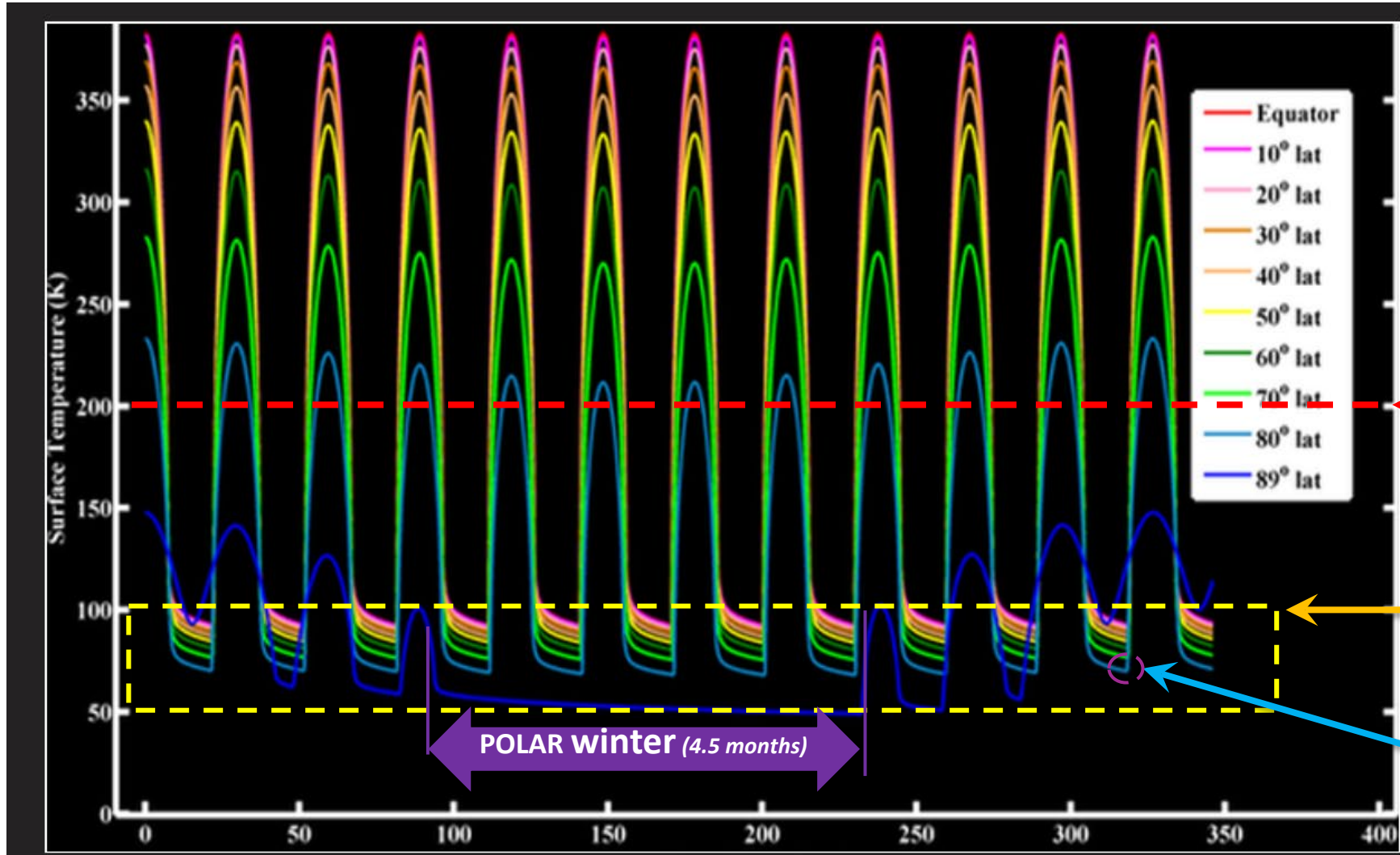


Cryogenic Hibernation Assessment of Commercial Cells for Space Flight Applications

PI & Presenter: Dr. Brian Holler, NASA GRC - LEX
IAPG CWG Spring 2026

The Extreme Lunar Environment

Thermal model calculations of monthly & annual lunar surface temperature variations at various latitudes



- **Day:** 150-400 K highs based on latitude
- **Night:** 50-100 K lows *all latitudes*
Duration (non-polar): ~354 hrs (~14.75 Earth days)
- **Lunar Polar Winter:** Sun is below horizon for ~4.5 months

Li-ion battery approx. freeze temperature

Lunar night is extremely cold *at all latitudes*

Lunar dawn

CLPS

COMMERCIAL LUNAR PAYLOAD SERVICES

National Aeronautics and
Space Administration



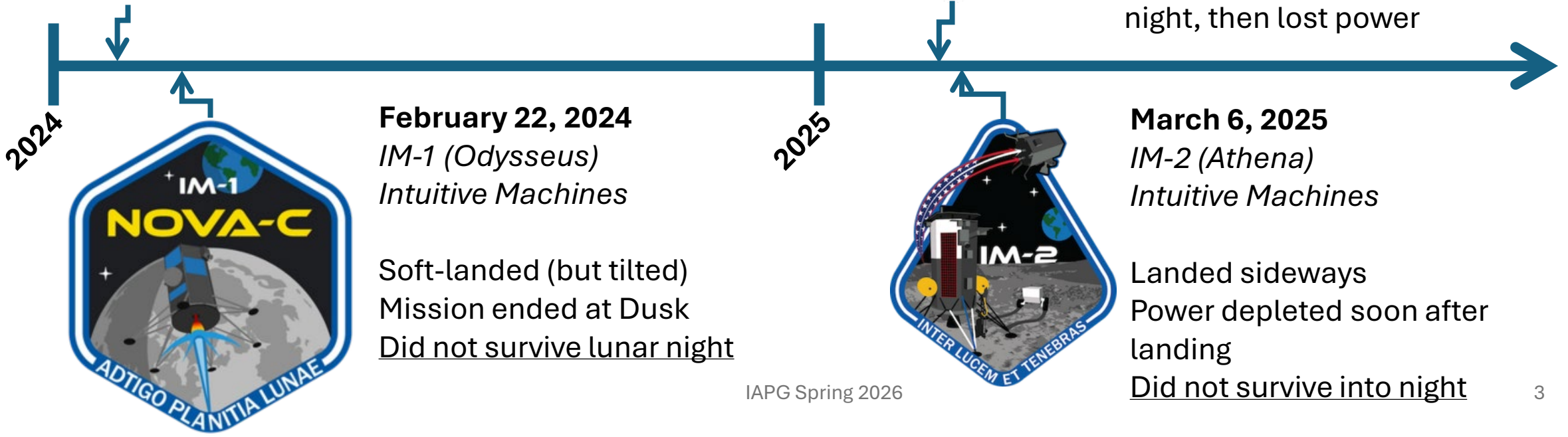
January 2024
Peregrine Mission One
Astrobotic

Mission Failure before landing
Did not reach lunar surface



March 2, 2025
Blue Ghost 1
Firefly Aerospace

Landed successfully, operated
for the lunar day
Operated ~5 hours into lunar
night, then lost power



February 22, 2024
IM-1 (Odysseus)
Intuitive Machines

Soft-landed (but tilted)
Mission ended at Dusk
Did not survive lunar night



March 6, 2025
IM-2 (Athena)
Intuitive Machines

Landed sideways
Power depleted soon after
landing
Did not survive into night



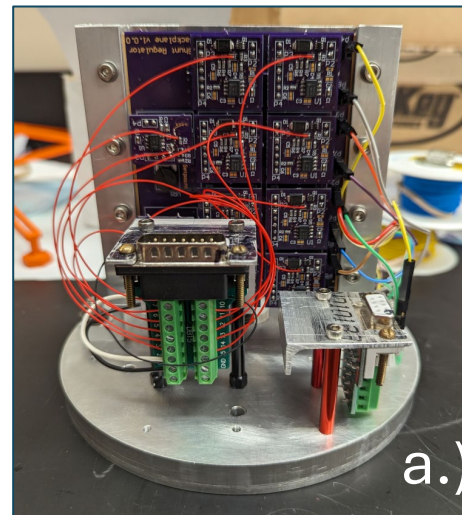
Project Overview

Description and Motivation:

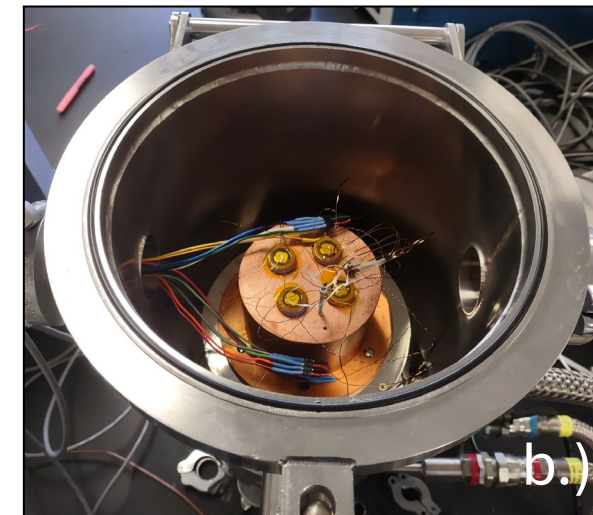
- Batteries and battery management systems are not designed to survive the extreme cold of the lunar night, which restricts available missions
- Develop and demonstrate a **novel, easily integrable BMS** to enable Li-ion battery hibernation passively through the lunar night with zero excess degradation
- Enables **longer mission time** (>1 lunar day) with **minimal impact on system size** or payload capacity
- Allows for a **significant launch cost savings** due to the reduced required battery mass when compared to current solutions

Primary Project Objectives:

- **Design** and **build** an integrated battery package/electronics system demonstrating lunar dawn mode to **enable Li-ion cell hibernation** through the lunar night (*Phase 1*)
- Experimentally determine **battery cell survival conditions** with a zero-degradation criterion to inform PHM inclusion in microcontroller embedded electronics (*Phase 2*)
- Perform **appropriate environmental tests** of each design to raise technology readiness level (TRL) from 3 to 5



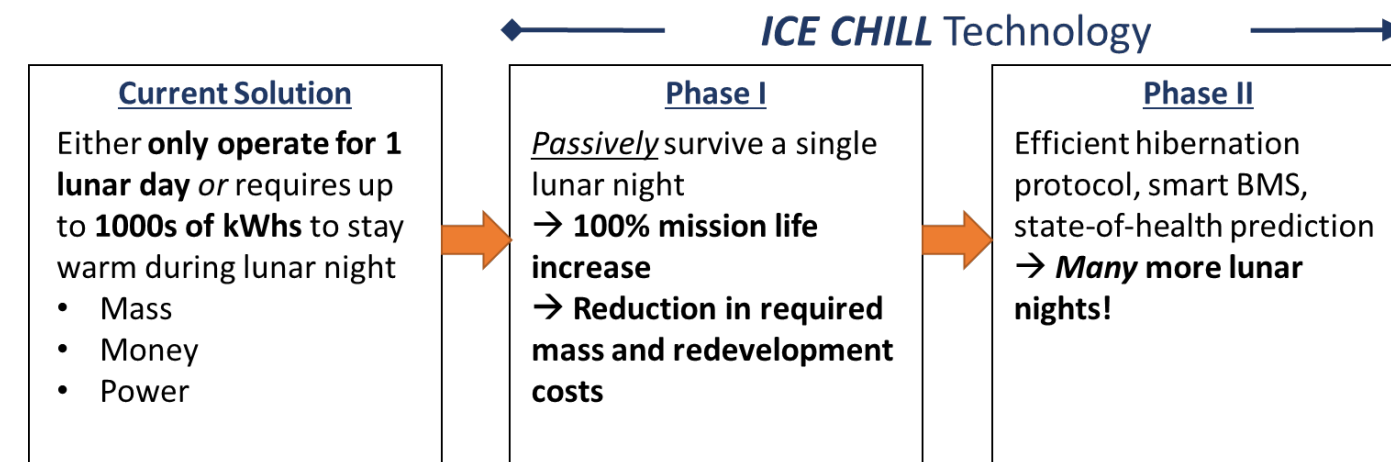
a.)



b.)

a.) Cryo-operable Solar Shunt Regulator to enable Lunar Dawn Recover Mode.

b.) Commercial 18650 Li-ion cells in cryochamber for low-temperature hibernation down to 50K.



