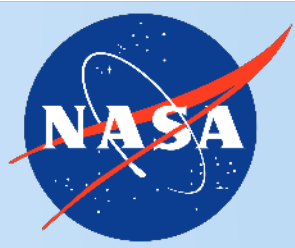


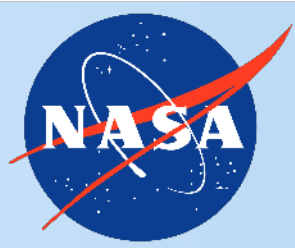
Integrated Computational Materials Engineering focused on Qual & Cert across NASA

Mallory James, MSFC; Andrew Kitahara, LaRC; Ed Glaessgen, LaRC
April 2025



A Sampling of Complementary ICME for Q&C Activities

- Computational Materials for Qualification and Certification (CM4QC) Steering Group
- Computational Materials Research Group at Langley
- Space Technology Research Institute – Institute for Model-Based Qualification & Certification of Additive Manufacturing



Today's Certification Approach

- Current approaches for certification rely heavily on **trial-and-error** where the outcomes from numerous combinations of parameters are evaluated experimentally.
- Iteration may occur on a particular part build numerous times to “dial-in” the recipe.
- Endless variations arise due to the sensitive nature of the metallurgical process and individual aspects of AM machines.
- Once qualified, the **AM process is not extensible**, limiting the ability to make design or manufacturing changes.

Example Laser Powder Bed Fusion Process

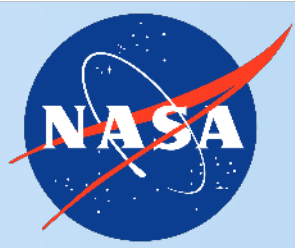


Iteration required on Steps 2-6 to Mature/Qualify Part and Processes

AM technology is demonstrating incredible benefits in wide-spread applications, design flexibility, affordability, schedule, performance, sustainability, and production on demand.

- **Certification is the #1 challenge:** complexity, unknowns, and delays are severely negating the benefits.
- ***Certification of remotely produced products is an unresolved technology gap.***





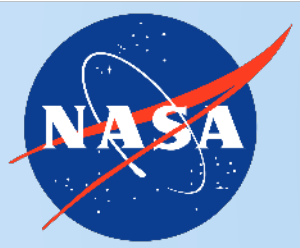
The Valley of Death

Mid Technology Readiness Level

The place where low TRL capabilities & technologies go to die

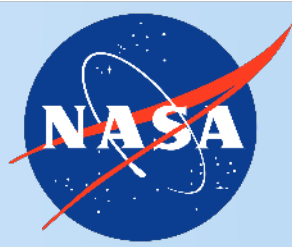
A Few (Rhetorical) Questions

- What is the gap between the State-of-the-Art and the State-of-Practice?
- How much foundational research is funded and published every year?
- How much of it languishes for decades, trapped in journal papers?
- Which components of this work (by NASA, others) can/should be matured as the foundation for new solutions for flight?



Computational Materials-informed Qualification and Certification Research Group

NASA Langley Research Center



Input from the Aviation Community

NASA/CR-2018-219771



Vision 2040: A Roadmap for Integrated, Multiscale Modeling and Simulation of Materials and Systems

*Xuan Liu and David Purner
Pratt & Whitney, East Hartford, Connecticut*

*Jared Kosters and Jack Holmes
Nextgent Group, Silver Spring, Maryland*

March 2018

NASA/TM-2019-220428



ARMD Workshop on Materials and Methods for Rapid Manufacturing for Commercial and Urban Aviation

*Jonathan B. Ransom, Edward H. Glaessgen, and Brian J. Jensen
Langley Research Center, Hampton, Virginia*

November 2019

NASA/TM-2021-15175



NASA / NIST / FAA Technical Interchange Meeting on Computational Materials Approaches for Qualification by Analysis for Aerospace Applications

*Edward H. Glaessgen
Langley Research Center, Hampton, Virginia*

*Lyle E. Levine, Paul W. Witherell, and M. Alkan Donmez
National Institute of Standards and Technology, Gaithersburg, Maryland*

*Michael Gorelik
Federal Aviation Administration, Scottsdale, Arizona*

*Nathan A. Ashmore
Boeing Research and Technology, St. Louis, Missouri*

*Richard R. Barto
Lockheed Martin Advanced Technology Laboratories, Cherry Hill, New Jersey*

*Corbett C. Battaille
Sandia National Laboratory, Albuquerque, New Mexico*

*Harry R. Millwater
University of Texas at San Antonio, San Antonio, Texas*

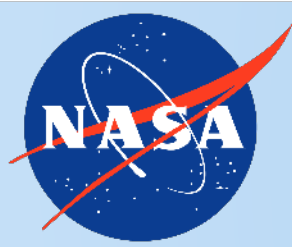
*Gerard J. Nanni
Bell, Fort Worth, Texas*

*Anthony D. Rollett
Carnegie Mellon University, Pittsburgh, Pennsylvania*

*Edwin J. Schwalbach
Air Force Research Laboratory, Dayton, Ohio*

*Vasishth Venkatesh
Pratt and Whitney, East Hartford, Connecticut*

May 2021



Computational Materials for Qualification and Certification (CM4QC) Steering Group

Formed through a NASA-FAA collaboration

Members: 10 industry, 8 government orgs, 8 universities/labs

Primary goals

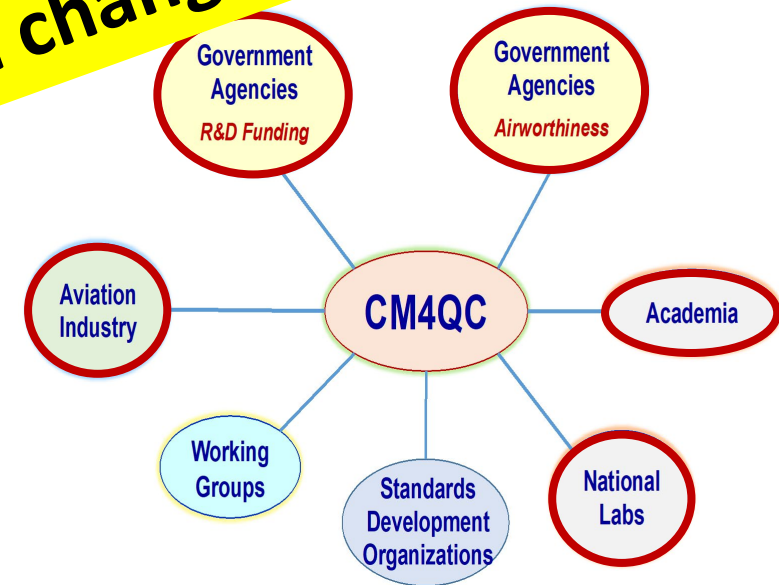
- Provide **coordination for and focus to investments** made by industry & government in the development of computational materials-based approaches for Q&C of process integration
- Identify key considerations and enablers required to **increase the acceptance of CM methods** used for Q&C of structural materials

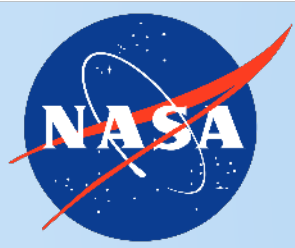
Operation

- High TRL: Understanding of the problem and key regulatory requirements
- Mid TRL: System-level research to develop computational materials & modeling capabilities
- Low TRL: Development of computational materials & modeling capabilities

Not “certification by analysis”
...Computational Materials-Informed Q&C
→ CM4QC is a paradigm change

Expected initial outcome –
Multi-year implementation plan





Steering Group Membership

Government

National Aeronautics and Space Administration*

Federal Aviation Administration*

National Institute of Standards and Technology*

Sandia National Laboratory*

U.S. Air Force Research Laboratory

U.S. Army Aviation Command

U.S. Navy Air Systems Command

Oak Ridge National Laboratory

Industry

BAE Systems Inc*

The Boeing Company*

Lockheed-Martin

Raytheon / Pratt & Whitney

General Electric Aviation

Spirit Aerosystems

Honeywell Aerospace

Southwest Research Institute

Northrop-Grumman Corp

Textron Aviation

Academia

Univ Texas San Antonio*

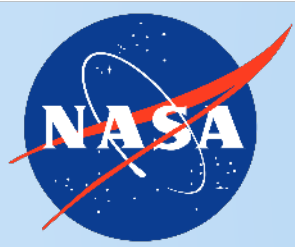
Carnegie Mellon Univ.*

Vanderbilt Univ.

Pennsylvania State Univ.

Northwestern Univ.

*CM4QC Leadership Team

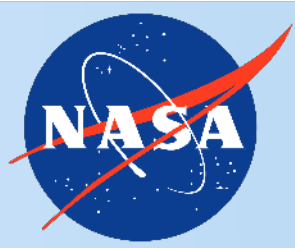


Highlights from the CM4QC Roadmap

Maturation of computational materials approaches in qualification and certification for metal additive manufacturing (as an example of process-intensive materials) in aviation

Produced by the Computational Materials for Qualification and Certification (CM4QC) Steering Group

The purpose of this group is not to establish Federal policy, but to enable a free-ranging discussion of emerging issues. Participants have provided non-consensus, non-voting advice, informational background, and technical and administrative assistance.



Computational Materials-informed Qualification and Certification Research Group

NASA and On-site Contractors

Sang-Hyon Chu ¹	Peter W. Spaeth ¹
Erik L. Frankforter ¹	David W. Wagner ¹
Andrew R. Kitahara ²	George R. Weber ¹
Patrick E. Leser ¹	Vesselin Yamakov ²
John Andrew Newman ¹	Saikumar Yeratapally ³
Joshua D. Pribe ²	Joseph N. Zalameda ¹
Brodan Richter ¹	

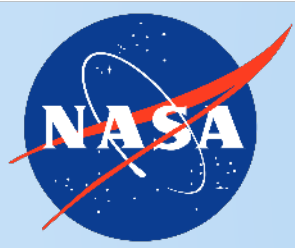
Adjuncts and Students

Evan B. Adcock
Claire E. Borden
Eric R. Burke¹
Devin E. Burns¹
Jared S. Fell¹
Ariel Gluck
Brian T. Gockel
Erick R. Ramirez
Sierra Stevenson
Seth T. Strayer

¹ National Aeronautics and Space Administration, Langley Research Center, Hampton, VA 23681

² Analytical Mechanics Associates, Hampton, VA 23666

³ Science and Technology Corporation, Hampton, VA 23666



Computational Materials-Informed Qualification and Certification of Additively Manufactured Flight Hardware

Q&C: Qualification and certification
AM: Additive manufacturing



Photograph

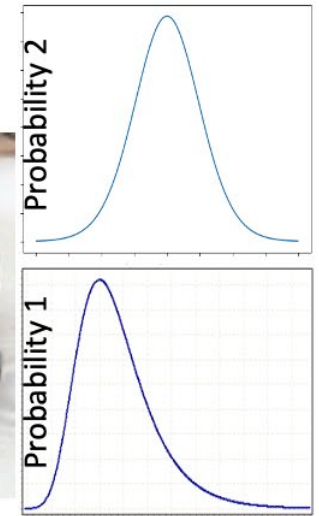


5 [mm]

Porosity



AM Component



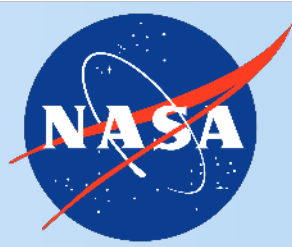
Process

Structure

Property/Performance

Small deviations in processing parameters may result in large differences in performance
Current approaches for Q&C of metallic materials (including AM) are entirely based on test data

Goal is to change the paradigm for Q&C: Develop a computational materials-informed ecosystem for quantifying sources of variability in fatigue performance of additively manufactured metallic materials through integrated multi-scale, multi-physics simulation, characterization and monitoring

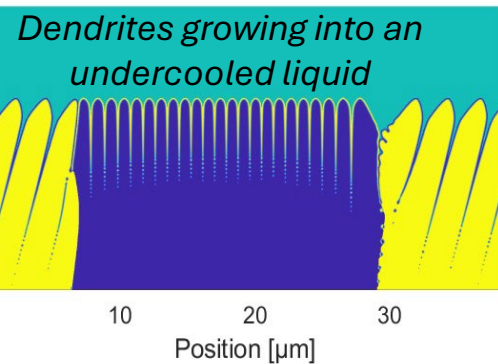
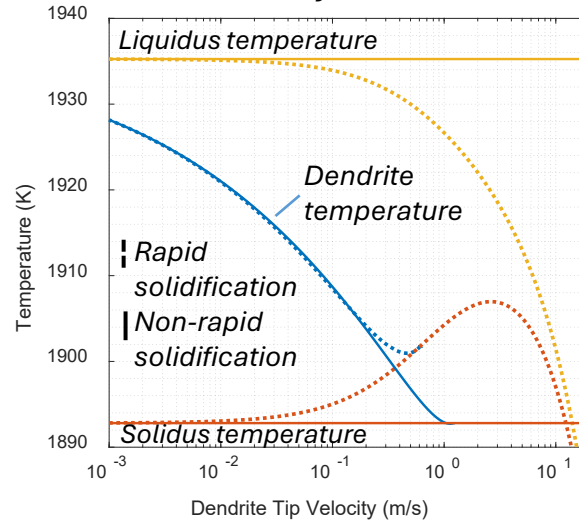


