

Integration of Structural Analysis and Manufacturing Process Planning for Global Optimization with Automated Fiber Placement

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Background and Introduction

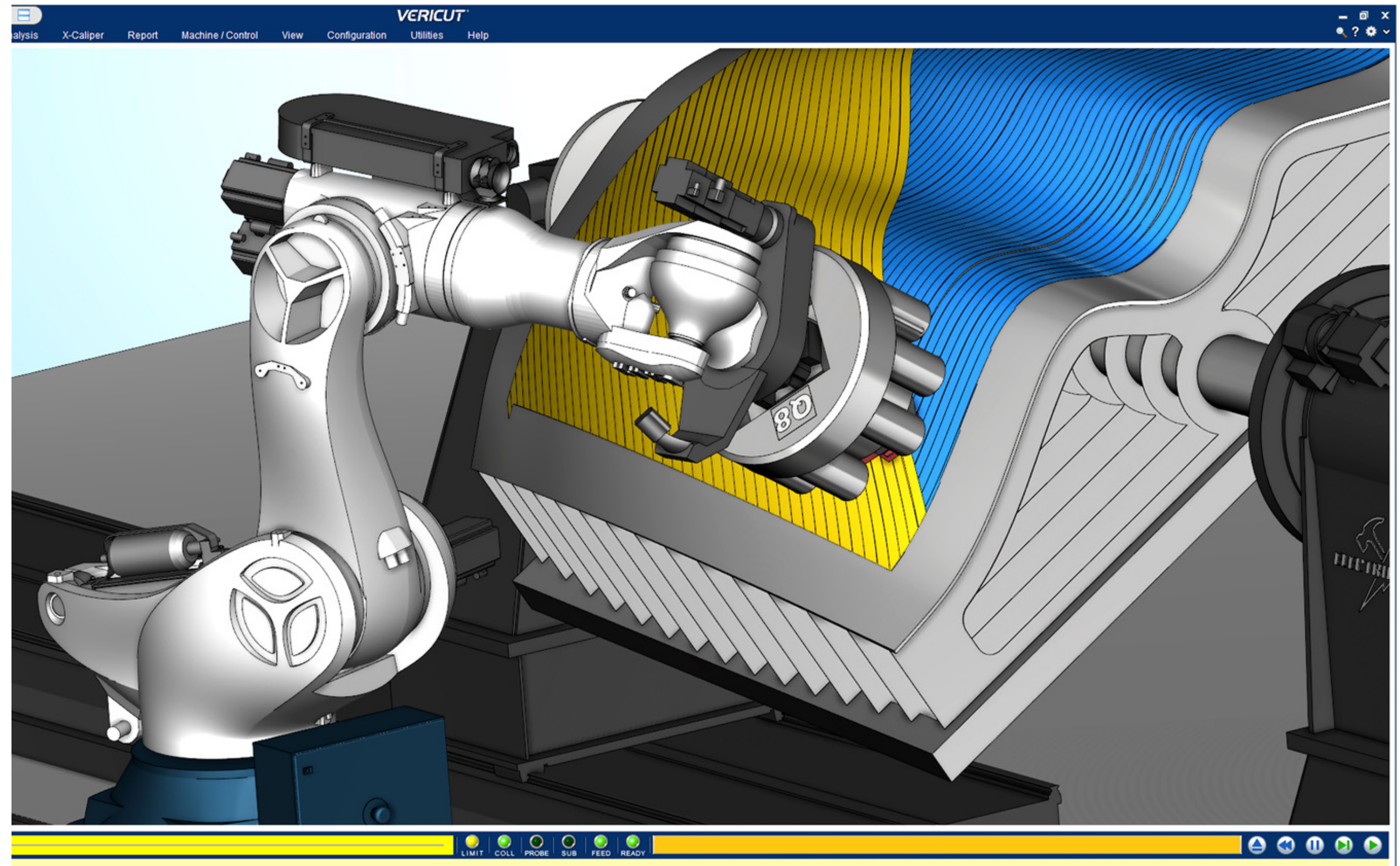
Software Integration

Optimization Approaches

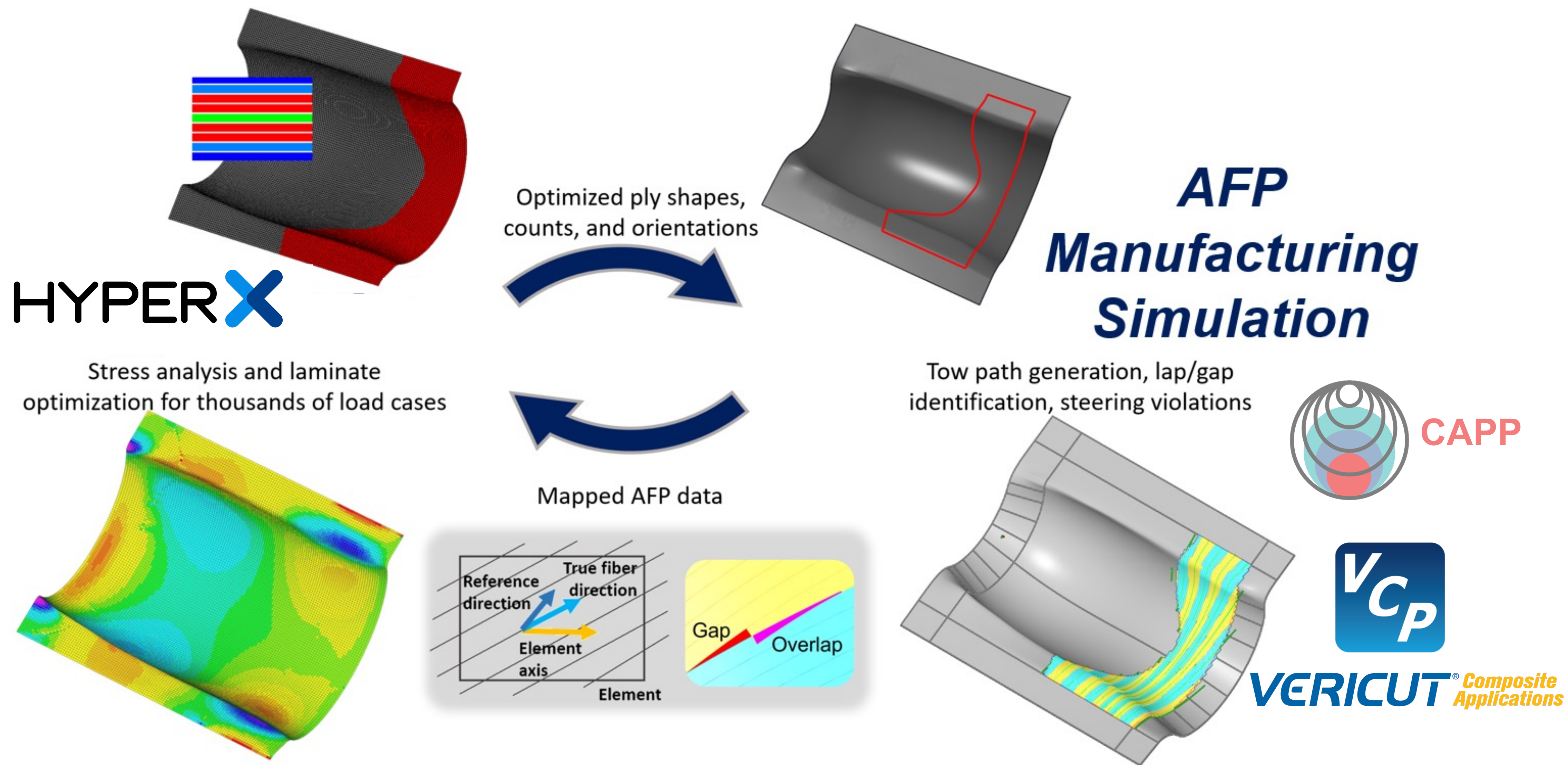
Verification Example with Wind Blade Section

Conclusions

- NASA's Hi-Rate Composite Aircraft Manufacturing (**HiCAM**) project is focused on addressing an aviation industry need for more rapid production of composite aircraft to meet increasing global demand for lightweight transport aircraft.
- Much of the technology being considered in HiCAM uses Automated Fiber Placement (**AFP**)
- The Design For Manufacturing (**DFM**) HiCAM task is focused on development of software to optimize AFP structures for weight and manufacturability