Concept of Operations

Key

Cryo-tolerant

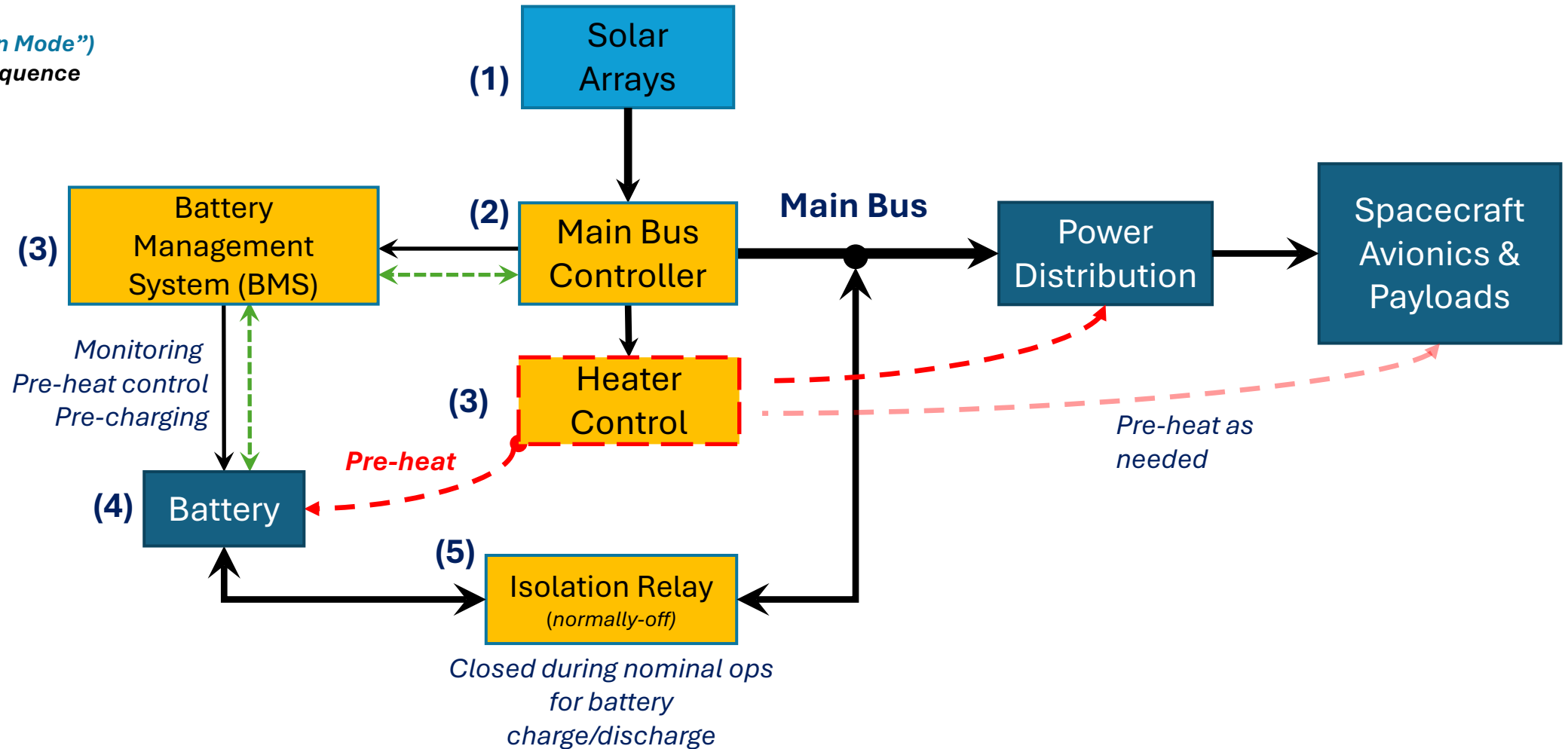
Cryo-operable ("Dawn Mode")

(#) – Dawn start-up sequence

→ Power

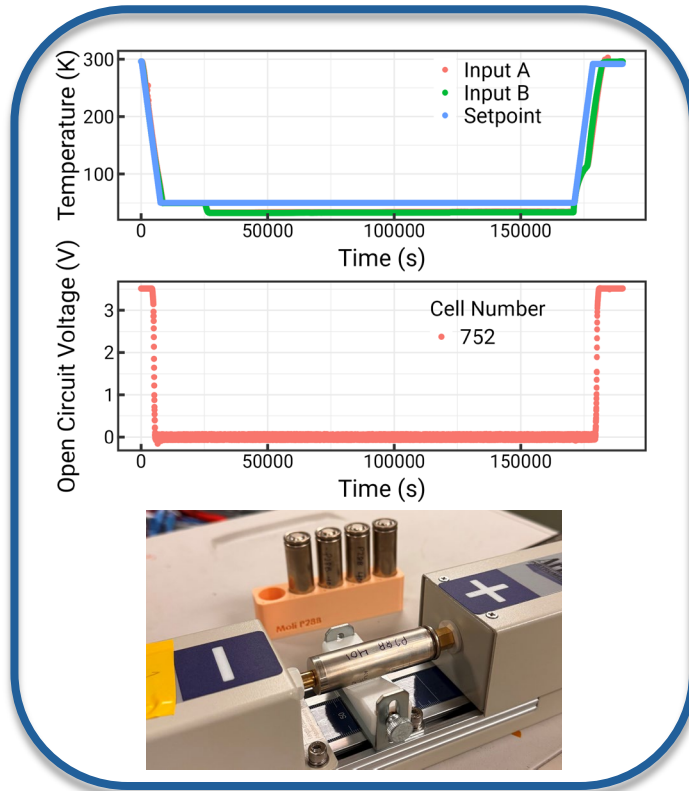
→ Thermal

→ C&DH

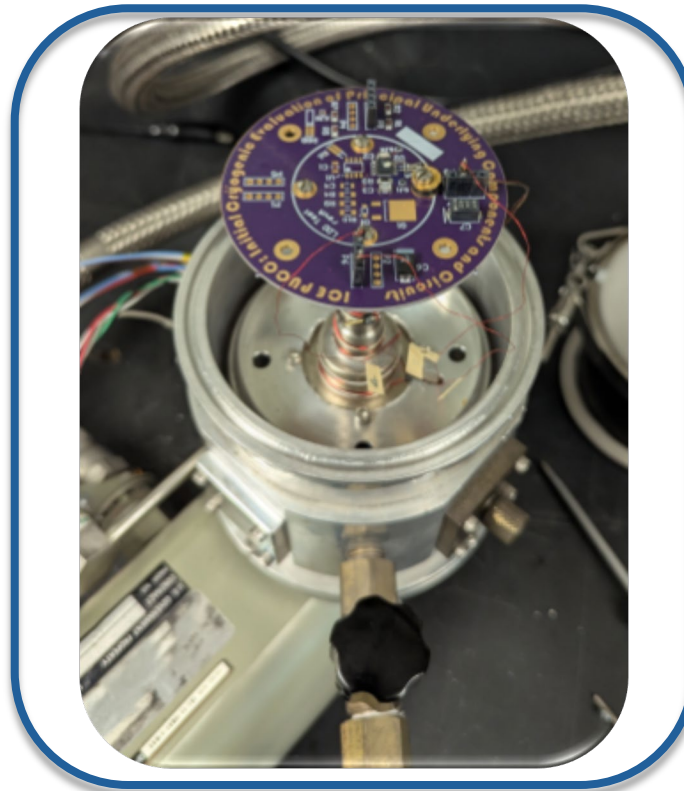


Technical Status Overview

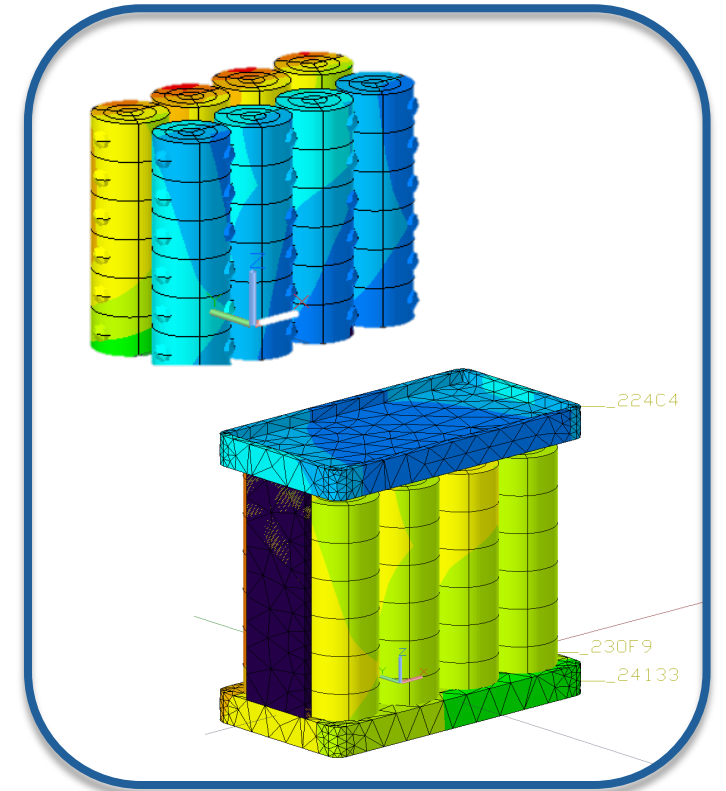
Battery Characterization



Electronics Development

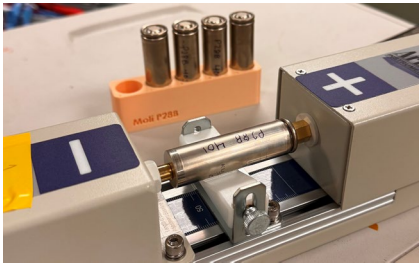
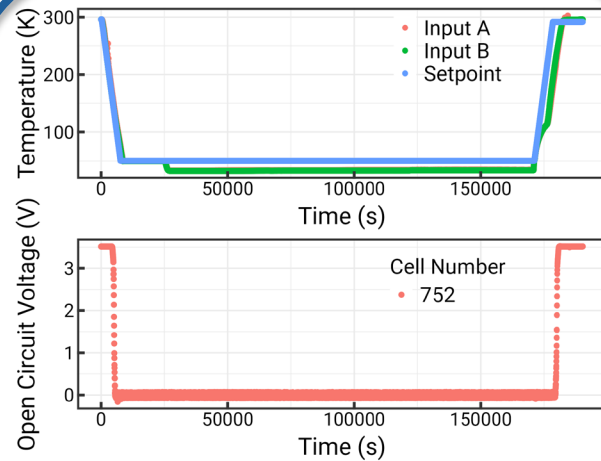


