

National Aeronautics and
Space Administration

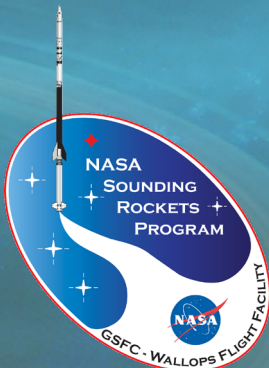


Sounding Rockets

Ernie Bowden

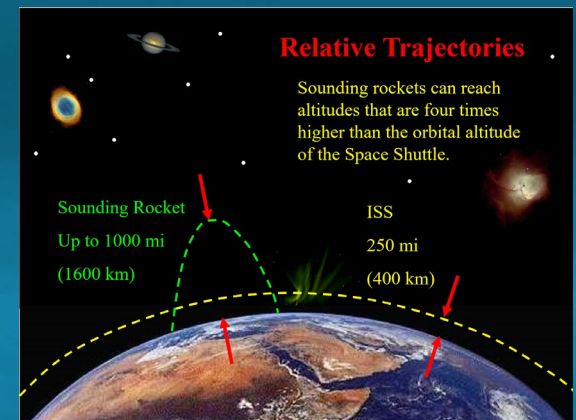
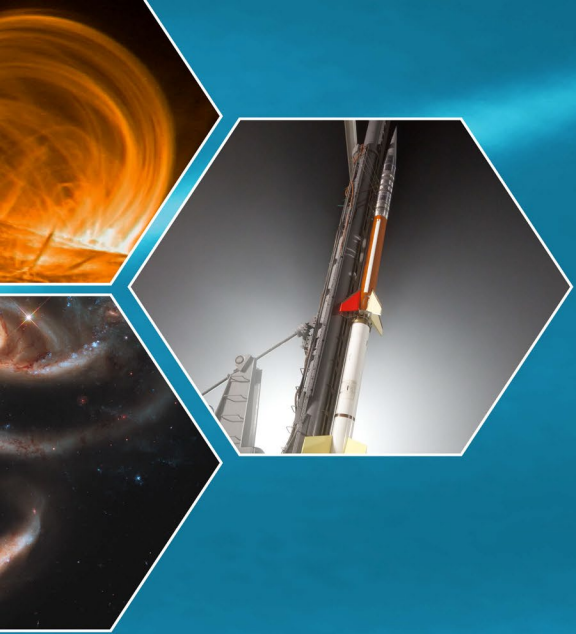
BrainLink 2024

November 21st, 2024



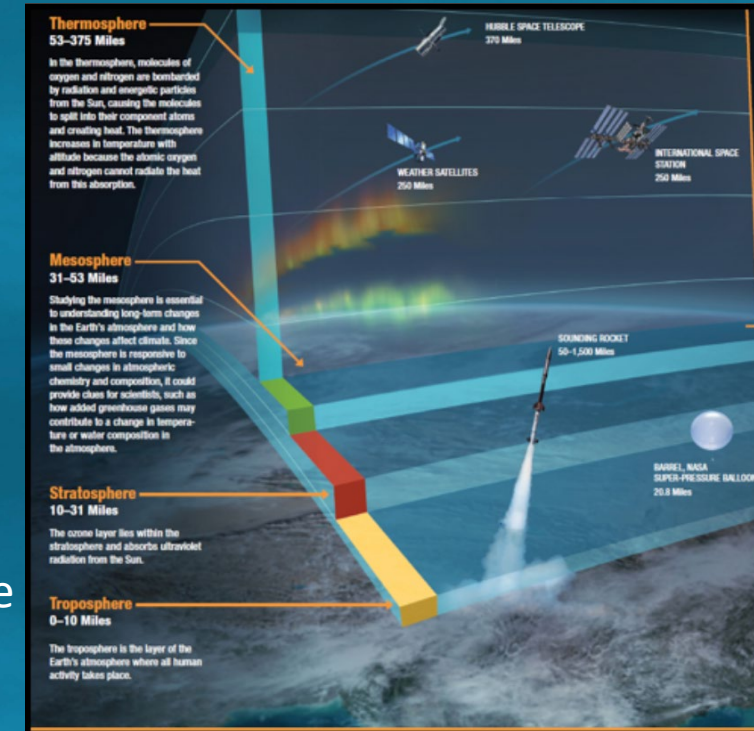
Sounding Rockets Introduction

- A sounding rocket is a vehicle that carries scientific payloads on a suborbital trajectory
- Suborbital trajectories are parabolic and do not reach orbital velocities
- Sounding rockets are spin-stabilized and often unguided
- Sounding rockets carry payloads up to 2,000 pounds to sub-orbital apogee altitudes between 100 km to 1,400 km (62 – 870 miles)
 - Up to 20 minutes of ballistic flight
 - Altitudes can reach 4x higher than the ISS orbits
- Target of 85% mission assurance success
 - Enables the program to conduct cutting edge science and test new technologies and new instruments



Why Sounding Rockets?

