QUALIFICATION TEST REPORT

BUMP PROTECTION HAT

(SUBASSEMBLY OF
T020/M509 HEAD PROTECTIVE ASSEMBLY)

CONTRACT NAS 9-11905

TYPE II DOCUMENT

DRL T-706, LINE ITEM 61

(NASA-CR-140347) QUALIFICATION TEST
REPORT BUMP PROTECTION HAT (SUBASSEMBLY
OF T020/M509 HEAD PROTECTIVE ASSEMBLY)
(General Electric Co.) 39 p HC $3.75

PREPARED BY: D. B. Willis
Product Engineer

APPROVED BY: G. E. Wood
Project Manager

GENERAL ELECTRIC
APOLLO AND GROUND SYSTEMS
HOUSTON PROGRAMS
Houston, Texas
ABSTRACT

The BPH (Bump Protection Hat), P/N 793-102, S/N 1001, was subjected to impact testing in accordance with QTP-72-1037, Qualification Test Procedure, and GE-TPS-C-010. The BPH underwent three impacts at 35 foot-pounds of energy. The impacts generated stress cracks, but no penetration. All impacts resulted in deflections of less than one-half inch.

The BPH successfully passed the testing and is considered to be qualified for Skylab and the Rescue Vehicle.
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APPENDIX A
APPENDIX B
APPENDIX C
1.0 INTRODUCTION

In accordance with QTP-72-1026, Qualification Test Plan for Skylab T020/M509 Head Protective Assembly, the only test required to qualify the assembly is to perform impact testing on the BPH (Bump Protection Hat). This report documents the performance of the required impact test.

Testing was performed in accordance with QTP-72-1037, Impact Test Qualification Test Procedure for the BPH (see Appendix A). Testing was performed on GE-TPS-C-010 (Appendix B). One deviation was made to the test procedure. Figure 3 of the test procedure called for impacts to be performed on one side, the front, and the back of the helmet. The actual impacts were performed on one side, the front, and the top of the helmet. The top was impacted instead of the back for two reasons: (1) It was felt that a top impact would be more realistic than a back impact. (2) The test facility would have to be modified to perform a back impact.

Testing was performed on the side and top of the helmet on October 30, 1972. The pendulum then had to be modified in order to perform the front impact. Front impact was performed November 16, 1972.

2.0 TEST APPARATUS

The test equipment is shown in Figures 1 and 2 of the test procedure (Appendix A). Photographs number 1, 4, and 8 of Appendix C show the test setup for the three impacts performed. The stop watch used during testing was not calibrated, as indicated on the Test Summary Sheet (Attachment A of QTP-72-1037).

3.0 PROCEDURE

Test procedures were according to QTP-72-1037 (Appendix A) and GE-TPS-C-010 (Appendix B).

4.0 RESULTS

All three impacts caused stress cracks on the outside of the BPH. The side impact also caused stress cracks on the inside of the BPH. There were no stress cracks on the inside of the BPH for the top and front impacts. Analysis of the side impact cracks showed that the outside stress cracks were not the same as the inside cracks, hence, there was no penetration of the helmet. All cracks were acceptable according to the pass-fail criteria of QTP-72-1026.
4.0 RESULTS (Continued)

Analysis of the high-speed film for shell deflection showed the following results for the three impacts:

a. Side Impact - Less than 3/8" deflection, though the exact amount could not be determined.


c. Front Impact - Less than 1/2" deflection, though the exact amount could not be determined because the impact was a glancing blow.

Maximum allowable deflection was 1/2". All three deflections were within the allowable amount.

One DR was generated at post-test on the stress cracks (Appendix B). No FIAR's were generated. The DR is closed.

5.0 CONCLUSIONS

All three impacts successfully passed the test from the standpoint of both cracks and deflection.

It was agreed by the Test Conductor, Technical Monitor, and NASA Quality Control that the stop watch not being calibrated was no problem. Stop watches are normally not calibrated for such tests. Human reflex error is a much more significant factor in timing the period of oscillation.

