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HIGH-TEMPERATURE, HIGH-POWER-DENSITY THERMIonic ENERGY CONVERSION FOR SPACE

by James F. Morris
Lewis Research Center
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November 23, 1977
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HIGH-TEMPERATURE, HIGH-POWER-DENSITY

THERMIONIC ENERGY CONVERSION FOR SPACE

by James F. Morris
Lewis Research Center

SUMMARY

Theoretic converter outputs and efficiencies indicate the need to consider thermionic energy conversion (TEC) with greater power densities and higher temperatures within reasonable limits for space missions. This parametric presentation of converter-output power density, voltage, and efficiency as functions of current density covers 1400-to-2000K emitters with 725-to-1000K collectors. The results encourage utilization of TEC with hotter-than-1650K emitters and greater-than-6W/cm² outputs to attain better efficiencies, greater voltages, and higher waste-heat-rejection temperatures for multihundred-kilowatt space-power applications. For example 1800K, 30 A/cm² TEC operation for NEP compared with the 1650K, 5 A/cm² case should allow much lower radiator weights, substantially fewer and/or smaller emitter heat pipes, significantly reduced reactor and shield-related weights, many fewer converters and associated current-collecting bus bars, less power conditioning, and lower transmission losses. Integration of these effects should yield considerably reduced NEP specific weights. So true overall system optimization with parametric TEC inputs is desirable.

THERMIONIC ENERGY CONVERSION (TEC) FOR SPACE

Reliable, efficient, durable electric-generation systems with high power-to-weight ratios are essential for future space missions--particularly those with near-megawatt requirements. Such power-system qualities characterize thermionic converters. In addition TEC embodies simplicity, light weights, small volumes, negligible mechanical stresses, no moving parts, modularity for space safety, great power densities, and high temperatures which allow low-mass radiators. Thermionic converters are also adaptable: They generate electricity directly from thermal energy of nuclear, solar, or chemical origin.

At present the major space TEC application appears to be nuclear electric propulsion (NEP) (refs. 1 to 3). But analyses that properly recognize the high-temperature, high-power-density advantages of TEC may prove it valuable for solar, radioisotope, and topping utilization in space also. Unfortunately, though, some design-feasibility
studies assume without optimization that low or intermediate temperatures and small power densities are required for space TEC (refs. 1 to 3).

The present report offers some theoretic results that emphasize the need to consider greater power densities and higher temperatures within reasonable limits for TEC in space. Converter outputs and efficiencies for 1400-to-2000K emitters with 725-to-1000K collectors make this point.

SOME TEC BACKGROUND AND THEORY

George Hatsopoulos and Elias Gyftopoulos, long-term international TEC experts, as well as B. Ya. Moyzhes and G. Ye. Pikus, two other world-renowned TEC contributors, elaborate on the thermionic-converter heat engine in their reference works (refs. 4 and 5): For such a device the heat supplied isothermally at absolute temperature \( T_h \) is \( \int dQ_h = \int T_h dS_h = T_h \int dS_h \), where \( dS_h \) is the entropy decrease of the source. Similarly the heat rejected isothermally at absolute temperature \( T_c \) is \( \int dQ_c = \int T_c dS_c = T_c \int dS_c \), where \( dS_c \) is the entropy increase of the sink. Then according to Carnot the ideal heat-engine efficiency is

\[
\eta = \frac{\int dQ_h - \int dQ_c}{\int dQ_h} = \frac{T_h - T_c}{T_h} \left( \frac{\int dS_c}{\int dS_h} - 1 \right) = \frac{T_h - T_c}{T_h} = \eta_c
\]

From this basic principle comes the expectation that in general raising the emitter temperature or lowering the collector temperature tends to increase TEC efficiency. Local exceptions to this corollary may occur for optimizations of specific converters. But with freedom of selection for electrode types and materials, enhancement modes, and operating conditions this temperature generalization for TEC efficiency prevails.

Occasionally, disseminated information apparently contends with the idea that TEC efficiencies generally rise with increasing emitter temperatures (ref. 3). At such times reaffirmation of the validity of Nicolas Carnot's thermodynamic legacy seems appropriate. But merely pointing to the preceding equation is perhaps somewhat simplistic. So the present report relies on TEC output and efficiency calculations based on the assumptions used to produce pages IV-15 to IV-18 of ref. 3: "Back emission should be limited to 10%" for 1400, 1650, and 1800K emitters (2000K included also) with 725, 925, and 1000K collectors. However the present analysis deletes the ref. 3 assumptions that "converter power density should be set at 5 to 6 W/cm²" and that the highest emitter temperature should be used only with the highest collector temperature. Also, assumed interelectrode
losses near zero by FY 81 (ref. 6) allow estimates of collector work functions.

