DECOUPLING IN THE
DESIGN AND SYNTHESIS OF
MULTIVARIABLE CONTROL SYSTEMS

by Peter L. Falb and William A. Wolovich

Electronics Research Center
Cambridge, Mass.
DECOUPLING IN THE DESIGN AND SYNTHESIS OF
MULTIVARIABLE CONTROL SYSTEMS

By Peter L. Falb and William A. Wolovich
Electronics Research Center
Cambridge, Mass.

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

For sale by the Clearinghouse for Federal Scientific and Technical Information
Springfield, Virginia 22151 – CFSTI price $3.00
DECOUPLING IN THE DESIGN AND SYNTHESIS OF MULTIVARIABLE CONTROL SYSTEMS

By Peter L. Falb* and William A. Wolovich
Electronics Research Center

SUMMARY

Necessary and sufficient conditions for the "decoupling" of an m-input, m-output time-invariant linear system using state variable feedback are determined. Given a system which satisfies these conditions, i.e., which can be decoupled by state variable feedback, the class $\Phi$ of all feedback matrices which decouple the system is characterized. The characterization of $\Phi$ is used to determine the number of closed loop poles which can be specified for the decoupled system and to develop a synthesis technique for the realization of desired closed loop pole configurations. Transfer matrix consequences of decoupling are examined and practical implications discussed through numerical examples.

1. INTRODUCTION

The development of techniques for the design of multivariable control systems is of considerable practical importance. A particular design approach involves the use of feedback to achieve closed loop control system stability. In conjunction with this approach, it is often of interest to know whether or not it is possible to have inputs control outputs independently, i.e., a single input influencing a single output. This is, in heuristic terms, the problem of decoupling.

* Division of Applied Mathematics, Brown University, Providence, Rhode Island, and NASA Electronics Research Center, Cambridge, Massachusetts.
The problem of decoupling a time-invariant linear system using state variable feedback and the relation of this problem to control system design have been discussed by several authors (refs. 1, 2, 3). Morgan (ref. 1) considered the question of decoupling for systems, the state equations of which had a somewhat special form. His chief result, which is a special case of the main theorem of this note, was the following: the time-invariant linear system

\[ \dot{x} = Ax + Bu, \quad y = Cx \]

can be decoupled if the matrix $CB$ is non-singular. Rekasius (ref. 2) extended Morgan's result and outlined an essentially trial-and-error procedure for specifying a certain number of the system's poles while decoupling the system. Neither Morgan nor Rekasius gave a clear proof of sufficiency, and they did not consider the question of necessity.

In this note, a necessary and sufficient condition for decoupling will be given; a characterization of the class of feedback matrices which decouple a system will be determined; the number of closed loop poles which can be specified while decoupling will also be determined; and a synthesis procedure for obtaining desired closed loop pole configurations will be developed. In line with these objectives, the remainder of the note is divided into the following sections:

II. Definitions
III. The Main Theorem
IV. The Class of Decoupling Matrices
V. A Synthesis Procedure
VI. Decoupling by Output Feedback
VII. A Practical Example
VIII. Concluding Remarks

In section II, precise definitions of decoupling and state variable feedback are given. Then the basic necessary and sufficient condition for decoupling is proved.
in section III. Using the main theorem, a description of all the decoupling matrices is presented (section IV). Next, the questions of synthesis and closed loop pole placement are examined (section V). In section VI, state variable feedback is replaced by output feedback and the relevant theory developed. The practical potential of the methods is indicated in the discussion of a VSTOL stability augmentation system in section VII. Finally, various concluding comments are made in section VIII.

SYMBOLS

\( n \) system order (integer \( \geq 1 \))
\( m \) dimension of input/output vector (integer \( 1 \leq m \leq n \))
\( x \) state vector \((n \times 1)\)
\( A \) system matrix \((n \times n)\)
\( B \) matrix multiplier of the control vector \((n \times m)\)
\( C \) matrix multiplier of the state vector \((m \times n)\)
\( u \) control vector \((m \times 1)\)
\( y \) output vector \((m \times 1)\)
\( F \) feedback matrix multiplier of the state \((m \times n)\)
\( G \) matrix multiplier of the external input \((m \times m)\)
\( \omega \) external input vector \((m \times 1)\)
\( C_i \) a vector \((1 \times n)\), the \(i\)-th row of \(C\)
\( 0 \) the zero (null) matrix
\( d_i \) a positive integer denoting the minimum \(j\) for which \(C_i A^j B \neq 0\)
\( y_i \) the \(i\)-th element of the output vector \(y\)
\( p_k(F) \) coefficients of the characteristic polynomial of \((A + BF)\)
\( \text{tr}(.) \) the trace of the matrix (\( \cdot \))
\( L_i(F, G) \) \((m \times n)\) matrix valued function of the external input \(\omega\)
\( B^* \) \((n \times m)\) matrix used to define decoupling

the decoupling matrix \((m \times m)\)
**II. DEFINITIONS**

Consider the time-invariant linear system

\[ \dot{x} = Ax + Bu, \quad y = Cx \]  

(1)

where \( x \) is an \( n \) vector called the state, \( u \) is an \( m \) vector called the control (or input), \( y \) is an \( m \) vector called the output, and \( A, B, C \) are \( n \times n, n \times m, \) and \( m \times n \) matrices, respectively. It is assumed that \( m \leq n \). If \( F \) is an \( m \times n \) matrix and \( G \) is a non-singular \( m \times m \) matrix, then the substitution of

\[ u = Fx + G\omega \]  

(2)

where \( \omega \) represents the new \( m \) vector control (Fig. 1), into (1) shall be called **linear state variable feedback**.
Figure 1. Multivariable Feedback System