- NASA Sounding Rockets Program offers a quick-turn around and low-cost access to space for the science community
 - Utilize military surplus rocket motors
 - Accept higher technical risk
- Can place instruments directly into regions where the science is occurring for in situ measurements
 - World-wide mobile operations
- Can measure regions of the atmosphere (mesosphere and lower thermosphere) that are too low to be sampled directly by orbiting spacecraft and too high to be sampled directly by balloons or aircraft
- Excellent platform to develop, fly, and validate new science instruments that will eventually fly on a satellite
- Excellent way to train pipeline of scientists for future orbital missions



Nature of the Program

- Located at NASA Wallops Flight Facility, Virginia
- Funded by NASA Science Mission Directorate, Heliophysics Division
- Support all NASA Science Divisions: Heliophysics, Astrophysics, and other NASA Directorates, e.g. Space Technology
- Also support other U.S. Government Agencies, e.g. Department of Defense
- Frequent internal technology development missions
- Frequent Education missions (RockOn, RockSat)
- 40-50 missions in development at any given time
- 15-20 flights/year at locations around the world
- Execution: NASA Sounding Rocket Operations Contract (NSROC)

Mission Identification

- Customer Funded
 - NASA enters into an agreement with other US government and commercial entities.
 - Customer is responsible for funding all payload development and launch operations costs.
- NASA Grants
 - Individual Principal Investigators (typically from a university) apply for a NASA grant via the Research Opportunities in Space and Earth Sciences (ROSES) in two proposal categories 1) Heliophysics Low-Cost Access to Space (LCAS) and 2) Astrophysics Research and Analysis (APRA). PI's are US citizens, but foreign nationals can be Co-Investigators.
 - The NASA grant funds the development of the science instruments and science team support. The NASA Sounding Rocket Program funds the launch vehicle, payload development, and launch operation services (including range costs).
- Programmatic Developments
 - The Sounding Rocket Program funds bi-annual technology development missions.
 - The Sounding Rocket Program funds 2 annual student education missions.

Services Provided



Mission Management & Operations

- Interface to mission stakeholders
- Team oversight
- Planning & Scheduling
- Progress Reporting



Engineering Support

- Electrical
- Mechanical
- Launch Vehicles & Propulsion
- Guidance, Navigation & Control
- Flight Performance
- Environmental Testing



Manufacturing

- Machine Shop
- Electrical Manufacturing
- Launch Vehicle Assembly
- Mechanical Assembly
- Quality Control
- Functional Testing



Payload systems

- Telemetry
- Power
- Recovery system
- Guidance, Navigation & Control systems
- Shutter door

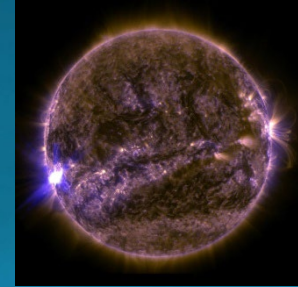
Launch Operations



- Vehicles
- Launchers
- Remote site operations

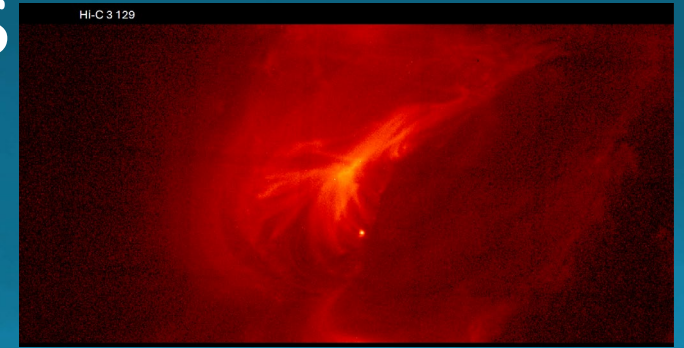
Types of Missions

- Geospace (Plasma Physics)
- Solar Telescopes
- Astronomical Telescopes
- High Speed Aerodynamics and Propulsion
- Reentry and Descent
- Technology Development
- Educational

