General Disclaimer

One or more of the Following Statements may affect this Document

- This document has been reproduced from the best copy furnished by the organizational source. It is being released in the interest of making available as much information as possible.

- This document may contain data, which exceeds the sheet parameters. It was furnished in this condition by the organizational source and is the best copy available.

- This document may contain tone-on-tone or color graphs, charts and/or pictures, which have been reproduced in black and white.

- This document is paginated as submitted by the original source.

- Portions of this document are not fully legible due to the historical nature of some of the material. However, it is the best reproduction available from the original submission.

Produced by the NASA Center for Aerospace Information (CASI)
IMPROVED ULTRASONIC STANDARD REFERENCE BLOCKS

NATIONAL BUREAU OF STANDARDS

PREPARED FOR
AIR FORCE MATERIALS LABORATORY

FEBRUARY 1976
IMPROVED ULTRASONIC STANDARD REFERENCE BLOCKS

NATIONAL BUREAU OF STANDARDS
WASHINGTON, D.C. 20234

FEBRUARY 1976

TECHNICAL REPORT AFML-TR-75-180
INTERIM REPORT FOR PERIOD 1 JANUARY 1974 - 31 DECEMBER 1974

Approved for public release; distribution unlimited
Notice

When Government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related Government procurement operation, the United States Government thereby incurs no responsibility nor any obligation whatsoever; and the fact that the government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto.

This report has been reviewed by the Information Office (OI) and is releasable to the National Technical Information Service (NTIS). At NTIS it will be available to the general public, including foreign nations.

This technical report has been reviewed and is approved for publication.

LEE R. GULLEY, JR.
Project Engineer

FOR THE COMMANDER

THOMAS D. COOPER, CHIEF
Aeronautical Systems Branch
Systems Support Division

Copies of this report should not be returned unless return is required by security considerations, contractual obligations, or notice on a specific document.

AIR FORCE - 9 JUNE 76 - 150
**REPORT DOCUMENTATION PAGE**

<table>
<thead>
<tr>
<th>1. REPORT NUMBER</th>
<th>2. GOVT ACCESSION NO.</th>
<th>3. RECIPIENT'S CATALOG NUMBER</th>
</tr>
</thead>
<tbody>
<tr>
<td>AMRL-TR-75-180</td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>4. TITLE (and Subtitle)</th>
<th>5. TYPE OF REPORT &amp; PERIOD COVERED</th>
<th>6. PERFORMING ORG. REPORT NUMBER</th>
</tr>
</thead>
<tbody>
<tr>
<td>IMPROVED ULTRASONIC STANDARD REFERENCE BLOCKS</td>
<td>Interim Report</td>
<td>NBSIR 75-685</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>7. AUTHOR(s)</th>
<th>8. CONTRACT OR GRANT NUMBER(s)</th>
</tr>
</thead>
</table>

<table>
<thead>
<tr>
<th>9. PERFORMING ORGANIZATION NAME AND ADDRESS</th>
<th>10. PROGRAM ELEMENT, PROJECT, TASK AREA &amp; WORK UNIT NUMBERS</th>
</tr>
</thead>
<tbody>
<tr>
<td>National Bureau of Standards</td>
<td>Project No. 7381</td>
</tr>
<tr>
<td>Washington, D. C. 20234</td>
<td>Task Area 07</td>
</tr>
<tr>
<td></td>
<td>Work Unit 43</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>11. CONTROLLING OFFICE NAME AND ADDRESS</th>
<th>12. REPORT DATE</th>
</tr>
</thead>
<tbody>
<tr>
<td>Air Force Materials Laboratory</td>
<td>1966</td>
</tr>
<tr>
<td>National Aeronautics and Space Administration (LeRC)</td>
<td></td>
</tr>
<tr>
<td>and Army Materials and Mechanics Research Center</td>
<td></td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>13. NUMBER OF PAGES</th>
<th>14. MONITORING AGENCY NAME &amp; ADDRESS (if different from Controlling Office)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Air Force Materials Laboratory</td>
</tr>
<tr>
<td></td>
<td>Wright-Patterson AFB, Ohio 45433</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>15. SECURITY CLASS. (of this report)</th>
<th>16. DISTRIBUTION STATEMENT (of this report)</th>
</tr>
</thead>
<tbody>
<tr>
<td>Unclassified</td>
<td>Approved for public release; distribution unlimited</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)</th>
</tr>
</thead>
</table>

<table>
<thead>
<tr>
<th>18. SUPPLEMENTARY NOTES</th>
<th>19. KEY WORDS (Continue on reverse side if necessary and identify by block number)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Aluminum ultrasonic standards; ASTM-type reference blocks; fabrication of refer-</td>
</tr>
<tr>
<td></td>
<td>ence blocks; immersion testing; longitudinal waves; metallurgical variables;</td>
</tr>
<tr>
<td></td>
<td>nondestructive testing, pulse-echo; steel ultrasonic standards; titanium ultra-</td>
</tr>
<tr>
<td></td>
<td>sonic standards; ultrasonics.</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>20. ABSTRACT (Continue on reverse side if necessary and identify by block number)</th>
</tr>
</thead>
<tbody>
<tr>
<td>A program to improve the quality, reproducibility and reliability of non-</td>
</tr>
<tr>
<td>destructive testing through the development of improved ASTM-type ultrasonic</td>
</tr>
<tr>
<td>reference standards is described. Reference blocks of aluminum, steel, and</td>
</tr>
<tr>
<td>titanium alloys are to be considered. Equipment representing the state-of-the-</td>
</tr>
<tr>
<td>art in laboratory and field ultrasonic equipment was obtained and evaluated.</td>
</tr>
<tr>
<td>RF and spectral data on ten sets of ultrasonic reference blocks have been taken</td>
</tr>
<tr>
<td>as part of a task to quantify the variability in response from nominally iden-</td>
</tr>
</tbody>
</table>
tical blocks. Techniques for residual stress, preferred orientation, and micro-structural measurements were refined and are applied to a reference block rejected by the manufacturer during fabrication in order to evaluate the effect of metallurgical condition on block response. New fabrication techniques for reference blocks are discussed and ASTM activities are summarized.
FOREWORD

This report was prepared by the National Bureau of Standards, Washington, D.C. The authors are Dr. D. G. Eitzen, project leader, G. F. Sushinsky, and D. J. Chwirut of the Engineering Mechanics Section, and G. J. Bechtoldt and Dr. A. W. Ruff of the Microstructure Characterization Section.

The following people have made valuable contributions to this program. Their assistance is gratefully acknowledged: G. D. Boswell, R. G. Elmer, M. Greenspan, C. E. Tschiegg, and F. R. Brockenridge, all of NBS; C. A. Burley and D. Feitig of the Reynolds Metals Company; and M. A. Kearney of Automation Industries. In addition, reference blocks have already been loaned by the following companies for evaluation under task 3, Naval Research Laboratory, Reynolds Metals Company, Westinghouse Electric Corporation, and Wyman-Gordon Company.

Sponsorship and financial assistance were received from the Air Force Materials Laboratory, National Aeronautics and Space Administration (Lewis Research Center), and the Army Materials and Mechanics Research Center. These organizations were technically represented by Lee R. Gulley, Jr., (AFML/IXA), Robert L. Davies, and George Darcy, respectively. The work was administered through the Air Force Materials Laboratory with Lee R. Gulley, Jr., as project monitor. The work was authorized under USAF Delivery Order F-33615-74-M6752 in support of Project No. 7381, Task No. 738107, and was performed during the period 1 January, 1974 to 31 December, 1974.

This report was submitted by the authors July 1975.
## TABLE OF CONTENTS

<table>
<thead>
<tr>
<th>Section</th>
<th>Page</th>
</tr>
</thead>
<tbody>
<tr>
<td>1. INTRODUCTION</td>
<td>1</td>
</tr>
<tr>
<td>2. PROGRAM OUTLINE</td>
<td>4</td>
</tr>
<tr>
<td>3. ACTIVITY SUMMARY</td>
<td></td>
</tr>
<tr>
<td>3.1 Literature Survey</td>
<td>7</td>
</tr>
<tr>
<td>3.2 Ultrasonic Measurement Facility</td>
<td></td>
</tr>
<tr>
<td>3.2.1 Immersion System</td>
<td>8</td>
</tr>
<tr>
<td>3.2.2 Flaw Detection Equipment</td>
<td>8</td>
</tr>
<tr>
<td>3.2.3 Spectrum Analyzer</td>
<td>10</td>
</tr>
<tr>
<td>3.2.4 Accessory Equipment</td>
<td>10</td>
</tr>
<tr>
<td>3.3 Comparison of Nominally Identical Blocks</td>
<td>11</td>
</tr>
<tr>
<td>3.4 Metallurgical Considerations</td>
<td>13</td>
</tr>
<tr>
<td>3.4.1 Orientation Texture</td>
<td>27</td>
</tr>
<tr>
<td>3.4.2 Reference Block Microstructure</td>
<td>37</td>
</tr>
<tr>
<td>3.4.3 Ultrasonic Inspection of Rejected Block</td>
<td>42</td>
</tr>
<tr>
<td>3.4.4 Residual Stress Measurements</td>
<td>42</td>
</tr>
<tr>
<td>3.5 Fabrication Considerations</td>
<td>48</td>
</tr>
<tr>
<td>3.6 Effects of Ultrasonic Measuring Systems</td>
<td>50</td>
</tr>
<tr>
<td>3.7 Single Material Standard</td>
<td>50</td>
</tr>
<tr>
<td>3.8 ASTM Participation</td>
<td>50</td>
</tr>
<tr>
<td>4. IMPORTANT CONCLUSIONS TO DATE</td>
<td>53</td>
</tr>
<tr>
<td>APPENDIX A</td>
<td>56</td>
</tr>
<tr>
<td>APPENDIX B</td>
<td>62</td>
</tr>
<tr>
<td>APPENDIX C</td>
<td>63</td>
</tr>
<tr>
<td>APPENDIX D</td>
<td>66</td>
</tr>
<tr>
<td>APPENDIX E</td>
<td>68</td>
</tr>
<tr>
<td>REFERENCES</td>
<td>79</td>
</tr>
<tr>
<td>Figure</td>
<td>Description</td>
</tr>
<tr>
<td>--------</td>
<td>-----------------------------------------------------------------------------</td>
</tr>
<tr>
<td>Figure 1</td>
<td>ASTM E-127 ultrasonic standard reference block. All dimensions in inches (1 in=25.4 mm).</td>
</tr>
<tr>
<td>Figure 2</td>
<td>Estimated Program Timetable.</td>
</tr>
<tr>
<td>Figure 3</td>
<td>Ultrasonic measurement facility showing 1) oscilloscope, 2) laboratory pulser/receiver, gate, and peak-detector, 3) flaw detector, 4) immersion system, 5) spectrum analyzer, and 6) ultrasonic standard reference blocks.</td>
</tr>
<tr>
<td>Figure 4</td>
<td>Distance - amplitude data for No. 3 blocks at 2.25 MHz.</td>
</tr>
<tr>
<td>Figure 5</td>
<td>Distance - amplitude data for No. 5 blocks at 2.25 MHz.</td>
</tr>
<tr>
<td>Figure 6</td>
<td>Distance - amplitude data for No. 8 blocks at 2.25 MHz.</td>
</tr>
<tr>
<td>Figure 7</td>
<td>Distance - amplitude data for No. 3 blocks at 5.0 MHz.</td>
</tr>
<tr>
<td>Figure 8</td>
<td>Distance - amplitude data for No. 5 blocks at 5.0 MHz.</td>
</tr>
<tr>
<td>Figure 9</td>
<td>Distance - amplitude data for No. 8 blocks at 5.0 MHz.</td>
</tr>
<tr>
<td>Figure 10</td>
<td>Distance - amplitude data for No. 3 blocks at 10 MHz.</td>
</tr>
<tr>
<td>Figure 11</td>
<td>Distance - amplitude data for No. 5 blocks at 10 MHz.</td>
</tr>
<tr>
<td>Figure 12</td>
<td>Distance - amplitude data for No. 8 blocks at 10 MHz.</td>
</tr>
<tr>
<td>Figure 13</td>
<td>RF signals and spectra from two nominally identical 5-0275 reference blocks at three frequencies.</td>
</tr>
<tr>
<td>Figure 14</td>
<td>(200) pole figure for 7075-T631 aluminum.</td>
</tr>
<tr>
<td>Figure 15</td>
<td>(111) pole figure for 7075-T631 aluminum.</td>
</tr>
<tr>
<td>Figure 16</td>
<td>(10.1) pole figure for commercial titanium sheet.</td>
</tr>
<tr>
<td>Figure 17</td>
<td>(00.2) pole figure for annealed commercial titanium sheet.</td>
</tr>
<tr>
<td>Figure 18</td>
<td>(10.0) pole figure for annealed commercial titanium sheet.</td>
</tr>
<tr>
<td>Figure 19</td>
<td>(200) pole figure for slice from aluminum reference block.</td>
</tr>
<tr>
<td>Figure 20</td>
<td>(111) pole figure for slice from aluminum reference block.</td>
</tr>
<tr>
<td>Figure 21</td>
<td>(220) pole figure for slice from aluminum reference block.</td>
</tr>
</tbody>
</table>
Figure 22 - Ultrasonic and residual stress data on rejected aluminum reference block. Note 1 Pa=1.45x10^-4 lb/ft², 1 mm=0.039 in. 38
Figure 23 - Optical micrograph of surface S2 as polished. 39
Figure 24 - Optical micrograph of surface S2 after first etching. 39
Figure 25 - Optical micrograph of surface S2 after longer etching. 40
Figure 26 - Optical micrograph of surface S2 after repolishing. 40
Figure 27 - Higher magnification optical micrograph of surface S2 after repolishing. 41
Figure 28 - Waveforms and spectra of back surface reflections of block 1. See Table 3 and Figure 22. 43
Figure 29 - Waveforms and spectra of back surface reflections of block 2. See Table 3 and Figure 22. 45
Figure 30 - Universal distance-amplitude curves for three sets of aluminum reference blocks. 54
Figure A-1 - Typical pulse waveforms from laboratory pulser. 61
Figure A-2 - Typical "steps" (switching transients) in the gated output. 61
IMPROVED ULTRASONIC STANDARD REFERENCE BLOCKS

