FLAT CONDUCTOR CABLE APPLICATIONS

By W. Angele
Process Engineering Laboratory

March 1972

George C. Marshall Space Flight Center
Marshall Space Flight Center, Alabama
This report contains brief descriptions, supplemented with artwork, of some of the numerous applications of flat conductor cable (FCC) systems. Both Government and commercial uses have been considered, with described applications designated as either aerospace, military, or commercial.

The document is designed to illustrate the number and variety of ways in which FCC is being applied and considered for future designs.

EDITOR'S NOTE

Use of trade names or names of manufacturers in this report does not constitute an official endorsement of such products or manufacturers, either express or implied, by the National Aeronautics and Space Administration or any other agency of the United States government.
PREFACE

This document was designed and prepared under the direction and with the assistance of Mr. W. Angele of NASA. He was supported in this effort by Candace Swanson, Hayes International Corporation, in fulfillment of requirements specified in a Technical Directive issued by the Process Engineering Laboratory, NASA/MSFC.

In this effort to further the development of flat conductor cable technology by making facts available to the interested public, many known producers of FCC systems were mailed requests for assistance. Unfortunately, response was negligible; often a result of various company policies.

Special thanks must be extended to the many individuals within NASA and private industry who gave generously of their time and knowledge, providing the necessary data and artwork for this document. These people, and the organizations they represent, realize the gains to the field and to themselves which must ultimately result when the unrestricted dissemination of knowledge occurs.
# TABLE OF CONTENTS

<table>
<thead>
<tr>
<th>Section I: Introduction</th>
<th>Page</th>
</tr>
</thead>
<tbody>
<tr>
<td>Purpose</td>
<td>3</td>
</tr>
<tr>
<td>Definition of Terms</td>
<td>3</td>
</tr>
<tr>
<td>History</td>
<td>4</td>
</tr>
<tr>
<td>Report Format</td>
<td>4</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>Section II: Aerospace Applications</th>
<th>Page</th>
</tr>
</thead>
<tbody>
<tr>
<td>Advanced Action Manipulator System (ADAMS)</td>
<td>10</td>
</tr>
<tr>
<td>Agena Space Booster</td>
<td>12</td>
</tr>
<tr>
<td>Apollo Lunar Surface Experiments Package (ALSEP)</td>
<td>14</td>
</tr>
<tr>
<td>Apollo Telescope Mount (ATM)</td>
<td>16</td>
</tr>
<tr>
<td>Electronics Control Equipment for Neutral Buoyancy System</td>
<td>20</td>
</tr>
<tr>
<td>Experiment Package for Zero-G Studies</td>
<td>22</td>
</tr>
<tr>
<td>KC-135 Aircraft Experiment Platform</td>
<td>24</td>
</tr>
<tr>
<td>Lunar Portable Magnetometer (LPM)</td>
<td>26</td>
</tr>
<tr>
<td>Lunar Surveyor</td>
<td>28</td>
</tr>
<tr>
<td>Pegasus Meteoroid-Detection Satellite</td>
<td>30</td>
</tr>
<tr>
<td>Research Space Vehicle Program</td>
<td>32</td>
</tr>
<tr>
<td>Saturn V Optical Tracking System GSE</td>
<td>34</td>
</tr>
<tr>
<td>Saturn 201 Instrumentation Unit</td>
<td>36</td>
</tr>
<tr>
<td>Saturn IVB Study</td>
<td>38</td>
</tr>
<tr>
<td>Solar Array for Skylab Orbital Workshop</td>
<td>40</td>
</tr>
<tr>
<td>Space Station Power System</td>
<td>42</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>Section III: Military Applications</th>
<th>Page</th>
</tr>
</thead>
<tbody>
<tr>
<td>Crystographic Equipment</td>
<td>48</td>
</tr>
<tr>
<td>E2C Airplane</td>
<td>50</td>
</tr>
<tr>
<td>F-111 Airplane GSE</td>
<td>52</td>
</tr>
<tr>
<td>F-111 Airplane: Mark II Avionics</td>
<td>54</td>
</tr>
<tr>
<td>F-14 Airplane</td>
<td>56</td>
</tr>
<tr>
<td>MK 48 Torpedo Fire Control System (FCS)</td>
<td>58</td>
</tr>
<tr>
<td>Naval Shipboard Equipment</td>
<td>60</td>
</tr>
<tr>
<td>Pershing Missile</td>
<td>62</td>
</tr>
<tr>
<td>Poseidon Missile Flight Programmer</td>
<td>64</td>
</tr>
<tr>
<td>Poseidon Missile GSE</td>
<td>66</td>
</tr>
<tr>
<td>Poseidon Missile Tunnel Wiring</td>
<td>68</td>
</tr>
</tbody>
</table>
# TABLE OF CONTENTS (Concluded)

