



Validation of a Scenario-Based Approach to Assess Gaps in Earth Observations

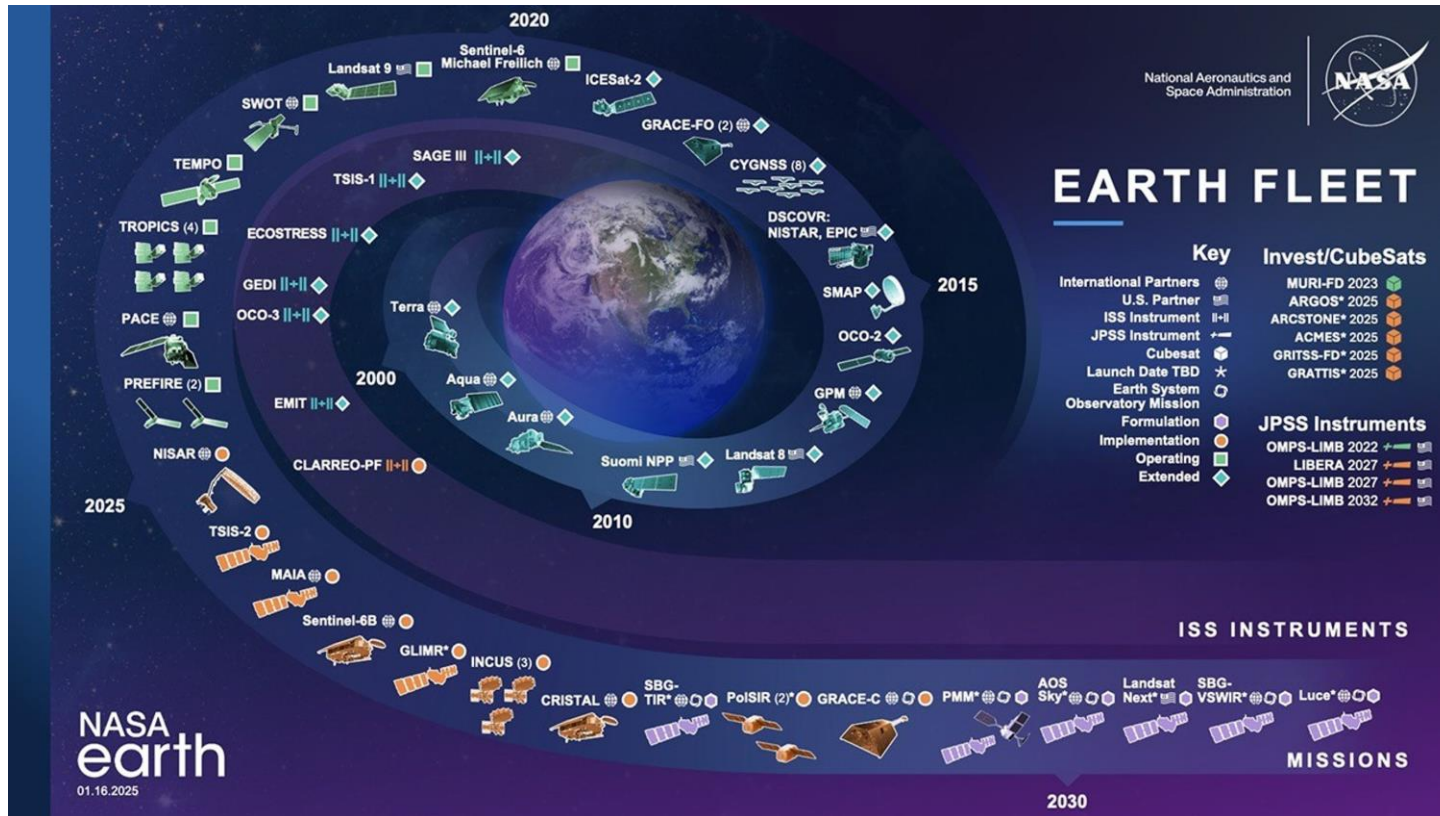
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Motivation



- Continuous Earth observation measurement records contribute to the accuracy of predictions and analysis of the Earth system
- The probability of gaps between Earth observation missions should be assessed well before their occurrence to support strategic decision making
- Factors contributing to the uncertainty of decision making:
 - Funding
 - Technology maturation
 - Shifting mission schedules



A Scenario-Based Approach to Assess Continuity Gaps in Earth Observations

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Abstract— Decision analysis processes outlined in systems engineering references, such as the Systems Engineering Body of Knowledge [1] or the NASA Systems Engineering Handbook [2], recommend following a series of steps to support decision-making for engineering applications. These steps are consistent across the literature and typically involve defining objectives, defining relevant criteria against which candidate alternatives can be assessed, selecting an evaluation method, assessing alternatives, and making a recommendation based on the assessment. These approaches are well-suited to assess systems for which sets of common objectives and constraints can be identified. A vast body of literature describes their application to aerospace engineering problems, and many methods have been developed since the middle of the 20th century to support the assessment of candidate alternatives for such systems [3, 4, 5, 6]. These approaches, however, show some limitations when decisions pertain to systems of systems. These collections of individual systems typically do not share a common set of objectives and constraints, tend to be highly complex, and the definition of assessable, high-value candidate alternatives poses a challenge due to interdependencies.

