}$$

## Constants:

$$T \text{ Range [K]: } \underline{295 < T \leq 1473}$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

Revision 2: 04-26-2023

Tabulated Property Data

| Temp.<br><br>( T )<br>K | Physical and Electrical                   |                                                          | Thermal                                      |                                                                                                     |      |                                                                                                  |      |                                                    | Elastic                             |                                   |                              |
|-------------------------|-------------------------------------------|----------------------------------------------------------|----------------------------------------------|-----------------------------------------------------------------------------------------------------|------|--------------------------------------------------------------------------------------------------|------|----------------------------------------------------|-------------------------------------|-----------------------------------|------------------------------|
|                         | Density<br><br>( ρ )<br>kg/m <sup>3</sup> | Electrical Resistivity<br><br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal Conductivity<br><br>( k )<br>W/(m-K) | Thermal Expansion<br><br>Axial<br>( dL/L <sub>0</sub> )<br>%<br>Diam.<br>( dL/L <sub>0</sub> )<br>% |      | Mean CTE<br><br>Axial<br>( ᾱ )<br>10 <sup>-6</sup> /K<br>Diam.<br>( ᾱ )<br>10 <sup>-6</sup> /K |      | Specific Heat<br><br>( C <sub>p</sub> )<br>J/(g-K) | Young's Modulus<br><br>( E )<br>GPa | Shear Modulus<br><br>( G )<br>GPa | Poisson's Ratio<br><br>( ν ) |
| 300                     | 6551                                      | 0.732                                                    | 13.41                                        | 0.00                                                                                                | 0.00 | 3.52                                                                                             | 7.50 | 0.286                                              | 90.8                                | 33.1                              | 0.37                         |
| 350                     | 6544                                      | 0.806                                                    | 13.68                                        | 0.02                                                                                                | 0.04 | 3.76                                                                                             | 7.50 | 0.292                                              | 87.7                                | 32.0                              | 0.37                         |
| 400                     | 6536                                      | 0.876                                                    | 13.99                                        | 0.04                                                                                                | 0.08 | 3.97                                                                                             | 7.50 | 0.297                                              | 84.7                                | 31.0                              | 0.37                         |
| 450                     | 6529                                      | 0.941                                                    | 14.34                                        | 0.07                                                                                                | 0.12 | 4.17                                                                                             | 7.50 | 0.302                                              | 81.7                                | 29.9                              | 0.37                         |
| 500                     | 6522                                      | 1.002                                                    | 14.74                                        | 0.09                                                                                                | 0.15 | 4.35                                                                                             | 7.50 | 0.307                                              | 78.8                                | 28.9                              | 0.36                         |
| 550                     | 6514                                      | 1.058                                                    | 15.19                                        | 0.12                                                                                                | 0.19 | 4.51                                                                                             | 7.50 | 0.312                                              | 75.9                                | 27.8                              | 0.36                         |
| 600                     | 6507                                      | 1.109                                                    | 15.67                                        | 0.14                                                                                                | 0.23 | 4.65                                                                                             | 7.50 | 0.317                                              | 73.1                                | 26.8                              | 0.36                         |
| 650                     | 6500                                      | 1.156                                                    | 16.21                                        | 0.17                                                                                                | 0.27 | 4.77                                                                                             | 7.50 | 0.322                                              | 70.3                                | 25.8                              | 0.36                         |
| 700                     | 6492                                      | 1.199                                                    | 16.79                                        | 0.20                                                                                                | 0.31 | 4.87                                                                                             | 7.50 | 0.327                                              | 67.6                                | 24.8                              | 0.37                         |
| 750                     | 6485                                      | 1.237                                                    | 17.41                                        | 0.23                                                                                                | 0.34 | 4.95                                                                                             | 7.50 | 0.333                                              | 65.0                                | 23.8                              | 0.37                         |
| 800                     | 6478                                      | 1.271                                                    | 18.08                                        | 0.25                                                                                                | 0.38 | 5.02                                                                                             | 7.50 | 0.338                                              | 62.5                                | 22.8                              | 0.37                         |
| 850                     | 6471                                      | 1.300                                                    | 18.79                                        | 0.28                                                                                                | 0.42 | 5.06                                                                                             | 7.50 | 0.343                                              | 60.1                                | 21.8                              | 0.38                         |
| 900                     | 6463                                      | 1.325                                                    | 19.55                                        | 0.31                                                                                                | 0.46 | 5.09                                                                                             | 7.50 | 0.348                                              | 57.7                                | 20.8                              | 0.39                         |
| 950                     | 6456                                      | 1.345                                                    | 20.36                                        | 0.33                                                                                                | 0.50 | 5.09                                                                                             | 7.50 | 0.353                                              | 55.5                                | 19.8                              | 0.40                         |
| 1000                    | 6449                                      | 1.361                                                    | 21.21                                        | 0.36                                                                                                | 0.53 | 5.08                                                                                             | 7.50 | 0.358                                              | 53.3                                | 18.8                              | 0.42                         |
| 1050                    | 6442                                      | 1.372                                                    | 22.10                                        | 0.38                                                                                                | 0.57 | 5.05                                                                                             | 7.50 | 0.363                                              | 51.2                                | 17.8                              | 0.44                         |
| 1100                    | 6438                                      | 1.379                                                    | 23.04                                        | 0.44                                                                                                |      | 5.08                                                                                             | 7.27 | 0.368                                              | 49.3                                | 16.9                              | 0.46                         |
| 1150                    | 6453                                      | 1.384                                                    | 24.02                                        | 0.46                                                                                                |      | 5.32                                                                                             | 5.95 | 0.375                                              | 47.4                                | 15.9                              | 0.49                         |
| 1200                    | 6472                                      | 1.339                                                    | 25.05                                        | 0.50                                                                                                |      | 5.54                                                                                             | 4.51 | 1.198                                              | 45.7                                | 15.0                              | 0.50                         |
| 1250                    | 6478                                      | 1.233                                                    | 26.12                                        | 0.55                                                                                                |      | 5.74                                                                                             | 3.92 | 0.502                                              | 44.0                                | 14.0                              | 0.50                         |
| 1300                    | 6469                                      | 1.183                                                    | 27.24                                        | 0.60                                                                                                |      | 5.92                                                                                             | 4.21 | 0.330                                              | 42.5                                | 13.1                              | 0.50                         |
| 1350                    | 6460                                      | 1.188                                                    | 28.40                                        | 0.64                                                                                                |      | 6.09                                                                                             | 4.47 | 0.330                                              | 41.1                                | 12.2                              | 0.50                         |
| 1400                    | 6450                                      | 1.195                                                    | 29.61                                        | 0.69                                                                                                |      | 6.24                                                                                             | 4.71 | 0.332                                              | 39.9                                | 11.2                              | 0.50                         |
| 1450                    | 6441                                      | 1.202                                                    | 30.86                                        | 0.74                                                                                                |      | 6.37                                                                                             | 4.93 | 0.333                                              | 38.8                                | 10.3                              | 0.50                         |
| 1500                    | 6432                                      | 1.209                                                    | 32.16                                        | 0.78                                                                                                |      | 6.49                                                                                             | 5.13 | 0.336                                              |                                     |                                   |                              |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

4 Other Nonferrous Metals & Alloys

4.2 Zirconium Alloys

4.2.1 Zircaloy

**Revision 0: 08-05-2020**

**References**

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## **5 Ferrous Alloys**

### **5.1 Austenitic Stainless Steels**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304 (SS304)

Revision 2.1: 08-25-2023

General

## Room Temperature Properties

|                                          |       |
|------------------------------------------|-------|
| Density, [kg/m <sup>3</sup> ]            | 8,018 |
| Melting Point, [K]                       | 1673  |
| Specific Heat, [J/(g-K)]                 | 0.476 |
| Thermal Conductivity, [W/(m-K)]          | 14.9  |
| Linear expansion coefficient, [μm/(m-K)] | 14.83 |
| Electrical resistivity, [μΩ-m]           | 0.72  |
| Young's Modulus, [GPa]                   | 197.0 |
| Shear Modulus, [GPa]                     | 77.8  |
| Poisson's Ratio, [-]                     | 0.265 |

## Composition

Table 5.1.1-1: Typical Composition ranges for 304-grade stainless steel (percent by weight)

| Grade       |      | Fe      | Cr   | Ni   | C    | Mn  | Si   | P     | S     | N    |
|-------------|------|---------|------|------|------|-----|------|-------|-------|------|
| <b>304</b>  | Min. | Balance | 18.0 | 8.0  | -    | -   | -    | -     | -     | -    |
|             | Max. |         | 20.0 | 10.5 | 0.08 | 2.0 | 0.75 | 0.045 | 0.030 | 0.10 |
| <b>304L</b> | Min. | Balance | 18.0 | 8.0  | -    | -   | -    | -     | -     | -    |
|             | Max. |         | 20.0 | 12   | 0.03 | 2.0 | 0.75 | 0.045 | 0.030 | 0.10 |
| <b>304H</b> | Min. | Balance | 18.0 | 8.0  | 0.04 | -   | -    | -     | -     | -    |
|             | Max. |         | 20.0 | 10.5 | 0.10 | 2.0 | 0.75 | 0.045 | 0.030 | -    |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Density with Temperature

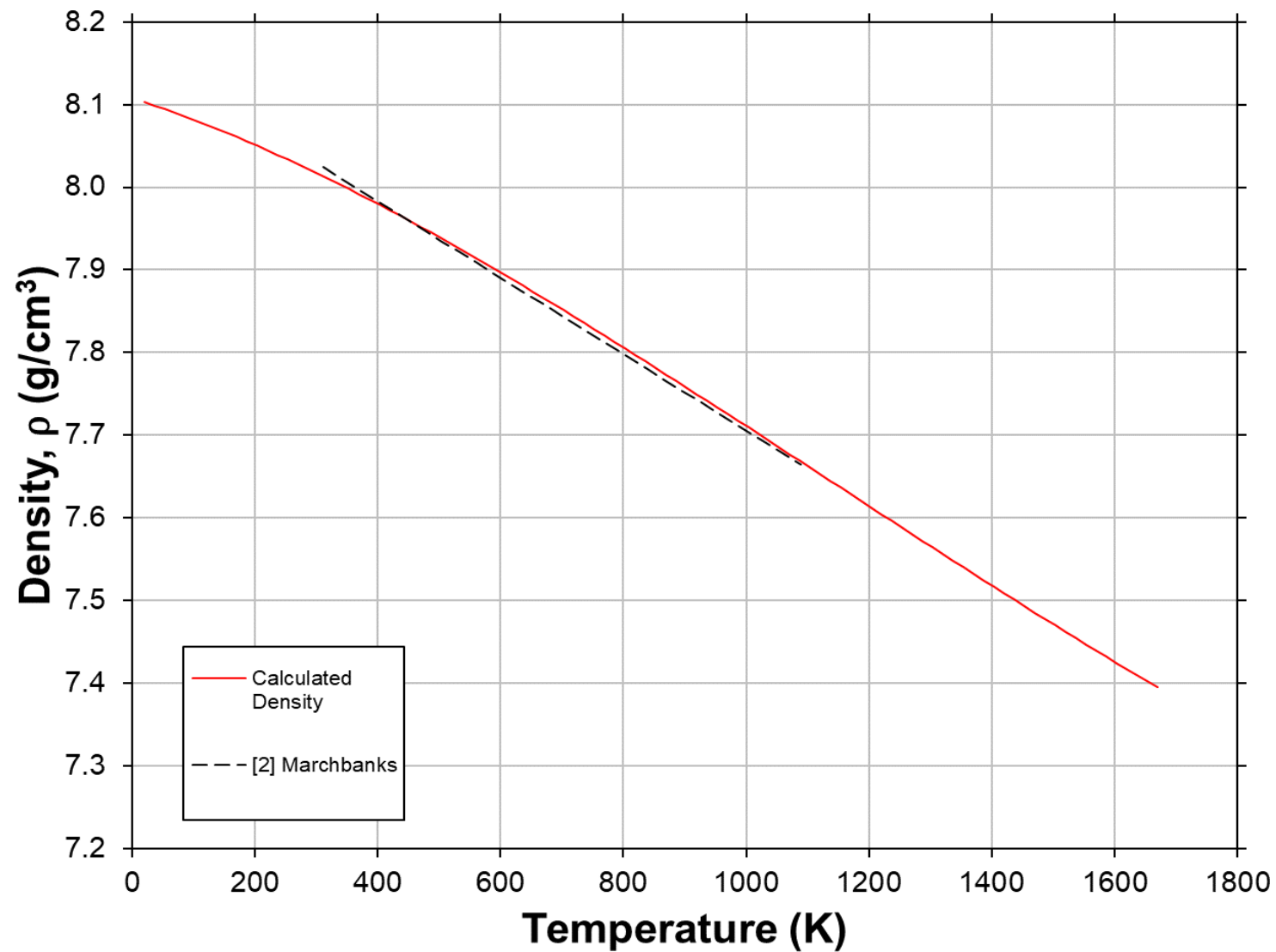


Figure 5.1.1-1: Density versus Temperature for SS304. Calculated from fitted trend of the Thermal Expansion data with comparison to Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Density with Temperature

## 100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 20                | ( -423.7 ) | 8103              | ( 505.9 )              | 700               | ( 800.3 )  | 7853              | ( 490.2 )              |
| 60                | ( -351.7 ) | 8093              | ( 505.2 )              | 750               | ( 890.3 )  | 7830              | ( 488.8 )              |
| 80                | ( -315.7 ) | 8087              | ( 504.9 )              | 800               | ( 980.3 )  | 7807              | ( 487.4 )              |
| 100               | ( -279.7 ) | 8082              | ( 504.5 )              | 850               | ( 1070.3 ) | 7783              | ( 485.9 )              |
| 150               | ( -189.7 ) | 8067              | ( 503.6 )              | 900               | ( 1160.3 ) | 7759              | ( 484.4 )              |
| 200               | ( -99.7 )  | 8051              | ( 502.6 )              | 950               | ( 1250.3 ) | 7735              | ( 482.9 )              |
| 250               | ( -9.7 )   | 8035              | ( 501.6 )              | 1000              | ( 1340.3 ) | 7711              | ( 481.4 )              |
| 300               | ( 80.3 )   | 8017              | ( 500.5 )              | 1050              | ( 1430.3 ) | 7687              | ( 479.9 )              |
| 350               | ( 170.3 )  | 7999              | ( 499.4 )              | 1100              | ( 1520.3 ) | 7663              | ( 478.4 )              |
| 400               | ( 260.3 )  | 7980              | ( 498.2 )              | 1200              | ( 1700.3 ) | 7614              | ( 475.3 )              |
| 450               | ( 350.3 )  | 7960              | ( 497.0 )              | 1300              | ( 1880.3 ) | 7566              | ( 472.3 )              |
| 500               | ( 440.3 )  | 7940              | ( 495.7 )              | 1400              | ( 2060.3 ) | 7518              | ( 469.3 )              |
| 550               | ( 530.3 )  | 7919              | ( 494.4 )              | 1500              | ( 2240.3 ) | 7471              | ( 466.4 )              |
| 600               | ( 620.3 )  | 7897              | ( 493.0 )              | 1600              | ( 2420.3 ) | 7426              | ( 463.6 )              |
| 650               | ( 710.3 )  | 7875              | ( 491.6 )              | 1670              | ( 2546.3 ) | 7395              | ( 461.7 )              |

**Application Notes:** Density is calculated here as a function of thermal expansion, as seen in the equation below to approximate the property trend with respect to temperature. The calculated trend is then compared against the density trend from reference [2].

### Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 8,020 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $20 \leq T \leq 1670$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

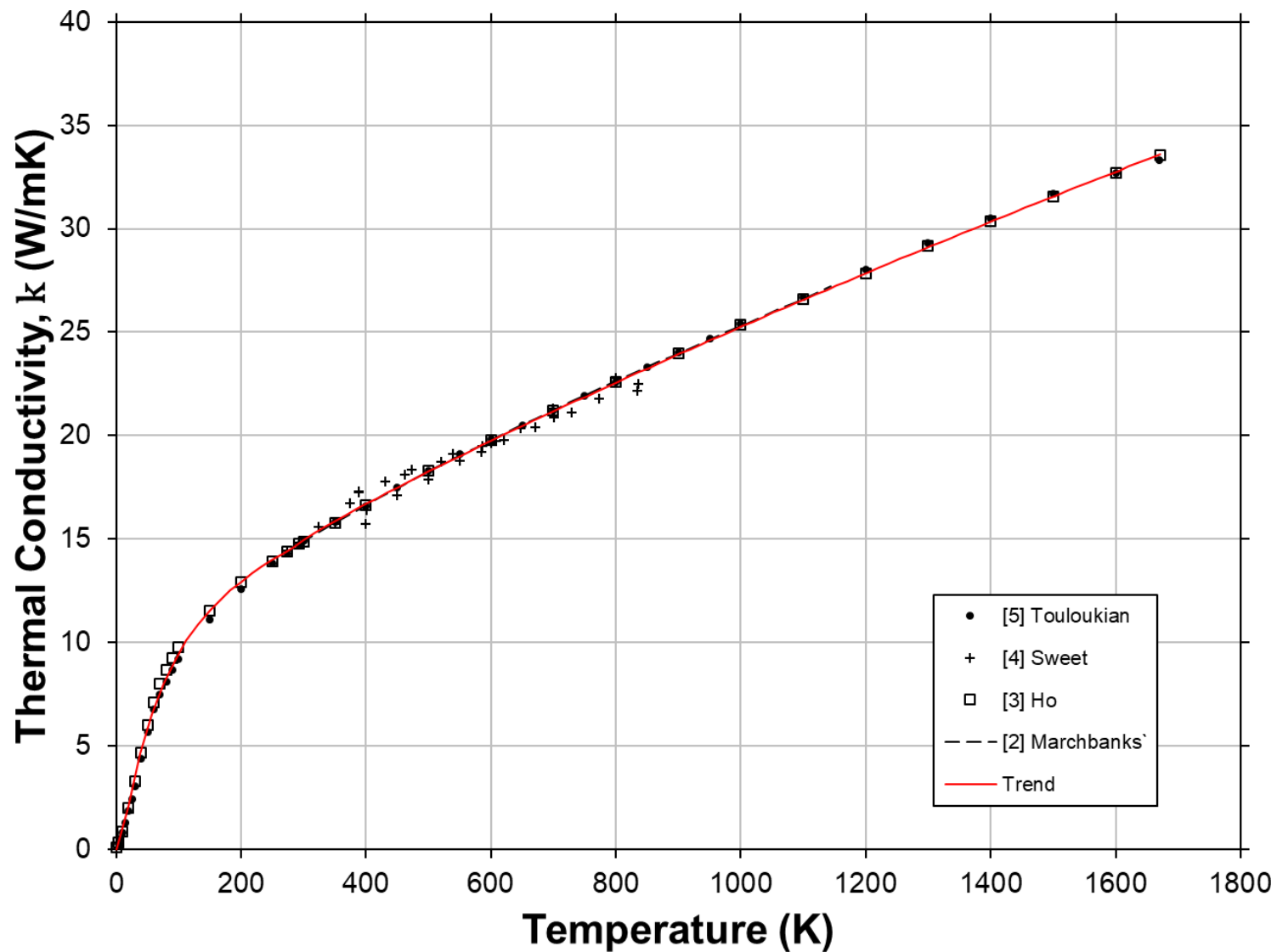


Figure 5.1.1-2: Thermal Conductivity versus Temperature of SS304 with comparison to Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m-K)                    | ((Btu-in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m-K)                    | ((Btu-in)/(ft <sup>2</sup> ·hr·°F)) |
| 20                | ( -423.7 ) | 2.10                       | ( 14.60 )                           | 650               | ( 710.3 )  | 20.46                      | ( 141.95 )                          |
| 60                | ( -351.7 ) | 6.82                       | ( 47.31 )                           | 700               | ( 800.3 )  | 21.17                      | ( 146.86 )                          |
| 80                | ( -315.7 ) | 8.32                       | ( 57.69 )                           | 750               | ( 890.3 )  | 21.87                      | ( 151.72 )                          |
| 100               | ( -279.7 ) | 9.47                       | ( 65.73 )                           | 800               | ( 980.3 )  | 22.56                      | ( 156.51 )                          |
| 150               | ( -189.7 ) | 11.53                      | ( 79.97 )                           | 850               | ( 1070.3 ) | 23.24                      | ( 161.25 )                          |
| 200               | ( -99.7 )  | 12.93                      | ( 89.71 )                           | 900               | ( 1160.3 ) | 23.92                      | ( 165.95 )                          |
| 250               | ( -9.7 )   | 13.99                      | ( 97.07 )                           | 1000              | ( 1340.3 ) | 25.25                      | ( 175.19 )                          |
| 300               | ( 80.3 )   | 14.95                      | ( 103.71 )                          | 1100              | ( 1520.3 ) | 26.56                      | ( 184.27 )                          |
| 350               | ( 170.3 )  | 15.86                      | ( 110.01 )                          | 1200              | ( 1700.3 ) | 27.84                      | ( 193.17 )                          |
| 400               | ( 260.3 )  | 16.70                      | ( 115.84 )                          | 1300              | ( 1880.3 ) | 29.10                      | ( 201.92 )                          |
| 450               | ( 350.3 )  | 17.49                      | ( 121.36 )                          | 1400              | ( 2060.3 ) | 30.34                      | ( 210.51 )                          |
| 500               | ( 440.3 )  | 18.26                      | ( 126.69 )                          | 1500              | ( 2240.3 ) | 31.56                      | ( 218.94 )                          |
| 550               | ( 530.3 )  | 19.01                      | ( 131.88 )                          | 1600              | ( 2420.3 ) | 32.75                      | ( 227.22 )                          |
| 600               | ( 620.3 )  | 19.74                      | ( 136.96 )                          | 1672              | ( 2549.9 ) | 33.60                      | ( 233.09 )                          |

**Application Notes:** The data for thermal conductivity is collected from references [2-5] and fitted with the equations below to approximate the property trend with respect to temperature.

## Fit Equation:

For temperature range:  $1 < T \leq 293$

$$k(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $293 < T \leq 1672$

$$k(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_2 / \left( \frac{T}{1000} \right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]  $1 < T \leq 293$

N = 2.199

A0 = 27080

A1 = 42.47

A2 = 1283

$293 < T \leq 1672$

B0 = 11.17

B1 = 15.21

B2 = -1.067

B\_2 = -0.06204



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Thermal Expansion with Temperature

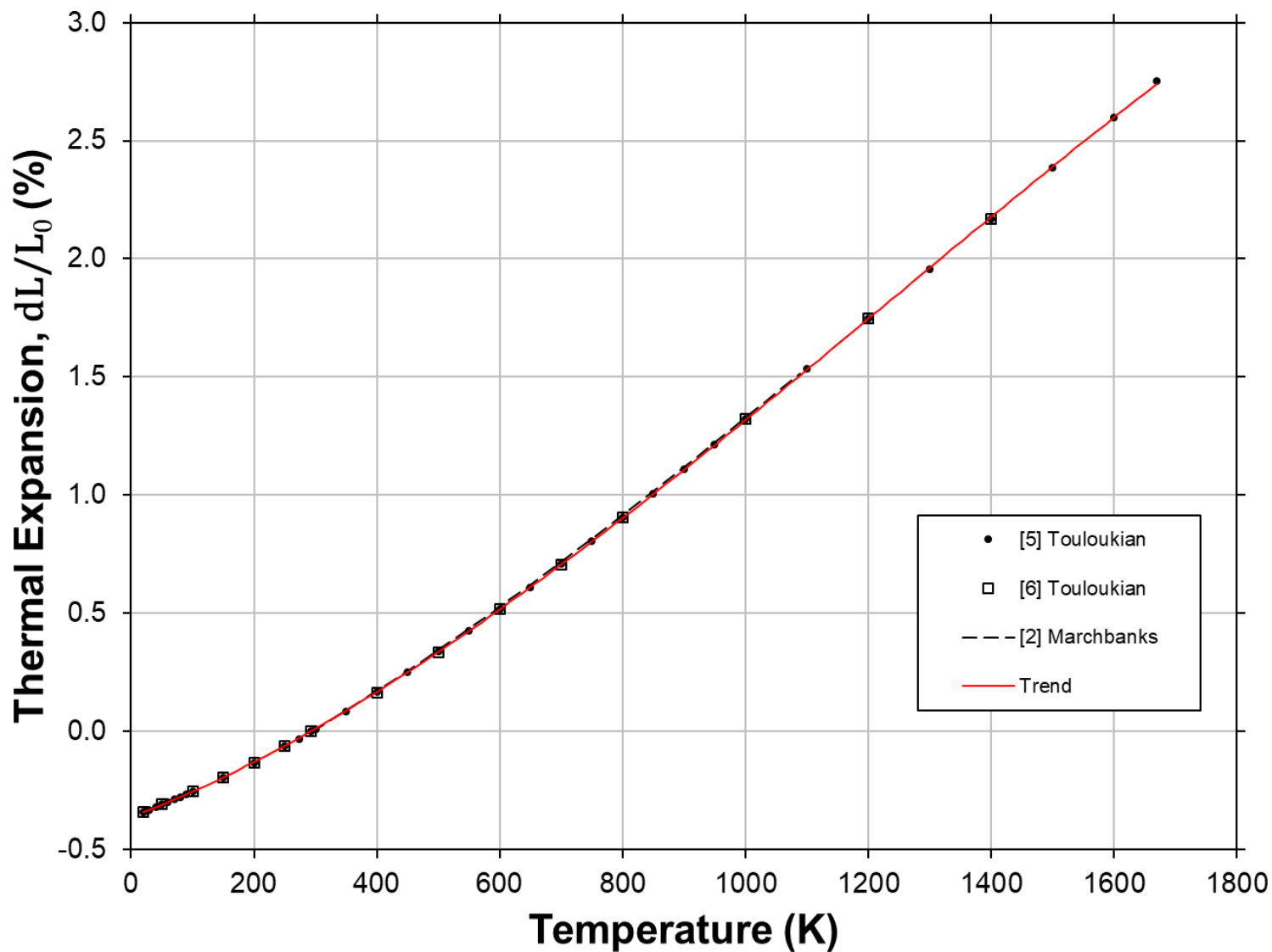


Figure 5.1.1-3: Thermal Expansion versus Temperature of SS304 with comparison to Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 20                | ( -423.7 ) | -0.343                                  | 700               | ( 800.3 )  | 0.705                                   |
| 60                | ( -351.7 ) | -0.301                                  | 750               | ( 890.3 )  | 0.804                                   |
| 80                | ( -315.7 ) | -0.278                                  | 800               | ( 980.3 )  | 0.903                                   |
| 100               | ( -279.7 ) | -0.255                                  | 850               | ( 1070.3 ) | 1.005                                   |
| 150               | ( -189.7 ) | -0.195                                  | 900               | ( 1160.3 ) | 1.108                                   |
| 200               | ( -99.7 )  | -0.130                                  | 950               | ( 1250.3 ) | 1.213                                   |
| 250               | ( -9.7 )   | -0.061                                  | 1000              | ( 1340.3 ) | 1.318                                   |
| 300               | ( 80.3 )   | 0.011                                   | 1050              | ( 1430.3 ) | 1.424                                   |
| 350               | ( 170.3 )  | 0.088                                   | 1100              | ( 1520.3 ) | 1.531                                   |
| 400               | ( 260.3 )  | 0.167                                   | 1200              | ( 1700.3 ) | 1.747                                   |
| 450               | ( 350.3 )  | 0.250                                   | 1300              | ( 1880.3 ) | 1.963                                   |
| 500               | ( 440.3 )  | 0.336                                   | 1400              | ( 2060.3 ) | 2.178                                   |
| 550               | ( 530.3 )  | 0.424                                   | 1500              | ( 2240.3 ) | 2.390                                   |
| 600               | ( 620.3 )  | 0.516                                   | 1600              | ( 2420.3 ) | 2.599                                   |
| 650               | ( 710.3 )  | 0.609                                   | 1670              | ( 2546.3 ) | 2.741                                   |

**Application Notes:** Data for thermal expansion was collected from references [2, 5, 6] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 20 \leq T \leq 1670$$

$$A0 = -0.363$$

$$A1 = 0.9848$$

$$A2 = 0.9543$$

$$A3 = -0.2581$$

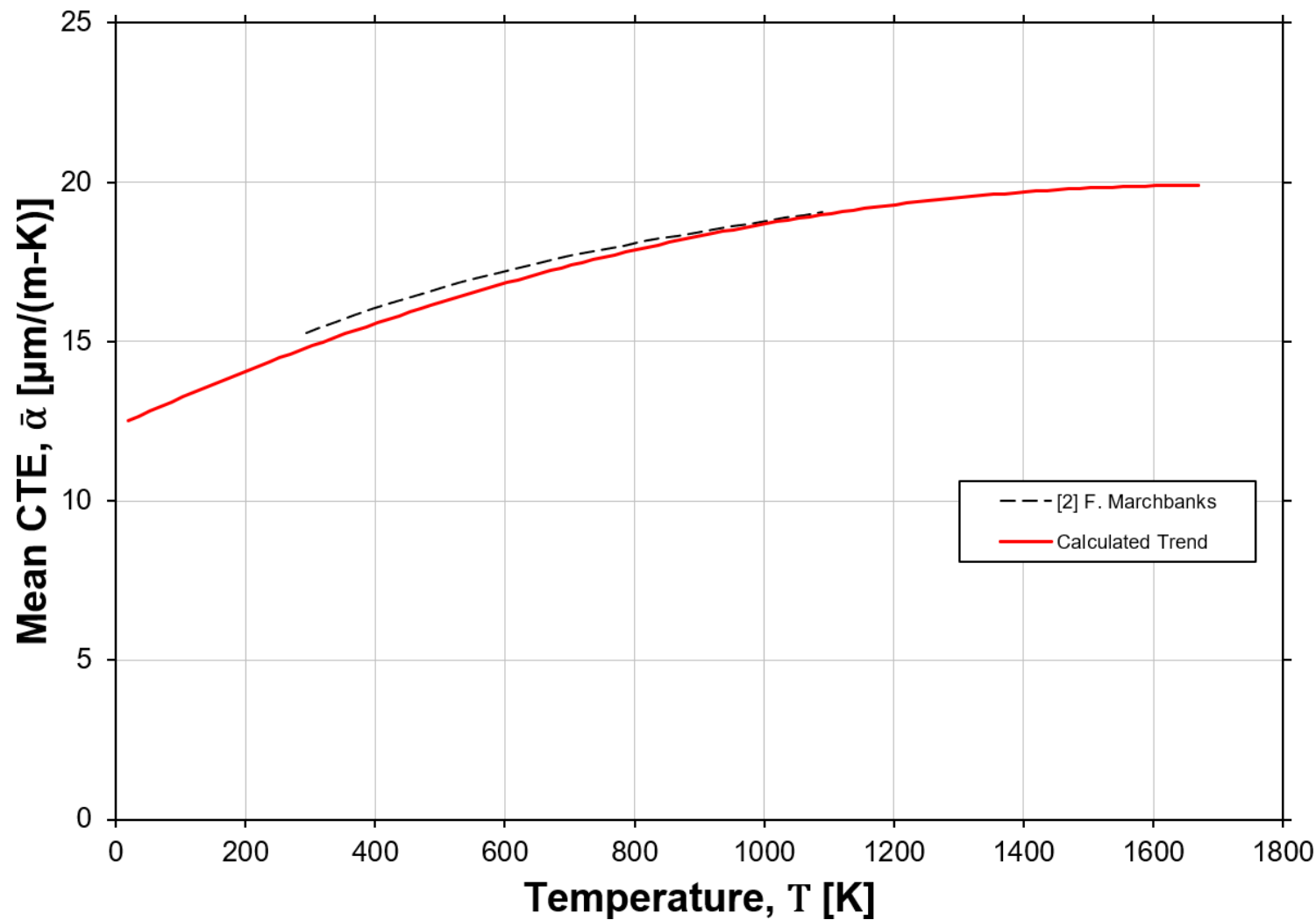


Figure 5.1.1-4: Mean Coefficient of Thermal Expansion versus Temperature of SS304. Calculated from fitted trend of the Thermal Expansion data for SS304 with comparison to Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 20                | ( -423.7 ) | 12.515                                | ( 6.953 )                                           | 700               | ( 800.3 )  | 17.378                                | ( 9.654 )                                           |
| 60                | ( -351.7 ) | 12.877                                | ( 7.154 )                                           | 750               | ( 890.3 )  | 17.630                                | ( 9.795 )                                           |
| 80                | ( -315.7 ) | 13.054                                | ( 7.252 )                                           | 800               | ( 980.3 )  | 17.869                                | ( 9.927 )                                           |
| 100               | ( -279.7 ) | 13.229                                | ( 7.350 )                                           | 850               | ( 1070.3 ) | 18.094                                | ( 10.052 )                                          |
| 150               | ( -189.7 ) | 13.656                                | ( 7.587 )                                           | 900               | ( 1160.3 ) | 18.305                                | ( 10.170 )                                          |
| 200               | ( -99.7 )  | 14.068                                | ( 7.815 )                                           | 950               | ( 1250.3 ) | 18.503                                | ( 10.280 )                                          |
| 250               | ( -9.7 )   | 14.464                                | ( 8.036 )                                           | 1000              | ( 1340.3 ) | 18.688                                | ( 10.382 )                                          |
| 300               | ( 80.3 )   | 14.846                                | ( 8.248 )                                           | 1050              | ( 1430.3 ) | 18.859                                | ( 10.477 )                                          |
| 350               | ( 170.3 )  | 15.213                                | ( 8.452 )                                           | 1100              | ( 1520.3 ) | 19.017                                | ( 10.565 )                                          |
| 400               | ( 260.3 )  | 15.566                                | ( 8.648 )                                           | 1200              | ( 1700.3 ) | 19.293                                | ( 10.718 )                                          |
| 450               | ( 350.3 )  | 15.903                                | ( 8.835 )                                           | 1300              | ( 1880.3 ) | 19.518                                | ( 10.843 )                                          |
| 500               | ( 440.3 )  | 16.227                                | ( 9.015 )                                           | 1400              | ( 2060.3 ) | 19.691                                | ( 10.940 )                                          |
| 550               | ( 530.3 )  | 16.536                                | ( 9.187 )                                           | 1500              | ( 2240.3 ) | 19.815                                | ( 11.008 )                                          |
| 600               | ( 620.3 )  | 16.831                                | ( 9.350 )                                           | 1600              | ( 2420.3 ) | 19.890                                | ( 11.050 )                                          |
| 650               | ( 710.3 )  | 17.111                                | ( 9.506 )                                           | 1670              | ( 2546.3 ) | 19.913                                | ( 11.063 )                                          |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated as a function of thermal expansion. Calculated values are then fitted with the equation below to approximate property trend with respect to temperature. This trend is compared against the property trend from reference [2].

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

T = Temperature [K]

## Constants:

T. Range [K]: 20 ≤ T ≤ 1670

A0 = 12.33

A1 = 9.298

A2 = -3.077

A3 = 0.1367



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Specific Heat with Temperature

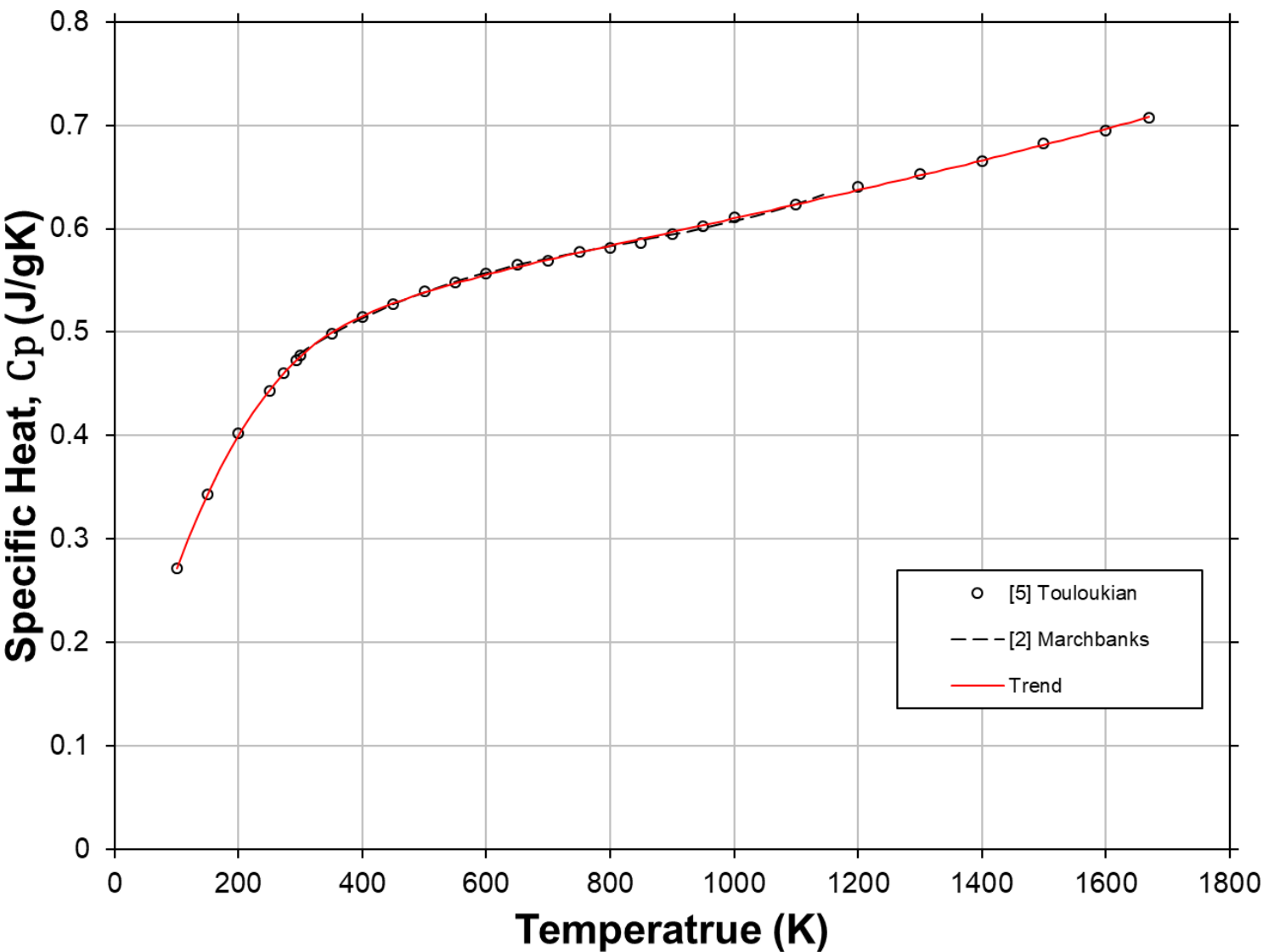


Figure 5.1.1-5: Specific Heat versus Temperature of SS304 with comparison to Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 100               | ( -279.7 ) | 0.272                            | ( 0.065 )       | 850               | ( 1070.3 ) | 0.590                            | ( 0.141 )       |
| 150               | ( -189.7 ) | 0.344                            | ( 0.082 )       | 900               | ( 1160.3 ) | 0.597                            | ( 0.143 )       |
| 200               | ( -99.7 )  | 0.401                            | ( 0.096 )       | 950               | ( 1250.3 ) | 0.603                            | ( 0.144 )       |
| 250               | ( -9.7 )   | 0.444                            | ( 0.106 )       | 1000              | ( 1340.3 ) | 0.610                            | ( 0.146 )       |
| 300               | ( 80.3 )   | 0.477                            | ( 0.114 )       | 1050              | ( 1430.3 ) | 0.617                            | ( 0.147 )       |
| 350               | ( 170.3 )  | 0.499                            | ( 0.119 )       | 1100              | ( 1520.3 ) | 0.623                            | ( 0.149 )       |
| 400               | ( 260.3 )  | 0.515                            | ( 0.123 )       | 1150              | ( 1610.3 ) | 0.630                            | ( 0.151 )       |
| 450               | ( 350.3 )  | 0.528                            | ( 0.126 )       | 1200              | ( 1700.3 ) | 0.637                            | ( 0.152 )       |
| 500               | ( 440.3 )  | 0.538                            | ( 0.129 )       | 1250              | ( 1790.3 ) | 0.644                            | ( 0.154 )       |
| 550               | ( 530.3 )  | 0.547                            | ( 0.131 )       | 1300              | ( 1880.3 ) | 0.651                            | ( 0.156 )       |
| 600               | ( 620.3 )  | 0.555                            | ( 0.133 )       | 1350              | ( 1970.3 ) | 0.659                            | ( 0.157 )       |
| 650               | ( 710.3 )  | 0.563                            | ( 0.135 )       | 1400              | ( 2060.3 ) | 0.666                            | ( 0.159 )       |
| 700               | ( 800.3 )  | 0.570                            | ( 0.136 )       | 1500              | ( 2240.3 ) | 0.681                            | ( 0.163 )       |
| 750               | ( 890.3 )  | 0.577                            | ( 0.138 )       | 1600              | ( 2420.3 ) | 0.697                            | ( 0.166 )       |
| 800               | ( 980.3 )  | 0.584                            | ( 0.139 )       | 1670              | ( 2546.3 ) | 0.708                            | ( 0.169 )       |

**Application Notes:** Data for specific heat is collected from references [2, 5] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$C_p(T) = A_0 + A_1 \cdot \left(\frac{T}{1000}\right) + A_2 \cdot \left(\frac{T}{1000}\right)^2 + A_3 \cdot \left(\frac{T}{1000}\right)^3 + A_2 / \left(\frac{T}{1000}\right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

| T. Range [K]: | <u>100 &lt; T ≤ 293</u> | <u>293 &lt; T ≤ 1670</u> |
|---------------|-------------------------|--------------------------|
| A0 =          | 0.07448                 | 0.5244                   |
| A1 =          | 2.368                   | 0.06317                  |
| A2 =          | -4.202                  | 0.02872                  |
| A3 =          | 2.584                   | 0                        |
| A_2 =         | 0                       | -0.006221                |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

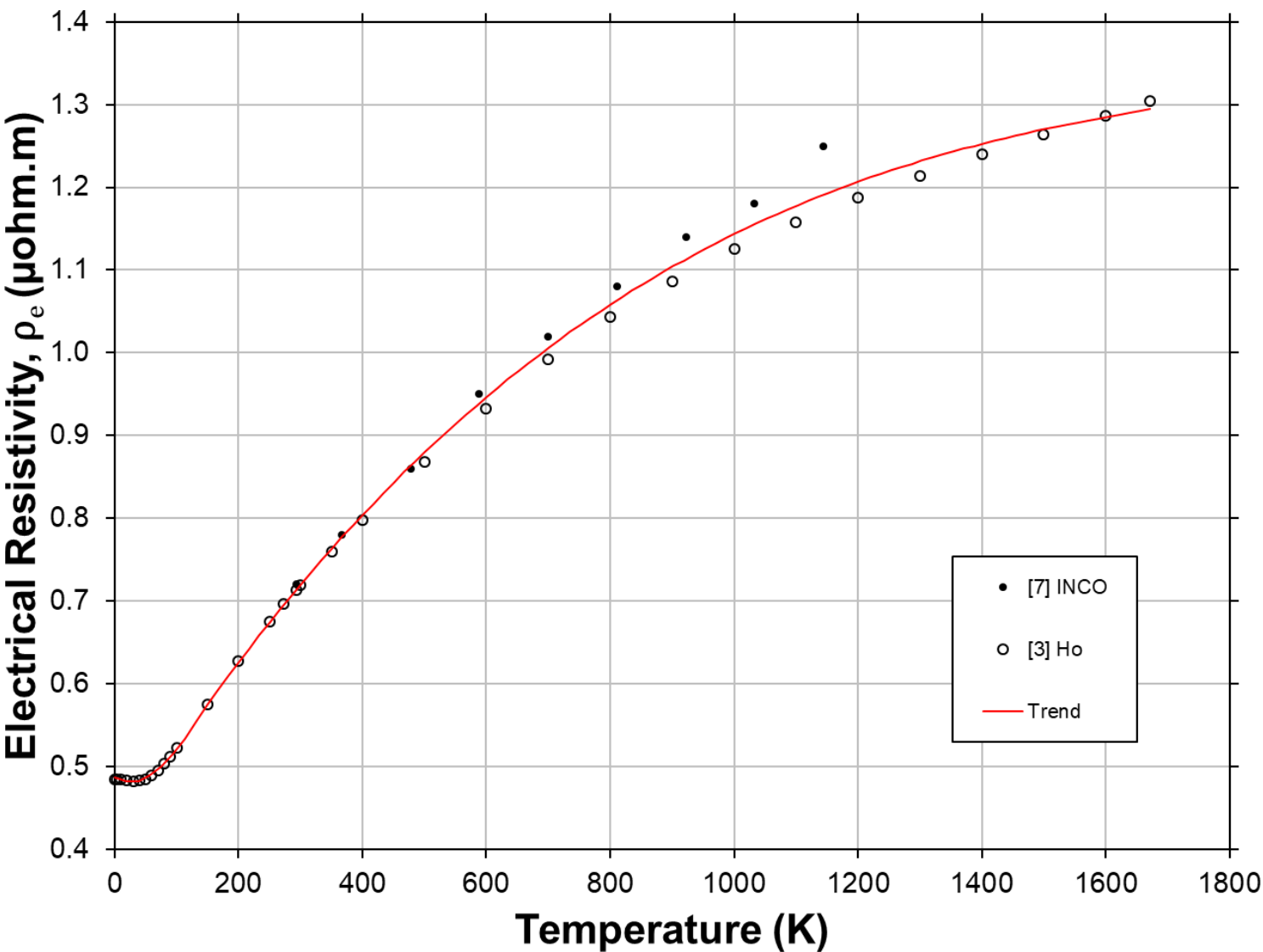


Figure 5.1.1-6: Electrical Resistivity versus Temperature of SS304.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 20                | ( -423.7 ) | 0.481                               | ( 18.95 )               | 700               | ( 800.3 )  | 1.006                               | ( 39.60 )               |
| 60                | ( -351.7 ) | 0.490                               | ( 19.31 )               | 750               | ( 890.3 )  | 1.033                               | ( 40.66 )               |
| 80                | ( -315.7 ) | 0.503                               | ( 19.82 )               | 800               | ( 980.3 )  | 1.058                               | ( 41.66 )               |
| 100               | ( -279.7 ) | 0.521                               | ( 20.49 )               | 850               | ( 1070.3 ) | 1.082                               | ( 42.59 )               |
| 150               | ( -189.7 ) | 0.575                               | ( 22.63 )               | 900               | ( 1160.3 ) | 1.104                               | ( 43.45 )               |
| 200               | ( -99.7 )  | 0.625                               | ( 24.63 )               | 950               | ( 1250.3 ) | 1.124                               | ( 44.26 )               |
| 250               | ( -9.7 )   | 0.674                               | ( 26.52 )               | 1000              | ( 1340.3 ) | 1.143                               | ( 45.01 )               |
| 300               | ( 80.3 )   | 0.719                               | ( 28.32 )               | 1050              | ( 1430.3 ) | 1.161                               | ( 45.71 )               |
| 350               | ( 170.3 )  | 0.763                               | ( 30.02 )               | 1100              | ( 1520.3 ) | 1.177                               | ( 46.36 )               |
| 400               | ( 260.3 )  | 0.804                               | ( 31.64 )               | 1200              | ( 1700.3 ) | 1.207                               | ( 47.51 )               |
| 450               | ( 350.3 )  | 0.842                               | ( 33.17 )               | 1300              | ( 1880.3 ) | 1.231                               | ( 48.48 )               |
| 500               | ( 440.3 )  | 0.879                               | ( 34.61 )               | 1400              | ( 2060.3 ) | 1.252                               | ( 49.31 )               |
| 550               | ( 530.3 )  | 0.914                               | ( 35.97 )               | 1500              | ( 2240.3 ) | 1.270                               | ( 50.00 )               |
| 600               | ( 620.3 )  | 0.946                               | ( 37.26 )               | 1600              | ( 2420.3 ) | 1.285                               | ( 50.59 )               |
| 650               | ( 710.3 )  | 0.977                               | ( 38.46 )               | 1672              | ( 2549.9 ) | 1.294                               | ( 50.96 )               |

**Application Notes:** Data for electrical resistivity is collected from references [3, 7] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

| T. Range [K]: | <u><math>1 &lt; T \leq 150</math></u> | <u><math>150 &lt; T \leq 1672</math></u> |
|---------------|---------------------------------------|------------------------------------------|
| A0 =          | 0.4873                                | 0.4075                                   |
| A1 =          | -0.4909                               | 1.197                                    |
| A2 =          | 10.31                                 | -0.5541                                  |
| A3 =          | -20.8                                 | 0.09297                                  |

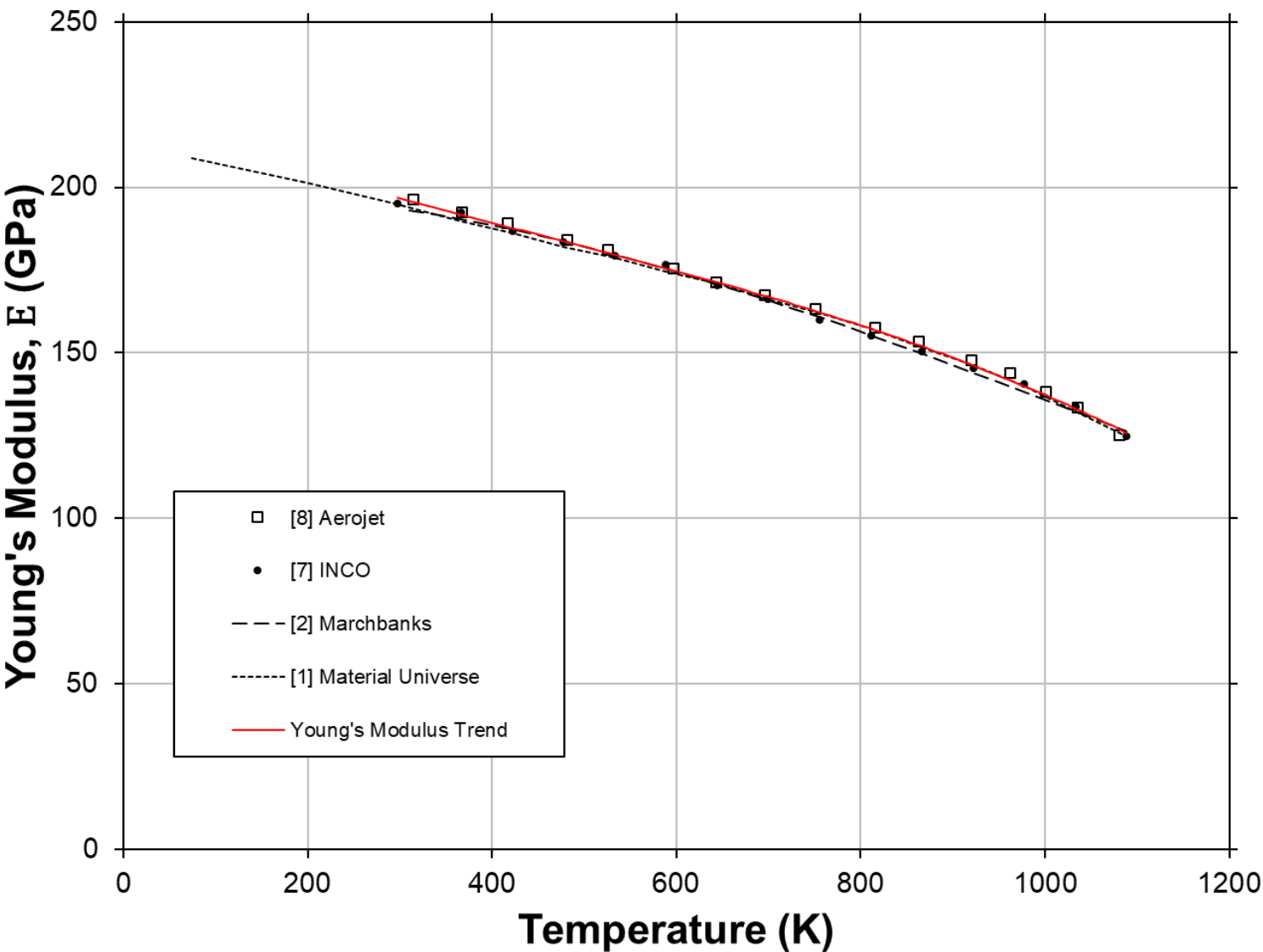


Figure 5.1.1-7: Young's Modulus versus Temperature of Inconel 718 with comparison to Marchbanks (1976) and Material Universe.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|-----------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )    | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 297               | ( 74.9 )  | 197.06                | ( 28.59 ) | 675               | ( 755.3 )  | 168.94                | ( 24.51 ) |
| 325               | ( 125.3 ) | 194.92                | ( 28.28 ) | 700               | ( 800.3 )  | 166.93                | ( 24.22 ) |
| 350               | ( 170.3 ) | 193.04                | ( 28.01 ) | 725               | ( 845.3 )  | 164.86                | ( 23.92 ) |
| 375               | ( 215.3 ) | 191.19                | ( 27.74 ) | 750               | ( 890.3 )  | 162.74                | ( 23.61 ) |
| 400               | ( 260.3 ) | 189.35                | ( 27.47 ) | 775               | ( 935.3 )  | 160.56                | ( 23.30 ) |
| 425               | ( 305.3 ) | 187.53                | ( 27.21 ) | 800               | ( 980.3 )  | 158.32                | ( 22.97 ) |
| 450               | ( 350.3 ) | 185.72                | ( 26.95 ) | 825               | ( 1025.3 ) | 156.00                | ( 22.64 ) |
| 475               | ( 395.3 ) | 183.91                | ( 26.69 ) | 850               | ( 1070.3 ) | 153.61                | ( 22.29 ) |
| 500               | ( 440.3 ) | 182.10                | ( 26.42 ) | 875               | ( 1115.3 ) | 151.13                | ( 21.93 ) |
| 525               | ( 485.3 ) | 180.28                | ( 26.16 ) | 900               | ( 1160.3 ) | 148.57                | ( 21.56 ) |
| 550               | ( 530.3 ) | 178.46                | ( 25.89 ) | 925               | ( 1205.3 ) | 145.91                | ( 21.17 ) |
| 575               | ( 575.3 ) | 176.61                | ( 25.63 ) | 950               | ( 1250.3 ) | 143.16                | ( 20.77 ) |
| 600               | ( 620.3 ) | 174.74                | ( 25.35 ) | 1000              | ( 1340.3 ) | 137.34                | ( 19.93 ) |
| 625               | ( 665.3 ) | 172.84                | ( 25.08 ) | 1050              | ( 1430.3 ) | 131.06                | ( 19.02 ) |
| 650               | ( 710.3 ) | 170.91                | ( 24.80 ) | 1089              | ( 1500.5 ) | 125.82                | ( 18.26 ) |

**Application Notes:** Data for Young's Modulus is collected from references [1, 2, 7, 8] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]: 297 < T ≤ 1089

A0 = 224.1

A1 = -109.8

A2 = 80.16

A3 = -57.12



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Shear Modulus with Temperature

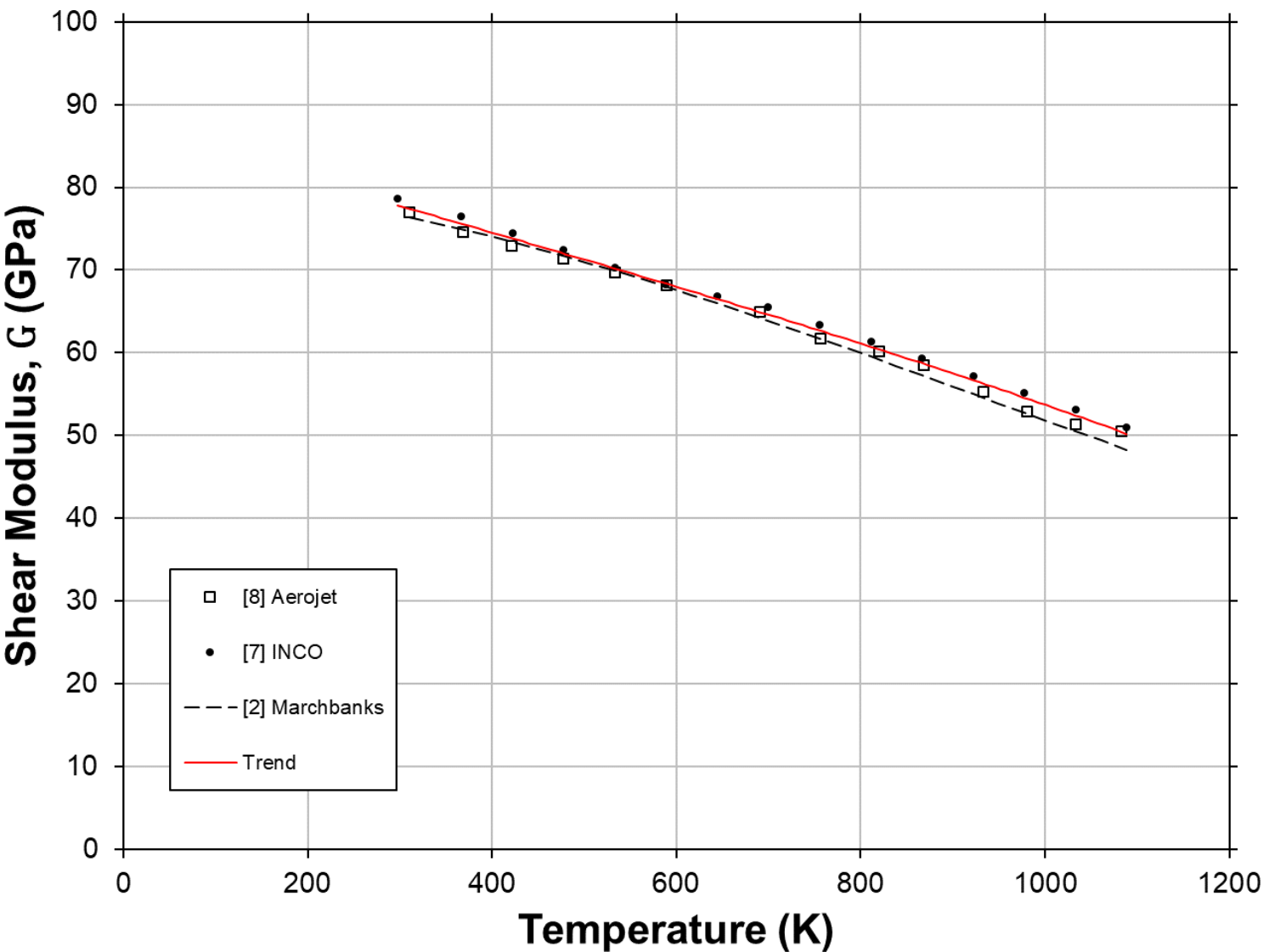


Figure 5.1.1-8: Shear Modulus versus Temperature of SS304 with comparison to Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Shear Modulus ( G ) |           | Temperature ( T ) |            | Shear Modulus ( G ) |          |
|-------------------|-----------|---------------------|-----------|-------------------|------------|---------------------|----------|
| K                 | ( °F )    | GPa                 | ( Msi )   | K                 | ( °F )     | GPa                 | ( Msi )  |
| 297               | ( 74.9 )  | 77.86               | ( 11.30 ) | 675               | ( 755.3 )  | 65.46               | ( 9.50 ) |
| 325               | ( 125.3 ) | 76.96               | ( 11.17 ) | 700               | ( 800.3 )  | 64.60               | ( 9.37 ) |
| 350               | ( 170.3 ) | 76.15               | ( 11.05 ) | 725               | ( 845.3 )  | 63.75               | ( 9.25 ) |
| 375               | ( 215.3 ) | 75.34               | ( 10.93 ) | 750               | ( 890.3 )  | 62.88               | ( 9.12 ) |
| 400               | ( 260.3 ) | 74.53               | ( 10.81 ) | 775               | ( 935.3 )  | 62.00               | ( 9.00 ) |
| 425               | ( 305.3 ) | 73.72               | ( 10.70 ) | 800               | ( 980.3 )  | 61.12               | ( 8.87 ) |
| 450               | ( 350.3 ) | 72.91               | ( 10.58 ) | 825               | ( 1025.3 ) | 60.23               | ( 8.74 ) |
| 475               | ( 395.3 ) | 72.10               | ( 10.46 ) | 850               | ( 1070.3 ) | 59.33               | ( 8.61 ) |
| 500               | ( 440.3 ) | 71.28               | ( 10.34 ) | 875               | ( 1115.3 ) | 58.42               | ( 8.48 ) |
| 525               | ( 485.3 ) | 70.46               | ( 10.22 ) | 900               | ( 1160.3 ) | 57.50               | ( 8.34 ) |
| 550               | ( 530.3 ) | 69.64               | ( 10.10 ) | 925               | ( 1205.3 ) | 56.58               | ( 8.21 ) |
| 575               | ( 575.3 ) | 68.81               | ( 9.98 )  | 950               | ( 1250.3 ) | 55.63               | ( 8.07 ) |
| 600               | ( 620.3 ) | 67.98               | ( 9.86 )  | 1000              | ( 1340.3 ) | 53.72               | ( 7.79 ) |
| 625               | ( 665.3 ) | 67.14               | ( 9.74 )  | 1050              | ( 1430.3 ) | 51.76               | ( 7.51 ) |
| 650               | ( 710.3 ) | 66.30               | ( 9.62 )  | 1089              | ( 1500.5 ) | 50.19               | ( 7.28 ) |

**Application Notes:** Data for shear modulus is collected from references [2, 7, 8] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$G(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]: 297 < T ≤ 1089

A0 = 87.66

A1 = -34.06

A2 = 5.073

A3 = -4.953

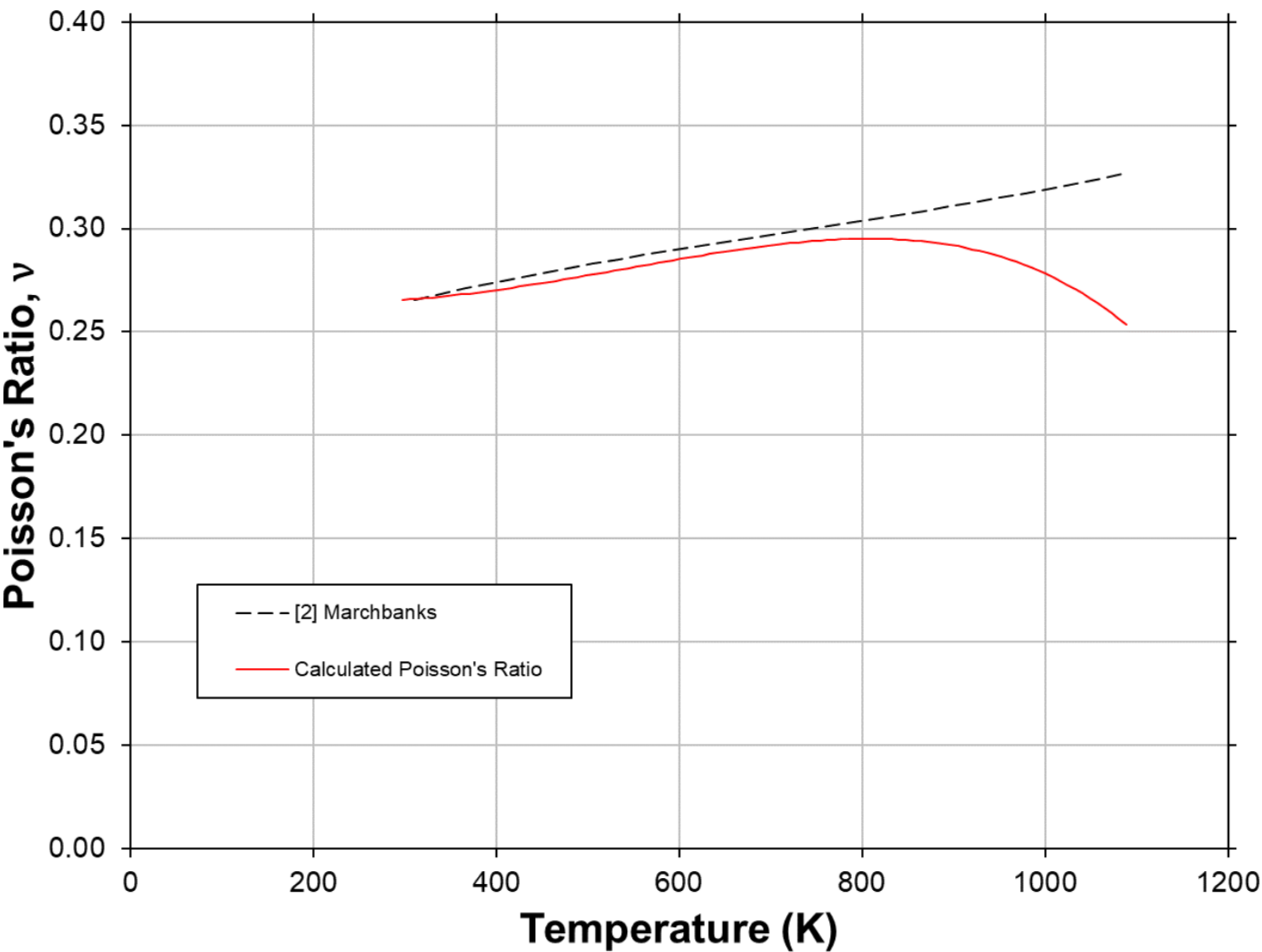


Figure 5.1.1-9: Poisson's Ratio versus Temperature of SS304. Calculated for the based Young's Modulus and Shear Modulus trends with comparison to Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|-----------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )    |                           | K                 | ( °F )     |                           |
| 297               | ( 74.9 )  | 0.265                     | 675               | ( 755.3 )  | 0.290                     |
| 325               | ( 125.3 ) | 0.266                     | 700               | ( 800.3 )  | 0.292                     |
| 350               | ( 170.3 ) | 0.268                     | 725               | ( 845.3 )  | 0.293                     |
| 375               | ( 215.3 ) | 0.269                     | 750               | ( 890.3 )  | 0.294                     |
| 400               | ( 260.3 ) | 0.270                     | 775               | ( 935.3 )  | 0.295                     |
| 425               | ( 305.3 ) | 0.272                     | 800               | ( 980.3 )  | 0.295                     |
| 450               | ( 350.3 ) | 0.274                     | 825               | ( 1025.3 ) | 0.295                     |
| 475               | ( 395.3 ) | 0.275                     | 850               | ( 1070.3 ) | 0.294                     |
| 500               | ( 440.3 ) | 0.277                     | 875               | ( 1115.3 ) | 0.293                     |
| 525               | ( 485.3 ) | 0.279                     | 900               | ( 1160.3 ) | 0.292                     |
| 550               | ( 530.3 ) | 0.281                     | 925               | ( 1205.3 ) | 0.290                     |
| 575               | ( 575.3 ) | 0.283                     | 950               | ( 1250.3 ) | 0.287                     |
| 600               | ( 620.3 ) | 0.285                     | 1000              | ( 1340.3 ) | 0.278                     |
| 625               | ( 665.3 ) | 0.287                     | 1050              | ( 1430.3 ) | 0.266                     |
| 650               | ( 710.3 ) | 0.289                     | 1089              | ( 1500.5 ) | 0.254                     |

**Application Notes:** Poisson's ratio is calculated as a function of Young's modulus and shear modulus, as seen in the equation below to approximate property trend with respect to temperature. This trend is then compared against property trend from reference [2].

## Fit Equations:

$$\nu(T) = E(T)/(2 \cdot G(T)) - 1$$

$$\nu(T) = \text{Poisson's Ratio}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $297 \leq T \leq 1089$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                          |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( α )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 8082                                  | 0.521                                                   | 9.47                                        | -0.255                                             | 13.23                                    | 0.272                                          | -                                  | -                                | -                           |
| 150                 | 8067                                  | 0.575                                                   | 11.53                                       | -0.195                                             | 13.66                                    | 0.344                                          | -                                  | -                                | -                           |
| 200                 | 8051                                  | 0.625                                                   | 12.93                                       | -0.130                                             | 14.07                                    | 0.401                                          | -                                  | -                                | -                           |
| 250                 | 8035                                  | 0.674                                                   | 13.99                                       | -0.061                                             | 14.46                                    | 0.444                                          | -                                  | -                                | -                           |
| 300                 | 8017                                  | 0.719                                                   | 14.95                                       | 0.011                                              | 14.85                                    | 0.477                                          | 196.83                             | 77.76                            | 0.27                        |
| 350                 | 7999                                  | 0.763                                                   | 15.86                                       | 0.088                                              | 15.21                                    | 0.499                                          | 193.04                             | 76.15                            | 0.27                        |
| 400                 | 7980                                  | 0.804                                                   | 16.70                                       | 0.167                                              | 15.57                                    | 0.515                                          | 189.35                             | 74.53                            | 0.27                        |
| 450                 | 7960                                  | 0.842                                                   | 17.49                                       | 0.250                                              | 15.90                                    | 0.528                                          | 185.72                             | 72.91                            | 0.27                        |
| 500                 | 7940                                  | 0.879                                                   | 18.26                                       | 0.336                                              | 16.23                                    | 0.538                                          | 182.10                             | 71.28                            | 0.28                        |
| 550                 | 7919                                  | 0.914                                                   | 19.01                                       | 0.424                                              | 16.54                                    | 0.547                                          | 178.46                             | 69.64                            | 0.28                        |
| 600                 | 7897                                  | 0.946                                                   | 19.74                                       | 0.516                                              | 16.83                                    | 0.555                                          | 174.74                             | 67.98                            | 0.29                        |
| 650                 | 7875                                  | 0.977                                                   | 20.46                                       | 0.609                                              | 17.11                                    | 0.563                                          | 170.91                             | 66.30                            | 0.29                        |
| 700                 | 7853                                  | 1.006                                                   | 21.17                                       | 0.705                                              | 17.38                                    | 0.570                                          | 166.93                             | 64.60                            | 0.29                        |
| 750                 | 7830                                  | 1.033                                                   | 21.87                                       | 0.804                                              | 17.63                                    | 0.577                                          | 162.74                             | 62.88                            | 0.29                        |
| 800                 | 7807                                  | 1.058                                                   | 22.56                                       | 0.903                                              | 17.87                                    | 0.584                                          | 158.32                             | 61.12                            | 0.30                        |
| 850                 | 7783                                  | 1.082                                                   | 23.24                                       | 1.005                                              | 18.09                                    | 0.590                                          | 153.61                             | 59.33                            | 0.29                        |
| 900                 | 7759                                  | 1.104                                                   | 23.92                                       | 1.108                                              | 18.31                                    | 0.597                                          | 148.57                             | 57.50                            | 0.29                        |
| 950                 | 7735                                  | 1.124                                                   | 24.59                                       | 1.213                                              | 18.50                                    | 0.603                                          | 143.16                             | 55.63                            | 0.29                        |
| 1000                | 7711                                  | 1.143                                                   | 25.25                                       | 1.318                                              | 18.69                                    | 0.610                                          | 137.34                             | 53.72                            | 0.28                        |
| 1100                | 7663                                  | 1.177                                                   | 26.56                                       | 1.531                                              | 19.02                                    | 0.623                                          | -                                  | -                                | -                           |
| 1200                | 7614                                  | 1.207                                                   | 27.84                                       | 1.747                                              | 19.29                                    | 0.637                                          | -                                  | -                                | -                           |
| 1300                | 7566                                  | 1.231                                                   | 29.10                                       | 1.963                                              | 19.52                                    | 0.651                                          | -                                  | -                                | -                           |
| 1400                | 7518                                  | 1.252                                                   | 30.34                                       | 2.178                                              | 19.69                                    | 0.666                                          | -                                  | -                                | -                           |
| 1500                | 7471                                  | 1.270                                                   | 31.56                                       | 2.390                                              | 19.82                                    | 0.681                                          | -                                  | -                                | -                           |
| 1600                | 7426                                  | 1.285                                                   | 32.75                                       | 2.599                                              | 19.89                                    | 0.697                                          | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.1 Stainless Steel 304

**Revision 0: 08-05-2020**

**References**

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- [2] F. Marchbanks, M.A. Moen, R.E. Irvin, Nuclear Systems Materials Handbook: Design Data, International Conference on the Mechanical Behavior of Materials, Oak Ridge National Laboratory, Boston, MA, USA, 1976.
- [3] C.Y. Ho, T.K. Chu, Electrical Resistivity and Thermal Conductivity of Nine Selected AISI stainless Steels, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1977.
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## **5 Ferrous Alloys**

### **5.1 Austenitic Stainless Steels**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347 (SS347)

Revision 2.1: 08-25-2023

General

## Room Temperature Properties

|                                           |              |
|-------------------------------------------|--------------|
| Theoretical Density, [kg/m <sup>3</sup> ] | 7,959        |
| Melting Point, [K]                        | 1650 to 1720 |
| Specific Heat, [J/(g-K)]                  | 0.455        |
| Thermal Conductivity, [W/(m-K)]           | 14.5         |
| Linear expansion coefficient, [μm/(m-K)]  | 15.69        |
| Electrical resistivity, [μΩ-m]            | 0.74         |
| Young's Modulus, [GPa]                    | 198.3        |
| Shear Modulus, [GPa]                      | 78.7         |
| Poisson's Ratio, [-]                      | 0.260        |

## Composition

Table 5.1.2-1: Typical Composition ranges for 347-grade stainless steel (percent by weight) [1].

| Grade |      | Fe      | Cr   | Ni   | C    | Mn  | Si   | P     | S     | Nb+Ta |
|-------|------|---------|------|------|------|-----|------|-------|-------|-------|
| 347   | Min. | Balance | 17.0 | 9.0  | -    | -   | -    | -     | -     | 10xC  |
|       | Max. |         | 19.0 | 13.0 | 0.08 | 2.0 | 0.75 | 0.045 | 0.030 | 1.0   |
| 347H  | Min. | Balance | 17.0 | 9.0  | 0.04 | -   | -    | -     | -     | 8xC   |
|       | Max. |         | 19.0 | 13.0 | 0.10 | 2.0 | 0.75 | 0.045 | 0.030 | 1.0   |



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Density with Temperature

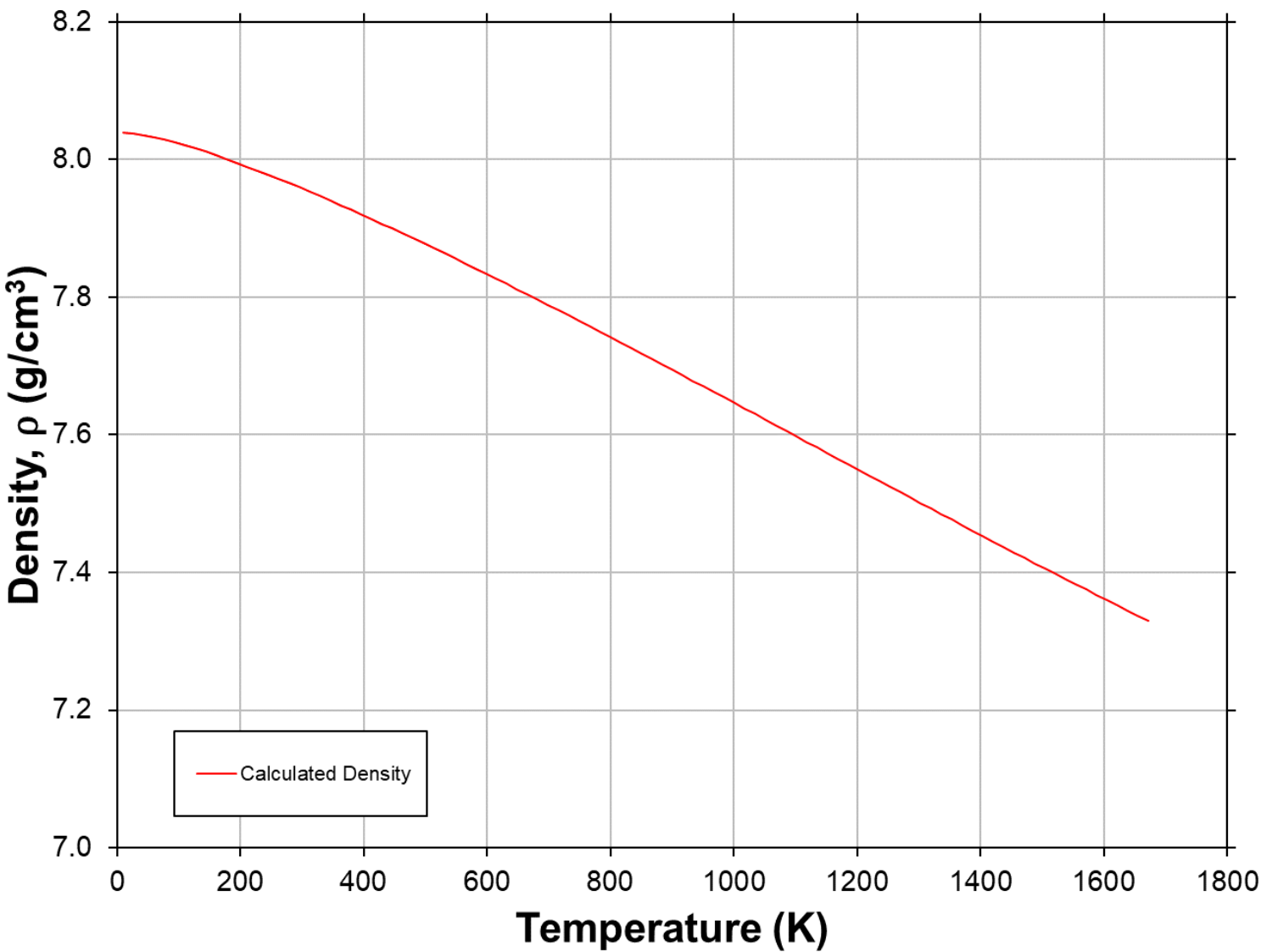


Figure 5.1.2-1: Density versus Temperature for SS347. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Density with Temperature

## 100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 20                | ( -423.7 ) | 8038              | ( 501.8 )              | 700               | ( 800.3 )  | 7788              | ( 486.2 )              |
| 60                | ( -351.7 ) | 8032              | ( 501.5 )              | 750               | ( 890.3 )  | 7765              | ( 484.8 )              |
| 80                | ( -315.7 ) | 8028              | ( 501.2 )              | 800               | ( 980.3 )  | 7742              | ( 483.3 )              |
| 100               | ( -279.7 ) | 8024              | ( 500.9 )              | 850               | ( 1070.3 ) | 7718              | ( 481.8 )              |
| 150               | ( -189.7 ) | 8009              | ( 500.0 )              | 900               | ( 1160.3 ) | 7694              | ( 480.4 )              |
| 200               | ( -99.7 )  | 7993              | ( 499.0 )              | 950               | ( 1250.3 ) | 7671              | ( 478.9 )              |
| 250               | ( -9.7 )   | 7975              | ( 497.9 )              | 1000              | ( 1340.3 ) | 7647              | ( 477.4 )              |
| 300               | ( 80.3 )   | 7958              | ( 496.8 )              | 1050              | ( 1430.3 ) | 7622              | ( 475.9 )              |
| 350               | ( 170.3 )  | 7939              | ( 495.6 )              | 1100              | ( 1520.3 ) | 7598              | ( 474.4 )              |
| 400               | ( 260.3 )  | 7919              | ( 494.4 )              | 1200              | ( 1700.3 ) | 7550              | ( 471.3 )              |
| 450               | ( 350.3 )  | 7898              | ( 493.1 )              | 1300              | ( 1880.3 ) | 7502              | ( 468.3 )              |
| 500               | ( 440.3 )  | 7877              | ( 491.8 )              | 1400              | ( 2060.3 ) | 7454              | ( 465.4 )              |
| 550               | ( 530.3 )  | 7855              | ( 490.4 )              | 1500              | ( 2240.3 ) | 7408              | ( 462.5 )              |
| 600               | ( 620.3 )  | 7833              | ( 489.0 )              | 1600              | ( 2420.3 ) | 7362              | ( 459.6 )              |
| 650               | ( 710.3 )  | 7811              | ( 487.6 )              | 1672              | ( 2549.9 ) | 7330              | ( 457.6 )              |

**Application Notes:** Density trend is calculated as a function of thermal expansion as seen in the equation below to approximate property trend with respect to temperature.

### Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 7,960 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $10 \leq T \leq 1672$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

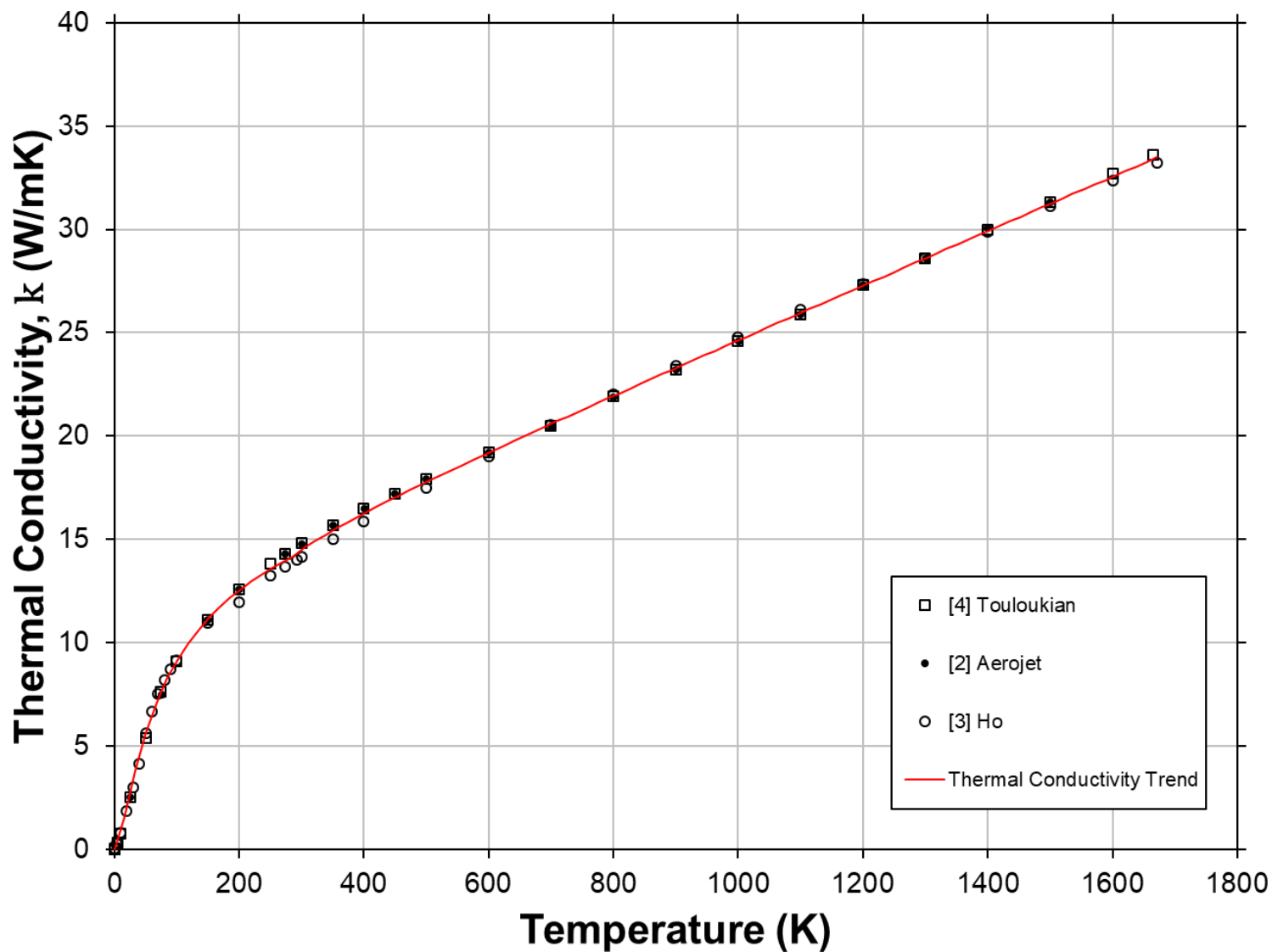


Figure 5.1.2-2: Thermal Conductivity versus Temperature of SS347. Displaying fitted trend of the data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       |
|-------------------|------------|----------------------------|---------------------------------------|-------------------|------------|----------------------------|---------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) |
| 20                | ( -423.7 ) | 1.98                       | ( 13.72 )                             | 600               | ( 620.3 )  | 19.20                      | ( 133.23 )                            |
| 60                | ( -351.7 ) | 6.48                       | ( 44.97 )                             | 700               | ( 800.3 )  | 20.58                      | ( 142.82 )                            |
| 80                | ( -315.7 ) | 7.95                       | ( 55.16 )                             | 800               | ( 980.3 )  | 21.94                      | ( 152.25 )                            |
| 100               | ( -279.7 ) | 9.10                       | ( 63.11 )                             | 900               | ( 1160.3 ) | 23.29                      | ( 161.58 )                            |
| 150               | ( -189.7 ) | 11.14                      | ( 77.26 )                             | 1000              | ( 1340.3 ) | 24.62                      | ( 170.84 )                            |
| 200               | ( -99.7 )  | 12.53                      | ( 86.91 )                             | 1100              | ( 1520.3 ) | 25.95                      | ( 180.07 )                            |
| 250               | ( -9.7 )   | 13.57                      | ( 94.15 )                             | 1200              | ( 1700.3 ) | 27.28                      | ( 189.26 )                            |
| 300               | ( 80.3 )   | 14.51                      | ( 100.67 )                            | 1300              | ( 1880.3 ) | 28.60                      | ( 198.42 )                            |
| 350               | ( 170.3 )  | 15.43                      | ( 107.05 )                            | 1400              | ( 2060.3 ) | 29.92                      | ( 207.57 )                            |
| 400               | ( 260.3 )  | 16.26                      | ( 112.80 )                            | 1500              | ( 2240.3 ) | 31.23                      | ( 216.70 )                            |
| 450               | ( 350.3 )  | 17.03                      | ( 118.18 )                            | 1600              | ( 2420.3 ) | 32.55                      | ( 225.81 )                            |
| 500               | ( 440.3 )  | 17.78                      | ( 123.33 )                            | 1672              | ( 2549.9 ) | 33.49                      | ( 232.37 )                            |

**Application Notes:** Data for thermal conductivity is collected from references [2-4], and fitted with the equations below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $1 < T \leq 293$

$$k(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $293 < T \leq 1672$

$$k(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_{-2} / \left( \frac{T}{1000} \right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

Temperature  $1 < T \leq 293$   $293 < T \leq 1672$

Range [K]:

|      |       |                   |          |
|------|-------|-------------------|----------|
| N =  | 2.184 | B0 =              | 11.52    |
| A0 = | 24520 | B1 =              | 13.24    |
| A1 = | 46.76 | B2 =              | -0.04806 |
| A2 = | 1197  | B <sub>-2</sub> = | -0.08805 |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Thermal Expansion with Temperature

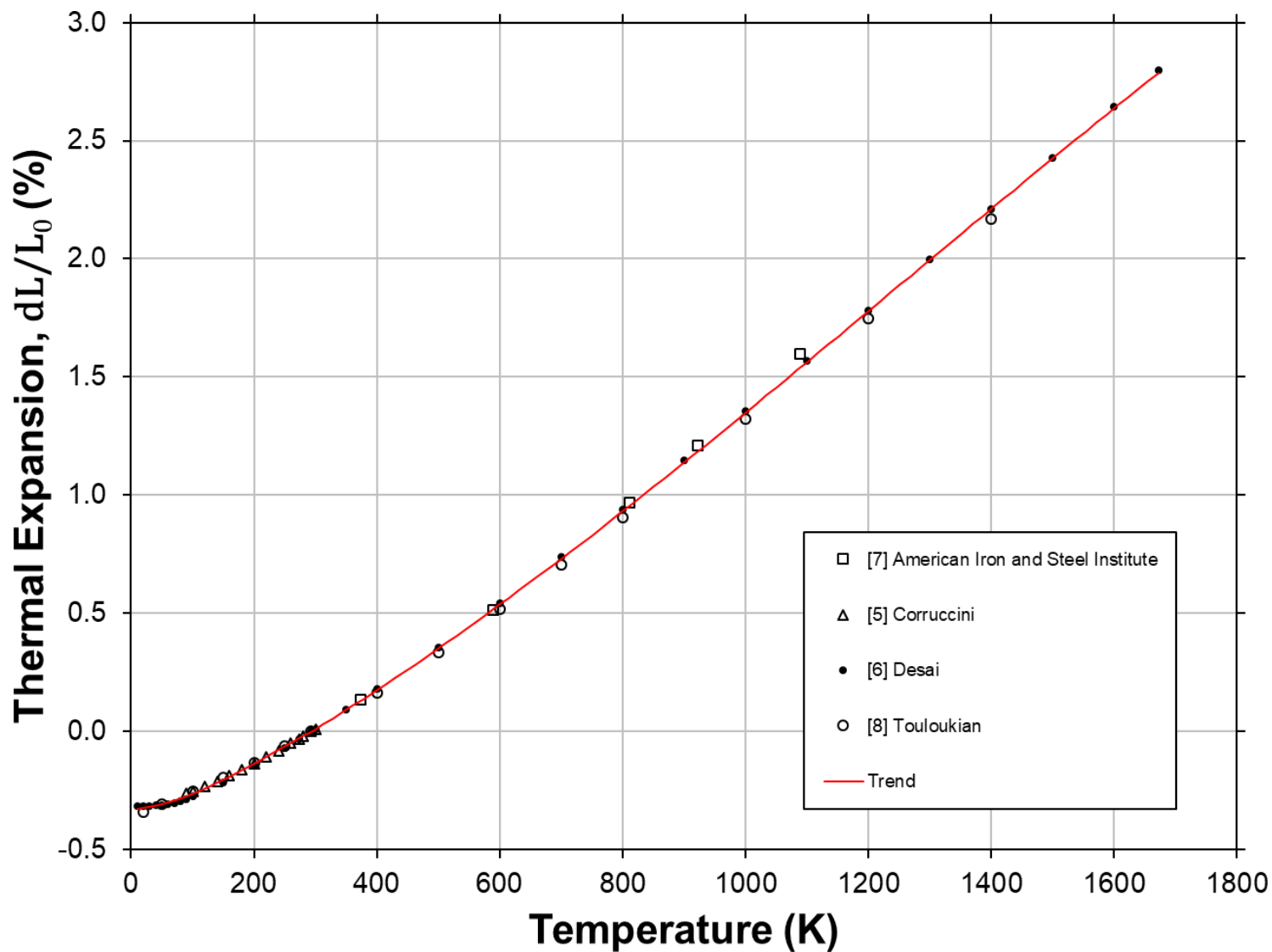


Figure 5.1.2-3: Thermal Expansion versus Temperature of SS347. Displaying fitted trend of the data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 20                | ( -423.7 ) | -0.325                                  | 700               | ( 800.3 )  | 0.730                                   |
| 60                | ( -351.7 ) | -0.302                                  | 750               | ( 890.3 )  | 0.830                                   |
| 80                | ( -315.7 ) | -0.285                                  | 800               | ( 980.3 )  | 0.931                                   |
| 100               | ( -279.7 ) | -0.265                                  | 850               | ( 1070.3 ) | 1.034                                   |
| 150               | ( -189.7 ) | -0.206                                  | 900               | ( 1160.3 ) | 1.137                                   |
| 200               | ( -99.7 )  | -0.137                                  | 950               | ( 1250.3 ) | 1.242                                   |
| 250               | ( -9.7 )   | -0.064                                  | 1000              | ( 1340.3 ) | 1.348                                   |
| 300               | ( 80.3 )   | 0.008                                   | 1050              | ( 1430.3 ) | 1.455                                   |
| 350               | ( 170.3 )  | 0.090                                   | 1100              | ( 1520.3 ) | 1.562                                   |
| 400               | ( 260.3 )  | 0.174                                   | 1200              | ( 1700.3 ) | 1.778                                   |
| 450               | ( 350.3 )  | 0.261                                   | 1300              | ( 1880.3 ) | 1.995                                   |
| 500               | ( 440.3 )  | 0.351                                   | 1400              | ( 2060.3 ) | 2.211                                   |
| 550               | ( 530.3 )  | 0.442                                   | 1500              | ( 2240.3 ) | 2.426                                   |
| 600               | ( 620.3 )  | 0.536                                   | 1600              | ( 2420.3 ) | 2.638                                   |
| 650               | ( 710.3 )  | 0.632                                   | 1672              | ( 2549.9 ) | 2.788                                   |

**Application Notes:** Thermal expansion is collected from references [5-8], and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion} [\%]$$

$$T = \text{Temperature} [K]$$

## Constants:

| Temperature Range [K]: | <u>10 &lt; T ≤ 293</u> | <u>293 &lt; T ≤ 1672</u> |
|------------------------|------------------------|--------------------------|
| A0 =                   | -0.3304                | -0.4153                  |
| A1 =                   | 0.1796                 | 1.199                    |
| A2 =                   | 5.497                  | 0.7665                   |
| A3 =                   | -7.823                 | -0.202                   |

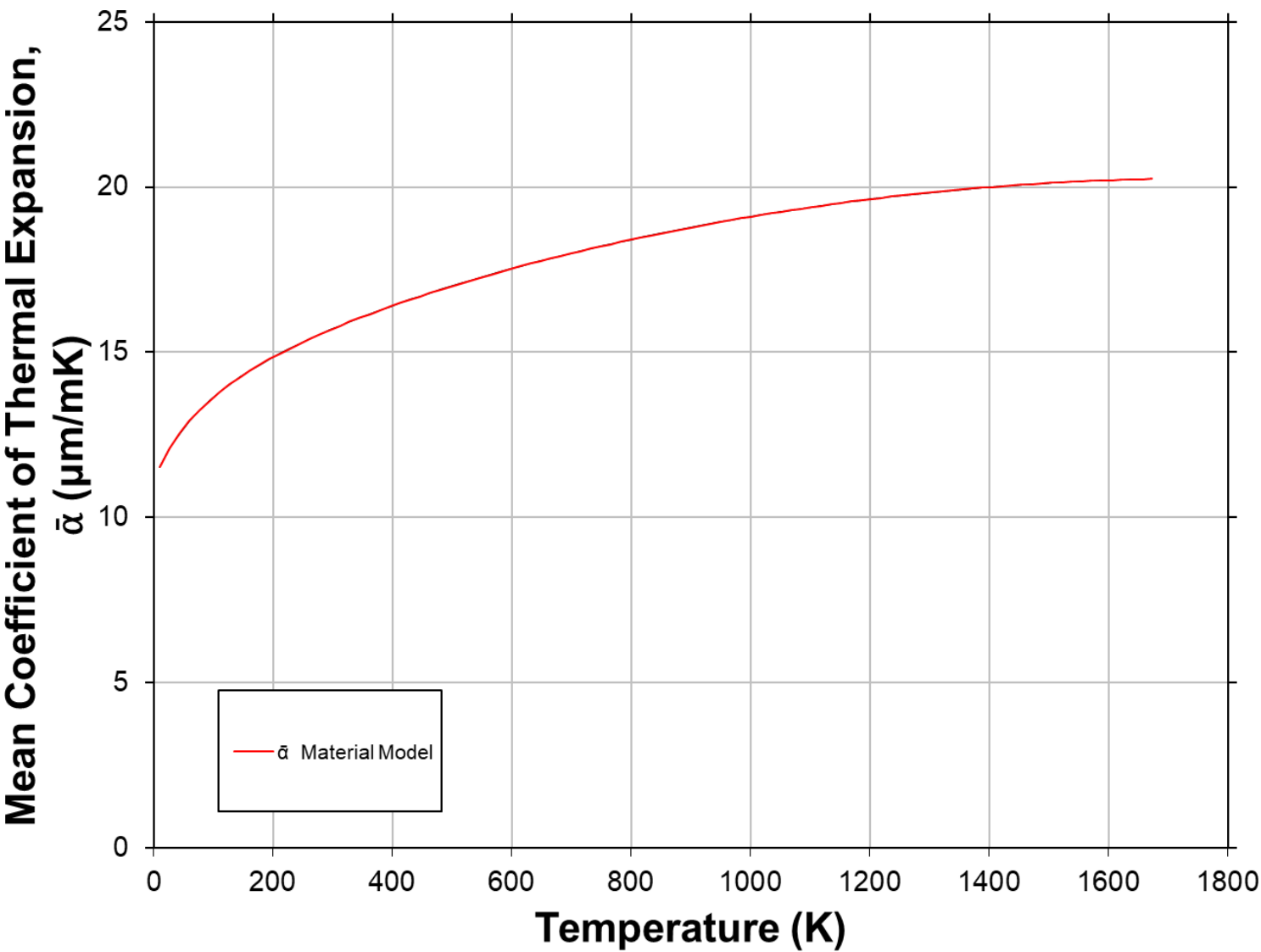


Figure 5.1.2-4: Mean Coefficient of Thermal Expansion versus Temperature of SS347. Calculated from Thermal Expansion data for SS347.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 20                | ( -423.7 ) | 11.871                                | ( 6.595 )                                           | 700               | ( 800.3 )  | 17.996                                | ( 9.998 )                                           |
| 60                | ( -351.7 ) | 12.912                                | ( 7.173 )                                           | 750               | ( 890.3 )  | 18.209                                | ( 10.116 )                                          |
| 80                | ( -315.7 ) | 13.298                                | ( 7.388 )                                           | 800               | ( 980.3 )  | 18.411                                | ( 10.228 )                                          |
| 100               | ( -279.7 ) | 13.630                                | ( 7.572 )                                           | 850               | ( 1070.3 ) | 18.600                                | ( 10.334 )                                          |
| 150               | ( -189.7 ) | 14.307                                | ( 7.948 )                                           | 900               | ( 1160.3 ) | 18.779                                | ( 10.433 )                                          |
| 200               | ( -99.7 )  | 14.846                                | ( 8.248 )                                           | 950               | ( 1250.3 ) | 18.946                                | ( 10.526 )                                          |
| 250               | ( -9.7 )   | 15.304                                | ( 8.502 )                                           | 1000              | ( 1340.3 ) | 19.103                                | ( 10.613 )                                          |
| 300               | ( 80.3 )   | 15.708                                | ( 8.727 )                                           | 1050              | ( 1430.3 ) | 19.250                                | ( 10.694 )                                          |
| 350               | ( 170.3 )  | 16.072                                | ( 8.929 )                                           | 1100              | ( 1520.3 ) | 19.386                                | ( 10.770 )                                          |
| 400               | ( 260.3 )  | 16.406                                | ( 9.114 )                                           | 1200              | ( 1700.3 ) | 19.628                                | ( 10.904 )                                          |
| 450               | ( 350.3 )  | 16.715                                | ( 9.286 )                                           | 1300              | ( 1880.3 ) | 19.830                                | ( 11.017 )                                          |
| 500               | ( 440.3 )  | 17.004                                | ( 9.447 )                                           | 1400              | ( 2060.3 ) | 19.993                                | ( 11.107 )                                          |
| 550               | ( 530.3 )  | 17.275                                | ( 9.597 )                                           | 1500              | ( 2240.3 ) | 20.118                                | ( 11.177 )                                          |
| 600               | ( 620.3 )  | 17.529                                | ( 9.738 )                                           | 1600              | ( 2420.3 ) | 20.205                                | ( 11.225 )                                          |
| 650               | ( 710.3 )  | 17.769                                | ( 9.872 )                                           | 1672              | ( 2549.9 ) | 20.244                                | ( 11.247 )                                          |

**Application Notes:** Mean coefficient of thermal expansion is calculated as a function of thermal expansion. Calculated data is then fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = \left( A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right) / \left( A_0 + \left( \frac{T}{1000} \right) \right)$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

Temp. Range [K]:  $10 \leq T \leq 1672$

$A_0 = 1.119$   
 $A_1 = 15.57$   
 $A_2 = 6.144$   
 $A_3 = -1.806$   
 $A_0 = 0.1007$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Specific Heat with Temperature

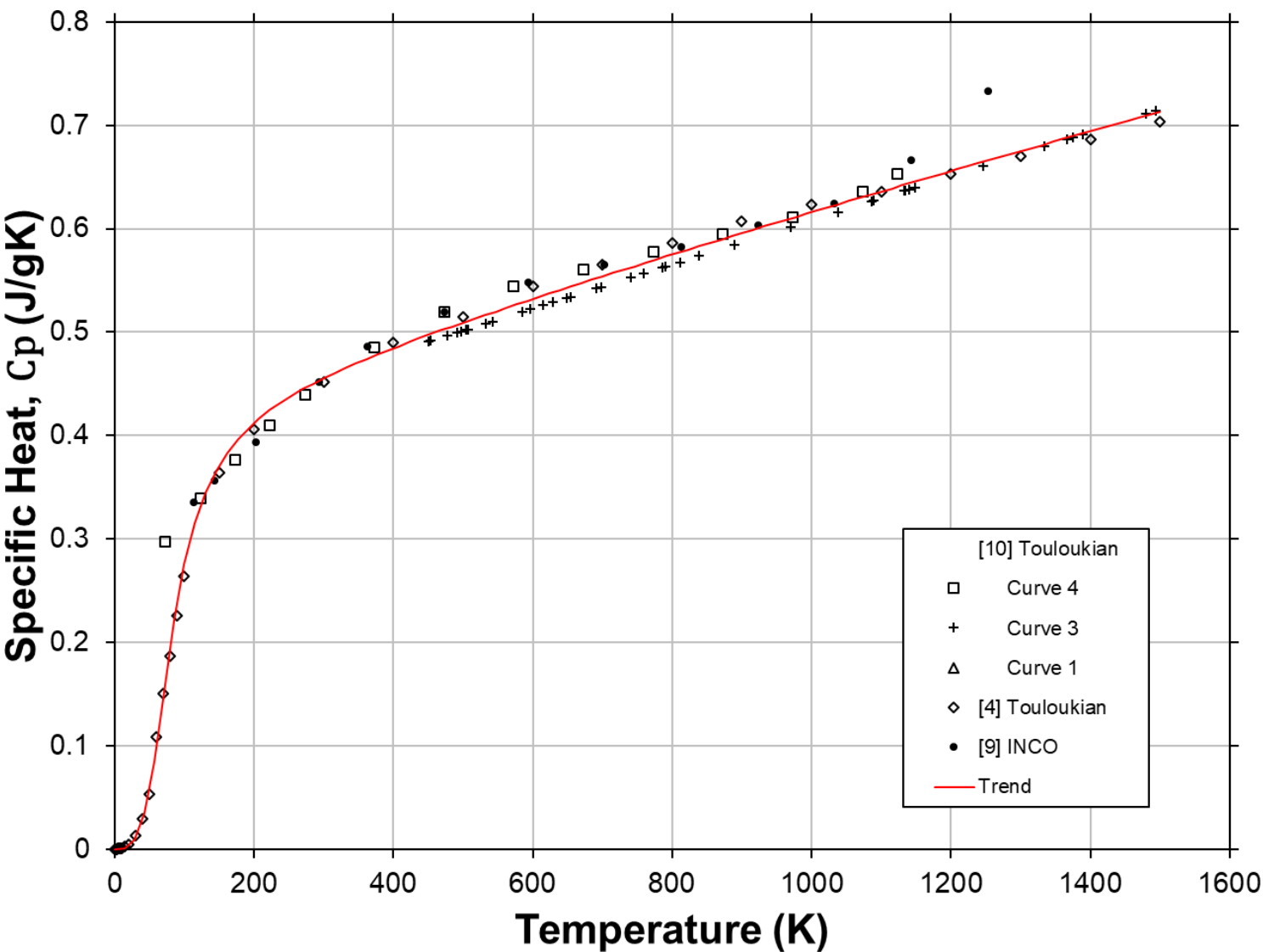


Figure 5.1.2-5: Specific Heat versus Temperature of SS347. Displaying fitted trend and data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 100               | ( -279.7 ) | 0.275                            | ( 0.066 )       | 650               | ( 710.3 )  | 0.543                            | ( 0.130 )       |
| 150               | ( -189.7 ) | 0.371                            | ( 0.089 )       | 700               | ( 800.3 )  | 0.554                            | ( 0.132 )       |
| 200               | ( -99.7 )  | 0.412                            | ( 0.099 )       | 800               | ( 980.3 )  | 0.575                            | ( 0.137 )       |
| 250               | ( -9.7 )   | 0.437                            | ( 0.104 )       | 900               | ( 1160.3 ) | 0.596                            | ( 0.142 )       |
| 300               | ( 80.3 )   | 0.455                            | ( 0.109 )       | 1000              | ( 1340.3 ) | 0.616                            | ( 0.147 )       |
| 350               | ( 170.3 )  | 0.471                            | ( 0.113 )       | 1100              | ( 1520.3 ) | 0.636                            | ( 0.152 )       |
| 400               | ( 260.3 )  | 0.484                            | ( 0.116 )       | 1200              | ( 1700.3 ) | 0.656                            | ( 0.157 )       |
| 450               | ( 350.3 )  | 0.497                            | ( 0.119 )       | 1300              | ( 1880.3 ) | 0.675                            | ( 0.161 )       |
| 500               | ( 440.3 )  | 0.509                            | ( 0.122 )       | 1400              | ( 2060.3 ) | 0.694                            | ( 0.166 )       |
| 550               | ( 530.3 )  | 0.521                            | ( 0.124 )       | 1500              | ( 2240.3 ) | 0.713                            | ( 0.170 )       |
| 600               | ( 620.3 )  | 0.532                            | ( 0.127 )       |                   |            |                                  |                 |

**Application Notes:** Data for specific heat is collected from references [4, 9, 10] and fitted with the equations below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $1 < T \leq 100$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $100 < T \leq 1500$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_{_2} / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

| Temperature Range [K]: | $1 < T \leq 100$ | $100 < T \leq 1500$        |
|------------------------|------------------|----------------------------|
| N =                    | 2.755            | B0 = 0.4071                |
| A0 =                   | 107.6            | B1 = 0.2227                |
| A1 =                   | -17.14           | B2 = -0.01218              |
| A2 =                   | 140.1            | B <sub>2</sub> = -0.001561 |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

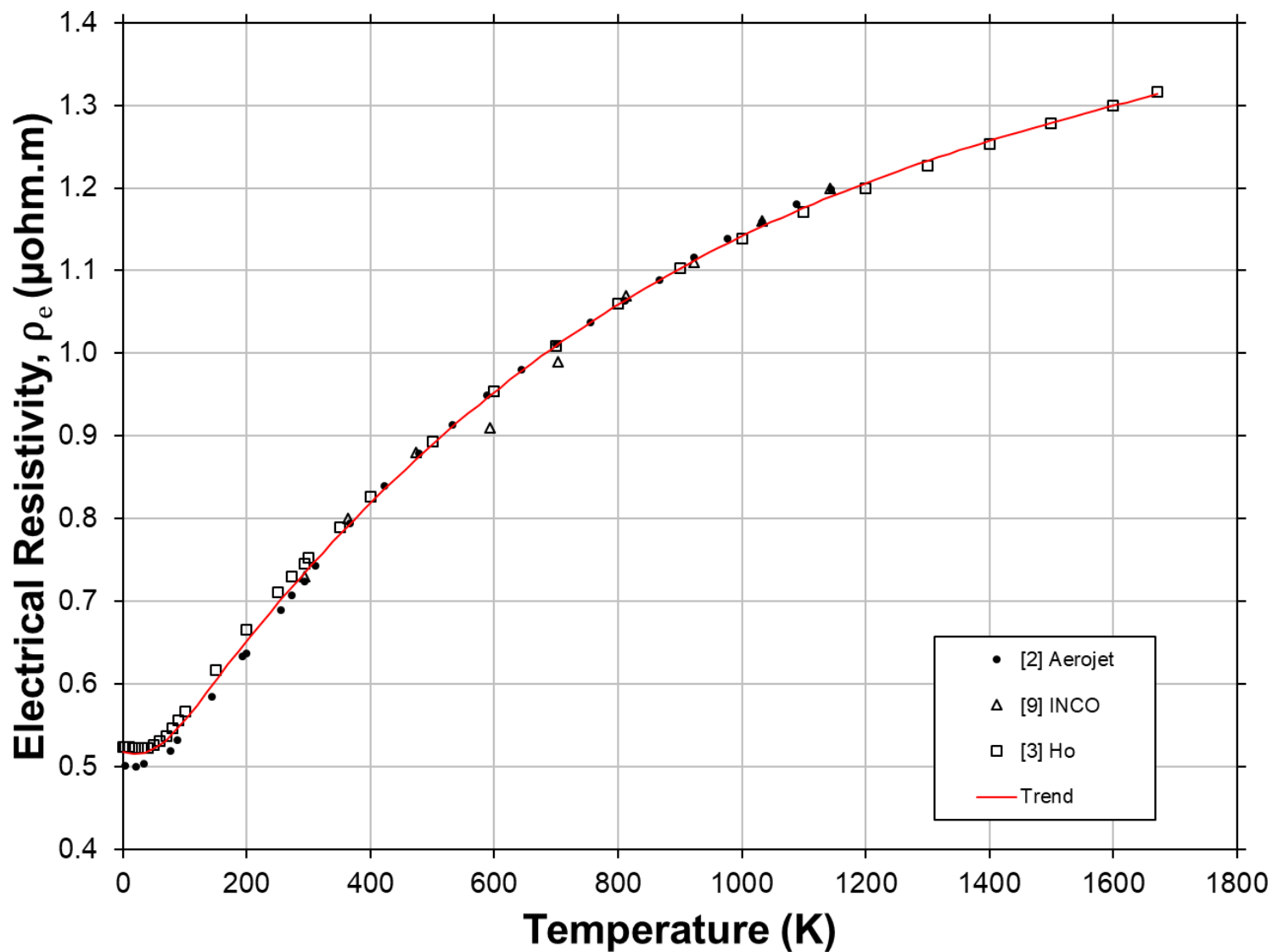


Figure 5.1.2-6: Electrical Resistivity versus Temperature of SS347.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 20                | ( -423.7 ) | 0.515                               | ( 20.28 )               | 700               | ( 800.3 )  | 1.009                               | ( 39.73 )               |
| 60                | ( -351.7 ) | 0.526                               | ( 20.73 )               | 750               | ( 890.3 )  | 1.035                               | ( 40.74 )               |
| 80                | ( -315.7 ) | 0.539                               | ( 21.24 )               | 800               | ( 980.3 )  | 1.059                               | ( 41.69 )               |
| 100               | ( -279.7 ) | 0.556                               | ( 21.88 )               | 850               | ( 1070.3 ) | 1.082                               | ( 42.58 )               |
| 150               | ( -189.7 ) | 0.605                               | ( 23.81 )               | 900               | ( 1160.3 ) | 1.103                               | ( 43.42 )               |
| 200               | ( -99.7 )  | 0.652                               | ( 25.68 )               | 950               | ( 1250.3 ) | 1.123                               | ( 44.21 )               |
| 250               | ( -9.7 )   | 0.697                               | ( 27.45 )               | 1000              | ( 1340.3 ) | 1.142                               | ( 44.95 )               |
| 300               | ( 80.3 )   | 0.740                               | ( 29.14 )               | 1050              | ( 1430.3 ) | 1.159                               | ( 45.64 )               |
| 350               | ( 170.3 )  | 0.781                               | ( 30.73 )               | 1100              | ( 1520.3 ) | 1.176                               | ( 46.30 )               |
| 400               | ( 260.3 )  | 0.819                               | ( 32.25 )               | 1200              | ( 1700.3 ) | 1.206                               | ( 47.49 )               |
| 450               | ( 350.3 )  | 0.855                               | ( 33.68 )               | 1300              | ( 1880.3 ) | 1.233                               | ( 48.54 )               |
| 500               | ( 440.3 )  | 0.890                               | ( 35.03 )               | 1400              | ( 2060.3 ) | 1.257                               | ( 49.49 )               |
| 550               | ( 530.3 )  | 0.922                               | ( 36.31 )               | 1500              | ( 2240.3 ) | 1.279                               | ( 50.36 )               |
| 600               | ( 620.3 )  | 0.953                               | ( 37.52 )               | 1600              | ( 2420.3 ) | 1.299                               | ( 51.16 )               |
| 650               | ( 710.3 )  | 0.982                               | ( 38.66 )               | 1672              | ( 2549.9 ) | 1.314                               | ( 51.71 )               |

**Application Notes:** Data for electrical resistivity is collected from references [2, 3, 9], and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

| T. Range [K]: | $1 < T \leq 150$ | $150 < T \leq 1672$ |
|---------------|------------------|---------------------|
| A0 =          | 0.5185           | 0.4477              |
| A1 =          | -0.3548          | 1.125               |
| A2 =          | 9.398            | -0.532              |
| A3 =          | -21.16           | 0.101               |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Young's Modulus with Temperature

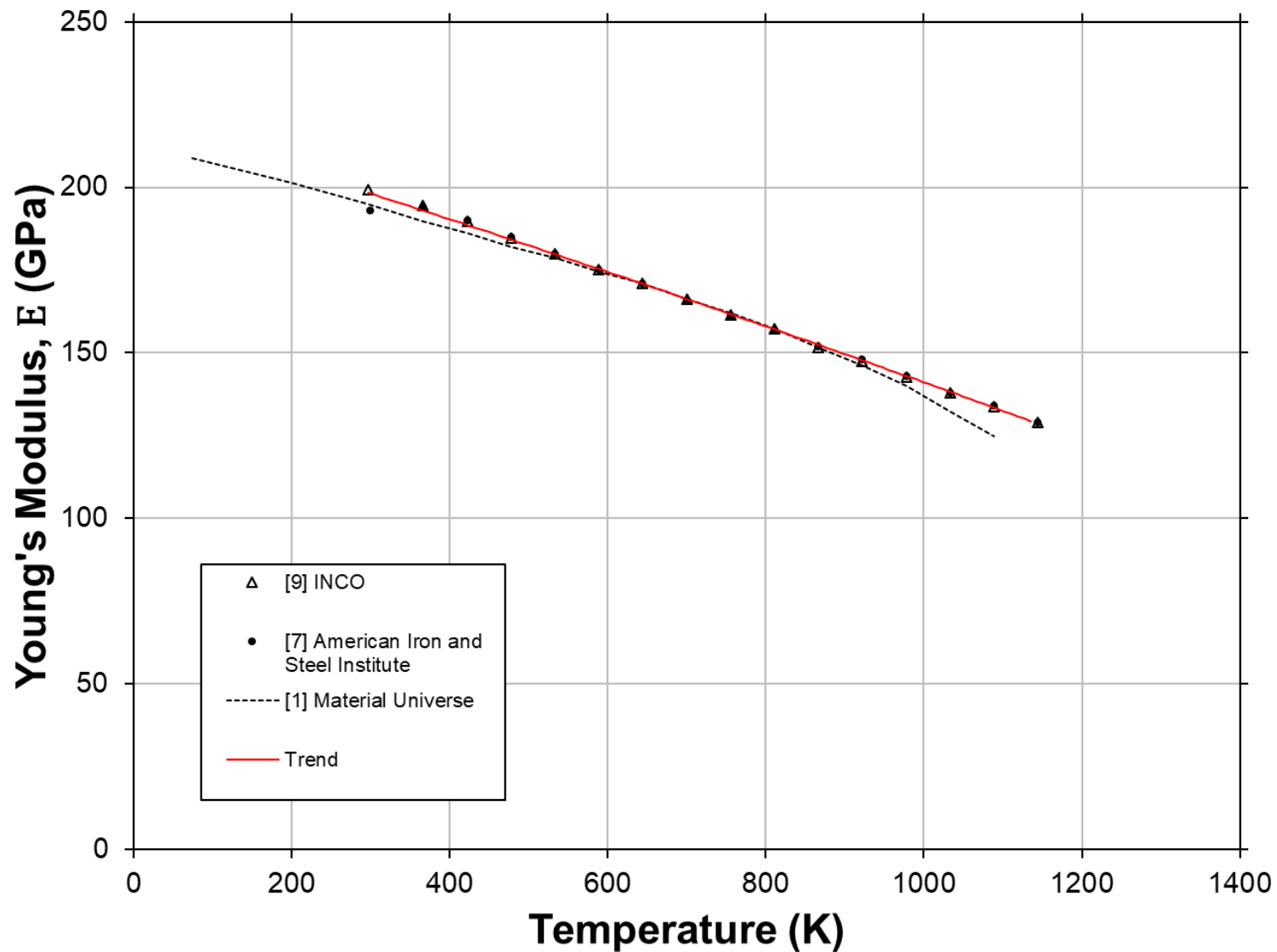


Figure 5.1.2-7: Young's Modulus versus Temperature of SS347. Displaying fitted trend and data with comparison to Materials Universe.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|-----------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )    | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 297               | ( 74.9 )  | 198.35                | ( 28.78 ) | 675               | ( 755.3 )  | 168.32                | ( 24.42 ) |
| 325               | ( 125.3 ) | 196.19                | ( 28.47 ) | 700               | ( 800.3 )  | 166.28                | ( 24.13 ) |
| 350               | ( 170.3 ) | 194.25                | ( 28.19 ) | 725               | ( 845.3 )  | 164.22                | ( 23.83 ) |
| 375               | ( 215.3 ) | 192.30                | ( 27.90 ) | 750               | ( 890.3 )  | 162.16                | ( 23.53 ) |
| 400               | ( 260.3 ) | 190.34                | ( 27.62 ) | 775               | ( 935.3 )  | 160.09                | ( 23.23 ) |
| 425               | ( 305.3 ) | 188.38                | ( 27.33 ) | 800               | ( 980.3 )  | 158.01                | ( 22.93 ) |
| 450               | ( 350.3 ) | 186.41                | ( 27.05 ) | 825               | ( 1025.3 ) | 155.92                | ( 22.62 ) |
| 475               | ( 395.3 ) | 184.43                | ( 26.76 ) | 850               | ( 1070.3 ) | 153.83                | ( 22.32 ) |
| 500               | ( 440.3 ) | 182.44                | ( 26.47 ) | 875               | ( 1115.3 ) | 151.72                | ( 22.02 ) |
| 525               | ( 485.3 ) | 180.45                | ( 26.18 ) | 900               | ( 1160.3 ) | 149.61                | ( 21.71 ) |
| 550               | ( 530.3 ) | 178.45                | ( 25.89 ) | 925               | ( 1205.3 ) | 147.50                | ( 21.40 ) |
| 575               | ( 575.3 ) | 176.44                | ( 25.60 ) | 950               | ( 1250.3 ) | 145.37                | ( 21.09 ) |
| 600               | ( 620.3 ) | 174.42                | ( 25.31 ) | 1000              | ( 1340.3 ) | 141.10                | ( 20.47 ) |
| 625               | ( 665.3 ) | 172.40                | ( 25.01 ) | 1100              | ( 1520.3 ) | 132.46                | ( 19.22 ) |
| 650               | ( 710.3 ) | 170.37                | ( 24.72 ) | 1144              | ( 1599.5 ) | 128.62                | ( 18.66 ) |

**Application Notes:** Data for Young's modulus is collected from references [1, 7, 9], and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$E(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]: 297 < T ≤ 1144

A0 = 220.7

A1 = -73.42

A2 = -6.182



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Shear Modulus with Temperature

