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Transient Difference Solutions of the Inhomogeneous Wave Equation: Simulation of the Green’s Function

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Prepared for the Eighth Aeroacoustics Conference sponsored by the American Institute of Aeronautics and Astronautics Atlanta, Georgia, April 11–13, 1983
TRANSIENT DIFFERENCE SOLUTIONS OF THE INHOMOGENEOUS WAVE EQUATION:
SIMULATION OF THE GREEN'S FUNCTION

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Abstract
A time-dependent finite difference formulation to the inhomogeneous wave equation is derived for plane wave propagation with harmonic noise sources. The difference equation and boundary conditions are developed along with the techniques to simulate the Dirac delta function associated with a concentrated noise source. Example calculations are presented for the Green's function and distributed noise sources. For the example considered, the desired Fourier transformed acoustic pressures are determined from the transient pressures by use of a ramping function and an integration technique, both of which eliminates the nonharmonic pressure associated with the initial transient.

Nomenclature

\[ \begin{align*}
  a_0 & \quad \text{ambient density, kg/m}^3 \\
  o & \quad \frac{(c \text{ at/ax})}{c} \\
  w & \quad \text{dimensionless frequency } (n_w^*/f^*) \\
  w_1 & \quad 2\pi \\
  n_2 & \quad n_2w \\
  w^* & \quad \text{dimensionless angular frequency, rad/sec} \\
  a^* & \quad \text{dimensionless quality} \\
  k & \quad \text{time index} \\
  w^* & \quad \text{dimensionless angular frequency, rad/sec} \\
  \text{Superscripts} & \quad \text{Subscripts} \\
  a & \quad \text{dimensional quality} \\
  n & \quad \text{harmonic number} \\
  \text{Introduction} & \quad \text{Subscripts} \\
  i & \quad \text{space index} \\
  n & \quad \text{harmonic number} \\
  \end{align*} \]

Finite difference and finite element numerical solutions to the homogeneous wave equation have been successfully used to predict harmonic radiation from turbofan engine inlets \(^{(1,2)}\) as well as internal propagation in a wide variety of complex hard and soft wall ducts, as cited in the summaries of references \(^{3,4}\). In these solutions, the noise sources were decoupled from the propagation problem which reduced the more general inhomogeneous wave equation \(^{(ref. 5, eq. 1.20)}\) to a simpler homogeneous form. In the homogeneous form, a known or estimated pressure profile is used as a boundary condition at some location, such as the fan face in a turbofan engine, and the propagation of this pressure wave is predicted from a solution of the wave equation or the more general linearized gas dynamic equations.

Some inhomogeneous problems of current interest, such as a turboprop noise field, require three dimensional solutions to the sound field. Consequently, the transient finite difference solution to the wave equation will be employed in the present paper. Customarily, in the bulk of the past numerical studies \(^{(3,9)}\), the pressure and acoustic velocities were assumed to be simple harmonic functions of time. In this case, the time independent wave equations can be used. The matrices associated with the numerical solutions to the time independent equation (called steady-
where \( L_f \) is the Fourier transform operator (ref. 14, p. 28) and \( H \) represents the unit step function. Symbols with an asterisk represent dimensional quantities. The second purpose of this paper is to develop the steps necessary to adapt the transient solution to simulate the Fourier transform form of the acoustic solutions, where all the acoustic parameters vary as \( e^{-i\omega t} \). The steady-state numerical solutions exactly simulate the analytical results. Unfortunately, this is not the case for the transient solution. Basically, the analytical and steady-state numerical theories assume the harmonic forcing function has begun at time equal to minus infinity. On the other hand, the transient solution abruptly begins the harmonic oscillation at time equal to zero. Clearly,

\[
L_f \left[ e^{-i\omega t} \right] \cdot L_f \left[ H(t) e^{-i\omega t} \right] = (1)
\]

where \( L_f \) is the Fourier transform operator (ref. 14, p. 28) and \( H \) represents the unit step function. Symbols with an asterisk represent dimensional quantities. The second purpose of this paper is to develop the steps necessary to adapt the transient solution to simulate the Fourier transform form of the solution.