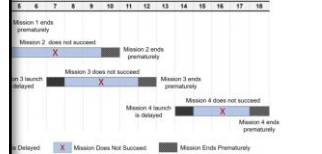
The formulation, development, operation, and funding of Earth observing missions are such that gaps may occur between missions, thereby impacting the continuity of measurements. These Earth observing mission architectures comprise systems of systems for which a common set of objectives and constraints can be challenging to define. When tasked with assessing gaps that may occur for a series of spacecraft missions that address similar system parameters, the authors therefore propose an approach and method beyond traditional decision-making. This approach does not require a specification of all common objectives, rather, it assesses the impact of possible decisions on the system-of-systems and characterizes its possible future states. It also offers the flexibility to adjust assumptions, make inputs, and refine supporting models over time, while enabling the rapid, early identification of challenges and key decision points. By simulating the impact of potential decisions on the entirety of the system-of-systems, the approach enables the quantification of multiple decisions that typically occur at the level of the individual system.

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This paper provides an overview of the approach that was developed to assess continuity gaps for a series of operational and planned spacecraft. Each observing mission, a description of its application to the problem at hand, and a discussion of currently known limitations.

1. INTRODUCTION
The complexity associated with formulating and developing Earth observing missions, compounded with operational and funding uncertainty, is such that gaps may occur between missions. Characterizing these potential gaps and assessing their likelihood to occur are essential to support strategic decision-making. Earlier this year, the authors were tasked with assessing the likelihood of gaps to occur for a series of spacecraft missions, either in operation or in formulation, that address similar system parameters in order to inform decision-making at the portfolio level. Gaps can result from a variety of causes, which can be technical or programmatic in nature. The assessment therefore needed to account for the current and forecasted programmatic environment, existing technical capabilities, and anticipated technology developments. Interdependencies between causes of potential gaps also needed to be considered in the assessment.

adverse event for its mission architecture and three missions in it of the MELDs. An delay is shown in Fig. 1. The adverse event are failure type. These can provide more detailed initiating events are the lowest tier of the diagram. While failure types and their decomposition are generalizable across spacecraft missions with only minor adjustments, initiating events are mission-specific. Multiple initiating events can be linked to a single failure cause, and if no initiating event is linked to a failure cause, that failure cause is not used in the model.



A thorough review of project documents, to include analysis results obtained by the various mission teams, supports the definition of the potential adverse events and their associated initiating events, including the assignment of probabilities to these events.

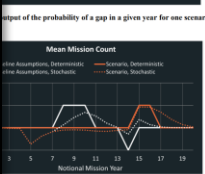
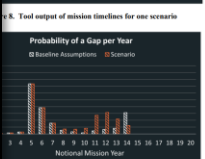
This paper provides an overview of the approach that was developed to assess continuity gaps for a series of operational and planned spacecraft. Each observing mission, a description of its application to the problem at hand, and a discussion of currently known limitations.

for a multi-mission architecture and associated potential adverse events to occur. MELDs are modeled as nodes contributing to a variation are modeled as nodes associated with a continuous outcome which has a distribution function. Nodes associated with a discrete outcome which has a distribution function are modeled as nodes associated with a discrete outcome which has a distribution function. Nodes associated with a continuous outcome which has a distribution function are modeled as nodes associated with a continuous outcome which has a distribution function.

Monte Carlo simulation techniques are applied to estimate the probability of a gap to occur given the absolute uncertainty of initiating events in the MELDs. For each simulation run, discrete event nodes are evaluated by sampling a uniform distribution to generate a random number between 0 and 1, then comparing it to the event probability to determine whether or not the event occurs. Continuous event nodes are evaluated by using the Inverse Transform method to sample the node's distribution; again, a random number is generated between 0 and 1, and the value of the inverse CDF at that number gives a randomly sampled number from that distribution. This process is repeated for each mission considered in the analysis; the outcomes are applied to mission timelines and result in either excluding the mission altogether, advancing launch dates, and/or terminating the mission earlier than planned. An updated hypothetical mission manifest is then produced from the combination of these outcomes.