Let \( d_1, d_2, \ldots, d_m \) be given by

\[
d_i = \min \{ j : C_i A^j B \neq 0, \ j = 0, 1, \ldots, n-1 \}, \ \text{or}
\]

\[
d_i = n-1 \text{ if } C_i A^j B = 0 \text{ for all } j
\]

where \( C_i \) denotes the \( i \)-th row of \( C \). Then, a simple calculation shows that

\[
C_i (A + BF)^k = C_i A^k, \quad k = 0, 1, \ldots, d_i
\]

\[
C_i (A + BF)^k = C_i A^{d_i-1} (A + BF), \quad k = d_i + 1, \ldots, n
\]

for \( i = 1, \ldots, m \). Application of the state variable feedback (2) and repeated differentiation together with (4) yield the relations

\[
y_i = C_i x = C_i (A + BF)^0 x
\]

\[
y_i = C_i Ax = C_i (A + BF)x
\]

\[
\vdots
\]

\[
y_i = C_i A^{d_i-1} x = C_i (A + BF)^{d_i} x
\]

\[
(y_1^{(d_i+1)}) \quad (C_i (A + BF)^{d_i+1} x + C_i (A + BF)^{d_i} B G \omega 
\]

\[
\vdots
\]

\[
y_1^{(n)} = C_i (A + BF)^{n} x + C_i (A + BF)^{n-1} B G \omega + \ldots + C_i (A + BF)^{d_i} B G \omega
\]
where $y_i$, $i = 1, \ldots, m$, is the $i$-th component of $y$. In view of the Cayley-Hamilton theorem,

$$
(A + BF)^n = \sum_{k=0}^{n-1} p_k(F) (A + BF)^k
$$

(6)

where the $p_k(F)$ are scalars depending upon $F$. Thus, $x$ can be eliminated from the final relation of (5) to give

$$
y^{(n)}_i - \sum_{k=0}^{n-1} p_k(F)y^{(k)}_i = \text{tr}(L^i\{F, G\} \Omega)
$$

(7)

where $\text{tr}(\cdot)$ denotes the trace of a matrix, $\Omega$ is the $m \times n$ matrix given by

$$
\Omega = \begin{bmatrix}
\omega & \omega^{(1)} & \ldots & \omega^{(n-1)} \\
\end{bmatrix}
$$

(8)

and $L^i\{F, G\}$ is the $n \times m$ matrix given by

$$
L^i\{F, G\} = \begin{bmatrix}
C_i \left[ (A+BF)^{n-1} - p_{n-1}(F)(A+BF)^{n-2} - \ldots - p_{d_i} + 1(F)(A+BF)^{d_i} \right] & \text{BG} \\
C_i \left[ (A+BF)^{n-2} - p_{n-1}(F)(A+BF)^{n-3} - \ldots - p_{d_i} + 2(F)(A+BF)^{d_i} \right] & \text{BG} \\
\vdots & \vdots \\
C_i \left[ (A+BF)^{d_i} \right] & \text{BG} \\
0 & \\
\end{bmatrix}
$$

(9)

where $0$ is a zero matrix consistent with the order of $L^i\{F, G\}$. If $E_{ij}$ denotes the $m \times m$ matrix with 1 as $ij$-th entry and zeros elsewhere, then $E_{ij} \Omega$ is an $m \times n$
matrix with $i$-th row identical to the $i$-th row of $Ω$ and with all other rows zero.
The matrix $E_{i\Omega}$ will be denoted by $Ω^i$. The following definition can now be made:

Definition. - The matrices $F$ and $G$, with $G$ nonsingular, decouple the system (1) if

$$y_i^{(n)} - \sum_{k=0}^{n-1} p_k(F)y_i^{(k)} = tr(L_i^{i,F,G}Ω) = tr(L_i^{i,F,G}Ω^i)$$

(10)

for $i = 1, \ldots, m$ and if

$$tr(L_i^{i,F,G}Ω) \neq 0$$

(11)

for $i = 1, \ldots, m$.

Note that this is a precise definition which does not involve vague statements about inputs controlling outputs independently.
III. THE MAIN THEOREM

With the definitions of section II, it is now possible to state and prove a theorem which gives a necessary and sufficient condition for decoupling.

Theorem. Let $B^*$ be the $m \times m$ matrix given by

$$
B^* = \begin{bmatrix}
  d_1 & A_1 & B \\
  C_1 & A & B \\
  & \vdots & \vdots \\
  C_m & A & mB \\
\end{bmatrix}
$$

Then there is a pair of matrices $F$ and $G$ which decouple the system (1) if and only if

$$
\det B^* \neq 0
$$

i.e., if and only if $B^*$ is nonsingular.

Proof: Suppose first that $B^*$ is nonsingular. Then it is claimed that the pair

$$
F^* = -B^{-1}A^*, \quad G^* = B^{-1}
$$

where

$$
A^* = \begin{bmatrix}
  d_1 + 1 \\
  C_1 A + 1 \\
  \vdots \\
  C_m A_{d + 1} \\
\end{bmatrix}
$$

decouples (1). In view of (4),

$$
C_i(A + BF^*)^j = C_i^j + C_i^j A^j BF^*
$$
But $C_i A \frac{d_i}{B}$ is simply the $i$-th row of $B^*$, and so it follows that

$$C_i A \frac{d_i}{B} F^* = -B_i B^*^{-1} A^* = -A_i = -C_i A \frac{d_i}{B} + 1$$

(17)

where $B_i$ and $A_i$ are the $i$-th rows of $B^*$ and $A^*$ respectively. Thus

$$C_i (A + BF^*)^k = 0$$

(18)

for any positive integer $k$. In a similar way, it follows that

$$C_i (A + BF^*) \frac{d_i}{B} B^*_1 B^*^{-1} = B_i B^*^{-1}$$

(19)

and hence that

$$I \{ F^*, G^* \}^{-1} = \begin{bmatrix}
-p_{d_1} + 1(F^*)B_i B^*^{-1} \\
-p_{d_1} + 2(F^*)B_i B^*^{-1} \\
\vdots \\
B_i B^*^{-1} \\
0
\end{bmatrix}$$