Finally, analytical predictions for noise propagation from both steady and unsteady noise sources, such as a propeller, rely on the integration of a unit source represented by a free field Green's function \( g \) and the product of the magnitude of the source \( F \) (ref. 14, p. 743) such that

\[
P = \int_A F_x g(x, \alpha) \, dA_0
\]

Consequently, predictions of the free field Green's function as well as the effect of a distributed source are developed herein in two examples for code verification and assessment of numerical accuracy and stability.

**Governing Equations and Boundary Conditions**

The governing differential equation and boundary conditions are introduced in this section of the report. First, the scalar inhomogeneous wave equation is developed. Next, the Green function form of the equation is developed and will be used for code verification. Finally, an expression for the Sommerfeld exit boundary condition is presented.

**Inhomogeneous One Dimensional Wave Equation**

In the absence of mean flow, the dimensionless form of the plane wave linearized gas equations can be written as (ref. 5, eq. 1.11)

\[
\frac{dP}{dt} + c \frac{dU}{dx} = Q
\]

**Continuity (isentropic)**

\[
\frac{dP}{dt} + c \frac{dU}{dx} = 0
\]

**Momentum**

\[
\frac{dU}{dt} + c \frac{dP}{dx} = F_x
\]

The parameters used to reduce the pressures and velocities to their dimensionless form are given in the Nomenclature. The usual notation for acoustic pressure and velocity are used. Here, \( Q \) represents a dimensionless mass source and \( F_x \) a dimensionless body force. Differentiating equation (3) with respect to time and equation (4) with respect to \( x \), and combining yields the dimensionless wave equation

\[
\frac{1}{c^2} \frac{d^2P}{dt^2} - \frac{d^2P}{dx^2} = S(x,t)
\]

where the acoustic source is

\[
S(x,t) = \frac{1}{c^2} \frac{dQ}{dt} - \frac{dF_x}{dx}
\]

The first term on the right-hand side of equation (6) represents the monopole contribution while the second term represents the dipole contribution. The source will be assumed to be harmonic in nature \( (e^{-i\omega t}) \) such that equation (5) can be written as

\[
\frac{1}{c^2} \frac{d^2P}{dt^2} - \frac{d^2P}{dx^2} = s(x) e^{-i\omega t}
\]

where

\[
\omega = n \alpha
\]

Only the first harmonic, \( n = 1 \), is considered herein. However, in the numerical data reduction section, a procedure will be presented to handle a multiple frequency source problem. Because of the introduction of complex notation for the noise source, all the dependent variables are complex.

**Green Function Equation**

For the purpose of checking the numerical theory, the simplest form of the source is desired.
Consequently, $s(x)$ is assumed to be of the form

$$s(x) = s(x_s)(x - x_s)$$  \hspace{1cm} (9)

and

$$s(x_s) = 1$$  \hspace{1cm} (10)

Thus, equation (7) can be written as

$$\frac{1}{c^2} \frac{\partial^2 p}{\partial t^2} - \frac{\partial^2 p}{\partial x^2} = s(x - x_s) e^{-i\omega t} H(t)$$  \hspace{1cm} (11)

This source is analogous to the analytical form treated by Morse and Ingard (Ref. 14, p. 133) for which a closed form analytical solution is presented.

If the signal $e^{-i\omega t}$ is assumed to be applied from time minus infinity, the time dependent pressure in equation (11) represents the Green function solution for a concentrated source. Thus, equation (11) can be written as

$$\frac{1}{c^2} \frac{\partial^2 p}{\partial t^2} - \frac{\partial^2 p}{\partial x^2} = s(x - x_s) e^{-i\omega t}$$  \hspace{1cm} (12)

In this steady-state case, all the acoustic parameters vary as $e^{-i\omega t}$, that is

$$G(x,t) = g(x) e^{-i\omega t}$$  \hspace{1cm} (13)

and the differential equation (8) becomes

$$\frac{\partial^2 g}{\partial x^2} + \left(\frac{\omega}{c}\right)^2 g = -s(x - x_s) e^{-i\omega t}$$  \hspace{1cm} (14)

Later, numerical solutions of equation (11) will be compared to analytical solutions of equation (14).

