



Lunar Browser trajectory tool

A NASA Ames Flight Dynamics team tool

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For more detailed information on the Lunar Browser tool, see this AIAA SciTech paper:

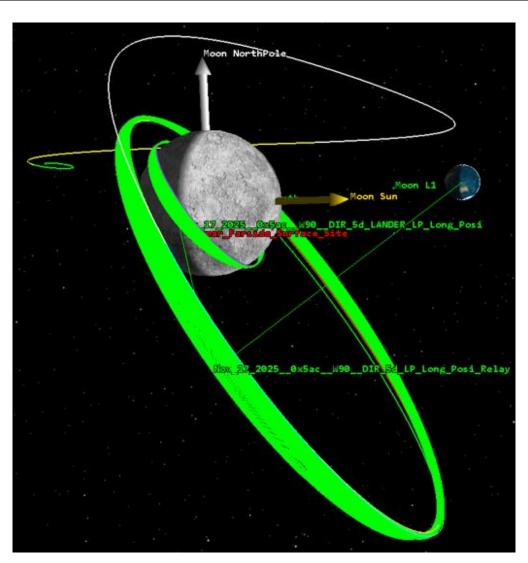
Transfers from TLI to Lunar Frozen Orbits with Applications to NASA's CLPS & ARTEMIS Programs (aiaa.org)



To learn more about the NASA Ames Flight Dynamics Team:

Ames Flight Dynamics (FD) - NASA





Genova, A. L., Morrison-Fogel, D., & Levinson-Muth, P., "Transfers from TLI to Lunar Frozen Orbits with Applications to NASA's CLPS & Artemis Programs," AAS/AIAA Space Flight Mech. Mtg., Orlando, FL, Jan. 8-12, 2024



NASA Ames Flight Dynamics Team

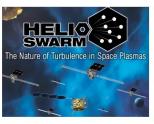


Full mission lifecycle support

NASA Life-Cycle FORMULATION TO IMPLEMENTATION Phases **Project Life-Cycle** Pre-Phase A: Phase A: Phase B: Phase C: Phase D: Phase E: Phase F: Phases Concept Studies Concept and Preliminary Design Final Design and System Assembly, Operations and Closeout Technology and Technology Fabrication Integration & Test, Sustainment Development Completion Launch & Checkout

Credits: Adapted from NASA NPR 7120 5E

Flight heritage



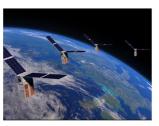
HelioSwarm

HelioSwarm will unlock the mystery of how turbulence heats space plasma, which is matter that makes up key elements in the Universe, like the Sun, stars, solar wind and even the Earth's upper atmosphere.



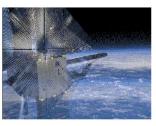
BioSentinel

Launched on Artemis-1, BioSentinel is a 6U CubeSat that carries a biology experiment into deep space for the first time in 50 years.



Starling

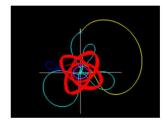
Starling will demonstrate technologies to enable multipoint science data collection by several small spacecraft flying in swarms.



Advanced Composite Solar Sail System (ACS3)

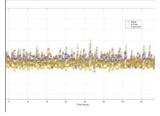
ACS3, technology demonstration uses composite materials in its novel, lightweight booms that deploy from a Cubesat.

Four Main Core Areas



Trajectory Design

Trajectory design is one of the main activities performed by the Ames Flight Dynamics team. Trajectories are designed by capitalizing on the fundamental laws of orbital mechanics to achieve a given set of mission objectives.



Navigation

The Ames Flight Dynamics team supports space missions from early concept studies through flight operations. In our most recent past mission support activity, Starling's and Biosentinel's mission operations ran from 2023.



Orbit Analysis

The FD team conducts all the analysis related to any spacecraft mission requirements. This includes producing valuable data for other subsystems, and studying the orbit evolution and payload performance with high fidelity propagations.



Formation Flying

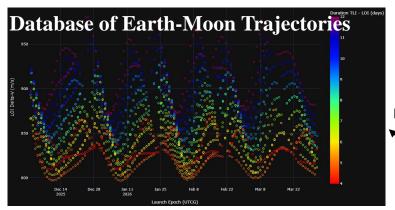
The focus of formation flying is to maintain a targeted orbit configuration of various spacecraft. Having multiple satellites fly in a specific geometry avoids the technical and financial challenge of building one satellite of equivalent size.

Innovative in-house tools

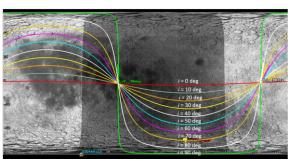
- Trajectory Browser
- Mauna
- Vector Analysis Tool
- Flite
- ROMA
- Lunar Browser







Lunar Browser can be operated via GUI with filters & logic control



Earth-to-Moon transfer trajectory visualizations & animations

Direct output

Interface

Applicable studies

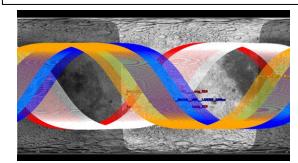
Lunar Browser

Populates Global Solution Space of Earth-Moon transfer trajectories & communication windows over multi-year period to evaluate complex and diverse CLPS requirements, proposals, and missions

>100,000 trajectories generated to date

Lunar Landing Windows

Extended solutions with high-fidelity phasing lunar orbits preceding lighting-constrained landings

