STARTUP ANALYSIS
FOR A HIGH-TEMPERATURE
GAS-LOADED HEAT PIPE

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A model for the rapid startup of a high-temperature gas-loaded heat pipe is presented. A two-dimensional diffusion analysis is used to determine the rate of energy transport by the vapor between the hot and cold zones of the pipe. The vapor transport rate is then incorporated in a simple thermal model of the startup of a radiation-cooled heat pipe. Numerical results for an argon-lithium system show that radial diffusion to the cold wall can produce large vapor flow rates during a rapid startup. The results also show that startup is not initiated until the vapor pressure \( p_v \) in the hot zone reaches a precise value proportional to the initial gas pressure \( p_i \). Through proper choice of \( p_i \), startup can be delayed until \( p_v \) is large enough to support a heat-transfer rate sufficient to overcome a thermal load on the heat pipe.
A model for the rapid startup of a high-temperature gas-loaded heat pipe is presented. A two-dimensional diffusion analysis is used to determine the rate of energy transport by the vapor between the hot and cold zones of the pipe. The vapor transport rate is then incorporated in a simple thermal model of the startup of a radiation-cooled heat pipe. Numerical results for an argon-lithium system show that radial diffusion to the cold wall can produce large vapor flow rates during a rapid startup. The results also show that startup is not initiated until the vapor pressure $p_v$ in the hot zone reaches a precise value proportional to the initial gas pressure $p_i$. Through proper choice of $p_i$, startup can be delayed until $p_v$ is large enough to support a heat-transfer rate sufficient to overcome a thermal load on the heat pipe.

INTRODUCTION

A gas-loaded heat pipe contains a fixed quantity of a noncondensible gas in addition to its working fluid. The introduction of a gas into an ordinary heat pipe has been suggested as a means of aiding its startup. When a high-temperature heat pipe is closely coupled to a thermal sink, the very low initial vapor pressure may make it incapable of delivering the energy demanded during startup (ref. 1). Some technique must be found to decrease the coupling between the pipe and the sink if startup is to be successful.

During operation of a gas-loaded heat pipe, vapor flows from the evaporator to the condensor and the noncondensible gas is carried with it. The gas collects in the end of the condensor and decreases the effective heat-rejection area. As the temperature, and hence the vapor pressure, in the evaporator is increased, the gas is compressed to a smaller volume. If a sufficient quantity of gas is introduced, the heat pipe can be completely decoupled from the thermal sink until the vapor pressure is high enough to support large energy transport rates.
Before a gas-loaded heat pipe is incorporated in a larger system it is necessary to be able to predict its performance during startup. To this end a two-dimensional analysis of the diffusion of vapor through the gas in the condenser is performed in the present work. The energy transport rate obtained in this study is combined with a simple thermal model of the startup of a radiation-cooled heat pipe. Numerical results are presented for a long lithium heat pipe with argon as the noncondensible gas. They should be of value in the design of experiments with high temperature gas-loaded heat pipes.

