

High-Fidelity Modeling Approach for Coupling Thermal, Fluid, and Neutronic Analysis in an NTP Engine

*Nuclear and Emerging Technologies
for Space Conference*

*Looking Forward to Future Fission
Power Needs*

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Introduction

- For modern nuclear thermal propulsion (NTP) designs, modeling and simulation is vital for progressing lower TRL stages of development
 - Analysis of parameter spaces
 - Design tradeoffs
 - Can provide continual support through hardware development and testing
- A complete model of an NTP system must accurately couple thermal and neutronic effects
 - High thermal power density causes rapid changes in core temperature, in turn causing tangible effects on core power distribution
 - Fluid behavior (particularly density) also causes major changes
 - Rapid temperature increase causes density decrease, which reduces moderating power



Motivation

- **Goal: Design an NTP modeling methodology that can accurately couple conduction, thermal-hydraulics, and neutronics**
 - Use open-source software
 - Analyze a steady-state, full-power operational environment
 - Accurately reproduce publicly available test data
- **Test data: XE Prime reactor**, developed under the Nuclear Engine for Rocket Vehicle Application (NERVA) program [1]
 - Only NTP reactor to ever reach a flight configuration test
- **Software: Idaho National Laboratory's (INL) Multiphysics Object-Oriented Simulation Environment (MOOSE) [2]**
 - Including Thermal Hydraulics Module and associated software like Cardinal

[1] Robbins, W. H., and Finger, H. B., "An Historical Perspective of the NERVA Nuclear Rocket Engine Technology Program," *NASA Contractor Report 187154*, AIAA-91-3451, July 1991.

[2] Harbour, L., et al., "4.0 MOOSE: Enabling massively parallel Multiphysics simulation," *SoftwareX*, vol. 31, pp. 102264, 2025.



Methodology

A detailed space scene featuring two satellites. One satellite in the foreground has a gold-colored body and a blue solar panel, with a bright blue beam of light emanating from its antenna. Another satellite with yellow and blue components is further away, also emitting a blue beam. The Earth's horizon with city lights is visible at the bottom, and the Moon and Mars are in the dark, star-filled sky.

Software Tools

- **MOOSE**

- Conduction kernel for solid temperature calculations [2]
- Thermal Hydraulics Module for fluid mechanics [3]
- Multiapp, transfer systems to transfer boundary conditions between solutions [4]

- **OpenMC**

- Monte Carlo code – calculation of heat source distribution
- Has been shown to be useful for modeling advanced and experimental reactor designs [5]

- **Cardinal**

- Generates a mesh mirror used to link geometries across disparate codes [6]
- Used to link **MOOSE** and **OpenMC**

[2] Harbour, L., et al., “4.0 MOOSE: Enabling massively parallel Multiphysics simulation,” *SoftwareX*, vol. 31, pp. 102264, 2025.

[3] Hansel, J., Andrs, D., Charlot, L., Giudicelli, G., “The MOOSE Thermal Hydraulics Module,” *Journal of Open Source Software*, vol. 9, no. 94, pp. 6164, 2024.

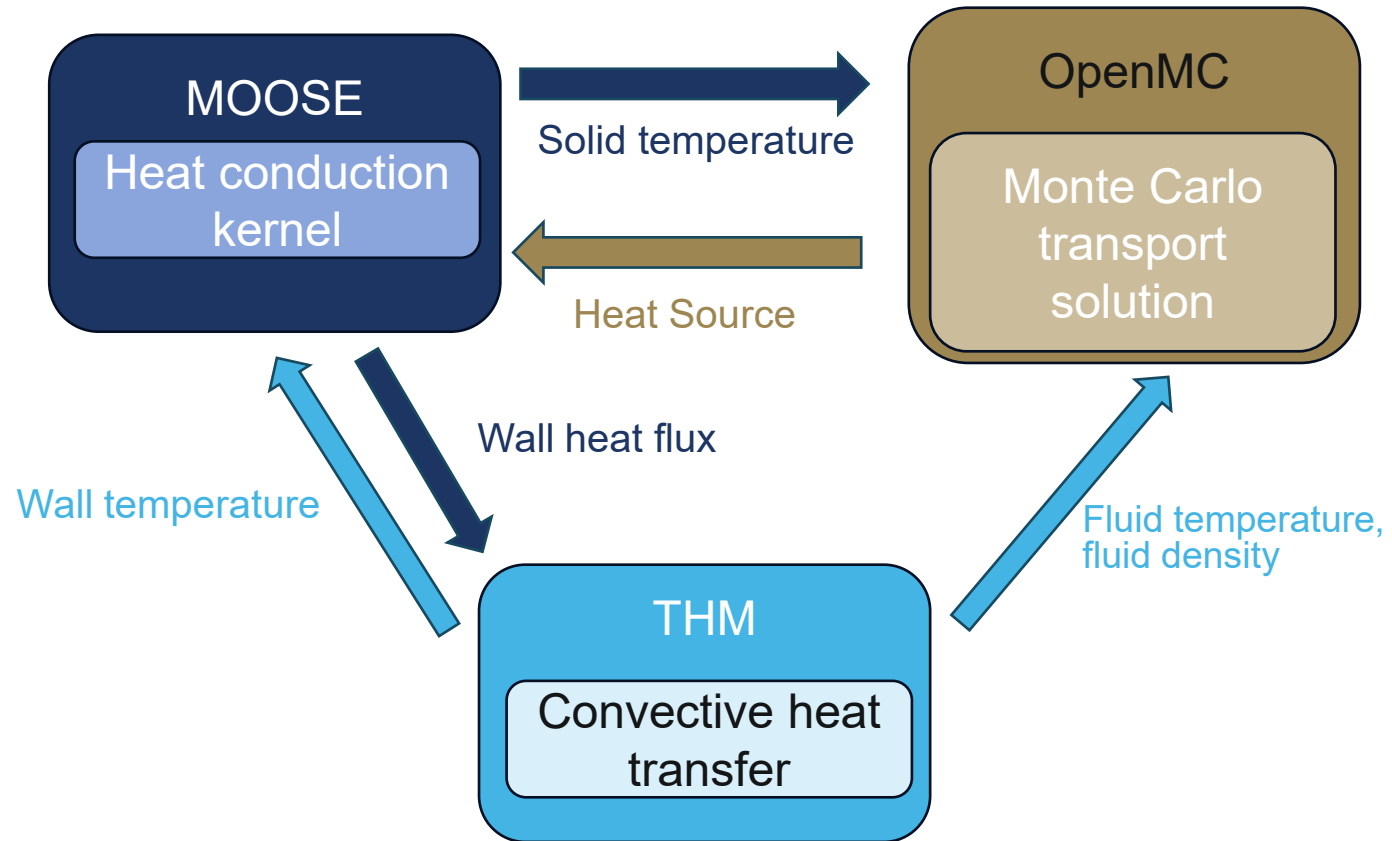
[4] Giudicelli, G., et al. “Data transfers for nuclear reactor multiphysics studies using the MOOSE framework,” *Frontiers in Nuclear Engineering*, vol. 4, 2025.

[5] Nguyen, K. H. N., and DeHart, M. D., “Modeling of the Advanced Test Reactor Using OpenMC, Cubit, and Griffin,” Idaho National Laboratory, INL/RPT—23-74489, August 2023.

[6] Novak, A. J., “Coupled Monte Carlo and thermal-fluid modeling of high temperature gas reactors using Cardinal,” *Annals of Nuclear Energy*, vol. 177, pp. 109310, 2022.