➤ A scenario-based approach was developed by M. Ivanco et al.^[1] to:

- 1) Estimate the probability of potential continuity gaps in a multi-mission architecture
- 2) Assess the probability in the context of multiple possible future states of the architecture



Deterministic and stochastic mission counts in a given year for one scenario



- Traditional NASA systems engineering decision analysis processes show limitations when decisions pertain to systems of systems (SoS)
- Ideas from the Military Decision-Making Process (MDMP) are incorporated
- A 7-step process defines the Scenario-Based Approach

1. Identification of adverse events

2. Modeling of possible causes of adverse events

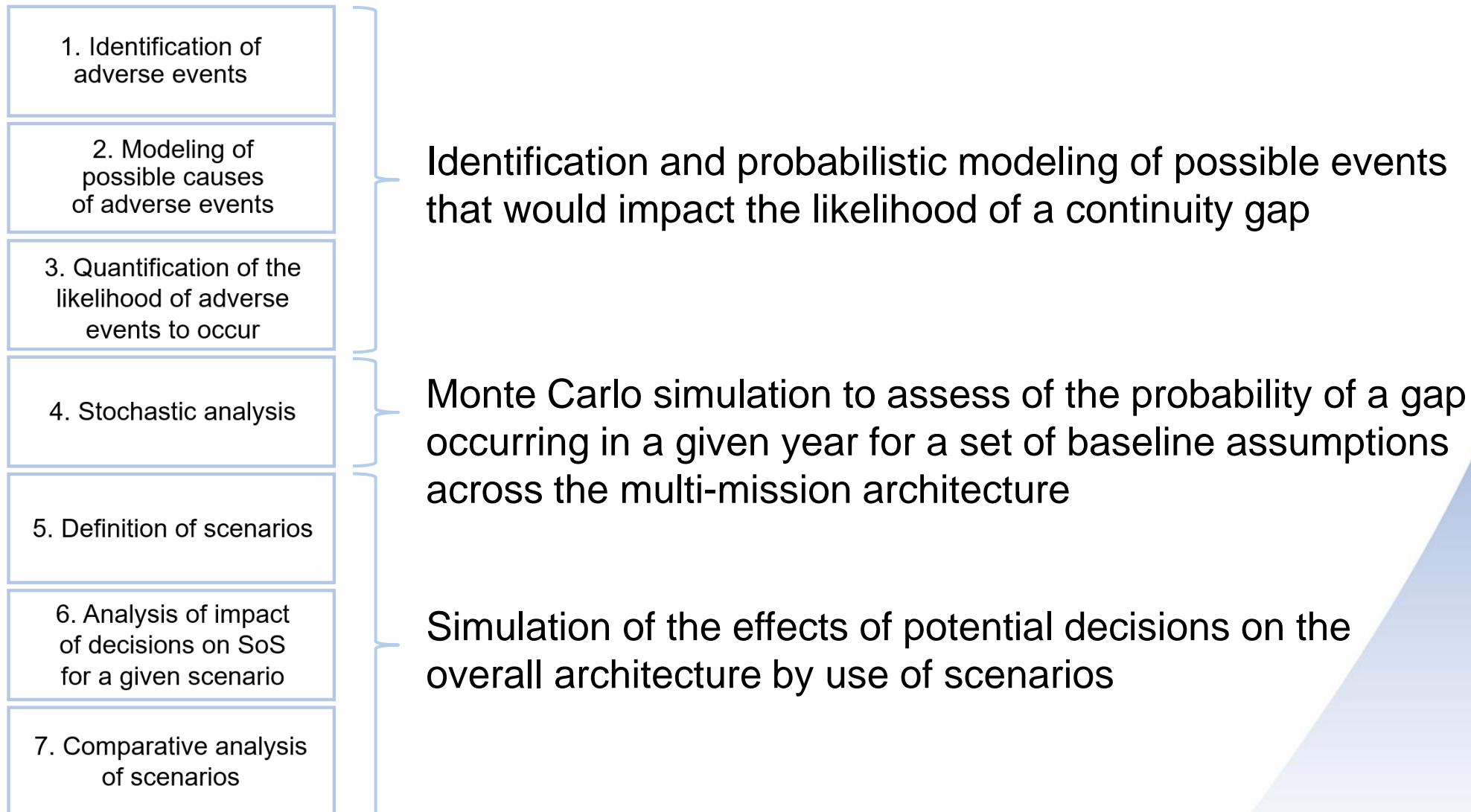
3. Quantification of the likelihood of adverse events to occur