Sommerfeld Radiation Boundary Condition

The two required boundary conditions upstream and downstream of the noise source can be expressed in terms of a specific acoustic impedance defined as

$$\zeta_e = \frac{p}{nU}$$  \hspace{1cm} (15)

Substituting equation (15) into equation (4) yields the following relationship for the exit pressure gradient

$$\frac{\partial p}{\partial x} = - \frac{1}{\zeta_e} \frac{\partial p}{\partial t}$$  \hspace{1cm} (16)

For the plane wave propagation to be considered herein, $\zeta_e = 1$ (dimensionally equal to $\rho c$) which is equivalent to propagation at an exit without acoustic reflections, commonly called the Sommerfeld radiation condition.

Initial Conditions

In the numerical calculations, for times equal to or less than zero, the acoustic field is assumed quiescent, that is, the acoustic pressure and velocities are taken to be zero. For times greater than zero, the application of the noise source (eq. (7)) will drive the pressures.

Difference Equations

Instead of a continuous solution for pressure in space and time, the finite-difference approximations will determine the pressure at isolated grid points in space as shown in Fig. 1 and at a discrete time steps at. Starting from the known initial conditions at $t = 0$ and the boundary conditions, the finite-difference algorithm will march-out the solution to later times.

The difference equations can be developed by an integration of the governing wave equation, equation (7), in the form

$$\int_{t-at/2}^{t+at/2} \int_{x-ax/2}^{x+ax/2} \left(\frac{1}{c^2} \frac{\partial^2 p}{\partial t^2} - \frac{\partial^2 p}{\partial x^2}\right) dx dt$$

The integration procedure is fully documented in reference 16 and is especially useful in developing equations at boundaries and for the special case of the unit impulse function.

Central Cell

Away from the boundaries, in the central cell of Fig. 1, the integration of the various terms in equation (17) yields

$$\frac{\Delta x}{\Delta t} \left(p^k_{i+1} - 2p^k_i + p^k_{i-1}\right) - \frac{\Delta t}{\Delta x} \left(p^k_i - 2p^k_{i+1} + p^k_{i-1}\right) e^{-i\omega t} = -s(x_i) e^{-i\omega t}$$  \hspace{1cm} (18)

where $i$ denotes the space index, $k$ the time index, and $\Delta x$ and $\Delta t$ are the space and time mesh spacing, respectively. All the spacings are assumed constant. The time $t^k$ is defined as $t^k = k\Delta t$.

Solving equation (18) for the new pressure $p^k_{i+1}$ yields

$$p^k_{i+1} = 2p^k_i - p^k_{i-1} - \frac{1}{c^2} \frac{\partial p}{\partial t} e^{-i\omega t}$$

where $a = (\text{cat} / \Delta x)^2$. As will be discussed shortly in conjunction with equation (24), the value of $a$ must be chosen less than or equal to 1 for numerical stability. Equation (19) is an algorithm which permits marching out solutions from known values of pressure at times associated with $k$ and $k-1$. The procedure is explicit for all the past values of $p^k_i$ and $p^k_{i-1}$ are computed. For the special case at $t = 0$, the difference equation for the start up is
\[ p_1 = p_0 + \Delta t \frac{\partial p_0}{\partial t} + \frac{1}{2} \left( p_{i+1} - 2p_i + p_{i-1} \right) \]
\[ s(x_i)^2 \Delta t^2 + \frac{s(x_i)^2}{2} \]  

(20)

where \( p_0 \) and \( \frac{\partial p}{\partial t} \) are taken as zero, and \( e^{-i\omega t} \) is taken as 1 and \( H(t) \) is assumed unity over the entire \( \Delta t/2 \) time step.

**Exit Boundary Cell**

The solution of equation (17) for the exit cells is fully documented when \( s(x) \) is zero (homogeneous solution). In this case, the algorithm for the exit pressure is

\[ p_{k+1} = \left( 2p_k - p_{k-1} + 2e p_k - p_{k-1} \right) \]
\[ + s(x_i)^2 \Delta t^2 e^{-i\omega t} \frac{s(x_i)^2}{1 + \omega} \]  

(21)

where \( \omega \) equals \((\omega_0 \Delta t) \). A similar equation exists at the entrance, in this case, \( k \) goes to 1 and \( k-1 \) becomes 2.