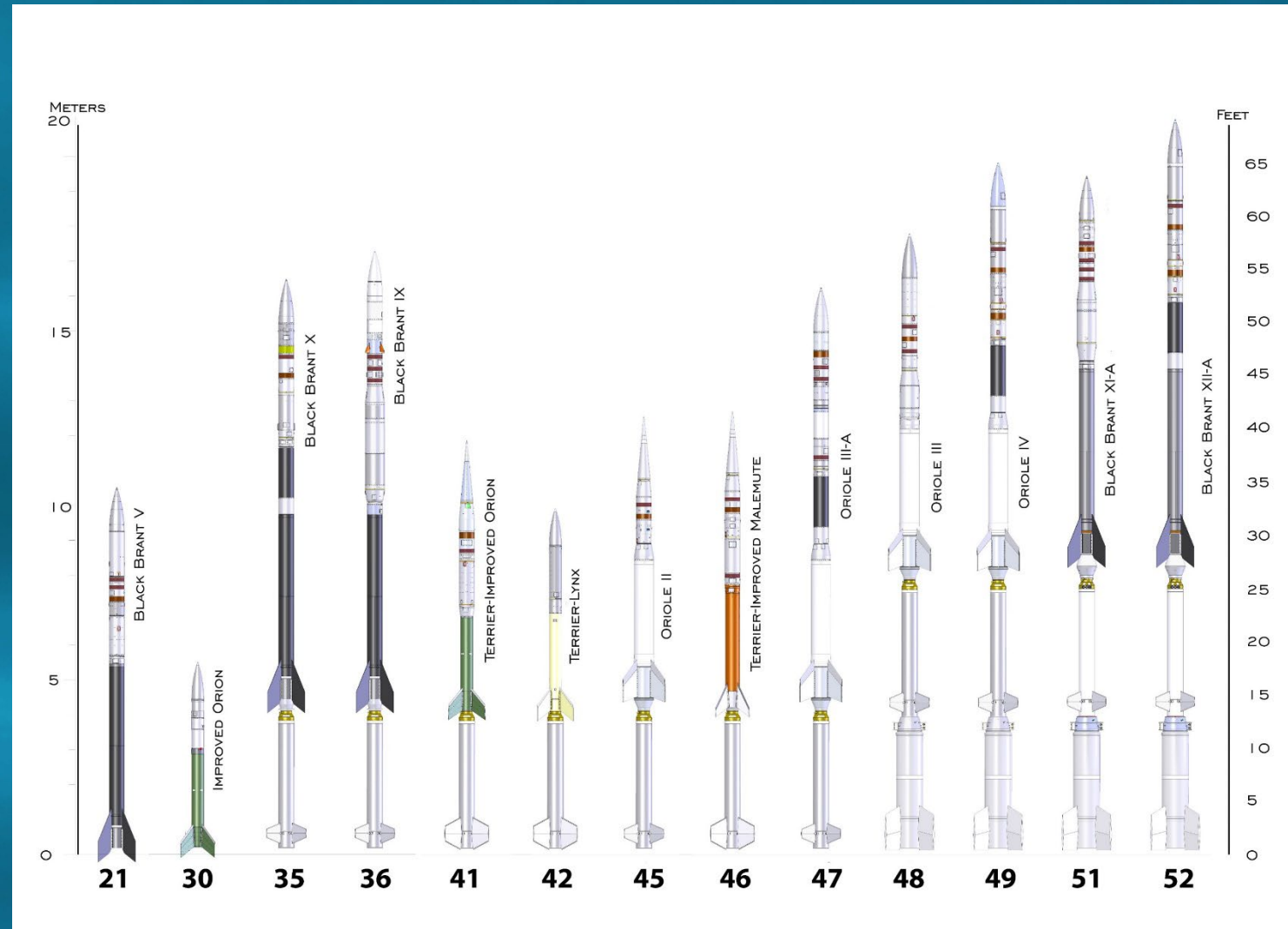


Recent Highlight Missions

- Solar Flare Campaign – FOXSI & Hi-C Flare
 - Co-observation of M-class flare with each other and SDO from Poker Flat Research Range (PFRR) Fairbanks, Alaska
- Endurance – Svalbard, Norway
 - First successful detection of Earth's ambipolar electric field
 - Results recently published in *Nature*
- Eclipse Campaign
 - Studied ionospheric changes during eclipse – with ejected sub-payloads
 - 3 missions from WSMR Oct '23
 - Prior to peak, at peak, and after peak
 - Recovered
 - 3 missions from WFF Apr '24
 - Prior to peak, at peak, and after peak