(20)

However, $B_i B^*^{-1} = \xi_i$, a row vector with 1 in the $i$-th place and zeros elsewhere, and so

$$\text{tr}(I \{ F^*, G^* \} \Omega) = -p_{d_1} + 1(F^*)\omega_1 -p_{d_1} + 2(F^*)\omega_1^{(1)} - \ldots + \omega_1^{(n-d_1-1)}$$

(21)

$$= \text{tr}(I \{ F^*, G^* \} \Omega) \neq 0$$

(22)

In other words, $F^*$ and $G^*$ decouple (1).
Now suppose that there is a pair of matrices $F, G$ which decouple (1). Then it follows from (4) that

$$C_i(A + BF)^{-1}BG = B_i^*G$$

for $i = 1, \ldots, m$. Since $C_i A^j B = 0$ for all $j$ would imply that $\text{tr}(L_i(F, G) G) = 0$ which would contradict the fact that $F$ and $G$ decouple (1), it is clear that $B_i^* \neq 0$ for $i = 1, \ldots, m$. As $G$ is nonsingular, $B_i^* G \neq 0$ for all $i$. Since (10) is satisfied, it follows that $B_i^* G$ is an $m$ row vector of the form $\alpha_i e_i$ with $\alpha_i \neq 0$ (otherwise there would be $\omega_j^{(k)}, j \neq i$ terms in $\text{tr}(L_i(F, G) G)$). Thus,

$$B_i^* G = \begin{bmatrix}
\alpha_1 \\
\alpha_2 \\
\vdots \\
\alpha_m
\end{bmatrix}$$

(24)

where $\alpha_i \neq 0$. Hence, $B_i^*$ is nonsingular since $G$ is.

The theorem just proved shows that $B_i^*$ is of paramount importance in the decoupling of (1) by state variable feedback. The basis for the choice of $F_i^*$ and $G_i^*$ in the proof of the theorem is the following observation: Since (5) implies that

$$y_1^{(d, 1)} + d_1 + \frac{1}{d_i + 1} x + C_i(A + BF)^{-1}BG\omega$$

(25)

which may also be written in the form

$$Y_i^* = (A_i^* + B_i^* F_i) x + B_i^* G_i\omega$$

(26)
where $\mathbf{y}^*$ is the $m$ vector with components $y_i^*$, it is clear that the choice $F = F^*$, $G = G^*$, leads to

$$\mathbf{y}^* = \omega$$

(27)

or, equivalently,

$$y_i^* = \omega_i$$

(28)

Caution: (28) does not represent the decoupled system since, in general, it involves the cancellation of zeros. The equations of the decoupled system are given by (10) or in state form as,

$$\dot{x} = (A + BF)x + BG\omega, \quad y = Cx$$

where $F, G$ are a decoupling pair.

It has now been established that the nonsingularity of $B^*$ is a necessary and sufficient condition for the existence of a pair of matrices $F, G$ which decouple (1). In the next section, the set of all pairs $F, G$ which decouple (1) will be characterized under the assumption that $B^*$ is nonsingular. This characterization leads to "answers" to the following two questions:

(a) The synthesis question; namely, how many closed loop poles can be specified for the decoupled system, how arbitrarily can they be specified, and how easily can an algorithm for specifying these poles be developed?

(b) The output feedback question; namely, when can feedback of the form $u = H\, y + G\, \omega$ decouple (1)?

**IV. THE CLASS OF DECOUPLING MATRICES**

Let $F$ be an $m \times n$ matrix, and let $G$ be a nonsingular $m \times m$ matrix. Under the assumption that (1) can be decoupled, necessary and sufficient conditions for
F, G to be a decoupling pair are determined in this section. These conditions turn out to be independent of G so that it will make sense to speak of the class Φ of matrices F which "decouple" (1).

Definition. Let \( Q^i(F) \) be the n x m matrix given by

\[
Q^i(F) = \begin{bmatrix}
C_i & (A + BF)^{n-1} & B \\
C_i & (A + BF)^{n-2} & B \\
& \vdots & \ddots & \ddots & \ddots \\
C_i & (A + BF) & B \\
& & & & 0
\end{bmatrix}
\]

for \( i = 1, \ldots, m \), where 0 is a zero matrix consistent with the order of \( Q^i(F) \).

Let \( P^i(F) \), for \( i = 1, \ldots, m \), be the n x n matrix given by

\[
P^i(F) = \begin{bmatrix}
1 & -p_{n-1}(F) & \cdots & -p_{d_i+1}(F) & | & 0 \\
0 & 1 & \cdots & -p_{d_i+2}(F) & | & 0 \\
& \vdots & \ddots & \ddots & \ddots & \vdots \\
0 & 0 & \cdots & 1 & | & 0 \\
& & & 0 & | & I
\end{bmatrix}
\]

where the \( p_k(F) \) are the coefficients of the characteristic polynomial of \( A + BF \), i.e.,:

\[
(A + BF)^n = \sum_{0}^{n-1} p_k(F)(A + BF)^k
\]

and I is an identity matrix consistent with the order of \( P^i(F) \).
Since \( P_i(F) \) is nonsingular, it follows that the rank of \( P_i(F)Q_i(F) \) is the same as the rank of \( Q_i(F) \). Note also that

\[
L_i\{F, G\} = P_i(F)Q_i(F)G
\]

where \( L_i\{F, G\} \) is defined by (9). Thus,

\[
\text{rank} \left[ L_i\{F, G\} \right] = \text{rank} \left[ Q_i(F) \right] \quad (33)
\]

for \( i = 1, \ldots, m \), since \( G \) is nonsingular. In view of the definition of decoupling, the following theorem can be established:

**Theorem.** - If the pair \( F, G \) decouples (1), then the rank of \( Q_i(F) \) is one for all \( i \); conversely, if the rank of \( Q_i(F) \) is one for all \( i \) and if \( B^* \) is nonsingular, then the pair \( F, B^{-1} \) decouples the system (1).