**Green's Function Equation**

Assuming \( s(x) \) of the form given by equations (9) and (10) and noting that \( H(t) \) is identical to 1, the integration of the source term in equation (17) yields

\[ \int_{t-\Delta t/2}^{t+\Delta t/2} s(x) e^{-i\omega t} dx dt \]
\[ -i\omega \frac{s(x)}{2} \int_{t-\Delta t/2}^{t+\Delta t/2} s(x) e^{-i\omega r} dx \]
\[ = e^{-i\omega \frac{s(x)}{2} \int_{t-\Delta t/2}^{t+\Delta t/2} s(x) e^{-i\omega r} dx} \]  

(22)

Therefore, by comparing the results of the integration in equation (22) to the right hand side of equation (18), if \( s(x_i) \) is chosen such that \( s(x_i) = 1/ax \)

(23)

then equation (19) will simulate the Green's function solution. Of course, \( s(x) \) equals zero for all \( x \) values not associated with the position of the concentrated source.

**Stability**

In the explicit time marching approach used here, round-off errors can grow in an unbounded fashion and destroy the solution if the time increment \( \Delta t \) is too large or the iteration scheme is improperly posed. The von Neumann method (Ref. 15, p. 104) applied to the homogeneous form of equation (7) required that

\[ \Delta t \frac{\omega_0}{2} \leq \frac{a}{c} \]  

(24)

This result was first derived by Courant, Friedrichs, and Levy in 1927 for a related method and, therefore, is known as the CFL condition. This criterion proved successful in the solution of equation (7). Note, for a two dimensional problem for which nonpropagating modes exist, the exit impedance can introduce an instability into the equation (1). In this case the eigenvalue approach (Ref. 16, p. 283) should be employed to determine the effect of the boundary conditions on stability.

**Numerical Data Reduction**

Recall, at the start of the numerical calculation, the acoustic pressures and velocities were assumed zero throughout the duct and a source begins a harmonic oscillation. The pressures will vary harmonically at all points in the duct. In order to conveniently interpret the numerical results, the time harmonic component of pressure should be removed so that only the spatial or Fourier transformed component of pressure remains.

**Homogeneous Equation**

Consider the solution for plane wave propagation in the absence of noise sources \( (S(x,t) = 0) \) and when the pressure at a boundary \( (x = 0) \) is given by

\[ p(t) = e^{-i\omega t} \quad -\infty < t < \infty \]  

(25)

In this steady-state case, all the acoustic parameters are proportional to \( e^{-i\omega t} \) and equation (7) reduces to the Helmholtz equation. The solution for pressure is (see Appendix A, Problem 1)

\[ p(x,t) = e^{-i\omega t} e^{-i\frac{\omega}{c} x} \]  

(26)

However, in the numerical simulation to this example, the boundary condition given in equation (25) becomes

\[ p(0,t) = e^{-i\omega t} H(t) \]  

(27)

In this case, the numerical solution will approximate (see Appendix A, Problem 2)

\[ p(x,t) = e^{-i\omega t} e^{-i\frac{\omega}{c} x} H(t-x/c) \]  

(28)

Although only a specific example has been considered, in homogeneous problems with forcing boundary conditions, such as equations (25) or (27), the pressure \( p(x,t) \) will always be proportional to \( e^{-i\omega t} \) provided sufficient time has elapsed so that the initial transient has passed \( (t > x/c) \). Therefore, the numerical data can be reduced to a simple spatial form by dividing by \( e^{-i\omega t} H(t) \); that is

\[ p(x) = \frac{e^{-i\omega t} p(x,t)}{H(t-x/c)} \]  

(29)

**Inhomogeneous Problems**

For many sources, such as a pure monopole, numerical data reduction for the inhomogeneous wave equation could employ equation (29) because the results will be solely harmonic in time, similar in form to equation (25) and (28). However, this will not always be the case as will now be shown. Consequently the data reduction scheme employed must always be scrutinized.