Thermal and Case Design

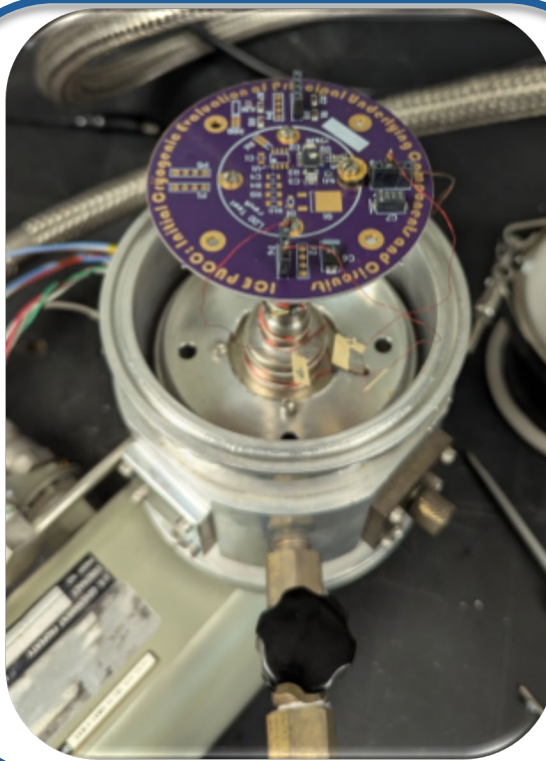


Technical Status Overview

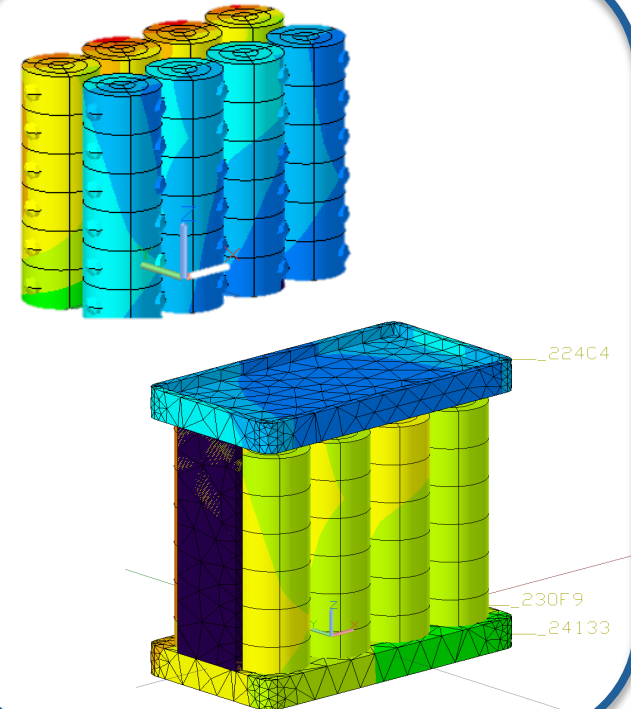
Battery Characterization



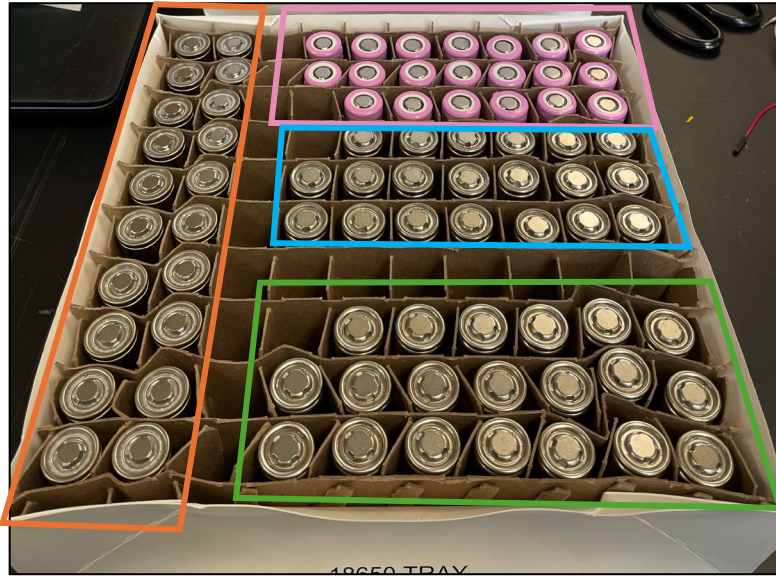
Electronics Development



Thermal and Case Design

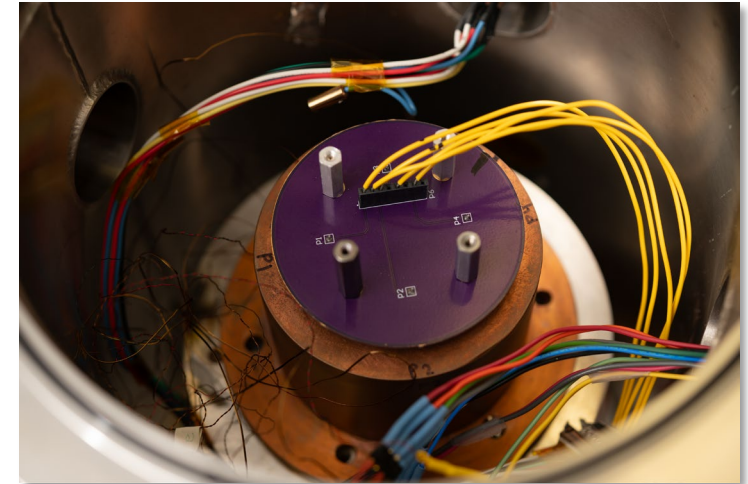
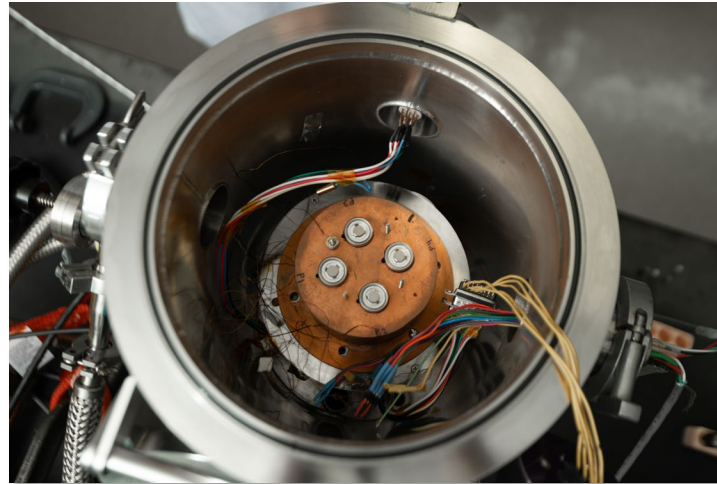
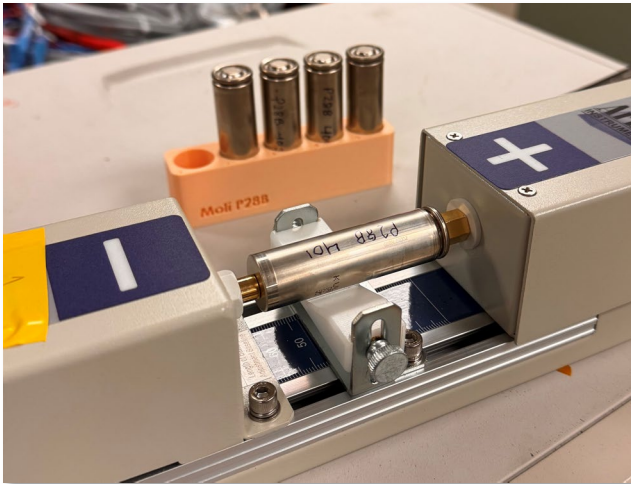