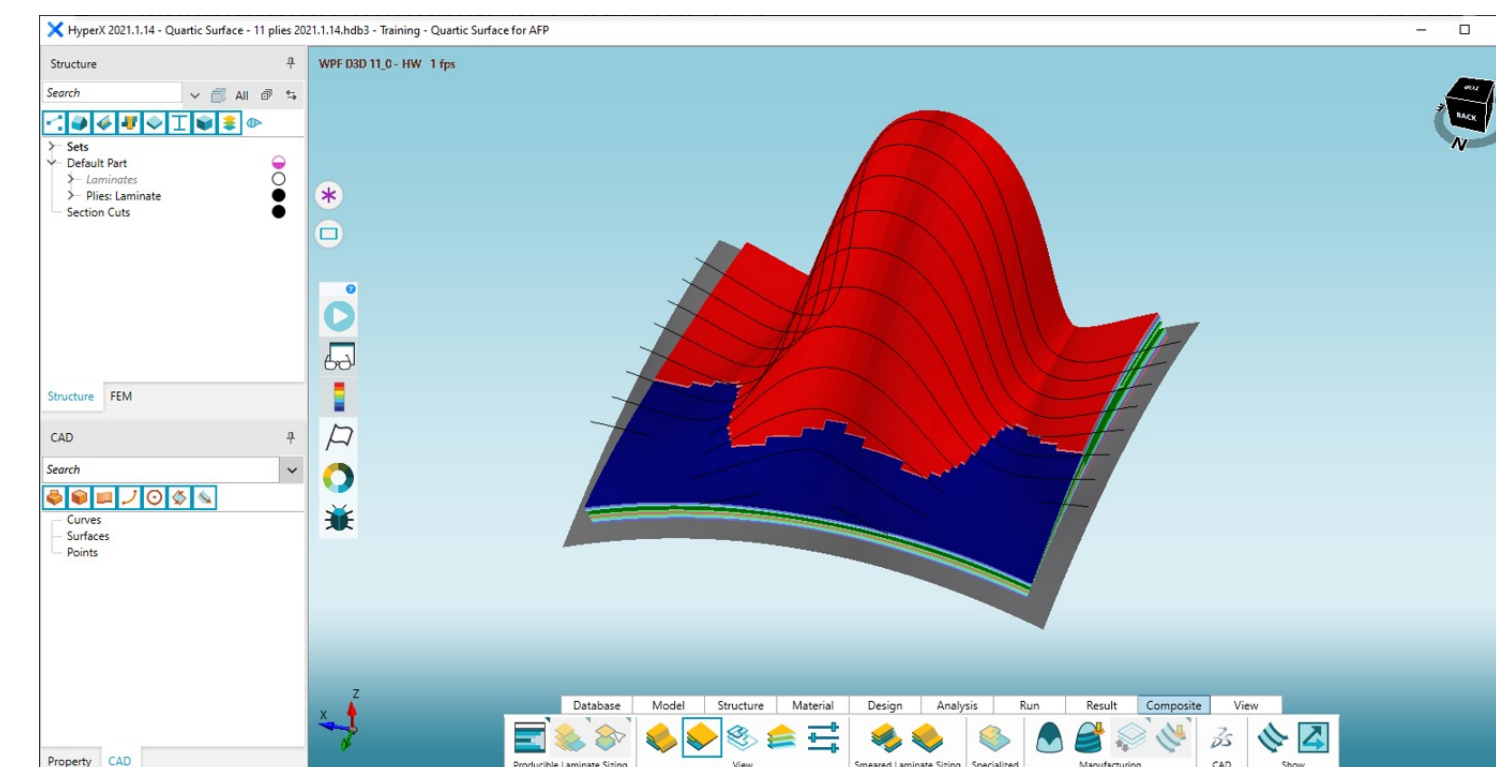


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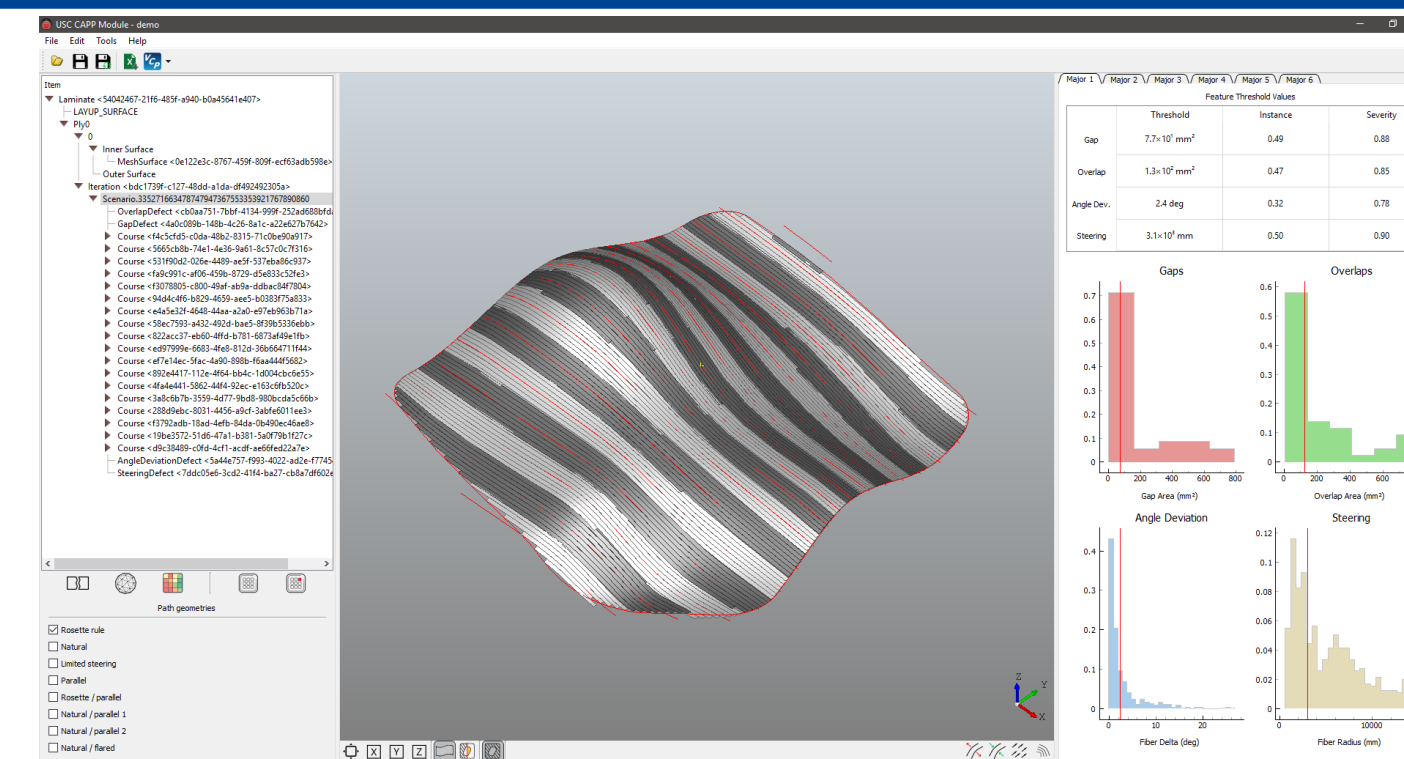




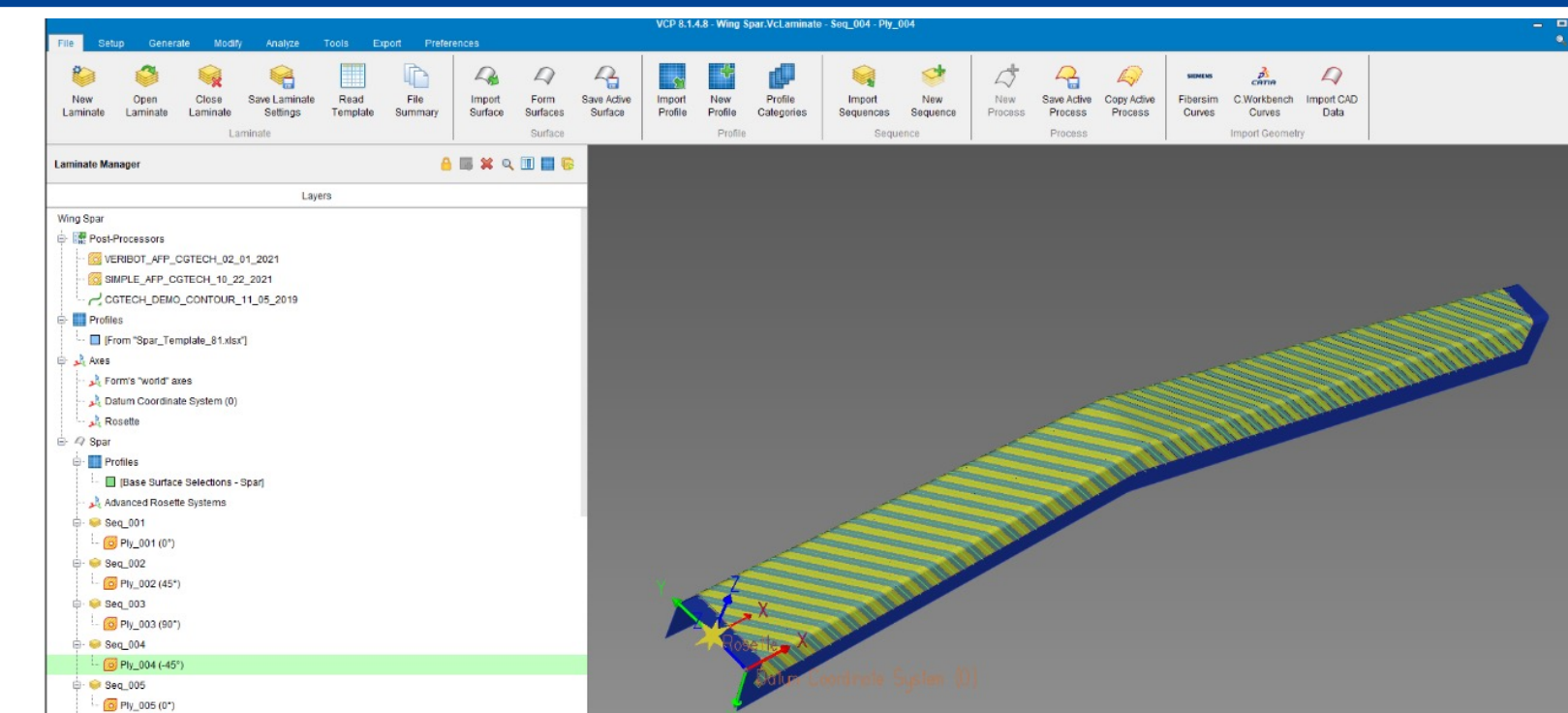
HyperX performs stress analysis and composite optimization for thousands of load cases and a variety of failure criteria. Generates optimized ply shapes, stacking sequences, and ply counts.