<table>
<thead>
<tr>
<th>Section</th>
<th>Page</th>
</tr>
</thead>
<tbody>
<tr>
<td>Standard Missile</td>
<td>69</td>
</tr>
<tr>
<td>Ships Inertial Navigation Systems (SINS)</td>
<td>70</td>
</tr>
<tr>
<td>Spartan Missile</td>
<td>72</td>
</tr>
<tr>
<td>Upstage Missile</td>
<td>74</td>
</tr>
<tr>
<td>Minuteman II and III Missiles</td>
<td>76</td>
</tr>
<tr>
<td>Mark 17 Reentry Vehicle, Minuteman III Missile</td>
<td>80</td>
</tr>
</tbody>
</table>

## SECTION IV. COMMERCIAL APPLICATIONS

- Boeing 747 Aircraft                                                  | 84   |
- Ceiling Lights                                                       | 86   |
- High Intensity Picture Light                                         | 88   |
- Microcircuit Prober                                                  | 90   |

## CONCLUSION

CONCLUSION                                                                 | 92   |

## APPENDIX. SOURCES OF ADDITIONAL INFORMATION

APPENDIX. SOURCES OF ADDITIONAL INFORMATION                                | 93   |

## BIBLIOGRAPHY OF SELECTED GOVERNMENT AND INDUSTRY DOCUMENTS

BIBLIOGRAPHY OF SELECTED GOVERNMENT AND INDUSTRY DOCUMENTS                | 95   |
SUMMARY

Flat conductor cable (FCC) has been selected or proposed for use in a wide variety of projects, aerospace and military fields of effort as well as industrial-commercial programs. In numerous cases; FCC has been able to provide definite advantages over conventional round wire cable. Several applications were such that no other cabling system could practically meet specifications.

In the body of this report, applications have been grouped according to type user, Government (aerospace or military) or commercial, a method quite useful for individual treatment of FCC uses. However, another way of obtaining an overall view of FCC applications is to consider a selected sampling according to type program requirements and advantages of use. This latter method has been chosen for the following summation.

Volume Reduction

As a result of its flat configuration and flexibility, FCC has a higher packaging density than conventional round wire cable. Because of this fact, FCC often is the only practical cable for a particular project. For example, the Apollo Telescope Mount's (ATM) gimbal system (see page 16) includes a cable arch which cannot house the required amount of round wire cable. In many complex shipboard instrumentation systems, such as the Ships Inertial Navigation System (SINS) (see page 70) and the MK 48 Torpedo Fire Control System (see page 58), space for required wiring is extremely limited. Again, FCC is uniquely able to meet specifications. Space is often a major consideration in airborne applications. The Navy's Standard and Poseidon missiles as well as E2C and F14 airplanes (described on pages 69, 64, 50 and 56) use FCC for some electronics equipment.

Weight Reduction

In aircraft as well as shipboard applications, weight is often a critical factor in cable selection. When McDonnell Douglas, under contract to NASA, replaced a 180-degree portion of a Saturn IVB aft skirt mockup cabling harness with flat cable (refer to page 38), they realized a weight savings of 76 percent. Boeing confirmed the weight savings advantages of FCC in a laboratory installation of the 747 airplane Auxiliary Power Unit (see page 84). Librascope uses FCC in numerous naval shipboard systems (noted on pages 58 and 60) largely because of resulting weight savings.
Cost Savings

In many instances, lower manufacturing costs and savings resulting from ease in installation of FCC systems has been a major basis for use. An assembly developed by ABT Associates and Non-Linear Systems (refer to page 86) can be simply installed for numerous low-voltage uses, such as home lighting, and result in considerable savings. Conversion to FCC for the Inertial Reference Unit in the Mark II avionics on the F-111 airplane (refer to page 54) actually saved 800 labor hours per assembly.