Battery Cryogenic Testing

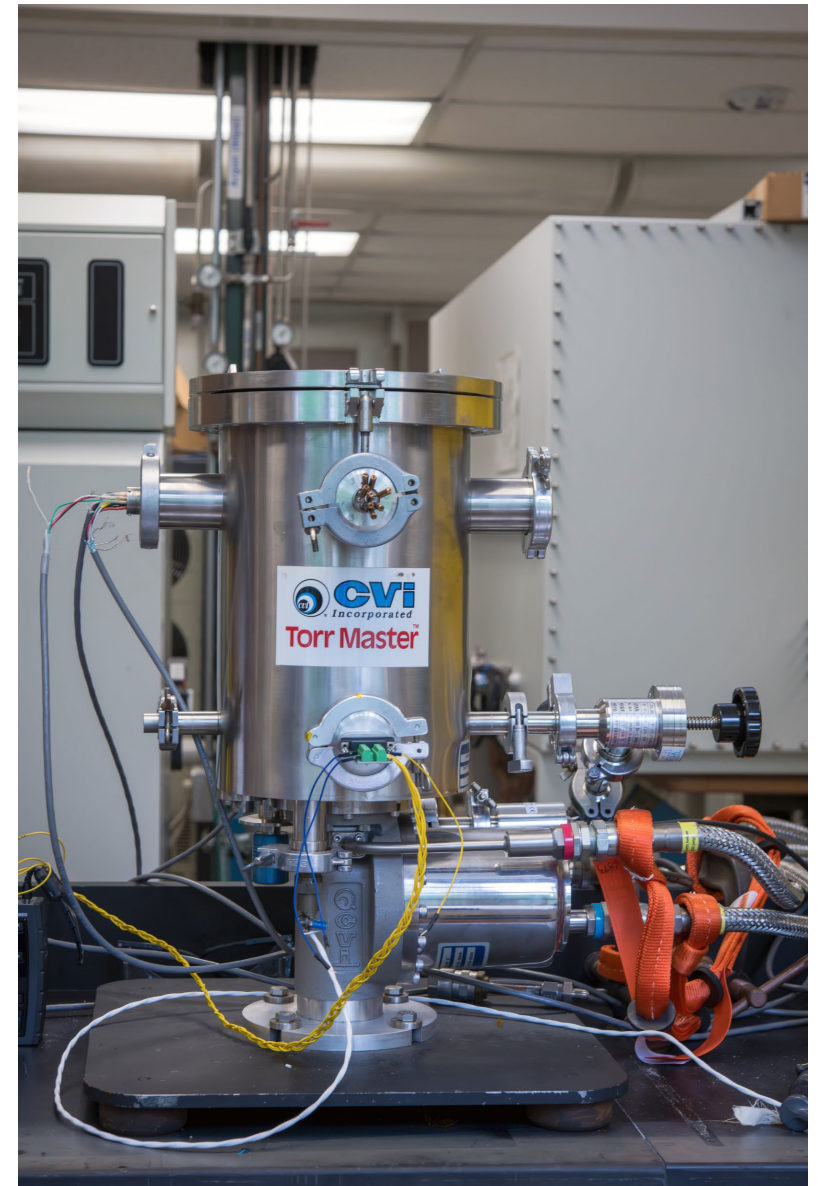
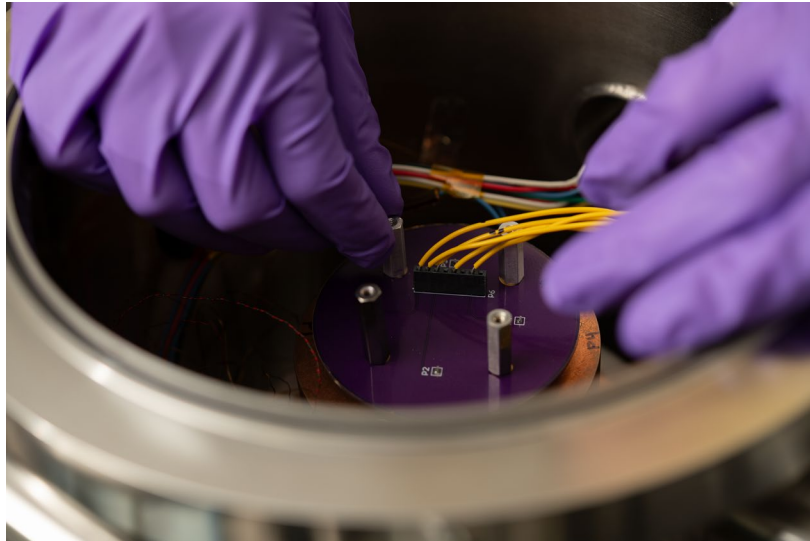
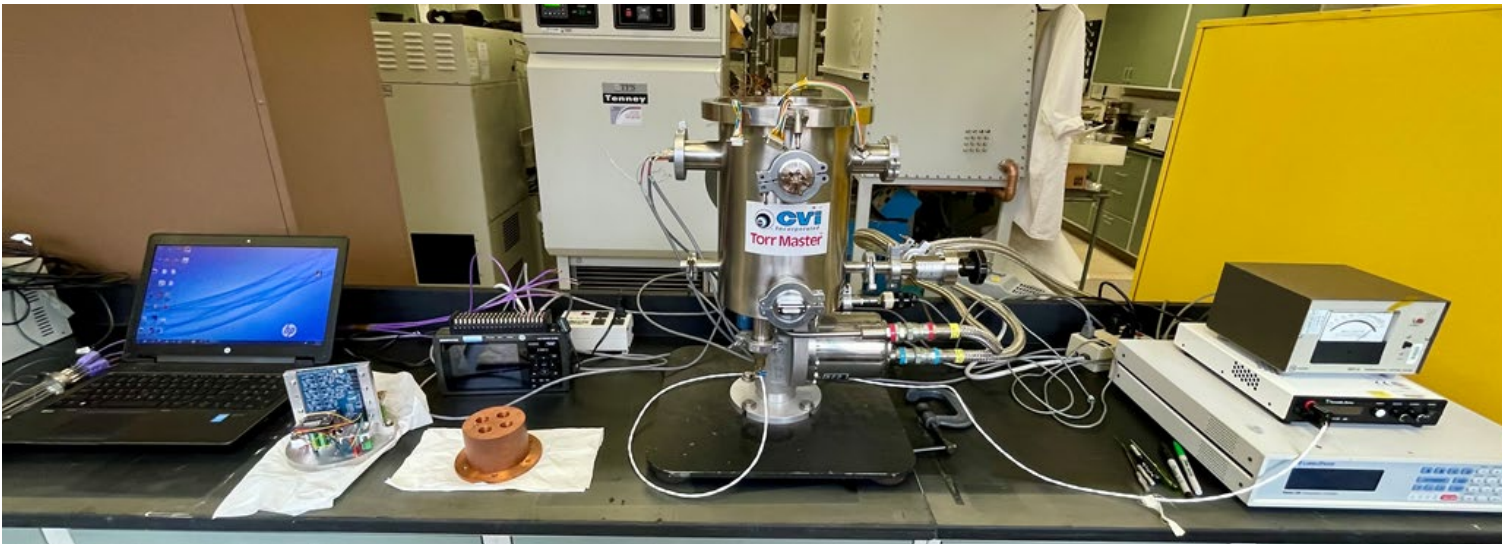


Manufacturer	Model	Capacity (Ah)
Samsung	Q30	3.0
LG	M36	3.6
Molicel	P28B	2.8
Molicel	M35A	3.6

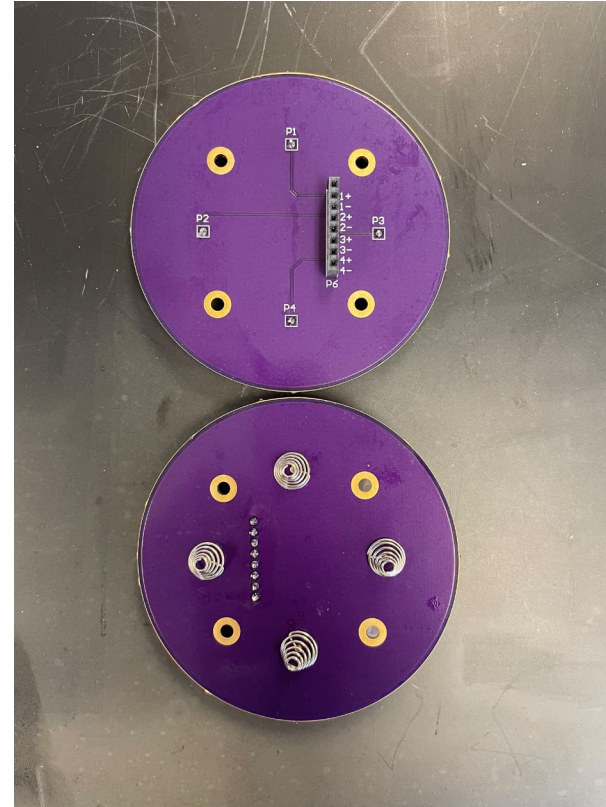
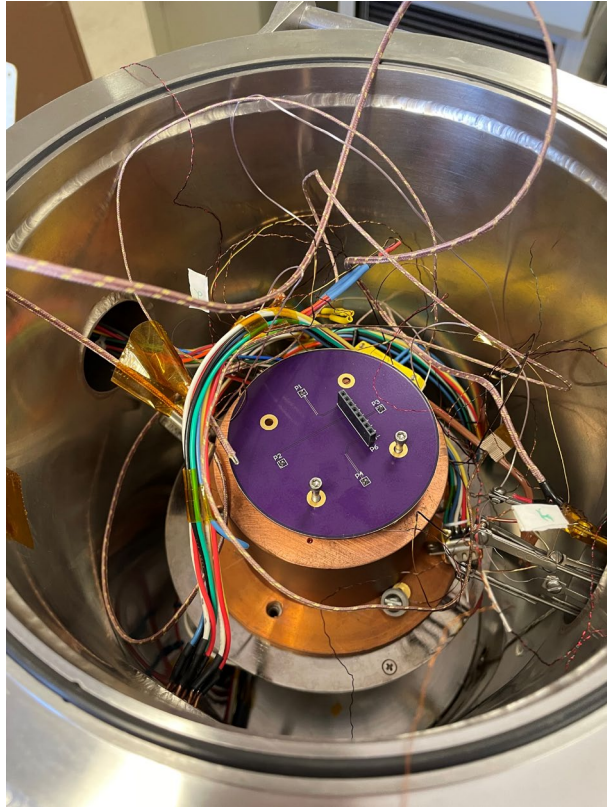
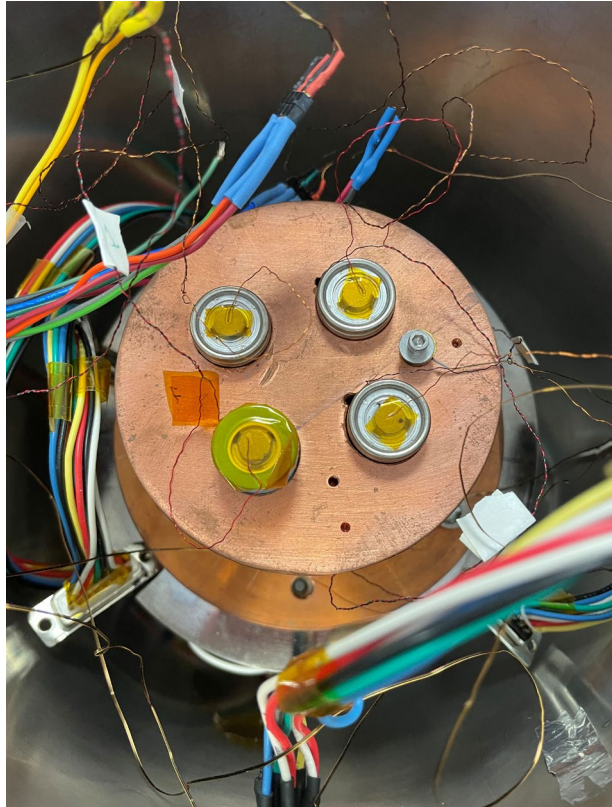
1. Preliminary Electrochemical Cycling, establishment of test charge level (0%, 100%)
2. Preliminary Electrochemical Impedance Spectroscopy (EIS)
3. Cryo-Hibernation Event
4. Post Electrochemical Cycling
5. Post EIS