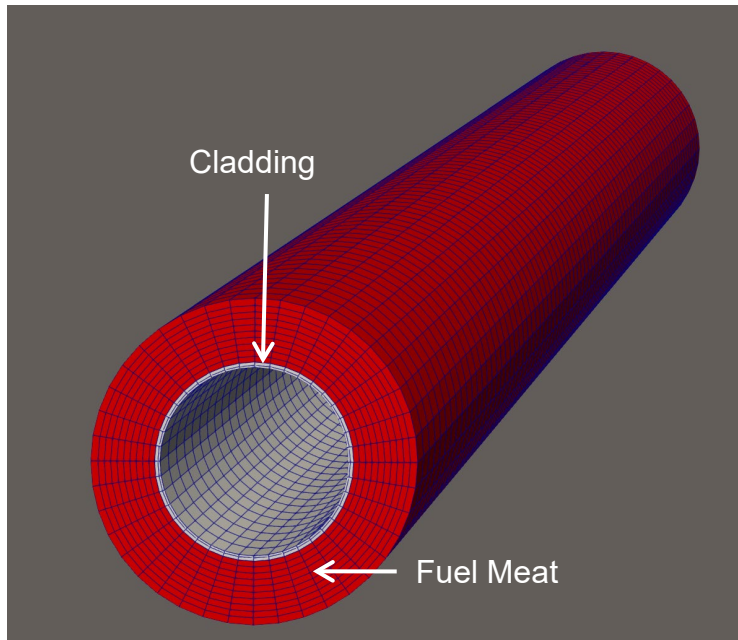


Coupling Sequence

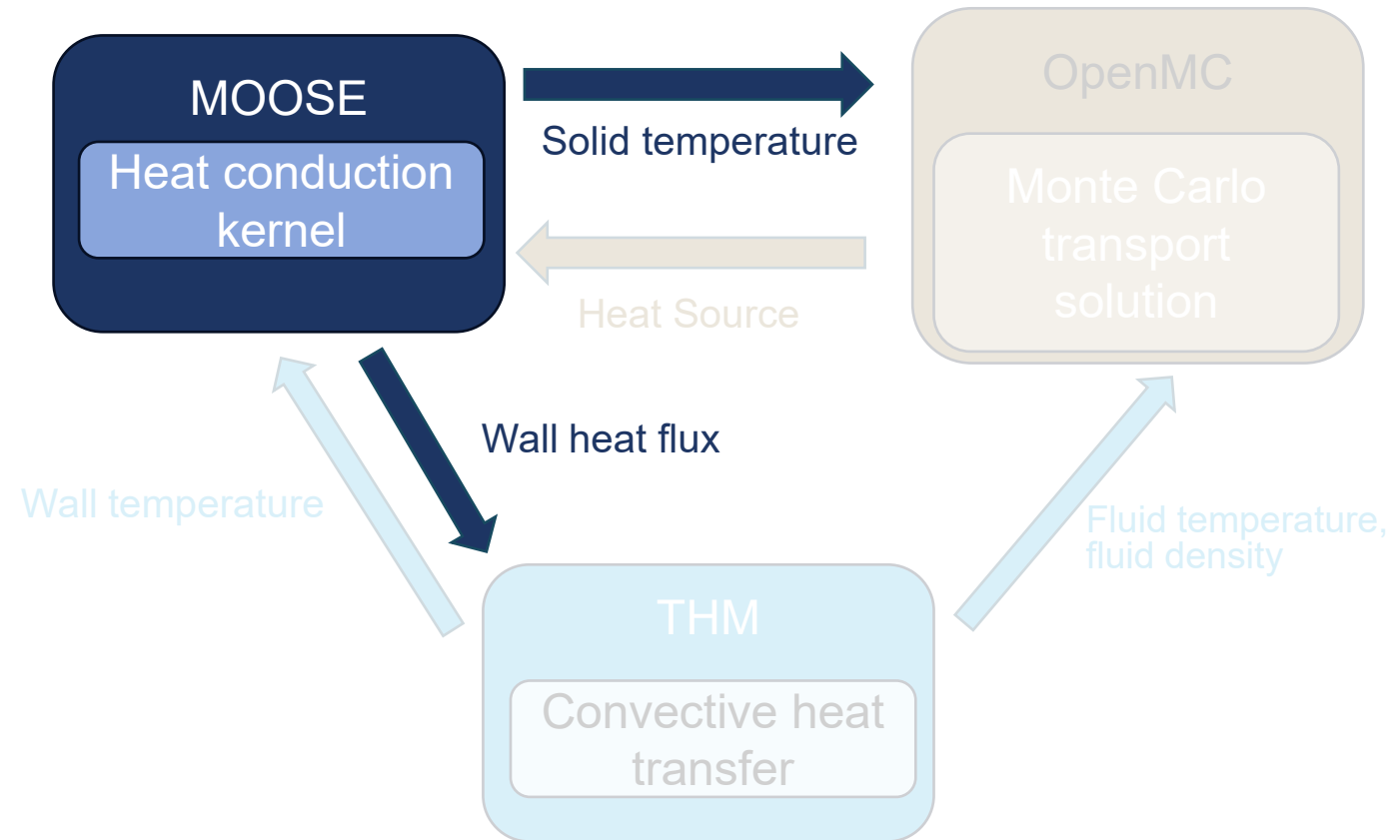


Solid Heat Conduction

- Heat conduction kernel
 - Uses the thermal energy conservation equation [2]



Solid mesh for a single coolant channel

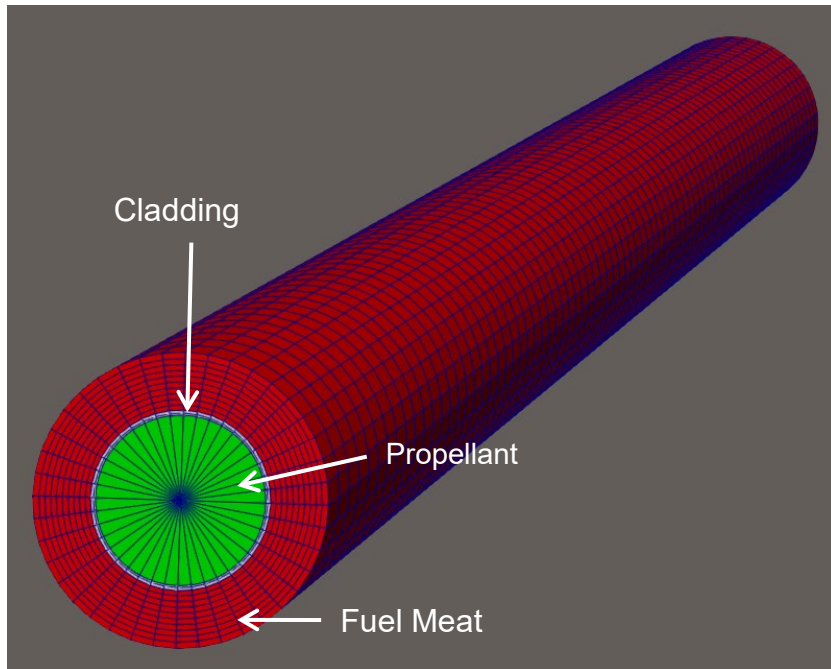


$$-\nabla \cdot (k \nabla T) = 0$$

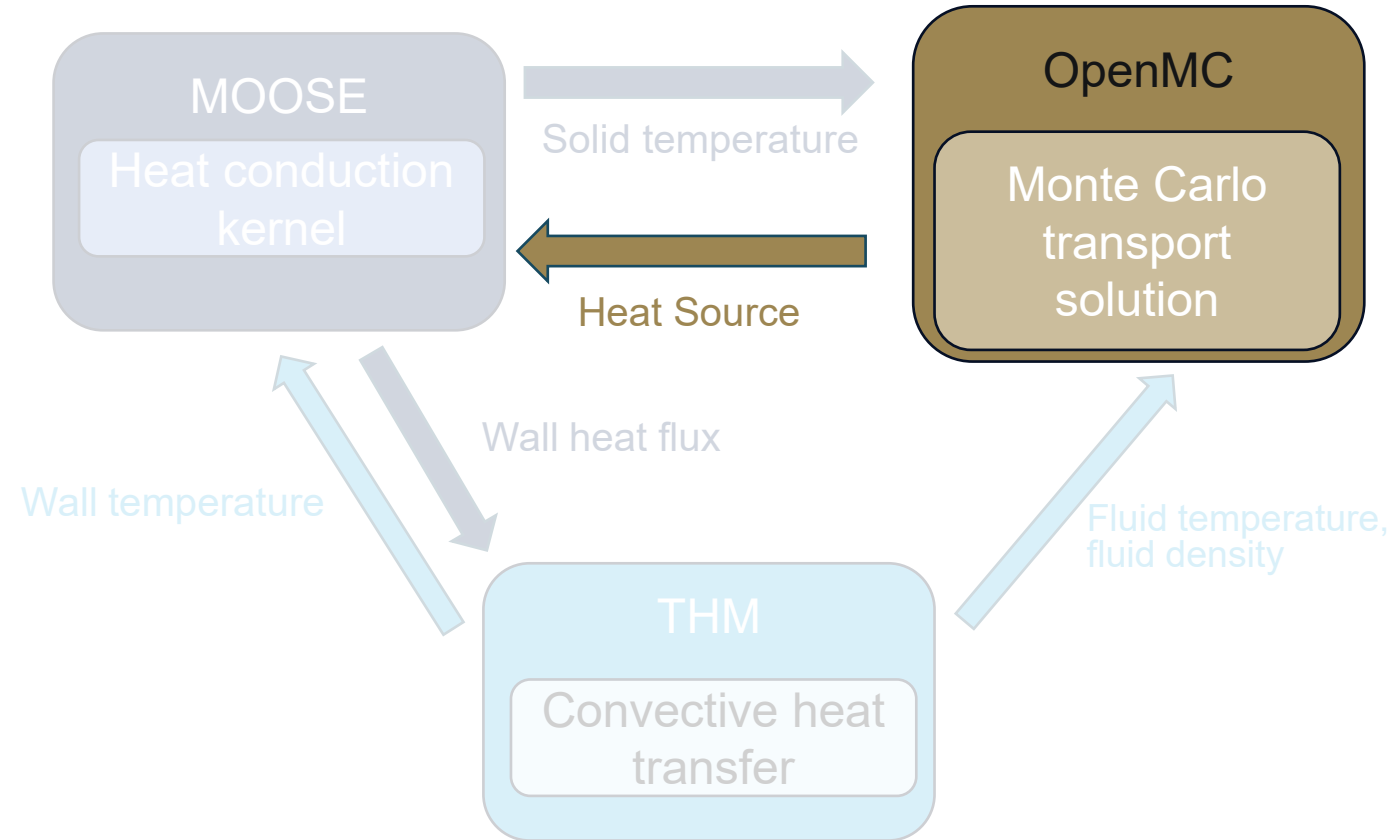
Neutronics

- Heat source

- **Nuclear heating** is calculated by OpenMC
- This becomes a **source term** added to the **thermal energy conservation** equation [2]