The appropriate converter outputs are the current density,

\[ J_0 = J_{SE} - J_R, \]

1) the electrode voltage,

\[ V_0 = \Phi_E - \Phi_C - V_D - V_A = \Phi_E - V_B - V_A, \]

2) the voltage at optimum-lead terminals,

\[ V_{OL} = V_0 - 2V_L, \]

3) the electrode power density,

\[ P_0 = J_0 V_0, \]

4) and the effective power density with optimum leads attached to the converter,

\[ P_{OL} = J_0 V_{OL} \]

5)

Here \( \Phi_E \) and \( \Phi_C \) are emitter and collector work functions, \( V_D \) is the interelectrode voltage drop, \( V_B = \Phi_C + V_A \) is the barrier index or total internal loss, \( V_A \) is the equivalent auxiliary input voltage (not used in the present calculations), and \( V_L \) is the voltage loss required for optimum leads.

The current-density components correspond to emitter saturation,

\[ J_{ES} = A (1-R_E) T_E^2 \exp \left(-\frac{\Phi_E}{kT_E}\right), \]

6) which has a collector-saturation counterpart,

\[ J_{CS} = A (1-R_C) T_C^2 \exp \left(-\frac{\Phi_C}{kT_C}\right), \]

7) and to the reverse flow \( J_R \), which includes reflections, backscattering, back emission, and other effects that diminish the output current density. In equations 6) and 7) \( A \) and \( k \) are Richardson and Boltzmann constants, \( T_E \) and \( T_C \) are emitter and collector temperatures, and \( R_E \) and \( R_C \) are emitter and collector reflection coefficients.

An important theoretic detail relates to a common inconsistency in the treatment of back emission (refs. 7 and 8): In generalized TEC terminology back emission subtracts from the emitter current in obtaining the net output current. This usual definition of back emission requires it to be only that part of the collector emission
that reaches the emitter and thereby diminishes the output current according to a net-flow balance at the converter boundaries. Thus back emission is not the saturated collector emission given by equation 7), regardless of $R_C$ modification, because the emission barrier is incorrect: This observation derives from the fact that, in the generally cited TEC power-producing mode, the emitter electron barrier (motive maximum) is a few tenths of a volt (the interelectrode voltage drop) above its collector counterpart. So during steady-state operation the preponderance of collector saturated emission cannot clear the emitter sheath, even in the absence of other deflecting encounters. Therefore most of the collector saturated emission must return to its source nullifying to a large extent its effect on the diminution of the net output current.

Unless the interelectrode loss is much closer to zero than to its currently common value of about a half volt, only a small fraction of the collector emission, the true back emission $J_{BE}$, will reach the emitter:

$$J_{BE} = A(1-R_{BE}) T_C^2 \exp \left( -\frac{V_B}{kT_C} \right)$$  \hspace{1cm} (8)

In this equation the effective back-emission reflection coefficient $R_{BE}$ comprises $R_C$ and similar coefficients for all interelectrode mechanisms that return collector-emitted electrons to their source—except those for noncollisional repulsion by the emitter sheath. Thus, using equation 8) without $R_{BE}$ produces a conservative estimate of the converter output current. Such an approximation seems reasonable for low cesium concentrations, reduced enhanced-mode pressures, and small interelectrode gaps. Of course, with zero interelectrode losses assumed (ref. 6 for FY 81) as well as negligible interelectrode-reflection effects, equations 7) and 8) become identical.

A simplified, yet reasonable estimate of TEC efficiency with optimum-lead losses ($\eta_{OL}$) embodies the previously discussed inputs (refs. 4 and 9):

$$\eta_{OL} = \frac{(J_{ES}-J_{BE})}{(J_{ES}(\phi_E+2kT_E)-J_{BE}(\phi_E+2kT_C)+5.7 \times 10^{-12} \left[ 0.05+7.5 \times 10^{-5}(T_E-1000) \right] (T_E^4-T_C^4))}$$  \hspace{1cm} (9)

Here the last term of the denominator approximates non-electronic thermal transport while the factor following the first 2 in the numerator represents the optimum-lead loss $V_{OL}$. Deleting $2V$ from equation 9) transforms that expression into one for the TEC electrode efficiency $\eta_{EC}$ used here to compute the optimum-lead loss. Of course, the electrode
efficiency is the true converter evaluation analogous to other power-generator performance ratings. But because of relatively high TEC current densities and low voltages the optimum-lead efficiency seems more pragmatic.

**TEC-PERFORMANCE TRENDS**

Theoretic TEC outputs and efficiencies for converters with 10-percent back emission and optimum leads appear parametrically in figures 1, 2, and 3 for 725, 925, and 1000K collectors. Each figure comprises plots of efficiency, voltage, and power density as functions of current density for 1400, 1650, 1800, and 2000K emitters.

