

# **GRC High Temperature Ceramic Matrix Composites (HTCMC)- Current Thrusts, Capabilities, & Challenges/Gaps**

**Dr. Jon Goldsby, Chief**

**Ceramic and Polymer Composites Branch**

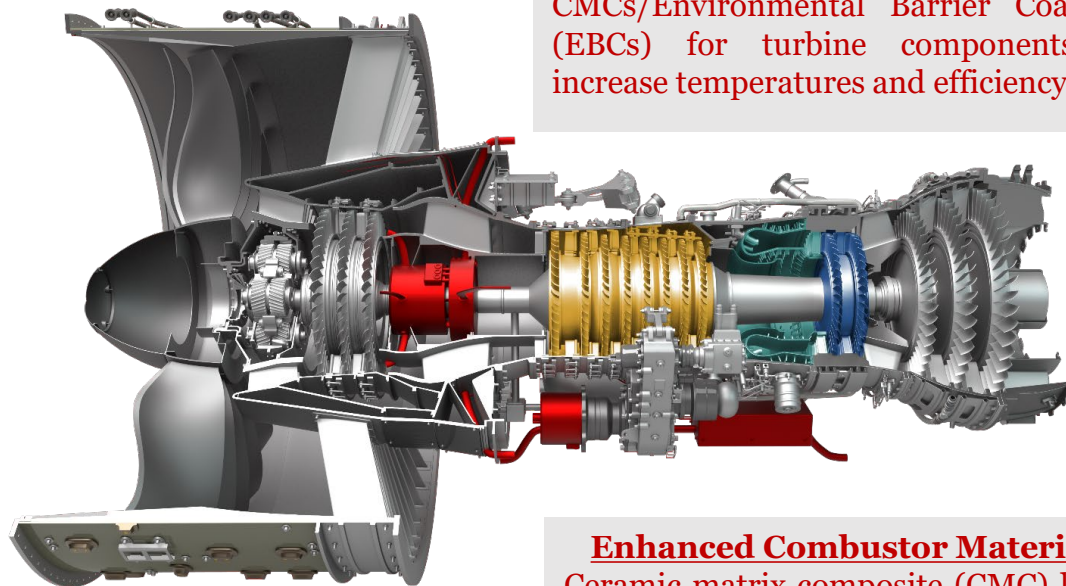
AFRL- NASA GRC High Temperature Materials  
Technical Interchange Mtg

**Apr 18, 2024**

# **Current Thrusts**

# Ceramic Matrix Composite (CMC) Components for Turbine Engine Applications

## Turbine Engines - Targeted Components (CMCs)



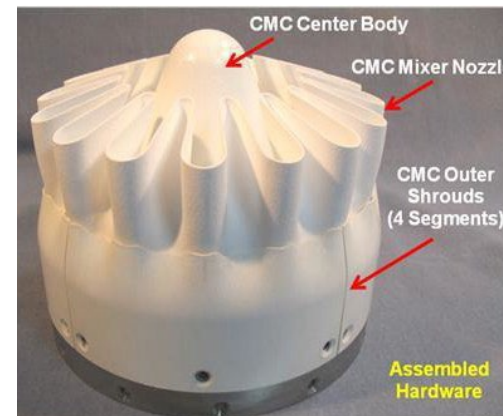
### High Pressure Turbine (HPT)

#### Materials

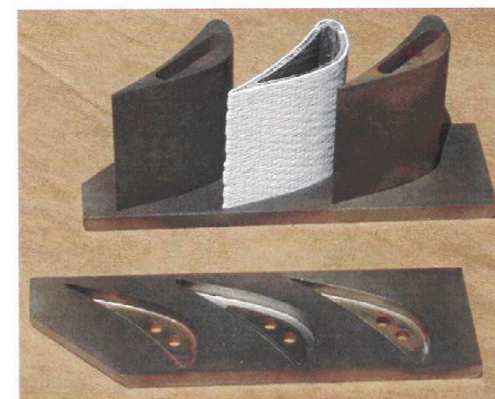
CMCs/Environmental Barrier Coatings (EBCs) for turbine components to increase temperatures and efficiency.

### Enhanced Combustor Materials

Ceramic matrix composite (CMC) liners for combustors to increase performance and durability.



Oxide/Oxide Mixer Nozzle

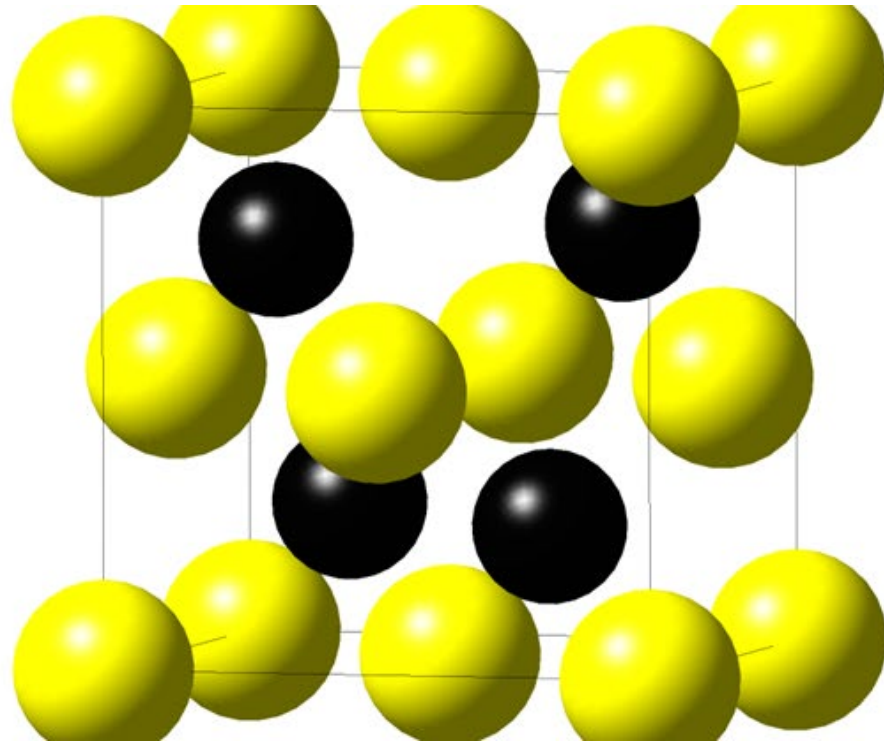


EBC Coated SiC/SiC Vanes



SiC/SiC Combustor Liners: Outer Liner and EBC Coated Inner Liner

# Silicon Carbide



Melting point: 2830 °C

Young's Modulus: 415 GPa

Density: 3.16 g/cm<sup>3</sup>

Hardness: 32 GPa (Vickers's)

Fracture Toughness: 3.160.3 MPa\*m<sup>1/2</sup>

Tensile Strength: 250 MPa

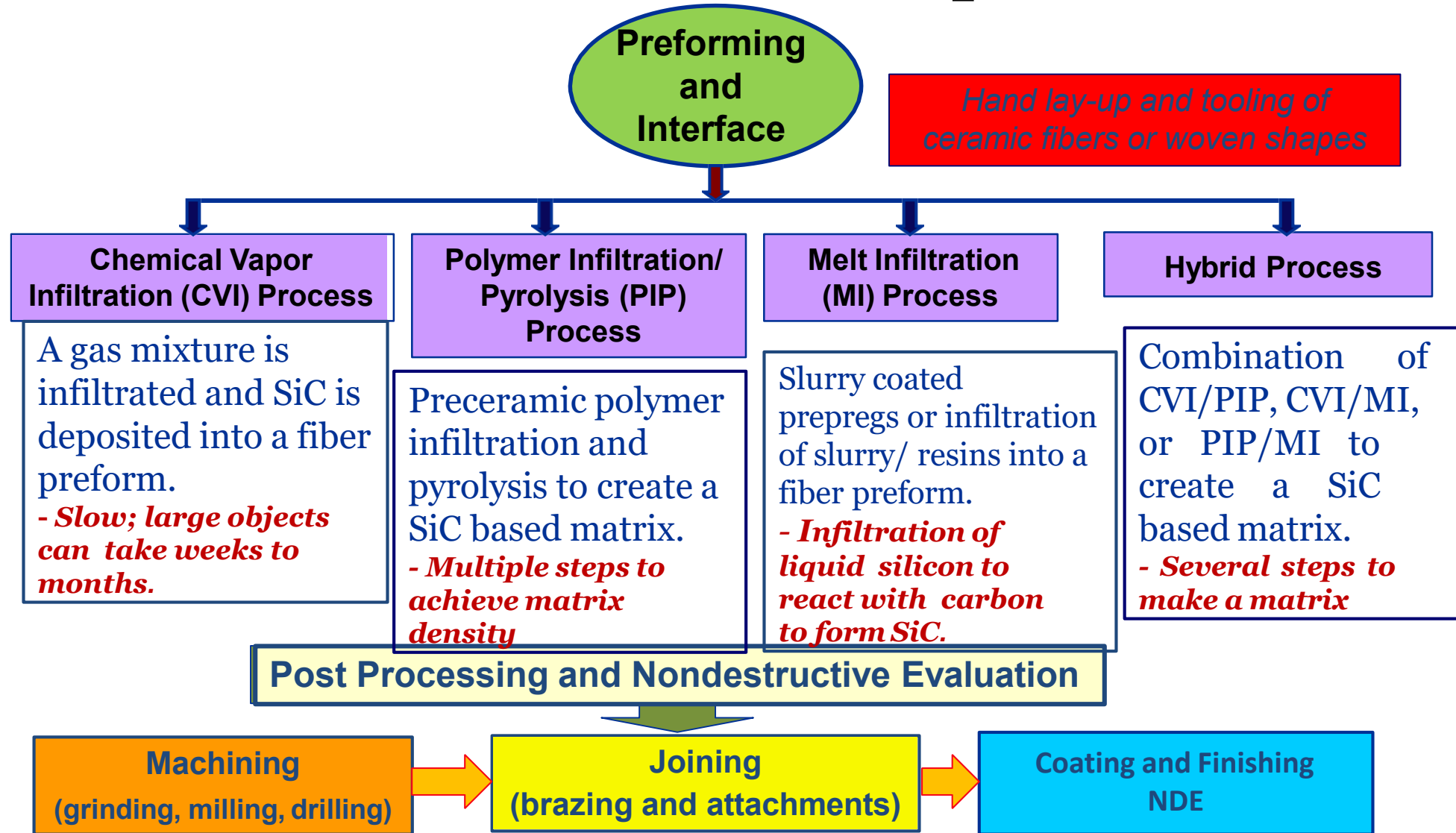
Thermal Conductivity: 114 W /m\*K

Crystal Structure: Fm-3m

# Applications

- Combustion and turbine section components.
- Heat exchangers, reformers, reactors, and filters for the chemical industry.
- Preheaters, recuperators, and radiant tubes.
- Thermal protection systems, thruster nozzles, reusable rocket nozzles, and turbopump components for space vehicles.
- Nuclear fission and fusion reactors as fuel cladding and radiation blankets.