Grain-scale Process-Structure Modeling

Grain-scale Prediction of Microstructure Evolution

Analytical modeling of dendritic growth

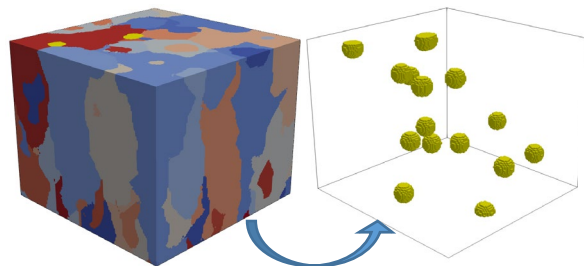
Dendrite velocity vs. undercooling



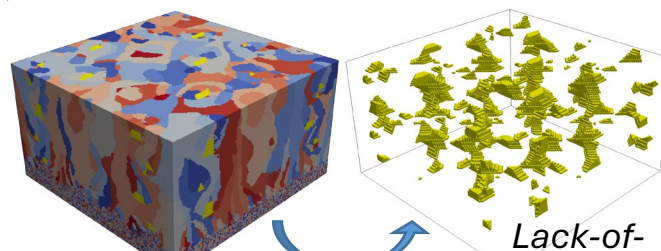
Phase-field modeling of dendritic growth (input to Monte Carlo simulation)

Defect modeling

Keyhole pores



(Geometric) Lack-of-fusion modeling

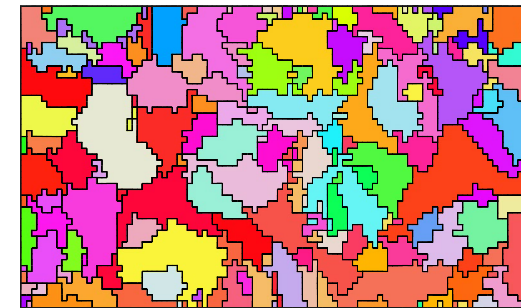


(Stochastic) Keyhole modeling

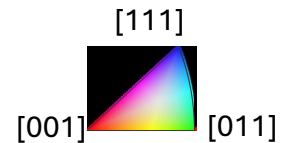
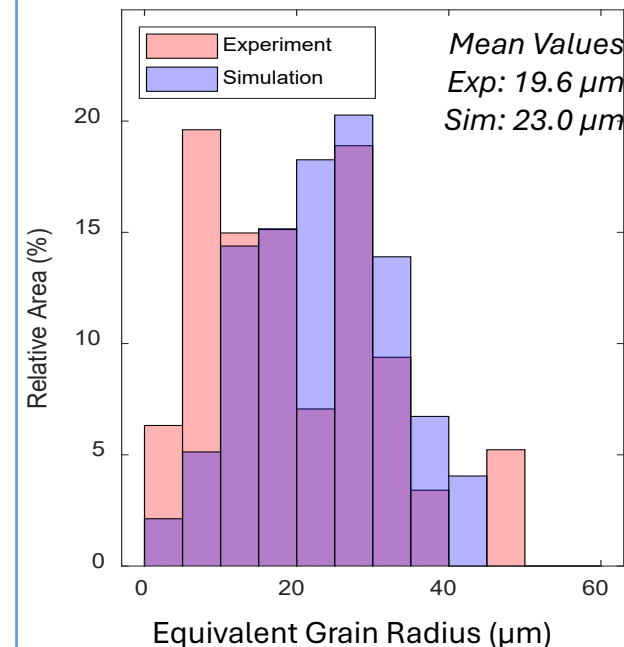
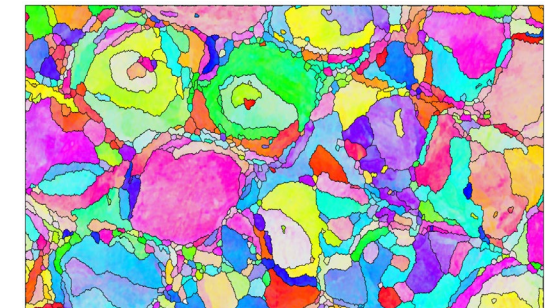
Lack-of-

Simulation vs. Experiment

Simulated Microstructure

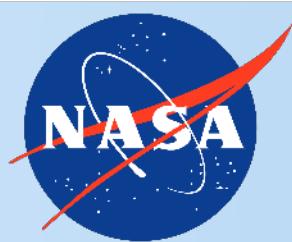


Experimental Microstructure



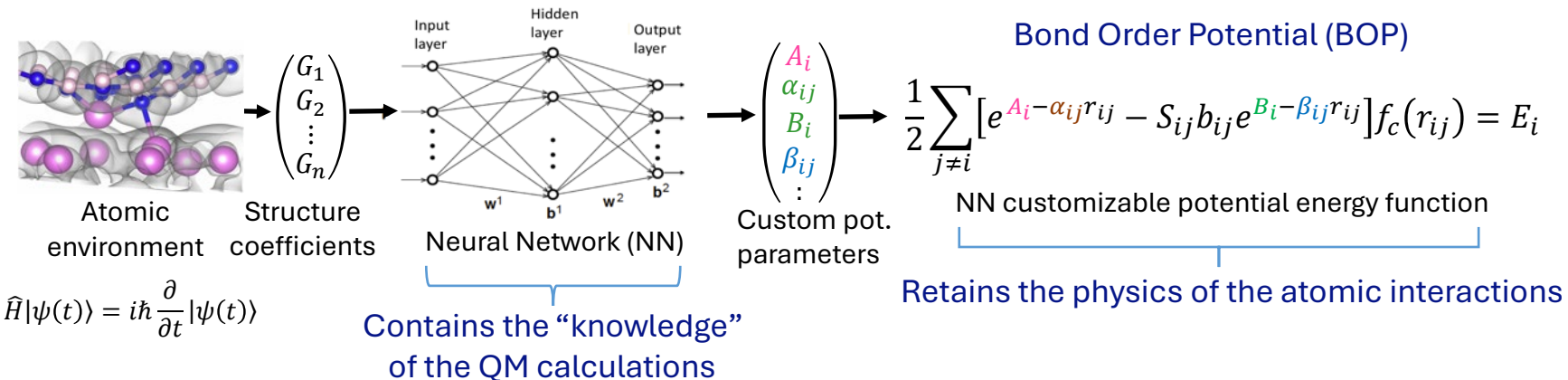
Observations:

- Calibrated microstructure evolution models can accurately capture simulate grain morphology
- Enables studying how process parameters affect resulting structure
- Provides a hand-off for structure-property-performance simulations



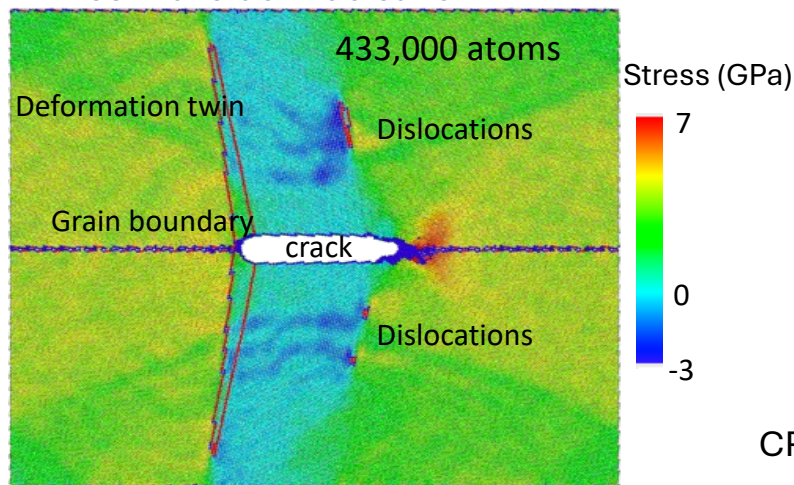
Physically Informed Neural Networks (PINN) Atomistic Simulations

Large scale atomistic simulations with quantum mechanics accuracy: machine learning and empirical potentials

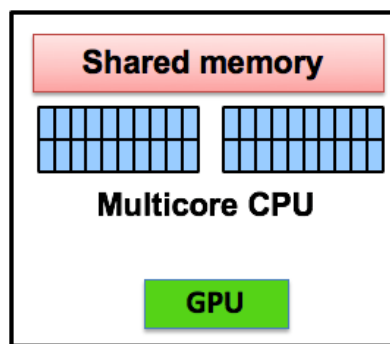


Quantum mechanics-based Density Functional Theory (DFT) calculations are used to train a NN

Fracture simulation with DFT precision:
central crack nucleation in Al



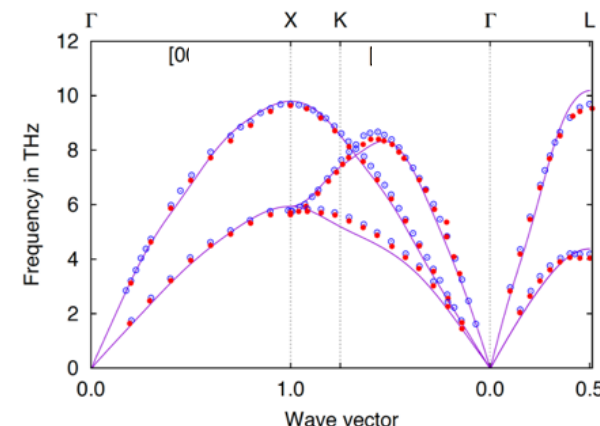
**Computing: optimized for
multicore processors with
graphic (GPU) accelerators**



427,330 atoms
40 Skylake cpus + 4 V100 GPU
CPU time for 12,000 integration steps: 14 cpu hours

Accuracy: PINN ~1% of DFT

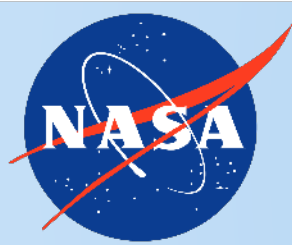
Phonon spectrum



Speed: PINN $\gg$ DFT

t (s) 100 Steps	PINN 16 cores	DFT 32 nodes
N=500 atoms	35 s	46,688 s
N=72,000	528 s	13.5 years*

*Extrapolation



Crystal Plasticity Framework and Applications

$$F = F^e F^p$$

elastic-plastic

$$L^p = \sum_{\alpha} \dot{\gamma}_{\alpha} (s_{\alpha} \otimes m_{\alpha}) \quad \text{slip rule}$$

$$\dot{\gamma}^{\alpha} = \dot{\gamma}_0 \frac{\tau_{\alpha}}{g} \left| \frac{\tau_{\alpha}}{g} \right|^{\frac{1}{m}-1}$$

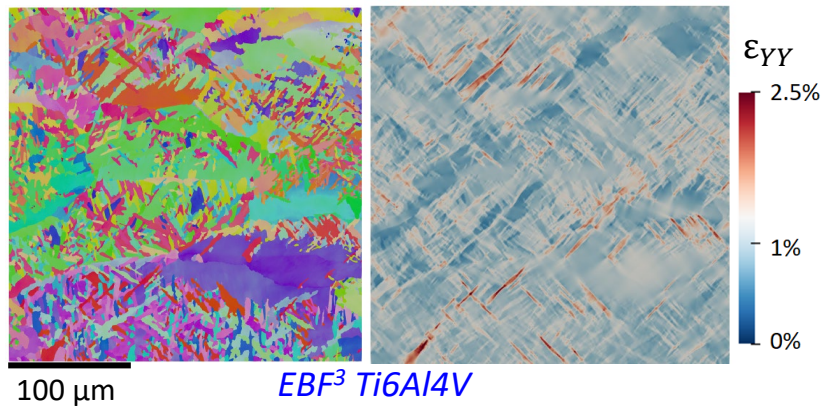
$$\dot{g} = G_0 \left(\frac{g_s - g}{g_s - g_0} \right) \dot{\gamma}_t \quad \text{Voce-Kocks hardening}$$

$$\dot{\gamma}_t = \sum_{\alpha} |\dot{\gamma}_{\alpha}|$$

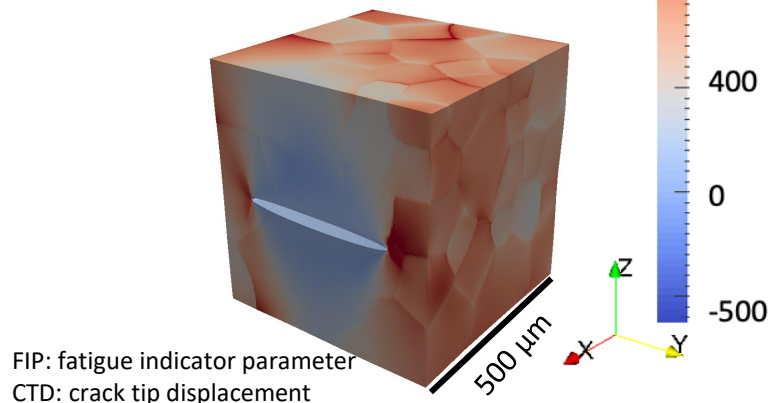
$$g_s = (g_0 + g_s^*) |\dot{\gamma}_t|^{\omega}$$