The Computer Aided Process Planning (CAPP) software automates process planning functions within AFP. Combining the automation with optimization schemes reduces the occurrence of geometry related fiber defects throughout a laminate.



Vericut Composites Programming (VCP) provides a suite of path planning tools to generate machine motion for multiple composites processes. VCP also contains tools to analyze these processes as well as pass data for analysis to other software.



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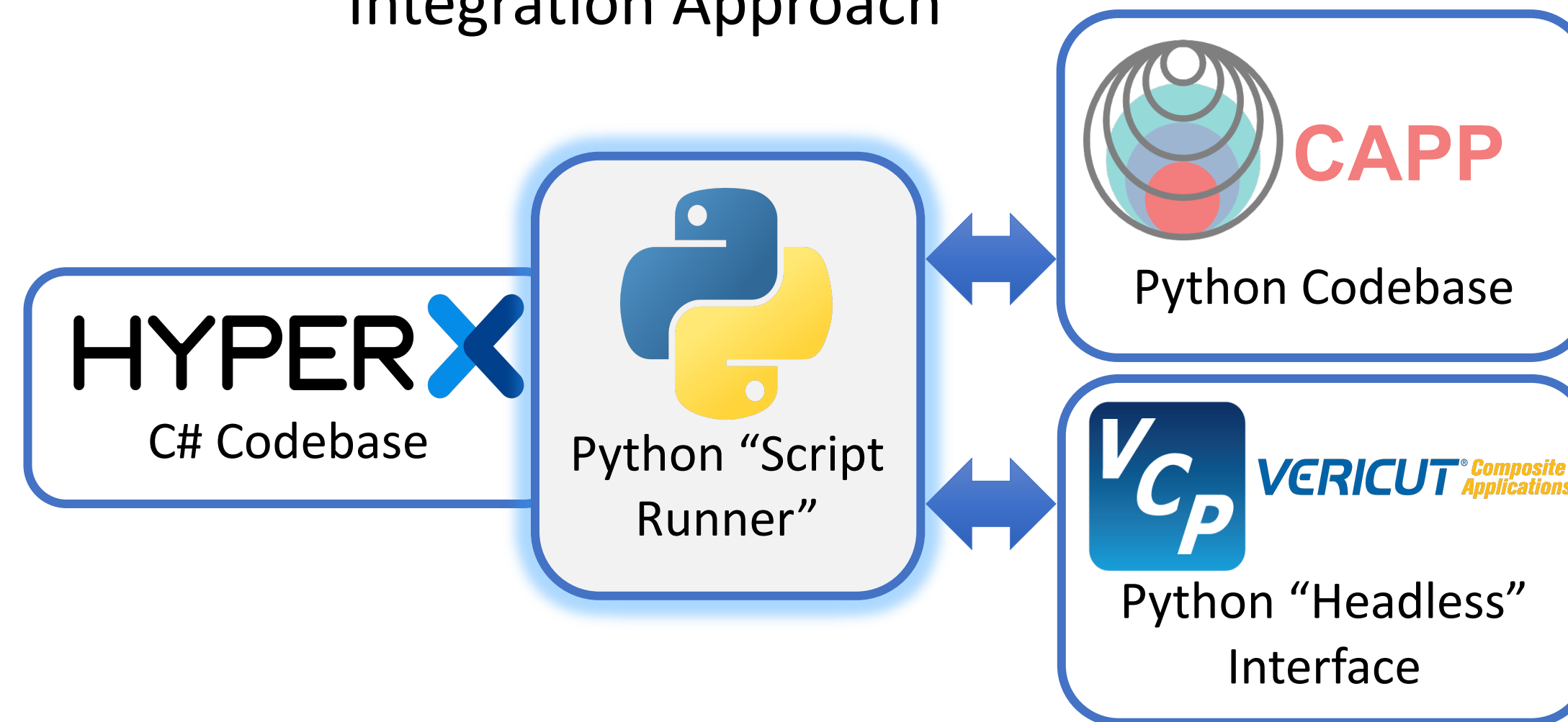
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Conclusions

- Integration with the three software accomplished via Python scripting
- A native integration platform was developed on top of HyperX – the “Script Runner”
- Allows custom UI development and process automation for HyperX as well as external tools