**Proof:** Suppose first that \( F, G \) decouples (1). Then

\[
\text{tr}(L_i\{F, G\} Q) = \text{tr}(L_i\{F, G\} Q^i) \neq 0 \quad (34)
\]

for all \( i \) where \( Q \) is the \( m \times n \) matrix given by

\[
Q = \begin{bmatrix}
\omega & \omega & \cdots & \omega^{(n-1)} \\
\omega & \omega & \cdots & \\
& \vdots & \ddots & \vdots \\
& & \vdots & \omega^{(n-1)}
\end{bmatrix}
\]

Since \( Q \) is arbitrary, the \( i \)-th column of \( L_i\{F, G\} \) is a non-zero vector, while every other column of \( L_i\{F, G\} \) is the zero vector. It follows that \( L_i\{F, G\} \) has rank one and hence, by (33), that \( \text{rank} \left[ Q_i(F) \right] = 1 \).
Now suppose that rank $Q_i^i(F) = 1$ for all $i$ and that $B^*$ is nonsingular.

Since

$$C_i (A + BF)^di \cdot B = C_i A^di \cdot B = B_i^* \neq 0$$

by the definition of $d_i$, where $B_i^*$ is the $i$-th row of $B^*$, it follows that

$$Q_i^i(F) = \begin{bmatrix}
\alpha_i B_i^* \\
\vdots \\
\alpha_i^i B_i^* \\
0
\end{bmatrix}$$

(37)

and hence that

$$Q_i^i(F) B^*^{-1} = \begin{bmatrix}
\alpha_i \\
\vdots \\
\alpha_i^i \\
1 \\
0
\end{bmatrix}$$

(38)

has only a non-zero $i$-th column. Thus,

$$\text{tr}(L_i^i \{ F, B^*^{-1}\ Omega \}) = \text{tr}(P^i(F)Q_i^i(F)B^*^{-1}\ Omega)$$

$$= \text{tr}(P^i(F)Q_i^i(F)B^*^{-1}\ Omega^i) \neq 0$$

(39)

and so the pair $F, B^*-1$ decouples (1).
Corollary. - If the pair $F, G$ decouples (1), then there is a diagonal matrix $\Lambda$ such that $G = \Lambda B^{-1}$

Proof. - If $F, G$ decouples (1), then $Q^i(F)$ is given by (37) and

$$Q^i(F)G = \begin{bmatrix}
    \lambda_i^1 & \cdots & 0 \\
    \lambda_i^2 & \cdots & 0 \\
    \cdots & \cdots & \cdots \\
    0 & \cdots & 0
\end{bmatrix}$$

where $\lambda_i^1 \neq 0$. It follows that $B^*G = \text{diag} \left[ \lambda^1, \ldots, \lambda^n \right]$, and the corollary is established.

Corollary. - If the pair $F, G$ decouples (1), then there is a diagonal matrix $\eta$ such that

$$FB = B^{-1} \{ \eta A^{**} - A^* \} B$$

where $A^{**}$ and $A^*$ are given by

$$A^{**} = \begin{bmatrix}
    d_1 \\
    C_1 A \\
    \vdots \\
    C_m A
\end{bmatrix}, \quad A^* = A^{**} A$$
Proof. - The corollary is an immediate consequence of the relations

\[
C_i(A + BF)^{d_i + 1} = C_i A^{d_i} B + C_i A^{d_i} BFB
\]

(43)

\[
C_i(A + BF)^{d_i + 1} = \gamma_i C_i(A + BF)^{d_i} B
\]

(44)

In summary, thus far it has been shown that the nonsingularity of $B^*$ is a necessary and sufficient condition for the existence of a decoupling pair $F, G$. Furthermore, the set of all pairs $F, G$ which decouple (1) consists of matrices $F$ such that rank $[Q_i(F)] = 1$ for all $i$ and $G$ such that $G = AB^*^{-1}$ where $A$ is diagonal and nonsingular. In order to clarify these points, an example will now be presented.

EXAMPLE: Let

\[
A = \begin{bmatrix}
1 & 1 & 0 \\
0 & 2 & 0 \\
0 & 1 & 3
\end{bmatrix}, \quad B = \begin{bmatrix}
-1 & 1 \\
0 & 0
\end{bmatrix}, \quad C = \begin{bmatrix}
1 & 0 & 0 \\
0 & 0 & 1
\end{bmatrix}
\]

(45)

Then

\[
B^* = \begin{bmatrix}
C_1 B \\
C_2 AB
\end{bmatrix} = \begin{bmatrix}
1 & 1 \\
-1 & 1
\end{bmatrix}
\]

(46)

Thus, $B^*$ is nonsingular, and the system can be decoupled. The set $\Phi$ of all $F$ which decouple the system (45) can now be obtained by determining all $2 \times 3$ matrices $F$. 