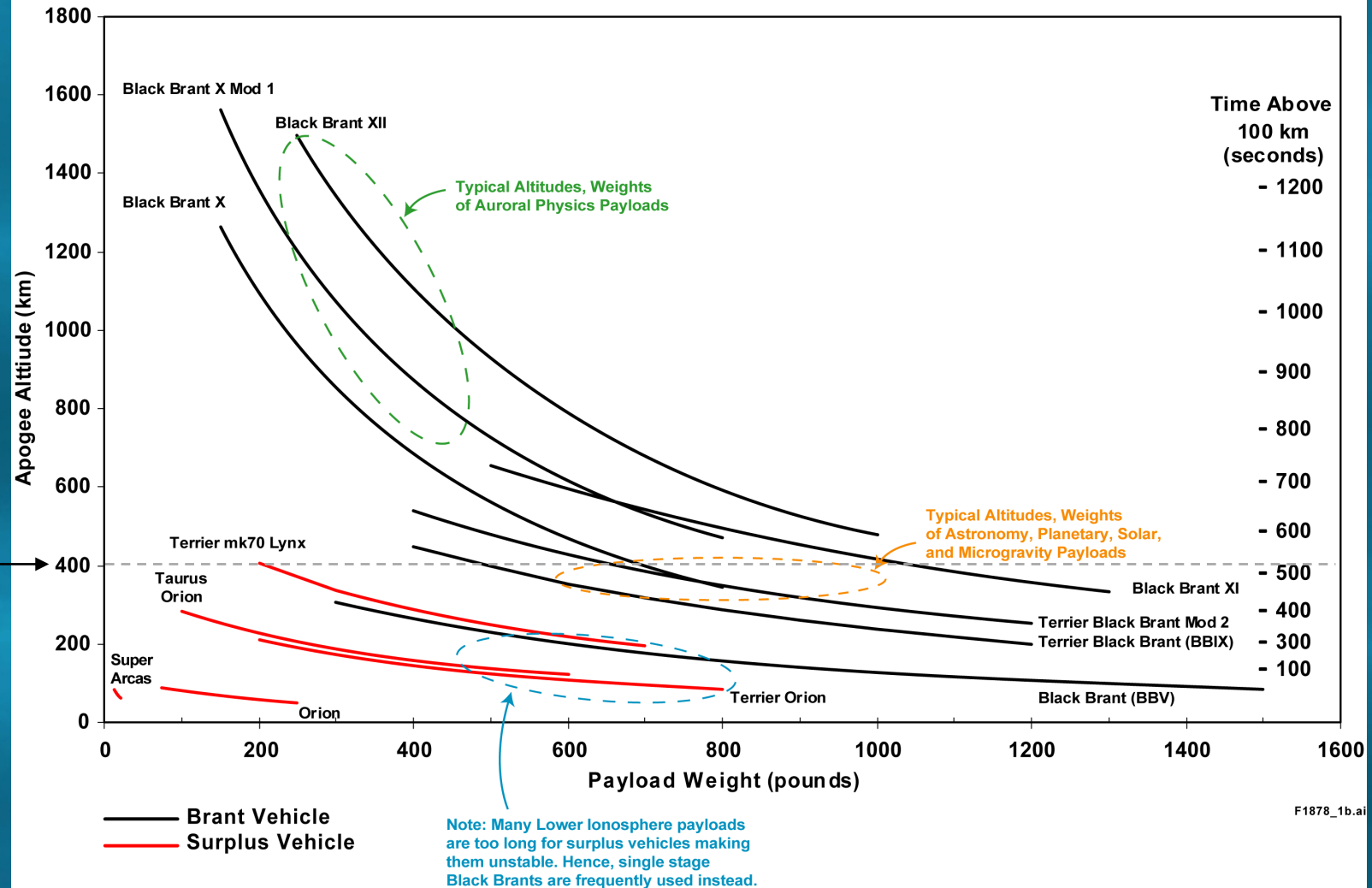


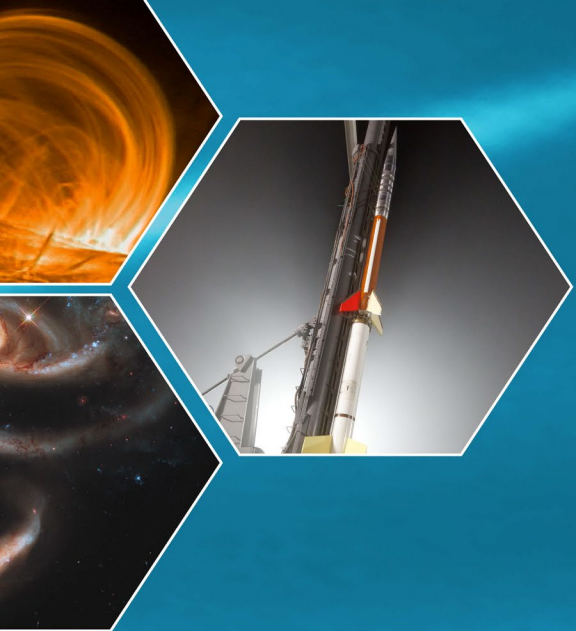
Launch Vehicles



Vehicle Performance

NASA Sounding Rocket Vehicle Performance

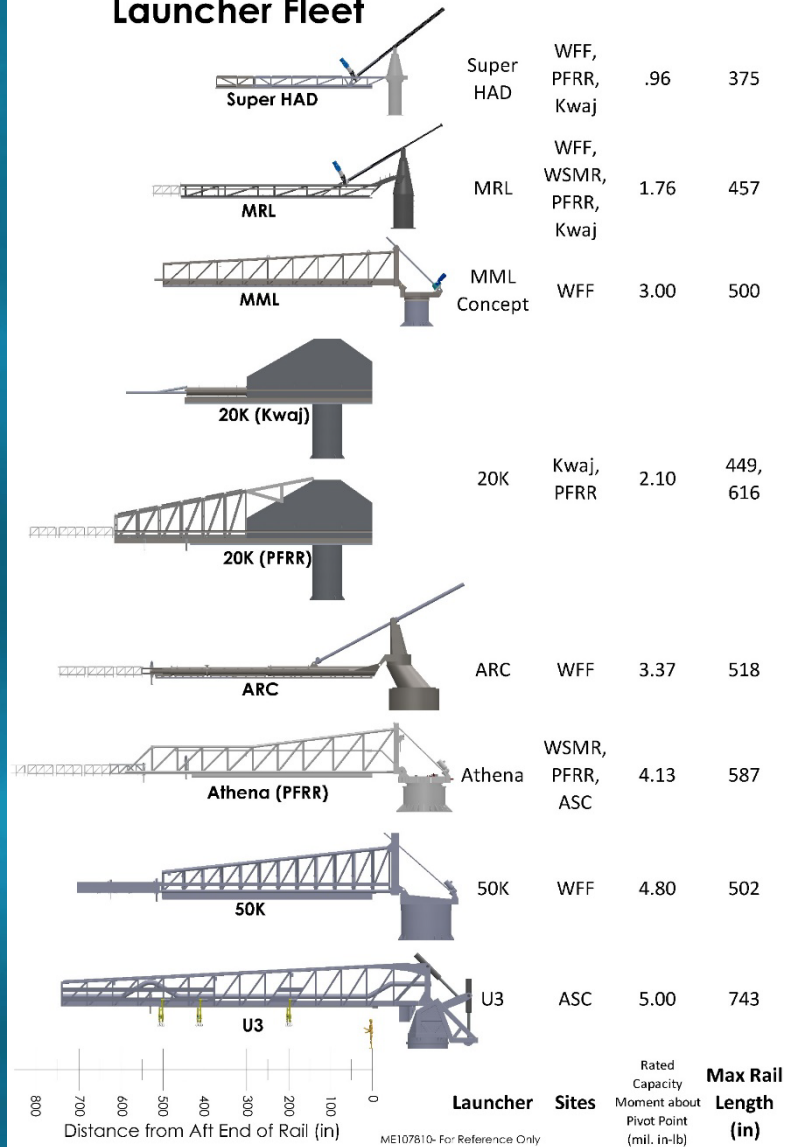




Launcher Fleet



NASA's Sounding Rocket Program Launcher Fleet



Payload Components*

Payload

2nd stage motor (Black Brant)

1st stage motor (Terrier)



Payload support systems include recovery systems, telemetry, boost guidance systems, Attitude Control Systems, deployment mechanisms, and a shutter door among others. Mission requirements determine which support systems are used.



Parachute Recovery

The parachute recovery system houses the parachute and deployment mechanism.

ACS

Attitude Control Systems (ACS) are used to align the payload in space.

Telemetry (TM)

The Telemetry system enables experiment data to be transmitted to a ground station.