- F is deformation gradient
- F^e, F^p are elastic and plastic portion of F , respectively
- L^p is the velocity gradient; $\dot{\gamma}_0$ is the reference slip rate
- $\dot{\gamma}^{\alpha}$ is the slip rate along slip system α
- $\dot{\gamma}_t$ is the absolute sum of slip rate on all slip systems
- g is the slip resistance
- m is the shear rate sensitivity parameter
- τ^{α} is the resolved shear stress acting on slip system α
- $\dot{g}$ is the rate of hardening; G_0 is hardening coefficient
- N_{ss} represents the number of slip systems
- ω is saturation rate exponent
- g_s^* is normalized reference saturation hardness

Example application 1: Study structure-property relations in AM materials

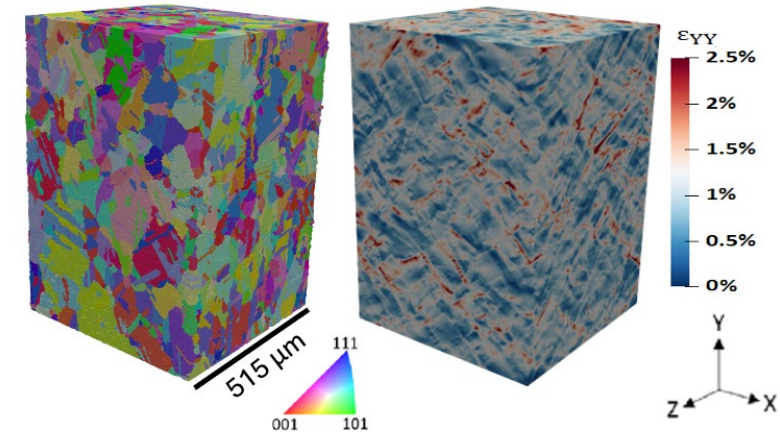


Example application 2: Study CTD/FIP relation using explicit cracks

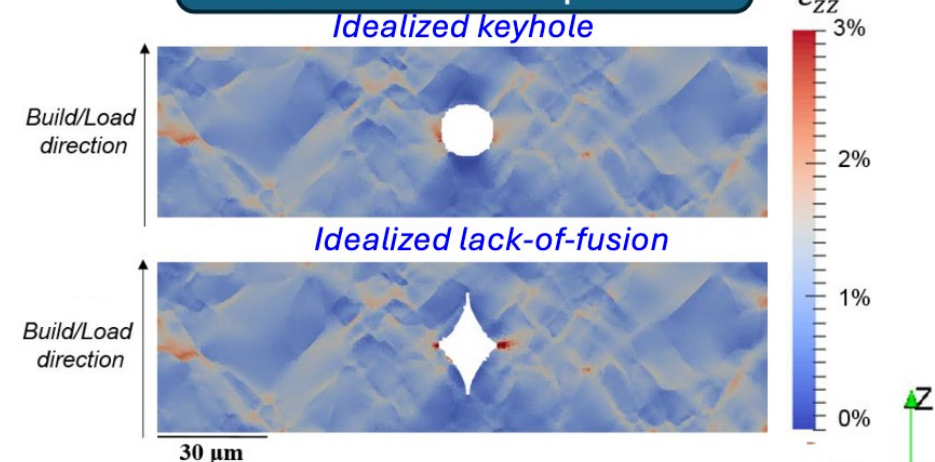


FIP: fatigue indicator parameter
CTD: crack tip displacement

L-PBF Inconel 625

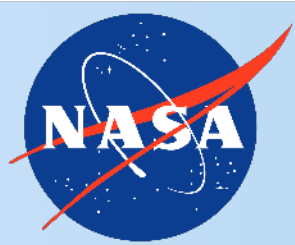


Example application 3: Strain localization near pores



Nodes: 4.5E6
Applied strain: 1%

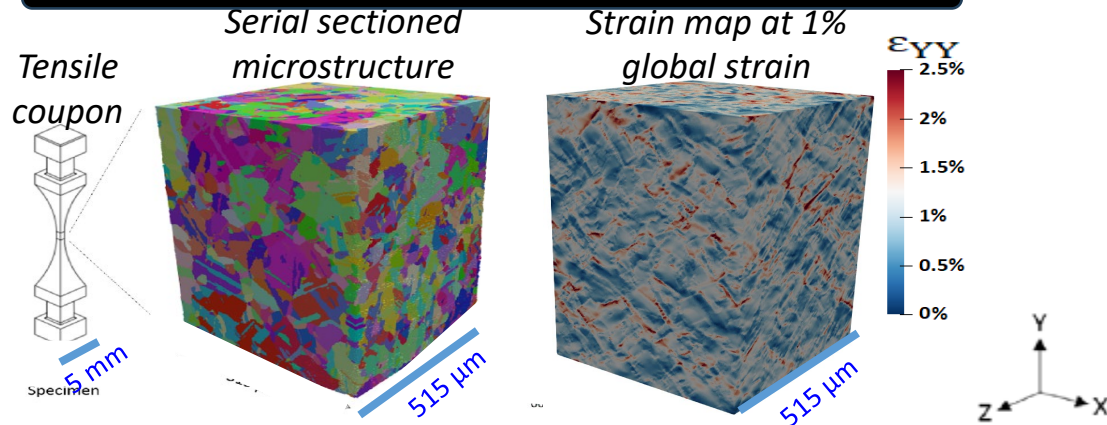
Processors: 400
Simulation time: ~14 hours



Microstructure Generation

Microstructure models from characterization

Method 1: Use characterization data directly

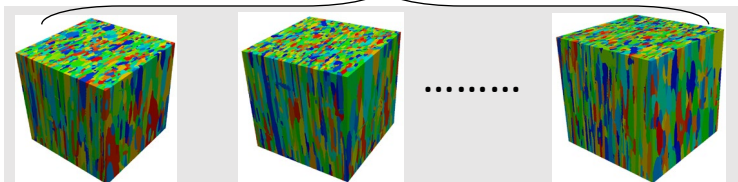


Method 2: Generate statistically equivalent microstructures based on characterization data

EBSD/XCT/HEDM/Serial sectioned microstructure

Statistics (grain size/shape/grain boundary character/texture etc.,)

Statistically equivalent microstructures

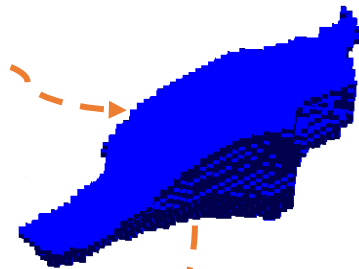
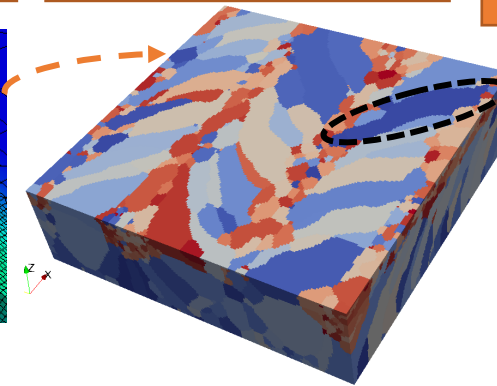
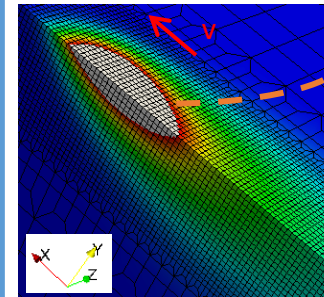


Process-specific 3D microstructure

Finite Element Thermal Analysis

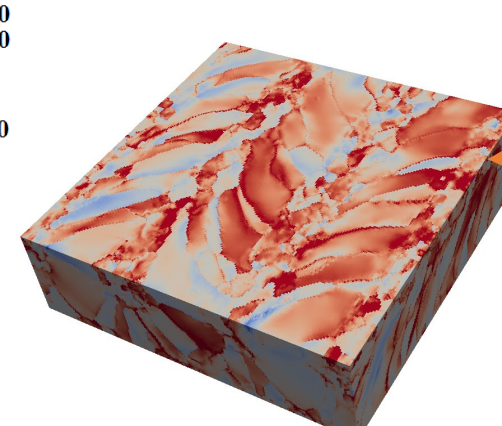
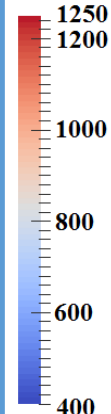
AM process simulation SPPARKS¹

Use Output from SPPARKS and generate .stl files of grains using DREAM. 3D²



DREAM. 3D³/Gmsh

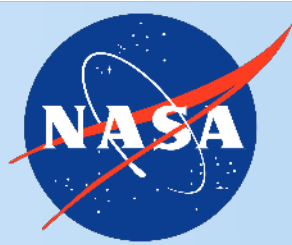
von Mises stress (MPa)



¹SPPARKS: spparks.sandia.gov

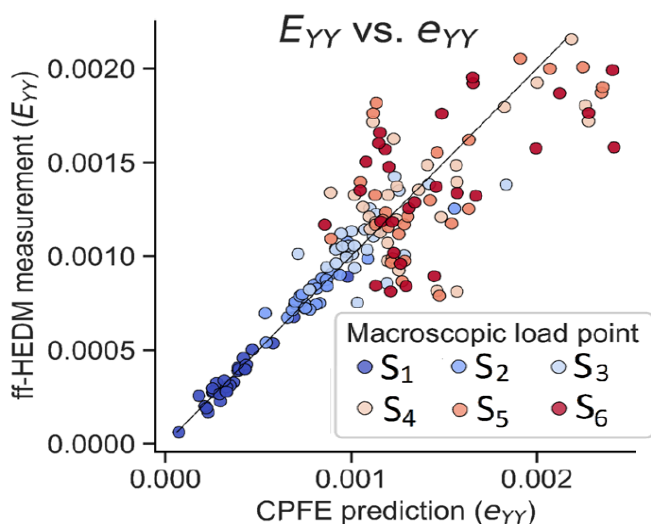
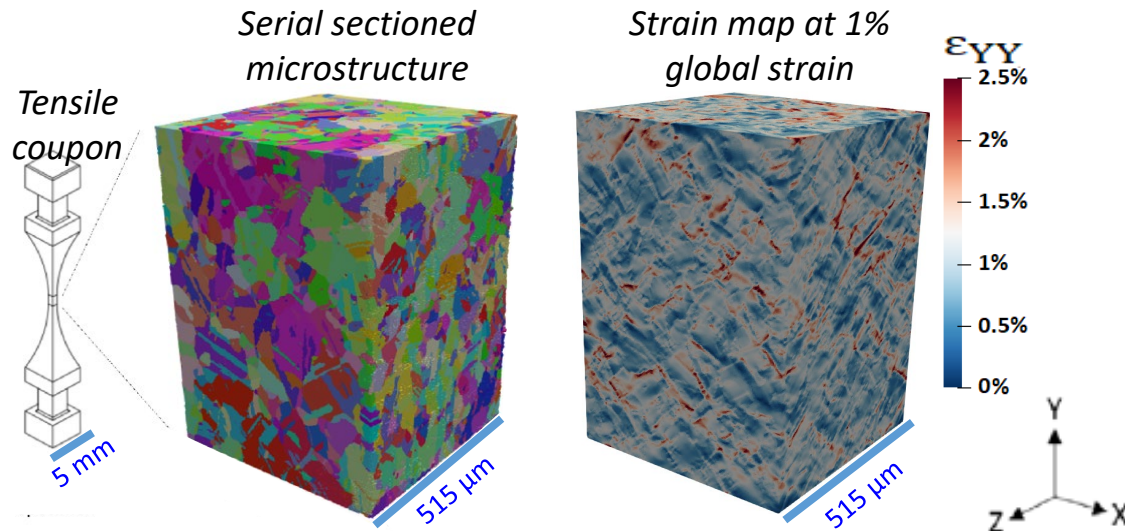
²DREAM.3D: dream3d.bluequartz.net