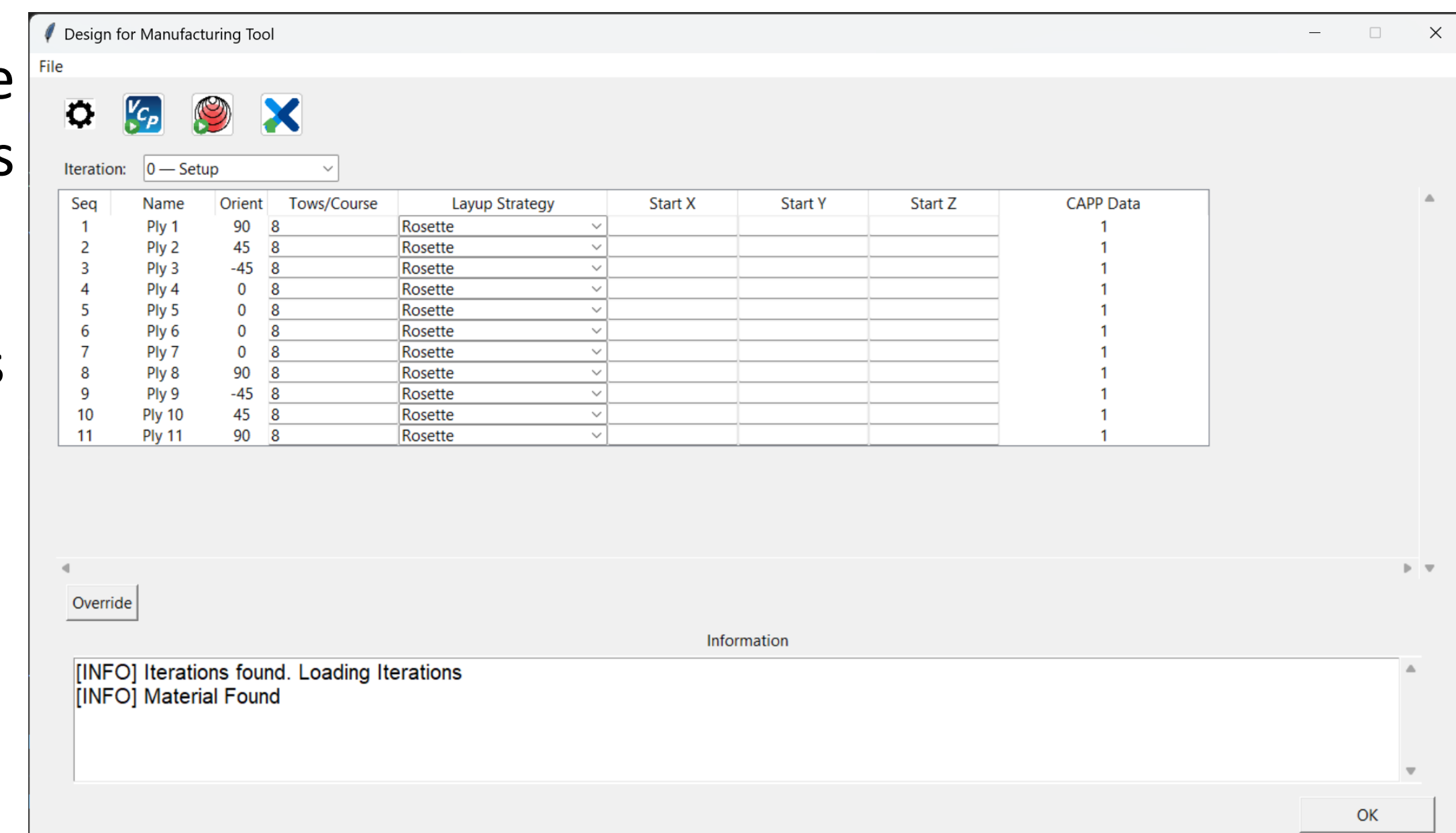
Integration Approach



Custom UI

Execute
external tools

Plies



Background and Introduction

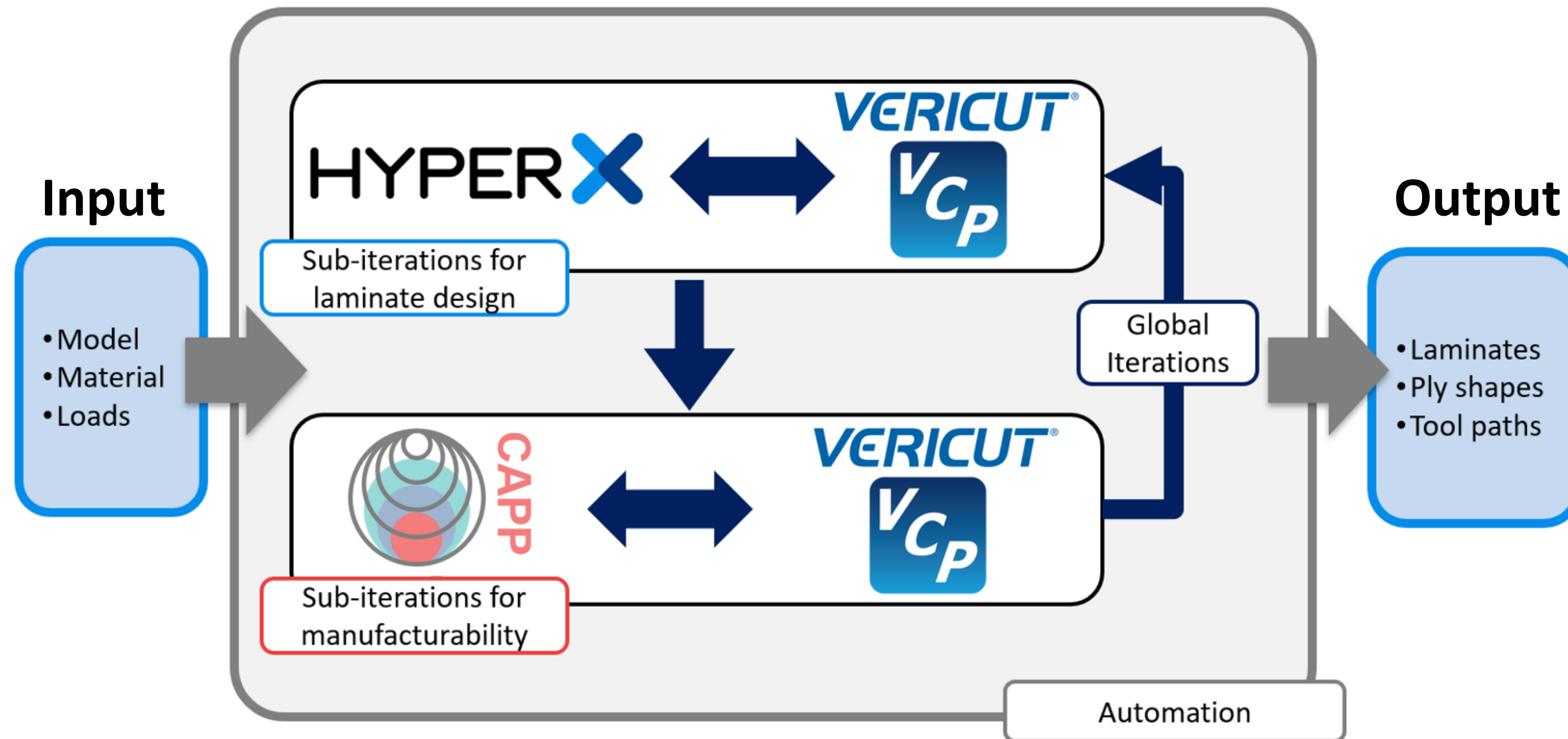
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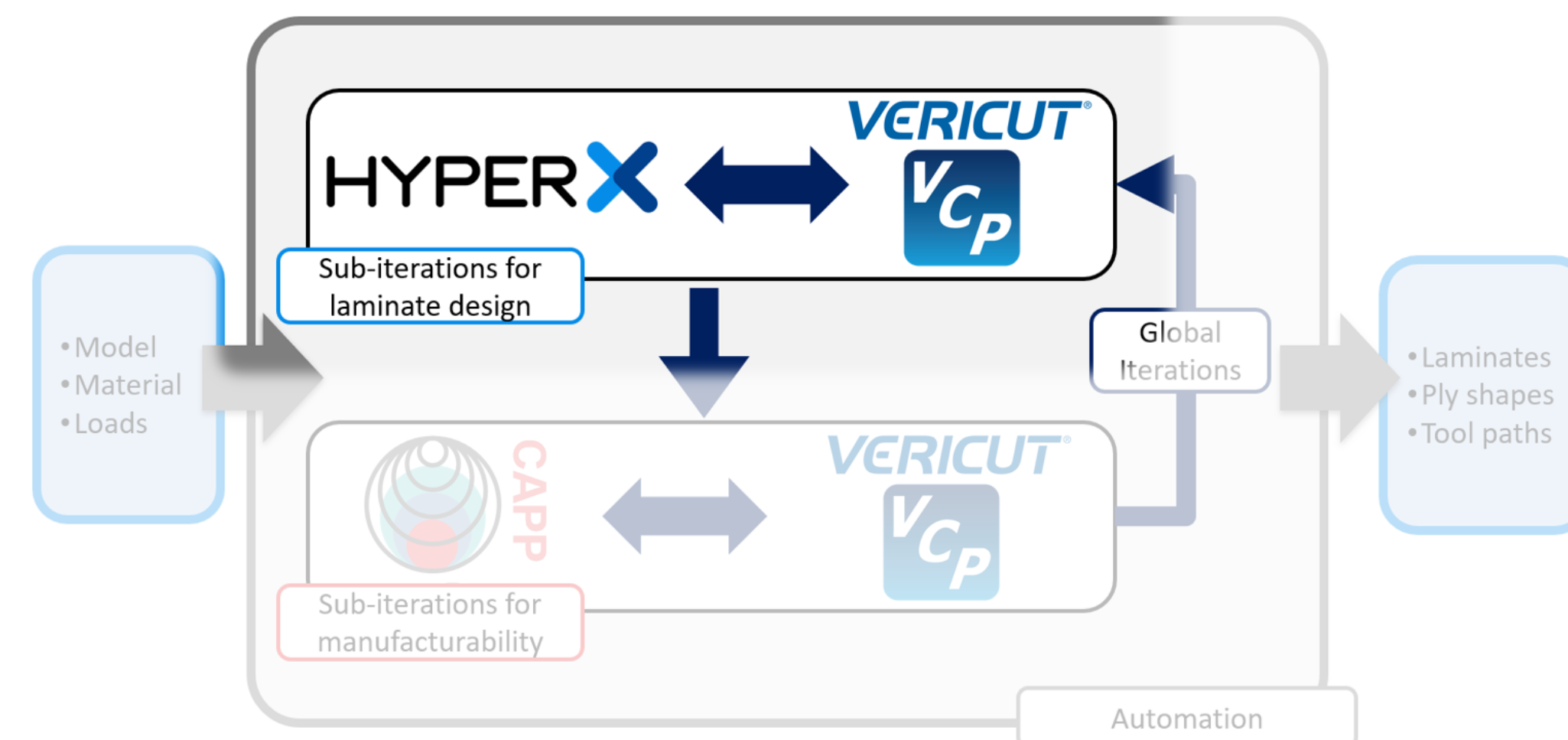
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Conclusions