Experiment Section

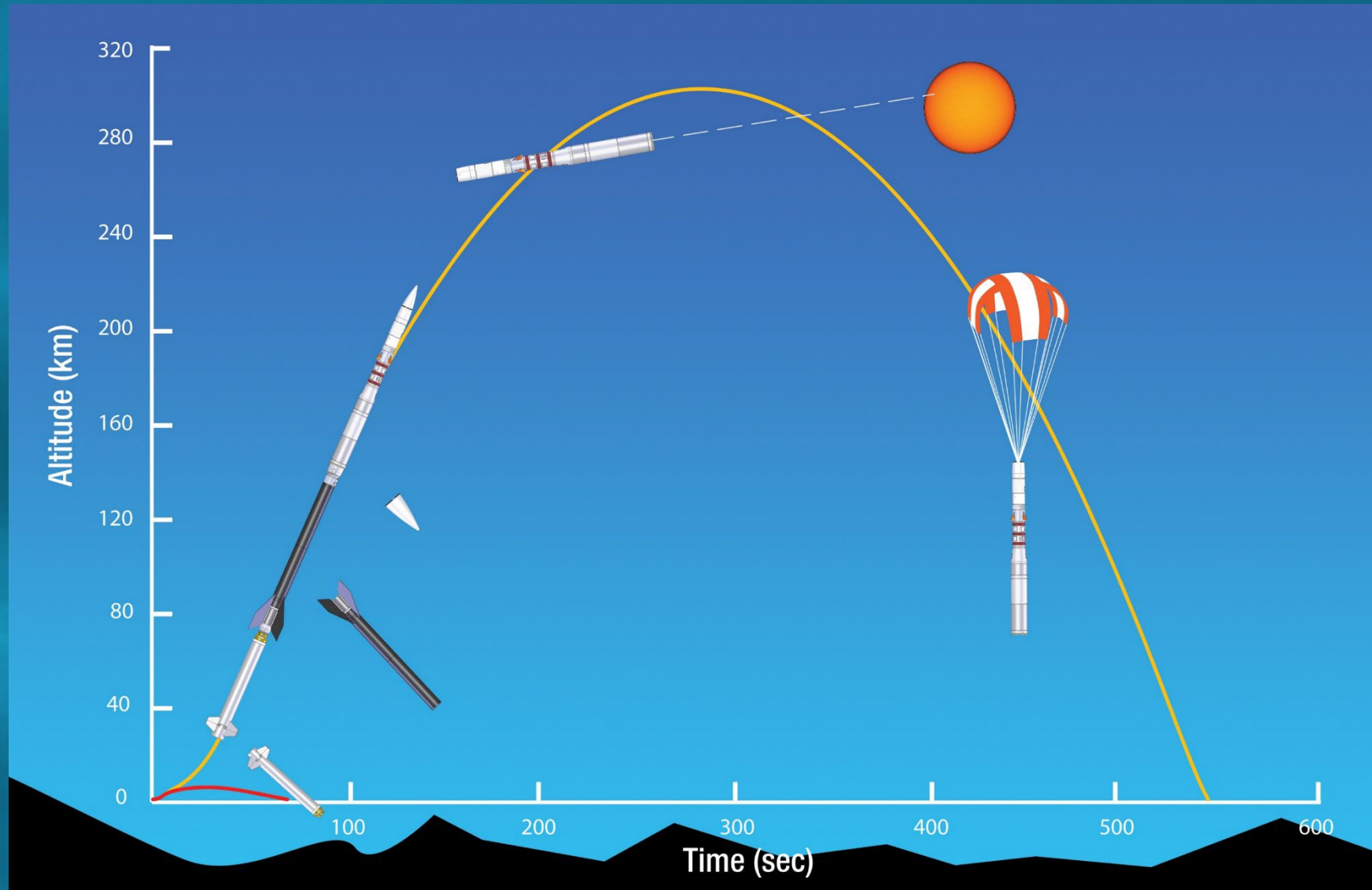
This section houses the scientific instruments.

Shutter Door

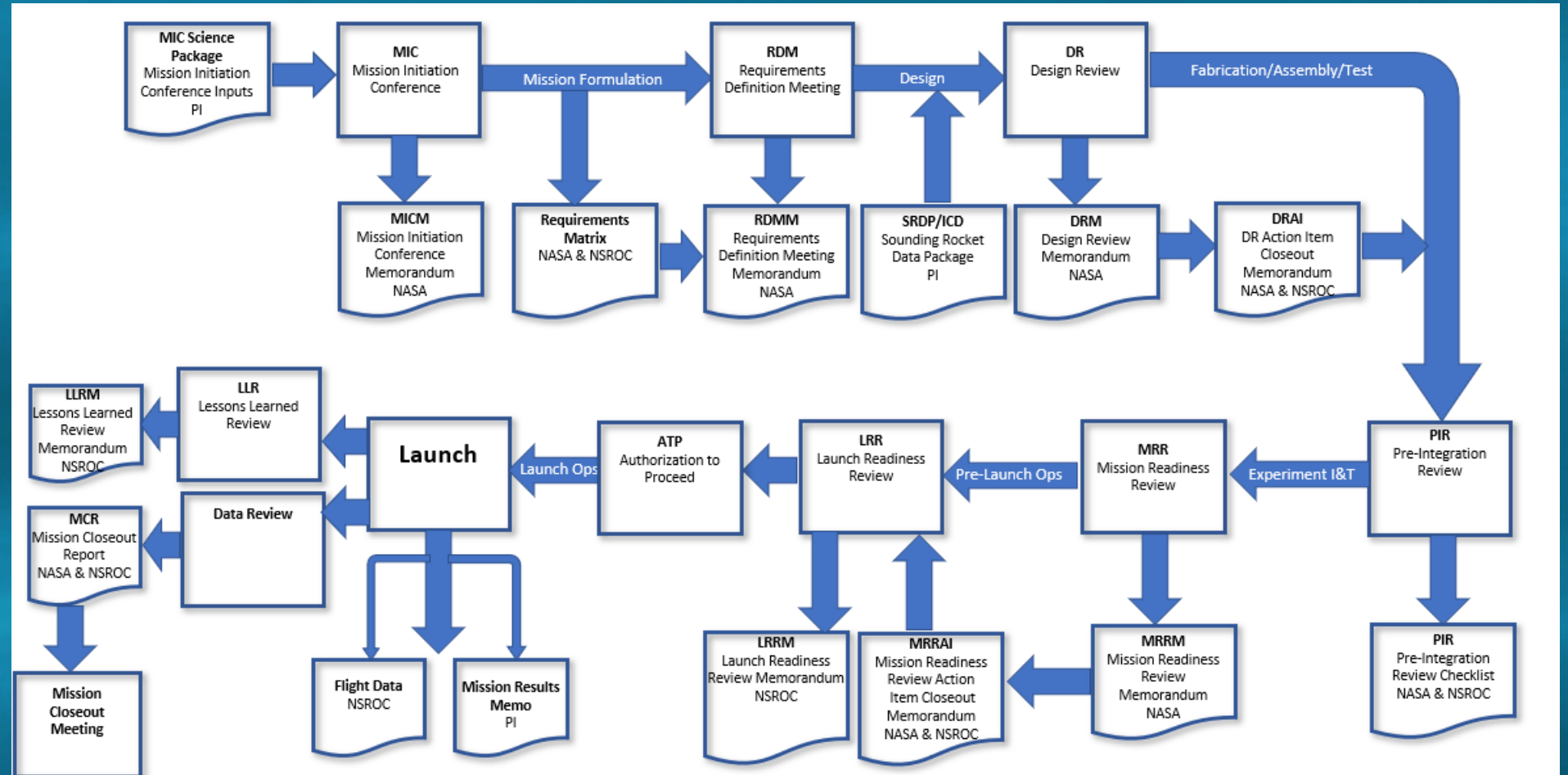
Used mainly for telescope payloads, the shutter door is opened in space allowing the telescope to see the target of investigation.