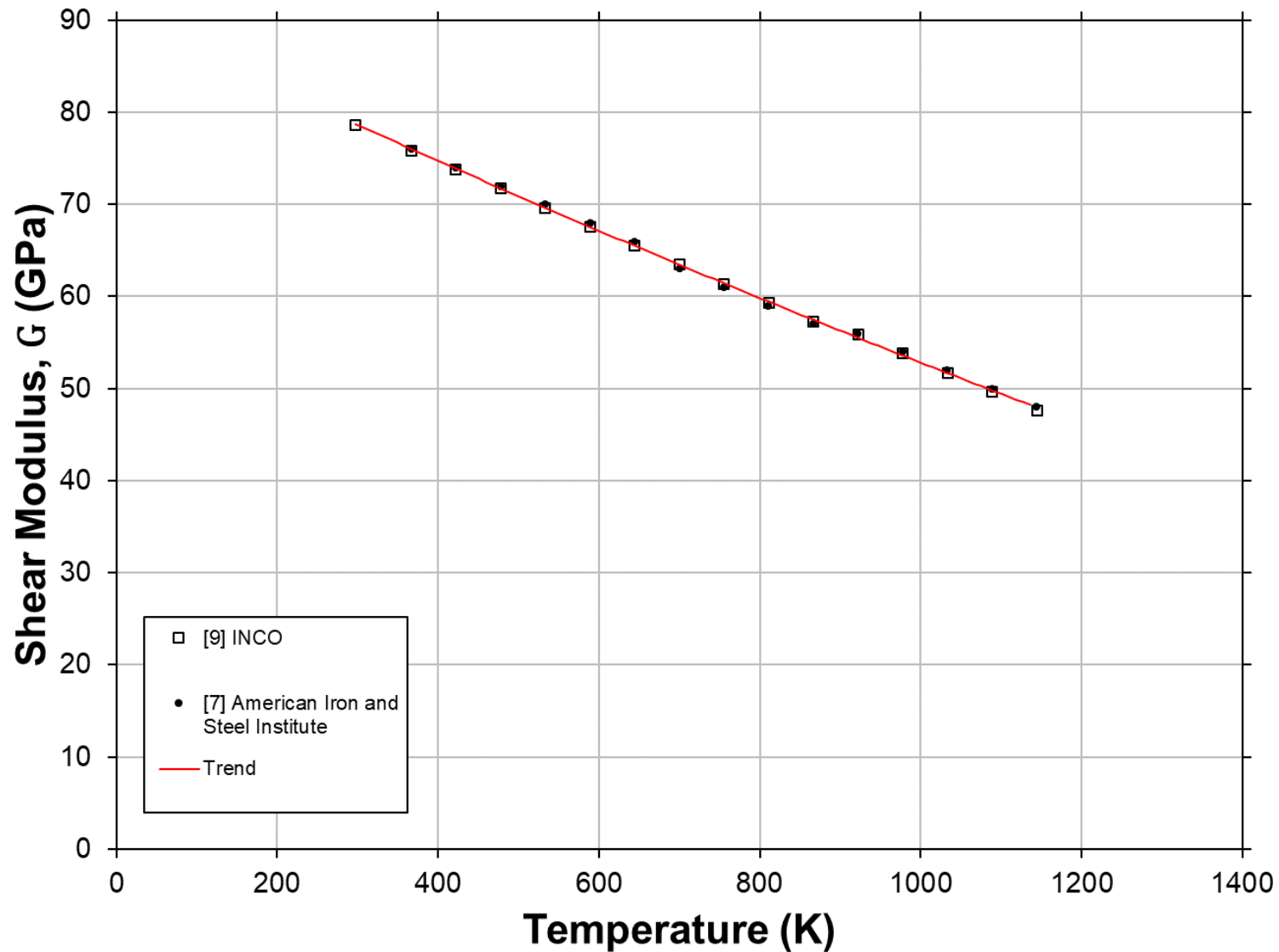


Figure 5.1.2-8: Shear Modulus versus Temperature of SS347. Displaying fitted trend and the data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Shear Modulus ( G ) |           | Temperature ( T ) |            | Shear Modulus ( G ) |          |
|-------------------|-----------|---------------------|-----------|-------------------|------------|---------------------|----------|
| K                 | ( °F )    | GPa                 | ( Msi )   | K                 | ( °F )     | GPa                 | ( Msi )  |
| 297               | ( 74.9 )  | 78.74               | ( 11.43 ) | 675               | ( 755.3 )  | 64.37               | ( 9.34 ) |
| 325               | ( 125.3 ) | 77.64               | ( 11.27 ) | 700               | ( 800.3 )  | 63.46               | ( 9.21 ) |
| 350               | ( 170.3 ) | 76.66               | ( 11.12 ) | 725               | ( 845.3 )  | 62.55               | ( 9.08 ) |
| 375               | ( 215.3 ) | 75.69               | ( 10.98 ) | 750               | ( 890.3 )  | 61.65               | ( 8.94 ) |
| 400               | ( 260.3 ) | 74.72               | ( 10.84 ) | 775               | ( 935.3 )  | 60.74               | ( 8.81 ) |
| 425               | ( 305.3 ) | 73.76               | ( 10.70 ) | 800               | ( 980.3 )  | 59.85               | ( 8.68 ) |
| 450               | ( 350.3 ) | 72.80               | ( 10.56 ) | 825               | ( 1025.3 ) | 58.96               | ( 8.55 ) |
| 475               | ( 395.3 ) | 71.85               | ( 10.43 ) | 850               | ( 1070.3 ) | 58.07               | ( 8.43 ) |
| 500               | ( 440.3 ) | 70.90               | ( 10.29 ) | 875               | ( 1115.3 ) | 57.19               | ( 8.30 ) |
| 525               | ( 485.3 ) | 69.95               | ( 10.15 ) | 900               | ( 1160.3 ) | 56.31               | ( 8.17 ) |
| 550               | ( 530.3 ) | 69.01               | ( 10.01 ) | 925               | ( 1205.3 ) | 55.43               | ( 8.04 ) |
| 575               | ( 575.3 ) | 68.08               | ( 9.88 )  | 950               | ( 1250.3 ) | 54.56               | ( 7.92 ) |
| 600               | ( 620.3 ) | 67.14               | ( 9.74 )  | 1000              | ( 1340.3 ) | 52.84               | ( 7.67 ) |
| 625               | ( 665.3 ) | 66.22               | ( 9.61 )  | 1100              | ( 1520.3 ) | 49.44               | ( 7.17 ) |
| 650               | ( 710.3 ) | 65.29               | ( 9.47 )  | 1144              | ( 1599.5 ) | 47.96               | ( 6.96 ) |

**Application Notes:** Data for shear modulus is collected from references [7, 9], and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$G(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]: 297 < T ≤ 1144

A0 = 90.74

A1 = -41.46

A2 = 3.556

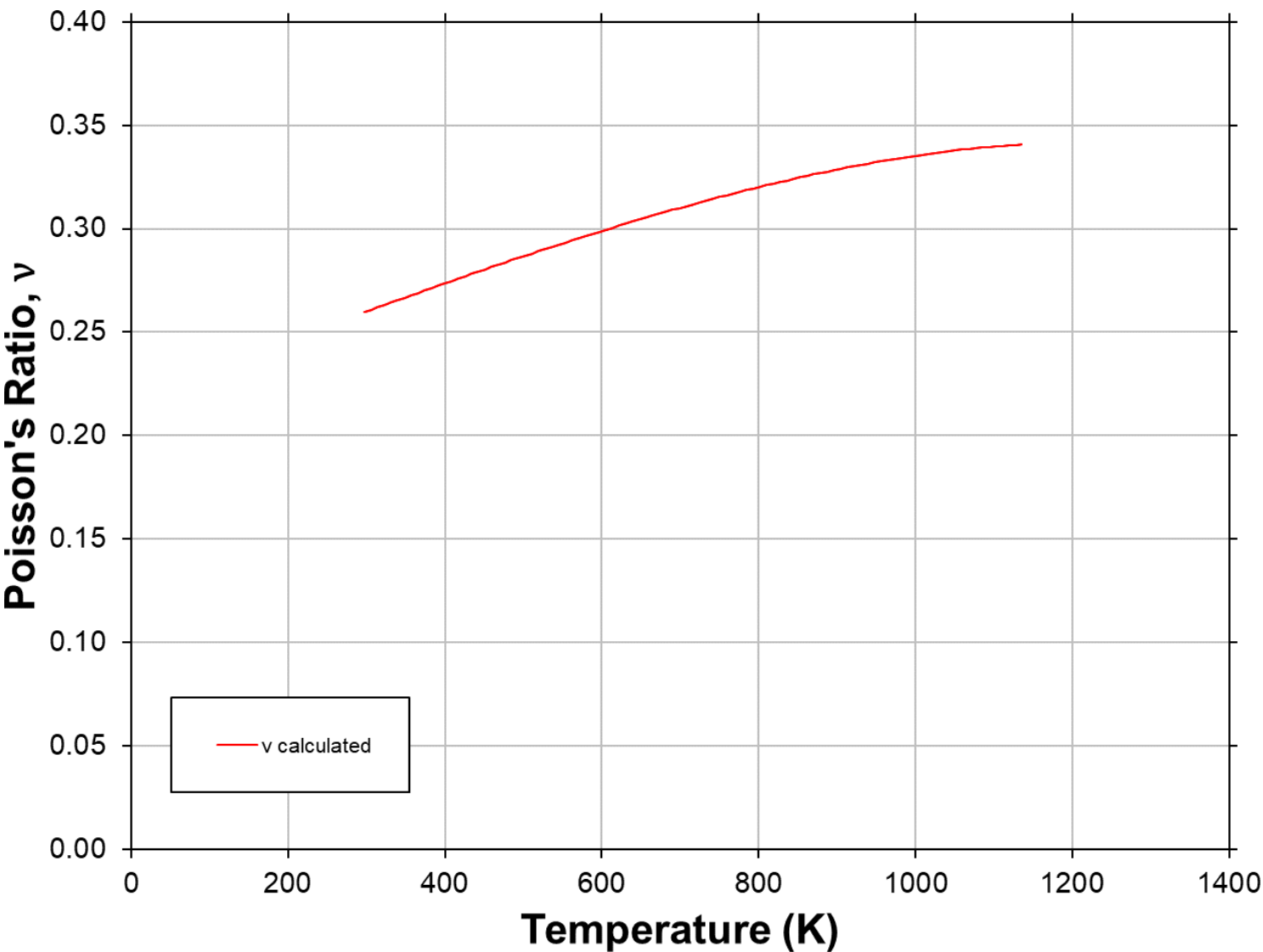


Figure 5.1.2-9: Poisson's Ratio versus Temperature of SS347. Calculated from the Young's Modulus and Shear Modulus fitted trends for SS347.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|-----------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )    |                           | K                 | ( °F )     |                           |
| 297               | ( 74.9 )  | 0.260                     | 675               | ( 755.3 )  | 0.307                     |
| 325               | ( 125.3 ) | 0.263                     | 700               | ( 800.3 )  | 0.310                     |
| 350               | ( 170.3 ) | 0.267                     | 725               | ( 845.3 )  | 0.313                     |
| 375               | ( 215.3 ) | 0.270                     | 750               | ( 890.3 )  | 0.315                     |
| 400               | ( 260.3 ) | 0.274                     | 775               | ( 935.3 )  | 0.318                     |
| 425               | ( 305.3 ) | 0.277                     | 800               | ( 980.3 )  | 0.320                     |
| 450               | ( 350.3 ) | 0.280                     | 825               | ( 1025.3 ) | 0.322                     |
| 475               | ( 395.3 ) | 0.283                     | 850               | ( 1070.3 ) | 0.325                     |
| 500               | ( 440.3 ) | 0.287                     | 875               | ( 1115.3 ) | 0.327                     |
| 525               | ( 485.3 ) | 0.290                     | 900               | ( 1160.3 ) | 0.329                     |
| 550               | ( 530.3 ) | 0.293                     | 925               | ( 1205.3 ) | 0.330                     |
| 575               | ( 575.3 ) | 0.296                     | 950               | ( 1250.3 ) | 0.332                     |
| 600               | ( 620.3 ) | 0.299                     | 1000              | ( 1340.3 ) | 0.335                     |
| 625               | ( 665.3 ) | 0.302                     | 1100              | ( 1520.3 ) | 0.340                     |
| 650               | ( 710.3 ) | 0.305                     | 1144              | ( 1599.5 ) | 0.341                     |

**Application Notes:** Poisson's Ratio is calculated as a function of Young's modulus and shear modulus trends, as seen in the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$\nu(T) = E(T)/(2 \cdot G(T)) - 1$$

$$\nu(T) = \text{Poisson's Ratio}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $297 < T \leq 1144$



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Yield Strength with Temperature

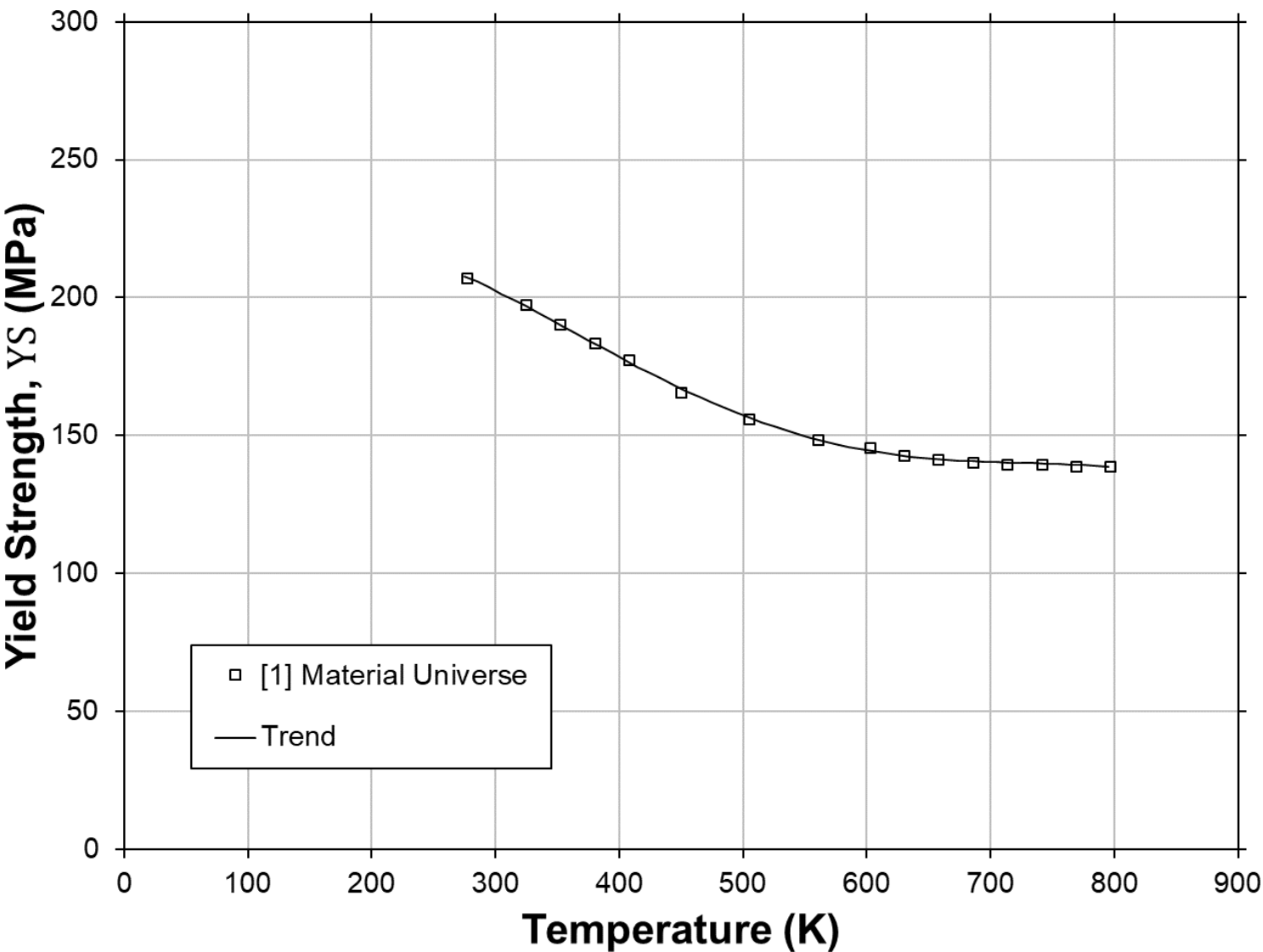


Figure 5.1.2-10: Yield Strength versus Temperature of SS347. Displaying fitted trend and data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Yield Strength with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Yield Strength ( YS ) |           | Temperature ( T ) |           | Yield Strength ( YS ) |           |
|-------------------|-----------|-----------------------|-----------|-------------------|-----------|-----------------------|-----------|
| K                 | ( °F )    | MPa                   | ( Ksi )   | K                 | ( °F )    | MPa                   | ( Ksi )   |
| 270               | ( 26.3 )  | 208.70                | ( 30.28 ) | 520               | ( 476.3 ) | 154.05                | ( 22.35 ) |
| 290               | ( 62.3 )  | 204.68                | ( 29.70 ) | 540               | ( 512.3 ) | 151.13                | ( 21.93 ) |
| 310               | ( 98.3 )  | 200.27                | ( 29.06 ) | 560               | ( 548.3 ) | 148.61                | ( 21.56 ) |
| 330               | ( 134.3 ) | 195.59                | ( 28.38 ) | 580               | ( 584.3 ) | 146.46                | ( 21.25 ) |
| 350               | ( 170.3 ) | 190.74                | ( 27.68 ) | 600               | ( 620.3 ) | 144.69                | ( 20.99 ) |
| 370               | ( 206.3 ) | 185.83                | ( 26.96 ) | 620               | ( 656.3 ) | 143.27                | ( 20.79 ) |
| 390               | ( 242.3 ) | 180.93                | ( 26.25 ) | 640               | ( 692.3 ) | 142.18                | ( 20.63 ) |
| 410               | ( 278.3 ) | 176.12                | ( 25.56 ) | 660               | ( 728.3 ) | 141.37                | ( 20.51 ) |
| 430               | ( 314.3 ) | 171.49                | ( 24.88 ) | 680               | ( 764.3 ) | 140.79                | ( 20.43 ) |
| 450               | ( 350.3 ) | 167.08                | ( 24.24 ) | 700               | ( 800.3 ) | 140.40                | ( 20.37 ) |
| 470               | ( 386.3 ) | 162.94                | ( 23.64 ) | 720               | ( 836.3 ) | 140.12                | ( 20.33 ) |
| 490               | ( 422.3 ) | 159.12                | ( 23.09 ) | 750               | ( 890.3 ) | 139.74                | ( 20.28 ) |
| 510               | ( 458.3 ) | 155.65                | ( 22.58 ) | 800               | ( 980.3 ) | 138.45                | ( 20.09 ) |

**Application Notes:** Data for yield strength is collected from reference [1], and fitted with the equation below to approximate property trends with respect to temperature.

## Fit Equation:

$$YS(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3 + A4 \cdot \left(\frac{T}{1000}\right)^4$$

$YS(T)$  = Yield Strength [MPa]

$T$  = Temperature (K)

## Constants:

T Range [K]: 270 < T ≤ 800

A0 = 147

A1 = 939.3

A2 = -3895

A3 = 5334

A4 = -2437



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Tensile Strength with Temperature

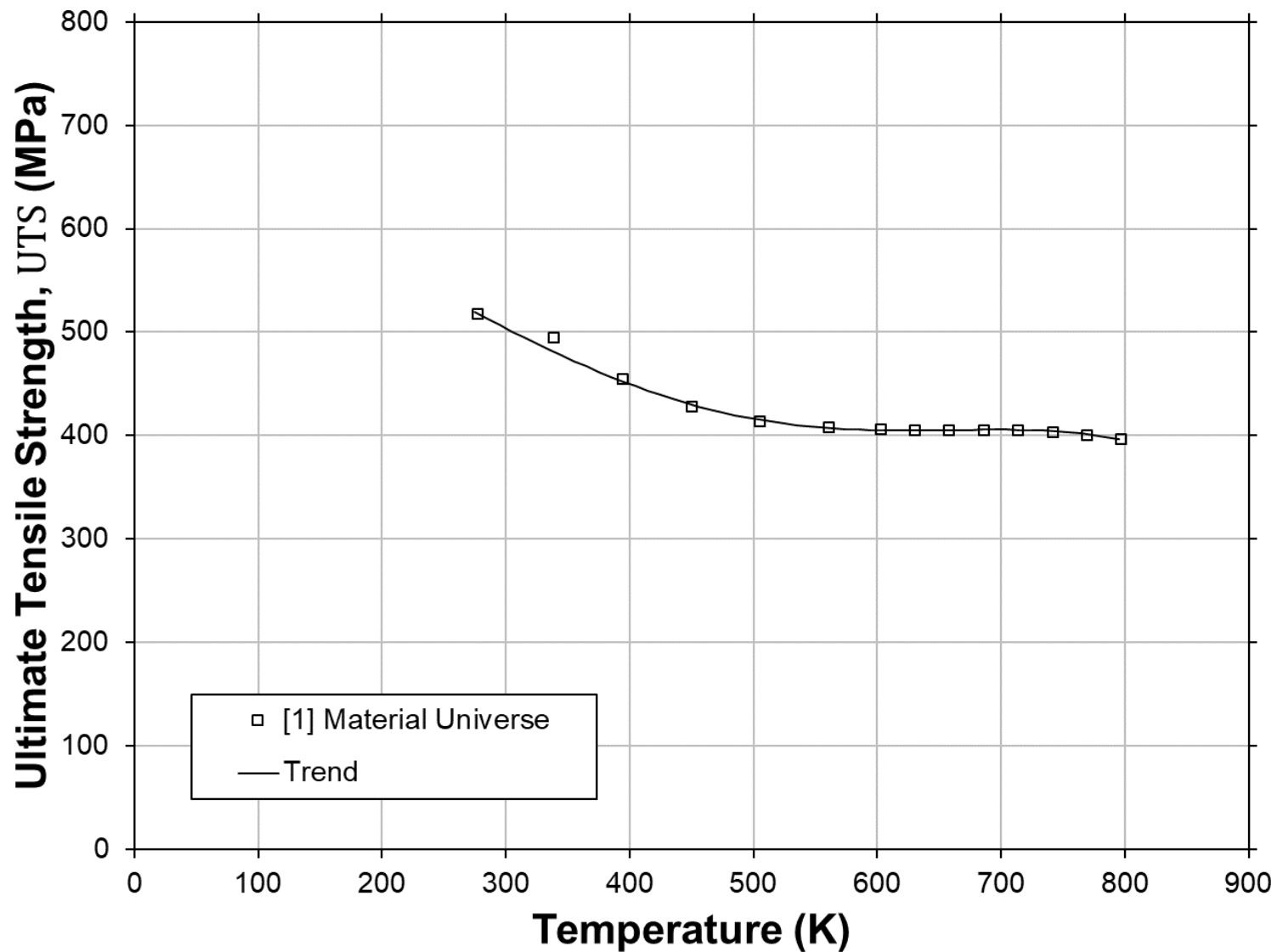


Figure 5.1.2-11: Ultimate Tensile Strength versus Temperature of SS347. Displaying fitted trend of the data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Tensile Strength with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Tensile Strength ( TS ) |           | Temperature ( T ) |           | Tensile Strength ( TS ) |           |
|-------------------|-----------|-------------------------|-----------|-------------------|-----------|-------------------------|-----------|
| K                 | ( °F )    | MPa                     | ( Ksi )   | K                 | ( °F )    | MPa                     | ( Ksi )   |
| 270               | ( 26.3 )  | 522.12                  | ( 75.76 ) | 520               | ( 476.3 ) | 412.41                  | ( 59.84 ) |
| 280               | ( 44.3 )  | 515.94                  | ( 74.86 ) | 540               | ( 512.3 ) | 409.44                  | ( 59.41 ) |
| 300               | ( 80.3 )  | 503.70                  | ( 73.09 ) | 560               | ( 548.3 ) | 407.29                  | ( 59.10 ) |
| 320               | ( 116.3 ) | 491.75                  | ( 71.35 ) | 580               | ( 584.3 ) | 405.87                  | ( 58.89 ) |
| 340               | ( 152.3 ) | 480.26                  | ( 69.69 ) | 600               | ( 620.3 ) | 405.09                  | ( 58.78 ) |
| 360               | ( 188.3 ) | 469.35                  | ( 68.10 ) | 620               | ( 656.3 ) | 404.81                  | ( 58.74 ) |
| 380               | ( 224.3 ) | 459.14                  | ( 66.62 ) | 640               | ( 692.3 ) | 404.90                  | ( 58.75 ) |
| 400               | ( 260.3 ) | 449.71                  | ( 65.25 ) | 660               | ( 728.3 ) | 405.18                  | ( 58.79 ) |
| 420               | ( 296.3 ) | 441.15                  | ( 64.01 ) | 680               | ( 764.3 ) | 405.49                  | ( 58.84 ) |
| 440               | ( 332.3 ) | 433.49                  | ( 62.90 ) | 700               | ( 800.3 ) | 405.60                  | ( 58.85 ) |
| 460               | ( 368.3 ) | 426.79                  | ( 61.93 ) | 720               | ( 836.3 ) | 405.31                  | ( 58.81 ) |
| 480               | ( 404.3 ) | 421.05                  | ( 61.09 ) | 750               | ( 890.3 ) | 403.57                  | ( 58.56 ) |
| 500               | ( 440.3 ) | 416.26                  | ( 60.40 ) | 800               | ( 980.3 ) | 394.91                  | ( 57.30 ) |

**Application Notes:** Data for ultimate tensile strength is collected from reference [1], and fitted with the equation below to approximate property trends with respect to temperature.

## Fit Equation:

$$TS(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3 + A4 \cdot \left(\frac{T}{1000}\right)^4$$

$TS(T)$  = Ultimate Tensile Strength (MPa)

$T$  = Temperature (K)

## Constants:

Temperature  
Range [K]:  $270 < T \leq 800$

A0 = 586.7

A1 = 633

A2 = -5429

A3 = 9529

A4 = -5133



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                          |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( α )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 8024                                  | 0.556                                                   | 9.10                                        | -0.265                                             | 13.63                                    | 0.275                                          | -                                  | -                                | -                           |
| 150                 | 8009                                  | 0.605                                                   | 11.14                                       | -0.206                                             | 14.31                                    | 0.371                                          | -                                  | -                                | -                           |
| 200                 | 7993                                  | 0.652                                                   | 12.53                                       | -0.137                                             | 14.85                                    | 0.412                                          | -                                  | -                                | -                           |
| 250                 | 7975                                  | 0.697                                                   | 13.57                                       | -0.064                                             | 15.30                                    | 0.437                                          | -                                  | -                                | -                           |
| 300                 | 7958                                  | 0.740                                                   | 14.51                                       | 0.008                                              | 15.71                                    | 0.455                                          | 198.12                             | 78.62                            | 0.26                        |
| 350                 | 7939                                  | 0.781                                                   | 15.43                                       | 0.090                                              | 16.07                                    | 0.471                                          | 194.25                             | 76.66                            | 0.27                        |
| 400                 | 7919                                  | 0.819                                                   | 16.26                                       | 0.174                                              | 16.41                                    | 0.484                                          | 190.34                             | 74.72                            | 0.27                        |
| 450                 | 7898                                  | 0.855                                                   | 17.03                                       | 0.261                                              | 16.72                                    | 0.497                                          | 186.41                             | 72.80                            | 0.28                        |
| 500                 | 7877                                  | 0.890                                                   | 17.78                                       | 0.351                                              | 17.00                                    | 0.509                                          | 182.44                             | 70.90                            | 0.29                        |
| 550                 | 7855                                  | 0.922                                                   | 18.50                                       | 0.442                                              | 17.27                                    | 0.521                                          | 178.45                             | 69.01                            | 0.29                        |
| 600                 | 7833                                  | 0.953                                                   | 19.20                                       | 0.536                                              | 17.53                                    | 0.532                                          | 174.42                             | 67.14                            | 0.30                        |
| 650                 | 7811                                  | 0.982                                                   | 19.90                                       | 0.632                                              | 17.77                                    | 0.543                                          | 170.37                             | 65.29                            | 0.30                        |
| 700                 | 7788                                  | 1.009                                                   | 20.58                                       | 0.730                                              | 18.00                                    | 0.554                                          | 166.28                             | 63.46                            | 0.31                        |
| 750                 | 7765                                  | 1.035                                                   | 21.27                                       | 0.830                                              | 18.21                                    | 0.564                                          | 162.16                             | 61.65                            | 0.32                        |
| 800                 | 7742                                  | 1.059                                                   | 21.94                                       | 0.931                                              | 18.41                                    | 0.575                                          | 158.01                             | 59.85                            | 0.32                        |
| 850                 | 7718                                  | 1.082                                                   | 22.62                                       | 1.034                                              | 18.60                                    | 0.585                                          | 153.83                             | 58.07                            | 0.32                        |
| 900                 | 7694                                  | 1.103                                                   | 23.29                                       | 1.137                                              | 18.78                                    | 0.596                                          | 149.61                             | 56.31                            | 0.33                        |
| 950                 | 7671                                  | 1.123                                                   | 23.96                                       | 1.242                                              | 18.95                                    | 0.606                                          | 145.37                             | 54.56                            | 0.33                        |
| 1000                | 7647                                  | 1.142                                                   | 24.62                                       | 1.348                                              | 19.10                                    | 0.616                                          | 141.10                             | 52.84                            | 0.34                        |
| 1100                | 7598                                  | 1.176                                                   | 25.95                                       | 1.562                                              | 19.39                                    | 0.636                                          | 132.46                             | 49.44                            | 0.34                        |
| 1200                | 7550                                  | 1.206                                                   | 27.28                                       | 1.778                                              | 19.63                                    | 0.656                                          | -                                  | -                                | -                           |
| 1300                | 7502                                  | 1.233                                                   | 28.60                                       | 1.995                                              | 19.83                                    | 0.675                                          | -                                  | -                                | -                           |
| 1400                | 7454                                  | 1.257                                                   | 29.92                                       | 2.211                                              | 19.99                                    | 0.694                                          | -                                  | -                                | -                           |
| 1500                | 7408                                  | 1.279                                                   | 31.23                                       | 2.426                                              | 20.12                                    | 0.713                                          | -                                  | -                                | -                           |
| 1600                | 7362                                  | 1.299                                                   | 32.55                                       | 2.638                                              | 20.20                                    | -                                              | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

5 Ferrous Alloys

5.1 Austenitic Stainless Steels

5.1.2 Stainless Steel 347

**Revision 0: 08-05-2020**

**References**

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## **6 Nuclear Materials**

### **6.1 Nuclear Fuel Materials**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 2.1: 08-25-2023

General

## Room Temperature Properties

|                                           |            |
|-------------------------------------------|------------|
| Molar Mass, [g/mol]                       | 252.04     |
| Theoretical Density, [kg/m <sup>3</sup> ] | 14,330     |
| Melting Point, [K]                        | 3123 ± 30* |
| Specific Heat, [J/(g-K)]                  | 0.189      |
| Thermal Conductivity, [W/(m-K)]           | 14.2       |
| Linear expansion coefficient, [μm/(m-K)]  | 7.25       |
| Electrical resistivity, [μΩ-m]            | 1.48       |
| Young's Modulus, [GPa]                    | 263.6      |
| Shear Modulus, [GPa]                      | 104.0      |
| Poisson's Ratio, [-]                      | 0.268      |

\* at nitrogen pressure ≥ 0.25 MPa (2.5 atm)

## Uranium – Nitrogen Phase Diagram

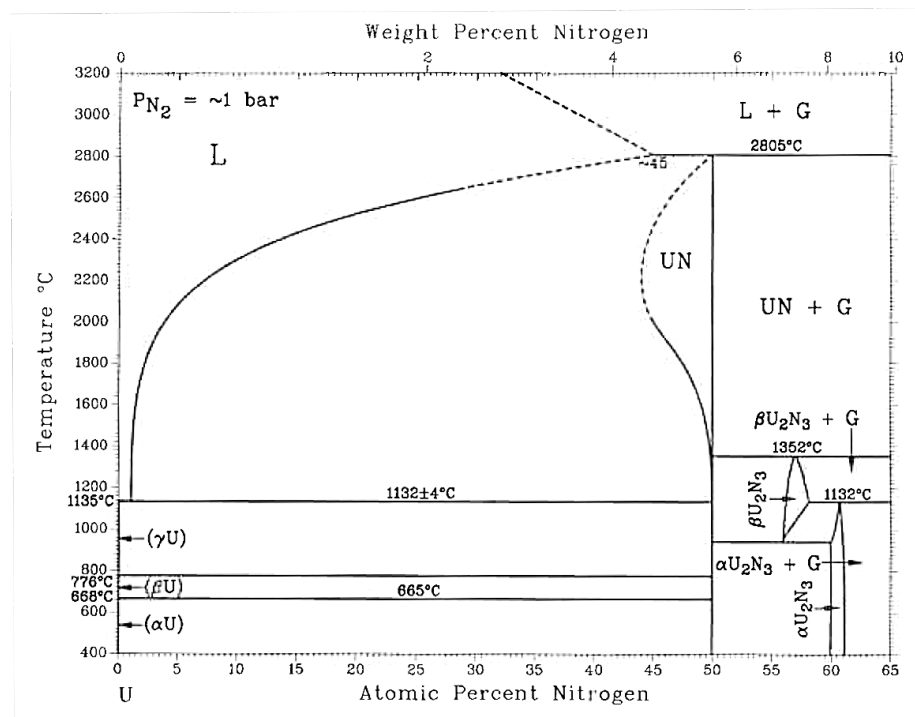


Figure 6.1.1-1: Uranium – Nitrogen Phase Diagram [1].  
Stoichiometric UN = 5.56 wt% Nitrogen



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Density with Temperature

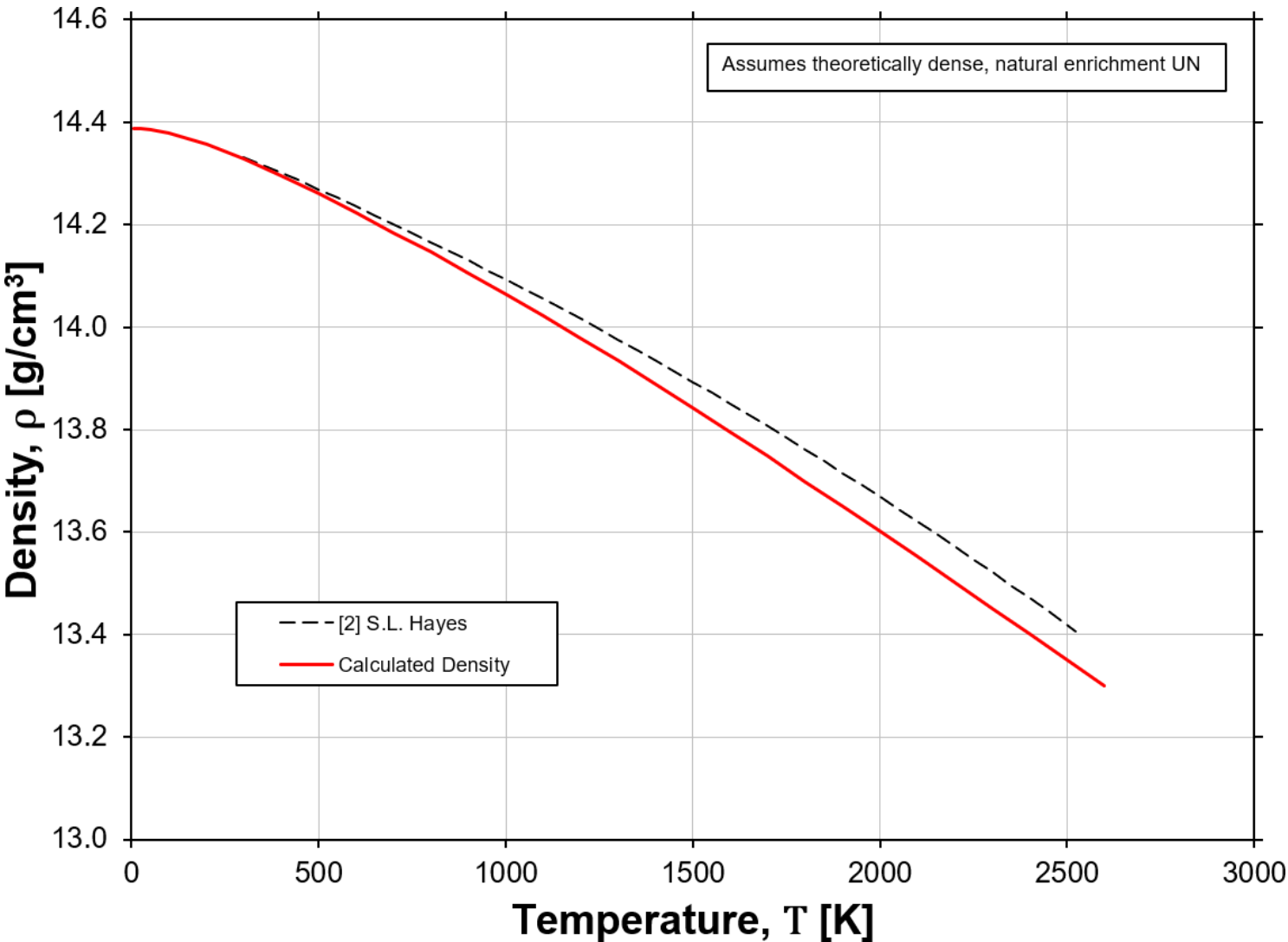


Figure 6.1.1-2: Density versus Temperature for UN. Calculated from fitted trend of the Thermal Expansion data with comparison to trend from Hayes (1990).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Density with Temperature

## 100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 5                 | ( -450.7 ) | 14387             | ( 898.2 )              | 1300              | ( 1880.3 ) | 13934             | ( 869.9 )              |
| 25                | ( -414.7 ) | 14387             | ( 898.1 )              | 1400              | ( 2060.3 ) | 13888             | ( 867.0 )              |
| 50                | ( -369.7 ) | 14385             | ( 898.1 )              | 1500              | ( 2240.3 ) | 13842             | ( 864.2 )              |
| 100               | ( -279.7 ) | 14379             | ( 897.7 )              | 1600              | ( 2420.3 ) | 13795             | ( 861.2 )              |
| 200               | ( -99.7 )  | 14357             | ( 896.3 )              | 1700              | ( 2600.3 ) | 13748             | ( 858.3 )              |
| 300               | ( 80.3 )   | 14329             | ( 894.5 )              | 1800              | ( 2780.3 ) | 13699             | ( 855.3 )              |
| 400               | ( 260.3 )  | 14295             | ( 892.4 )              | 1900              | ( 2960.3 ) | 13651             | ( 852.2 )              |
| 500               | ( 440.3 )  | 14260             | ( 890.2 )              | 2000              | ( 3140.3 ) | 13601             | ( 849.1 )              |
| 600               | ( 620.3 )  | 14223             | ( 887.9 )              | 2100              | ( 3320.3 ) | 13552             | ( 846.0 )              |
| 700               | ( 800.3 )  | 14185             | ( 885.6 )              | 2200              | ( 3500.3 ) | 13502             | ( 842.9 )              |
| 800               | ( 980.3 )  | 14146             | ( 883.1 )              | 2300              | ( 3680.3 ) | 13451             | ( 839.8 )              |
| 900               | ( 1160.3 ) | 14106             | ( 880.6 )              | 2400              | ( 3860.3 ) | 13401             | ( 836.6 )              |
| 1000              | ( 1340.3 ) | 14064             | ( 878.0 )              | 2500              | ( 4040.3 ) | 13350             | ( 833.4 )              |
| 1100              | ( 1520.3 ) | 14022             | ( 875.4 )              | 2600              | ( 4220.3 ) | 13299             | ( 830.2 )              |
| 1200              | ( 1700.3 ) | 13978             | ( 872.6 )              |                   |            |                   |                        |

**Application Notes:** Density is calculated here as a function of thermal expansion as seen in the equation below, to approximate property trend as a function of temperature. This trend is compared against trend from reference [2].

### Density Calculation:

$$\rho(T) = \rho_{RT}(1 - P)/(1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 14,355 \text{ [kg/m}^3\text{]}$$

$P$  = Fractional Porosity

$T$  = Temperature [K]

**Temperature Range:**  $5 \leq T \leq 2600$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

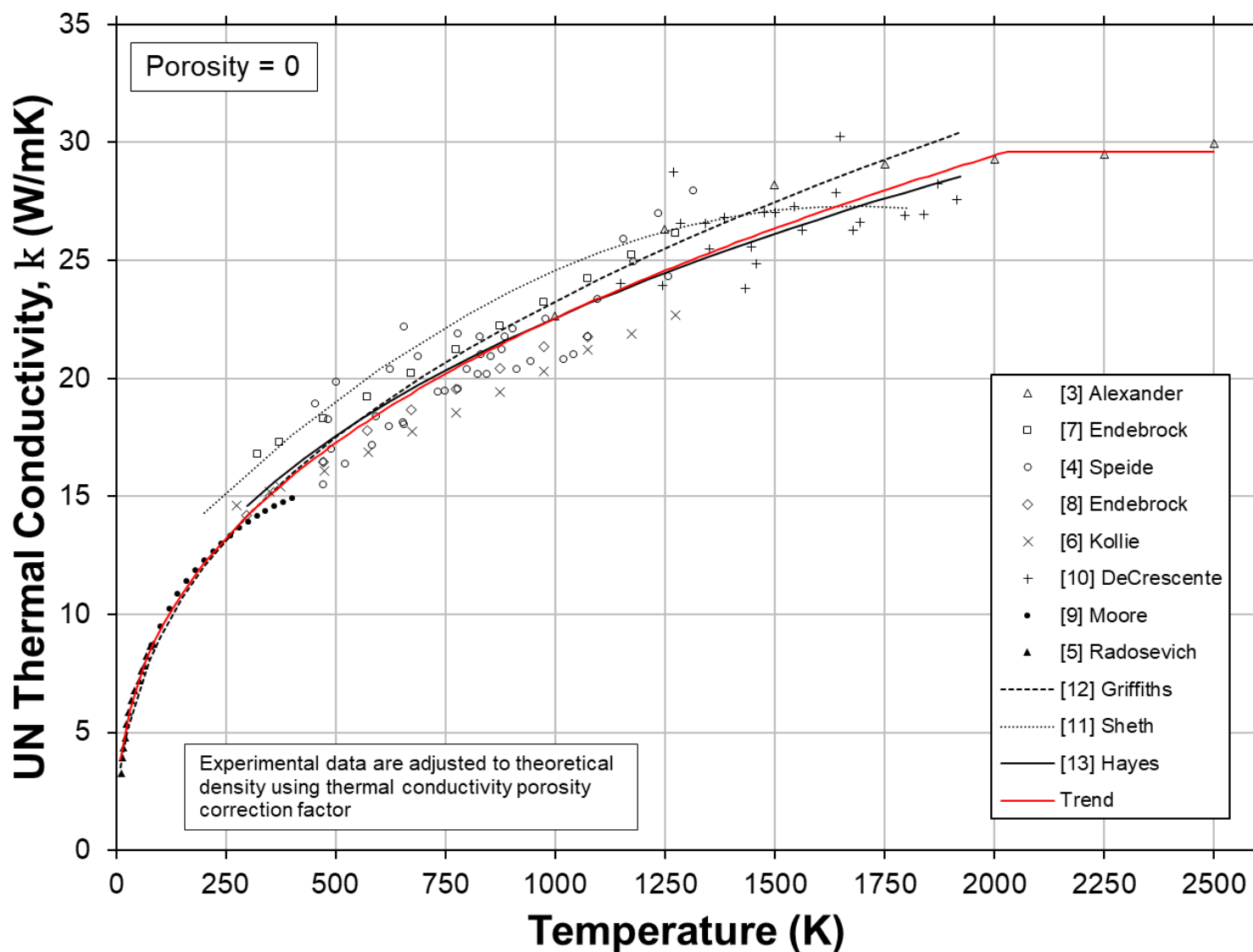


Figure 6.1.1-3: Thermal Conductivity versus Temperature of UN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 10                | ( -441.7 ) | 3.84                       | ( 26.63 )                           | 1200              | ( 1700.3 ) | 24.19                      | ( 167.82 )                          |
| 20                | ( -423.7 ) | 5.01                       | ( 34.76 )                           | 1300              | ( 1880.3 ) | 24.94                      | ( 173.06 )                          |
| 50                | ( -369.7 ) | 7.13                       | ( 49.45 )                           | 1400              | ( 2060.3 ) | 25.66                      | ( 178.06 )                          |
| 100               | ( -279.7 ) | 9.30                       | ( 64.55 )                           | 1500              | ( 2240.3 ) | 26.35                      | ( 182.85 )                          |
| 200               | ( -99.7 )  | 12.14                      | ( 84.26 )                           | 1600              | ( 2420.3 ) | 27.02                      | ( 187.44 )                          |
| 300               | ( 80.3 )   | 14.19                      | ( 98.48 )                           | 1700              | ( 2600.3 ) | 27.65                      | ( 191.86 )                          |
| 400               | ( 260.3 )  | 15.85                      | ( 110.00 )                          | 1800              | ( 2780.3 ) | 28.27                      | ( 196.13 )                          |
| 500               | ( 440.3 )  | 17.27                      | ( 119.85 )                          | 1900              | ( 2960.3 ) | 28.86                      | ( 200.25 )                          |
| 600               | ( 620.3 )  | 18.53                      | ( 128.55 )                          | 2000              | ( 3140.3 ) | 29.44                      | ( 204.24 )                          |
| 700               | ( 800.3 )  | 19.66                      | ( 136.40 )                          | 2100              | ( 3320.3 ) | 29.58                      | ( 205.23 )                          |
| 800               | ( 980.3 )  | 20.70                      | ( 143.59 )                          | 2200              | ( 3500.3 ) | 29.58                      | ( 205.23 )                          |
| 900               | ( 1160.3 ) | 21.65                      | ( 150.24 )                          | 2300              | ( 3680.3 ) | 29.58                      | ( 205.23 )                          |
| 1000              | ( 1340.3 ) | 22.55                      | ( 156.45 )                          | 2400              | ( 3860.3 ) | 29.58                      | ( 205.23 )                          |
| 1100              | ( 1520.3 ) | 23.39                      | ( 162.29 )                          | 2401              | ( 3862.1 ) | 29.58                      | ( 205.23 )                          |

**Application Notes:** Data for thermal conductivity is collected from references [3-13] and fitted with the equations below to approximate the property trend with respect to temperature and porosity.

## Fit Equation:

$$k(T) = A0 \cdot \left( \frac{T}{1000} \right)^N$$

$$k(T) = \text{Thermal Conductivity [W / (m · K)]}$$

$$k_p = k \cdot (1 - P) / (1 + P)$$

$$P = \text{Fractional Porosity (0 ≤ P ≤ 0.2)}$$

$$T = \text{Temperature [K]}$$

## Constants:

| T Range [K]: | <u>10 ≤ T ≤ 2025</u> | <u>2025 &lt; T ≤ 2500</u> |
|--------------|----------------------|---------------------------|
| A0 =         | 22.55                | 29.58                     |
| N =          | 0.3845               | 0                         |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Thermal Conductivity with Porosity

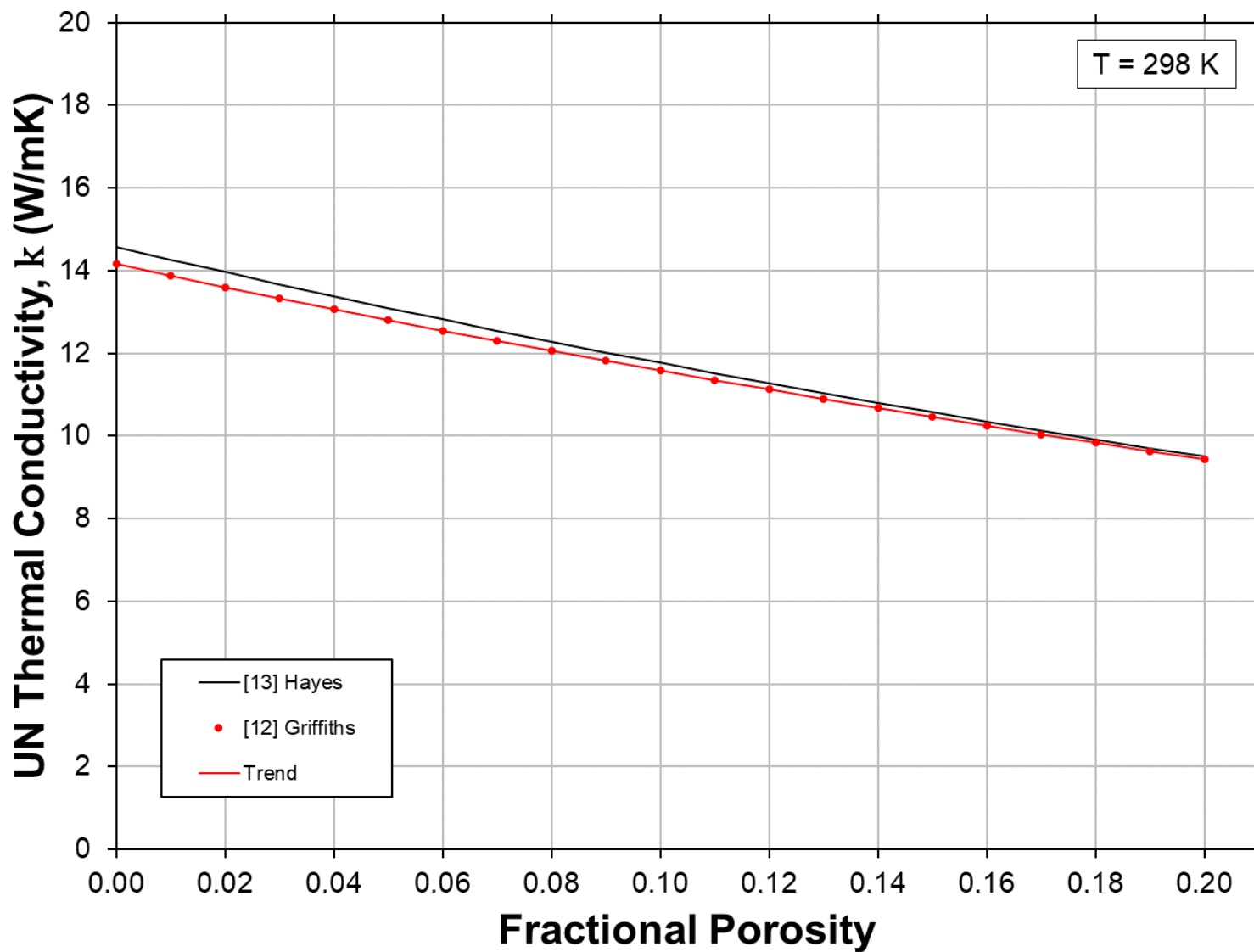


Figure 6.1.1-4: Thermal Conductivity versus Fractional Porosity of UN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Thermal Conductivity with Porosity

Room Temperature

| Porosity ( P ) | Thermal Conductivity ( k ) |                                     |
|----------------|----------------------------|-------------------------------------|
|                | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 0.00           | 14.16                      | ( 98.23 )                           |
| 0.02           | 13.60                      | ( 94.37 )                           |
| 0.04           | 13.07                      | ( 90.67 )                           |
| 0.06           | 12.55                      | ( 87.11 )                           |
| 0.08           | 12.06                      | ( 83.67 )                           |
| 0.10           | 11.58                      | ( 80.37 )                           |
| 0.12           | 11.12                      | ( 77.18 )                           |
| 0.14           | 10.68                      | ( 74.10 )                           |
| 0.16           | 10.25                      | ( 71.13 )                           |
| 0.18           | 9.84                       | ( 68.26 )                           |
| 0.20           | 9.44                       | ( 65.48 )                           |

**Application Notes:** Data for thermal conductivity is collected from references [3-13], and fitted with the equations below to approximate the property trend with respect to temperature and porosity.

## Fit Equation:

$$k(T) = A0 \cdot \left( \frac{T}{1000} \right)^N$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$$k_p = k \cdot (1 - P)/(1 + P)$$

$P$  = Fractional Porosity ( $0 \leq P \leq 0.2$ )

$T$  = Temperature [K]

## Constants:

| T Range [K]: | $10 \leq T \leq 2025$ | $2025 < T \leq 2500$ |
|--------------|-----------------------|----------------------|
| A0 =         | 22.55                 | 29.58                |
| N =          | 0.3845                | 0                    |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

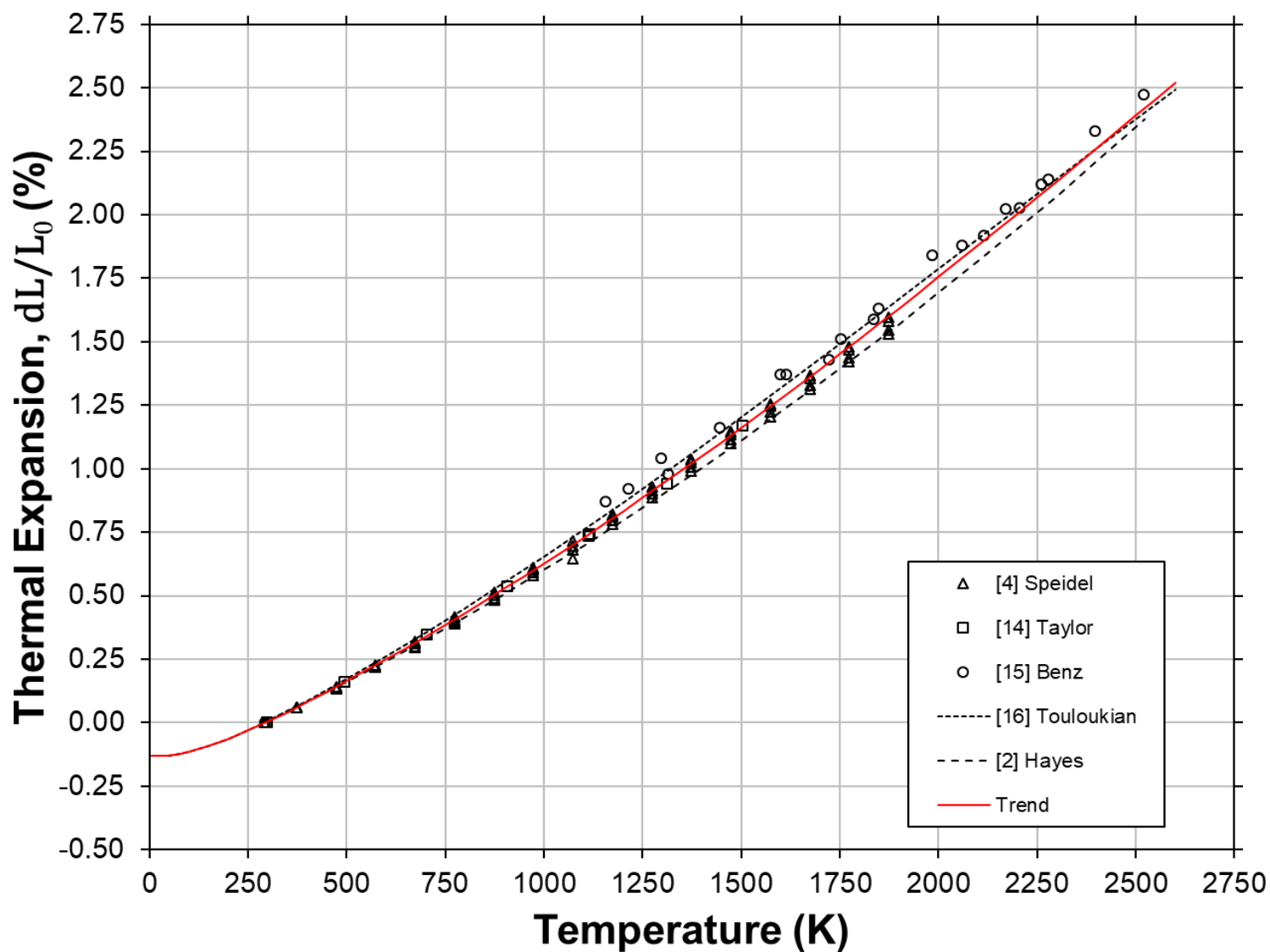


Figure 6.1.1-5: Thermal Expansion versus Temperature of UN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 5                 | ( -450.7 ) | -0.132                                  | 1300              | ( 1880.3 ) | 0.940                                   |
| 25                | ( -414.7 ) | -0.131                                  | 1400              | ( 2060.3 ) | 1.049                                   |
| 50                | ( -369.7 ) | -0.128                                  | 1500              | ( 2240.3 ) | 1.161                                   |
| 100               | ( -279.7 ) | -0.115                                  | 1600              | ( 2420.3 ) | 1.276                                   |
| 200               | ( -99.7 )  | -0.064                                  | 1700              | ( 2600.3 ) | 1.393                                   |
| 300               | ( 80.3 )   | 0.003                                   | 1800              | ( 2780.3 ) | 1.511                                   |
| 400               | ( 260.3 )  | 0.081                                   | 1900              | ( 2960.3 ) | 1.632                                   |
| 500               | ( 440.3 )  | 0.164                                   | 2000              | ( 3140.3 ) | 1.755                                   |
| 600               | ( 620.3 )  | 0.250                                   | 2100              | ( 3320.3 ) | 1.879                                   |
| 700               | ( 800.3 )  | 0.339                                   | 2200              | ( 3500.3 ) | 2.005                                   |
| 800               | ( 980.3 )  | 0.432                                   | 2300              | ( 3680.3 ) | 2.132                                   |
| 900               | ( 1160.3 ) | 0.527                                   | 2400              | ( 3860.3 ) | 2.260                                   |
| 1000              | ( 1340.3 ) | 0.626                                   | 2500              | ( 4040.3 ) | 2.390                                   |
| 1100              | ( 1520.3 ) | 0.728                                   | 2600              | ( 4220.3 ) | 2.521                                   |
| 1200              | ( 1700.3 ) | 0.832                                   |                   |            |                                         |

**Application Notes:** Data for thermal expansion is collected from references [2, 4, 14-16] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

| T Range [K]: | $5 \leq T \leq 293$ | $293 < T \leq 2600$ |
|--------------|---------------------|---------------------|
| A0 =         | -0.1313             | -0.2103             |
| A1 =         | -0.05692            | 0.65                |
| A2 =         | 2.515               | 0.2067              |
| A3 =         | -2.722              | -0.02025            |

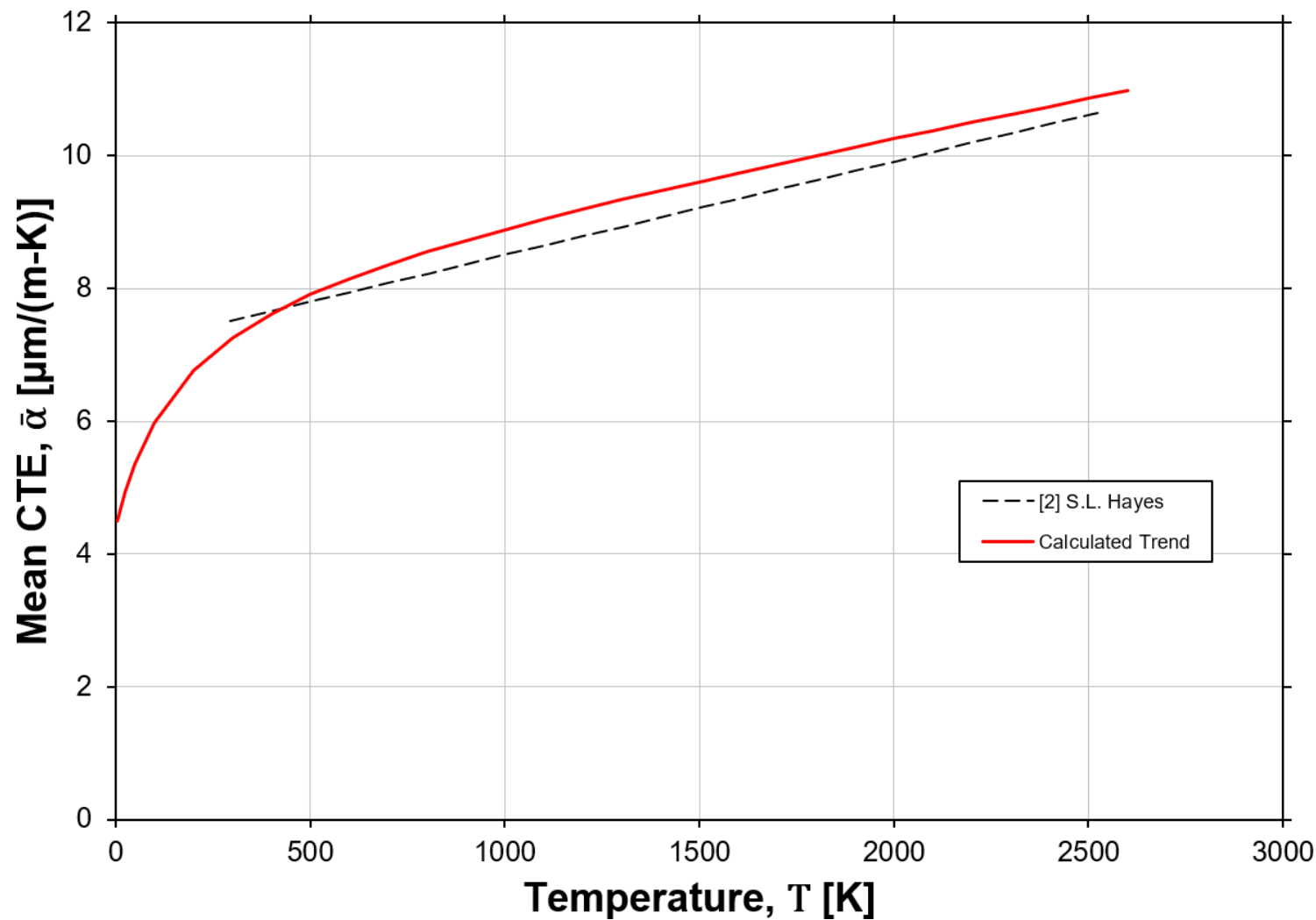


Figure 6.1.1-6: Mean Coefficient of Thermal Expansion versus Temperature of UN. Calculated from fitted trend of the Thermal Expansion data with comparison to trend from Hayes (1990).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 5                 | ( -450.7 ) | 4.503                                 | ( 2.502 )                                           | 1300              | ( 1880.3 ) | 9.333                                 | ( 5.185 )                                           |
| 25                | ( -414.7 ) | 4.926                                 | ( 2.736 )                                           | 1400              | ( 2060.3 ) | 9.472                                 | ( 5.262 )                                           |
| 50                | ( -369.7 ) | 5.350                                 | ( 2.972 )                                           | 1500              | ( 2240.3 ) | 9.608                                 | ( 5.338 )                                           |
| 100               | ( -279.7 ) | 5.976                                 | ( 3.320 )                                           | 1600              | ( 2420.3 ) | 9.741                                 | ( 5.412 )                                           |
| 200               | ( -99.7 )  | 6.763                                 | ( 3.757 )                                           | 1700              | ( 2600.3 ) | 9.872                                 | ( 5.484 )                                           |
| 300               | ( 80.3 )   | 7.260                                 | ( 4.033 )                                           | 1800              | ( 2780.3 ) | 10.000                                | ( 5.556 )                                           |
| 400               | ( 260.3 )  | 7.621                                 | ( 4.234 )                                           | 1900              | ( 2960.3 ) | 10.127                                | ( 5.626 )                                           |
| 500               | ( 440.3 )  | 7.907                                 | ( 4.393 )                                           | 2000              | ( 3140.3 ) | 10.252                                | ( 5.696 )                                           |
| 600               | ( 620.3 )  | 8.147                                 | ( 4.526 )                                           | 2100              | ( 3320.3 ) | 10.376                                | ( 5.764 )                                           |
| 700               | ( 800.3 )  | 8.357                                 | ( 4.643 )                                           | 2200              | ( 3500.3 ) | 10.499                                | ( 5.833 )                                           |
| 800               | ( 980.3 )  | 8.547                                 | ( 4.748 )                                           | 2300              | ( 3680.3 ) | 10.620                                | ( 5.900 )                                           |
| 900               | ( 1160.3 ) | 8.722                                 | ( 4.845 )                                           | 2400              | ( 3860.3 ) | 10.741                                | ( 5.967 )                                           |
| 1000              | ( 1340.3 ) | 8.885                                 | ( 4.936 )                                           | 2500              | ( 4040.3 ) | 10.861                                | ( 6.034 )                                           |
| 1100              | ( 1520.3 ) | 9.041                                 | ( 5.023 )                                           | 2600              | ( 4220.3 ) | 10.981                                | ( 6.100 )                                           |
| 1200              | ( 1700.3 ) | 9.190                                 | ( 5.105 )                                           |                   |            |                                       |                                                     |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated as a function thermal expansion data, then fitted with the equation below to approximate property trend with respect to temperature. Fitted trend is compared against trend from reference [2].

## Fit Equation:

$$\bar{\alpha}(T) = \left[ A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right] / \left[ 1 + A_0 \cdot \left( \frac{T}{1000} \right) \right]$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

T = Temperature [K]

## Constants:

T. Range [K]:  $5 \leq T \leq 2600$

A0 = 4.383

A1 = 50.92

A2 = 6.584

A\_0 = 5.965



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Specific Heat with Temperature

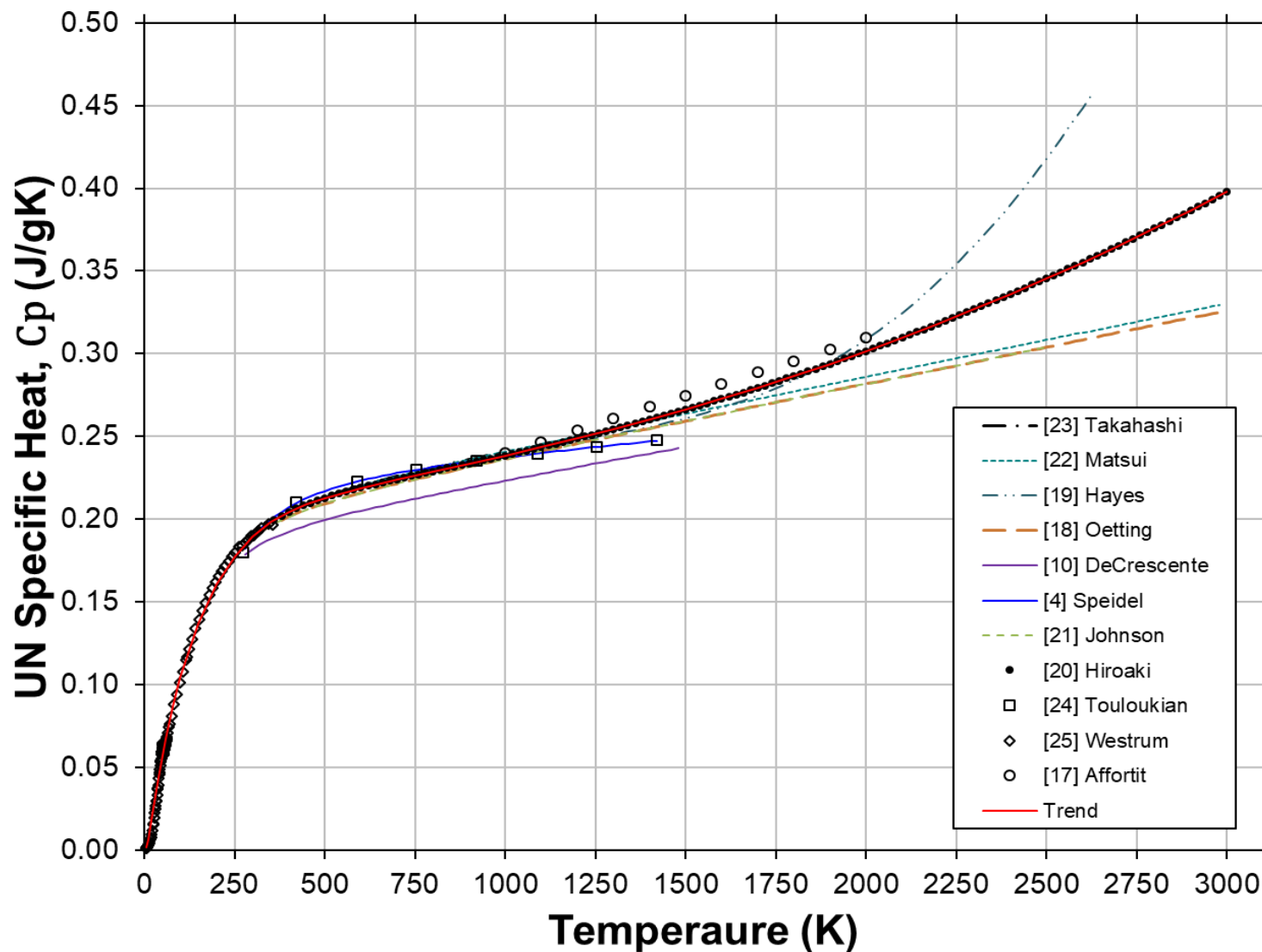


Figure 6.1.1-7: Specific Heat versus Temperature of UN. Displaying fitted trend and data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 5                 | ( -450.7 ) | 0.002                            | ( 0.000 )       | 1200              | ( 1700.3 ) | 0.249                            | ( 0.059 )       |
| 50                | ( -369.7 ) | 0.056                            | ( 0.013 )       | 1300              | ( 1880.3 ) | 0.254                            | ( 0.061 )       |
| 100               | ( -279.7 ) | 0.104                            | ( 0.025 )       | 1400              | ( 2060.3 ) | 0.260                            | ( 0.062 )       |
| 200               | ( -99.7 )  | 0.161                            | ( 0.038 )       | 1500              | ( 2240.3 ) | 0.266                            | ( 0.064 )       |
| 300               | ( 80.3 )   | 0.190                            | ( 0.045 )       | 1600              | ( 2420.3 ) | 0.273                            | ( 0.065 )       |
| 400               | ( 260.3 )  | 0.204                            | ( 0.049 )       | 1700              | ( 2600.3 ) | 0.279                            | ( 0.067 )       |
| 500               | ( 440.3 )  | 0.213                            | ( 0.051 )       | 1800              | ( 2780.3 ) | 0.287                            | ( 0.068 )       |
| 600               | ( 620.3 )  | 0.219                            | ( 0.052 )       | 2000              | ( 3140.3 ) | 0.302                            | ( 0.072 )       |
| 700               | ( 800.3 )  | 0.224                            | ( 0.054 )       | 2200              | ( 3500.3 ) | 0.318                            | ( 0.076 )       |
| 800               | ( 980.3 )  | 0.229                            | ( 0.055 )       | 2400              | ( 3860.3 ) | 0.336                            | ( 0.080 )       |
| 900               | ( 1160.3 ) | 0.234                            | ( 0.056 )       | 2600              | ( 4220.3 ) | 0.355                            | ( 0.085 )       |
| 1000              | ( 1340.3 ) | 0.238                            | ( 0.057 )       | 2800              | ( 4580.3 ) | 0.376                            | ( 0.090 )       |
| 1100              | ( 1520.3 ) | 0.243                            | ( 0.058 )       | 3000              | ( 4940.3 ) | 0.398                            | ( 0.095 )       |

**Application Notes:** Data for specific heat is collected from references [4, 10, 17-25] and fitted with the equations below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $5 \leq T < 293$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right]$$

For temperature range:  $293 \leq T \leq 3000$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_2 / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

| T. Range [K]: | $5 \leq T < 293$ | $293 \leq T \leq 3000$ |
|---------------|------------------|------------------------|
| N =           | 2.168            | B0 = 2.141E-01         |
| A0 =          | 2.784E+02        | B1 = 9.746E-03         |
| A1 =          | 9.191E+01        | B2 = 1.718E-02         |
| A2 =          | 7.568E+02        | B_2 = -2.607E-03       |
| A3 =          | 4.086E+02        |                        |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

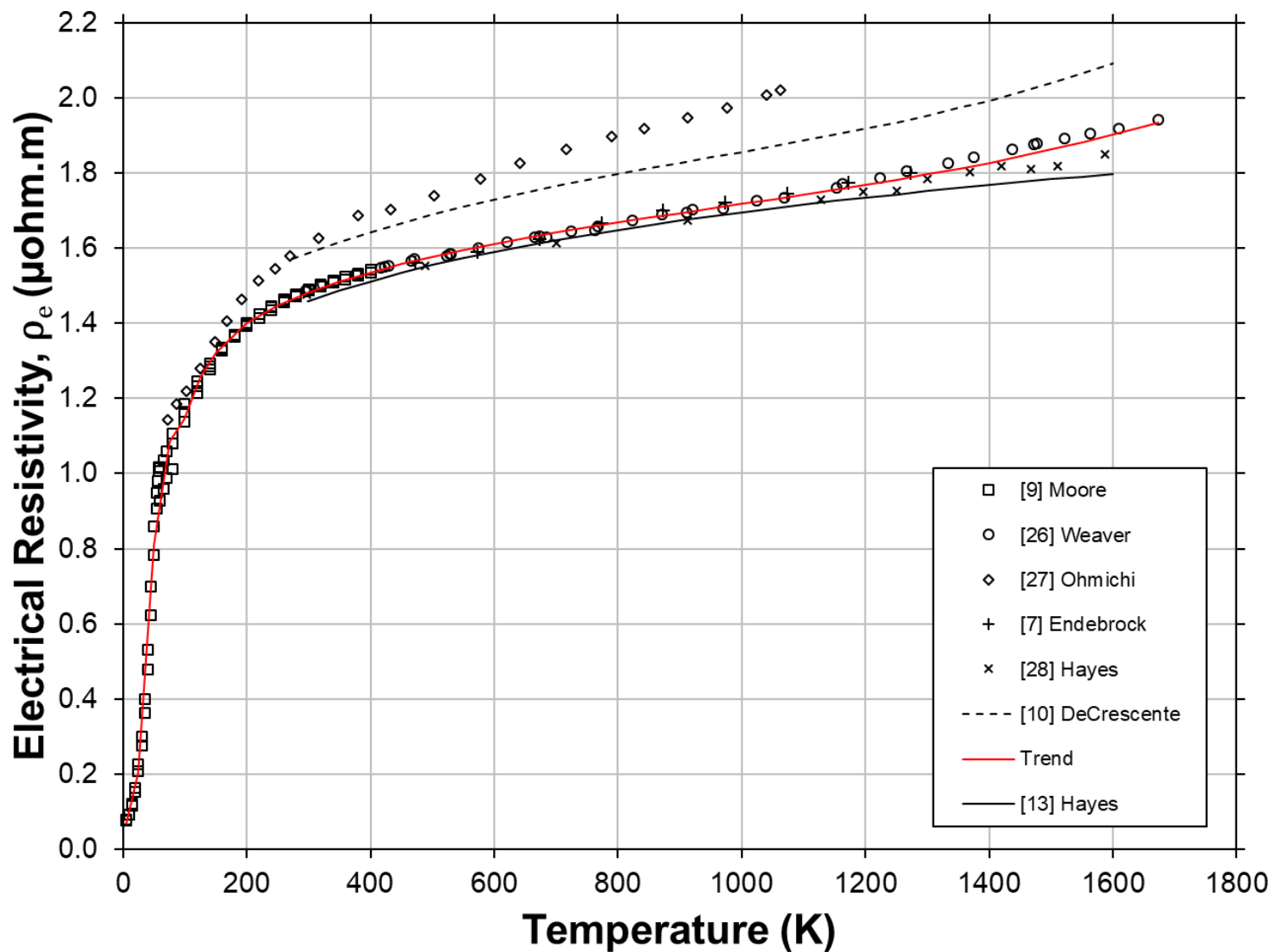


Figure 6.1.1-8: Electrical Resistivity versus Temperature of Uranium Nitride.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 5                 | ( -450.7 ) | 0.068                               | ( 2.66 )                | 1100              | ( 1520.3 ) | 1.742                               | ( 68.59 )               |
| 25                | ( -414.7 ) | 0.203                               | ( 8.00 )                | 1200              | ( 1700.3 ) | 1.768                               | ( 69.61 )               |
| 50                | ( -369.7 ) | 0.809                               | ( 31.87 )               | 1300              | ( 1880.3 ) | 1.796                               | ( 70.71 )               |
| 100               | ( -279.7 ) | 1.147                               | ( 45.15 )               | 1400              | ( 2060.3 ) | 1.827                               | ( 71.93 )               |
| 200               | ( -99.7 )  | 1.400                               | ( 55.10 )               | 1500              | ( 2240.3 ) | 1.862                               | ( 73.31 )               |
| 300               | ( 80.3 )   | 1.483                               | ( 58.38 )               | 1600              | ( 2420.3 ) | 1.902                               | ( 74.89 )               |
| 400               | ( 260.3 )  | 1.536                               | ( 60.47 )               | 1673              | ( 2551.7 ) | 1.935                               | ( 76.18 )               |
| 500               | ( 440.3 )  | 1.577                               | ( 62.09 )               |                   |            |                                     |                         |
| 600               | ( 620.3 )  | 1.611                               | ( 63.44 )               |                   |            |                                     |                         |
| 700               | ( 800.3 )  | 1.641                               | ( 64.61 )               |                   |            |                                     |                         |
| 800               | ( 980.3 )  | 1.668                               | ( 65.67 )               |                   |            |                                     |                         |
| 900               | ( 1160.3 ) | 1.693                               | ( 66.66 )               |                   |            |                                     |                         |
| 1000              | ( 1340.3 ) | 1.718                               | ( 67.62 )               |                   |            |                                     |                         |

**Application Notes:** Data for electrical resistivity is collected from references [7, 9, 10, 13, 26-28] and fitted with the equations below to best fit data and approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $5 \leq T < 100$

$$\rho_e(T) = \left[ A \cdot \left( \frac{T}{1000} \right)^N + A0 \right] / \left[ 1 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $100 \leq T \leq 1673$

$$\rho_e(T) = B0 + B1 \cdot \left( \frac{T}{1000} \right) + B2 \cdot \left( \frac{T}{1000} \right)^2 + B3 \cdot \left( \frac{T}{1000} \right)^3 + B\_2 / \left( \frac{T}{1000} \right)^2$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

| T. Range [K]: | $5 \leq T < 100$ | $100 \leq T \leq 1673$ |
|---------------|------------------|------------------------|
| N =           | 3                | B0 = 1.353             |
| A =           | 2.032E+03        | B1 = 6.467E-01         |
| A0 =          | 5.703E-02        | B2 = -4.299E-01        |
| A1 =          | -3.273E+01       | B3 = 1.504E-01         |
| A2 =          | 4.083E+02        | B\_2 = -2.668E-03      |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Young's Modulus with Porosity

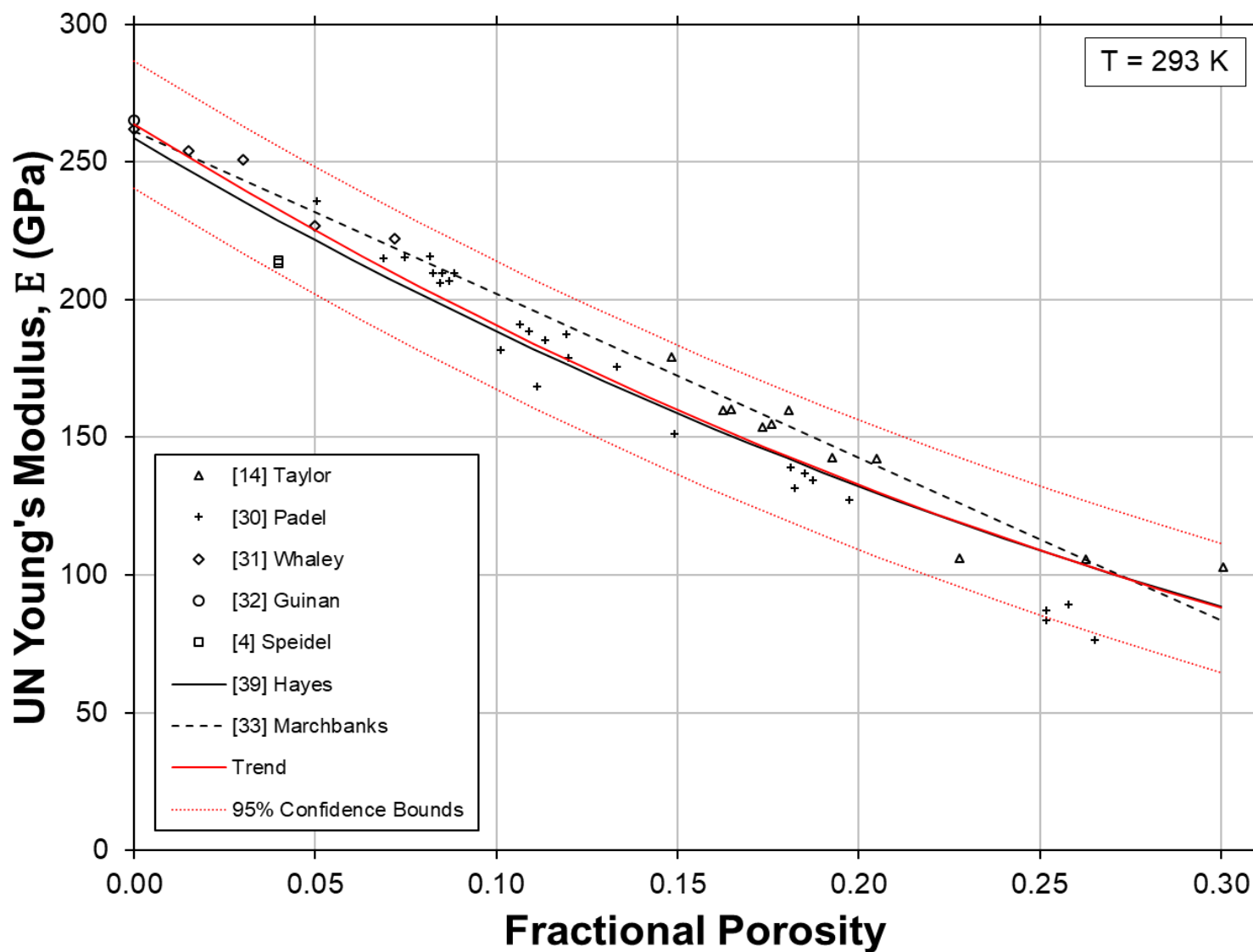


Figure 6.1.1-9: Young's Modulus versus Fractional Porosity of UN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Young's Modulus with Porosity

Room Temperature

| Porosity ( P ) | Young's Modulus ( E ) |           |
|----------------|-----------------------|-----------|
|                | GPa                   | ( Msi )   |
| 0.00           | 263.58                | ( 38.25 ) |
| 0.02           | 247.72                | ( 35.94 ) |
| 0.04           | 232.52                | ( 33.74 ) |
| 0.06           | 217.97                | ( 31.63 ) |
| 0.08           | 204.04                | ( 29.61 ) |
| 0.10           | 190.72                | ( 27.67 ) |
| 0.12           | 178.00                | ( 25.83 ) |
| 0.14           | 165.87                | ( 24.07 ) |
| 0.16           | 154.30                | ( 22.39 ) |
| 0.18           | 143.30                | ( 20.79 ) |
| 0.20           | 132.83                | ( 19.27 ) |
| 0.22           | 122.90                | ( 17.83 ) |
| 0.24           | 113.47                | ( 16.46 ) |
| 0.26           | 104.55                | ( 15.17 ) |
| 0.28           | 96.11                 | ( 13.95 ) |
| 0.30           | 88.15                 | ( 12.79 ) |

**Application Notes:** Data for Young's modulus is collected from references [4, 14, 29-34] and fitted with the equation below to approximate property trend with respect to temperature and porosity.

## Fit Equations:

$$E(T, P) = A \cdot (1 - P)^N [A_0 - A_1 \cdot (T - 273)]$$

$$E(T, P) = \text{Young's Modulus [GPa]}$$

$$T = \text{Temperature [K] for } 273 \leq T \leq 1700$$

$$P = \text{Fractional Porosity for } 0 \leq P \leq 0.3$$

## Constants:

$$A = 263.547$$

$$N = 3.071$$

$$A_0 = 1.003$$

$$A_1 = 1.148\text{E-}04$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Young's Modulus with Temperature

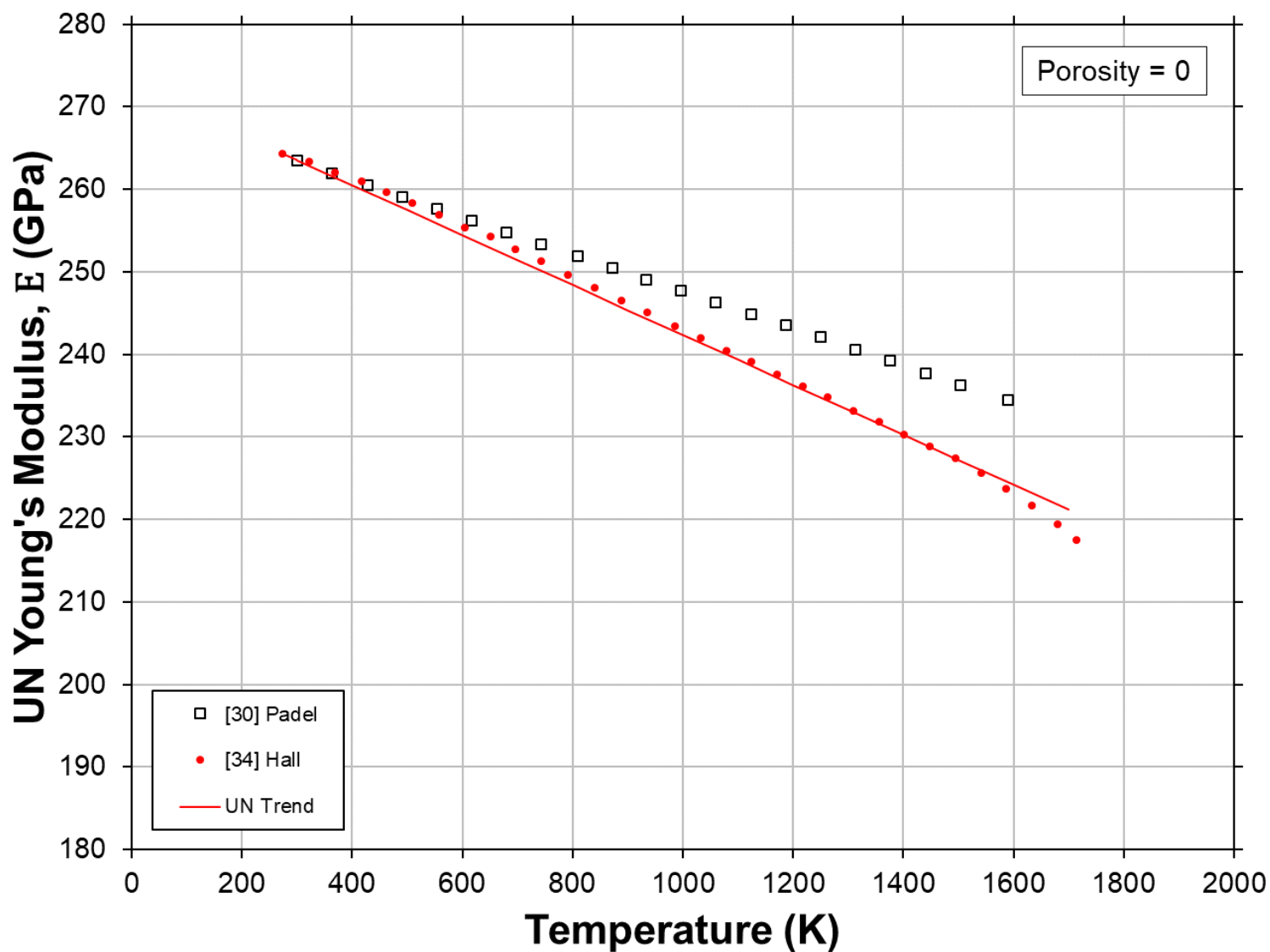


Figure 6.1.1-10: Young's Modulus versus Temperature of UN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 273               | ( 31.7 )   | 264.34                | ( 38.36 ) | 1000              | ( 1340.3 ) | 242.34                | ( 35.16 ) |
| 300               | ( 80.3 )   | 263.52                | ( 38.24 ) | 1050              | ( 1430.3 ) | 240.83                | ( 34.94 ) |
| 350               | ( 170.3 )  | 262.01                | ( 38.02 ) | 1100              | ( 1520.3 ) | 239.32                | ( 34.72 ) |
| 400               | ( 260.3 )  | 260.50                | ( 37.80 ) | 1150              | ( 1610.3 ) | 237.80                | ( 34.51 ) |
| 450               | ( 350.3 )  | 258.98                | ( 37.58 ) | 1200              | ( 1700.3 ) | 236.29                | ( 34.29 ) |
| 500               | ( 440.3 )  | 257.47                | ( 37.36 ) | 1250              | ( 1790.3 ) | 234.78                | ( 34.07 ) |
| 550               | ( 530.3 )  | 255.96                | ( 37.14 ) | 1300              | ( 1880.3 ) | 233.27                | ( 33.85 ) |
| 600               | ( 620.3 )  | 254.44                | ( 36.92 ) | 1350              | ( 1970.3 ) | 231.75                | ( 33.63 ) |
| 650               | ( 710.3 )  | 252.93                | ( 36.70 ) | 1400              | ( 2060.3 ) | 230.24                | ( 33.41 ) |
| 700               | ( 800.3 )  | 251.42                | ( 36.48 ) | 1450              | ( 2150.3 ) | 228.73                | ( 33.19 ) |
| 750               | ( 890.3 )  | 249.91                | ( 36.26 ) | 1500              | ( 2240.3 ) | 227.21                | ( 32.97 ) |
| 800               | ( 980.3 )  | 248.39                | ( 36.04 ) | 1550              | ( 2330.3 ) | 225.70                | ( 32.75 ) |
| 850               | ( 1070.3 ) | 246.88                | ( 35.82 ) | 1600              | ( 2420.3 ) | 224.19                | ( 32.53 ) |
| 900               | ( 1160.3 ) | 245.37                | ( 35.60 ) | 1650              | ( 2510.3 ) | 222.68                | ( 32.31 ) |
| 950               | ( 1250.3 ) | 243.85                | ( 35.38 ) | 1700              | ( 2600.3 ) | 221.16                | ( 32.09 ) |

**Application Notes:** Data for Young's modulus is collected from references [4, 14, 29-34] and fitted with the equations below to best fit data and approximate property trend with respect to temperature and porosity.

## Fit Equations:

$$E(T, P) = A \cdot (1 - P)^N [A0 - A1 \cdot (T - 273)]$$

$$E(T, P) = \text{Young's Modulus [GPa]}$$

$$T = \text{Temperature [K] for } 273 \leq T \leq 1700$$

$$P = \text{Fractional Porosity for } 0 \leq P \leq 0.3$$

## Constants:

$$T. \text{ Range [K]: } 273 \leq T < 1700$$

$$A = 263.547$$

$$N = 3.071$$

$$A0 = 1.003$$

$$A1 = 1.148\text{E-}04$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Shear Modulus with Porosity

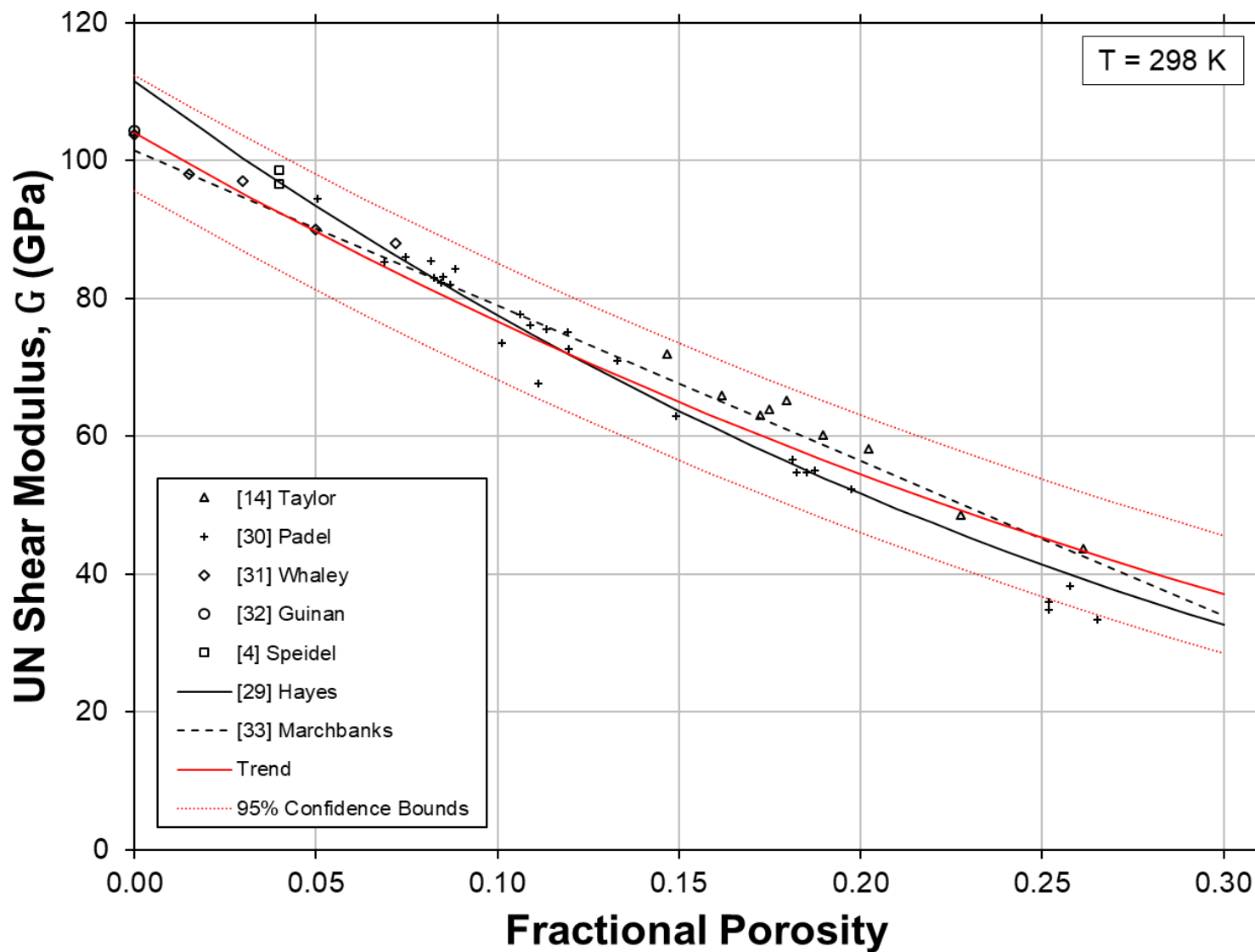


Figure 6.1.1-11: Shear Modulus versus Fractional Porosity of UN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Shear Modulus with Porosity

Room Temperature

| Porosity ( P ) | Shear Modulus ( G ) |           |
|----------------|---------------------|-----------|
|                | GPa                 | ( Msi )   |
| 0.00           | 104.01              | ( 15.09 ) |
| 0.02           | 98.11               | ( 14.24 ) |
| 0.04           | 92.43               | ( 13.41 ) |
| 0.06           | 86.96               | ( 12.62 ) |
| 0.08           | 81.72               | ( 11.86 ) |
| 0.10           | 76.68               | ( 11.13 ) |
| 0.12           | 71.86               | ( 10.43 ) |
| 0.14           | 67.23               | ( 9.76 )  |
| 0.16           | 62.81               | ( 9.11 )  |
| 0.18           | 58.58               | ( 8.50 )  |
| 0.20           | 54.54               | ( 7.91 )  |
| 0.22           | 50.69               | ( 7.36 )  |
| 0.24           | 47.02               | ( 6.82 )  |
| 0.26           | 43.53               | ( 6.32 )  |
| 0.28           | 40.21               | ( 5.83 )  |
| 0.30           | 37.06               | ( 5.38 )  |

**Application Notes:** Data for shear modulus is collected from references [4, 14, 29-33] and fitted with the equation below to approximate property trend with respect to temperature and porosity.

## Fit Equations:

$$G(T,P) = A \cdot (1 - P)^N [A0 - A1 \cdot (T - 273)]$$

$$G(T,P) = \text{Shear Modulus [GPa]}$$

$$T = \text{Temperature [K]}$$

$$P = \text{Fractional Porosity}$$

## Constants:

$$\text{Porosity (P): } 0 \leq P \leq 0.3$$

$$A = 103.999$$

$$N = 2.893$$

$$A0 = 1.003$$

$$A1 = 1.148\text{E-}04$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Shear Modulus with Temperature

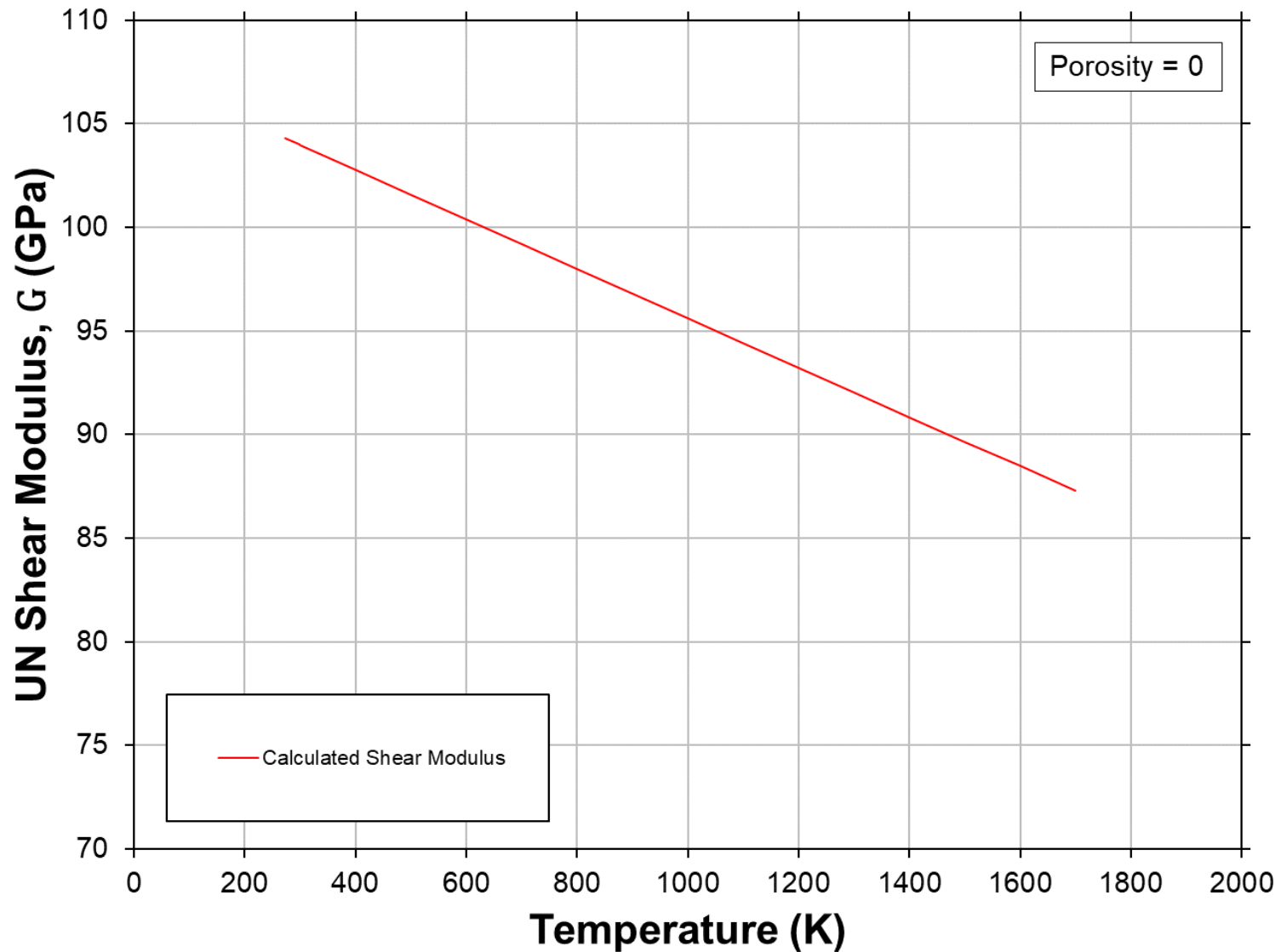


Figure 6.1.1-12: Shear Modulus versus Temperature of UN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( G ) |           | Temperature ( T ) |            | Shear Modulus ( G ) |           |
|-------------------|------------|---------------------|-----------|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   | K                 | ( °F )     | GPa                 | ( Msi )   |
| 273               | ( 31.7 )   | 104.31              | ( 15.14 ) | 1000              | ( 1340.3 ) | 95.63               | ( 13.88 ) |
| 300               | ( 80.3 )   | 103.99              | ( 15.09 ) | 1050              | ( 1430.3 ) | 95.03               | ( 13.79 ) |
| 350               | ( 170.3 )  | 103.39              | ( 15.00 ) | 1100              | ( 1520.3 ) | 94.44               | ( 13.70 ) |
| 400               | ( 260.3 )  | 102.79              | ( 14.92 ) | 1150              | ( 1610.3 ) | 93.84               | ( 13.62 ) |
| 450               | ( 350.3 )  | 102.20              | ( 14.83 ) | 1200              | ( 1700.3 ) | 93.24               | ( 13.53 ) |
| 500               | ( 440.3 )  | 101.60              | ( 14.74 ) | 1250              | ( 1790.3 ) | 92.65               | ( 13.44 ) |
| 550               | ( 530.3 )  | 101.00              | ( 14.66 ) | 1300              | ( 1880.3 ) | 92.05               | ( 13.36 ) |
| 600               | ( 620.3 )  | 100.41              | ( 14.57 ) | 1350              | ( 1970.3 ) | 91.45               | ( 13.27 ) |
| 650               | ( 710.3 )  | 99.81               | ( 14.48 ) | 1400              | ( 2060.3 ) | 90.86               | ( 13.18 ) |
| 700               | ( 800.3 )  | 99.21               | ( 14.40 ) | 1450              | ( 2150.3 ) | 90.26               | ( 13.10 ) |
| 750               | ( 890.3 )  | 98.62               | ( 14.31 ) | 1500              | ( 2240.3 ) | 89.66               | ( 13.01 ) |
| 800               | ( 980.3 )  | 98.02               | ( 14.22 ) | 1550              | ( 2330.3 ) | 89.06               | ( 12.92 ) |
| 850               | ( 1070.3 ) | 97.42               | ( 14.14 ) | 1600              | ( 2420.3 ) | 88.47               | ( 12.84 ) |
| 900               | ( 1160.3 ) | 96.83               | ( 14.05 ) | 1650              | ( 2510.3 ) | 87.87               | ( 12.75 ) |
| 950               | ( 1250.3 ) | 96.23               | ( 13.96 ) | 1700              | ( 2600.3 ) | 87.27               | ( 12.66 ) |

**Application Notes:** Data for shear modulus is collected from references [4, 14, 29-33] and fitted with the equation below to approximate property trend with respect to temperature and porosity.

## Fit Equations:

$$G(T, P) = A \cdot (1 - P)^N [A0 - A1 \cdot (T - 273)]$$

$$G(T, P) = \text{Shear Modulus [GPa]}$$

$$T = \text{Temperature [K]}$$

$$P = \text{Fractional Porosity}$$

## Constants:

Temperature Range [K]:  $273 \leq T \leq 1700$

$$A = 103.999$$

$$N = 2.893$$

$$A0 = 1.003$$

$$A1 = 1.148\text{E-}04$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Poisson's Ratio with Porosity

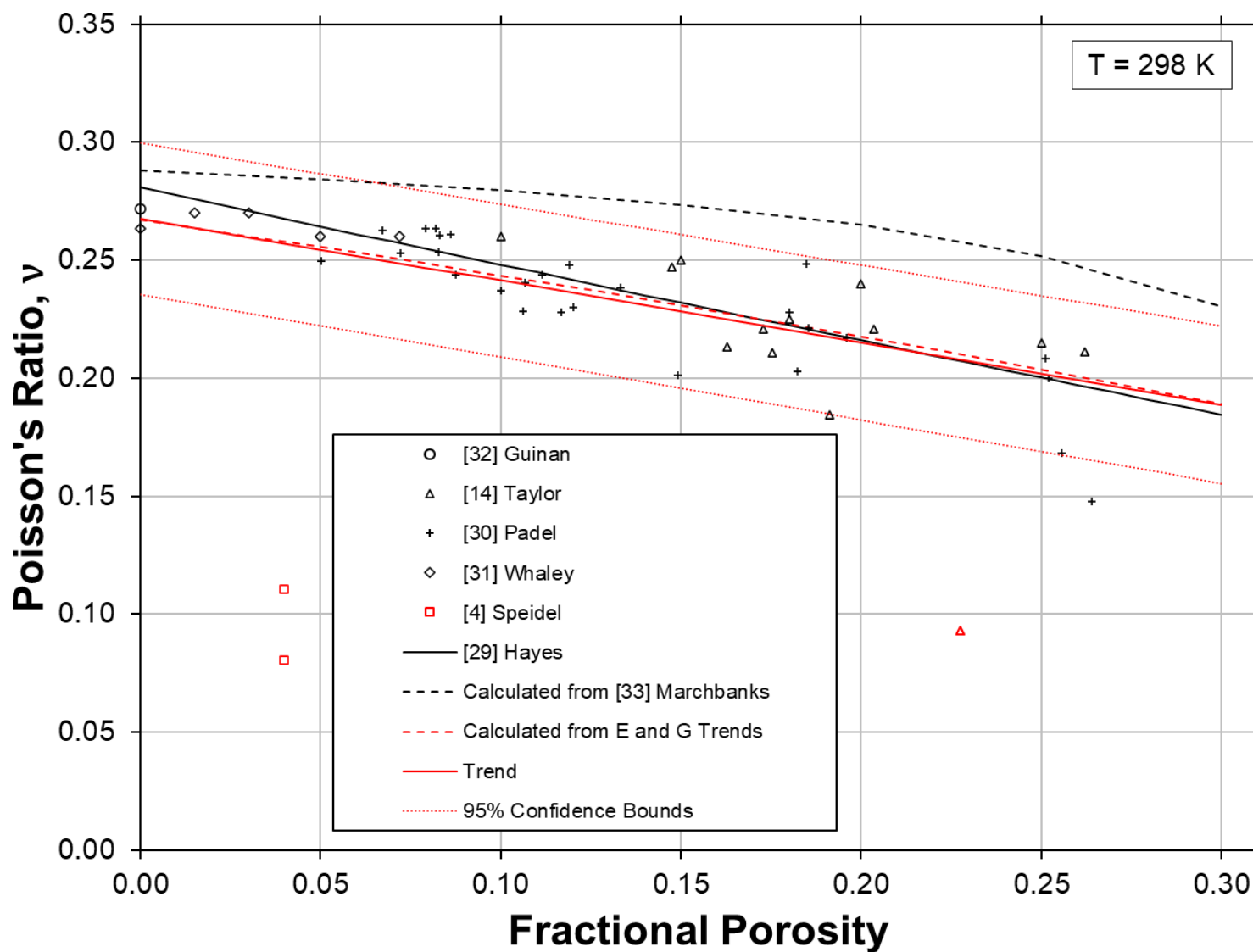


Figure 6.1.1-13: Poisson's Ratio versus Fractional Porosity of UN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Poisson's Ratio with Porosity

Room Temperature

| Porosity ( P ) | Poisson's Ratio ( $\nu$ ) |
|----------------|---------------------------|
| 0.00           | 0.268                     |
| 0.02           | 0.262                     |
| 0.04           | 0.257                     |
| 0.06           | 0.252                     |
| 0.08           | 0.247                     |
| 0.10           | 0.241                     |
| 0.12           | 0.236                     |
| 0.14           | 0.231                     |
| 0.16           | 0.226                     |
| 0.18           | 0.220                     |
| 0.20           | 0.215                     |
| 0.22           | 0.210                     |
| 0.24           | 0.205                     |
| 0.26           | 0.199                     |
| 0.28           | 0.194                     |
| 0.30           | 0.189                     |

**Application Notes:** Data for Poisson's Ratio is collected from references [4, 14, 29-33]. Property trend is calculated as a function of Young's modulus and shear modulus as seen in calculated equations below, and compared against the fitted trend equation below with respect to porosity. Only calculated trend is shown with respect to temperature.

## Fit Equations:

$$\nu(P) = A \cdot (1 - P)^N$$

## Calculated:

$$\nu(P) = E(T, P) / (2 \cdot G(T, P)) - 1$$

$\nu(P)$  = Poisson's Ratio

$P$  = Fractional Porosity

## Constants:

Porosity (P):  $0 \leq P \leq 0.3$

A = 0.2676

N = 0.9772



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

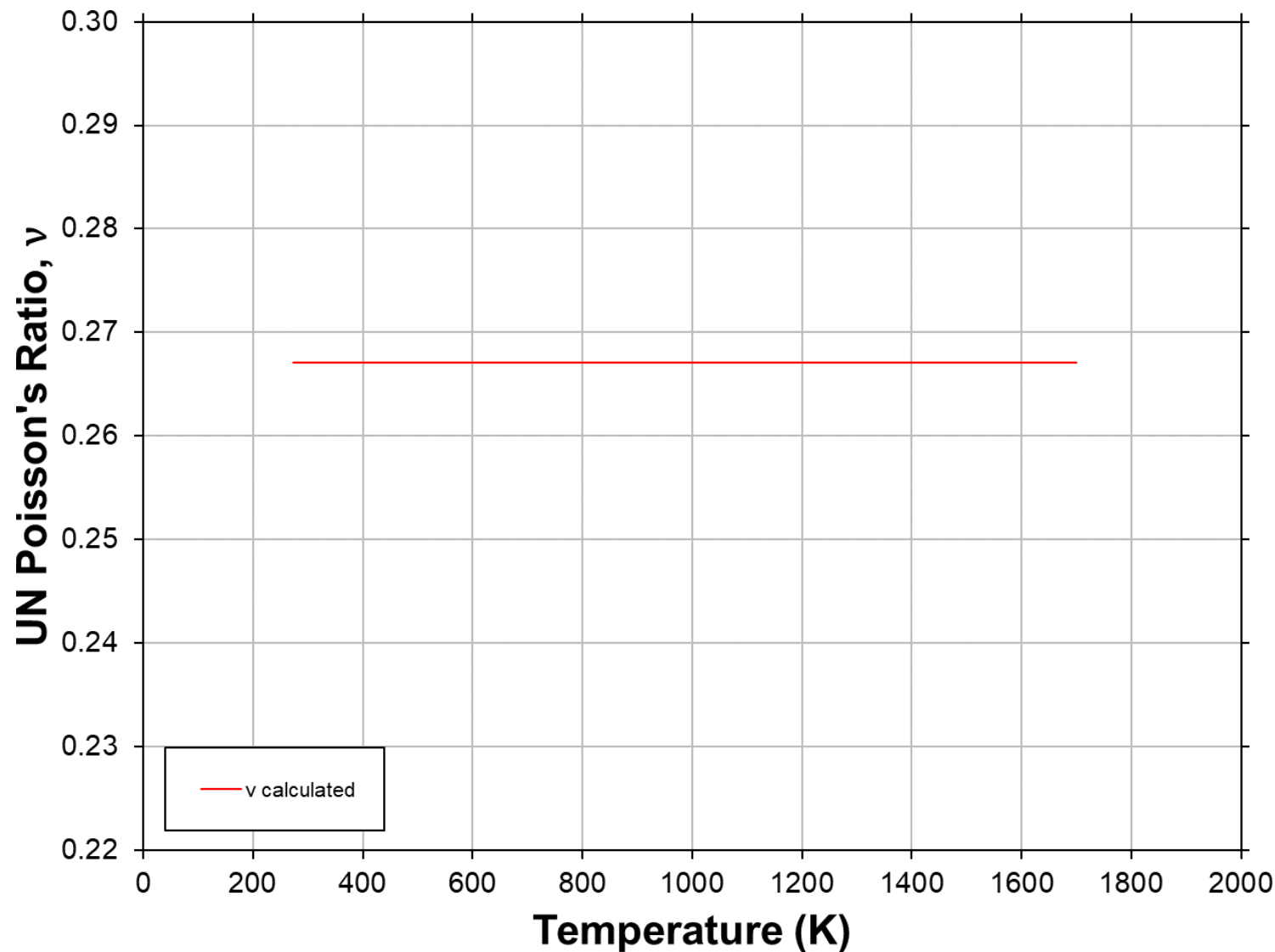


Figure 6.1.1-14: Poisson's Ratio versus Temperature of UN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|------------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )     |                           | K                 | ( °F )     |                           |
| 273               | ( 31.7 )   | 0.268                     | 1000              | ( 1340.3 ) | 0.268                     |
| 300               | ( 80.3 )   | 0.268                     | 1050              | ( 1430.3 ) | 0.268                     |
| 350               | ( 170.3 )  | 0.268                     | 1100              | ( 1520.3 ) | 0.268                     |
| 400               | ( 260.3 )  | 0.268                     | 1150              | ( 1610.3 ) | 0.268                     |
| 450               | ( 350.3 )  | 0.268                     | 1200              | ( 1700.3 ) | 0.268                     |
| 500               | ( 440.3 )  | 0.268                     | 1250              | ( 1790.3 ) | 0.268                     |
| 550               | ( 530.3 )  | 0.268                     | 1300              | ( 1880.3 ) | 0.268                     |
| 600               | ( 620.3 )  | 0.268                     | 1350              | ( 1970.3 ) | 0.268                     |
| 650               | ( 710.3 )  | 0.268                     | 1400              | ( 2060.3 ) | 0.268                     |
| 700               | ( 800.3 )  | 0.268                     | 1450              | ( 2150.3 ) | 0.268                     |
| 750               | ( 890.3 )  | 0.268                     | 1500              | ( 2240.3 ) | 0.268                     |
| 800               | ( 980.3 )  | 0.268                     | 1550              | ( 2330.3 ) | 0.268                     |
| 850               | ( 1070.3 ) | 0.268                     | 1600              | ( 2420.3 ) | 0.268                     |
| 900               | ( 1160.3 ) | 0.268                     | 1650              | ( 2510.3 ) | 0.268                     |
| 950               | ( 1250.3 ) | 0.268                     | 1700              | ( 2600.3 ) | 0.268                     |

**Application Notes:** Data for Poisson's Ratio is collected from references [4, 14, 29-33]. Property trend is calculated as a function of Young's modulus and shear modulus as seen in calculated equations below.