D. C. Eitzen, G. F. Sushinsky, and D. J. Chwirut
and C. J. Bechtoldt and A. W. Ruff

1. INTRODUCTION

In a wide range of technical activities, a greater dependence on nondestructive testing and evaluation (NDT and E) methods is being witnessed. The causes for this greater dependence on NDT and E methods include increased structural performance requirements, the use of defect-sensitive materials, changes in design philosophy, and increased requirements for the determination of the condition and changes in the condition of materials in service. The world-wide shortages of materials and energy have created pressure for the adoption of a "keep it in service if possible" attitude to replace the old "remove and replace on schedule" philosophy.

In particular, ultrasonic methods are being increasingly relied upon to evaluate material and structural condition. Characteristically, the NDT and E activities are performed at interfaces between different operational groups, e.g., material supplier - user, subcontractor - contractor, and part production - assembly. Lack of agreement in the results of ultrasonic evaluations at such interfaces can, in part, be traced to a lack of standard methodology and a lack of basic measurement standards since the techniques are highly dependent on reference standards. The incompatibility of measurements by different operational groups results in uncertainties regarding the actual material condition. These uncertainties lead to performance penalties due to increased design uncertainties and either unnecessary piece rejection or inadequate service performance. In addition to the performance penalties, serious economic inequities often result from the lack of reference standards or measurement inaccuracies.

A program to improve the widely used system of ASTM-type reference blocks for longitudinal ultrasonic testing was started in January 1974. The procedures for fabricating and checking these blocks are covered in two ASTM documents, E 127-64 "Standard Recommended Practice for Fabricating and Checking Aluminum Alloy Ultrasonic Standard Reference Blocks" [1], and E 428-71, "Standard Recommended Practice for Fabrication and Control of Steel Reference Blocks Used in Ultrasonic Inspection" [2]. Both of these documents are widely referenced in government and industry purchasing specifications and many other ASTM documents. One of the above documents is also sometimes used as a guide for the fabrication of titanium alloy ultrasonic reference blocks. However, both the authors and users of these documents admit that both contain serious shortcomings, but, partly

*Figures in brackets indicate literature references at the end of this paper.
because of corporate interests or priorities and a lack of institutional mission, no one has produced acceptable improvements through the voluntary standards systems. In fact, E 127 is scheduled to be dropped in January 1976 because it is unworkable in its present form yet no acceptable alternative has been produced to date. A stop-gap alternative may be approved later this year, but it is far from a total solution to the problem.

The ASTM-type reference blocks are cylindrical blocks with flat-bottomed holes drilled along the block axis, see Figure 1. A pulsed stress wave produced by a piezoelectric transducer enters normal to the undrilled end of the block and travels through the block. The flat end of the drilled hole acts as a reflector and returns some of the energy to the transducer which converts this energy into an electrical signal. This reflected signal, displayed on a cathode ray tube (CRT), becomes a reference signal for the evaluation of material of unknown condition. Sets of reference blocks with different hole diameters and different lengths are used to standardize ultrasonic measurement systems. Measurements made with these systems then provide a basis for estimating flaw severity and possible material rejection.

The problem with the reference blocks, simply stated, is this: using a single ultrasonic measuring system, the ultrasonic response from nominally identical reference blocks varies unacceptably. The extent of this variation has been reported to be as great as 300 percent in titanium. This causes, for example, different materials suppliers and users to inspect to different levels of acceptability, resulting in unjust competition between suppliers and increased costs due to unnecessary rejection and recycling (supplier over inspection) or wasted transportation costs following user rejection (supplier under inspection). The NB3 program is intended to investigate systematically the ASTM-type standard reference block system, to isolate if possible the causes of the variability, and to develop a new system of standards that will allow different organizations to make consistent measurements compatible with each other. It is envisioned that the output from this program could take one of three forms:

1) New methods documents to revise or replace ASTM E 127 and E 428 that would allow the NDT community to fabricate standard reference blocks that introduce acceptably small variability into the measurement system,

2) a system whereby certified standard reference blocks would be fabricated, and sold by the National Bureau of Standards through, for example, the Standard Reference Materials Program,

3) a calibration service whereby one set of blocks is defined as THE STANDARD SET. Users’ blocks could then be referenced to this set following prescribed procedures.
Hole A, bottom must be flat within 0.001 per 0.125 dia., and finished size must be the required dia. ±0.0005

Hole must be straight and perpendicular to entry surface within 0°-30' and located within 0.010 of longitudinal axis.

Flat Counterbore 0.250 x 0.062 deep

Except as noted

Figure 1—ASTM E-127 ultrasonic standard reference block. All dimensions in inches [1 in = 25.4 mm].
This program is centered in the Mechanics and Metallurgy Divisions of the National Bureau of Standards with consultation and support from other Divisions where appropriate.

2. PROGRAM OUTLINE

The objective of the program is to affect near-term improvements in the quality, reproducibility and reliability of ultrasonic nondestructive testing through the development of improved ASTM-type reference blocks. The materials to be used for the development of standards include aluminum, titanium and steel. The program is a two year effort to include the following nine tasks:

Task 1. Literature Search - A thorough search and review of all technical literature regarding ultrasonic test standards will be conducted prior to commencement of any major subsequent tasks. Results of the review will be used where applicable to accelerate or modify subsequent tasks.

Task 2. Ultrasonic Measurement Facility - State-of-the-art ultrasonic equipment and associated electronics appropriate for pulse-echo contact and immersion evaluations will be obtained. This equipment will be evaluated using current standardization methodology. This evaluation will be performed with a view towards the establishment of standard methods which are more definitive than those currently available. This equipment is intended to form the core of an ultrasonic reference block calibration facility, if established.

Task 3. Comparison of Nominally Identical Blocks - Nominally identical blocks from commercial sources and from the field will be evaluated for the distribution of ultrasonic response using the equipment of task 2. This task will serve to assess the extent of variability of ultrasonic responses from nominally identical blocks. The results of this evaluation will have an effect on the methods used to identify the causes of the deviations in blocks.

Task 4. Metallurgical Considerations - The current state of knowledge of the effects of the metallurgical conditions of materials on their ultrasonic characteristics will be reviewed. A limited number of confirmation experiments will be performed. Additional tests on materials of other metallurgical consistency will be undertaken to determine their ultrasonic response characteristics. This knowledge will be applied to the selection of materials for the fabrication of a master set of ASTM-type ultrasonic reference blocks.

Task 5. Fabrication Considerations - A number of nominally identical reference blocks with closely controlled metallurgical properties and fabrication techniques will be obtained. The blocks will be closely examined metrologically and the distribution of ultrasonic response will be determined using the measurements laboratory of Task 2. Several forming techniques will be used including the conventional drilling technique, the use of raw stock formed by powder metallurgy, and the
use of two-piece blocks. Comparison of the distributions in response of these blocks with the results of the evaluation of nominally identical field blocks (Task 3) will indicate whether significant reductions in the deviation of ultrasonic response of blocks can be anticipated in the near-term.

Task 6. Effects of Ultrasonic Measuring Systems - The results of previous round-robin on ASTM-type reference blocks will be checked to determine whether different ultrasonic measuring systems obtain the same ranking and distribution of ultrasonic response from nominally identical blocks. An additional round-robin will be performed, if necessary. The cooperation of interested NDT users will be sought. The verification of the principle of standardization associated with this task is a necessary step toward the establishment of a rational calibration program.

Task 7. Master Reference Blocks - The results of the above tasks will be used to develop master ASTM-type reference standards for aluminum, steel, and titanium. The final alloy selections for the master standards will be based on metallurgical considerations, long-term availability, ultrasonic response, incidence of structural use, and in consultation with the sponsors.

Task 8. A Single-Material Standard - An effort will be made to establish the feasibility of an improved standards program through the use of a single-material master standard. A candidate for the single-material standard is considered to be blocks made of crown glass. This material can be controlled to have an impedance matching that of aluminum, has no crystalline structure, has a minimal defect count (which can be evaluated by light-scattering techniques), and is amenable to the most sophisticated metrological evaluation. Preliminary analyses and tests will establish the feasibility of a one-material standard as the basis for determining the ultrasonic response of reference blocks of various materials. Based on appropriate feasibility indications the development of a basic standard will be considered. Future work may then be proposed in order to establish this standard.

Task 9. Calibration Service - An ASTM-type reference block calibration service will be initiated if appropriate. A system will be established to quantify the responses of blocks in terms of the NBS master standards, thus providing a common basis for comparison and an objective evaluation. Blocks will be evaluated in terms of the Master Reference Blocks of Task 7. It is expected that any continuing calibration service will be self-supporting through fees collected from the users.

The estimated program timetable is as shown in Figure 2.
Figure 2 — Estimated Program Timetable
3. ACTIVITY SUMMARY

3.1 Literature Survey

An extensive search and review of the open literature regarding ultrasonic reference standards has resulted in a collection of over two hundred documents. The search has included four areas: General background information, ultrasonic measurement techniques, previous work directly on standards, and the relationship of metallurgical variables to ultrasonic response. Formal inputs to the search were received from:

Nondestructive Testing Information and Analysis Center,
Defense Documentation Center,
National Technical Information Service, and
Smithsonian Science Information Exchange.

Of these the input from NTIAC was the most comprehensive. The number of pieces of open literature requiring review was surprisingly large, but few speak directly and conclusively to the problem.

In addition to the open literature, several dozen private documents or communications have been analyzed. The search for unpublished or private communications has been more time consuming but often more substantive. Important information regarding ultrasonic reference standards has been obtained through exchanges with representatives from such organizations as Automation Industries, Krautkramer-Branson Inc., Reynolds Metals Company, Aluminum Company of America, Westinghouse Electric Corporation, Kaiser Aluminum Company, Air Force Materials Laboratory, Army Materials and Mechanics Research Center, Naval Research Labs, Titanium Metals Corporation, Boeing Airplane Company, Douglas Aircraft Co., General Dynamics, Grumman Aerospace Corp., The United Kingdom's Aeronautical Quality Assurance Directorate Labs and Atomic Energy Research Establishment Harwell Labs, and of course the American Society for Testing and Materials.

A conclusion as to what is the major cause(s) of the wide distribution of response from nominally identical blocks when examined with a given ultrasonic system was an important objective of the literature search. No conclusion could be drawn. There were significant but sometimes contradictory statements indicating material or metallurgical, dimensional and fabrication problems. Apparently this question will not be resolved until studies based on the results of Task 3 are completed. The review of previous and ongoing work did result in several, more positive conclusions. From work in the United Kingdom over the last ten years it is concluded that "calibrations" by a corrected comparison with a standard set of aluminum blocks can be made to within 1 dB, using state-of-the-art equipment, and that sufficient reductions in block disparity to the point where corrections are not required will be difficult [3]. From work at Grumman [4] on reference blocks for titanium it is concluded
that two piece blocks may provide improved standards for this material. From communications concerning work at Westinghouse and Automation Industries, there is a large disagreement about the size of the problem with steel reference blocks. An additional, important conclusion is that the most active concentrated help can be expected from members of ASTM committee E-7.06. The aluminum producers have been particularly cooperative thus far.

3.2 Ultrasonic Measurement Facility

Commercially available, state-of-the-art ultrasonic equipment and accessories suitable for contact and immersion testing have been assembled through loans, through purchases with project funds, and through the availability of NBS equipment for the project. This includes an immersion tank with a motorized scanning bridge and precision manipulator, flaw detection equipment with associated gating and amplifying circuitry, a spectrum analyzer, and other accessory equipment. The laboratory set-up is shown in Figure 3. Brief descriptions of this equipment are included below with more detailed specifications and characteristics given in Appendix A.