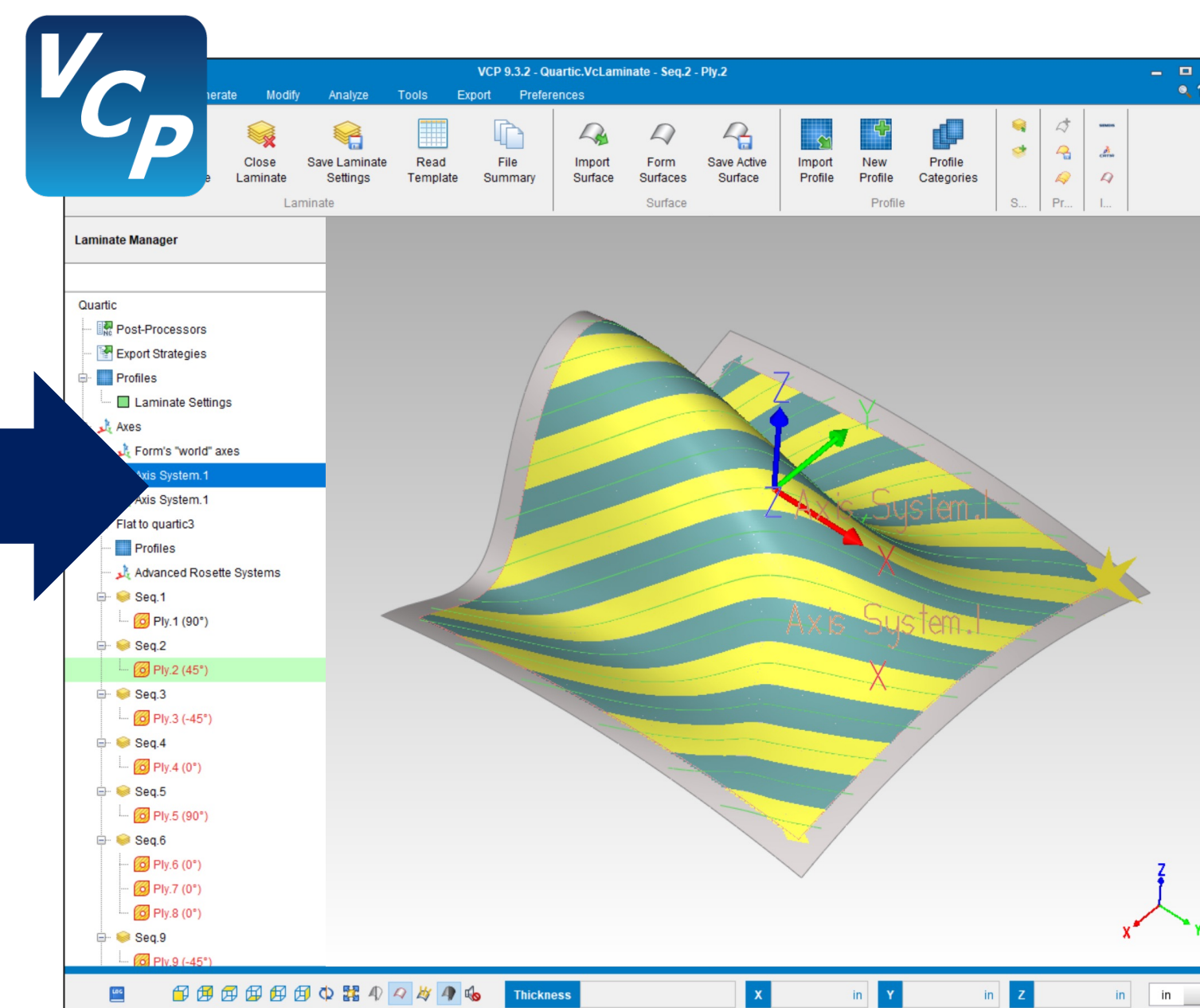
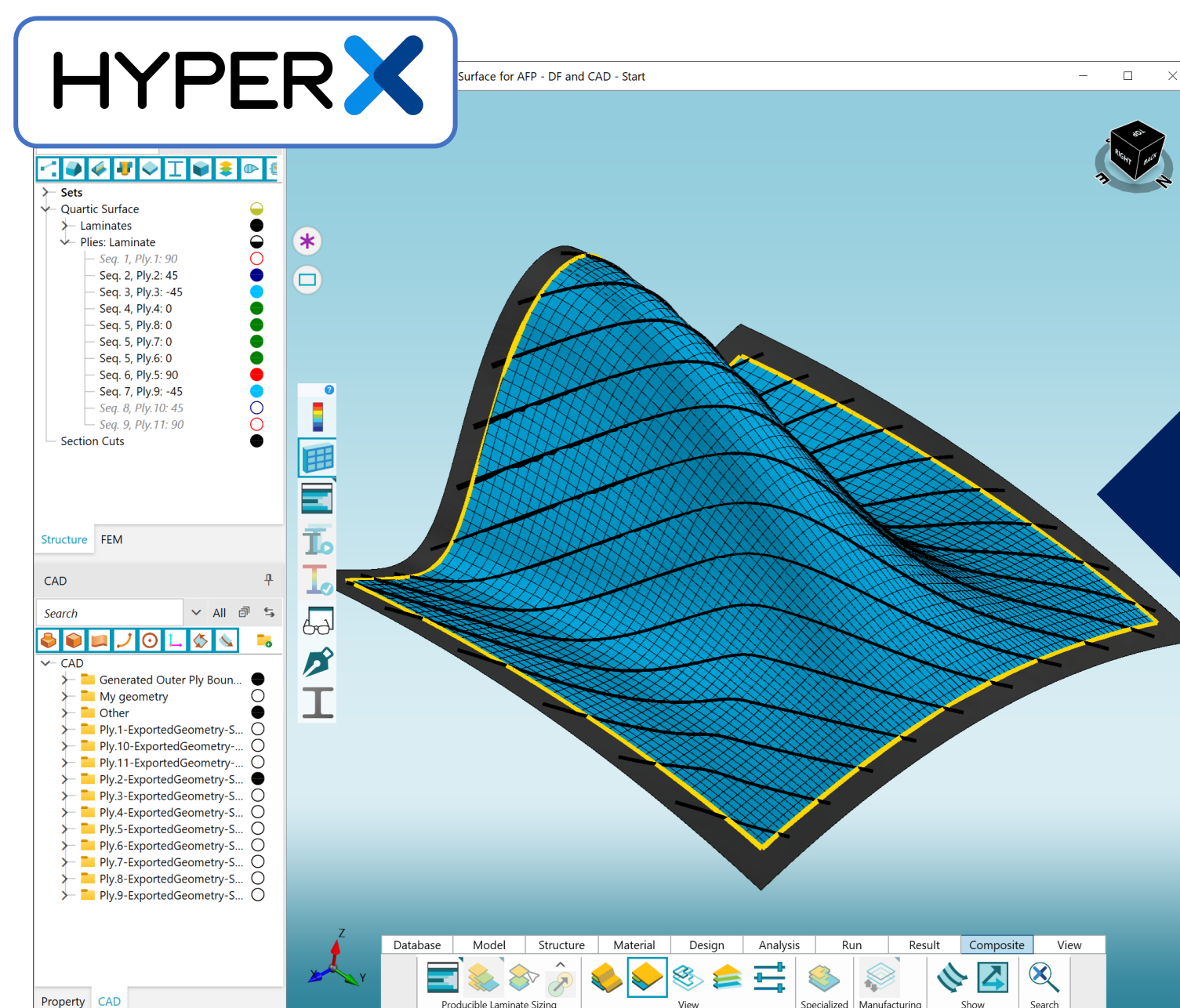
- Objective: develop laminate design that satisfies both stress and manufacturing reqs.
- Bi-level strategy selected due to complexity and runtime of software involved
 - Lower level: HyperX and CAPP iterate with VCP to optimize for stress and manufacturing respectively
 - Upper level: global iterations to converge the solution



- Objective: get HyperX analysis sync'd up with as-manufactured fiber directions
- Initially, HyperX assumes fiber directions based on FEM material axis
- VCP provides accurate fiber directions for each ply and HyperX re-optimizes the laminate



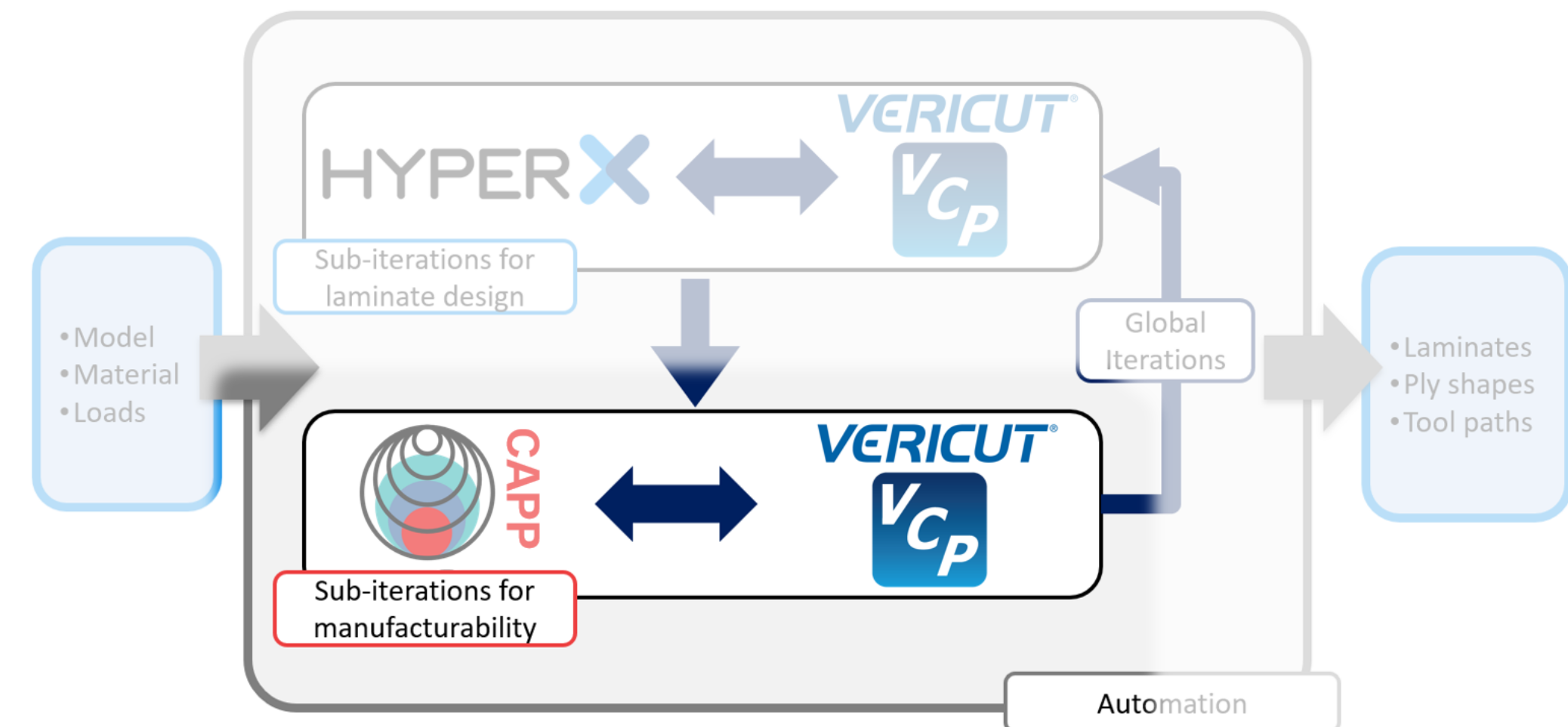
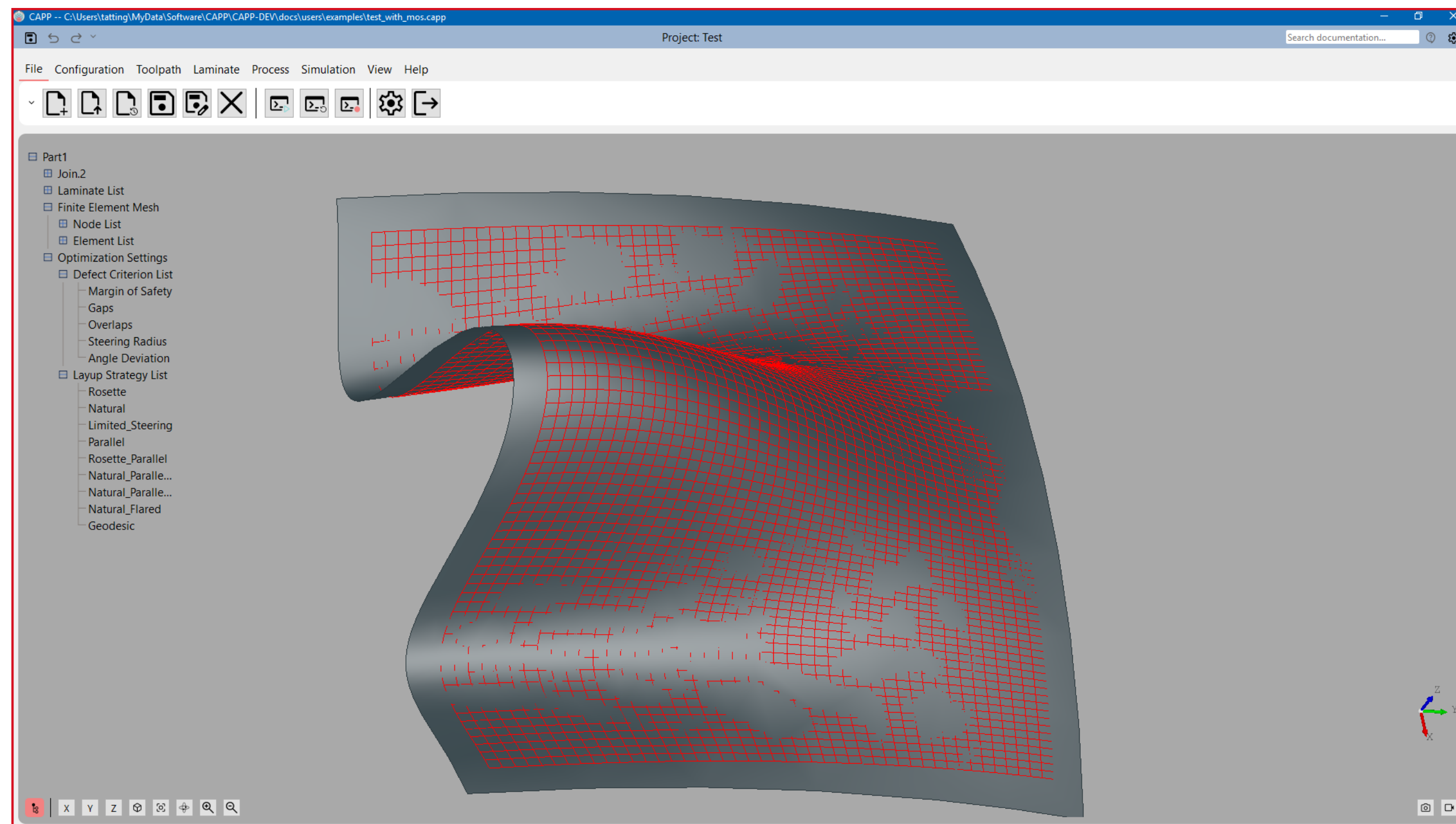
Uses FEA loads to optimize ply shapes, counts, and orientations