³ScIFEN: <https://software.nasa.gov/software/LAR-18720-1>



CP Simulation Validation Efforts

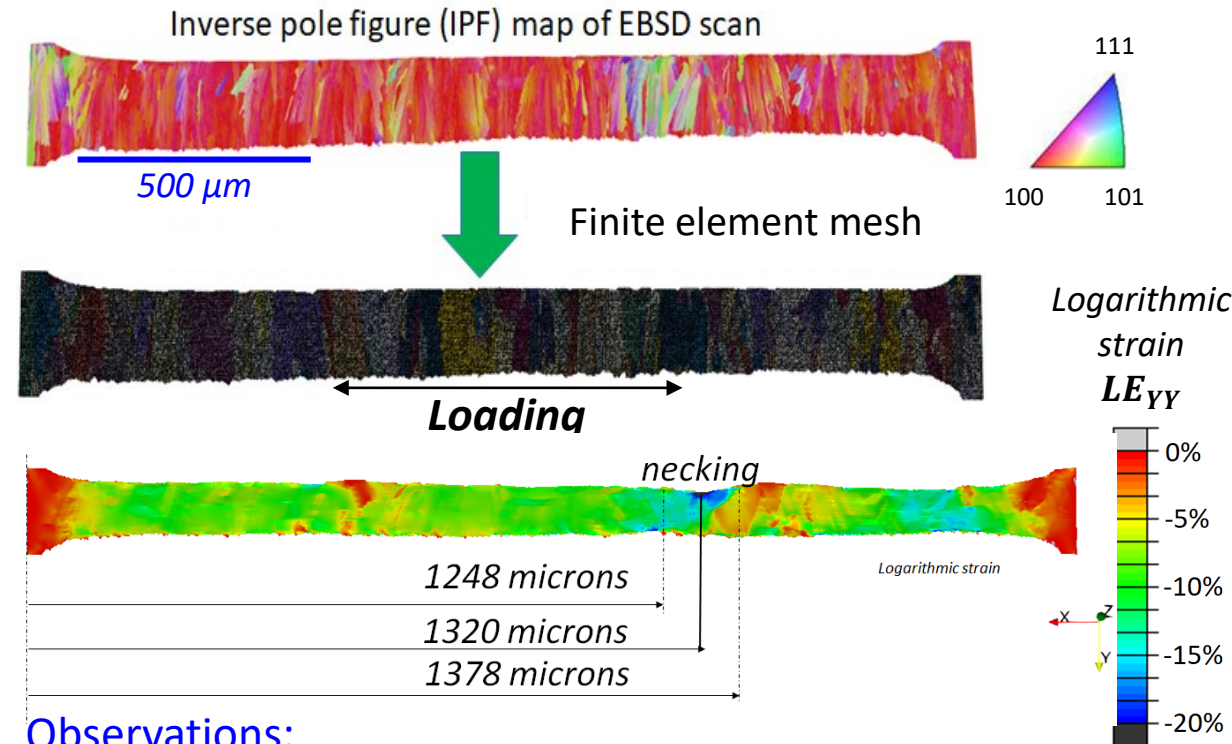
AFRL AM Modeling Challenge (2020)



Observations:

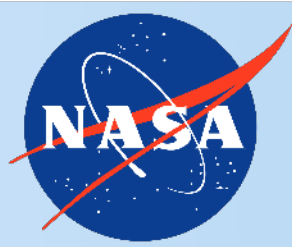
- CPFE predictions show good agreement with HEDM measurements for grain-average elastic strains in elastic regime
- Discrepancies observed in plastic regime
- Percentage error in elastic modulus is $\sim 1.5\%$

NIST AM-Bench Modeling Challenge (2022)

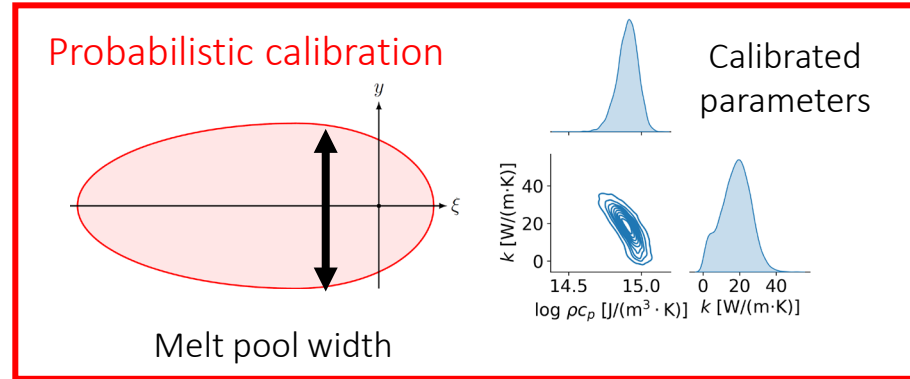
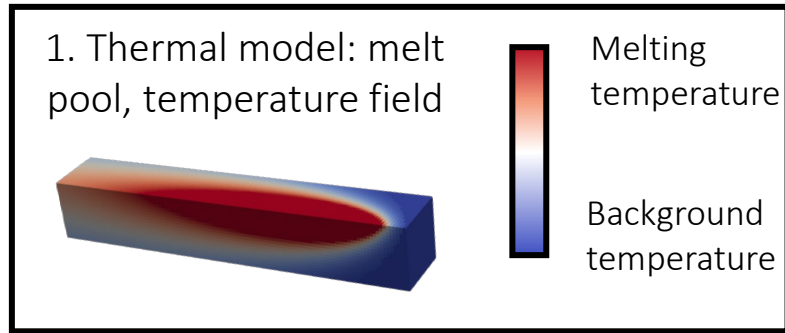


Observations:

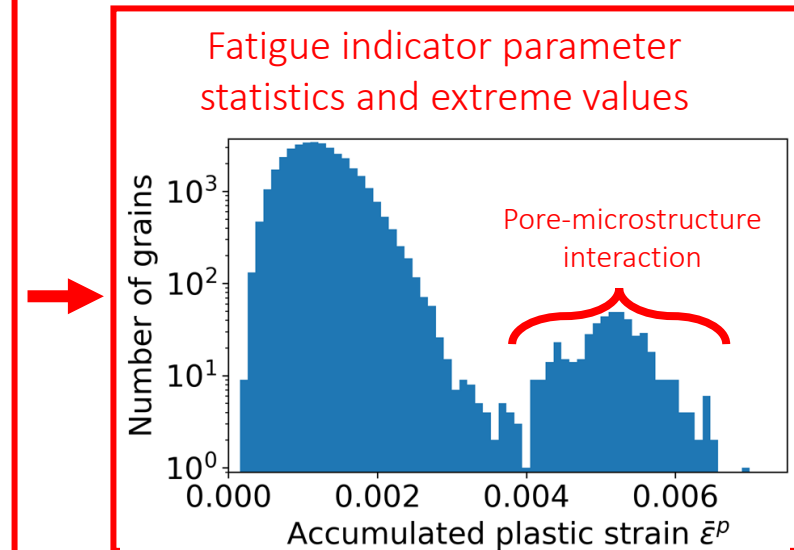
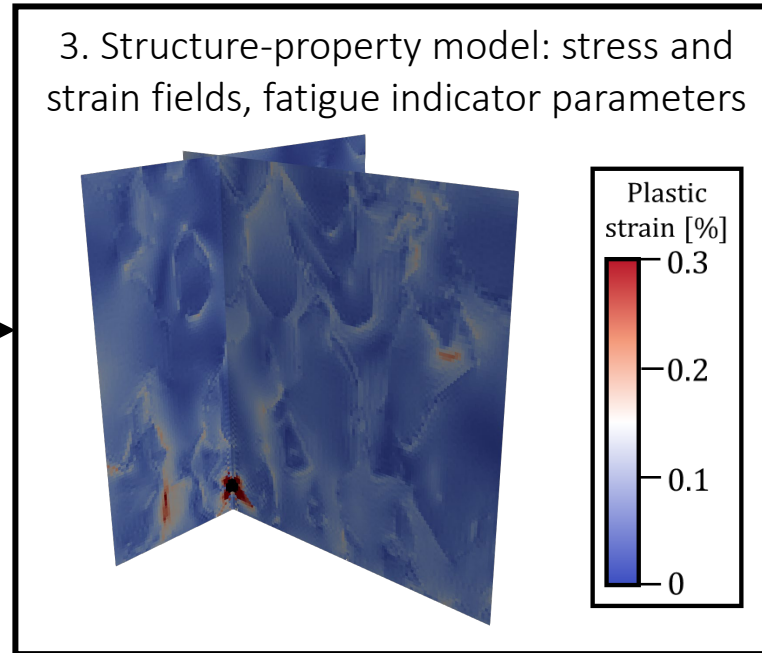
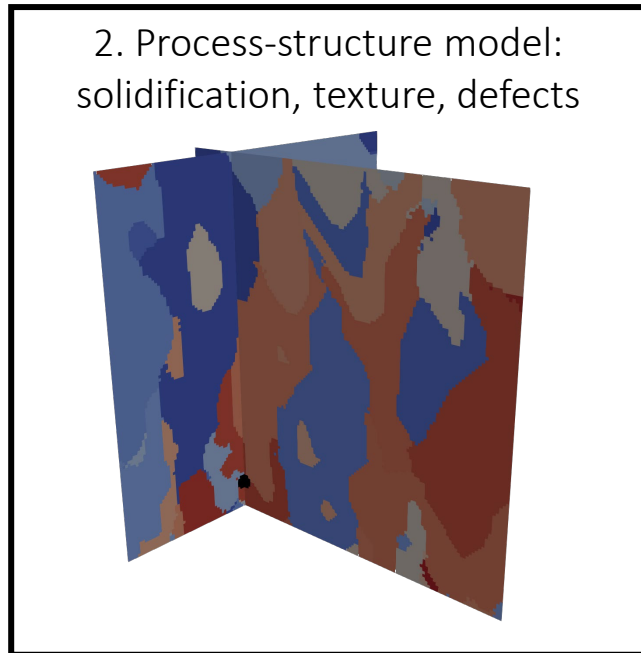
- Percentage error in predicted elastic modulus is $\sim 40\%$
 - Measured elastic modulus: 102 GPa
 - Predicted elastic modulus: 141 GPa
- High prediction errors observed across all participating teams
- Sources of discrepancy
 - Lack of full 3D microstructure (only 2D EBSD data was given)
 - Potential error in C_{ij} values (borrowed from literature)



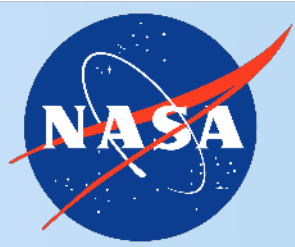
Process-Structure-Property Simulations with Uncertainty Quantification



Repeat simulations



POCs: Josh Pribe / Pat Leser



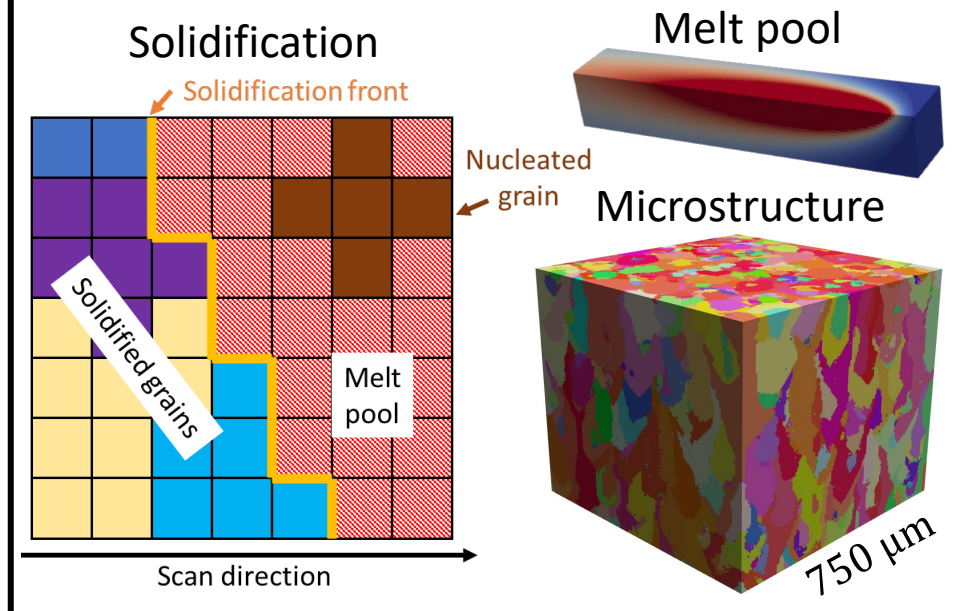
Global Sensitivity Analysis

Identifying inputs that need further calibration and process conditions that need tighter control using global sensitivity analysis of process-structure model