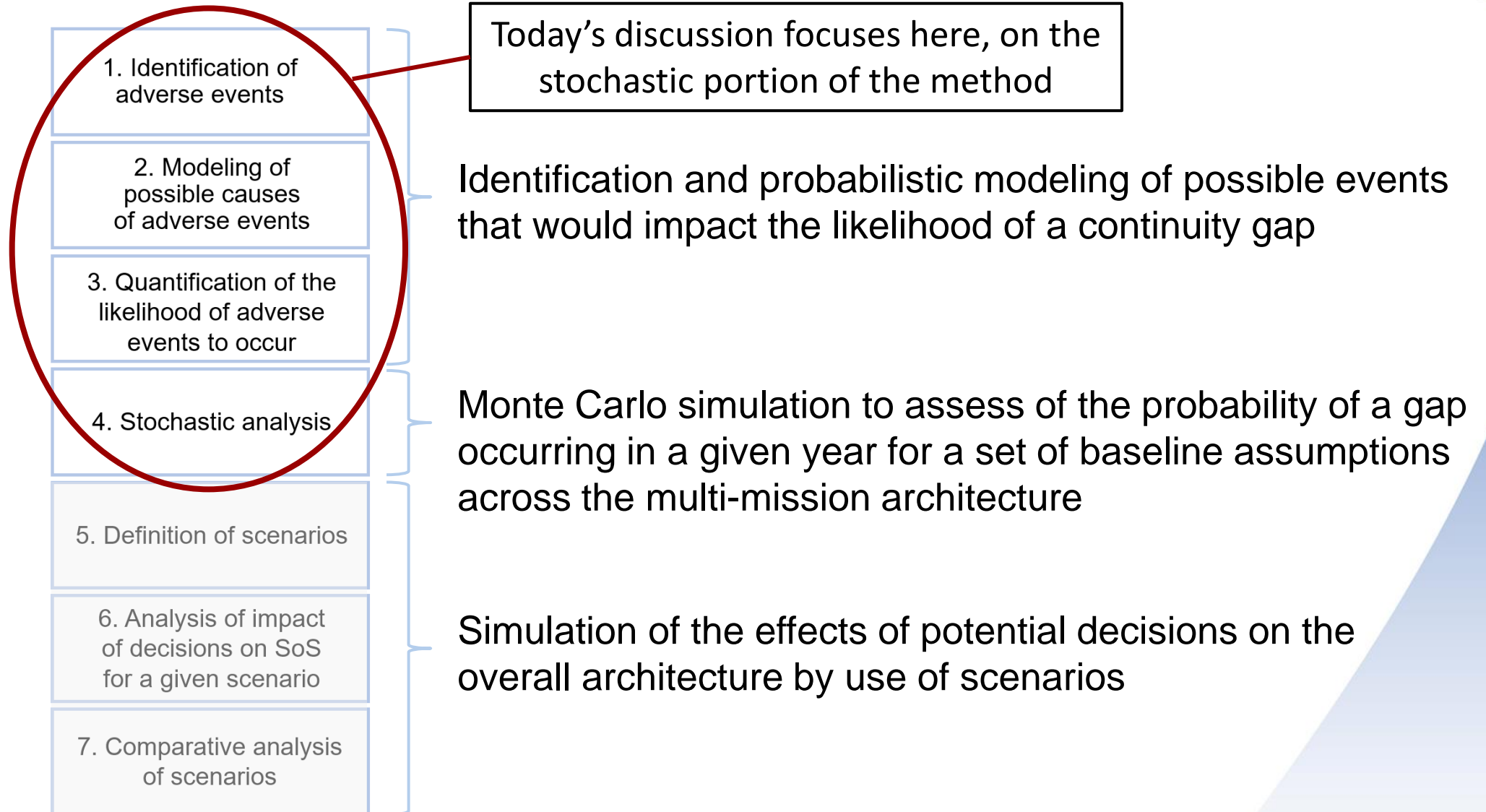
4. Stochastic analysis

5. Definition of scenarios

6. Analysis of impact of decisions on SoS for a given scenario

7. Comparative analysis of scenarios





Intent of Validation



- The stochastic method was applied retroactively to an existing multi-mission architecture, and the results were compared with actual outcomes of the missions