In the previously mentioned Saturn IVB aft skirt mockup fabricated in 1966 by McDonnell Douglas (see page 38), savings in materials were conservatively estimated at 33 percent. Using 1971 cable prices, material savings for this same mockup would be closer to 72 percent.

Movement

LINEAR MOTION

The configuration of FCC makes it extremely well suited to applications where two or more pieces of electrically interconnected hardware must move in a linear direction in relation to each other. Some of the FCC assemblies which have been employed to meet this requirement are discussed below.

Corrugated Cable

In the Hughes Lunar Surveyor spacecraft (refer to page 28), a cable permanently corrugated for elasticity was used which allowed an experiment package to be dropped onto the lunar surface. Numerous naval projects, such as the MK 48 Torpedo Fire Control System and the SINS (described on pages 58 and 70) have used permanently corrugated FCC to enable equipment drawers to be moved for servicing.

Another way of interconnecting moving drawer assemblies to a stationary console is through use of a temporary corrugated FCC configuration. In the closed drawer position, the cable forms corrugations in a minimum of space. When the drawer is extended, these straighten. Several consoles used in the operation of NASA/MSFC's Neutral Buoyancy Tank (see page 20) have this type FCC assembly for smooth drawer movement.

Reels

FCC can also be stowed on a reel and used to deploy equipment in an efficient and reliable manner. This type system has been used most successfully in the Apollo
Lunar Surface Experiments Package (refer to page 14) for deploying equipment packages up to 60 feet from the central data package, as well as the Lunar Portable Magnetometer (described on page 26). Military programs, like the Air Force GSE for the F-111 airplane (described on page 52) have also made use of FCC reel assemblies.

LIMITED ROTARY MOTION

Various configurations of FCC are used to maintain electrical connections between two pieces of electronic equipment which must rotate in relation to each other. The high flex life and low torque property of flat cable are definite assets for this type cable use.

Torque Compensating Loop

This configuration of interconnection is used in a gimbal system where electrical energy and signals must be conveyed at a minimum of torque. When applied to the ATM gimbal system (see page 16), the torque will be 90 percent less than that of a conventional round wire system, a considerable reduction.

Spring Coil

In NASA's Advanced Action Manipulator System (see page 10), a remotely operated electromechanical manipulator for performing work, flexible coils of FCC in the shoulder and elbow joints enable ±170 degrees rotation, as well as bending movement.

HINGE-LINE APPLICATIONS

FCC is ideally suited for use as a medium to convey electrical signals across a hinge line. Flat cable was used for this purpose in the wings of the highly successful Pegasus meteroid-detection satellite (refer to page 30) and has been proposed for similar aerospace applications. In a ground test unit for the Navy's Poseidon missile (see page 66), FCC assemblies provide door to cabinet and door to door connections.

Invisible Wiring

The flat, thin profile of FCC enables it to be applied to surfaces by an adhesive backing, painted or papered over, and thus provide almost invisible circuitry. Because of this feature, FCC has recently been selected for several low-voltage home wiring applications. The New York State Urban Development Corporation will use an FCC assembly, which includes a special switching device, for wiring to ceiling lights in an
experimental housing project (see page 86). This extremely promising innovation may well result in vast construction and installation cost reductions. Hobby Hill, Inc., of Chicago markets an FCC picture-light assembly (refer to page 88) which does not have the bulky, unsightly appearance of round wire cable. This system has numerous additional household uses.

Reliability

The reliability of FCC is attested to by the inclusion of FCC assemblies in numerous projects where severe operating conditions exist and high reliability is a critical requirement.

For example, a submarine environment, with its special conditions of shock, vibration, and temperature, makes its demands on a cabling system. Reliability was, thus, a major consideration when FCC was selected as an interconnecting medium in the Mark 48 Torpedo Fire Control System and in the SINS (described on pages 58 and 70). In addition, both applications presented severe mechanical and electrical problems because of the requirement that the circuitry must carry consistent electrical signals regardless of console drawer position.

