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Validation of the NASA Electrical Power System – Sizing and Analysis Tool (EPS-SAT)

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Introduction

- Revolutionary Vertical Lift Technology (RVLT) Project
 - The RVLT project invests in cutting-edge technology and tools development
 - Ensure electric vertical takeoff and landing (eVTOL) vehicle reliability and safety
 - Reduce environmental impacts and noise
 - Create sustainable transportation system with broad economic benefit (advanced air mobility (AAM))
 - RVLT Power Management and Distribution (PMAD) group at Glenn Research Center (GRC) has built testbeds and modeling tools to investigate eVTOL power systems
 - Power quality, including off-nominal (faulted) operation
 - Modeling, simulation, analysis, and tool development for trade studies
 - There exist few sizing and analysis tools for electric aircraft propulsion (electrical power system specifically)





NASA

Introduction

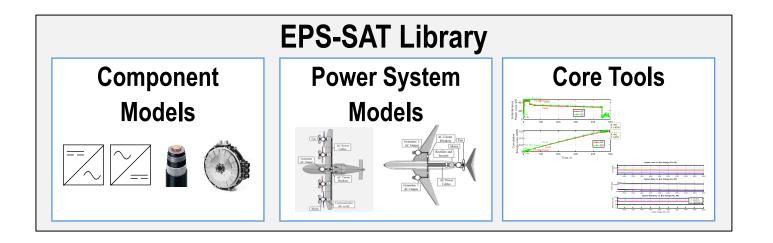
- Electric Power System Sizing and Analysis Tool (EPS-SAT)
 - Introduced to satisfy need for tool specifically intended for sizing and analysis of electrical power systems
 - System and component sizing
 - Power system architecture studies
 - System-level and component-level trades
 - Exposes design strengths and weaknesses
 - Can be used to direct investment dollars



EPS-SAT



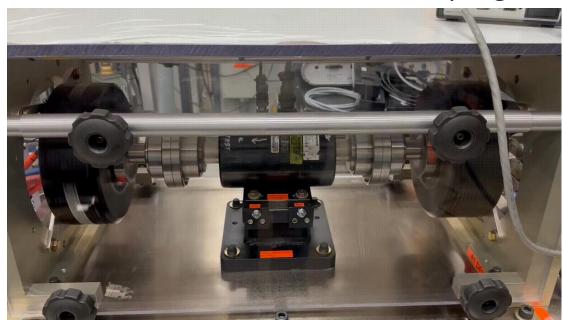
- Object-oriented MATLAB-based coding framework
 - Supports DC and single-phase or polyphase AC
 - Newton-Raphson solver for solving power flow equations
- Components and systems are sized in on-design mode
- Mission profiles and energy storage are specified in off-design mode
- Evaluate whether design closes given constraints (mass, range, system failure rate, etc.)



AREAL Testbed



- This was the hardware used for extracting mechanical and electrical validation data
- 200 kW, HVDC (max 1 kVDC)
- Reconfigurable and capable of both normal and faulted operation
- Physical PMSMs as well as motor emulator
- Bi-directional DC sources able to be programmed as dynamic sources and loads

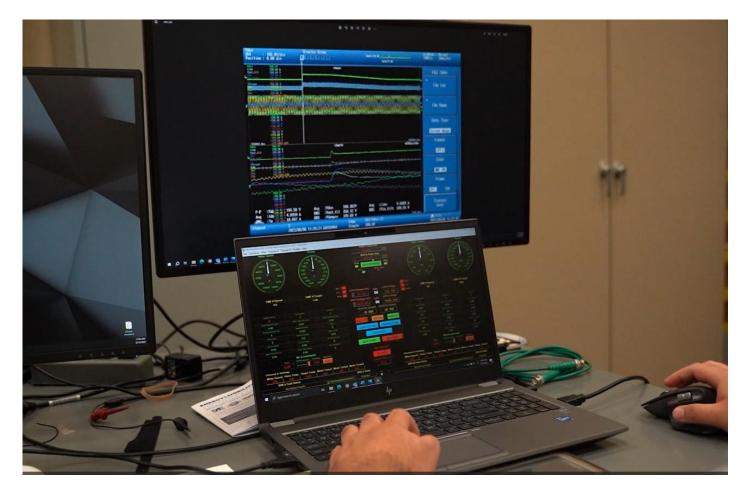






Testing and Data Processing

- Single-string configuration (single source and load)
- Steady-state values (718 VDC, 250:250:5500 rpm, 20:20:100 Nm)
 - 110 total
- Made mechanical (torque and speed) and electrical (voltage and current) measurements