16
such that \( \text{rank} \left[ \mathbf{Q}^i(\mathbf{F}) \right] = 1 \). In this example, this implies that the elements of \( \Phi \) must be of the form:

\[
\mathbf{F} = \begin{bmatrix}
  f_{11} & f_{12} & f_{13} \\
  f_{11} & -f_{12}^{-1} & -f_{13}
\end{bmatrix}
\]  

(47)

V. A SYNTHESIS PROCEDURE

The theorem presented in section IV does provide a procedure for determining \( \Phi \), the class of all feedback matrices \( \mathbf{F} \) which decouple (1). However, the direct application of the condition, \( \text{rank} \left[ \mathbf{Q}^i(\mathbf{F}) \right] = 1 \) for all \( i \), results only in constraints being placed upon certain of the \( m \times n \) parameters of \( \mathbf{F} \). What is still required is a procedure for specifying closed loop system poles while simultaneously decoupling (1) using an appropriate \( \mathbf{F} \in \Phi \). In this light, a synthesis procedure will now be presented for directly obtaining a feedback matrix \( \mathbf{F} \in \Phi \), the parameters of which are determined so as to yield desired closed loop pole structure.

In particular, suppose that \( \mathbf{M}_k \), \( k = 0, 1, \delta \) are given \( m \times m \) matrices, then the choice

\[
\mathbf{F} = \mathbf{B}^{-1} \left[ \sum_{k=0}^\delta \mathbf{M}_k \mathbf{C}^k \mathbf{A}^{-1} \right], \quad \mathbf{G} = \mathbf{B}^{-1}
\]

(48)

will, by (26), lead to

\[
\mathbf{y}^* = \sum_{k=0}^\delta \mathbf{M}_k \mathbf{C}^k \mathbf{x} + \omega
\]

(49)
If \( \delta = \max d_i \) and the \( M_k \) are suitably chosen, i.e.,

\[
M_k = \text{diag.} \left[ m_1^k, m_2^k, \ldots, m_m^k \right] \quad \text{for} \quad i = 1, 2, \ldots, m,
\]

then (30) may be written in the form

\[
y^*_\delta = \sum_{k=0}^{\delta} M_k y^{(k)} + \omega \tag{50}
\]

or

\[
y_i^{(d_i+1)} = m_0 y_i + m_1 y_i^{(1)} + \ldots + m_i y_i^{(d_i)} + \omega_i \tag{51}
\]

for \( i = 1, 2, \ldots, m \), which indicates that \( F \) and \( G \) decouple (1) and that \( m + \sum_{i=1}^{m} d_i \) of the closed loop poles can be varied by varying the \( M_k \). In this light, consider the following example:

**EXAMPLE:** Let

\[
A = \begin{bmatrix}
0 & 1 & 0 & 0 & 0 & 0 \\
0 & 0 & 1 & 0 & 0 & 0 \\
0 & 0 & -1 & 0 & 0 & 0 \\
0 & 0 & 0 & 0 & 1 & 0 \\
0 & 0 & 0 & 0 & 0 & 1 \\
0 & 0 & 0 & 0 & 0 & 0
\end{bmatrix}, \quad B = \begin{bmatrix}
0 \\
1 \\
0 \\
0 \\
0 \\
0
\end{bmatrix}, \quad C' = \begin{bmatrix}
1 & 0 \\
0 & 0 \\
0 & 0 \\
0 & 0 \\
0 & 0
\end{bmatrix} \tag{52}
\]

Now

\[
B^* = \begin{bmatrix}
0 & 1 \\
1 & 0
\end{bmatrix}, \quad A^* = \begin{bmatrix}
0 & 1 & 0 & 0 & 0 & 0 \\
0 & 0 & 0 & 0 & 0 & 0
\end{bmatrix} \tag{53}
\]
Since $B^*$ is nonsingular, the system can be decoupled. Setting, for example,

$$M_0 = M_1 = M_2 = \begin{bmatrix} 0 & 0 \\ 0 & 1 \end{bmatrix}$$

(54)

one obtains, using (29), the decoupled system

$$\begin{align*}
y_1^{(6)} &= y_1^{(2)} + 2y_1^{(3)} + 2y_1^{(4)} - \omega_1^{(1)} - 2\omega_1^{(2)} - 2\omega_1^{(3)} + \omega_1^{(5)} \\
y_2^{(6)} &= y_2^{(2)} + 2y_2^{(3)} + 2y_2^{(4)} + \omega_2^{(2)} + \omega_2^{(3)}
\end{align*}$$

(55)

Note that in this case $\det(sI - A - BF) = s^2(s^3 - s^2 - s - 1)$, where the poles representing $s^3 - s^2 - s - 1$ have been specified by the choice of the $M_k$. Other choices of the $M_k$ would lead to other closed loop pole configurations. Therefore, if $B^*$ is nonsingular, $m + \sum_{i=1}^{m} d_i$ of the system's closed loop poles can be arbitrarily specified ($d_i + 1$ at a time) while simultaneously decoupling the system using the synthesis procedure. The synthesis question is, therefore, partially resolved, although some points still require clarification. In particular, it will be shown that $m + \sum_{i=1}^{m} d_i$ can never exceed $n$, the number of system poles, and that it is sometimes possible to specify more than $m + \sum_{i=1}^{m} d_i$ while simultaneously decoupling the system.

Lemma. Let $K$ be the \( \begin{bmatrix} C_1 & \vdots & d_1 \\ C_2 & \vdots & d_2 \\ \vdots & \vdots & \vdots \\ C_m & \vdots & d_m \end{bmatrix} \)xn matrix given by

$$K = \begin{bmatrix} C_1 \\ \vdots \\ C_1A \\ \vdots \\ C_mA \end{bmatrix}$$

(56)
Then $\text{rank} \begin{bmatrix} K \end{bmatrix} = m + \sum_{1}^{m} d_{i}$ and hence $m + \sum_{1}^{m} d_{i} \leq n$.