The top of the helmet being impacted instead of the back is considered valid for the following reasons:

1. A top impact is much more likely than a back impact.

2. It was originally not required to impact the top. It was agreed by the test parties that a top impact is desirable.

3. The back area of the helmet is similar to the front area, which was impacted.

4. Side impact is the most severe since there is no liner in the impacted area to absorb part of the energy.

The BPH successfully completed qualification testing for Skylab and Rescue Vehicle.
APPENDIX A

TEST PROCEDURE
TEST SUMMARY SHEET
TEST DATA SHEETS
IMPACT TEST

QUALIFICATION TEST PROCEDURE

FOR SKYLAB

BUMP PROTECTION HAT

P/N 793-102

CONTRACT NAS 9-11905

Type I, Document

DRL T-706, Line Item 63

PREPARED BY: D. B. Willis
D. B. Willis
Product Engineer

APPROVED: C. E. Wood
C. E. Wood
Project Manager

APPROVED: / D. B. Willis
Technical Monitor

APPROVED: / J. B. Lanier
Branch Chief

APPROVED: / D. W. Bell
Chief, Crew Systems Division

APPROVED: / W. R. Rain
Quality Engineering

GENERAL ELECTRIC
APOLLO & GROUND SYSTEMS
HOUSTON PROGRAMS
Houston, Texas
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**DOCUMENT CHANGE/REVISION LOG**

Date: [BET 16 2037]
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<td>TEST PROCEDURE</td>
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1.0 INTRODUCTION

This document details the procedure for subjecting the BPH (Bump Protection Hat) to impact testing according to QTP-72-1026, Qualification Test Plan for Skylab T020/M509 Head Protective Assembly.

2.0 PURPOSE

The purpose of the test is to determine the shock absorbing characteristics of the BPH when impacted with a fixed hemispherically shaped anvil. The impact shall be at 12 ft./sec. with a mass of 0.48 slugs, according to QTP-72-1026, or an equivalent energy of 35 foot pounds. High speed photography shall be used to determine the shell deflection during impact.

3.0 APPLICATION

Any discrepancies, difficulties, replacement of components, deviations from procedures, etc., experienced during the test shall be documented on the Test Summary Sheet (Attachment A). The Test Conductor and the MSC Inspector shall sign the Test Summary Sheet. Any DR's or FIAR's generated shall be written in accordance with paragraph 5.6.2 of QTP-72-1026. Deviations from procedure shall be indicated on the Test Data Sheet (Attachment B).

4.0 REQUIREMENTS AND TEST PREPARATIONS

4.1 TEST EQUIPMENT AND FACILITY REQUIREMENTS

The test will be conducted by the special projects laboratory of the Crew Systems Division's Material Development Section. The test apparatus shall be a pendulum and wooden mannequin head assembly as shown in photograph 1.

The test article shall be one Bump Protection Hat, P/N 793-102, S/N 1001.

Support equipment shall include the following:

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<th>Item</th>
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<td>Communications Carrier</td>
<td>16536G-03 or -04</td>
<td>Class III</td>
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<tr>
<td>Camera</td>
<td>--</td>
<td>400 frames/sec.</td>
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<tr>
<td>Stop Watch</td>
<td>--</td>
<td>Calibrated</td>
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<tr>
<td>Grid</td>
<td>--</td>
<td>$\frac{1}{8}$&quot; squares or less</td>
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4.2 TEST EQUIPMENT VALIDATION

Prior to start of the test, a receiving inspection shall be performed on the BPH to assure no damage. The camera and stop watch shall be verified to be calibrated and functioning properly.

4.3 SAFETY PRECAUTIONS AND SPECIAL HANDLING

There are no special safety precautions involved in this test. Caution should be exercised while handling the BPH, camera, and stop watch to prevent damage to the hardware.