Airborne applications also place severe operating conditions on cabling systems and demand high reliability. Vibration effects often associated with bulky, stiff, round wire cables can be sharply reduced through the use of FCC systems. Flat conductor jumper cables used in the Mark II avionics in the Air Force F-111 airplane (see page 54) are corrugated, and thus provide float for vibration isolation. In an experiment platform for use in a KC-135 research aircraft (refer to page 24), the thin, flat configuration of FCC allows it to be fastened firmly to the structure. This results in less chances of destructive vibration between cable and connector.
SECTION I. INTRODUCTION
PURPOSE

Flat conductor cable (FCC) systems have been used extensively in both Government and commercial designs. This document presents some of the ways in which FCC has been successfully used or proposed for use. No attempt has been made to cover even a majority of examples in this rapidly expanding field but merely to show representative designs as well as some of the more innovative ideas.

Most of the applications described in this document do not represent optimum uses of FCC. Because FCC systems are just beginning to be used in production items, the full advantages of FCC application have not yet been realized. In several of the described applications, FCC is used to replace round wire cable (RWC) in an existing system, and thus some of the RWC design limits are automatically imposed on the new matrix. In other examples, only limited use of FCC has been made, often the result of insufficient development time and money, or perhaps a desire to first use FCC on a trial basis. The reader will also notice that many different cable systems have been developed for similar uses, for example electronic drawer applications. This is a natural consequence of different designers looking for the best ways of applying a relatively new system.

Although the benefits to be gained through use of FCC are well known, greater application of flat cable systems will occur as potential users become more familiar with existing uses and comprehensive application studies. As the awareness of these FCC systems increases, so will the usage of FCC, and as greater quantities of FCC are used, production costs will inevitably decline. The end result will be a system which not only is more suitable than other wiring for many projects, but which can be obtained at costs competitive with or substantially lower than those of other systems.

It is hoped that this document will add to the knowledge of the reader, and thus further the growth of FCC application.

DEFINITION OF TERMS

Flat conductor cable consists of flat parallel conductors laminated between thin flexible plastic insulating films, or otherwise held in a rectangular flat configuration. For the purposes of this report, the terms "FCC" and "flat cable" are considered to be synonymous with "flat conductor cable." Flat ribbon cable, containing round wires and sometimes referred to as flat cable, is not within the scope of this document and is not implied by the term "flat cable."
HISTORY

Efforts by Government and private industry have brought FCC technology to a highly advanced state. Since the earliest investigative work, initiated in 1956 at the Guidance and Control Laboratory of the Army Ballistic Missile Agency (now part of the Marshall Space Flight Center), FCC systems have been continually refined. Hardware has been developed, systems subjected to stringent testing, and military specifications have been written for both cables and connecting hardware. Much of this work has been accomplished at NASA Marshall Space Flight Center under the direction of Mr. Wilhelm Angele, recognized throughout the aerospace industry as the leading designer of FCC systems. In more recent years, private industry has done much to advance flat cable technology: designing simple and effective terminations, developing competitive production methods and designs for cable harnesses, and investigating unique packaging concepts.

The benefits of using FCC in many interconnecting systems have long been recognized. In a growing number of instances, highly flexible FCC is more compatible than round wire with modern circuit designs, component miniaturization trends, and new packaging techniques. Certainly FCC should always be considered in electrical system design where weight, space, cost reduction, flexibility, reliability, and uniform electrical characteristics are of prime importance. In case after case, results of carefully designed tests and comparative studies have lent strong support to merits attributed to FCC systems.

Nonetheless, even with the abundance of supportive evidence, the relative newcomer to the field is not completely convinced of the utility of FCC. He wants to see examples, actual cases of successful utilization.

REPORT FORMAT

The primary purpose of this document is to present varied and successful applications of flat conductor cable systems and to illustrate the numerous ways in which FCC can be used to great advantage. These include Government and commercial projects in which FCC was selected either because it met a requirement which no other interconnecting system could meet or because the benefits of use outweighed merits of other systems. To give some ideas as to future possibilities, descriptions of proposed applications and major studies have also been included.

Subject matter has been divided into three main sections:

Section II. Aerospace Applications

Section III. Military Applications

Section IV. Commercial Applications
Each section contains brief descriptions of applied and proposed FCC uses as well as major application studies. These descriptions are supplemented with photos and illustrations and with references to additional sources of information.