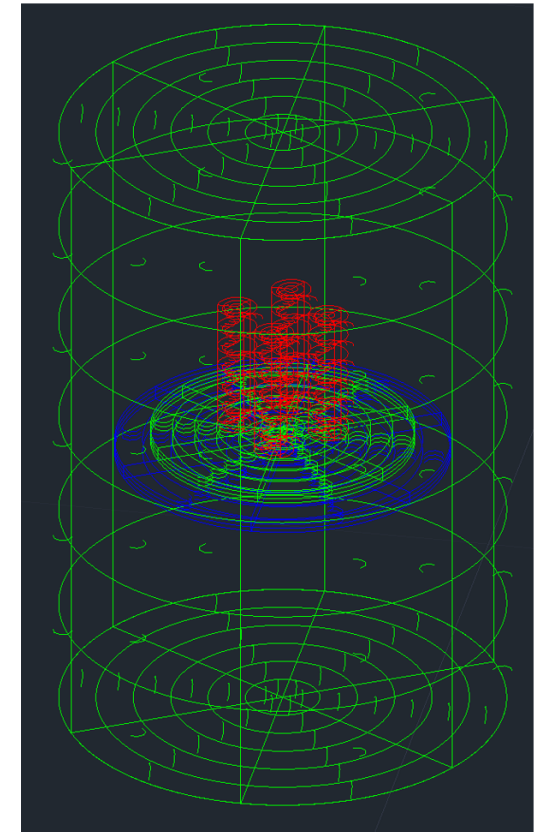
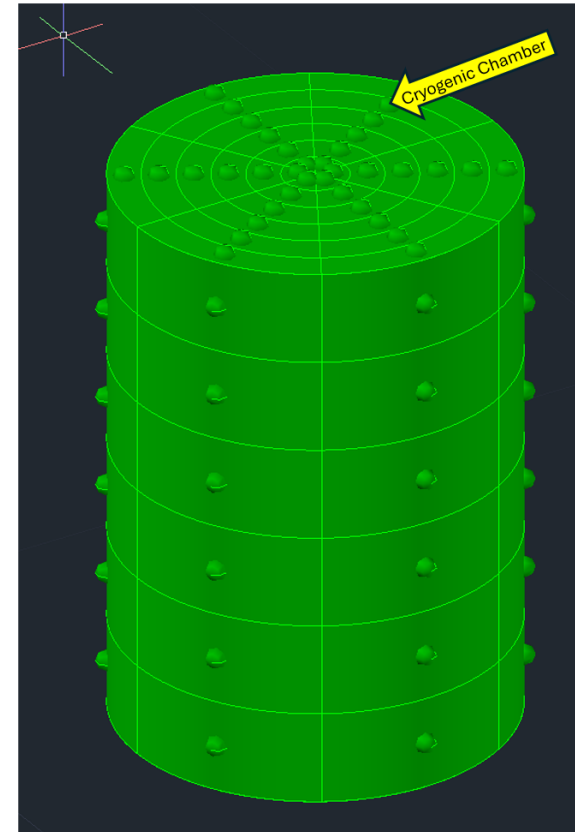
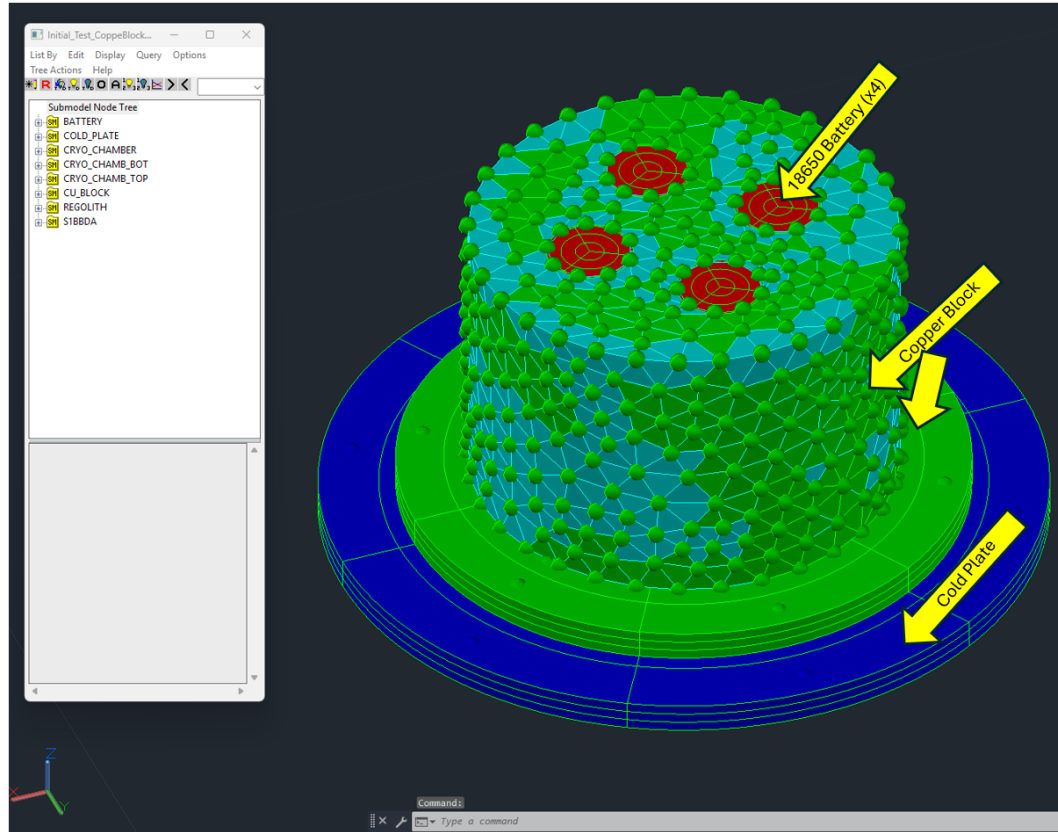
Cryochamber Test Setup



Cryochamber Test Setup



Gen 1 Thermal Model and Thermal Environment



Focused on Cryogenic Chamber environment and its lunar equivalent

Starts at 293.15K, cooldown rate 2K/min till it hits 50K, dwells at 50 K for 8 hours, then warms up in temperature either naturally or at 2 K/min.

Parametric Studies

Had concerns of degradation via temp gradients across the cell:

[In Situ Observation of Thermally Activated and Localized Li Leaching from Lithiated Graphite | ACS Energy Letters](#) -

Harrison Szeto, Vijay Kumar, and Yangying Zhu, *ACS Energy Letters* **2024** 9 (9), 4218-4224, DOI:

10.1021/acseenergylett.4c01571

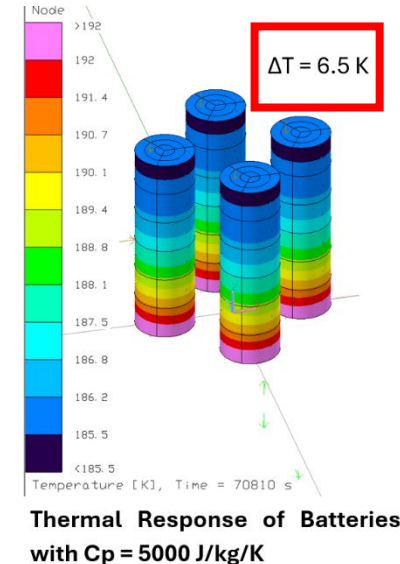
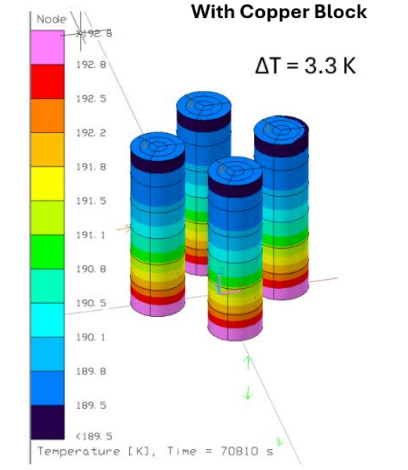
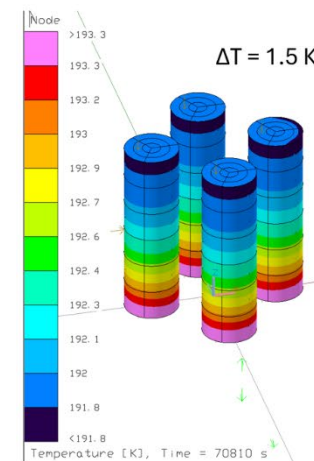
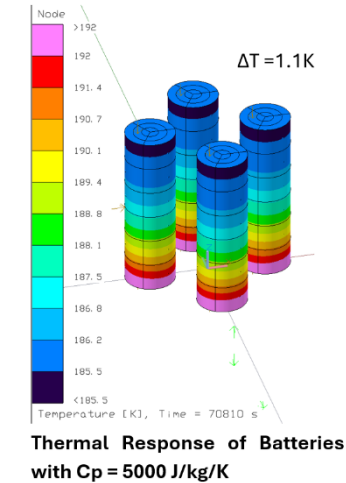
Investigated battery thermal properties and Copper Block contact

Heat Capacity (C_p)

- **Parametric sweep:** 800–5000 J/kg/K.
- **With Copper Block:** sensitivity at ~ 2500 J/kg/K ($\Delta T \approx 3$ K); ~ 6.5 K at 5000 J/kg/K.
- **Without Copper Block:** larger ΔT across batteries.
- **Realization:** Copper Block acts as a thermal dampener, masking gradients $\rightarrow \Delta T < 1$ K at high C_p .

Thermal Conductivity (κ)