4.4 TEST SETUP

The test equipment shall be set up as shown in Figures 1 and 2. The Communications Carrier shall be placed on the mannequin head and the BPH strapped over the Communications Carrier.

NOTE: Adjust the BPH suspension system to the mannequin head size.

4.5 TEST SETUP VALIDATION

The MSC Inspector shall verify proper test setup and stamp the Test Data Sheet accordingly, prior to commencing test.

5.0 TEST PROCEDURE

5.1 Prepare pendulum and wooden head assembly per Figures 1 and 2.

5.2 Suspend the impact head with the BPH and Communications Carrier mounted as a pendulum, and determine the period of oscillation. Determine period by swinging the pendulum through an arc of \(< 60^\circ\) and timing the period through 10 oscillations, using the stop watch to time.

5.3 Calculate the center of percussion of the impact assembly by the following formula: \( L = \frac{gt^2}{(2\pi)^2} \) where \( L \) is the length from the pivot point to center of percussion, \( g \) is acceleration due to gravity, and \( t \) is the period of oscillation of the pendulum. (See Figure 1.)
5.4 Weigh and record weight of the impact assembly with the BPH and Communications Carrier mounted on the head.

Weight

5.5 Calculate the release angle required for a 35 foot-pound impact by the following formulas:

a. \( E = mgh \) where \( m \) is the mass of the impact assembly, \( g \) is the acceleration due to gravity, and \( h \) is the vertical height when the center of percussion is raised.

b. \( \theta = \cos \left( \frac{L-h}{L} \right) \) where \( L \) and \( h \) are defined in 5.3 and 5.5 respectively.

5.6 Orient the impact assembly such that region 1 of Figure 3 will be impacted.

5.7 Pull impact assembly back to the desired release angle as calculated in step 5.5.

5.8 Begin countdown of 5-4-3-2-1 - Release. Turn camera on at count of 2.

5.9 Impact helmet. Do not let helmet strike the anvil more than once. Stop camera.

5.10 Orient the impact assembly such that region 2 of Figure 3 will be impacted.

5.11 Repeat steps 5.7 through 5.9.

5.12 Orient the impact assembly such that region 3 of Figure 3 will be impacted.

5.13 Repeat steps 5.7 through 5.9.

5.14 Remove the BPH and inspect for damage. Pass-fail criteria for the BPH is defined in paragraph 6.3.2.2.3 of QTP-72-1026.
FIG. 1

PIVOT POINT

PENDULUM HEAD

WOODEN HEAD

IMPACT ANVIL
2" Dia. Surface
Add weights if necessary to bring impact assembly to a total of 15.25 lbs, including BPH and Comm. Carrier weight.

BPH weights for necessary to bring to 15.25 lbs., including BPH and Comm. Carrier weight.
Region 1 may be on either side.

Region 3 is the same as region 2, except that it shall be on the back side.
TEST SUMMARY SHEET
 FOR
 IMPACT TEST OF BUMP PROTECTION HAT

1. Test Hardware P/N 793-102
   Test Hardware S/N 1001
   Date Test Performed 10/30/72

2. Indicate Test Procedure deviations below:
   a. Stop Watch not calibrated.
   b. Region three or Fig 3 replaced by a top of helmet impact (see TPS 6E-TPS-C-010)

3. Indicate DR's generated during test below:
   NASA-060 11/16/72

4. Indicate FIAR's generated during test below:
   None

5. Additional test data information:
   None

6. With the exceptions noted above, the test was successfully performed according to the procedures of QTP-72-1037, Impact Test Qualification Test Procedure for Skylab Bump Protection Hat.