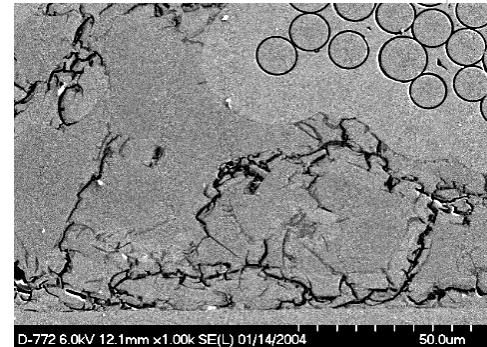
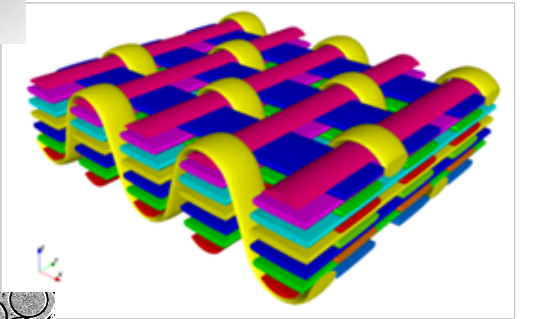
# Current Approaches for Manufacturing of Ceramic Matrix Composites



# GRC Has Extensive Expertise and Capabilities in High Temperature CMCs

NASA 2700° F CMC combines three technology advancements

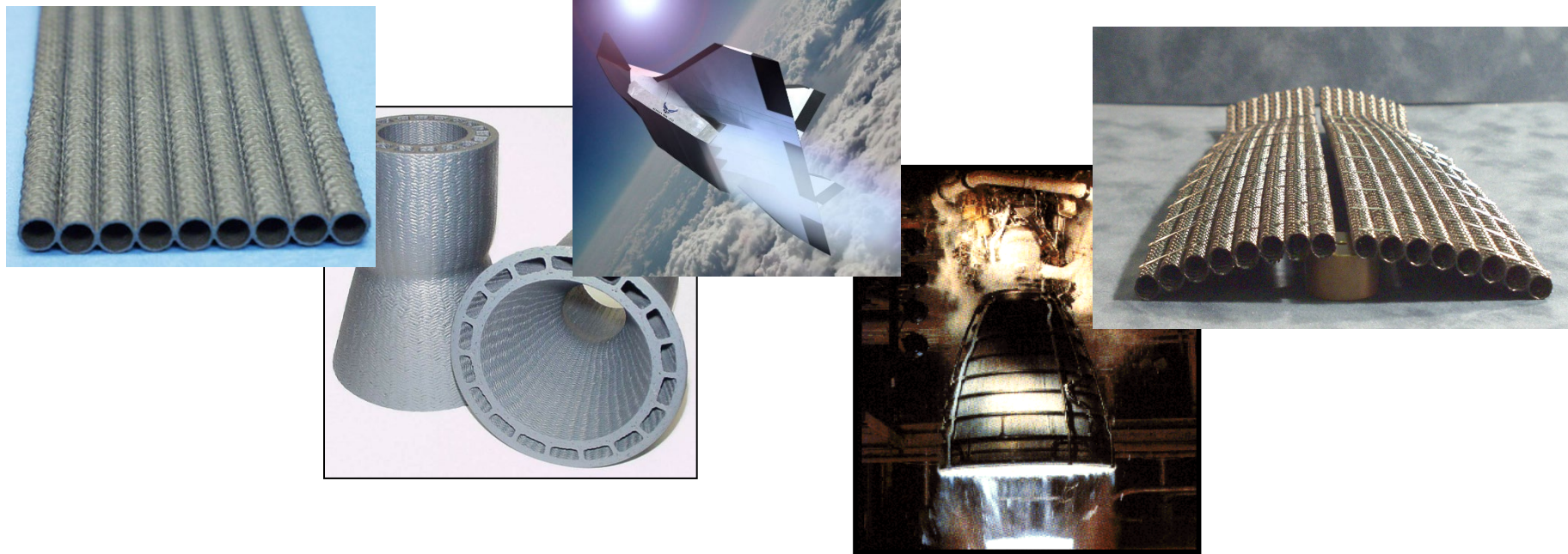
- Creep-resistant Sylramic-iBN fiber
- Advanced 3D fiber architecture
- Hybrid CVI-PIP SiC matrix



CMC research includes material development, life prediction and experimental validation

# Cooled CMC Development at NASA

**Test capabilities include actively cooled CMC subelements with built-in coolant channels**



- **Lighter weight than metallic designs: up to 50% weight reduction**
- **Increased operational margin: enhanced range and/or payload**

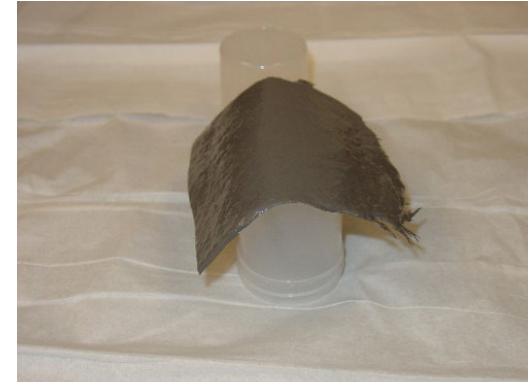
# Flexible Gaskets for Hypersonic Applications



GRC 11 Gaskets made from RTV foam



GRC 16 Gaskets made from Ablative polymer



GRC 17 Gaskets made from silicone based RTV polymer

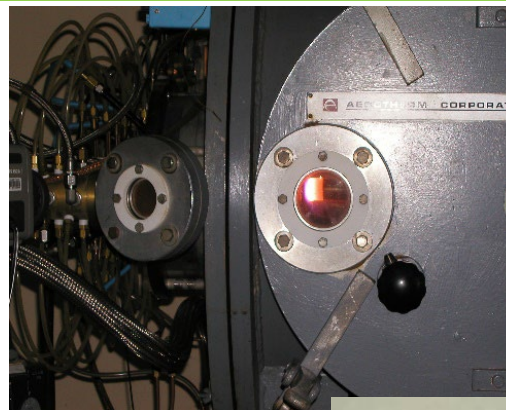
**These gaskets have shown excellent plasma performance in various facilities under re-entry and hypersonic conditions.**



QARE Testing at GRC



2"x2"



HYMETS (LaRC)



1" dia



ArcJet Testing at ARC and LCAT



# In-Space Manufacturing, Assembly, and Repair Technologies (I-SMART) for Space Structures

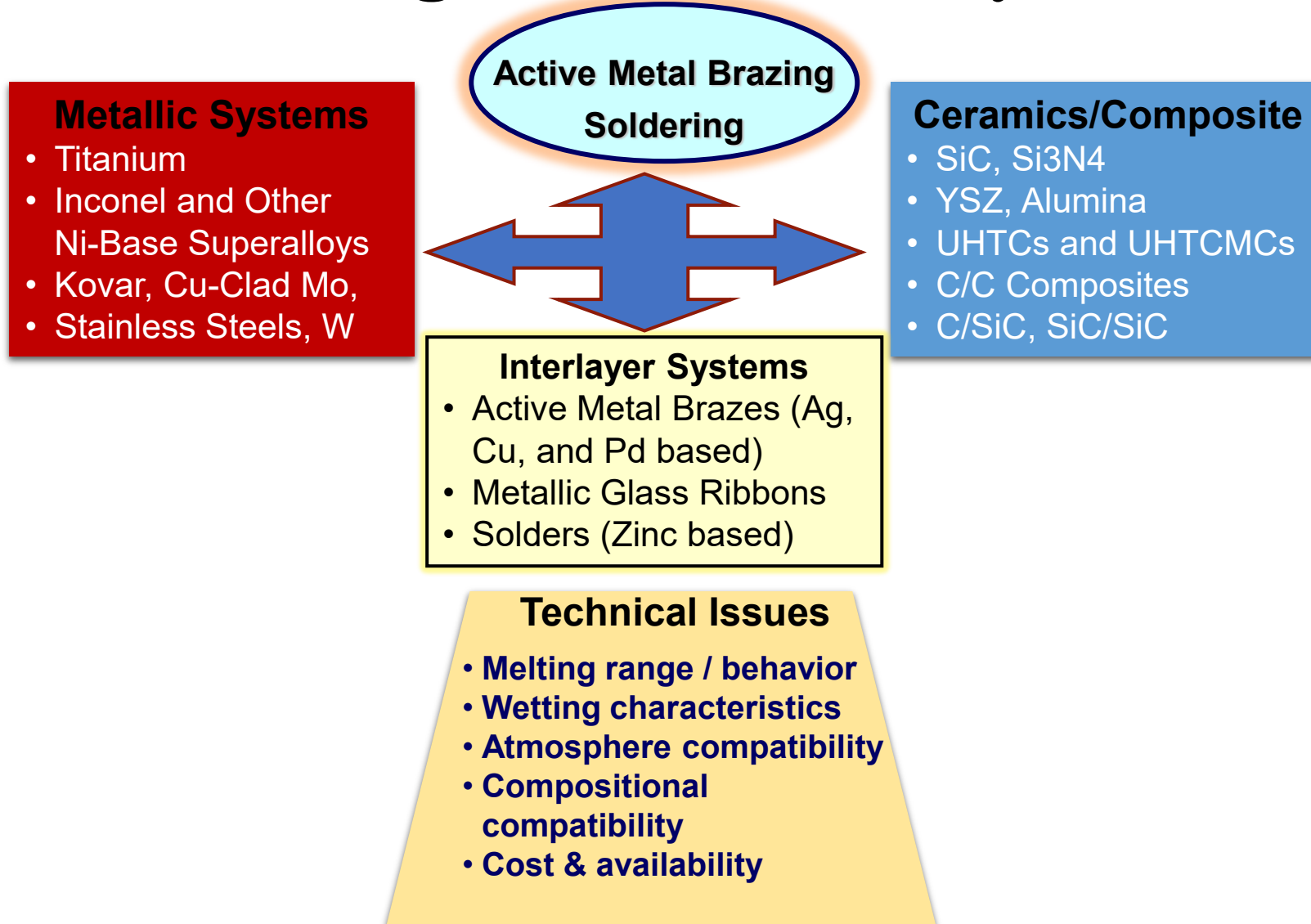
Developing refurbishment approaches for extending the reusability and for reducing the replacement and maintenance costs for the high-performance ceramic matrix composite (CMC) materials that are utilized for propulsion and airframe applications in hypersonic and aerospace vehicles

## Project Objective(s)

- Develop and optimize materials and processing for CMC repair approaches.
- To enable 2x the reusable life in repaired versus unrepaired CMC components.
- Demonstrate the ability to prolong CMC life so that mechanical properties and environmental durability are maintained.



