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**NASA
Technical
Memorandum**



NASA TM-82547

**FY 1983 SCIENTIFIC AND TECHNICAL REPORTS,
ARTICLES, PAPERS, AND PRESENTATIONS**

Compiled by Sarah S. Thacker
Management Operations Office

November 1983

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Space Administration

George C. Marshall Space Flight Center

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FOREWORD

In accordance with the NASA Space Act of 1958 the MSFC has provided for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof.

Since July 1, 1960, when the George C. Marshall Space Flight Center was organized, the reporting of scientific and engineering information has been considered a prime responsibility of the Center. Our credo has been that "research and development work is valuable, but only if its results can be communicated and made understandable to others."

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GEORGE C. MARSHALL SPACE FLIGHT CENTER
Marshall Space Flight Center, Alabama

FY 1983 SCIENTIFIC AND TECHNICAL REPORTS,
ARTICLES, PAPERS, AND PRESENTATIONS

TABLE OF CONTENTS

	Page
NASA TECHNICAL MEMORANDA	1
NASA TECHNICAL PAPERS.....	12
MSFC CONFERENCE PUBLICATIONS.....	16
NASA CONTRACTOR REPORTS	17
MSFC PAPERS CLEARED FOR PRESENTATION	31

TM-82514 April 1983
 MSFC Scan Stage 1 Workshop. James L. Green, J. H. Waite, J. F. E. Johnson, Joseph R. Doupnik, and Rod A. Heelis. Space Science Laboratory. N83-30095

This report describes the planning, implementation, and accomplishments of the first Space plasma Computer Analysis Network (SCAN) workshop held at Marshall Space Flight Center (MSFC). The purpose of the workshop was to identify specific cooperative scientific study topics within the discipline of Ionosphere Magnetosphere Coupling processes and to develop methods and procedures to accomplish this cooperative research using SCAN facilities. Cooperative scientific research was initiated in the areas of polar cusp composition, O⁺ polar outflow, and magnetospheric boundary morphology studies and an approach using a common metafile structure was adopted to facilitate the exchange of data and plots between the various workshop participants. The advantages of in person versus remote workshops were discussed also.

TM-82515 March 1983
 Atmospheric Environment for Space Shuttle (STS-5) Launch. D. L. Johnson, C. K. Hill, and G. W. Batts. Space Science Laboratory. N83-22908

This report presents a summary of selected atmospheric conditions observed near Space Shuttle STS-5 launch time on November 11, 1982, at Kennedy Space Center, Florida. Values of ambient pressure, temperature, moisture, ground winds, visual observations (cloud), and winds aloft are included. The sequence of pre-launch Jimsphere measured vertical wind profiles is given in this report. Also presented are the wind and thermodynamic parameters measured at the surface and aloft in the SRB descent/impact ocean area. Final meteorological tapes, which consists of wind and thermodynamic parameters versus altitude, for STS-5 vehicle ascent and SRB descent have been constructed. The STS-5 ascent meteorological data tape has been constructed by Marshall Space Flight Center in response to Shuttle task agreement No. 936-53-22-368 with Johnson Space Center.

TM-82516 February 1983
 Pseudo-Random Number Generator for the Sigma V Computer. Stanley N. Carroll. Systems Dynamics Laboratory. N83-23084

A technique is presented for developing a pseudo-random number generator based on the linear congruential form. The two numbers used for the generator are a prime number and a corresponding primitive root, where the prime is the largest prime number that can be accurately represented on a particular computer. The primitive root is selected by applying Marsaglia's lattice test. The technique presented has been applied to write a new random number program for the Sigma V computer. The new program, named S:RANDOM1, is judged to be superior to the older program named S:RANDOM. For applications requiring several independent random number generators, a table is included showing several acceptable primitive roots. The technique and programs described in the report can be applied to any computer having word length different from that of the Sigma V.

TM-82517 January 1983
 Attitude Control and Drag Compensation Propulsion System for the Gravity Probe-B Spacecraft. D. H. Blount. Structures and Propulsion Laboratory. N83-22315

An on-board propulsion system for attitude control and drag compensation is presented which uses helium boiloff gas from an experiment package dewar as propellant. This boiloff gas would normally be vented non-propulsively. Use of a small allowable temperature range in the dewar is exploited to store helium and accommodate incompatibilities in dewar heat leak and thruster demand flow over periods of more than one orbit. A relatively detailed thermodynamics analysis of the two-phase helium dewar and simulation of pressure loss through the helium distribution system is included.