DB Williams 11/27/72
Test Conductor

MSC Quality Inspector

ADM for L. N斯顿
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<th>PROCEDURE</th>
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| 4.2  | Receiving Inspection performed on BPH. Condition satisfactory.  
      | **BPH P/N** 793-102  
      | **BPH S/N** 1001 |
| 4.2  | Support equipment calibrated and functioning properly. |
| 4.5  | Test setup verified properly. |
| 5.2  | Period of oscillation determined. Period is 2.48 seconds. |
| 5.3  | Center of percussion determined. L is 5.01 ft. |
| 5.4  | Weight of impact assembly is 15.25 lbs. including BPH, Communication Carrier. |
| 5.5  | Release Angle = 57 degrees. |
| 5.6  | BPH oriented for region 1. |
| 5.9  | BPH impacted for region 1. |
| 5.10 | BPH oriented for region 2.  
      | **As specified on** GE-TPS-C-018  
      | **Note** |
| 5.11 | BPH impacted for region 2.  
      | **Note** |
| 5.12 | BPH oriented for region 3. |
| 5.13 | BPH impacted region 3. |
| 5.14 | BPH inspected for damage. Record any DR's generated at Post-Test here.  
<pre><code>  | **DR# NASA-060 11/16/72** |
</code></pre>
<p>|      | **Record any PLAR's generated at Post-Test here. ** |
|      | <strong>NONE</strong> |</p>
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<td>BPH S/N 1001</td>
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<td>Support equipment calibrated and functioning properly.</td>
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<td>4.5</td>
<td>Test setup verified proper.</td>
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<td>5.2</td>
<td>Period of oscillation determined. Period is 2.46 seconds.</td>
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<td>5.3</td>
<td>Center of percussion determined. L is 4.95 ft.</td>
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<td>5.4</td>
<td>Weight of impact assembly is 15.25 lbs., including BPH, Communication Carrier.</td>
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<td>5.5</td>
<td>Release Angle = 58 degrees.</td>
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<td>5.6</td>
<td>BPH oriented for region 1.</td>
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<tr>
<td>5.9</td>
<td>BPH impacted for region 1.</td>
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<td>5.10</td>
<td>BPH oriented for region 2.</td>
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<td>5.11</td>
<td>BPH impacted for region 2.</td>
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<tr>
<td>5.12</td>
<td>BPH oriented for region 3. (Region 2 of Fig. 3)</td>
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<td>BPH impacted region 3.</td>
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<td>BPH inspected for damage. Record any DR's generated at Post-Test here.</td>
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Record any FIAR's generated at Post-Test here.

None
**Test Preparation Sheet**

**NASA - Manned Spacecraft Center**

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**Contractor**

**NASA - Manned Spacecraft Center**

**Ref. To Procedures For Required Signatures**

**Contractor**

**Date**

**NASA**

**Date**
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### 4a. (Continued)
Point of the Bump Hat

### 4b. Delete the region 3 impact (A total of two impacts shall be performed instead of the original three per ATP-72-1037).

### 5. At the completion of testing, return the Bump Hat and the Comm Carrier to 6E Receiving.


### 8. Transfer Bump Hat S/N 1001 and Comm Carrier S/N 108 to Bldg 7 Bay Ave for impact test.


### 10. Perform the procedures indicated in Document ATP-72-1037, dated Oct. 19, 1972, with the following exception: Perform one impact only, being region 2 of Fig. 3 of ATP-72-1037.
11. At the completion of testing, remove the floral liner as required from the Bump Hat to inspect the impacted areas from the inside. Record any applicable DR's or FIA's generated as a result of inspection, by entering the number DR# NASA-060 in the applicable area of the Test Data Sheet of ATP-72-1037.