At the end of this document, a bibliography of selected Government and industry documents relating to FCC systems has been included for further study.
SECTION II. AEROSPACE APPLICATIONS
SECTION II. AEROSPACE APPLICATIONS

Advanced Action Manipulator System (ADAMS)
Agena Space Booster
Apollo Lunar Surface Experiments Package (ALSEP)
Apollo Telescope Mount (ATM)
Electronics Control Equipment for Neutral Buoyancy System
Experiment Package for Zero-g Studies
KC-135 Aircraft Experiment Platform
Lunar Portable Magnetometer (LPM)
Lunar Surveyor
Pegasus Meteoroid-Detection Satellite
Research Space Vehicle Program
Saturn-V Optical Tracking System GSE
Saturn 201 Instrumentation Unit
Saturn IVB Study
Solar Array for Skylab Orbital Workshop
Space Station Power System
ADVANCED ACTION MANIPULATOR SYSTEM (ADAMS)

FCC is used for electrical circuits which cross the shoulder and elbow axes.

The ADAMS (Fig. 1) is one of a group of low cost, electromechanical manipulators which are under development for the performance of tasks such as handling, loading, transferring, stacking, and assembling in space and in other harsh environments. The basic system will be coordinated with various MSFC and other NASA organizations to comprise ultimate systems. FCC was selected to replace original round wire cables because of greater flexibility, reliability, and packaging density factors.

Two FCC flexible spring coil assemblies, provided by NASA/MSFC, are used in the manipulator. The center coil of one is located in the shoulder joint (Fig. 2), allowing ±170 degrees rotation and an elevation of +200 degrees from the vertical down position of the upper arm. The second coil is centered in the elbow joint, and again allows rotation of ±170 degrees. Elbow pitch of -60 degrees to +140 degrees is also possible. Each cable assembly consists of two layered cables, terminated at one end to a standard NASA circular 1/2-inch plug, and at the other end to an FCC to round wire transition. Plugs of both assemblies connect to a feed-through receptacle below the shoulder joint, thus enabling quick disconnect of the ADAMS from the base structure. The flat cable used is Kapton/Teflon insulated, 1/2 inch wide, and contains six 4 x 40 mil conductors on a 75-mil baseline center to center spacing.

For additional information, refer to the Appendix, item number 1.
Figure 1. Overall view of ADAMS.

Figure 2. ADAMS shoulder spring coil assembly.
Figure 4. Shielded and corrugated FCC assemblies (Digital Sensors, Inc., is now part of Ansley Electronics Corporation).
Figure 5. Model of ALSEP system.

Figure 6. Model of ALSEP central station (lower portion shown).
Figure 7. FCC routing and securing in the ATM roll adapter.
In the basic ATM FCC to round wire transition, two layers of unshielded 32-conductor signal cable are threaded through the back of a conductor spacer, the stripped conductors are folded into the spacer groove, and an insulator is inserted. Round wires (#20 AWG) are then laid on and soldered to the flat conductors, and the soldered assembly is potted. The standard conductor spacer (shown above) is used in all of the ATM's transitions, both shielded and unshielded, signal cable and power cable. The transition spacer is modified to accommodate 8 conductors for power FCC to round wire (#12, #16, or #20 AWG) transitions.
Observe that all of the drawer movement occurs in the FCC, thus allowing electrical components and panel round wire to remain in a fixed position.

Figure 12. Panel assembly in closed position.
Figure 13. Experiment package prepared for use in the KC-135 aircraft.
Figure 14. Flat conductor power cable terminations to bus bars.

Figure 15. FCC power circuitry in fuse boxes. (Note simple bracket wall fixtures and ease with which FCC is folded to form required angles.)
Figure 16. Portable magnetometer and stowage assemblies.

Figure 17. Apollo 14 portable magnetometer on the moon.
Figure 18. FCC harness on Hughes Lunar Surveyor V Spacecraft, flown in 1967.
Figure 19. Pegasus spacecraft.
Figure 20. FCC wire-wrap distributor (Digital Sensors, Inc., is now part of Ansley Electronics Corporation).
Figure 21. FCC interconnect reels on Saturn V GSE (closed position).
Figure 22. Flat cable on measuring rack. (Note simple cable-folding techniques.)

Figure 23. Flat cable connection to black box.
Figure 24. FCC versus RWC comparison, Saturn IVB aft skirt mockup.
Figure 25. Solar array 16-connector wing junction box.