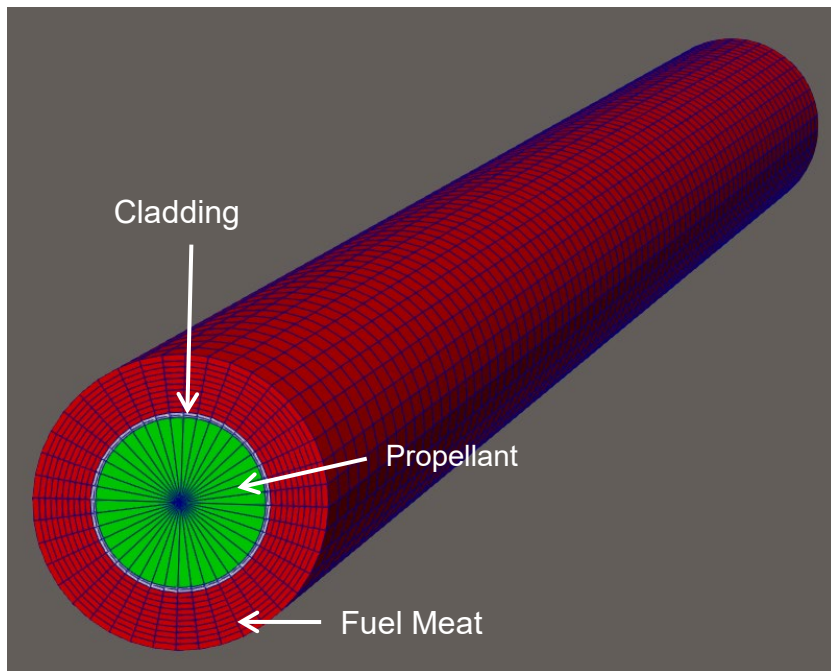
Solid & fluid mesh for a single coolant channel



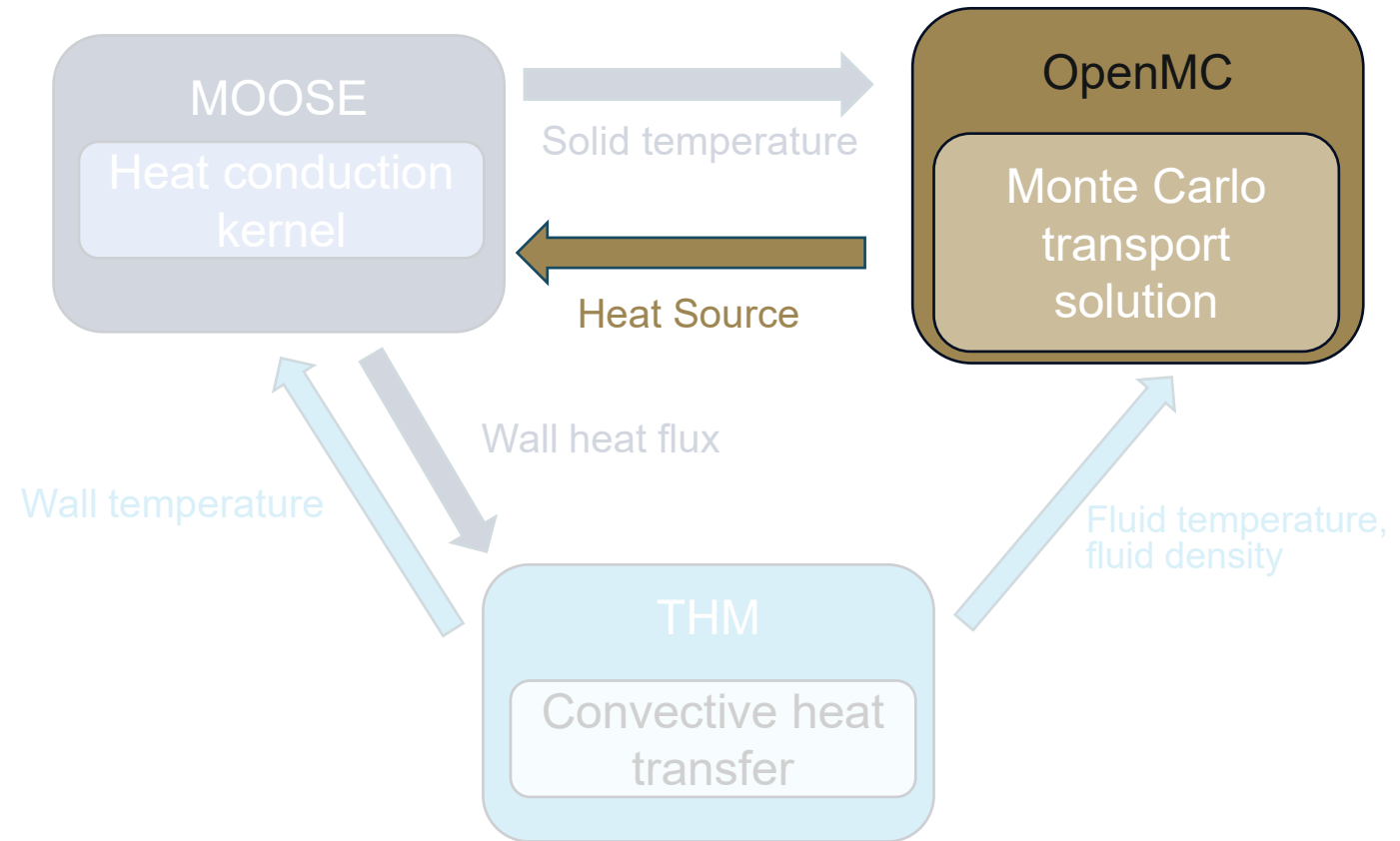
$$-\nabla \cdot (k \nabla T) + \textit{Source} = 0$$

Neutronics

- ****Assumptions**
 - Photon heating need not be considered
 - Source is constant in each material region within each axial layer
 - Source only present in fuel



Solid & fluid mesh for a single coolant channel



$$-\nabla \cdot (k \nabla T) + \textit{Source} = 0$$

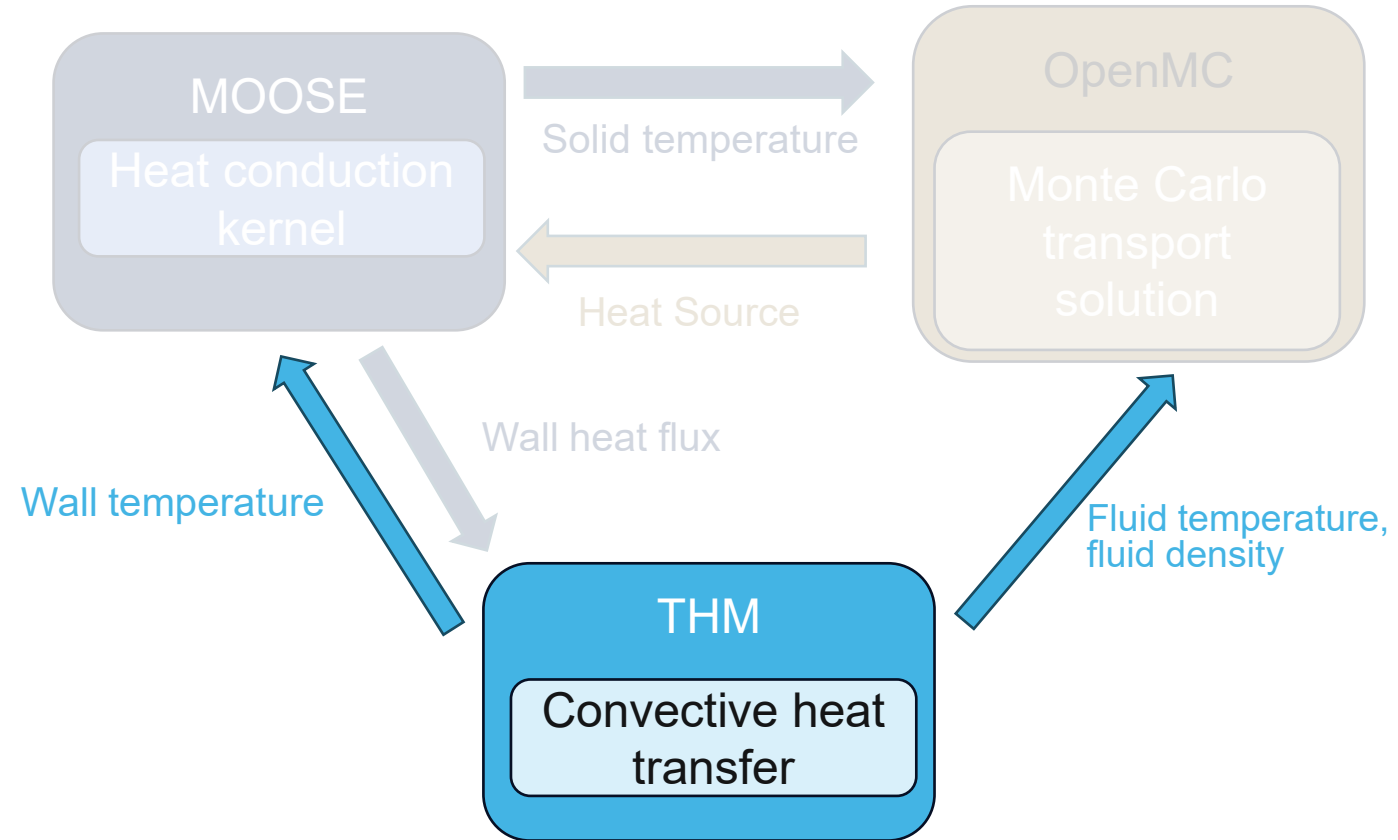
Thermal Hydraulics

- Flow is modeled as 1D, single-phase flow paths
- Split into 100 axial segments of equal length
- Segments are **governed by compressible Euler equations [3]**:

$$\frac{\partial \rho A}{\partial t} + \frac{\partial \rho u A}{\partial x} = 0$$

$$\frac{\partial \rho u A}{\partial t} + \frac{\partial (\rho u^2 + p) A}{\partial x} = p \frac{\partial A}{\partial x} - F A + \rho g_x A$$

$$\frac{\partial \rho E A}{\partial t} + \frac{\partial u(\rho E + p) A}{\partial x} = \rho u g_x A + q''' A$$