For the example source considered in this paper, the analytical solution to the steady Green's function equation (14) is (see Appendix A, Problem 3)

\[ g(x) = p(x) = \frac{e^{-i\omega t} p(x,t)}{e^{-i\frac{\omega}{c} x}} = \frac{e^{-i\omega t} p(x,t)}{e^{-i\frac{\omega}{c} x}} H(t-x/c) \]  

(30)
while the numerical simulation employing equation (11) yields (See Appendix A, Problem 4)

\[ P(x,t) = \frac{4c}{\omega_1} e^{-i\omega_1 x - x_5/c} - \frac{4c}{\omega_1} t > x/c \]  

(31)

Dividing by \( e^{-i\omega_1 t} \) according to equation (29) yields

\[ P(x,t) = \frac{4c}{\omega_1} e^{-i\omega_1 x - x_5/c} - \frac{4c}{\omega_1} t > x/c \]  

(32)

As mentioned in the introduction, it is desirable that the numerical solutions should approximate equation (30). Therefore procedures are developed to eliminate the dc component contribution in the numerical prediction of \( p \). In this example, two new schemes for data reduction will now be employed to simulate equation (30).

Ramping Solution

The dc component in equation (31) could result from the sudden onset of the source at time \( t = 0 \), in contrast to the Fourier transform solution for which the source has been applied from a time of minus infinity. To reduce the dc component, an exponential ramp function \( (1 - e^{-bt}) \) has been applied to the source such that

\[ S(x,t) = s(x-x_5)(1 - e^{-bt})e^{-i\omega_1 t} H(t) \]  

(33)

In this case, the numerical simulation for the Green's function solution becomes (see Appendix A, Problem 5)

\[ p(x) = \frac{4c}{\omega_1} e^{-i\omega_1 x - x_5/c} - \frac{4c}{\omega_1} \left( \frac{b}{\omega_1} \right) e^{-i\omega_1 t} \]  

(34)

Equation (34) will approximate the analytical Green's function solution provided \( \omega_1 \) is chosen to be small compared to unity.

Integral Solution

For a large class of inhomogeneous problems, the use of equation (29) presents some additional difficulties. Consider, for example, a typical forcing function for the steady or unsteady (circular-ferential variations of steady flow) loading noises on a propeller source (see Appendix A, Problem 5)

\[ p(x) = \frac{4c}{\omega_1} e^{-i\omega_1 x - x_5/c} - \frac{4c}{\omega_1} \left( \frac{b}{\omega_1} \right) e^{-i\omega_1 t} \]  

(35)

Besides yielding the individual higher harmonics, the orthogonal properties of equation (35) will eliminate the steady offset involved in the numerical example considered herein.

In employing equation (35), for convenience, the \( a_t \) from equation (28) is adjusted as follows to obtain an even number of \( a_t \)'s in one period:

\[ a_t = \text{FLD}(\text{IFLX}(1/a_{x_t}) + 1) \]  

(36)

where \( \text{IFLX} \) converts a floating point number to an integer, and where

\[ a_{cycle} = \text{IFLX}(1/a_{x_t}) + 1 \]  

(37)

Consequently, the following algorithm was used to numerically evaluate equation (35) to calculate \( p(x) \) for \( n = 1 \).

\[ \sum_{j=0}^{a_t} a_{cycle} \left( p_{j+1} e^{i\omega_1 t} + p_{j-1} e^{-i\omega_1 t} \right) \]  

(38)

Sample Problems - Comparisons

Two examples are now presented to illustrate the numerical techniques. Both examples have analytical solutions; therefore, a direct comparison between the numerical and analytical techniques can be made. The first example presents a simulation of the Green's function. The second example presents a solution for a distributed noise source.