Figure 26. Model of wing portion depicting typical cable routing during deployment.
Figure 27. Conceptual drawing of space station attached to power system by flat cable.
SECTION III. MILITARY APPLICATIONS
SECTION III. MILITARY APPLICATIONS

Crystographic Equipment
E2C Airplane
F-111 Airplane GSE
F-111 Airplane: Mark II Avionics
F-14 Airplane
MK 48 Torpedo Fire Control System (FCS)
Naval Shipboard Equipment
Pershing Missile
Poseidon Missile Flight Programmer
Poseidon Missile GSE
Poseidon Missile Tunnel Wiring
Standard Missile
Ships Inertial Navigation Systems (SINS)
Spartan Missile
Upstage Missile
Minuteman II and III Missiles
Mark 17 Reentry Vehicle, Minuteman III Missile
Figure 28. Flat conductor cable with plate and insert type connectors (Digital Sensors, Inc., is now part of Ansley Electronics Corporation).
Figure 29. Flat conductor jumper assembly for computer interconnectors (Digital Sensors, Inc., is now part of Ansley Electronics Corporation).
Figure 30. Self-retracting flat cable assembly
(Digital Sensors, Inc., is now part of Ansley Electronics Corporation).
Figure 31. Jumper cable assemblies (Digital Sensors, Inc., is now part of Ansley Electronics Corporation).

Figure 32. Inertial navigation assembly with flat conductor cable (Ansley Electronics Corporation).
Figure 33. Flat conductor jumper cable assemblies for electronic equipment interconnections (Digital Sensors, Inc., is now part of Ansley Electronics Corporation).
Figure 34. MK 48 Torpedo Fire Control System
(Librascope Division of the Singer Company).
Figure 35. Conceptual packaging drawing.

Figure 36. Mockup of a typical drawer in service position.

Figure 37. A completed 6-foot long 80-layer FCC harness 100 percent shielded.

Figure 38. A completed 9-foot long 180-layer FCC harness.
Figure 40. Installation of flat cable on vehicle gyro platform.

Figure 41. A flat cable installation.
Figure 42. Heavy duty FCC power-handling assembly
(Digital Sensors, Inc., is now part of Ansley Electronics Corporation).
Figure 43. Digital test and checkout equipment with FCC hinges (Ansley Electronics Corporation).
FCC interconnecting harnesses are used in the ordnance section.

General Dynamics used FCC harnesses (Fig. 44) in the ordnance section of the Navy's Standard Missile. Kapton/Teflon-insulated, shielded and unshielded cables were terminated through transition devices to Hughes Aircraft miniature center-lock screw connectors. Flat cable was used to reduce to a minimum the diametrical space required for the interconnecting harnesses.

For additional information, refer to the Appendix, item number 19.

Figure 44. FCC harness for the Standard Missile (General Dynamics).
Figure 45. SINS shielded corrugated FCC assembly
(Digital Sensors, Inc., is now part of Ansley Electronics Corporation).
Figure 46. Crimped flat conductor to round wire transition using Thomas and Betts copper molded splices.
Figure 47. 6-inch flat jumper cable in Upstage Guidance Command Unit (McDonnell Douglas).

Figure 48. Detail of a typical 6-inch jumper cable interconnection (McDonnell Douglas).
Figure 49. Gyro harness assembly (Digital Sensors, Inc., is now part of Ansley Electronics Corporation).
Figure 50. Miniature cable assembly (Ansley Electronics Corporation).
SECTION IV. COMMERCIAL APPLICATIONS
SECTION IV. COMMERCIAL APPLICATIONS

Boeing 747 Aircraft

Ceiling Lights

High Intensity Picture Light

Microcircuit Prober
Figure 51. Airplane power feeder system (The Boeing Company).

Figure 52. FCC feeder installation (The Boeing Company).
Figure 53. FCC-Switchpack System (ABT Associates and Non-Linear Systems).
ATTACH CONNECTOR TO WALL
Strip release paper from connector. Attach to wall back of picture. Strip release paper from CONDUCT-O-TAPE.

ATTACH TRANSFORMER TO BASEBOARD OR WALL
Strip off green release paper.* Press to wall — it will hold. Insert thin wire into bottom connectors.

ATTACH HANG-O-BRACKET TO FRAME

CONNECT WIRE TO TOP CONNECTOR
Hang up picture and picture light.