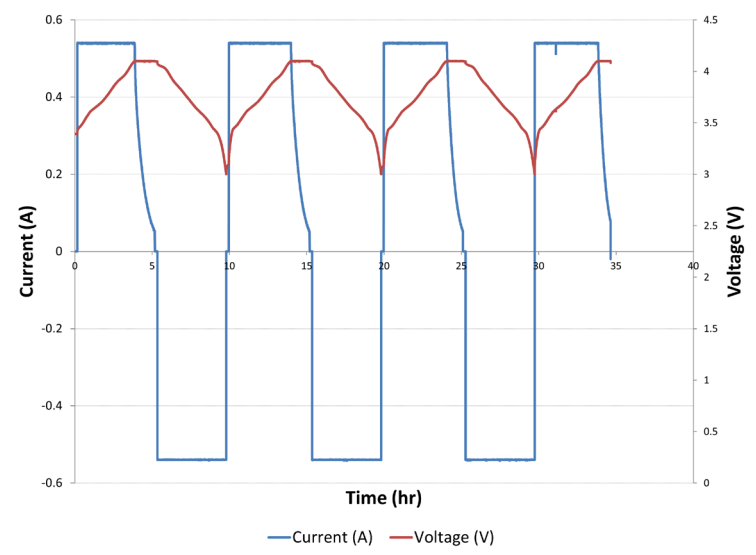
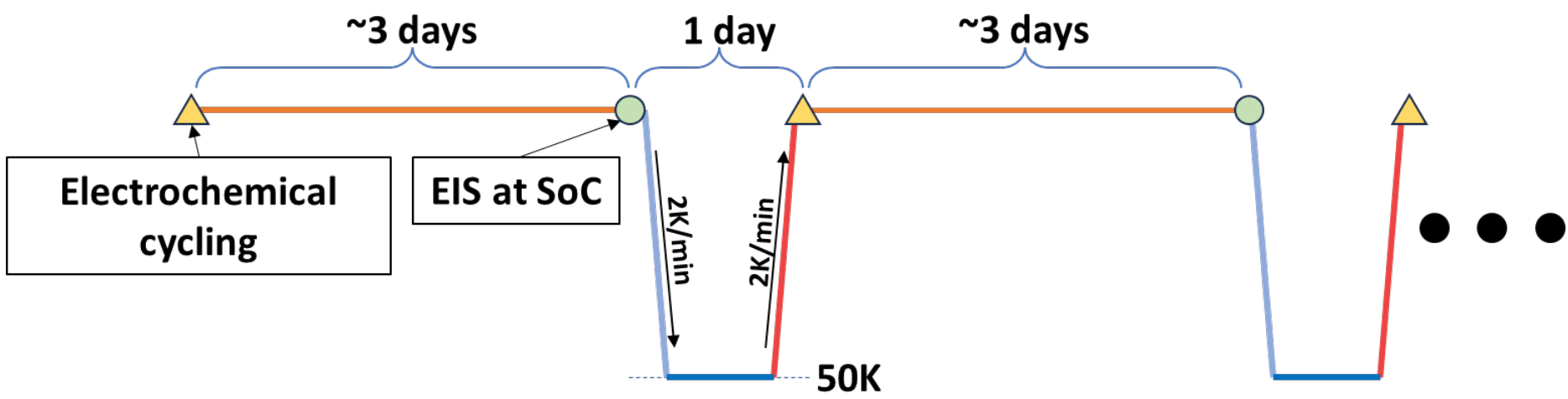
- **Isotropic sweep:** 1–100 W/m/K without Copper Block.
- **Literature:** ~ 13 W/m/K.
- Even extreme values (100 W/m/K) $\rightarrow$ negligible impact



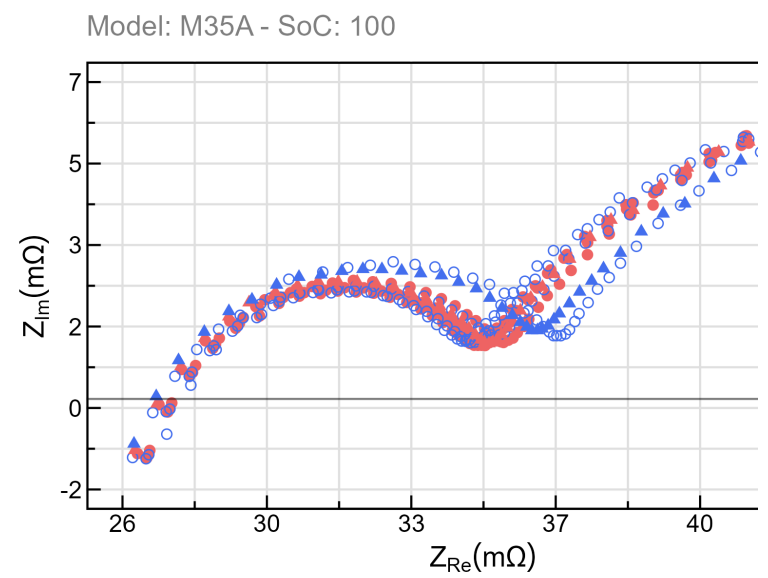
Passive Battery Hibernation Testing

Combinatorial Test Matrix

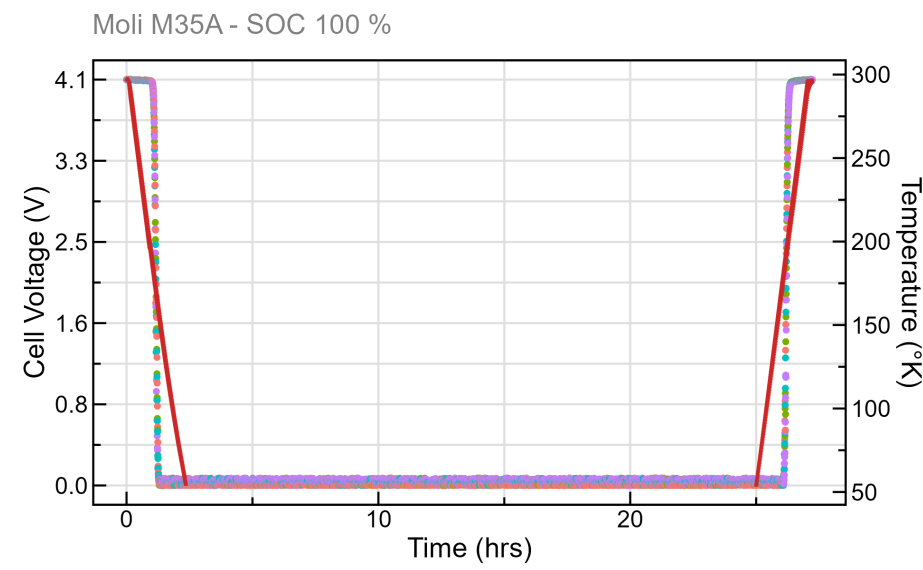
- State of Charge
- Temperature Ramp Rate
- Dwell Time