## Calculated:

$$\nu(T,P) = E(T,P)/(2 \cdot G(T,P)) - 1$$

$$\nu(T,P) = \text{Poisson's Ratio}$$

$$G(T,P) = \text{Shear Modulus [GPa]}$$

$$E(T,P) = \text{Young's Modulus [GPa]}$$

$$P = \text{Fractional Porosity}$$

$$T = \text{Temperature [K]}$$

## Constants:

Temperature Range [K]:  $273 \leq T \leq 1700$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical                    |                                                                 | Thermal                                     |                                           |                                                       |                                       | Elastic                            |                                  |                                 |
|---------------------|--------------------------------------------|-----------------------------------------------------------------|---------------------------------------------|-------------------------------------------|-------------------------------------------------------|---------------------------------------|------------------------------------|----------------------------------|---------------------------------|
|                     | Density<br>( $\rho$ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( $\rho_e$ )<br>$\mu\Omega\cdot m$ | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( $dL/L_0$ )<br>% | Mean CTE<br>( $\bar{\alpha}$ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( $C_p$ )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( $\nu$ ) |
| 100                 | 14379                                      | 1.147                                                           | 9.30                                        | -0.115                                    | 5.98                                                  | 0.104                                 | -                                  | -                                | -                               |
| 200                 | 14357                                      | 1.400                                                           | 12.14                                       | -0.064                                    | 6.76                                                  | 0.161                                 | -                                  | -                                | -                               |
| 300                 | 14329                                      | 1.483                                                           | 14.19                                       | 0.003                                     | 7.26                                                  | 0.190                                 | 263.52                             | 103.99                           | 0.268                           |
| 400                 | 14295                                      | 1.536                                                           | 15.85                                       | 0.081                                     | 7.62                                                  | 0.204                                 | 260.50                             | 102.79                           | 0.268                           |
| 500                 | 14260                                      | 1.577                                                           | 17.27                                       | 0.164                                     | 7.91                                                  | 0.213                                 | 257.47                             | 101.60                           | 0.268                           |
| 600                 | 14223                                      | 1.611                                                           | 18.53                                       | 0.250                                     | 8.15                                                  | 0.219                                 | 254.44                             | 100.41                           | 0.268                           |
| 700                 | 14185                                      | 1.641                                                           | 19.66                                       | 0.339                                     | 8.36                                                  | 0.224                                 | 251.42                             | 99.21                            | 0.268                           |
| 800                 | 14146                                      | 1.668                                                           | 20.70                                       | 0.432                                     | 8.55                                                  | 0.229                                 | 248.39                             | 98.02                            | 0.268                           |
| 900                 | 14106                                      | 1.693                                                           | 21.65                                       | 0.527                                     | 8.72                                                  | 0.234                                 | 245.37                             | 96.83                            | 0.268                           |
| 1000                | 14064                                      | 1.718                                                           | 22.55                                       | 0.626                                     | 8.89                                                  | 0.238                                 | 242.34                             | 95.63                            | 0.268                           |
| 1100                | 14022                                      | 1.742                                                           | 23.39                                       | 0.728                                     | 9.04                                                  | 0.243                                 | 239.32                             | 94.44                            | 0.268                           |
| 1200                | 13978                                      | 1.768                                                           | 24.19                                       | 0.832                                     | 9.19                                                  | 0.249                                 | 236.29                             | 93.24                            | 0.268                           |
| 1300                | 13934                                      | 1.796                                                           | 24.94                                       | 0.940                                     | 9.33                                                  | 0.254                                 | 233.27                             | 92.05                            | 0.268                           |
| 1400                | 13888                                      | 1.827                                                           | 25.66                                       | 1.049                                     | 9.47                                                  | 0.260                                 | 230.24                             | 90.86                            | 0.268                           |
| 1500                | 13842                                      | 1.862                                                           | 26.35                                       | 1.161                                     | 9.61                                                  | 0.266                                 | 227.21                             | 89.66                            | 0.268                           |
| 1600                | 13795                                      | 1.902                                                           | 27.02                                       | 1.276                                     | 9.74                                                  | 0.273                                 | 224.19                             | 88.47                            | 0.268                           |
| 1700                | 13748                                      | -                                                               | 27.65                                       | 1.393                                     | 9.87                                                  | 0.279                                 | 221.16                             | 87.27                            | 0.268                           |
| 1800                | 13699                                      | -                                                               | 28.27                                       | 1.511                                     | 10.00                                                 | 0.287                                 | -                                  | -                                | -                               |
| 1900                | 13651                                      | -                                                               | 28.86                                       | 1.632                                     | 10.13                                                 | 0.294                                 | -                                  | -                                | -                               |
| 2000                | 13601                                      | -                                                               | 29.44                                       | 1.755                                     | 10.25                                                 | 0.302                                 | -                                  | -                                | -                               |
| 2100                | 13552                                      | -                                                               | 29.58                                       | 1.879                                     | 10.38                                                 | 0.310                                 | -                                  | -                                | -                               |
| 2200                | 13502                                      | -                                                               | 29.58                                       | 2.005                                     | 10.50                                                 | 0.318                                 | -                                  | -                                | -                               |
| 2300                | 13451                                      | -                                                               | 29.58                                       | 2.132                                     | 10.62                                                 | 0.327                                 | -                                  | -                                | -                               |
| 2400                | 13401                                      | -                                                               | 29.58                                       | 2.260                                     | 10.74                                                 | 0.336                                 | -                                  | -                                | -                               |
| 2600                | 13299                                      | -                                                               | -                                           | 2.521                                     | 10.98                                                 | 0.355                                 | -                                  | -                                | -                               |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

**Revision 2: 04-26-2023**

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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

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6.1.1 Uranium Nitride (UN)

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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.1 Uranium Nitride (UN)

**Revision 2: 04-26-2023**

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## **6 Nuclear Materials**

### **6.1 Nuclear Fuel Materials**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 2.1: 08-25-2023

General

## Room Temperature Properties

|                                           |           |
|-------------------------------------------|-----------|
| Molar Mass, [g/mol]                       | 250.04    |
| Theoretical Density, [kg/m <sup>3</sup> ] | 13,630    |
| Melting Point, [K]                        | 2773±30   |
| Boiling Point, [K]                        | 4691-4866 |
| Specific Heat, [J/(g-K)]                  | 0.202     |
| Heat of Fusion, [kJ/kg]                   | 195.6     |
| Heat of Vapourization, [kJ/kg]            | 2120      |
| Thermal Conductivity, [W/(m-K)]           | 20.8      |
| Linear expansion coefficient, [μm/(m-K)]  | 10.1      |
| Electrical resistivity, [μΩ-m]            | 0.727     |
| Young's Modulus, [GPa]                    | 224.9     |
| Shear Modulus, [GPa]                      | 87.3      |
| Poisson's Ratio, [-]                      | 0.288     |

## Uranium – Carbon Phase Diagram

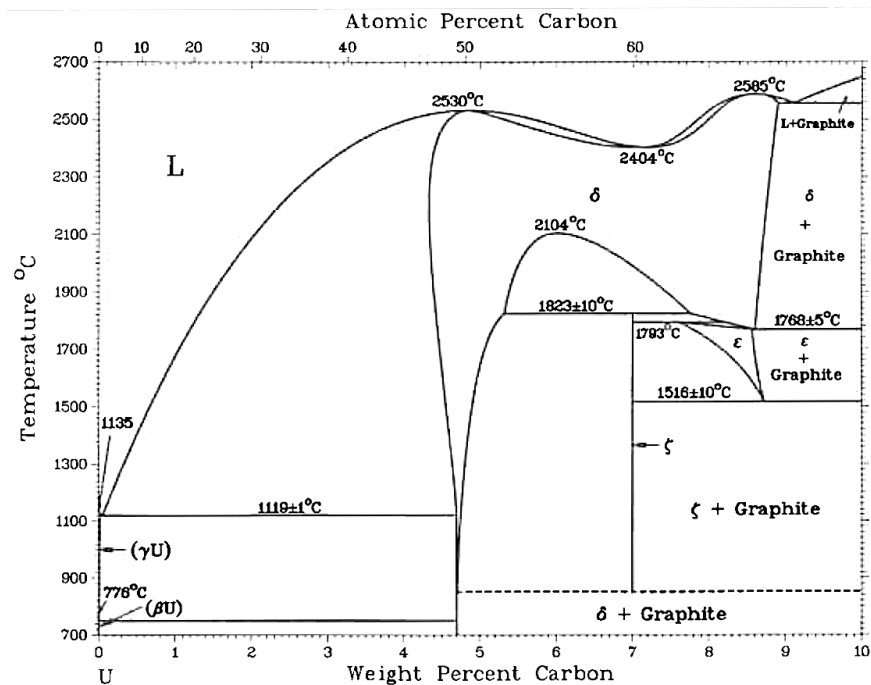


Figure 6.1.2-1: Uranium – Carbon Phase Diagram [1].  
Stoichiometric UC = 4.80 wt% Carbon.



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Density with Temperature

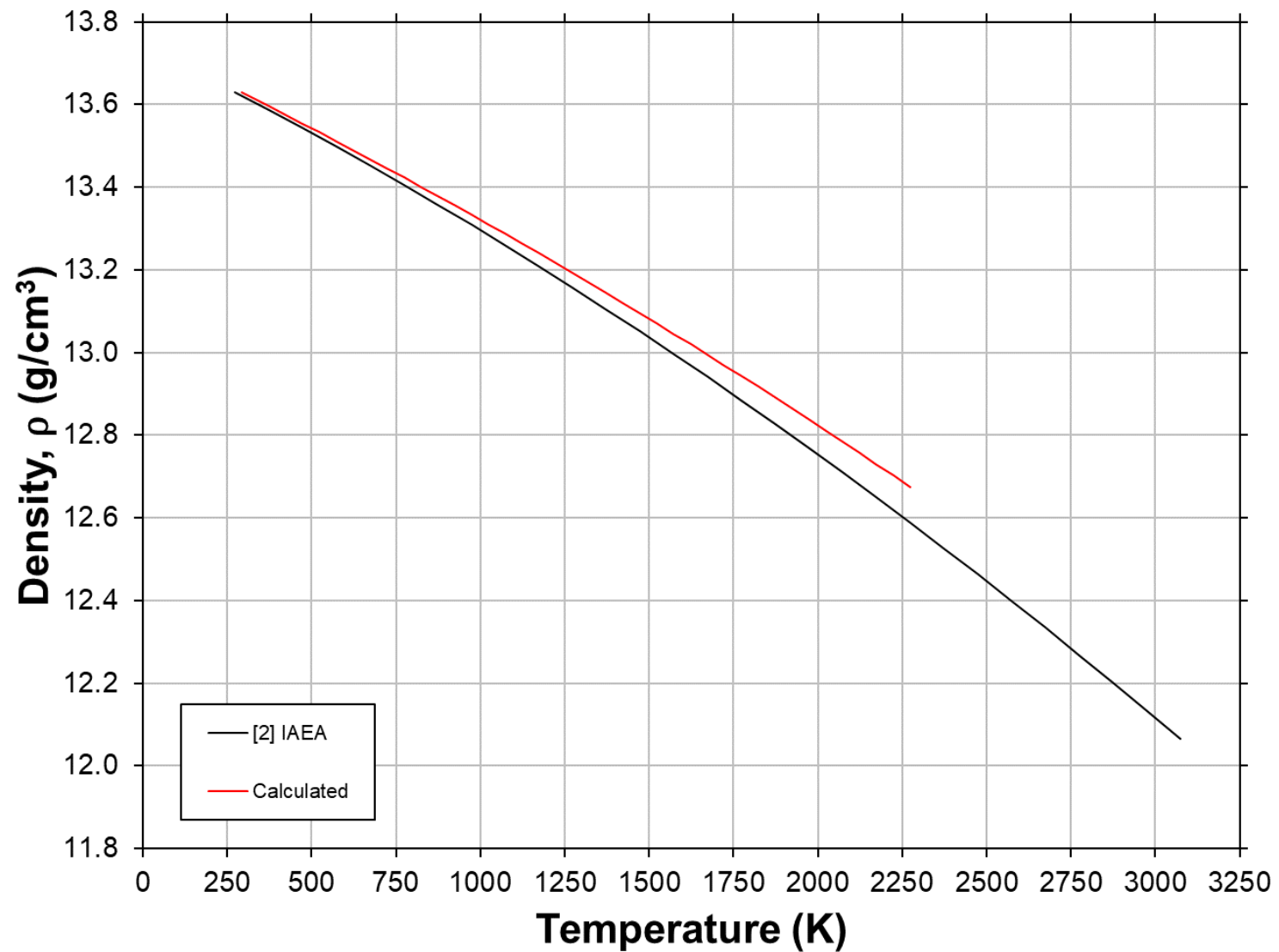


Figure 6.1.2-2: Density versus Temperature for UC. Calculated from fitted trend of the Thermal Expansion data with comparison to trend from the IAEA (2008).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Density with Temperature

## 100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 293               | ( 67.7 )   | 13630             | ( 850.9 )              | 1050              | ( 1430.3 ) | 13298             | ( 830.2 )              |
| 350               | ( 170.3 )  | 13606             | ( 849.4 )              | 1100              | ( 1520.3 ) | 13275             | ( 828.7 )              |
| 400               | ( 260.3 )  | 13585             | ( 848.1 )              | 1150              | ( 1610.3 ) | 13251             | ( 827.3 )              |
| 450               | ( 350.3 )  | 13564             | ( 846.8 )              | 1200              | ( 1700.3 ) | 13228             | ( 825.8 )              |
| 500               | ( 440.3 )  | 13543             | ( 845.5 )              | 1300              | ( 1880.3 ) | 13180             | ( 822.8 )              |
| 550               | ( 530.3 )  | 13522             | ( 844.1 )              | 1400              | ( 2060.3 ) | 13131             | ( 819.8 )              |
| 600               | ( 620.3 )  | 13500             | ( 842.8 )              | 1500              | ( 2240.3 ) | 13082             | ( 816.7 )              |
| 650               | ( 710.3 )  | 13478             | ( 841.4 )              | 1600              | ( 2420.3 ) | 13032             | ( 813.6 )              |
| 700               | ( 800.3 )  | 13456             | ( 840.1 )              | 1700              | ( 2600.3 ) | 12981             | ( 810.4 )              |
| 750               | ( 890.3 )  | 13434             | ( 838.7 )              | 1800              | ( 2780.3 ) | 12929             | ( 807.2 )              |
| 800               | ( 980.3 )  | 13412             | ( 837.3 )              | 1900              | ( 2960.3 ) | 12877             | ( 803.9 )              |
| 850               | ( 1070.3 ) | 13390             | ( 835.9 )              | 2000              | ( 3140.3 ) | 12824             | ( 800.6 )              |
| 900               | ( 1160.3 ) | 13367             | ( 834.5 )              | 2100              | ( 3320.3 ) | 12770             | ( 797.2 )              |
| 950               | ( 1250.3 ) | 13344             | ( 833.1 )              | 2200              | ( 3500.3 ) | 12715             | ( 793.8 )              |
| 1000              | ( 1340.3 ) | 13321             | ( 831.6 )              | 2273              | ( 3631.7 ) | 12675             | ( 791.3 )              |

**Application Notes:** Density is calculated here as a function of thermal expansion as seen in the equation below to approximate property trend as a function of temperature. This trend is compared against trend from reference [2].

### Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 13,630 \text{ [kg/m}^3\text{]}$$

$P$  = Fractional Porosity

$T$  = Temperature [K]

**Temperature Range:**  $293 \leq T \leq 2273$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

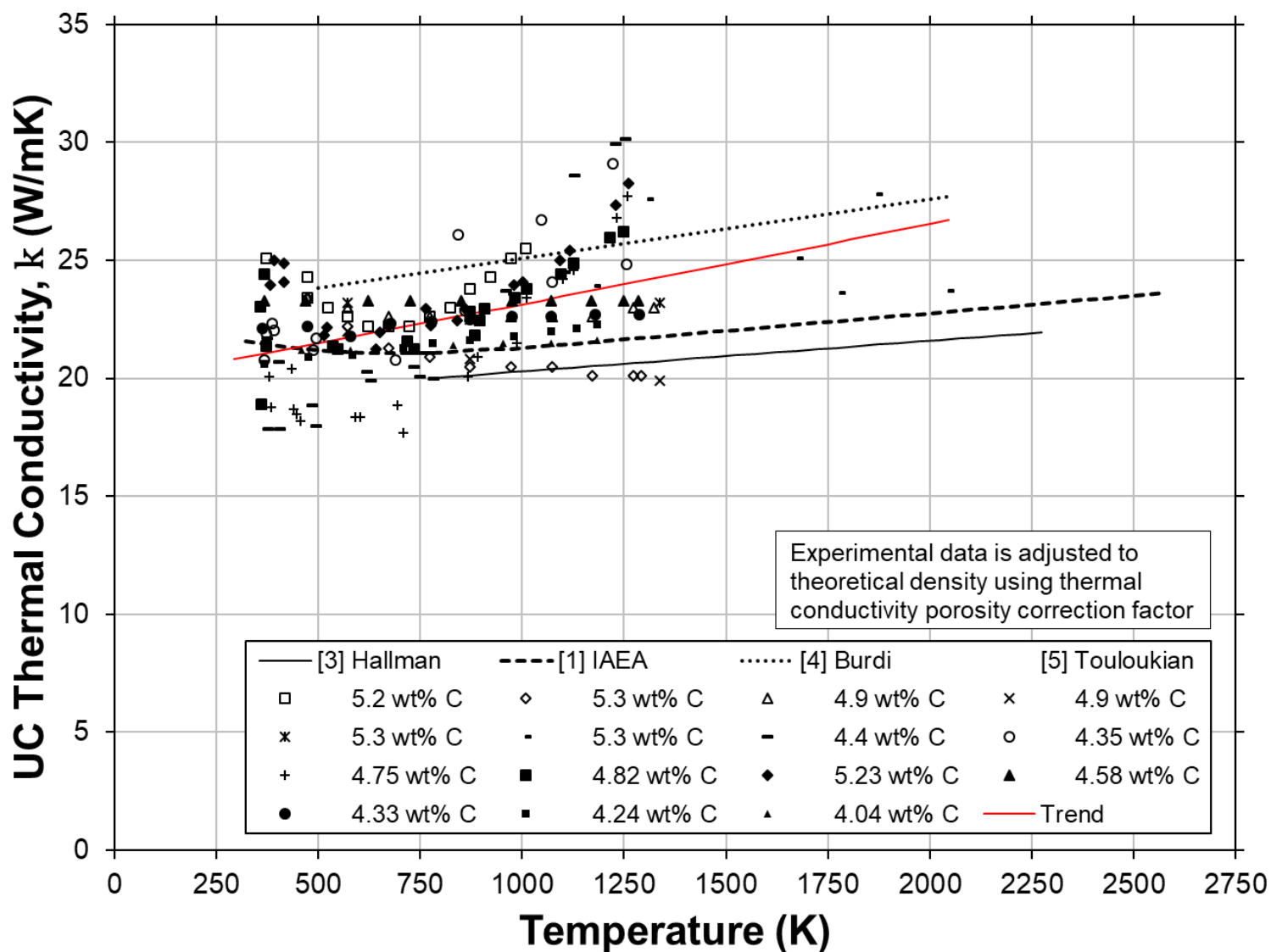


Figure 6.1.2-3: Thermal Conductivity versus Temperature of UC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 293               | ( 67.7 )   | 20.81                      | ( 144.40 )                          | 1050              | ( 1430.3 ) | 23.31                      | ( 161.70 )                          |
| 350               | ( 170.3 )  | 21.00                      | ( 145.68 )                          | 1100              | ( 1520.3 ) | 23.47                      | ( 162.86 )                          |
| 400               | ( 260.3 )  | 21.16                      | ( 146.81 )                          | 1150              | ( 1610.3 ) | 23.64                      | ( 164.03 )                          |
| 450               | ( 350.3 )  | 21.32                      | ( 147.94 )                          | 1200              | ( 1700.3 ) | 23.81                      | ( 165.20 )                          |
| 500               | ( 440.3 )  | 21.49                      | ( 149.07 )                          | 1250              | ( 1790.3 ) | 23.98                      | ( 166.37 )                          |
| 550               | ( 530.3 )  | 21.65                      | ( 150.21 )                          | 1300              | ( 1880.3 ) | 24.15                      | ( 167.55 )                          |
| 600               | ( 620.3 )  | 21.81                      | ( 151.34 )                          | 1350              | ( 1970.3 ) | 24.32                      | ( 168.72 )                          |
| 650               | ( 710.3 )  | 21.98                      | ( 152.48 )                          | 1400              | ( 2060.3 ) | 24.49                      | ( 169.90 )                          |
| 700               | ( 800.3 )  | 22.14                      | ( 153.63 )                          | 1500              | ( 2240.3 ) | 24.83                      | ( 172.27 )                          |
| 750               | ( 890.3 )  | 22.31                      | ( 154.77 )                          | 1600              | ( 2420.3 ) | 25.17                      | ( 174.64 )                          |
| 800               | ( 980.3 )  | 22.47                      | ( 155.92 )                          | 1700              | ( 2600.3 ) | 25.52                      | ( 177.03 )                          |
| 850               | ( 1070.3 ) | 22.64                      | ( 157.07 )                          | 1800              | ( 2780.3 ) | 25.86                      | ( 179.43 )                          |
| 900               | ( 1160.3 ) | 22.81                      | ( 158.22 )                          | 1900              | ( 2960.3 ) | 26.21                      | ( 181.84 )                          |
| 950               | ( 1250.3 ) | 22.97                      | ( 159.38 )                          | 2000              | ( 3140.3 ) | 26.56                      | ( 184.26 )                          |
| 1000              | ( 1340.3 ) | 23.14                      | ( 160.54 )                          | 2045              | ( 3221.3 ) | 26.71                      | ( 185.35 )                          |

**Application Notes:** Data for thermal conductivity is collected from references [2-5] and fitted with the equation below to approximate the property trend with respect to temperature and porosity.

## Fit Equation:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$$k_p = k \cdot (1 - P)/(1 + P)$$

$P$  = Fractional Porosity ( $0 \leq P \leq 0.2$ )

$T$  = Temperature [K]

## Constants:

T Range [K]: 293 ≤ T ≤ 2045

A0 = 19.87

A1 = 3.194

A2 = 0.07475



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

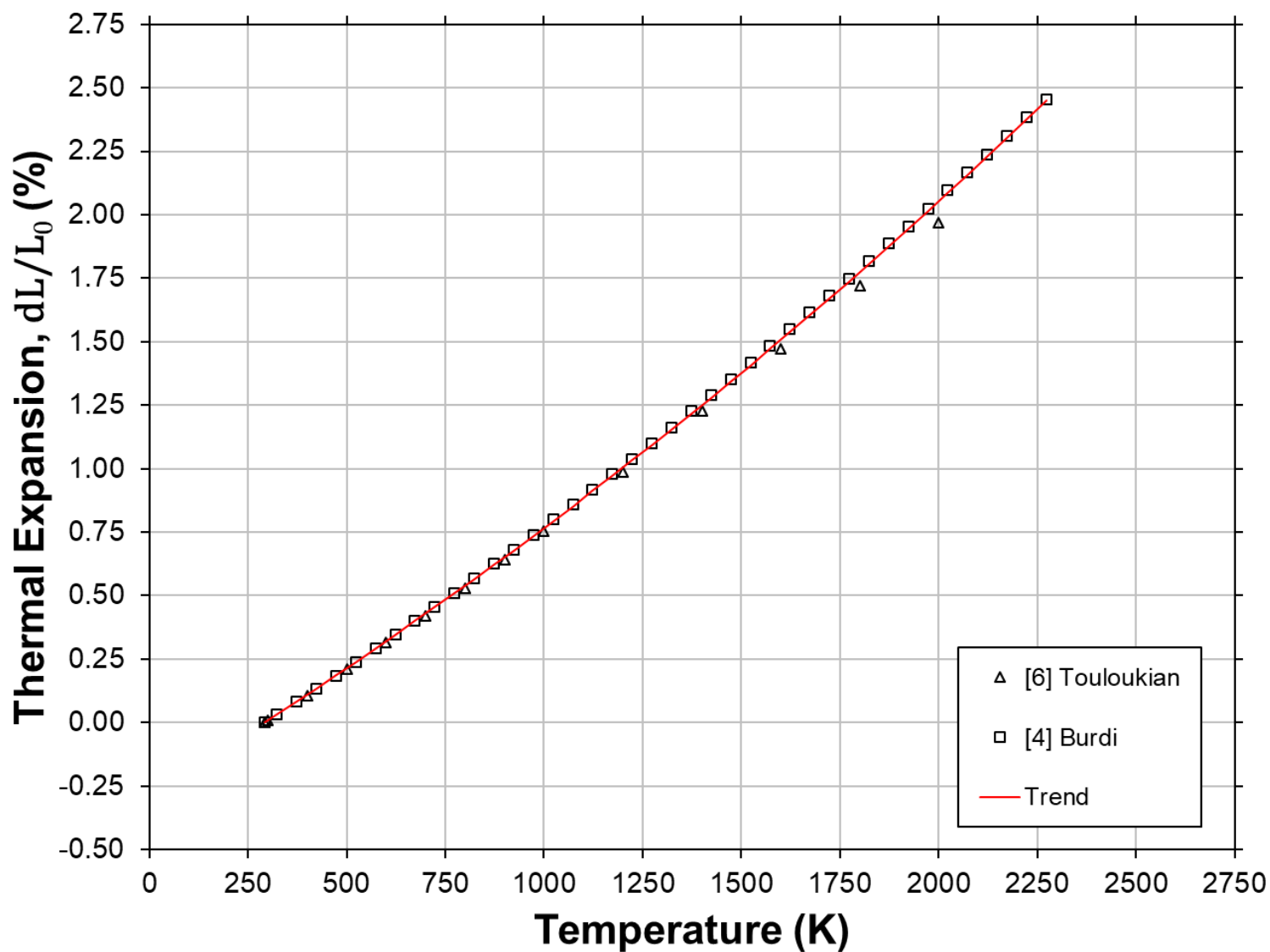


Figure 6.1.2-4: Thermal Expansion versus Temperature of UC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 293               | ( 67.7 )   | 0.000                                   | 1050              | ( 1430.3 ) | 0.825                                   |
| 350               | ( 170.3 )  | 0.058                                   | 1100              | ( 1520.3 ) | 0.884                                   |
| 400               | ( 260.3 )  | 0.109                                   | 1150              | ( 1610.3 ) | 0.944                                   |
| 450               | ( 350.3 )  | 0.161                                   | 1200              | ( 1700.3 ) | 1.004                                   |
| 500               | ( 440.3 )  | 0.214                                   | 1300              | ( 1880.3 ) | 1.126                                   |
| 550               | ( 530.3 )  | 0.267                                   | 1400              | ( 2060.3 ) | 1.250                                   |
| 600               | ( 620.3 )  | 0.320                                   | 1500              | ( 2240.3 ) | 1.377                                   |
| 650               | ( 710.3 )  | 0.374                                   | 1600              | ( 2420.3 ) | 1.507                                   |
| 700               | ( 800.3 )  | 0.429                                   | 1700              | ( 2600.3 ) | 1.639                                   |
| 750               | ( 890.3 )  | 0.484                                   | 1800              | ( 2780.3 ) | 1.774                                   |
| 800               | ( 980.3 )  | 0.539                                   | 1900              | ( 2960.3 ) | 1.912                                   |
| 850               | ( 1070.3 ) | 0.595                                   | 2000              | ( 3140.3 ) | 2.052                                   |
| 900               | ( 1160.3 ) | 0.652                                   | 2100              | ( 3320.3 ) | 2.196                                   |
| 950               | ( 1250.3 ) | 0.709                                   | 2200              | ( 3500.3 ) | 2.342                                   |
| 1000              | ( 1340.3 ) | 0.767                                   | 2273              | ( 3631.7 ) | 2.451                                   |

**Application Notes:** Data for thermal expansion is collected from references [4, 6] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 293 < T \leq 2273$$

$$A0 = -0.2887$$

$$A1 = 0.9588$$

$$A2 = 0.08756$$

$$A3 = 0.009167$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

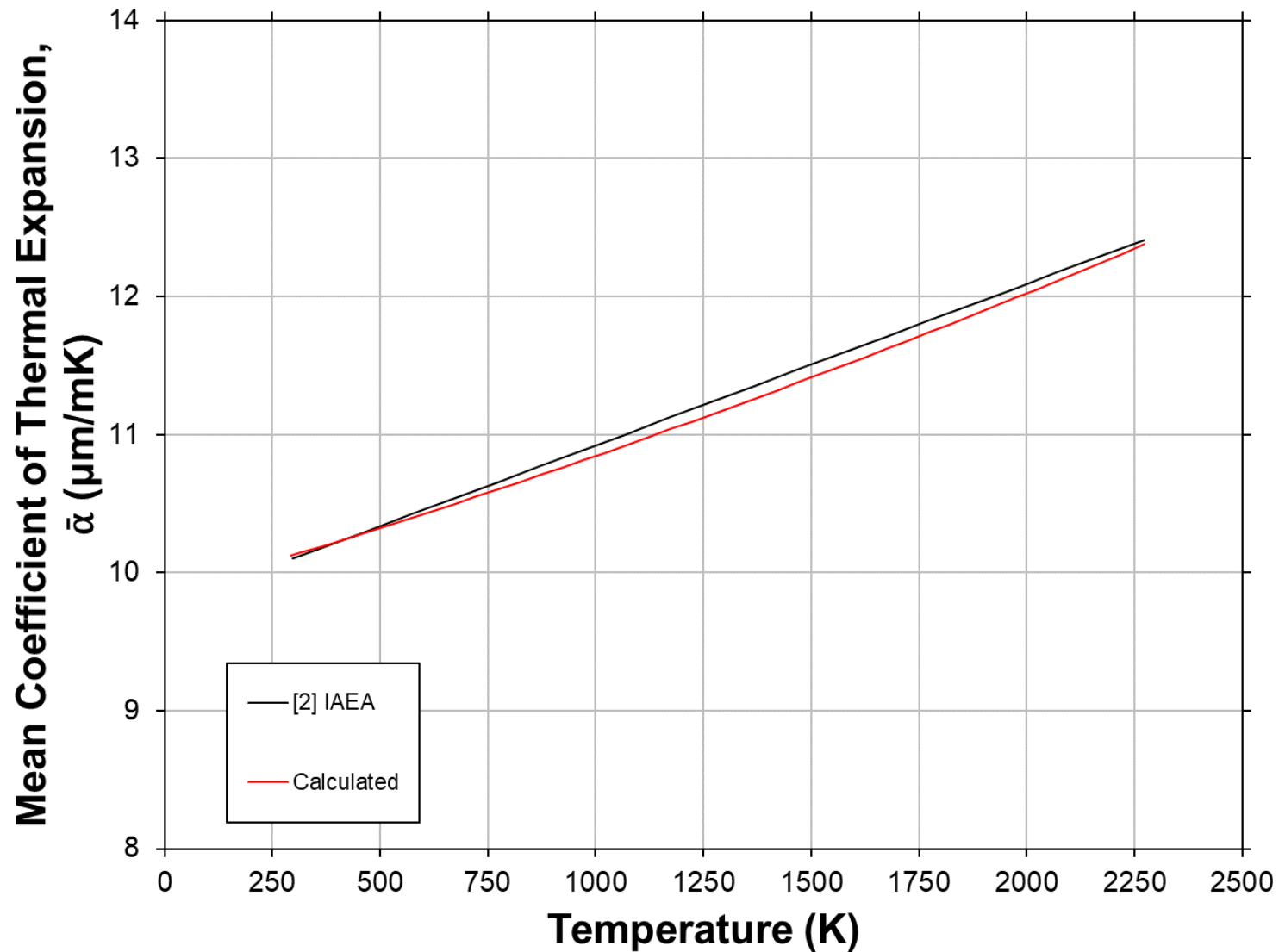


Figure 6.1.2-5: Mean Coefficient of Thermal Expansion versus Temperature of UC. Calculated from fitted trend of the Thermal Expansion data with comparison to IAEA (2008).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 293               | ( 67.7 )   | 10.121                                | ( 5.623 )                                           | 1050              | ( 1430.3 ) | 10.900                                | ( 6.056 )                                           |
| 350               | ( 170.3 )  | 10.176                                | ( 5.653 )                                           | 1100              | ( 1520.3 ) | 10.955                                | ( 6.086 )                                           |
| 400               | ( 260.3 )  | 10.225                                | ( 5.681 )                                           | 1150              | ( 1610.3 ) | 11.011                                | ( 6.117 )                                           |
| 450               | ( 350.3 )  | 10.274                                | ( 5.708 )                                           | 1200              | ( 1700.3 ) | 11.067                                | ( 6.148 )                                           |
| 500               | ( 440.3 )  | 10.324                                | ( 5.736 )                                           | 1300              | ( 1880.3 ) | 11.180                                | ( 6.211 )                                           |
| 550               | ( 530.3 )  | 10.374                                | ( 5.763 )                                           | 1400              | ( 2060.3 ) | 11.295                                | ( 6.275 )                                           |
| 600               | ( 620.3 )  | 10.425                                | ( 5.791 )                                           | 1500              | ( 2240.3 ) | 11.412                                | ( 6.340 )                                           |
| 650               | ( 710.3 )  | 10.476                                | ( 5.820 )                                           | 1600              | ( 2420.3 ) | 11.531                                | ( 6.406 )                                           |
| 700               | ( 800.3 )  | 10.527                                | ( 5.848 )                                           | 1700              | ( 2600.3 ) | 11.651                                | ( 6.473 )                                           |
| 750               | ( 890.3 )  | 10.579                                | ( 5.877 )                                           | 1800              | ( 2780.3 ) | 11.774                                | ( 6.541 )                                           |
| 800               | ( 980.3 )  | 10.632                                | ( 5.906 )                                           | 1900              | ( 2960.3 ) | 11.898                                | ( 6.610 )                                           |
| 850               | ( 1070.3 ) | 10.684                                | ( 5.936 )                                           | 2000              | ( 3140.3 ) | 12.023                                | ( 6.680 )                                           |
| 900               | ( 1160.3 ) | 10.738                                | ( 5.965 )                                           | 2100              | ( 3320.3 ) | 12.151                                | ( 6.751 )                                           |
| 950               | ( 1250.3 ) | 10.791                                | ( 5.995 )                                           | 2200              | ( 3500.3 ) | 12.280                                | ( 6.822 )                                           |
| 1000              | ( 1340.3 ) | 10.846                                | ( 6.025 )                                           | 2273              | ( 3631.7 ) | 12.376                                | ( 6.876 )                                           |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated as a function thermal expansion data, then fitted with the equation below to approximate property trend with respect to temperature. Fitted trend is compared against trend from reference [2].

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]: 293 ≤ T ≤ 2273

A0 = 9.847

A1 = 0.909

A2 = 0.0896



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Specific Heat with Temperature

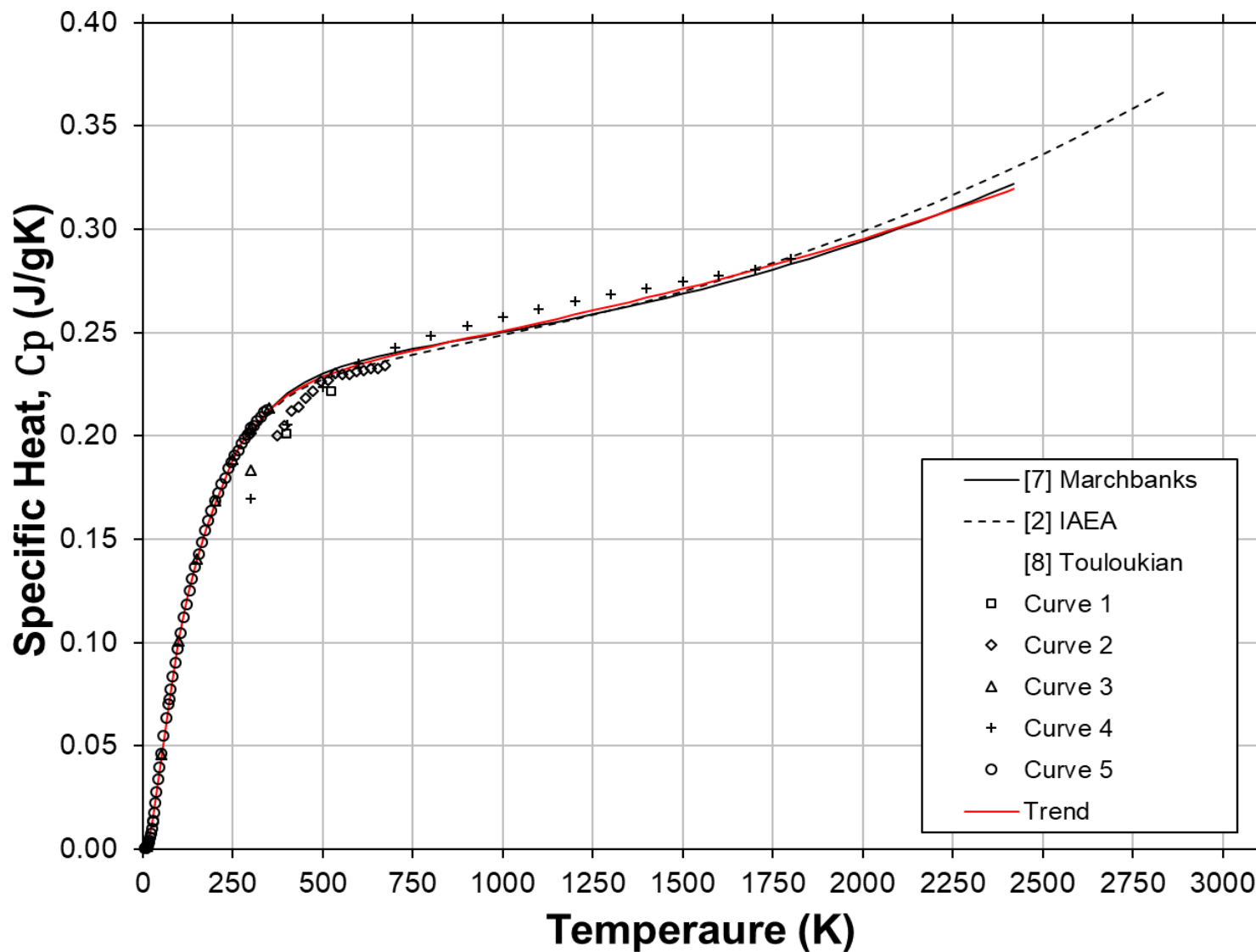


Figure 6.1.2-6: Specific Heat versus Temperature of UC. Displaying fitted trend and data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 25                | ( -414.7 ) | 0.012                            | ( 0.003 )       | 1200              | ( 1700.3 ) | 0.259                            | ( 0.062 )       |
| 50                | ( -369.7 ) | 0.043                            | ( 0.010 )       | 1300              | ( 1880.3 ) | 0.263                            | ( 0.063 )       |
| 100               | ( -279.7 ) | 0.101                            | ( 0.024 )       | 1400              | ( 2060.3 ) | 0.267                            | ( 0.064 )       |
| 200               | ( -99.7 )  | 0.167                            | ( 0.040 )       | 1500              | ( 2240.3 ) | 0.271                            | ( 0.065 )       |
| 300               | ( 80.3 )   | 0.203                            | ( 0.048 )       | 1600              | ( 2420.3 ) | 0.276                            | ( 0.066 )       |
| 400               | ( 260.3 )  | 0.220                            | ( 0.053 )       | 1700              | ( 2600.3 ) | 0.280                            | ( 0.067 )       |
| 500               | ( 440.3 )  | 0.229                            | ( 0.055 )       | 1800              | ( 2780.3 ) | 0.285                            | ( 0.068 )       |
| 600               | ( 620.3 )  | 0.235                            | ( 0.056 )       | 1900              | ( 2960.3 ) | 0.290                            | ( 0.069 )       |
| 700               | ( 800.3 )  | 0.239                            | ( 0.057 )       | 2000              | ( 3140.3 ) | 0.295                            | ( 0.071 )       |
| 800               | ( 980.3 )  | 0.243                            | ( 0.058 )       | 2100              | ( 3320.3 ) | 0.301                            | ( 0.072 )       |
| 900               | ( 1160.3 ) | 0.247                            | ( 0.059 )       | 2200              | ( 3500.3 ) | 0.306                            | ( 0.073 )       |
| 1000              | ( 1340.3 ) | 0.251                            | ( 0.060 )       | 2300              | ( 3680.3 ) | 0.312                            | ( 0.075 )       |
| 1100              | ( 1520.3 ) | 0.255                            | ( 0.061 )       | 2400              | ( 3860.3 ) | 0.318                            | ( 0.076 )       |

**Application Notes:** Data for specific heat is collected from references [2, 7, 8] and fitted with the equations below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $10 \leq T < 293$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $293 \leq T \leq 2418$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_{_2} / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

T. Range [K]:  $10 \leq T < 293$

N = 2.352

A0 = 90.53

A1 = 2.954

A2 = 267.2

$293 \leq T \leq 2418$

B0 = 2.331E-01

B1 = 1.053E-02

B2 = 1.051E-02

B\_2 = -3.091E-03

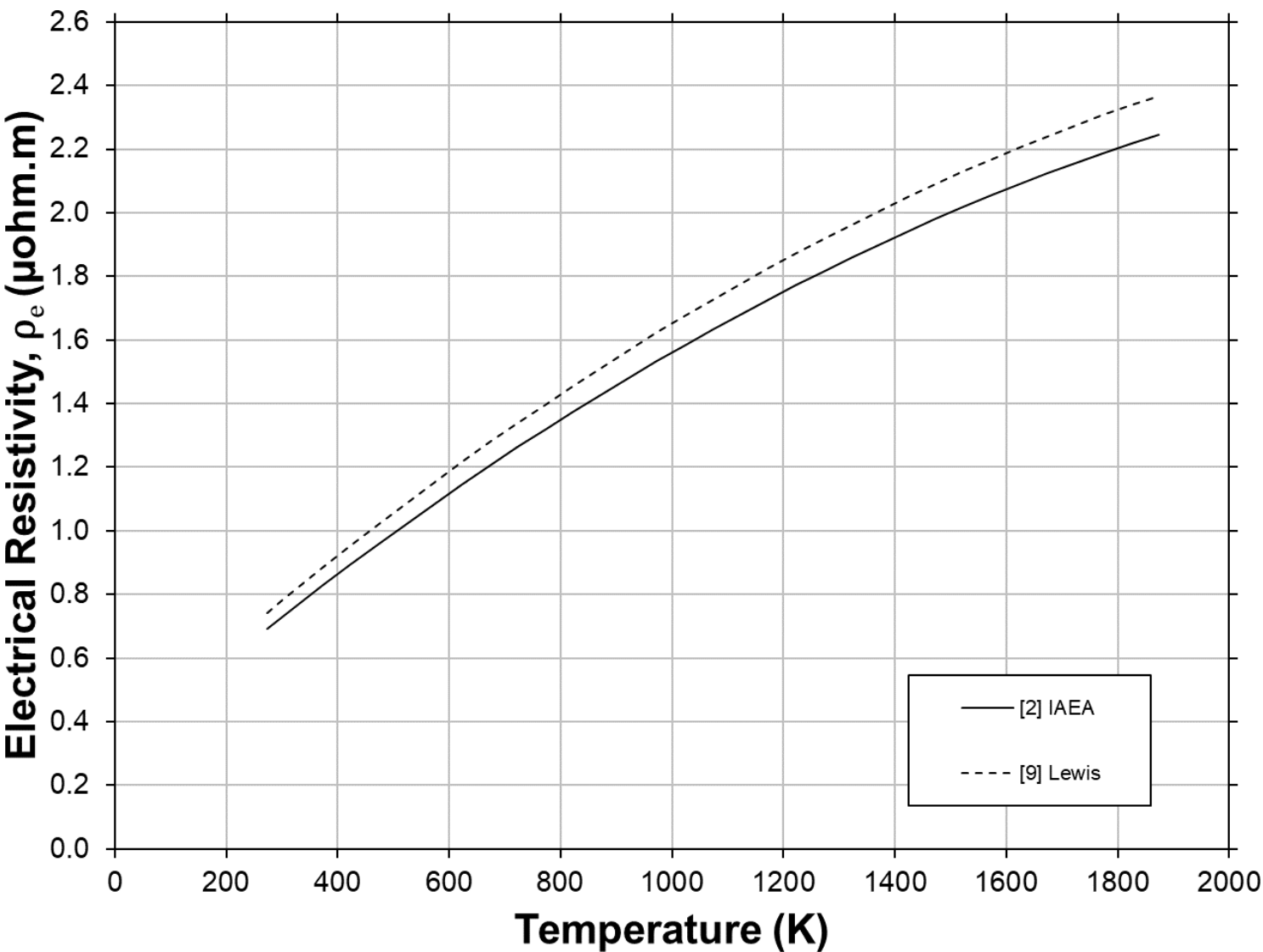


Figure 6.1.2-7: Electrical Resistivity versus Temperature of UC from IAEA (2008) and Lewis (1975).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 2: 04-26-2023

Electrical Resistivity with Temperature

100% Theoretical Density High Purity

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                     | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                     |
|-------------------|------------|-------------------------------------|---------------------|-------------------|------------|-------------------------------------|---------------------|
| K                 | ( °F )     | $\mu\Omega$ -m                      | ( $\mu\Omega$ -in ) | K                 | ( °F )     | $\mu\Omega$ -m                      | ( $\mu\Omega$ -in ) |
| 273               | ( 31.7 )   | 0.743                               | ( 29.26 )           | 1000              | ( 1340.3 ) | 1.651                               | ( 65.00 )           |
| 300               | ( 80.3 )   | 0.782                               | ( 30.78 )           | 1050              | ( 1430.3 ) | 1.703                               | ( 67.05 )           |
| 350               | ( 170.3 )  | 0.853                               | ( 33.57 )           | 1100              | ( 1520.3 ) | 1.754                               | ( 69.04 )           |
| 400               | ( 260.3 )  | 0.922                               | ( 36.30 )           | 1150              | ( 1610.3 ) | 1.803                               | ( 70.99 )           |
| 450               | ( 350.3 )  | 0.990                               | ( 38.99 )           | 1200              | ( 1700.3 ) | 1.851                               | ( 72.88 )           |
| 500               | ( 440.3 )  | 1.057                               | ( 41.61 )           | 1250              | ( 1790.3 ) | 1.898                               | ( 74.71 )           |
| 550               | ( 530.3 )  | 1.122                               | ( 44.19 )           | 1300              | ( 1880.3 ) | 1.943                               | ( 76.50 )           |
| 600               | ( 620.3 )  | 1.187                               | ( 46.71 )           | 1350              | ( 1970.3 ) | 1.987                               | ( 78.23 )           |
| 650               | ( 710.3 )  | 1.249                               | ( 49.18 )           | 1400              | ( 2060.3 ) | 2.030                               | ( 79.91 )           |
| 700               | ( 800.3 )  | 1.311                               | ( 51.60 )           | 1450              | ( 2150.3 ) | 2.071                               | ( 81.54 )           |
| 750               | ( 890.3 )  | 1.371                               | ( 53.97 )           | 1500              | ( 2240.3 ) | 2.111                               | ( 83.11 )           |
| 800               | ( 980.3 )  | 1.429                               | ( 56.28 )           | 1600              | ( 2420.3 ) | 2.187                               | ( 86.10 )           |
| 850               | ( 1070.3 ) | 1.487                               | ( 58.54 )           | 1700              | ( 2600.3 ) | 2.257                               | ( 88.88 )           |
| 900               | ( 1160.3 ) | 1.543                               | ( 60.74 )           | 1800              | ( 2780.3 ) | 2.323                               | ( 91.44 )           |
| 950               | ( 1250.3 ) | 1.598                               | ( 62.90 )           | 1873              | ( 2911.7 ) | 2.367                               | ( 93.18 )           |

**Application Notes:** Data for electrical resistivity is collected from reference [9] and is fitted with the equation below to approximate property trend with respect to temperature for high purity UC. The trend is compared against data from reference [2].

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot T + A2 \cdot T^2$$

$$\rho_e(T) = \text{Electrical Resistivity } [\mu\Omega \cdot m]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T. \text{ Range } [K]: \quad 273 \leq T < 1873$$

$$A0 = \quad 0.329$$

$$A1 = \quad 1.59E-03$$

$$A2 = \quad -2.68E-07$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Young's Modulus with Porosity

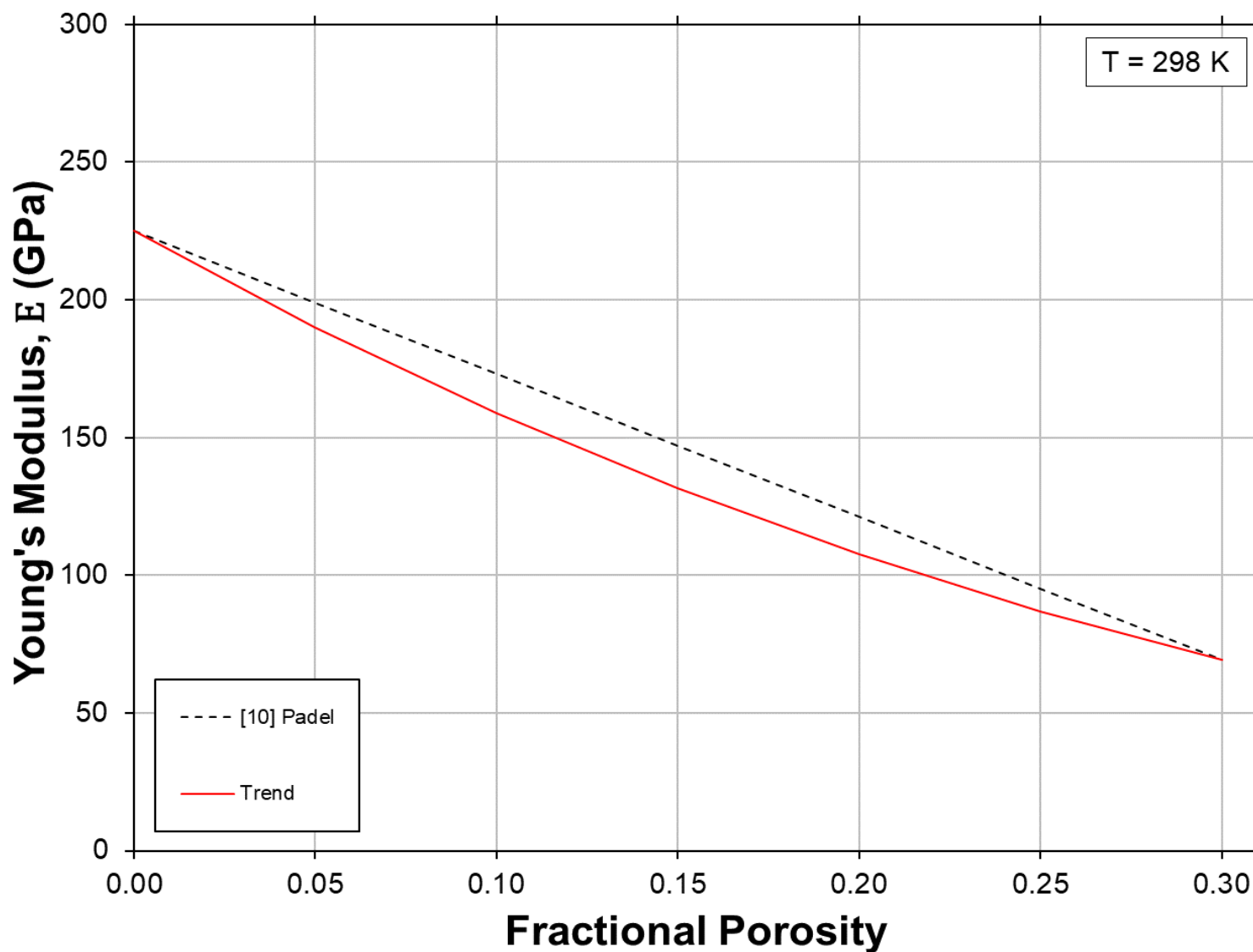


Figure 6.1.2-8: Young's Modulus versus Fractional Porosity of UC with comparison to Padel (1969).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Young's Modulus with Porosity

Room Temperature

| Porosity ( P ) | Young's Modulus ( E ) |           |
|----------------|-----------------------|-----------|
|                | GPa                   | ( Msi )   |
| 0.00           | 224.90                | ( 32.63 ) |
| 0.02           | 210.39                | ( 30.53 ) |
| 0.04           | 196.54                | ( 28.52 ) |
| 0.06           | 183.34                | ( 26.60 ) |
| 0.08           | 170.77                | ( 24.78 ) |
| 0.10           | 158.82                | ( 23.04 ) |
| 0.12           | 147.46                | ( 21.40 ) |
| 0.14           | 136.68                | ( 19.83 ) |
| 0.16           | 126.46                | ( 18.35 ) |
| 0.18           | 116.79                | ( 16.95 ) |
| 0.20           | 107.64                | ( 15.62 ) |
| 0.22           | 99.01                 | ( 14.37 ) |
| 0.24           | 90.87                 | ( 13.19 ) |
| 0.26           | 83.21                 | ( 12.07 ) |
| 0.28           | 76.02                 | ( 11.03 ) |
| 0.30           | 69.26                 | ( 10.05 ) |

**Application Notes:** Data for Young's modulus is collected from references [7, 10] and compared against the equation below to approximate property trend with respect to porosity at T=298K.

## Fit Equations:

$$E(T, P) = A \cdot (1 - P)^N [1 - A_0 \cdot (T - 298)]$$

$$E(T, P) = \text{Young's Modulus [GPa]}$$

$$T = \text{Temperature [K] for } 298 \leq T \leq 1600$$

$$P = \text{Fractional Porosity for } 0 \leq P \leq 0.3$$

## Constants:

$$A = 224.9$$

$$N = 3.302$$

$$A_0 = 1.00\text{E-}04$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Young's Modulus with Temperature

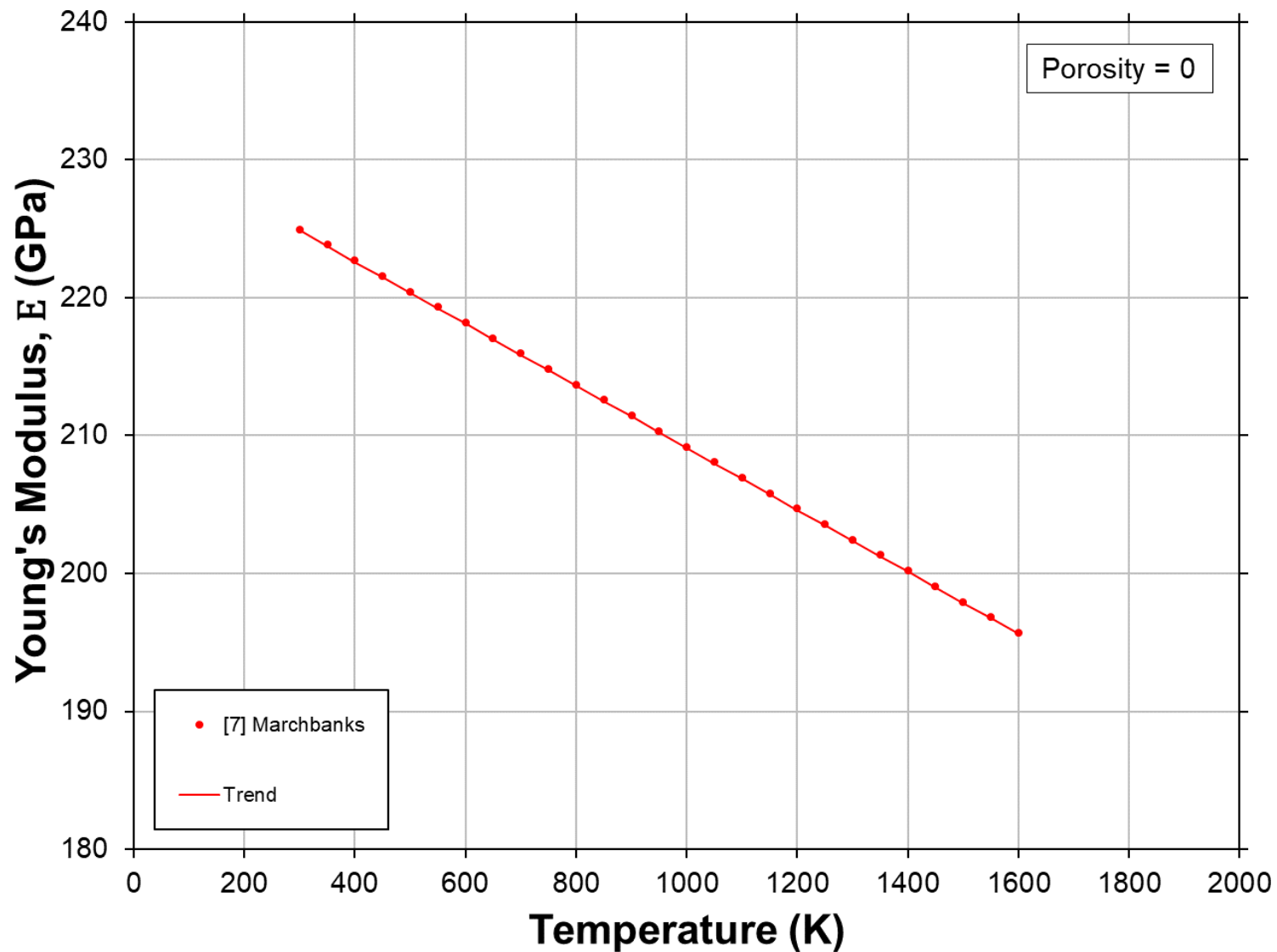


Figure 6.1.2-9: Young's Modulus versus Temperature of UC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 298               | ( 76.7 )   | 224.90                | ( 32.63 ) | 1000              | ( 1340.3 ) | 209.11                | ( 30.34 ) |
| 300               | ( 80.3 )   | 224.86                | ( 32.63 ) | 1050              | ( 1430.3 ) | 207.99                | ( 30.18 ) |
| 350               | ( 170.3 )  | 223.73                | ( 32.46 ) | 1100              | ( 1520.3 ) | 206.86                | ( 30.02 ) |
| 400               | ( 260.3 )  | 222.61                | ( 32.30 ) | 1150              | ( 1610.3 ) | 205.74                | ( 29.85 ) |
| 450               | ( 350.3 )  | 221.48                | ( 32.14 ) | 1200              | ( 1700.3 ) | 204.61                | ( 29.69 ) |
| 500               | ( 440.3 )  | 220.36                | ( 31.97 ) | 1250              | ( 1790.3 ) | 203.49                | ( 29.53 ) |
| 550               | ( 530.3 )  | 219.23                | ( 31.81 ) | 1300              | ( 1880.3 ) | 202.37                | ( 29.36 ) |
| 600               | ( 620.3 )  | 218.11                | ( 31.65 ) | 1350              | ( 1970.3 ) | 201.24                | ( 29.20 ) |
| 650               | ( 710.3 )  | 216.98                | ( 31.48 ) | 1400              | ( 2060.3 ) | 200.12                | ( 29.04 ) |
| 700               | ( 800.3 )  | 215.86                | ( 31.32 ) | 1450              | ( 2150.3 ) | 198.99                | ( 28.87 ) |
| 750               | ( 890.3 )  | 214.73                | ( 31.16 ) | 1500              | ( 2240.3 ) | 197.87                | ( 28.71 ) |
| 800               | ( 980.3 )  | 213.61                | ( 30.99 ) | 1550              | ( 2330.3 ) | 196.74                | ( 28.55 ) |
| 850               | ( 1070.3 ) | 212.49                | ( 30.83 ) | 1600              | ( 2420.3 ) | 195.62                | ( 28.38 ) |
| 900               | ( 1160.3 ) | 211.36                | ( 30.67 ) |                   |            |                       |           |
| 950               | ( 1250.3 ) | 210.24                | ( 30.51 ) |                   |            |                       |           |

**Application Notes:** Data for Young's modulus is collected from references [7, 10] and compared against the equation below to approximate property trend with respect to temperature and porosity.

## Fit Equations:

$$E(T, P) = A \cdot (1 - P)^N [1 - A0 \cdot (T - 298)]$$

$$E(T, P) = \text{Young's Modulus [GPa]}$$

$$T = \text{Temperature [K] for } 298 \leq T \leq 1600$$

$$P = \text{Fractional Porosity for } 0 \leq P \leq 0.3$$

## Constants:

$$A = 224.9$$

$$N = 3.302$$

$$A0 = 1.00\text{E-}04$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Shear Modulus with Porosity

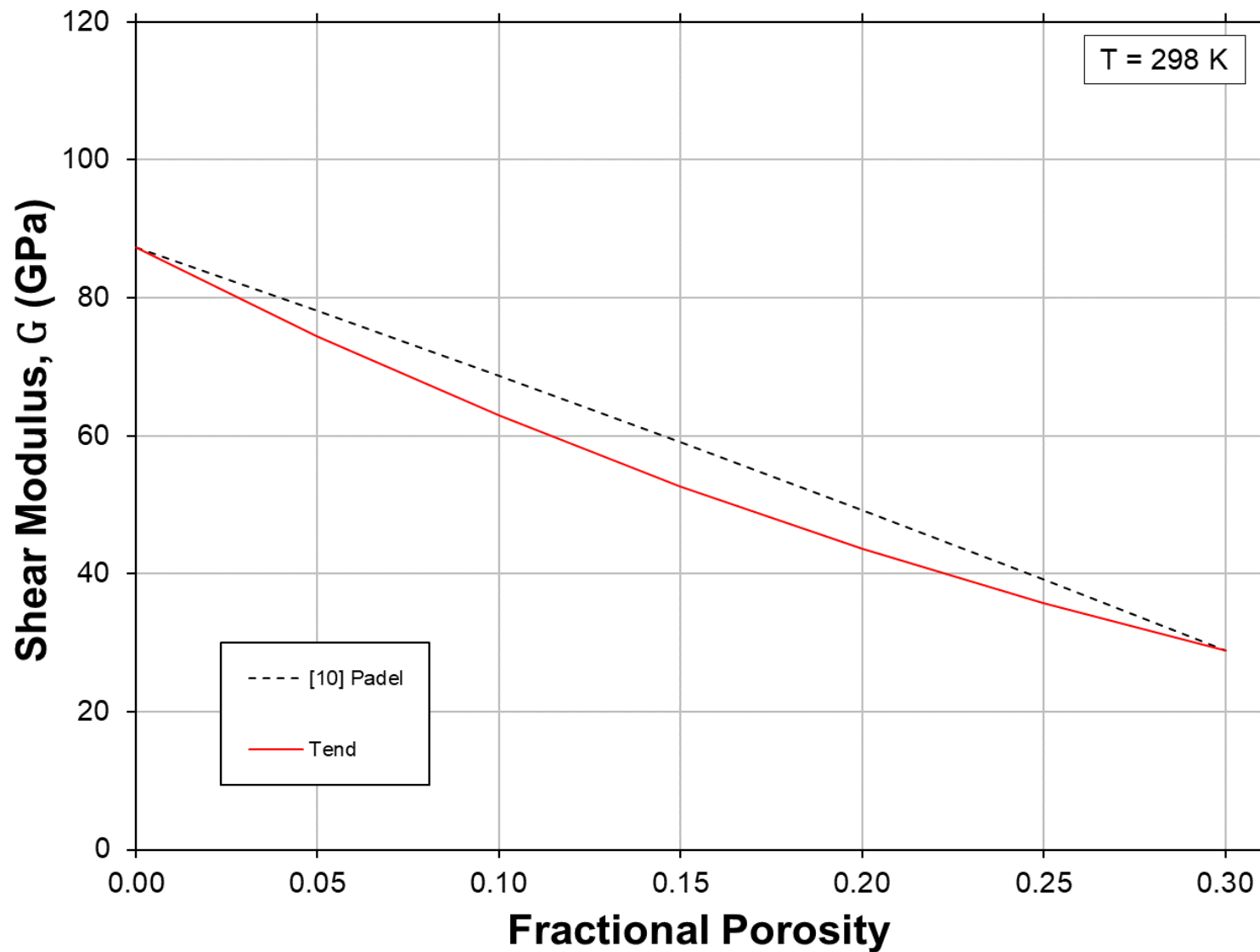


Figure 6.1.2-10: Shear Modulus versus Fractional Porosity of UC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Shear Modulus with Porosity

Room Temperature

| Porosity ( P ) | Shear Modulus ( G ) |           |
|----------------|---------------------|-----------|
|                | GPa                 | ( Msi )   |
| 0.00           | 87.30               | ( 12.67 ) |
| 0.02           | 81.99               | ( 11.90 ) |
| 0.04           | 76.89               | ( 11.16 ) |
| 0.06           | 72.02               | ( 10.45 ) |
| 0.08           | 67.36               | ( 9.77 )  |
| 0.10           | 62.92               | ( 9.13 )  |
| 0.12           | 58.67               | ( 8.51 )  |
| 0.14           | 54.62               | ( 7.93 )  |
| 0.16           | 50.77               | ( 7.37 )  |
| 0.18           | 47.10               | ( 6.83 )  |
| 0.20           | 43.62               | ( 6.33 )  |
| 0.22           | 40.32               | ( 5.85 )  |
| 0.24           | 37.19               | ( 5.40 )  |
| 0.26           | 34.23               | ( 4.97 )  |
| 0.28           | 31.44               | ( 4.56 )  |
| 0.30           | 28.80               | ( 4.18 )  |

**Application Notes:** Data for shear modulus is collected from reference [7, 10] and fitted with the equation below to approximate property trend with respect to temperature and porosity.

## Fit Equations:

$$G(T,P) = A \cdot (1 - P)^N \cdot [1 - A0 \cdot (T - 298)] / [1 - A1 \cdot (T - 298)]$$

$$G(T,P) = \text{Shear Modulus [GPa]}$$

$$T = \text{Temperature [K]}$$

$$P = \text{Fractional Porosity}$$

## Constants:

$$A = 87.3$$

$$N = 3.109$$

$$A0 = 1.00\text{E-}04$$

$$A1 = 4.40\text{E-}05$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Shear Modulus with Temperature

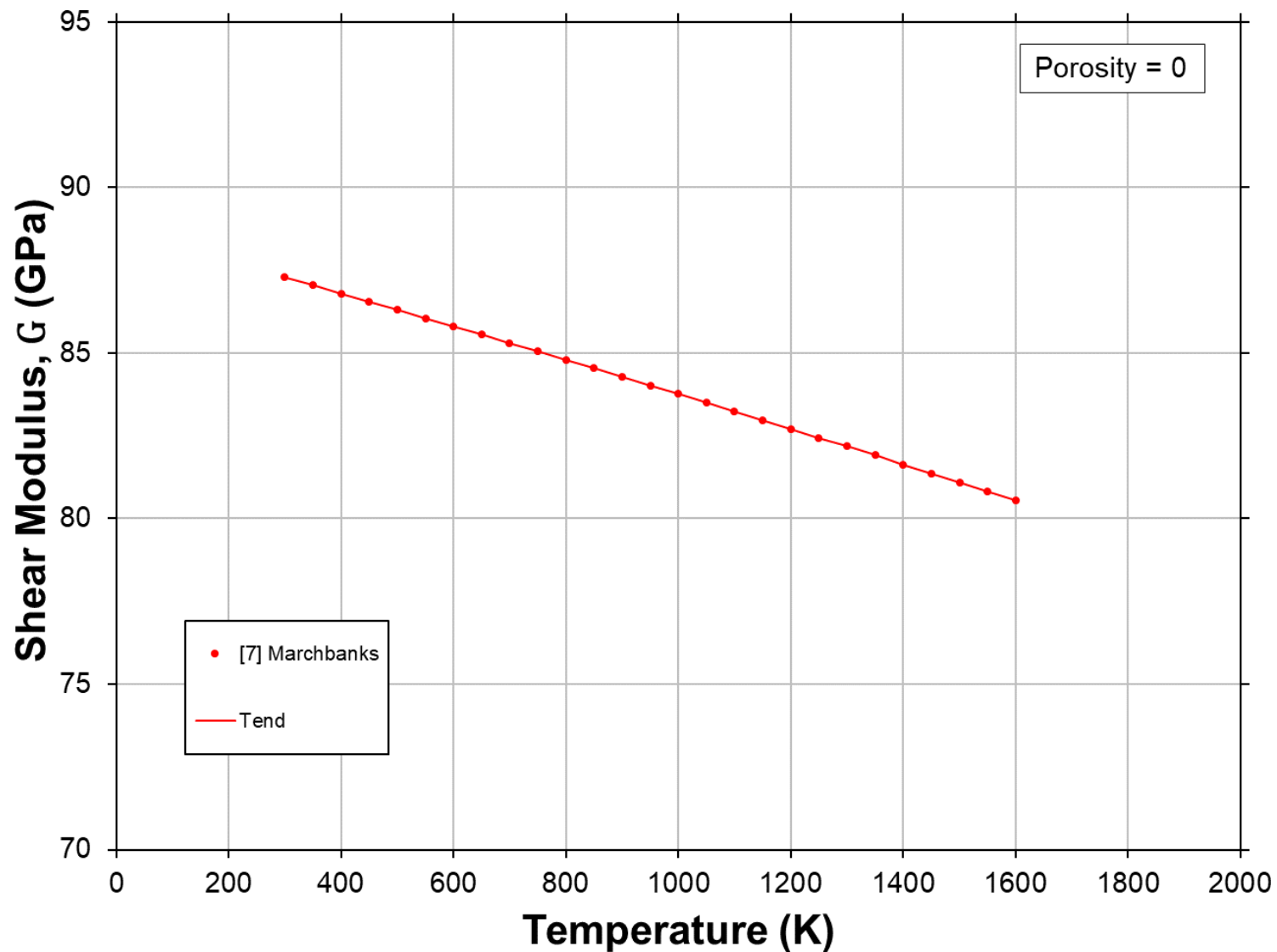


Figure 6.1.2-11: Shear Modulus versus Temperature of UC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( G ) |           | Temperature ( T ) |            | Shear Modulus ( G ) |           |
|-------------------|------------|---------------------|-----------|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   | K                 | ( °F )     | GPa                 | ( Msi )   |
| 298               | ( 76.7 )   | 87.30               | ( 12.67 ) | 1000              | ( 1340.3 ) | 83.76               | ( 12.15 ) |
| 300               | ( 80.3 )   | 87.29               | ( 12.67 ) | 1050              | ( 1430.3 ) | 83.50               | ( 12.12 ) |
| 350               | ( 170.3 )  | 87.05               | ( 12.63 ) | 1100              | ( 1520.3 ) | 83.24               | ( 12.08 ) |
| 400               | ( 260.3 )  | 86.80               | ( 12.59 ) | 1150              | ( 1610.3 ) | 82.97               | ( 12.04 ) |
| 450               | ( 350.3 )  | 86.55               | ( 12.56 ) | 1200              | ( 1700.3 ) | 82.71               | ( 12.00 ) |
| 500               | ( 440.3 )  | 86.30               | ( 12.52 ) | 1250              | ( 1790.3 ) | 82.44               | ( 11.96 ) |
| 550               | ( 530.3 )  | 86.05               | ( 12.49 ) | 1300              | ( 1880.3 ) | 82.18               | ( 11.92 ) |
| 600               | ( 620.3 )  | 85.80               | ( 12.45 ) | 1350              | ( 1970.3 ) | 81.91               | ( 11.88 ) |
| 650               | ( 710.3 )  | 85.55               | ( 12.41 ) | 1400              | ( 2060.3 ) | 81.64               | ( 11.85 ) |
| 700               | ( 800.3 )  | 85.30               | ( 12.38 ) | 1450              | ( 2150.3 ) | 81.37               | ( 11.81 ) |
| 750               | ( 890.3 )  | 85.05               | ( 12.34 ) | 1500              | ( 2240.3 ) | 81.10               | ( 11.77 ) |
| 800               | ( 980.3 )  | 84.79               | ( 12.30 ) | 1550              | ( 2330.3 ) | 80.82               | ( 11.73 ) |
| 850               | ( 1070.3 ) | 84.53               | ( 12.27 ) | 1600              | ( 2420.3 ) | 80.55               | ( 11.69 ) |
| 900               | ( 1160.3 ) | 84.28               | ( 12.23 ) |                   |            |                     |           |
| 950               | ( 1250.3 ) | 84.02               | ( 12.19 ) |                   |            |                     |           |

**Application Notes:** Data for Young's modulus is collected from references [7, 10] and compared against the equation below to approximate property trend with respect to temperature and porosity.

## Fit Equations:

$$G(T,P) = A \cdot (1 - P)^N \cdot [1 - A0 \cdot (T - 298)] / [1 - A1 \cdot (T - 298)]$$

$$G(T,P) = \text{Shear Modulus [GPa]}$$

$$T = \text{Temperature [K]}$$

$$P = \text{Fractional Porosity}$$

## Constants:

$$A = 87.3$$

$$N = 3.109$$

$$A0 = 1.00\text{E-}04$$

$$A1 = 4.40\text{E-}05$$

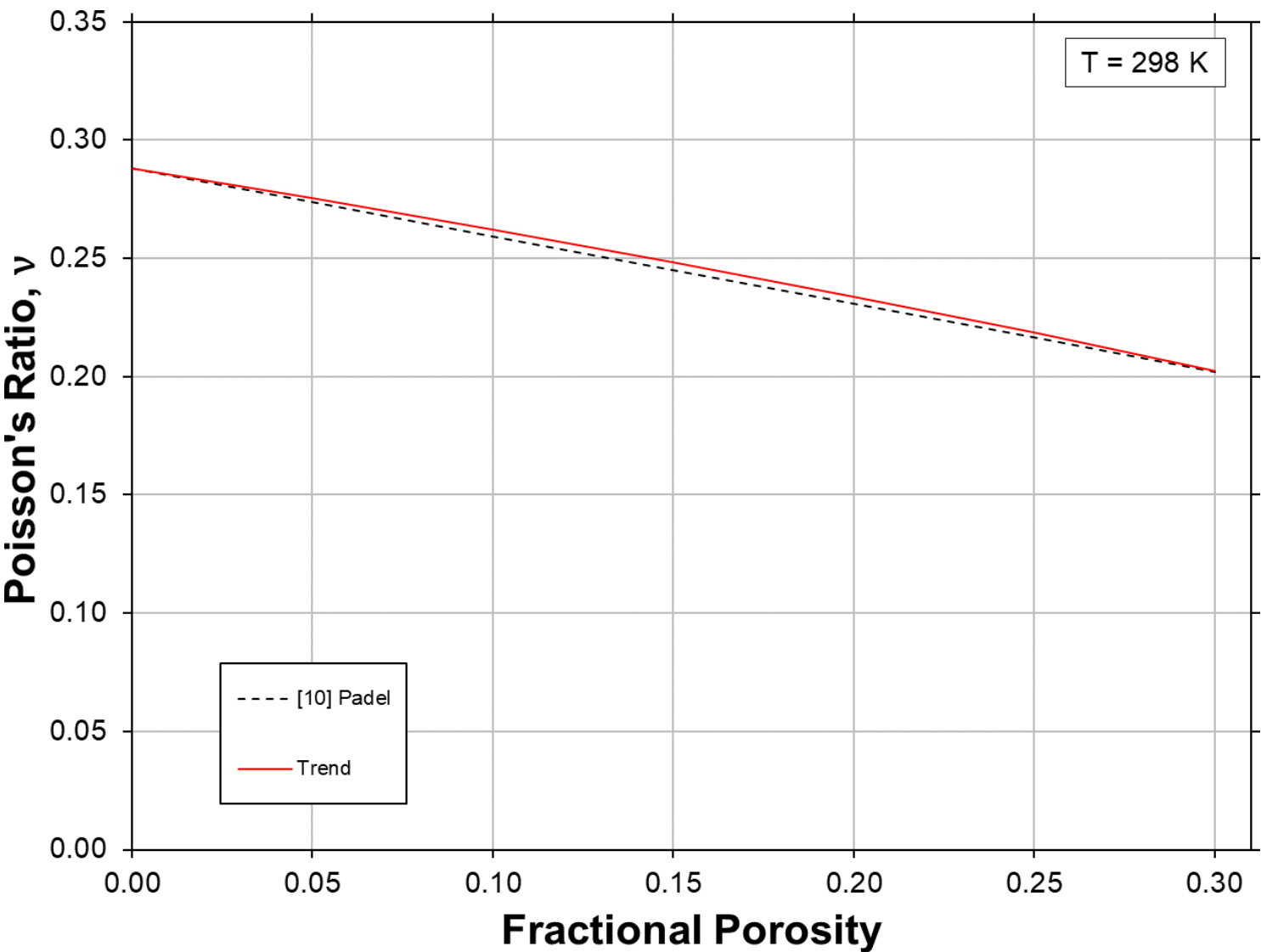


Figure 6.1.2-12: Poisson's Ratio versus Fractional Porosity of UC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Poisson's Ratio with Porosity

Room Temperature

| Porosity ( P ) | Poisson's Ratio ( $\nu$ ) |
|----------------|---------------------------|
| 0.00           | 0.288                     |
| 0.02           | 0.283                     |
| 0.04           | 0.278                     |
| 0.06           | 0.273                     |
| 0.08           | 0.268                     |
| 0.10           | 0.262                     |
| 0.12           | 0.257                     |
| 0.14           | 0.251                     |
| 0.16           | 0.245                     |
| 0.18           | 0.240                     |
| 0.20           | 0.234                     |
| 0.22           | 0.228                     |
| 0.24           | 0.222                     |
| 0.26           | 0.215                     |
| 0.28           | 0.209                     |
| 0.30           | 0.202                     |

**Application Notes:** Data for Poisson's Ratio is calculated from Young's modulus and Shear modulus data collected from references [7, 10] using the equation below to approximate property trend with respect to temperature and porosity.

## Fit Equations:

$$\nu(T, P) = E(T, P) / (2 \cdot G(T, P)) - 1$$

$$\nu(T, P) = \text{Poisson's Ratio}$$

$$E(T, P) = \text{Young's modulus}$$

$$G(T, P) = \text{Shear modulus}$$

$$T = \text{Temperature [K]}, P = \text{Fractional Porosity}$$

## Constants:

$$\text{Temperature [K]} = 298$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

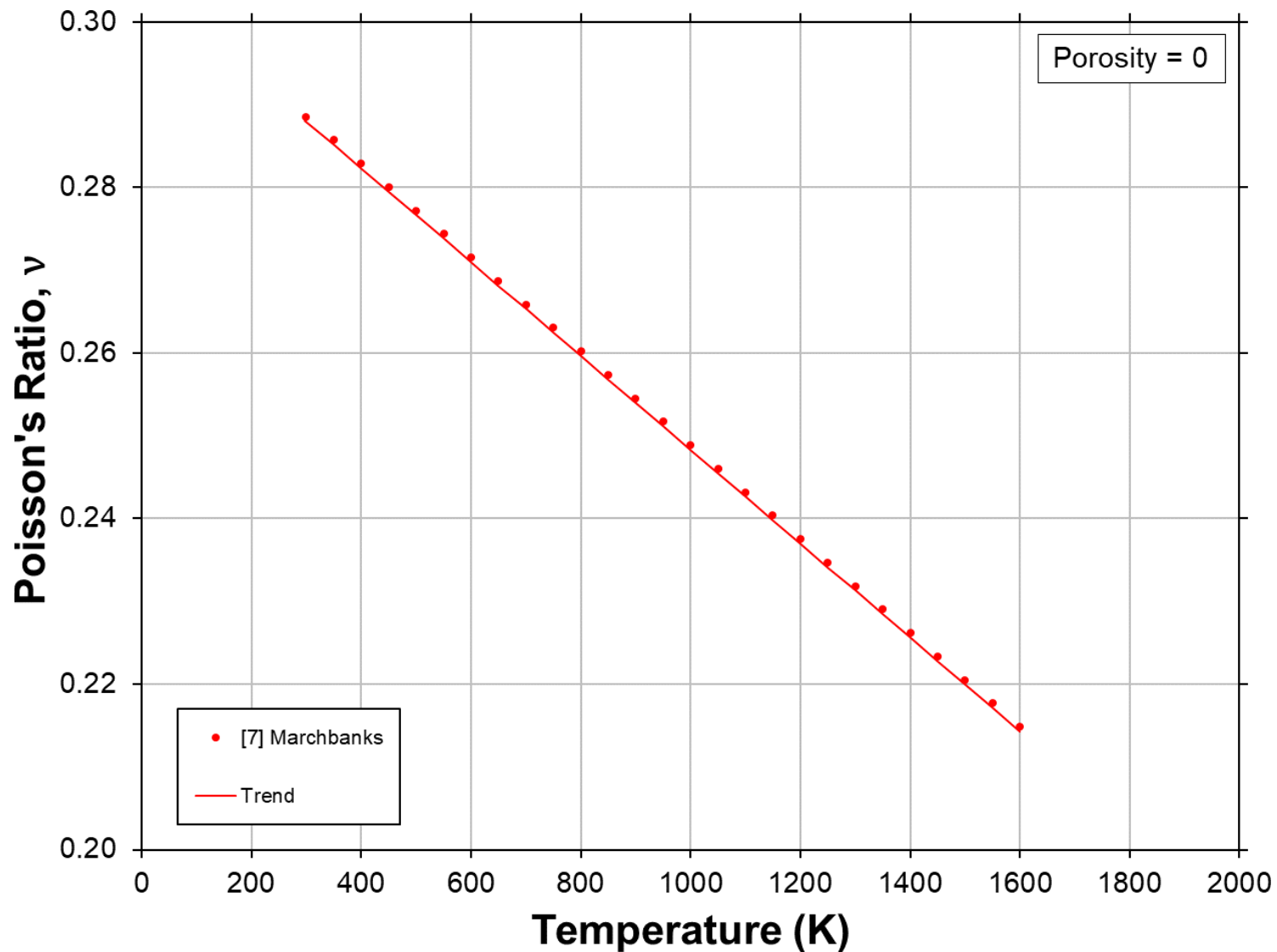


Figure 6.1.2-13: Poisson's Ratio versus Temperature of UC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|------------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )     |                           | K                 | ( °F )     |                           |
| 298               | ( 76.7 )   | 0.288                     | 1000              | ( 1340.3 ) | 0.248                     |
| 300               | ( 80.3 )   | 0.288                     | 1050              | ( 1430.3 ) | 0.245                     |
| 350               | ( 170.3 )  | 0.285                     | 1100              | ( 1520.3 ) | 0.243                     |
| 400               | ( 260.3 )  | 0.282                     | 1150              | ( 1610.3 ) | 0.240                     |
| 450               | ( 350.3 )  | 0.279                     | 1200              | ( 1700.3 ) | 0.237                     |
| 500               | ( 440.3 )  | 0.277                     | 1250              | ( 1790.3 ) | 0.234                     |
| 550               | ( 530.3 )  | 0.274                     | 1300              | ( 1880.3 ) | 0.231                     |
| 600               | ( 620.3 )  | 0.271                     | 1350              | ( 1970.3 ) | 0.228                     |
| 650               | ( 710.3 )  | 0.268                     | 1400              | ( 2060.3 ) | 0.226                     |
| 700               | ( 800.3 )  | 0.265                     | 1450              | ( 2150.3 ) | 0.223                     |
| 750               | ( 890.3 )  | 0.262                     | 1500              | ( 2240.3 ) | 0.220                     |
| 800               | ( 980.3 )  | 0.260                     | 1550              | ( 2330.3 ) | 0.217                     |
| 850               | ( 1070.3 ) | 0.257                     | 1600              | ( 2420.3 ) | 0.214                     |
| 900               | ( 1160.3 ) | 0.254                     |                   |            |                           |
| 950               | ( 1250.3 ) | 0.251                     |                   |            |                           |

**Application Notes:** Data for Poisson's Ratio is calculated from fitted Young's modulus and Shear modulus data collected from references [7, 10], using the equation below to approximate property trend with respect to temperature and porosity.

## Fit Equations:

$$\nu(T, P) = E(T, P) / (2 \cdot G(T, P)) - 1$$

$$\nu(T, P) = \text{Poisson's Ratio}$$

$$E(T, P) = \text{Young's modulus}$$

$$G(T, P) = \text{Shear modulus}$$

$$T = \text{Temperature [K]}, P = \text{Fractional Porosity}$$

## Constants:

$$\text{Fractional Porosity (P)} = \varnothing$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

Revision 2: 04-26-2023

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 300                 | 13627                                 | 0.782                                                   | 20.83                                       | 0.007                                              | 10.13                                     | 0.203                                          | 224.86                             | 87.29                            | 0.288                       |
| 350                 | 13606                                 | 0.853                                                   | 21.00                                       | 0.058                                              | 10.18                                     | 0.213                                          | 223.73                             | 87.05                            | 0.285                       |
| 400                 | 13585                                 | 0.922                                                   | 21.16                                       | 0.109                                              | 10.22                                     | 0.220                                          | 222.61                             | 86.80                            | 0.282                       |
| 450                 | 13564                                 | 0.990                                                   | 21.32                                       | 0.161                                              | 10.27                                     | 0.225                                          | 221.48                             | 86.55                            | 0.279                       |
| 500                 | 13543                                 | 1.057                                                   | 21.49                                       | 0.214                                              | 10.32                                     | 0.229                                          | 220.36                             | 86.30                            | 0.277                       |
| 550                 | 13522                                 | 1.122                                                   | 21.65                                       | 0.267                                              | 10.37                                     | 0.232                                          | 219.23                             | 86.05                            | 0.274                       |
| 600                 | 13500                                 | 1.187                                                   | 21.81                                       | 0.320                                              | 10.42                                     | 0.235                                          | 218.11                             | 85.80                            | 0.271                       |
| 650                 | 13478                                 | 1.249                                                   | 21.98                                       | 0.374                                              | 10.48                                     | 0.237                                          | 216.98                             | 85.55                            | 0.268                       |
| 700                 | 13456                                 | 1.311                                                   | 22.14                                       | 0.429                                              | 10.53                                     | 0.239                                          | 215.86                             | 85.30                            | 0.265                       |
| 750                 | 13434                                 | 1.371                                                   | 22.31                                       | 0.484                                              | 10.58                                     | 0.241                                          | 214.73                             | 85.05                            | 0.262                       |
| 800                 | 13412                                 | 1.429                                                   | 22.47                                       | 0.539                                              | 10.63                                     | 0.243                                          | 213.61                             | 84.79                            | 0.260                       |
| 850                 | 13390                                 | 1.487                                                   | 22.64                                       | 0.595                                              | 10.68                                     | 0.245                                          | 212.49                             | 84.53                            | 0.257                       |
| 900                 | 13367                                 | 1.543                                                   | 22.81                                       | 0.652                                              | 10.74                                     | 0.247                                          | 211.36                             | 84.28                            | 0.254                       |
| 1000                | 13321                                 | 1.651                                                   | 23.14                                       | 0.767                                              | 10.85                                     | 0.251                                          | 209.11                             | 83.76                            | 0.248                       |
| 1100                | 13275                                 | 1.754                                                   | 23.47                                       | 0.884                                              | 10.96                                     | 0.255                                          | 206.86                             | 83.24                            | 0.243                       |
| 1200                | 13228                                 | 1.851                                                   | 23.81                                       | 1.004                                              | 11.07                                     | 0.259                                          | 204.61                             | 82.71                            | 0.237                       |
| 1300                | 13180                                 | 1.943                                                   | 24.15                                       | 1.126                                              | 11.18                                     | 0.263                                          | 202.37                             | 82.18                            | 0.231                       |
| 1400                | 13131                                 | 2.030                                                   | 24.49                                       | 1.250                                              | 11.30                                     | 0.267                                          | 200.12                             | 81.64                            | 0.226                       |
| 1500                | 13082                                 | 2.111                                                   | 24.83                                       | 1.377                                              | 11.41                                     | 0.271                                          | 197.87                             | 81.10                            | 0.220                       |
| 1600                | 13032                                 | 2.187                                                   | 25.17                                       | 1.507                                              | 11.53                                     | 0.276                                          | 195.62                             | 80.55                            | 0.214                       |
| 1700                | 12981                                 | 2.257                                                   | 25.52                                       | 1.639                                              | 11.65                                     | 0.280                                          | -                                  | -                                | -                           |
| 1800                | 12929                                 | 2.323                                                   | 25.86                                       | 1.774                                              | 11.77                                     | 0.285                                          | -                                  | -                                | -                           |
| 1900                | 12877                                 | -                                                       | 26.21                                       | 1.912                                              | 11.90                                     | 0.290                                          | -                                  | -                                | -                           |
| 2000                | 12824                                 | -                                                       | 26.56                                       | 2.052                                              | 12.02                                     | 0.295                                          | -                                  | -                                | -                           |
| 2200                | 12715                                 | -                                                       | -                                           | 2.342                                              | 12.28                                     | 0.306                                          | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.1 Nuclear Fuel Materials

6.1.2 Uranium Carbide (UC)

**Revision 2: 04-26-2023**

**References**

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## **6 Nuclear Materials**

### **6.1 Nuclear Fuel Materials**



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

|                          |                            |                        |
|--------------------------|----------------------------|------------------------|
| 6 Nuclear Materials      | 6.1 Nuclear Fuel Materials | 6.1.3 <i>See Annex</i> |
| Revision 3.1: 08-05-2024 |                            | General                |

Data for this material can be found in the SNP-HDBK-0008 Annex.

## **6 Nuclear Materials**

### **6.2 Neutron Control Materials**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 2.1: 08-25-2023

General

### $\text{ZrH}_2$ Room Temperature Properties

|                                                              |                    |
|--------------------------------------------------------------|--------------------|
| Molar Mass, [g/mol]                                          | 93.24              |
| Theoretical Density, [ $\text{kg/m}^3$ ]                     | 5,586              |
| Melting Point, [K]                                           | *                  |
| Heat Capacity, [J/(g-K)]                                     | 0.331              |
| Thermal Conductivity, [W/(m-K)]                              | 67.7               |
| Linear expansion coefficient, [ $\mu\text{m}/(\text{m-K})$ ] | 8.34               |
| Electrical resistivity, [ $\mu\Omega\text{-m}$ ]             | 0.158 <sup>a</sup> |
| Young's Modulus, [GPa]                                       | 77.4 <sup>b</sup>  |
| Shear Modulus, [GPa]                                         | 27.5 <sup>b</sup>  |
| Poisson's Ratio, [-]                                         | 0.41 <sup>b</sup>  |

\* Thermal decomposition is dependent on temperature, pressure and atmosphere

<sup>a</sup> Extrapolated from fitted curve

<sup>b</sup> Based on average density functional theory (DFT) by P.F. Weck [1]



## Zirconium – Hydrogen Phase Diagram

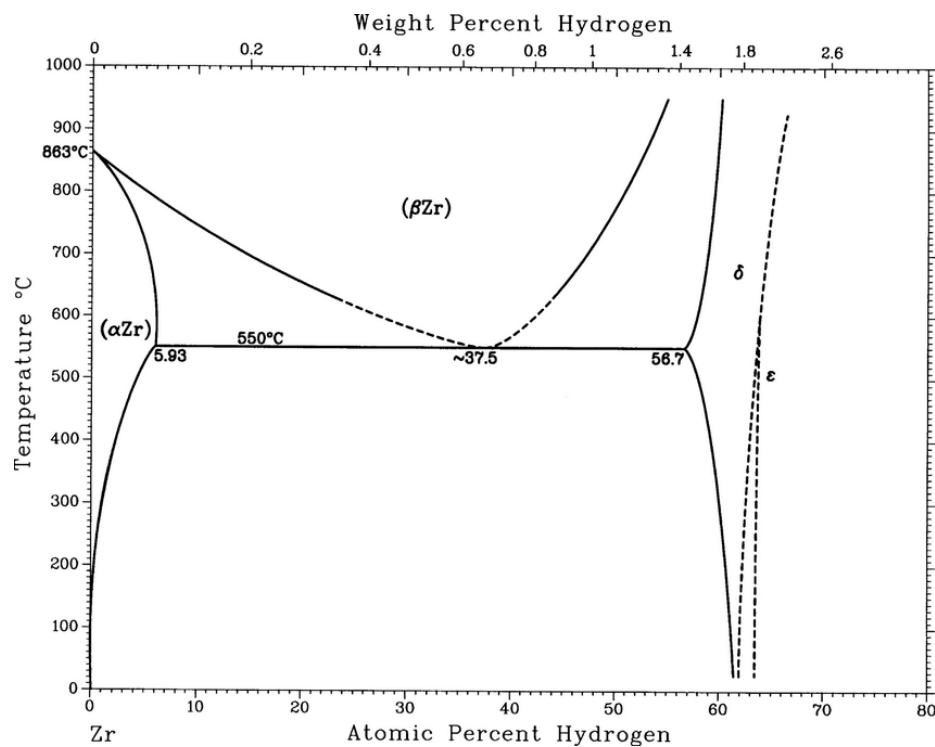


Figure 6.2.1-1: Zirconium – Hydrogen Phase Diagram [2].

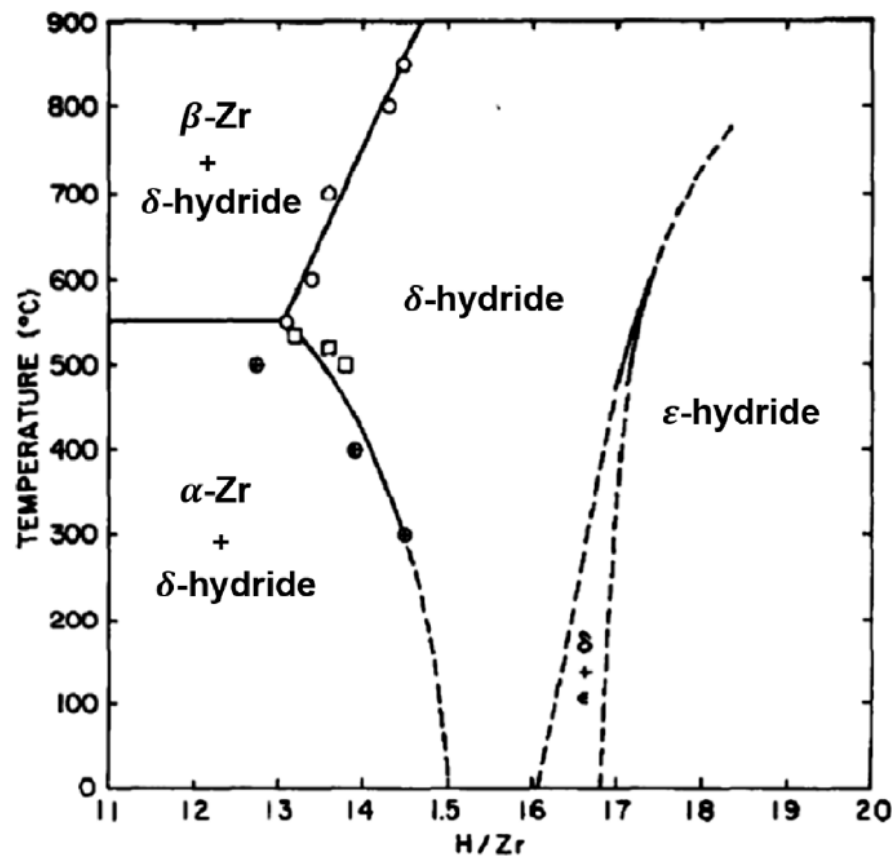


Figure 6.2.1-2: Zirconium – Hydrogen Phase Diagram [3].



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 3: 01-25-2024

Density with H/Zr Ratio

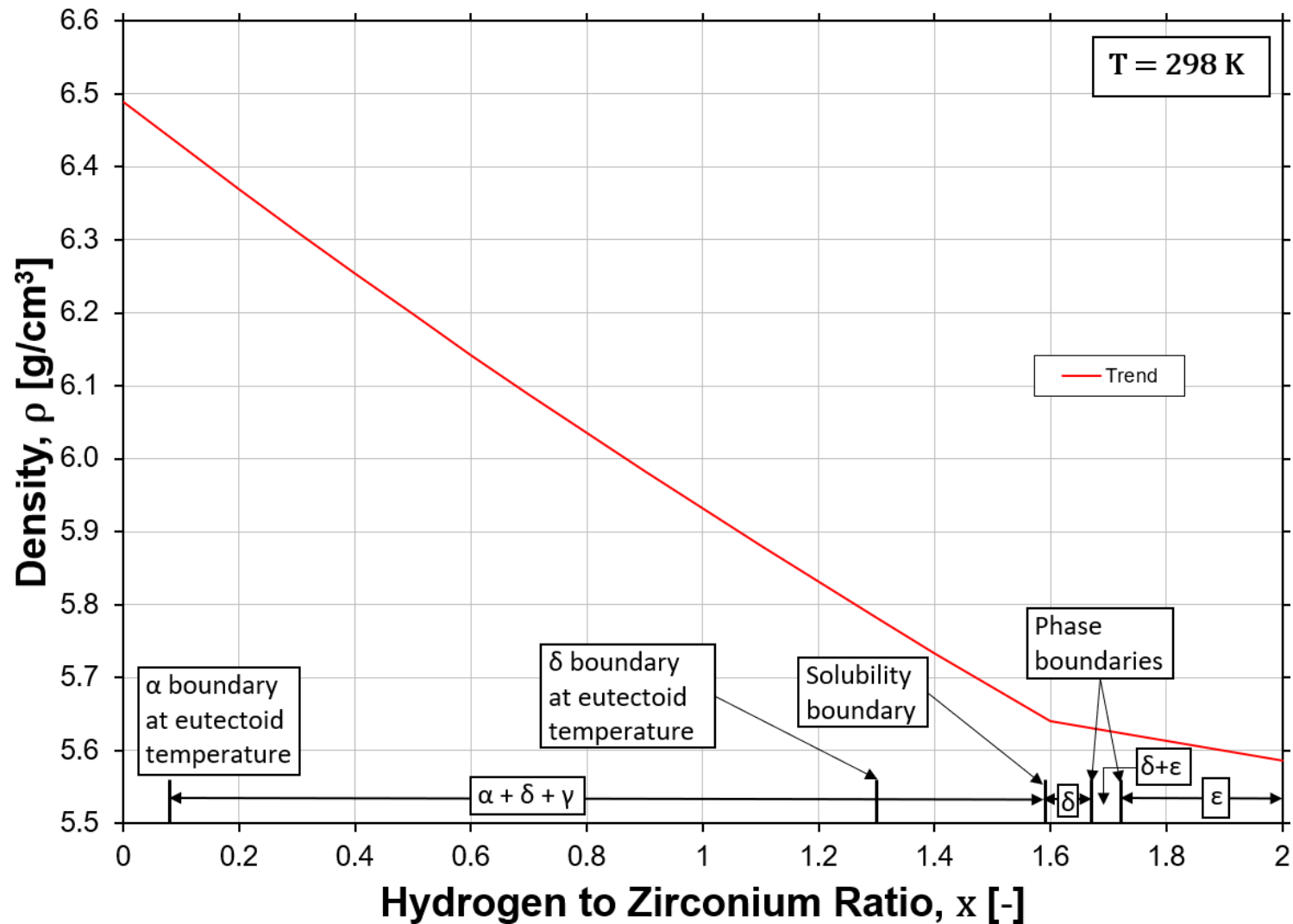


Figure 6.2.1-3: Density versus Hydrogen to Zirconium ratio ( $x$ ), adapted from M.T. Simnad [4]. Different phases of  $\text{ZrH}_x$  are also identified.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 3: 01-25-2024

Density with H/Zr Ratio

## 100% Theoretical Density

| H/Zr Ratio( x ) | Room Temperature Density ( ρ <sub>RT</sub> ) |                        | H/Zr Ratio( x ) | Room Temperature Density ( ρ <sub>RT</sub> ) |                        |
|-----------------|----------------------------------------------|------------------------|-----------------|----------------------------------------------|------------------------|
|                 | kg/m <sup>3</sup>                            | ( lb/ft <sup>3</sup> ) |                 | kg/m <sup>3</sup>                            | ( lb/ft <sup>3</sup> ) |
| 0.00            | 6489                                         | ( 405.1 )              | 1.10            | 5881                                         | ( 367.1 )              |
| 0.10            | 6429                                         | ( 401.3 )              | 1.20            | 5831                                         | ( 364.0 )              |
| 0.20            | 6369                                         | ( 397.6 )              | 1.30            | 5782                                         | ( 361.0 )              |
| 0.30            | 6311                                         | ( 394.0 )              | 1.40            | 5734                                         | ( 358.0 )              |
| 0.40            | 6254                                         | ( 390.4 )              | 1.50            | 5687                                         | ( 355.0 )              |
| 0.50            | 6198                                         | ( 386.9 )              | 1.60            | 5640                                         | ( 352.1 )              |
| 0.60            | 6143                                         | ( 383.5 )              | 1.70            | 5626                                         | ( 351.2 )              |
| 0.70            | 6088                                         | ( 380.1 )              | 1.80            | 5613                                         | ( 350.4 )              |
| 0.80            | 6035                                         | ( 376.8 )              | 1.90            | 5600                                         | ( 349.6 )              |
| 0.90            | 5983                                         | ( 373.5 )              | 2.00            | 5587                                         | ( 348.8 )              |
| 1.00            | 5931                                         | ( 370.3 )              |                 |                                              |                        |

**Application Notes:** Density as a function of H/Zr ratio is calculated using the equations below to approximate property trend with H/Zr ratio. The equation for room temperature density is taken from [4]. Density with respect to temperature is calculated as a function of thermal expansion, as shown below, for (H/Zr) from 1.6 to 2.0. Data in the table is for T = 298 K.

### Density Calculation:

$$\rho_{RT}(x) = 1000/(A0 + A1 \cdot x)$$

$$\rho(T) = \rho_{RT}(x)/(1 + dL/L_0(T)/100)^3$$

$$\rho_{RT}(x) = \text{Room Temperature Density [kg/m}^3\text{]}$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$x = \text{Hydrogen/Zirconium ratio}$$

$$T = \text{Temperature [K]}$$

### Constants:

$$T \text{ Range: } 293 \leq T \leq 977$$

$$H/Zr \text{ Ratio: } 0 \leq x \leq 1.6 \quad 1.6 \leq x \leq 2.0$$

$$A0 = 0.1541 \quad 0.1706$$

$$A1 = 0.0145 \quad 0.0042$$

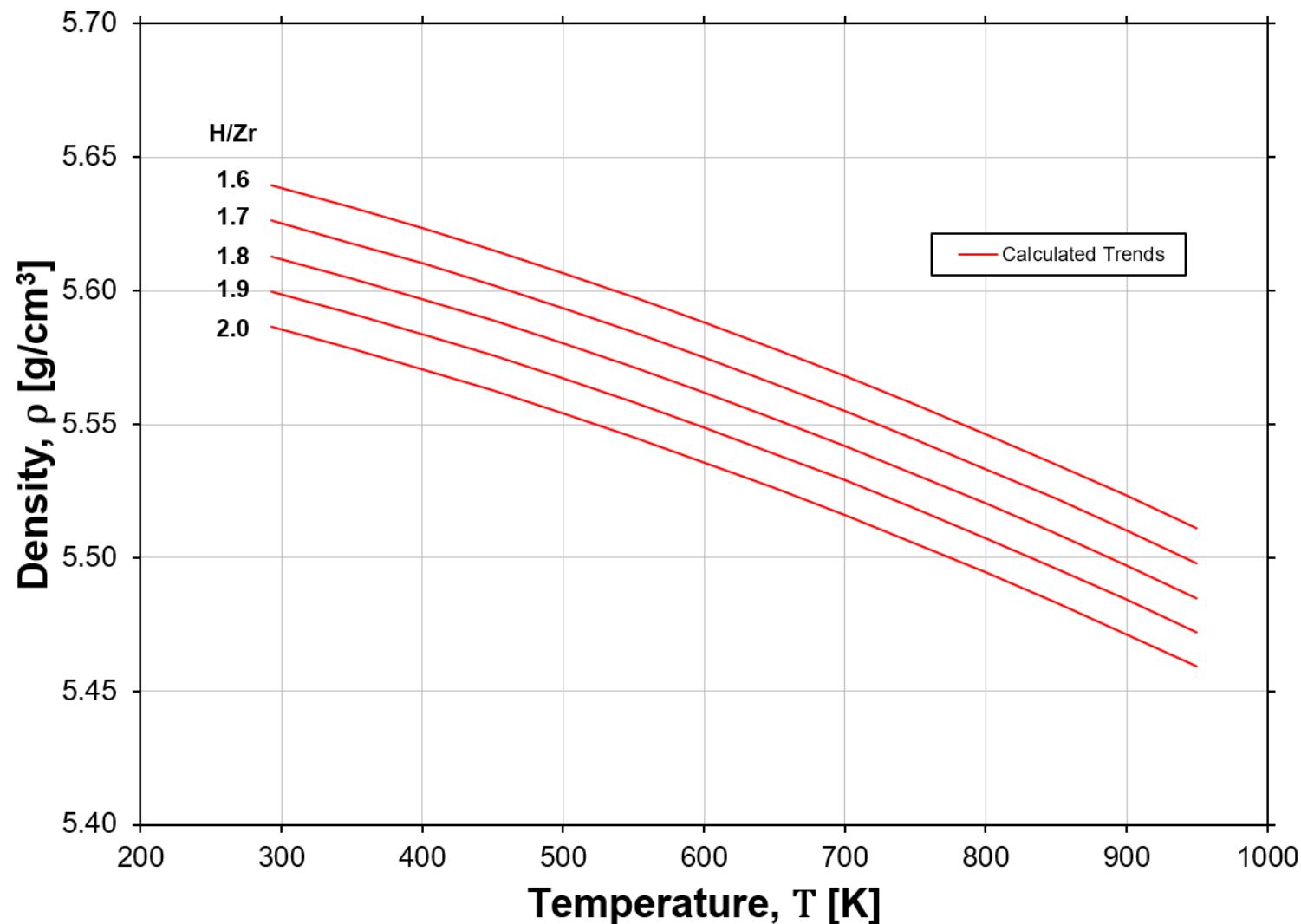


Figure 6.2.1-4: Density versus Temperature for Zirconium Hydrides at various H/Zr ratios. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 3: 01-25-2024

Density with Temperature

## 100% Theoretical Density

| Temperature ( T ) |           | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|-----------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )    | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 300               | ( 80.3 )  | 5586              | ( 348.7 )              | 650               | ( 710.3 )  | 5526              | ( 345.0 )              |
| 325               | ( 125.3 ) | 5582              | ( 348.5 )              | 675               | ( 755.3 )  | 5521              | ( 344.7 )              |
| 350               | ( 170.3 ) | 5578              | ( 348.3 )              | 700               | ( 800.3 )  | 5516              | ( 344.4 )              |
| 375               | ( 215.3 ) | 5575              | ( 348.0 )              | 725               | ( 845.3 )  | 5511              | ( 344.0 )              |
| 400               | ( 260.3 ) | 5571              | ( 347.8 )              | 750               | ( 890.3 )  | 5505              | ( 343.7 )              |
| 425               | ( 305.3 ) | 5567              | ( 347.5 )              | 775               | ( 935.3 )  | 5500              | ( 343.4 )              |
| 450               | ( 350.3 ) | 5563              | ( 347.3 )              | 800               | ( 980.3 )  | 5494              | ( 343.0 )              |
| 475               | ( 395.3 ) | 5558              | ( 347.0 )              | 825               | ( 1025.3 ) | 5489              | ( 342.7 )              |
| 500               | ( 440.3 ) | 5554              | ( 346.7 )              | 850               | ( 1070.3 ) | 5483              | ( 342.3 )              |
| 525               | ( 485.3 ) | 5550              | ( 346.5 )              | 875               | ( 1115.3 ) | 5477              | ( 341.9 )              |
| 550               | ( 530.3 ) | 5545              | ( 346.2 )              | 900               | ( 1160.3 ) | 5471              | ( 341.6 )              |
| 575               | ( 575.3 ) | 5541              | ( 345.9 )              | 925               | ( 1205.3 ) | 5465              | ( 341.2 )              |
| 600               | ( 620.3 ) | 5536              | ( 345.6 )              | 950               | ( 1250.3 ) | 5459              | ( 340.8 )              |
| 625               | ( 665.3 ) | 5531              | ( 345.3 )              | 975               | ( 1295.3 ) | 5453              | ( 340.4 )              |

**Application Notes:** Density as a function of H/Zr ratio is calculated using the equations below to approximate property trend with H/Zr ratio. The equation for room temperature density is taken from [4]. Density with respect to temperature is calculated as a function of thermal expansion, as shown below, for (H/Zr) from 1.6 to 2.0. Data in the table is for x = 2.0.

### Density Calculation:

$$\rho_{RT}(x) = 1000 / (A0 + A1 \cdot x)$$

$$\rho(T) = \rho_{RT}(x) / (1 + dL/L_0(T)/100)^3$$

$$\rho_{RT}(x) = \text{Room Temperature Density [kg/m}^3\text{]}$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$x = \text{Hydrogen/Zirconium ratio}$$

$$T = \text{Temperature [K]}$$

### Constants:

$$T \text{ Range: } 293 \leq T \leq 977$$

$$H/Zr \text{ Ratio: } 0 \leq x \leq 1.6 \quad 1.6 \leq x \leq 2.0$$

$$A0 = 0.1541 \quad 0.1706$$

$$A1 = 0.0145 \quad 0.0042$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

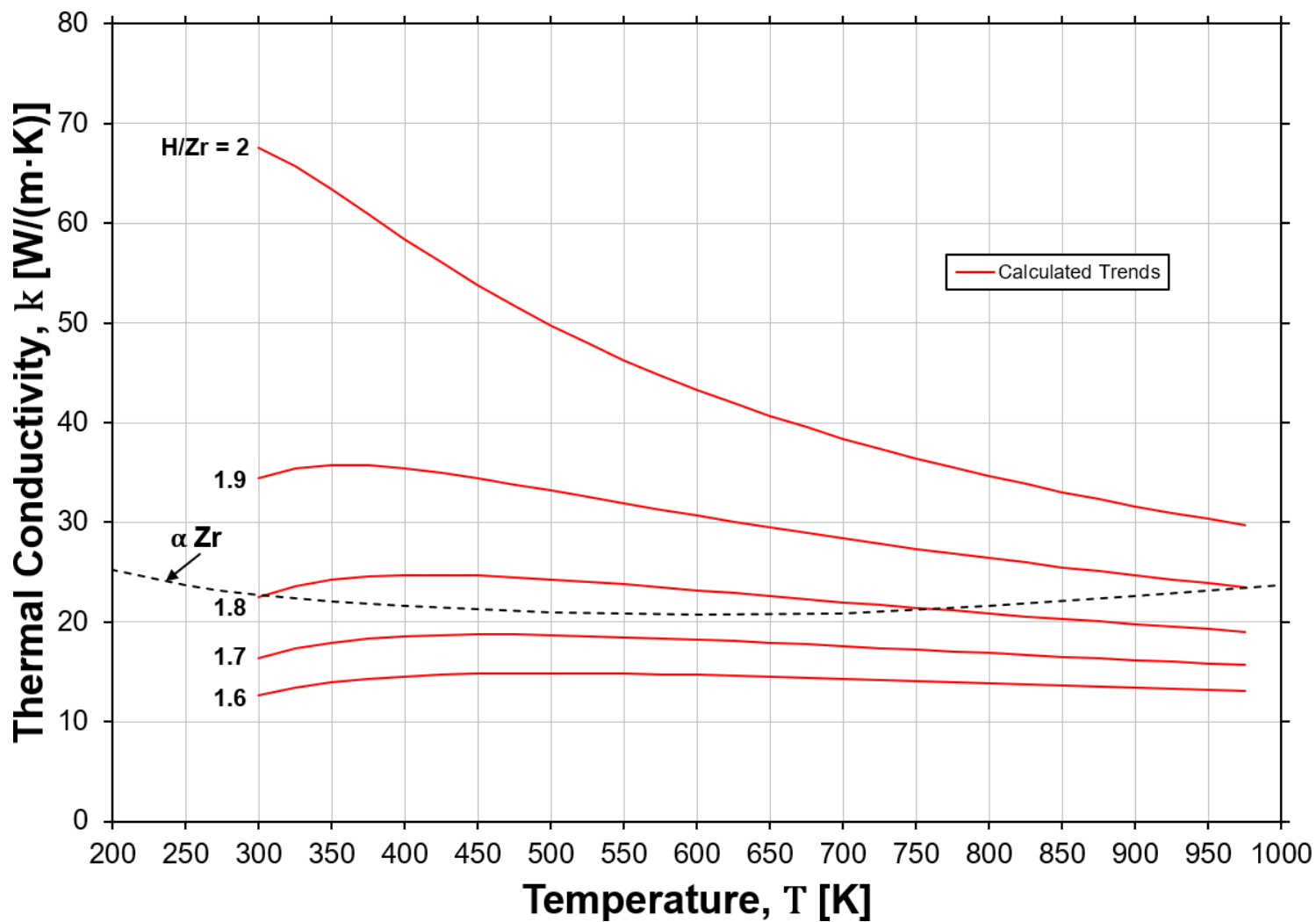


Figure 6.2.1-5: Thermal Conductivity versus Temperature of Zirconium Hydrides.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Thermal Conductivity ( k ) |                                       | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       |
|-------------------|-----------|----------------------------|---------------------------------------|-------------------|------------|----------------------------|---------------------------------------|
| K                 | ( °F )    | W/(m-K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) | K                 | ( °F )     | W/(m-K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) |
| 300               | ( 80.3 )  | 67.57                      | ( 468.80 )                            | 575               | ( 575.3 )  | 44.73                      | ( 310.34 )                            |
| 325               | ( 125.3 ) | 65.72                      | ( 455.99 )                            | 600               | ( 620.3 )  | 43.29                      | ( 300.32 )                            |
| 350               | ( 170.3 ) | 63.38                      | ( 439.71 )                            | 650               | ( 710.3 )  | 40.69                      | ( 282.28 )                            |
| 375               | ( 215.3 ) | 60.88                      | ( 422.39 )                            | 700               | ( 800.3 )  | 38.41                      | ( 266.53 )                            |
| 400               | ( 260.3 ) | 58.41                      | ( 405.23 )                            | 750               | ( 890.3 )  | 36.42                      | ( 252.65 )                            |
| 425               | ( 305.3 ) | 56.03                      | ( 388.76 )                            | 800               | ( 980.3 )  | 34.64                      | ( 240.34 )                            |
| 450               | ( 350.3 ) | 53.80                      | ( 373.24 )                            | 850               | ( 1070.3 ) | 33.05                      | ( 229.33 )                            |
| 475               | ( 395.3 ) | 51.71                      | ( 358.74 )                            | 900               | ( 1160.3 ) | 31.62                      | ( 219.41 )                            |
| 500               | ( 440.3 ) | 49.76                      | ( 345.26 )                            | 950               | ( 1250.3 ) | 30.33                      | ( 210.42 )                            |
| 525               | ( 485.3 ) | 47.96                      | ( 332.74 )                            | 975               | ( 1295.3 ) | 29.72                      | ( 206.23 )                            |
| 550               | ( 530.3 ) | 46.28                      | ( 321.13 )                            |                   |            |                            |                                       |

**Application Notes:** The thermal conductivity trend is calculated from the thermal diffusivity, specific heat, and density trends, as shown in the equation below, to approximate property trend with respect to hydrogen-zirconium ratio and temperature. Data in the table is for H/Zr = 2.0.