SYMBOLS

\[ A_i \] expansion coefficient, eq. (18)
\[ A_r \] parameter, eq. (31)
\[ A_v \] parameter, eq. (32)
\[ a_i \] expansion coefficient, eq. (23)
\[ B_i \] expansion coefficient, eq. (19)
\[ b_i \] expansion coefficient, eq. (24)
\[ C \] heat capacity per unit length
\[ c \] total molar density
\[ c_i \] initial molar concentration of noncondensible gas
\[ D \] outside diameter of heat pipe
\[ D_\theta \] parameter, eq. (36)
\[ D_l \] parameter, eq. (37)
\[ \phi \] binary diffusion coefficient
\[ E_{-1} \] exponential integral
\[ F_c(g) \] function, eq. (16)
\[ F_h(g) \] function, eq. (17)
\[ g \] dimensionless vapor flux
\[ h_{fg} \] heat of vaporization per unit mass
\[ J_0 \] Bessel function
\[ L \] length of hot zone
\[ L_e \] length of evaporator
\( L_p \) length of heat pipe
\( l \) dimensionless \( L \)
\( l_e \) dimensionless \( L_e \)
\( l_p \) dimensionless \( L_p \)
\( M_v \) molecular weight of vapor
\( N_o \) vapor flux, eq. (3)
\( N_v \) molar vapor flux
\( p_i \) initial pressure of noncondensible gas
\( p_v(T) \) vapor pressure
\( Q \) heat input
\( Q_T \) heat lost by radiation
\( Q_v \) energy transported by vapor
\( q \) dimensionless \( Q \)
\( q_r \) dimensionless \( Q_T \)
\( q_v \) dimensionless \( Q_v \)
\( R \) radius
\( R_w \) inside radius of wick
\( r \) dimensionless \( R \)
\( T \) temperature of hot zone
\( T_c \) temperature of cold zone
\( t \) time
\( t_i \) time at which \( Q \) is switched on
\( x_g \) mole fraction of noncondensible gas
\( x_v \) mole fraction of vapor
\( y \) concentration variable, eq. (9)
\( Z \) distance along pipe
\( z \) dimensionless \( Z \)
\( \alpha_i \) zeros of \( J_0^' \)
\( \beta_i \) zeros of \( J_0 \)
\( \gamma \) Euler constant
DESCRIPTION OF THE MODEL AND ASSUMPTIONS

Thermal Model

The thermal model has already been applied to the startup of an ordinary radiation-cooled heat pipe (refs. 2 and 3). When the power is turned on, a hot zone of uniform wall temperature $T$ forms in the evaporator. Once the vapor pressure $p_v(T)$ is high enough, the front edge of this zone begins to move into the condensor. Despite the steep wall-temperature gradient at the front, it is assumed that axial heat conduction in the wall is negligible compared to energy transport by vapor diffusion. The remaining cold zone retains its initial temperature $T_c$.

This model is in definite opposition to the results of the steady-state analysis of reference 4. There it is shown that at the front axial heat conduction in the wall is much greater than energy transport by the diffusing vapor. The present calculations confirm that the vapor flow rate past the front becomes quite small as steady state is approached. They also show, however, that very large diffusion rates accompany the motion of the temperature front during startup. Such rates are incompatible with a stationary front.

Diffusion Model

The goal of the diffusion analysis is to determine the vapor flow rate past the temperature front as a function of the length and temperature of the hot zone. To simplify
the calculations the local condensation rate upstream of the front is assumed to be negligible, whereas the cold wall is taken to act as a perfect sink for the vapor. The transition region between these two boundary conditions is considered to be vanishingly thin. The nonuniform vapor velocity in the hot zone must produce a secondary flow of noncondensible gas. This effect is neglected, however; the upstream velocity is assumed uniform over the cross section, and the gas is taken to be stagnant throughout the pipe. The axial pressure drop is neglected, and the total pressure, gas plus vapor, is set equal to the vapor pressure $p_v(T)$ in the evaporator. This restricts the analysis to cases where the vapor velocity is much less than the speed of sound during startup. In a long cylindrical heat pipe the vapor concentration should relax to a quasi-steady profile in a time of order $(R_w/2.4)^2/\mathcal{D}$, where $R_w$ is the inside radius of the wick and $\mathcal{D}$ is the binary diffusion coefficient for the vapor-gas mixture. If the distance moved by the temperature front during this time is small, then steady-flow equations can be used in the diffusion analysis. In an argon-lithium heat pipe at 1300 K, with $R_w = 1$ centimeter, the relaxation time is about 10 milliseconds. For a typical case (fig. 3) the maximum frontal motion during this time is 0.02 centimeter which is small. Finally, the product $c\mathcal{D}$, where $c$ is the total molar density, is assumed constant and evaluated at the average of the temperatures $T$ and $T_c$.