Electrochemical Cycling



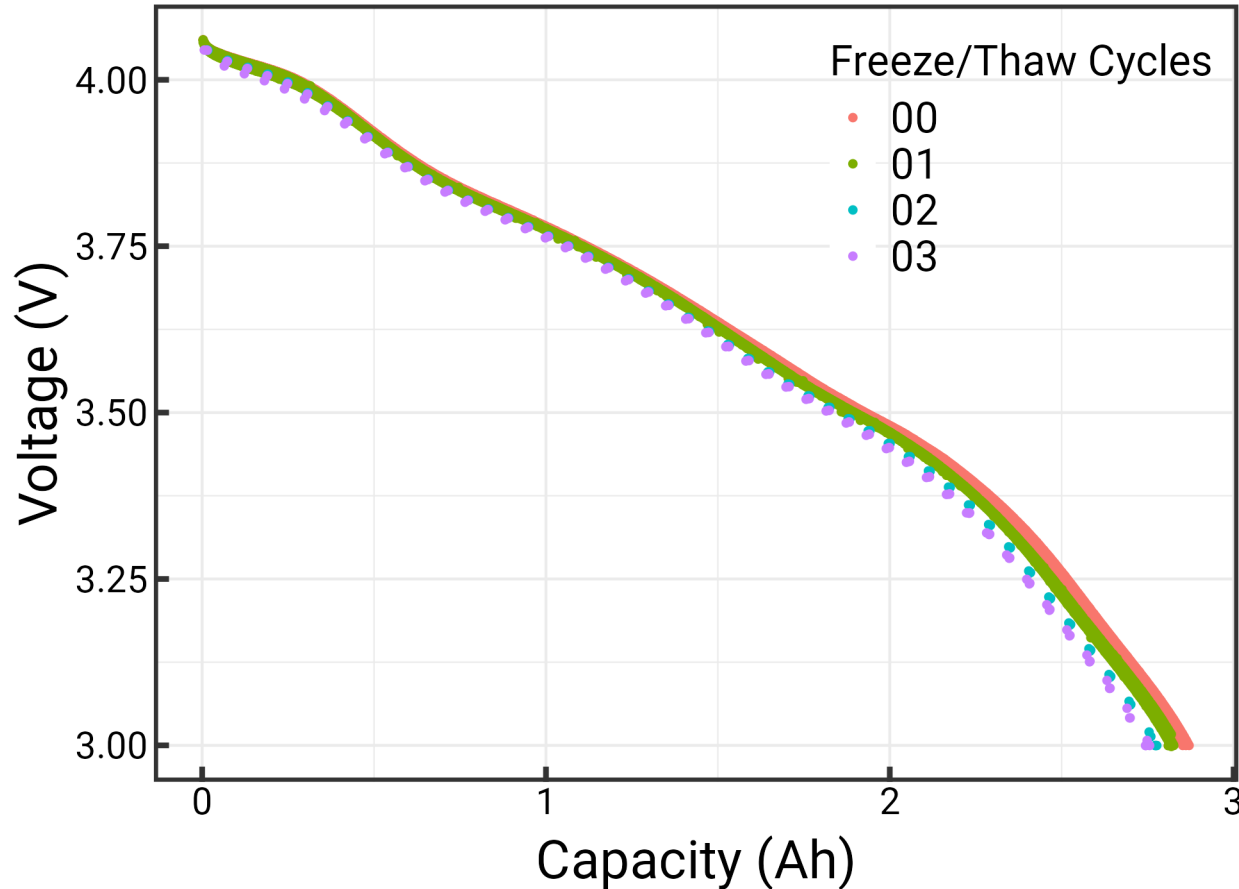
EIS



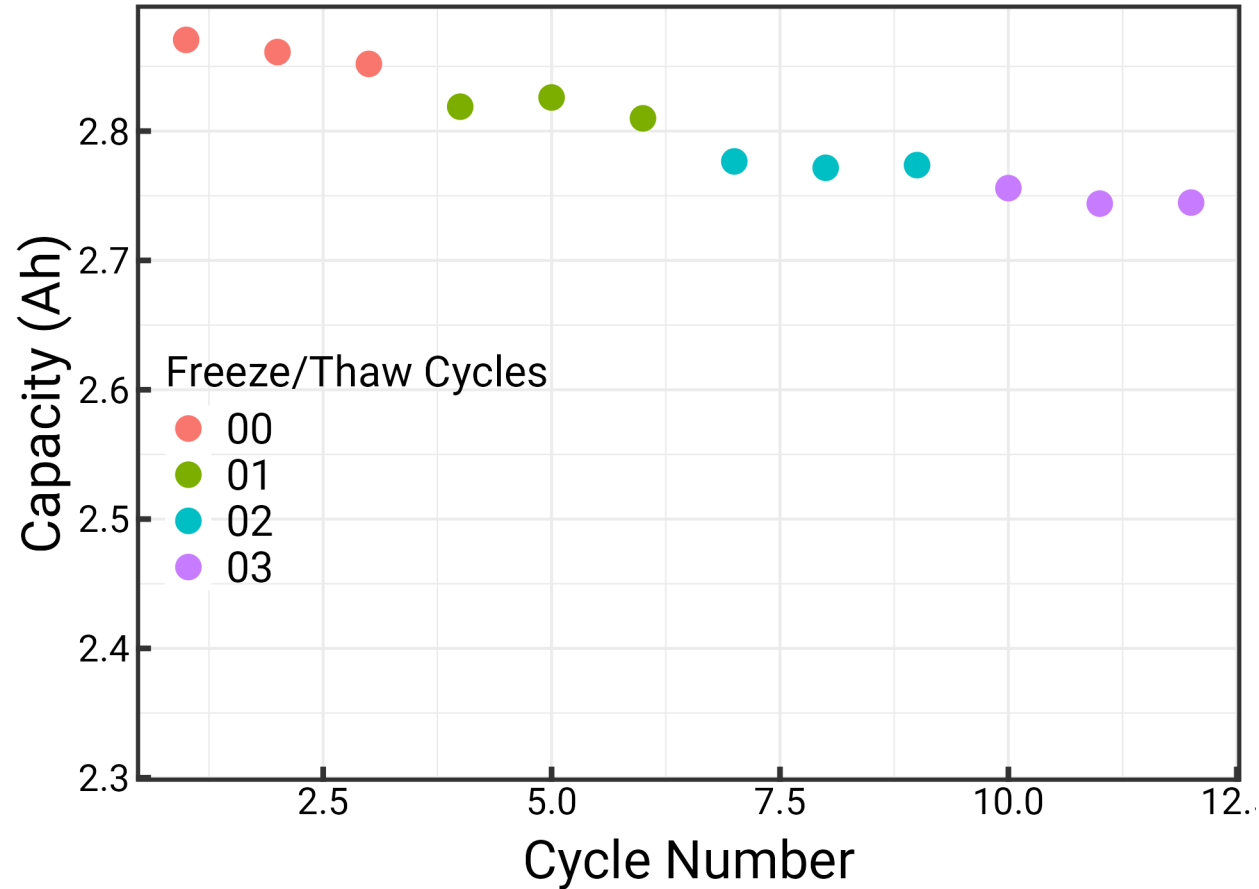
Cryo-Hibernation

Post-Hibernation Comparison of Cell Behaviors: Cycling

C/5 Discharge Cycles for Cell 507



Lifetime Capacity Fade for Cell 507

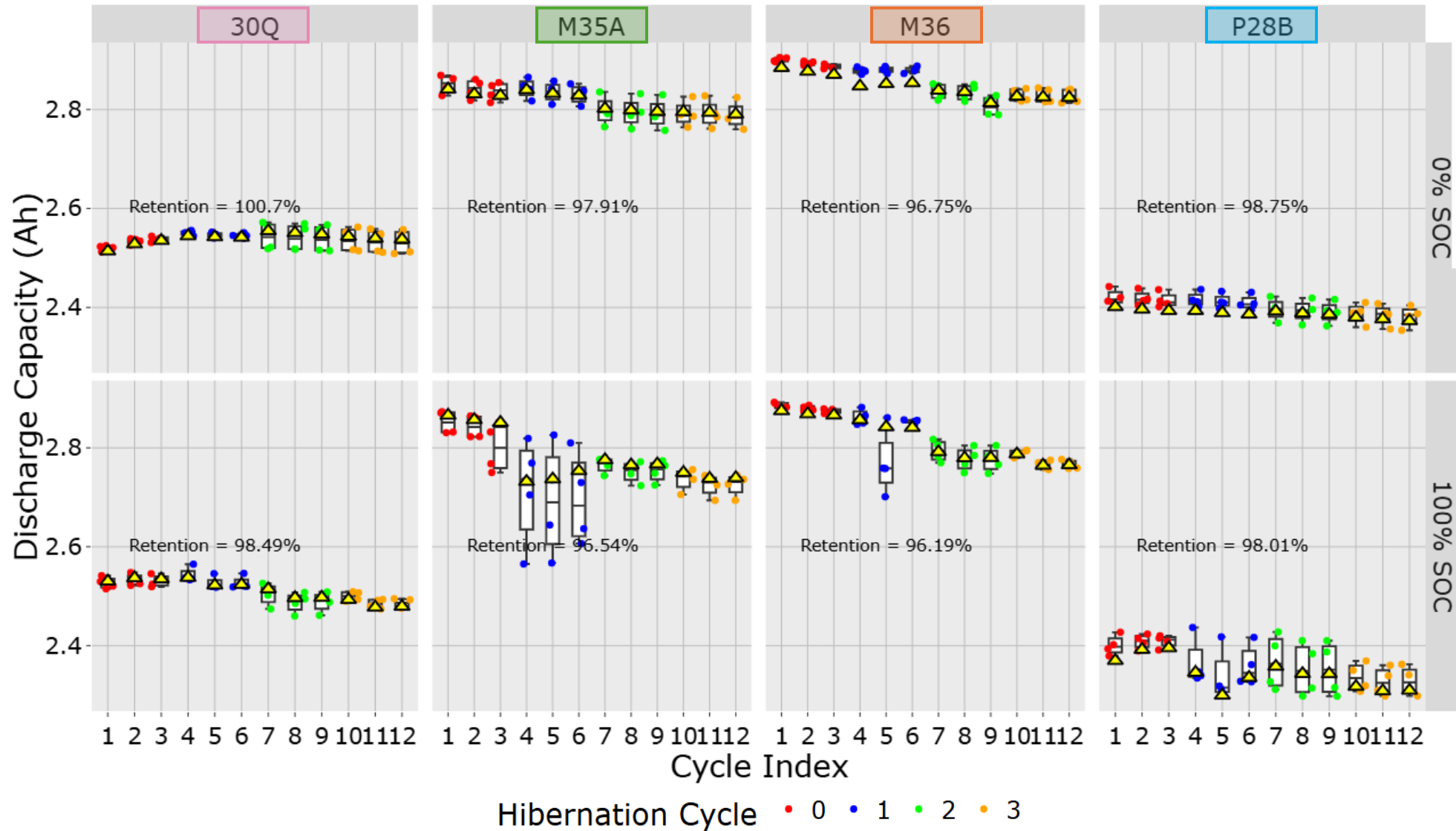


This is representative data: All tested cells showed similar behavior

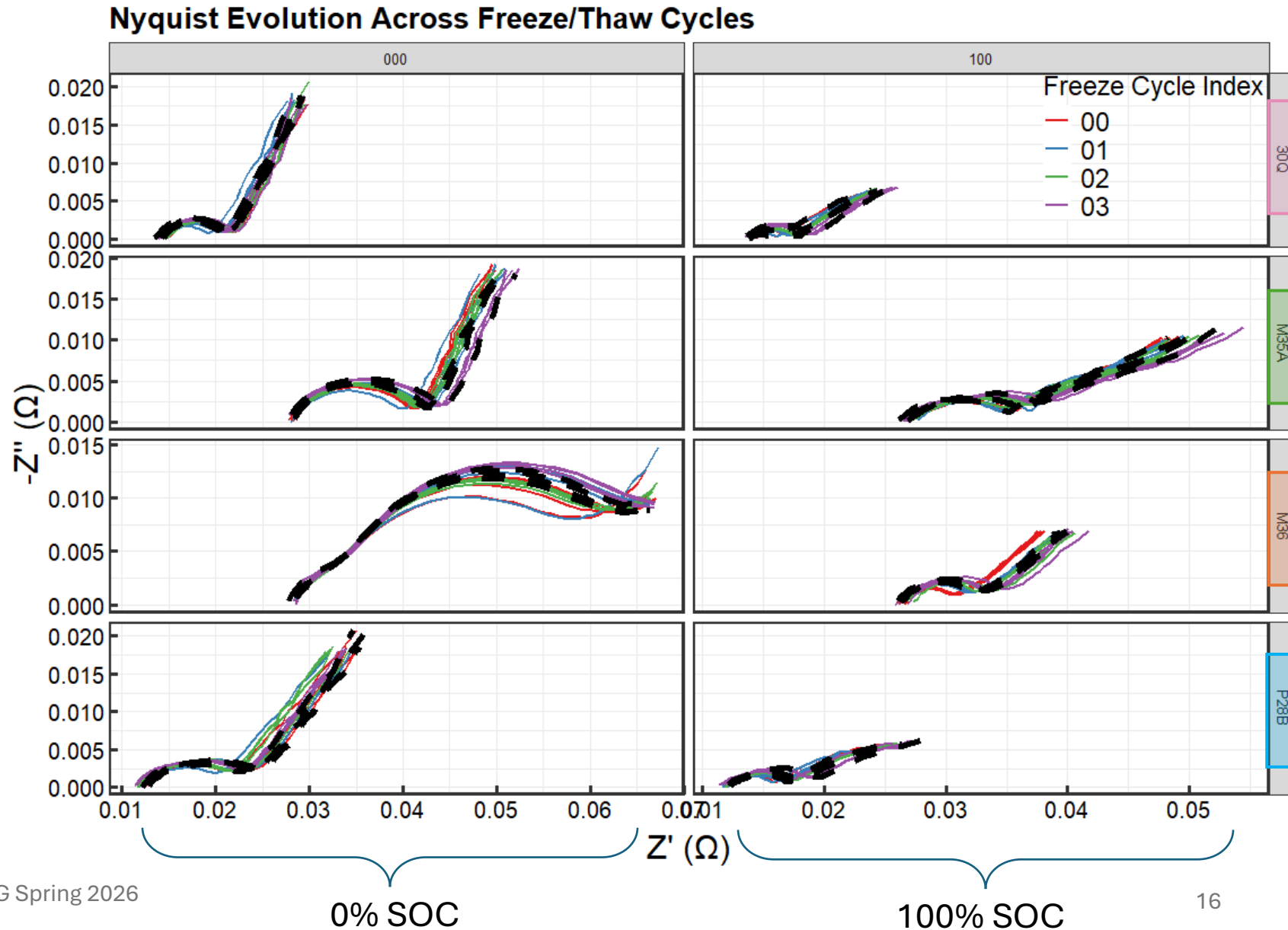
Post-Hibernation Comparison of Cell Behaviors: Cycling

Battery Cycle Testing Test:

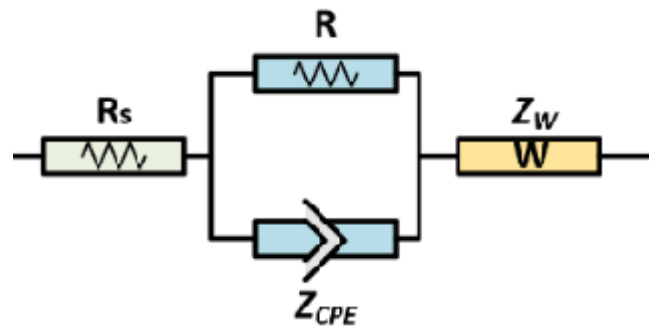
1. Rest 10 minutes
2. Variable Reset
3. C/5 Charge to 4.1V
4. Taper at 4.1V until 50mA
5. Rest 10 minutes
6. Discharge to 3.0V at C/5
7. Repeat steps 1-6 for a total of 3 cycles
8. Charge to designated SoC%