Performance Map Extraction



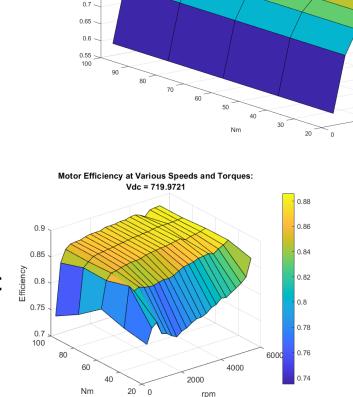
- Motor, inverter, and electric engine efficiency plots generated from motor and inverter power calculations
- Two wattmeter method incorporating FFT

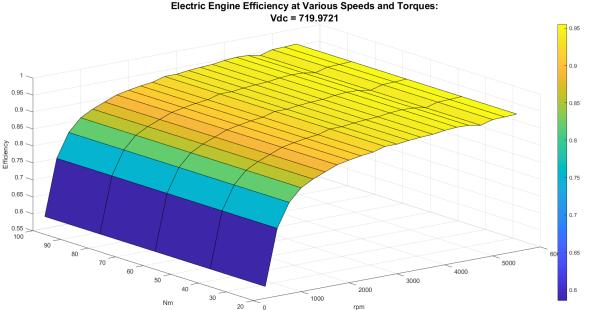
 $P_{AC,1} = \Sigma(real(fft(V_{AC}).fft(I_A^*)))$

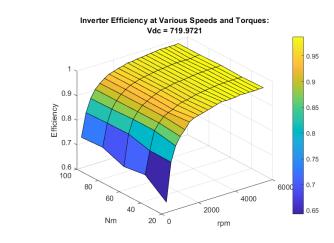
 $P_{AC,2} = \Sigma(real(fft(V_{BC}).fft(I_B^*)))$

 $\boldsymbol{P}_{\boldsymbol{A}\boldsymbol{C},\boldsymbol{T}\boldsymbol{O}\boldsymbol{T}} = \boldsymbol{P}_{\boldsymbol{A}\boldsymbol{C},1} + \boldsymbol{P}_{\boldsymbol{A}\boldsymbol{C},2}$