**Proof:** Let $k_{i}$ denote the $i$-th row of $K$ and $r_{i}$, arbitrary scalars such that

$$\sum_{1}^{\nu} r_{i} k_{i} = 0 \quad (57)$$

where

$$\nu = m + \sum_{1}^{m} d_{i} \quad (58)$$

In order to establish the lemma, one need only show that (57) implies that each $r_{i} = 0$. However, this follows directly from (57) by successive post multiplication by $B$, $AB$, ..., $A^\delta B$, and from the fact that $B^*$ is nonsingular.

Now let $p$ denote the number of closed loop poles which can be specified while decoupling, and let $f$ denote the number of free parameters (entries) in a decoupling matrix $F$ (for example, $f = 3$ in (47). Then the lemma and (51) combine to give

$$m + \sum_{1}^{m} d_{i} \leq p \leq n, \quad m + \sum_{1}^{m} d_{i} \leq f \quad (59)$$

Moreover, if $m + \sum_{1}^{m} d_{i} = n$, then all $n$ of the closed-loop poles can be arbitrarily positioned while simultaneously decoupling the system. Also, if $f = m + \sum_{1}^{m} d_{i}$, then (51) gives direct physical significance to the free parameters in $F$. If $f > m + \sum_{1}^{m} d_{i}$ (or $n$), then it may be possible to specify more than $m + \sum_{1}^{m} d_{i}$ of the closed loop poles. In this situation, it is often advantageous to calculate $C(sI - A - BF)^{-1} B B^*^{-1}$ with $f$ entries in $F$ remaining arbitrary. The following examples illustrate these ideas and some of the difficulties involved in their application.
EXAMPLE: Let
\[
A = \begin{bmatrix}
0 & 1 & 0 & 0 \\
0 & 0 & 1 & 0 \\
0 & 0 & 0 & 0 \\
0 & 0 & 0 & 1
\end{bmatrix}, \quad B = \begin{bmatrix}
0 & 0 \\
0 & 0 \\
1 & 0 \\
0 & 1
\end{bmatrix}, \quad C' = \begin{bmatrix}
1 & 0 \\
0 & 0 \\
0 & 1
\end{bmatrix}
\]

Then
\[
B^* = \begin{bmatrix}
C_1 A^2 B \\
C_2 B
\end{bmatrix} = \mathbf{I}
\]

and \( m + \sum_{i=1}^{m} d_i = 4 = n \). Thus, all the closed loop poles can be specified by using the synthesis procedure.

\[
F = \begin{bmatrix}
f_{11} & f_{12} & f_{13} & 0 \\
0 & 0 & 0 & f_{24}
\end{bmatrix}
\]

Moreover, application of the theorem (section IV) shows that (62) represents the most general form for a decoupling \( F \) so that \( f = 4 = n \). The general form of the decoupled transfer matrix is

\[
\begin{bmatrix}
\frac{(s-f_{24})}{(s-f_{24})(s^3-f_{13}s^2-f_{12}s-f_{11})} & 0 \\
0 & \frac{(s^3-f_{13}s^2-f_{12}s-f_{11})}{(s-f_{24})(s^3-f_{13}s^2-f_{12}s-f_{11})}
\end{bmatrix}
\]
EXAMPLE: Let

\[
A = \begin{bmatrix}
0 & 1 & 0 \\
2 & 3 & 0 \\
1 & 1 & 1
\end{bmatrix}, \quad B = \begin{bmatrix}
0 & 0 \\
1 & 0 \\
0 & 1
\end{bmatrix}, \quad C = \begin{bmatrix}
1 & 1 & 0 \\
0 & 0 & 1
\end{bmatrix}
\]

(64)

\[
B^* = \begin{bmatrix}
C_1 B \\
\vdots \\
C_2 B
\end{bmatrix} = I
\]

(65)

Thus, \(B^*\) is nonsingular, and the system (64) can be decoupled. It can be shown that the elements of \(\Phi\) must be such that

\[
F = \begin{bmatrix}
f_{11} & f_{12} & 0 \\
-1 & -1 & f_{23}
\end{bmatrix}
\]

(66)

so that \(f = n = 3 > 2 = m + \sum d_i\). Moreover, the closed-loop transfer matrix is given by

\[
C(sI - A - BF)^{-1}BB^*-1 = \begin{bmatrix}
(s + 1)(s - f_{23} - 1) & 0 \\
0 & s^2 - (f_{12} + 3)s - (f_{11} + 2)
\end{bmatrix}
\]

\[
\frac{(s - f_{23} - 1)(s^2 - (f_{12} + 3)s - (f_{11} + 2))}{(s - f_{23} - 1)(s^2 - (f_{12} + 3)s - (f_{11} + 2))}
\]

(67)

so that all of the closed loop poles can be specified. Note that application of the synthesis procedure in this case would allow one to specify only two of the three closed loop poles.
VI. DECOUPLING BY OUTPUT FEEDBACK

Since output feedback is only a special case of state variable feedback, i.e.:

\[ u = H y + G \omega \approx H C x + G \omega \] (68)

with HC replacing F, it follows immediately that (1) can be decoupled using output feedback if, and only if, (a) B* is nonsingular and (b) there is an m x m matrix H such that \( \text{rank} \left[ Q^i (HC) \right] = 1 \) for \( i = 1, \ldots, m \). These conditions provide a suitable test of whether or not a system can be decoupled using output feedback.

EXAMPLE: Let

\[
A = \begin{bmatrix} 1 & 1 & 0 \\ 0 & 2 & 0 \\ 0 & 1 & 3 \end{bmatrix}, \quad B = \begin{bmatrix} 1 \\ 1 \\ -1 \end{bmatrix}, \quad C = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & 1 \end{bmatrix}
\] (69)

Then

\[
B^* = \begin{bmatrix} 1 & 1 \\ -1 & 1 \end{bmatrix}
\] (70)

is nonsingular so that the system defined by (69) can be decoupled. However, it is not possible to decouple this system using output feedback. To see this, observe that the theorem and (39) imply that an F which decouples must be of the form

\[
F = \begin{bmatrix} f_{11} & f_{12} & f_{13} \\ f_{11} & -f_{12} & -f_{13} \end{bmatrix}
\] (71)
and that $HC$ must be of the form

$$ HC = \begin{bmatrix} h_{11} & 0 & h_{12} \\ h_{21} & 0 & h_{22} \end{bmatrix} \quad (72) $$

Equations (71) and (72) lead to the contradictory requirement that $f_{12} = 0$ and $f_{12} = -1$. This example illustrates the point that decoupling by state variable feedback need not imply decoupling by output feedback.