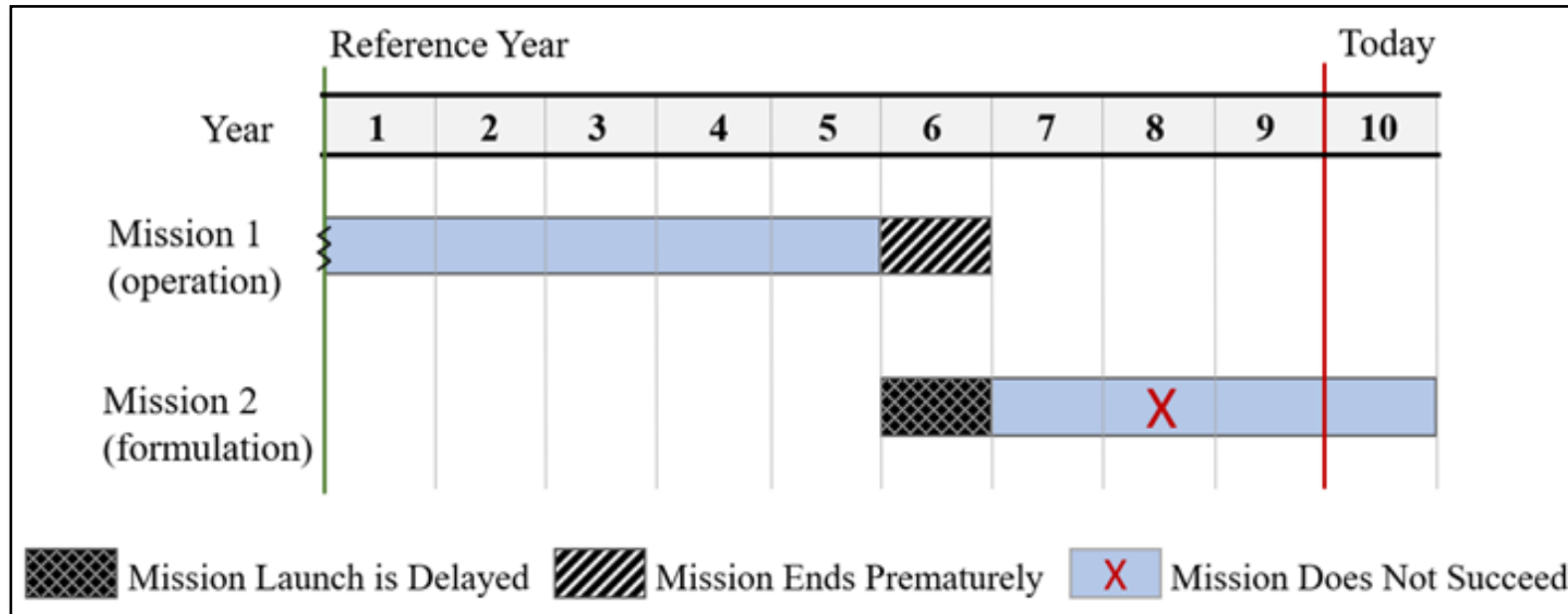
- Intent:
 - Initial effort investigating the validity and adequacy of the method
 - Provide insight into the effectiveness of the distributions and models used to assess the probability of gaps
 - Identify model improvements

- Not Intended:
 - An exhaustive validation of all use cases of the method

Validation Approach



- Mission formulation documents, used to inform model distributions, were gathered from the development phase of a mission that is in operation today
- Analysis was conducted using information that was available 5 years prior to the start of Mission 2