Uncertain inputs

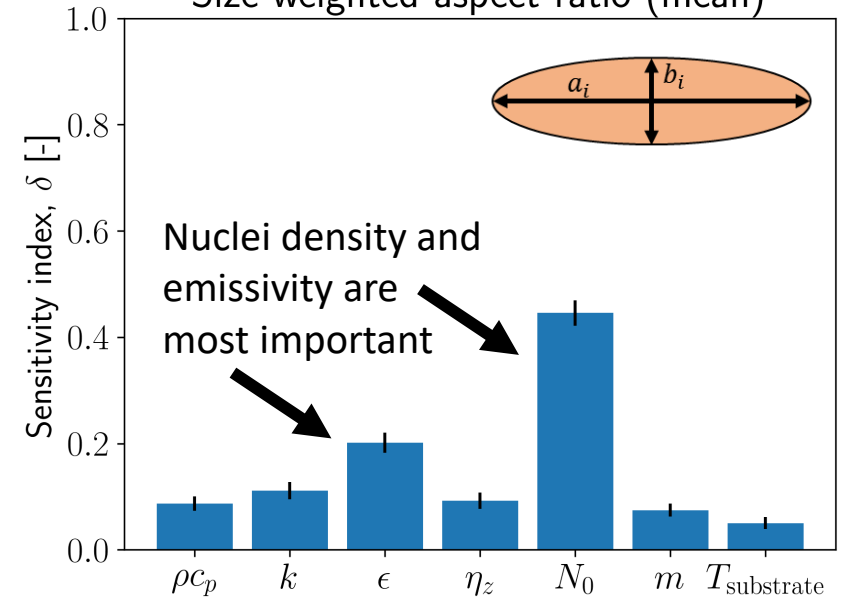
Material properties, process conditions

Thermal and process-structure simulations

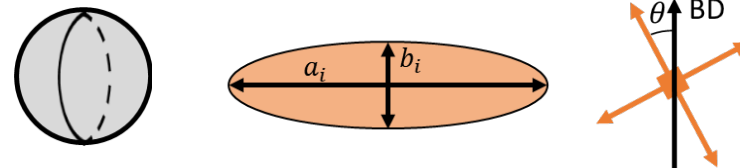


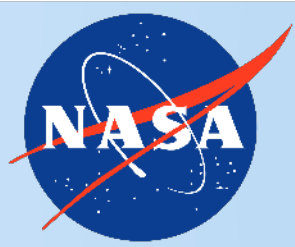
Sensitivity indices

Size-weighted aspect ratio (mean)



Microstructure quantities of interest





Configurable Architecture Additive Testbed (CAAT)

Configurable architecture AM testbed including in-situ non-destructive evaluation (NDE) for build quality assessment, informing qualification / certification models, processing condition measurements, and comparison to post build inspections with data fusion

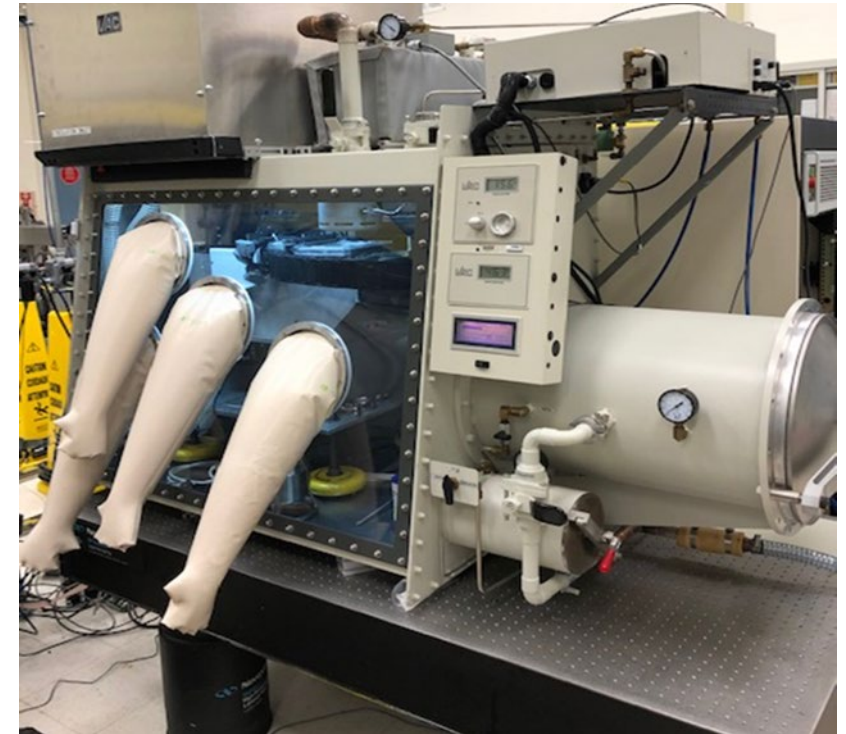
Approach: Achieve high-fidelity understanding and measurements of process to structure relationships so that quality of build assessments can be informed in industry relevant conditions

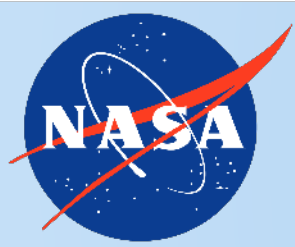
Establish: the Configurable Architecture Additive Testbed (CAAT) Lab

The CAAT lab studies the LPBF additive manufacturing using:

- 1) CAAT process capabilities:
 - 1) Max power: 700W
 - 2) Max speed: 2.5 m/s
 - 3) Minimum spot-size: 35 μm
- 2) Suite of in-situ process monitoring: Fully registered
- 3) In-situ monitoring to calculate, calibrate, and validate process metrics at the part scale
- 4) Connection with higher fidelity models
 - 1) thermal measures integrate with microstructure models

The CAAT lab is well suited to develop in-situ NDE based on the P-S-P models for LPBF additive manufacturing



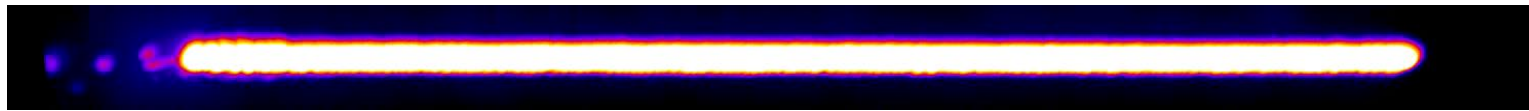


In-situ NDE for Metal Additive Manufacturing¹

Near Infrared 750 nm Band Imagery Reconstructed to Follow Build Path²



Near Infrared 750 nm Band Imagery Reconstructed for NDE Inspection



In-situ NDE Inspection Image using Principal Component Analysis



X-ray CT Validation of Detected Damage



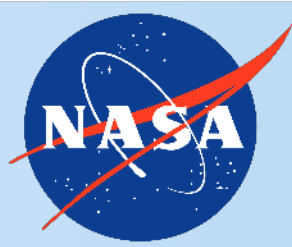
Use Co-axial to the
Laser Thermal
Imaging to Monitor
Temperature

Reconstruct Thermal
Imagery to Follow
the Laser Build Path
and Obtain Layer by
Layer Temperature

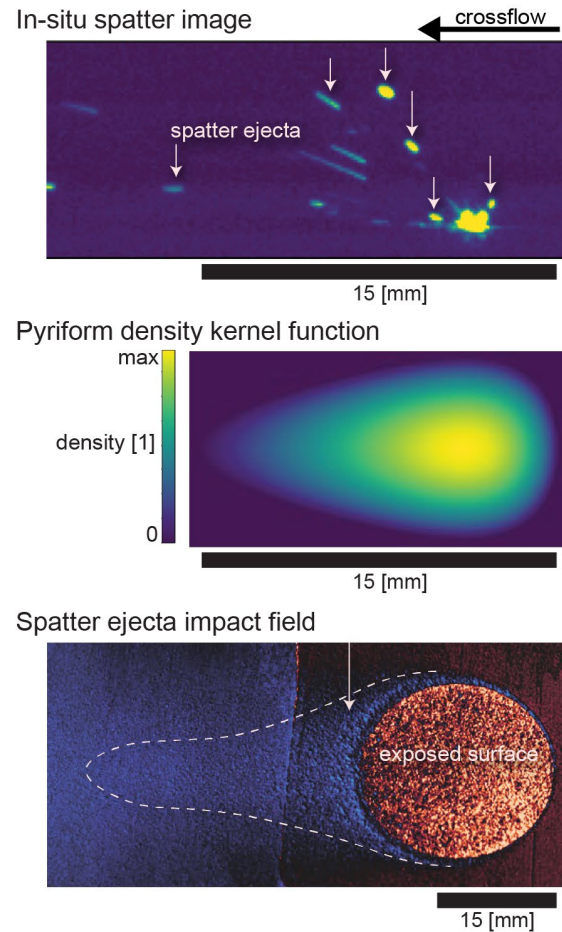
Process the Layer
Temperature
Responses for
Changes in Transient
Cool Down to Detect
Defects

¹ NASA US Patent 11.027.332, "In-situ Characterization and Inspection of Additive Manufacturing Deposits using Transient Infrared Thermography"

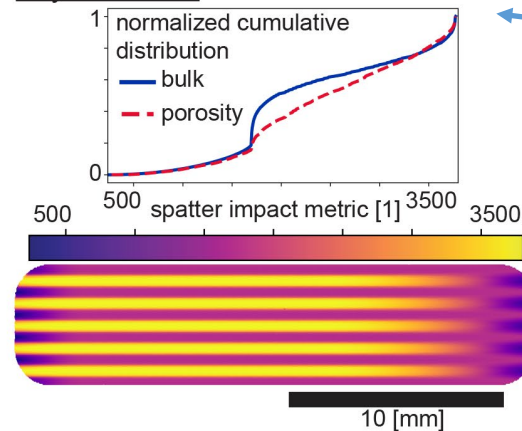
² Joseph N. Zalameda, Samuel J. A. Hocker, Peter W. Spaeth, Brandon Widener, "Near-infrared in-situ measurements for the detection of keyhole porosity in autogenous Ti-6Al-4V welds," Proc. SPIE 13047, Thermosense: Thermal Infrared Applications XLVI, 130470X (7 June 2024); <https://doi.org/10.1117/12.3014612>.



Spatter and Defects¹

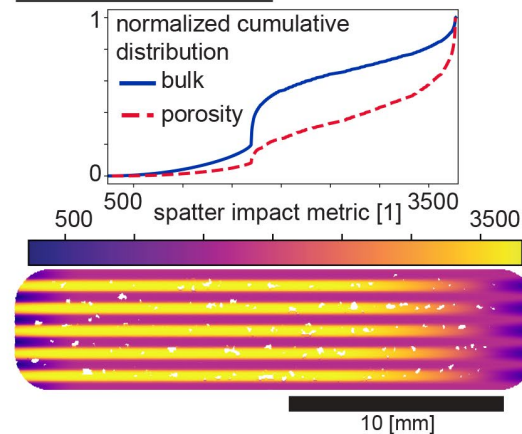


Keyhole mode



X-ray CT
Validated
Defects

Lack of fusion mode



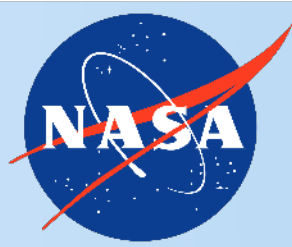
X-ray CT
Validated
Defects

Process defects are statistically correlated with the spatter impact metric.²

¹ NASA US Patent Application 18/224,935, "A Method and a Pyriform Process Metric to Predict and Mitigate Spatter- Induced Defects in Powder Bed Fusion-Laser Beam Metals AM"

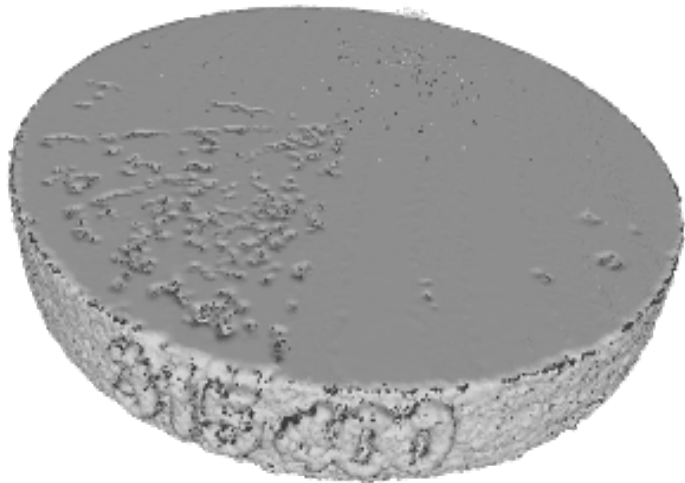
² Hocker, Samuel J.A., Andrew R. Kitahara, Brodan Richter, Sang-Hyon Chu, Brandon Widener, Peter W. Spaeth, Joseph N. Zalameda, and Edward H. Glaessgen. "A Novel Additive Manufacturing Process Metric for Predicting Spatter Related Porosity in Laser Powder Bed Fusion." Integrating Materials and Manufacturing Innovation (Nov 2024).

<https://doi.org/10.1007/s40192-024-00376-0>.