**Representative telescope payload*

Simple Flight Profile



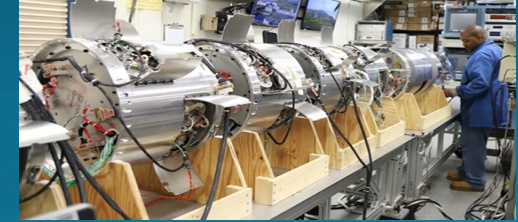
Sounding Rocket Mission Lifecycle



Mission lifecycle averages ~2 years

Integration and Testing

- Integration
 - Subsystem (NASA/NSROC)
 - Handshakes with instrument
 - Full payload integration
- Testing & Evaluation
 - Sequence Testing
 - GPS Outdoor
 - Magnetic Calibration
 - Electro Magnetic Interference (EMI)
 - Mass Properties
 - Vibration
 - Balance
 - Bend
 - Deployment



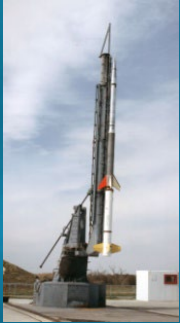


SOUNDING ROCKETS PROGRAM WORLDWIDE LAUNCH SITES 1959 - PRESENT

World-Wide Operations



Poker



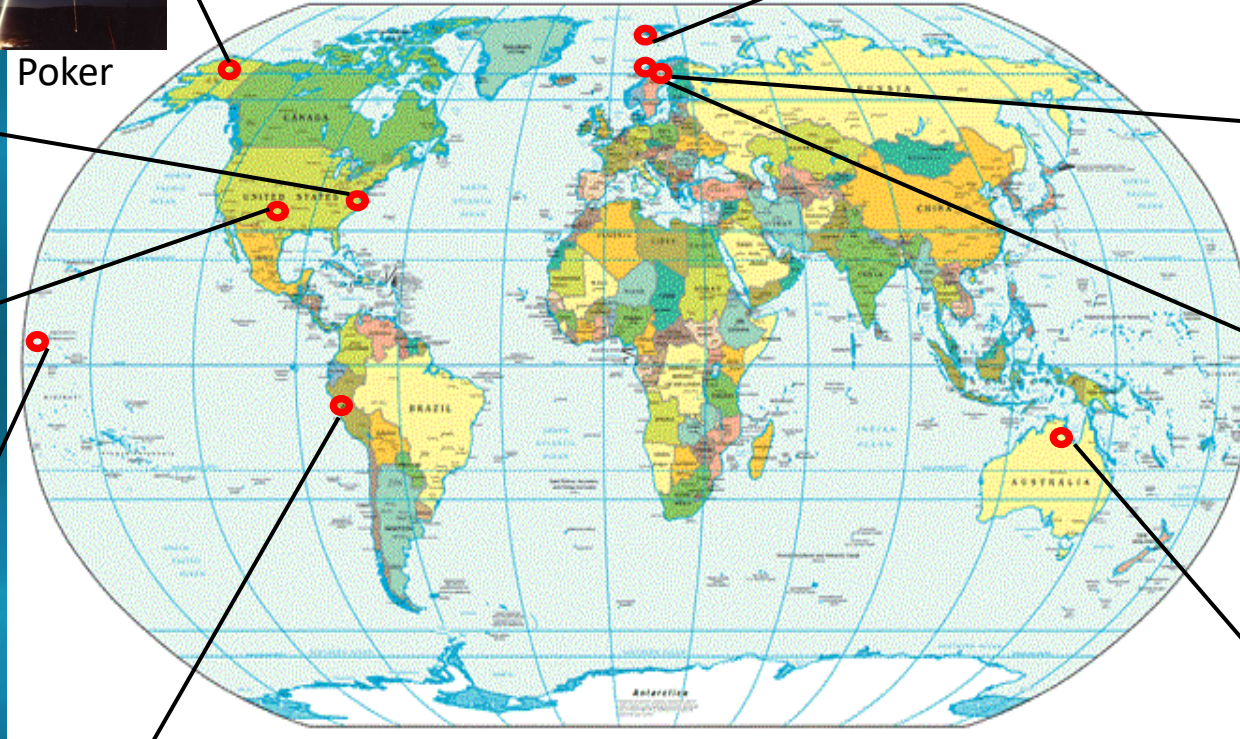
Wallops



White Sands



Kwajalein



Peru

The Sounding Rocket Program

“goes to where the science is...”