• Sanity check on EPS-SAT electric engine efficiencies



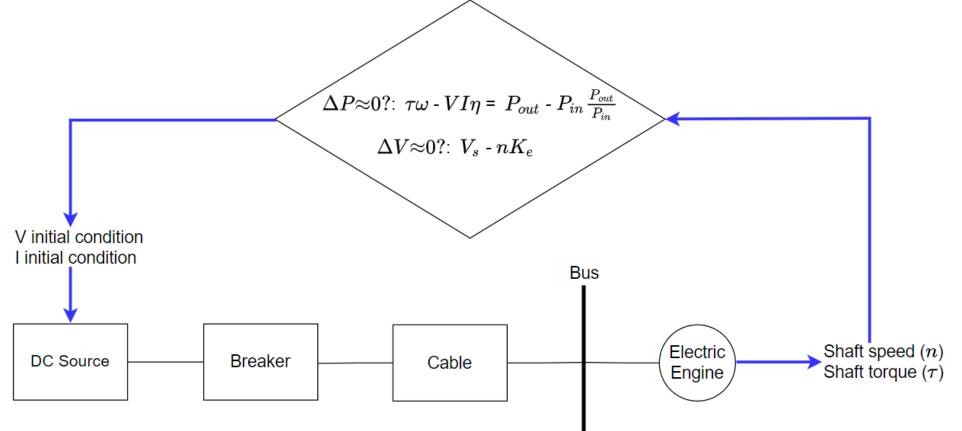




EPS-SAT Electric Engine Model Calculations



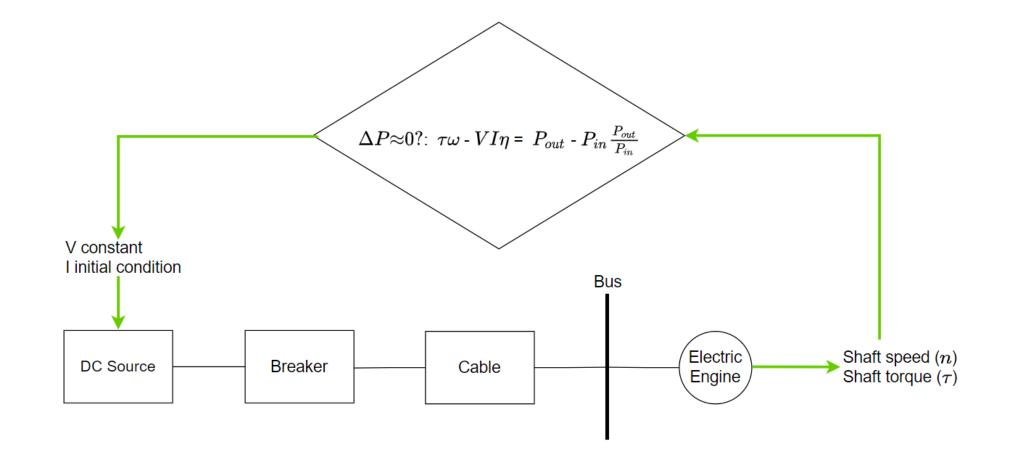
- On-design: given speed and torque setpoints, vary input voltage and current
 - Calculated electric engine input power varied until it equals DC input power (efficiency LUT)
 - Calculated electric engine input voltage (bus voltage) varied until it equals rated voltage (back EMF constant)



EPS-SAT Electric Engine Model Calculations



- Off-design: given speed, torque, and bus voltage, vary input current
 - Calculated electric engine input power varied until it equals DC input power (efficiency LUT)



Validation Strategy

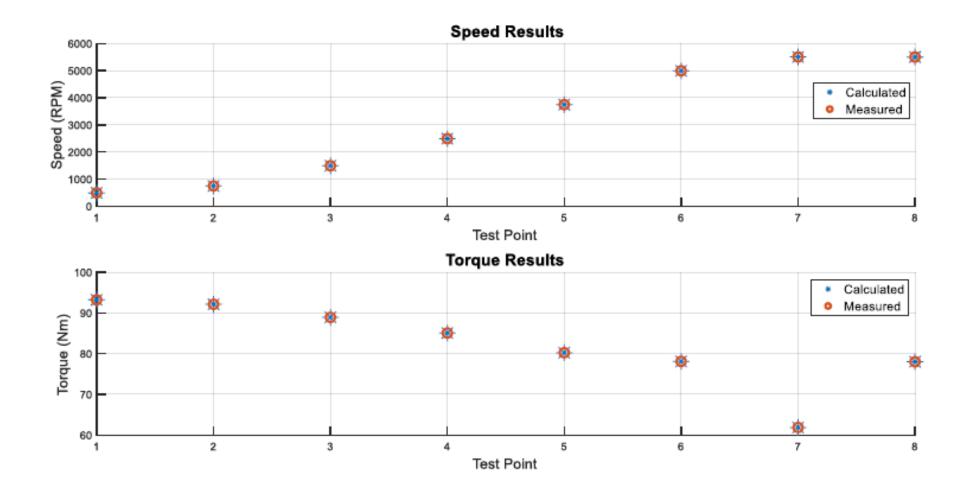
- AREAL hardware data points were the baseline
- Ran AREAL data points in EPS-SAT
- Evaluated results
 - Input the speed and torque at max continuous speed and torque
 - Expected:
 - EPS-SAT to produce matching voltage and current at the source terminals of the electric engine
 - Speed and torque at the load to match
 - Electric engine efficiency to match