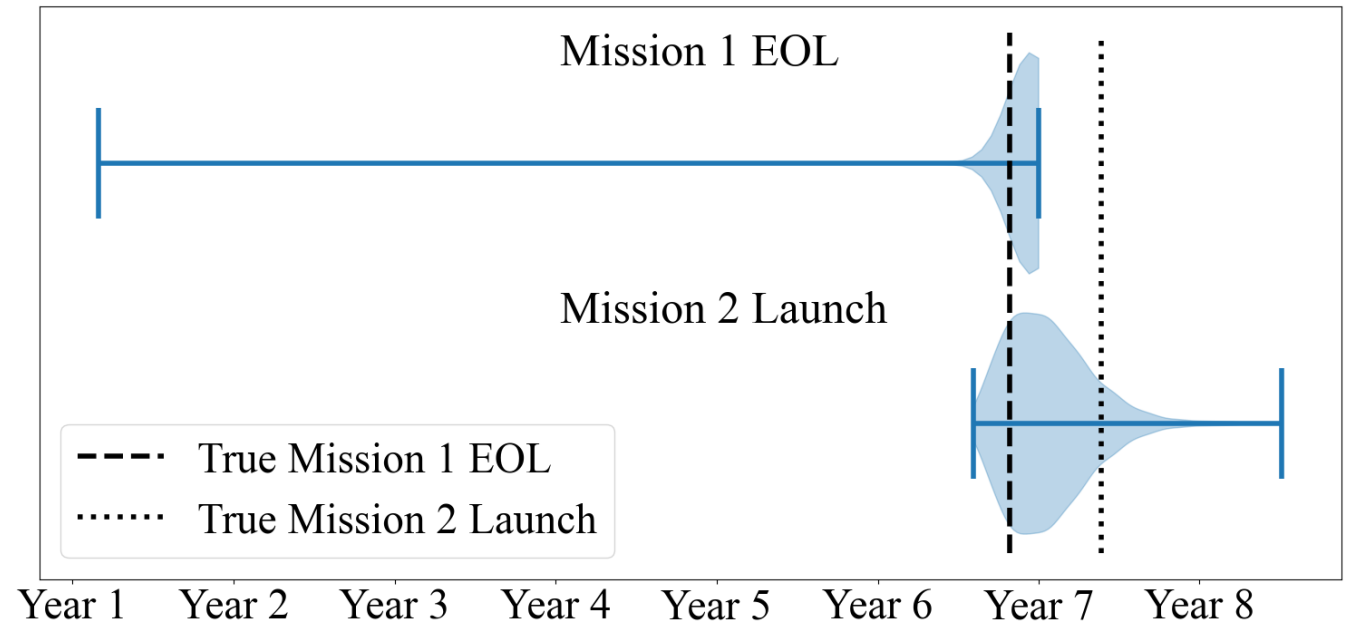


Comparison of Model Results to True Dates



The simulation predicts the likelihood of a continuity gap by probabilistically estimating Mission 1 end of life (EOL) and Mission 2 launch dates

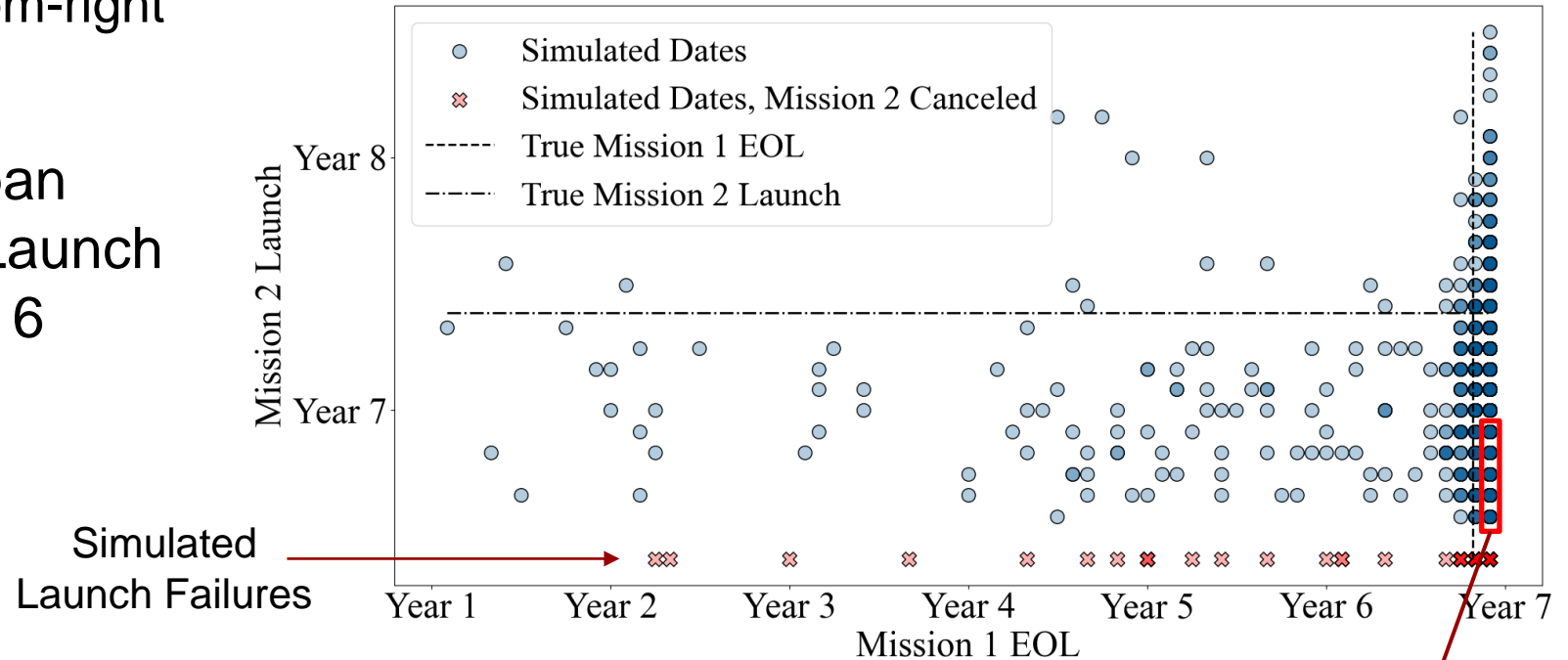
- Mean EOL estimate is within 2 days of the true EOL date
- Mean launch estimate predicts a 177-day launch delay
 - Actual launch delay = 294 days