## Fit Equation:

$$k(x, T) = \alpha \cdot C_p \cdot \rho$$

$$k(x, T) = \text{Thermal Conductivity [W/m} \cdot \text{K]}$$

$$\alpha(x, T) = \text{Thermal Diffusivity [m}^2\text{/s]}$$

$$C_p(x, T) = \text{Specific Heat [J/kg} \cdot \text{K]}$$

$$\rho(x, T) = \text{Density [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

$$x = \text{Hydrongen/Zirconium ratio}$$

**Temperature Range:**  $293 \leq T \leq 977$

**H/Zr Ratio:**  $1.6 \leq x \leq 2.0$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 0: 08-05-2020

Thermal Expansion with Temperature

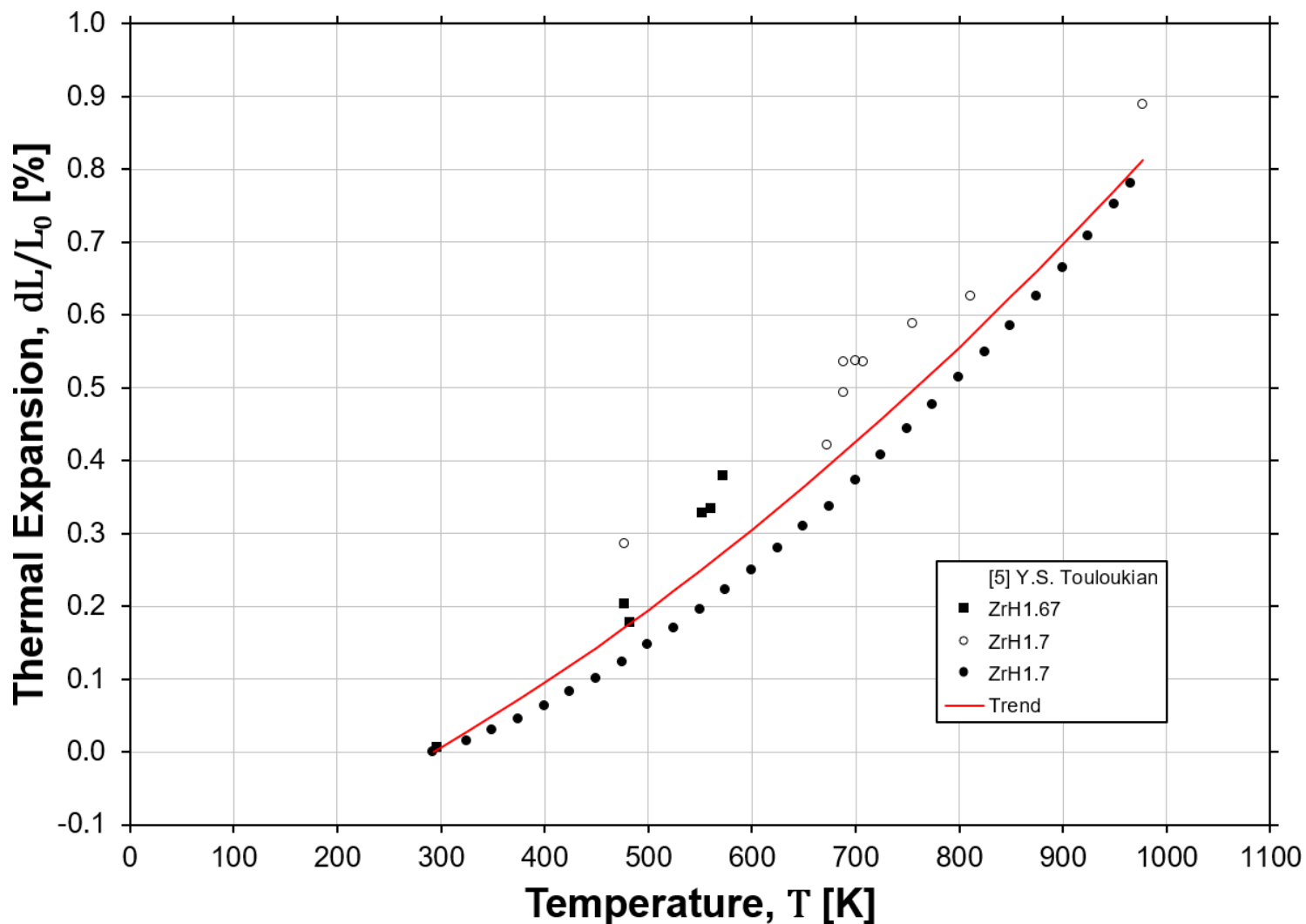


Figure 6.2.1-6: Thermal Expansion versus temperature of Zirconium Hydride.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|-----------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )    | %                                       | K                 | ( °F )     | %                                       |
| 300               | ( 80.3 )  | 0.006                                   | 650               | ( 710.3 )  | 0.364                                   |
| 325               | ( 125.3 ) | 0.027                                   | 675               | ( 755.3 )  | 0.394                                   |
| 350               | ( 170.3 ) | 0.049                                   | 700               | ( 800.3 )  | 0.425                                   |
| 375               | ( 215.3 ) | 0.072                                   | 725               | ( 845.3 )  | 0.457                                   |
| 400               | ( 260.3 ) | 0.095                                   | 750               | ( 890.3 )  | 0.489                                   |
| 425               | ( 305.3 ) | 0.119                                   | 775               | ( 935.3 )  | 0.522                                   |
| 450               | ( 350.3 ) | 0.144                                   | 800               | ( 980.3 )  | 0.556                                   |
| 475               | ( 395.3 ) | 0.169                                   | 825               | ( 1025.3 ) | 0.590                                   |
| 500               | ( 440.3 ) | 0.195                                   | 850               | ( 1070.3 ) | 0.625                                   |
| 525               | ( 485.3 ) | 0.221                                   | 875               | ( 1115.3 ) | 0.661                                   |
| 550               | ( 530.3 ) | 0.248                                   | 900               | ( 1160.3 ) | 0.697                                   |
| 575               | ( 575.3 ) | 0.276                                   | 925               | ( 1205.3 ) | 0.734                                   |
| 600               | ( 620.3 ) | 0.305                                   | 950               | ( 1250.3 ) | 0.772                                   |
| 625               | ( 665.3 ) | 0.334                                   | 975               | ( 1295.3 ) | 0.810                                   |

**Application Notes:** Data for thermal expansion is collected from reference [5], and it is fitted with the equation below to approximate property trend with respect to temperature. While the trend applies for a hydrogen/zirconium ratio of 1.67 to 1.7, the trend is used for ratios of 1.6 to 2.0 for use in density calculations; density, in turn, is used in thermal conductivity calculations.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$$dL/L_0(T) = \text{Thermal Expansion} [\%]$$

$$T = \text{Temperature} [K]$$

## Constants:

$$T \text{ Range: } 293 \leq T \leq 977$$

$$A0 = -0.1997$$

$$A1 = 0.5292$$

$$A2 = 0.5194$$



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

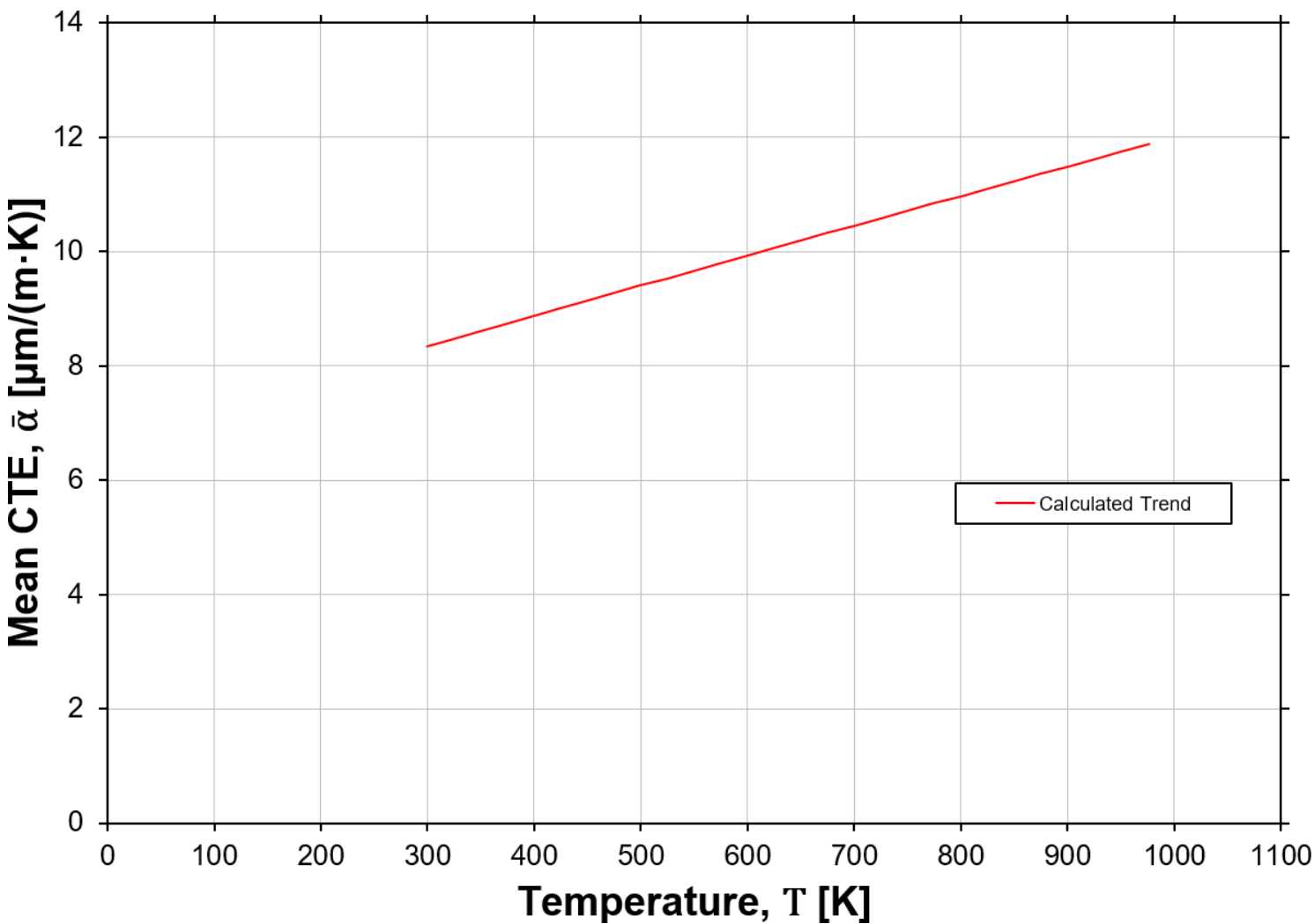


Figure 6.2.1-7: Mean Coefficient of Thermal Expansion versus Temperature of Zirconium Hydride. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Mean CTE ( $\bar{\alpha}$ )           |                                                   | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                   |
|-------------------|-----------|---------------------------------------|---------------------------------------------------|-------------------|------------|---------------------------------------|---------------------------------------------------|
| K                 | ( °F )    | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^\circ\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^\circ\text{F})$ ) |
| 300               | ( 80.3 )  | 8.342                                 | ( 4.635 )                                         | 650               | ( 710.3 )  | 10.193                                | ( 5.663 )                                         |
| 325               | ( 125.3 ) | 8.476                                 | ( 4.709 )                                         | 675               | ( 755.3 )  | 10.323                                | ( 5.735 )                                         |
| 350               | ( 170.3 ) | 8.609                                 | ( 4.783 )                                         | 700               | ( 800.3 )  | 10.453                                | ( 5.807 )                                         |
| 375               | ( 215.3 ) | 8.743                                 | ( 4.857 )                                         | 725               | ( 845.3 )  | 10.583                                | ( 5.880 )                                         |
| 400               | ( 260.3 ) | 8.876                                 | ( 4.931 )                                         | 750               | ( 890.3 )  | 10.713                                | ( 5.952 )                                         |
| 425               | ( 305.3 ) | 9.008                                 | ( 5.005 )                                         | 775               | ( 935.3 )  | 10.843                                | ( 6.024 )                                         |
| 450               | ( 350.3 ) | 9.141                                 | ( 5.078 )                                         | 800               | ( 980.3 )  | 10.972                                | ( 6.095 )                                         |
| 475               | ( 395.3 ) | 9.273                                 | ( 5.152 )                                         | 825               | ( 1025.3 ) | 11.101                                | ( 6.167 )                                         |
| 500               | ( 440.3 ) | 9.405                                 | ( 5.225 )                                         | 850               | ( 1070.3 ) | 11.230                                | ( 6.239 )                                         |
| 525               | ( 485.3 ) | 9.537                                 | ( 5.298 )                                         | 875               | ( 1115.3 ) | 11.358                                | ( 6.310 )                                         |
| 550               | ( 530.3 ) | 9.668                                 | ( 5.371 )                                         | 900               | ( 1160.3 ) | 11.487                                | ( 6.382 )                                         |
| 575               | ( 575.3 ) | 9.800                                 | ( 5.444 )                                         | 925               | ( 1205.3 ) | 11.615                                | ( 6.453 )                                         |
| 600               | ( 620.3 ) | 9.931                                 | ( 5.517 )                                         | 950               | ( 1250.3 ) | 11.743                                | ( 6.524 )                                         |
| 625               | ( 665.3 ) | 10.062                                | ( 5.590 )                                         | 975               | ( 1295.3 ) | 11.871                                | ( 6.595 )                                         |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated from thermal expansion and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range:  $293 \leq T \leq 977$

A0 = 6.721

A1 = 5.459

A2 = -0.1818



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 0: 08-05-2020

Thermal Diffusivity with Temperature

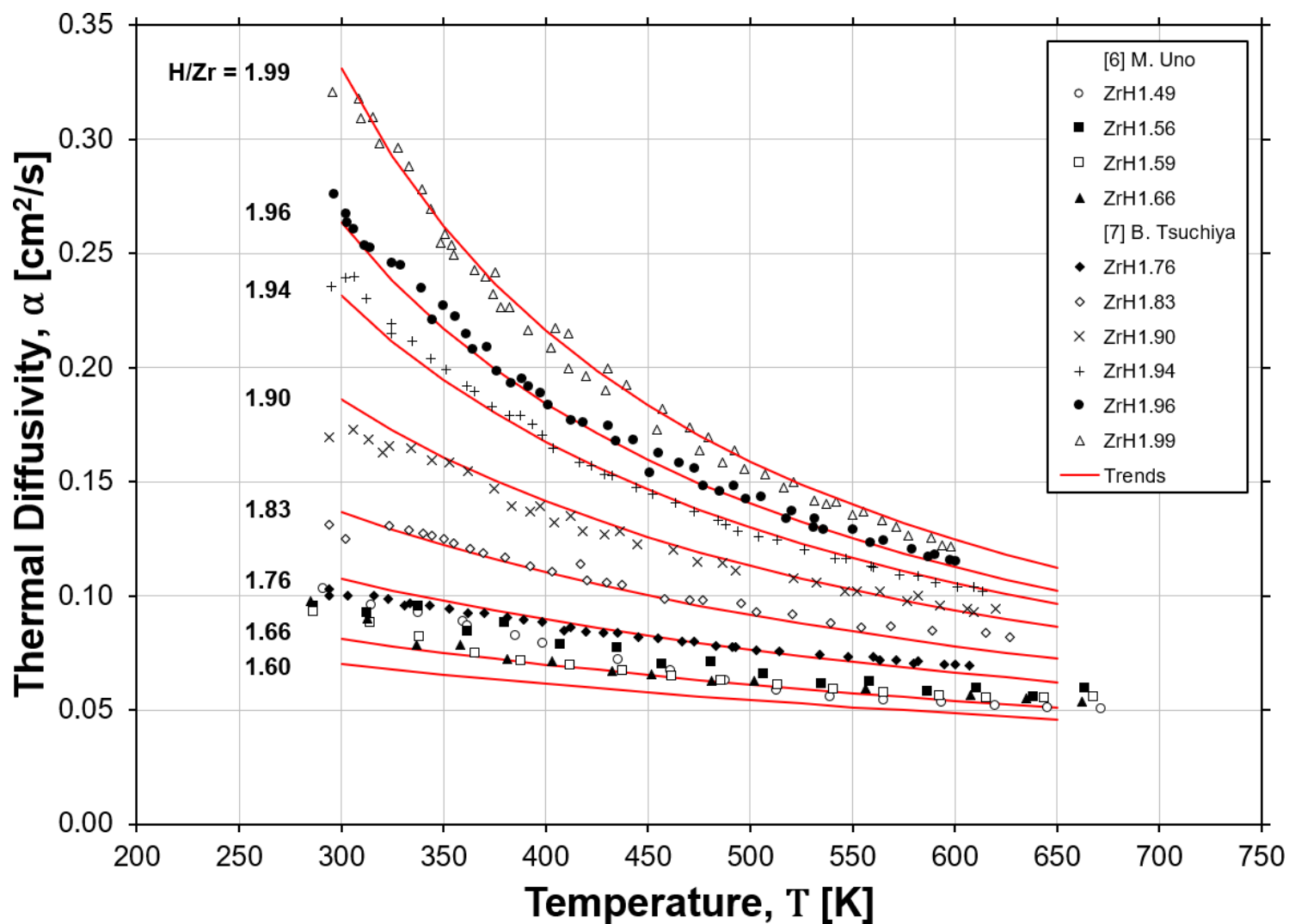


Figure 6.2.1-8: Thermal Diffusivity versus Temperature of Zirconium Hydrides.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 0: 08-05-2020

Thermal Diffusivity with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Thermal Diffusivity (α) |                        | Temperature ( T ) |            | Thermal Diffusivity (α) |                 |
|-------------------|-----------|-------------------------|------------------------|-------------------|------------|-------------------------|-----------------|
| K                 | ( °F )    | cm <sup>2</sup> /s      | ( in <sup>2</sup> /s ) | K                 | ( °F )     | μm/(m-K)                | ( μin/(in-°F) ) |
| 300               | ( 80.3 )  | 0.361                   | ( 0.0560 )             | 650               | ( 710.3 )  | 0.116                   | ( 0.0180 )      |
| 325               | ( 125.3 ) | 0.316                   | ( 0.0490 )             | 675               | ( 755.3 )  | 0.110                   | ( 0.0171 )      |
| 350               | ( 170.3 ) | 0.281                   | ( 0.0436 )             | 700               | ( 800.3 )  | 0.105                   | ( 0.0163 )      |
| 375               | ( 215.3 ) | 0.253                   | ( 0.0392 )             | 725               | ( 845.3 )  | 0.100                   | ( 0.0155 )      |
| 400               | ( 260.3 ) | 0.229                   | ( 0.0355 )             | 750               | ( 890.3 )  | 0.096                   | ( 0.0149 )      |
| 425               | ( 305.3 ) | 0.210                   | ( 0.0325 )             | 775               | ( 935.3 )  | 0.092                   | ( 0.0142 )      |
| 450               | ( 350.3 ) | 0.193                   | ( 0.0299 )             | 800               | ( 980.3 )  | 0.088                   | ( 0.0136 )      |
| 475               | ( 395.3 ) | 0.179                   | ( 0.0277 )             | 825               | ( 1025.3 ) | 0.084                   | ( 0.0131 )      |
| 500               | ( 440.3 ) | 0.166                   | ( 0.0258 )             | 850               | ( 1070.3 ) | 0.081                   | ( 0.0126 )      |
| 525               | ( 485.3 ) | 0.155                   | ( 0.0241 )             | 875               | ( 1115.3 ) | 0.078                   | ( 0.0121 )      |
| 550               | ( 530.3 ) | 0.146                   | ( 0.0226 )             | 900               | ( 1160.3 ) | 0.075                   | ( 0.0117 )      |
| 575               | ( 575.3 ) | 0.137                   | ( 0.0212 )             | 925               | ( 1205.3 ) | 0.073                   | ( 0.0112 )      |
| 600               | ( 620.3 ) | 0.129                   | ( 0.0200 )             | 950               | ( 1250.3 ) | 0.070                   | ( 0.0109 )      |
| 625               | ( 665.3 ) | 0.122                   | ( 0.0190 )             | 975               | ( 1295.3 ) | 0.068                   | ( 0.0105 )      |

**Application Notes:** Data for thermal diffusivity is collected from references [6, 7] and fitted with the equation below to approximate the property trend with respect to temperature and H/Zr ratio. Data in table is for H/Zr = 2.0.

## Fit Equation:

$$\alpha(x, T) = A0 / \{ T + A1(A2 - x) + A3 \} + A4$$

$$\alpha(x, T) = \text{Thermal Diffusivity (cm}^2/\text{s)}$$

$x$  = Hydrogen/Zirconium ratio

$T$  = Temperature [K]

## Constants:

$$\text{T. Range: } 300 \leq T \leq 977$$

$$\text{H/Zr ratio (x): } 1.6 \leq X \leq 2.0$$

$$A0 = 67.9$$

$$A1 = 1620$$

$$A2 = 2.0$$

$$A3 = -118$$

$$A4 = -0.0116$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 3: 01-25-2024

Specific Heat with Temperature

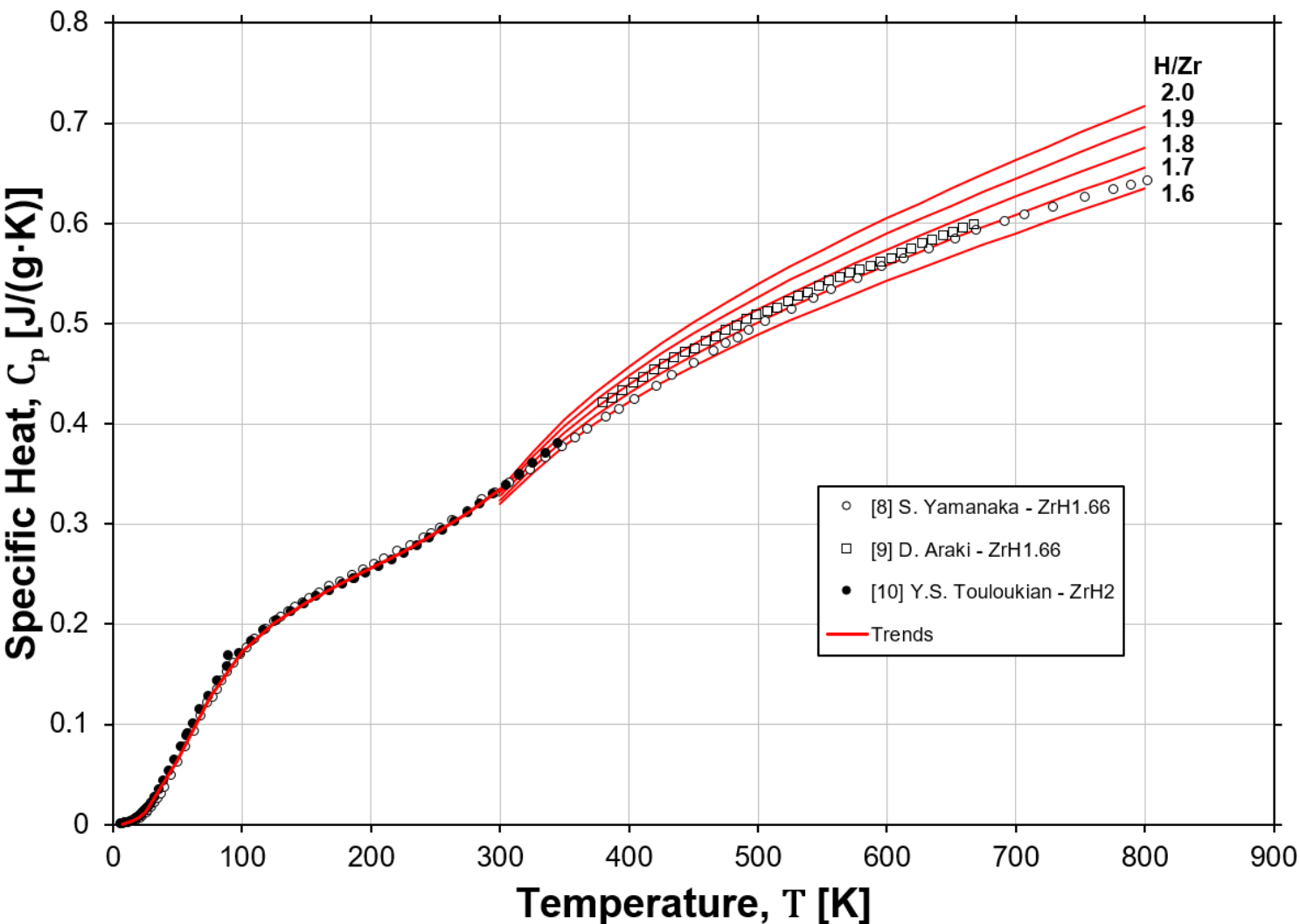


Figure 6.2.1-9: Specific Heat versus Temperature of Zirconium Hydrides. Trends above 300 K are from M. Uno [6].



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 3: 01-25-2024

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 7                 | ( -447.1 ) | 0.001                            | ( 0.000 )       | 350               | ( 170.3 )  | 0.404                            | ( 0.097 )       |
| 25                | ( -414.7 ) | 0.012                            | ( 0.003 )       | 400               | ( 260.3 )  | 0.457                            | ( 0.109 )       |
| 50                | ( -369.7 ) | 0.064                            | ( 0.015 )       | 450               | ( 350.3 )  | 0.501                            | ( 0.120 )       |
| 75                | ( -324.7 ) | 0.127                            | ( 0.030 )       | 500               | ( 440.3 )  | 0.539                            | ( 0.129 )       |
| 100               | ( -279.7 ) | 0.172                            | ( 0.041 )       | 550               | ( 530.3 )  | 0.573                            | ( 0.137 )       |
| 125               | ( -234.7 ) | 0.201                            | ( 0.048 )       | 600               | ( 620.3 )  | 0.605                            | ( 0.145 )       |
| 150               | ( -189.7 ) | 0.221                            | ( 0.053 )       | 650               | ( 710.3 )  | 0.635                            | ( 0.152 )       |
| 175               | ( -144.7 ) | 0.239                            | ( 0.057 )       | 700               | ( 800.3 )  | 0.663                            | ( 0.158 )       |
| 200               | ( -99.7 )  | 0.256                            | ( 0.061 )       | 750               | ( 890.3 )  | 0.690                            | ( 0.165 )       |
| 225               | ( -54.7 )  | 0.273                            | ( 0.065 )       | 800               | ( 980.3 )  | 0.717                            | ( 0.171 )       |
| 250               | ( -9.7 )   | 0.291                            | ( 0.070 )       | 850               | ( 1070.3 ) | 0.743                            | ( 0.178 )       |
| 275               | ( 35.3 )   | 0.311                            | ( 0.074 )       | 900               | ( 1160.3 ) | 0.768                            | ( 0.184 )       |
| 293               | ( 67.7 )   | 0.326                            | ( 0.078 )       | 950               | ( 1250.3 ) | 0.794                            | ( 0.190 )       |
| 300               | ( 80.3 )   | 0.335                            | ( 0.080 )       | 977               | ( 1298.9 ) | 0.807                            | ( 0.193 )       |

**Application Notes:** Data for specific heat is collected from references [8-10]. Trends above 300 K are from [6], while the trend below 300 K was generated using data from [8, 10]. The equations below approximate the trend with respect to temp. and H/Zr ratio. Data in table is for H/Zr = 2.0.

## Fit Equation:

For temperature range:  $7 \leq T < 300$

$$C_p(T) = \left( D0 \cdot \left( \frac{T}{1000} \right)^N \right) / \left( 1 + D1 \cdot \left( \frac{T}{1000} \right) + D2 \cdot \left( \frac{T}{1000} \right)^2 + D3 \cdot \left( \frac{T}{1000} \right)^3 \right)$$

For temperature range:  $300 \leq T \leq 977$

$$C_p[J/(mol \cdot K)] = A0 + A1 \cdot x + (B0 + B1 \cdot x)T - (C0 + C1 \cdot x)/T^2$$

$$C_p(T) = \text{Specific Heat } [J/(g \cdot K)] = C_p[J/(mol \cdot K)]/MW$$

$x$  = Hydrogen/Zirconium ratio;  $T$  = Temperature [K]

$MW$  = Molecular Weight =  $91.224 + 1.008 \cdot x$

## Constants:

| T. Range:       | $7 \leq T < 300$       | $300 \leq T \leq 977$ |
|-----------------|------------------------|-----------------------|
| H/Zr ratio (x): | $1.66 \leq x \leq 2.0$ | $1.6 \leq x \leq 2.0$ |
| N =             | 2.325                  | A0 = 25.02            |
| D0 =            | 47.25                  | A1 = 4.746            |
| D1 =            | -16.25                 | B0 = .003103          |
| D2 =            | 219                    | B1 = .02008           |
| D3 =            | -266                   | C0 = 1.943 E05        |
|                 |                        | C1 = 6.358 E05        |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 0: 08-05-2020

Electrical Resistivity with H/Zr Ratio

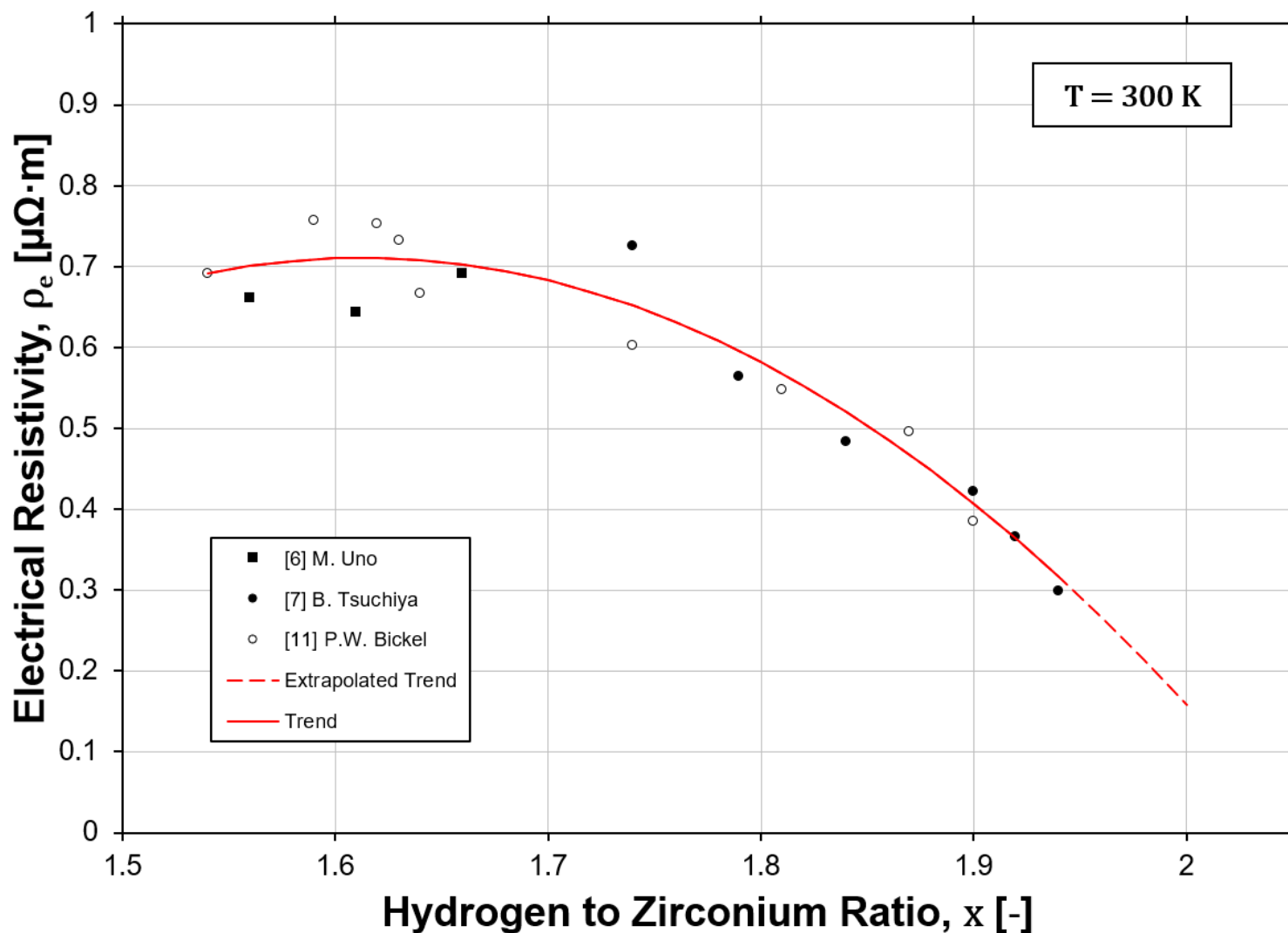


Figure 6.2.1-10: Electrical Resistivity versus Hydrogen to Zirconium ratio ( $x$ ).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 0: 08-05-2020

Electrical Resistivity with H/Zr Ratio

100% Theoretical Density

| H/Zr Ratio ( x ) | Electrical Resistivity ( ρ <sub>e</sub> ) |           | H/Zr Ratio ( x ) | Electrical Resistivity ( ρ <sub>e</sub> ) |           |
|------------------|-------------------------------------------|-----------|------------------|-------------------------------------------|-----------|
|                  | μΩ-m                                      | ( μΩ-in ) |                  | μΩ-m                                      | ( μΩ-in ) |
| 1.54             | 0.691                                     | ( 27.21 ) | 1.76             | 0.631                                     | ( 24.84 ) |
| 1.56             | 0.700                                     | ( 27.58 ) | 1.78             | 0.608                                     | ( 23.93 ) |
| 1.58             | 0.707                                     | ( 27.83 ) | 1.80             | 0.582                                     | ( 22.90 ) |
| 1.60             | 0.710                                     | ( 27.96 ) | 1.82             | 0.553                                     | ( 21.75 ) |
| 1.62             | 0.711                                     | ( 27.98 ) | 1.84             | 0.521                                     | ( 20.49 ) |
| 1.64             | 0.708                                     | ( 27.88 ) | 1.86             | 0.486                                     | ( 19.12 ) |
| 1.66             | 0.703                                     | ( 27.66 ) | 1.88             | 0.448                                     | ( 17.62 ) |
| 1.68             | 0.694                                     | ( 27.33 ) | 1.90             | 0.407                                     | ( 16.01 ) |
| 1.70             | 0.683                                     | ( 26.88 ) | 1.92             | 0.363                                     | ( 14.29 ) |
| 1.72             | 0.668                                     | ( 26.32 ) | 1.94             | 0.316                                     | ( 12.44 ) |
| 1.74             | 0.651                                     | ( 25.64 ) |                  |                                           |           |

**Application Notes:** Data for electrical resistivity is collected from references [6, 7, 11] and fitted with the equation below to approximate the property trend with respect to atomic ratio.

## Fit Equation:

$$\rho_e(x) = A0 + A1 \cdot x + A2 \cdot x^2$$

$$\rho_e(T) = \text{Electrical Resistivity } [\mu\Omega \cdot m]$$

$x$  = Hydrogen/Zirconium ratio

## Constants:

H/Zr Ratio  
Range (x):  $1.54 \leq x \leq 1.94$

A0 = -8.886

A1 = 11.9

A2 = -3.689



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

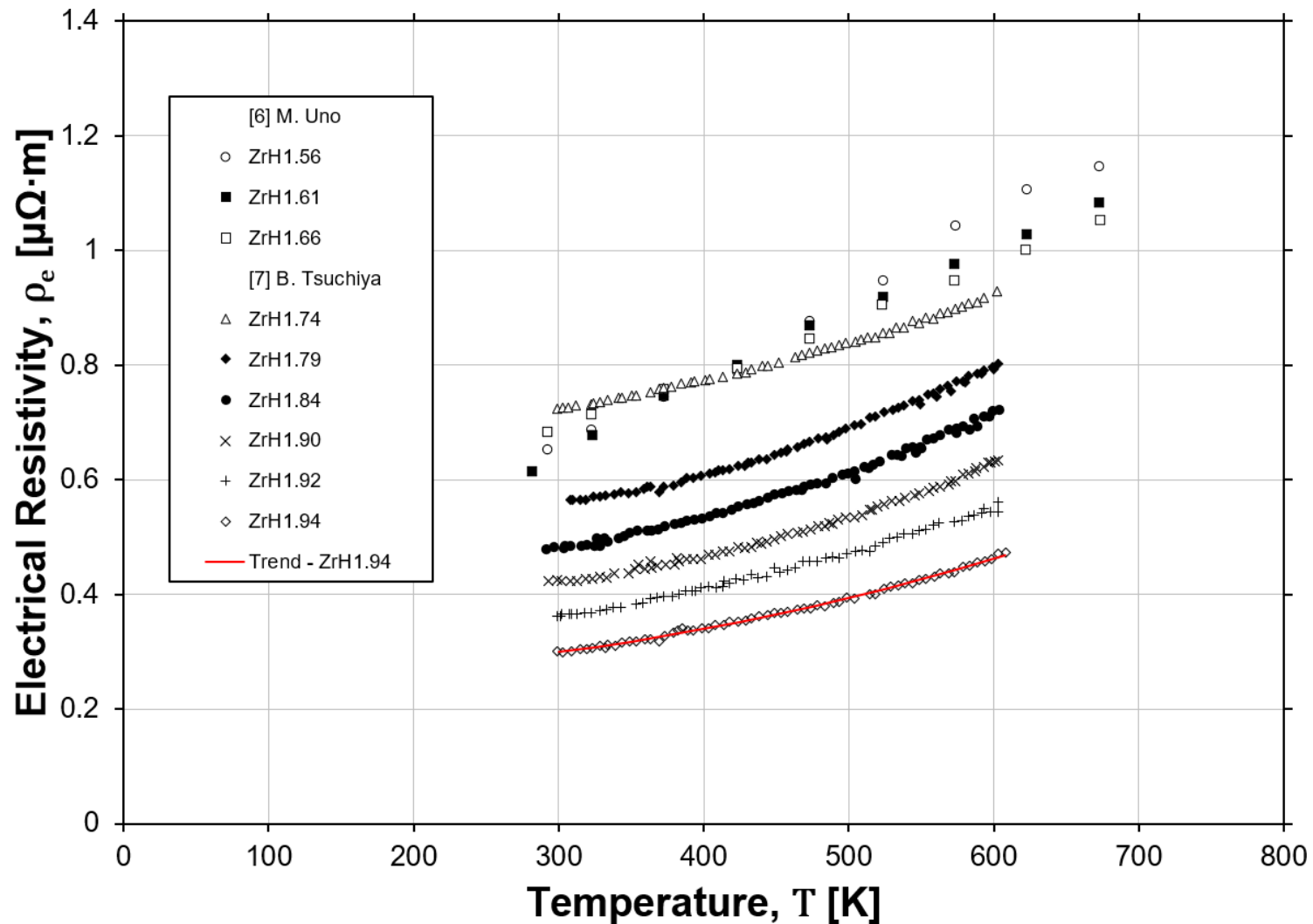


Figure 6.2.1-11: Electrical Resistivity versus Temperature of Zirconium Hydrides.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Electrical Resistivity ( ρ <sub>e</sub> ) |           | Temperature ( T ) |           | Electrical Resistivity ( ρ <sub>e</sub> ) |           |
|-------------------|-----------|-------------------------------------------|-----------|-------------------|-----------|-------------------------------------------|-----------|
| K                 | ( °F )    | μΩ-m                                      | ( μΩ-in ) | K                 | ( °F )    | μΩ-m                                      | ( μΩ-in ) |
| 300               | ( 80.3 )  | 0.299                                     | ( 11.77 ) | 480               | ( 404.3 ) | 0.382                                     | ( 15.04 ) |
| 320               | ( 116.3 ) | 0.306                                     | ( 12.05 ) | 500               | ( 440.3 ) | 0.394                                     | ( 15.51 ) |
| 340               | ( 152.3 ) | 0.314                                     | ( 12.35 ) | 520               | ( 476.3 ) | 0.407                                     | ( 16.01 ) |
| 360               | ( 188.3 ) | 0.322                                     | ( 12.66 ) | 540               | ( 512.3 ) | 0.420                                     | ( 16.53 ) |
| 380               | ( 224.3 ) | 0.330                                     | ( 13.01 ) | 560               | ( 548.3 ) | 0.433                                     | ( 17.07 ) |
| 400               | ( 260.3 ) | 0.340                                     | ( 13.37 ) | 580               | ( 584.3 ) | 0.448                                     | ( 17.63 ) |
| 420               | ( 296.3 ) | 0.349                                     | ( 13.75 ) | 600               | ( 620.3 ) | 0.463                                     | ( 18.21 ) |
| 440               | ( 332.3 ) | 0.360                                     | ( 14.16 ) | 608               | ( 634.7 ) | 0.469                                     | ( 18.45 ) |
| 460               | ( 368.3 ) | 0.371                                     | ( 14.59 ) |                   |           |                                           |           |

**Application Notes:** Data for electrical resistivity is collected from references [6, 7]. Data from [7] is fitted with the equation below to approximate the property trend with respect to temperature for H/Zr = 1.94.

## Fit Equation:

$x = \text{Hydrogen/Zirconium ratio} = 1.94$

$$\rho_e(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$\rho_e(T) = \text{Electrical Resistivity } [\mu\Omega \cdot m]$

$T = \text{Temperature } [K]$

## Constants:

T. Range:  $298 \leq T < 608$

A0 = 0.2613

A1 = -0.08387

A2 = 0.6988



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 0: 08-05-2020

Young's Modulus with H/Zr Ratio

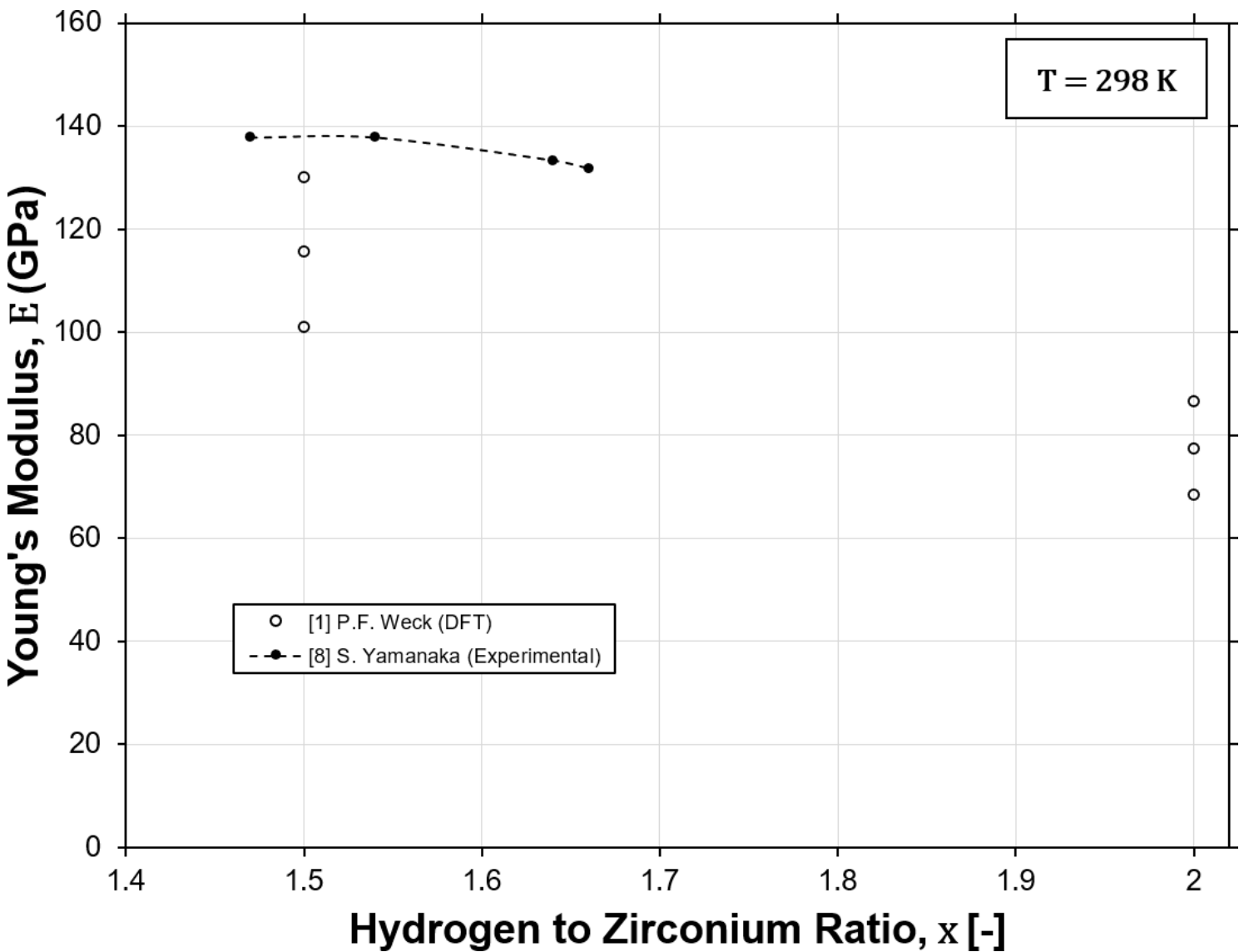


Figure 6.2.1-12: Young's Modulus versus Hydrogen to Zirconium ratio ( $x$ ).



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 0: 08-05-2020

Shear Modulus with H/Zr Ratio

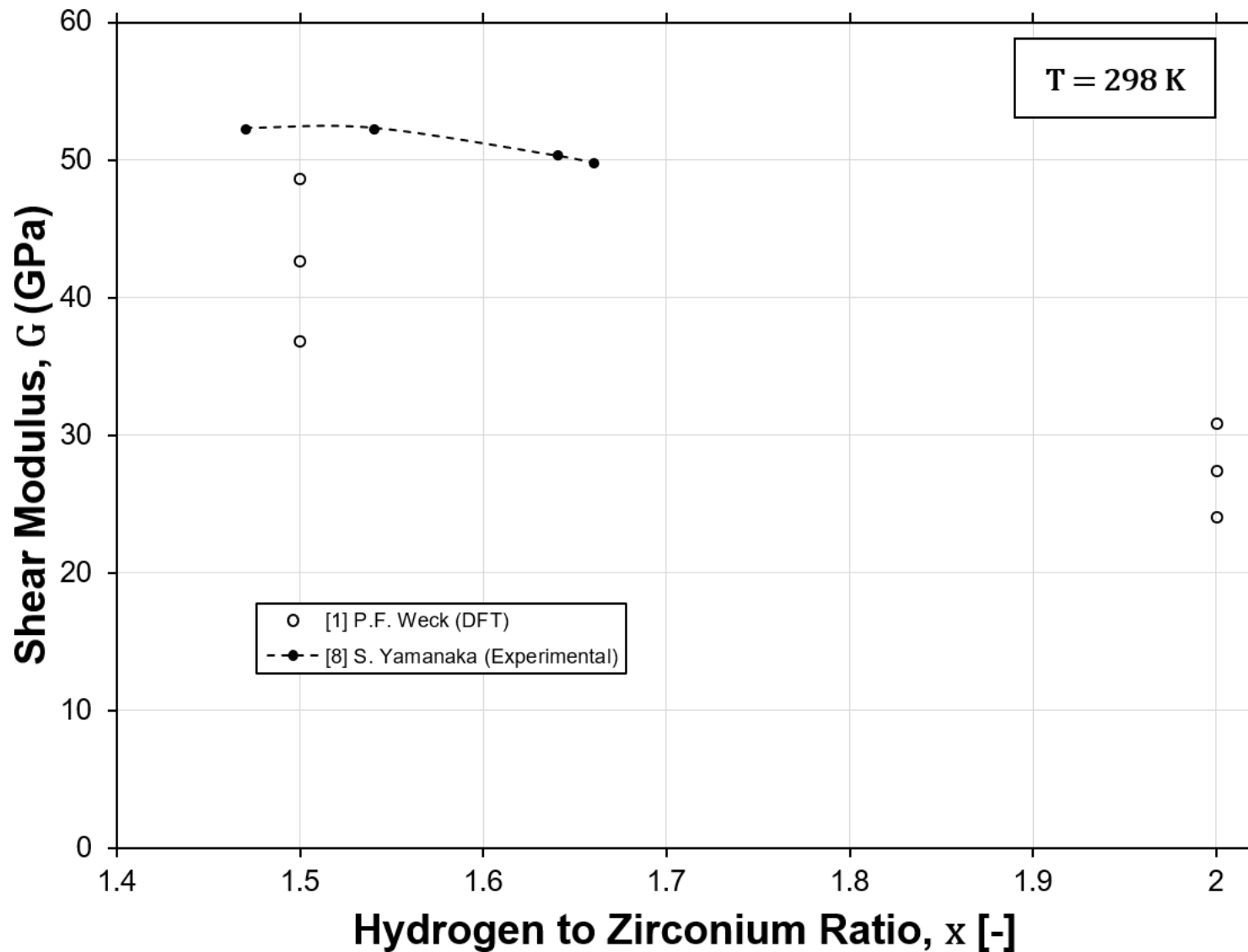


Figure 6.2.1-13: Shear Modulus for Hydrogen to Zirconium ratio ( $x$ ).



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride ( $\text{ZrH}_x$ )

Revision 0: 08-05-2020

Poisson's Ratio with H/Zr Ratio

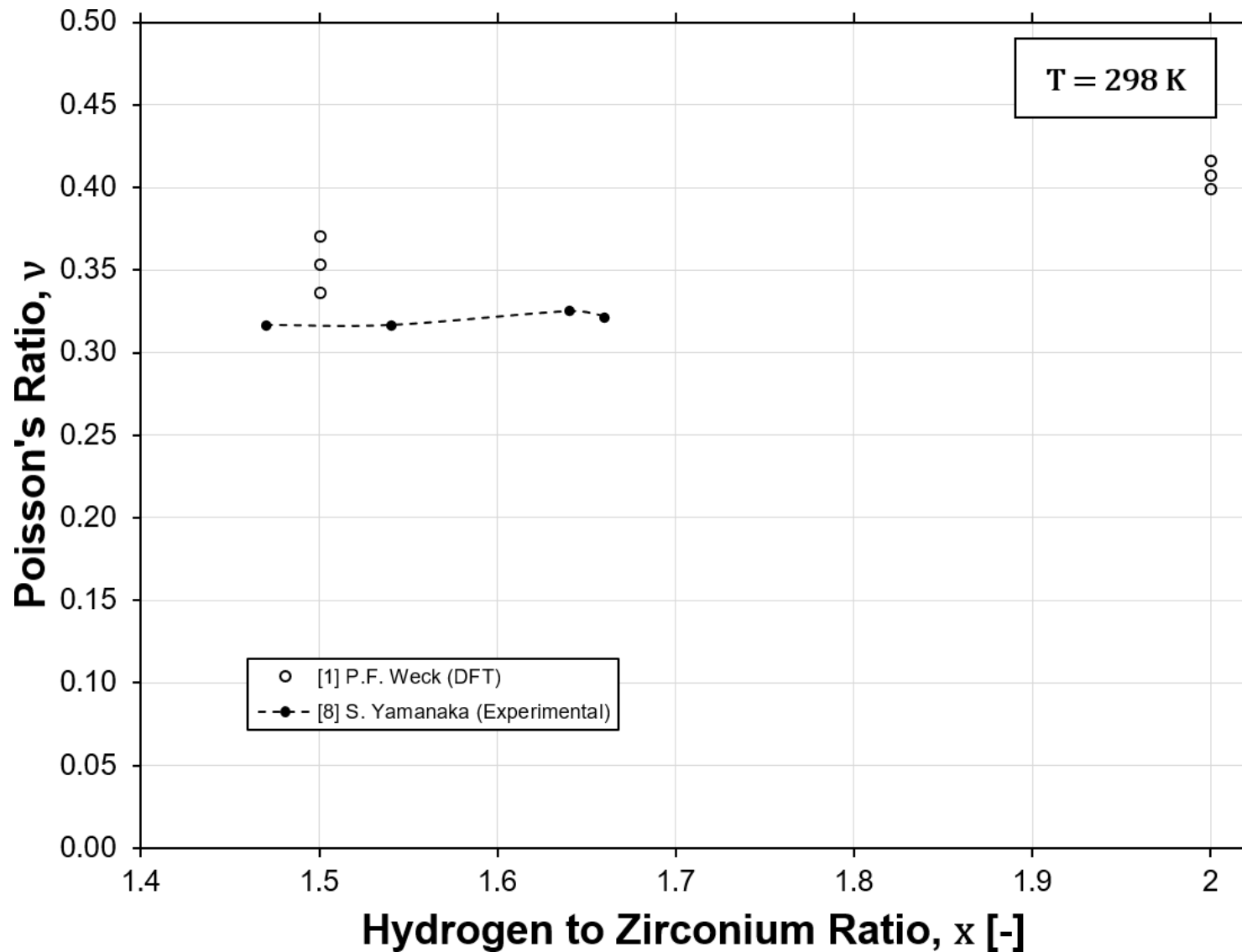


Figure 6.2.1-14: Poisson's Ratio for Hydrogen to Zirconium ratio ( $x$ ).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

Revision 3: 01-25-2024

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 293                 | 5587                                  | -                                                       | 67.91                                       | 0.00                                               | -                                         | 0.326                                          | -                                  | -                                | -                           |
| 300                 | 5586                                  | 0.299                                                   | 67.57                                       | 0.01                                               | 8.34                                      | 0.335                                          | 77.40                              | 27.50                            | 0.41                        |
| 320                 | 5583                                  | 0.306                                                   | 66.15                                       | 0.02                                               | 8.45                                      | 0.365                                          | -                                  | -                                | -                           |
| 340                 | 5580                                  | 0.314                                                   | 64.35                                       | 0.04                                               | 8.56                                      | 0.392                                          | -                                  | -                                | -                           |
| 360                 | 5577                                  | 0.322                                                   | 62.38                                       | 0.06                                               | 8.66                                      | 0.416                                          | -                                  | -                                | -                           |
| 380                 | 5574                                  | 0.330                                                   | 60.38                                       | 0.08                                               | 8.77                                      | 0.438                                          | -                                  | -                                | -                           |
| 400                 | 5571                                  | 0.340                                                   | 58.41                                       | 0.10                                               | 8.88                                      | 0.457                                          | -                                  | -                                | -                           |
| 420                 | 5567                                  | 0.349                                                   | 56.50                                       | 0.11                                               | 8.98                                      | 0.476                                          | -                                  | -                                | -                           |
| 440                 | 5564                                  | 0.360                                                   | 54.67                                       | 0.13                                               | 9.09                                      | 0.493                                          | -                                  | -                                | -                           |
| 460                 | 5561                                  | 0.371                                                   | 52.94                                       | 0.15                                               | 9.19                                      | 0.509                                          | -                                  | -                                | -                           |
| 480                 | 5558                                  | 0.382                                                   | 51.31                                       | 0.17                                               | 9.30                                      | 0.525                                          | -                                  | -                                | -                           |
| 500                 | 5554                                  | 0.394                                                   | 49.76                                       | 0.19                                               | 9.41                                      | 0.539                                          | -                                  | -                                | -                           |
| 520                 | 5551                                  | 0.407                                                   | 48.31                                       | 0.22                                               | 9.51                                      | 0.553                                          | -                                  | -                                | -                           |
| 540                 | 5547                                  | 0.420                                                   | 46.94                                       | 0.24                                               | 9.62                                      | 0.567                                          | -                                  | -                                | -                           |
| 560                 | 5543                                  | 0.433                                                   | 45.65                                       | 0.26                                               | 9.72                                      | 0.580                                          | -                                  | -                                | -                           |
| 580                 | 5540                                  | 0.448                                                   | 44.43                                       | 0.28                                               | 9.83                                      | 0.593                                          | -                                  | -                                | -                           |
| 600                 | 5536                                  | 0.463                                                   | 43.29                                       | 0.30                                               | 9.93                                      | 0.605                                          | -                                  | -                                | -                           |
| 650                 | 5526                                  | -                                                       | 40.69                                       | 0.36                                               | 10.19                                     | 0.635                                          | -                                  | -                                | -                           |
| 700                 | 5516                                  | -                                                       | 38.41                                       | 0.43                                               | 10.45                                     | 0.663                                          | -                                  | -                                | -                           |
| 750                 | 5505                                  | -                                                       | 36.42                                       | 0.49                                               | 10.71                                     | 0.690                                          | -                                  | -                                | -                           |
| 800                 | 5494                                  | -                                                       | 34.64                                       | 0.56                                               | 10.97                                     | 0.717                                          | -                                  | -                                | -                           |
| 850                 | 5483                                  | -                                                       | 33.05                                       | 0.63                                               | 11.23                                     | 0.743                                          | -                                  | -                                | -                           |
| 900                 | 5471                                  | -                                                       | 31.62                                       | 0.70                                               | 11.49                                     | 0.768                                          | -                                  | -                                | -                           |
| 950                 | 5459                                  | -                                                       | 30.33                                       | 0.77                                               | 11.74                                     | 0.794                                          | -                                  | -                                | -                           |
| 977                 | 5453                                  | -                                                       | 29.68                                       | 0.81                                               | 11.88                                     | 0.807                                          | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.1 Zirconium Hydride (ZrH<sub>x</sub>)

**Revision 3: 01-25-2024**

**References**

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## **6 Nuclear Materials**

### **6.2 Neutron Control Materials**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2.1: 08-25-2023

General

### Room Temperature Properties

|                                           |        |
|-------------------------------------------|--------|
| Atomic Mass, [amu]                        | 9.012  |
| Theoretical Density, [kg/m <sup>3</sup> ] | 1,848  |
| Melting Point, [K]                        | 1560   |
| Boiling Point, [K]                        | 2743   |
| Specific Heat, [J/(g-K)]                  | 1.807  |
| Heat of Fusion, [kJ/mol]                  | 7.95   |
| Heat of Vaporization, [kJ/mol]            | 297    |
| Thermal Conductivity, [W/(m-K)]           | 185.4  |
| Linear expansion coefficient, [μm/(m-K)]  | 11.1   |
| Electrical resistivity, [μΩ-m]            | 0.0458 |
| Young's Modulus, [GPa]                    | 288.5  |
| Shear Modulus, [GPa]                      | 136.6  |
| Poisson's Ratio, [-]                      | 0.056  |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Density with Temperature

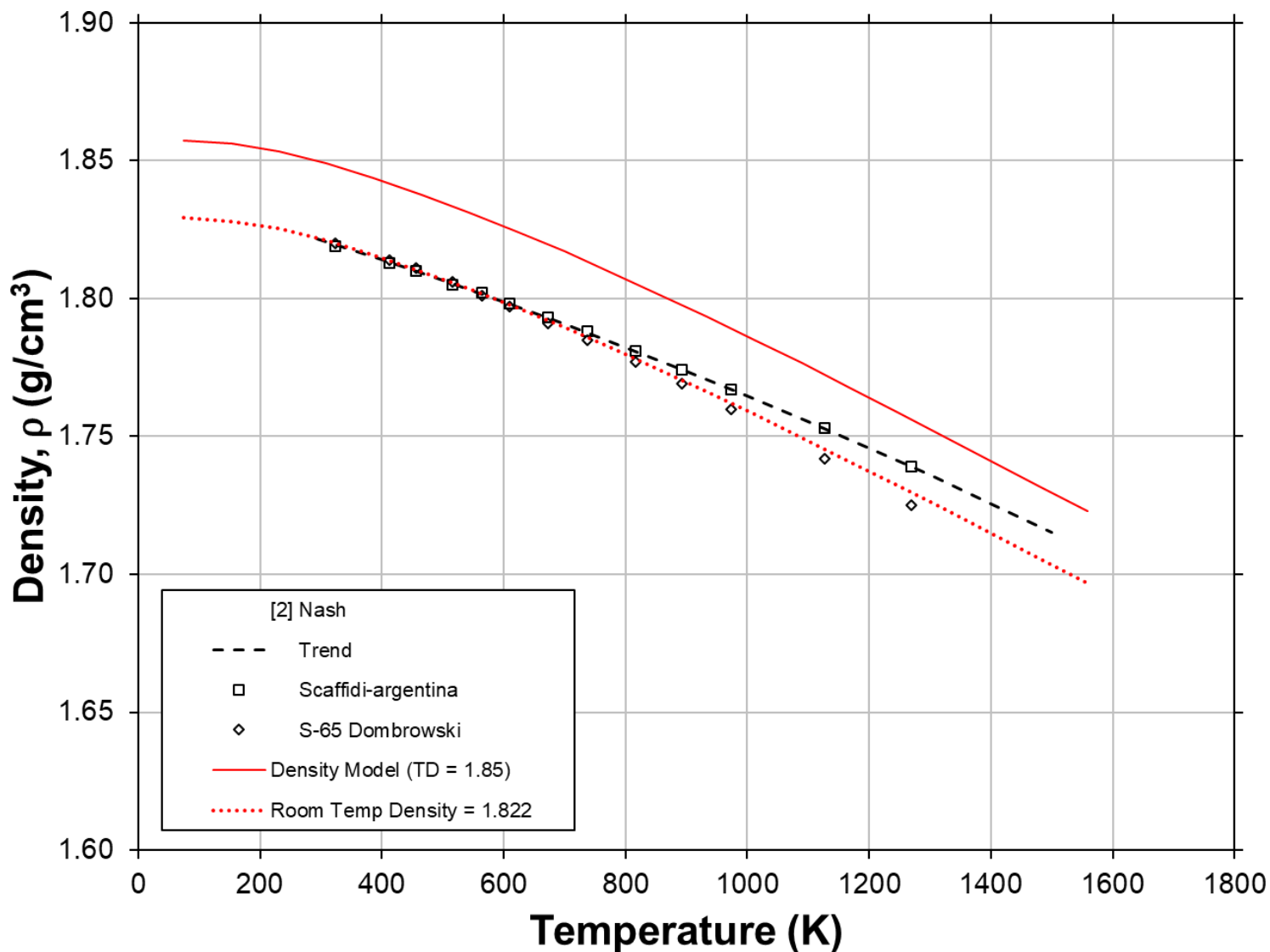


Figure 6.2.2-1: Density versus Temperature for Be. Trend calculated from fitted Thermal Expansion data and compared to trend from Nash (2005).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Density with Temperature

## 100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 75                | ( -324.7 ) | 1857              | ( 116.0 )              | 800               | ( 980.3 )  | 1807              | ( 112.8 )              |
| 100               | ( -279.7 ) | 1857              | ( 115.9 )              | 850               | ( 1070.3 ) | 1802              | ( 112.5 )              |
| 150               | ( -189.7 ) | 1856              | ( 115.9 )              | 900               | ( 1160.3 ) | 1797              | ( 112.2 )              |
| 200               | ( -99.7 )  | 1855              | ( 115.8 )              | 950               | ( 1250.3 ) | 1792              | ( 111.9 )              |
| 250               | ( -9.7 )   | 1853              | ( 115.7 )              | 1000              | ( 1340.3 ) | 1786              | ( 111.5 )              |
| 300               | ( 80.3 )   | 1850              | ( 115.5 )              | 1050              | ( 1430.3 ) | 1781              | ( 111.2 )              |
| 350               | ( 170.3 )  | 1846              | ( 115.3 )              | 1100              | ( 1520.3 ) | 1775              | ( 110.8 )              |
| 400               | ( 260.3 )  | 1843              | ( 115.0 )              | 1150              | ( 1610.3 ) | 1770              | ( 110.5 )              |
| 450               | ( 350.3 )  | 1839              | ( 114.8 )              | 1200              | ( 1700.3 ) | 1764              | ( 110.1 )              |
| 500               | ( 440.3 )  | 1835              | ( 114.5 )              | 1250              | ( 1790.3 ) | 1758              | ( 109.8 )              |
| 550               | ( 530.3 )  | 1830              | ( 114.3 )              | 1300              | ( 1880.3 ) | 1753              | ( 109.4 )              |
| 600               | ( 620.3 )  | 1826              | ( 114.0 )              | 1350              | ( 1970.3 ) | 1747              | ( 109.1 )              |
| 650               | ( 710.3 )  | 1822              | ( 113.7 )              | 1400              | ( 2060.3 ) | 1741              | ( 108.7 )              |
| 700               | ( 800.3 )  | 1817              | ( 113.4 )              | 1500              | ( 2240.3 ) | 1730              | ( 108.0 )              |
| 750               | ( 890.3 )  | 1812              | ( 113.1 )              | 1558              | ( 2344.7 ) | 1723              | ( 107.6 )              |

**Application Notes:** Density Model and Room Temp Density are calculated as functions of thermal expansion utilizing theoretical density of beryllium, and measured room temperature density of beryllium, respectively. The equation below is used to generate these trends. Experimental data for density is collected from references [2-4].

### Density Calculation:

$$\rho(T) = \rho_{RT}(1 - P)/(1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 1,848 \text{ [kg/m}^3\text{]}$$

$P$  = Fractional Porosity

$T$  = Temperature [K]

**Temperature Range:**  $75 \leq T \leq 1558$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

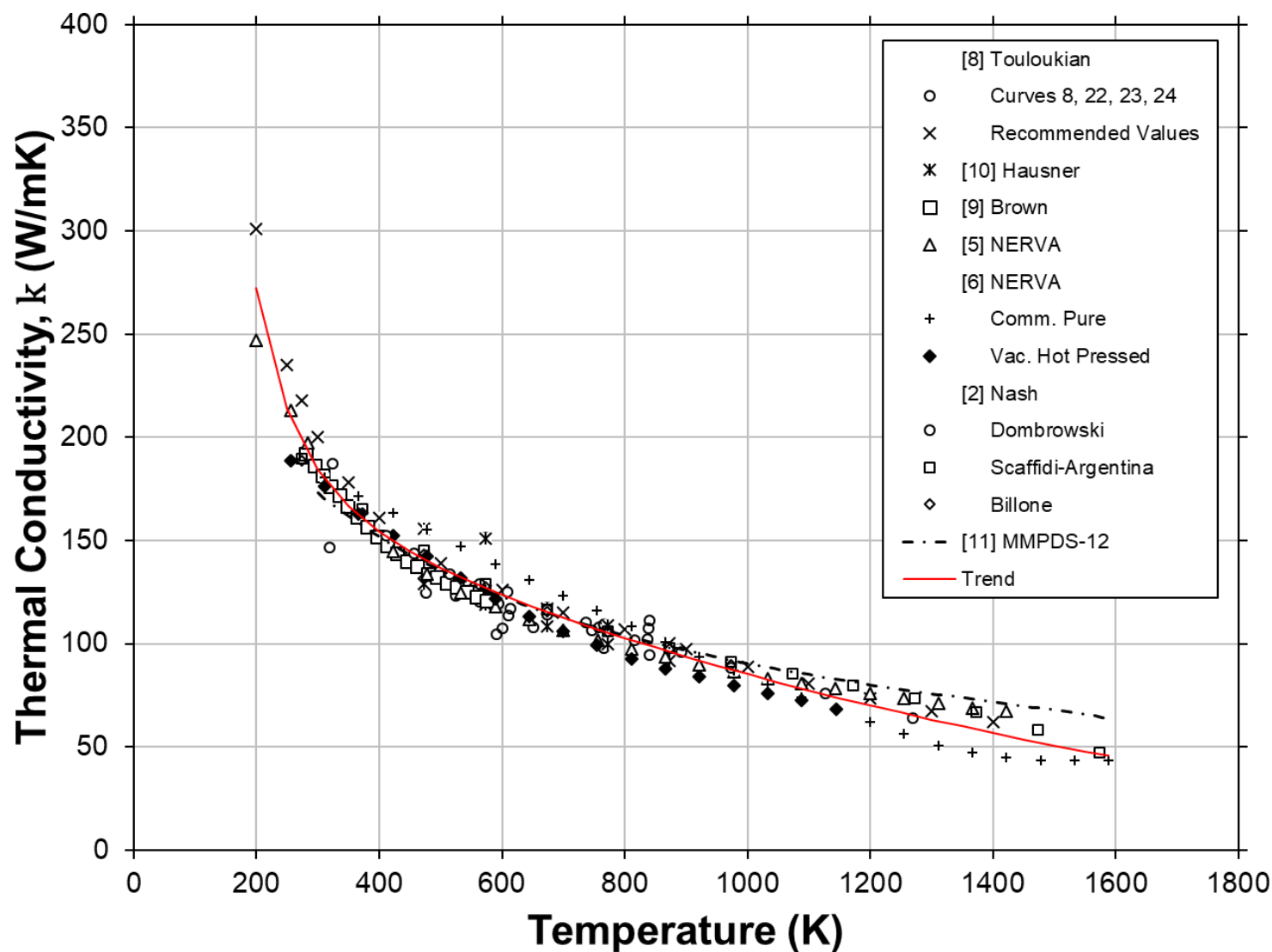


Figure 6.2.2-2: Thermal Conductivity versus Temperature of Beryllium. Displaying fitted trend of the data with comparison to MMPDS-12, S-Basis trend.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 200               | ( -99.7 )  | 272.06                     | ( 1887.62 )                         | 900               | ( 1160.3 ) | 93.78                      | ( 650.63 )                          |
| 225               | ( -54.7 )  | 237.49                     | ( 1647.72 )                         | 950               | ( 1250.3 ) | 89.50                      | ( 620.94 )                          |
| 250               | ( -9.7 )   | 214.01                     | ( 1484.82 )                         | 1000              | ( 1340.3 ) | 85.36                      | ( 592.25 )                          |
| 300               | ( 80.3 )   | 184.53                     | ( 1280.31 )                         | 1050              | ( 1430.3 ) | 81.36                      | ( 564.50 )                          |
| 350               | ( 170.3 )  | 166.63                     | ( 1156.09 )                         | 1100              | ( 1520.3 ) | 77.49                      | ( 537.63 )                          |
| 400               | ( 260.3 )  | 154.18                     | ( 1069.72 )                         | 1150              | ( 1610.3 ) | 73.74                      | ( 511.61 )                          |
| 450               | ( 350.3 )  | 144.62                     | ( 1003.36 )                         | 1200              | ( 1700.3 ) | 70.11                      | ( 486.42 )                          |
| 500               | ( 440.3 )  | 136.71                     | ( 948.54 )                          | 1250              | ( 1790.3 ) | 66.59                      | ( 462.04 )                          |
| 550               | ( 530.3 )  | 129.85                     | ( 900.91 )                          | 1300              | ( 1880.3 ) | 63.19                      | ( 438.44 )                          |
| 600               | ( 620.3 )  | 123.68                     | ( 858.08 )                          | 1350              | ( 1970.3 ) | 59.90                      | ( 415.61 )                          |
| 650               | ( 710.3 )  | 117.99                     | ( 818.62 )                          | 1400              | ( 2060.3 ) | 56.72                      | ( 393.54 )                          |
| 700               | ( 800.3 )  | 112.67                     | ( 781.68 )                          | 1450              | ( 2150.3 ) | 53.65                      | ( 372.23 )                          |
| 750               | ( 890.3 )  | 107.63                     | ( 746.72 )                          | 1500              | ( 2240.3 ) | 50.69                      | ( 351.67 )                          |
| 800               | ( 980.3 )  | 102.82                     | ( 713.38 )                          | 1550              | ( 2330.3 ) | 47.83                      | ( 331.85 )                          |
| 850               | ( 1070.3 ) | 98.21                      | ( 681.40 )                          | 1589              | ( 2400.5 ) | 45.68                      | ( 316.90 )                          |

**Application Notes:** Data for thermal conductivity is collected from references [2, 5-10] and fitted with the equation below to approximate property trend with respect to temperature. Fitted trend is compared to trend from reference [11].

## Fit Equation:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + AN \cdot \left(\frac{T}{1000}\right)^N$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $200 \leq T \leq 1589$

A0 = 182.6

A1 = -118.9

A2 = 20.47

AN = 1.192

N = -2.825



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

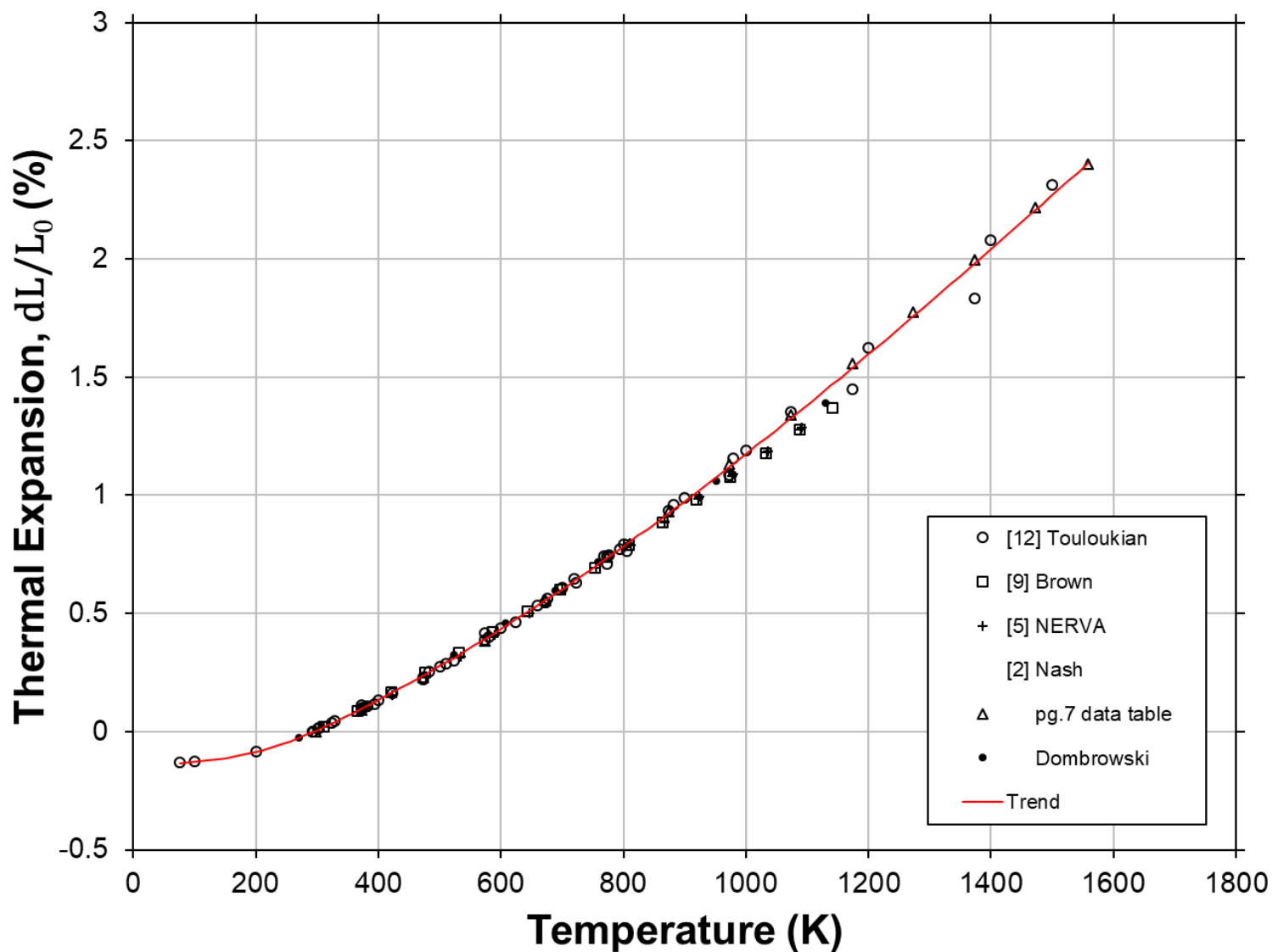


Figure 6.2.2-3: Thermal Expansion versus Temperature of Beryllium, displaying fitted trend and collected data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 75                | ( -324.7 ) | -0.132                                  | 800               | ( 980.3 )  | 0.784                                   |
| 100               | ( -279.7 ) | -0.127                                  | 850               | ( 1070.3 ) | 0.878                                   |
| 150               | ( -189.7 ) | -0.112                                  | 900               | ( 1160.3 ) | 0.975                                   |
| 200               | ( -99.7 )  | -0.085                                  | 950               | ( 1250.3 ) | 1.074                                   |
| 250               | ( -9.7 )   | -0.046                                  | 1000              | ( 1340.3 ) | 1.175                                   |
| 300               | ( 80.3 )   | 0.006                                   | 1050              | ( 1430.3 ) | 1.278                                   |
| 350               | ( 170.3 )  | 0.068                                   | 1100              | ( 1520.3 ) | 1.382                                   |
| 400               | ( 260.3 )  | 0.134                                   | 1150              | ( 1610.3 ) | 1.489                                   |
| 450               | ( 350.3 )  | 0.204                                   | 1200              | ( 1700.3 ) | 1.597                                   |
| 500               | ( 440.3 )  | 0.277                                   | 1250              | ( 1790.3 ) | 1.707                                   |
| 550               | ( 530.3 )  | 0.354                                   | 1300              | ( 1880.3 ) | 1.817                                   |
| 600               | ( 620.3 )  | 0.434                                   | 1350              | ( 1970.3 ) | 1.929                                   |
| 650               | ( 710.3 )  | 0.517                                   | 1400              | ( 2060.3 ) | 2.042                                   |
| 700               | ( 800.3 )  | 0.603                                   | 1500              | ( 2240.3 ) | 2.270                                   |
| 750               | ( 890.3 )  | 0.692                                   | 1558              | ( 2344.7 ) | 2.404                                   |

**Application Notes:** Data for thermal expansion is collected from references [2, 5, 9, 12] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

| T Range [K]: | <u>75 ≤ T ≤ 293</u> | <u>293 &lt; T ≤ 1558</u> |
|--------------|---------------------|--------------------------|
| A0 =         | -0.1385             | -0.2791                  |
| A1 =         | 0.02396             | 0.6761                   |
| A2 =         | 0.5172              | 0.9682                   |
| A3 =         | 3.451               | -0.1906                  |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

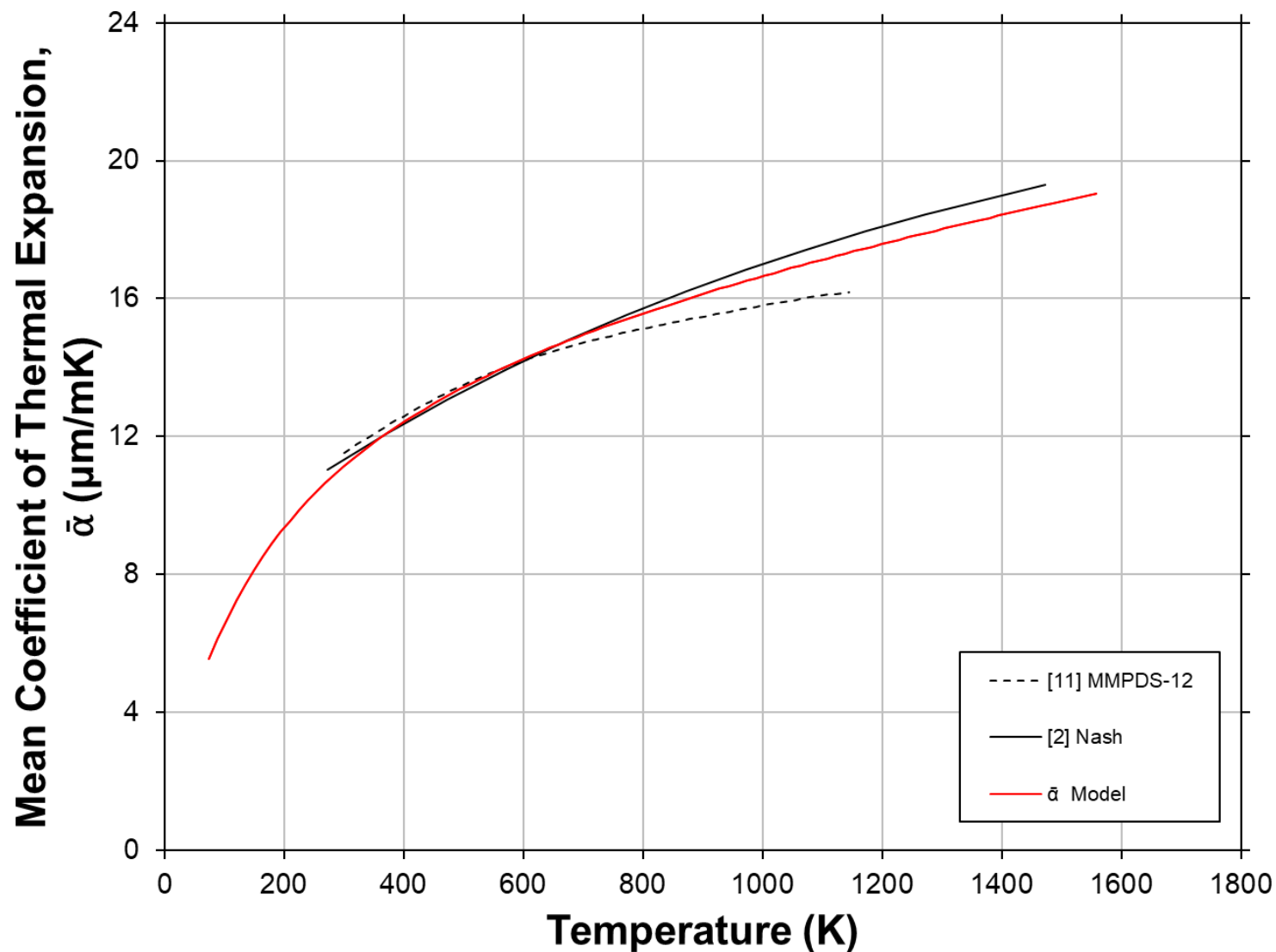


Figure 6.2.2-4: Mean Coefficient of Thermal Expansion versus Temperature of Beryllium. Calculated from fitted trend of the Thermal Expansion data and compared against trends from MMPDS-12 and Nash (2005).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2: 04-26-2023

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 75                | ( -324.7 ) | 5.535                                 | ( 3.075 )                                           | 800               | ( 980.3 )  | 15.572                                | ( 8.651 )                                           |
| 100               | ( -279.7 ) | 6.530                                 | ( 3.628 )                                           | 850               | ( 1070.3 ) | 15.859                                | ( 8.810 )                                           |
| 150               | ( -189.7 ) | 8.119                                 | ( 4.510 )                                           | 900               | ( 1160.3 ) | 16.133                                | ( 8.963 )                                           |
| 200               | ( -99.7 )  | 9.342                                 | ( 5.190 )                                           | 950               | ( 1250.3 ) | 16.396                                | ( 9.109 )                                           |
| 250               | ( -9.7 )   | 10.323                                | ( 5.735 )                                           | 1000              | ( 1340.3 ) | 16.649                                | ( 9.249 )                                           |
| 300               | ( 80.3 )   | 11.135                                | ( 6.186 )                                           | 1050              | ( 1430.3 ) | 16.893                                | ( 9.385 )                                           |
| 350               | ( 170.3 )  | 11.824                                | ( 6.569 )                                           | 1100              | ( 1520.3 ) | 17.130                                | ( 9.517 )                                           |
| 400               | ( 260.3 )  | 12.423                                | ( 6.902 )                                           | 1150              | ( 1610.3 ) | 17.360                                | ( 9.644 )                                           |
| 450               | ( 350.3 )  | 12.951                                | ( 7.195 )                                           | 1200              | ( 1700.3 ) | 17.584                                | ( 9.769 )                                           |
| 500               | ( 440.3 )  | 13.424                                | ( 7.458 )                                           | 1250              | ( 1790.3 ) | 17.803                                | ( 9.890 )                                           |
| 550               | ( 530.3 )  | 13.853                                | ( 7.696 )                                           | 1300              | ( 1880.3 ) | 18.017                                | ( 10.009 )                                          |
| 600               | ( 620.3 )  | 14.247                                | ( 7.915 )                                           | 1350              | ( 1970.3 ) | 18.226                                | ( 10.126 )                                          |
| 650               | ( 710.3 )  | 14.611                                | ( 8.117 )                                           | 1400              | ( 2060.3 ) | 18.431                                | ( 10.240 )                                          |
| 700               | ( 800.3 )  | 14.951                                | ( 8.306 )                                           | 1500              | ( 2240.3 ) | 18.832                                | ( 10.462 )                                          |
| 750               | ( 890.3 )  | 15.270                                | ( 8.483 )                                           | 1558              | ( 2344.7 ) | 19.058                                | ( 10.588 )                                          |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated as a function of thermal expansion and fitted with the equation below to approximate property trend with respect to temperature. Calculated trend is compared against references [2, 11].

## Fit Equation:

$$\bar{\alpha}(T) = \left[ A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right] / \left[ A_0 + \left( \frac{T}{1000} \right) \right]$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]: 75 ≤ T ≤ 1558

A0 = 0.2195

A1 = 16.75

A2 = 2.919

A<sub>0</sub> = 0.1946



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Specific Heat with Temperature

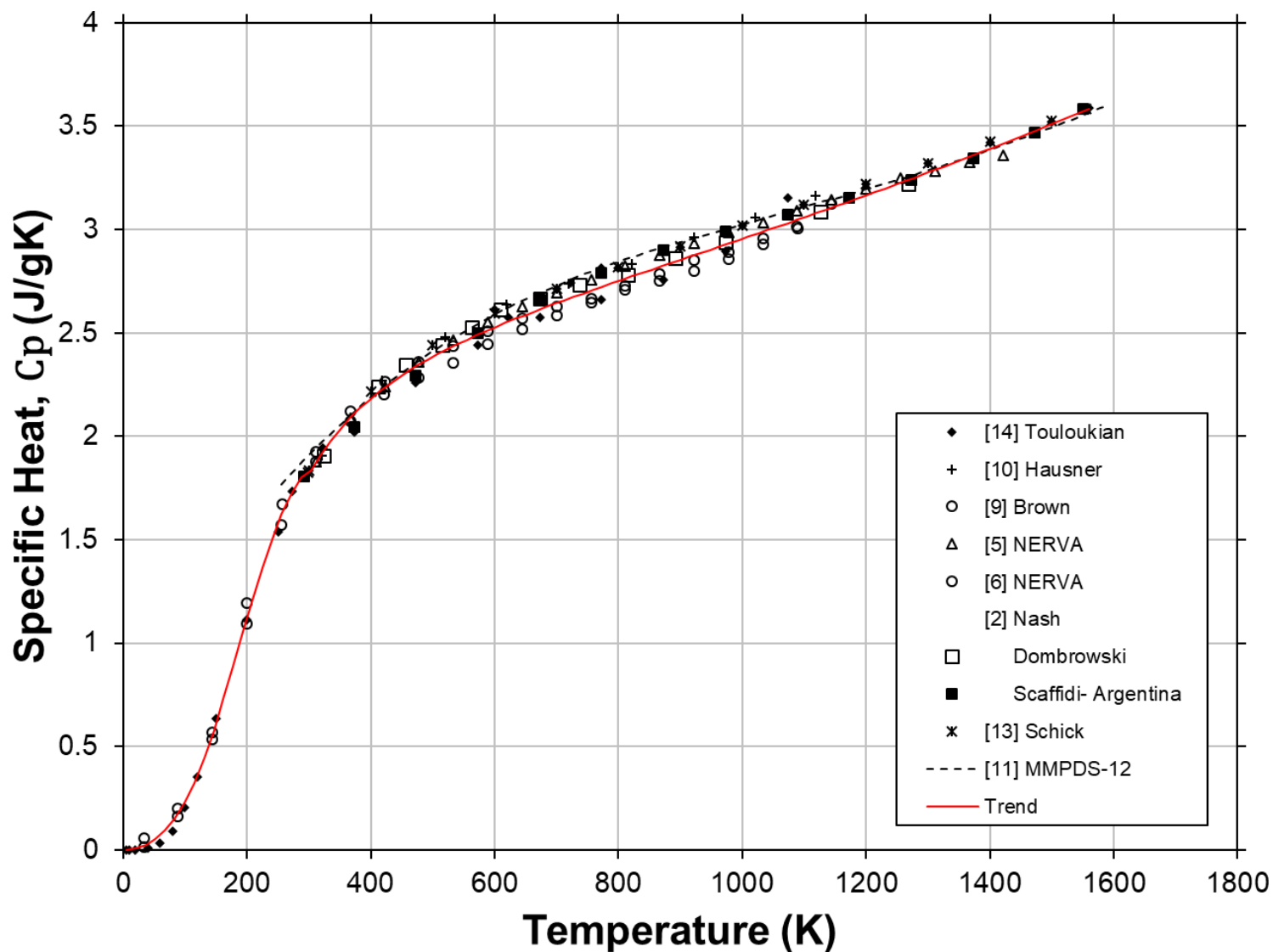


Figure 6.2.2-5: Specific Heat versus Temperature of Be fitted trend and data with comparison to MMPDS-12.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 25                | ( -414.7 ) | 0.010                            | ( 0.002 )       | 650               | ( 710.3 )  | 2.588                            | ( 0.618 )       |
| 50                | ( -369.7 ) | 0.046                            | ( 0.011 )       | 700               | ( 800.3 )  | 2.644                            | ( 0.632 )       |
| 100               | ( -279.7 ) | 0.233                            | ( 0.056 )       | 750               | ( 890.3 )  | 2.698                            | ( 0.645 )       |
| 150               | ( -189.7 ) | 0.613                            | ( 0.146 )       | 800               | ( 980.3 )  | 2.751                            | ( 0.657 )       |
| 200               | ( -99.7 )  | 1.122                            | ( 0.268 )       | 850               | ( 1070.3 ) | 2.802                            | ( 0.670 )       |
| 250               | ( -9.7 )   | 1.577                            | ( 0.377 )       | 900               | ( 1160.3 ) | 2.853                            | ( 0.682 )       |
| 300               | ( 80.3 )   | 1.818                            | ( 0.434 )       | 1000              | ( 1340.3 ) | 2.955                            | ( 0.706 )       |
| 350               | ( 170.3 )  | 2.032                            | ( 0.486 )       | 1100              | ( 1520.3 ) | 3.059                            | ( 0.731 )       |
| 400               | ( 260.3 )  | 2.182                            | ( 0.521 )       | 1200              | ( 1700.3 ) | 3.165                            | ( 0.757 )       |
| 450               | ( 350.3 )  | 2.294                            | ( 0.548 )       | 1300              | ( 1880.3 ) | 3.276                            | ( 0.783 )       |
| 500               | ( 440.3 )  | 2.384                            | ( 0.570 )       | 1400              | ( 2060.3 ) | 3.390                            | ( 0.810 )       |
| 550               | ( 530.3 )  | 2.460                            | ( 0.588 )       | 1500              | ( 2240.3 ) | 3.510                            | ( 0.839 )       |
| 600               | ( 620.3 )  | 2.527                            | ( 0.604 )       | 1560              | ( 2348.3 ) | 3.584                            | ( 0.857 )       |

**Application Notes:** Data for specific heat is collected from references [2, 5, 6, 13, 14] and fitted with the equations below to approximate property trend with respect to temperature. Fitted trend is compared against trend from [11].

## Fit Equation:

For temperature range:  $5 \leq T < 293$

$$C_p(T) = \left[ A0 \cdot \left( \frac{T}{1000} \right)^2 \right] / \left[ 1 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $293 \leq T \leq 1560$

$$C_p(T) = B0 + B1 \cdot \left( \frac{T}{1000} \right) + B2 \cdot \left( \frac{T}{1000} \right)^2 + B\_2 / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$T = \text{Temperature [K]}$

## Constants:

T. Range [K]:

$5 \leq T < 293$

$293 \leq T \leq 1560$

A0 = 14.23

B0 = 2.401

A1 = -5.341

B1 = 0.3375

A2 = 14.39

B2 = 0.2804

B\_2 = -0.06388



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

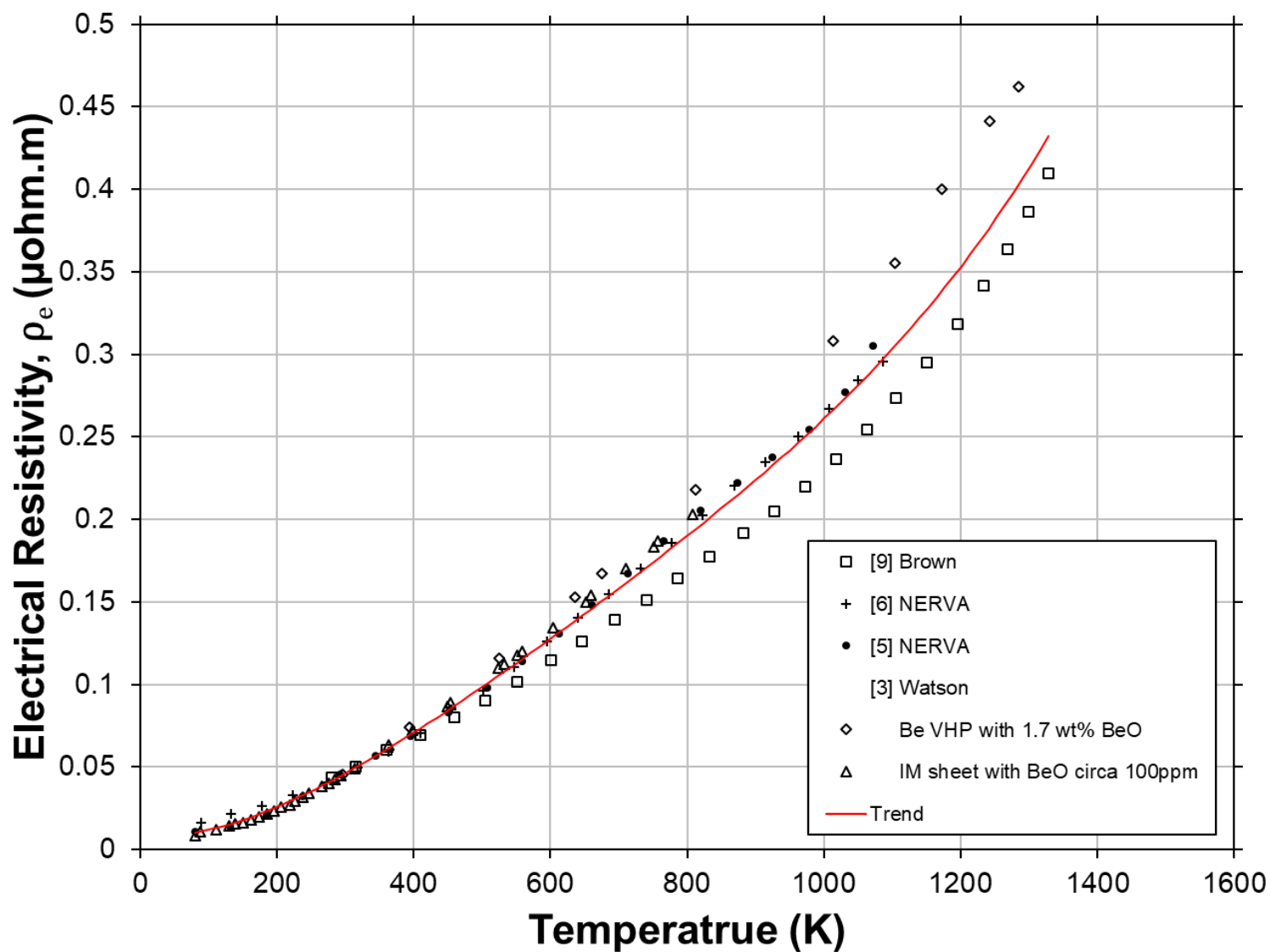


Figure 6.2.2-6: Electrical Resistivity versus Temperature of Beryllium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 79                | ( -317.5 ) | 0.010                               | ( 0.39 )                | 800               | ( 980.3 )  | 0.190                               | ( 7.48 )                |
| 100               | ( -279.7 ) | 0.012                               | ( 0.47 )                | 850               | ( 1070.3 ) | 0.207                               | ( 8.14 )                |
| 150               | ( -189.7 ) | 0.018                               | ( 0.70 )                | 900               | ( 1160.3 ) | 0.224                               | ( 8.82 )                |
| 200               | ( -99.7 )  | 0.026                               | ( 1.02 )                | 950               | ( 1250.3 ) | 0.242                               | ( 9.53 )                |
| 250               | ( -9.7 )   | 0.035                               | ( 1.39 )                | 1000              | ( 1340.3 ) | 0.261                               | ( 10.28 )               |
| 300               | ( 80.3 )   | 0.046                               | ( 1.82 )                | 1050              | ( 1430.3 ) | 0.281                               | ( 11.08 )               |
| 350               | ( 170.3 )  | 0.058                               | ( 2.29 )                | 1100              | ( 1520.3 ) | 0.303                               | ( 11.94 )               |
| 400               | ( 260.3 )  | 0.071                               | ( 2.79 )                | 1150              | ( 1610.3 ) | 0.327                               | ( 12.87 )               |
| 450               | ( 350.3 )  | 0.084                               | ( 3.32 )                | 1200              | ( 1700.3 ) | 0.353                               | ( 13.89 )               |
| 500               | ( 440.3 )  | 0.098                               | ( 3.88 )                | 1250              | ( 1790.3 ) | 0.381                               | ( 15.01 )               |
| 550               | ( 530.3 )  | 0.113                               | ( 4.44 )                | 1300              | ( 1880.3 ) | 0.413                               | ( 16.25 )               |
| 600               | ( 620.3 )  | 0.128                               | ( 5.03 )                | 1329              | ( 1932.5 ) | 0.433                               | ( 17.03 )               |
| 650               | ( 710.3 )  | 0.143                               | ( 5.62 )                |                   |            |                                     |                         |
| 700               | ( 800.3 )  | 0.158                               | ( 6.23 )                |                   |            |                                     |                         |
| 750               | ( 890.3 )  | 0.174                               | ( 6.85 )                |                   |            |                                     |                         |

**Application Notes:** Data for electrical resistivity is collected from references [3, 5, 6, 9] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3 + A4 \cdot \left(\frac{T}{1000}\right)^4$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $79 \leq T < 1329$

A0 = 7.141E-03

A1 = -9.177E-03

A2 = 6.214E-01

A3 = -5.928E-01

A4 = 2.345E-01



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2: 04-26-2023

Young's Modulus with Temperature

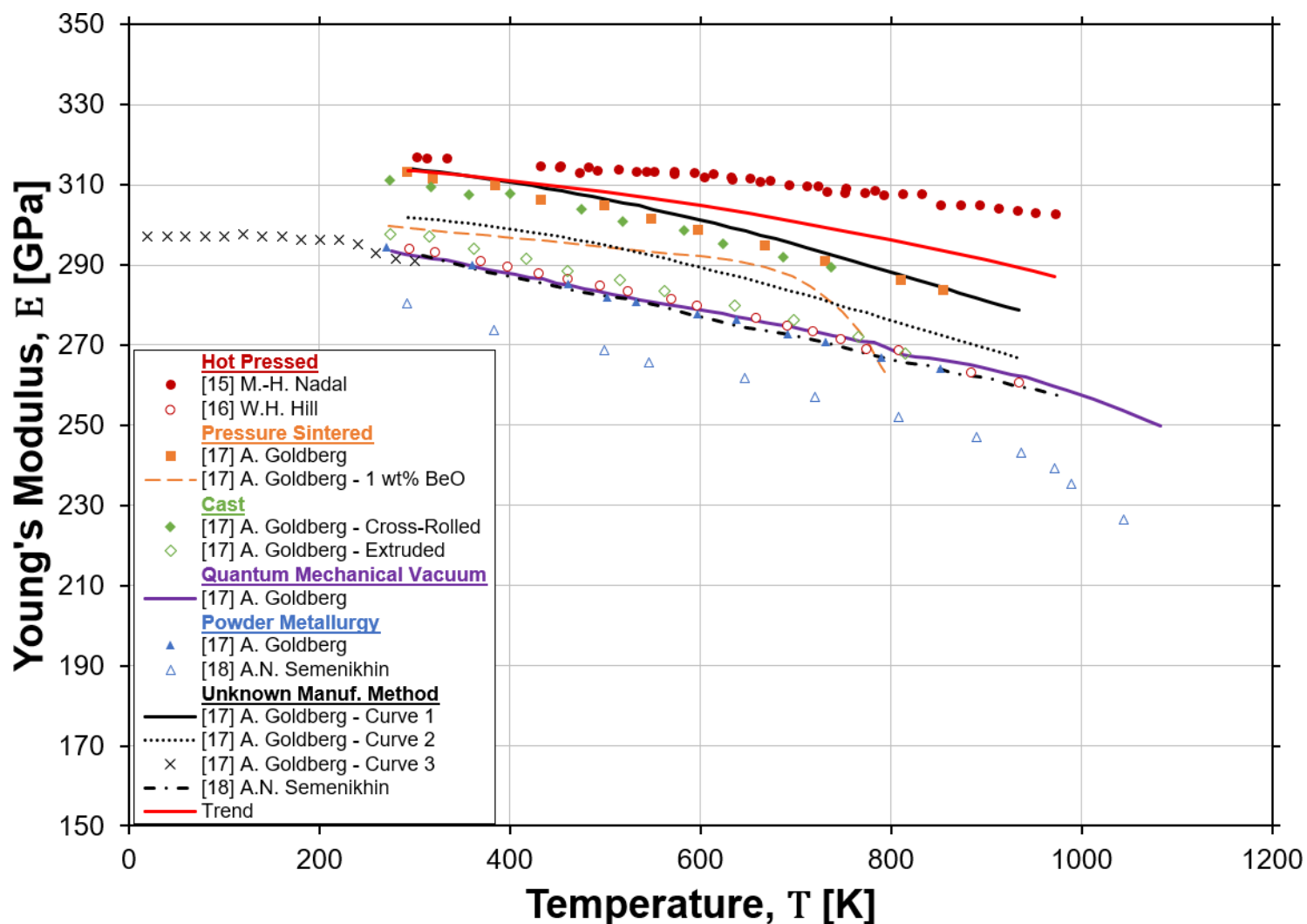


Figure 6.2.2-7: Young's Modulus versus Temperature for Beryllium. Data is sorted by manufacturing method, but the trend is solely based on datasets which contained both Young's modulus and shear modulus data: hot pressed data from Nadal and cast, cross-rolled data from Goldberg.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2: 04-26-2023

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|-----------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )    | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 293               | ( 67.7 )  | 313.63                | ( 45.51 ) | 580               | ( 584.3 )  | 305.52                | ( 44.33 ) |
| 300               | ( 80.3 )  | 313.49                | ( 45.49 ) | 600               | ( 620.3 )  | 304.78                | ( 44.22 ) |
| 320               | ( 116.3 ) | 313.06                | ( 45.42 ) | 620               | ( 656.3 )  | 304.03                | ( 44.11 ) |
| 340               | ( 152.3 ) | 312.61                | ( 45.36 ) | 640               | ( 692.3 )  | 303.25                | ( 44.00 ) |
| 360               | ( 188.3 ) | 312.14                | ( 45.29 ) | 660               | ( 728.3 )  | 302.46                | ( 43.89 ) |
| 380               | ( 224.3 ) | 311.65                | ( 45.22 ) | 680               | ( 764.3 )  | 301.64                | ( 43.77 ) |
| 400               | ( 260.3 ) | 311.13                | ( 45.14 ) | 700               | ( 800.3 )  | 300.79                | ( 43.65 ) |
| 420               | ( 296.3 ) | 310.59                | ( 45.07 ) | 720               | ( 836.3 )  | 299.93                | ( 43.52 ) |
| 440               | ( 332.3 ) | 310.03                | ( 44.99 ) | 740               | ( 872.3 )  | 299.05                | ( 43.39 ) |
| 460               | ( 368.3 ) | 309.45                | ( 44.90 ) | 750               | ( 890.3 )  | 298.60                | ( 43.33 ) |
| 480               | ( 404.3 ) | 308.85                | ( 44.81 ) | 800               | ( 980.3 )  | 296.26                | ( 42.99 ) |
| 500               | ( 440.3 ) | 308.23                | ( 44.72 ) | 850               | ( 1070.3 ) | 293.79                | ( 42.63 ) |
| 520               | ( 476.3 ) | 307.58                | ( 44.63 ) | 900               | ( 1160.3 ) | 291.19                | ( 42.25 ) |
| 540               | ( 512.3 ) | 306.92                | ( 44.53 ) | 950               | ( 1250.3 ) | 288.44                | ( 41.85 ) |
| 560               | ( 548.3 ) | 306.23                | ( 44.43 ) | 972               | ( 1289.9 ) | 287.20                | ( 41.67 ) |

**Application Notes:** Data for Young's modulus is collected from references [15-18]. However, only datasets which contained both Young's modulus and shear modulus data were used to make the trend below. The fit equation is an approximation of Young's modulus with respect to temperature. For more information, please see the Additional Application Notes section at the end of this chapter.

## Fit Equation:

$$E(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T. Range [K]: 293 ≤ T ≤ 972

A0 = 317.3

A1 = -4.553

A2 = -27.18



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2: 04-26-2023

Shear Modulus with Temperature

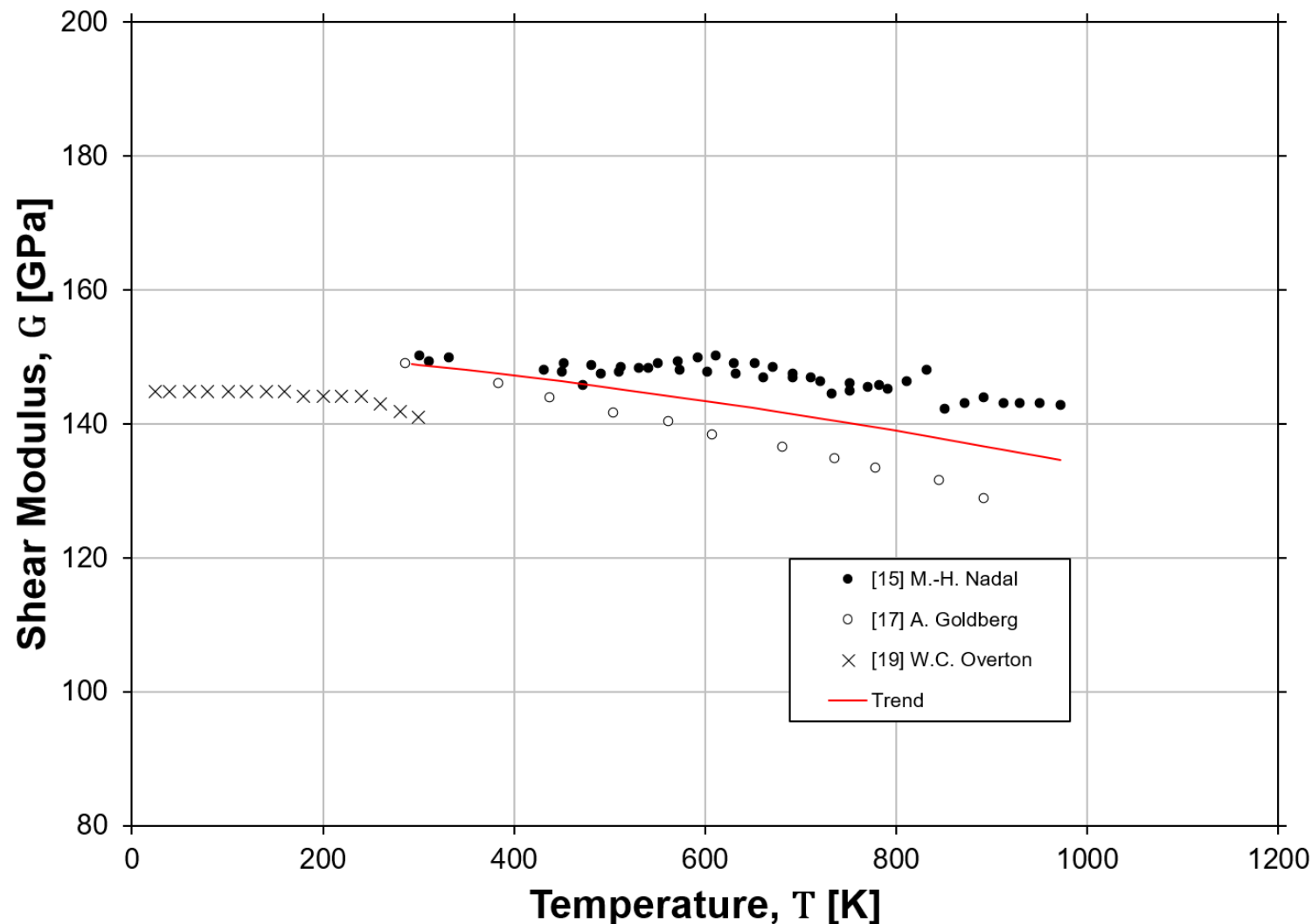


Figure 6.2.2-8: Shear Modulus versus Temperature for Beryllium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2: 04-26-2023

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Shear Modulus ( G ) |           | Temperature ( T ) |            | Shear Modulus ( G ) |           |
|-------------------|-----------|---------------------|-----------|-------------------|------------|---------------------|-----------|
| K                 | ( °F )    | GPa                 | ( Msi )   | K                 | ( °F )     | GPa                 | ( Msi )   |
| 293               | ( 67.7 )  | 148.92              | ( 21.61 ) | 580               | ( 584.3 )  | 143.86              | ( 20.87 ) |
| 300               | ( 80.3 )  | 148.81              | ( 21.59 ) | 600               | ( 620.3 )  | 143.45              | ( 20.81 ) |
| 320               | ( 116.3 ) | 148.51              | ( 21.55 ) | 620               | ( 656.3 )  | 143.04              | ( 20.75 ) |
| 340               | ( 152.3 ) | 148.19              | ( 21.50 ) | 640               | ( 692.3 )  | 142.62              | ( 20.69 ) |
| 360               | ( 188.3 ) | 147.87              | ( 21.46 ) | 660               | ( 728.3 )  | 142.19              | ( 20.63 ) |
| 380               | ( 224.3 ) | 147.54              | ( 21.41 ) | 680               | ( 764.3 )  | 141.75              | ( 20.57 ) |
| 400               | ( 260.3 ) | 147.20              | ( 21.36 ) | 700               | ( 800.3 )  | 141.31              | ( 20.50 ) |
| 420               | ( 296.3 ) | 146.86              | ( 21.31 ) | 720               | ( 836.3 )  | 140.86              | ( 20.44 ) |
| 440               | ( 332.3 ) | 146.51              | ( 21.26 ) | 740               | ( 872.3 )  | 140.41              | ( 20.37 ) |
| 460               | ( 368.3 ) | 146.15              | ( 21.21 ) | 750               | ( 890.3 )  | 140.18              | ( 20.34 ) |
| 480               | ( 404.3 ) | 145.79              | ( 21.15 ) | 800               | ( 980.3 )  | 139.00              | ( 20.17 ) |
| 500               | ( 440.3 ) | 145.41              | ( 21.10 ) | 850               | ( 1070.3 ) | 137.77              | ( 19.99 ) |
| 520               | ( 476.3 ) | 145.04              | ( 21.04 ) | 900               | ( 1160.3 ) | 136.51              | ( 19.81 ) |
| 540               | ( 512.3 ) | 144.65              | ( 20.99 ) | 950               | ( 1250.3 ) | 135.20              | ( 19.62 ) |
| 560               | ( 548.3 ) | 144.26              | ( 20.93 ) | 972               | ( 1289.9 ) | 134.61              | ( 19.53 ) |

**Application Notes:** Data for shear modulus is collected from references [15, 17, 19]. Datasets from [15, 17] were fitted with the equation below to approximate property trend with respect to temperature. For more information, please see the Additional Application Notes section at the end of this chapter.

## Fit Equation:

$$G(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $293 \leq T \leq 972$

A0 = 152.6

A1 = -9.985

A2 = -8.773



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2: 04-26-2023

Poisson's Ratio with Temperature

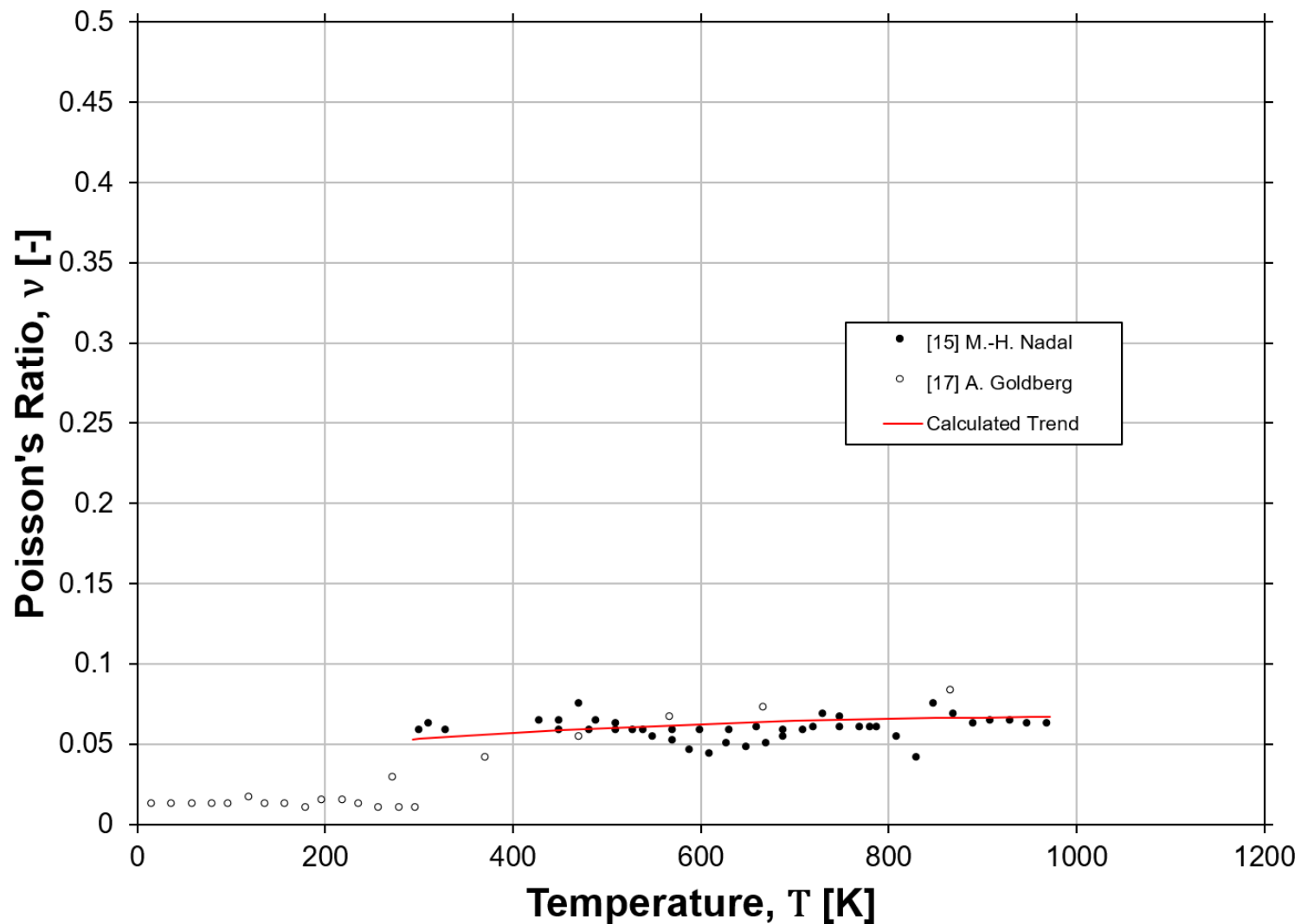


Figure 6.2.2-9: Poisson's Ratio versus Temperature for Beryllium. Calculated from trends for Young's modulus and shear modulus.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2: 04-26-2023

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Poisson's Ratio ( ν ) | Temperature ( T ) |            | Poisson's Ratio ( ν ) |
|-------------------|-----------|-----------------------|-------------------|------------|-----------------------|
| K                 | ( °F )    |                       | K                 | ( °F )     |                       |
| 293               | ( 67.7 )  | 0.053                 | 580               | ( 584.3 )  | 0.062                 |
| 300               | ( 80.3 )  | 0.053                 | 600               | ( 620.3 )  | 0.062                 |
| 320               | ( 116.3 ) | 0.054                 | 620               | ( 656.3 )  | 0.063                 |
| 340               | ( 152.3 ) | 0.055                 | 640               | ( 692.3 )  | 0.063                 |
| 360               | ( 188.3 ) | 0.055                 | 660               | ( 728.3 )  | 0.064                 |
| 380               | ( 224.3 ) | 0.056                 | 680               | ( 764.3 )  | 0.064                 |
| 400               | ( 260.3 ) | 0.057                 | 700               | ( 800.3 )  | 0.064                 |
| 420               | ( 296.3 ) | 0.057                 | 720               | ( 836.3 )  | 0.065                 |
| 440               | ( 332.3 ) | 0.058                 | 740               | ( 872.3 )  | 0.065                 |
| 460               | ( 368.3 ) | 0.059                 | 750               | ( 890.3 )  | 0.065                 |
| 480               | ( 404.3 ) | 0.059                 | 800               | ( 980.3 )  | 0.066                 |
| 500               | ( 440.3 ) | 0.060                 | 850               | ( 1070.3 ) | 0.066                 |
| 520               | ( 476.3 ) | 0.060                 | 900               | ( 1160.3 ) | 0.067                 |
| 540               | ( 512.3 ) | 0.061                 | 950               | ( 1250.3 ) | 0.067                 |
| 560               | ( 548.3 ) | 0.061                 | 972               | ( 1289.9 ) | 0.067                 |

**Application Notes:** Data for Poisson's Ratio is collected from references [15, 17]. Property trend is calculated as a function of Young's modulus and shear modulus, then converted to an equivalent equation for readers' convenience. For more information, please see the Additional Application Notes section at the end of this chapter.

## Fit Equation:

$$\nu(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$\nu(T) = \text{Poisson's Ratio } [-]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 293 \leq T \leq 972$$

$$A0 = 0.03986$$

$$A1 = 0.05156$$

$$A2 = -0.02185$$

$$A3 = -0.002744$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Yield Strength with Temperature

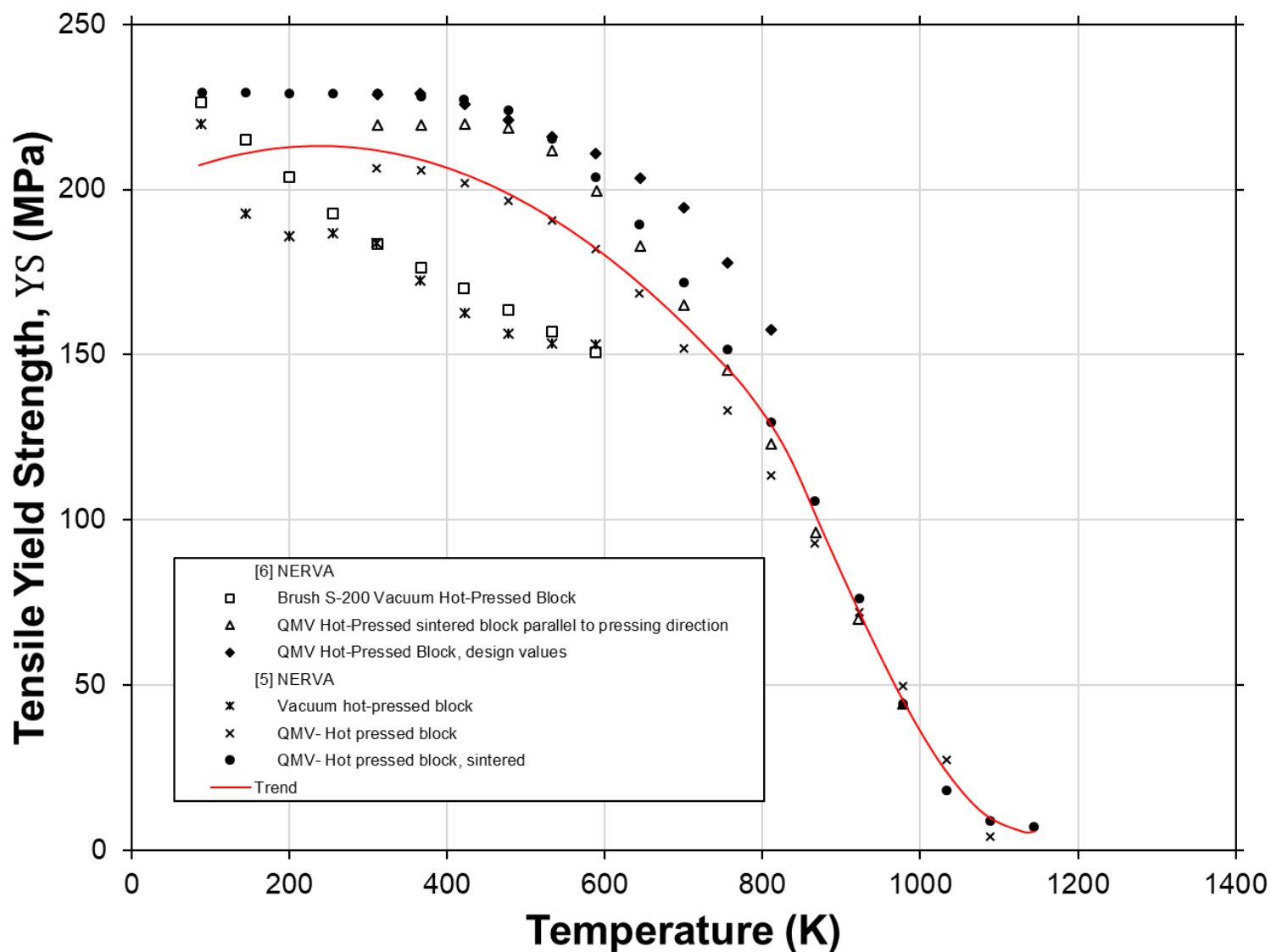


Figure 6.2.2-10: Yield Strength versus Temperature of Beryllium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Yield Strength with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Yield Strength ( YS ) |           | Temperature ( T ) |            | Yield Strength ( YS ) |           |
|-------------------|------------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | MPa                   | ( Ksi )   | K                 | ( °F )     | MPa                   | ( Ksi )   |
| 85                | ( -306.7 ) | 223.99                | ( 32.50 ) | 450               | ( 350.3 )  | 199.72                | ( 28.98 ) |
| 100               | ( -279.7 ) | 221.40                | ( 32.12 ) | 475               | ( 395.3 )  | 198.54                | ( 28.81 ) |
| 125               | ( -234.7 ) | 217.61                | ( 31.58 ) | 500               | ( 440.3 )  | 197.08                | ( 28.60 ) |
| 150               | ( -189.7 ) | 214.42                | ( 31.11 ) | 525               | ( 485.3 )  | 195.26                | ( 28.33 ) |
| 175               | ( -144.7 ) | 211.76                | ( 30.73 ) | 550               | ( 530.3 )  | 193.03                | ( 28.01 ) |
| 200               | ( -99.7 )  | 209.58                | ( 30.41 ) | 575               | ( 575.3 )  | 190.32                | ( 27.61 ) |
| 225               | ( -54.7 )  | 207.80                | ( 30.15 ) | 600               | ( 620.3 )  | 187.06                | ( 27.14 ) |
| 250               | ( -9.7 )   | 206.37                | ( 29.94 ) | 625               | ( 665.3 )  | 183.21                | ( 26.58 ) |
| 275               | ( 35.3 )   | 205.21                | ( 29.78 ) | 650               | ( 710.3 )  | 178.68                | ( 25.93 ) |
| 300               | ( 80.3 )   | 204.28                | ( 29.64 ) | 700               | ( 800.3 )  | 167.38                | ( 24.29 ) |
| 325               | ( 125.3 )  | 203.50                | ( 29.53 ) | 800               | ( 980.3 )  | 133.98                | ( 19.44 ) |
| 350               | ( 170.3 )  | 202.82                | ( 29.43 ) | 900               | ( 1160.3 ) | 83.79                 | ( 12.16 ) |
| 375               | ( 215.3 )  | 202.16                | ( 29.33 ) | 1000              | ( 1340.3 ) | 36.00                 | ( 5.22 )  |
| 400               | ( 260.3 )  | 201.46                | ( 29.23 ) | 1100              | ( 1520.3 ) | 8.01                  | ( 1.16 )  |
| 425               | ( 305.3 )  | 200.67                | ( 29.12 ) | 1150              | ( 1610.3 ) | 6.16                  | ( 0.89 )  |

**Application Notes:** Data for yield strength is collected from references [5, 6] and fitted with the equations below to approximate property trends with respect to temperature. For more information, please see the Additional Application Notes section at the end of this chapter.

