

FLUTTER OF DARRIEUS WIND TURBINE BLADES: CORRELATION OF THEORY AND EXPERIMENT

Norman D. Ham

Department of Aeronautics and Astronautics
Massachusetts Institute of Technology
Cambridge, Mass. 02139

Acquisition of frequency data for the Alcoa 17-M LC and 1238229 wind turbines has permitted some limited correlation of the flutter analysis of Ref. 1 with experimental data for flutter of Darrieus blades from Ref. 2.

The natural frequencies of the non-rotating critical blade bending modes can be written as

$$\omega_{20} = \lambda_2 \sqrt{\frac{EI_F}{mR^4}} \text{ rad./sec.}$$

$$\bar{\omega}_{20} = \bar{\lambda}_2 \sqrt{\frac{GJ}{mR^4}} \text{ rad./sec.}$$

where $\lambda_2, \bar{\lambda}_2$ are frequency parameters, and
 EI_F = blade flatwise bending stiffness, lb-ft.
 GJ = blade torsion stiffness, lb-ft.²
 m = blade running mass, slugs/ft.
 R = radius of circular arc portion of blade

Also, the effect of turbine rotational speed Ω is given by

$$\omega_2^2 = \omega_{20}^2 + K_2 \Omega^2$$

$$\bar{\omega}_2^2 = \bar{\omega}_{20}^2 + \bar{K}_2 \Omega^2$$

where K_2, \bar{K}_2 are Southwell coefficients.

Data provided by Alcoa has made possible the following tabulation for turbines of $h/D = 1.5$:

TABLE I

TYPE	λ_2	$\bar{\lambda}_2$	K_2	\bar{K}_2
17-Meter LC	5.72	23.7	2.74	4.57
1238229	10.0	23.5	4.32	3.99

Proceedings of the Second DOE/NASA Wind Turbine Dynamics Workshop, Cleveland, February 1981.

The bracing struts of Type 1238229 appear to have a significant effect on the flatwise bending parameters λ_2 and K_2 . It is suspected that significant mode shape variations may also be related to the clamping effect of the struts.

Experimental flutter data for wind tunnel models of Darrieus wind turbines of $h/D = 1.5$ are shown in Fig. 1 taken from Ref. 2. The correlating parameters are the reduced flutter speed

$$\frac{U_c \ell}{t} \sqrt{\frac{\rho_B}{G}}$$

where U_c = blade tangential velocity at flutter = $\Omega_F R_c$
 ℓ = blade length
 t = blade section maximum thickness
 ρ_B = blade material density
 G = blade material torsion modulus

and the density ratio

$$\frac{\rho_B A_B}{\pi \rho c^2}$$

where A_B = blade material cross-section area
 ρ = air density
 c = blade chord

These parameters were calculated for the 17-Meter LC and 1238229 wind turbines using their theoretical flutter speeds Ω_F of 108 rpm and 57 rpm respectively as calculated by the method of Ref. 1. The results are plotted in Fig. 1. The theory is seen to give a conservative prediction of flutter speed.

An attempt was made to predict the experimental flutter speed directly for the particular case of the aluminum blade of 0012 section of Fig. 1, using blade frequencies estimated by means of the parameters tabulated in TABLE I for the 17-Meter LC turbine, and the flutter parameter values $C_2 = -6.5$ and $m_B/m_g = 0.75$ for that strutless turbine. Again, the theoretical prediction is seen to be conservative.

REFERENCES

1. Ham, N.D., "Flutter of Darrieus Wind Turbine Blades", from "Wind Turbine Structural Dynamics", NASA Publication CONF-2034, November 1977.
2. Templin, R.J. and South, P., "Canadian Wind Energy Program", Proc. of the VAWT Technology Workshop, SAND76-5686, Sandia Laboratories, Albuquerque, N.M., July 1976.

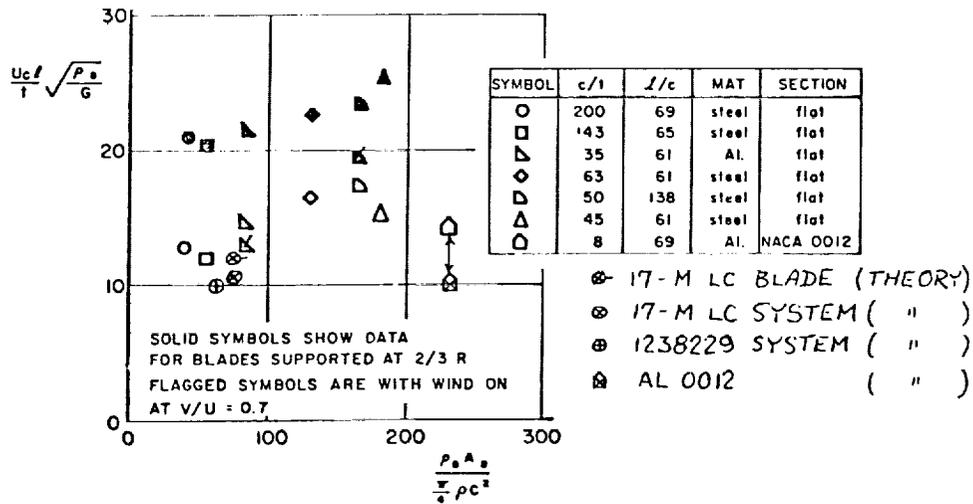


Fig.1. Variation of blade flutter parameter with density ratio for turbine with h/d = 1.5.