It should be noted that although a system may be decoupled using output feedback, some of the flexibility of specifying closed loop poles, as with state variable feedback, will in general be lost. For example, consider the system described by (60), with the most general $H$ given by

$$ H = \begin{bmatrix} h_{11} & 0 \\ 0 & h_{22} \end{bmatrix} \quad (73) $$

Since $\det(sI - A - BHC) = (s - 1 - h_{22})(s^3 - h_{11})$, output feedback will not be adequate to stabilize the system, although state variable feedback does provide a higher degree of flexibility (63).

EXAMPLE: Consider the system described by (64). It has been shown (67) that state variable feedback can be used to decouple the system while simultaneously specifying all three closed loop poles. Application of the theorem (section IV) and (39) imply that any 2 x 2 matrix $H$ of the form

$$ H = \begin{bmatrix} h_{11} & 0 \\ -1 & h_{22} \end{bmatrix} \quad (74) $$
will define an output feedback which decouples this system. From (74), it follows that
\[
\det(sI - A - BHC)(s - 1 - h_22)(s^2 - (h_{11} + 3)s - (h_{11} + 2))
\]
and hence that the system can be stabilized using output feedback (e.g., \( h_{22} = -1, h_{11} = -5 \)), although the poles are not completely arbitrary.

EXAMPLE: Let
\[
A = \begin{bmatrix}
1 & 1 & 0 \\
0 & 1 & 0 \\
0 & 0 & 1
\end{bmatrix}, \quad B = \begin{bmatrix}
0 & 1 \\
1 & 0 \\
1 & 0
\end{bmatrix}, \quad C' = \begin{bmatrix}
1 & 0 \\
1 & 1 \\
-1 & 0
\end{bmatrix}
\]

Then
\[
B^* = CB = \begin{bmatrix}
0 & 1 \\
1 & 0
\end{bmatrix}
\]

and \( m + \sum_{i=1}^{m} d_i = 2 < 3 = n \). It can be shown using the theorem that any decoupling \( F \) is of the form
\[
F = \begin{bmatrix}
0 & f_{12} & f_{13} \\
0 & F_{21} & f_{22} \\
f_{21} & f_{22} & -1 - f_{22}
\end{bmatrix}
\]

so that \( f = 4 \). However, the closed loop transfer matrix is given by
\[
\begin{bmatrix}
(s - 1)(s - f_{12} - f_{13} - 1) & 0 \\
0 & (s - 1)(s - f_{21} - 1) \\
(s - 1)(s - f_{21} - 1)(s - f_{12} - f_{13} - 1)
\end{bmatrix}
\]
so that \( p = 2; \) i.e., only two of the closed loop poles can be specified. It can also be shown for this example that output feedback leads to the transfer matrix

\[
\begin{bmatrix}
(s-1) \left( s - h_{12} - 1 \right) & 0 \\
0 & (s-1) \left( s - h_{21} - 1 \right)
\end{bmatrix}
\]

(79)

so that output and state variable feedback are equivalent. As previous examples illustrate, this is not true in general.

VII. A PRATICAL EXAMPLE

An area in which decoupling techniques may be of interest is the design of flight control and stability augmentation systems. Consider, for example, the following linearized longitudinal equations of motion for a lift-fan V/STOL vehicle* in a hovering condition (ref. 4).

\[
\begin{bmatrix}
\dot{u} \\
\dot{\theta} \\
\dot{\theta} \\
\dot{w} \\
\Delta \dot{x} \\
\Delta \dot{z}
\end{bmatrix} =
\begin{bmatrix}
X_u & X_\theta & 0 & 0 & 0 & 0 \\
0 & 0 & 1 & 0 & 0 & 0 \\
M_u & 0 & M_\theta & M_w & 0 & 0 \\
0 & 0 & Z_\theta & Z_w & 0 & 0 \\
1 & 0 & 0 & 0 & 0 & 0 \\
0 & 0 & 0 & 1 & 0 & 0
\end{bmatrix}
\begin{bmatrix}
X_{cv} & 0 & 0 \\
0 & 0 & 0 \\
M_{cv} & M_{\theta nf} & 0 \\
0 & Z_\theta & Z_{cs} \\
0 & 0 & 0 \\
0 & 0 & 0
\end{bmatrix}
\begin{bmatrix}
\delta_v \\
\delta_{nf} \\
\delta_s
\end{bmatrix}
\]

(80)

* similar to the XV-5A
where the quantities are:

- \( u \) - incremental longitudinal (x) velocity change
- \( \theta \) - incremental pitch angle
- \( \dot{\theta} \) - pitch rate
- \( w \) - incremental vertical (z) velocity change
- \( \Delta x \) - incremental position error
- \( \Delta z \) - incremental altitude error
- \( \delta_v \) - incremental collective fan input
- \( \delta_{nf} \) - incremental nose fan input
- \( \delta_s \) - incremental fan stagger input

The relevant outputs in this example are \( \theta, \Delta x, \) and \( \Delta z \), and the subscripted capitals (e.g., \( X_u \)) are the relevant stability derivatives.