Thermal Hydraulics

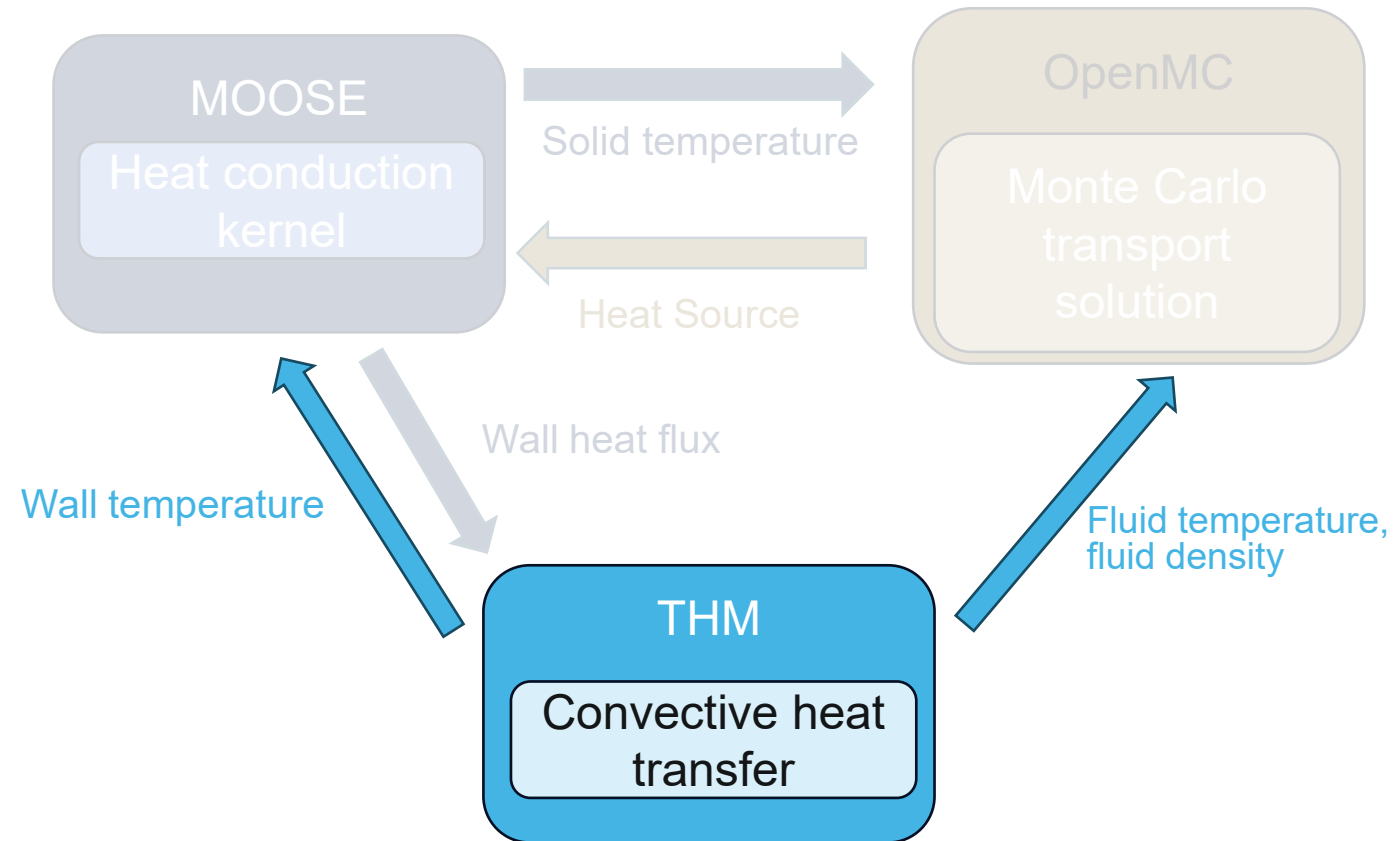
- Reynolds and Prandtl numbers solved using **hydrogen fluid properties** [7]

- Used to calculate Nusselt number with a **Dittus-Boelter correlation** [8]

$$Nu = 0.023Re^{0.8}Pr^{0.4}$$

- ****Assumptions**

- Hydrogen behaves as an ideal gas
- Dittus-Boelter correlation is adequate
- Wall temperature and wall heat flux are channel-averaged for each axial layer



[7] Giudicelli, G., et al., "The MOOSE fluid properties module," Computer Physics Communications, vol. 307, pp. 109407, 2025.

[8] Winterton, R. H. S., "Where did the Dittus and Boelter Equation come from?" International Journal of Heat and Mass Transfer, vol. 41, no. 4, pp. 809-810. January 1998.

Thermal Hydraulics

- Reynolds and Prandtl numbers solved using hydrogen fluid properties

- Used to calculate Darcy friction factor with a **Churchill correlation [3]**

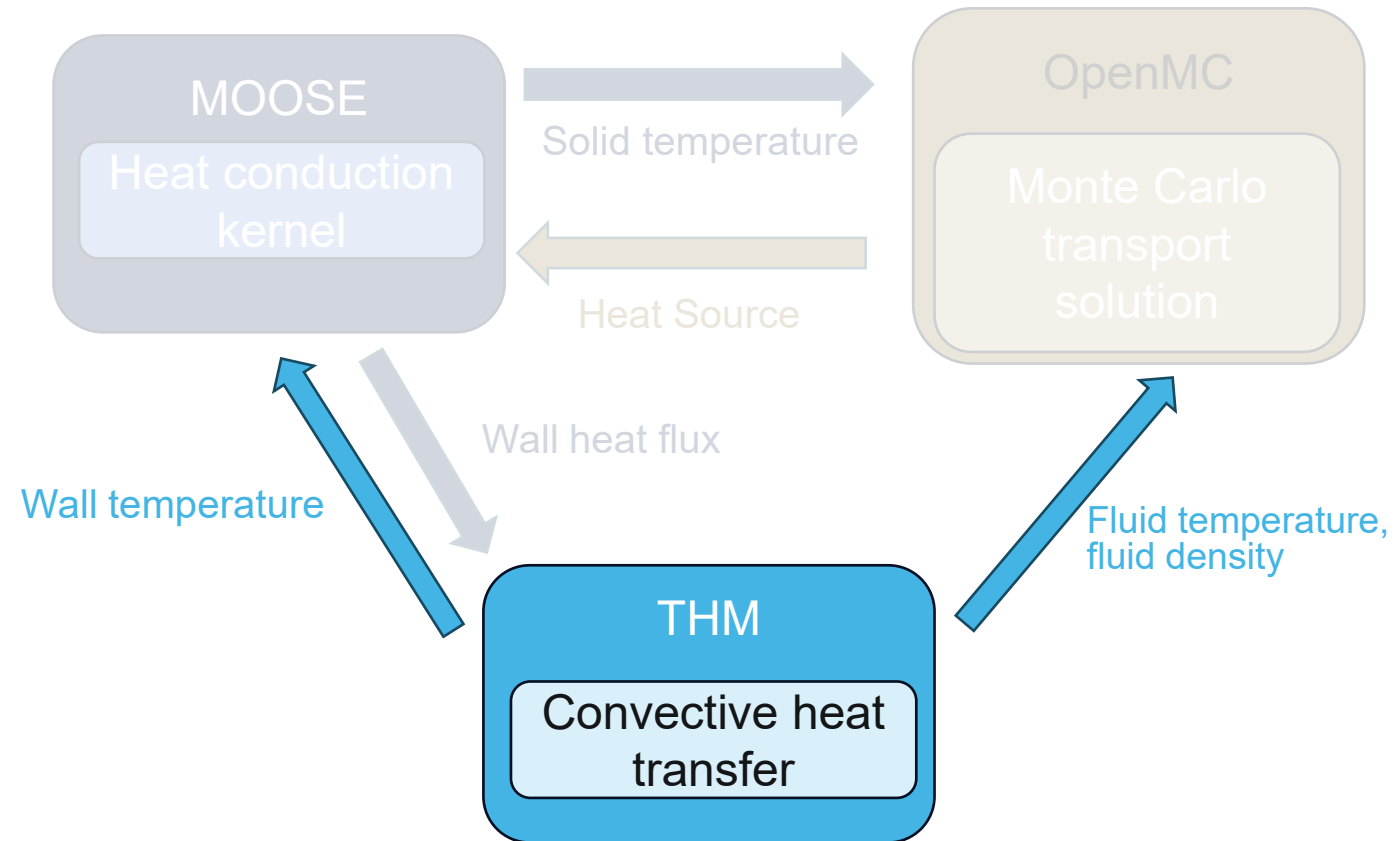
$$F = \frac{f \rho u |u| A}{2 D_h}$$

$$f = 8 \left[\left(\frac{8}{Re} \right)^{12} + \frac{1}{(A + B)^{1.5}} \right]^{\frac{1}{12}}$$