# Integration of Metals to Ceramics and Composites Using Metallic Interlayers



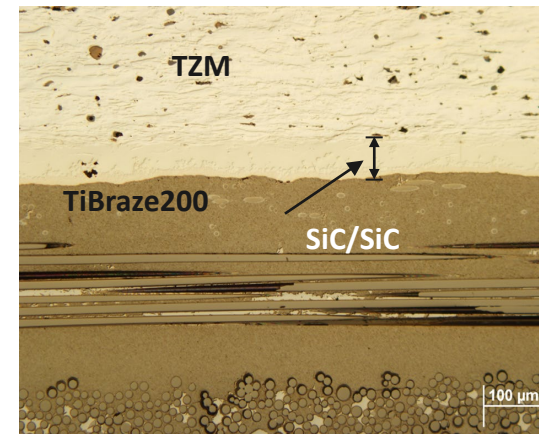
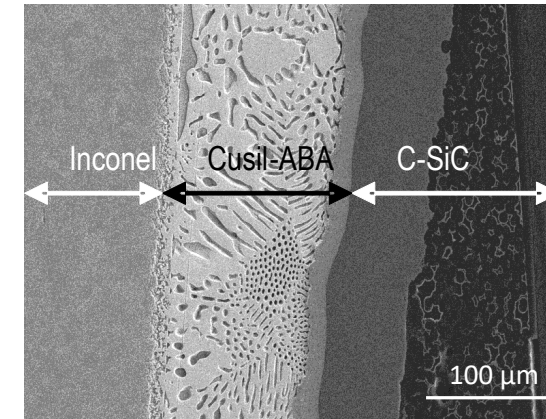
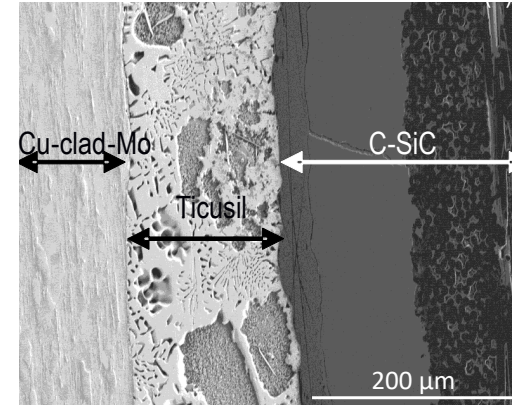
# Joining and Integration of Composites to Metals for Applications in Hypersonic Systems

## Objectives:

- Develop joining and integration technologies for incorporating high temperature carbon-carbon and ceramic matrix composite components with metallic systems.
  - Thermal protection,
  - Hot structures,
  - Control surfaces,
  - Heat exchangers, and combustor walls

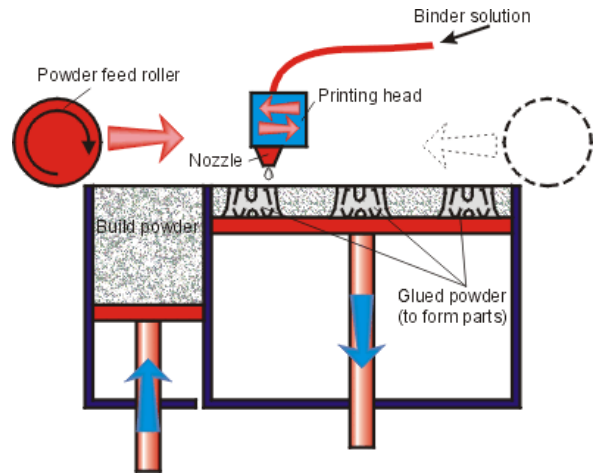
## Approach:

- Develop brazing for relevant material pairings (e.g. SiC/SiC, C/SiC, or C/C to Kovar, TZM, Inconel, or others) with proper selection of joining interlayer(s) and optimized brazing conditions. Develop tailored and functionally graded brazes. Conduct braze design and stress analysis.
- Address challenges in brazing more complex shapes such as tube to tube and complex geometries.

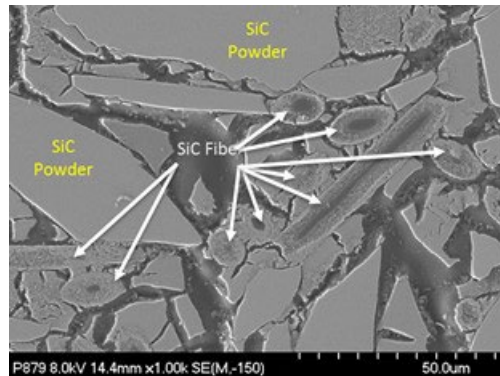


# Capabilities

# Additive Manufacturing of Ceramics Using Binder Jetting



**Binder Jetting process**



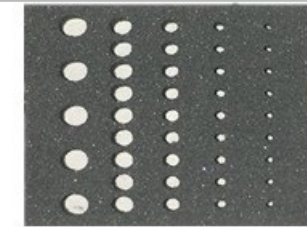
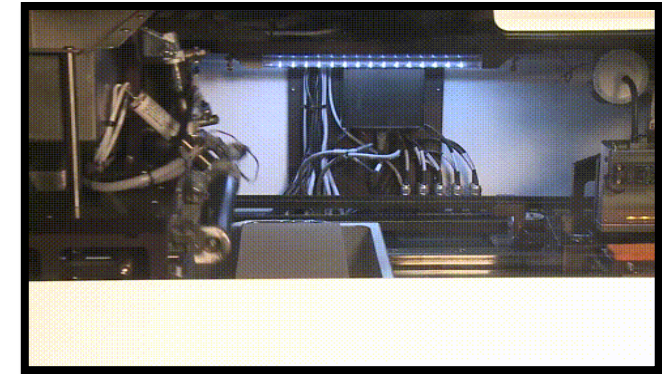
**Chopped Fiber Reinforced Ceramic Matrix Composite**



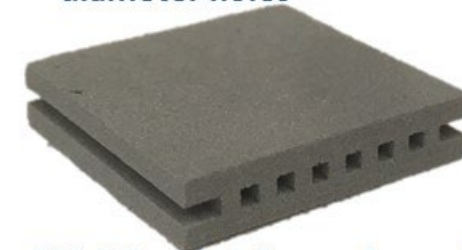
**ExOne's Innovent System**



**High pressure turbine nozzle segments: cooled doublet vane sections.**

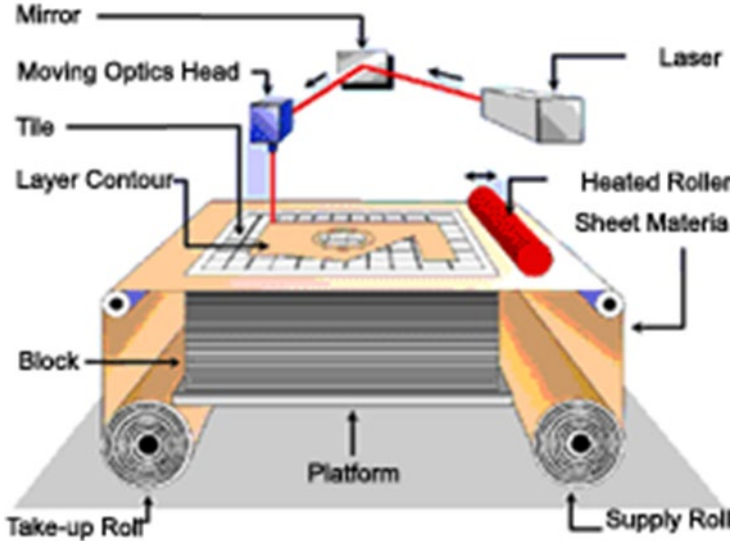


**1 in  
AM SiC with 500  $\mu$ m  
diameter holes**



**AM SiC with 1.5 mm channels**

# Laminated Object Manufacturing (LOM) For Silicon Carbide-Based Composites



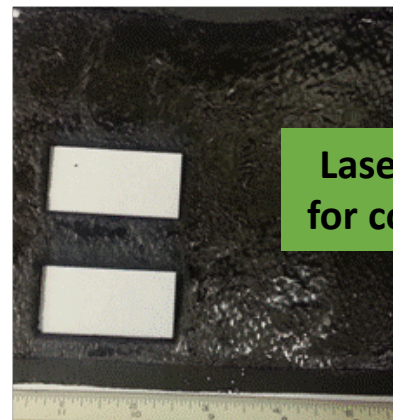
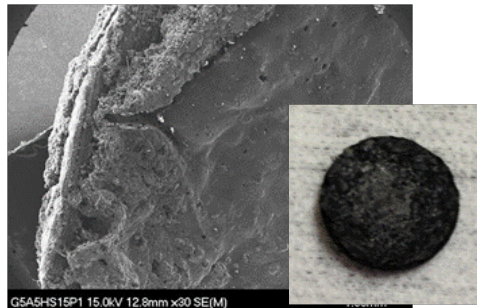
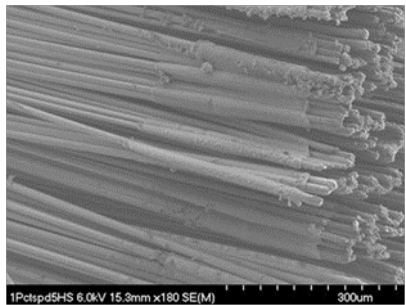
**LOM allows for continuous fiber reinforced CMCs.**



Universal Laser System (Two 60 watt laser heads and a work area of 32"x18")