## Fit Equations:

$$YS(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$YS(T)$  = Yield Strength [MPa]

$T$  = Temperature [K]

## Constants:

| T Range [K]: | <u><math>85 \leq T &lt; 810</math></u> | <u><math>810 &lt; T \leq 1150</math></u> |
|--------------|----------------------------------------|------------------------------------------|
| A0 =         | 243.8                                  | -1084                                    |
| A1 =         | -290.4                                 | 5158                                     |
| A2 =         | 731.4                                  | -6552                                    |
| A3 =         | -675                                   | 2514                                     |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Tensile Strength with Temperature

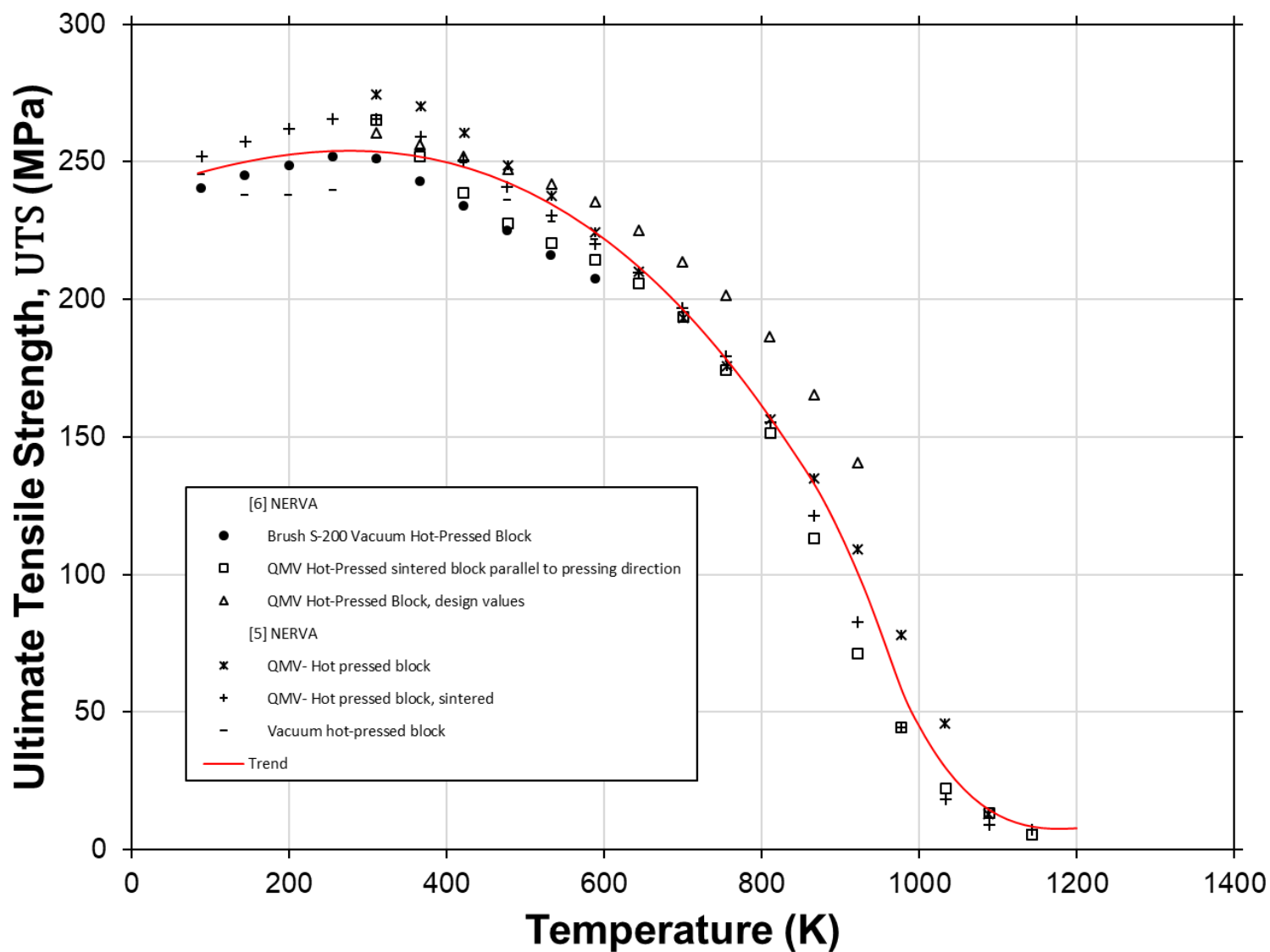


Figure 6.2.2-11: Ultimate Tensile Strength versus Temperature of Beryllium.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 0: 08-05-2020

Tensile Strength with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Tensile Strength ( TS ) |           | Temperature ( T ) |            | Tensile Strength ( TS ) |           |
|-------------------|------------|-------------------------|-----------|-------------------|------------|-------------------------|-----------|
| K                 | ( °F )     | MPa                     | ( Ksi )   | K                 | ( °F )     | MPa                     | ( Ksi )   |
| 85                | ( -306.7 ) | 245.78                  | ( 35.66 ) | 450               | ( 350.3 )  | 245.51                  | ( 35.62 ) |
| 100               | ( -279.7 ) | 246.93                  | ( 35.83 ) | 475               | ( 395.3 )  | 242.74                  | ( 35.22 ) |
| 125               | ( -234.7 ) | 248.68                  | ( 36.08 ) | 500               | ( 440.3 )  | 239.54                  | ( 34.76 ) |
| 150               | ( -189.7 ) | 250.21                  | ( 36.31 ) | 525               | ( 485.3 )  | 235.88                  | ( 34.23 ) |
| 175               | ( -144.7 ) | 251.50                  | ( 36.49 ) | 550               | ( 530.3 )  | 231.75                  | ( 33.63 ) |
| 200               | ( -99.7 )  | 252.53                  | ( 36.64 ) | 575               | ( 575.3 )  | 227.14                  | ( 32.96 ) |
| 225               | ( -54.7 )  | 253.29                  | ( 36.75 ) | 600               | ( 620.3 )  | 222.04                  | ( 32.22 ) |
| 250               | ( -9.7 )   | 253.77                  | ( 36.82 ) | 625               | ( 665.3 )  | 216.42                  | ( 31.40 ) |
| 275               | ( 35.3 )   | 253.95                  | ( 36.85 ) | 650               | ( 710.3 )  | 210.27                  | ( 30.51 ) |
| 300               | ( 80.3 )   | 253.81                  | ( 36.83 ) | 700               | ( 800.3 )  | 196.32                  | ( 28.49 ) |
| 325               | ( 125.3 )  | 253.34                  | ( 36.76 ) | 800               | ( 980.3 )  | 161.43                  | ( 23.42 ) |
| 350               | ( 170.3 )  | 252.53                  | ( 36.64 ) | 900               | ( 1160.3 ) | 116.38                  | ( 16.89 ) |
| 375               | ( 215.3 )  | 251.36                  | ( 36.47 ) | 1000              | ( 1340.3 ) | 45.00                   | ( 6.53 )  |
| 400               | ( 260.3 )  | 249.80                  | ( 36.25 ) | 1100              | ( 1520.3 ) | 12.61                   | ( 1.83 )  |
| 425               | ( 305.3 )  | 247.86                  | ( 35.96 ) | 1200              | ( 1700.3 ) | 7.74                    | ( 1.12 )  |

**Application Notes:** Data for ultimate tensile strength is collected from references [5, 6] and fitted with the equations below to approximate property trends with respect to temperature. For more information, please see the Additional Application Notes section at the end of this chapter.

## Fit Equations:

$$TS(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$TS(T)$  = Tensile Strength [MPa]

$T$  = Temperature [K]

## Constants:

T Range [K]:       $85 \leq T < 921$        $921 \leq T \leq 1200$

A0 =                      238                      6472

A1 =                      102.8                      -15800

A2 =                      -118.3                      12850

A3 =                      -162.3                      -3477



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2: 04-26-2023

Additional Application Notes

### Young's Modulus, Shear Modulus, and Poisson's Ratio Application Notes:

Starting with revision 2 of this handbook, the beryllium Young's modulus and shear modulus datasets have been updated to increase the accuracy of trends. In previous versions, static modulus data was displayed in each plot. Static modulus testing methods are known for the broad uncertainty intervals of their results, particularly at elevated temperatures. As a result, moduli values are often greatly underestimated by static testing. Luckily, more recent testing has been performed [15, 17], contributing toward a more realistic trend. All datasets shown in the Young's modulus, shear modulus, and Poisson's ratio plots for beryllium were gathered using dynamic modulus testing instead of static modulus testing. However, since a wide spread was still observed in Young's modulus data, only sources that measured Young's modulus and shear modulus in a single experiment were used to generate the Young's modulus trend. Over time, and where dynamic modulus data is available, improvements can be implemented for other materials' moduli trends as well.

While dynamic modulus testing generally provides more consistent values than static modulus testing, note that the dynamic modulus and static modulus represent two different scenarios. Dynamic modulus values are more appropriate for modeling a system experiencing vibration, while static modulus values become more useful for structures experiencing static loading. Other industries, such as the construction industry, employ equations to relate static and dynamic moduli values [20].

Poisson's ratio is calculated from Young's modulus and shear modulus as usual. Note that both the data and trend shown in the Poisson's ratio plot are approximately constant, and they are very close to zero. A Poisson's ratio of zero indicates that stretching or compressing one dimension (i.e. the length) of a body has no impact on the other dimensions (i.e. the width and height). Versions 0 and 1 of this handbook suggested that the Poisson's ratio for beryllium could be negative at high temperatures, but this is no longer expected.

### General Application Notes:

For all materials, properties will vary slightly by lot due to material composition and other factors. Beryllium, however, is more sensitive to these variables. Others have reported significant differences between beryllium pieces. BeO content, grain size, and manufacturing methods contribute to these differences [5, 6]. For mechanical properties in particular, additional testing with consistent material variables should be performed to determine the accuracy of this handbook's trends.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

Revision 2: 04-26-2023

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                          |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|--------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's Ratio<br>( ν ) |
| 100                 | 1857                                  | 0.012                                                   | -                                           | -0.127                                             | 6.53                                      | 0.233                                          | -                                  | -                                | -                        |
| 200                 | 1855                                  | 0.026                                                   | 272.06                                      | -0.085                                             | 9.34                                      | 1.122                                          | -                                  | -                                | -                        |
| 300                 | 1850                                  | 0.046                                                   | 184.53                                      | 0.006                                              | 11.13                                     | 1.818                                          | 313.49                             | 148.81                           | 0.053                    |
| 350                 | 1846                                  | 0.058                                                   | 166.63                                      | 0.068                                              | 11.82                                     | 2.032                                          | 312.38                             | 148.03                           | 0.055                    |
| 400                 | 1843                                  | 0.071                                                   | 154.18                                      | 0.134                                              | 12.42                                     | 2.182                                          | 311.13                             | 147.20                           | 0.057                    |
| 450                 | 1839                                  | 0.084                                                   | 144.62                                      | 0.204                                              | 12.95                                     | 2.294                                          | 309.75                             | 146.33                           | 0.058                    |
| 500                 | 1835                                  | 0.098                                                   | 136.71                                      | 0.277                                              | 13.42                                     | 2.384                                          | 308.23                             | 145.41                           | 0.060                    |
| 550                 | 1830                                  | 0.113                                                   | 129.85                                      | 0.354                                              | 13.85                                     | 2.460                                          | 306.57                             | 144.45                           | 0.061                    |
| 600                 | 1826                                  | 0.128                                                   | 123.68                                      | 0.434                                              | 14.25                                     | 2.527                                          | 304.78                             | 143.45                           | 0.062                    |
| 650                 | 1822                                  | 0.143                                                   | 117.99                                      | 0.517                                              | 14.61                                     | 2.588                                          | 302.86                             | 142.40                           | 0.063                    |
| 700                 | 1817                                  | 0.158                                                   | 112.67                                      | 0.603                                              | 14.95                                     | 2.644                                          | 300.79                             | 141.31                           | 0.064                    |
| 750                 | 1812                                  | 0.174                                                   | 107.63                                      | 0.692                                              | 15.27                                     | 2.698                                          | 298.60                             | 140.18                           | 0.065                    |
| 800                 | 1807                                  | 0.190                                                   | 102.82                                      | 0.784                                              | 15.57                                     | 2.751                                          | 296.26                             | 139.00                           | 0.066                    |
| 850                 | 1802                                  | 0.207                                                   | 98.21                                       | 0.878                                              | 15.86                                     | 2.802                                          | 293.79                             | 137.77                           | 0.066                    |
| 900                 | 1797                                  | 0.224                                                   | 93.78                                       | 0.975                                              | 16.13                                     | 2.853                                          | 291.19                             | 136.51                           | 0.067                    |
| 950                 | 1792                                  | 0.242                                                   | 89.50                                       | 1.074                                              | 16.40                                     | 2.904                                          | 288.44                             | 135.20                           | 0.067                    |
| 1000                | 1786                                  | 0.261                                                   | 85.36                                       | 1.175                                              | 16.65                                     | 2.955                                          | -                                  | -                                | -                        |
| 1050                | 1781                                  | 0.281                                                   | 81.36                                       | 1.278                                              | 16.89                                     | 3.007                                          | -                                  | -                                | -                        |
| 1100                | 1775                                  | 0.303                                                   | 77.49                                       | 1.382                                              | 17.13                                     | 3.059                                          | -                                  | -                                | -                        |
| 1150                | 1770                                  | 0.327                                                   | 73.74                                       | 1.489                                              | 17.36                                     | 3.112                                          | -                                  | -                                | -                        |
| 1200                | 1764                                  | 0.353                                                   | 70.11                                       | 1.597                                              | 17.58                                     | 3.165                                          | -                                  | -                                | -                        |
| 1300                | 1753                                  | 0.413                                                   | 63.19                                       | 1.817                                              | 18.02                                     | 3.276                                          | -                                  | -                                | -                        |
| 1400                | 1741                                  | -                                                       | 56.72                                       | 2.042                                              | 18.43                                     | 3.390                                          | -                                  | -                                | -                        |
| 1500                | 1730                                  | -                                                       | 50.69                                       | 2.270                                              | 18.83                                     | 3.510                                          | -                                  | -                                | -                        |
| 1550                | 1724                                  | -                                                       | 47.83                                       | 2.385                                              | 19.03                                     | 3.571                                          | -                                  | -                                | -                        |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

**Revision 2: 04-26-2023**

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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

6 Nuclear Materials

6.2 Neutron Control Materials

6.2.2 Beryllium (Be)

**Revision 2: 04-26-2023**

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## **7 Refractory Ceramics**

### **7.1 Oxides**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 2.1: 08-25-2023

General

### Room Temperature Properties

|                                           |           |
|-------------------------------------------|-----------|
| Molar Mass, [g/mol]                       | 25.01     |
| Theoretical Density, [kg/m <sup>3</sup> ] | 3,010     |
| Melting Point, [K]                        | 2823 ± 25 |
| Boiling Point, [K]                        | 4533      |
| α - β phase transition [K]                | 2322      |
| Specific Heat, [J/(g-K)]                  | 1.00      |
| Thermal Conductivity, [W/(m-K)]           | 275.33    |
| Linear expansion coefficient, [μm/(m-K)]  | 4.87      |
| Young's Modulus, [GPa]                    | 344.7     |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Density with Temperature

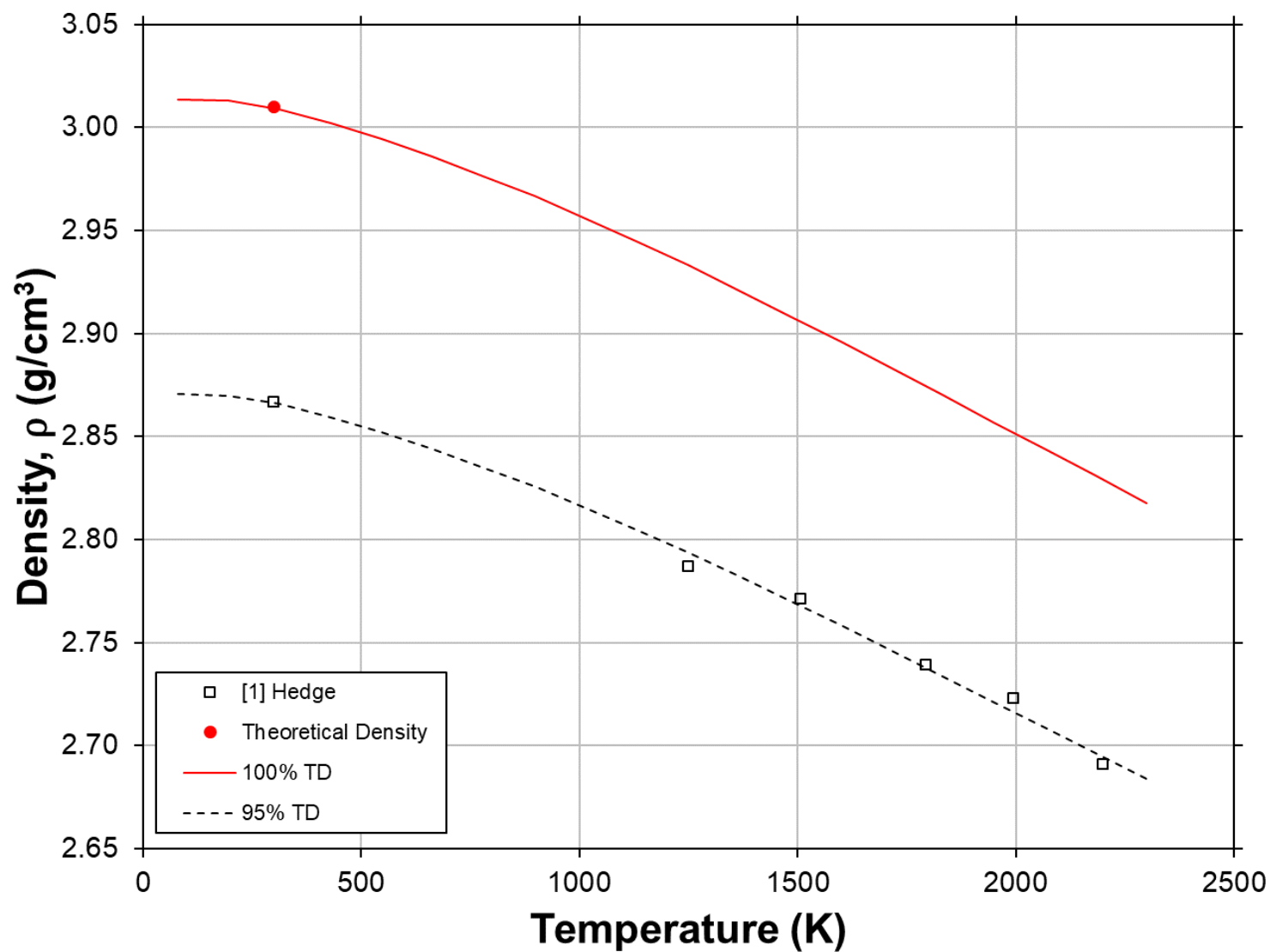


Figure 7.1.1-1: Density versus Temperature for Beryllia. Calculated from fitted trend of the Thermal Expansion data based on 100% TD (red) and 95% TD (dotted black). Compared against data from Hedge (1963).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 79                | ( -317.5 ) | 3014              | ( 188.1 )              | 900               | ( 1160.3 ) | 2966              | ( 185.2 )              |
| 100               | ( -279.7 ) | 3014              | ( 188.1 )              | 1000              | ( 1340.3 ) | 2957              | ( 184.6 )              |
| 150               | ( -189.7 ) | 3014              | ( 188.1 )              | 1100              | ( 1520.3 ) | 2948              | ( 184.0 )              |
| 200               | ( -99.7 )  | 3013              | ( 188.1 )              | 1200              | ( 1700.3 ) | 2938              | ( 183.4 )              |
| 250               | ( -9.7 )   | 3012              | ( 188.0 )              | 1300              | ( 1880.3 ) | 2928              | ( 182.8 )              |
| 300               | ( 80.3 )   | 3009              | ( 187.9 )              | 1400              | ( 2060.3 ) | 2917              | ( 182.1 )              |
| 350               | ( 170.3 )  | 3007              | ( 187.7 )              | 1500              | ( 2240.3 ) | 2907              | ( 181.5 )              |
| 400               | ( 260.3 )  | 3004              | ( 187.5 )              | 1600              | ( 2420.3 ) | 2896              | ( 180.8 )              |
| 450               | ( 350.3 )  | 3001              | ( 187.3 )              | 1700              | ( 2600.3 ) | 2885              | ( 180.1 )              |
| 500               | ( 440.3 )  | 2997              | ( 187.1 )              | 1800              | ( 2780.3 ) | 2874              | ( 179.4 )              |
| 550               | ( 530.3 )  | 2994              | ( 186.9 )              | 1900              | ( 2960.3 ) | 2863              | ( 178.7 )              |
| 600               | ( 620.3 )  | 2991              | ( 186.7 )              | 2000              | ( 3140.3 ) | 2851              | ( 178.0 )              |
| 650               | ( 710.3 )  | 2987              | ( 186.5 )              | 2100              | ( 3320.3 ) | 2840              | ( 177.3 )              |
| 700               | ( 800.3 )  | 2983              | ( 186.2 )              | 2200              | ( 3500.3 ) | 2829              | ( 176.6 )              |
| 800               | ( 980.3 )  | 2975              | ( 185.7 )              | 2300              | ( 3680.3 ) | 2818              | ( 175.9 )              |

**Application Notes:** Density trends are calculated as a function of thermal expansion data, as seen in the equation below for 100% theoretical density. Trend for 95% TD is compared against data from reference [1].

## Density Calculation:

$$\rho(T) = \rho_{RT}(1 - P)/(1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 3,010 \text{ [kg/m}^3\text{]}$$

*P* = Fractional Porosity

*T* = Temperature [K]

**Temperature Range:**  $79 \leq T \leq 2301$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

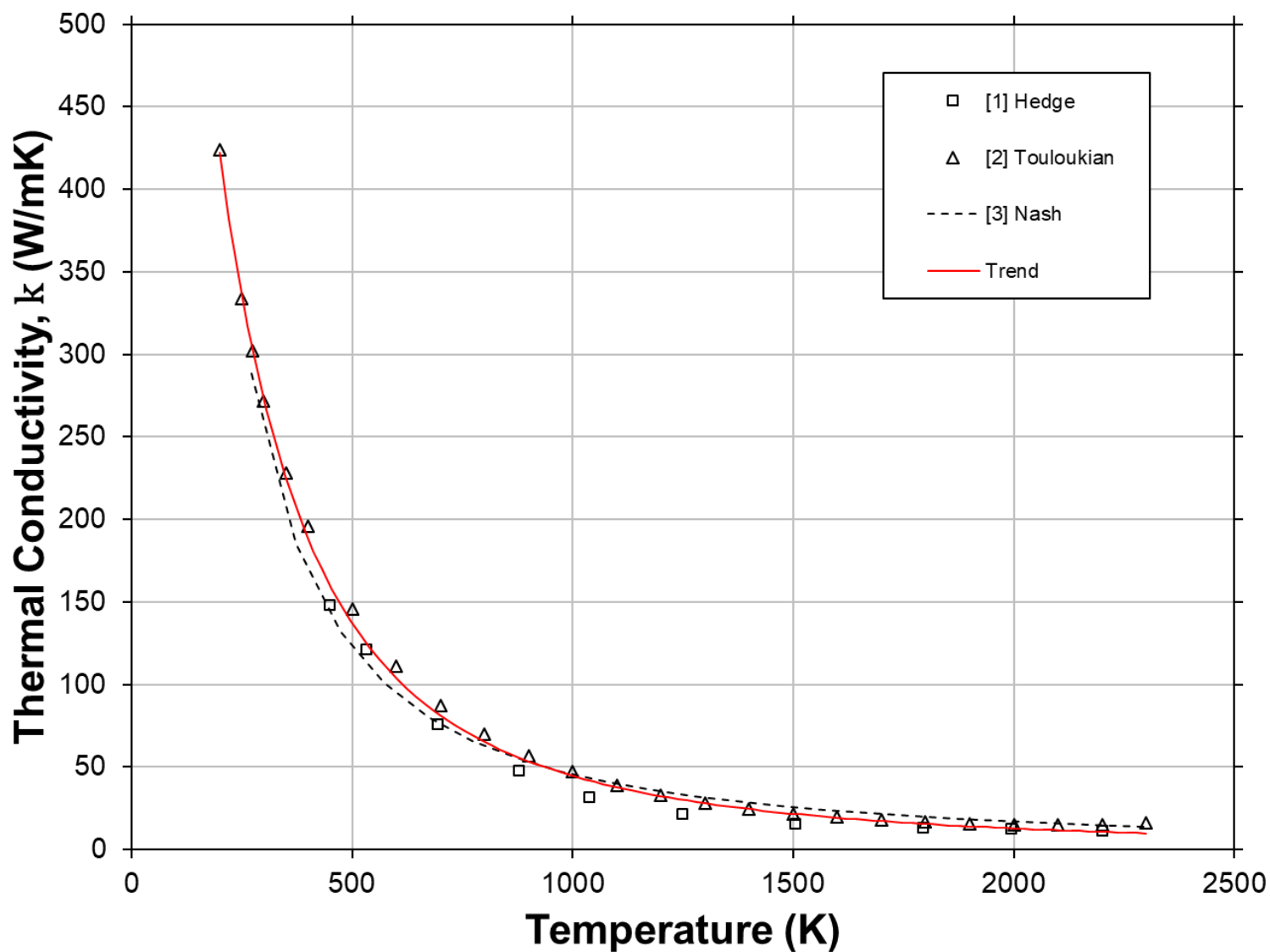


Figure 7.1.1-2: Thermal Conductivity versus Temperature of Beryllia. Displaying fitted trend of the data with comparison to trend from Nash (2005).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 200               | ( -99.7 )  | 421.93                     | ( 2927.41 )                         | 950               | ( 1250.3 ) | 48.78                      | ( 338.42 )                          |
| 250               | ( -9.7 )   | 336.46                     | ( 2334.39 )                         | 1000              | ( 1340.3 ) | 44.64                      | ( 309.73 )                          |
| 300               | ( 80.3 )   | 273.13                     | ( 1895.02 )                         | 1100              | ( 1520.3 ) | 37.80                      | ( 262.25 )                          |
| 350               | ( 170.3 )  | 225.36                     | ( 1563.60 )                         | 1200              | ( 1700.3 ) | 32.41                      | ( 224.87 )                          |
| 400               | ( 260.3 )  | 188.67                     | ( 1309.03 )                         | 1300              | ( 1880.3 ) | 28.10                      | ( 194.93 )                          |
| 450               | ( 350.3 )  | 160.00                     | ( 1110.12 )                         | 1400              | ( 2060.3 ) | 24.59                      | ( 170.58 )                          |
| 500               | ( 440.3 )  | 137.24                     | ( 952.22 )                          | 1500              | ( 2240.3 ) | 21.69                      | ( 150.52 )                          |
| 550               | ( 530.3 )  | 118.92                     | ( 825.06 )                          | 1600              | ( 2420.3 ) | 19.28                      | ( 133.79 )                          |
| 600               | ( 620.3 )  | 103.96                     | ( 721.31 )                          | 1700              | ( 2600.3 ) | 17.25                      | ( 119.71 )                          |
| 650               | ( 710.3 )  | 91.62                      | ( 635.66 )                          | 1800              | ( 2780.3 ) | 15.53                      | ( 107.74 )                          |
| 700               | ( 800.3 )  | 81.32                      | ( 564.21 )                          | 1900              | ( 2960.3 ) | 14.05                      | ( 97.48 )                           |
| 750               | ( 890.3 )  | 72.64                      | ( 504.01 )                          | 2000              | ( 3140.3 ) | 12.77                      | ( 88.62 )                           |
| 800               | ( 980.3 )  | 65.27                      | ( 452.86 )                          | 2100              | ( 3320.3 ) | 11.66                      | ( 80.92 )                           |
| 850               | ( 1070.3 ) | 58.96                      | ( 409.05 )                          | 2200              | ( 3500.3 ) | 10.69                      | ( 74.18 )                           |
| 900               | ( 1160.3 ) | 53.51                      | ( 371.25 )                          | 2300              | ( 3680.3 ) | 9.84                       | ( 68.25 )                           |

**Application Notes:** Data for thermal conductivity is collected from references [1, 2] and fitted with the equation below to approximate the property trend with respect to temperature. This trend is compared against trend from reference [3].

## Fit Equation:

$$k(T) = \left[ A_0 + A_1 \cdot \left( \frac{T}{1000} \right) \right] / \left[ A_{-0} + A_{-1} \cdot \left( \frac{T}{1000} \right) + \left( \frac{T}{1000} \right)^2 \right]$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]: 200 ≤ T < 2301

A0 = 57.1

A1 = 0.3584

A<sub>-0</sub> = 0.0476

A<sub>-1</sub> = 0.2395



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

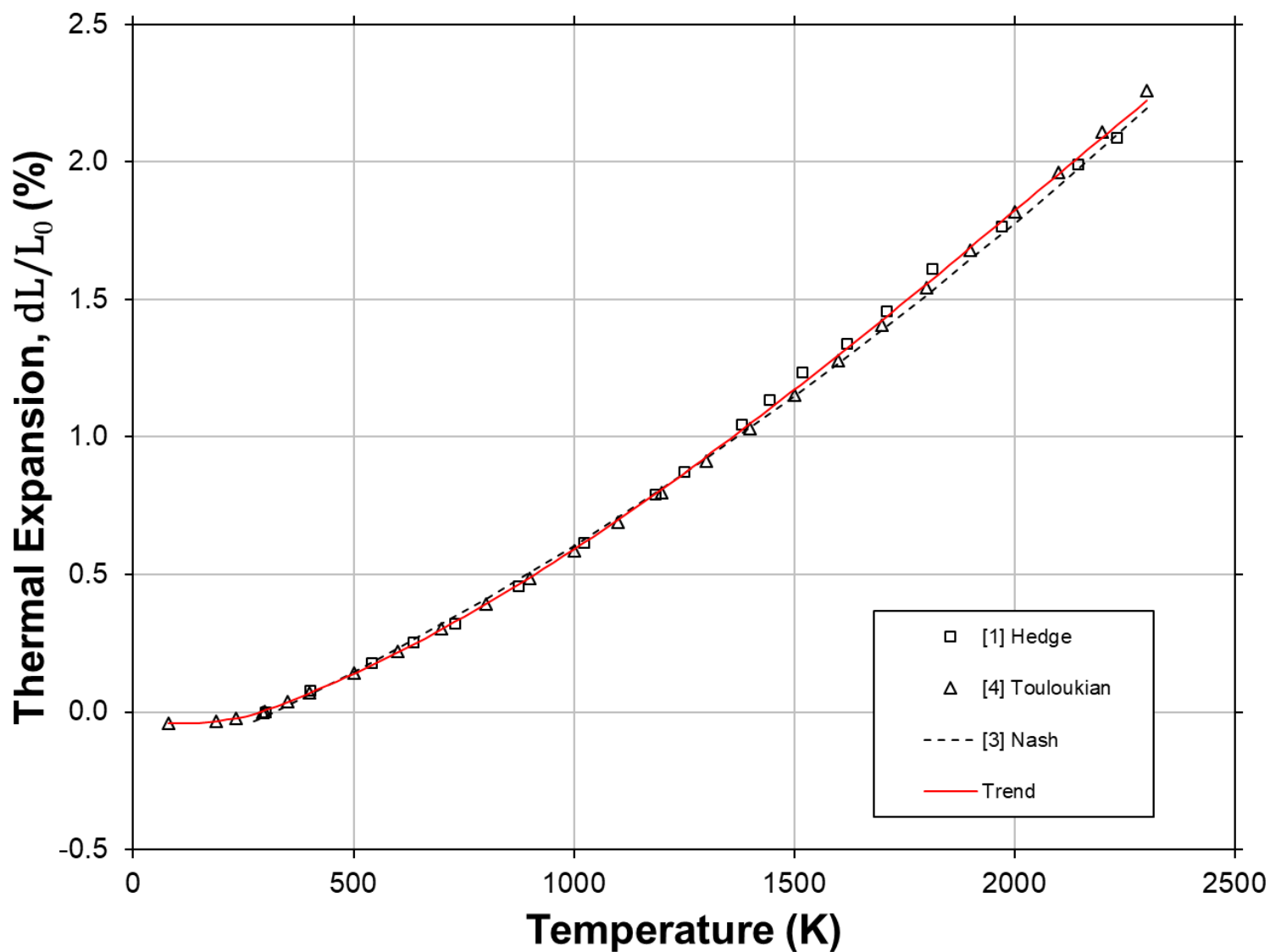


Figure 7.1.1-3: Thermal Expansion versus Temperature of Beryllia. Displaying fitted trend of the data with comparison to Nash (2005).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 79                | ( -317.5 ) | -0.041                                  | 900               | ( 1160.3 ) | 0.489                                   |
| 100               | ( -279.7 ) | -0.042                                  | 1000              | ( 1340.3 ) | 0.591                                   |
| 150               | ( -189.7 ) | -0.039                                  | 1100              | ( 1520.3 ) | 0.699                                   |
| 200               | ( -99.7 )  | -0.031                                  | 1200              | ( 1700.3 ) | 0.811                                   |
| 250               | ( -9.7 )   | -0.018                                  | 1300              | ( 1880.3 ) | 0.927                                   |
| 300               | ( 80.3 )   | 0.006                                   | 1400              | ( 2060.3 ) | 1.047                                   |
| 350               | ( 170.3 )  | 0.036                                   | 1500              | ( 2240.3 ) | 1.171                                   |
| 400               | ( 260.3 )  | 0.069                                   | 1600              | ( 2420.3 ) | 1.297                                   |
| 450               | ( 350.3 )  | 0.103                                   | 1700              | ( 2600.3 ) | 1.426                                   |
| 500               | ( 440.3 )  | 0.139                                   | 1800              | ( 2780.3 ) | 1.556                                   |
| 550               | ( 530.3 )  | 0.177                                   | 1900              | ( 2960.3 ) | 1.688                                   |
| 600               | ( 620.3 )  | 0.216                                   | 2000              | ( 3140.3 ) | 1.821                                   |
| 650               | ( 710.3 )  | 0.258                                   | 2100              | ( 3320.3 ) | 1.955                                   |
| 700               | ( 800.3 )  | 0.301                                   | 2200              | ( 3500.3 ) | 2.089                                   |
| 800               | ( 980.3 )  | 0.392                                   | 2300              | ( 3680.3 ) | 2.222                                   |

**Application Notes:** Data for thermal expansion is collected from references [1, 4] and fitted with the equation below to approximate property trend with respect to temperature. This trend is compared against trend from reference [3].

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

| T Range [K]: | <u>79 ≤ T ≤ 293</u> | <u>293 &lt; T ≤ 2301</u> |
|--------------|---------------------|--------------------------|
| A0 =         | -0.03153            | -0.1294                  |
| A1 =         | -0.1969             | 0.3148                   |
| A2 =         | 0.9181              | 0.481                    |
| A3 =         | 0.3533              | -0.07535                 |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

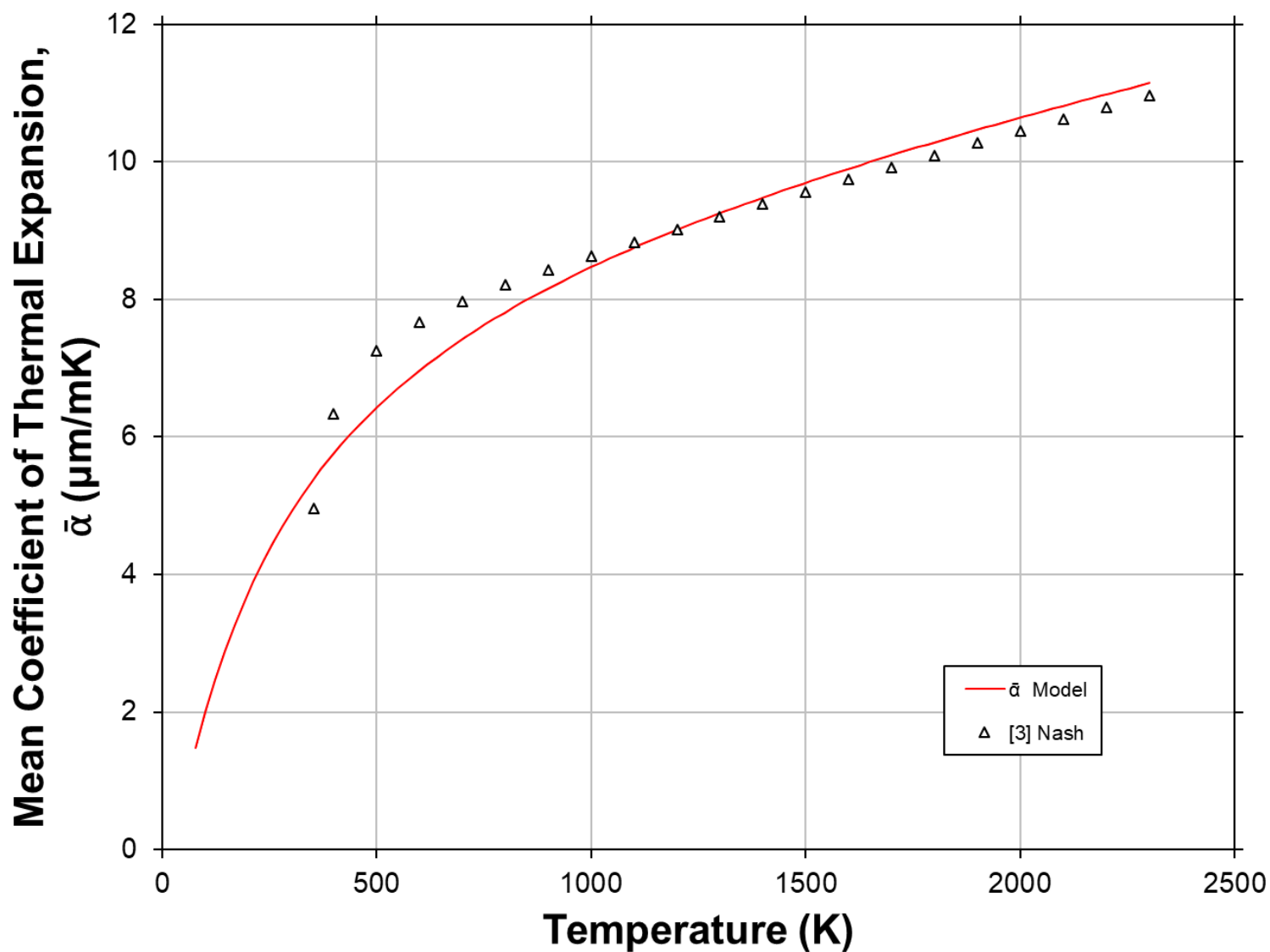


Figure 7.1.1-4: Mean Coefficient of Thermal Expansion versus Temperature of Beryllia calculated trend with comparison to Nash (2005).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 79                | ( -317.5 ) | 1.484                                 | ( 0.825 )                                           | 900               | ( 1160.3 ) | 8.161                                 | ( 4.534 )                                           |
| 100               | ( -279.7 ) | 1.970                                 | ( 1.094 )                                           | 1000              | ( 1340.3 ) | 8.471                                 | ( 4.706 )                                           |
| 150               | ( -189.7 ) | 2.940                                 | ( 1.634 )                                           | 1100              | ( 1520.3 ) | 8.752                                 | ( 4.862 )                                           |
| 200               | ( -99.7 )  | 3.718                                 | ( 2.065 )                                           | 1200              | ( 1700.3 ) | 9.012                                 | ( 5.007 )                                           |
| 250               | ( -9.7 )   | 4.357                                 | ( 2.421 )                                           | 1300              | ( 1880.3 ) | 9.254                                 | ( 5.141 )                                           |
| 300               | ( 80.3 )   | 4.895                                 | ( 2.719 )                                           | 1400              | ( 2060.3 ) | 9.481                                 | ( 5.267 )                                           |
| 350               | ( 170.3 )  | 5.356                                 | ( 2.975 )                                           | 1500              | ( 2240.3 ) | 9.696                                 | ( 5.387 )                                           |
| 400               | ( 260.3 )  | 5.756                                 | ( 3.198 )                                           | 1600              | ( 2420.3 ) | 9.900                                 | ( 5.500 )                                           |
| 450               | ( 350.3 )  | 6.110                                 | ( 3.394 )                                           | 1700              | ( 2600.3 ) | 10.096                                | ( 5.609 )                                           |
| 500               | ( 440.3 )  | 6.425                                 | ( 3.570 )                                           | 1800              | ( 2780.3 ) | 10.284                                | ( 5.714 )                                           |
| 550               | ( 530.3 )  | 6.709                                 | ( 3.727 )                                           | 1900              | ( 2960.3 ) | 10.466                                | ( 5.815 )                                           |
| 600               | ( 620.3 )  | 6.968                                 | ( 3.871 )                                           | 2000              | ( 3140.3 ) | 10.643                                | ( 5.913 )                                           |
| 650               | ( 710.3 )  | 7.204                                 | ( 4.002 )                                           | 2100              | ( 3320.3 ) | 10.814                                | ( 6.008 )                                           |
| 700               | ( 800.3 )  | 7.423                                 | ( 4.124 )                                           | 2200              | ( 3500.3 ) | 10.981                                | ( 6.101 )                                           |
| 800               | ( 980.3 )  | 7.815                                 | ( 4.342 )                                           | 2300              | ( 3680.3 ) | 11.145                                | ( 6.191 )                                           |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated from thermal expansion trend. Calculated data is fitted with the equation below to approximate property trend with respect to temperature. Trend is compared against data collected from reference [3].

## Fit Equation:

$$\bar{\alpha}(T) = \left[ A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right] / \left[ A_0 + \left( \frac{T}{1000} \right) \right]$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]: 79 ≤ T ≤ 2301

A0 = -0.2658

A1 = 9.845

A2 = 1.186

A<sub>0</sub> = 0.2709



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Specific Heat with Temperature

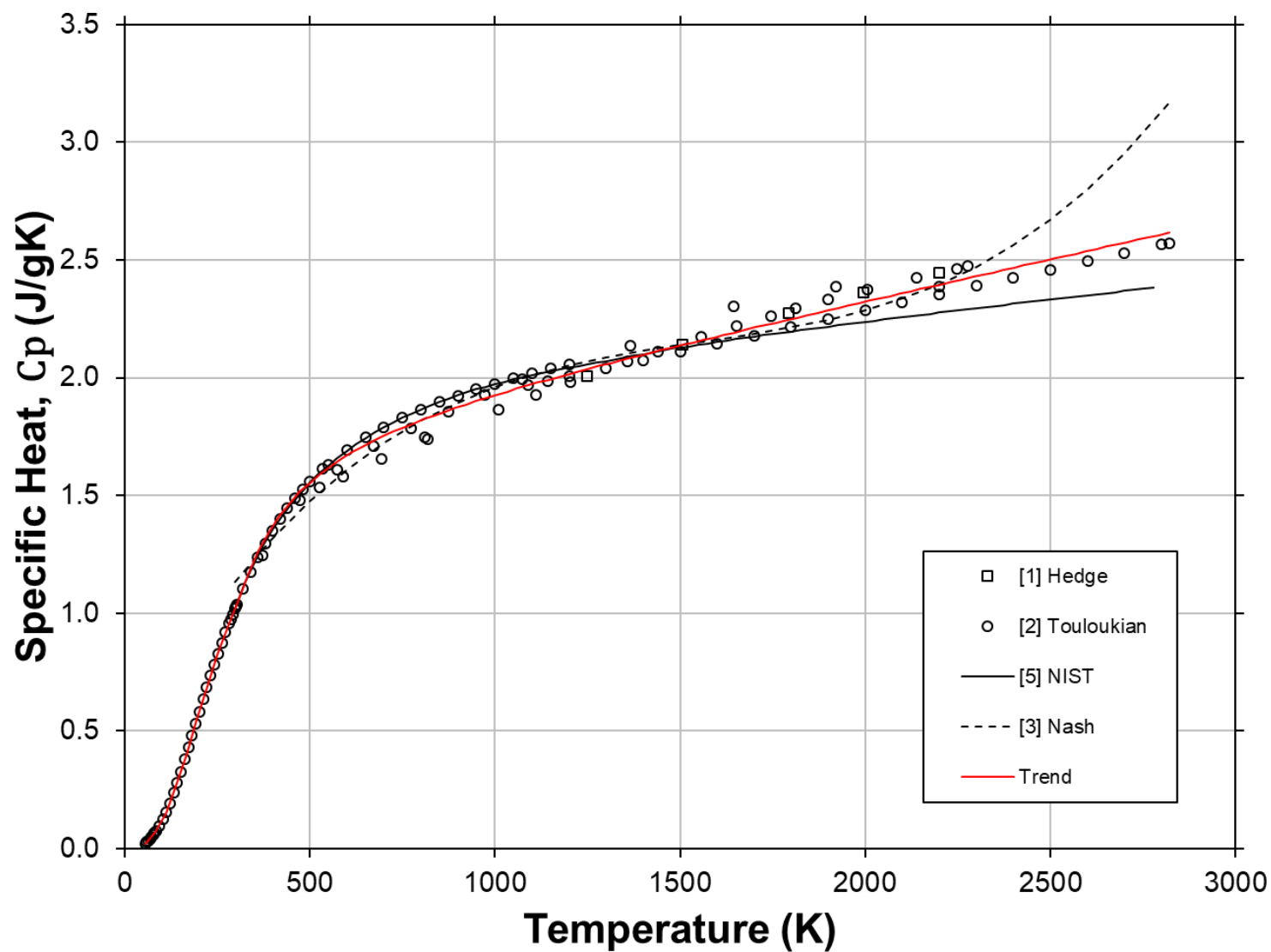


Figure 7.1.1-5: Specific Heat versus Temperature of Beryllia. Displaying fitted trend of the data with comparison to Nash (2005) and the NIST Webbook.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 55                | ( -360.7 ) | 0.022                            | ( 0.005 )       | 1100              | ( 1520.3 ) | 1.972                            | ( 0.471 )       |
| 100               | ( -279.7 ) | 0.115                            | ( 0.027 )       | 1200              | ( 1700.3 ) | 2.016                            | ( 0.482 )       |
| 150               | ( -189.7 ) | 0.318                            | ( 0.076 )       | 1300              | ( 1880.3 ) | 2.057                            | ( 0.492 )       |
| 200               | ( -99.7 )  | 0.573                            | ( 0.137 )       | 1400              | ( 2060.3 ) | 2.097                            | ( 0.501 )       |
| 250               | ( -9.7 )   | 0.818                            | ( 0.195 )       | 1500              | ( 2240.3 ) | 2.137                            | ( 0.511 )       |
| 300               | ( 80.3 )   | 1.014                            | ( 0.242 )       | 1600              | ( 2420.3 ) | 2.175                            | ( 0.520 )       |
| 400               | ( 260.3 )  | 1.371                            | ( 0.328 )       | 1700              | ( 2600.3 ) | 2.213                            | ( 0.529 )       |
| 500               | ( 440.3 )  | 1.554                            | ( 0.371 )       | 1800              | ( 2780.3 ) | 2.250                            | ( 0.538 )       |
| 600               | ( 620.3 )  | 1.670                            | ( 0.399 )       | 2000              | ( 3140.3 ) | 2.323                            | ( 0.555 )       |
| 700               | ( 800.3 )  | 1.753                            | ( 0.419 )       | 2200              | ( 3500.3 ) | 2.395                            | ( 0.573 )       |
| 800               | ( 980.3 )  | 1.820                            | ( 0.435 )       | 2400              | ( 3860.3 ) | 2.467                            | ( 0.590 )       |
| 900               | ( 1160.3 ) | 1.876                            | ( 0.448 )       | 2600              | ( 4220.3 ) | 2.538                            | ( 0.607 )       |
| 1000              | ( 1340.3 ) | 1.926                            | ( 0.460 )       | 2820              | ( 4616.3 ) | 2.616                            | ( 0.625 )       |

**Application Notes:** Data for specific heat is collected from references [1, 2] and fitted with the equations below to approximate property trend with respect to temperature. This trend is compared against trends from references [3, 5]

## Fit Equation:

For temperature range:  $55 \leq T < 293$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $293 \leq T \leq 2820$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

T. Range [K]:  $55 \leq T < 293$

$$N = 2.715$$

$$A_0 = 47.44$$

$$A_1 = -4.321$$

$$A_2 = 22.82$$

$293 \leq T \leq 2820$

$$B_0 = 1.645$$

$$B_1 = 0.3473$$

$$B_2 = -0.06613$$



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Young's Modulus with Temperature

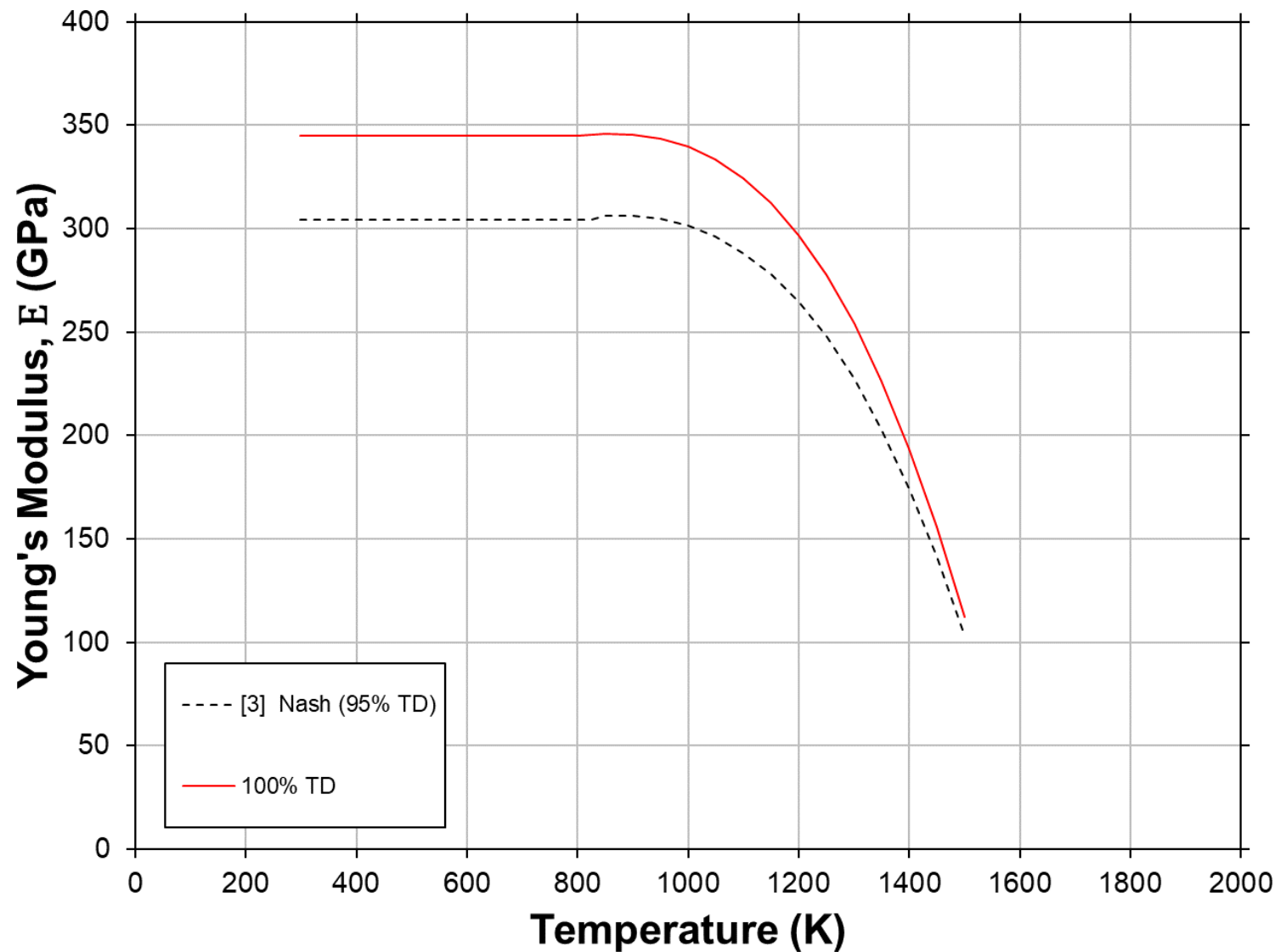


Figure 7.1.1-6: Young's Modulus versus Temperature of Beryllia for 100% from 95% TD as adapted from Nash (2005).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|-----------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )    | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 298               | ( 76.7 )  | 344.70                | ( 50.02 ) | 800               | ( 980.3 )  | 344.70                | ( 50.02 ) |
| 300               | ( 80.3 )  | 344.70                | ( 50.02 ) | 850               | ( 1070.3 ) | 345.72                | ( 50.16 ) |
| 325               | ( 125.3 ) | 344.70                | ( 50.02 ) | 900               | ( 1160.3 ) | 345.40                | ( 50.12 ) |
| 350               | ( 170.3 ) | 344.70                | ( 50.02 ) | 950               | ( 1250.3 ) | 343.45                | ( 49.83 ) |
| 375               | ( 215.3 ) | 344.70                | ( 50.02 ) | 1000              | ( 1340.3 ) | 339.53                | ( 49.27 ) |
| 400               | ( 260.3 ) | 344.70                | ( 50.02 ) | 1050              | ( 1430.3 ) | 333.27                | ( 48.36 ) |
| 425               | ( 305.3 ) | 344.70                | ( 50.02 ) | 1100              | ( 1520.3 ) | 324.31                | ( 47.06 ) |
| 450               | ( 350.3 ) | 344.70                | ( 50.02 ) | 1150              | ( 1610.3 ) | 312.29                | ( 45.31 ) |
| 475               | ( 395.3 ) | 344.70                | ( 50.02 ) | 1200              | ( 1700.3 ) | 296.85                | ( 43.07 ) |
| 500               | ( 440.3 ) | 344.70                | ( 50.02 ) | 1250              | ( 1790.3 ) | 277.63                | ( 40.28 ) |
| 550               | ( 530.3 ) | 344.70                | ( 50.02 ) | 1300              | ( 1880.3 ) | 254.27                | ( 36.89 ) |
| 600               | ( 620.3 ) | 344.70                | ( 50.02 ) | 1350              | ( 1970.3 ) | 226.41                | ( 32.85 ) |
| 650               | ( 710.3 ) | 344.70                | ( 50.02 ) | 1400              | ( 2060.3 ) | 193.70                | ( 28.11 ) |
| 700               | ( 800.3 ) | 344.70                | ( 50.02 ) | 1450              | ( 2150.3 ) | 155.76                | ( 22.60 ) |
| 750               | ( 890.3 ) | 344.70                | ( 50.02 ) | 1500              | ( 2240.3 ) | 112.24                | ( 16.29 ) |

**Application Notes:** Equations for Young's modulus are taken from reference [3], and shown below to approximate property trend as a function of both temperature and porosity.

## Fit Equations:

For temperature range:  $298 \leq T \leq 800$

$$E(T) = A0 + A1 \cdot P$$

For temperature range:  $800 < T \leq 1500$

$$E(T) = (A0 + A1 \cdot P) \cdot \left( B0 + B1 \cdot \left( \frac{T}{1000} \right) + B2 \cdot \left( \frac{T}{1000} \right)^2 + B3 \cdot \left( \frac{T}{1000} \right)^3 \right)$$

$$E(T) = \text{Young's Modulus [GPa]}$$

$T = \text{Temperature [K]}$

$P = \text{Frantional Porosity, } 0 < P < 0.30$

## Constants:

T. Range [K]:  $298 \leq T \leq 800$

$$A0 = 344.7$$

$$A1 = -805.7$$

$800 < T \leq 1500$

$$B0 = 1.313$$

$$B1 = -1.757$$

$$B2 = 2.822$$

$$B3 = -1.393$$



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Compressive Strength with Temperature

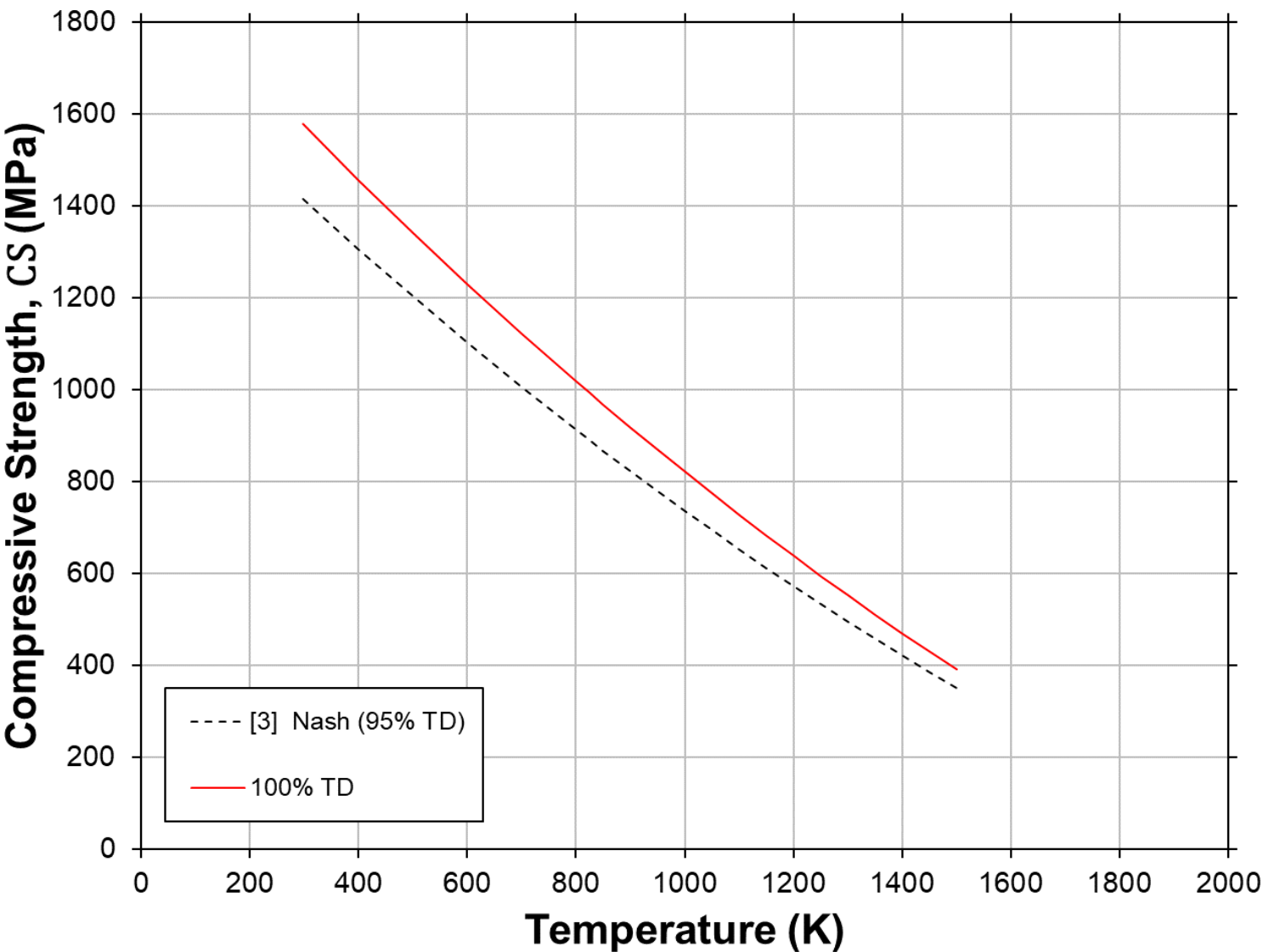


Figure 7.1.1-7: Compressive Strength versus Temperature of Beryllia, 100% TD and 95% TD as adapted from Nash (2005).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Compressive Strength with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Compressive Strength ( CS ) |            | Temperature ( T ) |            | Compressive Strength ( CS ) |            |
|-------------------|-----------|-----------------------------|------------|-------------------|------------|-----------------------------|------------|
| K                 | ( °F )    | MPa                         | ( Ksi )    | K                 | ( °F )     | MPa                         | ( Ksi )    |
| 298               | ( 76.7 )  | 1577.52                     | ( 228.90 ) | 800               | ( 980.3 )  | 1018.15                     | ( 147.73 ) |
| 300               | ( 80.3 )  | 1575.11                     | ( 228.55 ) | 850               | ( 1070.3 ) | 967.44                      | ( 140.37 ) |
| 325               | ( 125.3 ) | 1545.11                     | ( 224.20 ) | 900               | ( 1160.3 ) | 917.62                      | ( 133.15 ) |
| 350               | ( 170.3 ) | 1515.34                     | ( 219.88 ) | 950               | ( 1250.3 ) | 868.72                      | ( 126.05 ) |
| 375               | ( 215.3 ) | 1485.79                     | ( 215.59 ) | 1000              | ( 1340.3 ) | 820.71                      | ( 119.09 ) |
| 400               | ( 260.3 ) | 1456.48                     | ( 211.33 ) | 1050              | ( 1430.3 ) | 773.62                      | ( 112.25 ) |
| 425               | ( 305.3 ) | 1427.38                     | ( 207.11 ) | 1100              | ( 1520.3 ) | 727.42                      | ( 105.55 ) |
| 450               | ( 350.3 ) | 1398.52                     | ( 202.92 ) | 1150              | ( 1610.3 ) | 682.14                      | ( 98.98 )  |
| 475               | ( 395.3 ) | 1369.88                     | ( 198.77 ) | 1200              | ( 1700.3 ) | 637.75                      | ( 92.54 )  |
| 500               | ( 440.3 ) | 1341.46                     | ( 194.65 ) | 1250              | ( 1790.3 ) | 594.28                      | ( 86.23 )  |
| 550               | ( 530.3 ) | 1285.32                     | ( 186.50 ) | 1300              | ( 1880.3 ) | 551.70                      | ( 80.05 )  |
| 600               | ( 620.3 ) | 1230.07                     | ( 178.48 ) | 1350              | ( 1970.3 ) | 510.04                      | ( 74.01 )  |
| 650               | ( 710.3 ) | 1175.74                     | ( 170.60 ) | 1400              | ( 2060.3 ) | 469.27                      | ( 68.09 )  |
| 700               | ( 800.3 ) | 1122.30                     | ( 162.85 ) | 1450              | ( 2150.3 ) | 429.42                      | ( 62.31 )  |
| 750               | ( 890.3 ) | 1069.78                     | ( 155.22 ) | 1500              | ( 2240.3 ) | 390.46                      | ( 56.66 )  |

**Application Notes:** Equations for compressive strength are taken from reference [3], and shown below to approximate property trend as a function of both temperature and porosity.

## Fit Equations:

$$CS(T) = (A0 + A1 \cdot P) \cdot \left( B0 + B1 \cdot \left( \frac{T}{1000} \right) + B2 \cdot \left( \frac{T}{1000} \right)^2 \right)$$

$CS(T)$  = Compressive Strength [MPa]

$T$  = Temperature [K]

$P$  = Frantional Porosity,  $0 < P < 0.30$

## Constants:

T Range [K]:  $298 \leq T \leq 1500$

|      |       |      |         |
|------|-------|------|---------|
| A0 = | 1585  | B0 = | 1.232   |
| A1 = | -3273 | B1 = | -0.8284 |
|      |       | B2 = | 0.1142  |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical                    |                                                             | Thermal                                     |                                           |                                                       |                                       | Elastic                            |                                  |                                 |
|---------------------|--------------------------------------------|-------------------------------------------------------------|---------------------------------------------|-------------------------------------------|-------------------------------------------------------|---------------------------------------|------------------------------------|----------------------------------|---------------------------------|
|                     | Density<br>( $\rho$ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( $\rho_e$ )<br>$\mu\Omega$ -m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( $dL/L_0$ )<br>% | Mean CTE<br>( $\bar{\alpha}$ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( $C_p$ )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( $\nu$ ) |
| 100                 | 3014                                       |                                                             | -                                           | -0.042                                    | 1.97                                                  | 0.115                                 | -                                  |                                  |                                 |
| 200                 | 3013                                       |                                                             | 421.93                                      | -0.031                                    | 3.72                                                  | 0.573                                 | -                                  |                                  |                                 |
| 250                 | 3012                                       |                                                             | 336.46                                      | -0.018                                    | 4.36                                                  | 0.818                                 | -                                  |                                  |                                 |
| 300                 | 3009                                       |                                                             | 273.13                                      | 0.006                                     | 4.89                                                  | 1.014                                 | 344.70                             |                                  |                                 |
| 350                 | 3007                                       |                                                             | 225.36                                      | 0.036                                     | 5.36                                                  | 1.227                                 | 344.70                             |                                  |                                 |
| 400                 | 3004                                       |                                                             | 188.67                                      | 0.069                                     | 5.76                                                  | 1.371                                 | 344.70                             |                                  |                                 |
| 500                 | 2997                                       |                                                             | 137.24                                      | 0.139                                     | 6.43                                                  | 1.554                                 | 344.70                             |                                  |                                 |
| 600                 | 2991                                       |                                                             | 103.96                                      | 0.216                                     | 6.97                                                  | 1.670                                 | 344.70                             |                                  |                                 |
| 700                 | 2983                                       |                                                             | 81.32                                       | 0.301                                     | 7.42                                                  | 1.753                                 | 344.70                             |                                  |                                 |
| 800                 | 2975                                       |                                                             | 65.27                                       | 0.392                                     | 7.82                                                  | 1.820                                 | 344.70                             |                                  |                                 |
| 900                 | 2966                                       |                                                             | 53.51                                       | 0.489                                     | 8.16                                                  | 1.876                                 | 345.40                             |                                  |                                 |
| 1000                | 2957                                       |                                                             | 44.64                                       | 0.591                                     | 8.47                                                  | 1.926                                 | 339.53                             |                                  |                                 |
| 1100                | 2948                                       |                                                             | 37.80                                       | 0.699                                     | 8.75                                                  | 1.972                                 | 324.31                             |                                  |                                 |
| 1200                | 2938                                       |                                                             | 32.41                                       | 0.811                                     | 9.01                                                  | 2.016                                 | 296.85                             |                                  |                                 |
| 1300                | 2928                                       |                                                             | 28.10                                       | 0.927                                     | 9.25                                                  | 2.057                                 | 254.27                             |                                  |                                 |
| 1400                | 2917                                       |                                                             | 24.59                                       | 1.047                                     | 9.48                                                  | 2.097                                 | 193.70                             |                                  |                                 |
| 1500                | 2907                                       |                                                             | 21.69                                       | 1.171                                     | 9.70                                                  | 2.137                                 | 112.24                             |                                  |                                 |
| 1600                | 2896                                       |                                                             | 19.28                                       | 1.297                                     | 9.90                                                  | 2.175                                 | -                                  |                                  |                                 |
| 1700                | 2885                                       |                                                             | 17.25                                       | 1.426                                     | 10.10                                                 | 2.213                                 | -                                  |                                  |                                 |
| 1800                | 2874                                       |                                                             | 15.53                                       | 1.556                                     | 10.28                                                 | 2.250                                 | -                                  |                                  |                                 |
| 1900                | 2863                                       |                                                             | 14.05                                       | 1.688                                     | 10.47                                                 | 2.287                                 | -                                  |                                  |                                 |
| 2000                | 2851                                       |                                                             | 12.77                                       | 1.821                                     | 10.64                                                 | 2.323                                 | -                                  |                                  |                                 |
| 2100                | 2840                                       |                                                             | 11.66                                       | 1.955                                     | 10.81                                                 | 2.359                                 | -                                  |                                  |                                 |
| 2200                | 2829                                       |                                                             | 10.69                                       | 2.089                                     | 10.98                                                 | 2.395                                 | -                                  |                                  |                                 |
| 2300                | 2818                                       |                                                             | 9.84                                        | 2.222                                     | 11.14                                                 | 2.431                                 | -                                  |                                  |                                 |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.1 Beryllium Oxide (BeO)

**Revision 0: 08-05-2020**

**References**

- [1] J.C. Hedge, C. Kostenko, J.I. Lang, Thermal Properties of Refractory Alloys, IIT Research Institute, Chicago, IL, 1963.
- [2] Y.S. Touloukian, E.H. Buyco, Specific Heat - Nonmetallic Solids, Thermophysical Properties of Matter - the TPRC Data Series, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1970.
- [3] J. Nash, Reflector and Shield Material Properties for Project Prometheus, Knolls Atomic Power Laboratory (KAPL), Niskayuna, NY, 2005.
- [4] Y.S. Touloukian, R.K. Kirby, E.R. Taylor, T.Y.R. Lee, Thermal Expansion - Nonmetallic Solids, Thermophysical Properties of Matter - the TPRC Data Series, Volume 13, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1977.
- [5] NIST Webbook. <https://webbook.nist.gov> (Accessed July 2019).

## **7 Refractory Ceramics**

### **7.1 Oxides**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 2.1: 08-25-2023

General

### Room Temperature Properties

|                                           |              |
|-------------------------------------------|--------------|
| Theoretical Density, [kg/m <sup>3</sup> ] | 6014         |
| Melting Point, [K]                        | 2823 to 2973 |
| Specific Heat, [J/(g-K)]                  | 0.465        |
| Thermal Conductivity, [W/(m-K)]           | 1.9          |
| Linear expansion coefficient, [μm/(m-K)]  | 6.85         |
| Young's Modulus, [GPa]                    | 226.1        |
| Shear Modulus, [GPa]                      | 86.4         |
| Poisson's Ratio, [-]                      | 0.308        |

### Zirconia - Yttria Phase Diagram

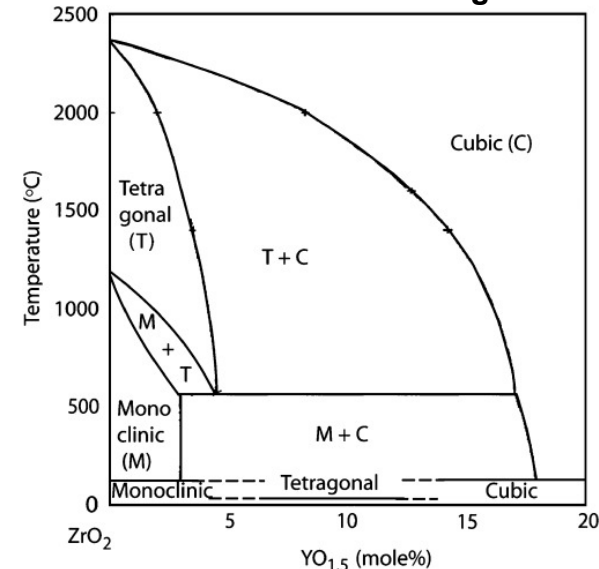


Figure 7.1.2-1: Zirconia - Yttria Phase Diagram [1]

### Composition

Table 7.1.2-1: Typical Composition ranges for yttria stabilized zirconia (percent by weight).

| Grade |                     | ZrO <sub>2</sub> | Y <sub>2</sub> O <sub>3</sub> | Other |
|-------|---------------------|------------------|-------------------------------|-------|
| YSZ   | Low Yttria Content  | 97               | 3                             | -     |
|       | High Yttria Content | 90               | 10                            | 1     |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Density with Temperature

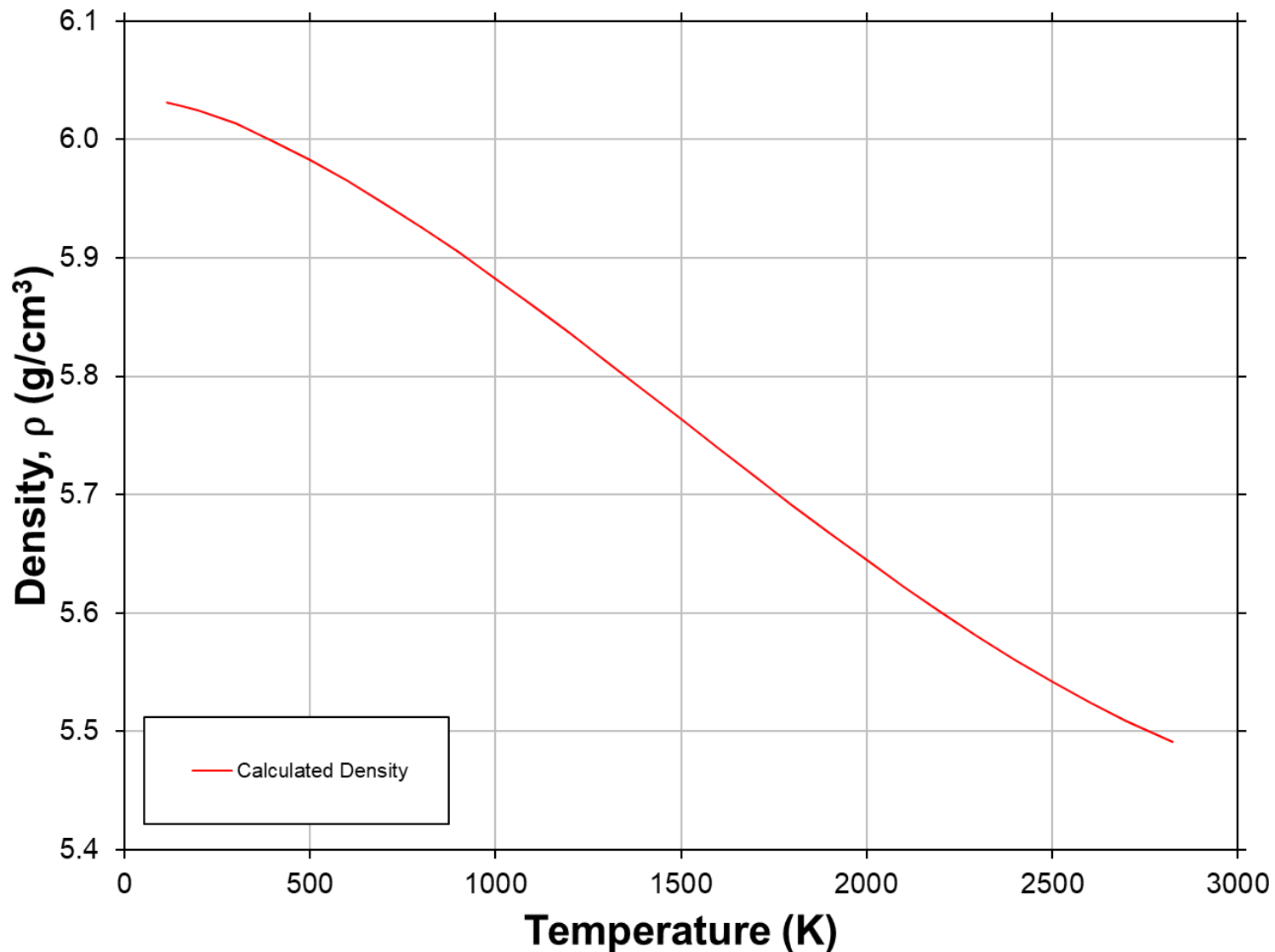


Figure 7.1.2-2: Density versus Temperature for YSZ. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 115               | ( -252.7 ) | 6031              | ( 376.5 )              | 1600              | ( 2420.3 ) | 5739              | ( 358.3 )              |
| 200               | ( -99.7 )  | 6024              | ( 376.1 )              | 1700              | ( 2600.3 ) | 5715              | ( 356.8 )              |
| 300               | ( 80.3 )   | 6013              | ( 375.4 )              | 1800              | ( 2780.3 ) | 5691              | ( 355.3 )              |
| 400               | ( 260.3 )  | 5999              | ( 374.5 )              | 1900              | ( 2960.3 ) | 5668              | ( 353.8 )              |
| 500               | ( 440.3 )  | 5983              | ( 373.5 )              | 2000              | ( 3140.3 ) | 5645              | ( 352.4 )              |
| 600               | ( 620.3 )  | 5965              | ( 372.4 )              | 2100              | ( 3320.3 ) | 5622              | ( 351.0 )              |
| 700               | ( 800.3 )  | 5946              | ( 371.2 )              | 2200              | ( 3500.3 ) | 5601              | ( 349.7 )              |
| 800               | ( 980.3 )  | 5926              | ( 370.0 )              | 2300              | ( 3680.3 ) | 5580              | ( 348.4 )              |
| 900               | ( 1160.3 ) | 5905              | ( 368.6 )              | 2400              | ( 3860.3 ) | 5560              | ( 347.1 )              |
| 1000              | ( 1340.3 ) | 5883              | ( 367.3 )              | 2500              | ( 4040.3 ) | 5542              | ( 346.0 )              |
| 1100              | ( 1520.3 ) | 5860              | ( 365.8 )              | 2600              | ( 4220.3 ) | 5525              | ( 344.9 )              |
| 1200              | ( 1700.3 ) | 5836              | ( 364.4 )              | 2700              | ( 4400.3 ) | 5509              | ( 343.9 )              |
| 1300              | ( 1880.3 ) | 5812              | ( 362.9 )              | 2800              | ( 4580.3 ) | 5495              | ( 343.0 )              |
| 1400              | ( 2060.3 ) | 5788              | ( 361.4 )              | 2824              | ( 4623.5 ) | 5491              | ( 342.8 )              |
| 1500              | ( 2240.3 ) | 5764              | ( 359.8 )              |                   |            |                   |                        |

**Application Notes:** Density trend is calculated as a function of thermal expansion as seen in the equation below to approximate property trend with respect to temperature.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 6014 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $115 \leq T \leq 2824$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

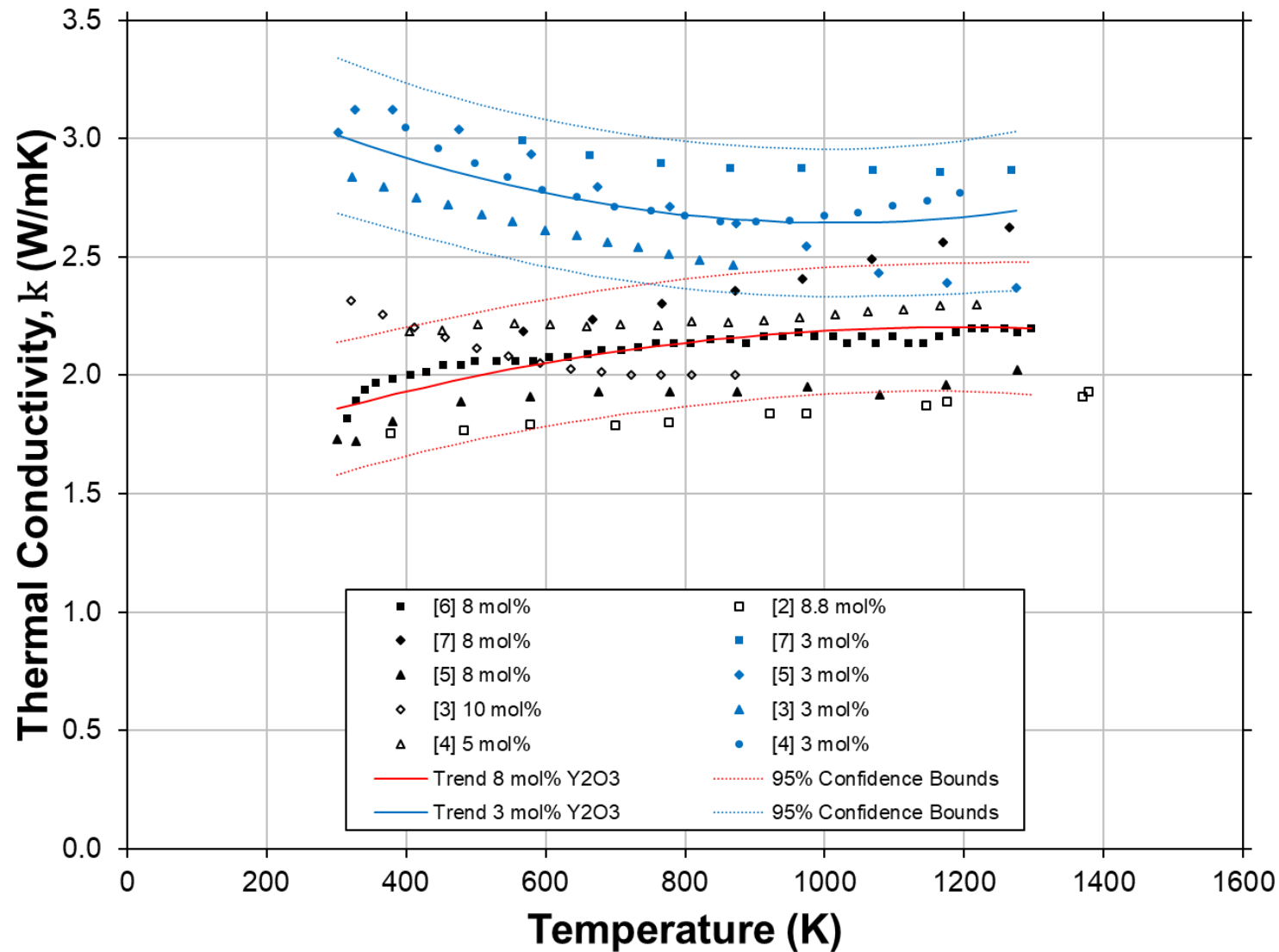


Figure 7.1.2-3: Thermal Conductivity versus Temperature of YSZ. Fitted trends of the data with 95% confidence bounds for 8 mol% and 3 mol% of  $Y_2O_3$ .



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

## 7 Refractory Ceramics

## 7.1 Oxides

## 7.1.2 Zirconia (Yttria Stabilized) (YSZ)

Revision 2: 04-26-2023

## Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Thermal Conductivity ( k ) |                                       | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       |
|-------------------|-----------|----------------------------|---------------------------------------|-------------------|------------|----------------------------|---------------------------------------|
| K                 | ( °F )    | W/(m·K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) |
| 300               | ( 80.3 )  | 1.86                       | ( 12.89 )                             | 650               | ( 710.3 )  | 2.08                       | ( 14.41 )                             |
| 325               | ( 125.3 ) | 1.88                       | ( 13.03 )                             | 700               | ( 800.3 )  | 2.10                       | ( 14.56 )                             |
| 350               | ( 170.3 ) | 1.90                       | ( 13.16 )                             | 750               | ( 890.3 )  | 2.12                       | ( 14.71 )                             |
| 375               | ( 215.3 ) | 1.91                       | ( 13.28 )                             | 800               | ( 980.3 )  | 2.14                       | ( 14.83 )                             |
| 400               | ( 260.3 ) | 1.93                       | ( 13.40 )                             | 850               | ( 1070.3 ) | 2.15                       | ( 14.94 )                             |
| 425               | ( 305.3 ) | 1.95                       | ( 13.52 )                             | 900               | ( 1160.3 ) | 2.17                       | ( 15.04 )                             |
| 450               | ( 350.3 ) | 1.97                       | ( 13.63 )                             | 950               | ( 1250.3 ) | 2.18                       | ( 15.12 )                             |
| 475               | ( 395.3 ) | 1.98                       | ( 13.74 )                             | 1000              | ( 1340.3 ) | 2.19                       | ( 15.18 )                             |
| 500               | ( 440.3 ) | 2.00                       | ( 13.85 )                             | 1050              | ( 1430.3 ) | 2.20                       | ( 15.23 )                             |
| 525               | ( 485.3 ) | 2.01                       | ( 13.95 )                             | 1100              | ( 1520.3 ) | 2.20                       | ( 15.27 )                             |
| 550               | ( 530.3 ) | 2.03                       | ( 14.05 )                             | 1150              | ( 1610.3 ) | 2.20                       | ( 15.29 )                             |
| 575               | ( 575.3 ) | 2.04                       | ( 14.15 )                             | 1200              | ( 1700.3 ) | 2.20                       | ( 15.29 )                             |
| 600               | ( 620.3 ) | 2.05                       | ( 14.24 )                             | 1250              | ( 1790.3 ) | 2.20                       | ( 15.28 )                             |
| 625               | ( 665.3 ) | 2.06                       | ( 14.33 )                             | 1297              | ( 1874.9 ) | 2.20                       | ( 15.26 )                             |

**Application Notes:** Data for thermal conductivity was collected from references [2-7] and fitted with equation (7.1.2-2) to approximate the property trend with respect to temperature for 3 mol% and 8 mol% Y<sub>2</sub>O<sub>3</sub> added to Zirconia. Data in table is trend for 8 mol% Y<sub>2</sub>O<sub>3</sub>.

### Fit Equation:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

### Constants:

| Y <sub>2</sub> O <sub>3</sub> mol%: | 3 mol%                   | 8 mol%                   |
|-------------------------------------|--------------------------|--------------------------|
| Temperature Range [K]:              | <u>300 ≤ T &lt; 1297</u> | <u>300 ≤ T &lt; 1297</u> |
| A0 =                                | 3.395                    | 1.586                    |
| A1 =                                | -1.481                   | 1.039                    |
| A2 =                                | 0.7307                   | -0.4368                  |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

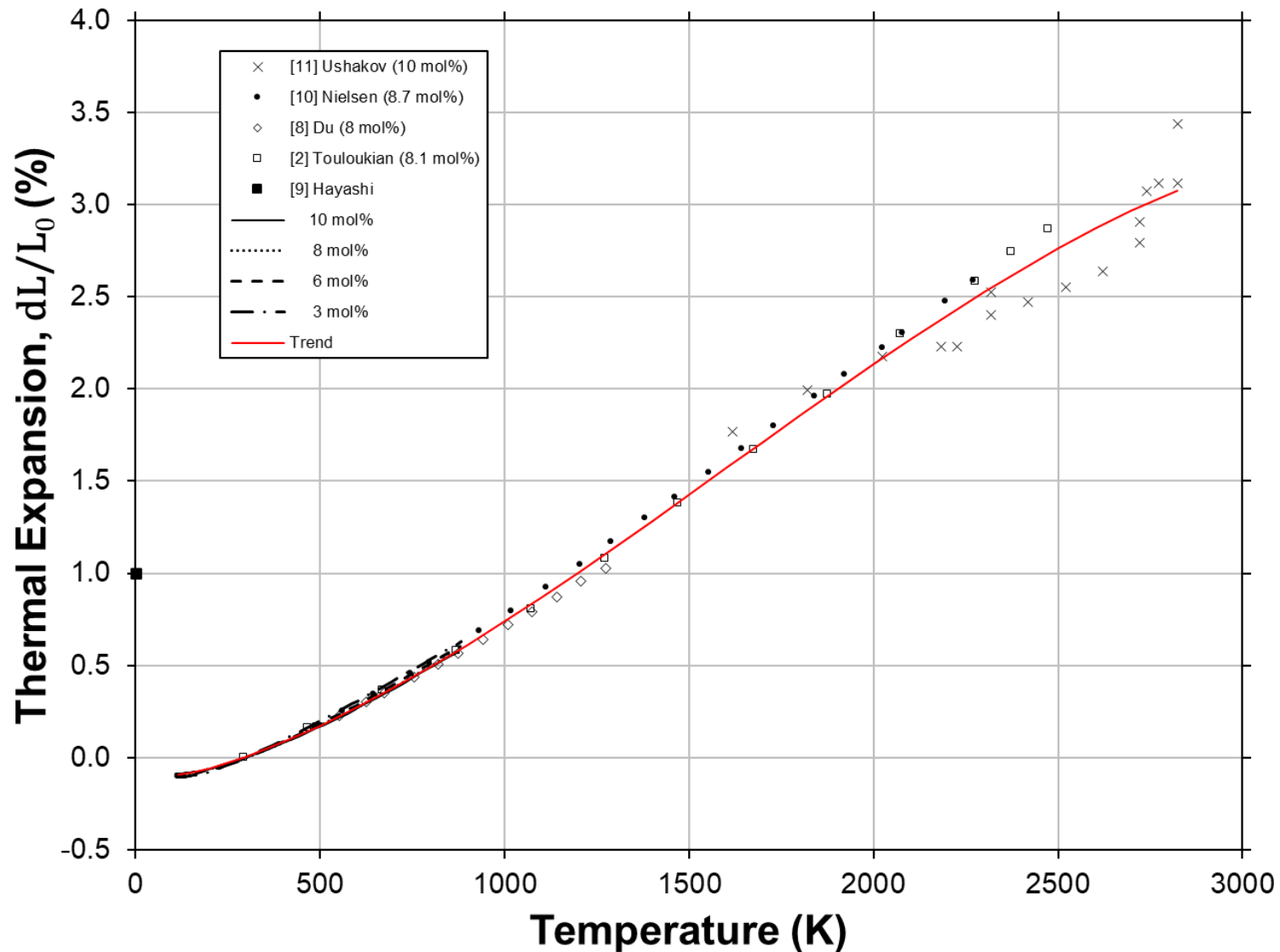


Figure 7.1.2-4: Thermal Expansion versus Temperature of YSZ.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 115               | ( -252.7 ) | -0.093                                  | 1600              | ( 2420.3 ) | 1.570                                   |
| 200               | ( -99.7 )  | -0.058                                  | 1700              | ( 2600.3 ) | 1.713                                   |
| 300               | ( 80.3 )   | 0.004                                   | 1800              | ( 2780.3 ) | 1.856                                   |
| 400               | ( 260.3 )  | 0.083                                   | 1900              | ( 2960.3 ) | 1.997                                   |
| 500               | ( 440.3 )  | 0.173                                   | 2000              | ( 3140.3 ) | 2.135                                   |
| 600               | ( 620.3 )  | 0.271                                   | 2100              | ( 3320.3 ) | 2.270                                   |
| 700               | ( 800.3 )  | 0.378                                   | 2200              | ( 3500.3 ) | 2.401                                   |
| 800               | ( 980.3 )  | 0.492                                   | 2300              | ( 3680.3 ) | 2.528                                   |
| 900               | ( 1160.3 ) | 0.613                                   | 2400              | ( 3860.3 ) | 2.648                                   |
| 1000              | ( 1340.3 ) | 0.739                                   | 2500              | ( 4040.3 ) | 2.762                                   |
| 1100              | ( 1520.3 ) | 0.870                                   | 2600              | ( 4220.3 ) | 2.869                                   |
| 1200              | ( 1700.3 ) | 1.005                                   | 2700              | ( 4400.3 ) | 2.967                                   |
| 1300              | ( 1880.3 ) | 1.143                                   | 2800              | ( 4580.3 ) | 3.057                                   |
| 1400              | ( 2060.3 ) | 1.284                                   | 2824              | ( 4623.5 ) | 3.077                                   |
| 1500              | ( 2240.3 ) | 1.427                                   |                   |            |                                         |

**Application Notes:** Data for thermal expansion is collected from references [2, 8-11] and fitted with the equation below to approximate property trend with respect to temperature. The thermal expansion fit is valid between 8 and 10 mol% yttria.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$dL/L_0(T)$  = Thermal Expansion, Percent [%]

$T$  = Temperature [K]

## Constants:

| Temperature<br>Range [K]: | $115 \leq T \leq 293$ | $293 < T \leq 2824$ |
|---------------------------|-----------------------|---------------------|
| A0 =                      | -0.1168               | -0.1689             |
| A1 =                      | 0.08241               | 0.392               |
| A2 =                      | 1.056                 | 0.6514              |
| A3 =                      | 0                     | -0.1357             |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

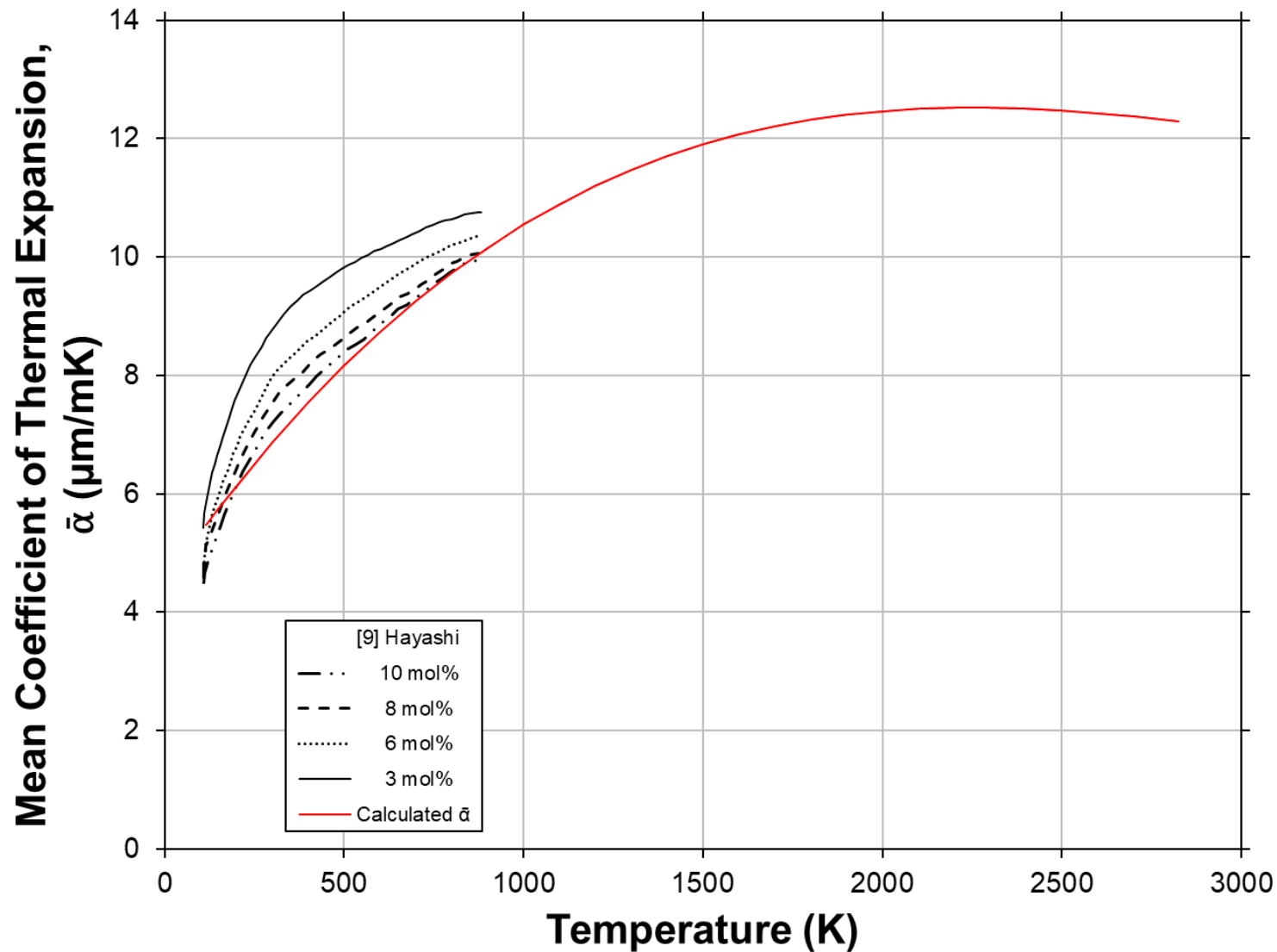


Figure 7.1.2-5: Mean Coefficient of Thermal Expansion verse Temperature of YSZ. Calculated from fitted trend of the Thermal Expansion data with comparison to Hayashi (2005).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 115               | ( -252.7 ) | 5.470                                 | ( 3.039 )                                           | 1600              | ( 2420.3 ) | 12.071                                | ( 6.706 )                                           |
| 200               | ( -99.7 )  | 6.135                                 | ( 3.408 )                                           | 1700              | ( 2600.3 ) | 12.209                                | ( 6.783 )                                           |
| 300               | ( 80.3 )   | 6.865                                 | ( 3.814 )                                           | 1800              | ( 2780.3 ) | 12.320                                | ( 6.844 )                                           |
| 400               | ( 260.3 )  | 7.540                                 | ( 4.189 )                                           | 1900              | ( 2960.3 ) | 12.404                                | ( 6.891 )                                           |
| 500               | ( 440.3 )  | 8.163                                 | ( 4.535 )                                           | 2000              | ( 3140.3 ) | 12.465                                | ( 6.925 )                                           |
| 600               | ( 620.3 )  | 8.734                                 | ( 4.852 )                                           | 2100              | ( 3320.3 ) | 12.503                                | ( 6.946 )                                           |
| 700               | ( 800.3 )  | 9.256                                 | ( 5.142 )                                           | 2200              | ( 3500.3 ) | 12.521                                | ( 6.956 )                                           |
| 800               | ( 980.3 )  | 9.732                                 | ( 5.407 )                                           | 2300              | ( 3680.3 ) | 12.521                                | ( 6.956 )                                           |
| 900               | ( 1160.3 ) | 10.162                                | ( 5.646 )                                           | 2400              | ( 3860.3 ) | 12.504                                | ( 6.947 )                                           |
| 1000              | ( 1340.3 ) | 10.549                                | ( 5.861 )                                           | 2500              | ( 4040.3 ) | 12.474                                | ( 6.930 )                                           |
| 1100              | ( 1520.3 ) | 10.895                                | ( 6.053 )                                           | 2600              | ( 4220.3 ) | 12.430                                | ( 6.906 )                                           |
| 1200              | ( 1700.3 ) | 11.201                                | ( 6.223 )                                           | 2700              | ( 4400.3 ) | 12.376                                | ( 6.876 )                                           |
| 1300              | ( 1880.3 ) | 11.470                                | ( 6.372 )                                           | 2800              | ( 4580.3 ) | 12.314                                | ( 6.841 )                                           |
| 1400              | ( 2060.3 ) | 11.703                                | ( 6.502 )                                           | 2824              | ( 4623.5 ) | 12.298                                | ( 6.832 )                                           |
| 1500              | ( 2240.3 ) | 11.903                                | ( 6.613 )                                           |                   |            |                                       |                                                     |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated as a function of thermal expansion. This calculated data is then fitted with the equation below to approximate property trend with respect to temperature and compared to data from reference [9]. The coefficient of thermal expansion fit is valid between 8 and 10 mol% yttria.

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

Temp. Range [K]:  $115 \leq T \leq 2824$

A0 = 4.503

A1 = 8.752

A2 = -3.026

A3 = 0.3202



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Specific Heat with Temperature

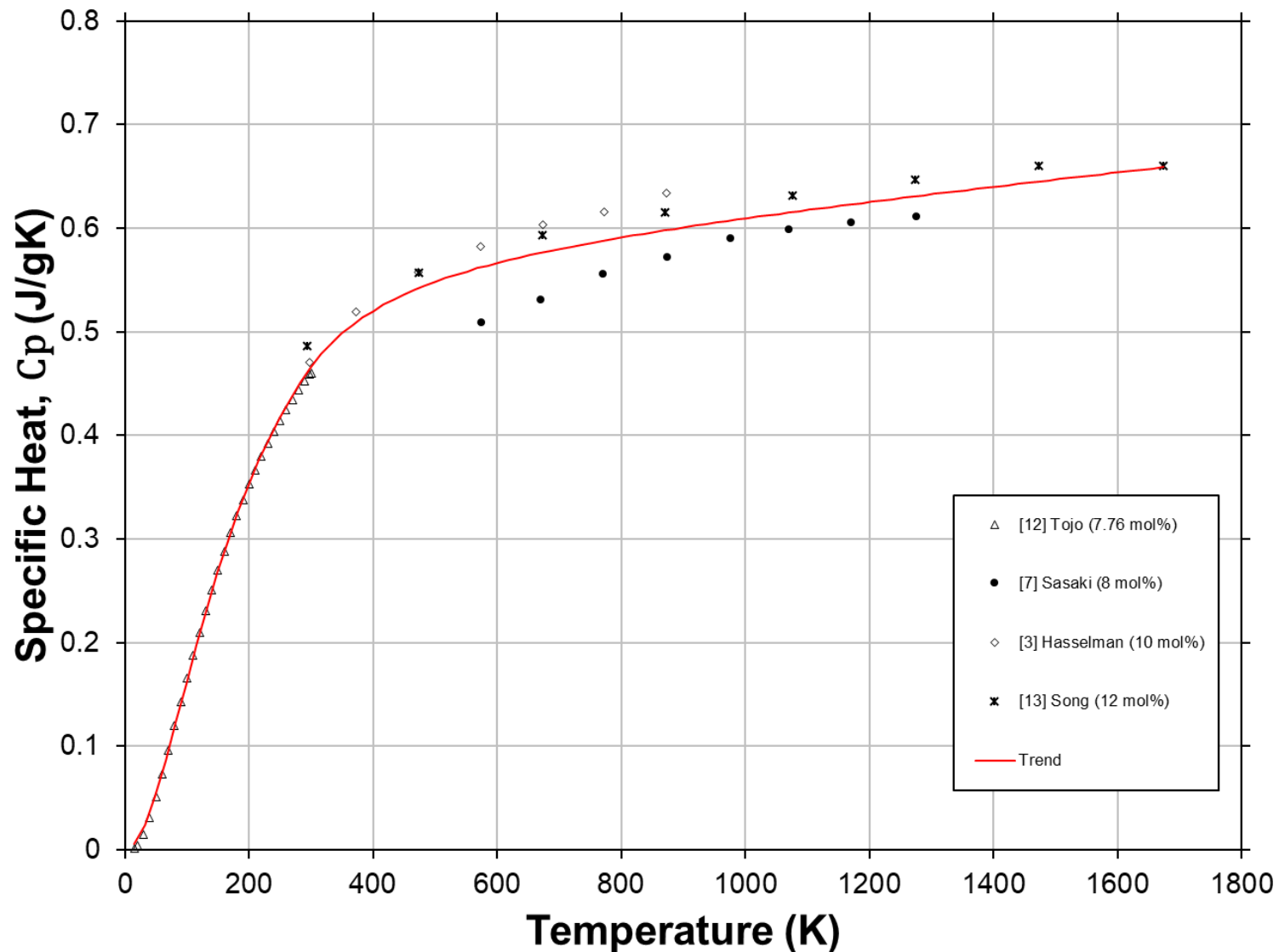


Figure 7.1.2-6: Specific Heat versus Temperature of YSZ.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 50                | ( -369.7 ) | 0.055                            | ( 0.013 )       | 750               | ( 890.3 )  | 0.586                            | ( 0.140 )       |
| 100               | ( -279.7 ) | 0.164                            | ( 0.039 )       | 800               | ( 980.3 )  | 0.591                            | ( 0.141 )       |
| 150               | ( -189.7 ) | 0.269                            | ( 0.064 )       | 850               | ( 1070.3 ) | 0.596                            | ( 0.142 )       |
| 200               | ( -99.7 )  | 0.354                            | ( 0.085 )       | 900               | ( 1160.3 ) | 0.601                            | ( 0.144 )       |
| 250               | ( -9.7 )   | 0.418                            | ( 0.100 )       | 950               | ( 1250.3 ) | 0.605                            | ( 0.145 )       |
| 300               | ( 80.3 )   | 0.467                            | ( 0.112 )       | 1000              | ( 1340.3 ) | 0.609                            | ( 0.146 )       |
| 350               | ( 170.3 )  | 0.498                            | ( 0.119 )       | 1050              | ( 1430.3 ) | 0.614                            | ( 0.147 )       |
| 400               | ( 260.3 )  | 0.520                            | ( 0.124 )       | 1100              | ( 1520.3 ) | 0.618                            | ( 0.148 )       |
| 450               | ( 350.3 )  | 0.536                            | ( 0.128 )       | 1200              | ( 1700.3 ) | 0.625                            | ( 0.149 )       |
| 500               | ( 440.3 )  | 0.548                            | ( 0.131 )       | 1300              | ( 1880.3 ) | 0.633                            | ( 0.151 )       |
| 550               | ( 530.3 )  | 0.558                            | ( 0.133 )       | 1400              | ( 2060.3 ) | 0.640                            | ( 0.153 )       |
| 600               | ( 620.3 )  | 0.567                            | ( 0.135 )       | 1500              | ( 2240.3 ) | 0.647                            | ( 0.155 )       |
| 650               | ( 710.3 )  | 0.574                            | ( 0.137 )       | 1600              | ( 2420.3 ) | 0.654                            | ( 0.156 )       |
| 700               | ( 800.3 )  | 0.580                            | ( 0.139 )       | 1674              | ( 2553.5 ) | 0.659                            | ( 0.158 )       |

**Application Notes:** Data for specific heat is collected from references [3, 7, 12, 13] and fitted with the equations below to approximate property trend with respect to temperature. The specific heat fit is valid between 8 and 12 mol% yttria.

## Fit Equation:

For temperature range:  $15 \leq T < 293$

$$C_p(T) = \left[ A0 \cdot \left( \frac{T}{1000} \right)^2 \right] / \left[ 1 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $293 \leq T \leq 1674$

$$C_p(T) = B0 + B1 \cdot \left( \frac{T}{1000} \right) + B_2 / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

T = Temperature [K]

## Constants:

T. Range [K]:  $55 \leq T < 293$

A0 = 27.02

A1 = 2.726

A2 = 37.79

$293 \leq T \leq 1674$

B0 = 0.5549

B1 = 0.06425

B<sub>2</sub> = -0.009664



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

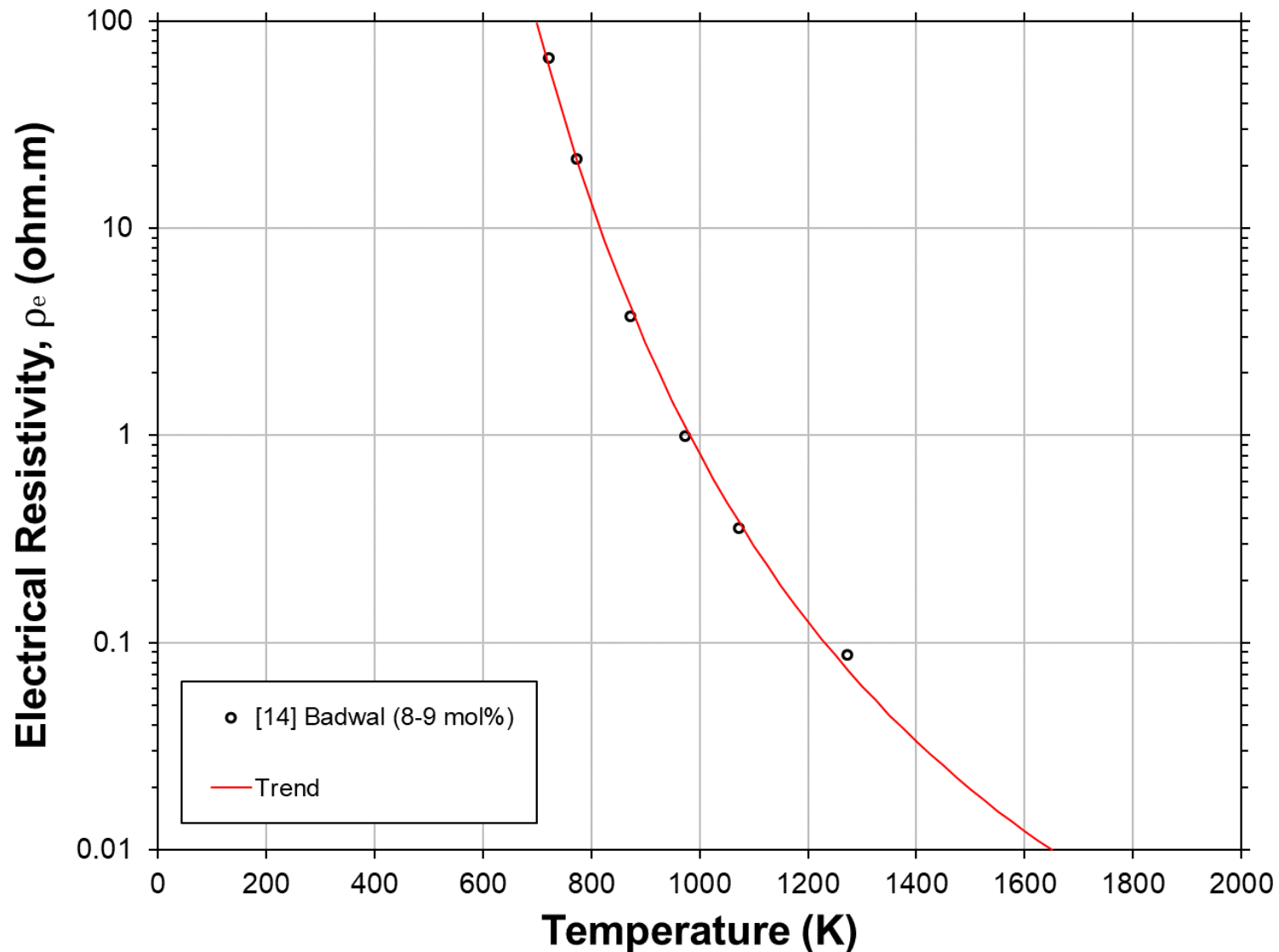


Figure 7.1.2-7: Electrical Resistivity versus Temperature of YSZ.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                  | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                  |
|-------------------|------------|-------------------------------------|------------------|-------------------|------------|-------------------------------------|------------------|
| K                 | ( °F )     | $\Omega$ -m                         | ( $\Omega$ -in ) | K                 | ( °F )     | $\Omega$ -m                         | ( $\Omega$ -in ) |
| 700               | ( 800.3 )  | 97.258                              | ( 3829.04 )      | 1025              | ( 1385.3 ) | 0.619                               | ( 24.35 )        |
| 725               | ( 845.3 )  | 56.110                              | ( 2209.06 )      | 1050              | ( 1430.3 ) | 0.477                               | ( 18.79 )        |
| 750               | ( 890.3 )  | 33.580                              | ( 1322.06 )      | 1075              | ( 1475.3 ) | 0.373                               | ( 14.67 )        |
| 775               | ( 935.3 )  | 20.774                              | ( 817.86 )       | 1100              | ( 1520.3 ) | 0.294                               | ( 11.59 )        |
| 800               | ( 980.3 )  | 13.243                              | ( 521.36 )       | 1125              | ( 1565.3 ) | 0.235                               | ( 9.25 )         |
| 825               | ( 1025.3 ) | 8.675                               | ( 341.55 )       | 1150              | ( 1610.3 ) | 0.189                               | ( 7.45 )         |
| 850               | ( 1070.3 ) | 5.827                               | ( 229.39 )       | 1175              | ( 1655.3 ) | 0.154                               | ( 6.06 )         |
| 875               | ( 1115.3 ) | 4.003                               | ( 157.61 )       | 1200              | ( 1700.3 ) | 0.126                               | ( 4.97 )         |
| 900               | ( 1160.3 ) | 2.808                               | ( 110.57 )       | 1225              | ( 1745.3 ) | 0.104                               | ( 4.11 )         |
| 925               | ( 1205.3 ) | 2.008                               | ( 79.07 )        | 1250              | ( 1790.3 ) | 0.087                               | ( 3.43 )         |
| 950               | ( 1250.3 ) | 1.462                               | ( 57.55 )        | 1273              | ( 1831.7 ) | 0.074                               | ( 2.92 )         |
| 975               | ( 1295.3 ) | 1.081                               | ( 42.58 )        |                   |            |                                     |                  |
| 1000              | ( 1340.3 ) | 0.812                               | ( 31.98 )        |                   |            |                                     |                  |

**Application Notes:** Data for electrical resistivity is collected from reference [14] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$\rho_e(T) = \exp \left( A0 + \frac{1000}{T} * A1 \right)$$

$\rho_e(T)$  = Electrical Resistivity [ $\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

Temperature  
Range [K]:  $700 \leq T \leq 1273$

A0 = -11.37399

A1 = 11.16595



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Young's Modulus with Temperature

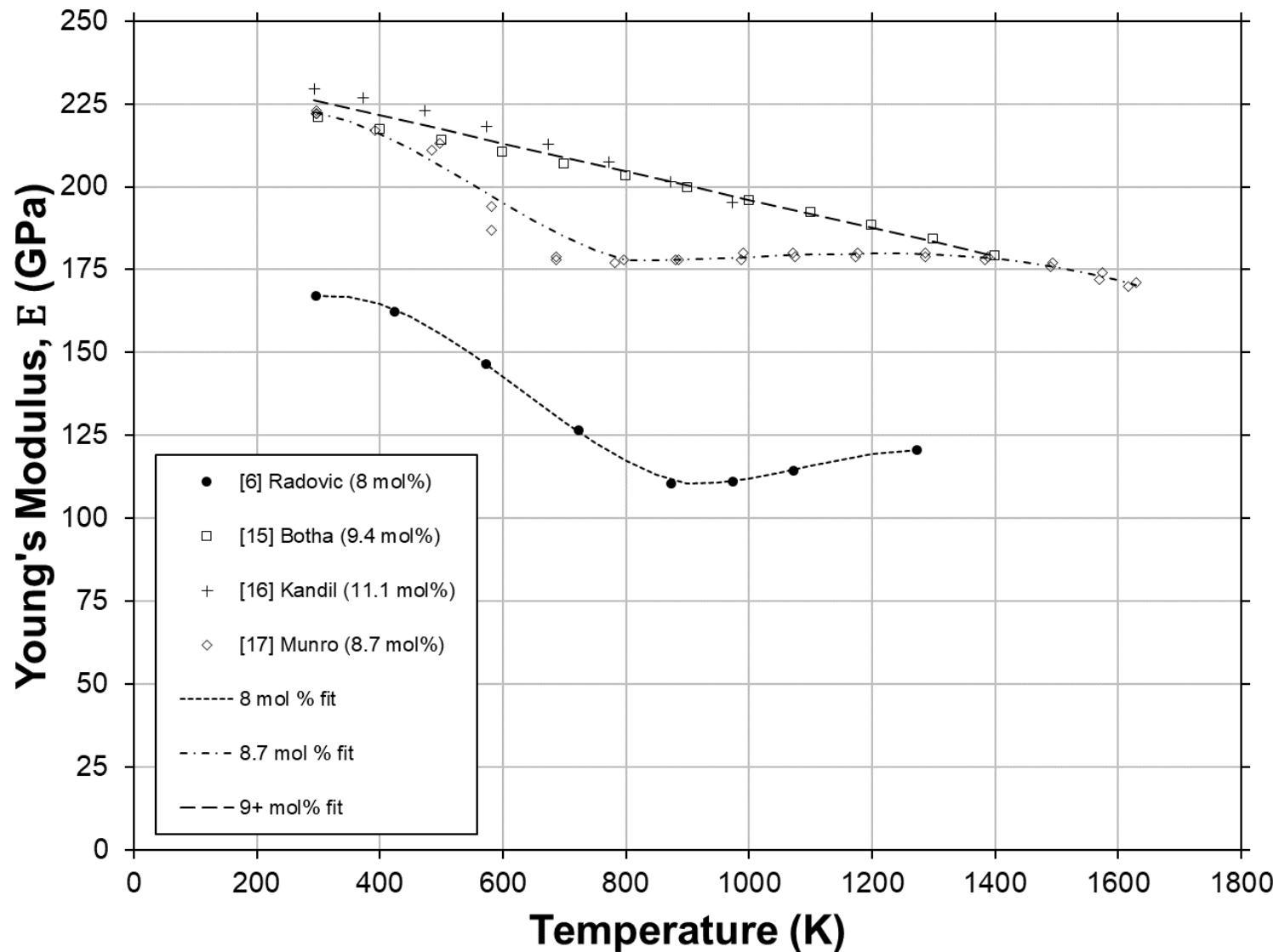


Figure 7.1.2-8: Young's Modulus versus Temperature of YSZ.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

## 7 Refractory Ceramics

## 7.1 Oxides

## 7.1.2 Zirconia (Yttria Stabilized) (YSZ)

Revision 0: 08-05-2020

## Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|-----------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )    | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 300               | ( 80.3 )  | 222.30                | ( 32.26 ) | 850               | ( 1070.3 ) | 177.96                | ( 25.82 ) |
| 350               | ( 170.3 ) | 219.75                | ( 31.89 ) | 900               | ( 1160.3 ) | 178.17                | ( 25.85 ) |
| 400               | ( 260.3 ) | 216.04                | ( 31.35 ) | 950               | ( 1250.3 ) | 178.47                | ( 25.90 ) |
| 450               | ( 350.3 ) | 211.44                | ( 30.68 ) | 1000              | ( 1340.3 ) | 178.81                | ( 25.95 ) |
| 500               | ( 440.3 ) | 206.23                | ( 29.92 ) | 1050              | ( 1430.3 ) | 179.16                | ( 26.00 ) |
| 550               | ( 530.3 ) | 200.69                | ( 29.12 ) | 1100              | ( 1520.3 ) | 179.47                | ( 26.04 ) |
| 600               | ( 620.3 ) | 195.11                | ( 28.31 ) | 1150              | ( 1610.3 ) | 179.71                | ( 26.08 ) |
| 650               | ( 710.3 ) | 189.76                | ( 27.53 ) | 1200              | ( 1700.3 ) | 179.83                | ( 26.09 ) |
| 700               | ( 800.3 ) | 184.92                | ( 26.83 ) | 1250              | ( 1790.3 ) | 179.79                | ( 26.09 ) |
| 750               | ( 890.3 ) | 180.88                | ( 26.25 ) | 1300              | ( 1880.3 ) | 179.56                | ( 26.05 ) |
| 800               | ( 980.3 ) | 177.88                | ( 25.81 ) |                   |            |                       |           |

**Application Notes:.** Data for Young's modulus is collected from references [6, 15-17] and fitted with the equations below to approximate property trend with respect to temperature. Fitted curves follow the data trends from 8 mol%, 8.7 mol%, and 9-11.1 mol% (noted as 9+ in the figure) yttria in zirconia. Table data represents modulus values for 8.7 mol%  $Y_2O_3$ .

### Fit Equation:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

### Constants:

| $Y_2O_3$ mol%: | 8 mol%              | 8.7 mol%            | 9-11.1 mol%         |
|----------------|---------------------|---------------------|---------------------|
| T. Range [K]:  | $295 \leq T < 911$  | $295 \leq T < 790$  | $295 \leq T < 1410$ |
| A0 =           | 104.5               | 197.4               | 238.6               |
| A1 =           | 441                 | 237.1               | -42.5               |
| A2 =           | -922.2              | -625.7              | 0                   |
| A3 =           | 488.6               | 373.6               | 0                   |
|                | $911 \leq T < 1300$ | $790 \leq T < 1630$ |                     |
| A0 =           | 561.4               | 226.7               | -                   |
| A1 =           | -1299               | -156.7              | -                   |
| A2 =           | 1220                | 162.7               | -                   |
| A3 =           | -370.4              | -53.89              | -                   |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Shear Modulus with Temperature

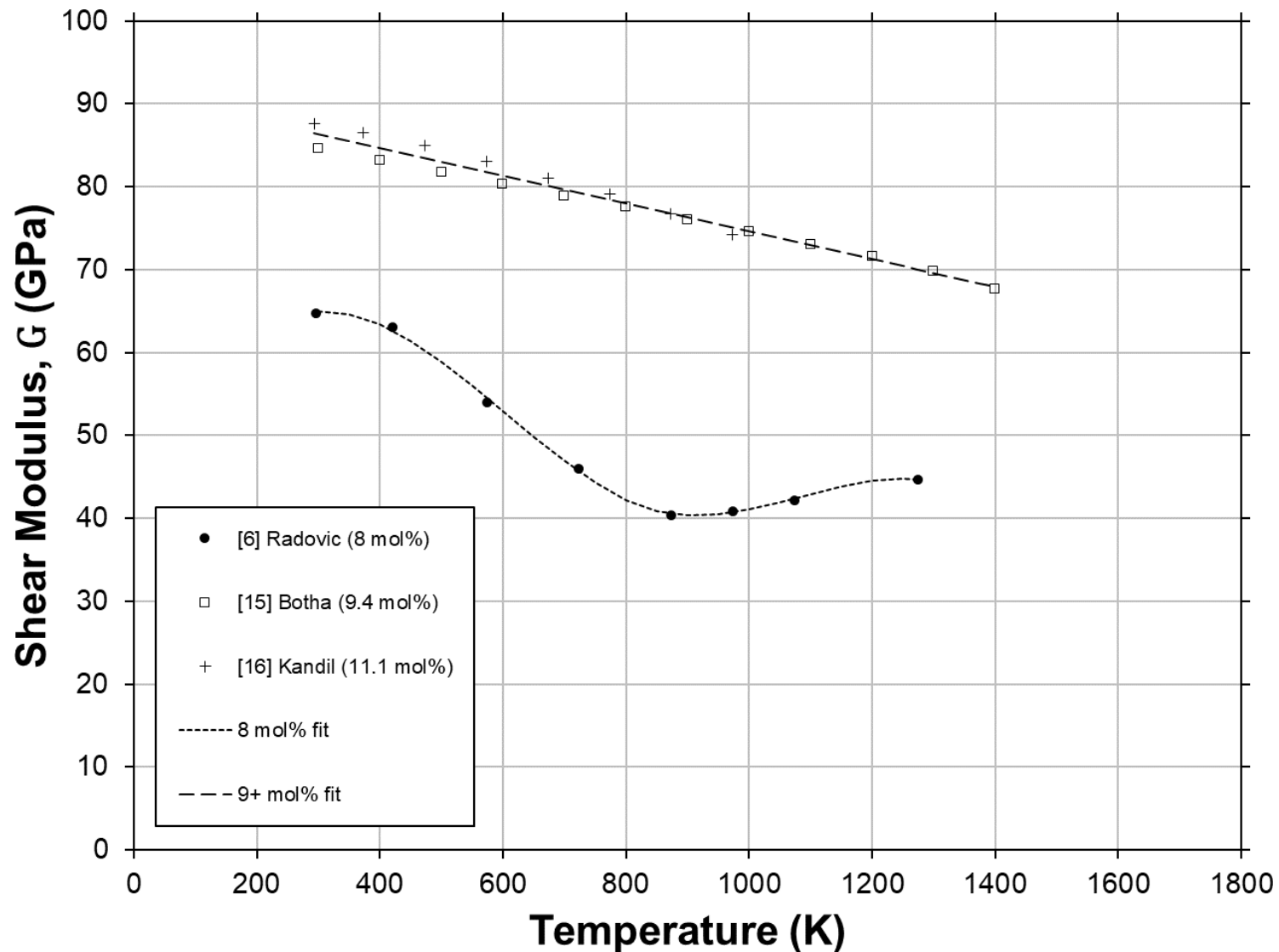


Figure 7.1.2-9: Shear Modulus versus Temperature of YSZ



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Shear Modulus ( G ) |          | Temperature ( T ) |            | Shear Modulus ( G ) |          |
|-------------------|-----------|---------------------|----------|-------------------|------------|---------------------|----------|
| K                 | ( °F )    | GPa                 | ( Msi )  | K                 | ( °F )     | GPa                 | ( Msi )  |
| 300               | ( 80.3 )  | 64.93               | ( 9.42 ) | 850               | ( 1070.3 ) | 40.80               | ( 5.92 ) |
| 350               | ( 170.3 ) | 64.61               | ( 9.38 ) | 900               | ( 1160.3 ) | 40.38               | ( 5.86 ) |
| 400               | ( 260.3 ) | 63.39               | ( 9.20 ) | 950               | ( 1250.3 ) | 40.51               | ( 5.88 ) |
| 450               | ( 350.3 ) | 61.43               | ( 8.91 ) | 1000              | ( 1340.3 ) | 41.10               | ( 5.96 ) |
| 500               | ( 440.3 ) | 58.92               | ( 8.55 ) | 1050              | ( 1430.3 ) | 41.97               | ( 6.09 ) |
| 550               | ( 530.3 ) | 56.03               | ( 8.13 ) | 1100              | ( 1520.3 ) | 42.94               | ( 6.23 ) |
| 600               | ( 620.3 ) | 52.95               | ( 7.68 ) | 1150              | ( 1610.3 ) | 43.85               | ( 6.36 ) |
| 650               | ( 710.3 ) | 49.84               | ( 7.23 ) | 1200              | ( 1700.3 ) | 44.53               | ( 6.46 ) |
| 700               | ( 800.3 ) | 46.90               | ( 6.80 ) | 1250              | ( 1790.3 ) | 44.80               | ( 6.50 ) |
| 750               | ( 890.3 ) | 44.29               | ( 6.43 ) | 1273              | ( 1831.7 ) | 44.73               | ( 6.49 ) |
| 800               | ( 980.3 ) | 42.20               | ( 6.12 ) |                   |            |                     |          |

**Application Notes:** Data for shear modulus is collected from references [6, 15, 16] and fitted with the equations below to approximate property trend with respect to temperature. Fitted curves follow the data trends from 8 mol% and 9-11.1 mol% (noted as 9+ in the figure) yttria in zirconia. Table data represents shear modulus for 8.0 mol%  $Y_2O_3$ .