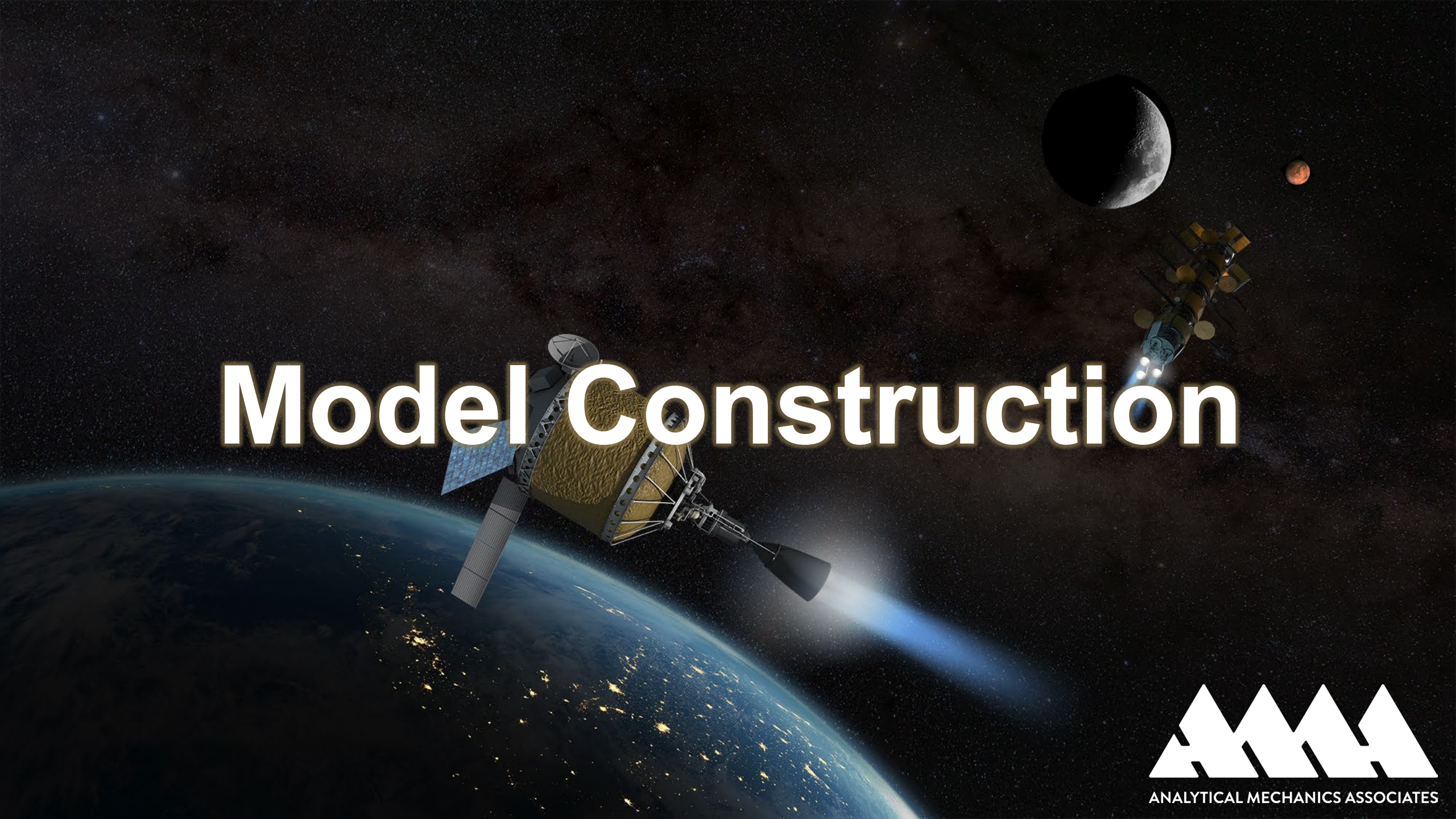
$$A = \left[2.457 \ln \left(\frac{1}{\left(\frac{7}{Re} \right)^{0.9} + 0.27 \frac{\epsilon}{D_h}} \right) \right]^{16}$$

$$B = \left(\frac{37530}{Re} \right)^{16}$$

****Assumption: Channel roughness is constant**

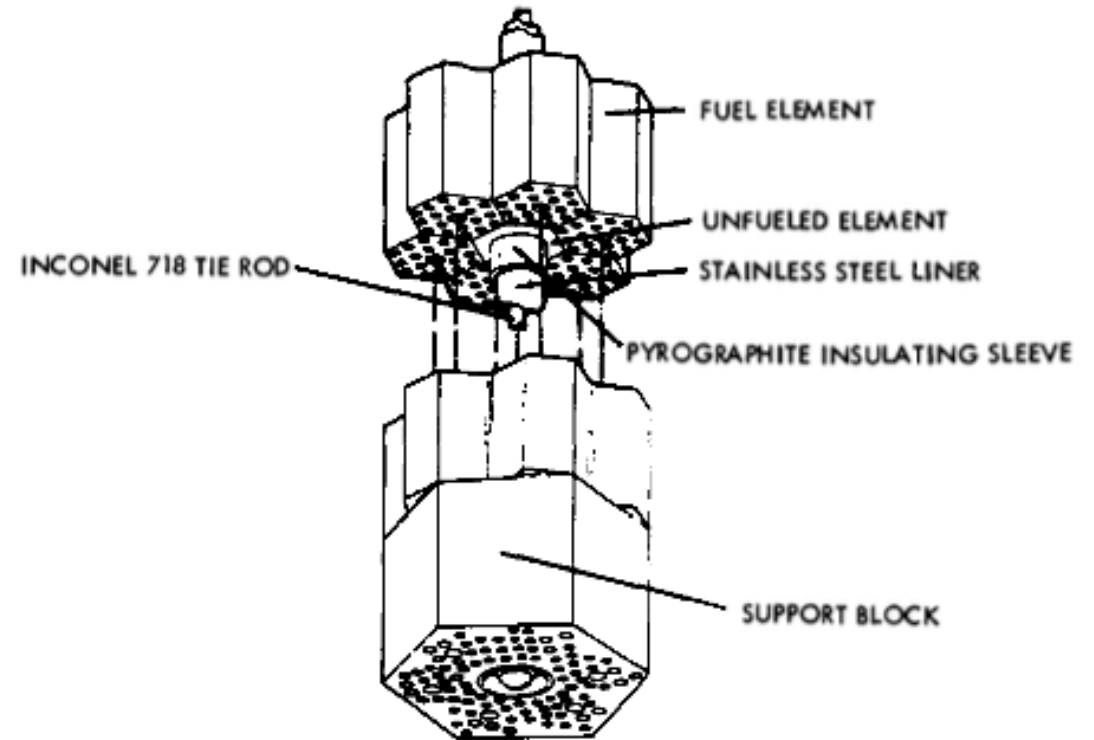


Model Construction

A detailed space scene featuring two satellites in orbit above Earth. The satellite in the foreground has a gold-colored body and a blue solar panel, with a bright light beam emanating from its antenna. The second satellite, further away, has a yellow and blue body. The Earth's horizon is visible at the bottom, showing city lights. The background includes the Moon and Mars against a starry sky.

Regular cluster

- XE Prime reactor – active core comprises 199 “regular clusters”
 - Single unfueled element surrounded on all sides by fueled elements
 - Elements are hexagonal



XE Prime regular cluster [9]

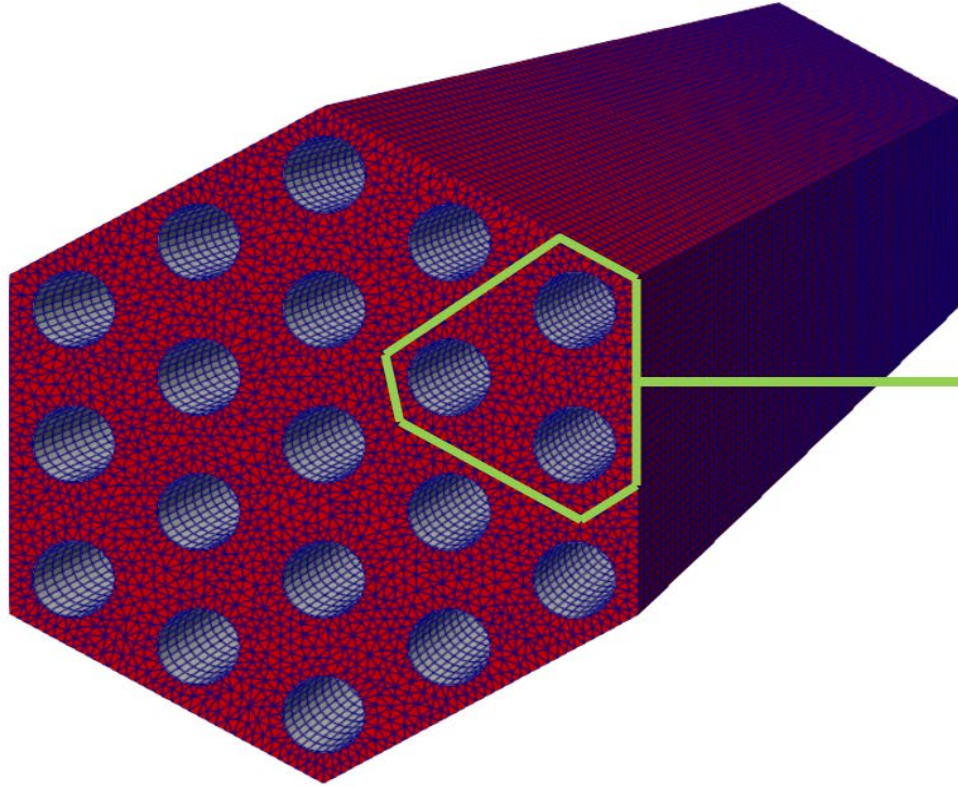
Fueled Element

- Graphite matrix loaded with UC fuel pellets [9]
 - Pellets enriched to 93.15 wt%
 - 19 equally spaced coolant channels coated in NbC
 - 52 inches long, with one inch of bottom tip unfueled
 - Only fueled portion is modeled
 - Densities calculated from XE Prime as-built data
 - Material properties based on Aerojet databooks for the NERVA program