3.2.1 Immersion System

The immersion system consists of a tank with transparent walls and dimensions of approximately 38x21x18 in (97x53x46 cm).* It is equipped with a motorized bridge and carriage, search tube, motorized manipulator, and mini-manipulator. It provides precision control of search unit positioning in the X, Y and Z directions, as well as angular positioning in two vertical planes normal to the tank bottom. A dry paper X-Y recorder is provided.

3.2.2 Flaw Detection Equipment

Two field inspection type flaw detection units, on loan from AFML and NRL, are currently available for use in the laboratory. These units feature a tuned, narrow band pulser and receiver combination. Their nominal operating frequencies are 1.0, 2.25, 5.0, and 10.0 MHz. A video (as opposed to RF) presentation on the CRT is featured. Gating and amplifying modules have also been borrowed. A third unit with updated features is on order. In addition to the above features, this third unit has a "calibrated" dB sensitivity control, an improved CRT display and improved gating and amplifying circuitry. These units are suitable for checking ultrasonic reference blocks per ASTM "Standard Recommended Practice for Fabricating and Checking Aluminum Alloy Ultrasonic Standard Reference Blocks" [1].

*Units for physical quantities in this paper are given in both the U.S. Customary Units and the International System Units (SI).
Figure 3 - Ultrasonic measurement facility showing 1) oscilloscope, 2) laboratory pulser/receiver, gate, and peak-detector, 3) flaw detector, 4) immersion system, 5) spectrum analyzer, and 6) ultrasonic standard reference blocks.
A flaw detector suitable for collecting more detailed laboratory data was also acquired. This unit consists of a power supply-frame, and a broadband pulser receiver combination, stepless gate, and peak detection and quantizing modules. Ultrasonic RF signals are displayed on a 100 MHz bandwidth storage oscilloscope equipped with two wide-band amplifiers. The stepless gate, peak detector, and quantizer provide much of the necessary electronic signal processing for quantitative flaw and search unit characterization. Signals are routed from the receiver through the stepless gate where signals reflected from discontinuities other than the one of interest are eliminated from the repetitive pulse train and the desired wave packet is isolated. This signal can then be used for spectrum analysis or further processed by the peak detector and quantizer. The peak detector converts the positive peak amplitude of the signal to a proportional DC voltage. This can then be quantized into discrete DC voltages based on incremental signal amplitude changes. Such processing is suitable for beam profiling, attenuation measurements or gray-tone C-scan recordings.

3.2.3 Spectrum Analyzer

Spectrum analysis is performed on ultrasonic signals received by the transducer after being processed through the gate circuitry. Signals analyzed are those reflected from special targets (e.g. steel balls, flat quartz blocks) or defects. This information is necessary for the evaluation of search unit characteristics and potentially helpful in determining defect size and orientation [5, 6]. In this program the information will be applied to the determination of the size and orientation of flat bottomed holes in reference blocks. The spectrum analyzer consists of a storage CRT display, and separate IF and RF plug-in modules. The frequency range extends from 0 to 110 MHz with both logarithmic and linear sensitivity displays. Signals processed through the stepless gate can be monitored for spectral content using this instrument.

3.2.4 Accessory Equipment

Search units for use in contact and immersion longitudinal pulse-echo testing made by three different manufacturers have been obtained. The units were chosen on the basis of crystal diameter and nominal center frequency to cover a representative range of those used in ultrasonic work. Special emphasis was placed on obtaining a few quartz search units suitable for work on standard artifacts in accordance with ASTM E 127 [1] and on units from which to choose for Task 3.

Seven sets of ultrasonic reference standards have been purchased from three different manufacturers. These consist of three "Distance/Area
Amplitude" sets (basic sets), purchased directly from the Defense Supply Agency (the source of most Air Force field blocks) and four "Distance Amplitude" sets. The "Distance Amplitude" sets consist of 2 sets of "number 3" blocks from the same manufacturer and one set each of "number 5 and 8" blocks from the third supplier. This sample will provide a measure of the inconsistency of products manufactured by different producers as well as the variability of the standards produced by the same manufacturer. These sets constitute part of the data base to be established at this laboratory. In addition they provide convenient working standards for activities in the program such as the evaluation of new fabrication techniques and consistency checks when different transducers are used.

3.3 Comparison of Nominally Identical Blocks

An important step toward decreasing the disparity in the ASTM-type ultrasonic reference blocks is a survey of nominally identical blocks. The purpose of the survey is to quantify the extent of variability in field blocks presently being used by the NDT community. Participation was enlisted from the membership of ASTM E-07.06, the ultrasonics subcommittee, and from the general NDT community through an appeal in the Nondestructive Testing Information and Analysis Center Newsletter. A list of organizations which have formally offered the loan of reference blocks for data gathering purposes is presented in Appendix B. Five borrowed sets of aluminum blocks have been inspected to date. Additional blocks, aluminum, titanium and steel, have been scheduled. In addition five sets of purchased blocks, 2 distance amplitude sets from one manufacturer and three basic sets from another manufacturer (through the Defense Supply Agency) have been evaluated. Pulse-echo ultrasonic response data were taken from the blocks at three test frequencies, 2.25, 5 and 10 MHz, using the immersion tank with temperature controlled distilled water, wide-band pulser/receiver, stepless gate, oscilloscope and spectrum analyzer previously described. The ultrasonic measurement system was used only in its linear range as determined from the response from steel balls. Some characteristics of the search units used for the data are given in Table 1. All search units on hand at the start of data taking were checked for symmetry, location of the Y₁₀ point (point of separation of near and far fields) [7], center frequency and frequency envelope, RF waveform and sensitivity. The search units used to gather the reference block data were selected on the basis of the above factors.

In taking the data on the blocks, all pulser/receiver settings were put at a repeatable position. The gain was set using the reflected signal from a selected steel ball positioned and maximized at the measured Y₁₀ point. The standardization points for the particular test conditions

*These reference standards are commonly referred to as "number x" blocks where x represents the diameter of the "flat bottomed hole in 64 THS of an inch (1 in - 2.54 cm)."
<table>
<thead>
<tr>
<th>Nominal Center Frequency</th>
<th>Crystal Diameter</th>
<th>Measured Y/2 Point</th>
<th>Transducer Type</th>
<th>Serial No.</th>
</tr>
</thead>
<tbody>
<tr>
<td>2.25 MHz</td>
<td>0.50 in (12.7 mm)</td>
<td>2.5 in (64 mm)</td>
<td>A306</td>
<td>3529</td>
</tr>
<tr>
<td>5.0 MHz</td>
<td>0.50 in (12.7 mm)</td>
<td>5.0 in (127 mm)</td>
<td>A309</td>
<td>3042</td>
</tr>
<tr>
<td>10.0 MHz</td>
<td>0.25 in (6.4 mm)</td>
<td>2.5 in (64 mm)</td>
<td>A312</td>
<td>4263</td>
</tr>
</tbody>
</table>
are given in Table 2. The standardization points served only as a basis for comparison of blocks with a given hole size and were chosen to give the response nearest to the block with a 0.50 in (13 mm) metal travel distance from the first set tested. After the pulser/receiver settings were selected, the transducer was positioned so that the \( \frac{1}{2} \) point was at the ultrasound entry surface of the block. Then the return signal from this surface was maximized by angulating the transducer. The oscilloscope time delay was used to expand the signal reflected from the hole bottom. This signal was tapped off to the stepless gate, which can be set so that the output from this module contains only the signal of interest. Thus, only the return signal from the flat-bottomed hole is fed into the spectrum analyzer. The data collected for each block includes photo-recordings of the RF waveform and the spectrum of the signal from the hole bottom, and a recording of the peak-to-peak voltage and all pertinent equipment settings. Since only one of the ten sets of blocks evaluated was an Area-Amplitude set, the peak-to-peak voltage data is plotted against metal travel distance for number 3, 5, and 8 hole sizes at each of the three frequencies, Figures 4-12. Figure 13 presents typical photo-recordings of the RF signal and the signal spectrum.

As can be seen from Figures 4-12, several anomalies were noted in this study. The unserial-numbered set of number 5 blocks (Figures 5, 8, 11) give very inconsistent response, and in fact the response increases with increasing metal distance at 5 and 10 MHz with differences between nominally identical blocks that are in excess of 700%. The scatter among nominally identical blocks appears to average about 20-30 percent. It appears that several widely variant blocks will be available for inspection under tasks 4 and 5, in order to determine the actual causes of the variability. One might conjecture, however, that at least a part of the disparity is due to a material condition since some sets of blocks appear to give a consistently higher response than others, and one might expect more random variances from the mean to result from other suspected variables such as improper geometry.

3.4 Metallurgical Considerations

This task is concerned with the evaluation of the distribution in response of ultrasonic reference blocks and the material from which they are fabricated, particularly in connection with metallurgical and microstructural parameters. The emphasis has been on aluminum alloys so far, however, titanium has been examined briefly and will be studied further in the next year. Some emphasis will also be placed on steel. An aluminum reference block rejected by the manufacturer during fabrication is currently under close examination. Correlations are being sought between ultrasonic response anomalies and microstructural features. Techniques for residual stress measurements, preferred orientation measurements and microstructural measurements have been refined and are being applied in this task.
### TABLE 2 - STANDARDIZATION POINTS FOR ALUMINUM BLOCKS

<table>
<thead>
<tr>
<th>Block Hole Size</th>
<th>Test Frequency</th>
<th>Ball Diameter</th>
<th>Amplitude</th>
</tr>
</thead>
<tbody>
<tr>
<td>in</td>
<td>mm</td>
<td>in</td>
<td>mm</td>
</tr>
<tr>
<td>0.047(#3) 1.19</td>
<td>2.25</td>
<td>0.0625</td>
<td>1.588</td>
</tr>
<tr>
<td>0.047 1.19</td>
<td>5.0</td>
<td>0.1875</td>
<td>4.762</td>
</tr>
<tr>
<td>0.047 1.19</td>
<td>10.0</td>
<td>0.2812</td>
<td>7.144</td>
</tr>
<tr>
<td>0.078(#5) 1.98</td>
<td>2.25</td>
<td>0.1250</td>
<td>3.175</td>
</tr>
<tr>
<td>0.078 1.98</td>
<td>5.0</td>
<td>0.4375</td>
<td>11.112</td>
</tr>
<tr>
<td>0.078 1.98</td>
<td>10.0</td>
<td>0.6250</td>
<td>15.875</td>
</tr>
<tr>
<td>0.125(#8) 3.18</td>
<td>2.25</td>
<td>0.3125</td>
<td>7.938</td>
</tr>
<tr>
<td>0.125 3.18</td>
<td>5.0</td>
<td>1.0000</td>
<td>25.400</td>
</tr>
<tr>
<td>0.125 3.18</td>
<td>10.0</td>
<td>1.0625</td>
<td>26.988</td>
</tr>
</tbody>
</table>
Figure 4 — Distance – amplitude data for No. 3 blocks at 2.25 MHz

STD PT. 0.0625 in (1.588 mm)
ball = 1.2 V
Water Distance = 2.5 in (64 mm)
Figure 5 - Distance-amplitude data for No. 5 blocks at 2.25 MHz.
Figure 6 - Distance-amplitude data for No. 8 blocks at 2.25 MHz.
Figure 7 - Distance-amplitude data for No. 3 blocks at 5.0 MHz.
Figure 8 - Distance-amplitude data for No. 5 blocks at 5.0 MHz

STD. PT. - 0.4375 in (11.12 mm)
ball = 1.2 V
Water distance = 5.0 in (127 mm)
Figure 9 – Distance–amplitude data for No. 8 blocks at 5.0 MHz.

Metal Distance, mm

STD. PT. – 1.000 in (25.40 mm)
ball = 1.20 V
Water Distance – 5.0 in (127 mm)
Figure 10 - Distance—amplitude data for No. 3 blocks at 10 MHz.
Figure 77 — Distance-amplitude data for No. 5 blocks at 10 MHz.

STD. PT—0.625 in (15.875mm) ball= 0.60 V
Water Distance= 2.5 in (64 mm)
Figure 12 — Distance—amplitude data for No. 8 blocks at 10 MHz.
Figure 13 - RF signals and spectra from two nominally identical 5-0275 reference blocks at three frequencies.
Figure 13 - (con't).
Figure 13 - (con't).
3.4.1 Orientation Texture

Measurements of preferred orientation textures have been conducted as part of this study in order to determine the degree of nonrandomness and variability present in relevant specimens. A brief description of the texture measurement process is as follows. The data-collecting X-ray scaler has a memory which permits the accumulation of new diffraction data while the previous collection is transferred to punched tape. The pole figure device operates in the spiral mode (reflection method) which involves the rotation of the sample in its plane (alpha angle) every 16 minutes while moving off the center 5° (beta angle).