Without exception, for a given collector temperature, all performance curves for higher emitter temperatures rise above those for the lower emitter temperatures. This observation would have gratified Nicolas Carnot.

The efficiency curves reach values very close to their maxima above 5 A/cm² for the 1400K emitters; 20 A/cm² for 1650K emitters; 30 A/cm², 1800K; and 40 A/cm², 2000K.

The two preceding paragraphs imply that studies of any TEC system should evaluate parametrically the effects of converters with emitters hotter than 1650K and current densities greater than 5 A/cm² (refs. 1 to 3). Table 1 for 925K collectors (refs. 2 and 3) further emphasizes this observation. The underlined Table 1 entries indicate output and efficiency improvements (for converters with optimum leads) resulting from raising the emitter temperature from 1650K to 1800K at 5 A/cm² and at 30 A/cm².

These underlined values also reveal the significant output and efficiency gains for TEC operation at 1800K and 30 A/cm² as compared with 1650K and 5 A/cm² (refs. 1 to 3): The 28.5% increase in optimum-lead efficiency means lighter radiators and either more output power or smaller nuclear reactors and lighter shield-dependent weights for NEP. The 10.8% higher optimum-lead voltage requires less power conditioning capability and results in lower transmission-line losses for a given quantity of output power. The 560% gain in effective output power density allows many fewer converters and associated current-collecting bus bars for a given output-power level. And of course the higher emitter temperature (coupled with greater efficiency) enables the use of substantially fewer and/or smaller emitter heat pipes. This reduction in turn should produce significant decreases
in shielding-related as well as reactor weights. The higher emitter
temperature can also make possible considerably increased collector
temperatures if parametric studies indicate the need for lower
radiator weights (the T^4 influence).

The previously enumerated advantages of 1800K, 30 A/cm^2 TEC
operation over the 1650K, 5 A/cm^2 case have obviously strong effects
on NEP specific-weight reductions. So the importance of true overall
system optimization with parametric TEC inputs should not be under-
estimated.

Omitted tabulations similar to those of Table 1 are also available
for collector temperatures of 725K and 1000K. And as figures 1 to 3
attest, the order of performance remains unchanged: For a given
collector temperature the highest emitter temperature produces the
best TEC performance; the lowest emitter temperature gives the poorest
TEC performance.

If the only emitter, collector combinations considered were
1400K with 725K, 1650K with 925K, and 1800K with 1000K all at 5.5
W/cm^2 as in reference 3, the TEC-output relationshi; would appear
quite different from those in figures 1 to 3. But a parametric TEC-
optimization study should evaluate each collector temperature with
each emitter temperature. When existing converter-component capa-
bilities preclude such pairings, appropriately directed technology
advancements may render them possible in the near future.

Reference 3 states that "the higher temperature converters are
limited to higher work function materials, and thus eventually extra-
polate to lower operating efficiencies." But the 1800K emitter work
functions in the table are obtainable with cesiated tungsten, for
example, without invoking oxygenation. Such work functions are even
more readily accessible with rhenium and still more easily attainable
with iridium.

As for the collector work functions in the preceding table, they
are well within reach of cesiated, oxygenated tungsten: This collector
has a work-function minimum of 1.21 eV according to recent measure-
ments (ref. 9). Unoxgenated minimum cesiated work functions run
1.45 eV for rhenium (ref. 4) and probably 1.4 or lower for 111 iridium
(refs. 7, 8, and 10 to 14). And tungsten, rhenium, and iridium are all
satisfactory for 1800K-emitter service.
Incidentally the calculations for figures 1 to 3 give results rather centrally located among those from other TEC efficiency models for 10% back emission and zero arc drop (Private communication with G. D. Fitzpatrick of Rasor Associates, Inc.). The variation occurs because of differences in loss approximations. A comparison of TEC efficiencies appear in Table 2.

Table 2 lists extremes of conditions primarily to compare TEC-efficiency models over wide ranges. But these values also strongly imply the desirability of high-temperature, high-power-density thermionic energy conversion for space.
REFERENCES


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Figure 1. - Optimum-lead TEC efficiency ($\eta_{OL}$), voltage ($V_{OL}$), and power ($P_{OL}$) versus current ($J_O$) for four emitter temperatures ($T_E$) at a collector temperature of 725 K with 10 percent back emission.
Figure 2 - Optimum-lead TEC efficiency \( n_{OL} \), voltage \( V_{OL} \), and power \( P_{OL} \) versus current \( J_0 \) for four emitter temperatures \( T_E \) at a collector temperature of 925 K with 10 percent back emission.
Figure 3. - Optimum-lead TEC efficiency ($\eta_{OL}$), voltage ($V_{OL}$), and power ($P_{OL}$) versus current ($J_0$) for four emitter temperatures ($T_E$) at a collector temperature of 1000 K and 10 percent back emission.