11/29/72
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<td>2. Delete step 4b of original TPS</td>
<td></td>
</tr>
<tr>
<td>3. Revise step 5 of original TPS to read as follows: At the completion of testing for regimes 1 and 2, return the Bump Hat and the Corn Cover to 6E Receiving and hold until test facility is modified to perform region 3 test</td>
<td></td>
</tr>
</tbody>
</table>


REFER TO PROCEDURES FOR REQUIRED SIGNATURES

Contractor Date

NASA - MANNED SPACECRAFT CENTER

NASA—MSC—Cont., Houston, Texas.
## DISCREPANCY REPORT/MATERIAL REVIEW RECORD

### NASA - MANNED SPACECRAFT CENTER

<table>
<thead>
<tr>
<th>Category</th>
<th>DISCREPANCY REPORT/MATERIAL REVIEW RECORD</th>
</tr>
</thead>
<tbody>
<tr>
<td>1. Category</td>
<td>GE</td>
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<tr>
<td>2. Article Name</td>
<td>Bump Protection Hat</td>
</tr>
<tr>
<td>3. Drawing Number</td>
<td>793-102</td>
</tr>
<tr>
<td>4. Serial/Lot Number</td>
<td>1001</td>
</tr>
<tr>
<td>5. RECORD NUMBER</td>
<td>DR/MRR NASA-060</td>
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<tr>
<td>6. Contractor's Name</td>
<td>GENERAL ELECTRIC</td>
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<tr>
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</tr>
<tr>
<td>8. Contractor's Serial Number</td>
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</tr>
<tr>
<td>9. Supplier's Name</td>
<td>SIERRA ENG, C.</td>
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<tr>
<td>10. Supplier's Drawing Number</td>
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<tr>
<td>11. Supplier's Serial No.</td>
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<td>12. Next Higher Assy</td>
<td>NASA-MAN -- PROTECTION KIT</td>
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<tr>
<td>13. REF. Document No.</td>
<td>GE-TPS-C-010</td>
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<tr>
<td>14. SpaceXcraft</td>
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<tr>
<td>15. Family</td>
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<tr>
<td>16. Function</td>
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<tr>
<td>17. Cause/Origin</td>
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<tr>
<td>18. System</td>
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<td>19. Disposition</td>
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</table>

### DISCREPANCY

**REFERENCE** STEP #11 of GE-TPS-C-010, INSPECTION AFTER IMPACT REVEALED STRESS CRACKS AT ALL THREE IMPACT LOCATIONS.

### DISPOSITION

**INSTRUCTIONS**

1. Pass - fail criteria in the Qual Test Plan, QTP-72-1026, para. 6.3.2.2.3, states that stress cracks are acceptable, but penetration cracks are not. None of the cracks penetrated the helmet. Hence the cracks are acceptable for passing Qual. Qualification testing is complete. Downgrade the Bump Hat to Class III per TPS-GE-TPS-C-010.

### MRB APPROVAL SIGNATURES

**MRB APPROVAL SIGNATURES**

<table>
<thead>
<tr>
<th>31. System Engineer (Contractor)</th>
<th>DATE</th>
<th>32. System Engineer (NASA)</th>
<th>DATE</th>
</tr>
</thead>
<tbody>
<tr>
<td>L.E. Zinna 05</td>
<td>11/29/72</td>
<td>D. Breaker 03</td>
<td>11/29/72</td>
</tr>
<tr>
<td>33. Quality Control Rep. (Contractor)</td>
<td>DATE</td>
<td>34. Quality Control Rep. (NASA)</td>
<td>DATE</td>
</tr>
<tr>
<td>L. J. Green 04</td>
<td>11/29/72</td>
<td>D. Breaker 03</td>
<td>11/29/72</td>
</tr>
<tr>
<td>N. Rathbun 01</td>
<td>11/29/72</td>
<td>D. Breaker 03</td>
<td>11/29/72</td>
</tr>
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</table>
PHOTOGRAPH DESCRIPTIONS BY NUMBER

1. Test Setup for Side Impact
2. Side Impact Viewed from outside helmet
3. Side impact viewed from inside helmet
4. Test setup for top impact
5. Top of helmet viewed prior to impact
6. Top of helmet after impact (magnified 3 times)
7. Top inside of helmet after impact
8. Test setup for front impact
9. Front of helmet viewed prior to impact
10. Front of helmet after impact
11. Front inside of helmet after impact

NOTE: All high-speed film taken to determine deflection is in file at General Electric.