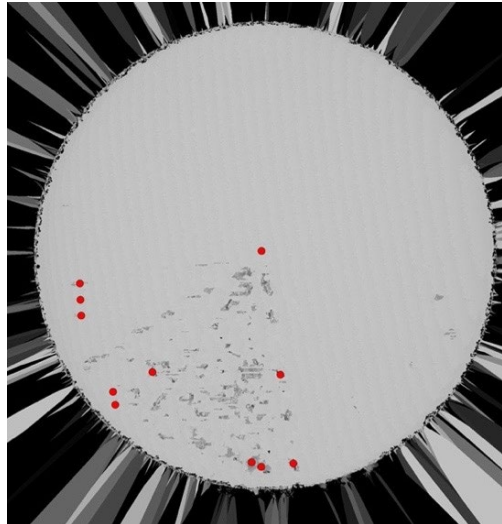


Improving X-Ray CT Defect Characterization by Integration with Optical Profilometry

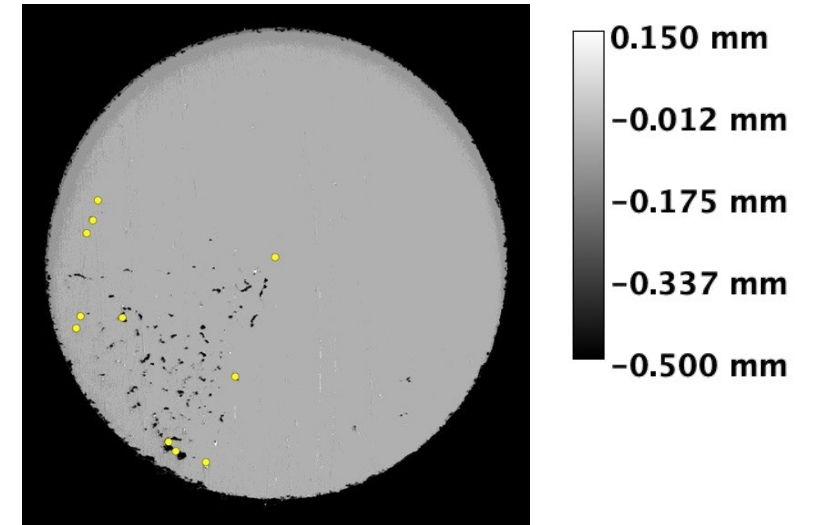
Component surface derived from X-ray CT segmentation



Surface depth estimated from X-ray CT segmentation



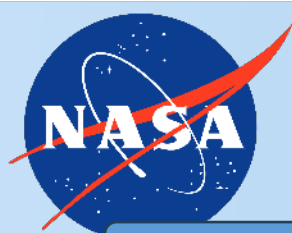
Comparison against high resolution surface profilometry



- Improvements to defect detection and metrological accuracy of X-Ray CT for AM metal porosity detection
 - Material agnostic algorithm, may be applied to other material and defect types
- Approach may be used to both characterize and optimize X-Ray CT volumetric defect detection

Patent Pending
E. Frankforter and P. Spaeth: “System and Method of Quantifying and Improving X-Ray Computed Tomography Measurement Accuracy”

X-Ray Computed Tomography (CT) Integration with Optical Profilometry Enables Improved Measurement Accuracy for Additive Manufacturing Porosity Defects

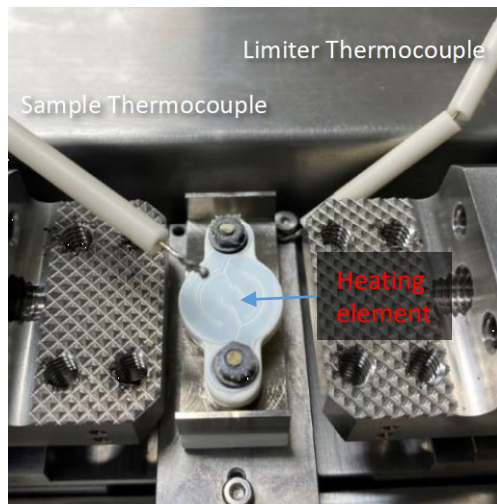


Testing at Elevated Temperatures

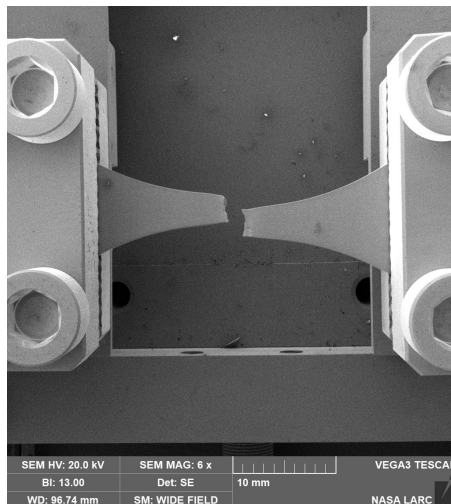
In-situ Mechanical Tests at Elevated Temperatures

High-Temperature Setup for In-Situ Tests

- A significant impact of **operation temperatures** on crack behaviors in Ni-based superalloys
- In-situ load stage system is rated up to 800°C
- **Thermal electron emission** (starting over 500°C) interferes SEM-DIC and EBSD
- Local heating induces thermal expansion in both the sample and the stage, causing vibration and drift of samples during loading
- Solutions to high-temperature in-situ: contact heater, thermal insulation to isolate the other parts, uniform thermal expansion



Heating element for in-situ stage

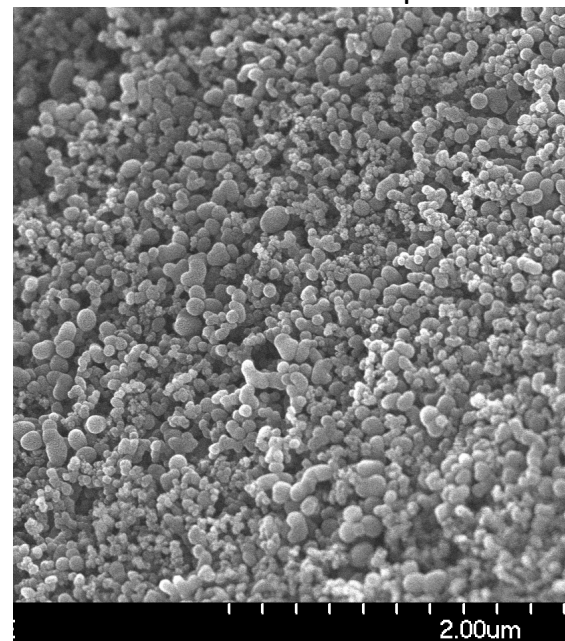


In-situ tensile test under SEM

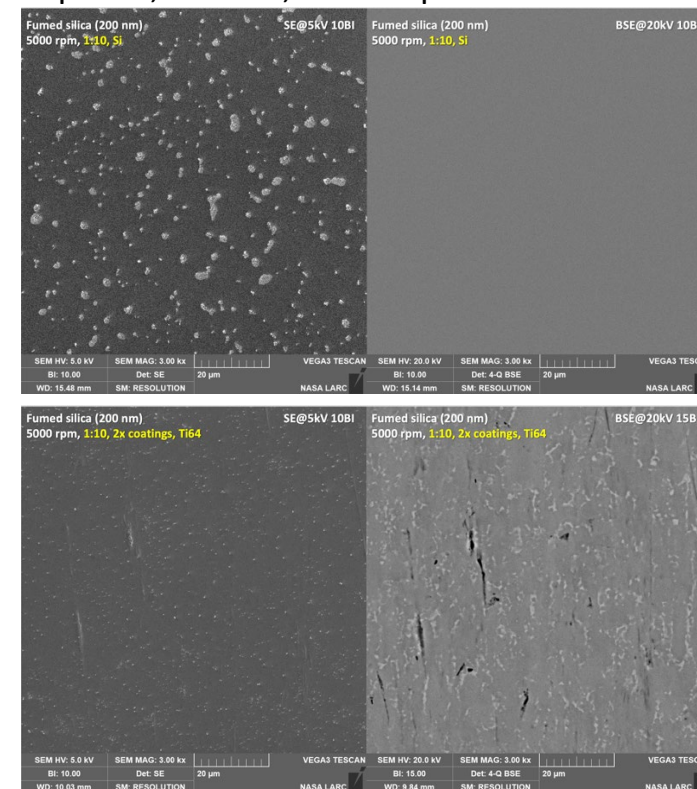
High-Temperature Micro Speckle Patterning

High-Temperature Speckle Pattern – Silica Nanoparticles

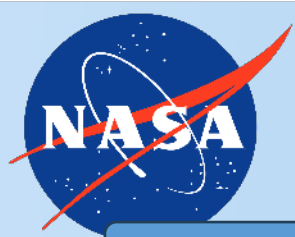
- High-temperature micro speckle patterns need **thermal stability** and **amorphous structure** for simultaneous acquisition of DIC and EBSD
- Fumed silica nanoparticles: amorphous, 200 nm, stable up to 1000°C



Fumed silica nanoparticles: amorphous, 200 nm in average size, thermally stable up to 1000°C



High-temperature speckle patterns of fumed silica nanoparticles on (top) silicon wafer and (bottom) Ti64 (left: SEM image; right: BSE image)

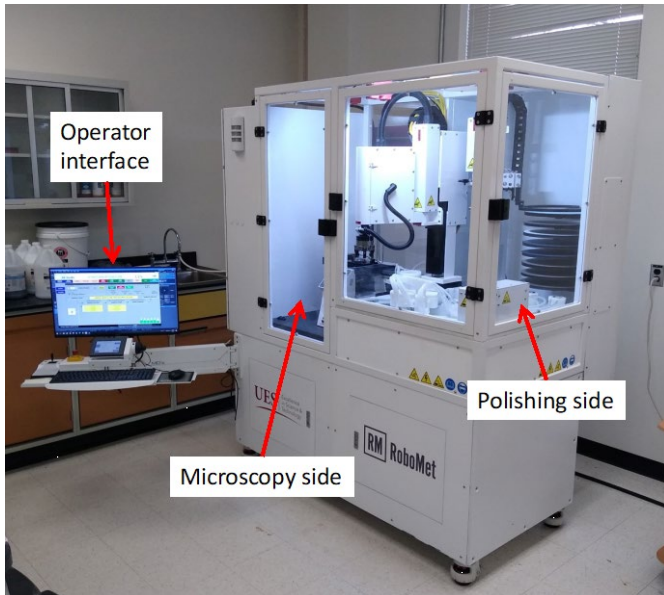


Microstructure Characterization and Residual Stress Analysis

Microstructure Characterization Tools

Microstructure Analysis and Fabrication

- Microstructure/composition analysis:
 - SEM: Secondary and Backscattered Electron Imaging
 - Electron Back Scatter Diffraction (EBSD)
 - Energy Dispersive Spectroscopy (EDS)
- Nanofabrication:
 - Plasma Focused Ion Beam (PFIB)
- 3D microstructure investigation
 - RoboMet to identify pores and cracks



RoboMet

- Automated Polishing and Microscopy Image Capturing
- Rapid sample preparations
- 3D microstructure imaging
- Serial sectioning for grain size and pore distribution in AM materials
- EBSD sample preps
- Interrogation volumes:
FIB 0.001 mm³ vs. RoboMet 1000 mm³

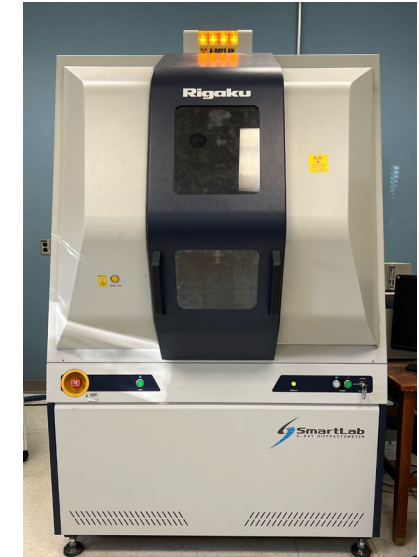
Residual Stress Analysis

Residual Stress Analysis of AM builds

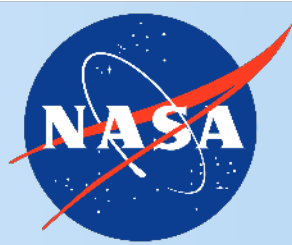
- Hole drilling method
 - Hole drilling with PFIB and measuring the resulting strains
 - Local residual stress can be measured
 - Destructive method
- XRD method
 - Nondestructive determination of surface/subsurface residual stress of AM build samples
 - Only shallow (<0.001 inch) surface layer is measured, requiring electrolytic polishing to remove top layers of subsurface



Helios PFIB for nanofabrication



XRD for residual stress analysis



The TRUST-AM Platform

New Technology Report: LAR-20454-1
“Trust and Reliability through Unified Simulation
and Testing for Additive Manufacturing”

Platform for prototyping Computational Materials-informed Qualification and Certification

State of the Discipline: Qualification and Certification (Q&C)