The output matrix \( C \) is thus defined, i.e.:

\[
C = \begin{bmatrix}
0 & 1 & 0 & 0 & 0 & 0 \\
0 & 0 & 0 & 0 & 1 & 0 \\
0 & 0 & 0 & 0 & 0 & 1 \\
\end{bmatrix}
\]  

(81)

In this example

\[
B^* = \begin{bmatrix}
C_{1AB} \\
C_{2AB} \\
C_{3AB}
\end{bmatrix} = \begin{bmatrix}
M_{cv} & M_{\theta nf} & 0 \\
X_{cv} & 0 & 0 \\
0 & Z_{\theta} & Z_{cs}
\end{bmatrix}
\]

(82)
and is nonsingular since it is assumed that \( Z_{cs} M_{\theta} X_{cv} \neq 0 \). Therefore, \( m + \Sigma_{i=1}^{3} d_{i} = 6 \), and hence all six of the closed-loop poles can be arbitrarily specified while simultaneously decoupling this system. It can be shown using the theorem that a decoupling \( F \) has 6 (i.e., \( f = 6 \)) free parameters. Thus, the synthesis procedure (section V) can be directly applied to give physical significance to these free parameters. For example, suppose that independent pitch, translation, and altitude control are desired, i.e.:

\[
\begin{align*}
\ddot{\theta} &= m^0_1 \dot{\theta} + m^1_1 \dot{\theta} + \omega_1 \\
\Delta \dot{x} &= m^0_2 \Delta x + m^1_2 \Delta x + \omega_2 \\
\Delta \dot{z} &= m^0_3 \Delta z + m^1_3 \Delta z + \omega_3
\end{align*}
\]

(83)

According to the synthesis procedure, \( F \) can be set equal to \( B^{-1} \left[ \frac{1}{\Sigma_{k} M_{\theta} C A^{k} - A^*} \right] \).

It can be shown that for this decoupling \( F \)

\[
A + BF = \begin{bmatrix}
1 & m^0_2 & 0 & 0 & 0 & m^0_2 \\
0 & m^0_2 & 0 & 0 & 0 & 0 \\
0 & m^0_1 & m^1_1 & 0 & 0 & 0 \\
0 & 0 & 0 & m^1_3 & 0 & m^0_3 \\
1 & 0 & 0 & 0 & 0 & 0 \\
0 & 0 & 0 & 1 & 0 & 0 \\
\end{bmatrix}
\]

(84)
If $G$ is now set equal to $B^*-1$, the closed-loop transfer matrix is:

$$C(sI-A-BF)^{-1}BB^{-1} = \begin{bmatrix}
\left(s^2 - m_2^1 s - m_2^0\right)\left(s^2 - m_3^1 s - m_3^0\right), & 0, & 0 \\
0, & \left(s^2 - m_1^1 s - m_1^0\right)\left(s^2 - m_3^1 s - m_3^0\right), & 0 \\
0, & 0, & \left(s^2 - m_1^1 s - m_1^0\right)\left(s^2 - m_2^1 s - m_2^0\right)
\end{bmatrix} \quad (85)$$

If the $m_i$ are suitably chosen, then, in effect, the pilot will be faced with the task of controlling three highly stable second-order systems. This example serves only to indicate a potential practical area of application for the ideas presented in this paper.

The above examples illustrate the techniques developed for synthesizing decoupling controllers for multivariable systems.

VIII. CONCLUDING REMARKS

The problem of decoupling a time-invariant linear system using state variable feedback has been considered. Necessary and sufficient conditions for "decoupling" have been determined in terms of the nonsingularity of a matrix $B^*$. The class $\Phi$ of all feedback matrices which decouple a system has been characterized, and a synthesis technique for the realization of desired closed loop pole configurations has been developed. In essence, the major theoretical questions relating to decoupling via state variable feedback have been resolved for time-invariant linear systems.

A number of interesting potential areas of future research arise from the results obtained here. In particular, the question of extending the theory to the
time-varying situation is of considerable interest. Some preliminary results relating to stabilization have already been obtained. The design of aircraft and VSTOL stability augmentation systems via decoupling techniques is a potential practical area of application as was mentioned in section VII. Practical implementation of the techniques presented in this note has begun, but much remains to be done before the theory is transformed into a practical design technique.

REFERENCES


Electronics Research Center
National Aeronautics and Space Administration
Cambridge, Massachusetts, August 1967
125-19-01-01

"The aeronautical and space activities of the United States shall be conducted so as to contribute . . . to the expansion of human knowledge of phenomena in the atmosphere and space. The Administration shall provide for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof."

—NATIONAL AERONAUTICS AND SPACE ACT OF 1958

NASA SCIENTIFIC AND TECHNICAL PUBLICATIONS

TECHNICAL REPORTS: Scientific and technical information considered important, complete, and a lasting contribution to existing knowledge.

TECHNICAL NOTES: Information less broad in scope but nevertheless of importance as a contribution to existing knowledge.

TECHNICAL MEMORANDUMS: Information receiving limited distribution because of preliminary data, security classification, or other reasons.

CONTRACTOR REPORTS: Scientific and technical information generated under a NASA contract or grant and considered an important contribution to existing knowledge.

TECHNICAL TRANSLATIONS: Information published in a foreign language considered to merit NASA distribution in English.

SPECIAL PUBLICATIONS: Information derived from or of value to NASA activities. Publications include conference proceedings, monographs, data compilations, handbooks, sourcebooks, and special bibliographies.

TECHNOLOGY UTILIZATION PUBLICATIONS: Information on technology used by NASA that may be of particular interest in commercial and other non-aerospace applications. Publications include Tech Briefs, Technology Utilization Reports and Notes, and Technology Surveys.

Details on the availability of these publications may be obtained from:

SCIENTIFIC AND TECHNICAL INFORMATION DIVISION

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

Washington, D.C. 20546