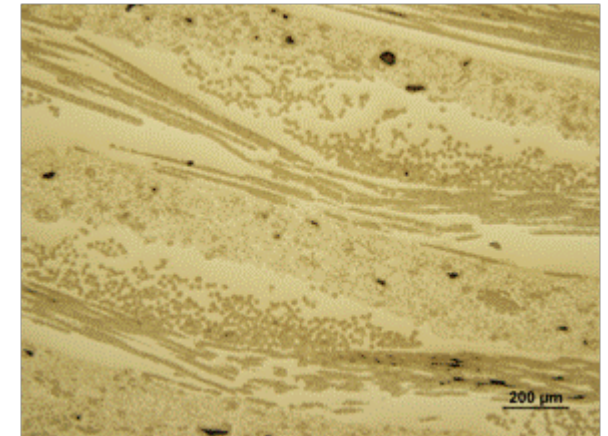
## Prepregs for Composite Processing

- A number of SiC (Hi-Nicalon S, uncoated) fabrics (~6"x6") were prepregged.
- These prepregs were used for optimization of laser cutting process.
- Baseline laser cutting data was also generated for different types of SiC fabrics (CG Nicalon, Hi-Nicalon, and Hi-Nicalon S)



Laser cut prepregs used for composite processing

**Fabrics and Prepregs cut at different laser powers/speeds**



**Silicon Infiltration:**  
1475°C, 30 minutes in vacuum

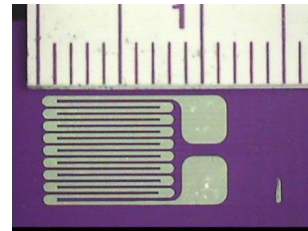
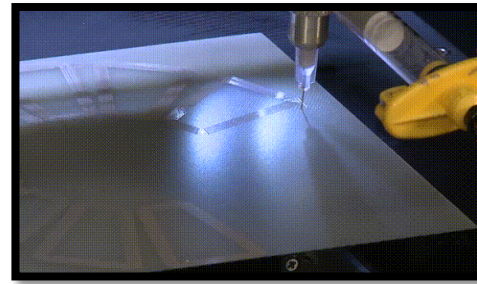
# Direct Writing Technologies for Sensors/Embedded Systems

## NScript Capabilities and Benefits

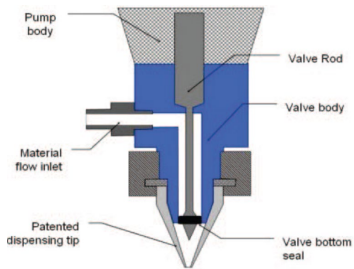
- 300x 300x 150mm Gantry XYZ Platform
- SmartPump with 100 Picoliter Volumetric Control
- Ability to host up to four separate materials.
- Print on curved surfaces and 3D structures.
- Precise motion control and micro-dispensing of materials.
- Direct writing with clean starts and stops (no contact or masks as for screen printing).
- Ability to print a wide variety of ceramic pastes (structural and functional), electronic pastes, adhesives, solders, bio-materials.



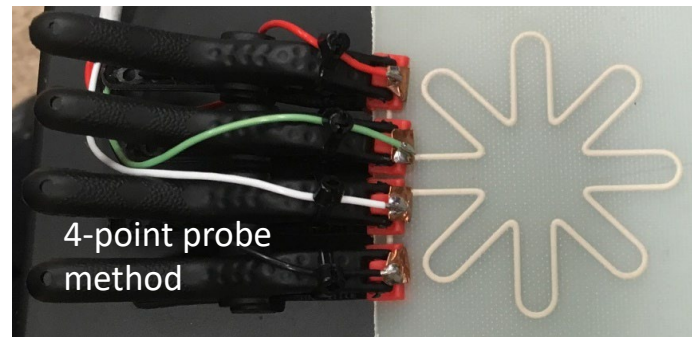
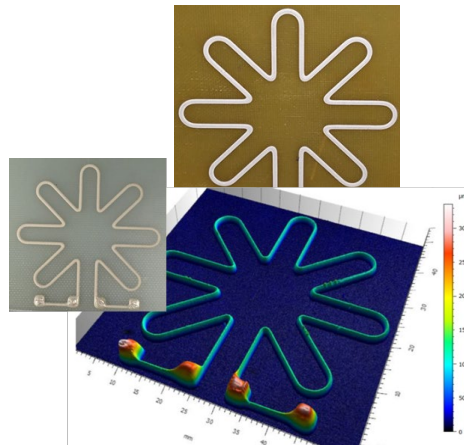
NScript Direct Write Printer



Printed strain gages.



Keyence VR 3200 Profilometer



4-point probe method



Thin Surface and Imbedded Thick 4-Pt Probe Windings

# ARCJoinT: Joining of Ceramic Components Using Affordable, Robust Ceramic Joining Technology (ARCJoinT)



**Apply Carbonaceous Mixture to Joint Areas**

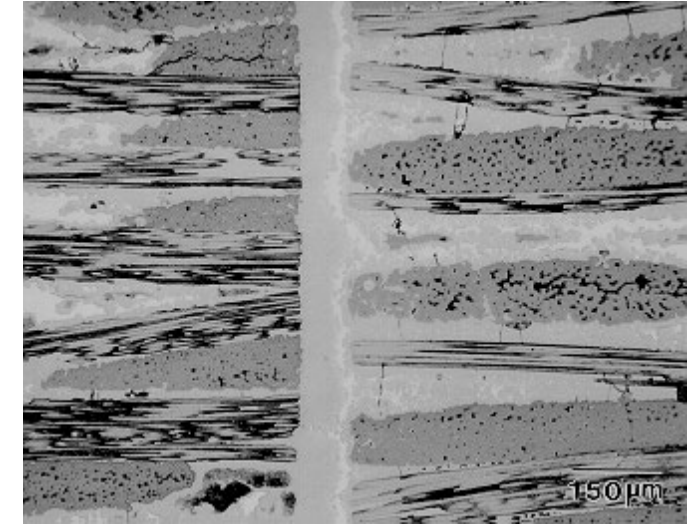
*Cure at 110-120°C for 10 to 20 minutes*

**Apply Silicon or Silicon-Alloy (paste, tape, or slurry)**

*Heat at 1250-1425°C for 10 to 15 minutes*

**Affordable and Robust Ceramic Joints with Tailorable Properties**

**1999 R&D 100 Award  
2000 NorTech Innovation Award**



**Joined MI C/SiC Composite**

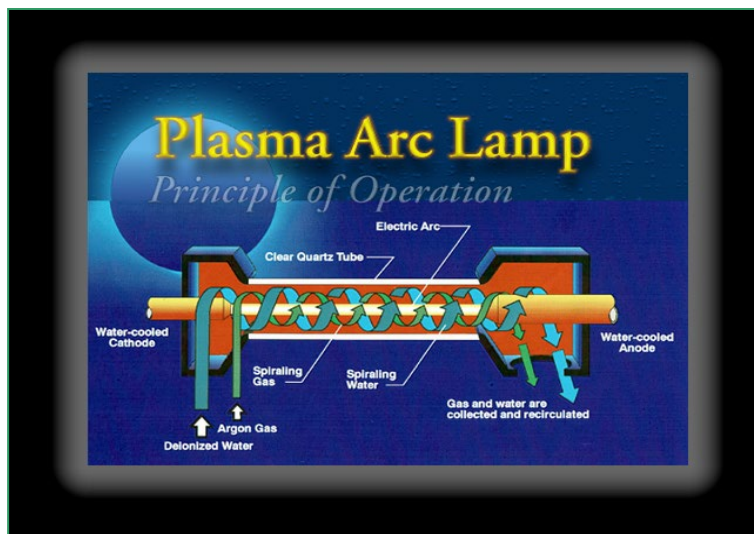
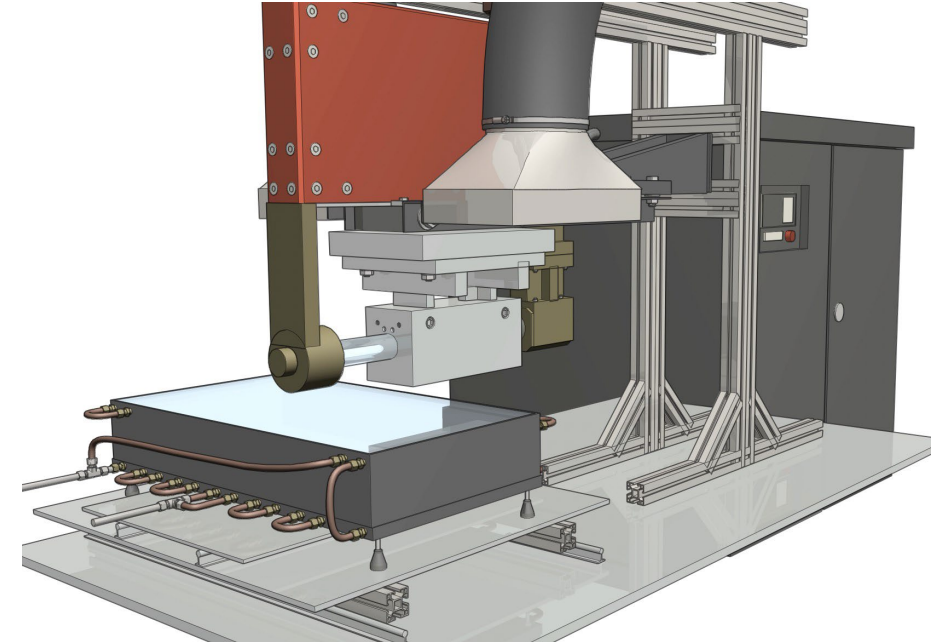
## Advantages

- **Joint interlayer properties are compatible with parent materials.**
- **Processing temperature around 1200-1450°C.**
- **No external pressure or high temperature tooling is required.**
- **Localized heating sources can be utilized.**
- **Adaptable to in-field installation, service, and repair.**