Generates tow paths on the tool surface within the prescribed ply boundaries

- The Computer Aided Process Planning (CAPP) software matchmakes the design and manufacturing parameters.
- The goal is to minimize the deviation between the as-manufactured and the as-designed structure..

CAPP Software Interface



- The principal functions are starting point, layup strategy and laminate-level optimizations.

Design for Manufacturing Tool

VCP CAPP HyperX Material

Mode: Basic

Strategy: Menu

Margin of Safety Manufacturability

50 50

Uniform Coverage

Fiber Steering Fiber Angle

MOS Threshold:

Gap 12.50 5 in.

Overlap 12.50 5 in.

Angle Dev. 12.50 5 Deg.

Steering 12.50 20 in.

OK

Design for Manufacturing Tool

VCP CAPP HyperX Material

Mode: Advanced

Strategy: Menu

Margin of Safety Manufacturability

50 50

Uniform Coverage

Fiber Steering Fiber Angle

MOS Threshold: 5

Sev. Inst. Thres.

Gap 6.25 = 6.25 5.0 in.

Overlap 6.25 = 6.25 5.0 in.

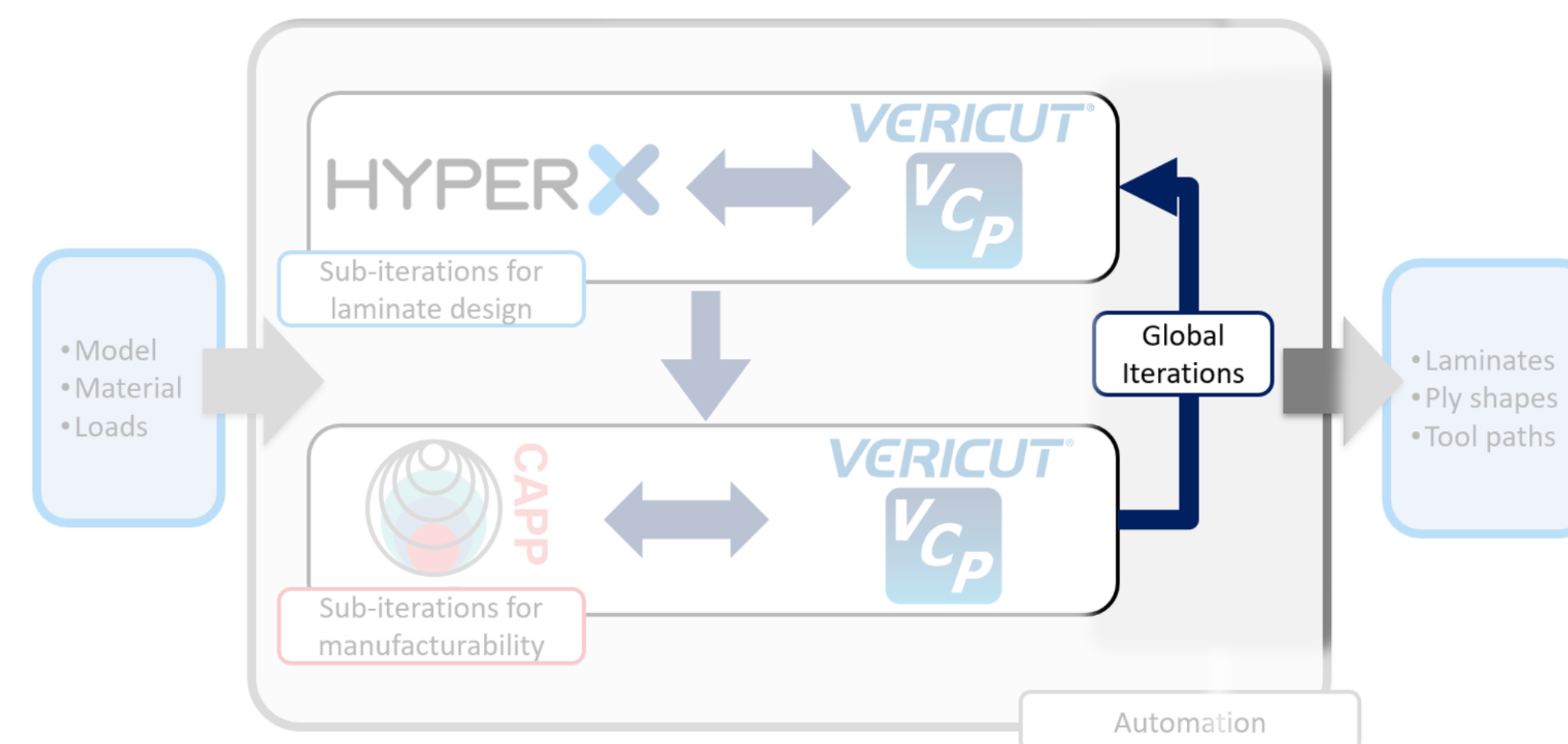
Angle Dev. 6.25 = 6.25 5.0 Deg.

Steering 6.25 = 6.25 20.0 in.