Example 1 - Green's Function

As our first example of the difference technique, consider the numerical solution of the Green's Function equation, (12). In this case, the finite difference equations (19) and (20) will be used in the solution with the source term equal to \( 1/a_x \) (eq. (23)) at \( x_s \) and zero elsewhere. The exact analytical solution is given by equation (32). Taking the position of the source \( x_s \) equal to zero, considering only positive values of \( x \) in the output, and dividing both sides of equation (32) by \( e^{-i\omega_1 t} \), equation (32) can be rewritten as (recall \( a_t = 2\pi\) )

\[ \sum_{j=0}^{a_t} a_{cycle} \left( p_{j+1} e^{i\omega_1 t} + p_{j-1} e^{-i\omega_1 t} \right) \]  

(39)

For a nondimensional speed of sound of unity and \( a_{x} \) of 0.00521, the numerical and analytical results are shown in Fig. 2 for times of 0.75, 1.0, 1.25, and 1.5. At \( t = 0.75 \) in Fig. 2a, the initial transient has progressed three-quarters of the way to the termination point of \( x \) equal to unity. At \( t \) greater or equal to 1.0 in Figs. 2b, c, and d, the time dependent oscillatory nature of the transient form of the solution is shown. As seen in all the Fig. 2 plots, the analytical and numerical results are in excellent agreement.

Recall, the steady-state Fourier transform solution to the Green's function equation is de-
The results of the integration technique, equation (38), are shown in Fig. 5. As seen in Fig. 5, the analytical and numerical results are in exact agreement at the end of four cycles \( t = 4 \).

In employing the ramping technique for the distributed source, it was necessary to reduce \( b \) to 0.1 to keep an eight percent error, as was used in Fig. 3. The reduced \( b \) leads to a solution time an order of magnitude greater than the integration technique, which limits the usefulness of the ramping technique.

**Concluding Remarks**

The transient difference technique was successfully formulated for the one dimensional inhomogeneous wave equation with a harmonic source. The Green's function and a distributed noise source are accurately predicted by the numerical theory. The desired Fourier transformed acoustic pressures are determined from the transient pressures by use of a ramping function and an integration technique both of which eliminate the nonharmonic pressures associated with the start-up of the particular inhomogeneous source function considered herein. The integration technique is an order of magnitude faster than the ramping technique; consequently, future numerical solution should employ this technique.

**Appendix A**

**Fourier Integral Solutions**

To verify the accuracy and applicability of the numerical techniques, closed form analytical solutions will now be developed for a variety of check problems. The complex exponential form of the Fourier integral representation will be used to develop the analytical expression.

The Fourier transform of \( P(x,t) \) is defined as (Ref. 14, p. 281)

\[
\hat{P}(x,\omega) = \frac{1}{2\pi} \int_{-\infty}^{\infty} P(x,t) e^{i\omega t} dt
\]

The Fourier transform of the wave equation (5) becomes

\[
\frac{\partial^2 p}{\partial x^2} + \frac{\omega^2}{c^2} p = \frac{1}{2\pi} \int_{-\infty}^{\infty} S(x,t) e^{i\omega t} dt
\]

The Fourier transforms to be considered have only simple poles and vanish at \( i \omega \rightarrow \pm \infty \). For the problems considered herein, the \( x \) and \( t \) variables can be incorporated in a new variable \( s = t - x/c \) and the inverse transform can be written as

\[
P(x,s) = \int B(\omega) e^{-i\omega s} d\omega
\]

In this case, the inverse can be written directly as (Ref. 14, p. 17, Eq. 1.2.15)

\[
x < 0 \quad P(x,s) = 2\pi i \int_{-\infty}^{\infty} B(\omega) e^{-i\omega s} d\omega
\]
For $s(x) = 0$, the solution of equation (A2) in a semi-infinite domain (no reflections) yields
\[ p(x, w) = C_+ e^{iwx/c} \]  \hspace{1cm}  \text{(A6)}

The constant $C_+$ (forward waves) is determined by the transformation of the boundary condition, equation (25) (Ref. 15, p. 321).
\[ p(0, w) = \delta(w - w_1) = C_+ \]  \hspace{1cm}  \text{(A7)}

The inverse solution $P(x, t)$ of equation (A6) with $C_+$ as given by equation (A7) can be determined directly from equation (26) in the body of this report.