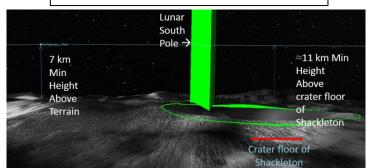


Lunar Terrain Analysis

Applicable

studies

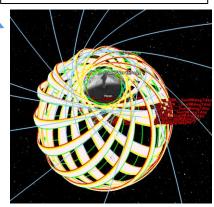
Highest resolution data available used with lighting, timing, & other constraints



Retrograde Northeast Solution Family: *RAAN (Moon-Sun) ~ 120 deg *Approx. 19-day wait time in lunar orbit before dawn landing on lunar farside *Apolune of retrograde lunar elliptical forzen orbit is positioned on the Earth-side during daylight operations *Retrograde Southeast Solution Family: *RAAN (Moon-Sun) ~ 270 deg *Approx. 7-day wait time in lunar orbit before dawn landing on lunar farside *Apolune of retrograde lunar elliptical forzen orbit is positioned on unar farside *Apolune of retrograde lunar elliptical forzen orbit is positioned on lunar farside during daylight operations

Relative Motion

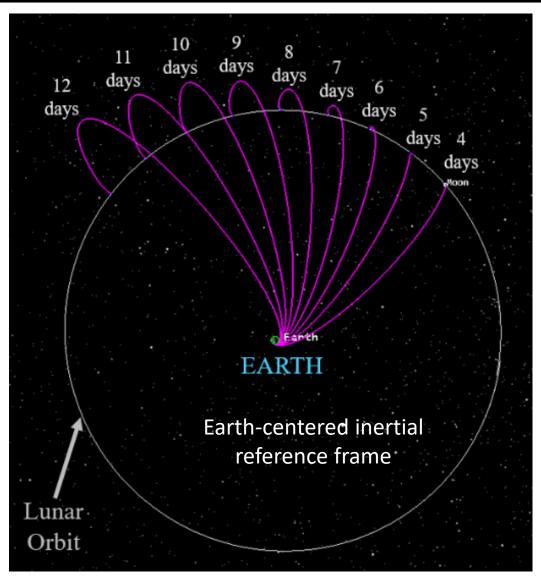
collision assessment & prevention

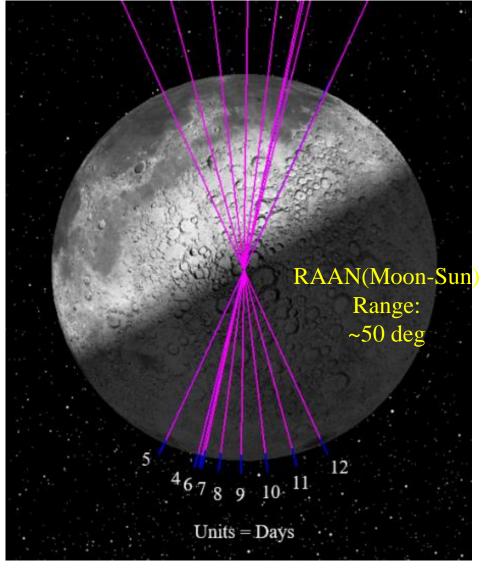




TLI to LOI transit duration range





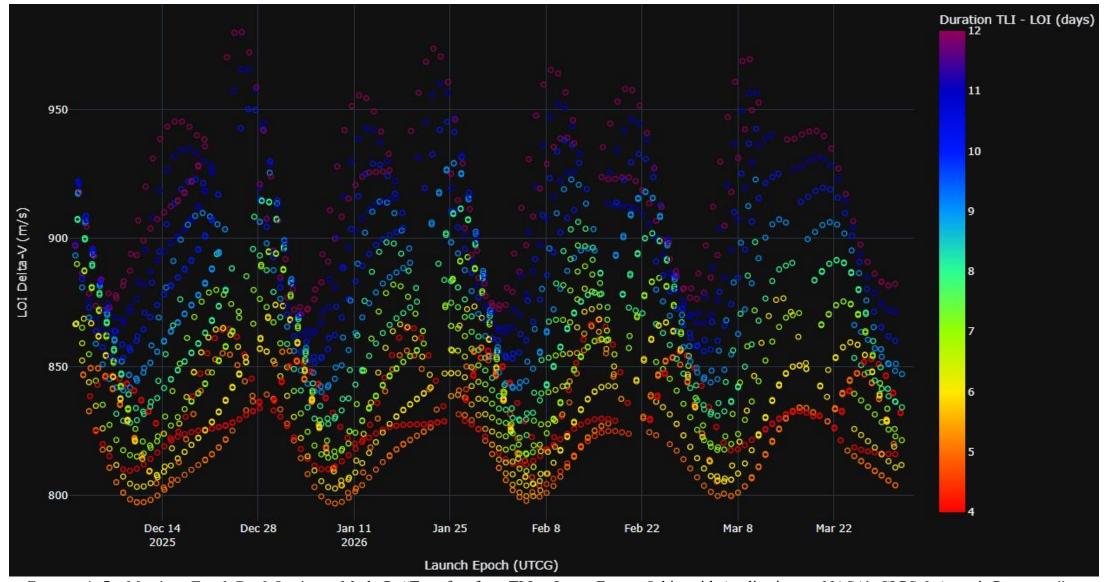


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LOI ΔV vs. Launch Epoch, from December 2025 to March 2026





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