DIFFUSION ANALYSIS

The molar flux of vapor $N_v$ through a stagnant gas is given by (ref. 5)

$$N_v = x_v N_v - c\mathcal{D} \nabla x_v = c\mathcal{D} \nabla \ln x_g$$

(1)

where $x_v$ and $x_g$ are the mole fractions of vapor and gas, respectively. Conservation of mass requires

$$\nabla^2 \ln x_g = 0$$

(2)

A cylindrical coordinate system is introduced with the main flow in the $Z$ direction and the origin at the interface between the hot and cold zones (fig. 1). It is assumed that the axial distance over which $x_v$ departs significantly from 0 or 1 is small compared to $L$ or $L_p - L$ where $L$ is the length of the hot zone and $L_p$ is the length of the heat pipe. Hence, the boundaries on equation (2) are taken at $Z = \pm \infty$. The boundary conditions for equation (2) are as follows:
Figure 1 - Coordinate system for diffusion analysis with main flow in Z-direction and origin at interface between hot and cold zones.

For \( Z < 0 \):

\[
\frac{c \partial}{\partial Z} \ln x_g - N_0 \quad \text{as } Z \to -\infty
\]  

\[
\frac{\partial}{\partial R} \ln x_g = 0 \quad \text{at } R = R_w
\]  

For \( Z > 0 \):

\[
x_g - 1 \quad \text{as } Z \to -\infty
\]

\[
x_g = 1 \quad \text{at } R = R_w
\]

where \( N_0 \) is an unknown constant flux. Conditions (4) and (6) represent no condensation and the perfect sink, respectively. The unknown \( N_0 \) is determined by specifying the initial molar concentration \( c_1 \) of the noncondensible gas at temperature \( T_c \). A mass balance on the gas gives

\[
c_1 L_p = \int_{-L}^{L} \frac{c x_g}{c x_g} dZ \approx \int_{-\infty}^{0} \frac{c x_g}{c x_g} dZ + \int_{0}^{\infty} \frac{c (x_g - 1)}{c} dZ + \int_{0}^{L_p - L} c dZ
\]

\[
\approx c_h \int_{-\infty}^{0} \frac{x_g}{x_g} dZ - c_c \int_{0}^{\infty} (1 - x_g) dZ + c_c (L_p - L)
\]

where the bar indicates an average over the cross section, and \( c_h \) and \( c_c \) are constants obtained by evaluating \( c \) at the temperatures \( T \) and \( T_c \), respectively. Note that the integrands in equation (7), as finally written, go rapidly to zero away from \( Z = 0 \).
Dimensionless variables are introduced:

\[ r = \frac{R}{R_w}, \quad z = \frac{Z}{R_w}, \quad g = \frac{N_o R_w}{c \Omega} \]

\[ l = \frac{L}{R_w}, \quad l_p = \frac{L_p}{R_w}, \quad \theta = \frac{T}{T_c} \]

and

\[ \lambda(\theta, l) = \frac{c_c(L_p - L) - c_i L_p}{c_c R_w} = l_p - l - \frac{l_p p_i}{p_v(T)} \]  \( (8) \)

where \( p_i \) is the initial gas pressure. A new dependent variable \( y \) is defined by

\[ \ln x_g = -gy \]  \( (9) \)

In cylindrical coordinates, equation (2) becomes

\[ \frac{1}{r} \frac{\partial}{\partial r} \left( r \frac{\partial y}{\partial r} \right) + \frac{\partial^2 y}{\partial z^2} = 0 \]  \( (10) \)

Equations (3) to (6) become the following:

For \( z < 0 \):

\[ \frac{\partial y}{\partial z} = -1 \quad \text{as } z \to -\infty \]  \( (11) \)

\[ \frac{\partial y}{\partial r} = 0 \quad \text{at } r = 1 \]  \( (12) \)