## Fit Equations:

$$G(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$G(T)$  = Shear Modulus [GPa]

$T$  = Temperature [K]

## Constants:

| $Y_2O_3$ mol%: | 8 mol%              | 9-11.1 mol%         |
|----------------|---------------------|---------------------|
| T. Range [K]:  | $275 \leq T < 870$  | $293 \leq T < 1400$ |
| A0 =           | 37.67               | 91.43               |
| A1 =           | 199.1               | -16.9               |
| A2 =           | -432.2              | 0.07455             |
| A3 =           | 238                 | 0                   |
|                | $870 \leq T < 1273$ |                     |
| A0 =           | 311.1               | -                   |
| A1 =           | -784.2              | -                   |
| A2 =           | 743.3               | -                   |
| A3 =           | -229.1              | -                   |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Poisson's ratio with Temperature

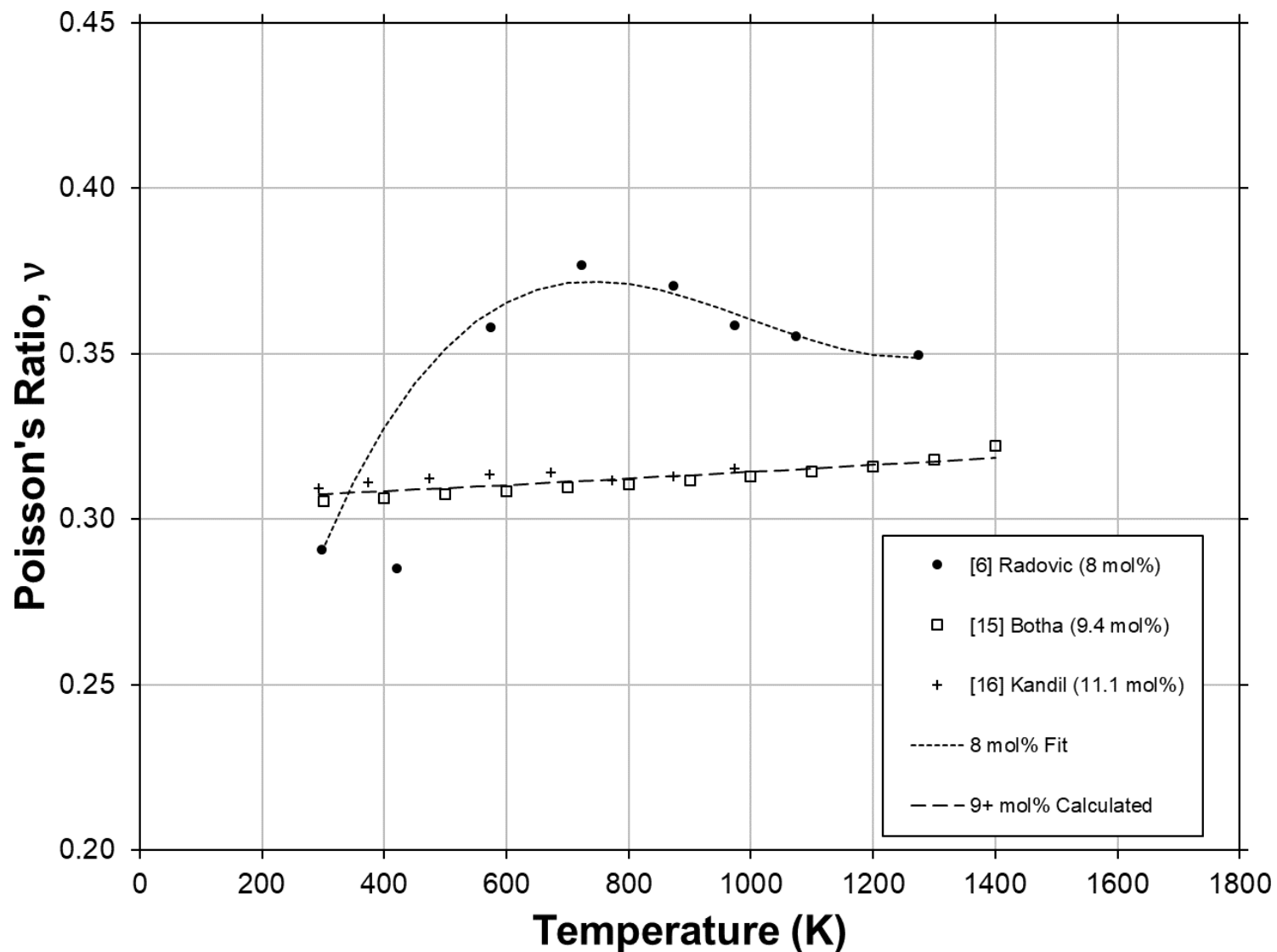


Figure 7.1.2-10: Poisson's Ratio versus Temperature of YSZ.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 0: 08-05-2020

Poisson's ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|-----------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )    |                           | K                 | ( °F )     |                           |
| 300               | ( 80.3 )  | 0.291                     | 700               | ( 800.3 )  | 0.371                     |
| 325               | ( 125.3 ) | 0.302                     | 750               | ( 890.3 )  | 0.372                     |
| 350               | ( 170.3 ) | 0.311                     | 800               | ( 980.3 )  | 0.371                     |
| 375               | ( 215.3 ) | 0.320                     | 850               | ( 1070.3 ) | 0.369                     |
| 400               | ( 260.3 ) | 0.328                     | 900               | ( 1160.3 ) | 0.367                     |
| 425               | ( 305.3 ) | 0.335                     | 950               | ( 1250.3 ) | 0.364                     |
| 450               | ( 350.3 ) | 0.341                     | 1000              | ( 1340.3 ) | 0.360                     |
| 475               | ( 395.3 ) | 0.347                     | 1050              | ( 1430.3 ) | 0.357                     |
| 500               | ( 440.3 ) | 0.352                     | 1100              | ( 1520.3 ) | 0.354                     |
| 525               | ( 485.3 ) | 0.356                     | 1150              | ( 1610.3 ) | 0.351                     |
| 550               | ( 530.3 ) | 0.360                     | 1200              | ( 1700.3 ) | 0.349                     |
| 575               | ( 575.3 ) | 0.363                     | 1250              | ( 1790.3 ) | 0.349                     |
| 600               | ( 620.3 ) | 0.365                     | 1273              | ( 1831.7 ) | 0.349                     |
| 625               | ( 665.3 ) | 0.368                     |                   |            |                           |
| 650               | ( 710.3 ) | 0.369                     |                   |            |                           |

**Application Notes:** Poisson's Ratio is calculated as a function of Young's modulus and shear modulus – as seen in the equation below – to approximate property trend with respect to temperature. The two curves displayed are calculated from the trends for Young's and Shear moduli at 8 mol% and 9-11.1 mol% (noted as 9+ in the figure) of yttria added to zirconia. Table data represents shear modulus for 8.0 mol%  $Y_2O_3$ .

## Fit Equations:

$$\nu(T) = E(T)/(2 \cdot G(T)) - 1$$

$$\nu(T) = \text{Poisson's Ratio}$$

$$T = \text{Temperature [K]}$$

## Constants:

$$T \text{ Range [K]: } 300 \leq T \leq 1273$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

Revision 2: 04-26-2023

Tabulated Property Data

| Temp.<br>(T)<br>K | Physical and Electrical                    |                                                          | Thermal                                          |                                           |                                                       |                                          | Elastic                                  |                                        |                                         |
|-------------------|--------------------------------------------|----------------------------------------------------------|--------------------------------------------------|-------------------------------------------|-------------------------------------------------------|------------------------------------------|------------------------------------------|----------------------------------------|-----------------------------------------|
|                   | Density<br>( $\rho$ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( $\rho_e$ )<br>$\Omega$ -m | Thermal Conductivity<br>8 mol%<br>(k)<br>W/(m-K) | Thermal<br>Expansion<br>( $dL/L_0$ )<br>% | Mean CTE<br>( $\bar{\alpha}$ )<br>10 <sup>-6</sup> /K | Specific<br>Heat<br>( $C_p$ )<br>J/(g-K) | Young's Modulus<br>9+ mol%<br>(E)<br>GPa | Shear Modulus<br>9+ mol%<br>(G)<br>GPa | Poisson's Ratio<br>9+ mol%<br>( $\nu$ ) |
| 200               | 6024                                       | -                                                        | -                                                | -0.06                                     | 6.13                                                  | 0.354                                    | -                                        | -                                      | -                                       |
| 300               | 6013                                       | -                                                        | 1.86                                             | 0.00                                      | 6.86                                                  | 0.467                                    | 225.9                                    | 86.4                                   | 0.308                                   |
| 400               | 5999                                       | -                                                        | 1.93                                             | 0.08                                      | 7.54                                                  | 0.520                                    | 221.6                                    | 84.7                                   | 0.308                                   |
| 500               | 5983                                       | -                                                        | 2.00                                             | 0.17                                      | 8.16                                                  | 0.548                                    | 217.4                                    | 83.0                                   | 0.309                                   |
| 600               | 5965                                       | -                                                        | 2.05                                             | 0.27                                      | 8.73                                                  | 0.567                                    | 213.1                                    | 81.3                                   | 0.310                                   |
| 700               | 5946                                       | 97.3                                                     | 2.10                                             | 0.38                                      | 9.26                                                  | 0.580                                    | 208.9                                    | 79.6                                   | 0.311                                   |
| 800               | 5926                                       | 13.2                                                     | 2.14                                             | 0.49                                      | 9.73                                                  | 0.591                                    | 204.6                                    | 78.0                                   | 0.312                                   |
| 900               | 5905                                       | 2.8                                                      | 2.17                                             | 0.61                                      | 10.16                                                 | 0.601                                    | 200.4                                    | 76.3                                   | 0.313                                   |
| 1000              | 5883                                       | 0.8                                                      | 2.19                                             | 0.74                                      | 10.55                                                 | 0.609                                    | 196.1                                    | 74.6                                   | 0.314                                   |
| 1100              | 5860                                       | 0.3                                                      | 2.20                                             | 0.87                                      | 10.89                                                 | 0.618                                    | 191.9                                    | 72.9                                   | 0.315                                   |
| 1200              | 5836                                       | 0.1                                                      | 2.20                                             | 1.01                                      | 11.20                                                 | 0.625                                    | 187.6                                    | 71.3                                   | 0.316                                   |
| 1300              | 5812                                       | -                                                        | -                                                | 1.14                                      | 11.47                                                 | 0.633                                    | 183.4                                    | 69.6                                   | 0.317                                   |
| 1400              | 5788                                       | -                                                        | -                                                | 1.28                                      | 11.70                                                 | 0.640                                    | 179.1                                    | 67.9                                   | 0.319                                   |
| 1500              | 5764                                       | -                                                        | -                                                | 1.43                                      | 11.90                                                 | 0.647                                    | -                                        | -                                      | -                                       |
| 1600              | 5739                                       | -                                                        | -                                                | 1.57                                      | 12.07                                                 | 0.654                                    | -                                        | -                                      | -                                       |
| 1700              | 5715                                       | -                                                        | -                                                | 1.71                                      | 12.21                                                 | -                                        | -                                        | -                                      | -                                       |
| 1800              | 5691                                       | -                                                        | -                                                | 1.86                                      | 12.32                                                 | -                                        | -                                        | -                                      | -                                       |
| 1900              | 5668                                       | -                                                        | -                                                | 2.00                                      | 12.40                                                 | -                                        | -                                        | -                                      | -                                       |
| 2000              | 5645                                       | -                                                        | -                                                | 2.14                                      | 12.46                                                 | -                                        | -                                        | -                                      | -                                       |
| 2100              | 5622                                       | -                                                        | -                                                | 2.27                                      | 12.50                                                 | -                                        | -                                        | -                                      | -                                       |
| 2200              | 5601                                       | -                                                        | -                                                | 2.40                                      | 12.52                                                 | -                                        | -                                        | -                                      | -                                       |
| 2300              | 5580                                       | -                                                        | -                                                | 2.53                                      | 12.52                                                 | -                                        | -                                        | -                                      | -                                       |
| 2400              | 5560                                       | -                                                        | -                                                | 2.65                                      | 12.50                                                 | -                                        | -                                        | -                                      | -                                       |
| 2600              | 5525                                       | -                                                        | -                                                | 2.87                                      | 12.43                                                 | -                                        | -                                        | -                                      | -                                       |
| 2800              | 5495                                       | -                                                        | -                                                | 3.06                                      | 12.31                                                 | -                                        | -                                        | -                                      | -                                       |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

**Revision 0: 08-05-2020**

**References**

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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.1 Oxides

7.1.2 Zirconia (Yttria Stabilized)  
(YSZ)

**Revision 0: 08-05-2020**

**References**

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## **7 Refractory Ceramics**

### **7.2 Carbides**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide ( $B_4C$ )

Revision 2.1: 08-25-2023

General

### Room Temperature Properties

|                                                              |       |
|--------------------------------------------------------------|-------|
| Molar Mass, [g/mol]                                          | 55.25 |
| Theoretical Density, [kg/m <sup>3</sup> ]                    | 2,500 |
| Melting Point, [K]                                           | 2723  |
| Boiling Point, [K]                                           | 3773  |
| Specific Heat, [J/(g-K)]                                     | 0.93  |
| Thermal Conductivity, [W/(m-K)]                              | 28.57 |
| Linear expansion coefficient, [ $\mu\text{m}/(\text{m-K})$ ] | 3.98  |
| Young's Modulus, [GPa]                                       | 450.0 |
| Shear Modulus, [GPa]                                         | 189.1 |
| Poisson's Ratio, [-]                                         | 0.190 |

### Boron – Carbon Phase Diagram

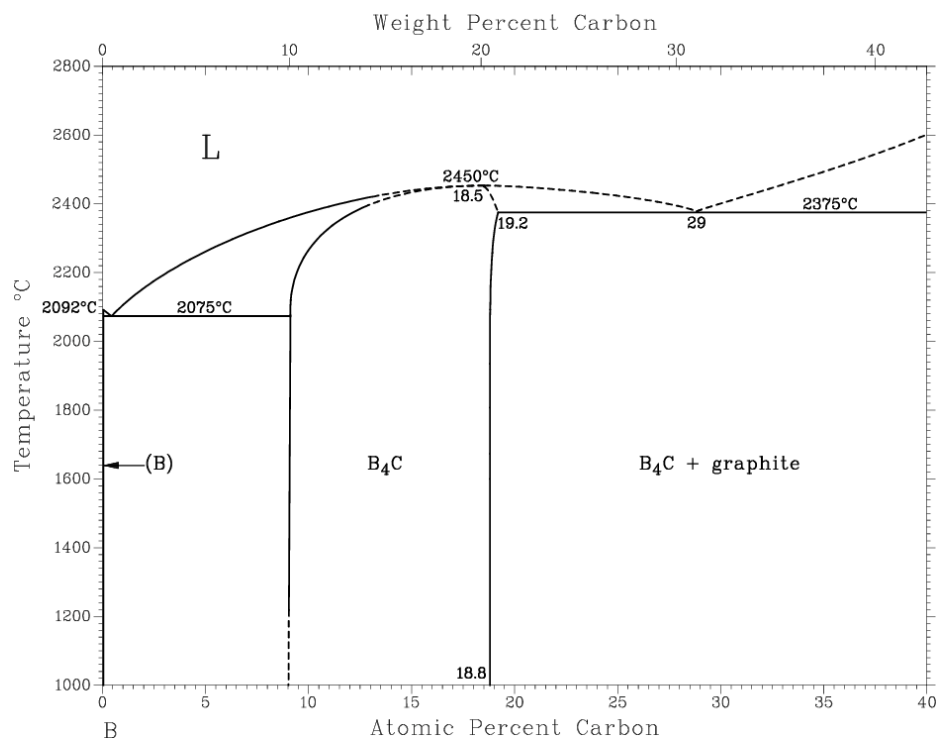


Figure 7.2.1-1: Boron – Carbon Phase Diagram



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide ( $B_4C$ )

Revision 0: 08-05-2020

Density with Temperature

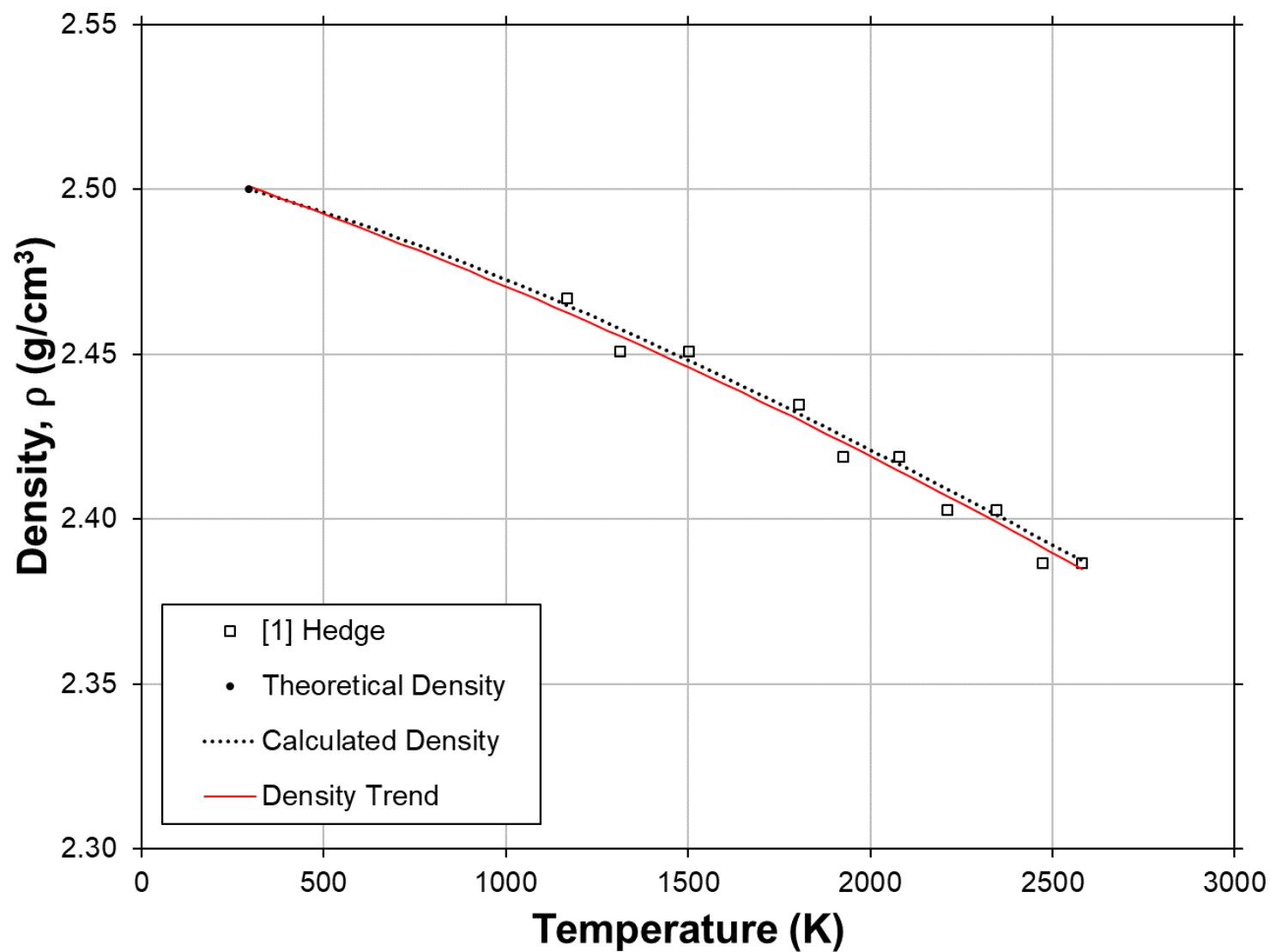


Figure 7.2.1-2: Density versus Temperature for  $B_4C$ . Displaying fitted trend of data with comparison to the density calculated from the fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 293               | ( 67.7 )   | 2500              | ( 156.1 )              | 1200              | ( 1700.3 ) | 2463              | ( 153.8 )              |
| 300               | ( 80.3 )   | 2500              | ( 156.1 )              | 1300              | ( 1880.3 ) | 2458              | ( 153.5 )              |
| 350               | ( 170.3 )  | 2498              | ( 156.0 )              | 1400              | ( 2060.3 ) | 2453              | ( 153.2 )              |
| 400               | ( 260.3 )  | 2497              | ( 155.9 )              | 1500              | ( 2240.3 ) | 2448              | ( 152.8 )              |
| 450               | ( 350.3 )  | 2495              | ( 155.8 )              | 1600              | ( 2420.3 ) | 2443              | ( 152.5 )              |
| 500               | ( 440.3 )  | 2493              | ( 155.6 )              | 1700              | ( 2600.3 ) | 2438              | ( 152.2 )              |
| 550               | ( 530.3 )  | 2491              | ( 155.5 )              | 1800              | ( 2780.3 ) | 2432              | ( 151.8 )              |
| 600               | ( 620.3 )  | 2489              | ( 155.4 )              | 1900              | ( 2960.3 ) | 2427              | ( 151.5 )              |
| 650               | ( 710.3 )  | 2487              | ( 155.3 )              | 2000              | ( 3140.3 ) | 2421              | ( 151.1 )              |
| 700               | ( 800.3 )  | 2486              | ( 155.2 )              | 2100              | ( 3320.3 ) | 2415              | ( 150.8 )              |
| 750               | ( 890.3 )  | 2483              | ( 155.0 )              | 2200              | ( 3500.3 ) | 2410              | ( 150.4 )              |
| 800               | ( 980.3 )  | 2481              | ( 154.9 )              | 2300              | ( 3680.3 ) | 2404              | ( 150.1 )              |
| 900               | ( 1160.3 ) | 2477              | ( 154.6 )              | 2400              | ( 3860.3 ) | 2398              | ( 149.7 )              |
| 1000              | ( 1340.3 ) | 2473              | ( 154.4 )              | 2500              | ( 4040.3 ) | 2392              | ( 149.3 )              |
| 1100              | ( 1520.3 ) | 2468              | ( 154.1 )              | 2581              | ( 4186.1 ) | 2388              | ( 149.1 )              |

**Application Notes:** Data for density is collected from reference [1] and fitted with density fit equation below to approximate property trend with respect to temperature. Density is also calculated as a function of thermal expansion as seen in density calculation equation and compared against data trend.

## Density Calculation Equation:

$$\rho(T) = \rho_{RT}(1 - P)/(1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 2,500 \text{ [kg/m}^3\text{]}$$

$P$  = Fractional Porosity

$T$  = Temperature [K]

## Density Fit:

$$\rho(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

## Constants:

T Range [K]:  $293 \leq T \leq 2581$

A0 = 2512

A1 = -36.44

A2 = -4.987



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide ( $B_4C$ )

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

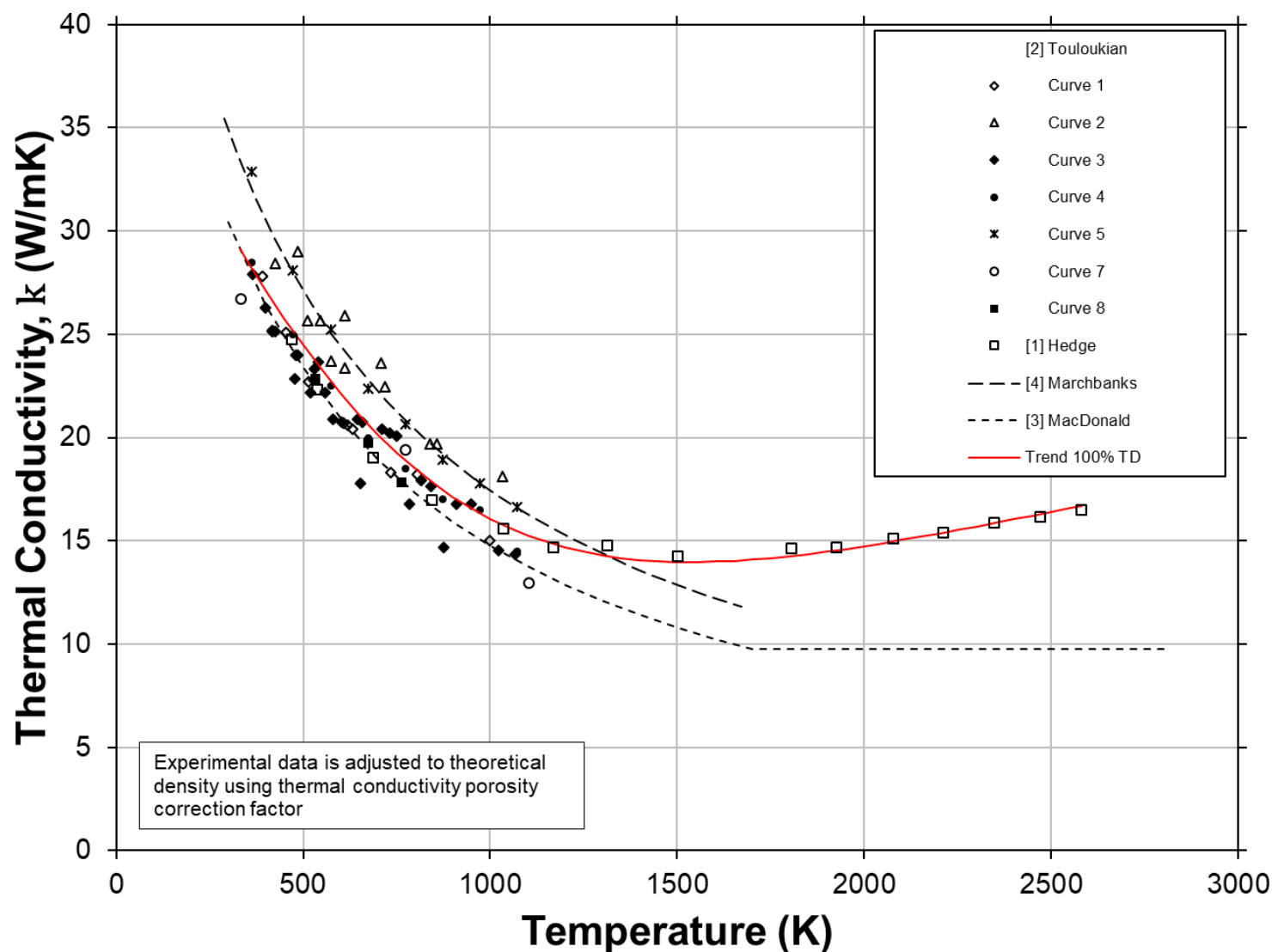


Figure 7.2.1-3: Thermal Conductivity versus Temperature of  $B_4C$ . Displaying fitted trend of the data with comparison to Marchbanks (1976) and MacDonald (1976) for 100% Theoretical density.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 330               | ( 134.3 )  | 29.16                      | ( 202.31 )                          | 1200              | ( 1700.3 ) | 14.69                      | ( 101.95 )                          |
| 350               | ( 170.3 )  | 28.57                      | ( 198.19 )                          | 1300              | ( 1880.3 ) | 14.31                      | ( 99.25 )                           |
| 400               | ( 260.3 )  | 27.13                      | ( 188.20 )                          | 1400              | ( 2060.3 ) | 14.08                      | ( 97.66 )                           |
| 450               | ( 350.3 )  | 25.76                      | ( 178.72 )                          | 1500              | ( 2240.3 ) | 13.98                      | ( 96.99 )                           |
| 500               | ( 440.3 )  | 24.47                      | ( 169.78 )                          | 1600              | ( 2420.3 ) | 13.99                      | ( 97.07 )                           |
| 550               | ( 530.3 )  | 23.26                      | ( 161.40 )                          | 1700              | ( 2600.3 ) | 14.09                      | ( 97.75 )                           |
| 600               | ( 620.3 )  | 22.14                      | ( 153.60 )                          | 1800              | ( 2780.3 ) | 14.26                      | ( 98.92 )                           |
| 650               | ( 710.3 )  | 21.10                      | ( 146.39 )                          | 1900              | ( 2960.3 ) | 14.48                      | ( 100.46 )                          |
| 700               | ( 800.3 )  | 20.14                      | ( 139.76 )                          | 2000              | ( 3140.3 ) | 14.74                      | ( 102.30 )                          |
| 750               | ( 890.3 )  | 19.27                      | ( 133.71 )                          | 2100              | ( 3320.3 ) | 15.04                      | ( 104.36 )                          |
| 800               | ( 980.3 )  | 18.48                      | ( 128.23 )                          | 2200              | ( 3500.3 ) | 15.36                      | ( 106.58 )                          |
| 850               | ( 1070.3 ) | 17.77                      | ( 123.30 )                          | 2300              | ( 3680.3 ) | 15.70                      | ( 108.92 )                          |
| 900               | ( 1160.3 ) | 17.14                      | ( 118.89 )                          | 2400              | ( 3860.3 ) | 16.05                      | ( 111.34 )                          |
| 1000              | ( 1340.3 ) | 16.08                      | ( 111.54 )                          | 2500              | ( 4040.3 ) | 16.40                      | ( 113.81 )                          |
| 1100              | ( 1520.3 ) | 15.27                      | ( 105.98 )                          | 2581              | ( 4186.1 ) | 16.69                      | ( 115.82 )                          |

**Application Notes:** Data for thermal conductivity is collected from references [1, 2] and fitted with the equation below to approximate property trend with respect to temperature and porosity. Trend for 100% TD is compared against references [3, 4].

## Fit Equation:

$$k(T) = \left( A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right) / \left( 1 + A_{-2} \cdot \left( \frac{T}{1000} \right)^2 \right)$$

$$k(T) = \text{Thermal Conductivity } [W / (m \cdot K)]$$

$$k_p = k \cdot (1 - P) / (1 + P)$$

$$P = \text{Fractional Porosity } (0 \leq P \leq 0.24)$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 330 \leq T \leq 2581$$

$$A_0 = 39.96$$

$$A_1 = -33.8$$

$$A_2 = 18.11$$

$$A_{-2} = 0.5096$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide ( $B_4C$ )

Revision 0: 08-05-2020

Thermal Expansion with Temperature

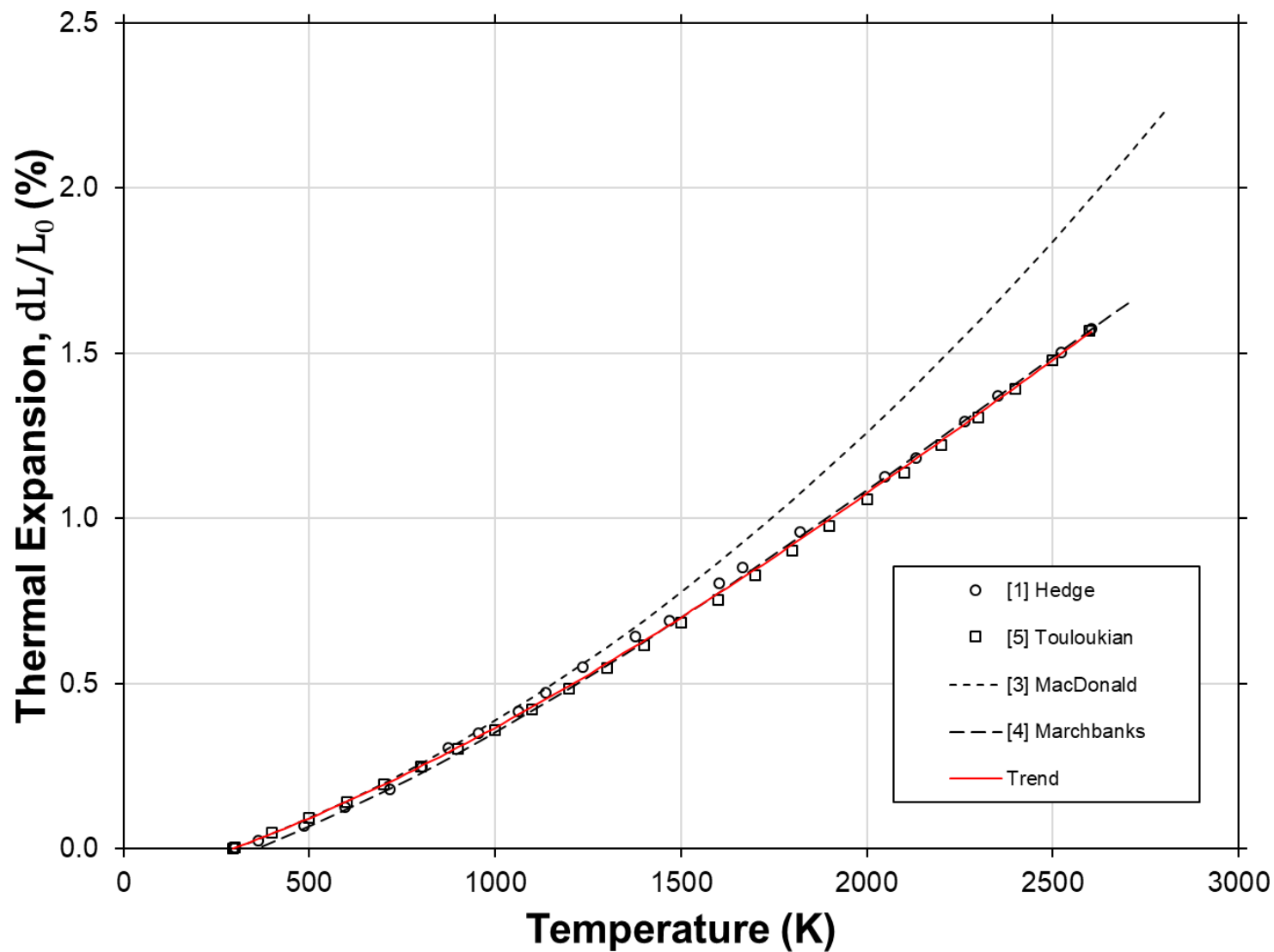


Figure 7.2.1-4: Thermal Expansion versus Temperature of  $B_4C$ . Displaying fitted trend of the data with comparison to Marchbanks (1976) and MacDonald (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 293               | ( 67.7 )   | 0.000                                   | 1200              | ( 1700.3 ) | 0.494                                   |
| 300               | ( 80.3 )   | 0.003                                   | 1300              | ( 1880.3 ) | 0.560                                   |
| 350               | ( 170.3 )  | 0.024                                   | 1400              | ( 2060.3 ) | 0.629                                   |
| 400               | ( 260.3 )  | 0.046                                   | 1500              | ( 2240.3 ) | 0.700                                   |
| 450               | ( 350.3 )  | 0.069                                   | 1600              | ( 2420.3 ) | 0.772                                   |
| 500               | ( 440.3 )  | 0.092                                   | 1700              | ( 2600.3 ) | 0.846                                   |
| 550               | ( 530.3 )  | 0.117                                   | 1800              | ( 2780.3 ) | 0.921                                   |
| 600               | ( 620.3 )  | 0.142                                   | 1900              | ( 2960.3 ) | 0.998                                   |
| 650               | ( 710.3 )  | 0.168                                   | 2000              | ( 3140.3 ) | 1.076                                   |
| 700               | ( 800.3 )  | 0.194                                   | 2100              | ( 3320.3 ) | 1.155                                   |
| 750               | ( 890.3 )  | 0.221                                   | 2200              | ( 3500.3 ) | 1.235                                   |
| 800               | ( 980.3 )  | 0.249                                   | 2300              | ( 3680.3 ) | 1.316                                   |
| 900               | ( 1160.3 ) | 0.306                                   | 2400              | ( 3860.3 ) | 1.397                                   |
| 1000              | ( 1340.3 ) | 0.367                                   | 2500              | ( 4040.3 ) | 1.479                                   |
| 1100              | ( 1520.3 ) | 0.429                                   | 2600              | ( 4220.3 ) | 1.562                                   |

**Application Notes:** Data for thermal expansion is collected from references [1, 5] and fitted with the equation below to approximate property trend with respect to temperature. Trend is compared against trends from references [3, 4].

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 293 \leq T \leq 2605$$

$$A0 = -0.1068$$

$$A1 = 0.3132$$

$$A2 = 0.1812$$

$$A3 = -0.02107$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide ( $B_4C$ )

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

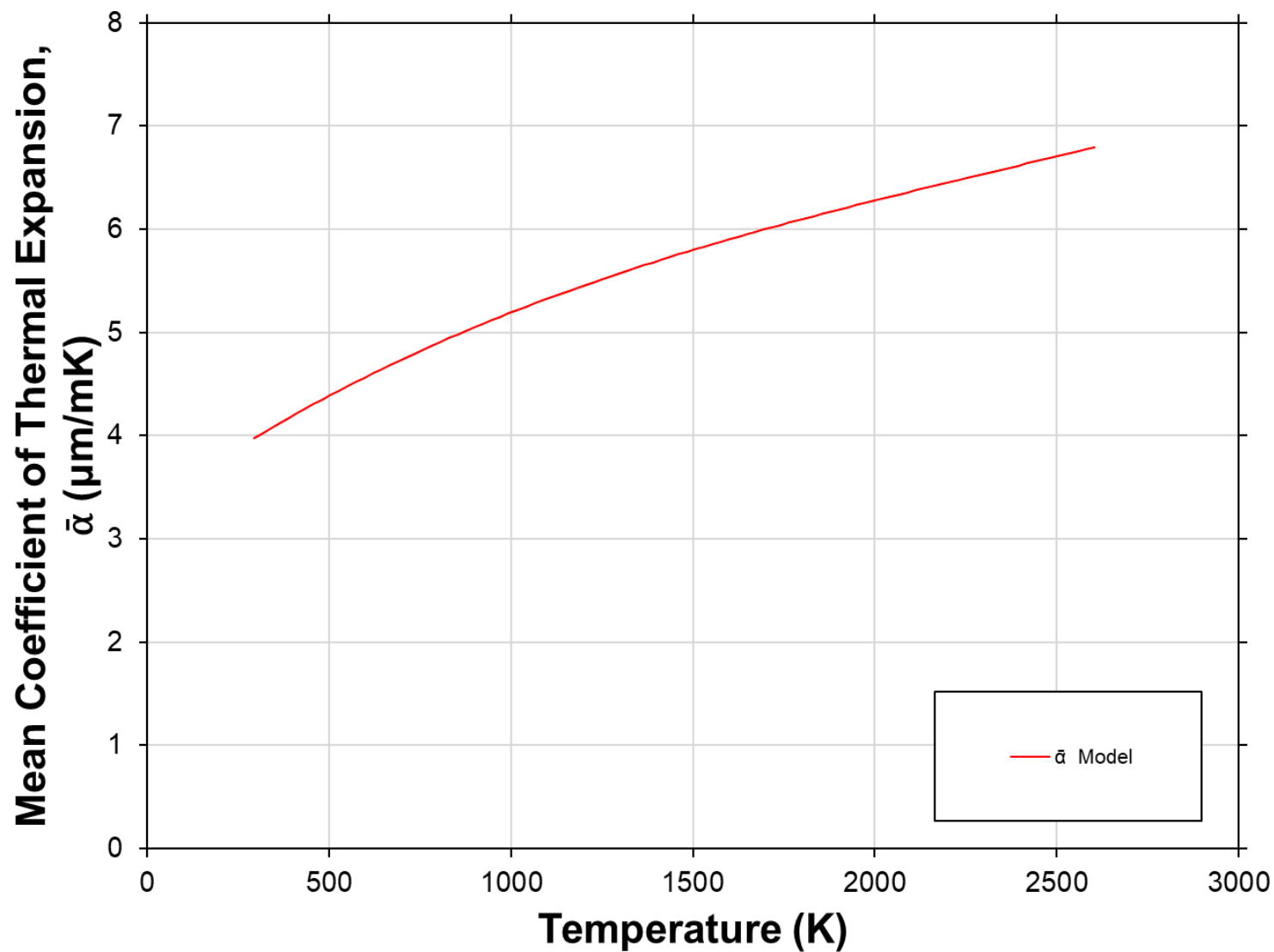


Figure 7.2.1-5: Mean Coefficient of Thermal Expansion versus Temperature of  $B_4C$ . Calculated from fitted trend of the Thermal Expansion data for  $B_4C$ .



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ ) |                 | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ ) |                 |
|-------------------|------------|-----------------------------|-----------------|-------------------|------------|-----------------------------|-----------------|
| K                 | ( °F )     | μm/(m·K)                    | ( μin/(in·°F) ) | K                 | ( °F )     | μm/(m·K)                    | ( μin/(in·°F) ) |
| 293               | ( 67.7 )   | 3.978                       | ( 2.210 )       | 1200              | ( 1700.3 ) | 5.456                       | ( 3.031 )       |
| 300               | ( 80.3 )   | 3.993                       | ( 2.218 )       | 1300              | ( 1880.3 ) | 5.577                       | ( 3.098 )       |
| 350               | ( 170.3 )  | 4.096                       | ( 2.275 )       | 1400              | ( 2060.3 ) | 5.691                       | ( 3.162 )       |
| 400               | ( 260.3 )  | 4.196                       | ( 2.331 )       | 1500              | ( 2240.3 ) | 5.800                       | ( 3.222 )       |
| 450               | ( 350.3 )  | 4.293                       | ( 2.385 )       | 1600              | ( 2420.3 ) | 5.904                       | ( 3.280 )       |
| 500               | ( 440.3 )  | 4.388                       | ( 2.438 )       | 1700              | ( 2600.3 ) | 6.003                       | ( 3.335 )       |
| 550               | ( 530.3 )  | 4.480                       | ( 2.489 )       | 1800              | ( 2780.3 ) | 6.098                       | ( 3.388 )       |
| 600               | ( 620.3 )  | 4.569                       | ( 2.538 )       | 1900              | ( 2960.3 ) | 6.190                       | ( 3.439 )       |
| 650               | ( 710.3 )  | 4.655                       | ( 2.586 )       | 2000              | ( 3140.3 ) | 6.279                       | ( 3.488 )       |
| 700               | ( 800.3 )  | 4.739                       | ( 2.633 )       | 2100              | ( 3320.3 ) | 6.366                       | ( 3.537 )       |
| 750               | ( 890.3 )  | 4.821                       | ( 2.678 )       | 2200              | ( 3500.3 ) | 6.452                       | ( 3.584 )       |
| 800               | ( 980.3 )  | 4.900                       | ( 2.722 )       | 2300              | ( 3680.3 ) | 6.537                       | ( 3.632 )       |
| 900               | ( 1160.3 ) | 5.051                       | ( 2.806 )       | 2400              | ( 3860.3 ) | 6.621                       | ( 3.679 )       |
| 1000              | ( 1340.3 ) | 5.194                       | ( 2.886 )       | 2500              | ( 4040.3 ) | 6.706                       | ( 3.726 )       |
| 1100              | ( 1520.3 ) | 5.329                       | ( 2.961 )       | 2600              | ( 4220.3 ) | 6.792                       | ( 3.773 )       |

**Application Notes:** Trend for mean coefficient of thermal expansion is calculated as a function of thermal expansion as seen in the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [μm/(m·K)]

T = Temperature [K]

## Constants:

T. Range [K]: 293 ≤ T ≤ 2605

A0 = 3.307

A1 = 2.486

A2 = -0.6976

A3 = 0.09882



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide ( $B_4C$ )

Revision 0: 08-05-2020

Specific Heat with Temperature

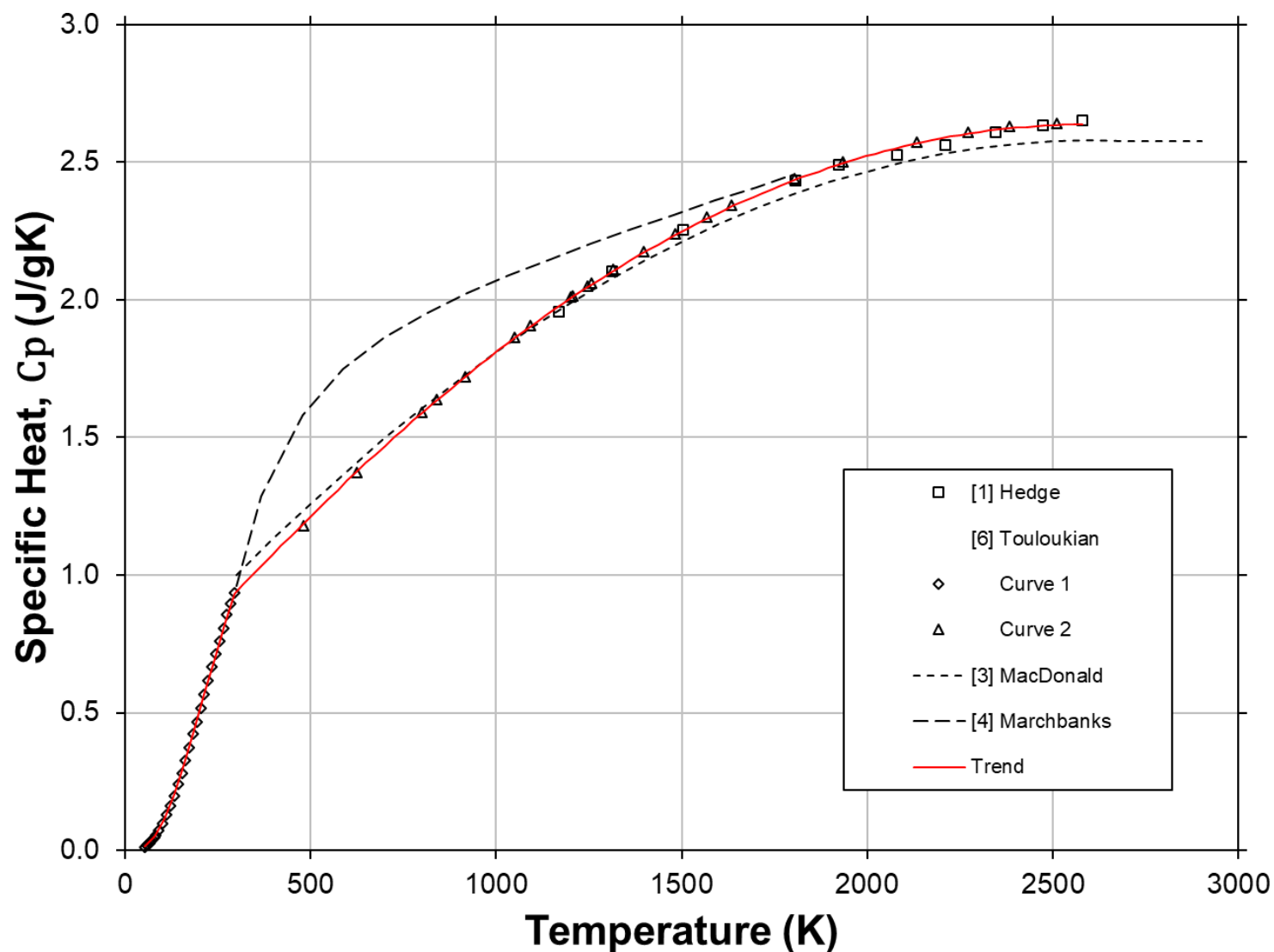


Figure 7.2.1-6: Specific Heat versus Temperature of  $B_4C$ . Displaying fitted trend of the data with 95% confidence bounds with comparison to Marchbanks (1976) and MacDonald (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 54                | ( -362.5 ) | 0.014                            | ( 0.003 )       | 1200              | ( 1700.3 ) | 2.004                            | ( 0.479 )       |
| 100               | ( -279.7 ) | 0.093                            | ( 0.022 )       | 1300              | ( 1880.3 ) | 2.092                            | ( 0.500 )       |
| 150               | ( -189.7 ) | 0.267                            | ( 0.064 )       | 1400              | ( 2060.3 ) | 2.173                            | ( 0.519 )       |
| 200               | ( -99.7 )  | 0.496                            | ( 0.119 )       | 1500              | ( 2240.3 ) | 2.248                            | ( 0.537 )       |
| 250               | ( -9.7 )   | 0.740                            | ( 0.177 )       | 1600              | ( 2420.3 ) | 2.316                            | ( 0.554 )       |
| 300               | ( 80.3 )   | 0.941                            | ( 0.225 )       | 1700              | ( 2600.3 ) | 2.378                            | ( 0.568 )       |
| 350               | ( 170.3 )  | 1.009                            | ( 0.241 )       | 1800              | ( 2780.3 ) | 2.433                            | ( 0.581 )       |
| 400               | ( 260.3 )  | 1.077                            | ( 0.257 )       | 1900              | ( 2960.3 ) | 2.481                            | ( 0.593 )       |
| 500               | ( 440.3 )  | 1.212                            | ( 0.290 )       | 2000              | ( 3140.3 ) | 2.523                            | ( 0.603 )       |
| 600               | ( 620.3 )  | 1.344                            | ( 0.321 )       | 2100              | ( 3320.3 ) | 2.558                            | ( 0.611 )       |
| 700               | ( 800.3 )  | 1.469                            | ( 0.351 )       | 2200              | ( 3500.3 ) | 2.587                            | ( 0.618 )       |
| 800               | ( 980.3 )  | 1.589                            | ( 0.380 )       | 2300              | ( 3680.3 ) | 2.609                            | ( 0.624 )       |
| 900               | ( 1160.3 ) | 1.702                            | ( 0.407 )       | 2400              | ( 3860.3 ) | 2.625                            | ( 0.627 )       |
| 1000              | ( 1340.3 ) | 1.810                            | ( 0.432 )       | 2500              | ( 4040.3 ) | 2.634                            | ( 0.630 )       |
| 1100              | ( 1520.3 ) | 1.910                            | ( 0.457 )       | 2581              | ( 4186.1 ) | 2.637                            | ( 0.630 )       |

**Application Notes:** Data for specific heat is collected from references [1, 6], and fitted with the equation below to approximate property trend with respect to temperature. Fitted trend is compared against trends from references [3, 4].

## Fit Equation:

$$C_p(T) = A_0 + A_1 \cdot \left(\frac{T}{1000}\right) + A_2 \cdot \left(\frac{T}{1000}\right)^2 + A_3 \cdot \left(\frac{T}{1000}\right)^3 + A_2 / \left(\frac{T}{1000}\right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

| T. Range [K]: | <u>54 ≤ T &lt; 293</u> | <u>293 ≤ T ≤ 2581</u> |
|---------------|------------------------|-----------------------|
| A0 =          | 0.0647                 | 0.4356                |
| A1 =          | -2.65                  | 1.7                   |
| A2 =          | 34.56                  | -0.3283               |
| A3 =          | -52.65                 | 0                     |
| A_2 =         | 0                      | 0.00223               |

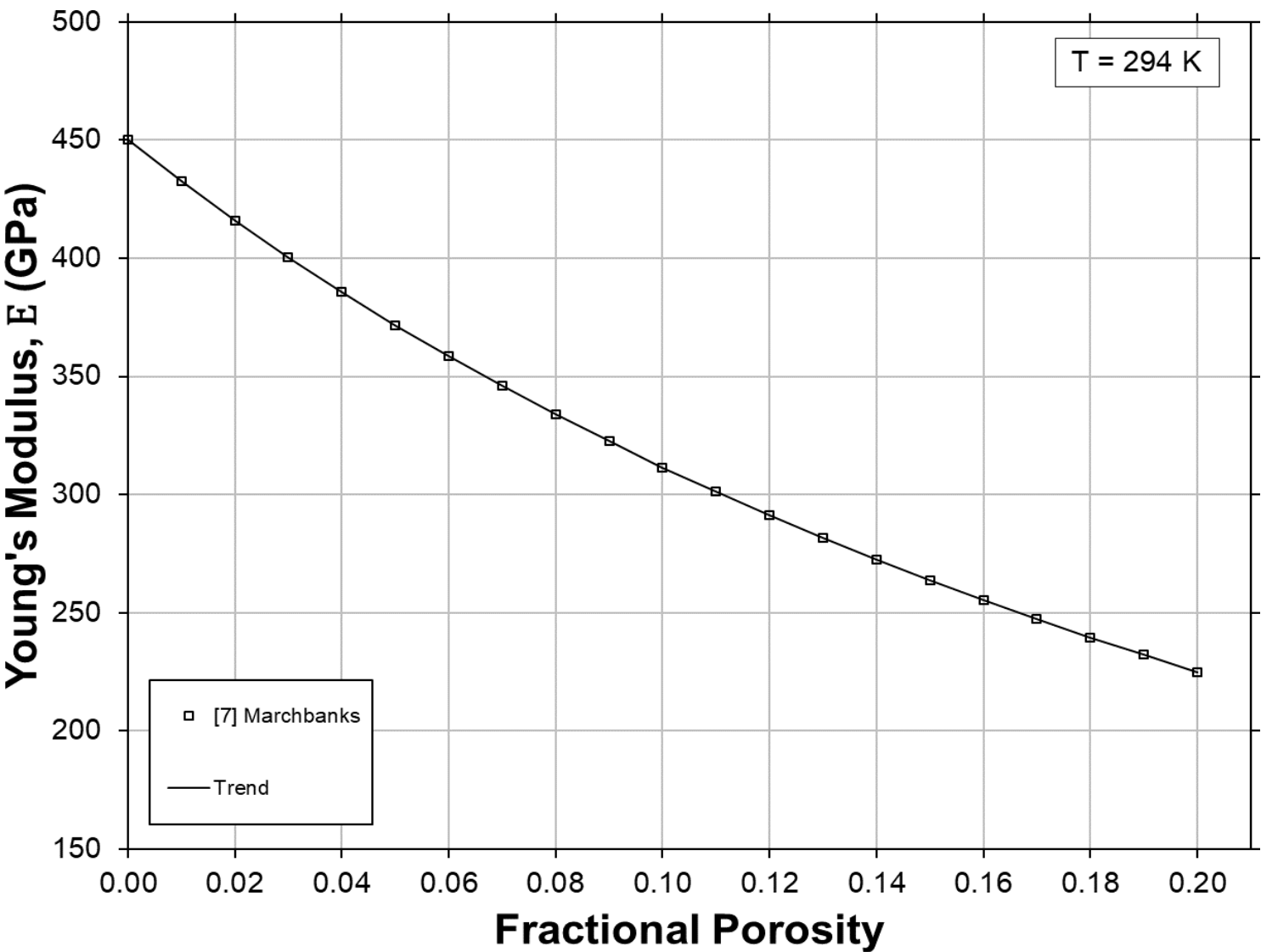


Figure 7.2.1-7: Young's Modulus versus Fractional Porosity of  $B_4C$  adapted from Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Young's Modulus with Porosity

Room Temperature

| Porosity ( P ) | Young's Modulus ( E ) |           |
|----------------|-----------------------|-----------|
|                | GPa                   | ( Msi )   |
| 0.00           | 450.00                | ( 65.29 ) |
| 0.02           | 416.04                | ( 60.37 ) |
| 0.04           | 385.73                | ( 55.97 ) |
| 0.06           | 358.49                | ( 52.02 ) |
| 0.08           | 333.89                | ( 48.45 ) |
| 0.10           | 311.56                | ( 45.21 ) |
| 0.12           | 291.20                | ( 42.25 ) |
| 0.14           | 272.56                | ( 39.55 ) |
| 0.16           | 255.43                | ( 37.06 ) |
| 0.18           | 239.64                | ( 34.77 ) |
| 0.20           | 225.03                | ( 32.65 ) |

**Application Notes:** Data for Young's modulus is collected from reference [7] and fitted with the equation below to approximate property trend with respect to both temperature and fractional porosity. Data in the table is for T = 294 K.

## Fit Equations:

$$E(T) = \left( \frac{1 - P}{1 + A \cdot P} \right) \cdot \left( A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right)$$

$E(T)$  = Young's Modulus [GPa]

$P$  = Fractional Porosity for  $0 \leq P < 0.2$

$T$  = Temperature [K]

## Constants:

Porosity Range:  $0 \leq P < 0.2$

Temperature [K] 294

$A = 2.999$

$A_0 = 452.1$

$A_1 = -7.101$

$A_2 = -1.540E-08$

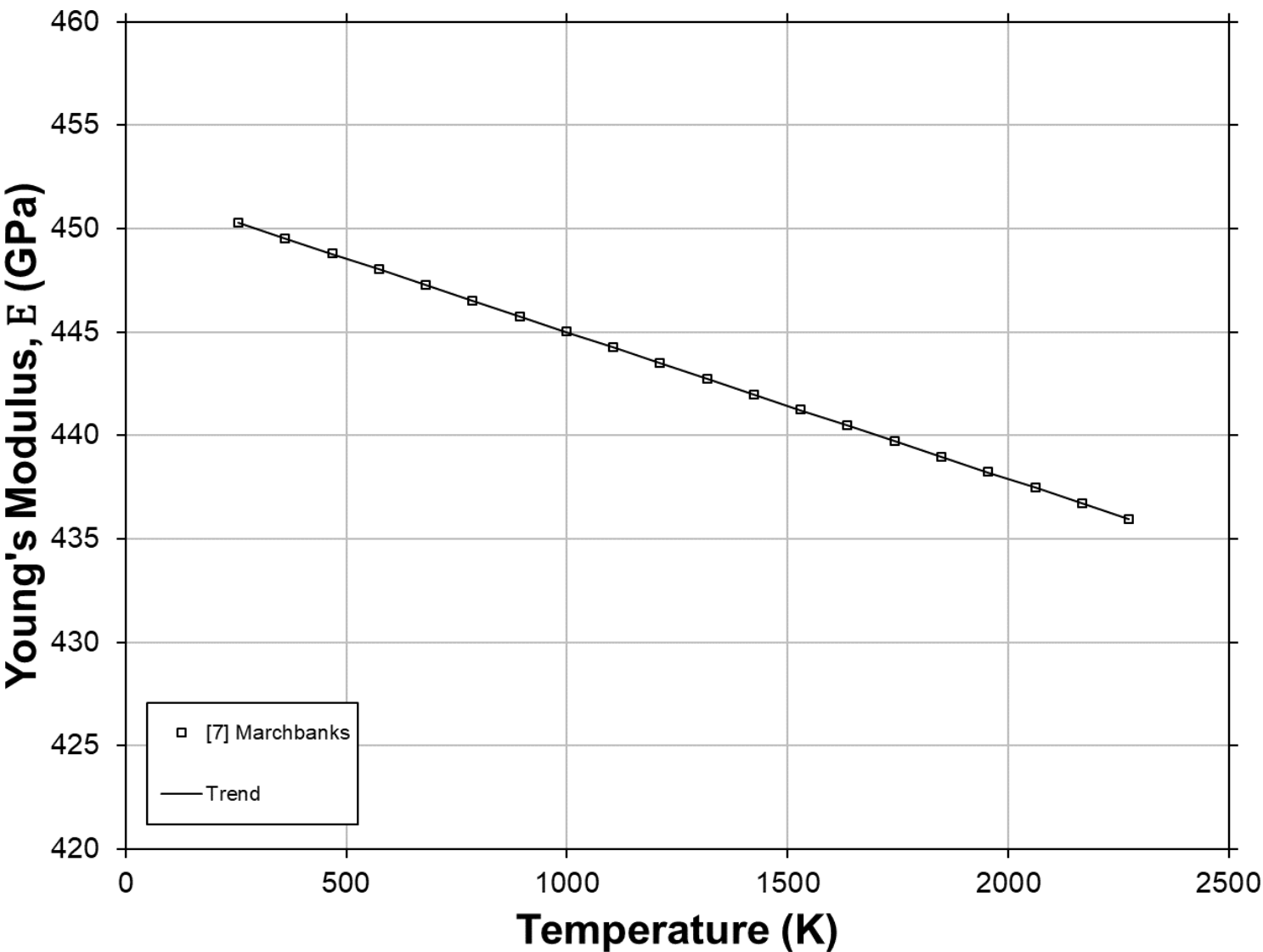


Figure 7.2.1-8: Young's Modulus versus Temperature of  $B_4C$  adapted from Marchbanks (1976)



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 255               | ( -0.7 )   | 450.29                | ( 65.34 ) | 1000              | ( 1340.3 ) | 445.00                | ( 64.57 ) |
| 300               | ( 80.3 )   | 449.97                | ( 65.29 ) | 1050              | ( 1430.3 ) | 444.64                | ( 64.52 ) |
| 350               | ( 170.3 )  | 449.61                | ( 65.24 ) | 1100              | ( 1520.3 ) | 444.29                | ( 64.47 ) |
| 400               | ( 260.3 )  | 449.26                | ( 65.19 ) | 1200              | ( 1700.3 ) | 443.58                | ( 64.36 ) |
| 450               | ( 350.3 )  | 448.90                | ( 65.14 ) | 1300              | ( 1880.3 ) | 442.87                | ( 64.26 ) |
| 500               | ( 440.3 )  | 448.55                | ( 65.08 ) | 1400              | ( 2060.3 ) | 442.16                | ( 64.16 ) |
| 550               | ( 530.3 )  | 448.19                | ( 65.03 ) | 1500              | ( 2240.3 ) | 441.45                | ( 64.05 ) |
| 600               | ( 620.3 )  | 447.84                | ( 64.98 ) | 1600              | ( 2420.3 ) | 440.74                | ( 63.95 ) |
| 650               | ( 710.3 )  | 447.48                | ( 64.93 ) | 1700              | ( 2600.3 ) | 440.03                | ( 63.85 ) |
| 700               | ( 800.3 )  | 447.13                | ( 64.88 ) | 1800              | ( 2780.3 ) | 439.32                | ( 63.75 ) |
| 750               | ( 890.3 )  | 446.77                | ( 64.83 ) | 1900              | ( 2960.3 ) | 438.61                | ( 63.64 ) |
| 800               | ( 980.3 )  | 446.42                | ( 64.78 ) | 2000              | ( 3140.3 ) | 437.90                | ( 63.54 ) |
| 850               | ( 1070.3 ) | 446.06                | ( 64.72 ) | 2100              | ( 3320.3 ) | 437.19                | ( 63.44 ) |
| 900               | ( 1160.3 ) | 445.71                | ( 64.67 ) | 2200              | ( 3500.3 ) | 436.48                | ( 63.33 ) |
| 950               | ( 1250.3 ) | 445.35                | ( 64.62 ) | 2274              | ( 3633.5 ) | 435.95                | ( 63.26 ) |

**Application Notes:** Data for Young's modulus is collected from reference [7] and fitted with the equation below to approximate property trend with respect to both temperature and fractional porosity. Data shown in table is for fractional porosity P=0.

## Fit Equations:

$$E(T) = \left( \frac{1 - P}{1 + A \cdot P} \right) \cdot \left( A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right)$$

$E(T)$  = Young's Modulus [GPa]

$P$  = Fractional Porosity for  $0 < P < 0.2$

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $255 \leq T < 2274$   
 Fractional Porosity (P) 0  
 $A = 2.999$   
 $A_0 = 452.1$   
 $A_1 = -7.101$   
 $A_2 = -1.540E-08$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide ( $B_4C$ )

Revision 0: 08-05-2020

Modulus of Rupture with Porosity

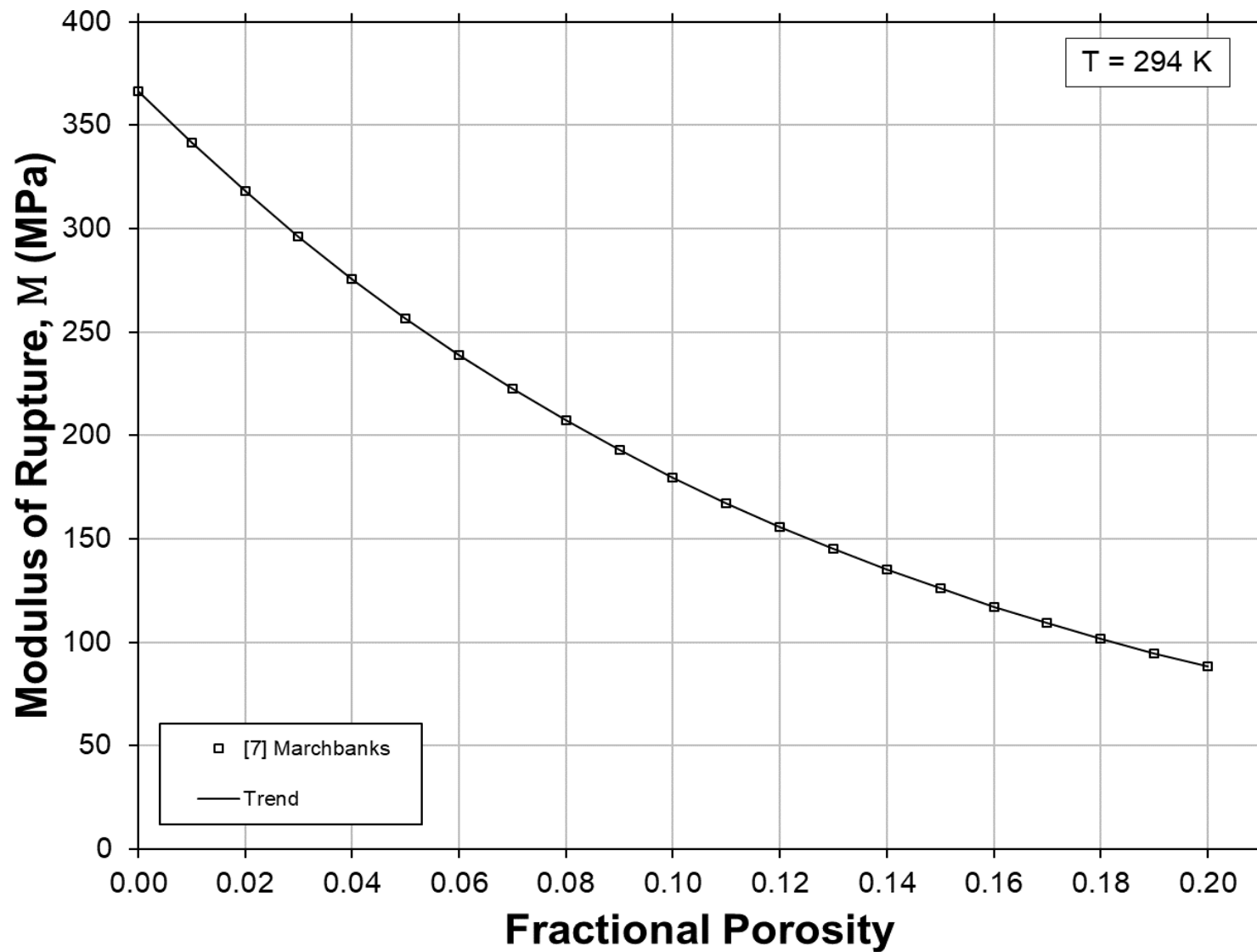


Figure 7.2.1-9: Modulus of Rupture verse Fractional Porosity of  $B_4C$  adapted from Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Modulus of Rupture with Porosity

Room Temperature

| Porosity ( P ) | Modulus of Rupture ( M ) |           |
|----------------|--------------------------|-----------|
|                | MPa                      | ( Ksi )   |
| 0.00           | 366.67                   | ( 53.20 ) |
| 0.02           | 317.98                   | ( 46.14 ) |
| 0.04           | 275.76                   | ( 40.01 ) |
| 0.06           | 239.15                   | ( 34.70 ) |
| 0.08           | 207.39                   | ( 30.09 ) |
| 0.10           | 179.86                   | ( 26.10 ) |
| 0.12           | 155.97                   | ( 22.63 ) |
| 0.14           | 135.26                   | ( 19.63 ) |
| 0.16           | 117.30                   | ( 17.02 ) |
| 0.18           | 101.73                   | ( 14.76 ) |
| 0.20           | 88.22                    | ( 12.80 ) |

**Application Notes:** Data for modulus of rupture has been collected from reference [7] and fitted with the equation below to approximate property trend with respect to both temperature and fractional porosity. Data shown in the table is for T = 294 K.

## Fit Equations:

$$M(T) = e^{(A \cdot P)} \cdot \left( A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 \right)$$

$M(T)$  = Modulus of Rupture

$P$  = Fractional Porosity for  $0 \leq P < 0.2$

$T$  = Temperature [K]

## Constants:

Porosity Range:  $0 \leq P < 0.2$

Temperature [K] 294

A = -7.123

A0 = 358.9

A1 = 41.76

A2 = -52.43



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide ( $B_4C$ )

Revision 0: 08-05-2020

Modulus of Rupture with Temperature

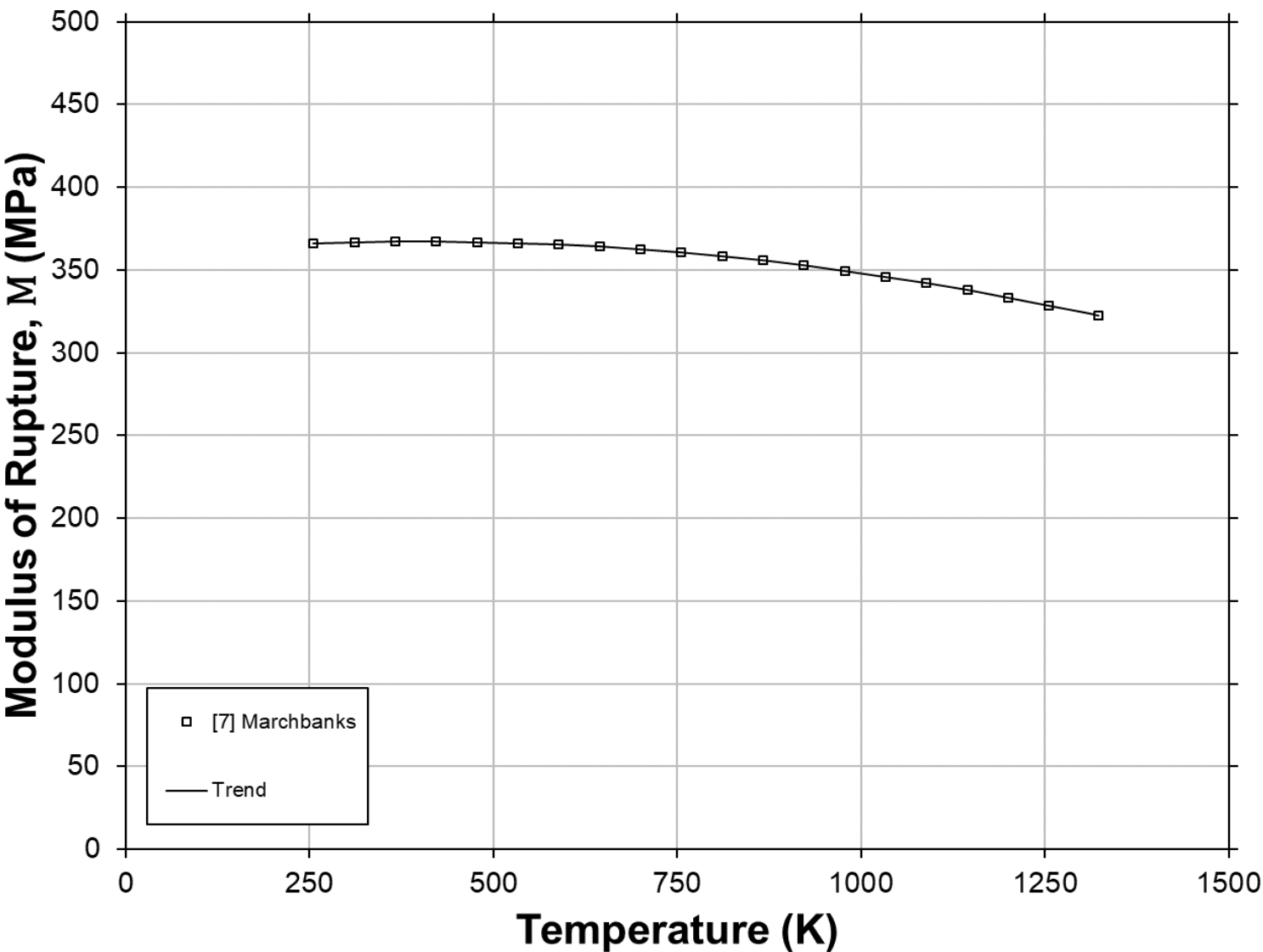


Figure 7.2.1-10: Modulus of Rupture versus Temperature of  $B_4C$  adapted from Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Modulus of Rupture with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Modulus of Rupture ( M ) |           | Temperature ( T ) |            | Modulus of Rupture ( M ) |           |
|-------------------|-----------|--------------------------|-----------|-------------------|------------|--------------------------|-----------|
| K                 | ( °F )    | MPa                      | ( Ksi )   | K                 | ( °F )     | MPa                      | ( Ksi )   |
| 296               | ( 73.1 )  | 366.67                   | ( 53.20 ) | 650               | ( 710.3 )  | 363.89                   | ( 52.80 ) |
| 300               | ( 80.3 )  | 366.71                   | ( 53.21 ) | 675               | ( 755.3 )  | 363.20                   | ( 52.70 ) |
| 325               | ( 125.3 ) | 366.93                   | ( 53.24 ) | 700               | ( 800.3 )  | 362.44                   | ( 52.59 ) |
| 350               | ( 170.3 ) | 367.09                   | ( 53.27 ) | 725               | ( 845.3 )  | 361.62                   | ( 52.47 ) |
| 375               | ( 215.3 ) | 367.19                   | ( 53.28 ) | 750               | ( 890.3 )  | 360.73                   | ( 52.34 ) |
| 400               | ( 260.3 ) | 367.22                   | ( 53.28 ) | 775               | ( 935.3 )  | 359.77                   | ( 52.20 ) |
| 425               | ( 305.3 ) | 367.18                   | ( 53.28 ) | 800               | ( 980.3 )  | 358.75                   | ( 52.06 ) |
| 450               | ( 350.3 ) | 367.07                   | ( 53.26 ) | 825               | ( 1025.3 ) | 357.67                   | ( 51.90 ) |
| 475               | ( 395.3 ) | 366.91                   | ( 53.24 ) | 850               | ( 1070.3 ) | 356.52                   | ( 51.73 ) |
| 500               | ( 440.3 ) | 366.67                   | ( 53.20 ) | 900               | ( 1160.3 ) | 354.02                   | ( 51.37 ) |
| 525               | ( 485.3 ) | 366.37                   | ( 53.16 ) | 1000              | ( 1340.3 ) | 348.23                   | ( 50.53 ) |
| 550               | ( 530.3 ) | 366.01                   | ( 53.11 ) | 1100              | ( 1520.3 ) | 341.40                   | ( 49.54 ) |
| 575               | ( 575.3 ) | 365.58                   | ( 53.05 ) | 1200              | ( 1700.3 ) | 333.51                   | ( 48.39 ) |
| 600               | ( 620.3 ) | 365.08                   | ( 52.97 ) | 1300              | ( 1880.3 ) | 324.58                   | ( 47.10 ) |
| 625               | ( 665.3 ) | 364.52                   | ( 52.89 ) | 1324              | ( 1923.5 ) | 322.28                   | ( 46.76 ) |

**Application Notes:** : Data for modulus of rupture has been collected from reference [7] and fitted with the equation below to approximate property trend with respect to both temperature and fractional porosity. Data shown in table is for fractional porosity P=0.

## Fit Equations:

$$M(T) = e^{(A \cdot P)} \cdot \left( A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 \right)$$

$M(T)$  = Modulus of Rupture

$P$  = Fractional Porosity for  $0 \leq P < 0.2$

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $255 \leq T < 1324$

Fractional Porosity (P) 0

A = -7.123

A0 = 358.9

A1 = 41.76

A2 = -52.43

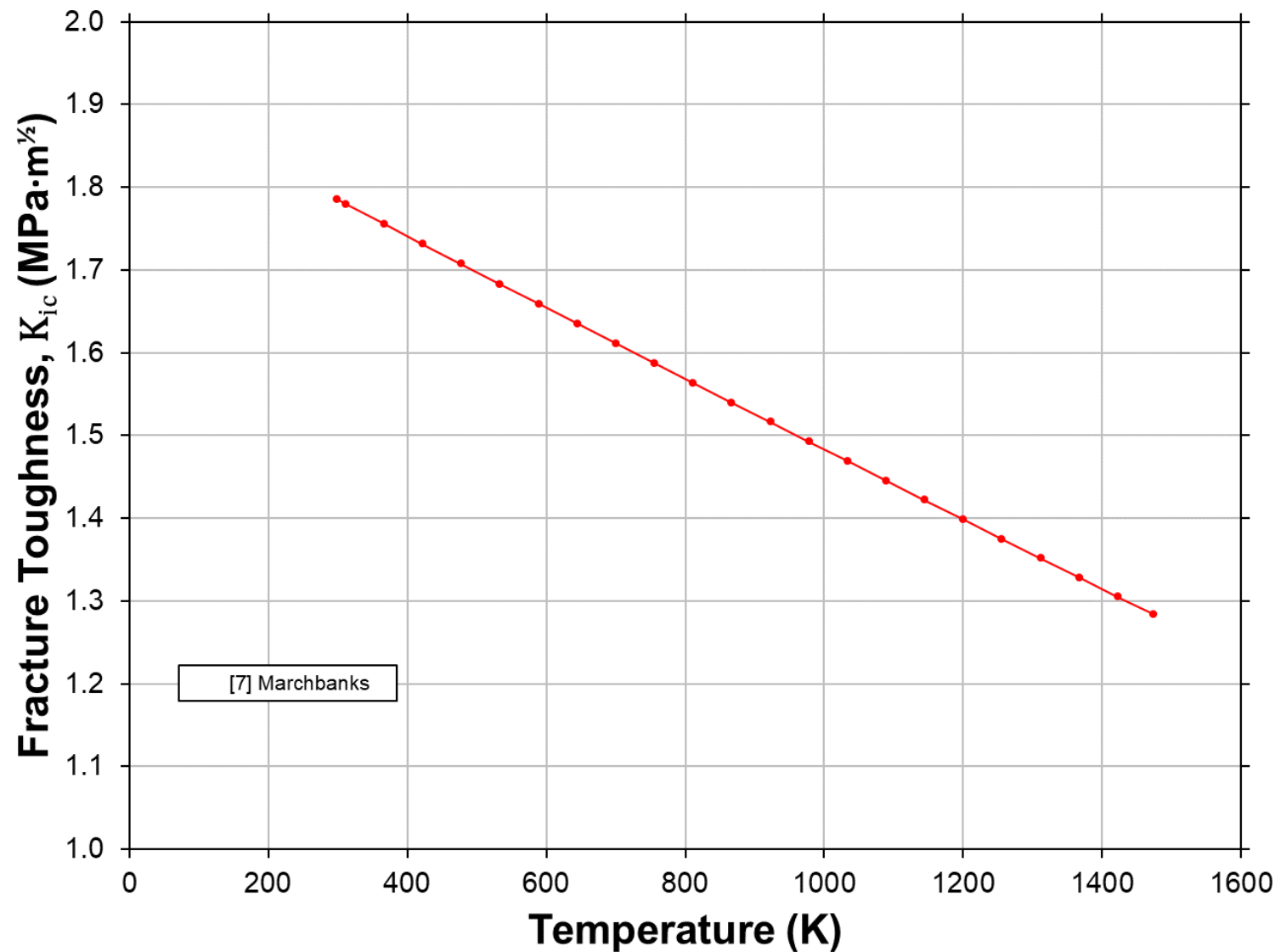


Figure 7.2.1-11: Fracture Toughness versus Temperature of B4C adapted from Marchbanks (1976).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Fracture Toughness with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Fracture Toughness ( K <sub>IC</sub> ) |                           | Temperature ( T ) |            | Fracture Toughness ( K <sub>IC</sub> ) |                           |
|-------------------|-----------|----------------------------------------|---------------------------|-------------------|------------|----------------------------------------|---------------------------|
| K                 | ( °F )    | MPa m <sup>1/2</sup>                   | ( Ksi in <sup>1/2</sup> ) | K                 | ( °F )     | MPa m <sup>1/2</sup>                   | ( Ksi in <sup>1/2</sup> ) |
| 296               | ( 73.1 )  | 1.79                                   | ( 1.63 )                  | 650               | ( 710.3 )  | 1.63                                   | ( 1.49 )                  |
| 300               | ( 80.3 )  | 1.78                                   | ( 1.62 )                  | 675               | ( 755.3 )  | 1.62                                   | ( 1.48 )                  |
| 325               | ( 125.3 ) | 1.77                                   | ( 1.61 )                  | 700               | ( 800.3 )  | 1.61                                   | ( 1.47 )                  |
| 350               | ( 170.3 ) | 1.76                                   | ( 1.60 )                  | 725               | ( 845.3 )  | 1.60                                   | ( 1.46 )                  |
| 375               | ( 215.3 ) | 1.75                                   | ( 1.59 )                  | 750               | ( 890.3 )  | 1.59                                   | ( 1.45 )                  |
| 400               | ( 260.3 ) | 1.74                                   | ( 1.58 )                  | 775               | ( 935.3 )  | 1.58                                   | ( 1.44 )                  |
| 425               | ( 305.3 ) | 1.73                                   | ( 1.57 )                  | 800               | ( 980.3 )  | 1.57                                   | ( 1.43 )                  |
| 450               | ( 350.3 ) | 1.72                                   | ( 1.56 )                  | 850               | ( 1070.3 ) | 1.55                                   | ( 1.41 )                  |
| 475               | ( 395.3 ) | 1.71                                   | ( 1.55 )                  | 900               | ( 1160.3 ) | 1.53                                   | ( 1.39 )                  |
| 500               | ( 440.3 ) | 1.70                                   | ( 1.54 )                  | 1000              | ( 1340.3 ) | 1.48                                   | ( 1.35 )                  |
| 525               | ( 485.3 ) | 1.69                                   | ( 1.54 )                  | 1100              | ( 1520.3 ) | 1.44                                   | ( 1.31 )                  |
| 550               | ( 530.3 ) | 1.68                                   | ( 1.53 )                  | 1200              | ( 1700.3 ) | 1.40                                   | ( 1.27 )                  |
| 575               | ( 575.3 ) | 1.67                                   | ( 1.52 )                  | 1300              | ( 1880.3 ) | 1.36                                   | ( 1.23 )                  |
| 600               | ( 620.3 ) | 1.65                                   | ( 1.51 )                  | 1400              | ( 2060.3 ) | 1.31                                   | ( 1.20 )                  |
| 625               | ( 665.3 ) | 1.64                                   | ( 1.50 )                  | 1474              | ( 2193.5 ) | 1.28                                   | ( 1.17 )                  |

**Application Notes:** Data and trend for fracture toughness is from reference [7]. The equation from this reference is shown below and approximates property trend with respect to temperature.

## Fit Equations:

$$K_{ic}(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$$K_{ic}(T) = \text{Fracture Toughness} [MPa \cdot \sqrt{m}]$$

$$T = \text{Temperature} [K]$$

## Constants:

$$T. \text{ Range } [K]: \quad 296 \leq T < 1474$$

$$A0 = 1.916E+00$$

$$A1 = -4.408E-01$$

$$A2 = 7.926E-03$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 300                 | 2500                                  |                                                         | -                                           | 0.003                                              | 3.99                                      | 0.941                                          | 450.0                              |                                  |                             |
| 350                 | 2498                                  |                                                         | 28.57                                       | 0.024                                              | 4.10                                      | 1.009                                          | 449.6                              |                                  |                             |
| 400                 | 2497                                  |                                                         | 27.13                                       | 0.046                                              | 4.20                                      | 1.077                                          | 449.3                              |                                  |                             |
| 450                 | 2495                                  |                                                         | 25.76                                       | 0.069                                              | 4.29                                      | 1.145                                          | 448.9                              |                                  |                             |
| 500                 | 2493                                  |                                                         | 24.47                                       | 0.092                                              | 4.39                                      | 1.212                                          | 448.5                              |                                  |                             |
| 550                 | 2491                                  |                                                         | 23.26                                       | 0.117                                              | 4.48                                      | 1.279                                          | 448.2                              |                                  |                             |
| 600                 | 2489                                  |                                                         | 22.14                                       | 0.142                                              | 4.57                                      | 1.344                                          | 447.8                              |                                  |                             |
| 700                 | 2486                                  |                                                         | 20.14                                       | 0.194                                              | 4.74                                      | 1.469                                          | 447.1                              |                                  |                             |
| 800                 | 2481                                  |                                                         | 18.48                                       | 0.249                                              | 4.90                                      | 1.589                                          | 446.4                              |                                  |                             |
| 1000                | 2473                                  |                                                         | 16.08                                       | 0.367                                              | 5.19                                      | 1.810                                          | 445.0                              |                                  |                             |
| 1100                | 2468                                  |                                                         | 15.27                                       | 0.429                                              | 5.33                                      | 1.910                                          | 444.3                              |                                  |                             |
| 1200                | 2463                                  |                                                         | 14.69                                       | 0.494                                              | 5.46                                      | 2.004                                          | 443.6                              |                                  |                             |
| 1300                | 2458                                  |                                                         | 14.31                                       | 0.560                                              | 5.58                                      | 2.092                                          | 442.9                              |                                  |                             |
| 1400                | 2453                                  |                                                         | 14.08                                       | 0.629                                              | 5.69                                      | 2.173                                          | 442.2                              |                                  |                             |
| 1500                | 2448                                  |                                                         | 13.98                                       | 0.700                                              | 5.80                                      | 2.248                                          | 441.4                              |                                  |                             |
| 1600                | 2443                                  |                                                         | 13.99                                       | 0.772                                              | 5.90                                      | 2.316                                          | 440.7                              |                                  |                             |
| 1700                | 2438                                  |                                                         | 14.09                                       | 0.846                                              | 6.00                                      | 2.378                                          | 440.0                              |                                  |                             |
| 1800                | 2432                                  |                                                         | 14.26                                       | 0.921                                              | 6.10                                      | 2.433                                          | 439.3                              |                                  |                             |
| 1900                | 2427                                  |                                                         | 14.48                                       | 0.998                                              | 6.19                                      | 2.481                                          | 438.6                              |                                  |                             |
| 2000                | 2421                                  |                                                         | 14.74                                       | 1.076                                              | 6.28                                      | 2.523                                          | 437.9                              |                                  |                             |
| 2100                | 2415                                  |                                                         | 15.04                                       | 1.155                                              | 6.37                                      | 2.558                                          | 437.2                              |                                  |                             |
| 2200                | 2410                                  |                                                         | 15.36                                       | 1.235                                              | 6.45                                      | 2.587                                          | 436.5                              |                                  |                             |
| 2300                | 2404                                  |                                                         | 15.70                                       | 1.316                                              | 6.54                                      | 2.609                                          | -                                  |                                  |                             |
| 2400                | 2398                                  |                                                         | 16.05                                       | 1.397                                              | 6.62                                      | 2.625                                          | -                                  |                                  |                             |
| 2500                | 2392                                  |                                                         | 16.40                                       | 1.479                                              | 6.71                                      | 2.634                                          | -                                  |                                  |                             |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.1 Boron Carbide (B<sub>4</sub>C)

**Revision 0: 08-05-2020**

**References**

- [1] J.C. Hedge, C. Kostenko, J.I. Lang, Thermal Properties of Refractory Alloys, IIT Research Institute, Chicago, IL, 1963.
- [2] Y.S. Touloukian, R.W. Powell, C.Y. Ho, P.G. Klemens, Thermal conductivity - Nonmetallic Solids, Thermophysical Properties of Matter - The TPRC Data Series, Volume 2, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1971.
- [3] P.E. MacDonald, L.B. Thompson, MATPRO: a handbook of materials properties for use in the analysis of light water reactor fuel rod behavior, SEE CODE- 9502158 Aerojet Nuclear Co., Idaho Falls, Idaho (USA). Idaho National Engineering Lab., 1976.
- [4] F. Marchbanks, M.A. Moen, R.E. Irvin, Nuclear Systems Materials Handbook: Design Data, International Conference on the Mechanical Behavior of Materials, Oak Ridge National Laboratory, Boston, MA, USA, 1976.
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- [6] Y.S. Touloukian, E.H. Buyco, Specific Heat - Nonmetallic Solids, Thermophysical Properties of Matter - the TPRC Data Series, Thermophysical and Electronic Properties Information Analysis Center, Lafayette, IN, 1970.
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## **7 Refractory Ceramics**

### **7.2 Carbides**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 2.1: 08-25-2023

General

## Room Temperature Properties

|                                           |         |
|-------------------------------------------|---------|
| Molar Mass, [g/mol]                       | 103.23  |
| Theoretical Density, [kg/m <sup>3</sup> ] | 6,730   |
| Melting Point, [K]                        | 3693±30 |
| Specific Heat, [J/(g-K)]                  | 0.363   |
| Thermal Conductivity, [W/(m-K)]           | 26.3    |
| Linear expansion coefficient, [μm/(m-K)]  | 5.10    |
| Electrical resistivity, [μΩ-m]            | 0.61    |
| Young's Modulus, [GPa]                    | 397.9   |
| Shear Modulus, [GPa]                      | 161.8   |
| Poisson's Ratio, [-]                      | 0.229   |

## Zirconium – Carbon Phase Diagram

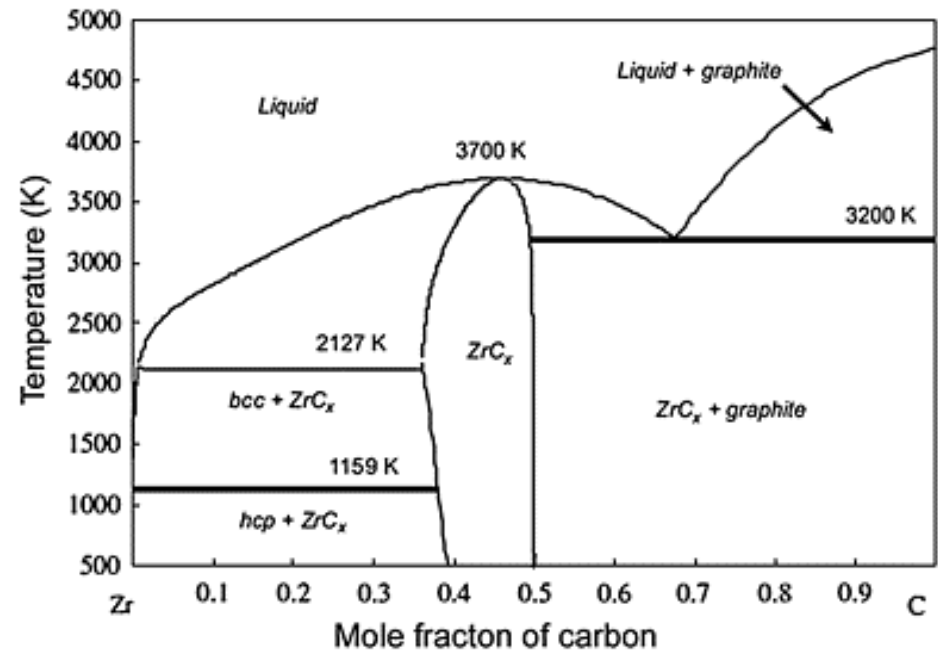


Figure 7.2.2-1: Zirconium – Carbon Phase Diagram [1].



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Density with Temperature

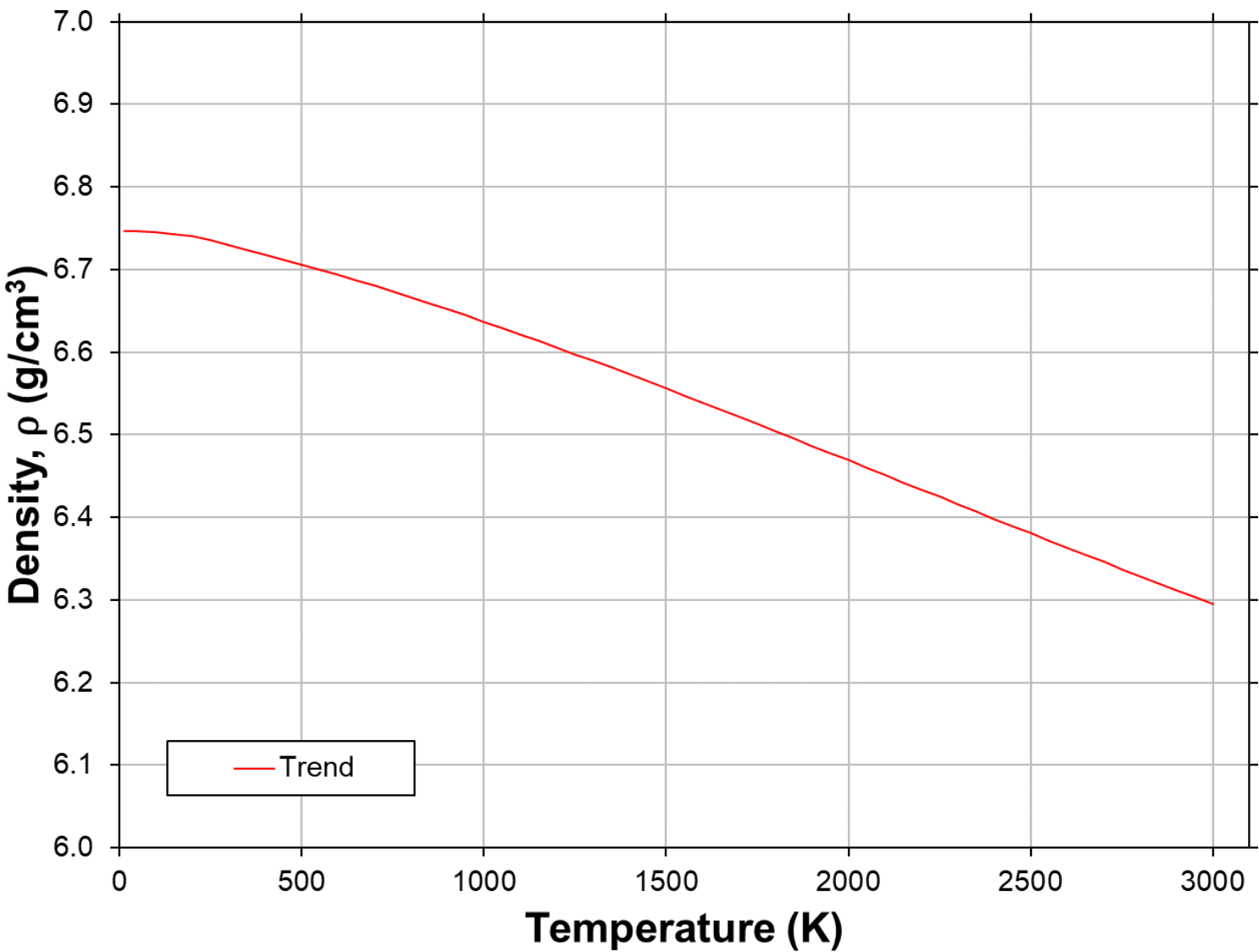


Figure 7.2.2-2: Density versus Temperature for ZrC. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 100               | ( -279.7 ) | 6745              | ( 421.1 )              | 1600              | ( 2420.3 ) | 6539              | ( 408.2 )              |
| 200               | ( -99.7 )  | 6740              | ( 420.8 )              | 1700              | ( 2600.3 ) | 6522              | ( 407.2 )              |
| 300               | ( 80.3 )   | 6729              | ( 420.1 )              | 1800              | ( 2780.3 ) | 6504              | ( 406.1 )              |
| 400               | ( 260.3 )  | 6718              | ( 419.4 )              | 1900              | ( 2960.3 ) | 6487              | ( 405.0 )              |
| 500               | ( 440.3 )  | 6706              | ( 418.7 )              | 2000              | ( 3140.3 ) | 6469              | ( 403.9 )              |
| 600               | ( 620.3 )  | 6694              | ( 417.9 )              | 2100              | ( 3320.3 ) | 6451              | ( 402.8 )              |
| 700               | ( 800.3 )  | 6680              | ( 417.0 )              | 2200              | ( 3500.3 ) | 6434              | ( 401.6 )              |
| 800               | ( 980.3 )  | 6666              | ( 416.2 )              | 2300              | ( 3680.3 ) | 6416              | ( 400.5 )              |
| 900               | ( 1160.3 ) | 6652              | ( 415.3 )              | 2400              | ( 3860.3 ) | 6398              | ( 399.4 )              |
| 1000              | ( 1340.3 ) | 6637              | ( 414.3 )              | 2500              | ( 4040.3 ) | 6381              | ( 398.3 )              |
| 1100              | ( 1520.3 ) | 6622              | ( 413.4 )              | 2600              | ( 4220.3 ) | 6363              | ( 397.3 )              |
| 1200              | ( 1700.3 ) | 6606              | ( 412.4 )              | 2700              | ( 4400.3 ) | 6346              | ( 396.2 )              |
| 1300              | ( 1880.3 ) | 6590              | ( 411.4 )              | 2800              | ( 4580.3 ) | 6329              | ( 395.1 )              |
| 1400              | ( 2060.3 ) | 6573              | ( 410.4 )              | 2900              | ( 4760.3 ) | 6312              | ( 394.1 )              |
| 1500              | ( 2240.3 ) | 6556              | ( 409.3 )              | 3000              | ( 4940.3 ) | 6296              | ( 393.0 )              |

**Application Notes:** Density is calculated as a function of Thermal Expansion, as seen in the equation below to approximate the property trend with respect to temperature.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 6,730 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $12 \leq T \leq 3000$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

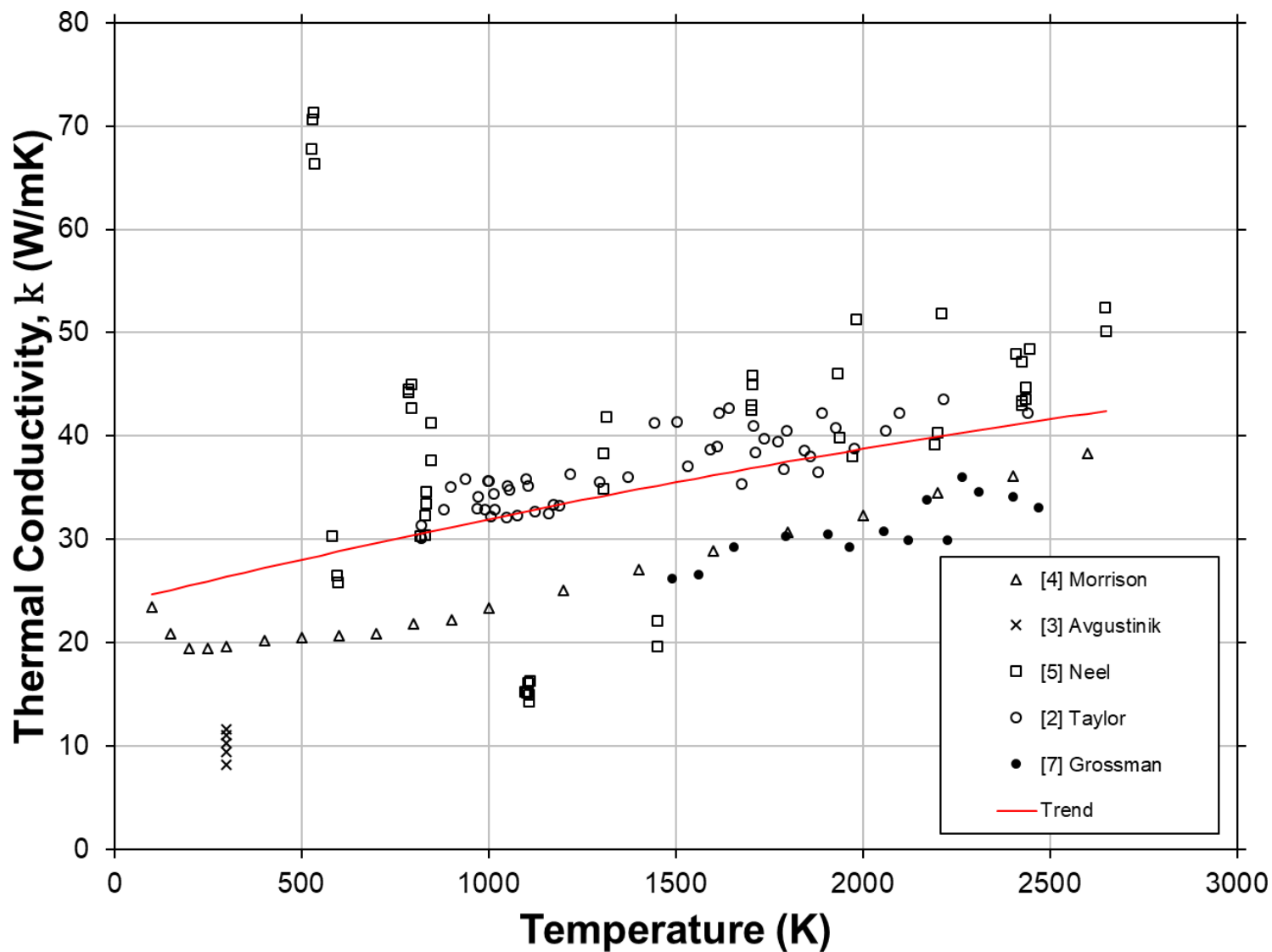
7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature





# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 100               | ( -279.7 ) | 24.64                      | ( 170.98 )                          | 1400              | ( 2060.3 ) | 34.85                      | ( 241.76 )                          |
| 150               | ( -189.7 ) | 25.08                      | ( 174.00 )                          | 1500              | ( 2240.3 ) | 35.53                      | ( 246.52 )                          |
| 200               | ( -99.7 )  | 25.51                      | ( 177.00 )                          | 1600              | ( 2420.3 ) | 36.20                      | ( 251.19 )                          |
| 300               | ( 80.3 )   | 26.37                      | ( 182.94 )                          | 1700              | ( 2600.3 ) | 36.86                      | ( 255.76 )                          |
| 400               | ( 260.3 )  | 27.21                      | ( 188.77 )                          | 1800              | ( 2780.3 ) | 37.51                      | ( 260.23 )                          |
| 500               | ( 440.3 )  | 28.03                      | ( 194.51 )                          | 1900              | ( 2960.3 ) | 38.14                      | ( 264.61 )                          |
| 600               | ( 620.3 )  | 28.85                      | ( 200.15 )                          | 2000              | ( 3140.3 ) | 38.75                      | ( 268.88 )                          |
| 700               | ( 800.3 )  | 29.65                      | ( 205.69 )                          | 2100              | ( 3320.3 ) | 39.36                      | ( 273.06 )                          |
| 800               | ( 980.3 )  | 30.43                      | ( 211.13 )                          | 2200              | ( 3500.3 ) | 39.95                      | ( 277.14 )                          |
| 900               | ( 1160.3 ) | 31.20                      | ( 216.48 )                          | 2300              | ( 3680.3 ) | 40.52                      | ( 281.13 )                          |
| 1000              | ( 1340.3 ) | 31.96                      | ( 221.73 )                          | 2400              | ( 3860.3 ) | 41.08                      | ( 285.02 )                          |
| 1100              | ( 1520.3 ) | 32.70                      | ( 226.89 )                          | 2500              | ( 4040.3 ) | 41.63                      | ( 288.81 )                          |
| 1200              | ( 1700.3 ) | 33.43                      | ( 231.94 )                          | 2600              | ( 4220.3 ) | 42.16                      | ( 292.50 )                          |
| 1300              | ( 1880.3 ) | 34.14                      | ( 236.90 )                          | 2650              | ( 4310.3 ) | 42.42                      | ( 294.31 )                          |

**Application Notes:** Data for thermal conductivity is collected from references [2-7] and fitted with the equation below to approximate the property trend with respect to temperature. There is a large scatter in this data, which is primarily experimental. Due to this scatter, the trend attempts to capture the behavior of lower points.

## Fit Equation:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$$k(T) = \text{Thermal Conductivity } [W / (m \cdot K)]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 100 \leq T \leq 2650$$

$$A0 = 23.76$$

$$A1 = 8.900$$

$$A2 = -0.7014$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

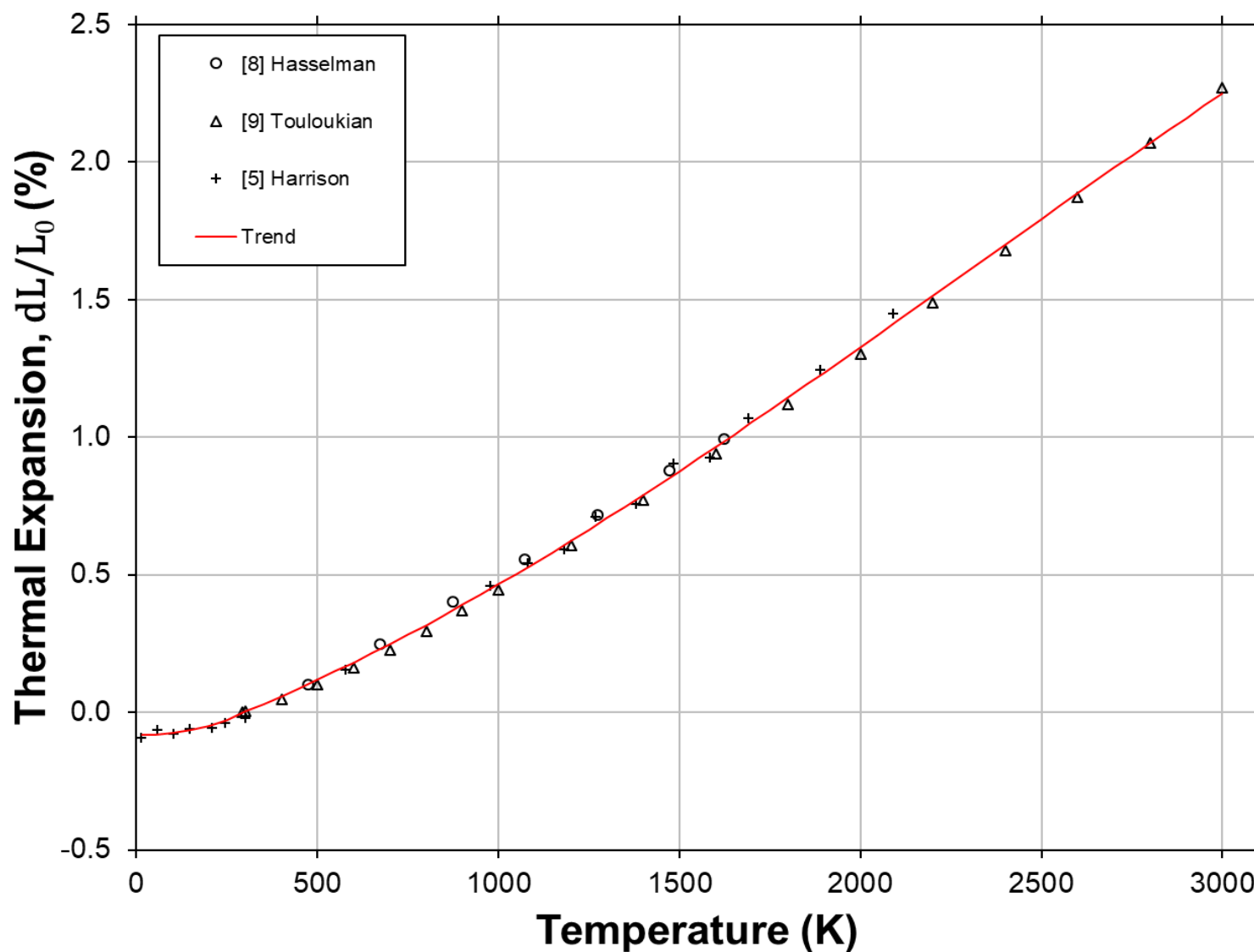


Figure 7.2.2-4: Thermal Expansion versus Temperature for ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 100               | ( -279.7 ) | -0.076                                  | 1600              | ( 2420.3 ) | 0.964                                   |
| 200               | ( -99.7 )  | -0.050                                  | 1700              | ( 2600.3 ) | 1.053                                   |
| 300               | ( 80.3 )   | 0.003                                   | 1800              | ( 2780.3 ) | 1.143                                   |
| 400               | ( 260.3 )  | 0.058                                   | 1900              | ( 2960.3 ) | 1.235                                   |
| 500               | ( 440.3 )  | 0.118                                   | 2000              | ( 3140.3 ) | 1.327                                   |
| 600               | ( 620.3 )  | 0.181                                   | 2100              | ( 3320.3 ) | 1.420                                   |
| 700               | ( 800.3 )  | 0.247                                   | 2200              | ( 3500.3 ) | 1.513                                   |
| 800               | ( 980.3 )  | 0.317                                   | 2300              | ( 3680.3 ) | 1.606                                   |
| 900               | ( 1160.3 ) | 0.390                                   | 2400              | ( 3860.3 ) | 1.700                                   |
| 1000              | ( 1340.3 ) | 0.465                                   | 2500              | ( 4040.3 ) | 1.793                                   |
| 1100              | ( 1520.3 ) | 0.543                                   | 2600              | ( 4220.3 ) | 1.886                                   |
| 1200              | ( 1700.3 ) | 0.623                                   | 2700              | ( 4400.3 ) | 1.978                                   |
| 1300              | ( 1880.3 ) | 0.705                                   | 2800              | ( 4580.3 ) | 2.069                                   |
| 1400              | ( 2060.3 ) | 0.790                                   | 2900              | ( 4760.3 ) | 2.160                                   |
| 1500              | ( 2240.3 ) | 0.876                                   | 3000              | ( 4940.3 ) | 2.249                                   |

**Application Notes:** Data for thermal expansion is collected from references [6, 8, 9] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

| T Range [K]: | $12 \leq T \leq 293$ | $293 < T \leq 3000$ |
|--------------|----------------------|---------------------|
| A0 =         | -0.07982             | -0.1399             |
| A1 =         | -0.07271             | 0.4103              |
| A2 =         | 1.118                | 0.2274              |
| A3 =         | 0                    | -0.03291            |

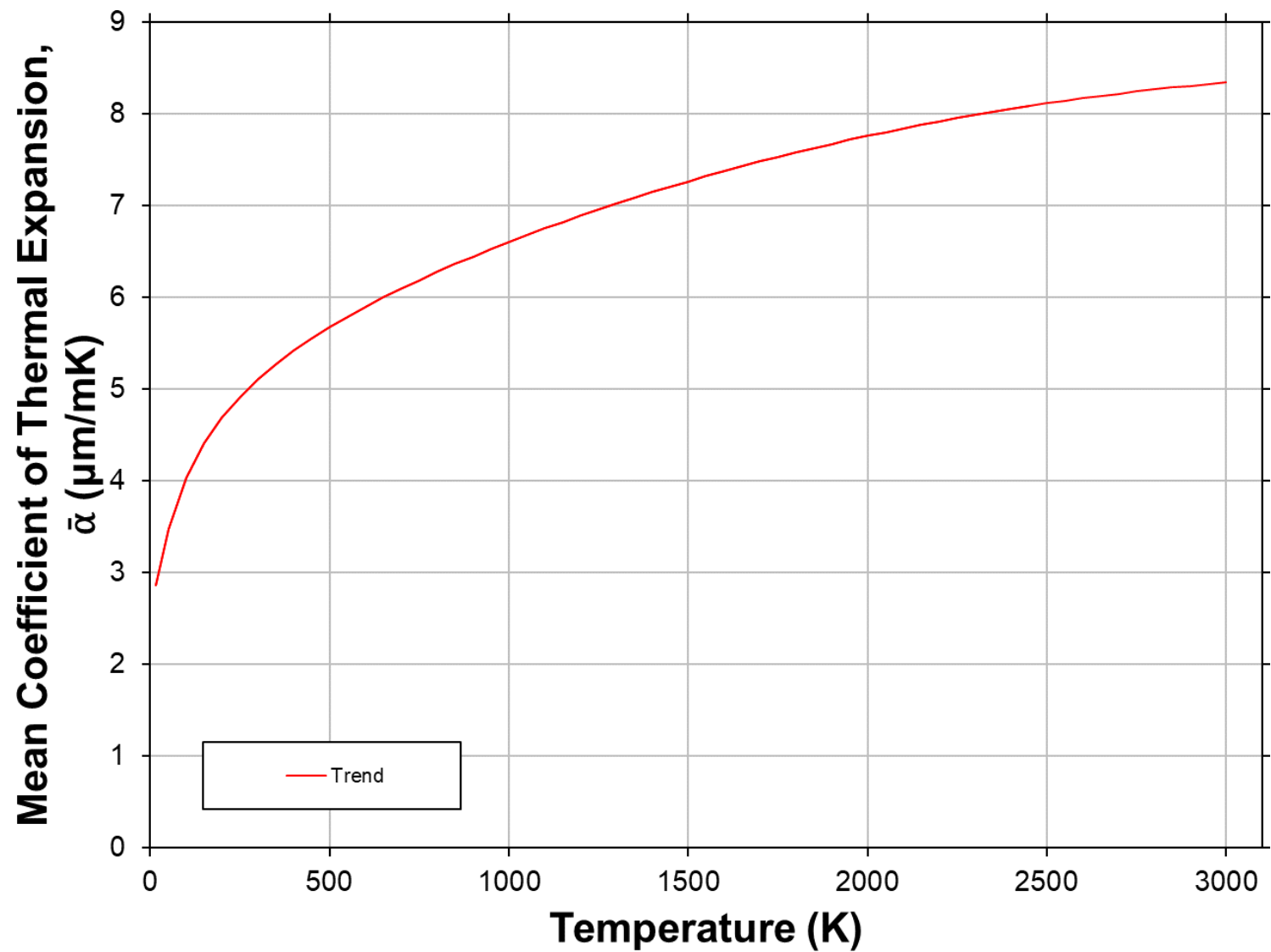


Figure 7.2.2-5: Mean Coefficient of Thermal Expansion versus Temperature for ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 100               | ( -279.7 ) | 4.028                                 | ( 2.238 )                                           | 1600              | ( 2420.3 ) | 7.372                                 | ( 4.096 )                                           |
| 200               | ( -99.7 )  | 4.687                                 | ( 2.604 )                                           | 1700              | ( 2600.3 ) | 7.477                                 | ( 4.154 )                                           |
| 300               | ( 80.3 )   | 5.103                                 | ( 2.835 )                                           | 1800              | ( 2780.3 ) | 7.576                                 | ( 4.209 )                                           |
| 400               | ( 260.3 )  | 5.415                                 | ( 3.008 )                                           | 1900              | ( 2960.3 ) | 7.669                                 | ( 4.261 )                                           |
| 500               | ( 440.3 )  | 5.672                                 | ( 3.151 )                                           | 2000              | ( 3140.3 ) | 7.757                                 | ( 4.309 )                                           |
| 600               | ( 620.3 )  | 5.894                                 | ( 3.275 )                                           | 2100              | ( 3320.3 ) | 7.838                                 | ( 4.355 )                                           |
| 700               | ( 800.3 )  | 6.093                                 | ( 3.385 )                                           | 2200              | ( 3500.3 ) | 7.915                                 | ( 4.397 )                                           |
| 800               | ( 980.3 )  | 6.276                                 | ( 3.486 )                                           | 2300              | ( 3680.3 ) | 7.986                                 | ( 4.437 )                                           |
| 900               | ( 1160.3 ) | 6.444                                 | ( 3.580 )                                           | 2400              | ( 3860.3 ) | 8.052                                 | ( 4.473 )                                           |
| 1000              | ( 1340.3 ) | 6.601                                 | ( 3.667 )                                           | 2500              | ( 4040.3 ) | 8.112                                 | ( 4.507 )                                           |
| 1100              | ( 1520.3 ) | 6.749                                 | ( 3.750 )                                           | 2600              | ( 4220.3 ) | 8.167                                 | ( 4.537 )                                           |
| 1200              | ( 1700.3 ) | 6.888                                 | ( 3.827 )                                           | 2700              | ( 4400.3 ) | 8.217                                 | ( 4.565 )                                           |
| 1300              | ( 1880.3 ) | 7.020                                 | ( 3.900 )                                           | 2800              | ( 4580.3 ) | 8.262                                 | ( 4.590 )                                           |
| 1400              | ( 2060.3 ) | 7.144                                 | ( 3.969 )                                           | 2900              | ( 4760.3 ) | 8.302                                 | ( 4.612 )                                           |
| 1500              | ( 2240.3 ) | 7.261                                 | ( 4.034 )                                           | 3000              | ( 4940.3 ) | 8.337                                 | ( 4.631 )                                           |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated as a function of thermal expansion. Calculated data was fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = \left[ A_0 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 + A_3 \cdot \left( \frac{T}{1000} \right)^3 \right] / \left[ 1 + A_{_0} \cdot \left( \frac{T}{1000} \right) \right]$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

T = Temperature [K]

## Constants:

T. Range [K]:  $12 \leq T \leq 3000$

A<sub>0</sub> = 2.49

A<sub>1</sub> = 49.67

A<sub>2</sub> = 15.26

A<sub>3</sub> = -2.158

A<sub>\_0</sub> = 8.886



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Specific Heat with Temperature

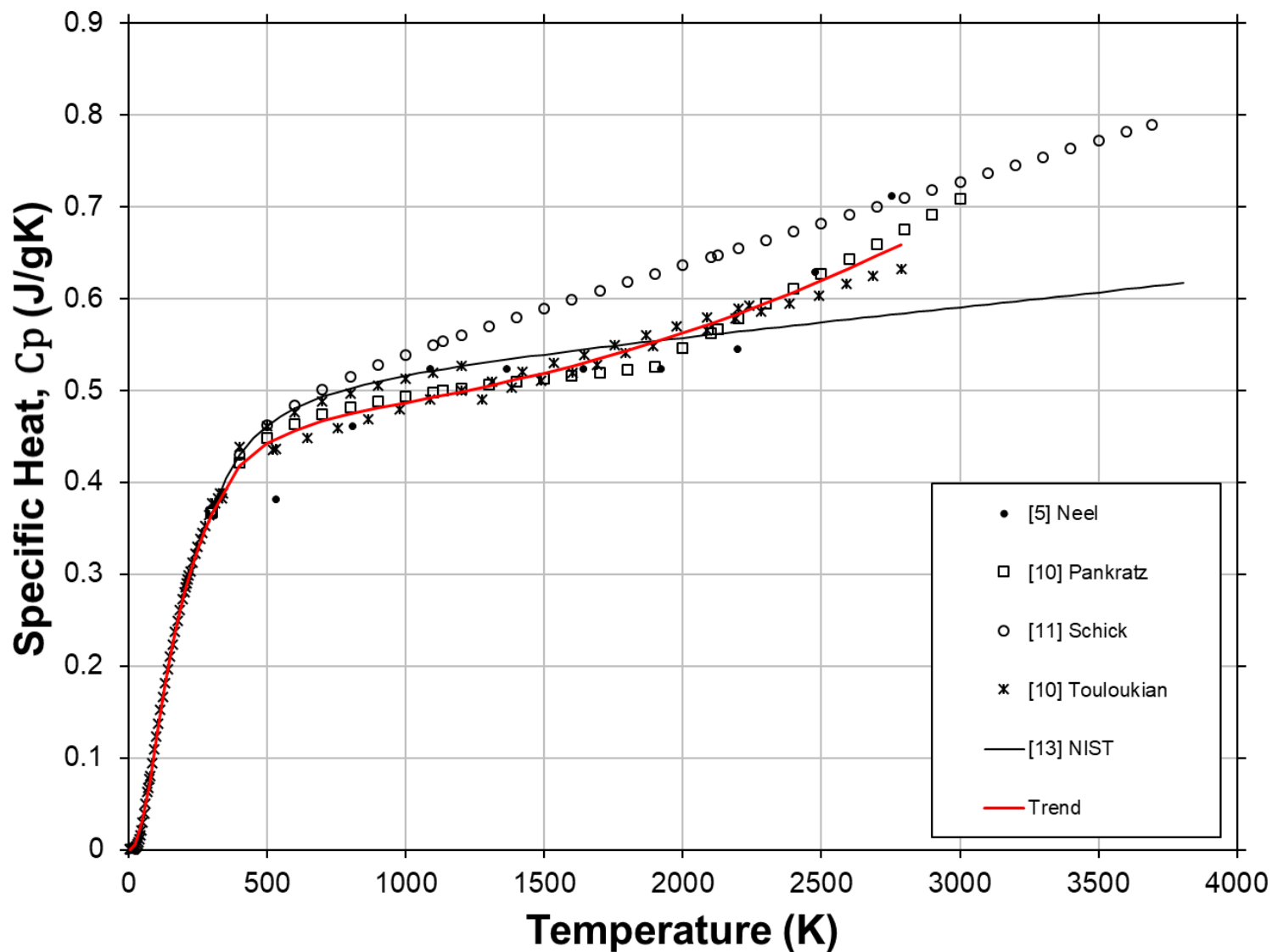


Figure 7.2.2-6: Specific Heat versus Temperature for ZrC. Displaying fit and data with comparison to NIST Webbook, and The Atomic Energy Review.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 100               | ( -279.7 ) | 0.122                            | ( 0.029 )       | 1400              | ( 2060.3 ) | 0.512                            | ( 0.122 )       |
| 200               | ( -99.7 )  | 0.278                            | ( 0.066 )       | 1500              | ( 2240.3 ) | 0.519                            | ( 0.124 )       |
| 300               | ( 80.3 )   | 0.364                            | ( 0.087 )       | 1600              | ( 2420.3 ) | 0.526                            | ( 0.126 )       |
| 400               | ( 260.3 )  | 0.417                            | ( 0.100 )       | 1700              | ( 2600.3 ) | 0.535                            | ( 0.128 )       |
| 500               | ( 440.3 )  | 0.442                            | ( 0.106 )       | 1800              | ( 2780.3 ) | 0.543                            | ( 0.130 )       |
| 600               | ( 620.3 )  | 0.457                            | ( 0.109 )       | 1900              | ( 2960.3 ) | 0.552                            | ( 0.132 )       |
| 700               | ( 800.3 )  | 0.466                            | ( 0.111 )       | 2000              | ( 3140.3 ) | 0.562                            | ( 0.134 )       |
| 800               | ( 980.3 )  | 0.474                            | ( 0.113 )       | 2100              | ( 3320.3 ) | 0.572                            | ( 0.137 )       |
| 900               | ( 1160.3 ) | 0.480                            | ( 0.115 )       | 2200              | ( 3500.3 ) | 0.583                            | ( 0.139 )       |
| 1000              | ( 1340.3 ) | 0.486                            | ( 0.116 )       | 2300              | ( 3680.3 ) | 0.595                            | ( 0.142 )       |
| 1100              | ( 1520.3 ) | 0.492                            | ( 0.118 )       | 2400              | ( 3860.3 ) | 0.607                            | ( 0.145 )       |
| 1200              | ( 1700.3 ) | 0.498                            | ( 0.119 )       | 2600              | ( 4220.3 ) | 0.632                            | ( 0.151 )       |
| 1300              | ( 1880.3 ) | 0.505                            | ( 0.121 )       | 2788              | ( 4558.7 ) | 0.658                            | ( 0.157 )       |

**Application Notes:** Data for specific heat is collected from references [5, 6, 10-12] and fitted with the equations below to approximate the property trend with respect to temperature. Fitted trend is compared to trends from references[13, 14].