Computer programs in BASIC language were written to manipulate the diffraction data gathered using the pole figure device on the diffractometer. The programs developed for the data analysis and plotting (called Poleft) are included as Appendix C and are annotated. The X-ray data were corrected for background and the strongest point assigned a value of 100. A Fourier series was fitted to the data points in groups and the beta angle then interpolated for each intensity value from 10 to 90 at intervals of 10. The program contains an algorithm which converts the polar coordinates to cartesian coordinates for plotting the pole figure. This data is written into a computer file which is utilized by a program called Polepl (see Appendix D) to plot the pole figure on a stereographic projection. An automatic X-Y plotter which permits conversational interaction during plotting was used.

Studies have been conducted on several aluminum alloy and titanium specimens cut from sheet stock. These specimens were used for technique development and to determine the range of measured variables. Two pole figures made from a specimen cut from an aluminum sheet, 7075-T631, are shown. The (200) pole figure, Figure 14, indicates a maximum intensity (density of poles) in the center, falling off less rapidly in the direction of rolling than in the transverse direction. A random orientation of grains in the sample would lead to a uniform pole figure having a constant intensity level. In this case a tendency is present for grain orientations such that (200) planes in those grains are parallel to the sheet surface. The maximum pole density (100 relative units) lies in the center corresponding to the normal to the sample surface. The (111) pole figure for this specimen, Figure 15, shows a secondary maximum (50 units) in the center but the maximum intensities are at about 55° from the center in the transverse direction. This value is the angular distance between the (100) and (111) planes; the two pole figures are consistent with each other. This aluminum alloy sheet, therefore, shows a strong preferred orientation.

Pole figure determinations were made on several titanium sheets. A (10.1) pole figure made on a 50% reduced sheet is shown in Figure 16. The pole figure shows a four-fold symmetry with the maximum intensities occurring at about 35° from the center. Texture measurements were also made of the same sheet in the annealed condition, 1355 °F (735 °C), 5 min. The maximum intensity in the (00.2) pole figure, Figure
Figure 14 - (200) pole figure for 7075-T631 aluminum.
Figure 15 - (111) pole figure for 7075-T631 aluminum
Figure 16 - (10.1) pole figure for commercial titanium sheet
17, occurred in the center. In the (10.1) pole figure, Figure 18, the maximum occurred at about 50° from the center. Comparison of Figures 16 and 18 indicates the strong differences in texture that can be expected in titanium as a result of different mechanical and thermal treatments.

X-ray diffraction measurements were taken from one surface of the slice (see Figure 22) sectioned from the rejected aluminum ultrasonic reference block. The pole figures corresponding to reflections (200), (111), and (220) are included with this report. The axis of the block is located at the center of these pole figures. The (200) pole figure, Figure 19, shows two-fold symmetry with the maximum intensity in the center. The intensity falls off to less than 10 units at a deviation of 5 degrees from the axis. Secondary maxima of 40 are located at 180 degrees to each other at a deviation from the axis of approximately 25 degrees. The relatively high intensities of the (200) poles at the axis of the slice imply a high density of (200) poles on the cylindrical surface of the block. This axial orientation texture is very strong as indicated by the rapid decrease in pole density within 5 degrees of the axis.

The (111) pole figure, Figure 20, shows maxima at 35 and 55 degrees from the axial position. These maxima would be expected at these locations on the basis of the (200) pole figure. The (111) pole density at 55 degrees from the center shows pseudo four-fold symmetry indicating that the (200) preferred orientation in the center has a secondary preferred orientation and is not distributed randomly about the axial position. This preferred orientation of the (111) poles has also been noted in the residual stress measurements since the intensities of peaks measured at other than zero inclination angle are found to vary with the rotation of specimens. The (222) peak is used with other diffraction peaks in obtaining the residual stress data.

The principal feature of the (220) pole figure, Figure 21, is the occurrence of maxima in restricted belts about 22.5 degrees from the center of the figure. These must be related to the (200) poles which reach secondary maxima in two large areas about 22.5 degrees from the center of the (200) pole figure. Secondary maxima of the (200) poles occur at 45 degrees and have four-fold symmetry. These are related to the very sharp maximum occurring in the center of the (200) pole figure.

This pole figure information suggests that this reference block has secondary preferred orientation around its axis. The texture must have occurred from early fabrication of the rod. Since ultrasonic attenuation is sensitive to crystal orientation in stressed crystals, then this texture may be contributing significantly to the ultrasonic response of the block. Variations, if any, of the texture throughout the block will be sought.

There is some scatter in the center of the pole figures which is due to grain size. Even with 0.6 in (15 mm) oscillation of the speci-
Figure 17 - (00.2) pole figure for annealed commercial titanium sheet.
Figure 18 - (10.0) pole figure for annealed commercial titanium sheet.
Figure 19 - (200) pole figure for slice from aluminum reference block.
Figure 20 - (111) pole figure for slice from aluminum reference block.
Figure 21 - (220) pole figure for slice from aluminum reference block.
men during X-ray measurements, the scatter is largest at low beta angles but disappears when the beta angle has passed 20 to 25 degrees. At increasing beta angle for the same slits, a larger area of the specimen is covered by the X-ray beam. This effect may also be due to the grain shape, that is, the diffracting planes examined may have spread out further parallel to the surface of the specimen than in other directions if the grains are elongated in the surface plane.

3.4.2 Reference Block Microstructure

The aluminum reference block rejected by the manufacturer after fabrication was sectioned after preliminary acoustic inspection in order to examine the uniformity of metallurgical microstructure throughout the block. As indicated in Figure 22, the block was sectioned into three principal parts. The end containing the flat-bottomed hole was cut off at a length of 1.2 in (30 mm) and subsequently, a slice 0.16 in (4 mm) thick was taken from the surface opposite the flat-bottomed hole for texture measurements. The remaining block, 2.4 in (61 mm) long, was examined ultrasonically and then sectioned into two portions, each 1.2 in (30 mm) long. These two portions were ultrasonically inspected in detail. All cuts were carefully made perpendicular to the axis using a narrow, thin circular saw blade. The newly cut surfaces were then metallographically polished using a series of progressively finer abrasives, finishing with 40 μm (1 μm) diamond followed by MgO powder. Care was taken to minimize deviations from a flat surface and rounding at the edges.

Two of these new surfaces were examined metallographically. Several etching solutions were used. The results did not differ substantially. Figure 23 is an optical micrograph of an as-polished surface (S2). Many voids and cavities are seen there. Etching the polished surface reveals the grain structure and other phases that are present in this alloy, Figure 24. The grain diameters generally are in the range from 4 to 20x10^-4 in (10 μm to 50 μm). At longer etching times another feature emerges in many of the grains as shown in Figure 25. "Star-like" features appear within the grains and are probably due to composition variations arising from solidification structures that remain from the initial ingot stage. This surface (S2) was lightly polished mechanically and reexamined without further etching. Figure 26 shows the remaining grain boundary outlines and many examples of voids and second phase regions in the alloy. At higher magnification, details can be seen in several of the second phase regions marked as A in Figure 27. The discrete pitting reactions at the grain boundaries (rather than continuous, uniform etching) suggest that discrete precipitates lie along the boundaries in nonuniform distributions.

The microstructure seen on the section surfaces is believed typical of the entire block. It is complex and nonuniform, containing many voids, cavities, foreign phase regions, and possible inclusions. Some original solidification structure remains, including possible alloy composition gradients. These structures would be expected to affect ultrasonic wave propagation and the lack of homogeneity in structure would produce nonuniform
Figure 22—Ultrasonic and residual stress data on rejected aluminum reference block.

Note 1 Pa = 1.45 x 10^-4 lbf/in², 1 mm = 0.039 in.
Figure 23 - Optical micrograph of surface S2 as polished.

Figure 24 - Optical micrograph of surface S2 after first etching.
Figure 25 - Optical micrograph of surface S2 after longer etching.

Figure 26 - Optical micrograph of surface S2 after repolishing.
Figure 27 - Higher magnification optical micrograph of surface S2 after repolishing.
ultrasonic response. Studies of other reference block specimens should be conducted to determine how frequently such nonuniform structures are found.

3.4.3 Ultrasonic Inspection of Rejected Block

The two 1,2 in (30 mm) blocks were inspected ultrasonically. They were scanned using a nominal 10 MHz longitudinal beam transducer in an immersion tank. Each was first scanned such that the area between the top surface of the block and the bottom surface was displayed. No acoustic anomalies were observed on these scans. It was suspected that the material was not uniformly attenuating. Therefore, a delayed presentation of the amplitude of the first back reflection was observed as the block was scanned. Water path distance was maintained at 3.0 in (7.6 cm), during the scanning operation. Scanning increments were set at 0.030 in (0.076 cm). Seventy-five scans were needed to traverse the 2 in (5 cm) blocks because of transducer beam spreading. Each block was scanned in two orthogonal directions. An arbitrary amplitude of +60 mV was chosen as the norm in checking for attenuation uniformity. Amplitude losses greater than 1/3, i.e., signals less than +40 mV were noted at several locations.

Signal amplitude increases (>60 mV) were also noted, particularly on block 1, Figure 22. Waveform and spectrum photographs (Figures 28 and 29 respectively) were taken at particular locations using the techniques described in Section 3.3. Further information is given in Table 3. Using a nominal 5 MHz search unit, amplitude losses were less than 15% and no location information was quantified.

3.4.4 Residual Stress Measurements

The stress measurements were made using the method outlined in SAE TR-182, "Measurement of Stress by X-rays", [8]. However, the determination of the peak position is done differently. The alpha-1 peak is separated by using a modified method of that outlined by Gangulee [9] in the separation of $\alpha_1 - \alpha_2$ doublets. The new points near the center are fitted to a parabola and the parabola maximum is taken as the peak position. The annotated program used to calculate the stress is included as Appendix E.

The residual stress results from the sectioned block surfaces are summarized in Figure 22 and are shown in more detail in Appendix E. The first results were obtained by fitting a parabola to five equally spaced points. The measurements were made using the 222 diffraction indices using chromium radiation. The intensity of the peaks varied with the angle of inclination and with rotation of the specimen. The reason for this effect is quite evident in view of the texture shown in the (111) pole figure, Figure 20. The texture caused some problems, especially at the 15° angle of inclination, in getting useful data. It is felt that separation of the doublets will give better results in the interpretation of data with widely varying intensities. There
Figure 28 - Waveforms and spectra of back surface reflections of block 1. See Table 3 and Figure 22.
Figure 28 - (con't).

(c) point 3
(d) point 4
Figure 29 - Waveforms and spectra of back surface reflections of block 2. See Table 3 and Figure 22.
Figure 29 - (con't).

(c) point 3
### Table 3 - Rejected Block Response

<table>
<thead>
<tr>
<th>Block</th>
<th>Location No.</th>
<th>Figure No.</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
<td>(1)</td>
<td>28 (a)</td>
<td>&quot;normal&quot; area (60 mV peak)</td>
</tr>
<tr>
<td>1</td>
<td>(2)</td>
<td>28 (b)</td>
<td>amplitude loss (36 mV peak) 40% loss in crosshatched area</td>
</tr>
<tr>
<td>1</td>
<td>(3)</td>
<td>28 (c)</td>
<td>amplitude gain (65 mV peak)</td>
</tr>
<tr>
<td>1</td>
<td>(4)</td>
<td>28 (d)</td>
<td>amplitude gain (75 mV peak)</td>
</tr>
<tr>
<td>2</td>
<td>(1)</td>
<td>29 (a)</td>
<td>&quot;normal&quot; area (60 mV peak)</td>
</tr>
<tr>
<td>2</td>
<td>(2)</td>
<td>29 (b)</td>
<td>amplitude loss (30 mV peak) 50% loss in crosshatched area</td>
</tr>
<tr>
<td>2</td>
<td>(3)</td>
<td>29 (c)</td>
<td>amplitude loss (50 mV peak) 15% loss</td>
</tr>
</tbody>
</table>

**Equipment Settings:**

- **Pulser/Receiver:**
  - Rep Rate: #3
  - Voltage: (150; .1)
  - Damping: min.
  - Filter: 3
  - Gain: 26.5

- **Spectrum analyzer**
  - C.F.: 10 MHz
  - Gain: linear
  - B.W.: 100 kHz
  - 1mV/div x .25
  - Atten: 0
  - Filter: 0
seems to be a significant variation in residual stress measured on sectioned surfaces of this block. Further investigation of additional surfaces, of the effect of surface preparation, and of other materials is indicated.