For \( z > 0 \):

\[ y = 0 \quad \text{as } z \to -\infty \]  \( (13) \)

\[ y = 0 \quad \text{at } r = 1 \]  \( (14) \)
Equation (7) becomes

$$F_c(g) - \theta^{-1} F_h(g) = \lambda(\theta, I) \quad (15)$$

with

$$F_c(g) = 2 \int_0^1 r \, dr \int_0^\infty (1 - e^{-gy}) \, dz \quad (16)$$

$$F_h(g) = 2 \int_0^1 r \, dr \int_0^\infty e^{-gy} \, dz \quad (17)$$

Once \( y(r, z) \) is determined, equation (15) can be solved numerically to obtain \( g \), and hence the mass flux as a function of \( T \) and \( I \).

The solution of equations (10) to (14) is given by the following:

\( z < 0 \):

$$y = -z + \sum_{i=1}^{\infty} A_i e^{\alpha_i z} J_o(\alpha_i r), \quad J_o'(\alpha_i) = 0 \quad (18)$$

\( z > 0 \):

$$y = \sum_{i=1}^{\infty} B_i e^{-\beta_i z} J_o(\beta_i r), \quad J_o'(\beta_i) = 0 \quad (19)$$

Note that \( \alpha_1 = 0 \) and \( J_o(0) = 1 \). The Bessel functions \( J_o(\alpha_i r) \) and \( J_o(\beta_i r) \) satisfy the following relations (ref. 6, ch. 11):

$$2 \int_0^1 r J_o(\alpha_i r) J_o(\alpha_j r) \, dr = \delta_{ij} \left[ J_o(\alpha_i) \right]^2 \quad (20)$$

$$2 \int_0^1 r J_o(\beta_i r) J_o(\beta_j r) \, dr = \delta_{ij} \left[ J_o'(\beta_i) \right]^2 \quad (21)$$

$$2 \int_0^1 r J_o(\alpha_i r) J_o(\beta_j r) \, dr = 2\beta_j \left( \alpha_i^2 - \beta_j^2 \right)^{-1} J_o(\alpha_i) J_o'(\beta_j) \quad (22)$$
where $\delta_{ii} = 1$ and $\delta_{ij} = 0$ for $i \neq j$. Requiring $y$ and $\partial y/\partial z$ to be continuous at $z = 0$ and applying relations (18) to (20) to the results give

$$a_j = -2 \sum_{i=1}^{\infty} b_i \beta_i \left( \alpha_i^2 - \beta_i^2 \right)^{-1}, \quad j = 1, \ldots, \infty$$  \hspace{1cm} (23)

$$b_j = 2 \sum_{i=1}^{\infty} a_i \alpha_i \left( \alpha_i^2 - \beta_i^2 \right)^{-1} + 2\beta_j^{-2}, \quad j = 1, \ldots, \infty$$  \hspace{1cm} (24)

where $a_i = A_i J_0(\alpha_i)$ and $b_i = -B_i J_0'(\beta_i)$. A finite subset of equations (23) and (24) with $j = 1, \ldots, n$ can be solved by iteration. Final values are obtained by increasing $n$ until the low order coefficients no longer change. For $n = 30$ the first few coefficients are

$$a_1 = 0.2537 \quad b_1 = 0.2578$$
$$a_2 = -0.0761 \quad b_2 = 0.0638$$
$$a_3 = -0.0336 \quad b_3 = 0.0308$$
$$a_4 = -0.0201 \quad b_4 = 0.0188$$
$$a_5 = -0.0138 \quad b_5 = 0.0129$$

The remaining coefficients decrease quite slowly, but the exponential factors in equations (16) and (17) ensure adequate convergence away from $z = 0$.