PLUG IVORY WIRE INTO OUTLET
Press switch on top of transformer. Press again to turn light off.

CONDUCT-O-TAPE can be painted or papered over.*

Figure 54. Installation of high intensity picture light assembly.
Figure 55. Microcircuit Prober Model MP-0200 with FCC.
APPENDIX

SOURCES OF ADDITIONAL INFORMATION

2. Ansley Electronics Corporation, 4100 N. Figueroa St., Los Angeles, California 90065.

3. Lockheed Missiles and Space Company, Sunnyvale, California.

4. Bendix Corporation, Aerospace Systems Division, 3300 Plymouth Road, Ann Arbor, Michigan 48107.


8. Hughes Aircraft, Centinela Ave. and Teak St., Culver City, California 90230.


12. Scope Inc., 2816 Fallfax Dr., Falls Church, Virginia 22042.


14. Convair Division of General Dynamics, Fort Worth, Texas.

15. Autonetics, Division of North American Rockwell, 3370 E. Miraloma Ave., Anaheim, California 92803.

SPECIFICATIONS AND STANDARDS


MIL-C-55543, Cable, Electrical, Flat Multiconductor, Flexible, Unshielded, November 15, 1968.


REPORTS


FLAT CONDUCTOR CABLE APPLICATIONS

By W. Angele

The information in this report has been reviewed for security classification. Review of any information concerning Department of Defense or Atomic Energy Commission programs has been made by the MSFC Security Classification Officer. This report, in its entirety, has been determined to be unclassified.

This document has also been reviewed and approved for technical accuracy.

M.P.L. SIEBEL
Director, Process Engineering Laboratory

\(5/30/72\)
DISTRIBUTION

INTERNAL

DIR
L. Wofford, Jr.
DEP-T
James Wiggins (6)
AD-S
PD-RV-MGR
Roy E. Godfrey
PD-RV-V
Hermon Might
PD-SA-DIR
W.G. Huber
PD-SS-MGR
W.A. Brooksbank
PM-HE-I
H. Thayer
PM-SAT-IU
Frederick Duerr
PM-SAT-S-II
William LaHatte (4)
PM/SE-ATM
E. Cagle
S&E-DIR
H.K. Weidner
S&E-ASTN-DIR
K. Heimburg
S&E-ASTN-EAS
E. Deuel (4)
S&E-ASTR-DIR
F. Moore
S&E-ASTR-CSE
H. Vick
S&E-ASTR-E
R. Aden
P. Youngblood
S&E-ASTR-EAA
J. Burson
S&E-ASTR-EB
E. Baggs
S&E-ASTR-EBC
W. Shockley (10)
B. McPeak
S&E-ASTR-EBF
J. Fuller
R. Milner
C. Crowell
S&E-ASTR-EP
C. Graff
S&E-ASTR-EPN
J. Miller
S&E-ASTR-G
C. Mandel
S&E-ASTR-GM
C. Cornelius
S&E-PE-M
W. Angele (100)
V. Caruso
W. Wilson
J. Williams
S&E-PE-EE
J. D Bennight
J. Hankins (100)
S&E-PE-MES
R. Herndon
J. Willis
J. Carden (100)
S&E-PE-MS
J. Splawn (1)
S&E-PE-MW
P. Parks
S&E-PE-MX
P. Schuerer
E. Van Orden
S&E-PE-P
W. Potter (1)
S&E-PE-RC
W. Vardaman
S&E-PE-T
W. Crumpton (1)
S&E-PE-TC
J. Lands (1)
S&E-QUAL-AP
J. Boehm
S&E-QUAL-AFC
J. Stroud
S&E-QUAL-ARM
R. V. Allen
S&E-QUAL-AR
H. Hoelzer
S&E-QUAL-ARM
W. Haeussermann
O. Hoberg
J. Mack
F. Weber
S&E-QUAL-AP
R. Currie
S.A. Johns
S&E-QUAL-H
F.C. Hammers
S&E-QUAL-I
L.L. McNair
S&E-QUAL-AP
J.O. Aberg
S&E-QUAL-QT
C.O. Brooks
S&E-QUAL-QP
J. Corder
S&E-QUAL-QPA
H. Kroeger
S&E-QUAL-QPA
M. Siebel (2)
S&E-QUAL-QT
F. Weckwarth (1)