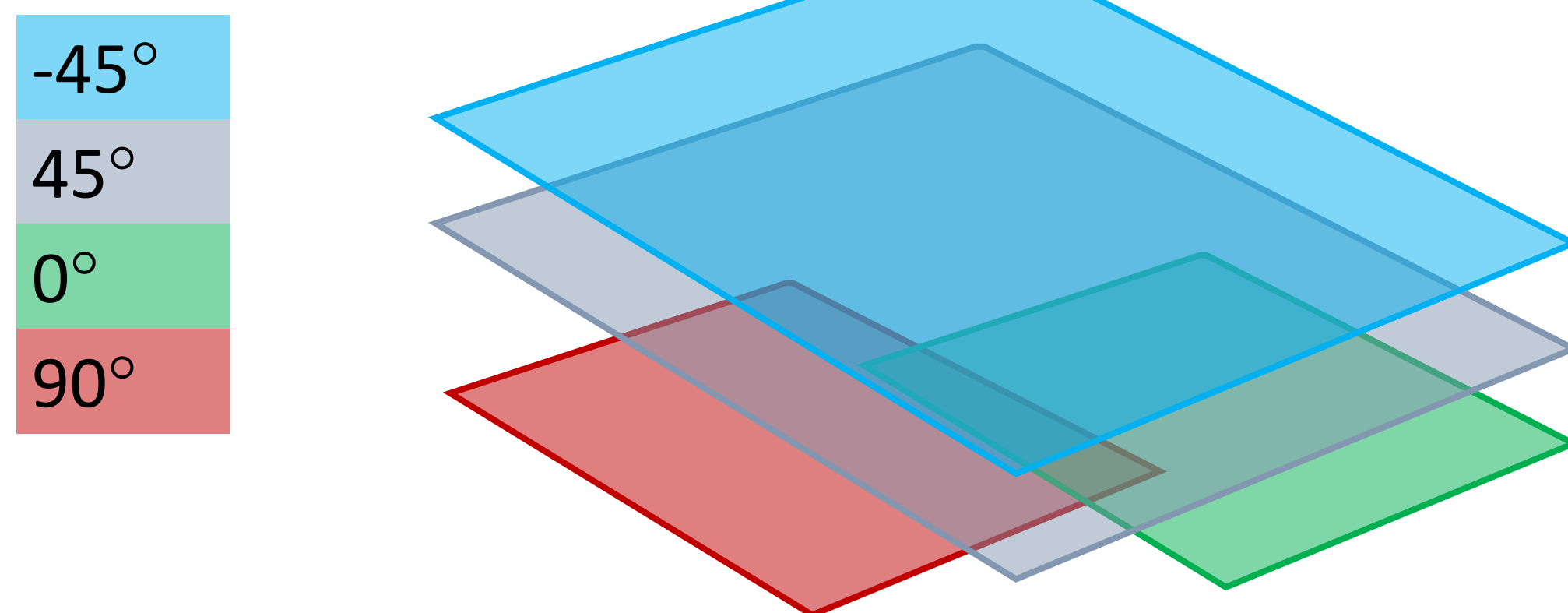
Rebalance Clear All

OK

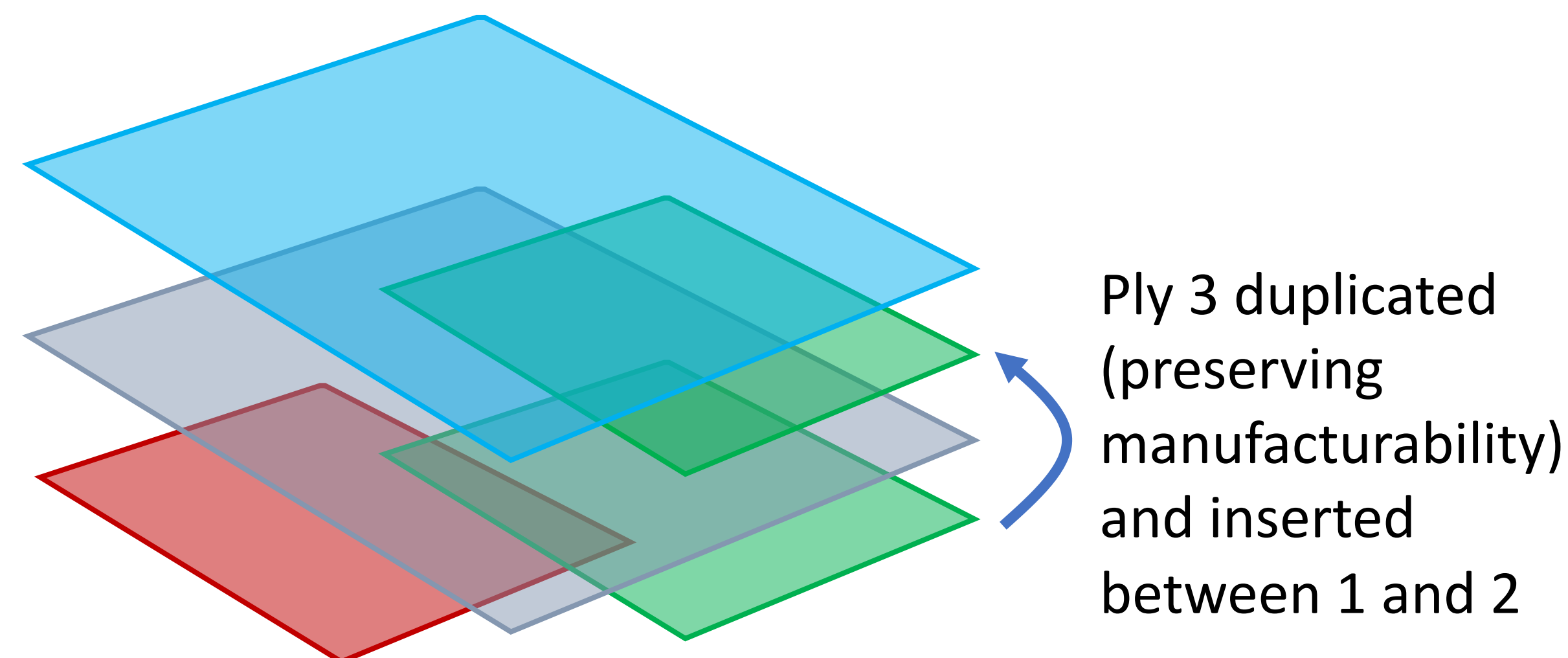
- HyperX or CAPP can make changes to the plies that the other software does not know about
- Global level iterations needed to converge the HyperX-VCP and CAPP-VCP sub-iterations
- After each sub-iteration, the global optimizer can insert new manufacturing-optimized plies to resolve negative stress margins, as needed



Iteration N Plies



Iteration N+1 Plies



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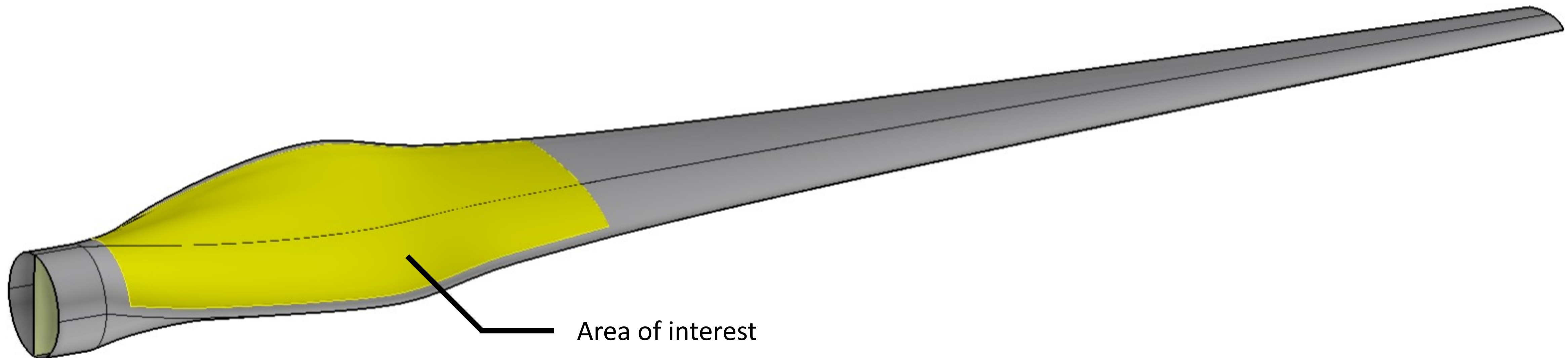
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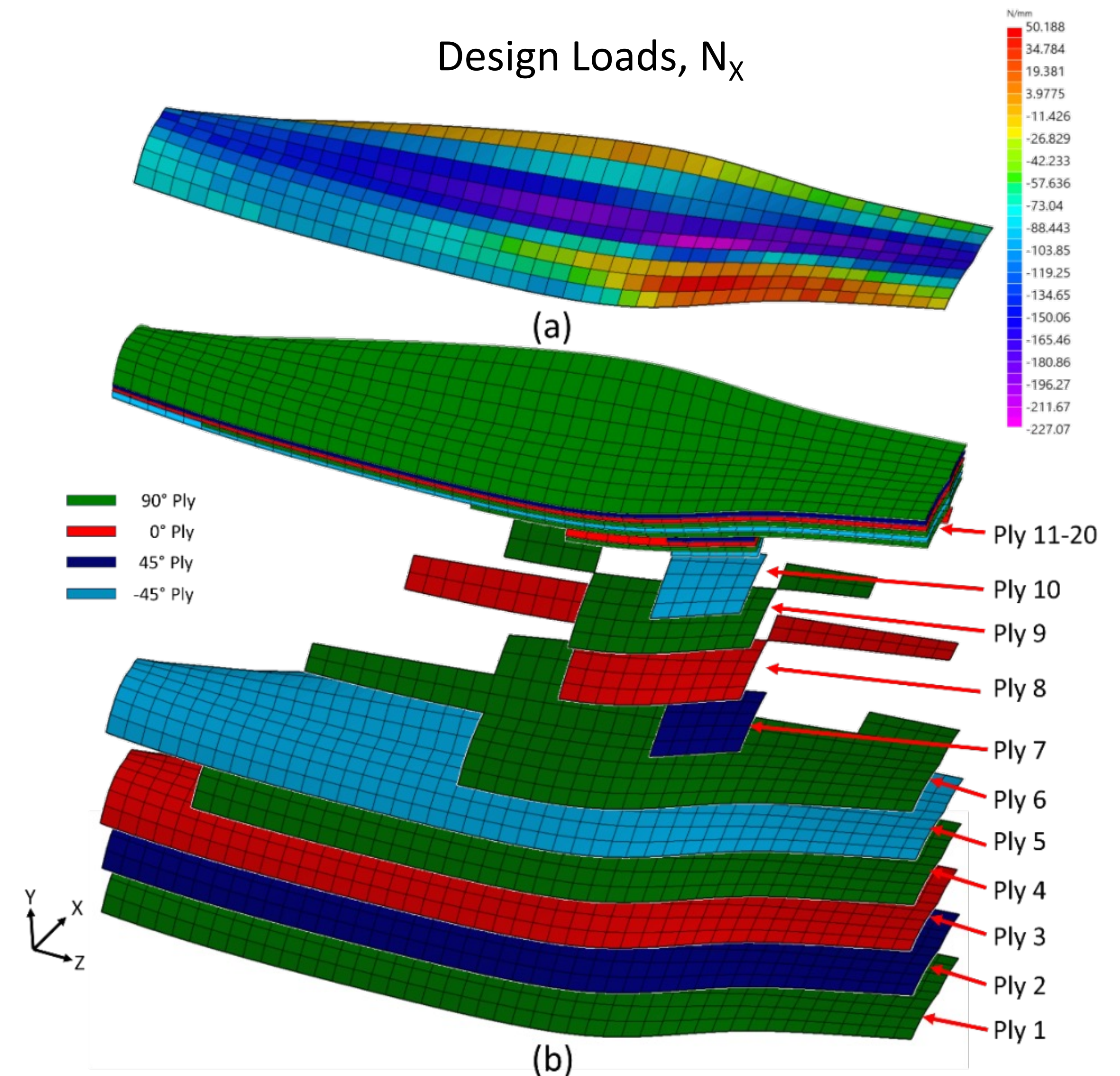
Verification Example with Wind Blade Section