Once $g$ has been found by solving equation (15), the energy transport by diffusion from the hot to the cold zone $Q_v$ is given by

$$Q_v = \pi R_w^2 M_v N_o h_{fg} = \pi R_w M_v c\varphi h_{fg}$$  \hspace{1cm} (25)

where $M_v$ is the molecular weight of the vapor and $h_{fg}$ is its heat of vaporization per unit mass. For $g$ of order one this corresponds to a very small energy transport. As an example, in an argon-lithium heat pipe at 1300 K with $R_w = 1$ centimeter and $g = 1$, equation (25) gives $Q_v \approx 7.4$ watts. For large transport rates to occur it is therefore necessary to have $g \gg 1$. From equations (9) and (19) this corresponds to the bulk of the gas being pushed far enough into the cold zone to enable the vapor to reach the cold surface by diffusing radially through a very thin layer of gas.

The functions $F_c(g)$ and $F_h(g)$ must be evaluated numerically. The results have been approximated by analytical expressions for subsequent use. For $g \gg 1$, however, $F_c$ is dominant and the following approximation is of interest. If only the first term in
the series of equation (19) is retained, equation (16) gives

\[ F_c(g) \approx 2 \int_0^1 r \, dr \int_0^\infty \left\{ 1 - \exp \left[ -gB_1J_0(\beta_1 r)e^{-\beta_1 z} \right]\right\} dz \]

\[ \approx 2 \int_0^1 r \, dr \beta_1^{-1} \int_0^\infty gB_1J_0(\beta_1 r) \left( 1 - e^{-t} \right) \frac{dt}{t} \]

\[ \approx 2 \int_0^1 r \, dr \beta_1^{-1} \left\{ \gamma + \ln [gB_1J_0(\beta_1 r)] + E_1 [gB_1J_0(\beta_1 r)] \right\} \]

\[ \approx \beta_1^{-1} (\ln g - 1.339) \]  

(26)

where \( \gamma = 0.5772 \). The radial integral was performed numerically. Note that the exponential integral (ref. 6, ch. 3) \( E_1(x) \) approaches zero for large \( x \). Hence, \( F_c(g) \) has only a weak dependence on \( g \) for \( g \gg 1 \).

**TIME-DEPENDENT THERMAL ANALYSIS**

The time-dependent equations are written for a radiation-cooled heat pipe with a shielded evaporator. The state of the heat pipe at time \( t \) is shown in figure 2, where \( Q \) is the heat input to the evaporator, \( Q_r \) the heat lost by radiation, and \( Q_v \) the energy transported by the vapor to the cold zone.

When the hot zone is expanding into the condensor, energy balances on control vol-

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Figure 2. - State of heat pipe at time t.
umes enclosing the hot and cold zones, respectively, at time $t$ give

$$CL \frac{dT}{dt} = Q - Q_r - Q_v \quad (27)$$

$$C(T - T_c) \frac{dL}{dt} = Q_v \quad (28)$$

where $C$ is the heat capacity per unit length of the wall, wick, and liquid. In terms of $\theta$ and $l$ these equations become

$$l \frac{d\theta}{dt} = q_r - q_v \quad (29)$$

$$\left(\theta - 1\right) \frac{dl}{dt} = q_v \quad (30)$$

with $q = Q/CR_wT_c$. The quantities $q_r$ and $q_v$ are given by

$$q_r = A_r \theta^4 (l - l_e) \quad (31)$$

$$q_v = A_v g \quad (32)$$

where $A_r = \pi D \epsilon \sigma T_c^2 / C$, $l_e = L_e / R_w$, and $A_v = \pi M_v (c \varrho) h_{fg} / CT_c$. Also, $D$ is the outside diameter of the pipe, $\epsilon$ the emissivity, and $\sigma$ the Stefan-Boltzmann constant. The formulation is completed by equation (15) with $\lambda(\theta, l)$ given by equation (8).

During the initial startup period, $q_v$ remains negligibly small until $p_v(T) > p_i$ and $\lambda > 0$. For $\lambda > 0$, there is an extremely rapid transient in which $q_v$ approaches $q$ and $d\theta/dt$ becomes small. Thereafter, both $l$ and $\theta$ increase at moderate rates, and $q_v$ decreases steadily as $q_r$ increases. This continues until steady state is approached. The behavior is illustrated in figure 3 for an argon-lithium heat pipe.