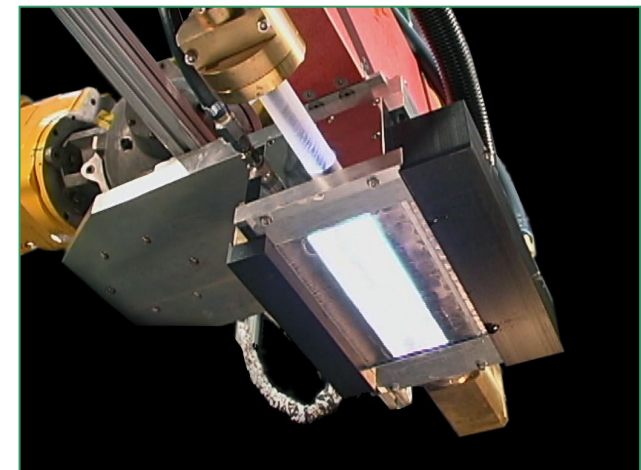
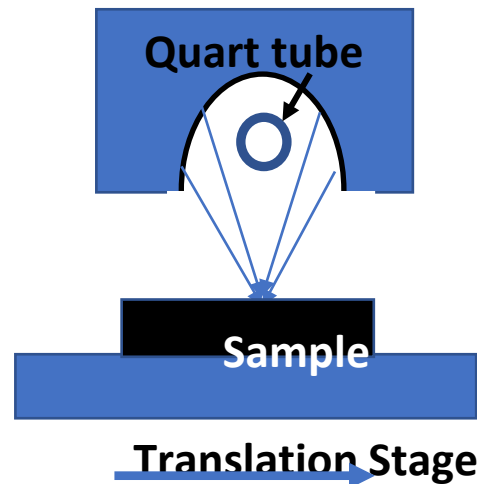
**Very good quality, high strength bonds are obtained.**

# 200kW Plasma Arc Lamp (Artificial Sun)

- Full spectrum stabilized arc generated within quartz tube.
- Reflector focuses radiant energy.
- Generates radiate 4000°F surface temperature
- Durability testing
- Material processing
- Thermal gradients



Reflector focuses Radiant Energy



# Advanced Vacuum Systems (AVS)

Vacuum Hot- Press

Tungsten heating 2000°C (max)

Ram force 110 tons

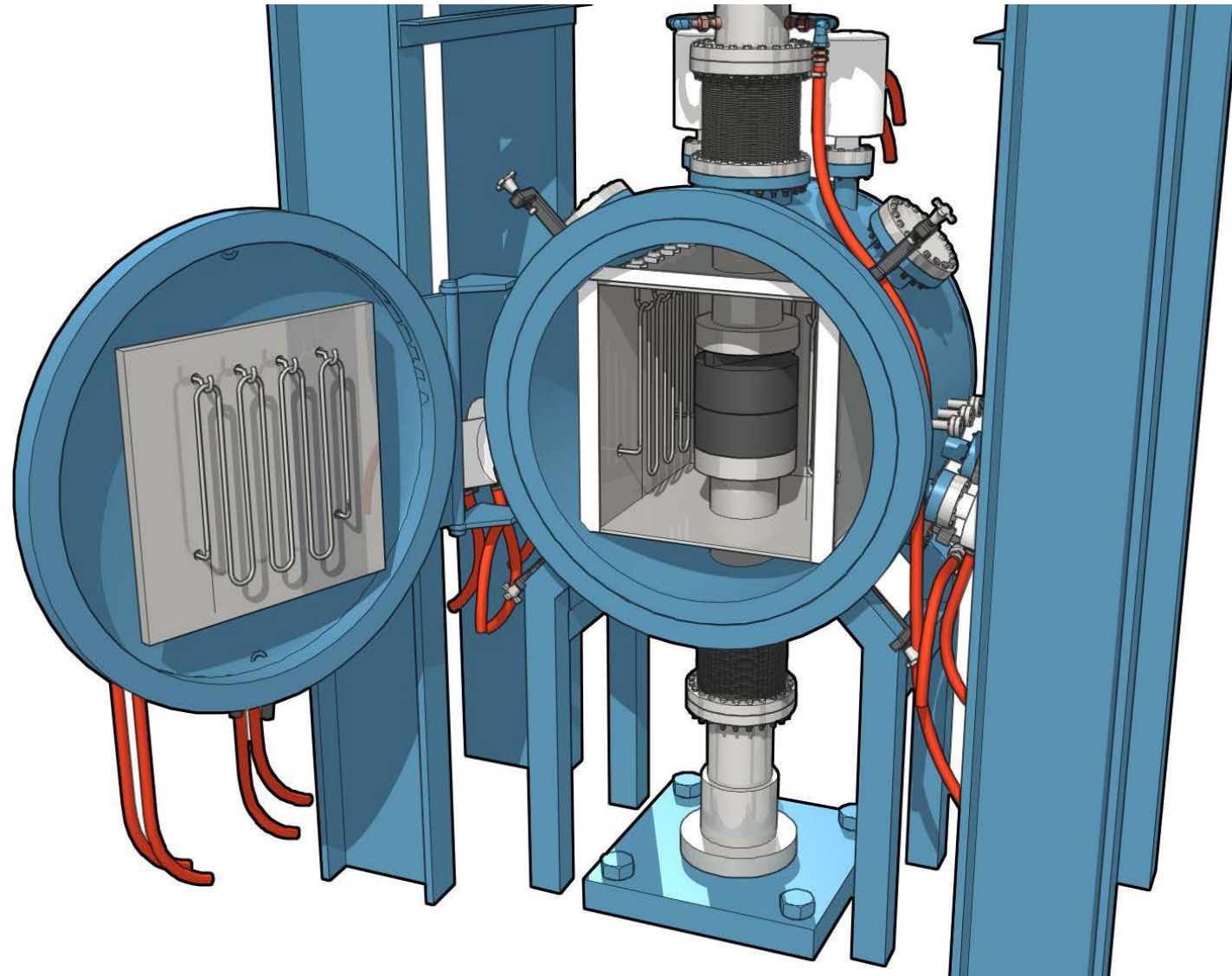
9-inch diameter die

- Diffusion Bonding
- Ceramic Processing
- Metal Processing
- Powder Densification
- Powder Metal Forming
- Thermal Forging
- Super Plastic Forming
- Refractory Metal Forming
- Thermal Sizing.

ATMOSPHERES

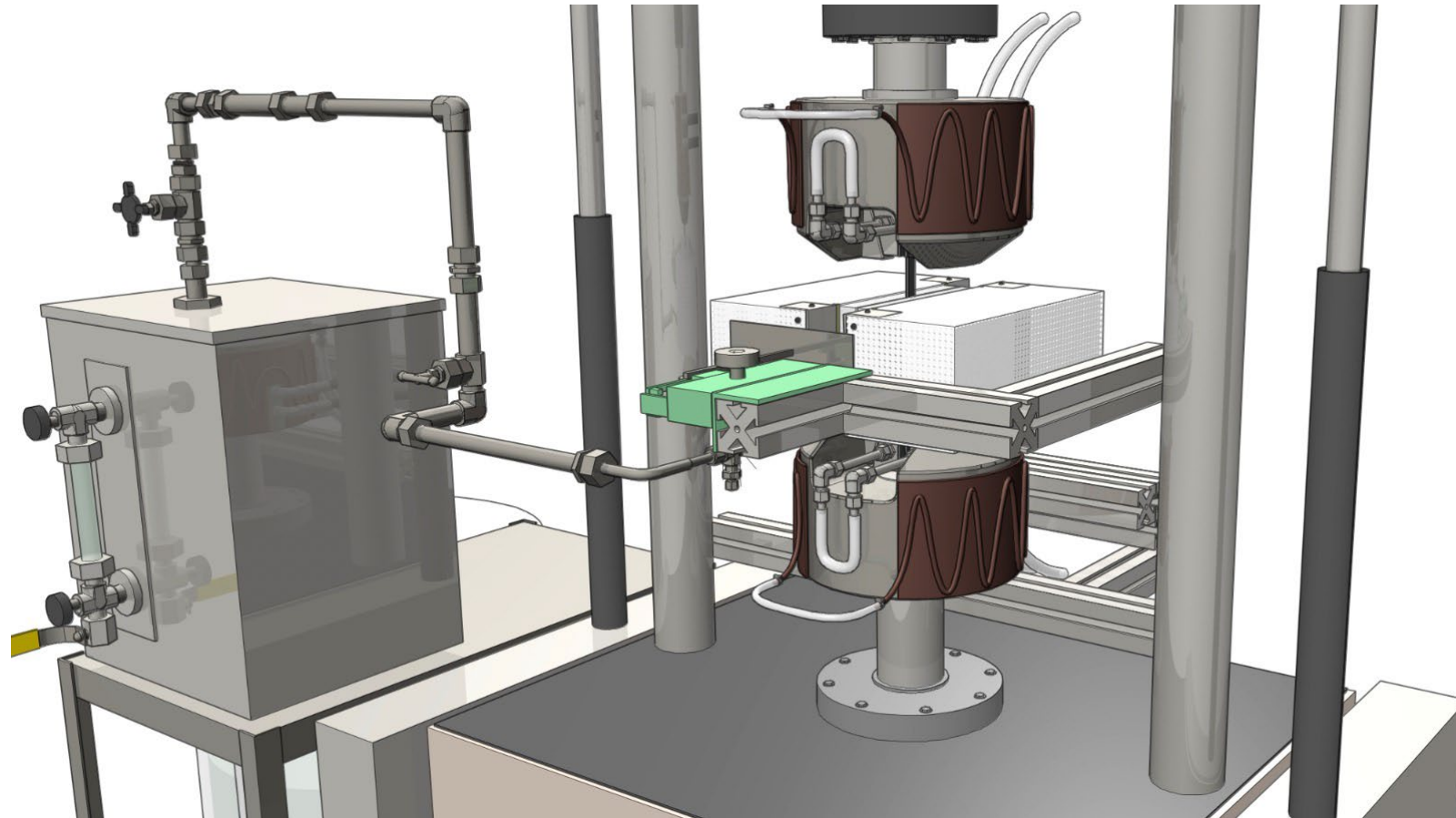
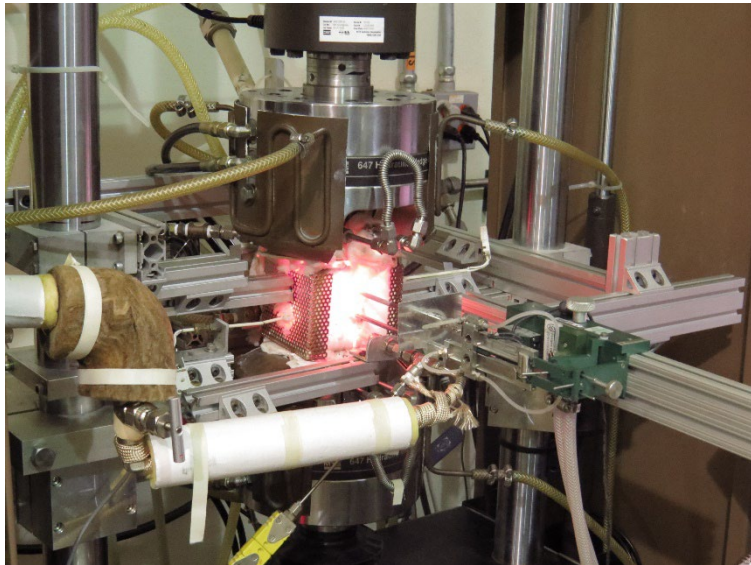
Vacuum, Inert, or Reducing Gas;