Industry needs learned from the CM4QC Steering Group

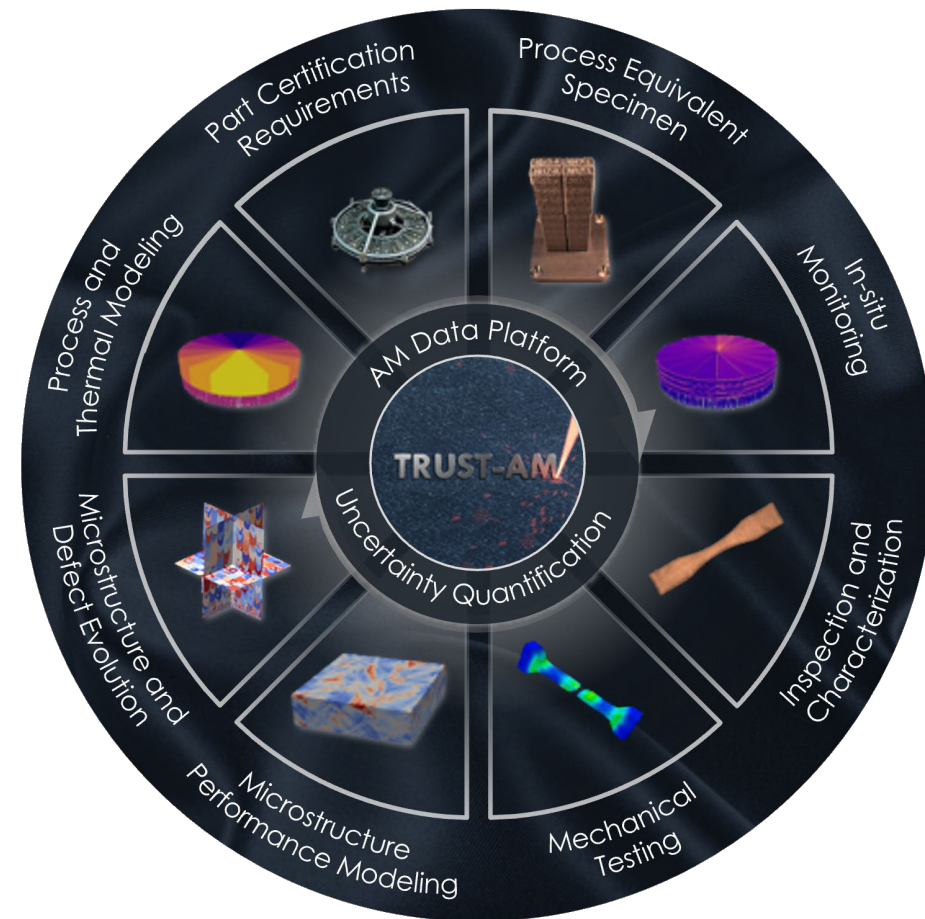
- Time and cost of Q&C limits adoption of additive manufacturing
- Manufacturing technology is outpacing regulatory environment

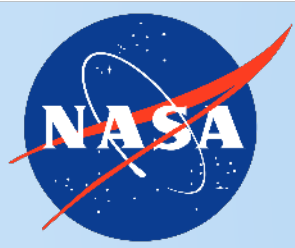
TRUST-AM

- An agile response from NASA's Transformational Tools and Technologies (T³) Project to address industry needs for digitally accelerated Q&C
- Augments physical testing requirements with modeling, simulation, uncertainty quantification, and data science
- A platform for prototyping Computational Materials-informed capabilities as part of a Q&C package

Impact

- Rapidly mature and transition Computational Materials research and development as the foundation for OEM-relevant engineering technologies
- Increase trust in modeling and simulation while reducing cost of Q&C





IMQCAM

Institute for Model-based Qualification & Certification of Additive Manufacturing

NASA Space Technology Research Institute

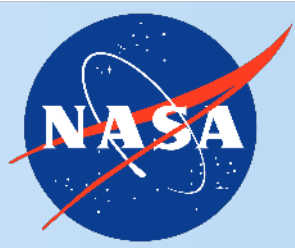
Co-Directors: A.D. (Tony) Rollett (CMU), PI & Somnath Ghosh (JHU)

Core Team: Sankaran Mahadevan, Caglar Oskay, Pranav M Karve (Vanderbilt); Jamie Guest, Jaafar El-Awady, Dave Elbert, Maggie Eminizer (JHU); Li Ma, Michael Presley, Steven Storck (JHU-APL), Harry Millwater (UTSA), Tao Sun (NWU), John Lewandowski (CWRU); Mohadeseh Mousavi-Taheri, Sneha Narra, Bryan Webler (CMU); David Furrer (P&W); James Sobotka (SwRI)

Central goal: Innovate a model-centric workflow that closes the critical gaps between current capabilities and what is necessary for efficient qualification and certification of parts by metals additive manufacturing , meeting NASA standards.

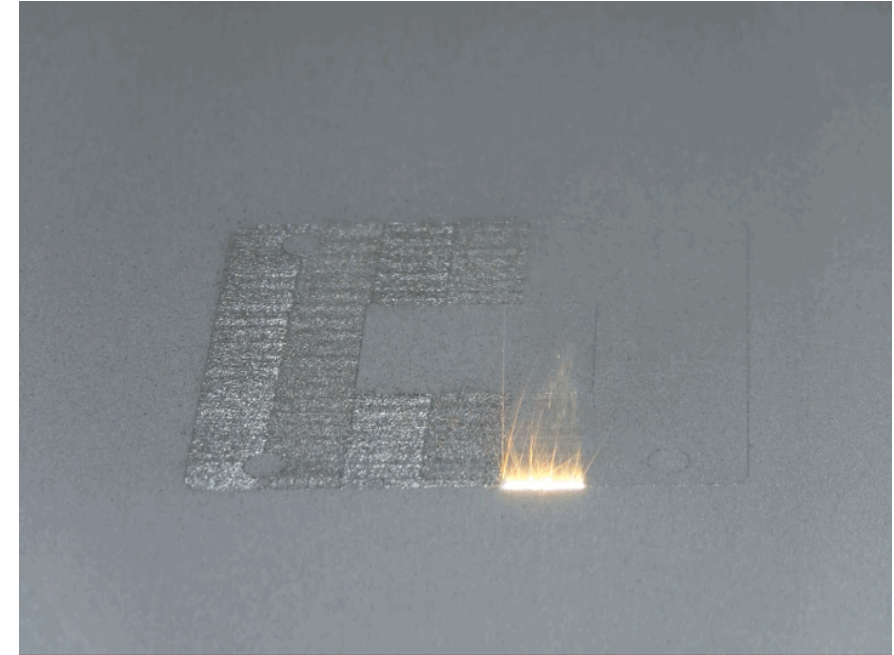
**Carnegie
Mellon
University**





Motivation for the Institute

- A new transformation of advanced AM technologies to build spaceflight hardware for NASA's exploration missions is upon us.
- This wave is enabled by dramatic changes in design and manufacturing paradigm brought about by AM.
- However, many of the inherent benefits of digitally oriented AM are being negated by the cost and schedule of today's all empirical development and certification cycles. [1]
- **The more we understand about the relationship between the central AM process, the material it produces and the way that the material performs, the better we can model it. In this way, ICME is a powerful tool to apply this process-structure-properties understanding to advance certification by analysis paradigms.**



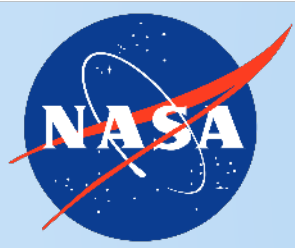
Credit: [2]

This effort will leverage digital twin technology to significantly decrease the resources needed in the current certification cycles further unlocking the value of AM.

[1] Vickers, J., AIAA Complex Aerospace Systems Exchange (CASE) Digital Twin Certification for Additive Manufacturing Discussion Paper

[2] NIST, <https://www.nist.gov/el/lpbf-thermography>

[3] E. Glaessgen and D. Stargel, The Digital Twin Paradigm for Future NASA and U.S. Air Force Vehicles, in 53rd AIAA/ASME/ASCE/AHS/ASC Structures, Structural Dynamics and Materials Conference (2012)



Institute Overview

What's an STRI?

Long-term sustained investment in research and technology critical to NASA's future. Highlights include:

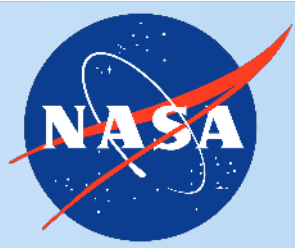
- **Empowered** university-led team
- Specific research objectives with **credible expected outcomes** in 5 years
- Talented, **diverse**, cross-disciplinary and fully **integrated** team
- Low to mid TRL

Award Details

- Expected duration: **5 years**
- Award amount up to **\$3M per year**
- Institutes expected (and *empowered*) to implement their own review internal processes
- NASA oversight – annual reviews and brief quarterly status reports

Research Areas

1. A furthered understanding of **materials-processes-structure-property relationships** for AM through advanced computational toolsets coupled with innovative experiments, laying the foundation for accelerated product certification.
2. Development of **uncertainty-quantified, physics-based models and simulations** to understand the factors that can affect the formation, distribution, and the effects of process-induced defects and to address other principal sources of variability.
3. Integration and application of methodology, software tools, artificial intelligence (AI), machine learning (ML), and/or databases, etc., based on computational and experimental tools to a **model-centric certification approach**.



Institute Participants

Industry

- Boeing
- GE
- Lockheed Martin
- Pratt & Whitney
- Raytheon
- RTX (Parent Company: Collins, PW, and Raytheon)
- Hexagon
- LIFT
- MMI
- Relativity Space
- Science and Technology Corporation



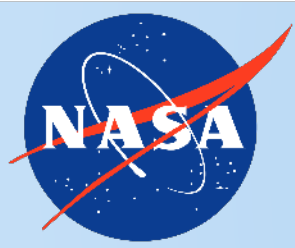
Group photograph from the IMQCAM Continuation Review Oct 3-4 at CMU. Photo credit: CMU

Other Government Agencies

- ARL/DEVCOM
- Air Force Research Lab
- NIST
- Sandia National Lab
- FAA
- NAVAIR

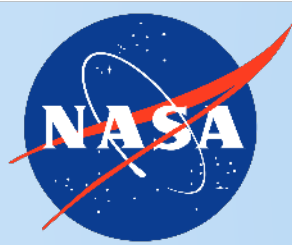
Partner Academic Institutions

- Carnegie Mellon University
- Johns Hopkins University (JHU)
- JHU-Applied Physics Laboratory
- Vanderbilt University
- University of Texas-San Antonio
- Southwest Research Institute
- University of Virginia
- Pratt & Whitney (PW)
- Case Western Reserve University



IMQCAM Goals

- Develop a digital twin of metals additive manufacturing that comprises an integrated set of verified, experimentally-validated, uncertainty-quantified computational models and simulation & design tools that will mirror the entire materials-processes-structure-property performance and life linkage in metal AM
- Demonstrated for at least two materials, LPBF Ti-6Al-4V and In718
- Evaluated by multiple companies against their own data, i.e., third-party validation
- Adoption of IMQCAM models by partner OEMs



Highlighted Progress and Focus Areas

Defect Models

- Interpass temperature, hatch spacing, and dwell times influence on defects
- Spatter-induced porosity
- Process window lack-of-fusion defect prediction
- Melt track variability
- Synchrotron experiments

Microstructure Models

Fatigue Crack Growth Models

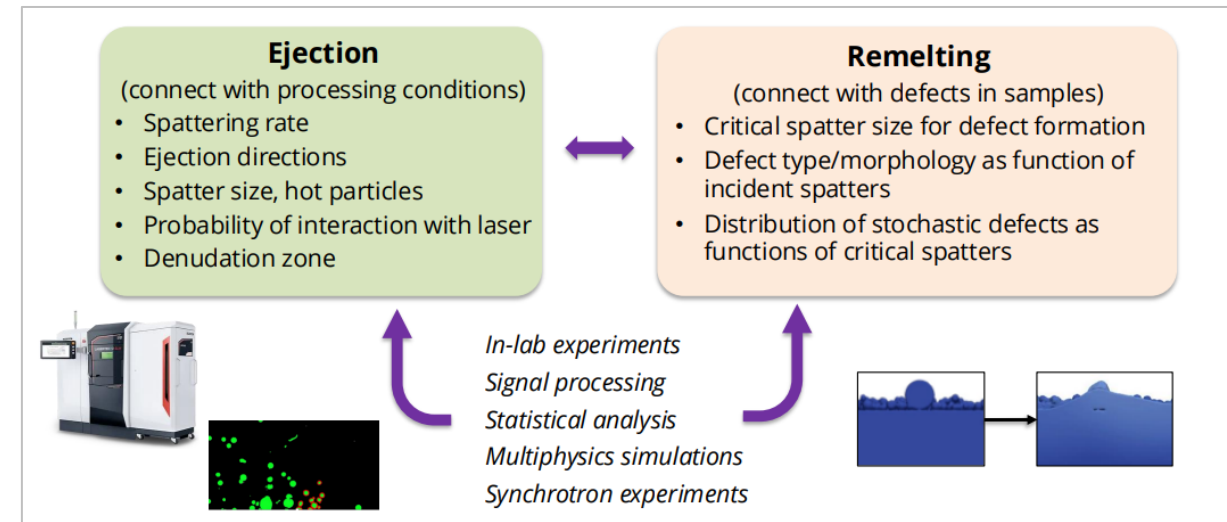
Uncertainty Quantification

Data Integration

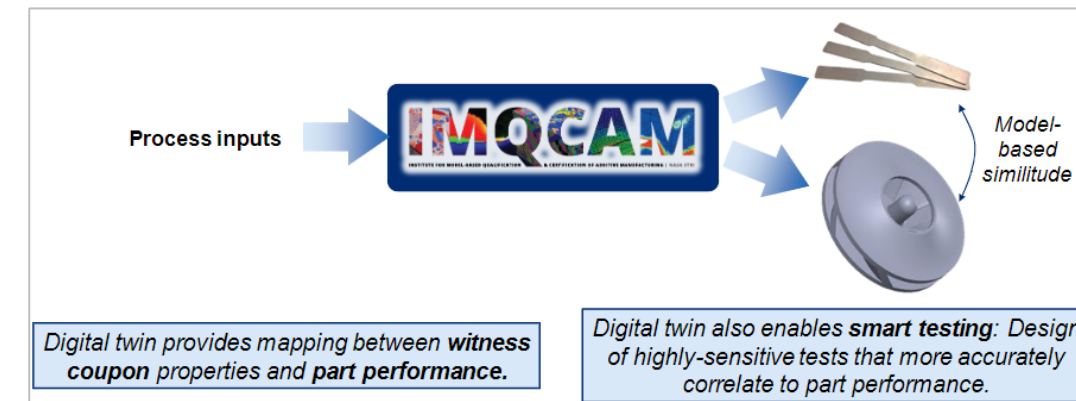
- Build, Characterization, & Powder Feedstock Data
- Automated, Linked Workflows
- Event Driven Integration

Qualification and Certification Framework

- Defined Application Focuses (e.g., location-specific properties)

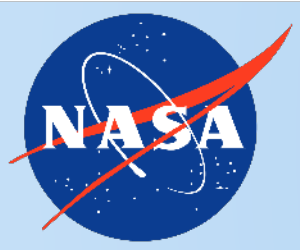


Modeling Spatter-induced Porosity Example
Credit: Sneha Narra, CMU and Tao Sun, Northwestern



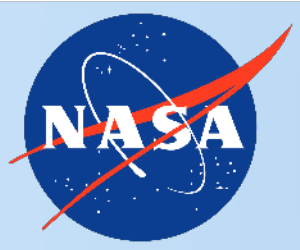
Location-Specific Properties Application Focus Area Example

Credit: Brendan Croom, APL

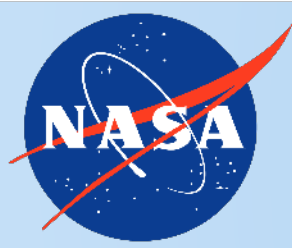


Questions?