## Fit Equation:

For temperature range:  $4 < T \leq 294$

$$C_p(T) = \left[ A_0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A_1 \cdot \left( \frac{T}{1000} \right) + A_2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $294 < T \leq 2788$

$$C_p(T) = B_0 + B_1 \cdot \left( \frac{T}{1000} \right) + B_2 \cdot \left( \frac{T}{1000} \right)^2 + B_{_2} / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

T. Range [K]:  $5 \leq T < 293$

N = 2.509

A0 = 52.73

A1 = -4.986

A2 = 83.5

$293 \leq T \leq 2788$

B0 = 0.4889

B1 = -0.02133

B2 = 0.02964

B<sub>2</sub> = -0.01088



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

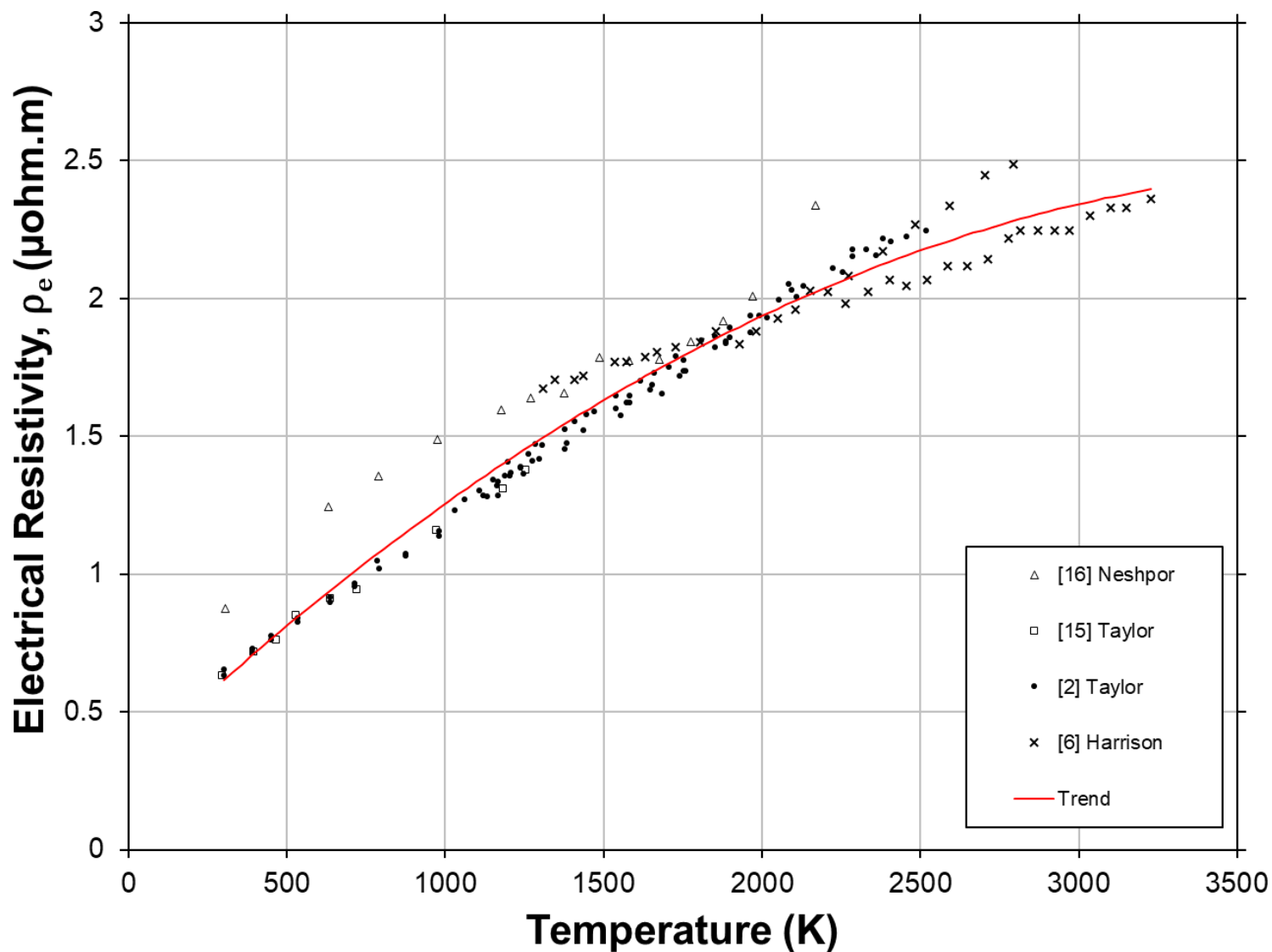


Figure 7.2.2-7: Electrical Resistivity versus Temperature for ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 290               | ( 62.3 )   | 0.605                               | ( 23.81 )               | 1800              | ( 2780.3 ) | 1.822                               | ( 71.74 )               |
| 400               | ( 260.3 )  | 0.715                               | ( 28.14 )               | 1900              | ( 2960.3 ) | 1.881                               | ( 74.04 )               |
| 500               | ( 440.3 )  | 0.812                               | ( 31.96 )               | 2000              | ( 3140.3 ) | 1.936                               | ( 76.23 )               |
| 600               | ( 620.3 )  | 0.906                               | ( 35.67 )               | 2100              | ( 3320.3 ) | 1.989                               | ( 78.32 )               |
| 700               | ( 800.3 )  | 0.997                               | ( 39.27 )               | 2200              | ( 3500.3 ) | 2.039                               | ( 80.29 )               |
| 800               | ( 980.3 )  | 1.086                               | ( 42.76 )               | 2300              | ( 3680.3 ) | 2.087                               | ( 82.16 )               |
| 900               | ( 1160.3 ) | 1.172                               | ( 46.15 )               | 2400              | ( 3860.3 ) | 2.132                               | ( 83.92 )               |
| 1000              | ( 1340.3 ) | 1.255                               | ( 49.43 )               | 2500              | ( 4040.3 ) | 2.173                               | ( 85.57 )               |
| 1100              | ( 1520.3 ) | 1.336                               | ( 52.59 )               | 2600              | ( 4220.3 ) | 2.213                               | ( 87.11 )               |
| 1200              | ( 1700.3 ) | 1.414                               | ( 55.65 )               | 2700              | ( 4400.3 ) | 2.249                               | ( 88.55 )               |
| 1300              | ( 1880.3 ) | 1.489                               | ( 58.61 )               | 2800              | ( 4580.3 ) | 2.283                               | ( 89.87 )               |
| 1400              | ( 2060.3 ) | 1.561                               | ( 61.45 )               | 2900              | ( 4760.3 ) | 2.314                               | ( 91.09 )               |
| 1500              | ( 2240.3 ) | 1.630                               | ( 64.18 )               | 3000              | ( 4940.3 ) | 2.342                               | ( 92.20 )               |
| 1600              | ( 2420.3 ) | 1.697                               | ( 66.81 )               | 3150              | ( 5210.3 ) | 2.379                               | ( 93.66 )               |
| 1700              | ( 2600.3 ) | 1.761                               | ( 69.33 )               | 3300              | ( 5480.3 ) | 2.410                               | ( 94.87 )               |

**Application Notes:** Data for electrical resistivity is collected from references [2, 6, 15, 16] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $290 \leq T < 3300$

A0 = 0.2991

A1 = 1.094

A2 = -0.1377

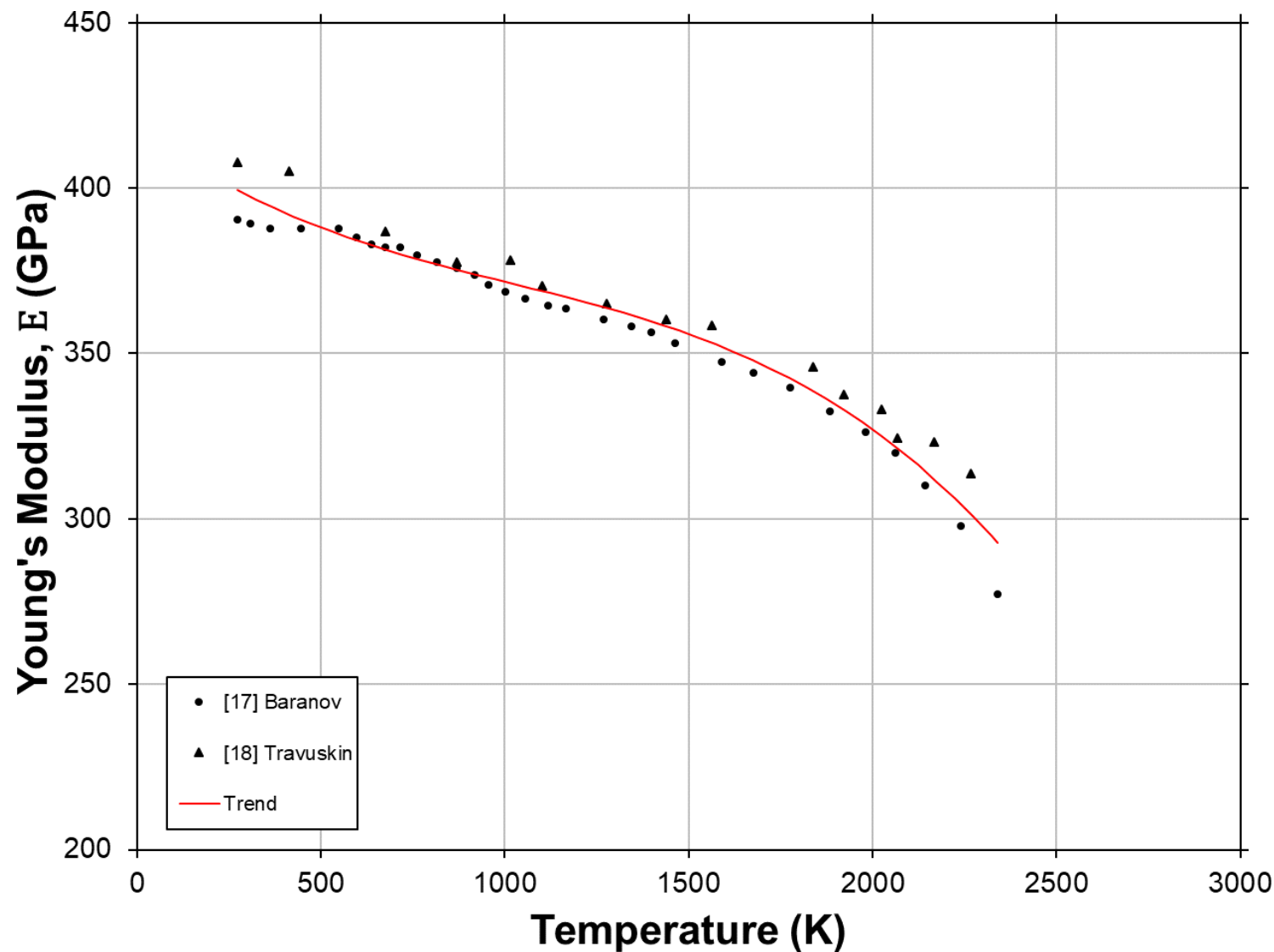


Figure 7.2.2-8: Young's Modulus versus Temperature for ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 273               | ( 31.7 )   | 399.37                | ( 57.95 ) | 1000              | ( 1340.3 ) | 371.66                | ( 53.93 ) |
| 300               | ( 80.3 )   | 397.83                | ( 57.72 ) | 1100              | ( 1520.3 ) | 368.90                | ( 53.53 ) |
| 350               | ( 170.3 )  | 395.11                | ( 57.33 ) | 1200              | ( 1700.3 ) | 366.06                | ( 53.12 ) |
| 400               | ( 260.3 )  | 392.58                | ( 56.96 ) | 1300              | ( 1880.3 ) | 363.03                | ( 52.68 ) |
| 450               | ( 350.3 )  | 390.22                | ( 56.62 ) | 1400              | ( 2060.3 ) | 359.68                | ( 52.19 ) |
| 500               | ( 440.3 )  | 388.02                | ( 56.30 ) | 1500              | ( 2240.3 ) | 355.92                | ( 51.64 ) |
| 550               | ( 530.3 )  | 385.96                | ( 56.00 ) | 1600              | ( 2420.3 ) | 351.64                | ( 51.02 ) |
| 600               | ( 620.3 )  | 384.03                | ( 55.72 ) | 1700              | ( 2600.3 ) | 346.71                | ( 50.31 ) |
| 650               | ( 710.3 )  | 382.22                | ( 55.46 ) | 1800              | ( 2780.3 ) | 341.04                | ( 49.48 ) |
| 700               | ( 800.3 )  | 380.52                | ( 55.21 ) | 1900              | ( 2960.3 ) | 334.50                | ( 48.54 ) |
| 750               | ( 890.3 )  | 378.90                | ( 54.98 ) | 2000              | ( 3140.3 ) | 327.00                | ( 47.45 ) |
| 800               | ( 980.3 )  | 377.35                | ( 54.75 ) | 2100              | ( 3320.3 ) | 318.42                | ( 46.20 ) |
| 850               | ( 1070.3 ) | 375.87                | ( 54.54 ) | 2200              | ( 3500.3 ) | 308.65                | ( 44.78 ) |
| 900               | ( 1160.3 ) | 374.44                | ( 54.33 ) | 2300              | ( 3680.3 ) | 297.57                | ( 43.18 ) |
| 950               | ( 1250.3 ) | 373.04                | ( 54.13 ) | 2340              | ( 3752.3 ) | 292.75                | ( 42.48 ) |

**Application Notes:** Data for Young's modulus is collected from references [17, 18] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equations:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Young's Modulus [GPa]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $273 \leq T < 2340$

A0 = 418.8

A1 = -85.2

A2 = 56.47

A3 = -18.41

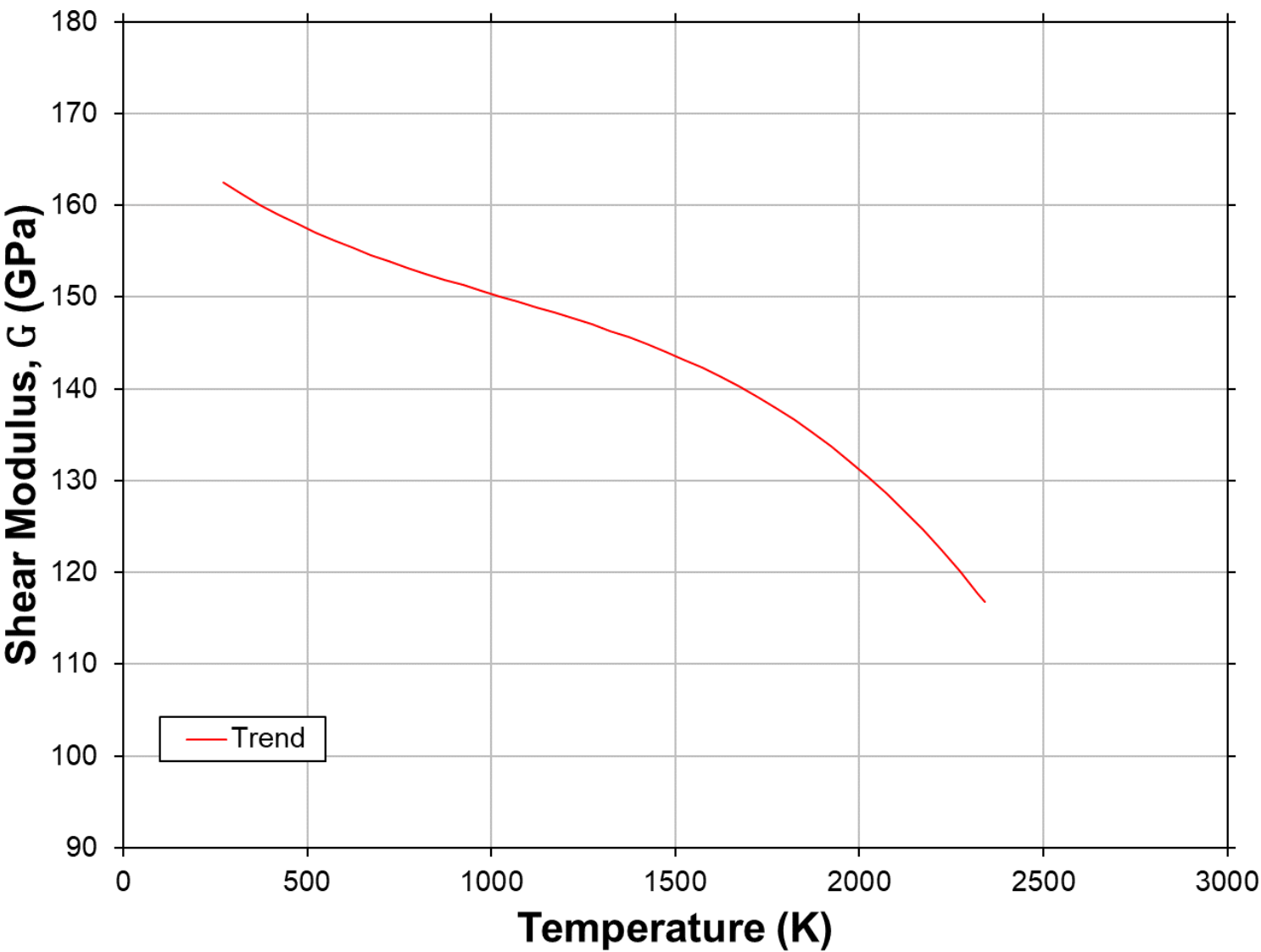


Figure 7.2.2-9: Shear Modulus versus Temperature for ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Shear Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Shear Modulus ( G ) |           | Temperature ( T ) |            | Shear Modulus ( G ) |           |
|-------------------|------------|---------------------|-----------|-------------------|------------|---------------------|-----------|
| K                 | ( °F )     | GPa                 | ( Msi )   | K                 | ( °F )     | GPa                 | ( Msi )   |
| 273               | ( 31.7 )   | 162.47              | ( 23.57 ) | 1000              | ( 1340.3 ) | 150.36              | ( 21.82 ) |
| 300               | ( 80.3 )   | 161.78              | ( 23.47 ) | 1100              | ( 1520.3 ) | 149.17              | ( 21.65 ) |
| 350               | ( 170.3 )  | 160.59              | ( 23.30 ) | 1200              | ( 1700.3 ) | 147.95              | ( 21.47 ) |
| 400               | ( 260.3 )  | 159.47              | ( 23.14 ) | 1300              | ( 1880.3 ) | 146.65              | ( 21.28 ) |
| 450               | ( 350.3 )  | 158.44              | ( 22.99 ) | 1400              | ( 2060.3 ) | 145.22              | ( 21.07 ) |
| 500               | ( 440.3 )  | 157.47              | ( 22.85 ) | 1500              | ( 2240.3 ) | 143.60              | ( 20.84 ) |
| 550               | ( 530.3 )  | 156.57              | ( 22.72 ) | 1600              | ( 2420.3 ) | 141.77              | ( 20.57 ) |
| 600               | ( 620.3 )  | 155.73              | ( 22.60 ) | 1700              | ( 2600.3 ) | 139.66              | ( 20.26 ) |
| 650               | ( 710.3 )  | 154.94              | ( 22.48 ) | 1800              | ( 2780.3 ) | 137.24              | ( 19.91 ) |
| 700               | ( 800.3 )  | 154.20              | ( 22.37 ) | 1900              | ( 2960.3 ) | 134.45              | ( 19.51 ) |
| 750               | ( 890.3 )  | 153.49              | ( 22.27 ) | 2000              | ( 3140.3 ) | 131.26              | ( 19.05 ) |
| 800               | ( 980.3 )  | 152.82              | ( 22.17 ) | 2100              | ( 3320.3 ) | 127.61              | ( 18.52 ) |
| 850               | ( 1070.3 ) | 152.18              | ( 22.08 ) | 2200              | ( 3500.3 ) | 123.47              | ( 17.92 ) |
| 900               | ( 1160.3 ) | 151.56              | ( 21.99 ) | 2300              | ( 3680.3 ) | 118.80              | ( 17.24 ) |
| 950               | ( 1250.3 ) | 150.96              | ( 21.90 ) | 2340              | ( 3752.3 ) | 116.77              | ( 16.94 ) |

**Application Notes:** Shear modulus is calculated as a function of Young's modulus and Poisson's Ratio using the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$G(T) = 0.5 \cdot E(T) / (1 + \nu(T))$$

$$G(T) = \text{Shear Modulus [GPa]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $273 \leq T \leq 2340$

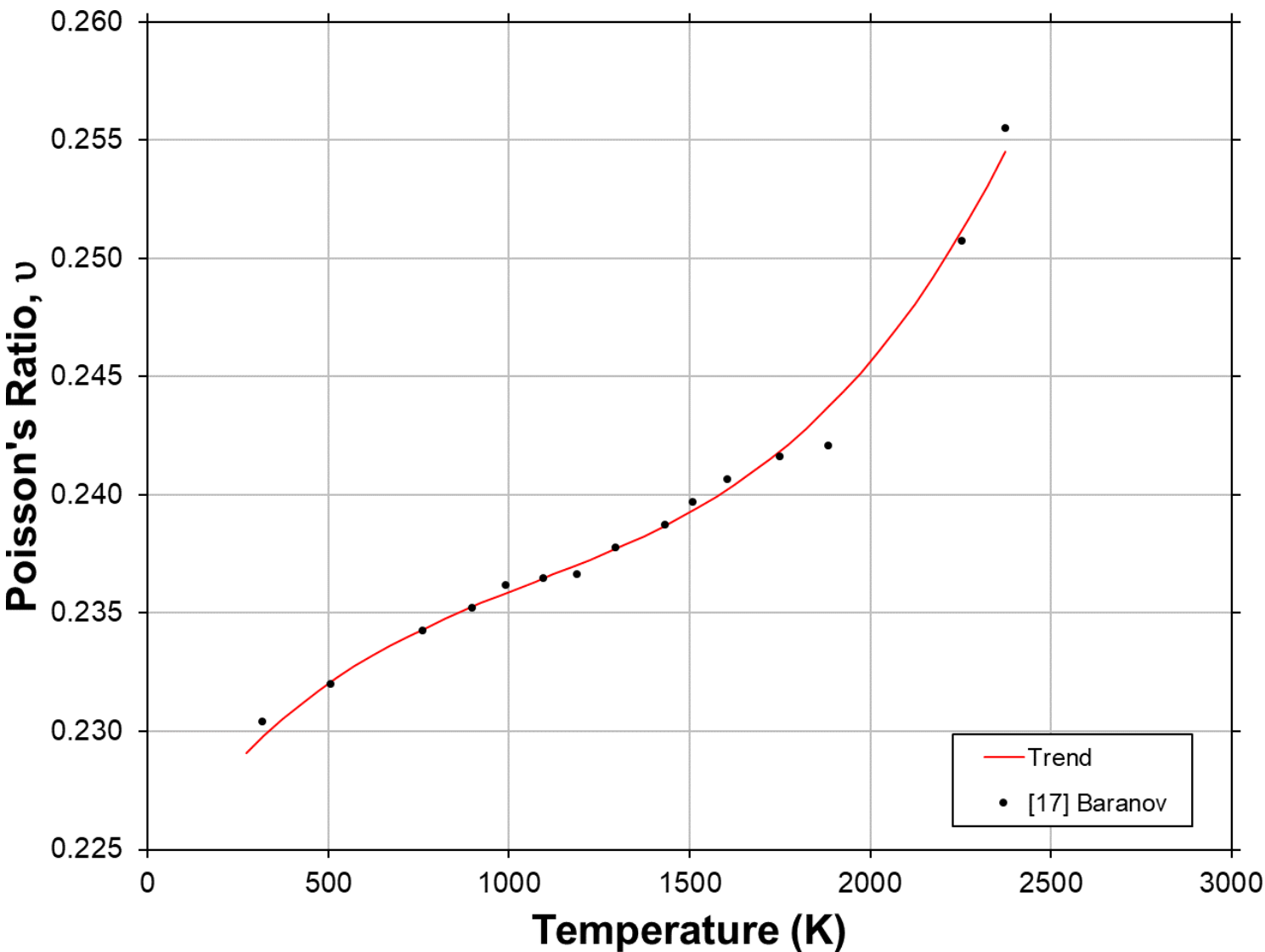


Figure 7.2.2-10: Poisson's Ratio versus Temperature for ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Poisson's Ratio with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) | Temperature ( T ) |            | Poisson's Ratio ( $\nu$ ) |
|-------------------|------------|---------------------------|-------------------|------------|---------------------------|
| K                 | ( °F )     |                           | K                 | ( °F )     |                           |
| 273               | ( 31.7 )   | 0.229                     | 1000              | ( 1340.3 ) | 0.236                     |
| 300               | ( 80.3 )   | 0.229                     | 1100              | ( 1520.3 ) | 0.236                     |
| 350               | ( 170.3 )  | 0.230                     | 1200              | ( 1700.3 ) | 0.237                     |
| 400               | ( 260.3 )  | 0.231                     | 1300              | ( 1880.3 ) | 0.238                     |
| 450               | ( 350.3 )  | 0.231                     | 1400              | ( 2060.3 ) | 0.238                     |
| 500               | ( 440.3 )  | 0.232                     | 1500              | ( 2240.3 ) | 0.239                     |
| 550               | ( 530.3 )  | 0.233                     | 1600              | ( 2420.3 ) | 0.240                     |
| 600               | ( 620.3 )  | 0.233                     | 1700              | ( 2600.3 ) | 0.241                     |
| 650               | ( 710.3 )  | 0.233                     | 1800              | ( 2780.3 ) | 0.242                     |
| 700               | ( 800.3 )  | 0.234                     | 1900              | ( 2960.3 ) | 0.244                     |
| 750               | ( 890.3 )  | 0.234                     | 2000              | ( 3140.3 ) | 0.246                     |
| 800               | ( 980.3 )  | 0.235                     | 2100              | ( 3320.3 ) | 0.248                     |
| 850               | ( 1070.3 ) | 0.235                     | 2200              | ( 3500.3 ) | 0.250                     |
| 900               | ( 1160.3 ) | 0.235                     | 2300              | ( 3680.3 ) | 0.252                     |
| 950               | ( 1250.3 ) | 0.236                     | 2375              | ( 3815.3 ) | 0.255                     |

**Application Notes:** Data for Poisson's Ratio is collected from reference [17] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equations:

$$\nu(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\nu(T)$  = Poisson's Ratio [-]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $273 \leq T < 2375$

A0 = 0.224

A1 = 0.02248

A2 = -0.01534

A3 = 0.004755



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Hardness with Temperature

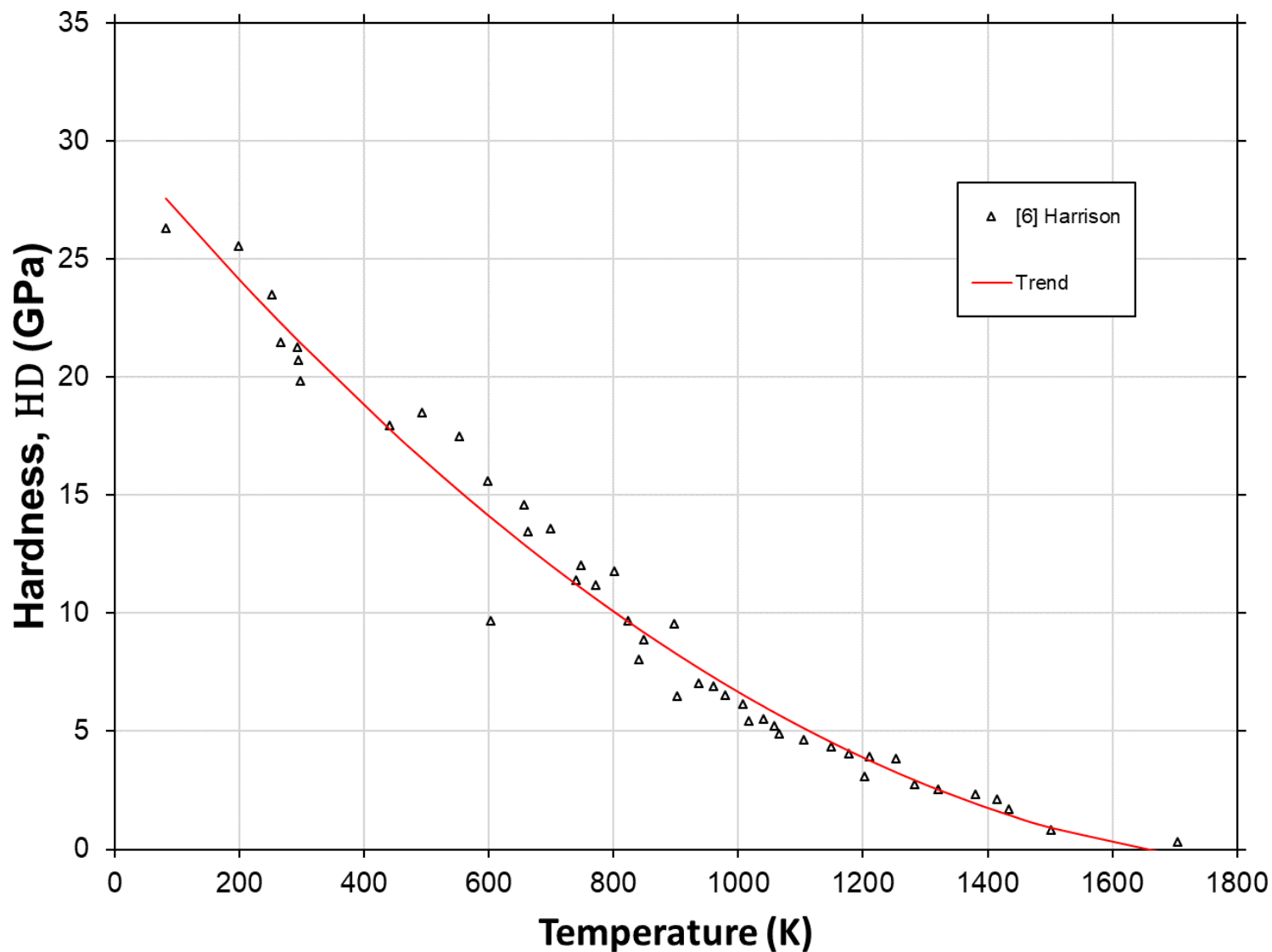


Figure 7.2.2-11: Hardness versus Temperature for ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Hardness with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Hardness (GPa ) | Temperature ( T ) |            | Hardness (GPa ) |
|-------------------|------------|-----------------|-------------------|------------|-----------------|
| K                 | ( °F )     |                 | K                 | ( °F )     |                 |
| 80                | ( -315.7 ) | 27.639          | 800               | ( 980.3 )  | 10.078          |
| 100               | ( -279.7 ) | 27.040          | 850               | ( 1070.3 ) | 9.165           |
| 150               | ( -189.7 ) | 25.569          | 900               | ( 1160.3 ) | 8.292           |
| 200               | ( -99.7 )  | 24.139          | 950               | ( 1250.3 ) | 7.458           |
| 250               | ( -9.7 )   | 22.748          | 1000              | ( 1340.3 ) | 6.665           |
| 300               | ( 80.3 )   | 21.397          | 1050              | ( 1430.3 ) | 5.911           |
| 350               | ( 170.3 )  | 20.086          | 1100              | ( 1520.3 ) | 5.198           |
| 400               | ( 260.3 )  | 18.814          | 1150              | ( 1610.3 ) | 4.524           |
| 450               | ( 350.3 )  | 17.583          | 1200              | ( 1700.3 ) | 3.890           |
| 500               | ( 440.3 )  | 16.391          | 1250              | ( 1790.3 ) | 3.295           |
| 550               | ( 530.3 )  | 15.239          | 1300              | ( 1880.3 ) | 2.741           |
| 600               | ( 620.3 )  | 14.127          | 1400              | ( 2060.3 ) | 1.751           |
| 650               | ( 710.3 )  | 13.055          | 1500              | ( 2240.3 ) | 0.921           |
| 700               | ( 800.3 )  | 12.023          | 1600              | ( 2420.3 ) | 0.250           |
| 750               | ( 890.3 )  | 11.030          |                   |            |                 |

**Application Notes:** Data for hardness is collected from reference [6] and fitted with the equation below to approximate the property trend with respect to temperature.

## Fit Equations:

$$HD(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$$HD(T) = \text{Hardness [GPa]}$$

$$T = \text{Temperature [K]}$$

## Constants:

$$\text{T. Range [K]: } 80 \leq T < 1600$$

$$A0 = 30.1$$

$$A1 = -31.4$$

$$A2 = 7.965$$



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 6745                                  | -                                                       | 24.64                                       | -0.08                                              | 4.03                                      | 0.122                                          | -                                  | -                                | -                           |
| 200                 | 6740                                  | -                                                       | 25.51                                       | -0.05                                              | 4.69                                      | 0.278                                          | -                                  | -                                | -                           |
| 300                 | 6729                                  | 0.615                                                   | 26.37                                       | 0.00                                               | 5.10                                      | 0.364                                          | 397.8                              | 161.8                            | 0.229                       |
| 400                 | 6718                                  | 0.715                                                   | 27.21                                       | 0.06                                               | 5.41                                      | 0.417                                          | 392.6                              | 159.5                            | 0.231                       |
| 500                 | 6706                                  | 0.812                                                   | 28.03                                       | 0.12                                               | 5.67                                      | 0.442                                          | 388.0                              | 157.5                            | 0.232                       |
| 600                 | 6694                                  | 0.906                                                   | 28.85                                       | 0.18                                               | 5.89                                      | 0.457                                          | 384.0                              | 155.7                            | 0.233                       |
| 700                 | 6680                                  | 0.997                                                   | 29.65                                       | 0.25                                               | 6.09                                      | 0.466                                          | 380.5                              | 154.2                            | 0.234                       |
| 800                 | 6666                                  | 1.086                                                   | 30.43                                       | 0.32                                               | 6.28                                      | 0.474                                          | 377.4                              | 152.8                            | 0.235                       |
| 900                 | 6652                                  | 1.172                                                   | 31.20                                       | 0.39                                               | 6.44                                      | 0.480                                          | 374.4                              | 151.6                            | 0.235                       |
| 1000                | 6637                                  | 1.255                                                   | 31.96                                       | 0.46                                               | 6.60                                      | 0.486                                          | 371.7                              | 150.4                            | 0.236                       |
| 1100                | 6622                                  | 1.336                                                   | 32.70                                       | 0.54                                               | 6.75                                      | 0.492                                          | 368.9                              | 149.2                            | 0.236                       |
| 1200                | 6606                                  | 1.414                                                   | 33.43                                       | 0.62                                               | 6.89                                      | 0.498                                          | 366.1                              | 148.0                            | 0.237                       |
| 1300                | 6590                                  | 1.489                                                   | 34.14                                       | 0.71                                               | 7.02                                      | 0.505                                          | 363.0                              | 146.6                            | 0.238                       |
| 1400                | 6573                                  | 1.561                                                   | 34.85                                       | 0.79                                               | 7.14                                      | 0.512                                          | 359.7                              | 145.2                            | 0.238                       |
| 1500                | 6556                                  | 1.630                                                   | 35.53                                       | 0.88                                               | 7.26                                      | 0.519                                          | 355.9                              | 143.6                            | 0.239                       |
| 1600                | 6539                                  | 1.697                                                   | 36.20                                       | 0.96                                               | 7.37                                      | 0.526                                          | 351.6                              | 141.8                            | 0.240                       |
| 1700                | 6522                                  | 1.761                                                   | 36.86                                       | 1.05                                               | 7.48                                      | 0.535                                          | 346.7                              | 139.7                            | 0.241                       |
| 1800                | 6504                                  | 1.822                                                   | 37.51                                       | 1.14                                               | 7.58                                      | 0.543                                          | 341.0                              | 137.2                            | 0.242                       |
| 1900                | 6487                                  | 1.881                                                   | 38.14                                       | 1.23                                               | 7.67                                      | 0.552                                          | 334.5                              | 134.5                            | 0.244                       |
| 2000                | 6469                                  | 1.936                                                   | 38.75                                       | 1.33                                               | 7.76                                      | 0.562                                          | 327.0                              | 131.3                            | 0.246                       |
| 2200                | 6434                                  | 2.039                                                   | 39.95                                       | 1.51                                               | 7.91                                      | 0.583                                          | 308.6                              | 123.5                            | 0.250                       |
| 2400                | 6398                                  | 2.132                                                   | 41.08                                       | 1.70                                               | 8.05                                      | 0.607                                          | -                                  | -                                | -                           |
| 2600                | 6363                                  | 2.213                                                   | 42.16                                       | 1.89                                               | 8.17                                      | 0.632                                          | -                                  | -                                | -                           |
| 2800                | 6329                                  | 2.283                                                   | -                                           | 2.07                                               | 8.26                                      | -                                              | -                                  | -                                | -                           |
| 3000                | 6296                                  | 2.342                                                   | -                                           | 2.25                                               | 8.34                                      | -                                              | -                                  | -                                | -                           |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

**Revision 0: 08-05-2020**

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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.2 Carbides

7.2.2 Zirconium Carbide (ZrC)

**Revision 0: 08-05-2020**

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## **7 Refractory Ceramics**

### **7.3 Nitrides**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 2.1: 08-25-2023

General

## Room Temperature Properties

|                                           |        |
|-------------------------------------------|--------|
| Molar Mass, [g/mol]                       | 105.23 |
| Theoretical Density, [kg/m <sup>3</sup> ] | 7,350  |
| Melting Point, [K]                        | 3253   |
| Specific Heat, [J/(g-K)]                  | 0.388  |
| Thermal Conductivity, [W/(m-K)]           | 6.2    |
| Linear expansion coefficient, [μm/(m-K)]  | 6.62   |
| Electrical resistivity, [μΩ-m]            | 0.22   |
| Young's Modulus, [GPa]                    | 366.8  |
| Shear Modulus, [GPa]                      | 149.9  |
| Poisson's Ratio, [-]                      | 0.223  |

## Zirconium – Nitrogen Phase Diagram

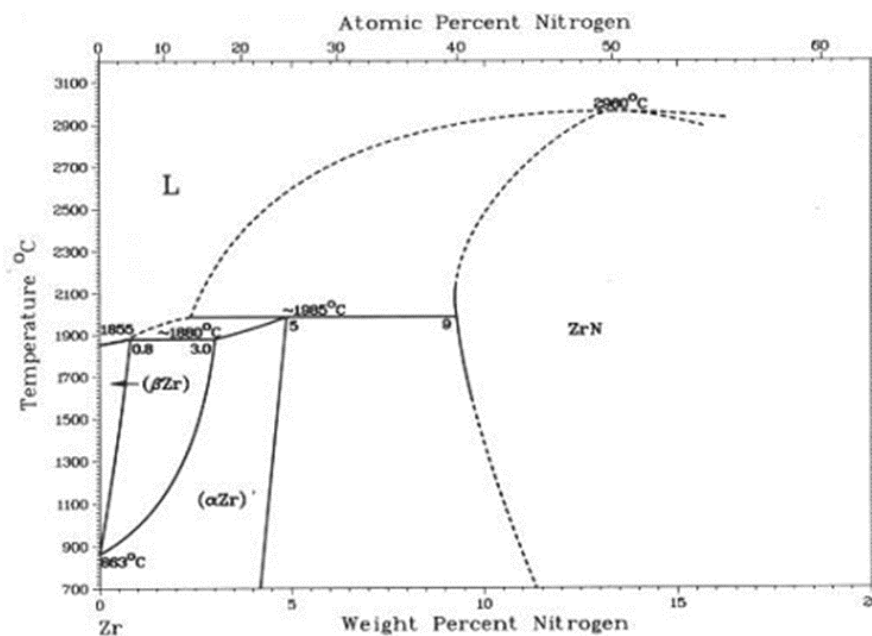


Figure 7.3.1-1: Zirconium – Nitrogen Phase Diagram [1].

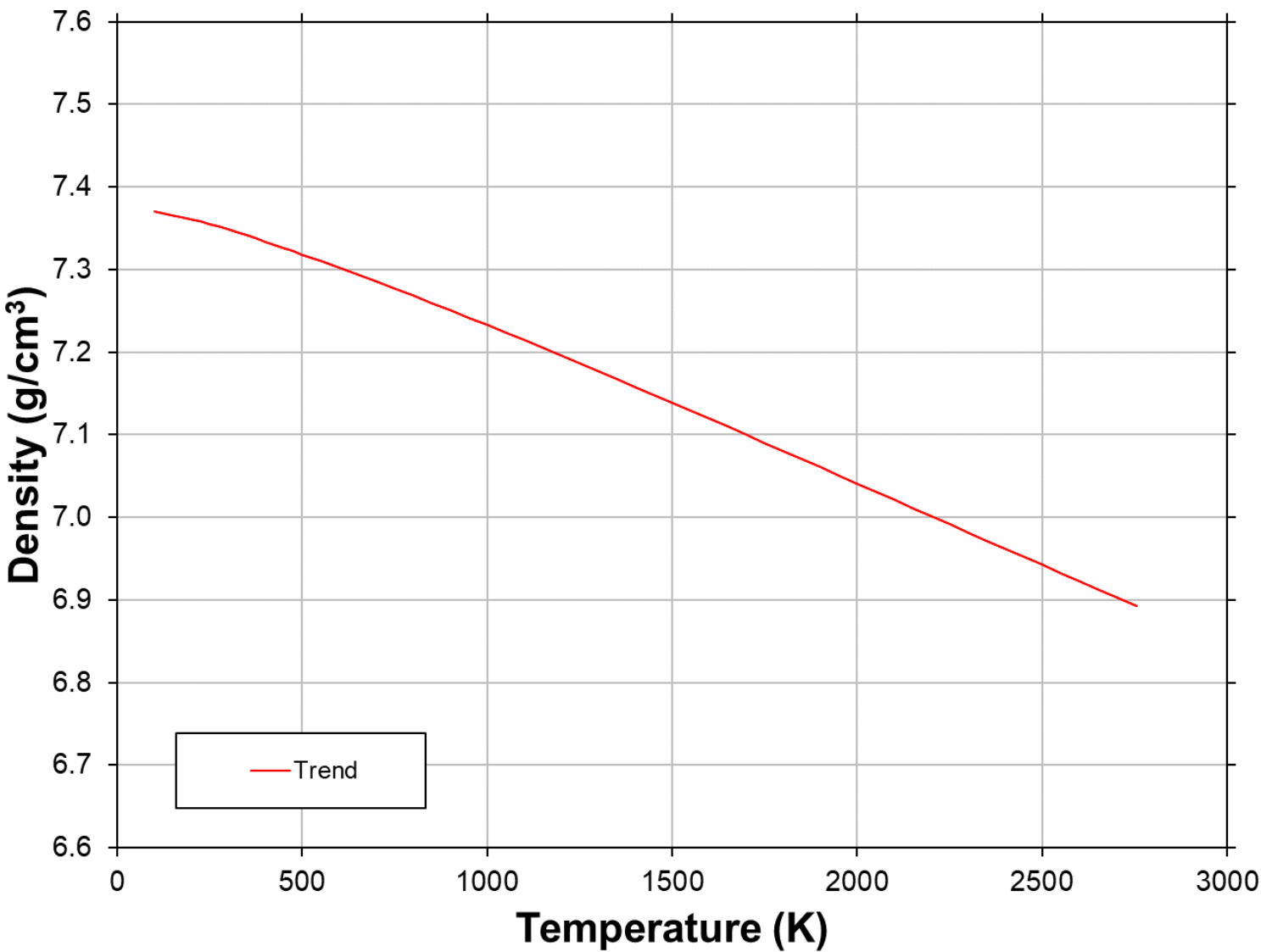


Figure 7.3.1-2: Density versus Temperature for ZrN. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 100               | ( -279.7 ) | 7370              | ( 460.1 )              | 1400              | ( 2060.3 ) | 7158              | ( 446.9 )              |
| 150               | ( -189.7 ) | 7366              | ( 459.9 )              | 1500              | ( 2240.3 ) | 7139              | ( 445.7 )              |
| 200               | ( -99.7 )  | 7361              | ( 459.5 )              | 1600              | ( 2420.3 ) | 7120              | ( 444.5 )              |
| 250               | ( -9.7 )   | 7355              | ( 459.2 )              | 1700              | ( 2600.3 ) | 7100              | ( 443.3 )              |
| 300               | ( 80.3 )   | 7349              | ( 458.8 )              | 1800              | ( 2780.3 ) | 7080              | ( 442.0 )              |
| 400               | ( 260.3 )  | 7334              | ( 457.9 )              | 1900              | ( 2960.3 ) | 7061              | ( 440.8 )              |
| 500               | ( 440.3 )  | 7318              | ( 456.9 )              | 2000              | ( 3140.3 ) | 7041              | ( 439.6 )              |
| 600               | ( 620.3 )  | 7302              | ( 455.9 )              | 2100              | ( 3320.3 ) | 7021              | ( 438.3 )              |
| 700               | ( 800.3 )  | 7285              | ( 454.8 )              | 2200              | ( 3500.3 ) | 7001              | ( 437.1 )              |
| 800               | ( 980.3 )  | 7268              | ( 453.8 )              | 2300              | ( 3680.3 ) | 6982              | ( 435.9 )              |
| 900               | ( 1160.3 ) | 7251              | ( 452.7 )              | 2400              | ( 3860.3 ) | 6962              | ( 434.6 )              |
| 1000              | ( 1340.3 ) | 7233              | ( 451.5 )              | 2500              | ( 4040.3 ) | 6942              | ( 433.4 )              |
| 1100              | ( 1520.3 ) | 7215              | ( 450.4 )              | 2600              | ( 4220.3 ) | 6923              | ( 432.2 )              |
| 1200              | ( 1700.3 ) | 7196              | ( 449.3 )              | 2700              | ( 4400.3 ) | 6904              | ( 431.0 )              |
| 1300              | ( 1880.3 ) | 7177              | ( 448.1 )              | 2755              | ( 4499.3 ) | 6893              | ( 430.3 )              |

**Application Notes:** Density is calculated as a function of thermal expansion as seen in the equation below to approximate property trend with respect to temperature.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 7,350 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $100 \leq T \leq 2755$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

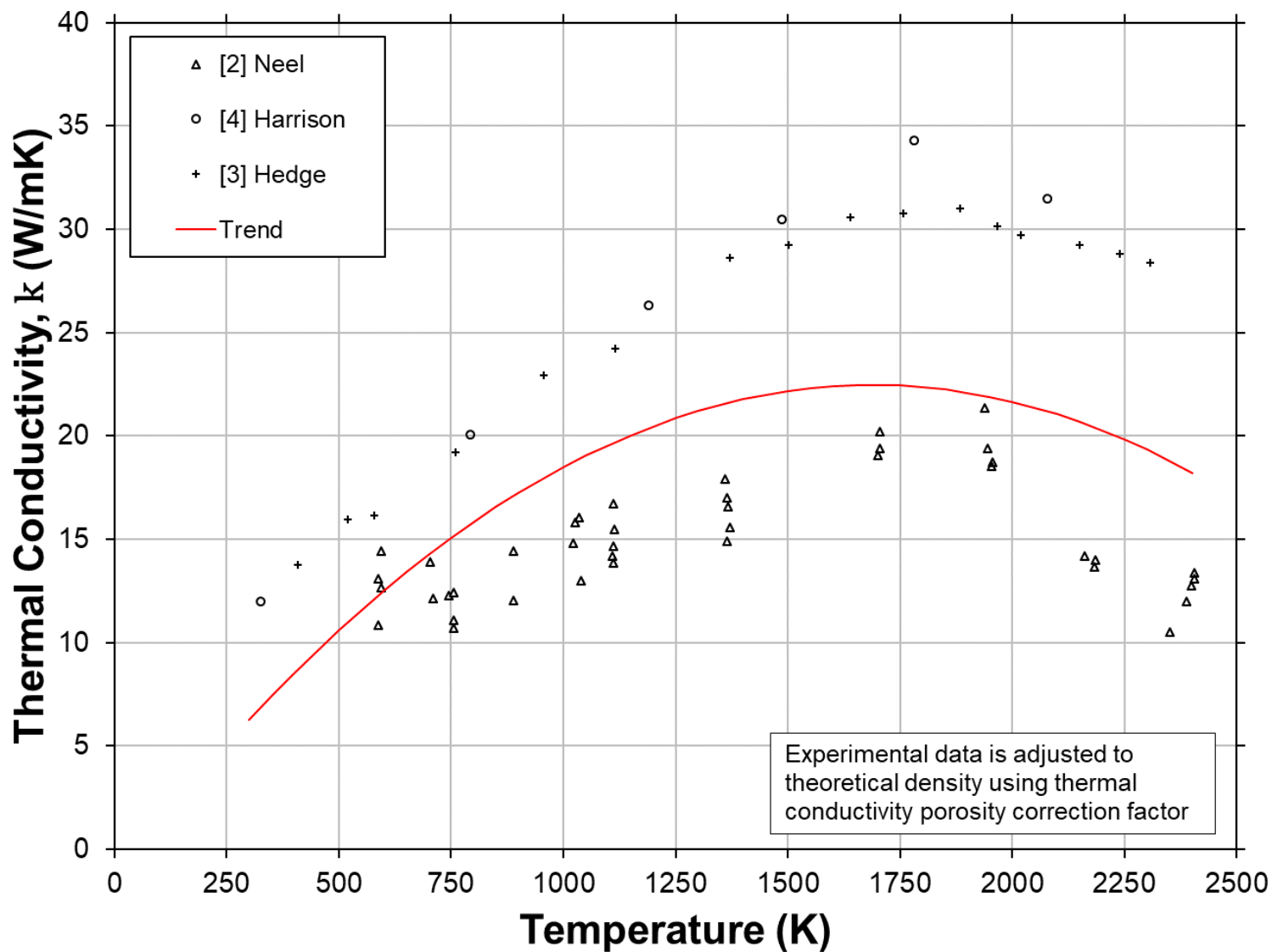


Figure 7.3.1-3: Thermal Conductivity versus Temperature of ZrN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 293               | ( 67.7 )   | 6.10                       | ( 42.34 )                           | 1000              | ( 1340.3 ) | 18.49                      | ( 128.28 )                          |
| 300               | ( 80.3 )   | 6.27                       | ( 43.48 )                           | 1100              | ( 1520.3 ) | 19.56                      | ( 135.73 )                          |
| 350               | ( 170.3 )  | 7.41                       | ( 51.43 )                           | 1200              | ( 1700.3 ) | 20.47                      | ( 142.01 )                          |
| 400               | ( 260.3 )  | 8.52                       | ( 59.09 )                           | 1300              | ( 1880.3 ) | 21.21                      | ( 147.12 )                          |
| 450               | ( 350.3 )  | 9.58                       | ( 66.46 )                           | 1400              | ( 2060.3 ) | 21.77                      | ( 151.07 )                          |
| 500               | ( 440.3 )  | 10.60                      | ( 73.54 )                           | 1500              | ( 2240.3 ) | 22.18                      | ( 153.85 )                          |
| 550               | ( 530.3 )  | 11.58                      | ( 80.33 )                           | 1600              | ( 2420.3 ) | 22.41                      | ( 155.47 )                          |
| 600               | ( 620.3 )  | 12.51                      | ( 86.82 )                           | 1700              | ( 2600.3 ) | 22.47                      | ( 155.92 )                          |
| 650               | ( 710.3 )  | 13.41                      | ( 93.02 )                           | 1800              | ( 2780.3 ) | 22.37                      | ( 155.20 )                          |
| 700               | ( 800.3 )  | 14.26                      | ( 98.93 )                           | 1900              | ( 2960.3 ) | 22.10                      | ( 153.32 )                          |
| 750               | ( 890.3 )  | 15.07                      | ( 104.55 )                          | 2000              | ( 3140.3 ) | 21.66                      | ( 150.27 )                          |
| 800               | ( 980.3 )  | 15.84                      | ( 109.88 )                          | 2100              | ( 3320.3 ) | 21.05                      | ( 146.05 )                          |
| 850               | ( 1070.3 ) | 16.56                      | ( 114.92 )                          | 2200              | ( 3500.3 ) | 20.27                      | ( 140.66 )                          |
| 900               | ( 1160.3 ) | 17.25                      | ( 119.66 )                          | 2300              | ( 3680.3 ) | 19.33                      | ( 134.11 )                          |
| 950               | ( 1250.3 ) | 17.89                      | ( 124.12 )                          | 2400              | ( 3860.3 ) | 18.22                      | ( 126.40 )                          |

**Application Notes:** Data for thermal conductivity is collected from references [2-4] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]: 293 ≤ T < 2400

A0 = -1.494

A1 = 28.39

A2 = -8.407



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

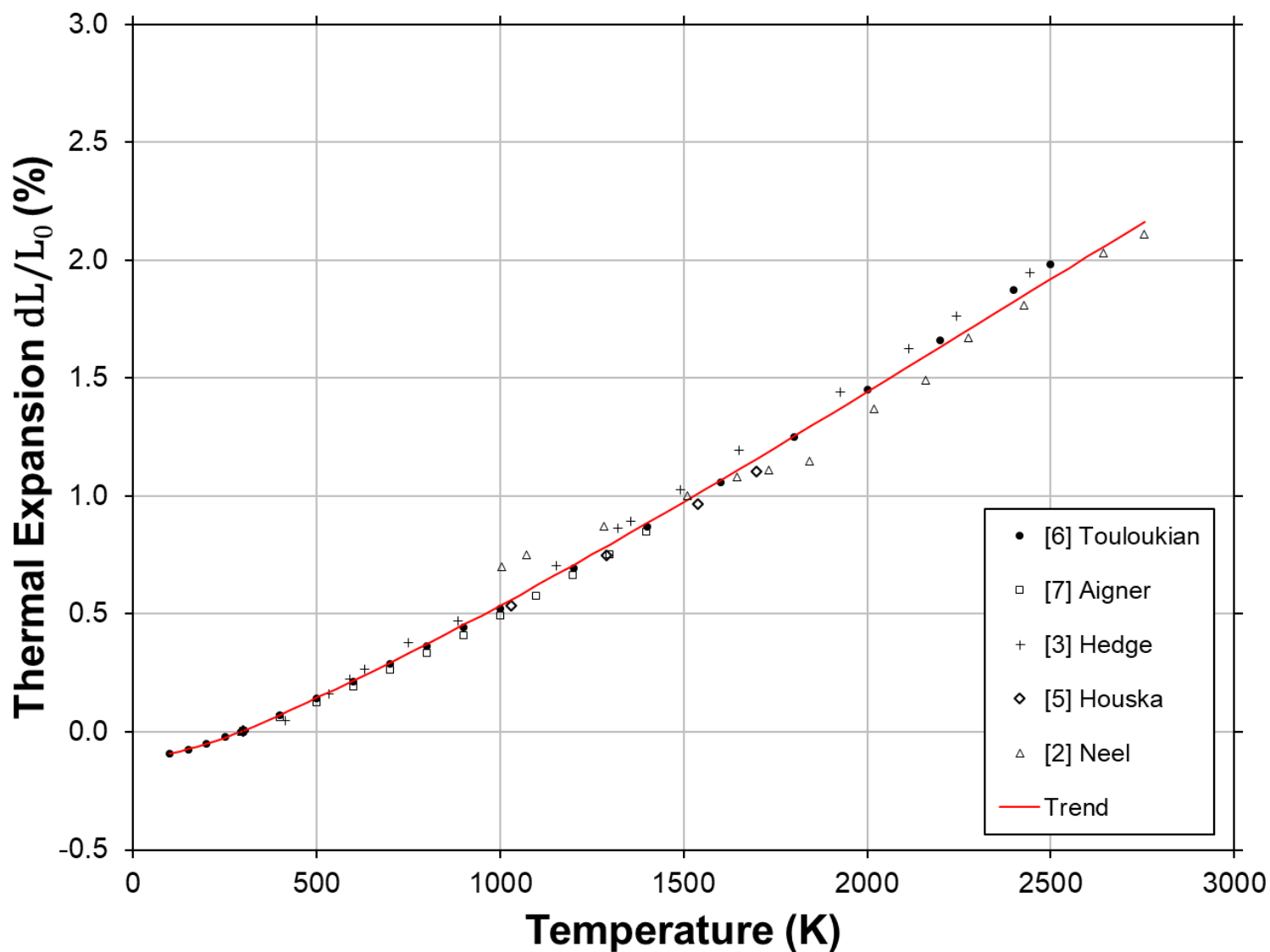


Figure 7.3.1-4: Thermal Expansion versus Temperature of ZrN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 100               | ( -279.7 ) | -0.093                                  | 1400              | ( 2060.3 ) | 0.885                                   |
| 150               | ( -189.7 ) | -0.072                                  | 1500              | ( 2240.3 ) | 0.975                                   |
| 200               | ( -99.7 )  | -0.049                                  | 1600              | ( 2420.3 ) | 1.067                                   |
| 250               | ( -9.7 )   | -0.024                                  | 1700              | ( 2600.3 ) | 1.160                                   |
| 300               | ( 80.3 )   | 0.004                                   | 1800              | ( 2780.3 ) | 1.253                                   |
| 400               | ( 260.3 )  | 0.073                                   | 1900              | ( 2960.3 ) | 1.347                                   |
| 500               | ( 440.3 )  | 0.144                                   | 2000              | ( 3140.3 ) | 1.442                                   |
| 600               | ( 620.3 )  | 0.218                                   | 2100              | ( 3320.3 ) | 1.537                                   |
| 700               | ( 800.3 )  | 0.295                                   | 2200              | ( 3500.3 ) | 1.633                                   |
| 800               | ( 980.3 )  | 0.373                                   | 2300              | ( 3680.3 ) | 1.729                                   |
| 900               | ( 1160.3 ) | 0.454                                   | 2400              | ( 3860.3 ) | 1.824                                   |
| 1000              | ( 1340.3 ) | 0.537                                   | 2500              | ( 4040.3 ) | 1.920                                   |
| 1100              | ( 1520.3 ) | 0.621                                   | 2600              | ( 4220.3 ) | 2.015                                   |
| 1200              | ( 1700.3 ) | 0.708                                   | 2700              | ( 4400.3 ) | 2.110                                   |
| 1300              | ( 1880.3 ) | 0.796                                   | 2755              | ( 4499.3 ) | 2.162                                   |

**Application Notes:** Data for thermal expansion is collected from references [2, 3, 5-7] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

| T Range [K]: | $100 \leq T < 293$ | $293 \leq T \leq 2755$ |
|--------------|--------------------|------------------------|
| A0 =         | -0.1259            | -0.1847                |
| A1 =         | 0.2824             | 0.5826                 |
| A2 =         | 0.4987             | 0.1624                 |
| A3 =         | 0                  | -0.02349               |

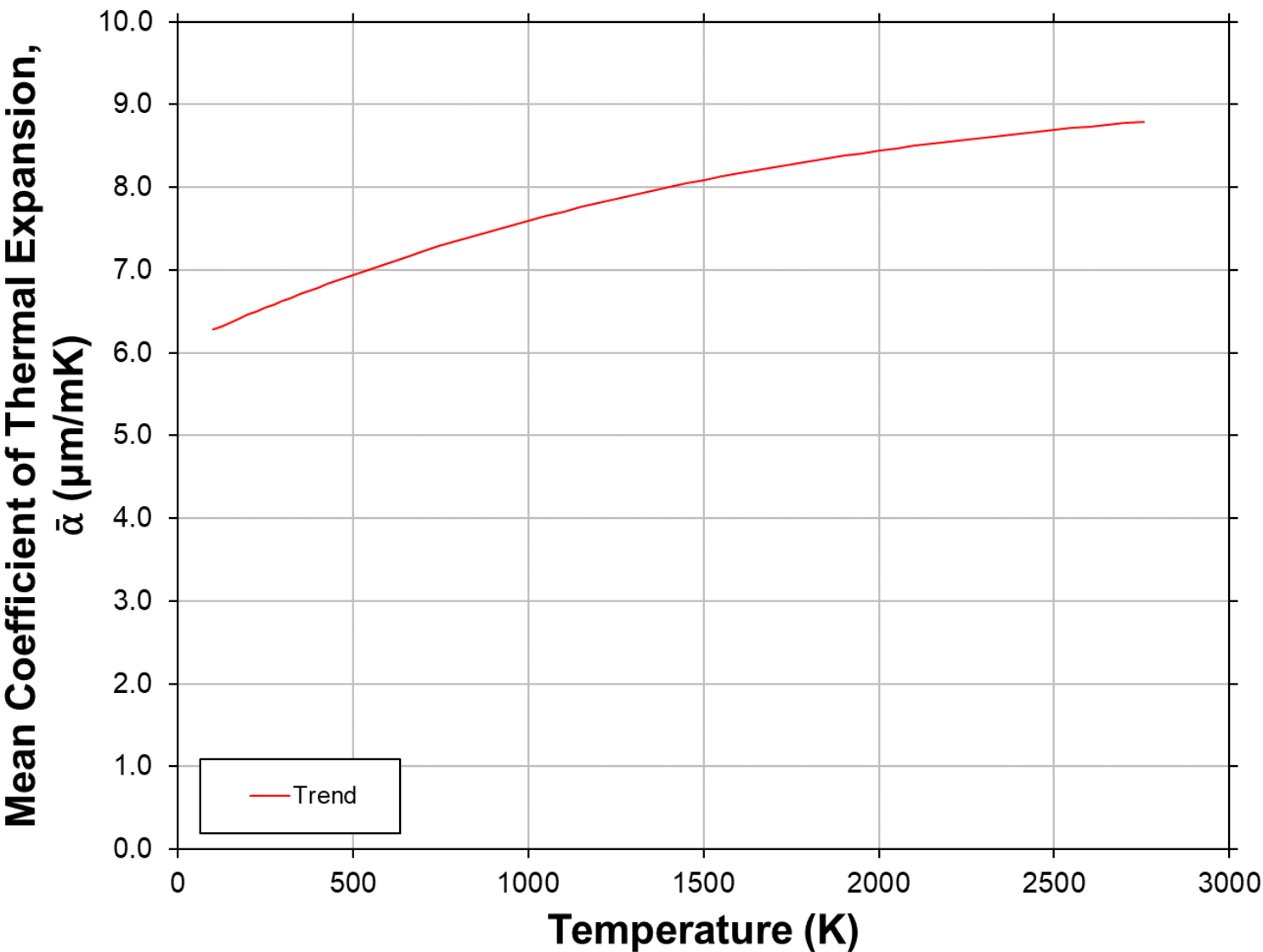


Figure 7.3.1-5: Mean Coefficient of Thermal Expansion versus Temperature of ZrN. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 100               | ( -279.7 ) | 6.277                                 | ( 3.487 )                                           | 1400              | ( 2060.3 ) | 8.001                                 | ( 4.445 )                                           |
| 150               | ( -189.7 ) | 6.368                                 | ( 3.538 )                                           | 1500              | ( 2240.3 ) | 8.087                                 | ( 4.493 )                                           |
| 200               | ( -99.7 )  | 6.456                                 | ( 3.587 )                                           | 1600              | ( 2420.3 ) | 8.168                                 | ( 4.538 )                                           |
| 250               | ( -9.7 )   | 6.542                                 | ( 3.635 )                                           | 1700              | ( 2600.3 ) | 8.243                                 | ( 4.580 )                                           |
| 300               | ( 80.3 )   | 6.627                                 | ( 3.681 )                                           | 1800              | ( 2780.3 ) | 8.314                                 | ( 4.619 )                                           |
| 400               | ( 260.3 )  | 6.789                                 | ( 3.771 )                                           | 1900              | ( 2960.3 ) | 8.380                                 | ( 4.656 )                                           |
| 500               | ( 440.3 )  | 6.942                                 | ( 3.857 )                                           | 2000              | ( 3140.3 ) | 8.442                                 | ( 4.690 )                                           |
| 600               | ( 620.3 )  | 7.088                                 | ( 3.938 )                                           | 2100              | ( 3320.3 ) | 8.500                                 | ( 4.722 )                                           |
| 700               | ( 800.3 )  | 7.227                                 | ( 4.015 )                                           | 2200              | ( 3500.3 ) | 8.553                                 | ( 4.752 )                                           |
| 800               | ( 980.3 )  | 7.357                                 | ( 4.087 )                                           | 2300              | ( 3680.3 ) | 8.603                                 | ( 4.780 )                                           |
| 900               | ( 1160.3 ) | 7.481                                 | ( 4.156 )                                           | 2400              | ( 3860.3 ) | 8.650                                 | ( 4.806 )                                           |
| 1000              | ( 1340.3 ) | 7.598                                 | ( 4.221 )                                           | 2500              | ( 4040.3 ) | 8.694                                 | ( 4.830 )                                           |
| 1100              | ( 1520.3 ) | 7.708                                 | ( 4.282 )                                           | 2600              | ( 4220.3 ) | 8.735                                 | ( 4.853 )                                           |
| 1200              | ( 1700.3 ) | 7.812                                 | ( 4.340 )                                           | 2700              | ( 4400.3 ) | 8.773                                 | ( 4.874 )                                           |
| 1300              | ( 1880.3 ) | 7.909                                 | ( 4.394 )                                           | 2755              | ( 4499.3 ) | 8.793                                 | ( 4.885 )                                           |

**Application Notes:** Data for mean coefficient of thermal expansion is calculated as a function of thermal expansion. Calculated data is then fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $100 \leq T \leq 2755$

A0 = 6.089

A1 = 1.926

A2 = -0.4595

A3 = 0.04236



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Specific Heat with Temperature

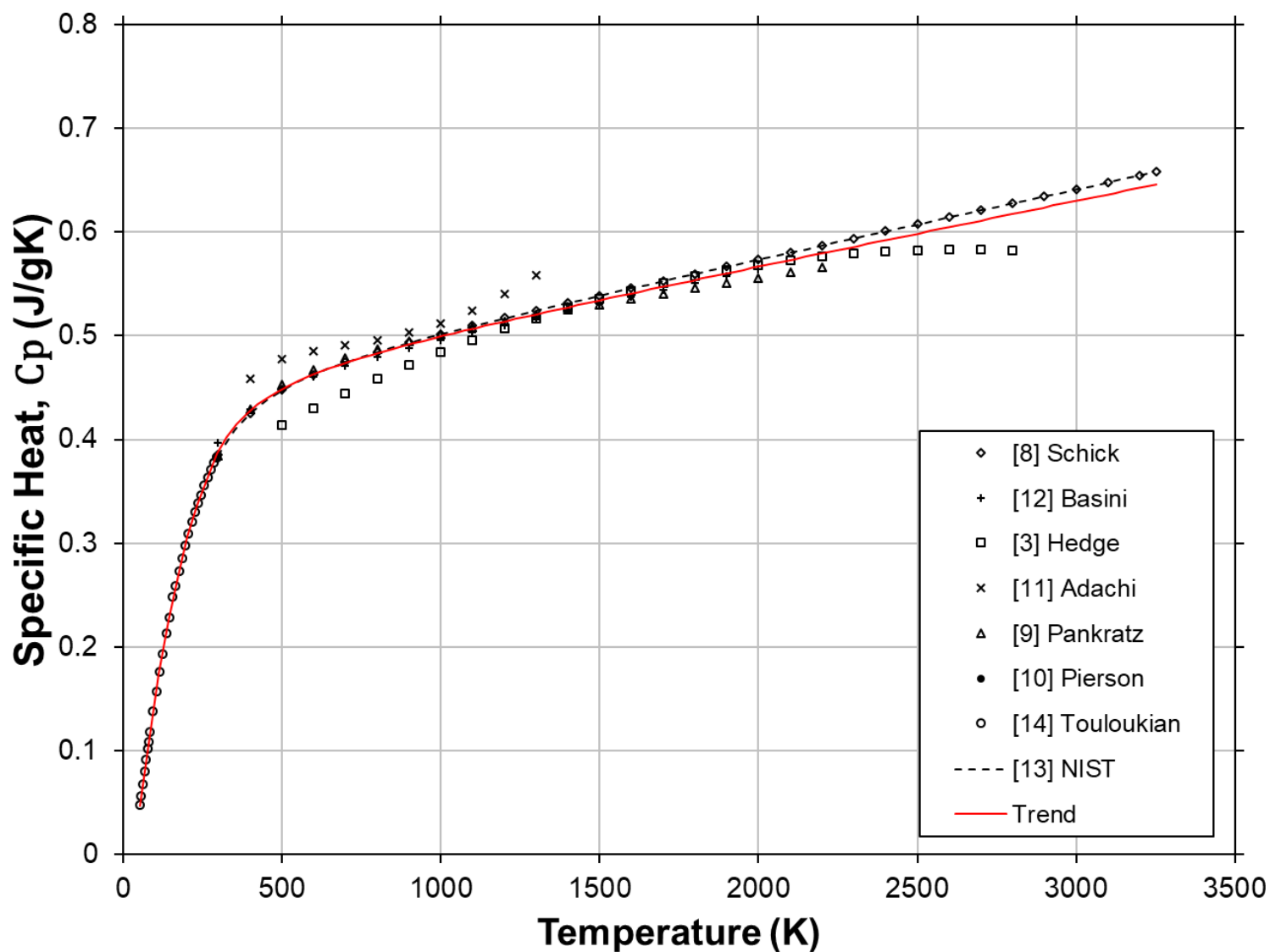


Figure 7.3.1-6: Specific Heat versus Temperature of ZrN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 53                | ( -364.3 ) | 0.047                            | ( 0.011 )       | 1300              | ( 1880.3 ) | 0.521                            | ( 0.124 )       |
| 100               | ( -279.7 ) | 0.149                            | ( 0.036 )       | 1400              | ( 2060.3 ) | 0.527                            | ( 0.126 )       |
| 200               | ( -99.7 )  | 0.302                            | ( 0.072 )       | 1500              | ( 2240.3 ) | 0.534                            | ( 0.128 )       |
| 300               | ( 80.3 )   | 0.390                            | ( 0.093 )       | 1600              | ( 2420.3 ) | 0.541                            | ( 0.129 )       |
| 400               | ( 260.3 )  | 0.427                            | ( 0.102 )       | 1800              | ( 2780.3 ) | 0.554                            | ( 0.132 )       |
| 500               | ( 440.3 )  | 0.448                            | ( 0.107 )       | 2000              | ( 3140.3 ) | 0.567                            | ( 0.135 )       |
| 600               | ( 620.3 )  | 0.463                            | ( 0.111 )       | 2200              | ( 3500.3 ) | 0.579                            | ( 0.138 )       |
| 700               | ( 800.3 )  | 0.474                            | ( 0.113 )       | 2400              | ( 3860.3 ) | 0.592                            | ( 0.141 )       |
| 800               | ( 980.3 )  | 0.483                            | ( 0.115 )       | 2600              | ( 4220.3 ) | 0.605                            | ( 0.145 )       |
| 900               | ( 1160.3 ) | 0.491                            | ( 0.117 )       | 2800              | ( 4580.3 ) | 0.617                            | ( 0.148 )       |
| 1000              | ( 1340.3 ) | 0.499                            | ( 0.119 )       | 3000              | ( 4940.3 ) | 0.630                            | ( 0.151 )       |
| 1100              | ( 1520.3 ) | 0.507                            | ( 0.121 )       | 3200              | ( 5300.3 ) | 0.642                            | ( 0.154 )       |
| 1200              | ( 1700.3 ) | 0.514                            | ( 0.123 )       | 3253              | ( 5395.7 ) | 0.646                            | ( 0.154 )       |

**Application Notes:** Data for specific heat is collected from references [3, 8-14] and fitted with the equations below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $53 \leq T \leq 294$

$$C_p(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

For temperature range:  $294 \leq T \leq 3253$

$$C_p(T) = B0 + B1 \cdot \left(\frac{T}{1000}\right) + B_2 / \left(\frac{T}{1000}\right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$T = \text{Temperature [K]}$

## Constants:

T. Range [K]:  $53 \leq T < 293$

A0 = -0.09452

A1 = 2.94

A2 = -5.398

A3 = 3.077

$293 \leq T \leq 3253$

B0 = 0.4432

B1 = 0.06247

B<sub>2</sub> = -0.006519



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

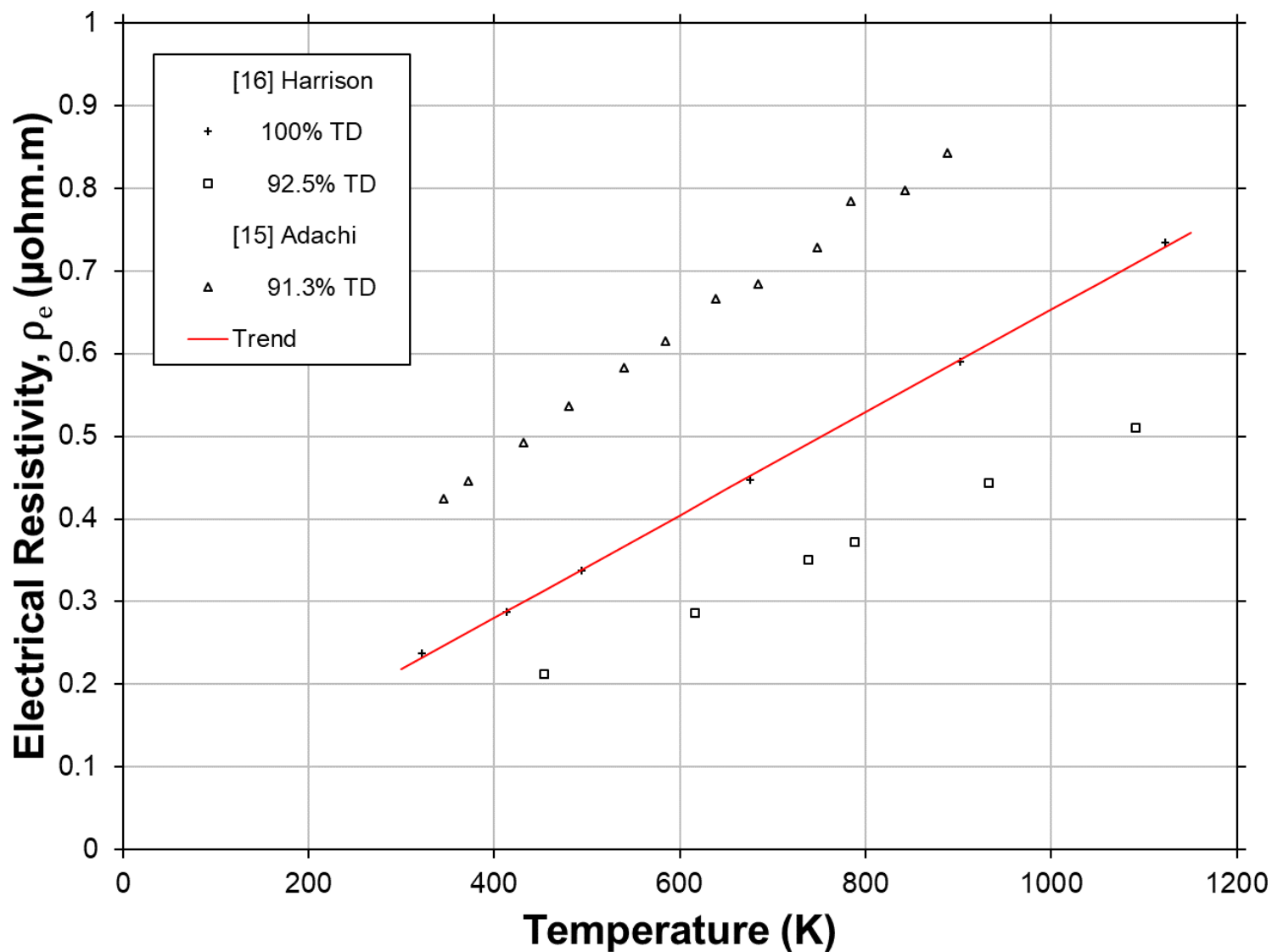


Figure 7.3.1-7: Reference data for Electrical Resistivity versus Temperature of ZrN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |           | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|-----------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )    | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 293               | ( 67.7 )  | 0.214                               | ( 8.42 )                | 650               | ( 710.3 )  | 0.436                               | ( 17.16 )               |
| 300               | ( 80.3 )  | 0.218                               | ( 8.59 )                | 675               | ( 755.3 )  | 0.451                               | ( 17.77 )               |
| 325               | ( 125.3 ) | 0.234                               | ( 9.21 )                | 700               | ( 800.3 )  | 0.467                               | ( 18.38 )               |
| 350               | ( 170.3 ) | 0.249                               | ( 9.82 )                | 725               | ( 845.3 )  | 0.482                               | ( 18.99 )               |
| 375               | ( 215.3 ) | 0.265                               | ( 10.43 )               | 750               | ( 890.3 )  | 0.498                               | ( 19.61 )               |
| 400               | ( 260.3 ) | 0.280                               | ( 11.04 )               | 775               | ( 935.3 )  | 0.514                               | ( 20.22 )               |
| 425               | ( 305.3 ) | 0.296                               | ( 11.65 )               | 800               | ( 980.3 )  | 0.529                               | ( 20.83 )               |
| 450               | ( 350.3 ) | 0.312                               | ( 12.26 )               | 825               | ( 1025.3 ) | 0.545                               | ( 21.44 )               |
| 475               | ( 395.3 ) | 0.327                               | ( 12.88 )               | 850               | ( 1070.3 ) | 0.560                               | ( 22.05 )               |
| 500               | ( 440.3 ) | 0.343                               | ( 13.49 )               | 900               | ( 1160.3 ) | 0.591                               | ( 23.28 )               |
| 525               | ( 485.3 ) | 0.358                               | ( 14.10 )               | 950               | ( 1250.3 ) | 0.622                               | ( 24.50 )               |
| 550               | ( 530.3 ) | 0.374                               | ( 14.71 )               | 1000              | ( 1340.3 ) | 0.653                               | ( 25.72 )               |
| 575               | ( 575.3 ) | 0.389                               | ( 15.32 )               | 1050              | ( 1430.3 ) | 0.684                               | ( 26.95 )               |
| 600               | ( 620.3 ) | 0.405                               | ( 15.94 )               | 1100              | ( 1520.3 ) | 0.716                               | ( 28.17 )               |
| 625               | ( 665.3 ) | 0.420                               | ( 16.55 )               | 1175              | ( 1655.3 ) | 0.762                               | ( 30.00 )               |

**Application Notes:** Data for electrical resistivity is collected from references [15, 16] and fitted with the equation below for 100% theoretical density to approximate property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right)$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $293 \leq T < 1175$

A0 = 0.03185

A1 = 0.6215



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 7370                                  | -                                                       | -                                           | -0.093                                             | 6.277                                     | 0.149                                          |                                    |                                  |                             |
| 200                 | 7361                                  | -                                                       | -                                           | -0.049                                             | 6.456                                     | 0.302                                          |                                    |                                  |                             |
| 300                 | 7349                                  | 0.218                                                   | 6.27                                        | 0.004                                              | 6.627                                     | 0.390                                          | 366.8                              | 149.9                            | 0.223                       |
| 400                 | 7334                                  | 0.280                                                   | 8.52                                        | 0.073                                              | 6.789                                     | 0.427                                          |                                    |                                  |                             |
| 500                 | 7318                                  | 0.343                                                   | 10.60                                       | 0.144                                              | 6.942                                     | 0.448                                          |                                    |                                  |                             |
| 600                 | 7302                                  | 0.405                                                   | 12.51                                       | 0.218                                              | 7.088                                     | 0.463                                          |                                    |                                  |                             |
| 700                 | 7285                                  | 0.467                                                   | 14.26                                       | 0.295                                              | 7.227                                     | 0.474                                          |                                    |                                  |                             |
| 800                 | 7268                                  | 0.529                                                   | 15.84                                       | 0.373                                              | 7.357                                     | 0.483                                          |                                    |                                  |                             |
| 900                 | 7251                                  | 0.591                                                   | 17.25                                       | 0.454                                              | 7.481                                     | 0.491                                          |                                    |                                  |                             |
| 1000                | 7233                                  | 0.653                                                   | 18.49                                       | 0.537                                              | 7.598                                     | 0.499                                          |                                    |                                  |                             |
| 1100                | 7215                                  | 0.716                                                   | 19.56                                       | 0.621                                              | 7.708                                     | 0.507                                          |                                    |                                  |                             |
| 1200                | 7196                                  | -                                                       | 20.47                                       | 0.708                                              | 7.812                                     | 0.514                                          |                                    |                                  |                             |
| 1300                | 7177                                  | -                                                       | 21.21                                       | 0.796                                              | 7.909                                     | 0.521                                          |                                    |                                  |                             |
| 1400                | 7158                                  | -                                                       | 21.77                                       | 0.885                                              | 8.001                                     | 0.527                                          |                                    |                                  |                             |
| 1500                | 7139                                  | -                                                       | 22.18                                       | 0.975                                              | 8.087                                     | 0.534                                          |                                    |                                  |                             |
| 1600                | 7120                                  | -                                                       | 22.41                                       | 1.067                                              | 8.168                                     | 0.541                                          |                                    |                                  |                             |
| 1700                | 7100                                  | -                                                       | 22.47                                       | 1.160                                              | 8.243                                     | 0.547                                          |                                    |                                  |                             |
| 1800                | 7080                                  | -                                                       | 22.37                                       | 1.253                                              | 8.314                                     | 0.554                                          |                                    |                                  |                             |
| 2000                | 7041                                  | -                                                       | 21.66                                       | 1.442                                              | 8.442                                     | 0.567                                          |                                    |                                  |                             |
| 2200                | 7001                                  | -                                                       | 20.27                                       | 1.633                                              | 8.553                                     | 0.579                                          |                                    |                                  |                             |
| 2400                | 6962                                  | -                                                       | 18.22                                       | 1.824                                              | 8.650                                     | 0.592                                          |                                    |                                  |                             |
| 2600                | 6923                                  | -                                                       | -                                           | 2.015                                              | 8.735                                     | 0.605                                          |                                    |                                  |                             |
| 2800                | -                                     | -                                                       | -                                           | -                                                  | -                                         | 0.617                                          |                                    |                                  |                             |
| 3000                | -                                     | -                                                       | -                                           | -                                                  | -                                         | 0.630                                          |                                    |                                  |                             |
| 3200                | -                                     | -                                                       | -                                           | -                                                  | -                                         | 0.642                                          |                                    |                                  |                             |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

**Revision 0: 08-05-2020**

**References**

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## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.1 Zirconium Nitride (ZrN)

**Revision 0: 08-05-2020**

**References**

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## **7 Refractory Ceramics**

### **7.3 Nitrides**



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 2.1: 08-25-2023

General

## Room Temperature Properties

|                                           |        |
|-------------------------------------------|--------|
| Molar Mass, [g/mol]                       | 192.50 |
| Theoretical Density, [kg/m <sup>3</sup> ] | 13,940 |
| Melting Point, [K]                        | 3660   |
| Specific Heat, [J/(g-K)]                  | 0.210  |
| Thermal Conductivity, [W/(m-K)]           | 9.0    |
| Linear expansion coefficient, [μm/(m-K)]  | 5.73   |
| Electrical resistivity, [μΩ-m]            | 0.30   |
| Young's Modulus, [GPa]                    | 404.1  |
| Shear Modulus, [GPa]                      | 172.4  |
| Poisson's Ratio, [-]                      | 0.172  |

## Hafnium – Nitrogen Phase Diagram

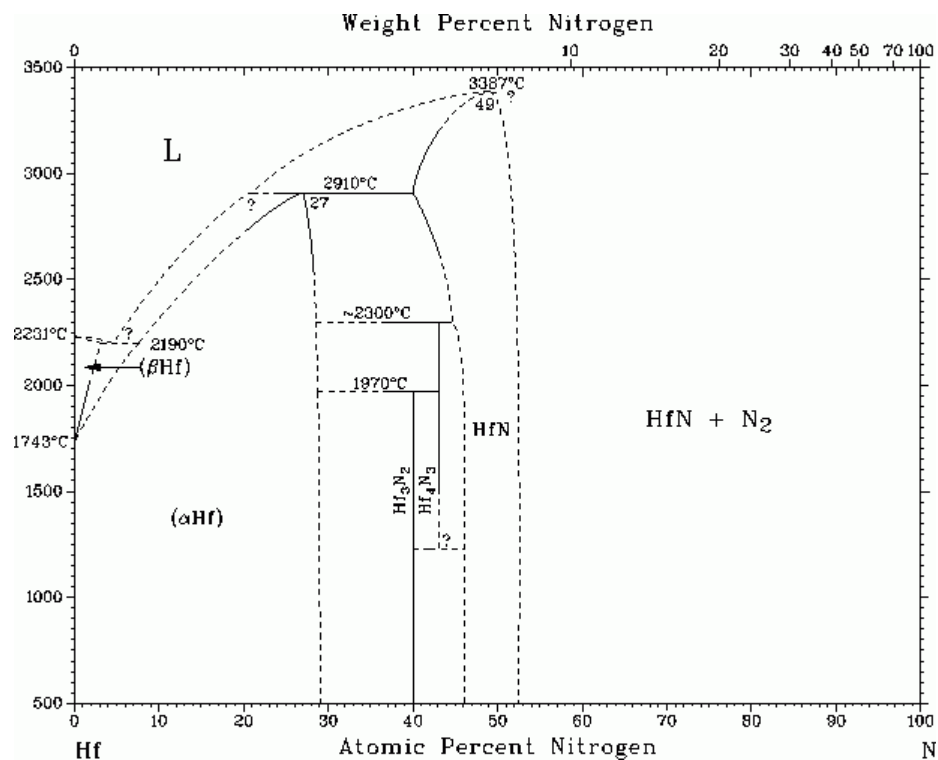


Figure 7.3.2-1: Hafnium – Nitrogen Phase Diagram.

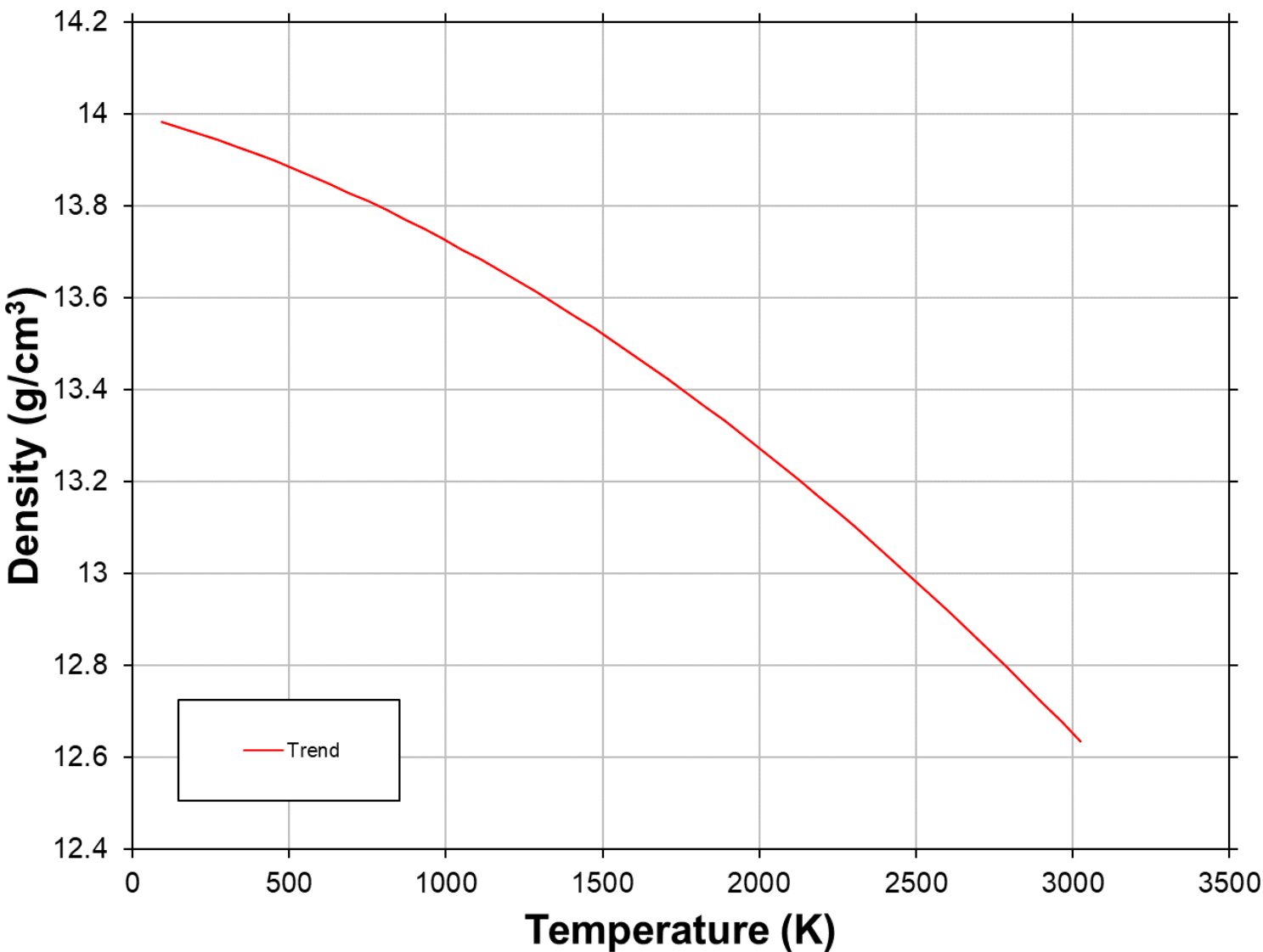


Figure 7.3.2-2: Density versus Temperature for HfN. Calculated from fitted trend of the Thermal Expansion data.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 0: 08-05-2020

Density with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Density ( ρ )     |                        | Temperature ( T ) |            | Density ( ρ )     |                        |
|-------------------|------------|-------------------|------------------------|-------------------|------------|-------------------|------------------------|
| K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) | K                 | ( °F )     | kg/m <sup>3</sup> | ( lb/ft <sup>3</sup> ) |
| 93                | ( -292.3 ) | 13983             | ( 872.9 )              | 1500              | ( 2240.3 ) | 13520             | ( 844.1 )              |
| 100               | ( -279.7 ) | 13981             | ( 872.9 )              | 1600              | ( 2420.3 ) | 13474             | ( 841.2 )              |
| 200               | ( -99.7 )  | 13960             | ( 871.5 )              | 1700              | ( 2600.3 ) | 13426             | ( 838.2 )              |
| 300               | ( 80.3 )   | 13936             | ( 870.1 )              | 1800              | ( 2780.3 ) | 13377             | ( 835.1 )              |
| 400               | ( 260.3 )  | 13911             | ( 868.5 )              | 1900              | ( 2960.3 ) | 13326             | ( 831.9 )              |
| 500               | ( 440.3 )  | 13885             | ( 866.8 )              | 2000              | ( 3140.3 ) | 13273             | ( 828.6 )              |
| 600               | ( 620.3 )  | 13856             | ( 865.0 )              | 2100              | ( 3320.3 ) | 13218             | ( 825.2 )              |
| 700               | ( 800.3 )  | 13826             | ( 863.1 )              | 2200              | ( 3500.3 ) | 13162             | ( 821.7 )              |
| 800               | ( 980.3 )  | 13794             | ( 861.1 )              | 2300              | ( 3680.3 ) | 13104             | ( 818.1 )              |
| 900               | ( 1160.3 ) | 13760             | ( 859.0 )              | 2400              | ( 3860.3 ) | 13044             | ( 814.3 )              |
| 1000              | ( 1340.3 ) | 13724             | ( 856.8 )              | 2500              | ( 4040.3 ) | 12983             | ( 810.5 )              |
| 1100              | ( 1520.3 ) | 13687             | ( 854.5 )              | 2600              | ( 4220.3 ) | 12920             | ( 806.6 )              |
| 1200              | ( 1700.3 ) | 13648             | ( 852.0 )              | 2700              | ( 4400.3 ) | 12855             | ( 802.5 )              |
| 1300              | ( 1880.3 ) | 13607             | ( 849.5 )              | 2850              | ( 4670.3 ) | 12755             | ( 796.3 )              |
| 1400              | ( 2060.3 ) | 13565             | ( 846.8 )              | 3025              | ( 4985.3 ) | 12634             | ( 788.8 )              |

**Application Notes:** Density is calculated as a function of thermal expansion, as seen in the equation below to approximate property trend with respect to temperature.

## Density Calculation:

$$\rho(T) = \rho_{RT} / (1 + dL/L_0(T)/100)^3$$

$$\rho(T) = \text{Density [kg/m}^3\text{]}$$

$$\text{Room Temperature Density } (\rho_{RT}) = 13,940 \text{ [kg/m}^3\text{]}$$

$$T = \text{Temperature [K]}$$

**Temperature Range:**  $93 \leq T \leq 3025$

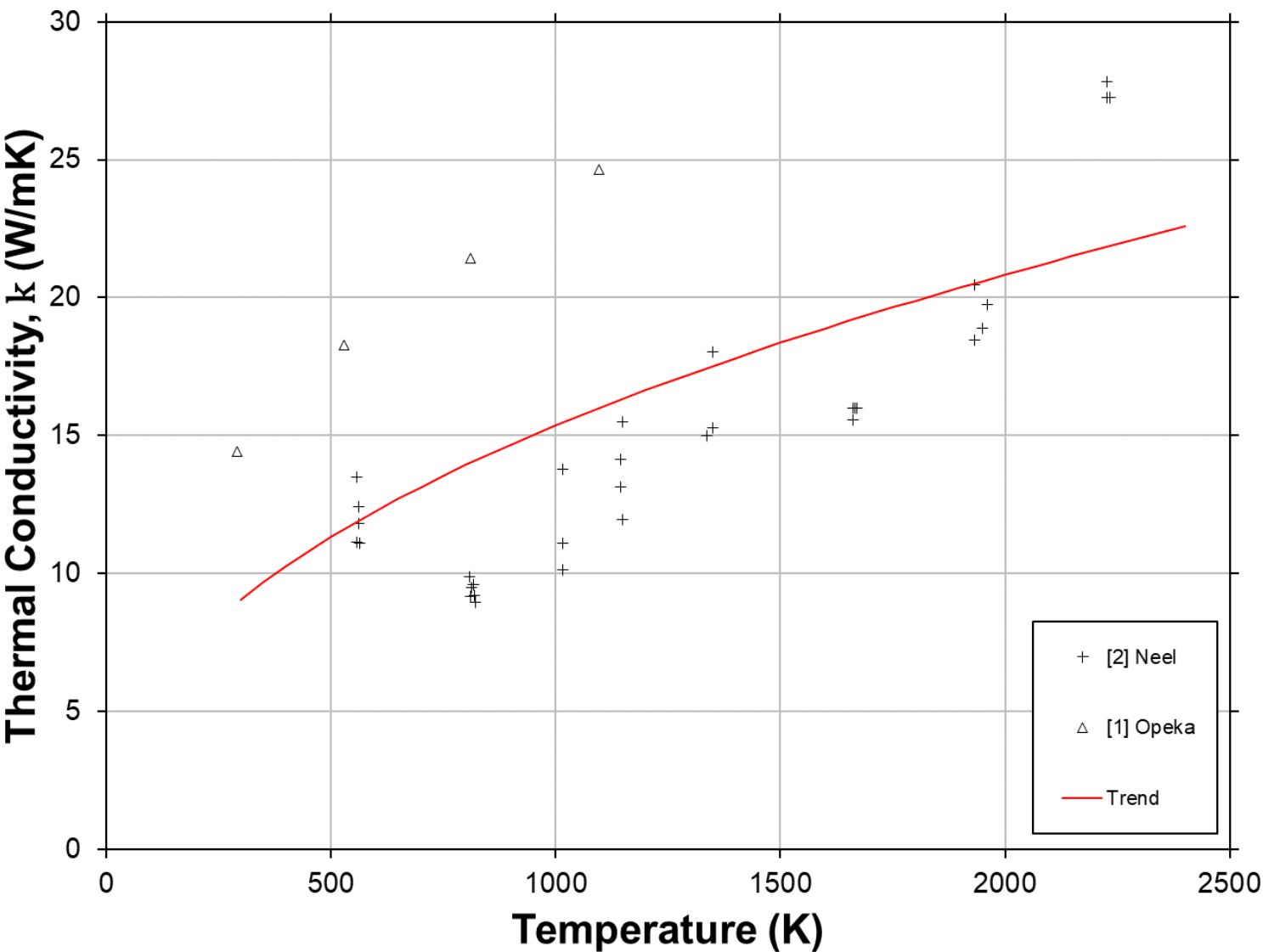


Figure 7.3.2-3: Thermal Conductivity versus Temperature of HfN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 0: 08-05-2020

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                     |
|-------------------|------------|----------------------------|-------------------------------------|-------------------|------------|----------------------------|-------------------------------------|
| K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) | K                 | ( °F )     | W/(m·K)                    | ((Btu·in)/(ft <sup>2</sup> ·hr·°F)) |
| 298               | ( 76.7 )   | 9.02                       | ( 62.56 )                           | 1050              | ( 1430.3 ) | 15.69                      | ( 108.87 )                          |
| 350               | ( 170.3 )  | 9.68                       | ( 67.15 )                           | 1100              | ( 1520.3 ) | 16.02                      | ( 111.12 )                          |
| 400               | ( 260.3 )  | 10.26                      | ( 71.21 )                           | 1200              | ( 1700.3 ) | 16.64                      | ( 115.45 )                          |
| 450               | ( 350.3 )  | 10.81                      | ( 75.00 )                           | 1300              | ( 1880.3 ) | 17.24                      | ( 119.59 )                          |
| 500               | ( 440.3 )  | 11.32                      | ( 78.55 )                           | 1400              | ( 2060.3 ) | 17.81                      | ( 123.55 )                          |
| 550               | ( 530.3 )  | 11.81                      | ( 81.92 )                           | 1500              | ( 2240.3 ) | 18.36                      | ( 127.36 )                          |
| 600               | ( 620.3 )  | 12.27                      | ( 85.11 )                           | 1600              | ( 2420.3 ) | 18.89                      | ( 131.03 )                          |
| 650               | ( 710.3 )  | 12.71                      | ( 88.16 )                           | 1700              | ( 2600.3 ) | 19.40                      | ( 134.57 )                          |
| 700               | ( 800.3 )  | 13.13                      | ( 91.08 )                           | 1800              | ( 2780.3 ) | 19.89                      | ( 137.99 )                          |
| 750               | ( 890.3 )  | 13.53                      | ( 93.89 )                           | 1900              | ( 2960.3 ) | 20.37                      | ( 141.32 )                          |
| 800               | ( 980.3 )  | 13.92                      | ( 96.59 )                           | 2000              | ( 3140.3 ) | 20.83                      | ( 144.54 )                          |
| 850               | ( 1070.3 ) | 14.30                      | ( 99.20 )                           | 2100              | ( 3320.3 ) | 21.28                      | ( 147.68 )                          |
| 900               | ( 1160.3 ) | 14.66                      | ( 101.73 )                          | 2200              | ( 3500.3 ) | 21.72                      | ( 150.73 )                          |
| 950               | ( 1250.3 ) | 15.02                      | ( 104.18 )                          | 2300              | ( 3680.3 ) | 22.15                      | ( 153.70 )                          |
| 1000              | ( 1340.3 ) | 15.36                      | ( 106.56 )                          | 2400              | ( 3860.3 ) | 22.57                      | ( 156.61 )                          |

**Application Notes:** Data for thermal conductivity is collected from references [1, 2] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$k(T) = A0 \cdot \left( \frac{T}{1000} \right)^N$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]: 298 ≤ T ≤ 2400

A0 = 15.358

N = 0.4399



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

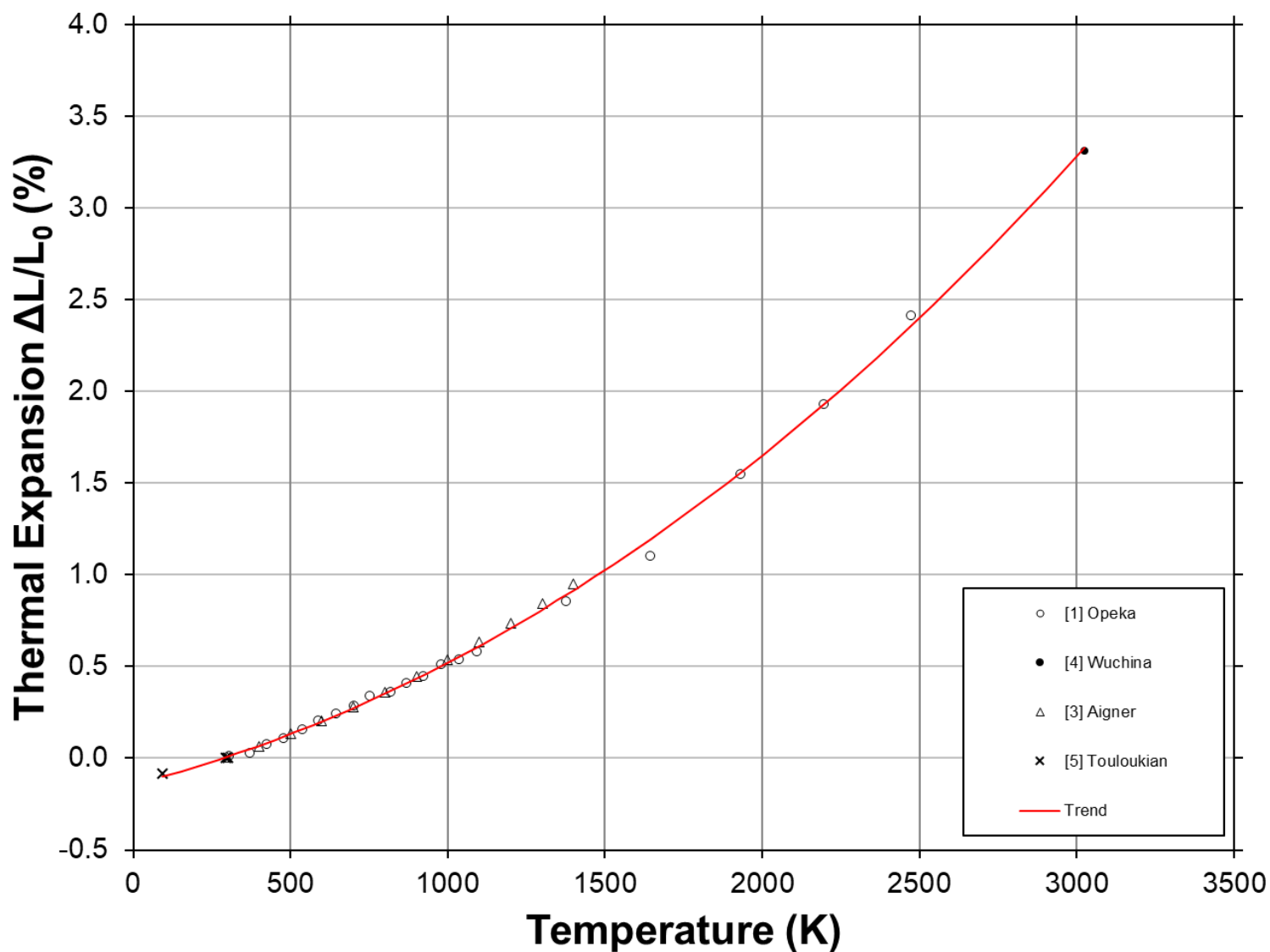


Figure 7.3.2-4: Thermal Expansion versus Temperature of HfN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 0: 08-05-2020

Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) | Temperature ( T ) |            | Thermal Expansion ( dL/L <sub>0</sub> ) |
|-------------------|------------|-----------------------------------------|-------------------|------------|-----------------------------------------|
| K                 | ( °F )     | %                                       | K                 | ( °F )     | %                                       |
| 93                | ( -292.3 ) | -0.102                                  | 1500              | ( 2240.3 ) | 1.024                                   |
| 100               | ( -279.7 ) | -0.099                                  | 1600              | ( 2420.3 ) | 1.139                                   |
| 200               | ( -99.7 )  | -0.047                                  | 1700              | ( 2600.3 ) | 1.259                                   |
| 300               | ( 80.3 )   | 0.008                                   | 1800              | ( 2780.3 ) | 1.384                                   |
| 400               | ( 260.3 )  | 0.068                                   | 1900              | ( 2960.3 ) | 1.514                                   |
| 500               | ( 440.3 )  | 0.133                                   | 2000              | ( 3140.3 ) | 1.649                                   |
| 600               | ( 620.3 )  | 0.201                                   | 2100              | ( 3320.3 ) | 1.789                                   |
| 700               | ( 800.3 )  | 0.275                                   | 2200              | ( 3500.3 ) | 1.934                                   |
| 800               | ( 980.3 )  | 0.352                                   | 2300              | ( 3680.3 ) | 2.084                                   |
| 900               | ( 1160.3 ) | 0.434                                   | 2400              | ( 3860.3 ) | 2.239                                   |
| 1000              | ( 1340.3 ) | 0.521                                   | 2500              | ( 4040.3 ) | 2.400                                   |
| 1100              | ( 1520.3 ) | 0.612                                   | 2600              | ( 4220.3 ) | 2.566                                   |
| 1200              | ( 1700.3 ) | 0.708                                   | 2700              | ( 4400.3 ) | 2.738                                   |
| 1300              | ( 1880.3 ) | 0.809                                   | 2850              | ( 4670.3 ) | 3.005                                   |
| 1400              | ( 2060.3 ) | 0.914                                   | 3025              | ( 4985.3 ) | 3.333                                   |

**Application Notes:** Data for thermal expansion is collected from references [1, 3-5] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$dL/L_0(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$dL/L_0(T) = \text{Thermal Expansion } [\%]$$

$$T = \text{Temperature } [K]$$

## Constants:

$$T \text{ Range } [K]: \quad 93 \leq T < 3025$$

$$A0 = -0.1461$$

$$A1 = 0.4525$$

$$A2 = 0.2065$$

$$A3 = 0.007964$$

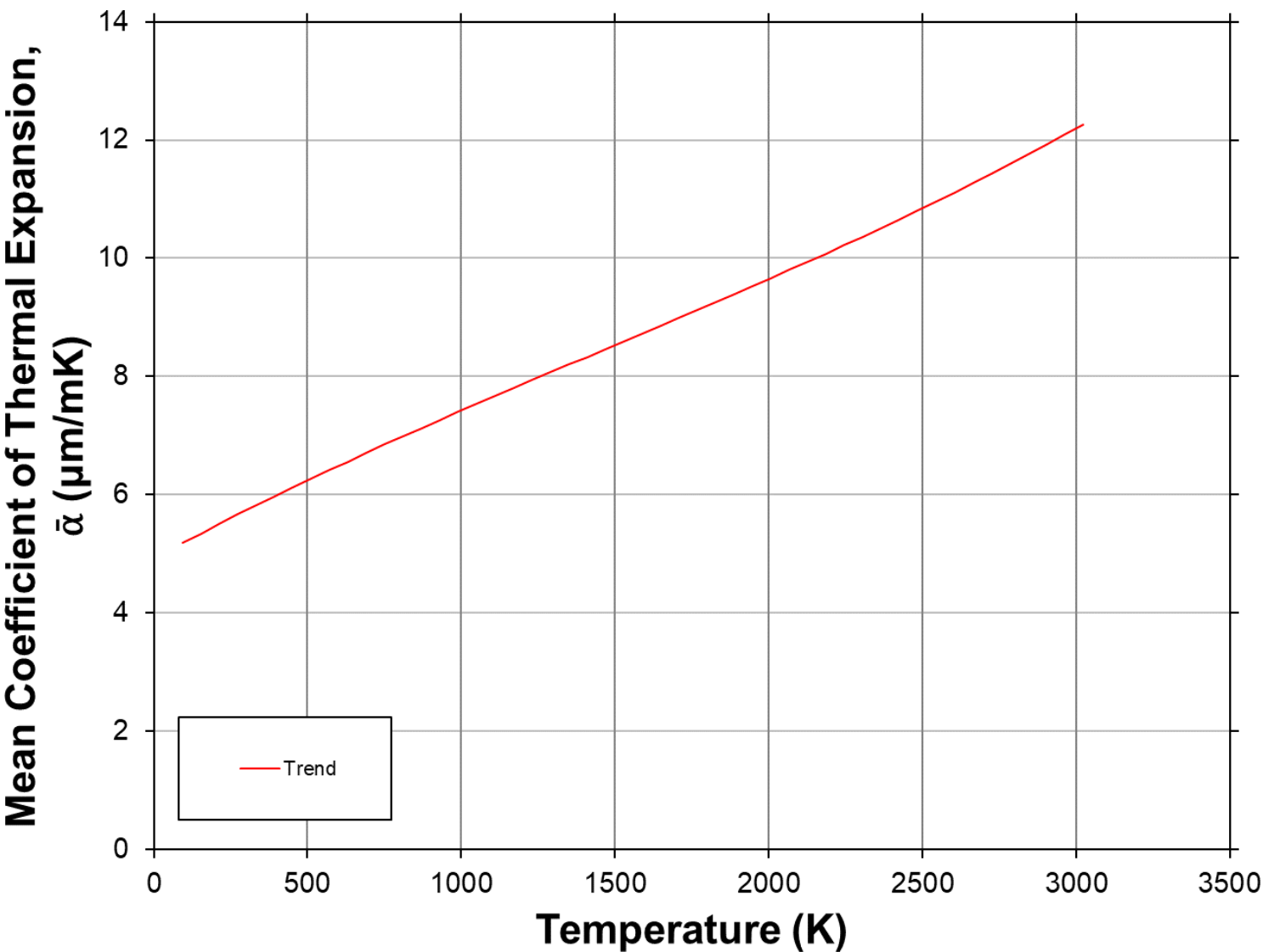


Figure 7.3.2-5: Mean Coefficient of Thermal Expansion versus Temperature of HfN as calculated from fitted trend of the Thermal Expansion.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 0: 08-05-2020

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     | Temperature ( T ) |            | Mean CTE ( $\bar{\alpha}$ )           |                                                     |
|-------------------|------------|---------------------------------------|-----------------------------------------------------|-------------------|------------|---------------------------------------|-----------------------------------------------------|
| K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) | K                 | ( °F )     | $\mu\text{m}/(\text{m}\cdot\text{K})$ | ( $\mu\text{in}/(\text{in}\cdot^{\circ}\text{F})$ ) |
| 93                | ( -292.3 ) | 5.179                                 | ( 2.877 )                                           | 1500              | ( 2240.3 ) | 8.532                                 | ( 4.740 )                                           |
| 100               | ( -279.7 ) | 5.199                                 | ( 2.888 )                                           | 1600              | ( 2420.3 ) | 8.753                                 | ( 4.863 )                                           |
| 200               | ( -99.7 )  | 5.470                                 | ( 3.039 )                                           | 1700              | ( 2600.3 ) | 8.974                                 | ( 4.985 )                                           |
| 300               | ( 80.3 )   | 5.734                                 | ( 3.186 )                                           | 1800              | ( 2780.3 ) | 9.197                                 | ( 5.109 )                                           |
| 400               | ( 260.3 )  | 5.991                                 | ( 3.328 )                                           | 1900              | ( 2960.3 ) | 9.421                                 | ( 5.234 )                                           |
| 500               | ( 440.3 )  | 6.242                                 | ( 3.468 )                                           | 2000              | ( 3140.3 ) | 9.648                                 | ( 5.360 )                                           |
| 600               | ( 620.3 )  | 6.487                                 | ( 3.604 )                                           | 2100              | ( 3320.3 ) | 9.879                                 | ( 5.488 )                                           |
| 700               | ( 800.3 )  | 6.727                                 | ( 3.737 )                                           | 2200              | ( 3500.3 ) | 10.113                                | ( 5.618 )                                           |
| 800               | ( 980.3 )  | 6.962                                 | ( 3.868 )                                           | 2300              | ( 3680.3 ) | 10.352                                | ( 5.751 )                                           |
| 900               | ( 1160.3 ) | 7.194                                 | ( 3.996 )                                           | 2400              | ( 3860.3 ) | 10.595                                | ( 5.886 )                                           |
| 1000              | ( 1340.3 ) | 7.422                                 | ( 4.123 )                                           | 2500              | ( 4040.3 ) | 10.845                                | ( 6.025 )                                           |
| 1100              | ( 1520.3 ) | 7.647                                 | ( 4.248 )                                           | 2600              | ( 4220.3 ) | 11.100                                | ( 6.167 )                                           |
| 1200              | ( 1700.3 ) | 7.870                                 | ( 4.372 )                                           | 2700              | ( 4400.3 ) | 11.362                                | ( 6.312 )                                           |
| 1300              | ( 1880.3 ) | 8.091                                 | ( 4.495 )                                           | 2850              | ( 4670.3 ) | 11.769                                | ( 6.539 )                                           |
| 1400              | ( 2060.3 ) | 8.312                                 | ( 4.618 )                                           | 3025              | ( 4985.3 ) | 12.268                                | ( 6.816 )                                           |

**Application Notes:** Trend for mean coefficient of thermal expansion is calculated from thermal expansion and shown in the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\bar{\alpha}(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\bar{\alpha}(T)$  = Coefficient of Thermal Expansion [ $\mu\text{m}/(\text{m}\cdot\text{K})$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]: 93 ≤ T ≤ 3025