Comparison of Model Results to True Dates



- Simulation tended to overestimate Mission 1 EOL and underestimate Mission 2 delay
 - 67% of points fall into bottom-right quadrant
- Mission 1 EOL estimates span Years 1-7, while Mission 2 Launch estimates appear after Year 6

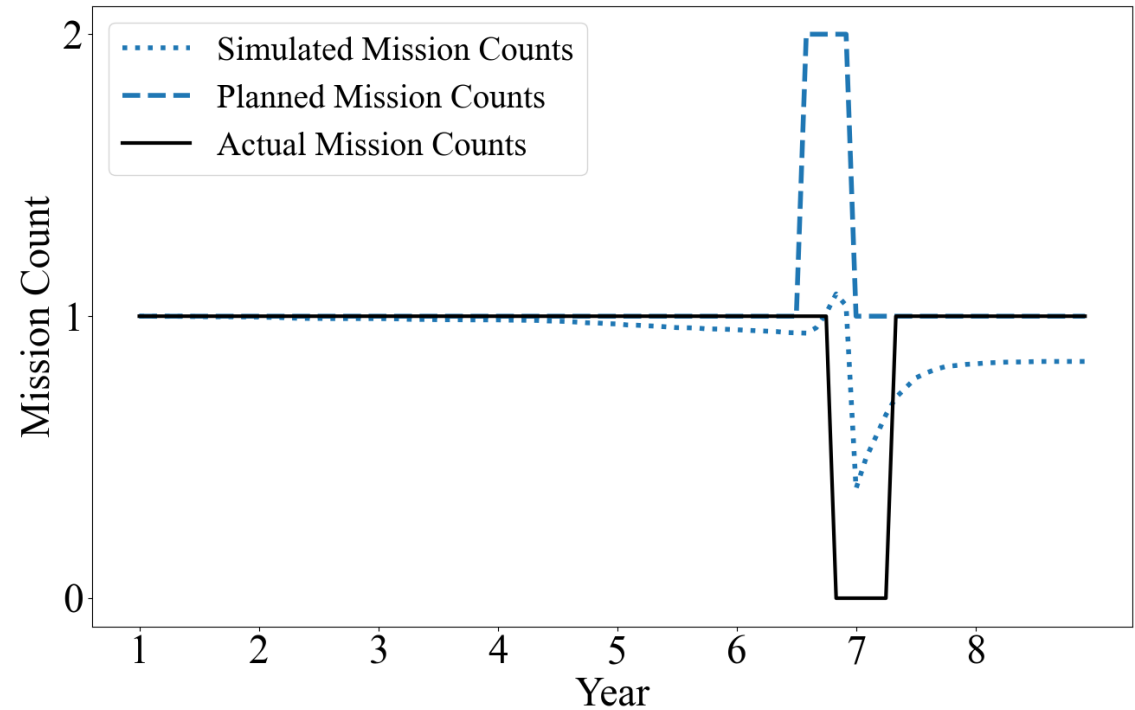


Comparison of Mission Counts



- Gaps between missions are identified by estimating the count of operational missions
 - Mission Count of 0 = Gap
- Simulated mission count represents the mean number of operational missions across all simulation runs for each time step

The stochastic model more closely predicts the true outcomes of the missions than what was planned during mission formulation