3.5 Fabrication Considerations

Two areas related to the physical (non-metallurgical) fabrication of reference blocks are being studied. The first is a study of the critical dimensions in E 127 blocks. Arrangements have been made with the Dimensional Technology Section at NBS to inspect selected blocks for hole diameter, hole depth, surface finish of the hole bottom, parallelism of the hole bottom and top surface, and corner radius. Blocks tested in Task 3 that exhibit anomalous response and are available for destruction will be tested for anomalous physical dimensions. Further work in this area is deferred pending further progress on Task 3. In a related experiment, eighteen No. 5 blocks, six each with 0.50, 3-00 and 5.75 in (12.7, 76.2, and 146.0 mm) metal travel distance were machined at NBS with the E 127-64 tolerances specified. The material was 7075-T651 aluminum alloy, of unknown origin, except that it was all from one heat. This temper was used because a supply of material from one heat was readily available, and hopefully metallurgical variables could be minimized. After the cylinders were machined, but before the flat-bottomed holes were drilled, the cylinders were inspected ultrasonically at 5 MHz to determine material uniformity. Among sets of nominally identical blocks, the back surface response along the cylinder centerline varied by no more than 10 percent among the six blocks. After the flat-bottomed holes were drilled, the blocks were cleaned and plugged temporarily per E 127, and inspected ultrasonically at 2.25, 5.0, 10.0, and 15.0 MHz. The results of these tests are given in Table 4. At 5 MHz, the scatter between similar blocks was always less than ±7 percent, most of which is attributed to material nonuniformity. At 2.25 MHz, the scatter is even less, but at 10 and 15 MHz it is somewhat greater. It therefore appears that, at least in this case, the machining of the flat-bottomed holes did not introduce significant disparity into the measurements. Dimensional metrology will be used to determine if the dimensions of these blocks are significantly more uniform than required by the tolerances as specified in ASTM E 127. Blocks with No. 3 and 8 holes will be similarly fabricated and checked.

A second subtask relating to fabrication is a feasibility study of making two piece blocks. If it is determined that inaccuracies in the dimensions of the flat-bottomed hole are a cause of ultrasonic variability, it may be beneficial to fabricate the reference block from two cylinders, one solid and one containing a through hole. This would greatly facilitate both the machining and metrology processes. The two cylinders would then be connected by an ultrasound-transmitting bond, such as wringing or diffusion bonding. The latter has been reported to be feasible for titanium, e.g. [4], but as yet no work on this has been done on the current program. Some experiments have been performed on wrung pieces of steel, aluminum, and quartz, the latter with a view toward Task 8. Test pieces with very flat surfaces (0.5 fringe or better)
<table>
<thead>
<tr>
<th>Block Size and Number</th>
<th>Ultrasonic Response, volts</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Test Frequency, MHz</td>
</tr>
<tr>
<td></td>
<td>2.25</td>
</tr>
<tr>
<td>5-0050-1</td>
<td>1.00</td>
</tr>
<tr>
<td>-2</td>
<td>1.02</td>
</tr>
<tr>
<td>-3</td>
<td>1.02</td>
</tr>
<tr>
<td>-4</td>
<td>1.02</td>
</tr>
<tr>
<td>-5</td>
<td>1.05</td>
</tr>
<tr>
<td>-6</td>
<td>1.03</td>
</tr>
<tr>
<td>5-0300-1</td>
<td>1.19</td>
</tr>
<tr>
<td>-2</td>
<td>1.27</td>
</tr>
<tr>
<td>-3</td>
<td>1.30</td>
</tr>
<tr>
<td>-4</td>
<td>1.22</td>
</tr>
<tr>
<td>-5</td>
<td>1.20</td>
</tr>
<tr>
<td>-6</td>
<td>1.20</td>
</tr>
<tr>
<td>5-0575-1</td>
<td>1.32</td>
</tr>
<tr>
<td>-2</td>
<td>1.30</td>
</tr>
<tr>
<td>-3</td>
<td>1.40</td>
</tr>
<tr>
<td>-4</td>
<td>1.34</td>
</tr>
<tr>
<td>-5</td>
<td>1.40</td>
</tr>
<tr>
<td>-6</td>
<td>1.30</td>
</tr>
</tbody>
</table>
and very fine surface finishes (2μ in or better) were fabricated and wrung together using dimensional gage block techniques. Ultrasonic data was taken at 5 MHz, and in some cases in the steel and quartz, the reflected energy received from the interface was less than 10 percent of that received from the back surface. No success has been achieved with the aluminum. This work will be continued or dropped, depending on whether or not hole geometry is found to be a significant cause of variability.

3.6 Effects of Ultrasonic Measuring Systems

Using current standardization procedures, consistent quantitative measurements from various systems are not possible unless the ratio of responses from two references is the same on both systems. For example, an area-amplitude set of blocks that is linear on the block manufacturer's system must also be linear on the user's system in order to be useful. To determine what effects different test instruments have on the relative response of aluminum blocks, an intercomparison of data was made between NBS and the Reynolds Metals Company. Data was taken on three distance-amplitude sets using the same 5 MHz, 0.375 in (9.5 mm) diameter quartz search unit and the same test procedures, but with different instruments, although the same model.

The data from the two labs are presented in Table 5. The variability between systems, including operator error, is in general less than 10 percent.

In addition, three runs were made at NBS on one set of blocks using the same test system but with different operators in an attempt to quantify operator error. This data is given in Table 6. The maximum deviation between readings was less than 5 percent of the average of three readings except for one point down to the values where the minimum resolvable increment was greater than 5 percent of the reading.

3.7 Single Material Standard

The feasibility of using a material with no grain structure, high homogeneity and good inspectability such as fused quartz or crown glass, as a single material standard is being considered. This idea was well received by the attendees at the NBS NDE Public Review and Workshop in December 1974. This approach might require the development of transforms relating the acoustic impedance, attenuation, and sound speeds of the master blocks and structural materials. Significant effort on this task is planned.

3.8 ASTM Participation

The NBS investigators have joined and become active in ASTM Committee E-7 on Nondestructive Testing, Sub Committee E-7.06 on Ultrasonics, and particularly section E-7.05.02 on Aluminum Reference Blocks. Close contact has been maintained with the chairman of E-7.05.02 and considerable
TABLE 5 - RESULTS OF DATA INTERCOMPARISON ON ULTRASONIC BLOCKS. SEARCH UNIT-5 MHz, 0.375 in QUARTZ (SN 50A 1338) WATER DISTANCE = 3.5 inches.

<table>
<thead>
<tr>
<th>Metal Distance</th>
<th>#3 Blocks</th>
<th>#5 Blocks</th>
<th>#8 Blocks</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Lab A</td>
<td>Lab B</td>
<td>Lab A</td>
</tr>
<tr>
<td>STD PT-0050</td>
<td>100</td>
<td>100</td>
<td>100</td>
</tr>
<tr>
<td>-0075</td>
<td>83</td>
<td>80</td>
<td>84</td>
</tr>
<tr>
<td>-0100</td>
<td>67</td>
<td>67</td>
<td>67</td>
</tr>
<tr>
<td>-0125</td>
<td>54</td>
<td>58</td>
<td>56</td>
</tr>
<tr>
<td>-0175</td>
<td>37</td>
<td>40</td>
<td>38</td>
</tr>
<tr>
<td>-0225</td>
<td>27</td>
<td>25</td>
<td>30</td>
</tr>
<tr>
<td>-0275</td>
<td>20</td>
<td>23</td>
<td>20</td>
</tr>
<tr>
<td>-0325</td>
<td>16</td>
<td>17</td>
<td>16</td>
</tr>
<tr>
<td>-0375</td>
<td>12</td>
<td>13</td>
<td>14</td>
</tr>
<tr>
<td>-0425</td>
<td>11</td>
<td>10</td>
<td>11</td>
</tr>
<tr>
<td>-0475</td>
<td>9</td>
<td>9</td>
<td>10</td>
</tr>
<tr>
<td>-0525</td>
<td>8</td>
<td>7</td>
<td>8</td>
</tr>
<tr>
<td>-0575</td>
<td>7</td>
<td>5</td>
<td>7</td>
</tr>
</tbody>
</table>

51
TABLE 6 - EVALUATION OF OPERATOR VARIABILITY. BLOCK SET 150-3 (#5 BLOCKS), 5 MHz 0.375 in QUARTZ SEARCH UNIT

<table>
<thead>
<tr>
<th>Block No.</th>
<th>Operator A</th>
<th>Operator B</th>
<th>Operator C</th>
<th>Avg.</th>
<th>Spread Among 3 Runs</th>
<th>Spread Avg. %</th>
</tr>
</thead>
<tbody>
<tr>
<td>5-0050</td>
<td>100</td>
<td>100</td>
<td>100</td>
<td>100</td>
<td>-</td>
<td>-</td>
</tr>
<tr>
<td>-0062</td>
<td>88</td>
<td>86</td>
<td>88</td>
<td>87.3</td>
<td>2</td>
<td>2.3</td>
</tr>
<tr>
<td>-0075</td>
<td>84</td>
<td>80</td>
<td>80</td>
<td>81.3</td>
<td>4</td>
<td>4.9</td>
</tr>
<tr>
<td>-0088</td>
<td>75</td>
<td>72</td>
<td>72</td>
<td>73</td>
<td>3</td>
<td>4.1</td>
</tr>
<tr>
<td>-0100</td>
<td>67</td>
<td>66</td>
<td>65</td>
<td>66</td>
<td>2</td>
<td>3.0</td>
</tr>
<tr>
<td>-0125</td>
<td>56</td>
<td>54</td>
<td>55</td>
<td>55</td>
<td>2</td>
<td>3.6</td>
</tr>
<tr>
<td>-0175</td>
<td>38</td>
<td>36</td>
<td>37</td>
<td>37</td>
<td>2</td>
<td>5.4</td>
</tr>
<tr>
<td>-0225</td>
<td>30</td>
<td>29</td>
<td>29</td>
<td>29.3</td>
<td>1</td>
<td>3.4</td>
</tr>
<tr>
<td>-0275</td>
<td>20</td>
<td>21</td>
<td>21</td>
<td>20.7</td>
<td>1</td>
<td>4.8</td>
</tr>
<tr>
<td>-0325</td>
<td>16</td>
<td>16</td>
<td>17</td>
<td>16.3</td>
<td>1</td>
<td>6.1</td>
</tr>
<tr>
<td>-0375</td>
<td>14</td>
<td>13</td>
<td>13</td>
<td>13.3</td>
<td>1</td>
<td>7.5</td>
</tr>
<tr>
<td>-0425</td>
<td>11</td>
<td>11</td>
<td>11</td>
<td>11</td>
<td>0</td>
<td>0</td>
</tr>
<tr>
<td>-0475</td>
<td>10</td>
<td>10</td>
<td>9</td>
<td>9.7</td>
<td>1</td>
<td>10.3</td>
</tr>
<tr>
<td>-0525</td>
<td>8</td>
<td>7</td>
<td>8</td>
<td>7.7</td>
<td>1</td>
<td>13.0</td>
</tr>
<tr>
<td>-0575</td>
<td>7</td>
<td>6</td>
<td>7</td>
<td>6.7</td>
<td>1</td>
<td>14.9</td>
</tr>
</tbody>
</table>
consultation has taken place regarding the revision of E 127 scheduled for ballot by E-7.06 and E-7 later this year. Experiments were performed to verify the validity of using a universal distance-amplitude curve to replace the three curves currently used (Figure 6 of [1]). If different standardization points are used for different size blocks, their responses can be compared to a single curve of higher amplitude than the number 3 and number 5 curves in the current document. In the current document, the maximum response expected from a number 3 hole is only 12 percent of the scope vertical linear limit, and resolution becomes a problem.

The data from three sets of blocks, one each No. 3, 5, and 8, when plotted on a universal distance-amplitude curve basis, are shown in Figure 30. The scatter between these data appears to be no worse than the scatter between data from blocks of the same size (Figures 4-12). Continued, long-term participation in activities of these groups is planned.

4. IMPORTANT CONCLUSIONS TO DATE

Based on the work completed to January 1975 the following conclusions are drawn:

1) No previous work has isolated the cause of block variability. The problems of dimensional, metallurgical and fabrication considerations must all be attacked. Work in the United Kingdom has suggested it would be difficult to fabricate blocks with less than ±1 db variability but that this tolerance can be achieved with assigned correction factors ("calibration").

2) Among the blocks evaluated to date, the "average" variation between nominally identical aluminum blocks is about 20-30 percent, but variations as high as 700 percent have been recorded.

3) Metallurgical studies were conducted on an aluminum block rejected by the manufacturer during fabrication. The block contained a high degree of preferred orientation texture, probably occurring as a result of the fabrication processing of the rod from which the block was made. The block microstructure was complex; voids, second phase regions and chemical concentration variations were all present. Significant variations in residual stress in this block were also found. All these factors probably contribute to the measured variation from -50 to 25 percent around the average back surface ultrasonic response of this block.