A0 = 4.919

A1 = 2.839

A2 = -0.4357

A3 = 0.09928



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 0: 08-05-2020

Specific Heat with Temperature

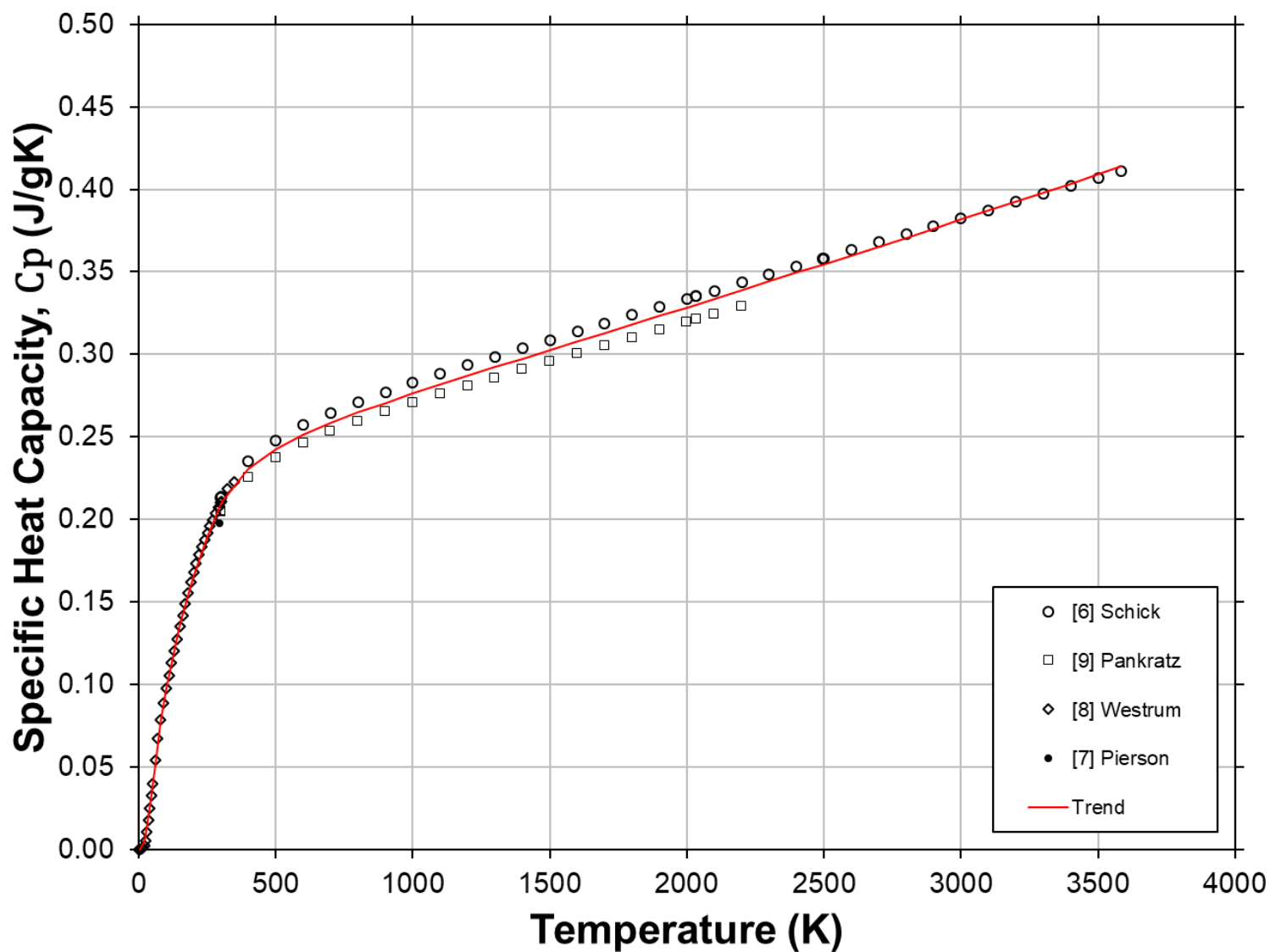


Figure 7.3.2-6: Specific Heat versus Temperature of HfN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 0: 08-05-2020

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 293               | ( 67.7 )   | 0.208                            | ( 0.050 )       | 1600              | ( 2420.3 ) | 0.308                            | ( 0.074 )       |
| 400               | ( 260.3 )  | 0.231                            | ( 0.055 )       | 1700              | ( 2600.3 ) | 0.313                            | ( 0.075 )       |
| 500               | ( 440.3 )  | 0.243                            | ( 0.058 )       | 1800              | ( 2780.3 ) | 0.318                            | ( 0.076 )       |
| 600               | ( 620.3 )  | 0.251                            | ( 0.060 )       | 1900              | ( 2960.3 ) | 0.323                            | ( 0.077 )       |
| 700               | ( 800.3 )  | 0.259                            | ( 0.062 )       | 2000              | ( 3140.3 ) | 0.328                            | ( 0.079 )       |
| 800               | ( 980.3 )  | 0.265                            | ( 0.063 )       | 2200              | ( 3500.3 ) | 0.339                            | ( 0.081 )       |
| 900               | ( 1160.3 ) | 0.271                            | ( 0.065 )       | 2400              | ( 3860.3 ) | 0.349                            | ( 0.084 )       |
| 1000              | ( 1340.3 ) | 0.276                            | ( 0.066 )       | 2600              | ( 4220.3 ) | 0.360                            | ( 0.086 )       |
| 1100              | ( 1520.3 ) | 0.282                            | ( 0.067 )       | 2800              | ( 4580.3 ) | 0.371                            | ( 0.089 )       |
| 1200              | ( 1700.3 ) | 0.287                            | ( 0.069 )       | 3000              | ( 4940.3 ) | 0.382                            | ( 0.091 )       |
| 1300              | ( 1880.3 ) | 0.292                            | ( 0.070 )       | 3200              | ( 5300.3 ) | 0.393                            | ( 0.094 )       |
| 1400              | ( 2060.3 ) | 0.297                            | ( 0.071 )       | 3400              | ( 5660.3 ) | 0.404                            | ( 0.096 )       |
| 1500              | ( 2240.3 ) | 0.303                            | ( 0.072 )       | 3583              | ( 5989.7 ) | 0.414                            | ( 0.099 )       |

**Application Notes:** Data for specific heat is collected from references [6-9] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

For temperature range:  $5 \leq T < 293$

$$C_p(T) = \left[ A0 \cdot \left( \frac{T}{1000} \right)^N \right] / \left[ 1 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2 \right]$$

For temperature range:  $293 \leq T \leq 3583$

$$C_p(T) = B0 + B1 \cdot \left( \frac{T}{1000} \right) + B2 \cdot \left( \frac{T}{1000} \right)^2 + B\_2 / \left( \frac{T}{1000} \right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

T. Range [K]:  $5 \leq T < 293$

N = 2.586

A0 = 120.7

A1 = -7.663

A2 = 296.5

$293 \leq T \leq 3583$

B0 = 0.2326

B1 = 0.04545

B2 = 0.001445

B\_2 = -0.003245

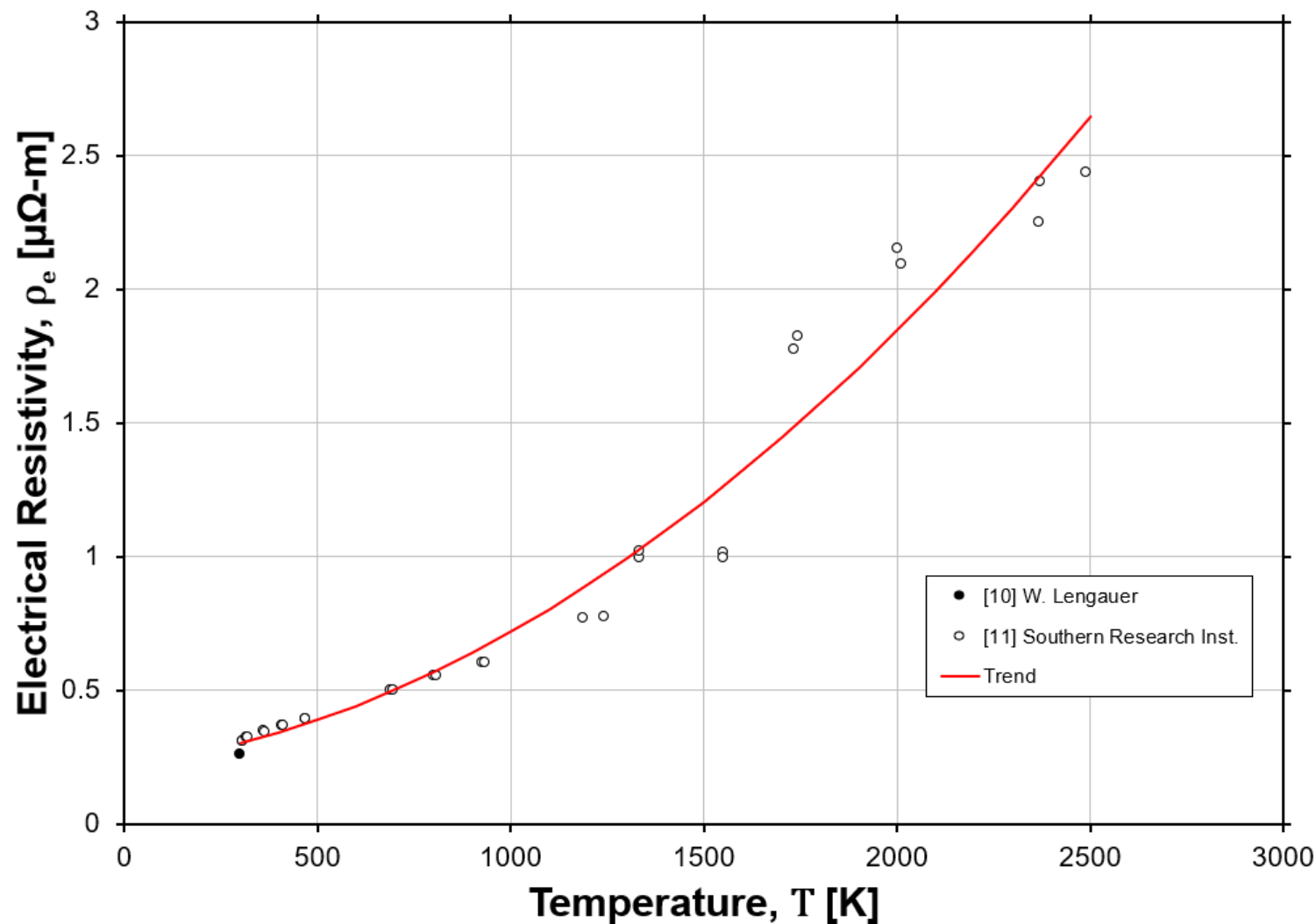


Figure 7.3.2-7: Electrical Resistivity versus Temperature of HfN.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 0: 08-05-2020

Electrical Resistivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         | Temperature ( T ) |            | Electrical Resistivity ( $\rho_e$ ) |                         |
|-------------------|------------|-------------------------------------|-------------------------|-------------------|------------|-------------------------------------|-------------------------|
| K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) | K                 | ( °F )     | $\mu\Omega\cdot m$                  | ( $\mu\Omega\cdot in$ ) |
| 290               | ( 62.3 )   | 0.296                               | ( 11.66 )               | 1100              | ( 1520.3 ) | 0.803                               | ( 31.61 )               |
| 300               | ( 80.3 )   | 0.300                               | ( 11.81 )               | 1200              | ( 1700.3 ) | 0.894                               | ( 35.19 )               |
| 350               | ( 170.3 )  | 0.320                               | ( 12.58 )               | 1300              | ( 1880.3 ) | 0.991                               | ( 39.02 )               |
| 400               | ( 260.3 )  | 0.341                               | ( 13.42 )               | 1400              | ( 2060.3 ) | 1.095                               | ( 43.10 )               |
| 450               | ( 350.3 )  | 0.364                               | ( 14.32 )               | 1500              | ( 2240.3 ) | 1.205                               | ( 47.43 )               |
| 500               | ( 440.3 )  | 0.388                               | ( 15.28 )               | 1600              | ( 2420.3 ) | 1.321                               | ( 52.00 )               |
| 550               | ( 530.3 )  | 0.414                               | ( 16.30 )               | 1700              | ( 2600.3 ) | 1.443                               | ( 56.82 )               |
| 600               | ( 620.3 )  | 0.442                               | ( 17.38 )               | 1800              | ( 2780.3 ) | 1.572                               | ( 61.88 )               |
| 650               | ( 710.3 )  | 0.471                               | ( 18.53 )               | 1900              | ( 2960.3 ) | 1.707                               | ( 67.19 )               |
| 700               | ( 800.3 )  | 0.501                               | ( 19.73 )               | 2000              | ( 3140.3 ) | 1.848                               | ( 72.75 )               |
| 750               | ( 890.3 )  | 0.533                               | ( 21.00 )               | 2100              | ( 3320.3 ) | 1.995                               | ( 78.55 )               |
| 800               | ( 980.3 )  | 0.567                               | ( 22.33 )               | 2200              | ( 3500.3 ) | 2.149                               | ( 84.61 )               |
| 850               | ( 1070.3 ) | 0.603                               | ( 23.72 )               | 2300              | ( 3680.3 ) | 2.309                               | ( 90.90 )               |
| 900               | ( 1160.3 ) | 0.640                               | ( 25.18 )               | 2400              | ( 3860.3 ) | 2.475                               | ( 97.45 )               |
| 1000              | ( 1340.3 ) | 0.718                               | ( 28.27 )               | 2500              | ( 4040.3 ) | 2.648                               | ( 104.24 )              |

**Application Notes:** Data for electrical resistivity is collected from references [10, 11] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$\rho_e(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$\rho_e(T)$  = Electrical Resistivity [ $\mu\Omega \cdot m$ ]

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $290 \leq T < 2500$

A0 = 0.2147

A1 = 0.1901

A2 = 0.3132



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

Revision 0: 08-05-2020

Tabulated Property Data

| Temp.<br>( T )<br>K | Physical and Electrical               |                                                         | Thermal                                     |                                                    |                                           |                                                | Elastic                            |                                  |                             |
|---------------------|---------------------------------------|---------------------------------------------------------|---------------------------------------------|----------------------------------------------------|-------------------------------------------|------------------------------------------------|------------------------------------|----------------------------------|-----------------------------|
|                     | Density<br>( ρ )<br>kg/m <sup>3</sup> | Electrical<br>Resistivity<br>( ρ <sub>e</sub> )<br>μΩ-m | Thermal<br>Conductivity<br>( k )<br>W/(m-K) | Thermal<br>Expansion<br>( dL/L <sub>0</sub> )<br>% | Mean CTE<br>( ᾱ )<br>10 <sup>-6</sup> /K | Specific Heat<br>( C <sub>p</sub> )<br>J/(g-K) | Young's<br>Modulus<br>( E )<br>GPa | Shear<br>Modulus<br>( G )<br>GPa | Poisson's<br>Ratio<br>( ν ) |
| 100                 | 13981                                 | -                                                       | -                                           | -0.10                                              | 5.20                                      | 0.098                                          |                                    |                                  |                             |
| 200                 | 13960                                 | -                                                       | -                                           | -0.05                                              | 5.47                                      | 0.166                                          | 404.1                              | 172.4                            | 0.172                       |
| 300                 | 13936                                 | 0.300                                                   | 9.04                                        | 0.01                                               | 5.73                                      | 0.210                                          |                                    |                                  |                             |
| 400                 | 13911                                 | 0.341                                                   | 10.26                                       | 0.07                                               | 5.99                                      | 0.231                                          |                                    |                                  |                             |
| 500                 | 13885                                 | 0.388                                                   | 11.32                                       | 0.13                                               | 6.24                                      | 0.243                                          |                                    |                                  |                             |
| 600                 | 13856                                 | 0.442                                                   | 12.27                                       | 0.20                                               | 6.49                                      | 0.251                                          |                                    |                                  |                             |
| 700                 | 13826                                 | 0.501                                                   | 13.13                                       | 0.27                                               | 6.73                                      | 0.259                                          |                                    |                                  |                             |
| 800                 | 13794                                 | 0.567                                                   | 13.92                                       | 0.35                                               | 6.96                                      | 0.265                                          |                                    |                                  |                             |
| 900                 | 13760                                 | 0.640                                                   | 14.66                                       | 0.43                                               | 7.19                                      | 0.271                                          |                                    |                                  |                             |
| 1000                | 13724                                 | 0.718                                                   | 15.36                                       | 0.52                                               | 7.42                                      | 0.276                                          |                                    |                                  |                             |
| 1100                | 13687                                 | 0.803                                                   | 16.02                                       | 0.61                                               | 7.65                                      | 0.282                                          |                                    |                                  |                             |
| 1200                | 13648                                 | 0.894                                                   | 16.64                                       | 0.71                                               | 7.87                                      | 0.287                                          |                                    |                                  |                             |
| 1300                | 13607                                 | 0.991                                                   | 17.24                                       | 0.81                                               | 8.09                                      | 0.292                                          |                                    |                                  |                             |
| 1400                | 13565                                 | 1.095                                                   | 17.81                                       | 0.91                                               | 8.31                                      | 0.297                                          |                                    |                                  |                             |
| 1500                | 13520                                 | 1.205                                                   | 18.36                                       | 1.02                                               | 8.53                                      | 0.303                                          |                                    |                                  |                             |
| 1600                | 13474                                 | 1.321                                                   | 18.89                                       | 1.14                                               | 8.75                                      | 0.308                                          |                                    |                                  |                             |
| 1800                | 13377                                 | 1.572                                                   | 19.89                                       | 1.38                                               | 9.20                                      | 0.318                                          |                                    |                                  |                             |
| 2000                | 13273                                 | 1.848                                                   | 20.83                                       | 1.65                                               | 9.65                                      | 0.328                                          |                                    |                                  |                             |
| 2200                | 13162                                 | 2.149                                                   | 21.72                                       | 1.93                                               | 10.11                                     | 0.339                                          |                                    |                                  |                             |
| 2400                | 13044                                 | 2.475                                                   | 22.57                                       | 2.24                                               | 10.60                                     | 0.349                                          |                                    |                                  |                             |
| 2600                | 12920                                 | -                                                       | -                                           | 2.57                                               | 11.10                                     | 0.360                                          |                                    |                                  |                             |
| 2800                | 12789                                 | -                                                       | -                                           | 2.91                                               | 11.63                                     | 0.371                                          |                                    |                                  |                             |
| 3000                | 12652                                 | -                                                       | -                                           | 3.28                                               | 12.20                                     | 0.382                                          |                                    |                                  |                             |
| 3200                | -                                     | -                                                       | -                                           | -                                                  | -                                         | 0.393                                          |                                    |                                  |                             |
| 3400                | -                                     | -                                                       | -                                           | -                                                  | -                                         | 0.404                                          |                                    |                                  |                             |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

7 Refractory Ceramics

7.3 Nitrides

7.3.2 Hafnium Nitride (HfN)

**Revision 0: 08-05-2020**

**References**

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## **8 Composites and Structures**

### **8.1 Carbide Based**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

Revision 0: 08-02-2021

General

### Room Temperature Properties

|                                           |             |               |
|-------------------------------------------|-------------|---------------|
| Theoretical Density, [kg/m <sup>3</sup> ] | 2,300-2,800 |               |
| Specific Heat, [J/(g-K)]                  | 0.670       |               |
| Thermal Conductivity, [W/(m-K)]           | 10.0-16.0   |               |
| Linear expansion coefficient, [μm/(m-K)]  | 2.47        |               |
| Electrical Conductivity, [MS/m]           | 365.00      |               |
| Young's Modulus, [GPa]                    | 200-280     |               |
| Shear Modulus, [GPa]                      | 80-120      |               |
| Poisson's Ratio                           | 0.13-0.25   |               |
| Proportional Limit Stress [MPa]           | 90-120      | Flat Specimen |
|                                           | 80-110      | Tube Axial    |
|                                           | 100-160     | Tube Hoop     |
| Ultimate Tensile Strength [MPa]           | 250-350     | Flat Specimen |
|                                           | 230-270     | Tube Axial    |
|                                           | 200-340     | Tube Hoop     |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

Revision 0: 08-02-2021

Thermal Conductivity with Temperature

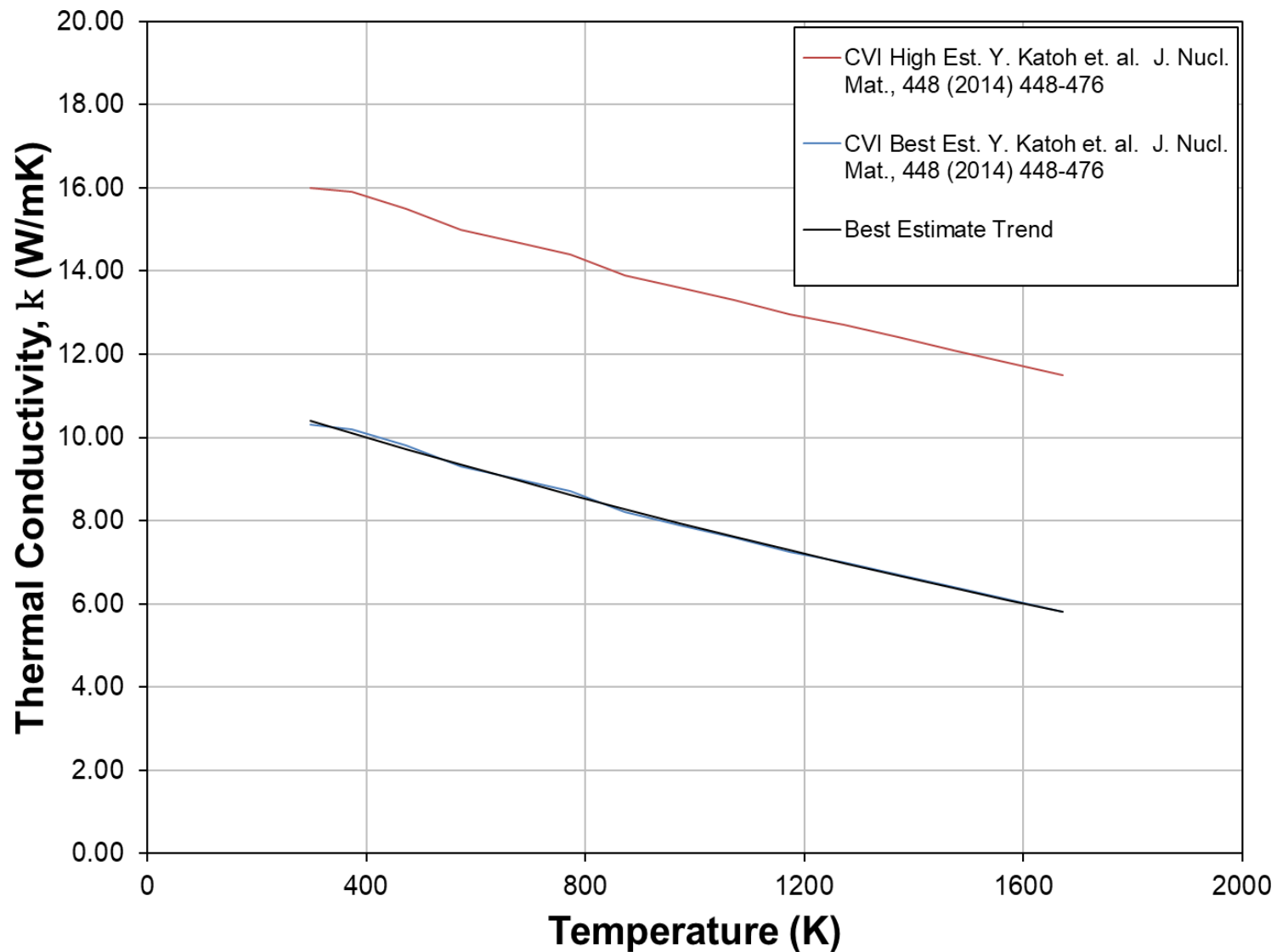


Figure 8.1.1-1: Thermal Conductivity vs. Temperature. Displaying fitted trend of data for best estimate values from Katoh (2104).



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

Revision 2: 04-26-2023

Thermal Conductivity with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       |
|-------------------|------------|----------------------------|---------------------------------------|-------------------|------------|----------------------------|---------------------------------------|
| K                 | ( °F )     | W/(m-K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) | K                 | ( °F )     | W/(m-K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) |
| 330               | ( 134.3 )  | 10.28                      | ( 71.33 )                             | 1000              | ( 1430.3 ) | 7.85                       | ( 54.45 )                             |
| 300               | ( 80.3 )   | 10.40                      | ( 72.15 )                             | 1050              | ( 1520.3 ) | 7.68                       | ( 53.30 )                             |
| 350               | ( 170.3 )  | 10.20                      | ( 70.78 )                             | 1100              | ( 1610.3 ) | 7.52                       | ( 52.17 )                             |
| 400               | ( 260.3 )  | 10.01                      | ( 69.43 )                             | 1150              | ( 1700.3 ) | 7.36                       | ( 51.06 )                             |
| 450               | ( 350.3 )  | 9.81                       | ( 68.09 )                             | 1200              | ( 1790.3 ) | 7.20                       | ( 49.97 )                             |
| 500               | ( 440.3 )  | 9.62                       | ( 66.77 )                             | 1250              | ( 1880.3 ) | 7.05                       | ( 48.88 )                             |
| 550               | ( 530.3 )  | 9.44                       | ( 65.47 )                             | 1300              | ( 1970.3 ) | 6.89                       | ( 47.82 )                             |
| 600               | ( 620.3 )  | 9.25                       | ( 64.18 )                             | 1350              | ( 2060.3 ) | 6.74                       | ( 46.77 )                             |
| 650               | ( 710.3 )  | 9.07                       | ( 62.91 )                             | 1400              | ( 2150.3 ) | 6.59                       | ( 45.74 )                             |
| 700               | ( 800.3 )  | 8.89                       | ( 61.65 )                             | 1450              | ( 2240.3 ) | 6.45                       | ( 44.72 )                             |
| 750               | ( 890.3 )  | 8.71                       | ( 60.41 )                             | 1500              | ( 2330.3 ) | 6.30                       | ( 43.72 )                             |
| 800               | ( 980.3 )  | 8.53                       | ( 59.19 )                             | 1550              | ( 2420.3 ) | 6.16                       | ( 42.74 )                             |
| 850               | ( 1070.3 ) | 8.36                       | ( 57.98 )                             | 1600              | ( 2420.3 ) | 6.02                       | ( 41.77 )                             |
| 900               | ( 1160.3 ) | 8.18                       | ( 56.78 )                             | 1650              | ( 2510.3 ) | 5.88                       | ( 40.82 )                             |
| 950               | ( 1250.3 ) | 8.01                       | ( 55.61 )                             | 1673              | ( 2551.7 ) | 5.82                       | ( 40.38 )                             |

**Application Notes:** Data for thermal conductivity is collected from reference [1] and fitted with the equation below to approximate property trend with respect to temperature. Best estimate data is for Hi-Nicalon Type S (HNLS) fiber / chemical vapor-infiltrated (CVI) SiC composite with either pyrolytic carbon (PyC) or multi-layer PyC/SiC interface.

**Fit Equation:**

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

**Constants:**

T Range [K]:  $298 \leq T \leq 1673$

A0 = 11.631

A1 = -4.244

A2 = 0.46103

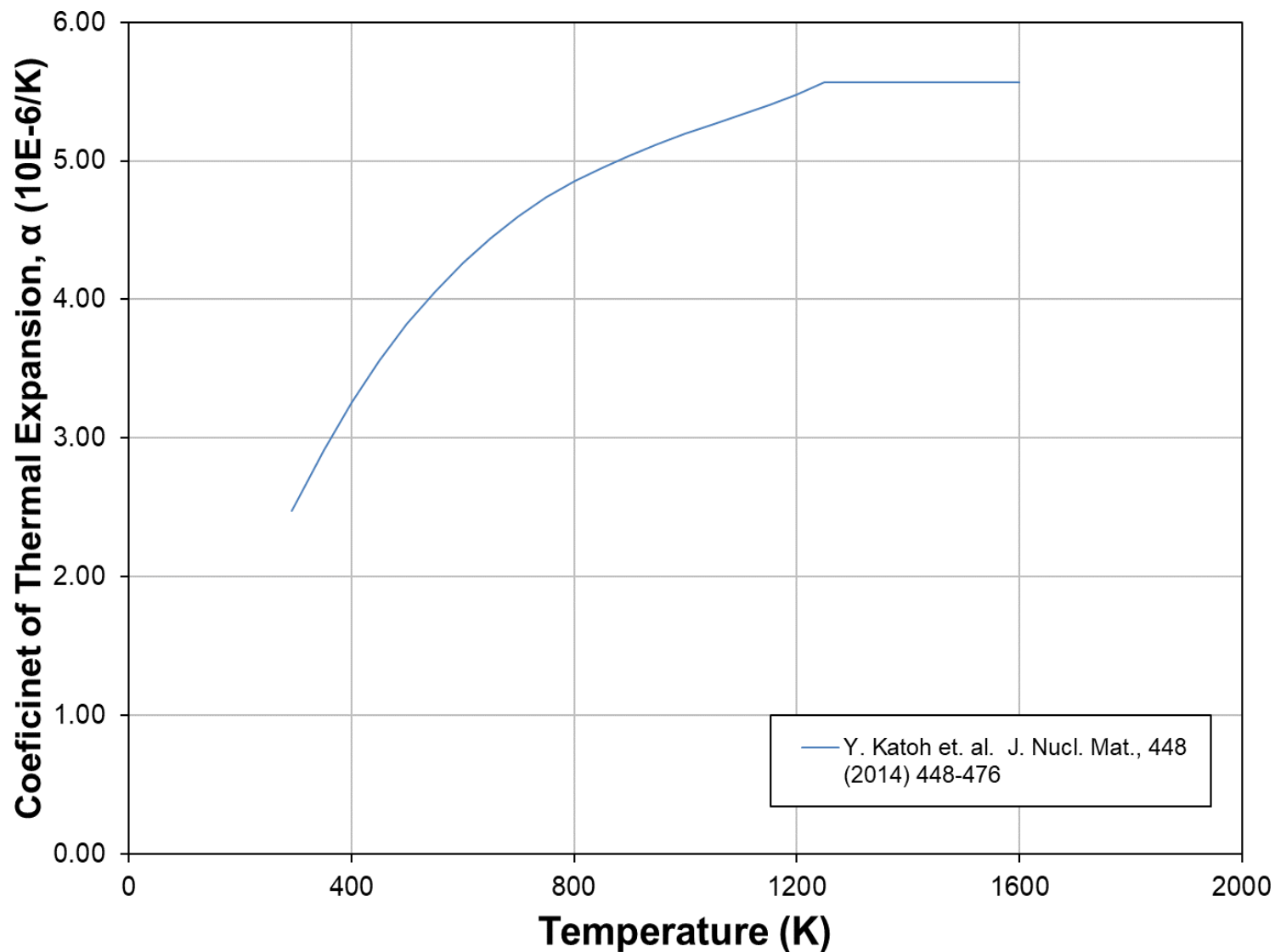


Figure 8.1.1-2: Thermal Expansion versus Temperature of SiC/SiC Composite.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

Revision 0: 08-02-2021

Coefficient of Thermal Expansion with Temperature

100% Theoretical Density

| Temperature ( T ) |            | CTE      |                 | Temperature ( T ) |            | CTE      |                 |
|-------------------|------------|----------|-----------------|-------------------|------------|----------|-----------------|
| K                 | ( °F )     | μm/(m-K) | ( μin/(in-°F) ) | K                 | ( °F )     | μm/(m-K) | ( μin/(in-°F) ) |
| 293               | ( 67.7 )   | 2.476    | ( 1.376 )       | 1150              | ( 1610.3 ) | 5.403    | ( 3.002 )       |
| 300               | ( 80.3 )   | 2.533    | ( 1.407 )       | 1200              | ( 1700.3 ) | 5.479    | ( 3.044 )       |
| 350               | ( 170.3 )  | 2.915    | ( 1.619 )       | 1250              | ( 1790.3 ) | 5.563    | ( 3.090 )       |
| 400               | ( 260.3 )  | 3.255    | ( 1.808 )       | 1300              | ( 1880.3 ) | 5.563    | ( 3.090 )       |
| 450               | ( 350.3 )  | 3.558    | ( 1.976 )       | 1350              | ( 1970.3 ) | 5.563    | ( 3.090 )       |
| 500               | ( 440.3 )  | 3.825    | ( 2.125 )       | 1400              | ( 2060.3 ) | 5.563    | ( 3.090 )       |
| 550               | ( 530.3 )  | 4.060    | ( 2.255 )       | 1450              | ( 2150.3 ) | 5.563    | ( 3.090 )       |
| 600               | ( 620.3 )  | 4.265    | ( 2.370 )       | 1500              | ( 2240.3 ) | 5.563    | ( 3.090 )       |
| 650               | ( 710.3 )  | 4.444    | ( 2.469 )       | 1550              | ( 2330.3 ) | 5.563    | ( 3.090 )       |
| 700               | ( 800.3 )  | 4.599    | ( 2.555 )       | 1600              | ( 2420.3 ) | 5.563    | ( 3.090 )       |
| 750               | ( 890.3 )  | 4.734    | ( 2.630 )       | 1650              | ( 2510.3 ) | 5.563    | ( 3.090 )       |
| 800               | ( 980.3 )  | 4.850    | ( 2.695 )       | 1700              | ( 2600.3 ) | 5.563    | ( 3.090 )       |
| 900               | ( 1160.3 ) | 5.040    | ( 2.800 )       | 1750              | ( 2690.3 ) | 5.563    | ( 3.090 )       |
| 1000              | ( 1340.3 ) | 5.193    | ( 2.885 )       | 1800              | ( 2780.3 ) | 5.563    | ( 3.090 )       |
| 1100              | ( 1520.3 ) | 5.332    | ( 2.962 )       |                   |            |          |                 |

**Application Notes:** Data for thermal expansion is collected from reference [1] and fitted using the equation below to approximate property trend with respect to temperature. CTE is assumed to be constant for T>1273K based on Snead et al. [2]. For this range, α(1250K) is used.

## Fit Equation:

$$\alpha(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\alpha(T)$  = Thermal Expansion Coefficient [μm/(m · K)]

T = Temperature [K]

## Constants:

T Range [K]:  $293 \leq T < 1273$        $1273 \leq T \leq 1800$   
A0 = -0.7765       $\alpha(T) = \alpha(1250) = 5.563$   
A1 = 14.35  
A2 = -12.209  
A3 = 3.8289



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

Revision 0: 08-02-2021

Specific Heat with Temperature

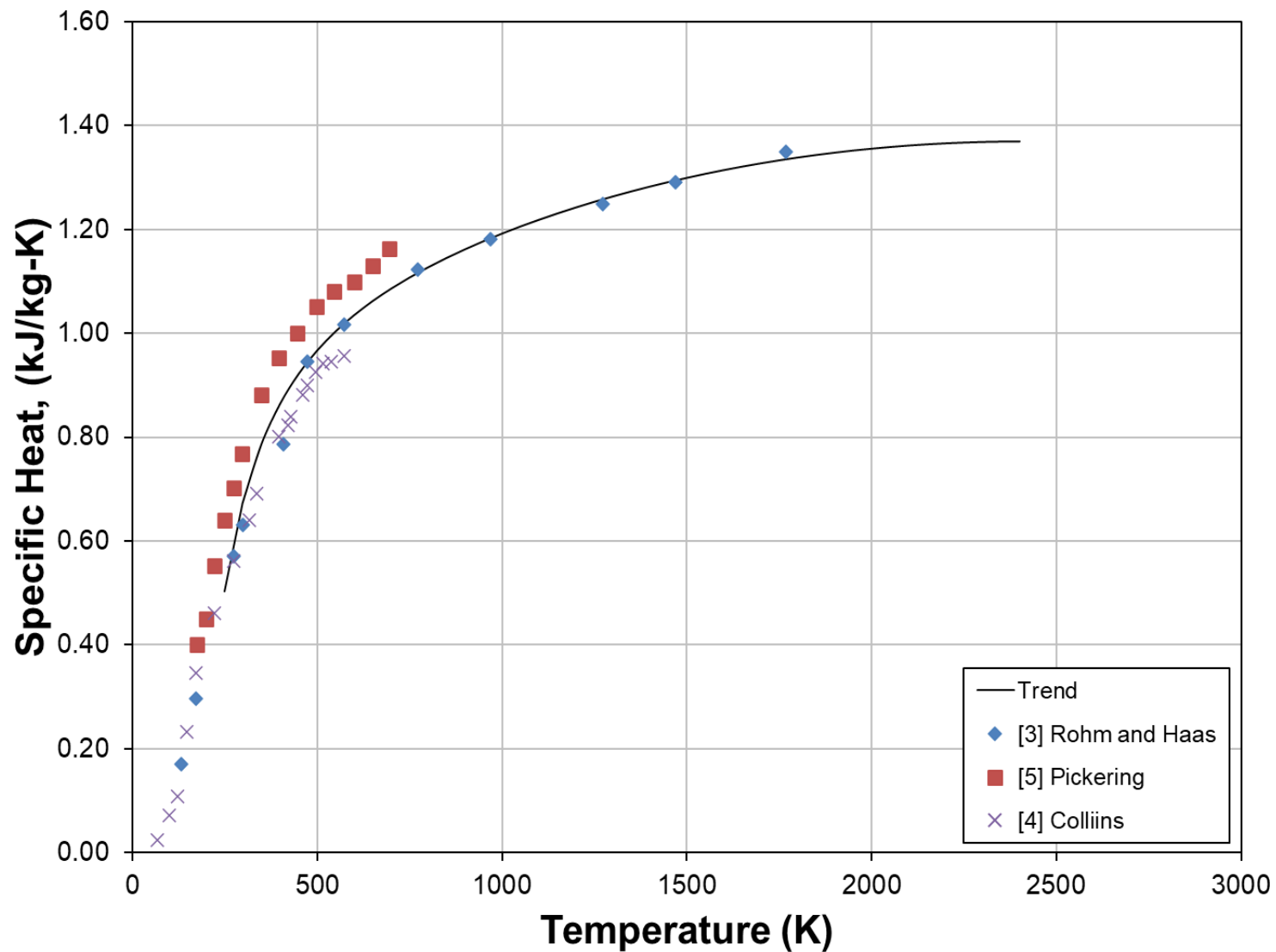


Figure 8.1.1-3: Specific Heat versus Temperature of SiC/SiC composite assuming behavior is the same as monolithic SiC. Trend line calculated from various test data [1]



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

Revision 2: 04-26-2023

Specific Heat with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) | K                 | ( °F )     | J/(g·K)                          | ( BTU/(lb·°F) ) |
| 250               | ( -9.7 )   | 0.504                            | ( 0.120 )       | 1000              | ( 1340.3 ) | 1.192                            | ( 0.285 )       |
| 300               | ( 80.3 )   | 0.677                            | ( 0.162 )       | 1100              | ( 1520.3 ) | 1.218                            | ( 0.291 )       |
| 350               | ( 170.3 )  | 0.787                            | ( 0.188 )       | 1200              | ( 1700.3 ) | 1.242                            | ( 0.297 )       |
| 400               | ( 260.3 )  | 0.864                            | ( 0.207 )       | 1300              | ( 1880.3 ) | 1.263                            | ( 0.302 )       |
| 450               | ( 350.3 )  | 0.922                            | ( 0.220 )       | 1400              | ( 2060.3 ) | 1.282                            | ( 0.306 )       |
| 500               | ( 440.3 )  | 0.967                            | ( 0.231 )       | 1500              | ( 2240.3 ) | 1.299                            | ( 0.310 )       |
| 550               | ( 530.3 )  | 1.004                            | ( 0.240 )       | 1600              | ( 2420.3 ) | 1.314                            | ( 0.314 )       |
| 600               | ( 620.3 )  | 1.035                            | ( 0.247 )       | 1700              | ( 2600.3 ) | 1.327                            | ( 0.317 )       |
| 650               | ( 710.3 )  | 1.062                            | ( 0.254 )       | 1800              | ( 2780.3 ) | 1.338                            | ( 0.320 )       |
| 700               | ( 800.3 )  | 1.086                            | ( 0.259 )       | 1900              | ( 2960.3 ) | 1.347                            | ( 0.322 )       |
| 750               | ( 890.3 )  | 1.107                            | ( 0.265 )       | 2000              | ( 3140.3 ) | 1.355                            | ( 0.324 )       |
| 800               | ( 980.3 )  | 1.127                            | ( 0.269 )       | 2100              | ( 3320.3 ) | 1.361                            | ( 0.325 )       |
| 850               | ( 1070.3 ) | 1.145                            | ( 0.274 )       | 2200              | ( 3500.3 ) | 1.365                            | ( 0.326 )       |
| 900               | ( 1160.3 ) | 1.161                            | ( 0.278 )       | 2300              | ( 3680.3 ) | 1.368                            | ( 0.327 )       |
| 950               | ( 1250.3 ) | 1.177                            | ( 0.281 )       | 2400              | ( 3860.3 ) | 1.369                            | ( 0.327 )       |

**Application Notes:** Specific heat values for monolithic SiC were compiled and summarized in reference [3-5] and fit using the equation below to approximate property trend as a function of temperature. Based on findings from reference [2], SiC/SiC composite specific heat trend with temperature is assumed to closely follow monolithic SiC.

## Fit Equation:

$$C_p(T) = A_0 + A_1 \cdot \left(\frac{T}{1000}\right) + A_2 \cdot \left(\frac{T}{1000}\right)^2 + A_{_2}/\left(\frac{T}{1000}\right)^2$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

T. Range [K]: 250 ≤ T < 2400

A<sub>0</sub> = 0.92565

A<sub>1</sub> = 0.3772

A<sub>2</sub> = -0.079259

A<sub>\_2</sub> = -0.031946



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

Revision 0: 08-02-2021

Young's Modulus with Temperature

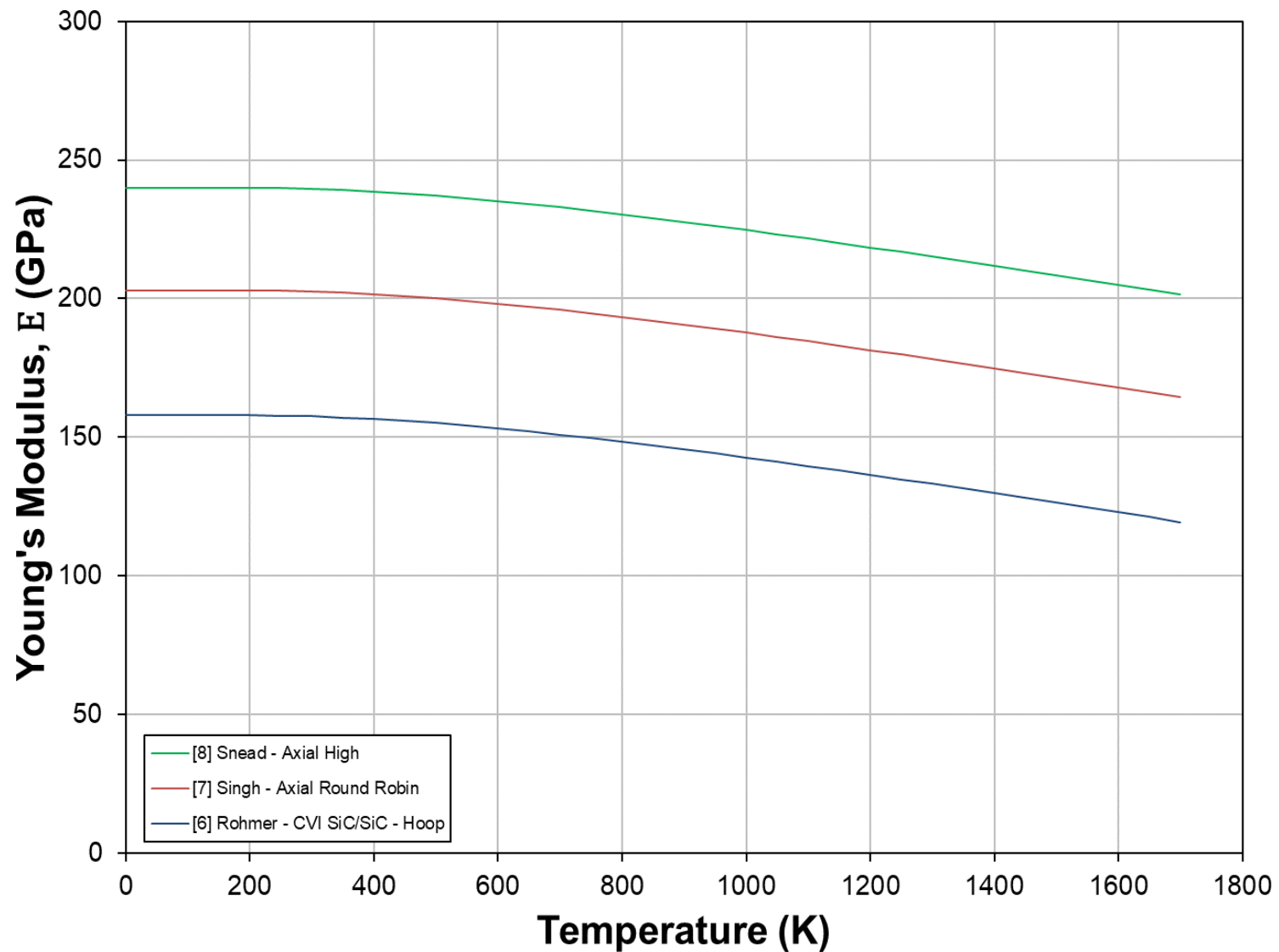


Figure 8.1.1-4: Elastic Modulus versus Temperature for SiC/SiC composite.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

Revision 0: 08-02-2021

Young's Modulus with Temperature

100% Theoretical Density

| Temperature ( T ) |            | Young's Modulus ( E ) |           | Temperature ( T ) |            | Young's Modulus ( E ) |           |
|-------------------|------------|-----------------------|-----------|-------------------|------------|-----------------------|-----------|
| K                 | ( °F )     | GPa                   | ( Msi )   | K                 | ( °F )     | GPa                   | ( Msi )   |
| 300               | ( 80.3 )   | 203.51                | ( 29.53 ) | 1100              | ( 1520.3 ) | 185.65                | ( 26.94 ) |
| 350               | ( 170.3 )  | 203.10                | ( 29.47 ) | 1150              | ( 1610.3 ) | 184.07                | ( 26.71 ) |
| 400               | ( 260.3 )  | 202.56                | ( 29.39 ) | 1200              | ( 1700.3 ) | 182.47                | ( 26.48 ) |
| 450               | ( 350.3 )  | 201.88                | ( 29.29 ) | 1250              | ( 1790.3 ) | 180.84                | ( 26.24 ) |
| 500               | ( 440.3 )  | 201.08                | ( 29.18 ) | 1300              | ( 1880.3 ) | 179.19                | ( 26.00 ) |
| 550               | ( 530.3 )  | 200.17                | ( 29.05 ) | 1350              | ( 1970.3 ) | 177.52                | ( 25.76 ) |
| 600               | ( 620.3 )  | 199.17                | ( 28.90 ) | 1400              | ( 2060.3 ) | 175.83                | ( 25.51 ) |
| 650               | ( 710.3 )  | 198.08                | ( 28.74 ) | 1450              | ( 2150.3 ) | 174.13                | ( 25.27 ) |
| 700               | ( 800.3 )  | 196.92                | ( 28.57 ) | 1500              | ( 2240.3 ) | 172.40                | ( 25.02 ) |
| 750               | ( 890.3 )  | 195.68                | ( 28.39 ) | 1550              | ( 2330.3 ) | 170.67                | ( 24.76 ) |
| 800               | ( 980.3 )  | 194.39                | ( 28.21 ) | 1600              | ( 2420.3 ) | 168.92                | ( 24.51 ) |
| 850               | ( 1070.3 ) | 193.04                | ( 28.01 ) | 1650              | ( 2510.3 ) | 167.16                | ( 24.25 ) |
| 900               | ( 1160.3 ) | 191.64                | ( 27.81 ) | 1700              | ( 2600.3 ) | 165.39                | ( 24.00 ) |
| 950               | ( 1250.3 ) | 190.20                | ( 27.60 ) | 1750              | ( 2690.3 ) | 163.60                | ( 23.74 ) |
| 1000              | ( 1340.3 ) | 188.71                | ( 27.38 ) | 1800              | ( 2780.3 ) | 161.81                | ( 23.48 ) |
| 1050              | ( 1430.3 ) | 187.20                | ( 27.16 ) |                   |            |                       |           |

**Application Notes:** Temperature dependence of SiC/SiC composite is assumed to follow the same trend as monolithic SiC [1] and shown in the equation below. Trend data generated using room temperature composite modulus values from [6-8].

## Fit Equation:

$$E(T) = E_0 - B \cdot T \exp\left(-\frac{T_0}{T}\right)$$

$E(T)$  = Modulus of Elasticity [GPa]

$E_0$  = Room Temperature Modulus

$T$  = Temperature [K]

## Constants:

T. Range [K]:  $0 \leq T \leq 1800$

$E_0$  = 204 GPa

$T_0$  = 962K

$B$  = 0.04 GPa/K



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

Revision 0: 08-02-2021

Additional Application Notes

### Thermal Conductivity Application Notes:

Through-thickness thermal conductivity for SiC/SiC composites can vary greatly and is dependent on fiber, interface and matrix. Room temperature ranges are:

- 8.5-18.1 W/m K for Hi Nicalon S / CVI-SiC composites with PyC or multi-layer PyC/SiC interfaces.
- 15.2-23.7 W/m K for Tyranno SA3 / CVI-SiC composites with PyC or multi-layer PyC/SiC interfaces.
- Even for composites with similar fiber, matrix and architecture, through-thickness thermal conductivity values can vary by as much as 40%. It is likely that the interface porosity, not total porosity is the main influence.
- For amorphous SiC/SiC composites, very little data exists regarding thermal conductivity as a function of temperature. Values ranging from 1.26-1.64 W/m K for polymer impregnation and pyrolysis (PIP) SiC/SiC with a PyC or multi-layer PyC/SiC interface have been reported by Zhao [9].

### Specific Heat Application Notes:

According to Snead et al. [7], heat capacity for crystalline SiC/SiC composites is expected to be essentially the same as that for high-purity monolithic SiC. Data used is collected from various sources for monolithic SiC.

Very little data exists for temperature dependent specific heat for SiC/SiC composites.

### Thermal Expansion Application Notes:

Experimental results from Katoh [1] indicates that CTE for SA3 and HNLS composites exhibit the same linear thermal expansion behavior with no detectable differences between the CVI composites and high-purity CVD SiC.

### Elastic Modulus Application Notes:

Elastic Modulus data shown assumes the composite behavior follows the same trend as monolithic SiC. The formula uses room temperature values for Elastic Modulus as the starting point. Very little experimental data exists to support this assumption.



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

**Revision 0: 08-02-2021**

**Additional Application Notes**

### **Proportional and Ultimate Stress Application Notes:**

Substantial room temperature tensile testing of specimens cut from plate plates has been performed, while limited axial and hoop strength testing has been performed for tubes [7]. The fiber type, the fiber architecture, the fiber interface coating and the matrix microstructure (including porosity) all influence the strength characteristics for SiC/SiC composites, including the proportional limit stress (PLS) and strain, the ultimate tensile strength (UTS) and the strain at the UTS. Consequently, mechanical properties used for analyses must be obtained for specimens prepared from composite components fabricated using the same materials and process steps as those intended for the final components. Until such data are available, strength data provided should be considered as reference values only.

Although some limited data for temperature dependent proportional limit stress are available [10], values are greatly influenced by fiber architecture as well as interface type and thickness. Therefore, the same comments apply regarding the need for application specific material testing. Additionally, elevated temperature testing data available in the literature in most cases are from tests performed in air where oxidation drives the resulting reduction in strength.



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.1 Crystalline SiC/SiC

**Revision 0: 08-02-2021**

**References**

- [1] Y. Katoh, K. Ozawa, C. Shih, et al., Continuous SiC fiber, CVI SiC matrix composites for nuclear applications: Properties and irradiation effects, Journal of Nuclear Materials 448(1-3) (2014) 448-476.
- [2] L.L. Snead, T. Nozawa, Y. Katoh, et al., Handbook of SiC properties for fuel performance modeling, Journal of Nuclear Materials 371(1) (2007) 329-377.
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- [6] E. Rohmer, E. Martin, C. Lorrette, Mechanical properties of SiC/SiC braided tubes for fuel cladding, Journal of Nuclear Materials 453(1) (2014) 16-21.
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- [8] G. Singh, S.T. Gonczy, E. Lara-Curzio, Y. Katoh, Interlaboratory Round Robin Study on Axial Tensile Properties of Tubular SiC/SiC Specimens, Oak Ridge National Lab. (ORNL), Oak Ridge, TN (United States), 2018.
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- [10] L. Li, Temperature-dependent proportional limit stress of SiC/SiC fiber-reinforced ceramic-matrix composites, High Temperature Materials and Processes 39(1) (2020) 209-218.

## **8 Composites and Structures**

### **8.1 Carbide Based**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 0: 08-02-2021

General

### Room Temperature Properties

|                                           |            |                                          |
|-------------------------------------------|------------|------------------------------------------|
| Theoretical Density, [kg/m <sup>3</sup> ] | 6,700      |                                          |
| Specific Heat, [J/(g-K)]                  | 0.350      |                                          |
| Thermal Conductivity, [W/(m-K)]           | 40-60      |                                          |
| Linear expansion coefficient, [μm/(m-K)]  | 4.50       |                                          |
| Electrical resistivity, [μΩ-m]            | 0.61       |                                          |
| Young's Modulus, [GPa]                    | 200-360    | Dependent on Carbon content and Porosity |
| Shear Modulus, [GPa]                      | 80-140     |                                          |
| Modulus of Rupture [MPa]                  | 100-130    |                                          |
| Ultimate Tensile Strength [MPa]           | 77-120     |                                          |
| Poisson's Ratio, [-]                      | 0.165-0.21 |                                          |



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 0: 08-02-2021

Thermal Conductivity with Temperature

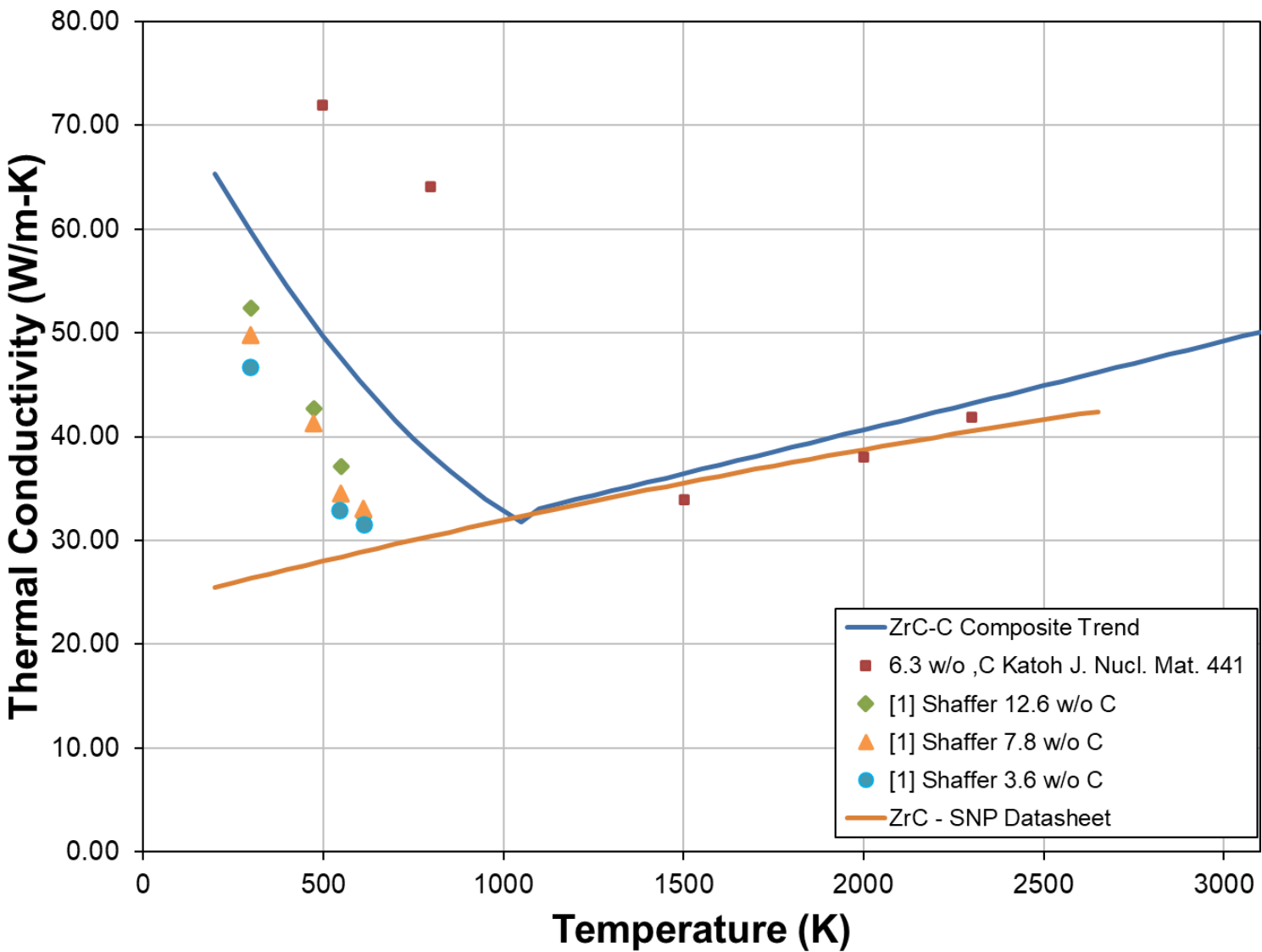


Figure 8.1.2-1: Thermal Conductivity vs. Temperature. Displaying fitted trend of data for best estimate values from Schaffer.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 0: 08-02-2021

Thermal Conductivity with Temperature

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       |
|-------------------|------------|----------------------------|---------------------------------------|-------------------|------------|----------------------------|---------------------------------------|
| K                 | ( °F )     | W/(m-K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) | K                 | ( °F )     | W/(m-K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) |
| 300               | ( 80.3 )   | 59.65                      | ( 413.84 )                            | 1100              | ( 1700.3 ) | 33.10                      | ( 229.62 )                            |
| 350               | ( 170.3 )  | 56.99                      | ( 395.42 )                            | 1200              | ( 1880.3 ) | 33.93                      | ( 235.41 )                            |
| 400               | ( 260.3 )  | 54.45                      | ( 377.79 )                            | 1300              | ( 2060.3 ) | 34.77                      | ( 241.22 )                            |
| 450               | ( 350.3 )  | 52.02                      | ( 360.95 )                            | 1400              | ( 2240.3 ) | 35.60                      | ( 247.03 )                            |
| 500               | ( 440.3 )  | 49.71                      | ( 344.91 )                            | 1500              | ( 2420.3 ) | 36.44                      | ( 252.85 )                            |
| 550               | ( 530.3 )  | 47.51                      | ( 329.65 )                            | 1600              | ( 2600.3 ) | 37.29                      | ( 258.69 )                            |
| 600               | ( 620.3 )  | 45.43                      | ( 315.18 )                            | 1700              | ( 2780.3 ) | 38.13                      | ( 264.53 )                            |
| 650               | ( 710.3 )  | 43.46                      | ( 301.50 )                            | 1800              | ( 2960.3 ) | 38.97                      | ( 270.39 )                            |
| 700               | ( 800.3 )  | 41.60                      | ( 288.62 )                            | 1900              | ( 3140.3 ) | 39.82                      | ( 276.25 )                            |
| 750               | ( 890.3 )  | 39.86                      | ( 276.52 )                            | 2000              | ( 3320.3 ) | 40.66                      | ( 282.13 )                            |
| 800               | ( 980.3 )  | 38.23                      | ( 265.22 )                            | 2100              | ( 3500.3 ) | 41.51                      | ( 288.02 )                            |
| 850               | ( 1070.3 ) | 36.71                      | ( 254.70 )                            | 2200              | ( 3680.3 ) | 42.36                      | ( 293.91 )                            |
| 900               | ( 1160.3 ) | 35.31                      | ( 244.98 )                            | 2300              | ( 3680.3 ) | 43.21                      | ( 299.82 )                            |
| 950               | ( 1250.3 ) | 34.02                      | ( 236.04 )                            | 2400              | ( 3860.3 ) | 44.07                      | ( 305.74 )                            |
| 1000              | ( 1340.3 ) | 32.85                      | ( 227.90 )                            | 2500              | ( 4040.3 ) | 44.92                      | ( 311.67 )                            |

**Application Notes:** Data for thermal conductivity is collected from reference [1] and fitted with the equation below to approximate property trend with respect to temperature.

## Fit Equation:

$$k(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]: 200 ≤ T ≤ 1100

A0 = 77.967

A1 = -67.901

A2 = 22.781

T Range [K]: 1100 ≤ T ≤ 3100

A0 = 24.012

A1 = 8.175

A2 = 0.07545



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 0: 08-02-2021

Coefficient of Thermal Expansion with Temperature

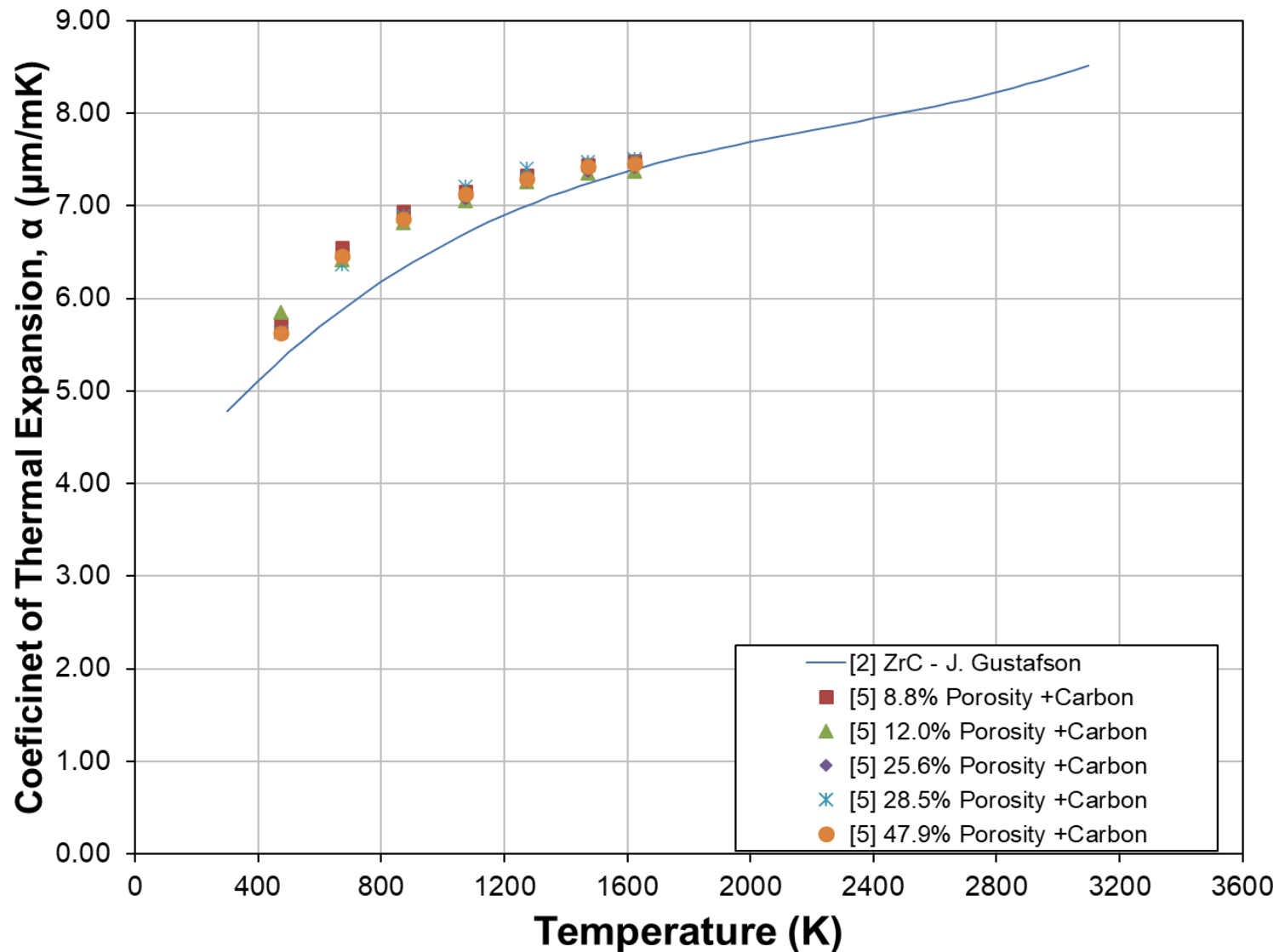


Figure 8.1.2-2: Thermal Expansion versus Temperature for ZrC-C composite based on dense ZrC data from Gustafson.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 0: 08-02-2021

Coefficient of Thermal Expansion with Temperature

| Temperature ( T ) |            | CTE      |                 | Temperature ( T ) |            | CTE      |                 |
|-------------------|------------|----------|-----------------|-------------------|------------|----------|-----------------|
| K                 | ( °F )     | μm/(m-K) | ( μin/(in-°F) ) | K                 | ( °F )     | μm/(m-K) | ( μin/(in-°F) ) |
| 300               | ( 80.3 )   | 4.781    | ( 2.656 )       | 1800              | ( 2780.3 ) | 7.545    | ( 4.192 )       |
| 400               | ( 260.3 )  | 5.112    | ( 2.840 )       | 1900              | ( 2960.3 ) | 7.620    | ( 4.234 )       |
| 500               | ( 440.3 )  | 5.415    | ( 3.008 )       | 2000              | ( 3140.3 ) | 7.690    | ( 4.272 )       |
| 600               | ( 620.3 )  | 5.692    | ( 3.162 )       | 2100              | ( 3320.3 ) | 7.756    | ( 4.309 )       |
| 700               | ( 800.3 )  | 5.946    | ( 3.303 )       | 2200              | ( 3500.3 ) | 7.819    | ( 4.344 )       |
| 800               | ( 980.3 )  | 6.176    | ( 3.431 )       | 2300              | ( 3680.3 ) | 7.882    | ( 4.379 )       |
| 900               | ( 1160.3 ) | 6.385    | ( 3.547 )       | 2400              | ( 3860.3 ) | 7.944    | ( 4.414 )       |
| 1000              | ( 1340.3 ) | 6.574    | ( 3.652 )       | 2500              | ( 4040.3 ) | 8.009    | ( 4.449 )       |
| 1100              | ( 1520.3 ) | 6.745    | ( 3.747 )       | 2600              | ( 4220.3 ) | 8.077    | ( 4.487 )       |
| 1200              | ( 1700.3 ) | 6.899    | ( 3.833 )       | 2700              | ( 4400.3 ) | 8.149    | ( 4.527 )       |
| 1300              | ( 1880.3 ) | 7.037    | ( 3.909 )       | 2800              | ( 4580.3 ) | 8.228    | ( 4.571 )       |
| 1400              | ( 2060.3 ) | 7.161    | ( 3.978 )       | 2900              | ( 4760.3 ) | 8.314    | ( 4.619 )       |
| 1500              | ( 2240.3 ) | 7.272    | ( 4.040 )       | 3000              | ( 4940.3 ) | 8.410    | ( 4.672 )       |
| 1600              | ( 2420.3 ) | 7.373    | ( 4.096 )       | 3100              | ( 5120.3 ) | 8.516    | ( 4.731 )       |
| 1700              | ( 2600.3 ) | 7.463    | ( 4.146 )       | 3200              | ( 5300.3 ) | 8.634    | ( 4.796 )       |

**Application Notes:** Since CTE is not significantly affected by porosity or additional carbon, data presented for thermal expansion is based on the fit equation below for dense, near stoichiometric ZrC [2].

## Fit Equation:

$$\alpha(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\alpha(T)$  = Thermal Expansion Coefficient [μm/(m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $0 \leq T \leq 3300$

A0 = 3.6129

A1 = 4.3674

A2 = -1.6475

A3 = 0.2416



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

|                             |                   |                                |
|-----------------------------|-------------------|--------------------------------|
| 8 Composites and Structures | 8.1 Carbide Based | 8.1.2 ZrC-C Composite          |
| Revision 0: 08-02-2021      |                   | Specific Heat with Temperature |

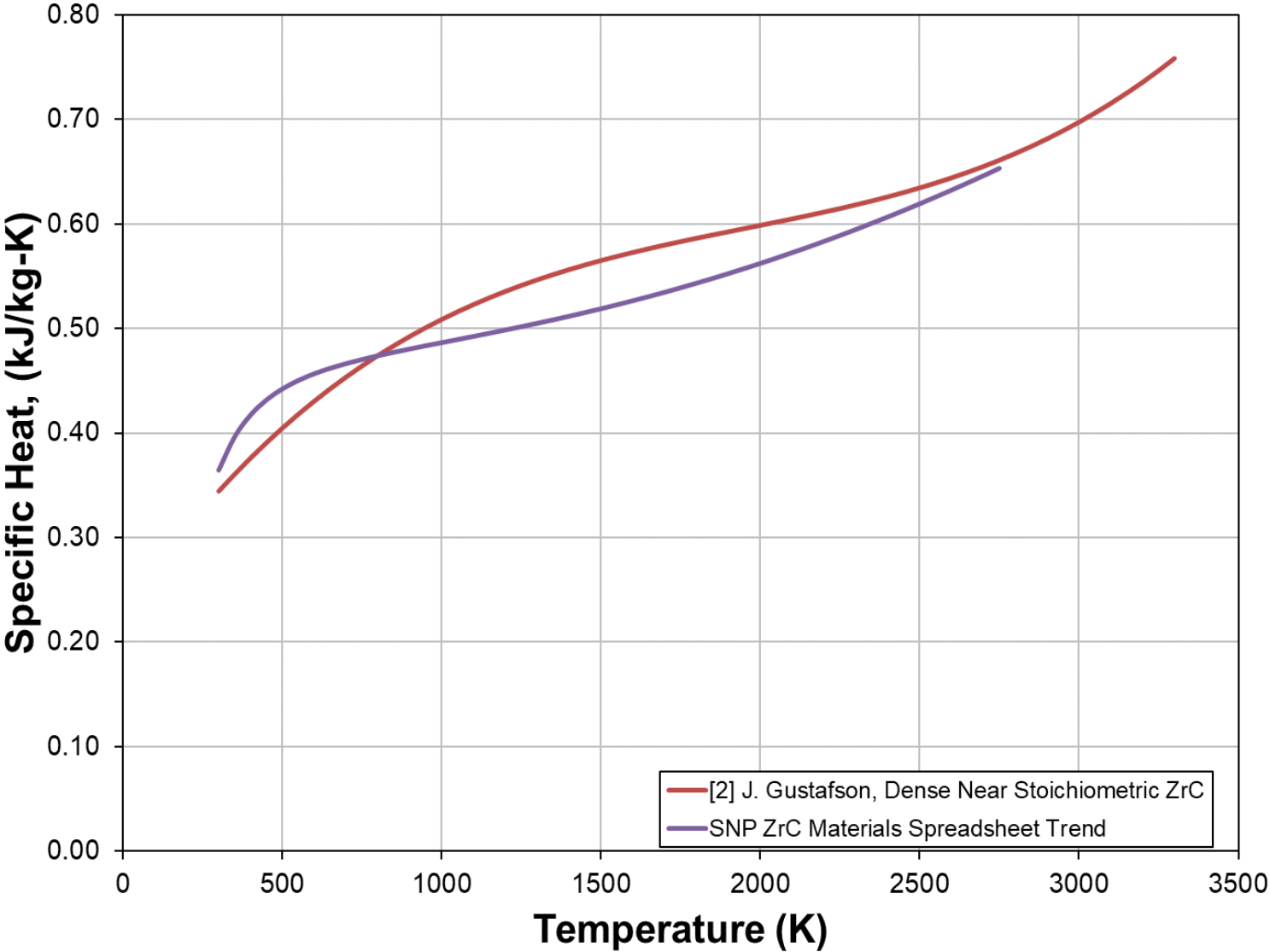


Figure 8.1.2-3: Estimated Specific Heat versus Temperature for ZrC-C composite based on dense, near stoichiometric ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 0: 08-02-2021

Specific Heat with Temperature

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g-K)                          | ( BTU/(lb-°F) ) | K                 | ( °F )     | J/(g-K)                          | ( BTU/(lb-°F) ) |
| 300               | ( 80.3 )   | 0.344                            | ( 0.082 )       | 1800              | ( 2780.3 ) | 0.586                            | ( 0.140 )       |
| 400               | ( 260.3 )  | 0.376                            | ( 0.090 )       | 1900              | ( 2960.3 ) | 0.593                            | ( 0.142 )       |
| 500               | ( 440.3 )  | 0.405                            | ( 0.097 )       | 2000              | ( 3140.3 ) | 0.599                            | ( 0.143 )       |
| 600               | ( 620.3 )  | 0.430                            | ( 0.103 )       | 2100              | ( 3320.3 ) | 0.605                            | ( 0.145 )       |
| 700               | ( 800.3 )  | 0.453                            | ( 0.108 )       | 2200              | ( 3500.3 ) | 0.611                            | ( 0.146 )       |
| 800               | ( 980.3 )  | 0.474                            | ( 0.113 )       | 2300              | ( 3680.3 ) | 0.618                            | ( 0.148 )       |
| 900               | ( 1160.3 ) | 0.492                            | ( 0.118 )       | 2400              | ( 3860.3 ) | 0.626                            | ( 0.150 )       |
| 1000              | ( 1340.3 ) | 0.509                            | ( 0.122 )       | 2500              | ( 4040.3 ) | 0.635                            | ( 0.152 )       |
| 1100              | ( 1520.3 ) | 0.523                            | ( 0.125 )       | 2600              | ( 4220.3 ) | 0.644                            | ( 0.154 )       |
| 1200              | ( 1700.3 ) | 0.536                            | ( 0.128 )       | 2700              | ( 4400.3 ) | 0.655                            | ( 0.157 )       |
| 1300              | ( 1880.3 ) | 0.547                            | ( 0.131 )       | 2800              | ( 4580.3 ) | 0.667                            | ( 0.160 )       |
| 1400              | ( 2060.3 ) | 0.556                            | ( 0.133 )       | 2900              | ( 4760.3 ) | 0.682                            | ( 0.163 )       |
| 1500              | ( 2240.3 ) | 0.565                            | ( 0.135 )       | 3000              | ( 4940.3 ) | 0.697                            | ( 0.167 )       |
| 1600              | ( 2420.3 ) | 0.573                            | ( 0.137 )       | 3100              | ( 5120.3 ) | 0.716                            | ( 0.171 )       |
| 1700              | ( 2600.3 ) | 0.580                            | ( 0.139 )       | 3200              | ( 5300.3 ) | 0.736                            | ( 0.176 )       |

**Application Notes:** Data for thermal conductivity is collected from reference [2] and fitted with the equation below for dense, near stoichiometric ZrC. Based on data from reference [3], specific heat is not significantly affected by additional carbon or porosity.

## Fit Equation:

$$C_p(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$C_p(T) = \text{Specific Heat [J/(g · K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

$$T \text{ Range [K]: } 0 \leq T \leq 3300$$

$$A0 = .22803$$

$$A1 = 0.4422$$

$$A2 = -0.1948$$

$$A3 = 0.03318$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 0: 08-02-2021

Young's Modulus with Temperature and Carbon + Porosity

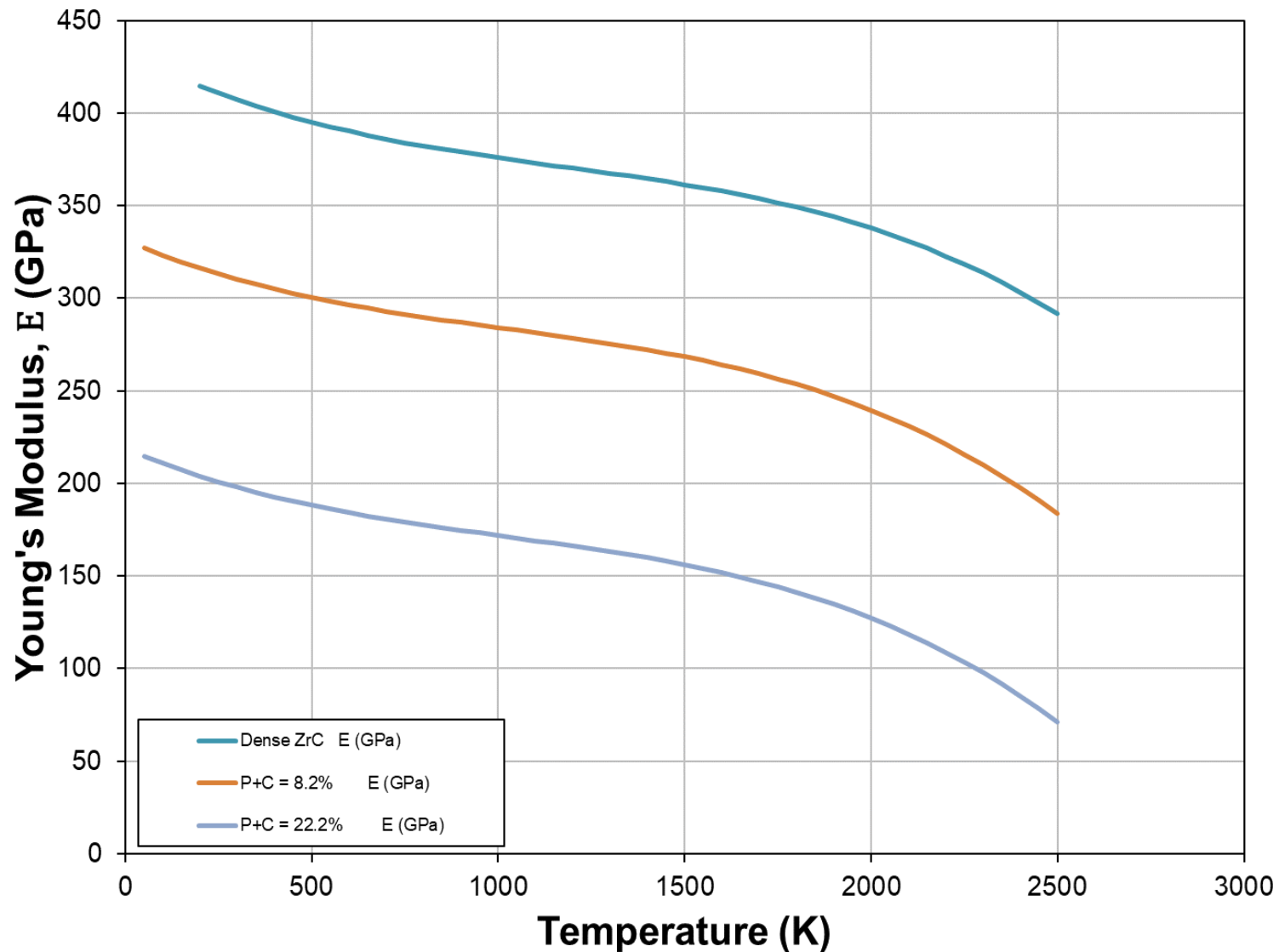


Figure 8.1.2-4:: : Estimated Elastic Modulus versus Temperature for ZrC-C Composite with ZrC shown for comparison.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 0: 08-02-2021

**Young's Modulus with Temperature and Carbon + Porosity**

| Temperature ( T ) |            | Dense ZrC | Young's Modulus ( E ) [GPa] |                   |
|-------------------|------------|-----------|-----------------------------|-------------------|
| K                 | ( °F )     |           | P+C = 8.2% E [GPa]          | P+C = 22% E [GPa] |
| 300               | ( 80.3 )   | 397.83    | 365.52                      | 247.58            |
| 500               | ( 440.3 )  | 388.02    | 356.50                      | 241.48            |
| 700               | ( 800.3 )  | 380.52    | 349.61                      | 236.81            |
| 900               | ( 1160.3 ) | 374.44    | 344.03                      | 233.03            |
| 1100              | ( 1520.3 ) | 368.90    | 338.95                      | 229.58            |
| 1300              | ( 1880.3 ) | 363.03    | 333.55                      | 225.93            |
| 1500              | ( 2240.3 ) | 355.92    | 327.02                      | 221.51            |
| 1700              | ( 2600.3 ) | 346.71    | 318.55                      | 215.77            |
| 1900              | ( 2960.3 ) | 334.50    | 307.34                      | 208.17            |
| 2100              | ( 3320.3 ) | 318.42    | 292.56                      | 198.16            |
| 2300              | ( 3680.3 ) | 297.57    | 273.41                      | 185.19            |
| 2500              | ( 4040.3 ) | 271.08    | 249.07                      | 168.70            |
| 2700              | ( 4400.3 ) | 238.06    | 218.73                      | 148.16            |
| 2900              | ( 4760.3 ) | 197.63    | 181.58                      | 122.99            |
| 3100              | ( 5120.3 ) | 148.90    | 136.81                      | 92.67             |
| 3300              | ( 5480.3 ) | 91.00     | 83.61                       | 56.63             |

**Application Notes:** Trend data based on empirical modulus relationship shown in the equation below from reference [4] and using temperature dependent modulus data for dense ZrC from reference [2]. Scaled values used for temperature dependent modulus with the addition of porosity and carbon.

## Fit Equation:

$$E(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$E(T)$  = Modulus of Elasticity [GPa]

$T$  = Temperature [K]

T Range [K]: 273 < T < 3300

A0 = 418.8

A1 = -85.2

A2 = 56.47

A3 = -18.41



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 0: 08-02-2021

RT Young's Modulus with % Carbon + Porosity

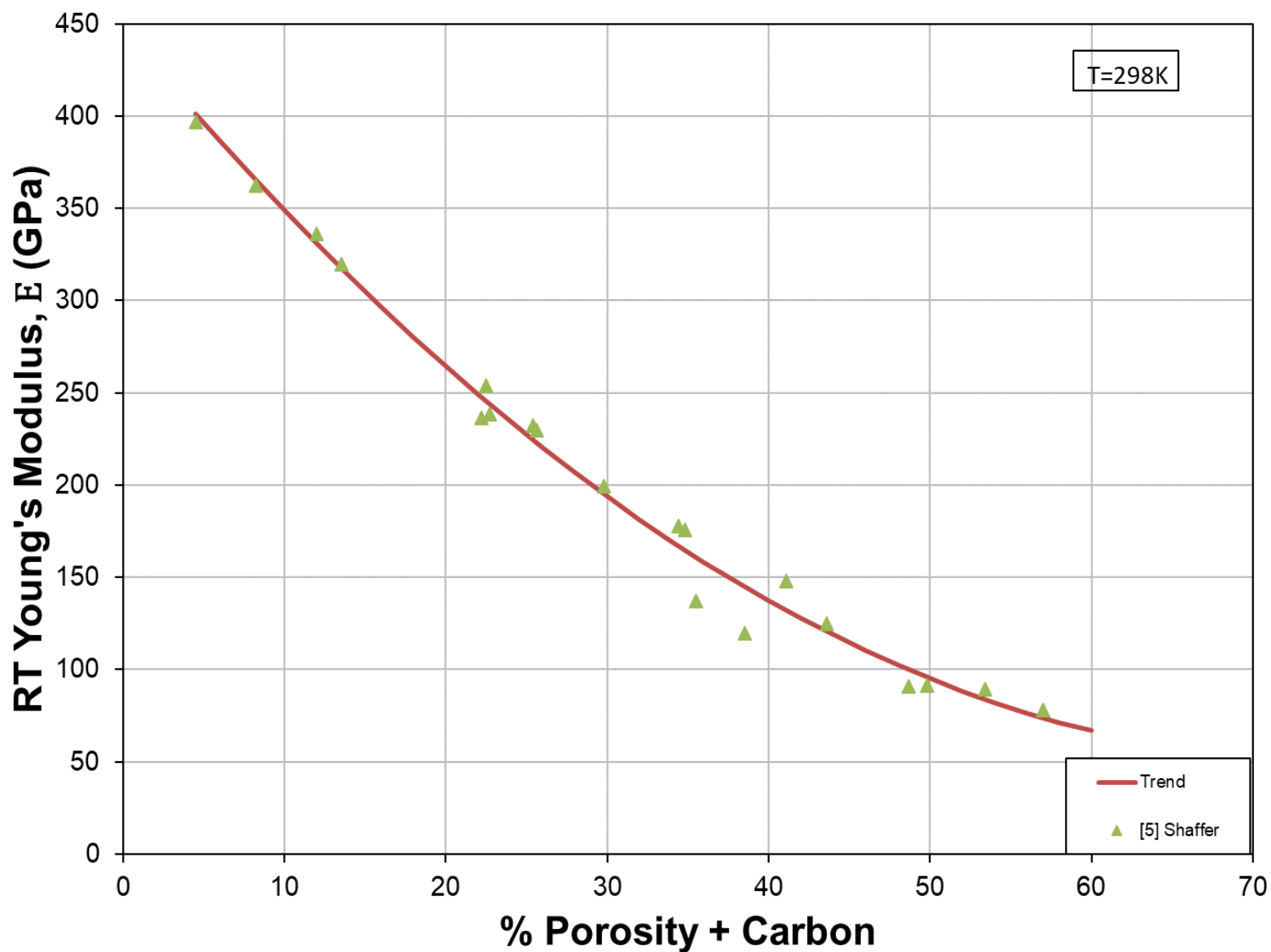


Figure 8.1.2-5: Estimated RT Elastic Modulus versus %Carbon and %Porosity for ZrC-C Composite.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

Revision 2: 04-26-2023

RT Young's Modulus with % Carbon + Porosity

| %Porosity + %Carbon | RT Modulus [GPa] | %Porosity + %Carbon | RT Modulus [GPa] |
|---------------------|------------------|---------------------|------------------|
| 4.500               | 401.29           | 26                  | 220.33           |
| 6.000               | 386.55           | 28                  | 206.80           |
| 8.000               | 367.40           | 30                  | 193.83           |
| 10.000              | 348.81           | 32                  | 181.43           |
| 12.000              | 330.78           | 34                  | 169.58           |
| 14.000              | 313.31           | 36                  | 158.30           |
| 16.000              | 296.41           | 38                  | 147.59           |
| 18.000              | 280.07           | 40                  | 137.43           |
| 20.000              | 264.29           | 42                  | 127.84           |
| 22.000              | 249.07           | 44                  | 118.81           |
| 24.000              | 234.42           | 46                  | 110.34           |

**Application Notes:** Data for Young's modulus is collected from [1, 5] and fitted using the equation below based on the assumption that porosity and excess carbon have similar effects on modulus based on findings from reference [4].

## Fit Equation:

$$E_p = A0 + A1 \cdot P + A2 \cdot P^2$$

$E_p$  = Porosity + Carbon Scaled Modulus of Elasticity [GPa]

$P$  = %Porosity + %Carbon

## Constants:

Porosity + Carbon Range:  $4.5\% \leq P \leq 60\%$

A0 = 447.39

A1 = -10.561

A2 = 0.0703



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.2 ZrC-C Composite

**Revision 0: 08-02-2021**

**References**

- [1] P. Schaffer, D. Hasselman, A. Chaerski, Factors Affecting Thermal Shock Resistance of Polyphase Ceramic Bodies - Part 2, The Carborundum Company, R&DD, 1962.
- [2] J. Gustafson, NASA GCD Feasibility NTP Material Property Handbook Development Status, 2019.
- [3] Y. Katoh, G. Vasudevamurthy, T. Nozawa, L.L. Snead, Properties of zirconium carbide for nuclear fuel applications, Journal of Nuclear Materials 441(1) (2013) 718-742.
- [4] V.M. Baranov, V.I. Knyazev, O.S. Korostin, The temperature dependence of the elastic constants of nonstoichiometric zirconium carbides, Strength of Materials 5(9) (1973) 1074-1077.
- [5] P. Schaffer, D. Hasselman, A. Chaerski, Factors Affecting Thermal Shock Resistance of Polyphase Ceramic Bodies - Part 1, The Carborundum Company, R&DD, 1961.

## **8 Composites and Structures**

### **8.1 Carbide Based**



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.3 High Porosity ZrC

Revision 2: 04-26-2023

General

### Room Temperature Properties

|                                           |         |              |
|-------------------------------------------|---------|--------------|
| Theoretical Density, [kg/m <sup>3</sup> ] | 1,300   | 80% porosity |
| Specific Heat, [J/(g-K)]                  | 0.340   |              |
| Thermal Conductivity, [W/(m-K)]           | 0.4-1.0 |              |
| Linear expansion coefficient, [μm/(m-K)]  | 4.5-5.0 |              |
| Young's Modulus, [GPa]                    | 55.0    |              |
| Compressive Yield Stress [MPa]            | 17      | 72% porosity |



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.3 High Porosity ZrC

Revision 0: 08-02-2021

Thermal Conductivity with Temperature

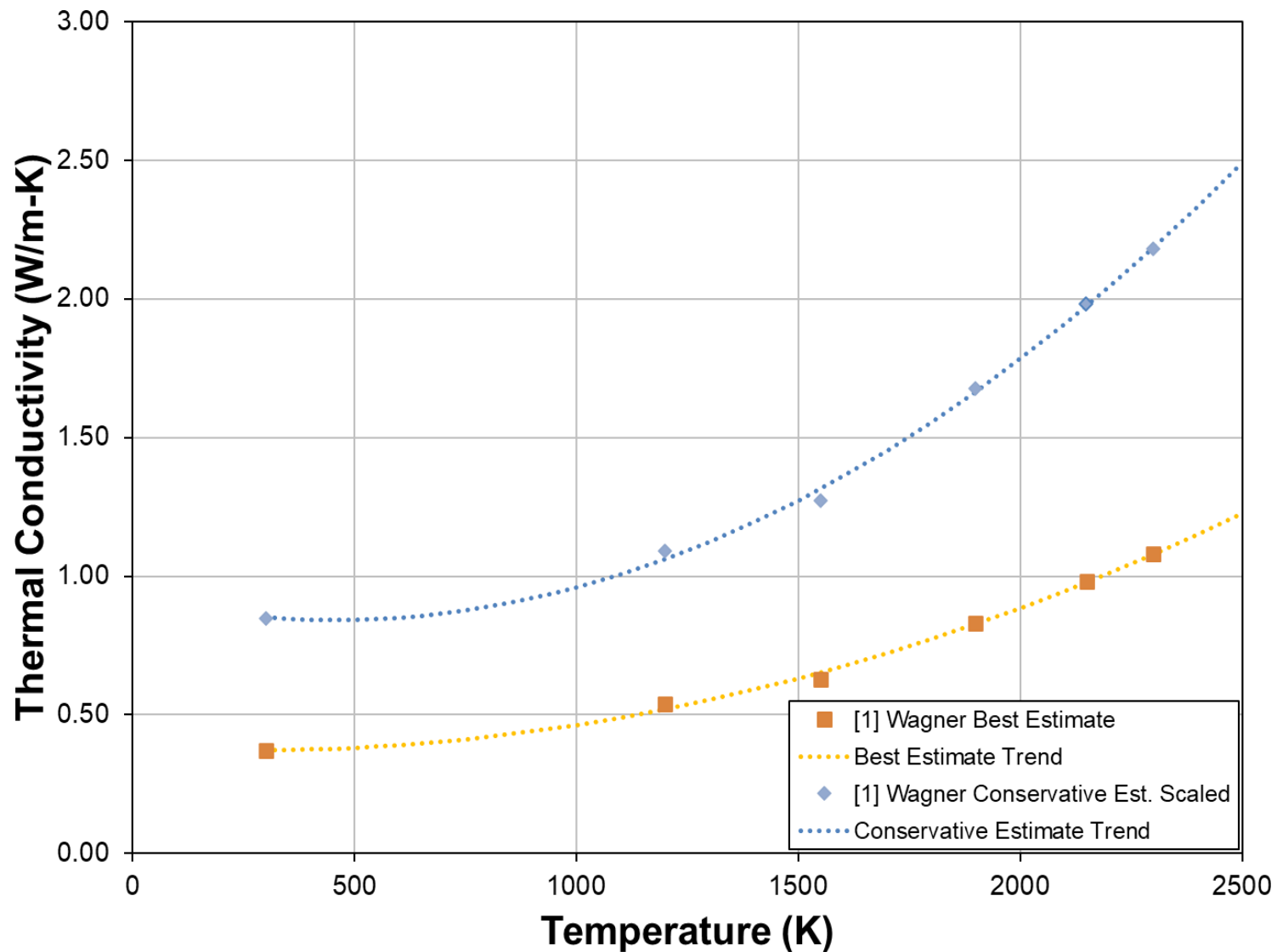


Figure 8.1.3-1: Thermal Conductivity vs. Temperature for high porosity ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.3 High Porosity ZrC

Revision 2: 04-26-2023

Thermal Conductivity with Temperature

| Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       | Temperature ( T ) |            | Thermal Conductivity ( k ) |                                       |
|-------------------|------------|----------------------------|---------------------------------------|-------------------|------------|----------------------------|---------------------------------------|
| K                 | ( °F )     | W/(m-K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) | K                 | ( °F )     | W/(m-K)                    | ((Btu-in.)/(ft. <sup>2</sup> -hr-°F)) |
| 300               | ( 80.3 )   | 0.37                       | ( 2.59 )                              | 1100              | ( 1700.3 ) | 0.49                       | ( 3.40 )                              |
| 350               | ( 170.3 )  | 0.37                       | ( 2.59 )                              | 1200              | ( 1880.3 ) | 0.52                       | ( 3.61 )                              |
| 400               | ( 260.3 )  | 0.38                       | ( 2.61 )                              | 1300              | ( 2060.3 ) | 0.55                       | ( 3.84 )                              |
| 450               | ( 350.3 )  | 0.38                       | ( 2.62 )                              | 1400              | ( 2240.3 ) | 0.59                       | ( 4.10 )                              |
| 500               | ( 440.3 )  | 0.38                       | ( 2.65 )                              | 1500              | ( 2420.3 ) | 0.63                       | ( 4.38 )                              |
| 550               | ( 530.3 )  | 0.39                       | ( 2.68 )                              | 1600              | ( 2600.3 ) | 0.68                       | ( 4.69 )                              |
| 600               | ( 620.3 )  | 0.39                       | ( 2.71 )                              | 1700              | ( 2780.3 ) | 0.72                       | ( 5.01 )                              |
| 650               | ( 710.3 )  | 0.40                       | ( 2.76 )                              | 1800              | ( 2960.3 ) | 0.77                       | ( 5.37 )                              |
| 700               | ( 800.3 )  | 0.40                       | ( 2.80 )                              | 1900              | ( 3140.3 ) | 0.83                       | ( 5.74 )                              |
| 750               | ( 890.3 )  | 0.41                       | ( 2.86 )                              | 2000              | ( 3320.3 ) | 0.89                       | ( 6.14 )                              |
| 800               | ( 980.3 )  | 0.42                       | ( 2.92 )                              | 2100              | ( 3500.3 ) | 0.95                       | ( 6.56 )                              |
| 850               | ( 1070.3 ) | 0.43                       | ( 2.98 )                              | 2200              | ( 3680.3 ) | 1.01                       | ( 7.01 )                              |
| 900               | ( 1160.3 ) | 0.44                       | ( 3.06 )                              | 2300              | ( 3680.3 ) | 1.08                       | ( 7.48 )                              |
| 950               | ( 1250.3 ) | 0.45                       | ( 3.13 )                              | 2400              | ( 3860.3 ) | 1.15                       | ( 7.98 )                              |
| 1000              | ( 1340.3 ) | 0.46                       | ( 3.22 )                              | 2500              | ( 4040.3 ) | 1.22                       | ( 8.50 )                              |

**Application Notes:** Data for thermal conductivity is collected from reference [1] and fitted with the equation below to approximate property trend with respect to temperature. Table data represents best estimate data for ZrC with a porosity of 72% and C/Zr ratio of 0.94.

## Fit Equation:

$$k(T) = A0 + A1 \cdot \left( \frac{T}{1000} \right) + A2 \cdot \left( \frac{T}{1000} \right)^2$$

$k(T)$  = Thermal Conductivity [W / (m · K)]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $300 \leq T \leq 2500$

A0 = 0.38532

A1 = -0.093246

A2 = 0.017156



SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

|                             |                   |                                                   |
|-----------------------------|-------------------|---------------------------------------------------|
| 8 Composites and Structures | 8.1 Carbide Based | 8.1.3 High Porosity ZrC                           |
| Revision 0: 08-02-2021      |                   | Coefficient of Thermal Expansion with Temperature |

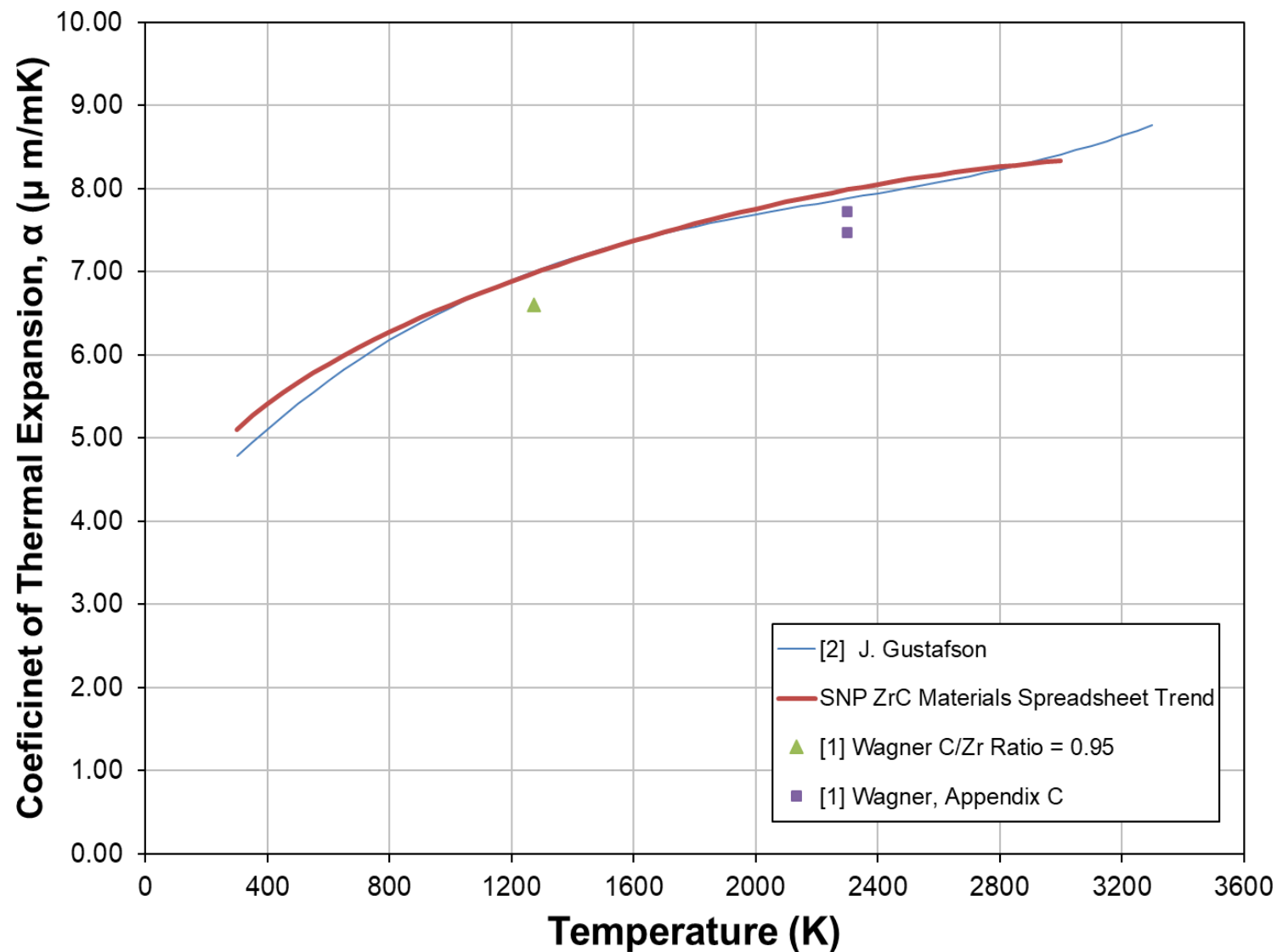


Figure 8.1.3-2: Thermal Expansion versus Temperature for porous ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.3 High Porosity ZrC

Revision 0: 08-02-2021

Coefficient of Thermal Expansion with Temperature

| Temperature ( T ) |            | CTE      |                 | Temperature ( T ) |            | CTE      |                 |
|-------------------|------------|----------|-----------------|-------------------|------------|----------|-----------------|
| K                 | ( °F )     | μm/(m-K) | ( μin/(in-°F) ) | K                 | ( °F )     | μm/(m-K) | ( μin/(in-°F) ) |
| 300               | ( 80.3 )   | 4.781    | ( 2.656 )       | 1800              | ( 2780.3 ) | 7.545    | ( 4.192 )       |
| 400               | ( 260.3 )  | 5.112    | ( 2.840 )       | 1900              | ( 2960.3 ) | 7.620    | ( 4.234 )       |
| 500               | ( 440.3 )  | 5.415    | ( 3.008 )       | 2000              | ( 3140.3 ) | 7.690    | ( 4.272 )       |
| 600               | ( 620.3 )  | 5.692    | ( 3.162 )       | 2100              | ( 3320.3 ) | 7.756    | ( 4.309 )       |
| 700               | ( 800.3 )  | 5.946    | ( 3.303 )       | 2200              | ( 3500.3 ) | 7.819    | ( 4.344 )       |
| 800               | ( 980.3 )  | 6.176    | ( 3.431 )       | 2300              | ( 3680.3 ) | 7.882    | ( 4.379 )       |
| 900               | ( 1160.3 ) | 6.385    | ( 3.547 )       | 2400              | ( 3860.3 ) | 7.944    | ( 4.414 )       |
| 1000              | ( 1340.3 ) | 6.574    | ( 3.652 )       | 2500              | ( 4040.3 ) | 8.009    | ( 4.449 )       |
| 1100              | ( 1520.3 ) | 6.745    | ( 3.747 )       | 2600              | ( 4220.3 ) | 8.077    | ( 4.487 )       |
| 1200              | ( 1700.3 ) | 6.899    | ( 3.833 )       | 2700              | ( 4400.3 ) | 8.149    | ( 4.527 )       |
| 1300              | ( 1880.3 ) | 7.037    | ( 3.909 )       | 2800              | ( 4580.3 ) | 8.228    | ( 4.571 )       |
| 1400              | ( 2060.3 ) | 7.161    | ( 3.978 )       | 2900              | ( 4760.3 ) | 8.314    | ( 4.619 )       |
| 1500              | ( 2240.3 ) | 7.272    | ( 4.040 )       | 3000              | ( 4940.3 ) | 8.410    | ( 4.672 )       |
| 1600              | ( 2420.3 ) | 7.373    | ( 4.096 )       | 3100              | ( 5120.3 ) | 8.516    | ( 4.731 )       |
| 1700              | ( 2600.3 ) | 7.463    | ( 4.146 )       | 3200              | ( 5300.3 ) | 8.634    | ( 4.796 )       |

**Application Notes:** Since CTE is not significantly affected by porosity, data presented for thermal expansion is based on the fit equation below for dense, near stoichiometric ZrC [2]. CTE is also assumed to be weakly influenced by C/Zr ratio from reference [3].

## Fit Equation:

$$\alpha(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$\alpha(T)$  = Thermal Expansion Coefficient [ $\mu\text{m}/(\text{m} \cdot \text{K})$ ]

$T$  = Temperature [K]

## Constants:

T Range [K]:  $0 \leq T \leq 3300$

A0 = 3.6129

A1 = 4.3674

A2 = - 1.6475

A3 = 0.2416



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

|                             |                   |                                |
|-----------------------------|-------------------|--------------------------------|
| 8 Composites and Structures | 8.1 Carbide Based | 8.1.3 High Porosity ZrC        |
| Revision 0: 08-02-2021      |                   | Specific Heat with Temperature |

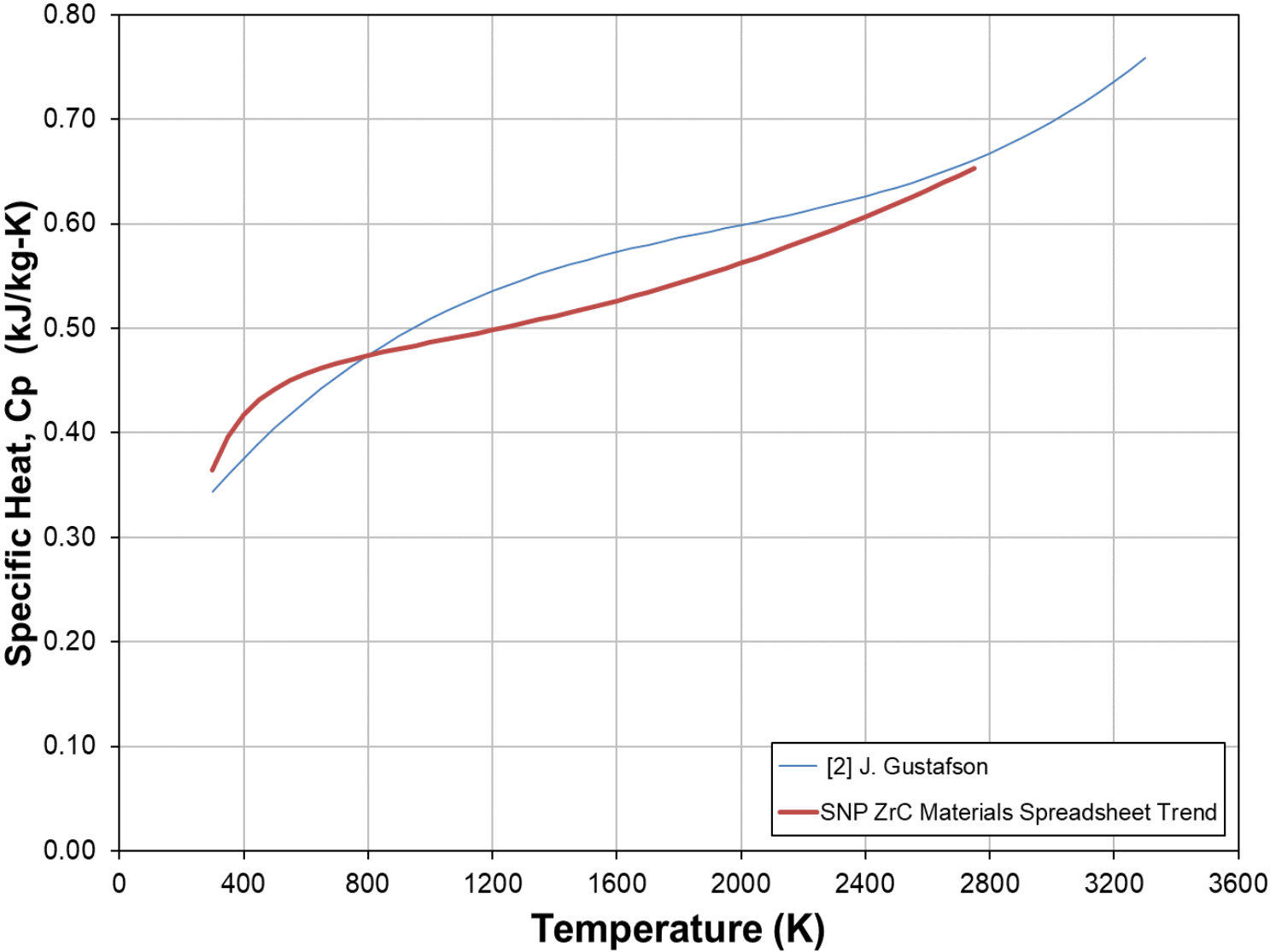


Figure 8.1.3-3: Estimated Specific Heat versus Temperature of porous ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.3 High Porosity ZrC

Revision 0: 08-02-2021

Specific Heat with Temperature

| Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 | Temperature ( T ) |            | Specific Heat ( C <sub>p</sub> ) |                 |
|-------------------|------------|----------------------------------|-----------------|-------------------|------------|----------------------------------|-----------------|
| K                 | ( °F )     | J/(g-K)                          | ( BTU/(lb-°F) ) | K                 | ( °F )     | J/(g-K)                          | ( BTU/(lb-°F) ) |
| 300               | ( 80.3 )   | 0.344                            | ( 0.082 )       | 1800              | ( 2780.3 ) | 0.586                            | ( 0.140 )       |
| 400               | ( 260.3 )  | 0.376                            | ( 0.090 )       | 1900              | ( 2960.3 ) | 0.593                            | ( 0.142 )       |
| 500               | ( 440.3 )  | 0.405                            | ( 0.097 )       | 2000              | ( 3140.3 ) | 0.599                            | ( 0.143 )       |
| 600               | ( 620.3 )  | 0.430                            | ( 0.103 )       | 2100              | ( 3320.3 ) | 0.605                            | ( 0.145 )       |
| 700               | ( 800.3 )  | 0.453                            | ( 0.108 )       | 2200              | ( 3500.3 ) | 0.611                            | ( 0.146 )       |
| 800               | ( 980.3 )  | 0.474                            | ( 0.113 )       | 2300              | ( 3680.3 ) | 0.618                            | ( 0.148 )       |
| 900               | ( 1160.3 ) | 0.492                            | ( 0.118 )       | 2400              | ( 3860.3 ) | 0.626                            | ( 0.150 )       |
| 1000              | ( 1340.3 ) | 0.509                            | ( 0.122 )       | 2500              | ( 4040.3 ) | 0.635                            | ( 0.152 )       |
| 1100              | ( 1520.3 ) | 0.523                            | ( 0.125 )       | 2600              | ( 4220.3 ) | 0.644                            | ( 0.154 )       |
| 1200              | ( 1700.3 ) | 0.536                            | ( 0.128 )       | 2700              | ( 4400.3 ) | 0.655                            | ( 0.157 )       |
| 1300              | ( 1880.3 ) | 0.547                            | ( 0.131 )       | 2800              | ( 4580.3 ) | 0.667                            | ( 0.160 )       |
| 1400              | ( 2060.3 ) | 0.556                            | ( 0.133 )       | 2900              | ( 4760.3 ) | 0.682                            | ( 0.163 )       |
| 1500              | ( 2240.3 ) | 0.565                            | ( 0.135 )       | 3000              | ( 4940.3 ) | 0.697                            | ( 0.167 )       |
| 1600              | ( 2420.3 ) | 0.573                            | ( 0.137 )       | 3100              | ( 5120.3 ) | 0.716                            | ( 0.171 )       |
| 1700              | ( 2600.3 ) | 0.580                            | ( 0.139 )       | 3200              | ( 5300.3 ) | 0.736                            | ( 0.176 )       |

**Application Notes:** The table data is a fit of equation below for dense, near stoichiometric ZrC from reference [2]. Based on data from reference [3], specific heat decreases as C/Zr ratio decreases, however, the reduction is small for materials at near stoichiometric ratios.

## Fit Equation:

$$C_p(T) = A0 + A1 \cdot \left(\frac{T}{1000}\right) + A2 \cdot \left(\frac{T}{1000}\right)^2 + A3 \cdot \left(\frac{T}{1000}\right)^3$$

$$C_p(T) = \text{Specific Heat [J/(g} \cdot \text{K)]}$$

$$T = \text{Temperature [K]}$$

## Constants:

$$T \text{ Range [K]: } 0 \leq T \leq 3300$$

$$A0 = .22803$$

$$A1 = 0.4422$$

$$A2 = -0.1948$$

$$A3 = 0.03318$$



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.3 High Porosity ZrC

Revision 0: 08-02-2021

Young's Modulus with Temperature and Porosity

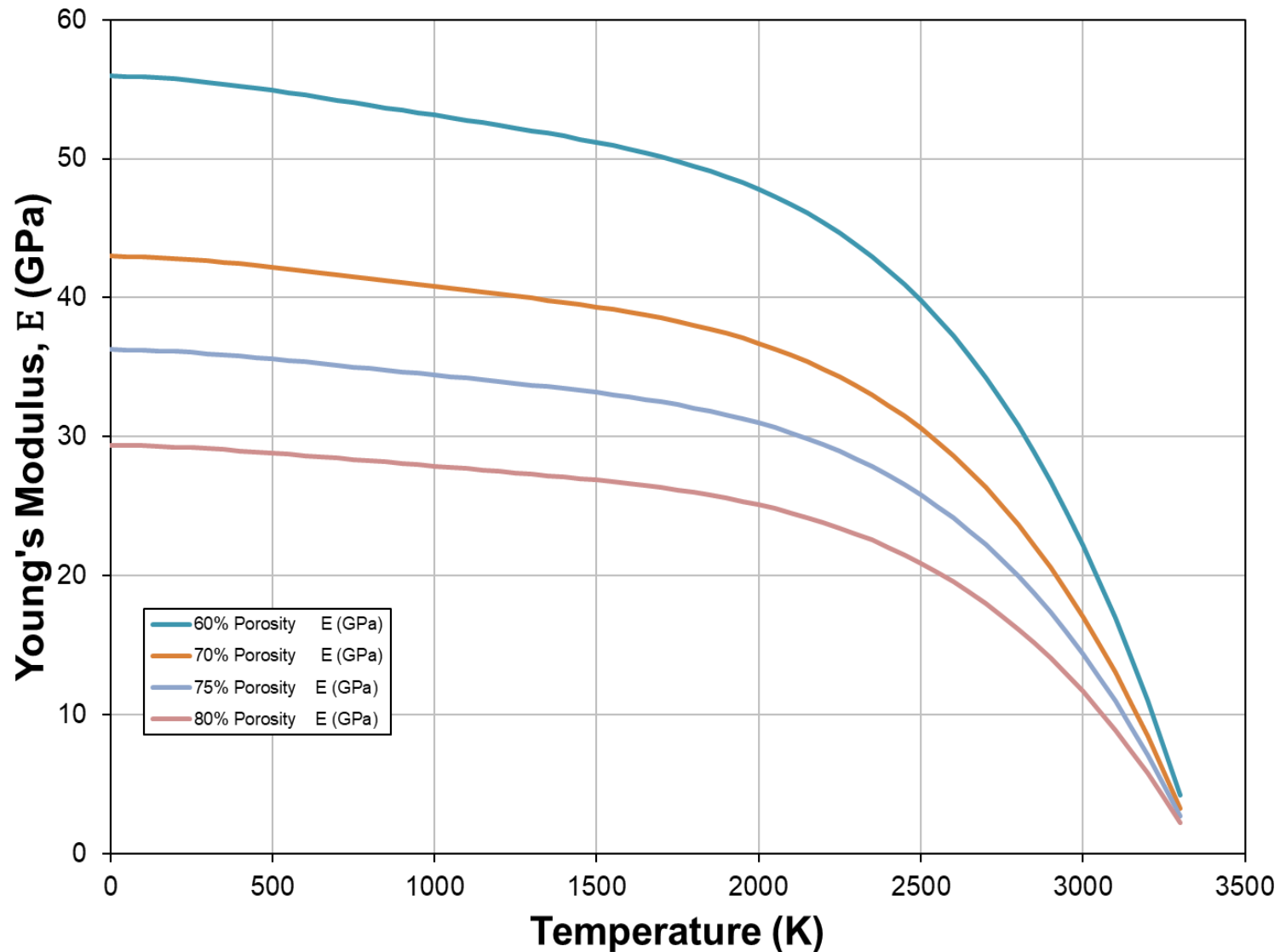


Figure 8.1.3-4: Estimated Elastic Modulus versus Temperature for Porous ZrC.



# SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.3 High Porosity ZrC

Revision 0: 08-02-2021

Young's Modulus with Temperature and Porosity

| Temperature ( T ) |            | Young's Modulus ( E ) [GPa] |              |              |              |              |
|-------------------|------------|-----------------------------|--------------|--------------|--------------|--------------|
| K                 | ( °F )     | Dense ZrC                   | 60% Porosity | 70% Porosity | 75% Porosity | 80% Porosity |
| 100               | ( -279.7 ) | 308.31                      | 55.89        | 42.95        | 36.23        | 29.34        |
| 200               | ( -99.7 )  | 307.47                      | 55.73        | 42.83        | 36.13        | 29.26        |
| 400               | ( 260.3 )  | 304.67                      | 55.23        | 42.44        | 35.80        | 28.99        |
| 600               | ( 620.3 )  | 301.02                      | 54.57        | 41.93        | 35.37        | 28.65        |
| 800               | ( 980.3 )  | 297.07                      | 53.85        | 41.38        | 34.91        | 28.27        |
| 1000              | ( 1340.3 ) | 293.08                      | 53.13        | 40.82        | 34.44        | 27.89        |
| 1200              | ( 1700.3 ) | 289.07                      | 52.40        | 40.27        | 33.97        | 27.51        |
| 1400              | ( 2060.3 ) | 284.78                      | 51.62        | 39.67        | 33.46        | 27.10        |
| 1600              | ( 2420.3 ) | 279.68                      | 50.70        | 38.96        | 32.86        | 26.62        |
| 1800              | ( 2780.3 ) | 273.00                      | 49.49        | 38.03        | 32.08        | 25.98        |
| 2000              | ( 3140.3 ) | 263.69                      | 47.80        | 36.73        | 30.98        | 25.09        |
| 2200              | ( 3500.3 ) | 250.43                      | 45.39        | 34.88        | 29.43        | 23.83        |
| 2400              | ( 3860.3 ) | 231.65                      | 41.99        | 32.27        | 27.22        | 22.04        |
| 2600              | ( 4220.3 ) | 205.50                      | 37.25        | 28.63        | 24.15        | 19.56        |
| 2800              | ( 4580.3 ) | 169.90                      | 30.80        | 23.67        | 19.96        | 16.17        |
| 3000              | ( 4940.3 ) | 122.46                      | 22.20        | 17.06        | 14.39        | 11.65        |
| 3200              | ( 5300.3 ) | 60.55                       | 10.98        | 8.43         | 7.11         | 5.76         |
| 3300              | ( 5480.3 ) | 23.27                       | 4.22         | 3.24         | 2.73         | 2.21         |

**Application Notes:** Table data generated by using temperature dependent modulus data from [2] as the input to the equation below from reference [4]. Due to the lack of experimental data, the model-based relationship in this equation should only serve as a guide.

## Fit Equation:

$$E_p = E_0 [1 - \exp(-r(1 - P))]$$

$E_p$  = Porosity Scaled Modulus of Elasticity [GPa]

$E_0$  = Modulus for dense ZrC

$P$  = Porosity

$r$  = constant = 0.5



## SPACE NUCLEAR PROPULSION MATERIAL PROPERTY HANDBOOK

8 Composites and Structures

8.1 Carbide Based

8.1.3 High Porosity ZrC

**Revision 0: 08-02-2021**

**References**

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