Partial Pressure to 5 PSI

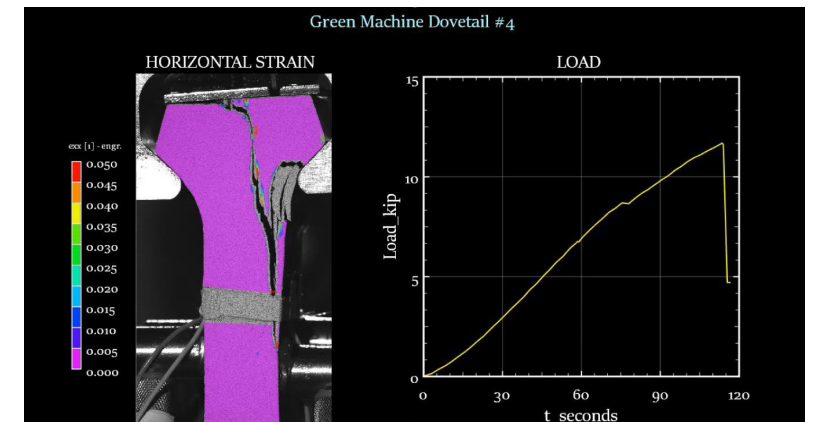
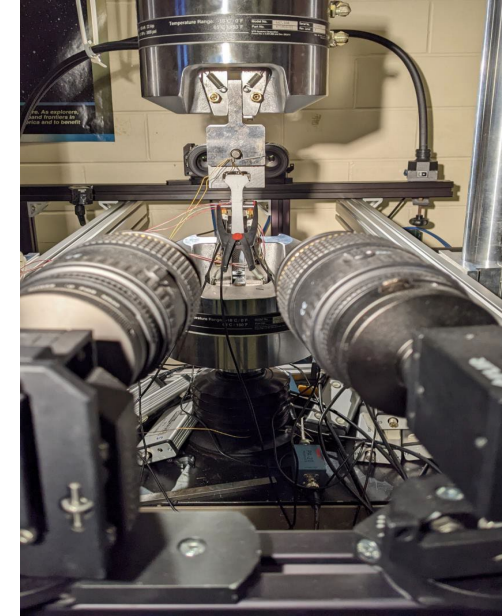
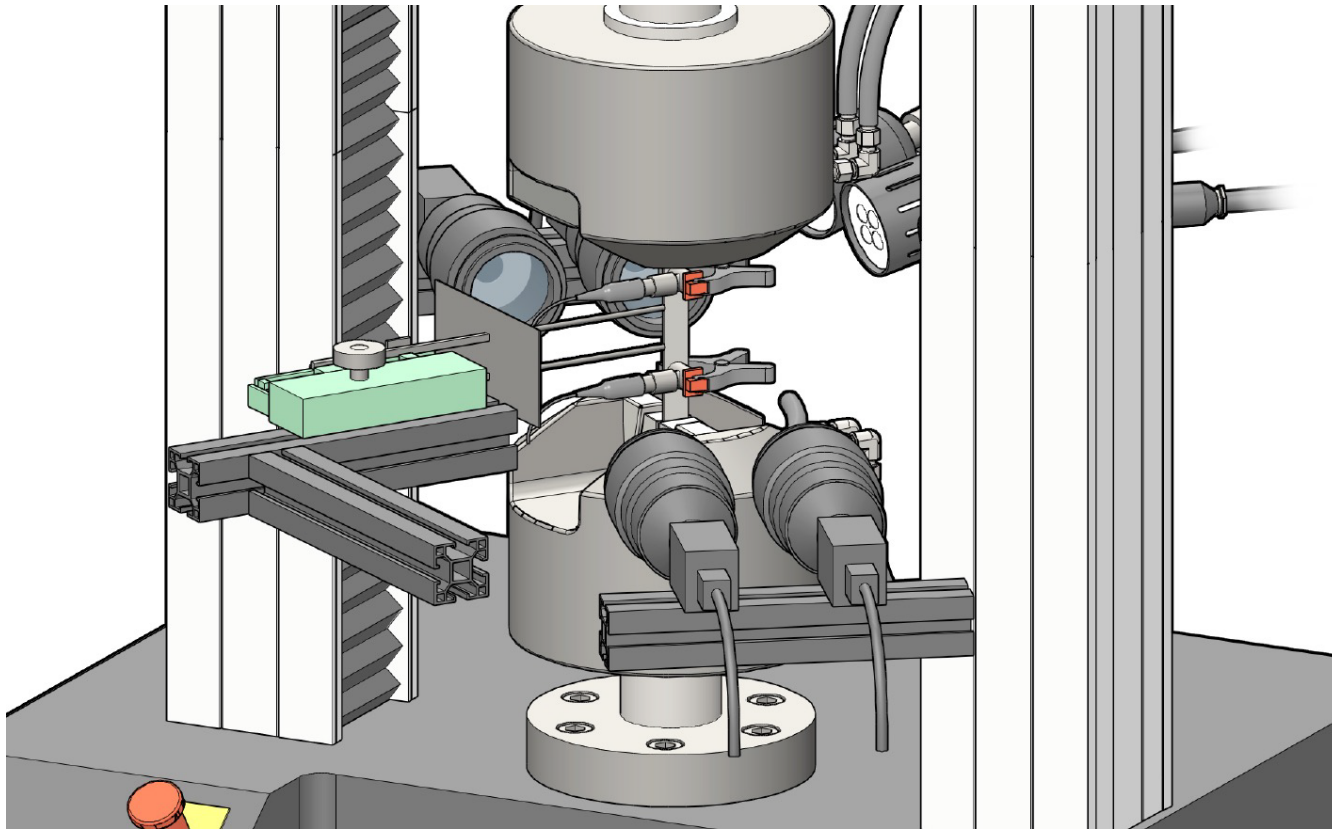


## Elevated temperature, tensile-dwell fatigue tests in steam environment

- Demonstrated capability to perform long duration (up to 300 hours) Sustained-Peak Low-Cycle Fatigue (SPLCF) tests in steam environment at 2400 °F.

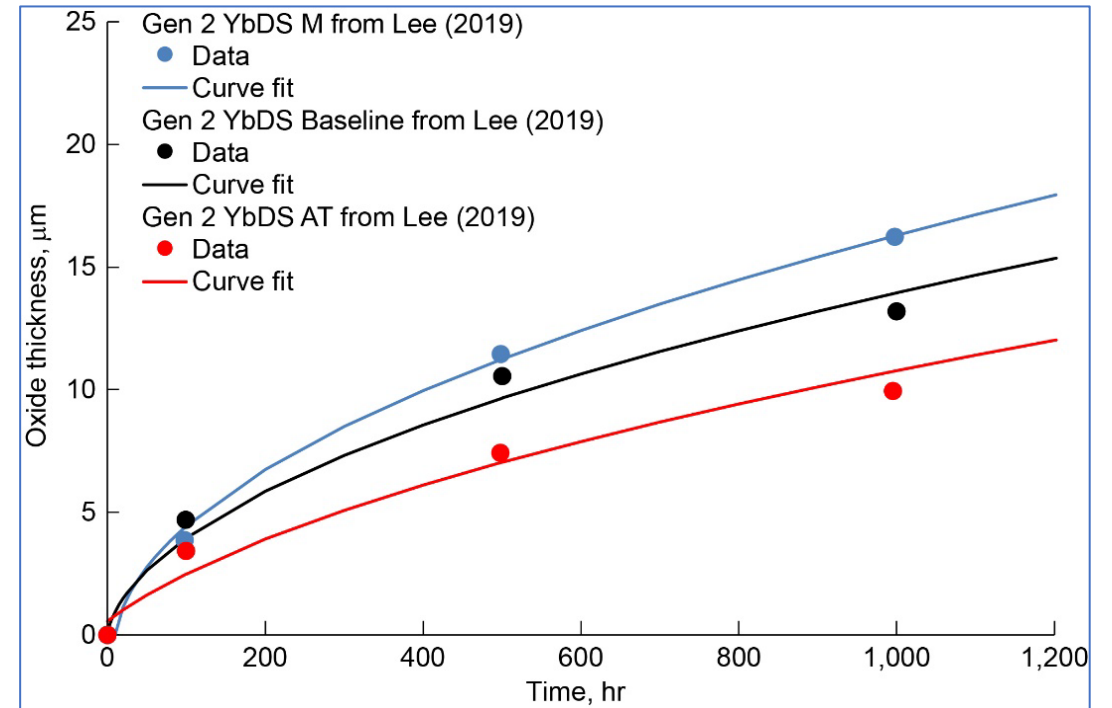


# Digital Image Correlation Measurement



# Estimates of the Permeability of Water in Rare Earth Silicates

This work is significant as no direct measurements of the permeability of water in rare earth silicates have been reported in the literature and steam oxidation is one of the greatest threats to the durability of EBC coated ceramic matrix composites in engine applications.