Svalbard



Kiruna, Sweden



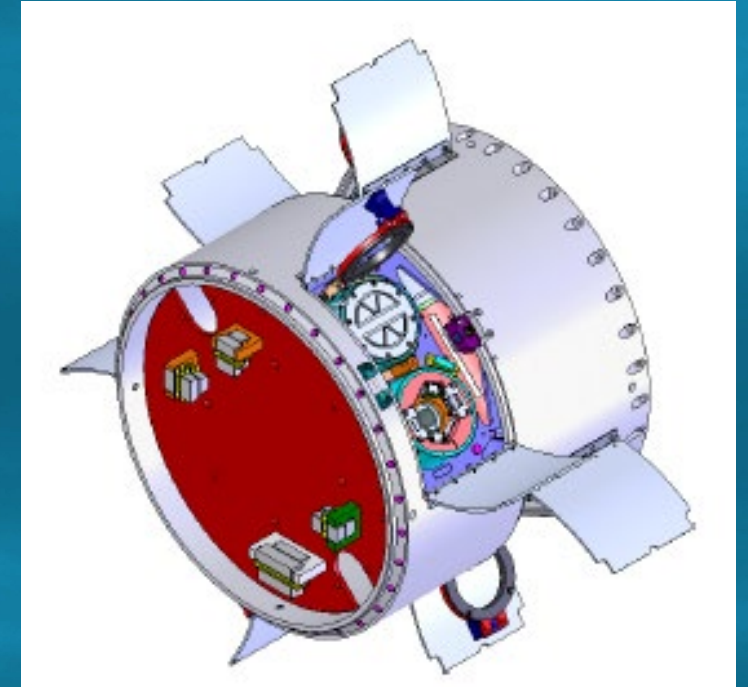
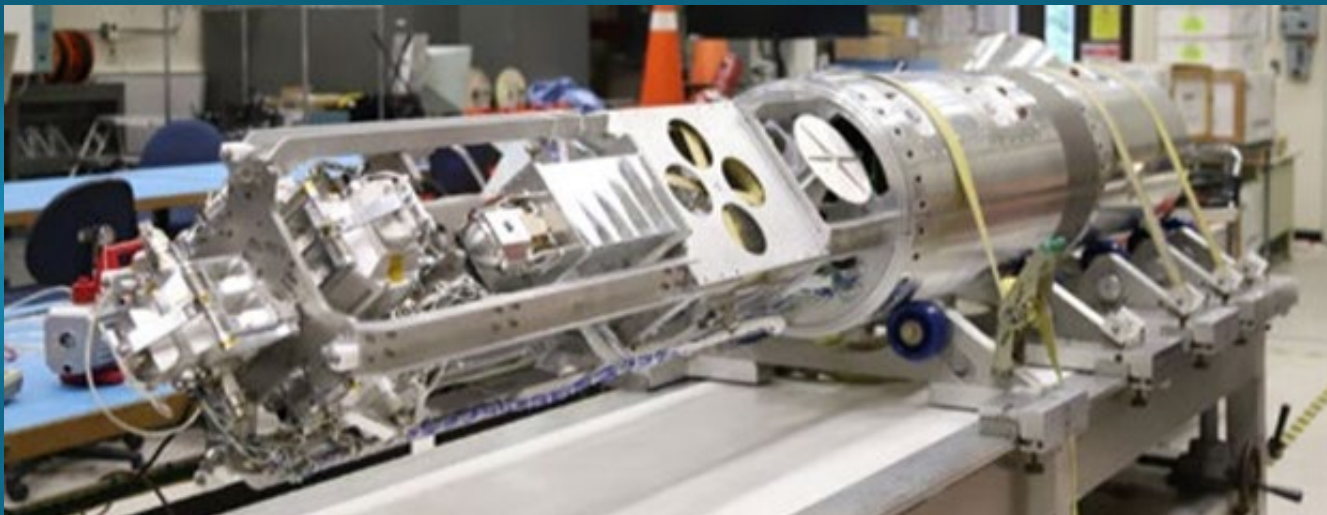
Andenes, Norway



Australia

Payload Systems - Mechanical

- Experiment Mounting
 - Vacuum sections, exposed deck structures, booms
- Shutter Doors
- Crush Bumpers
- Deployments
 - Booms
 - Ejectables – Spring or rocket propelled