4) Efforts to manufacture two-piece blocks wrung together have met with mixed success. Some success has been achieved with steel and quartz, little with aluminum. One-piece aluminum blocks have been fabricated at NBS from a uniform lot of material. The spread among six nominally identical blocks was less than 10 percent for three different sets.
Figure 30—Universal distance-amplitude curves for three sets of aluminum reference blocks.
5) The variability between data taken by three operators using the same blocks and the same equipment was measured to be less than about 5 percent. The deviations between readings from two operators using the same blocks and the same search unit but different systems was less than 10 percent.
APPENDIX A

Equipment Specifications

Some of the important specifications for the equipment identified in Section 3.2 are given below:

1. Ultrasonic Immersion System

a) Tank Dimensions:

<table>
<thead>
<tr>
<th></th>
<th>Length</th>
<th>Width</th>
<th>Depth</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>38 in</td>
<td>21 in</td>
<td>18 in</td>
</tr>
<tr>
<td></td>
<td>97 cm</td>
<td>53 cm</td>
<td>46 cm</td>
</tr>
</tbody>
</table>

b) Bridge and Carriage

Operating in the automatic scanning mode the bridge and carriage assembly is controllable within the following limits:

(1) Bridge Indexing
- adjustable from 0.001 in (0.03 mm) to 0.099 in (2.5 mm) in 0.001 in (0.03 mm) increments.

(2) Bridge Travel
- at least 38 in (97 cm)

(3) Carriage Speed
- continuously adjustable from approximately 0.5 in (1 cm) to 15 in (38 cm) per second.

(4) Carriage Travel
- adjustable from approximately 4.0 in (10.2 cm) to 12 in (30 cm) in 0.5 in (1 cm) increments.

c) Motorized Manipulator and Search Tube

(1) Vertical Indexing
- adjustable from 0.001 in (0.03 mm) to 0.099 in (2.5 mm) in 0.001 in (0.03 mm) increments.

(2) Vertical Travel
- at least 17.0 in (43 cm).

d) Auxiliary Manipulator

This mini-manipulator provides angular adjustment in two right angle vertical planes with tilt ranges of ±30 degrees. Uncertainty in angular adjustments is less than 1 degree.
e) X-Y Recorder

The X-Y recorder is a dry paper type using electrosensitive paper with an electrostatic paper hold down. The pen is mechanically driven to provide approximately 1 to 1 recordings.
   - platen size - 11 by 17 in (28 by 43 cm).

f) Temperature Control

An immersion heater with thermostatic control provides the capability of maintaining temperature in the 70 °F to 80 °F (21 °C to 27 °C) range with a time variation of 1 °F (.05 °C).

g) Water System

An internal water system consisting of a pump, filter, and water skimmer is provided.

2. Broadband Ultrasonic System

a) Pulser/Receiver

   (1) Pulser
      - output voltage selectively variable from 40 to 350 V into 50 ohms.
      - rise time - 5 - 15 ns measured between 10 and 90 percent amplitude points.
      - pulse width - 15 - 150 ns measured full width at half amplitude.
      - frequency - 0 - 30 MHz.
      - repetition rate - 500 to 5000 Hz internal oscillator; 0 to 10000 Hz external source.
      - damping resistance - 5 to 500 ohms.

   Typical pulses are shown in Figure A-1. These were taken at the narrowest pulse width setting with minimum damping. Figure A-1 represents the pulse used under normal operating procedures documented in this work.

   (2) Receiver
      - frequency range - 0 to 30 MHz
      - input impedance - 500 ohms
      - gain 10 to 70 dB
      - voltage output - maximum 2.5 V peak-to-peak

b) Gate

   - eliminates unwanted signals from a repetitive pulse train to isolate the desired wave packet without distorting the wave packet.
   - delay range - 0.2 to 1000 µs.
   - width range - 0.2 to 100 µs.
- switching transients - less than 10 mV (see Figure A-2 for measured transients).
- bandwidth - 0.2 to 50 MHz.

Several commercially available "stepless" gates were evaluated. The unit chosen represents the most versatile gate with the required specifications in gate delay, width and minimized switching transients. Switching transients of those evaluated typically ranged from less than 10 mV for the laboratory system to 50 mV using an inexpensive double balanced mixer.

c) Peak Detector
- converts the peak amplitude of ultrasonic pulses to a proportionate DC voltage in both the linear and logarithmic mode.
- input range - 0.01 to 1.0 V positive.
- input pulse width - 20 ns minimum.
- linearity - ±5 percent of peak amplitude or ±2 mV at input, whichever is greater.
- linear gain - adjustable from 0.5 to 16 times the input.
- logarithmic gain - adjustable from 40 to 1.25 dB full scale.
- DC offset - 0 to 5 V or 0 to 40 dB.
- output voltage - 0 to 1 VDC into 1000 ohms.
- decay time - 0.01, 0.1, and 1 seconds.

d) Quantizer
- enables a step-wise quantization of gated video signals into discrete DC voltages.
- input range - 0 to 10 V peak video signals.
- input pulse width - 200 ns minimum.
- quantization range - 5 to 80 increments into a total range of 40 dB.

3. Spectrum Analyzer

The spectrum analyzer consists of a storage CRT with separate IF and RF plug-in modules.
- frequency range - 0 to 110 MHz with adjustable center frequency.
- bandwidth - 0.01 to 300 kHz.
- scan width - 0.02 kHz to 10 MHz per division.
- scan time - 0.01 to 10 seconds per division.
- calibrated vertical reference level.
  - log +10 to -72 dBm per division.
  - linear 0.025 µV to 100 mV per division.
4. Search Units

Search units purchased for this work are listed in Table A-1 by crystal diameter and nominal center frequency. Both immersion and contact types are included. These represent but a sample of the available size-frequency combinations. Particular emphasis has been placed on the 0.375 in (0.953 cm) quartz crystal at 5.0 and 15.0 MHz in order to be compatible with ASTM E 127 specifications of 1964 and proposed modifications, as well as search unit size and frequency combinations suitable for use in Task 3.
TABLE A-1 - ULTRASONIC SEARCH UNITS

<table>
<thead>
<tr>
<th>Size in cm</th>
<th>Frequency (MHz)</th>
<th>2.25</th>
<th>5.0</th>
<th>10.0</th>
<th>15.0</th>
</tr>
</thead>
<tbody>
<tr>
<td>0.25</td>
<td>Immersion</td>
<td>4</td>
<td></td>
<td>1</td>
<td>1</td>
</tr>
<tr>
<td>0.375</td>
<td>Immersion</td>
<td>6</td>
<td>1</td>
<td>5</td>
<td></td>
</tr>
<tr>
<td>0.5</td>
<td>Contact</td>
<td>2</td>
<td>1</td>
<td>3</td>
<td>2</td>
</tr>
<tr>
<td>0.75</td>
<td>Contact</td>
<td></td>
<td></td>
<td>1(a)</td>
<td>1</td>
</tr>
</tbody>
</table>

(a) Focused
Figure A-1 - Typical pulse waveforms from laboratory pulser.

Figure A-2 - Typical "steps" (switching transients) in the gated output.
APPENDIX B

Organizations offering Loan of Reference Blocks

Aluminum Company of America
Battelle Memorial Institute
CBL Industries, Inc.
Curtiss-Wright Corporation
Kaiser Aluminum Company
Krautkramer-Branson, Inc.
LTV Aerospace Corporation
Met Lab Inc.
NASA, Lewis Research Center
Naval Research Labs
Naval Weapons Center
Pratt and Whitney Aircraft Company
Reynolds Metals Company
Westinghouse Electric Corporation
Wyman-Gordon Company
APPENDIX C

POLEFT Program

10 ' POLEFT REDUCES SCALER DATA OUTPUT FROM TEXTURE GONIOMETER
20 ' FOR USE IN PLOTTING POLE FIGURE BY POLEPL..
30 '###.## ####.## ####.## ####.## #### #### ##F##
40 t##.# ###.# ##.# ,##.# ### ### #.### #.### ##.# #^M.# ##.# #!^! +^## ###;
50 DIMYCISO86).#C(200)sD(500)sACSO)sBC50)sAS(72)sESC72)sBS(72)
60 DIN +I (too ),XC100)sZ(100),Q(100)
70 PRINT"TYPE IN INPUT AND OUTPUT FILE NAMES";
80 INPUT CS,DS ' PERMITS DESIGNATION OF FILES AT RUN TIME.
90 IF CS="STOP" THEN 1240
100 FILES CSaDS
110 SCRATCH#
120 INPUT#1sAS,BSsESsBOsB9 ' READ IN TITLE AND DESCRIPTION OF
130 PRINT AS 'JOBS BO=BACKGROUND CORRECTION, B9=HIGHEST
140 PRINT BS 'INTENSITY DUMP.
150 PRINT ES
160 MAT Q=ZER
170 E9=14 'E9=NUMBER OF COMPLETED REVOLUTIONS.( ALPHA AXIS)
180 X9=E9
190 E8=E6 'E8= NO. OF DUMPS / REVOLUTION
200 M6=360/(2*E9)
210 A1=5/E8
220 A2=A1/2
230 F=360/E8
240 B1=F/E8
250 P=3*141593/180
260 B9=B9-B0
270 PRINTB0:B9
280 FORN=ITCE9 ' READS DATA FROM FILE TO Y MATRIX
290 ForM=ITCE8
300 INPUT#1,Z
310 IFEND#1THEN350
320 Y(N,M)=(Z-BO)/B9*100
330 NEXTM
340 NEXTN ' END OF DATA READ.
350 FORM=1TCE8
360 FORM=1TCE9
370 C(N)=Y(N,M1)
380 C(2*E9+1-N)=Y(N,M1)
390 NEXTN
400 FORM=0TGX9 ' START OF CALCULATION OF FOURIER COEFFICIENTS.
410 C=M6/2
POLEFT Program

420 A(M)=0
430 B(M)=0
440 FOR N=1 TO E9*2
450 A(M)=A(M)+C(N)*SIN(P*M*C)
460 B(M)=B(M)+C(N)*COS(P*M*C)
470 C=0+M6
480 NEXT N
490 A(M)=A(M)/E9
500 B(M)=B(M)/E9
510 NEXT M ' FOURIER END.
520 BO=1000
530 B9=0
540 0=M6/2
550 S8=0
560 FOR N=X9+1 TO E9*X9 ' CALCULATES POINTS BETWEEN OBSERVED POINTS:
570 D(N)=B(C)/2
580 FORM=1+X9
590 D(N)=D(N)+A(M)*SIN(C*P*M)+B(M)*COS(C*P*M)
600 NEXT M
610 C=0+M6/X9
620 IF B9>D(N) THEN 650 ' FIND POSITION OF CROSS-OVER POINTS:
630 B9=D(N)
640 B3=N
650 IF B9<D(N) THEN 680
660 B4=N
670 B9=D(N)
680 NEXT N
690 W4=0
700 C=D(X9)
710 FOR N=X9+1 TO E9*X9
720 C1=D(N)
730 FORM=INT((C1+9.999)/10)*10+9STEP10
740 IFC=MTHEN820
750 IFC>M THEN 780
760 IFC1<=M THEN 830
770 GOT0820
780 IFC1>=M THEN 800
790 GOT0820
800 NEXT M
810 GOT0830
820 GOSUB1110
POLEFT Program

830 C=G1
840 NEXTN
850 FORN1=2TO4 "ROUTINE FOR ELIMINATING UNWANTED POINTS."
860 IFI(N1-1)<>=I(N1)THEN890
870 NEXTN1
880 GOTO1050
890 N5=N1-1
900 GOSUB 1210
910 N5=N1
920 GOSUB1210
930 N4=0
940 FORN2=N1+1TO4
950 IFI(N2-1)<>=I(N2)THEN980
960 N4=N4+1
970 GOTO1040
980 IF N4=0 THEN 1010
990 N5=N2-1
1000 GOSUB 1210
1010 N5=N2
1020 GOSUB1210
1030 N4=0
1040 NEXTN2
1050 NEXTM1
1060 FORN=1TO9 "PRINTS COUNT NO. OF POINTS AT EACH LEVEL.
1070 PRINTN;Q(N);
1080 NEXTN
1090 PRINT
1100 GOTO 70
1110 W4=W4+1 "ROUTINE TO FIND POLAR AND CARTESIAN COORDINATES."
1120 A=((N-1-X9)*5/X9*M1*A1)
1130 B=F*M1-B1
1140 D=10*TAN(P*A/B)
1150 X=D*COS(P*B)
1160 Y=D*SIN(P*B)
1170 X(W4)=X
1180 Y(W4)=Y
1190 I(W4)=M
1200 RETURN
1210 OUTPUT#2,1(N5);X(N5);Z(N5); OUTPUT TO FILE FOR PLOTTES.
1220 Q(I(N5)/10)=Q(I(N5)/10)+1 "COUNTS NO. OF POINTS AT EACH LEVEL."
1230 RETURN
1240 END
APPENDIX D