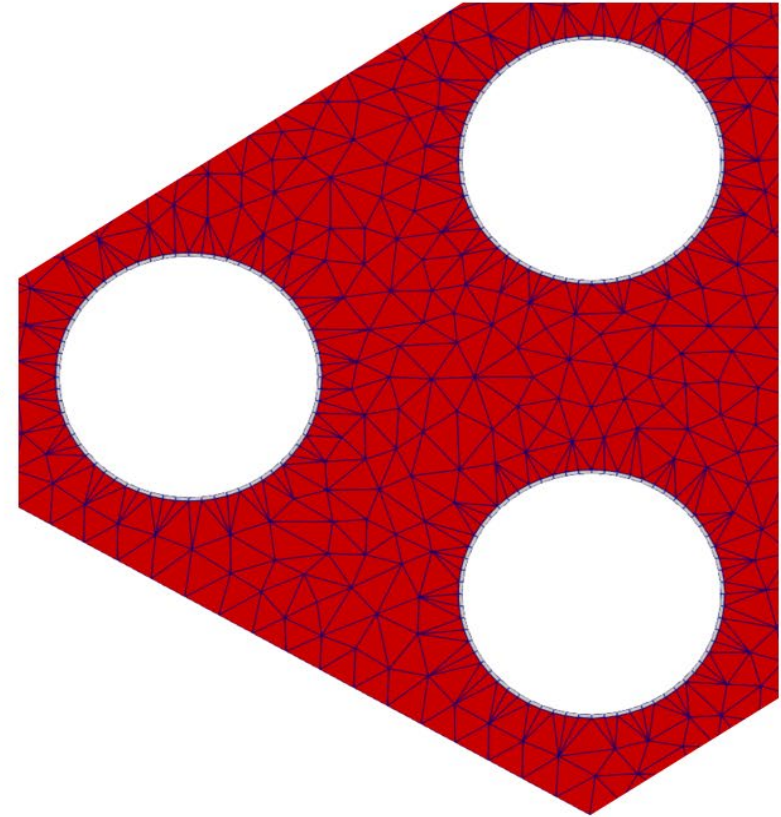
Fueled element model dimensions

Parameter [units]	Value
Element pitch [cm]	1.91262
Reamed channel radius [cm]	0.12573
Coated channel radius [cm]	0.1215644
Fueled length [cm]	1.2954
Channel pitch [cm]	0.43942
Fueled matrix density [g/cc]	2.217

Fueled Element



Fueled element mesh



Slice view of three channels



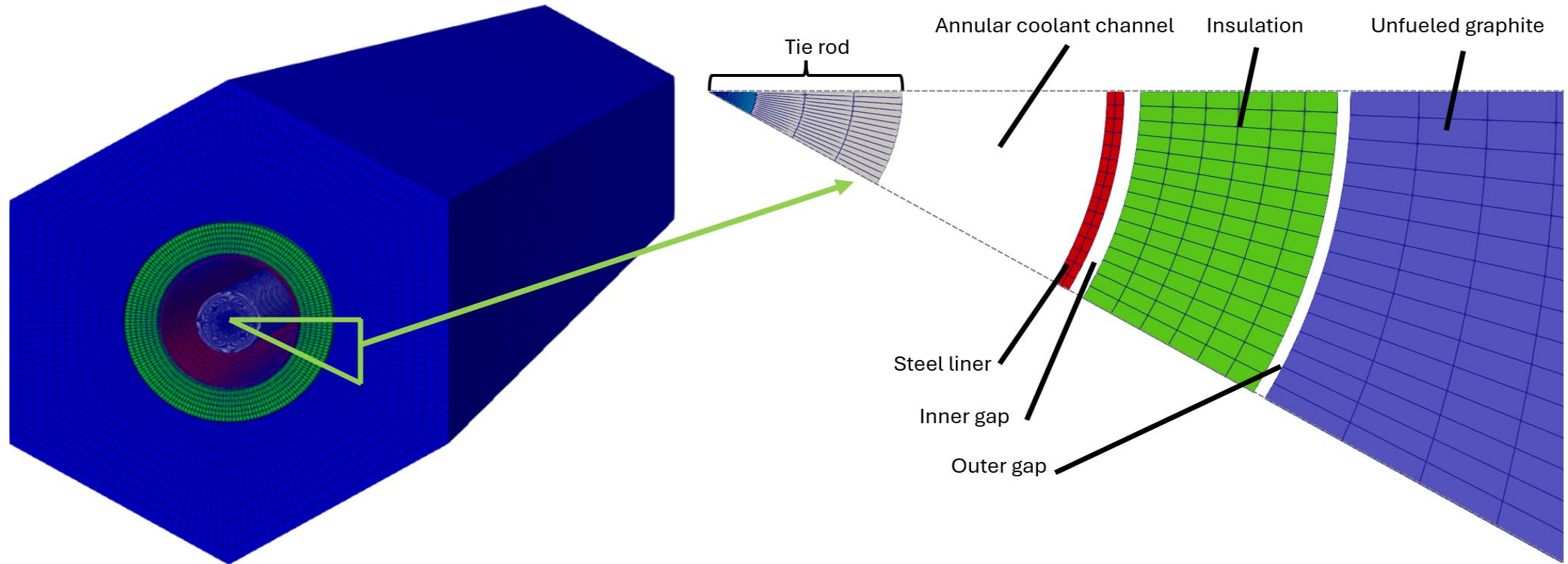
Unfueled Element

- Single, annular coolant channel [9]
 - Inner wall is a structural tie rod made of Inconel-718
 - Outer wall is stainless steel liner
 - Liner is wrapped in insulation layer of pyrolytic graphite
 - Channel is inserted into hexagon with single, large hole
 - Made of unfueled matrix material
 - Only first 51 inches are modeled to match length with fueled element
 - As-built data used again for density
 - Aeroject databook used again for material properties
 - Pyrographite has two given conductivities (in-plane vs. through-plane)
 - Through-plane is used for entire mesh

Unfueled element model dimensions

Parameter [units]	Value
Element pitch [cm]	1.91262
Tie rod radius [cm]	0.140335
Coolant channel outer radius [cm]	0.28702
Steel liner outer radius [cm]	0.29972
Insulation inner radius [cm]	0.31115
Insulation outer radius [cm]	0.45339
Unfueled graphite inner radius [cm]	0.46228
Element length [cm]	1.2954

Unfueled Element

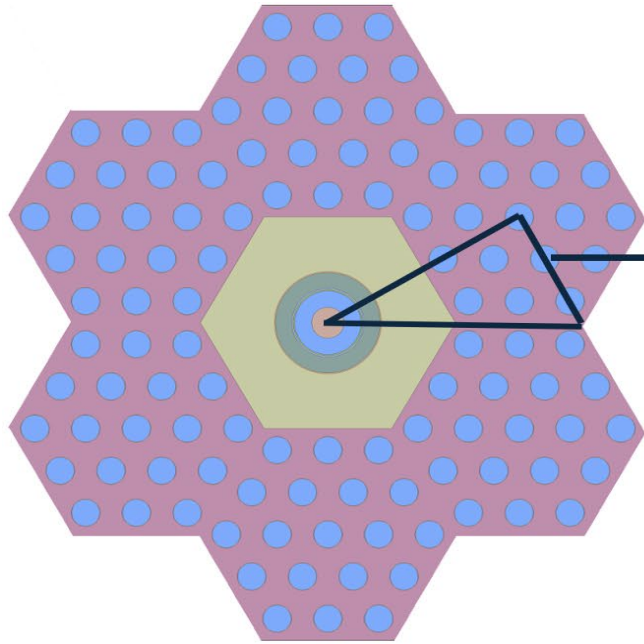


Unfueled element mesh

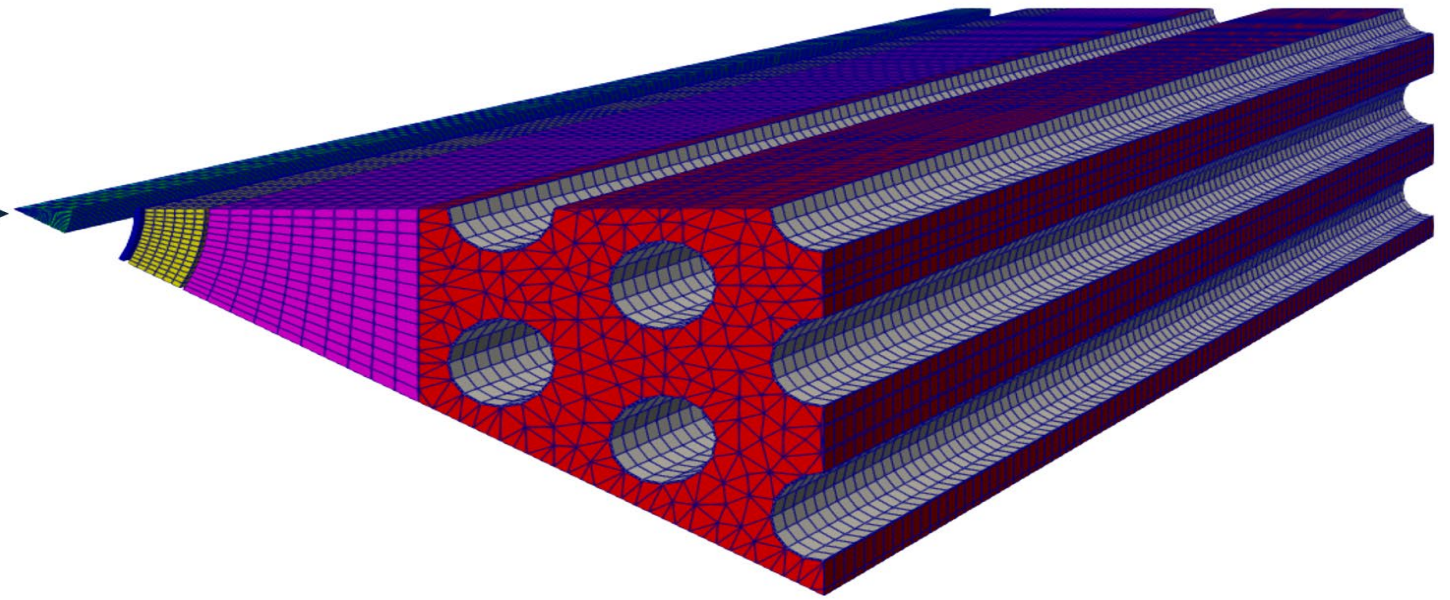
Annular channel slice view

Cluster Slice

- Rather than simulate an entire regular cluster, a 30° slice was made based on three axes of symmetry
 - Each is modeled thermally as an adiabatic wall and neutronically as a perfectly reflective surface