In this latter period, equation (15) causes a severe numerical difficulty. The function $\lambda(\theta, l)$ has a very strong dependence on $\theta$ and $l$, while $F_c(g)$ is a weak function of $g$ for $g >> 1$. Thus, $q_v$ is extremely sensitive to small changes in $\theta$ and $l$, which makes equations (29) and (30) numerically unstable for any reasonable size integration step ($>0.1$ sec).

Since $l(d\theta/dt)$ is small compared to $(q - q_r)$ in this period, equation (29) can be used to obtain an approximate value of $q_v$ and hence $g$. Equations (30) and (15) can then be solved simultaneously for $l$ and $\theta$. It is necessary, however, to first obtain a good ap-
Figure 3. - Startup behavior for argon-lithium heat pipe with constant heat input. Length of heat pipe, $L_p$, 300 centimeters; length of evaporator, $L_e$, 30 centimeters; heat input, $Q$, 5 kilowatts.
proximation to $\frac{d\theta}{dt}$. Differentiating equations (15) and (8) gives

$$
\left[ \lambda_e - \theta^{-2} F_h(g) \right] \frac{d\theta}{dt} - \frac{dl}{dt} = \left[ F'_c(g) - \theta^{-1} F'_h(g) \right] \frac{dg}{dt}
$$

(33)

with $\lambda = T_c \theta^2 p_1 p_v^2 / p_v$. Equation (29), with $l(d\theta/dt)$ assumed small and $q$ taken as constant, gives

$$
\frac{dg}{dt} \approx - \frac{A_r}{A_v} \left[ 4 \theta^3 (l - l_e) \frac{d\theta}{dt} + \theta^4 \frac{dl}{dt} \right]
$$

(34)

Equations (33) and (34) give

$$
D_\theta \frac{d\theta}{dt} - D_l \frac{dl}{dt} \approx 0
$$

(35)

with

$$
D_\theta = \lambda + \theta^{-2} F_h + \left( F'_c - \theta^{-1} F'_h \right) \frac{A_r}{A_v} \cdot 4 \theta^3 (l - l_e)
$$

(36)
Equations (29), (30), and (35) give

\[ D_L = 1 - \left( F_C - \theta^{-1} F_h \right) A_r \theta^{-4} \]  

(37)

Finally, equations (29) and (38) give

\[ \left[ l + (\theta - 1) \frac{D_\theta}{D_L} \right] \frac{d\theta}{dt} \approx q - q_r \]  

(38)

Equations (29), and (35) give

\[ q_v \approx (q - q_r) \frac{(\theta - 1)D_\theta}{(\theta - 1)D_\theta + lD_L} \]  

(39)

Equations (30), (15), and (39) form a system of one differential and two algebraic equations whose solution approximates that of equations (29), (30), and (15) when \( d\theta/dt \) is small. The stability of this approximate set of equations is much improved over that of the original set.

Both sets of equations have been incorporated in a numerical integration scheme. The initial conditions at \( t = 0 \) are taken to be \( l_o = l_e \) and \( \lambda(\theta_o, l_o) = 0 \). From equation (8) this gives

\[ p_v(T_o) = \frac{p_1 l_p}{(l_p - l_e)} \]  

(40)

At earlier times \( q_r = 0, q_v \approx 0, \) and \( \theta = 1 + q(t - t_1)/l_e \) where \( t_1 \) is the time when \( Q \) is switched on. The integration is started using the original set of equations and a very small integration step. When \( q_v \) is close to its maximum value, the integration is continued using the approximate equations. The numerical routine automatically varies the size of the integration step to achieve the maximum value consistent with accuracy and stability. In a typical case the stepsize varied from \( 10^{-4} \) second at the start to 10 seconds at the end of the integration. In a few cases the approximation was checked by using the original set of equations throughout the calculation. The agreement between the two methods was better than three significant figures in \( l \) and \( \theta \). Solutions have also been obtained for a startup in which \( Q \) is increased in steps until the final steady state is reached.
NUMERICAL RESULTS