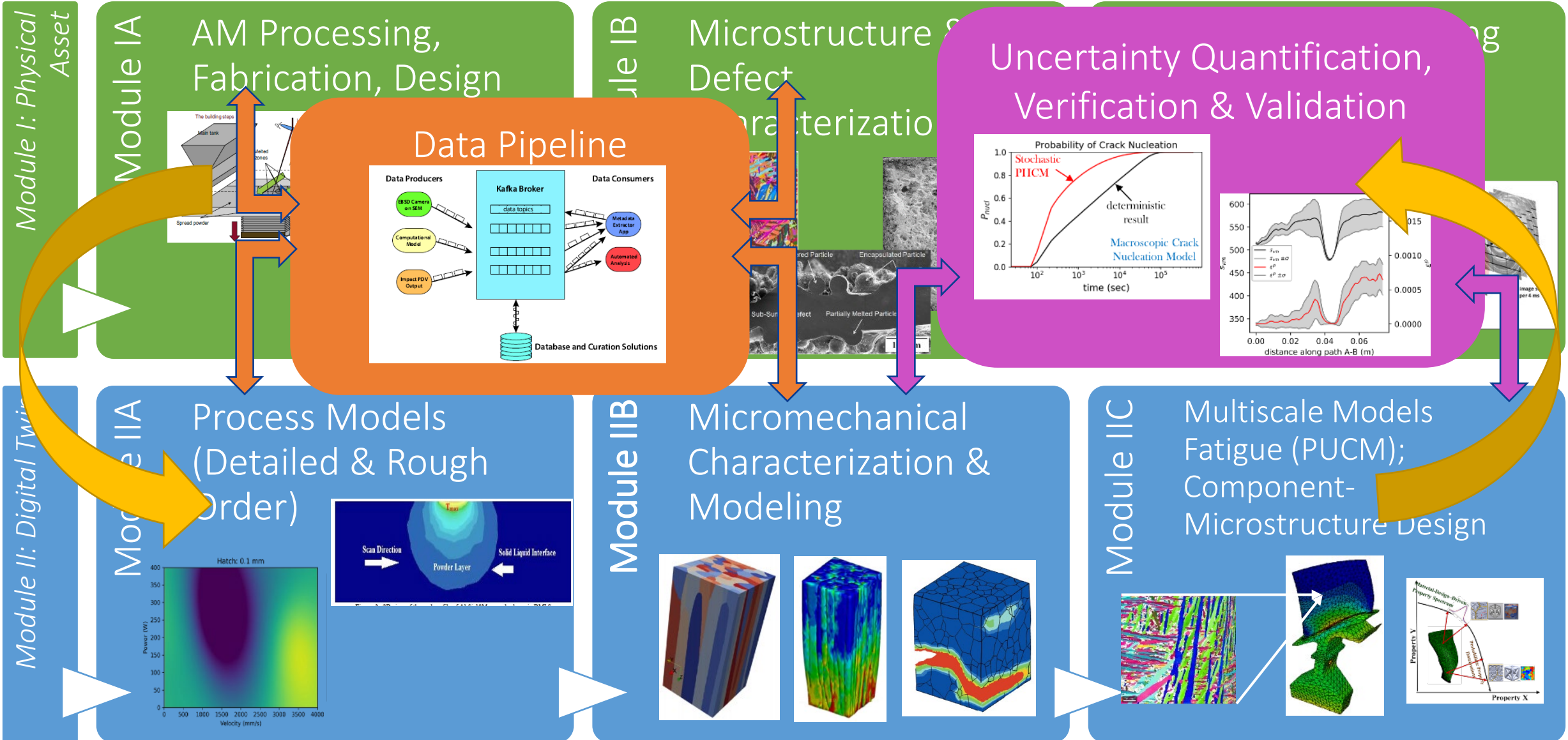


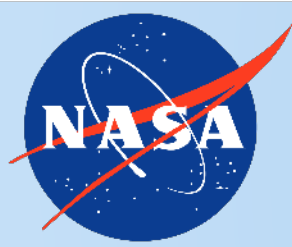


Backup



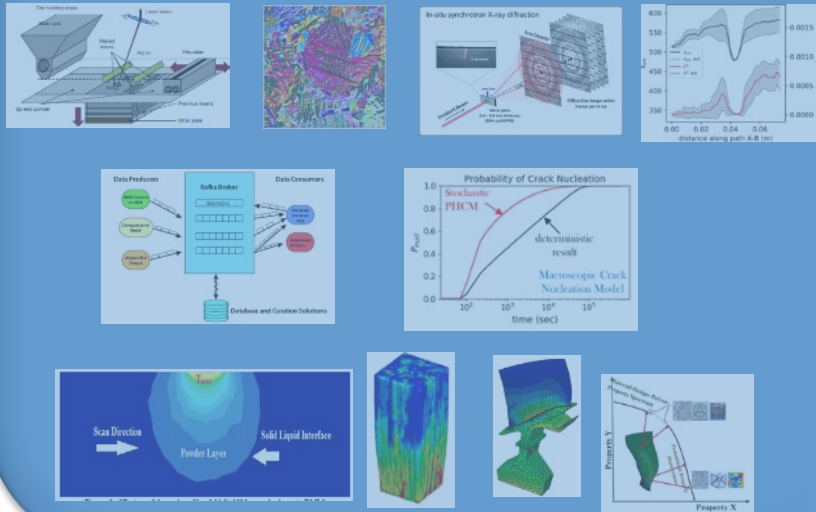
Task I: Processing, Testing, Modeling, Simulation and Design





Task II: Qualification, Certification, and Translation

Task I: Processing, Testing, Modeling, Simulation and Design



Module I

Data Coordination
& Dissemination

Software
Coordination &
Dissemination

Module II

Qualification/Certification

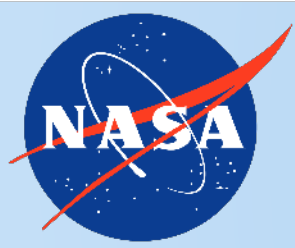
Using Experimental &
Simulation Data

Module III

Translation

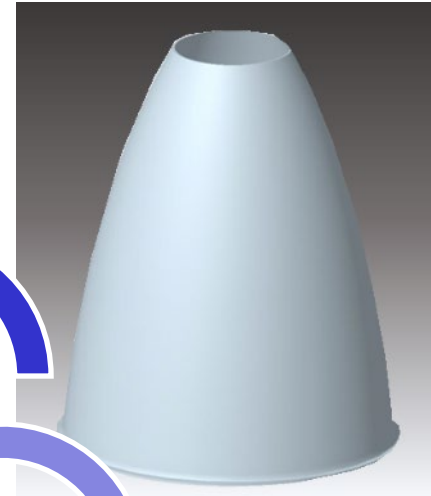
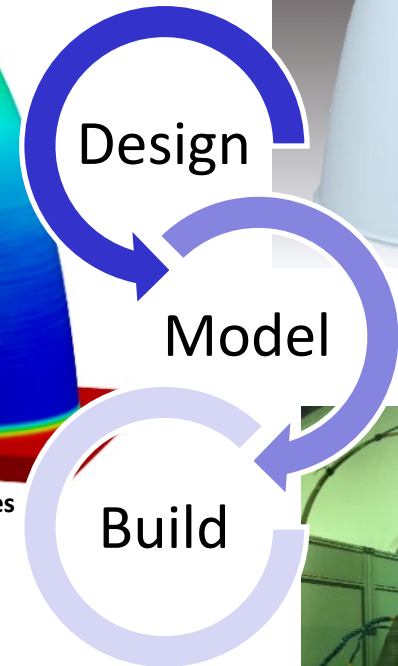
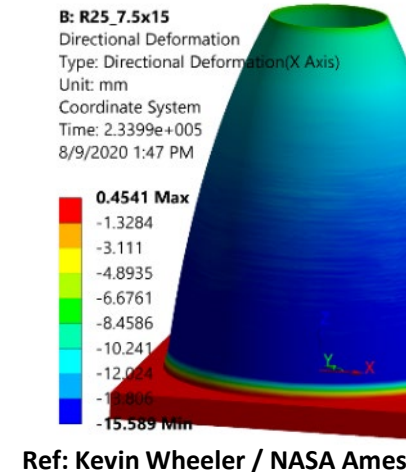
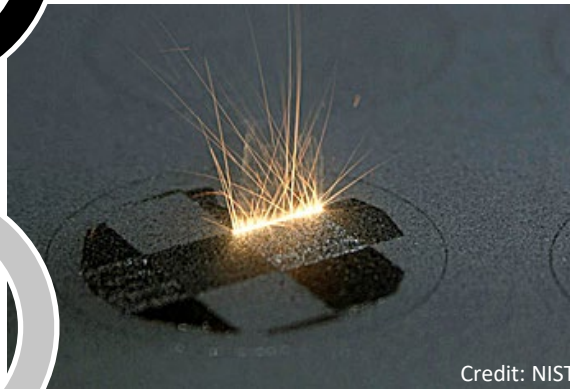
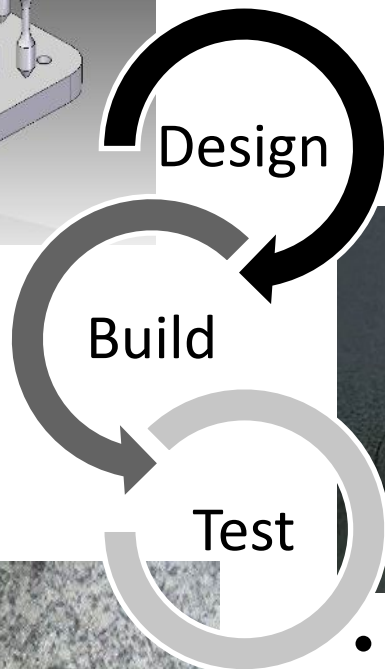
NASA Centers
Software Coordination

Industry/OEMs

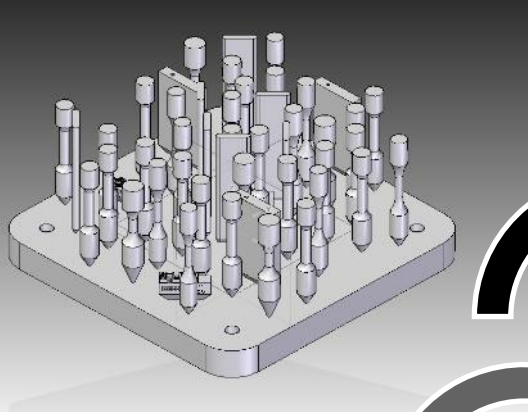


Why Modeling Matters

- Advancement of computational tools for certification enables a transition from Design-Build-Test paradigm to Design-Model-Build



- Coupon-level material characterization work in a digital space enables component design, analysis, and certification faster and more affordably with fewer build iterations



Traditional Experimental AM Certification

Material Characterization
Witness Testing Approach
Establish Process Control Factors
Post-processing Requirements
NDE Techniques

New Computational AM Certification

Modeled Sources of Variability
Effects of Defects and Surface Roughness
Design Materials for their Intended Usage
Dependence of Performance on Processing
Build Failure Prediction
Stress and Heat Simulation
AI/ML Tools for Process Control
Computational Materials Enables
Extensibility

**Computational
Materials**

Cost savings
from model-
based tools