# **Gaps and Challenges**

# Development of an Ultra- High Temperature Ceramic Fiber

- Polymer base precursor
- Small diameter to allow weaving
- Interphase coatings
- Temperature and time dependent characterization

# GRC High Temp CMC – Current Thrusts, Life Prediction, Capabilities, & Challenges/Gaps

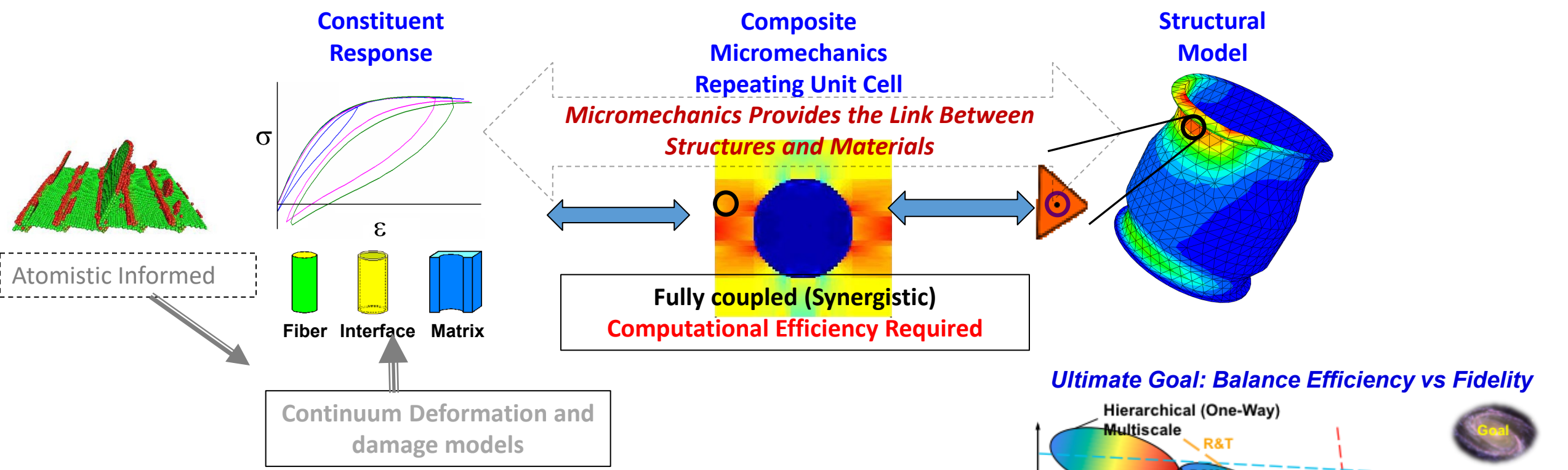
Steven M. Arnold, PhD, FASM

Technical Lead: Multiscale Multiphysics Modeling

T<sup>3</sup> Project: Material and Structures Discipline Lead



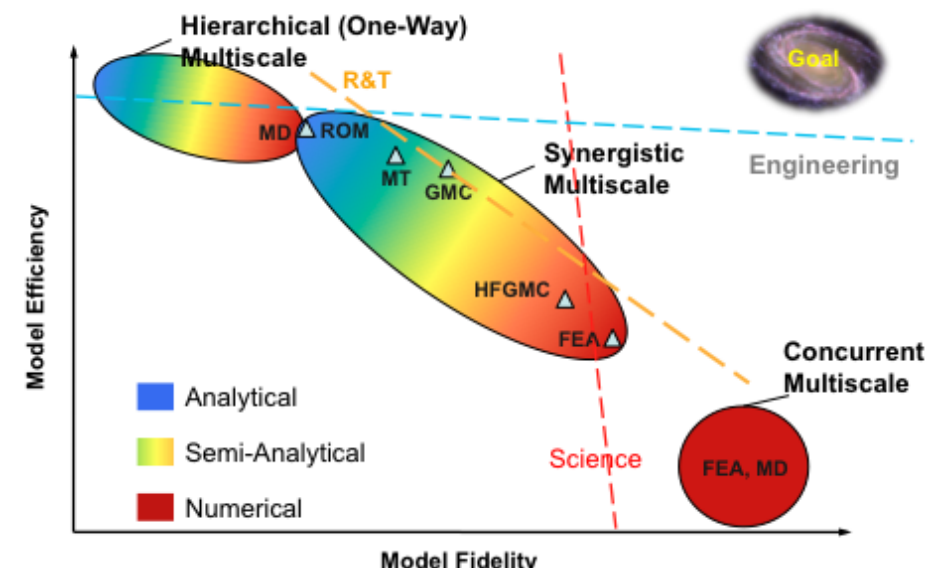
# NASA GRC's Multiscale Analysis Approach Links the Behavior Of A Composite Material/Structure To The Behavior Of Its Constituents



**Framework Admits Nonlinear Constitutive and Damage Models**

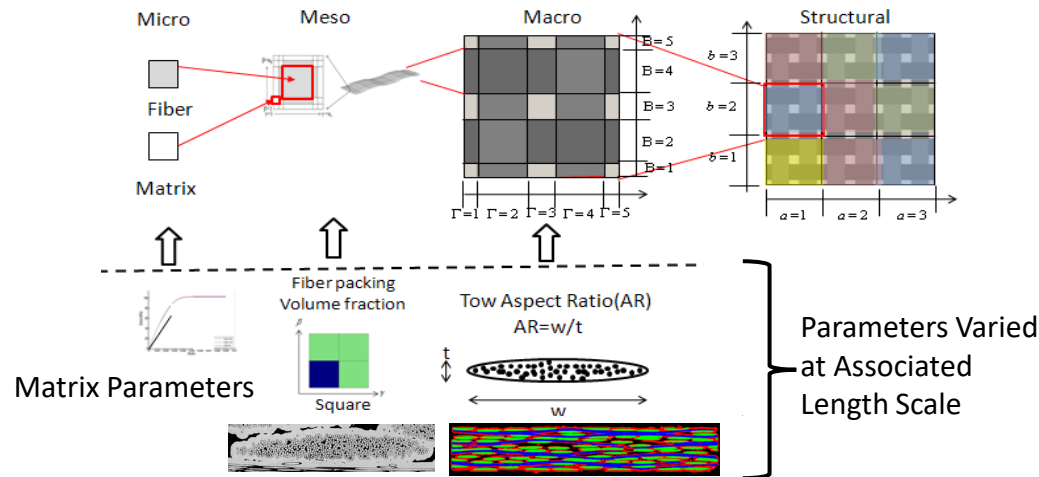
Abouti, J., Arnold, S.M., and Bednarczyk, B.A. (2013) *Micromechanics of Composite Materials: A Generalized Multiscale Analysis Approach*, Elsevier, Oxford, UK., pp 1-984.

**Ultimate Goal: Balance Efficiency vs Fidelity**

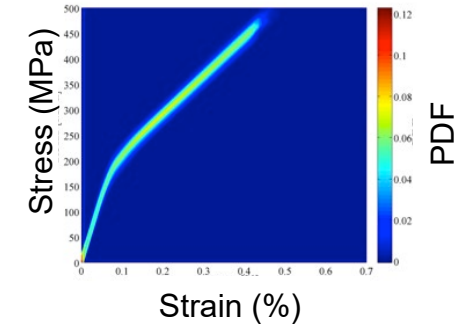
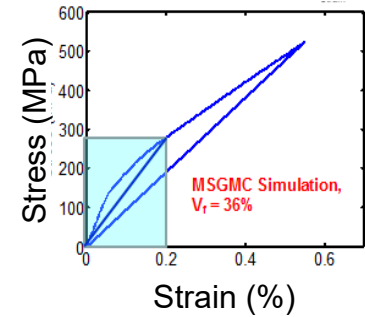


# Multiscale Modeling of CMC Behavior

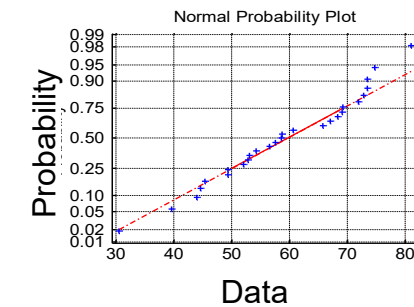
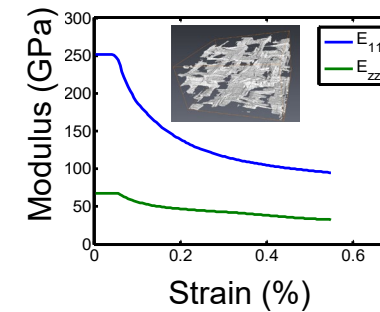
## Stochastic Nonlinear Response of Woven CMCs Using MSGMC



Liu, K.C. And Arnold, S.M. (2013), "Statistical Influence of Scale Specific Features on the Progressive Damage of Woven Ceramic Matrix Composites (CMCs)", Computers, Materials and Continua. Vol.35, No. 1, pp. 35-65.

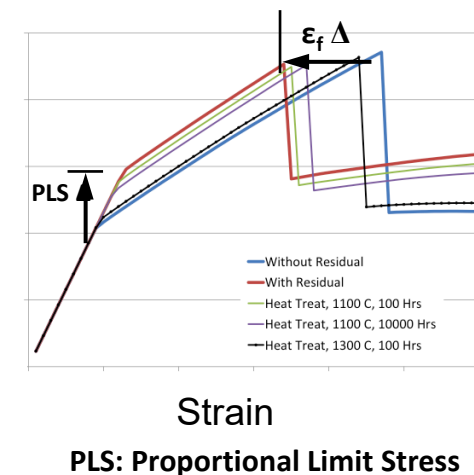
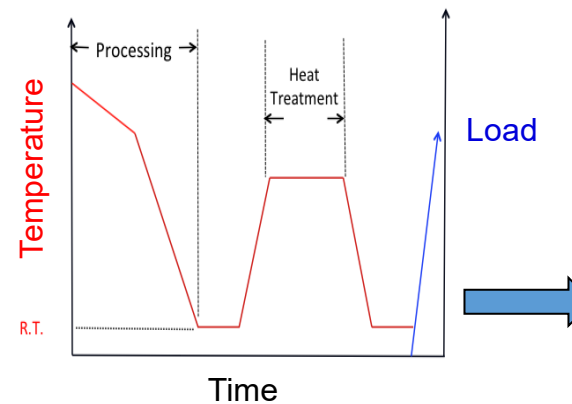
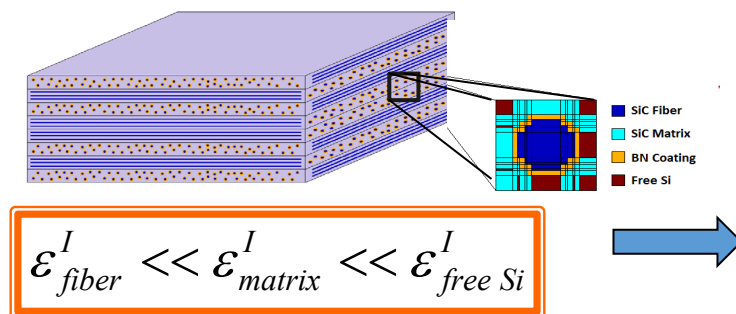


Stochastic simulation of a 5HS weave composite



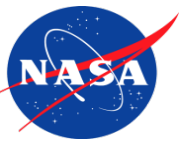
## History Dependent Response of Laminated CMCs Using GMC