Problem 2 - Harmonic Pressure $t > 0$

Again for $s(x) = 0$, the solution of equation (A2) in a semi-infinite domain is given by equation (A6). In this case however, the constant $C_+$ is determined by the transformation of the boundary condition equation (27) to give
\[ p(0, w) = \delta(w - w_1) = C_+ \]  \hspace{1cm}  \text{(A8)}

Considering equations (A6) and (A8), the only residue is $w_1$, which lies on the real axis. Therefore, the inverse can be found directly from equations (A4) and (A5) to yield $P(x, t)$ as given by equation (31) in the body of this report.

Problem 3 - Green's Function $- \infty < t < \infty$

For a source in the form as given in equation (12), the right hand side of equation (A2) takes the form
\[ \frac{\partial^2 P}{\partial t^2} + \frac{\omega^2}{c^2} P = -\delta(x - x_s) \delta(w - w_1) \]  \hspace{1cm}  \text{(A9)}

The homogeneous solution for nonreflected waves in this particular example can be written as (Ref. 14, p. 133)
\[ p(x, w) = A e^{iwx/x_1/c} \quad x > x_s \]
\[ = A e^{-iwx/x_1/c} \quad x < x_s \]  \hspace{1cm}  \text{(A10)}

for a pressure continuous across $x = x_s$. There is, of course, a discontinuity in the slope of $p(x, w)$ at $x = x_s$. The constant $A$ and therefore the magnitude of the discontinuity takes on the value $x_i - x_1$ to satisfy the right hand side of equation (A9) at $x = x_s$. Following the procedure outlined in great detail by Morse and Ingard (Ref. 14, p. 133) the constant $A$ is found to be
\[ A = \frac{ic}{w_1} \delta(w - w_1) \] (A11)

Again, the inverse can be easily found from the direct use of the Dirac delta function (Ref. 14, p. 31, eq. 1.3.24) to yield $P(x, t)$ as given by equation (30) in the body of this report.

Problem 4 - Modified Green's Function $t > 0$

For a source in the form as given in equation (11), the right hand side of equation (A2) takes the form
\[ \frac{\partial^2 P}{\partial t^2} + \frac{\omega^2}{c^2} P = -\delta(x - x_s) \]  \hspace{1cm}  \text{(A12)}

Using the approach outlined in Problem 3 the Fourier transformed pressure becomes
\[ p(x, w) = -C e^{iwx/x_1/c} \]  \hspace{1cm}  \text{(A13)}

In taking the inverse of equation (A12), three residues are at $0$, $w_1$, and $w_1 + ib$ and all lie on or below the real axis. Therefore, the inverse can again be found directly from equations (A4) and (A5) to yield $P(x, t)$ as given by equation (31) in the body of this report.

Problem 5 - Ramping Function $t > 0$

For a source in the form as given in equation (33) the right hand side of equation (A2) takes the form
\[ \frac{\partial^2 P}{\partial t^2} + \frac{\omega^2}{c^2} P = \delta(x - x_s) \]  \hspace{1cm}  \text{(A14)}

The solution for the Fourier transformed pressure becomes
\[ p(x, w) = C e^{iwx/x_1/c} \]  \hspace{1cm}  \text{(A15)}

In taking the inverse of equation (A14), three residues are at $0$, $w_1$, and $w_1 + ib$ and all lie on or below the real axis. Therefore, the inverse can again be found directly from equations (A4) and (A5) to yield
\[ P(x, t) = \frac{1}{2\omega_1} \left[ e^{i(x - x_s)/c} - e^{-i(x - x_s)/c} \right] \]  \hspace{1cm}  \text{(A16)}

Continued on the next page.
For long times, the last two terms in equation (A15) can be neglected. In addition, if \( b/w_1 \) is assumed to be small, equation (A15) can be approximated by equation (34) in the body of this report.

References

Figure 1. - Finite difference grid.

Figure 2. - Nature of Green's function and numerical solutions when source is turned on at time equal to zero.
Figure 3 - Green function solution to the inhomogeneous wave equation using a ramped source \( u(x, t) = u_0 \sin(x - \omega t) \).

Figure 4 - Green function solution to the inhomogeneous wave equation using the integration technique \( (\text{f} - 2 \lambda) \).
Figure 5. - Distributed source solution to the inhomogeneous wave equation using the integration technique (t=4).