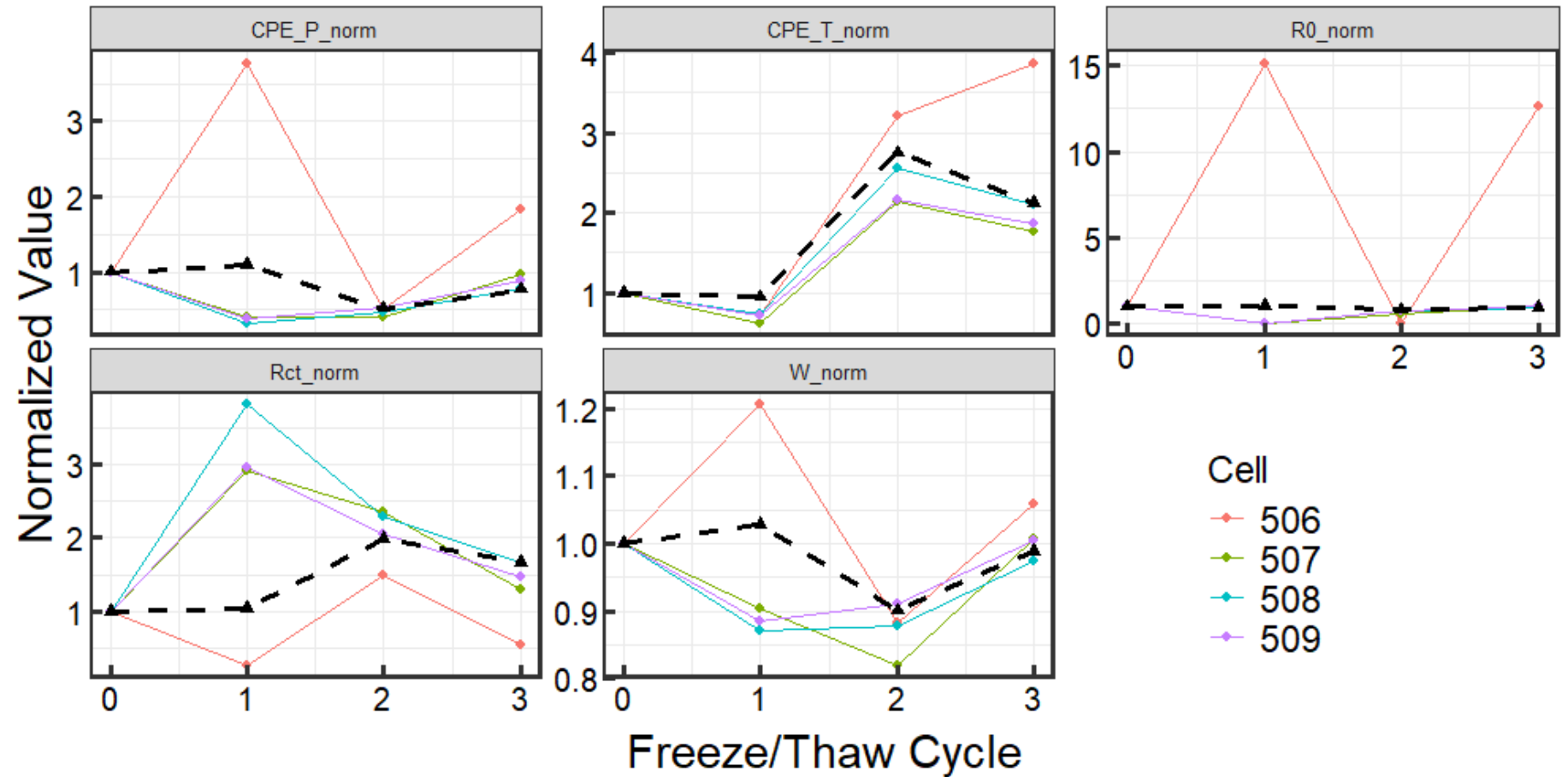
Post-Hibernation Comparison of Cell Behaviors: EIS



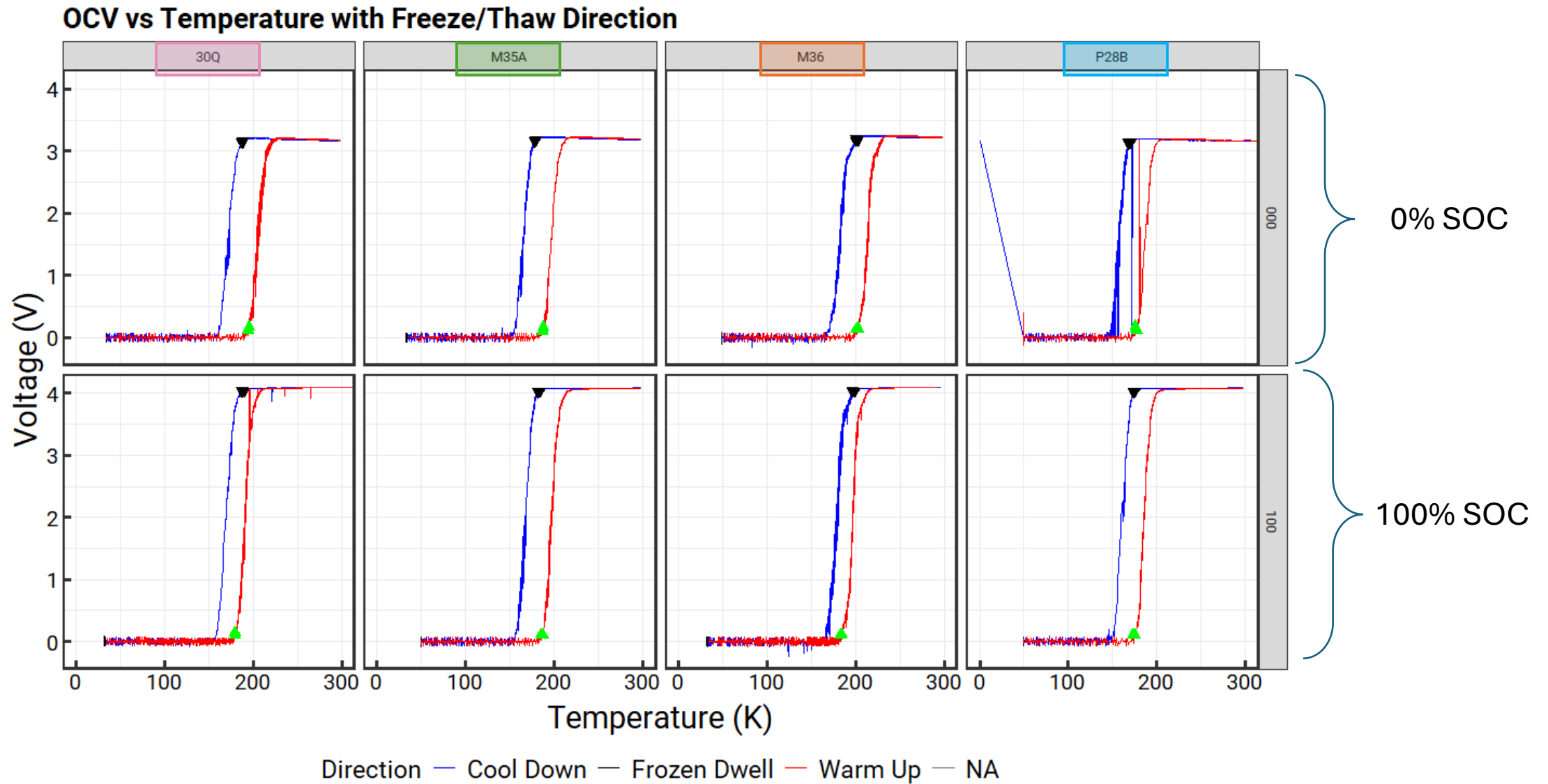
EIS Equivalent Circuit Model Parameter Fitting



Normalized Parameters for **M35A** at SoC 100



OCV Recovery Hysteresis



Cryogenic Testing Current Findings

Big Take-aways (so far)

- Completed full test matrix for **single-lunar-night assessment** across all four Strategic Reserve cell models (working on multiple-night assessment)
- **Validated thermal ramp rates:** No degradation observed when heating ≤ 2 K/min (consistent with lunar maximum rates).
- **Optimized dwell testing:** Demonstrated that extended low-temperature dwell (~14 days) is unnecessary; 1-day dwell sufficient once thermal equilibrium (~8 hr) is reached.
- **Confirmed hibernation success:** All cells recovered OCV and maintained minimal capacity degradation across all tested parameters
- **Preliminary insight:** Depleted cells (0% SoC, lithium stored in cathode) may be more favorable for lunar hibernation.

Upcoming Future Work – Putting it all Together

Battery Future Work

- Continue long-term cycling (≥ 100 cycles) to assess cell lifespan across multiple lunar days.
- **Down-select** to two best-performing cell models for higher testing throughput.
- Perform **non-destructive CT scans** (GLIMPSE) to study internal structures.
- Use long-term data to advance **BMS health and charge estimation models**.

Thermal Future Work

- Analyze **Gen 2 battery pack design**.
- Size heaters to achieve **2 K/min heat-up** (equator vs. South Pole).
- Develop thermal package to meet target heat-up rate.
- Validate **thermal runaway prevention**.

Electronics Future Work

- Expand cryogenic testing of circuits, components, and evaluation boards.
- Assess **microcontroller performance** at cryo temps; add heaters as needed.
- Build and validate a **full BMS** integrated with cells in vacuum chamber.

System Level Future Work

- Begin assembly of **8s1p battery pack** (28V, 3Ah, 80W system).
- Integrate BMS with battery system.
- Conduct **environmental testing** (radiation, EMI).
- Finalize and document operational protocols.

Thank you!

Single-Night Hibernation Works!

- Cells survived initial lunar night testing with minimal degradation

Technology Infusion Underway

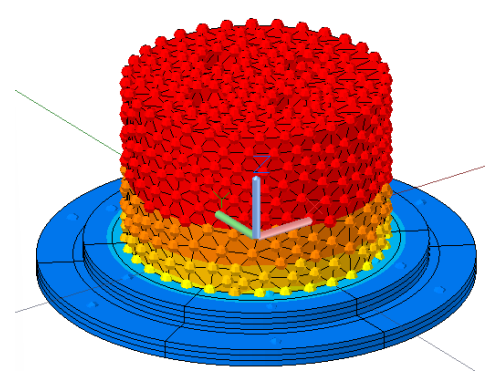
- Engaging with Moon-to-Mars programs, rover teams, and lunar infrastructure groups

Next Steps: Multi-Night & Integration

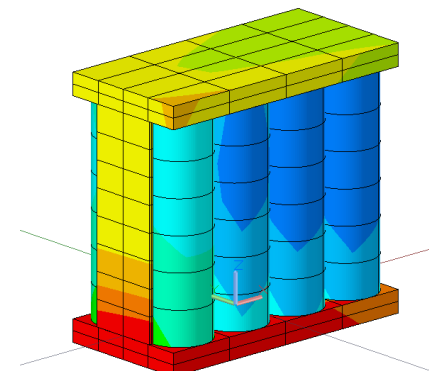
- Wrapping up multi-night evaluations and moving toward full battery + BMS integration



POC:
brian.a.holler@nasa.gov



0% SOC



100% SOC