The time-dependent theory of the preceding section has been applied to the startup of an argon-lithium heat pipe. The specifications of the pipe are as follows: \( L_p = 300 \) centimeters, \( L_e = 30 \) centimeters, \( D = 1.9 \) centimeters, \( R_w = 0.78 \) centimeter, \( C = 1.99 \) joules per centimeter per degrees Kelvin, and \( \epsilon = 0.3 \). The values for \( R_w, C, \) and \( \epsilon \) were obtained by selecting a 1.0 millimeter thick tantalum alloy wall, an 0.5-millimeter-thick annular liquid return, and an 0.2-millimeter-thick tungsten screen wick with 50-percent porosity. The binary diffusion coefficient was evaluated from the Chapman-Enskog expression (ref. 7) with measured values of the Lennard-Jones parameters for the argon-lithium interaction (ref. 8).

Figure 3 shows the results for startup with a constant heat input \( Q \) of 5 kilowatts and initial gas pressures \( p_i \) of 50, 100, and 200 torr. The switch on times \( t_s \) for these cases are -12.2, -13.0, and -13.9 seconds, respectively. At \( t = 0 \) the vapor energy transport \( Q_v \) increases from near zero to slightly less than \( Q \) in less than 0.02 second. In this period of time the vapor pressure \( p_v(T) \) increased by only 0.7 percent, but this is sufficient to raise the parameter \( \lambda \) (eq. (8)) from 0 to 2.2. It should be noted that it takes about the same length of time to establish a steady-state concentration profile in an argon-lithium system of the previously given dimensions at these pressures and temperatures. Nevertheless, the figures show that in each case startup is initiated at a very precise temperature \( T_0 \) given by equation (36). This fact can be used to eliminate startup difficulties in many applications. Since the heat-transfer capability of the heat pipe increases with the vapor pressure \( p_v \), the initial gas pressure \( p_i \) can be increased to the point where startup is not initiated until \( p_v \) is large enough to support a heat-transfer rate large enough to overcome the thermal load on the condensor.

Figure 4 shows the results for a startup in which \( Q \) is increased in steps to 25 kilowatts with \( p_i = 50 \) torr. In this way the vapor flow rate at the evaporator exit does not reach a very large value until \( p_v \) is large enough to sustain it.
Figure 4. - Startup behavior for an argon-lithium heat pipe with heat input $Q$ increased in steps from 2 to 25 kilowatts. Length of heat pipe, $L_p = 300$ centimeters; length of evaporator, $L_e = 30$ centimeters; initial pressure of noncondensible gas, 50 torr; time at which $Q$ is turned on, $t_1 = 30.4$ seconds.
CONCLUSIONS

A two-dimensional diffusion analysis has been used to determine the energy transported by the vapor between the hot and cold zones of a gas-loaded heat pipe. This theory has been combined with a simplified model of the startup process. Numerical results from the combined analysis show that very large diffusion rates occur during startup. Such a result cannot be obtained from a one-dimensional axial diffusion analysis since the radial flow is an essential part of the process.

The numerical results also show that startup is not initiated until the vapor pressure \( p_v(T) \) reaches a precise value which is proportional to the initial gas pressure \( p_i \). Hence, startup difficulties can be reduced by selecting a value for \( p_i \) such that the pipe will not start until \( p_v \) is large enough to support a heat-transfer rate sufficient to overcome the thermal load on the pipe.

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The aeronautical and space activities of the United States shall be conducted so as to contribute ‘. . . to the expansion of human knowledge of phenomena in the atmosphere and space. The Administration shall provide for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof.”

—National Aeronautics and Space Act of 1958

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