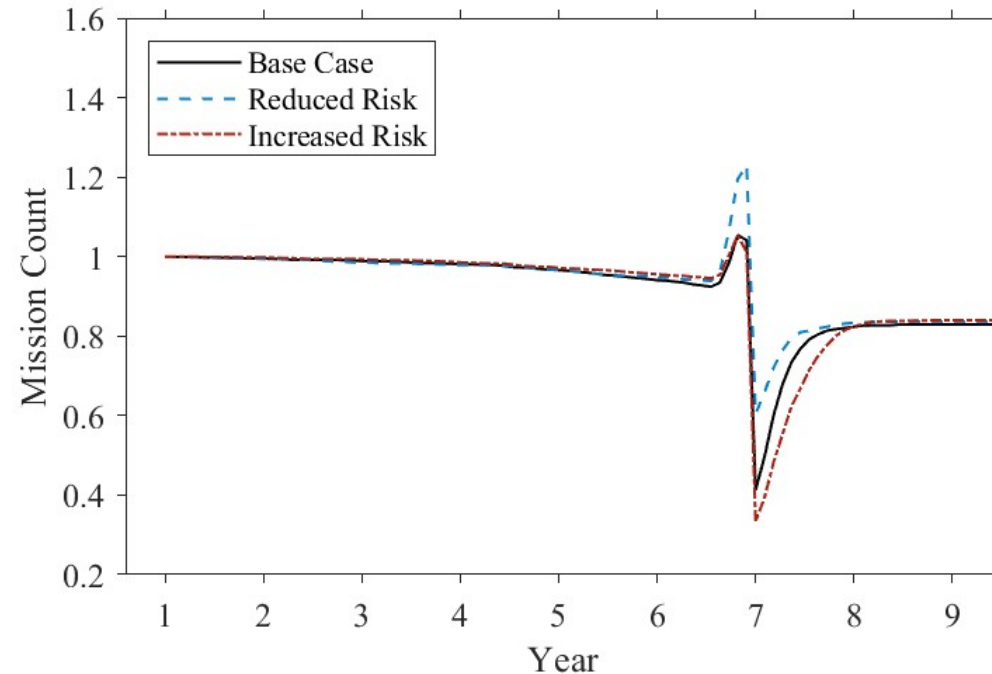


Sensitivity Analysis



How do adjustments to subjective input parameters affect model results?

- Should schedule risk be characterized as “Low,” “Low+,” or “Medium”?



Underestimating risk could have a larger impact on the model results than adding additional risk margin to the inputs

Model Improvements

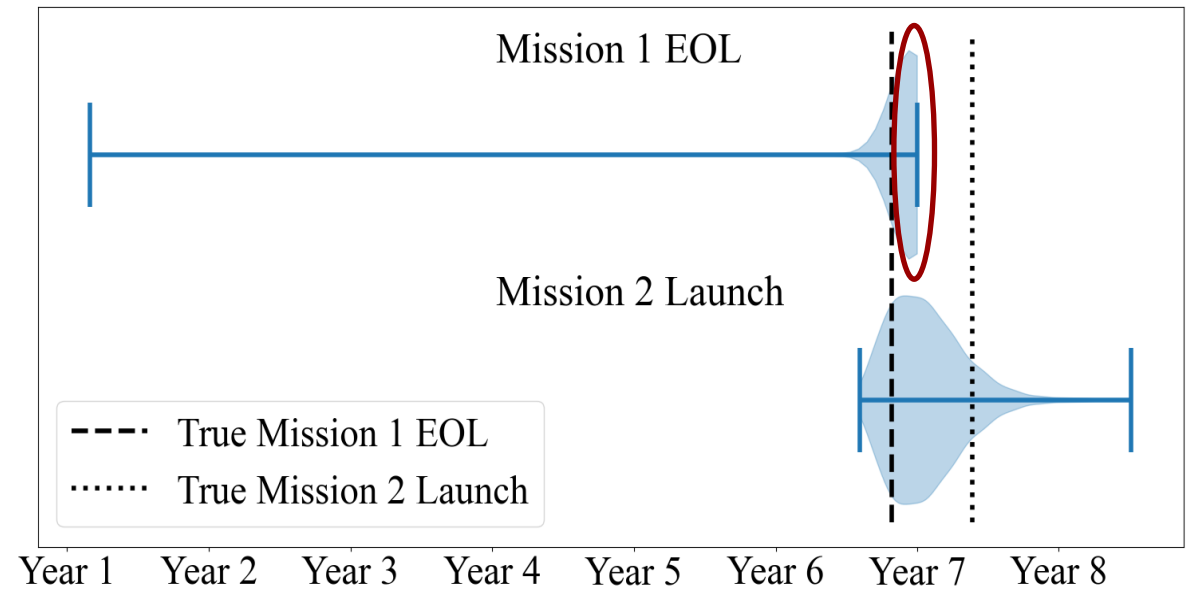


➤ Planned model improvements:

- 1) Modeling extended mission lifetimes
 - Addresses truncation of right side of probability density plot for Mission 1
- 2) Assess the relative ability of a mission to address a given science parameter, rather than using a binary mission count

➤ Proposed model improvements:

- 3) Include the capability to simulate the effects of redundant systems or other non-mission ending failures



Concluding Remarks

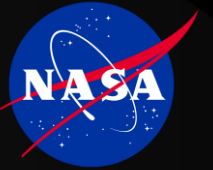


Simulation estimates are adequate in predicting mission timelines and likelihood of gaps for the selected mission architecture

- The model more closely predicted the true outcomes of the missions than what was planned during formulation
 - Mean EOL estimate for Mission 1 was within 2 days of the true EOL date
 - 177-day predicted delay of Mission 2 launch is indicative of the 294-day true delay
- The sensitivity analysis suggests that underestimating the risk of a system could have a larger impact on the model results than adding additional risk margin to the inputs

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