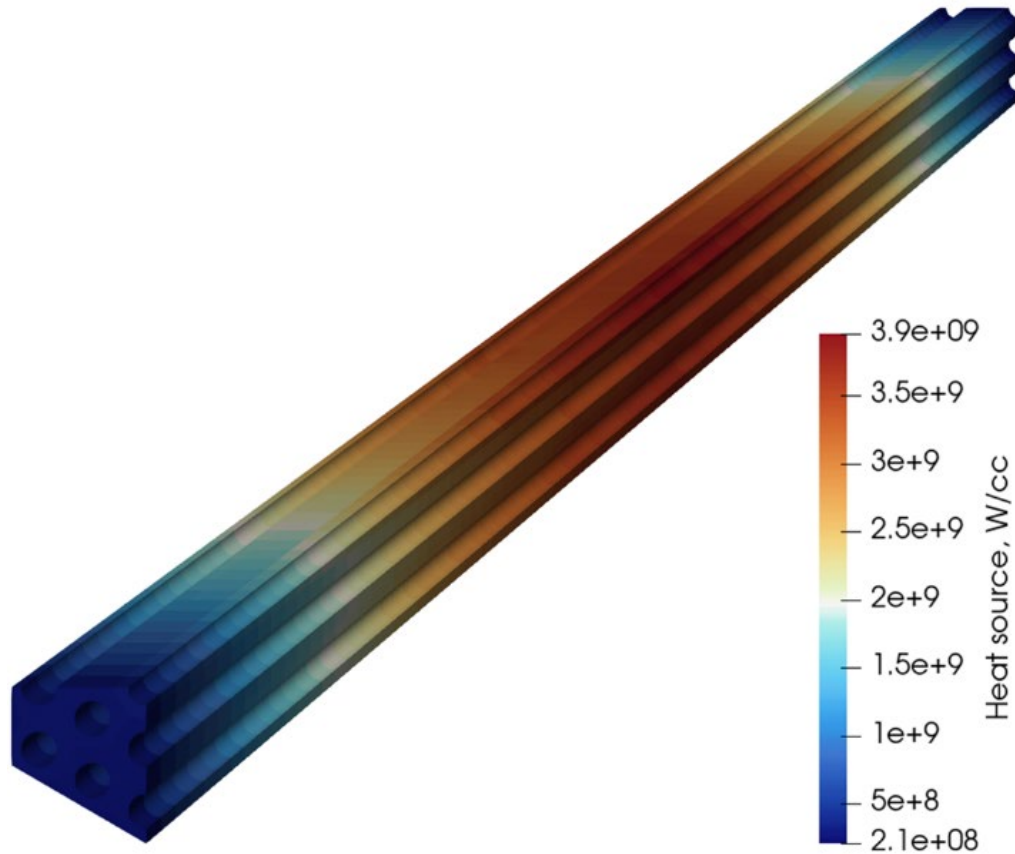
Regular cluster (top-down)