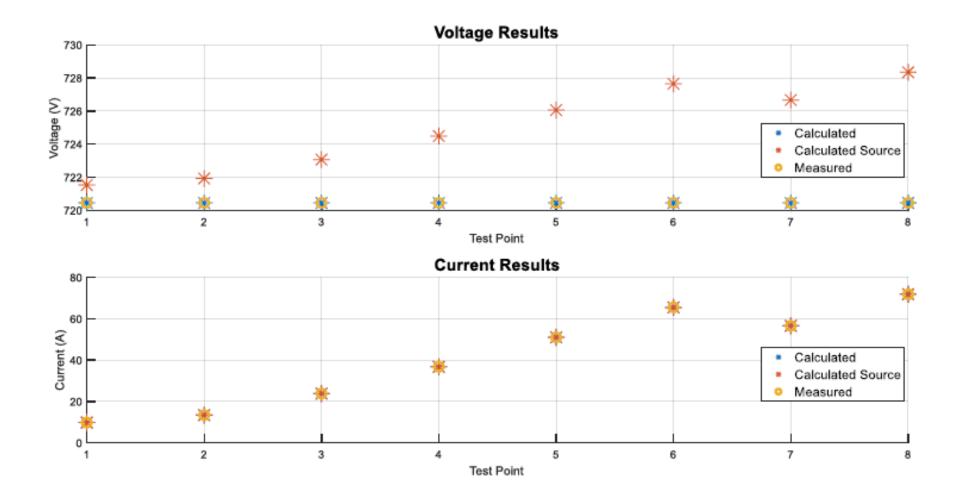
EPS-SAT and **AREAL** Validation





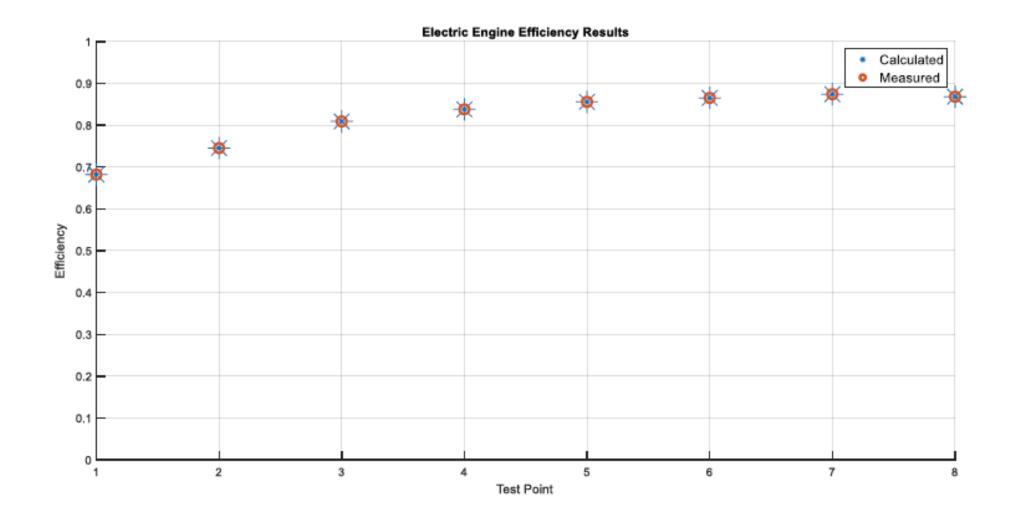
EPS-SAT and **AREAL** Validation





EPS-SAT and **AREAL** Validation







Future Work

- Add capability to model inverter and motor AC powers (and efficiencies)
- Compare realistic flight profile data measured in AREAL to flight profile data modeled in EPS-SAT
- Improve model for mass calculation



Conclusion

- AREAL single-string hardware configuration was modeled in EPS-SAT
- Single-string AREAL testbed steady state testing results at 718
 VDC validate EPS-SAT software calculations
 - Almost no difference between electric engine efficiencies, speeds, torques, DC voltages, and DC currents
- EPS-SAT is a useful tool for high-level, rapid design and evaluation of EAP systems

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