POLEPL Program

10 ' POLEPL PLOTS POINTS ON A POLE FIGURE FROM DATA STORED IN A
20 ' FILE CREATED BY POLEFT.
30 DATA 4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999,4999
40 PRINT" 	" PLTL"
50 FORM9=ITOS ' FORMS CENTER LINE FOR POLE FIGURE.
60 READAI,A1
70 IF Y9/P=INT(K9/9)THEN100
80 PRINTUSING99,A1,91
90 OUTC10
100 PRINTUSING1000,A1,91
110 NEXTM9
120 PRINT" " PLTT"
130 PRINT" ENTER NAME OF INPUT FILE";
140 INPUTA$ ' PERMITS DESIGNATION OF FILE AT RUN TIME CREATED
150 FILPAS ' BY POLEFT.
160 HEX INTENSITY PLOTS
170 DMC(11,3).P(11),C$(10)
180 LQ=3.392585093
190 P1=3.14159258
200 ' R=9
210 R6=6.49995
220 R4HEADDC(11,3) ' READS DATA FOR SYMBOLS AVAILABLE FOR PLOTTING.
240 DATA 270,90,120,90,270,90,120,90,270,90,120,90,270,90,120
250 DATA 420,90,120,90,420,90,120,90,420,90,120,90,420,90,120
260 R4HEADD(11) ' READ DATA TO DETERMINE SIZE OF SYMBOLS.
270 DATA 95,95,95,95,95,95,95,95,95,95,95,95,95,95,95,95,95,95,95,95
280 RH=7.5
290 R6=.753
300 RH=R749995/83
310 PRINT" RADIUS=""50/R6","POINT SIZE=""50/R6
320 W1=10*(INT(LQ(3)/LQ))
330 RH=INT(5+R6/W1)
340 FOR J=TOP
350 IFRA=(I+J)*TH:
360 NEXT J
370 W1=1*(I+W1)/10
380 :3=I+INT(5+R6/W1)
390 U9=
400 RA=0
410 U7=0
420 FORM9=INTC0STFP-10
430 PRINT"COLOR AND SYMBOL NO."
440 INPUT A$,J6 ' PERMITS SELECTION OF COLOR & SYMBOL FOR EACH
450 PRINT$J6 ' INTENSITY AT RUN TIME
460 IF J=THEN 490
470 H5=R6/P(J6)
480 N=J=$0
490 RESTOR#1
500 IFEND#1 THEN 770
510 INPUT1,19,17,17
520 IF 19<Y9 THEN 900 ' SEARCH FOR FILE FOR DATA WITH CERTAIN
530 N=N+1 ' INTENSITY VALUES IF PASSES TEST IN 510
POLEPL Program

540 IF J6<>0 THEN 660
550 U9=1
560 W8=X7
570 H8=Y7
580 GOSUB 860
590 IF U8=0 THEN 500
600 U9=1
610 IF U7=0 THEN 640
620 U7=0
630 PRINT "PLTL"
640 PRINT "POINT OFF SCALE. X ="; X7; " Y ="; Y7; " INTENSITY ="; I9
650 GOTO 500
660 U9=2
670 FOR C9=C(J6, 1) TO C(J6, 2) STEP C(J6, 3)
680 C8=C9*P1/180
690 W8=X7+R5*COS(C8)
700 H8=Y7+R5*SIN(C8)
710 IF J6>2 THEN 730
720 U9=1
730 130 SUB 60
740 IF U8=0 THEN 600
750 NEXT C9
760 GOSUB 860
770 IF U8=1 THEN 660
780 NEXT J6
790 GOSUB 860
800 PRINT "PLTL"
810 U7=0
820 IF N>0 THEN 840
830 PRINT "NO POINTS FOUND WITH"; I6; " <= INTENSITY <="; I7
840 NEXT M9
850 STOP
860 REMi SUBROUTINE 'SUBPLT' FOR PLOTTING
870 W9=INT(W6+W7*W8)
880 H9=INT(H6+H7*H8)
890 IF W9>9999 OR W9<0 OR H9>9999 OR H9<0 THEN 1020
900 IF U7<>0 THEN 930
910 U7=1
920 PRINT "PLTL"
930 IF U9=0 THEN 990
940 PRINT USING 950, W9, H9
950 t#####
960 IF U9=1 THEN 1030
970 U9=0
980 GOTO 1030
990 PRINT USING 1000, W9, H9
1000 t#####
1010 GOTO 1030
1020 U9=1
1030 RETURN
APPENDIX E

STRESS Program

5 'PROGRAM FOR CALCULATING STRESS
10 **=#### # # # # # # # # # # #
15 T9=.085959596
20 T8=.91782/T9
25 PRINT"ENTER NAME OF INPUT FILE"
30 INPUT$ 'PERMITS DESIGNATION OF FILE AT RUN TIME.
35 IFAS="STOP"THEN840
40 DIMC$[72],D$(72)
45 FILESAS
50 INPUT#1,C$,D$)
55 REMC$=D$
60 P9=&PI/180
65 Z9=0
70 M=0
75 INPUT#1,A1,S9 'READS STARTING ANGLE & ANGLE OF INCLINATION.
80 X(1)=T9/2
85 FORI=1TO10000 'START OF DATA READING ROUTINE.
90 INPUT#I;Y
95 IFENDTHEN135 'DETECTS FLAG IN FILE.
100 IFY<OTHEN135
105 S3=S9*P9
110 Y(I)=Y(I-1)-TAN(S9)*COT(Z9)/(1+TAN(S9)*COT(Z9))**2/(2*SIN(Z9)**2))
120 M=M+1 'ABOVE CALCULATES LP & PSI CORRECTIONS--SEE SAE TR-182
125 X(I+1)=X(I)+T9
130 NEXTI
135 FORI=1TOM
140 Y(2*M+1-I)=Y(I)
145 NEXTI
150 M=2*M
155 DIMC$(300),Q(300),A(300),B(300),C(300),D(300),X(300),I(600),Y(300)
160 P2&PI
165 FORN=0TOC/4 'ROUTINE FOR CALC. FOURIER COEF. SEE GANGULEE
170 P(N)=1+4*7*COS(P*N*T8/M)
175 Q(N)=47*SIN(P*N*T8/M)
180 M1=A=B=0
185 FORX=1TOM
190 XI=X-5
195 A=X+Y(X)*COS(P*N*X1/M)
200 B=B+Y(X)*SIN(P*N*X1/M)
205 NEXTX
210 A(N)=A/M2
215 B(N)=B/M2
STRESS Program

455 F1=W/W1
460 W1=W
465 GOTO 70
470 U(2)=U(2)-E2*U(3)
475 Z9=Z9+1
480 A2=U(2)/(2*U(3))-.5
485 A3=A2*T9/4+A1 'FINDS TWO-THETA ANGLE'
490 H(Z9)=S9
495 Z(Z9)=A3
500 PRINT USING 10, A2; S9, A3
505 PRINT"

" 510 IF Y<=-100 THEN 520
515 GOTO 70
520 "YOUNG'S MODULUS = #.#### N/MM$^2$; #.#### KG/MM$^2$; #.#### PD/IN$^3$"
525 "POISSON'S RATIO = #.####"
530 "STRESS COMPONENT = #.#### N/MM$^2$; #.#### KG/MM$^2$; #.#### PD/IN$^3$"
535 "PROBABLE ERROR OF LINEAR FIT = #.####"
540 "RELATIVE STANDARD DEVIATION OF STRESS = #.#### PER CENT"
545 "ERROR RELATIVE TO AVG. ORDINATE = #.#### PER CENT"
550 "STANDARD DEVIATION OF STRESS = #.#### N/MM$^2$"
555 PRINT"

" 560 C=D=E=F=S1=W=0
570 N=-33 'POISSON RATIO'
575 E1=10400000 'YOUNG'S MODULUS FOR 7075 AL• ALLOY'
580 E1=6894.75*E1
585 PRINTS
590 PRINT
595 PRINT
600 FOR I=1 TO Z9 'START OF STRESS ROUTINE'
605 S3=P9*H(I)
610 Y(I)=Z(I)-Z(I)
615 X(I)=SIN(S3)*SIN(S3)
620 C=C+Y(I)
STRESS Program

220 NEXTN
225 A(0)=A(0)/2
230 FORN=0 TO M/4 'ROUTINE FOR FINDING ALPHA1 FOURIER COEF.
235 E=P(N)*P(N)+Q(N)*Q(N)
240 C(N)=(A(N)*P(N)+Q(N)*B(N))/E
245 D(N)=(-A(N)*Q(N)+B(N)*P(N))/E
250 NEXTN
255 A3=0
260 FORX=2 TO M*2
265 XI=X+5
270 I=0
275 FORN=0 TO M/4 'CALCULATES ALPHA1 PEAK
280 I=I+C(N)*COS(P*N*X1/(M*4))+D(N)*SIN(P*N*X1/(M*4))
285 NEXTN
290 I(X)=I
295 IF A3=I THEN 310 'FIND HIGHEST POINT IN ALPHA1 PEAK
300 A3=I
305 A2=X
310 NEXTX
315 T7=T9/4*(A2-.5)+A1
320 PRINTUSING10,A2,S9,T7,T7+T8+T9
325 M=60
330 FORI=1 TO M 'START OF PARABOLA FITTING ROUTINE
335 Y(I)=I(A2-31+I)
340 X(I)=A2-31+I
345 P(I)=0
350 Q(I)=1
355 NEXTI
360 I=E2=F1=0
365 W1=M
370 W=E1=0
375 I=I+1
380 FORL=1 TO M
385 W=W+Y(L)*Q(L)
390 E1=E1+X(L)*Q(L)*Q(L)
395 NEXTL
400 U(I)=W/W1
405 IF I=3 THEN 470
410 E1=E1/W1
415 E2=E2+E1
420 W=0
425 FORL=1 TO M
430 V=(X(L)-E1)*Q(L)-F1*P(L)
435 P(L)=Q(L)
440 Q(L)=V
445 W=W+U*U
450 NEXTL

70
STRESS Program

665 D=D+X(I)*2
630 E=E+X(I)*Y(I)
635 F=F+X(I)
640 NEXTI
645 D5=D*Z9-F*F
650 Q1=(Z9*E-C*F)/D5
655 Q2=(C*D-E*F)/D5
660 PRINTUSING520,E1,E1/9806650,E1/6894.75
665 PRINT
670 PRINTUSING525,N
675 PRINT
680 S=-P9*Q1*E1/(2*(1+N)*TAN(*5*P9*Z(I)))
685 PRINTUSING530,S,S/9806650,S/6894.75
690 PRINT
695 IF S<0 THEN 710
700 PRINT " THIS IS A TENSILE STRESS IN THE PLANE OF THE SURFACE."
705 GOTO 715
710 PRINT " THIS IS A COMPRESSION STRESS IN THE PLANE OF THE SURFACE."
715 PRINT
720 PRINT
725 PRINT "DELTA 2*THETA FITTED TO LSE STRAIGHT LINE:";
730 PRINT " DEL 2TH = (";Q1;" + \times \sin(\psi_1)) \times (\sin(\psi_2)) + (";Q2;")\n735 PRINT
740 PRINT " ANGLE OF"
745 PRINT " INCLINATION 2 THETA \sin^2 DEL 2TH LSE FIT"
750 FOR I=1 TO Z9
755 Z=Q1*X(I)+Q9
760 PRINTUSING535,H(I),Z(I),X(I),Y(I),Z
765 SI=SI+(Y(I)-Z)^2
770 NEXTI
775 PRINT
780 S2=6754*SQR(S1/(Z9-2))
785 PRINTUSING540,S2
790 C=C/Z9
795 PRINTUSING550,ABS(100*S2/C)
800 PRINT
805 PRINT
810 S3=SQR(S1*M/(D5*(M-2)))
815 S4=ABS(S3*S/Q1)
820 PRINTUSING555,S4
825 PRINTUSING545,ABS(100*S4/S)
830 PRINT"

835 GOTO 25
840 END
RESIDUAL STRESS ANALYSIS

AL BLOCK #1; BOTTOM SIDE
SURFACE S3/2
YOUNG'S MODULUS = 7.2E+10 N/M²; 7.3E+03 KG/MM²; 1.04E+07 PD/IN²
POISSON'S RATIO = .33

STRESS COMPONENT=-1.2E+08 N/M²; -1.2E+01 KG/MM²; -1.7E+04 PD/IN²

THIS IS A COMPRESSIVE STRESS IN THE PLANE OF THE SURFACE.