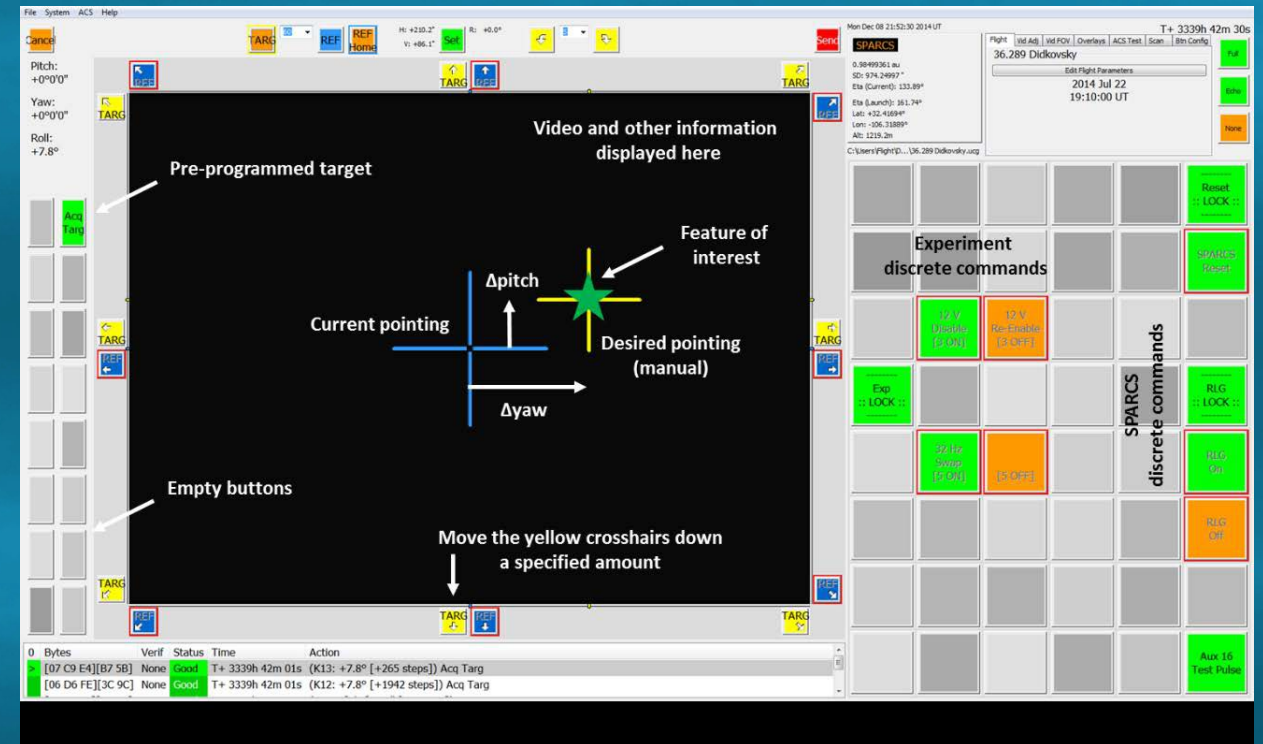
Payload Systems – Telemetry/Power

- Power
 - Unregulated battery power, switching between external power and battery
 - Various battery sizes to support range of power consumption
 - Regulated power
- In-flight events
 - Timed – relative to launch indication
 - Altitude Driven – requires navigation input
- Telemetry
 - PCM Matrix with variety of input types
 - Analog, Serial, Parallel, Asynchronous Serial, other device specific options
 - EVTm – Ethernet based
 - Baseband – Modulate an analog signal



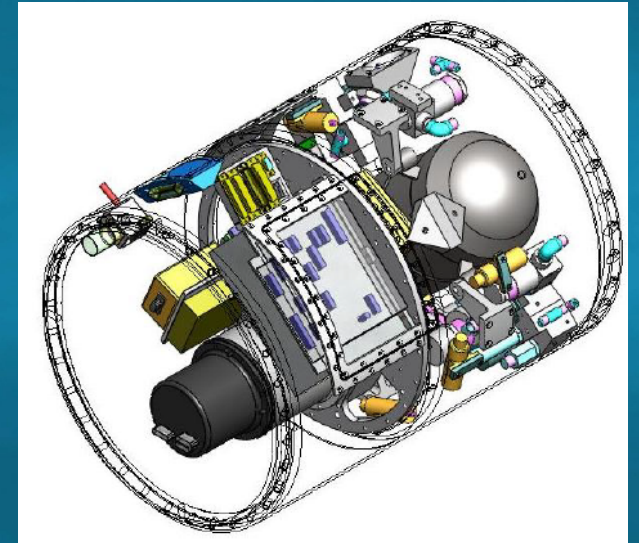
Payload Systems – Command Uplink

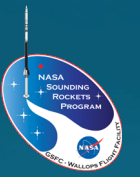
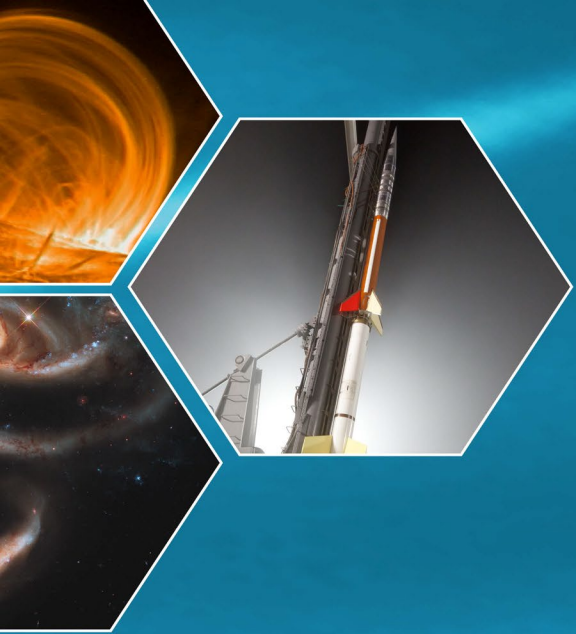
- Discrete commands – relay closures
- Experiment Serial commands
- ACS Targeting commands
 - Maneuver switching
 - Pointing refinement
 - Can incorporate science video into console display



Payload Systems - ACS

- Solar Pointing Attitude Rocket Control System (SPARCS)
 - Sun pointing – sun center or offset
 - Arc-second stability
 - Can be augmented with uplink to deliver arc-second accuracy
- Celestial Attitude Control System (CACS)
 - Night sky targets
 - Arc-second stability
 - Can be augmented with uplink to deliver arc-second accuracy
- NSROC Inertial Attitude Control System (NIACS)
 - Inertial targeting – including in-flight target calculation based on trajectory
 - Supports spinning and non-spinning control
 - 2-degree accuracy
 - Stability highly dependent on payload and mission configuration
- NMACS
 - Spinning alignment to sensed magnetic field
 - 2-degree accuracy
 - Stability highly dependent on payload and mission configuration





Thank You!