Conclusions

- Example demonstrates the HyperX-VCP iteration (CAPP-HyperX iteration still in work)
- 3m wind blade
- 1 load case (pressure distribution)
- High contour region selected as a good challenge problem

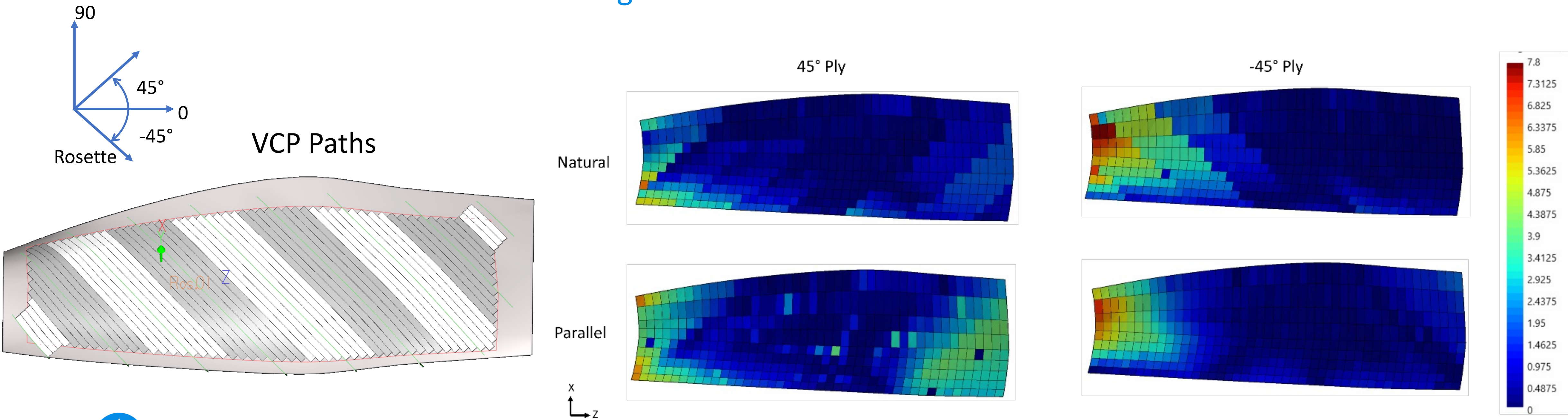


- Plies generated in HyperX
 - 20 Plies in total
 - Symmetrical Plies
- The internal stresses of the wind turbine blade
 - Stress concentrations at bottom right
 - Results on a thicker laminate in the stress concentrations
- Margin of safety, delta, and % differences was measured in the full plies.



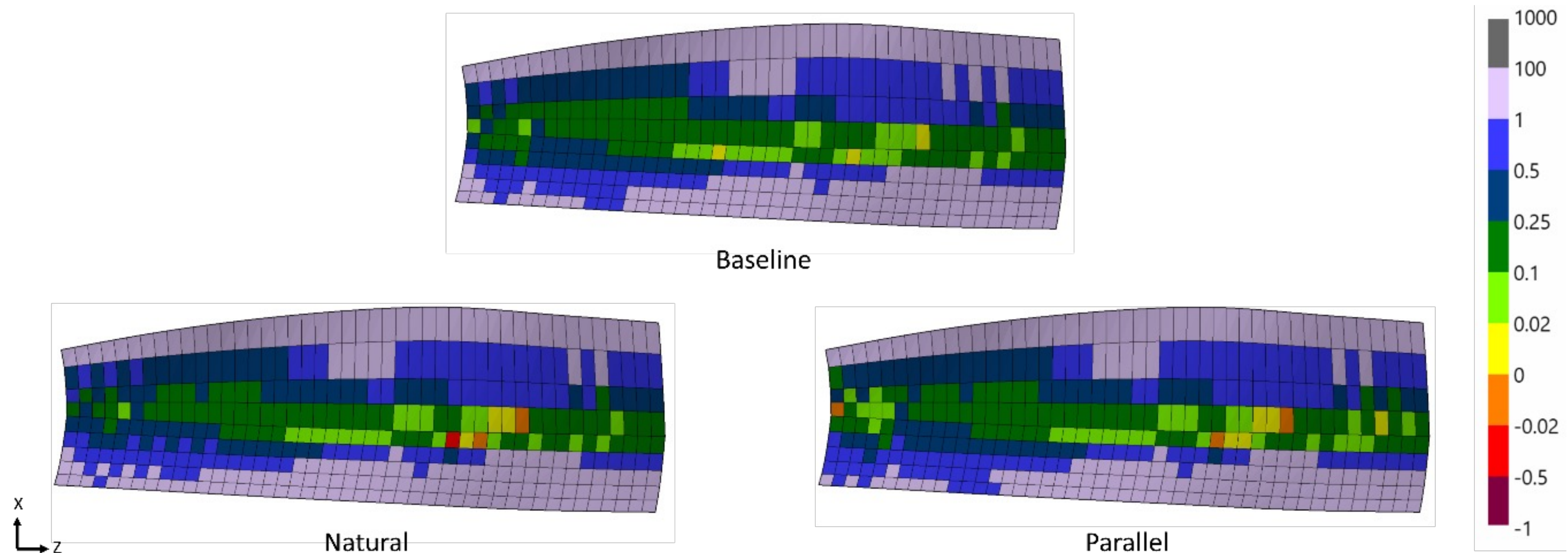
- Angle deviation in the complete plies shows a deviation between 0° to 7°.
- Average deviation - 1.0 ° – 1.6° in the 45 and -45 deg
- Large Deviations in:
 - Left hand side of the model
 - minor deviation in the bottom right

Tow path	Angle	Ply Number	Average Deviation	Maximum Deviation
Natural	45	2,19	1.52	6.28
	-45	5,16	1.03	7.14
	90	3,18	0.52	3.3
	0	1,20	1.04	7.18
Parallel	45	2,19	1.67	6.47
	-45	5,16	1.25	7.75
	90	3,18	0.98	5.9
	0	1,20	0.68	4.61

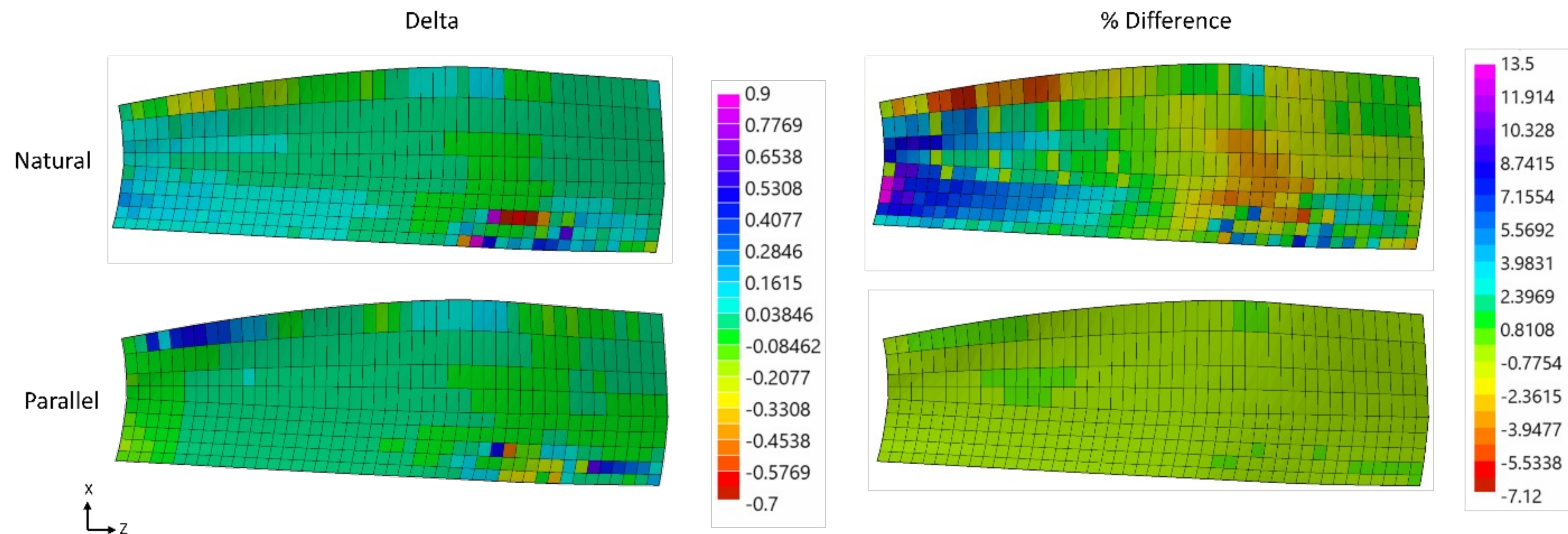


- The baseline of the wind turbine blade shows no negative margins of safety
- Negative margins are produced by both the natural and parallel tow path strategies
 - The natural paths produced negative margins at the center right
 - The parallel tow paths produced negative margins at the center left and right of the wind blade

Margin of Safety



- For the “as-manufactured” fiber directions, the delta for some elements does not show a large decrease or increase in MS.
- The % difference shows that the change of MS was significant in certain places but did not significantly impact the resulting MS.
- The negative MS experienced relatively small changes in MS.
 - Small changes in misalignment can affect the overall safety of the composite laminate.



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Conclusions

- Presented approach for integration of stress analysis/optimization (HyperX) with AFP manufacturing process planning (CAPP), including iteration with AFP path generation (CGTech)
- Optimization results in laminates that are weight-optimum and also optimized for manufacturability
- Demonstrated the approach on a section of a 3m wind blade. Automated iterations used to:
 - Generate AFP paths on structure
 - Automatically update stress analysis
 - Add plies to structure to alleviate negative margins caused by fiber deviations

Thank You!

Questions?

Acknowledgements

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