DELTA 2*THETA FITTED TO LSE STRAIGHT LINE:
DEL 2TH = ( 1.16305 )*(SIN PSI)² + ( .122302 )

<table>
<thead>
<tr>
<th>ANGLE OF INCLINATION</th>
<th>2 THETA</th>
<th>SIN²</th>
<th>DEL 2TH</th>
<th>LSE FIT</th>
</tr>
</thead>
<tbody>
<tr>
<td>0</td>
<td>155.305</td>
<td>.000</td>
<td>.000</td>
<td>.122</td>
</tr>
<tr>
<td>15</td>
<td>155.522</td>
<td>.067</td>
<td>.217</td>
<td>.200</td>
</tr>
<tr>
<td>30</td>
<td>155.781</td>
<td>.250</td>
<td>.476</td>
<td>.413</td>
</tr>
<tr>
<td>45</td>
<td>156.204</td>
<td>.500</td>
<td>.899</td>
<td>.704</td>
</tr>
<tr>
<td>60</td>
<td>156.147</td>
<td>.750</td>
<td>.842</td>
<td>.995</td>
</tr>
</tbody>
</table>

PROBABLE ERROR OF LINEAR FIT = .111
ERROR RELATIVE TO AVG. ORDI NATE = 22.7 PER CENT

STANDARD DEVIATION OF STRESS = 2.7E+07 N/M²
RELATIVE STAND. DEV. OF STRESS = 22.6 PER CENT
RESIDUAL STRESS ANALYSIS

Al block #1, top side center surface S1
Young's modulus = 7.2E+10 N/m²; 7.3E+03 kg/mm²; 1.04E+07 pd/in²
Poisson's ratio = .33

Stress component = -1.1E+06 N/m²; -1.2E+01 kg/mm²; -1.6E+04 pd/in²

This is a compressive stress in the plane of the surface.

Delta 2*theta fitted to LSE straight line:
\[ \delta 2\theta = (1.09988)\sin\psi + 5.90986E-2 \]

Angle of inclination 2 theta sin² delta 2 theta lse fit

<table>
<thead>
<tr>
<th>Angle of Inclination</th>
<th>2 Theta</th>
<th>Sin²</th>
<th>Delta 2 Theta</th>
<th>LSE Fit</th>
</tr>
</thead>
<tbody>
<tr>
<td>0</td>
<td>155.225</td>
<td>.000</td>
<td>.000</td>
<td>.059</td>
</tr>
<tr>
<td>15</td>
<td>155.410</td>
<td>.067</td>
<td>.182</td>
<td>.133</td>
</tr>
<tr>
<td>30</td>
<td>155.585</td>
<td>.250</td>
<td>.357</td>
<td>.334</td>
</tr>
<tr>
<td>45</td>
<td>155.834</td>
<td>.500</td>
<td>.606</td>
<td>.609</td>
</tr>
<tr>
<td>60</td>
<td>156.102</td>
<td>.750</td>
<td>.874</td>
<td>.884</td>
</tr>
</tbody>
</table>

Probable error of linear fit = .032
Error relative to average ordinate = 7.8 per cent

Standard deviation of stress = 7.7E+06 N/m²
Relative standard deviation of stress = 6.8 per cent
RESIDUAL STRESS ANALYSIS

AL BLOCK, TOP, POLISHED NOT ETCHED
SURFACE S1
YOUNG'S MODULUS $= 7.2 \times 10^{10} \text{ N/m}^2; 7.3 \times 10^3 \text{ kg/mm}^2; 1.04 \times 10^7 \text{ PD/IN}^2$

POISSON'S RATIO $= 0.33$

STRESS COMPONENT $= -1.4 \times 10^8 \text{ N/m}^2; -1.4 \times 10^1 \text{ kg/mm}^2; -2.0 \times 10^4 \text{ PD/IN}^2$

THIS IS A COMpressive STRESS IN THE PLANE OF THE SURFACE.

DELTA 2*THETA FITTED TO LSE STRAIGHT LINE:
DEL 2TH $= (1.32547 \times \sin \psi)^2 + (0.033)$

<table>
<thead>
<tr>
<th>ANGLE OF</th>
<th>2 THETA</th>
<th>SIN^2</th>
<th>DEL 2TH</th>
<th>LSE FIT</th>
</tr>
</thead>
<tbody>
<tr>
<td>0</td>
<td>155.183</td>
<td>0.000</td>
<td>0.000</td>
<td>0.033</td>
</tr>
<tr>
<td>15</td>
<td>155.380</td>
<td>0.067</td>
<td>0.197</td>
<td>0.122</td>
</tr>
<tr>
<td>30</td>
<td>155.584</td>
<td>0.250</td>
<td>0.401</td>
<td>0.364</td>
</tr>
<tr>
<td>45</td>
<td>155.699</td>
<td>0.500</td>
<td>0.516</td>
<td>0.696</td>
</tr>
<tr>
<td>60</td>
<td>156.311</td>
<td>0.750</td>
<td>1.128</td>
<td>1.027</td>
</tr>
</tbody>
</table>

PROBABLE ERROR OF LINEAR FIT $= 0.088$
ERROR RELATIVE TO AVG. ORDINATE $= 19.6$ PER CENT

STANDARD DEVIATION OF STRESS $= 2.2 \times 10^7 \text{ N/m}^2$
RELATIVE STANDARD DEVIATION OF STRESS $= 15.7$ PER CENT
RESIDUAL STRESS ANALYSIS

AL BLOCK, BOTTOM WITH HOLE AS REC'D
SURFACE 52
YOUNG'S MODULUS = 7.2E+10 N/M²; 7.3E+03 KG/MM²; 1.04E+07 PD/IN²

POISSON'S RATIO = .33

STRESS COMPONENT=-1.1E+08 N/M²; -1.1E+01 KG/MM²; -1.6E+04 PD/IN²

THIS IS A COMRESSIVE STRESS IN THE PLANE OF THE SURFACE.

DELTA 2*THETA FITTED TO LSE STRAIGHT LINE:
DEL 2TH = (1.06992)*(SIN PSI)² + (7.88835E-3)

<table>
<thead>
<tr>
<th>ANGLE OF INCLINATION</th>
<th>2 THETA</th>
<th>SIN²</th>
<th>DEL 2TH</th>
<th>LSE FIT</th>
</tr>
</thead>
<tbody>
<tr>
<td>0</td>
<td>155.336</td>
<td>0.000</td>
<td>0.000</td>
<td>0.008</td>
</tr>
<tr>
<td>15</td>
<td>155.427</td>
<td>0.067</td>
<td>0.091</td>
<td>0.080</td>
</tr>
<tr>
<td>30</td>
<td>155.583</td>
<td>0.250</td>
<td>0.247</td>
<td>0.275</td>
</tr>
<tr>
<td>45</td>
<td>155.928</td>
<td>0.500</td>
<td>0.592</td>
<td>0.543</td>
</tr>
<tr>
<td>60</td>
<td>156.122</td>
<td>0.750</td>
<td>0.786</td>
<td>0.810</td>
</tr>
</tbody>
</table>

PROBABLE ERROR OF LINEAR FIT = .025
ERROR RELATIVE TO AVG. ORDIINATE = 7.2 PER CENT

STANDARD DEVIATION OF STRESS = 6.0E+06 N/M²
RELATIVE STANDARD DEVIATION OF STRESS = 5.5 PER CENT
RESIDUAL STRESS ANALYSIS

AL BLOCK SLICE, SIDE OPPOSITE HOLE, CENTER
SURFACE S4\1
YOUNG'S MODULUS = 7.2E+10 N/M²; 7.3E+03 KG/MM²; 1.04E+07 PD/IN²

POISSON'S RATIO = .33

STRESS COMPONENT=-1.0E+08 N/M²; -1.0E+01 KG/MM²; -1.5E+04 PD/IN²

THIS IS A COMPRESSIVE STRESS IN THE PLANE OF THE SURFACE.

DELTA 2*THETA FITTED TO LSE STRAIGHT LINE:
\[ \text{DEL 2TH} = (0.986165)(\sin \psi)^2 + (4.67392E-2) \]

<table>
<thead>
<tr>
<th>ANGLE OF INCLINATION</th>
<th>2 THETA</th>
<th>SIN²</th>
<th>DEL 2TH</th>
<th>LSE FIT</th>
</tr>
</thead>
<tbody>
<tr>
<td>0</td>
<td>155.387</td>
<td>0.000</td>
<td>0.000</td>
<td>0.047</td>
</tr>
<tr>
<td>15</td>
<td>155.567</td>
<td>0.067</td>
<td>0.180</td>
<td>0.113</td>
</tr>
<tr>
<td>30</td>
<td>155.614</td>
<td>0.250</td>
<td>0.227</td>
<td>0.293</td>
</tr>
<tr>
<td>45</td>
<td>156.016</td>
<td>0.500</td>
<td>0.629</td>
<td>0.540</td>
</tr>
<tr>
<td>60</td>
<td>156.130</td>
<td>0.750</td>
<td>0.743</td>
<td>0.786</td>
</tr>
</tbody>
</table>

PROBABLE ERROR OF LINEAR FIT = .056
ERROR RELATIVE TO AV. ORDINATE = 15.9 PER CENT

STANDARD DEVIATION OF STRESS = 1.4E+07 N/M²
RELATIVE STANDARD DEVIATION OF STRESS = 13.6 PER CENT

J
RESIDUAL STRESS ANALYSIS

AL BLOCK, EDGE, OPPOSITE FROM HOLE SURFACE S1

YOUNG'S MODULUS = 7.2E+10 N/M\(^2\); 7.3E+03 KG/MM\(^2\); 1.04E+07 PD/IN\(^2\)

POISSON'S RATIO = 0.33

STRESS COMPONENT = -1.0E+08 N/M\(^2\); -1.0E+01 KG/MM\(^2\); -1.5E+04 PD/IN\(^2\)

THIS IS A COMPRESSIVE STRESS IN THE PLANE OF THE SURFACE*

DELTA 2*THETA FITTED TO LSE STRAIGHT LINE:

\[
\Delta 2\theta = ( -994081 )*( \sin \psi )^2 + ( -1.29431E-2 )
\]

<table>
<thead>
<tr>
<th>ANGLE OF INCLINATION</th>
<th>2*THETA</th>
<th>SIN^2</th>
<th>DEL 2*THETA</th>
<th>LSE FIT</th>
</tr>
</thead>
<tbody>
<tr>
<td>0</td>
<td>155.263</td>
<td>0.000</td>
<td>0.000</td>
<td>-0.013</td>
</tr>
<tr>
<td>15</td>
<td>155.319</td>
<td>0.067</td>
<td>0.056</td>
<td>0.054</td>
</tr>
<tr>
<td>30</td>
<td>155.491</td>
<td>0.250</td>
<td>0.328</td>
<td>0.326</td>
</tr>
<tr>
<td>45</td>
<td>155.717</td>
<td>0.500</td>
<td>0.484</td>
<td>0.484</td>
</tr>
<tr>
<td>60</td>
<td>156.018</td>
<td>0.750</td>
<td>0.733</td>
<td>0.733</td>
</tr>
</tbody>
</table>

PROBABLE ERROR OF LINEAR FIT = 0.016
ERROR RELATIVE TO AVG. ORDIJATE = 5.3 PER CENT

STANDARD DEVIATION OF STRESS = 3.9E+06 N/M\(^2\)
RELATIVE STANDARD DEV. OF STRESS = 3.8 PER CENT
RESIDUAL STRESS ANALYSIS

AL BLOCK, SLICE, SIDE OPPOSITE HOLE
SURFACE S5/2
YOUNG'S MODULUS = 7.2E+10 N/M²; 7.3E+03 KG/MM²; 1.04E+07 PD/IN²
POISSON'S RATIO = .33
STRESS COMPONENT = -2.6E+07 N/M²; -2.6E+00 KG/MM²; -3.8E+03 PD/IN²
THIS IS A COMPRESSIVE STRESS IN THE PLANE OF THE SURFACE.

DELTA 2*THETA FITTED TO LSE STRAIGHT LINE:
DEL 2TH = ( .254154 )*(SIN PSI)² + ( 5.13841E-2 )

<table>
<thead>
<tr>
<th>ANGLE OF INCLINATION</th>
<th>2 THETA</th>
<th>SIN²</th>
<th>DEL 2TH</th>
<th>LSE FIT</th>
</tr>
</thead>
<tbody>
<tr>
<td>0</td>
<td>155.506</td>
<td>.000</td>
<td>.000</td>
<td>.051</td>
</tr>
<tr>
<td>15</td>
<td>155.573</td>
<td>.067</td>
<td>.067</td>
<td>.068</td>
</tr>
<tr>
<td>30</td>
<td>155.703</td>
<td>.250</td>
<td>.197</td>
<td>.115</td>
</tr>
<tr>
<td>45</td>
<td>155.678</td>
<td>.500</td>
<td>.172</td>
<td>.178</td>
</tr>
<tr>
<td>60</td>
<td>155.725</td>
<td>.750</td>
<td>.219</td>
<td>.242</td>
</tr>
</tbody>
</table>

PROBABLE ERROR OF LINEAR FIT = .039
ERROR RELATIVE TO AVG. ORDINATE = 29.7 PER CENT

STANDARD DEVIATION OF STRESS = 9.4E+06 N/M²
RELATIVE STANDARD DEVIATION OF STRESS = 36.4 PER CENT

500 DATA EXHAUSTED
NOW AT 500
READY
REFERENCES