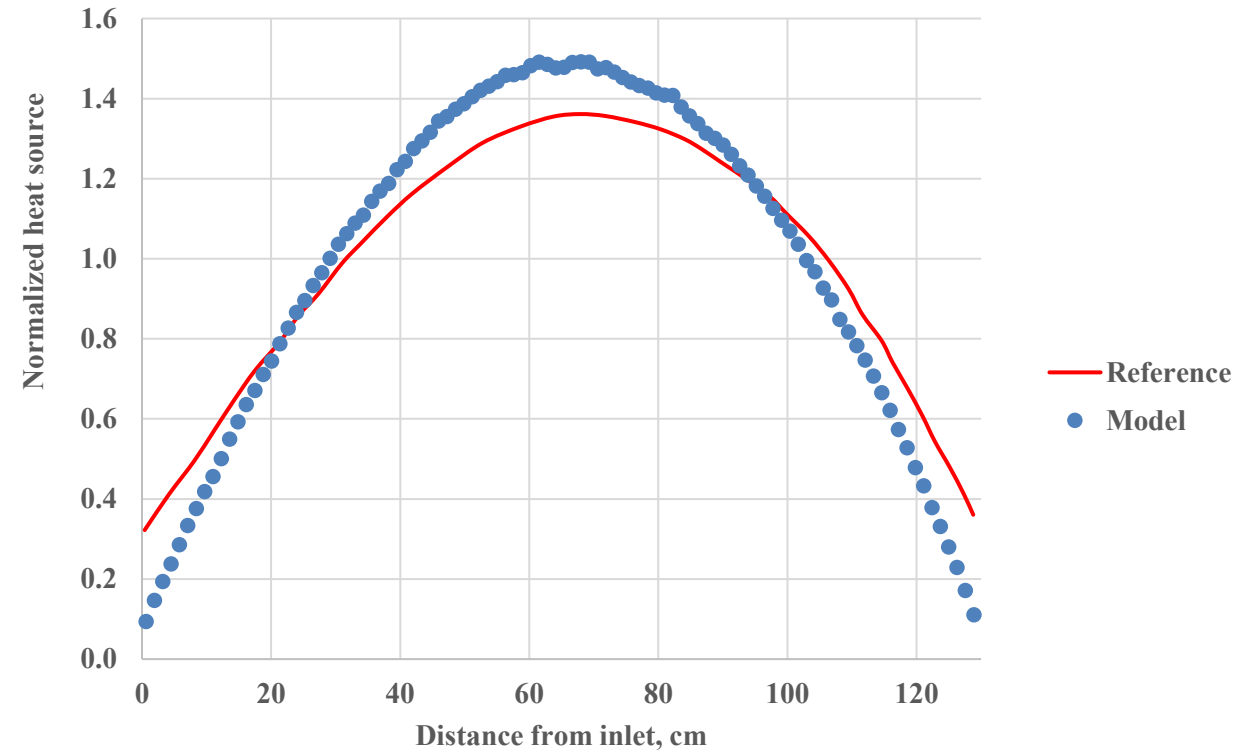
Regular cluster slice mesh

Results

Power Shape



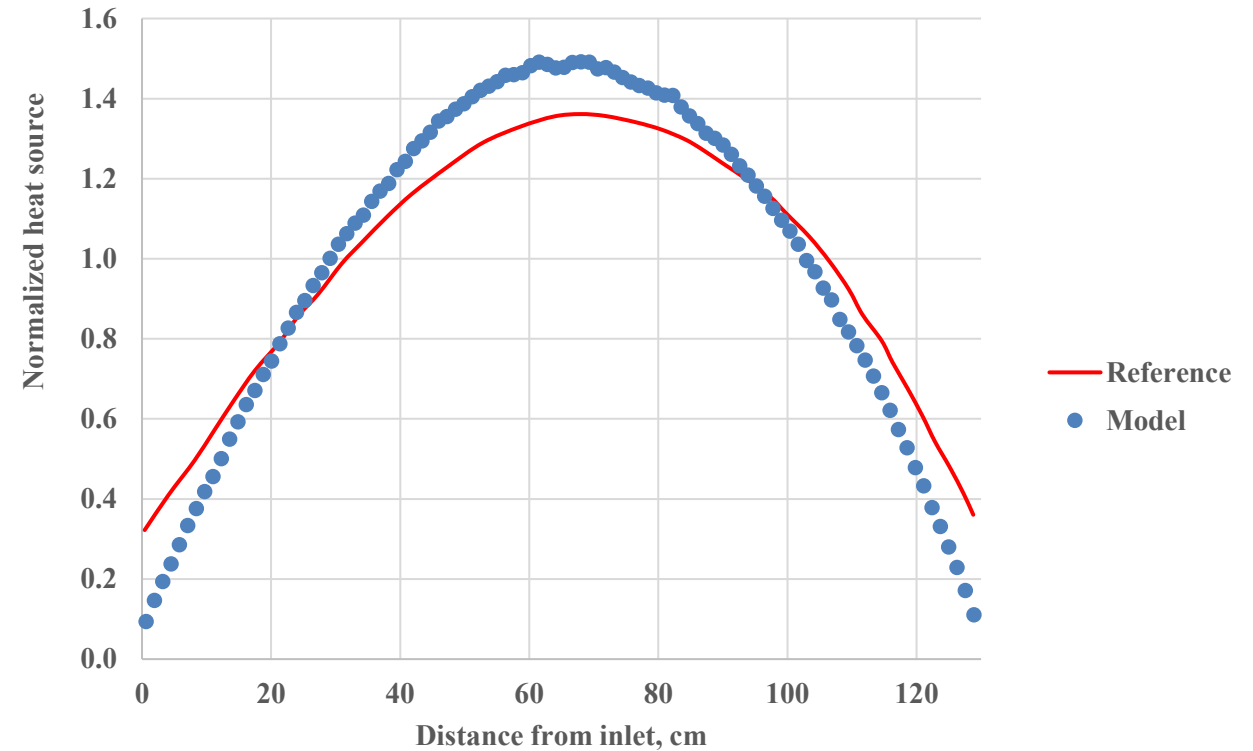
Volumetric heat source distribution



Axial heat source distribution

Power Shape

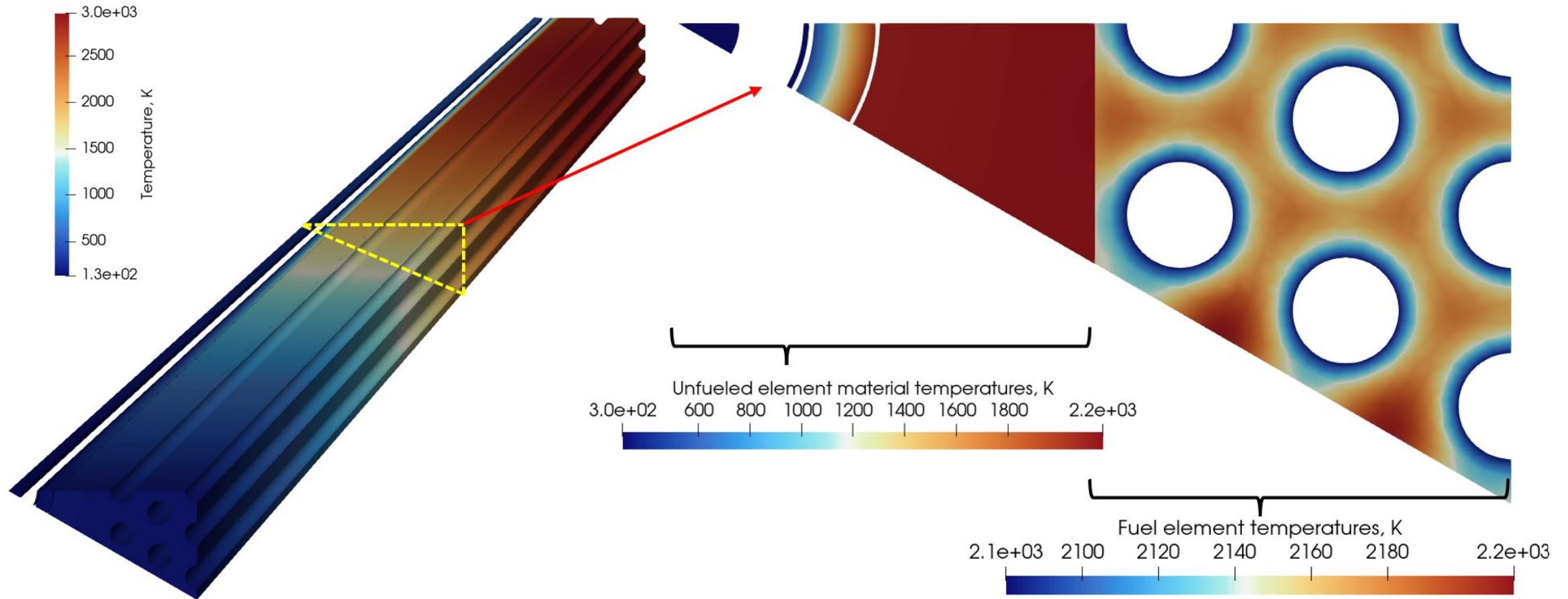
- Model peaks far more sharply than the reference
 - Lack of dome-side axial reflection and shielding
 - Lack of aft-side unfueled-tip
 - Slice model does not account for irregular cluster arrangements in active core periphery, nor any radial reflection



Axial heat source distribution



Solid Temperature



Solid temperature distribution

Mid-plane radial temperature distribution

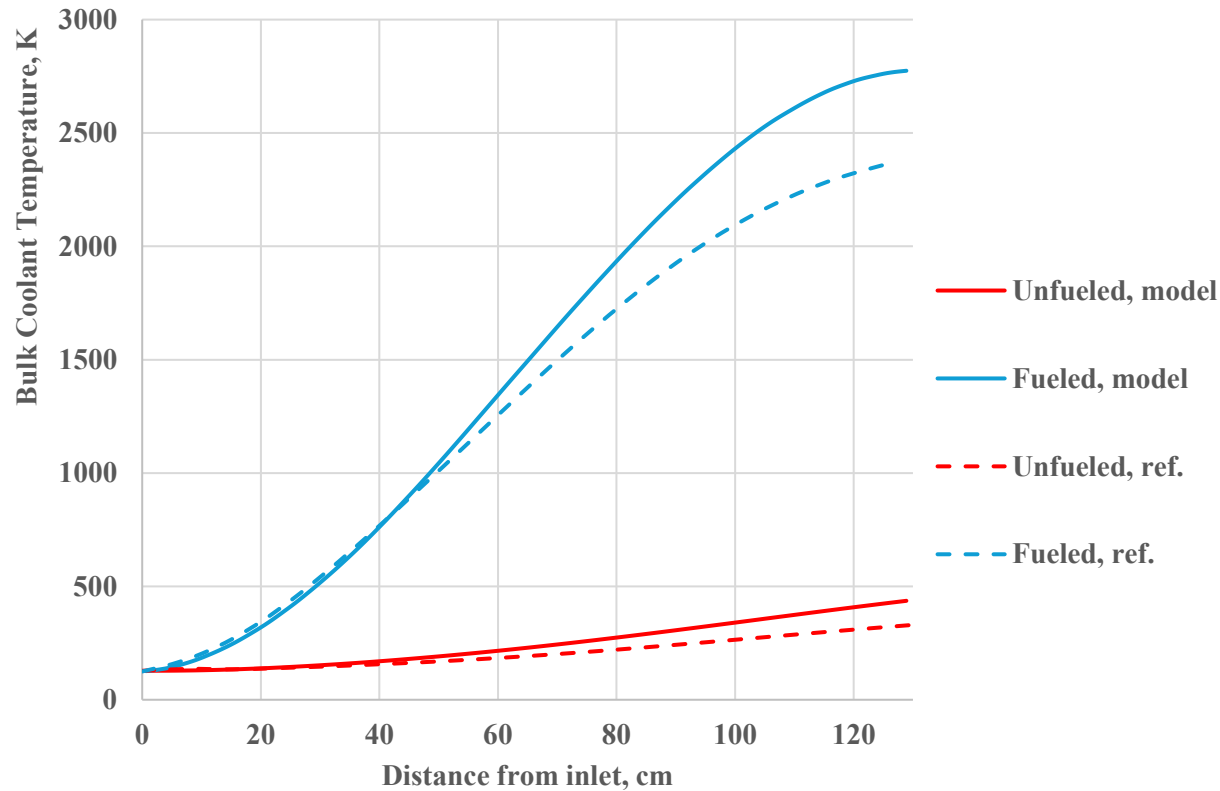
Solid Temperature

Max temperature of each mesh region

Region	Max temperature, reference [K]	Max temperature, model [K]	Error [%]
Fueled matrix	2473	2959	19.7
Channel coating	2392	2923	22.1
Unfueled matrix	2464	2932	19.0
Pyrolytic graphite	2368	2811	18.7
Steel liner	411.2	537.2	30.6
Tie rod	348.3	524.9	50.7

- All temperatures are way too high
 - Can be reduced by substituting Dittus-Boelter with Wolf-McCarthy [10]
 - Ideal gas hydrogen needs to be substituted with parahydrogen
 - Other minor issues
 - False axis of symmetry
 - Gap relationships use constant emissivity and default conductivity

Bulk Coolant Temperature



Bulk temperature axial distribution

- Temperatures are too high again, likely for similar reasons as with the solid temperatures
 - Dittus-Boelter needs to be substituted with Wolf-McCarthy
 - Ideal gas hydrogen needs to be substituted with parahydrogen

Conclusions

A detailed space scene featuring two satellites. In the foreground, a satellite with a gold-colored body and a blue solar panel is shown in orbit above the Earth's horizon. It has a black conical antenna pointing towards the right, from which a bright blue beam of light emanates. In the background, another satellite with a yellow and blue body is visible, also emitting a blue beam of light. The Earth's horizon is at the bottom, showing a blue atmosphere and yellow city lights. The background is a deep black space filled with stars, with a large, dark, cratered moon in the upper right and a small reddish planet in the distance.

Conclusions

- Major refinements to this model are needed
 - Nusselt number correlation
 - Fluid properties
 - Hydrogen gaps
 - Axial reflectors, unfueled tip, support block
- However, the approach still has merit
 - Successful coupling of three key physical behaviors
 - Can be expanded across multiple time steps



Next Steps

- Refine model
- Expand model to full core
 - Include irregular clusters on periphery, radial reflectors, control drums, etc.
- Expand methodology for transient analysis
 - Link PRK solver into solution sequence
 - Include reactivity effects of temperature changes, propellant density changes, control drum movements
- Link methodology to system-level analysis
 - Engine cycle model can dynamically update inlet/outlet conditions as time steps progress



Acknowledgements

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