Mital, S. K., Arnold, S.M., Bednarczyk, B.A. and Pineda E.J.; "Micromechanics-based Modeling of SiC/SiC Ceramic Matrix Composites and Structures", Special Issue: Ceramic Matrix Composites: Performance Evaluation and Application, Recent Progress in Materials, Vol. 5, No. 2, June 2023





# ImMAC Enables Multiscale Failure / Progressive Damage

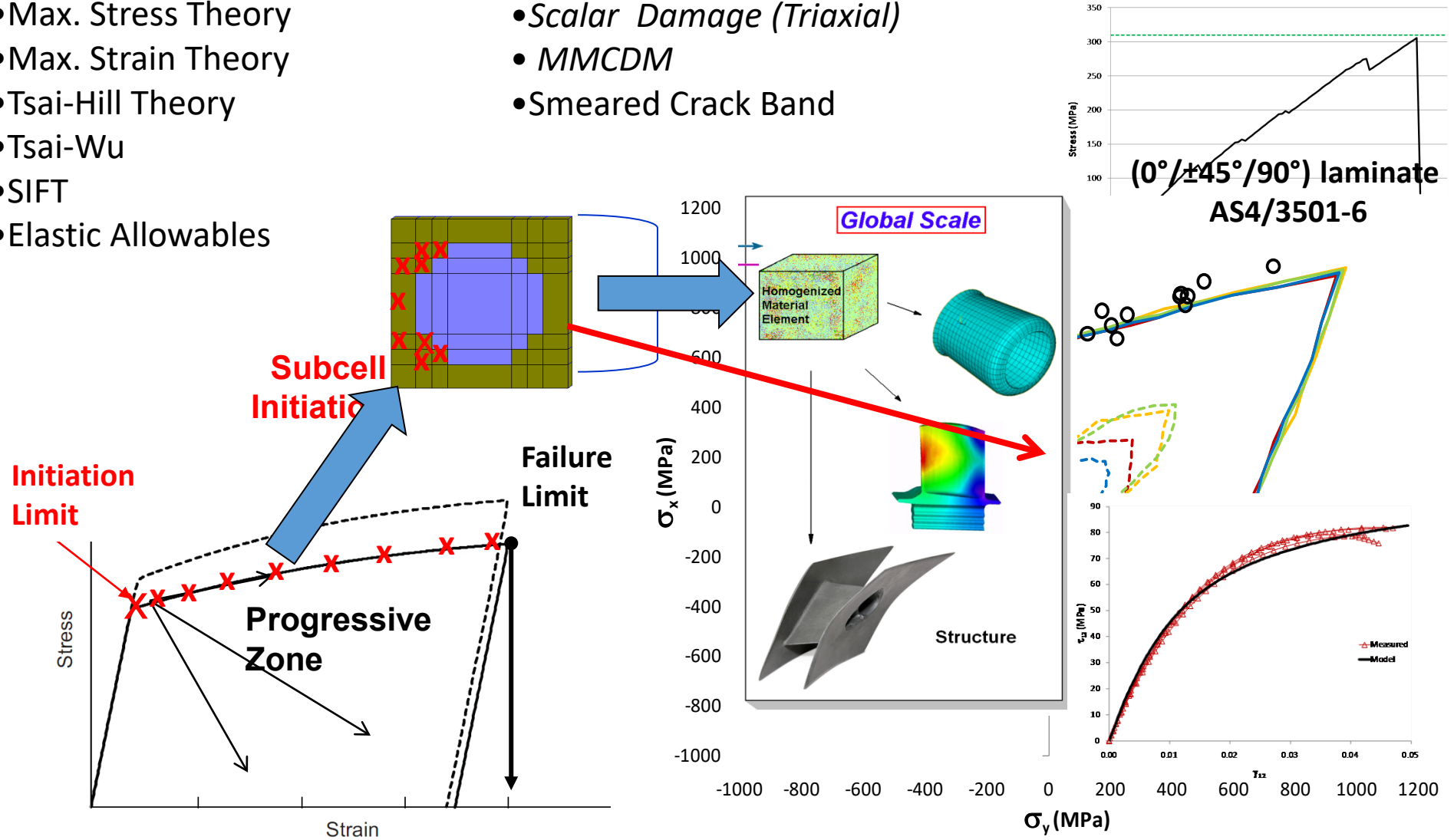


## Subcell Elimination Criterion

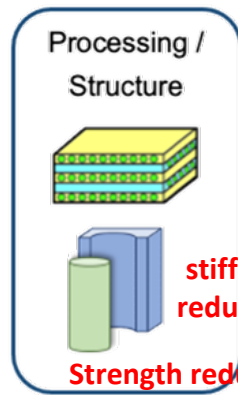
- Max. Stress Theory
- Max. Strain Theory
- Tsai-Hill Theory
- Tsai-Wu
- SIFT
- Elastic Allowables

## Progressive Damage Criterion

- *Scalar Damage (Triaxial)*
- *MMCDM*
- *Smearred Crack Band*



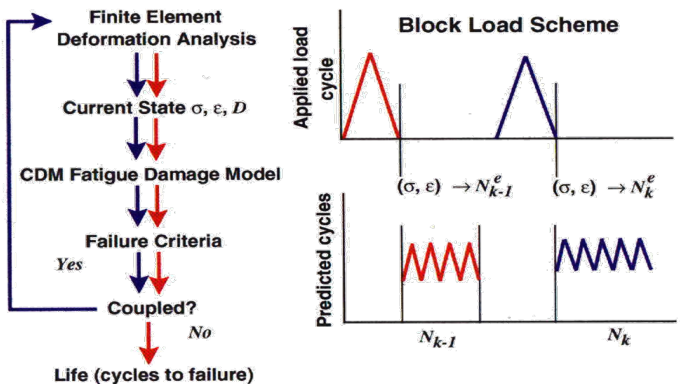
# Coupled Deformation and Damage Methodology Utilized For Fatigue Life Prediction



$$\frac{dD_E}{dN} = [1 - (1 - D_F)^{\beta+1}]^\alpha \left[ \frac{\hat{F}_m}{(1 - D_F)} \right]^\beta$$

$$\alpha = 1 - a \frac{\langle \hat{F}_l - 1 \rangle}{\langle 1 - \hat{F}_u \rangle}$$

## Life Prediction Scheme



- MAC/GMC offers macroscale fatigue life prediction by degrading the stiffness of the matrix and strength of the fiber (microscale damage) – **Physics based Model**
- Created a Recurrent Neural Network surrogate model to improve computational efficiency (**1000X Speedup**)
  - Currently restricted to 8-ply, symmetric CMC
  - Trained on both laminates with random angles and industry-standard layups (5k laminates; > 15K hrs generate training data)



### Fatigue Life Estimator v1.0

**Laminate Definition**

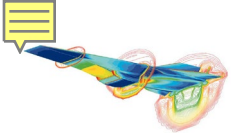
	Value	Lower Bound	Upper Bound
$\theta_1$ (°)	62	-90.0	90.0
$\theta_2$ (°)	-80	-90.0	90.0
$\theta_3$ (°)	75	-90.0	90.0
$\theta_4$ (°)	-5	-90.0	90.0
$V_f$	0.64	0.4	0.7

**Fiber Properties**

	Value	Lower Bound	Upper Bound
$E_L$ (GPa)	643	70.0	700.0
$E_T$ (GPa)	116	70.0	200.0
$\nu_L$	0.33	0.2	0.4
$\nu_T$	0.22	0.2	0.4
$G_L$ (GPa)	133	25.0	200.0
$SU_{11}$ (MPa)	9000	2500.0	9500.0
$SU_{22}$ (MPa)	1500	300.0	2500.0

**Matrix Properties**

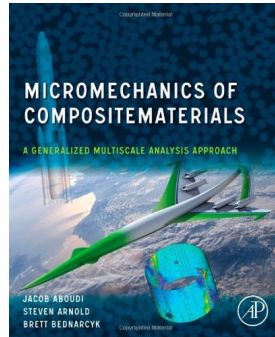
	Value	Lower Bound	Upper Bound
$E$ (GPa)	3.4	2.5	4.5
$\nu$	0.21	0.2	0.45
$\epsilon_1$	0.026	0.015	0.03
$\epsilon_2$	0.025	0.015	0.03
$\beta$	5.6	4.0	12.0
SFL (MPa)	30	15.0	30.0
XML (MPa)	110	80.0	160.0



# Key References



Aboudi, J., Arnold, S.M., and Bednarczyk, B.A. (2013)  
*Micromechanics of Composite Materials: A Generalized Multiscale Analysis Approach*, Elsevier, Oxford, UK., pp 1-984.



## Outline

- 1) Introduction
- 2) Constituent Material Modeling
- 3) Fundamentals of the Mechanics of Multiphase Materials
- 4) The Method of Cells Micromechanics
- 5) The Generalized Method of Cells Micromechanics
- 6) The High Fidelity Generalized Method of Cells Micromechanics
- 7) Multiscale Modeling of Composites
- 8) Fully Coupled Thermomechanical Analysis of Multiphase Composites
- 9) Finite Strain Micromechanical Modeling of Multiphase Composites
- 10) Micromechanical Analysis of Smart Composite Materials
- 11) Higher-Order Theory for Functionally Graded Materials
- 12) Wave Propagation in Multiphase and Porous Materials
- 13) Micromechanics Software

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“Written with both *students and practitioners* in mind and is coupled with a fully functional MATLAB code to enable solution of technologically relevant micromechanics problems. The many illustrative example problems and exercises highlight key concepts and rely heavily on the MATLAB code”



## Table of Contents

1. Introduction
2. Lamination Theory Using Macromechanics
3. Closed Form Micromechanics
4. Failure Criteria and Margins of Safety
5. The Generalized Method of Cells (GMC) Micromechanics Theory
6. The High-Fidelity Generalized Method of Cells (HFGMC) Micromechanics Theory
7. Progressive Damage and Failure

## Features:

- Thermoelastic Material Behavior
- Emphasis on Local fields via Strain and Stress Concentration Tensors;
- General **MATLAB open-source code provided**
  - four micromechanics theories MT, MOC, GMC, HFGMC
  - four failure criteria, along with consistent treatment of Margins of Safety (MoS)
  - Emphasis on PMC & CMC, order and disorder microstructures
  - <https://github.com/nasa/Practical-Micromechanics>
- Extensive Practical Examples; ~ 15 Exercises /Chapter (Solution Manual available to professors)