TM-82518 January 1983
 Gravity Probe-B Control Subsystem. John Farmer. Systems Dynamics Laboratory. N83-23336

NASA TECHNICAL MEMORANDA

The purpose of this document is to identify and briefly examine the control problems of the proposed Gravity Probe-B spacecraft. Suggestions are made for control thruster geometry, dynamics, and maximum output value. Maximum total thrust is also a consideration since all thrusters expel gaseous helium produced from a common liquid supply. Control philosophy and preliminary designs are presented for both the attitude and drag free control systems. A radial separation of the spacecraft center of mass and the center of the proof mass cavity produces cross coupling between the attitude and drag free systems. The low available thrust implies that this separation, throughout the mission, must be kept within very close tolerance. For this reason an on-board mass balance control system may be necessary.

Simulation was performed only to the extent necessary to show the control concepts to be feasible.

TM-82520 April 1983
Analysis of "Pull-Test" Tools and Their Limitations as Applied to Terminal Junction Blocks. James Lee Smith, Information and Electronics System Laboratory. N83-24553

Discovery of unlocked contacts in Deutsch Block terminal junctions in Solid Rocket Booster flight hardware prompted an investigation into "pull-test" techniques to help insure against possible failures. In this paper, internal frictional forces between socket and pin and between wire and grommet were examined. Pull-test force must be greater than internal friction yet less than the crimp strength of the pin or socket. For this reason, a 100 percent accurate test is impossible. Test tools were evaluated. Currently available tools are adequate for "pull-testing."

TM-82521 December 1982
Liquid Helium Management for Gravity Probe-B. Charles Schafer, Space Science Laboratory. N83-24830

The Gravity Probe-B (GP-B) experiment will be degraded if accelerations at a proof mass become larger than 10^{-10} g. This makes necessary the management of the configuration and

dynamical behavior of the large amount of liquid helium present in the GP-B spacecraft dewar. Three approaches to the solution of this problem are discussed. It is concluded that the most promising technique involves the use of baffles into which the liquid helium can be forced during a relatively high spacecraft rotation period, and in which the liquid helium will be held by capillary forces during the operational period when the rotation rate is much lower. Some likely baffle configurations are suggested.

TM-82522 January 1983
Significant Events in Low-Level Flow Conditions Hazardous to Aircraft. Margaret B. Alexander and Dennis W. Camp. Space Science Laboratory. N83-25268

Atmospheric parameters recorded at the NASA 150-Meter Ground Winds Tower Facility at Kennedy Space Center during high surface winds are analyzed to determine magnitude, frequency, duration, and simultaneity of occurrence of low-level flow conditions known to be hazardous to the ascent and descent of conventional aircraft and the Space Shuttle. Graphic and tabular presentations of mean and extreme values and simultaneous occurrences of turbulence (gustiness and gust factor), wind shear (speed and direction), and vertical motion (updrafts and downdrafts), along with associated temperature inversions are included as a function of tower height, layer and/or distance for six 5-sec intervals (one interval every 100 sec) of parameters sampled simultaneously at the rate of 10 speeds, directions and the temperatures per second during an approximately 10-min period (2143 47.0 to 2152 16.9 UT) on July 3, 1973.

TM-82523 April 1983
A Study Into the Loss of Lock of the Space Telescope Fine Guidance Sensor. Michael E. Polites, Systems Analysis and Integration Laboratory. N83-26768

This report documents the results of a study into the loss of lock phenomenon associated with the Space Telescope (ST) Fine Guidance Sensor (FGS). The primary cause of loss of lock has been found to be a combination of cosmic ray spikes and photon noise due to a 14.5 Mv star. The

NASA TECHNICAL MEMORANDA

probability of maintaining lock versus time is estimated both for the baseline FGS design and with parameter changes in the FGS firmware which will improve the probability of maintaining lock. The parameters varied are changeable in-flight from the ground and hence do not impact the design of the FGS hardware.

TM-82524 February 1983
STS-2, -3, -4 Induced Environment Contamination Monitor (IECM) Summary Report. Edited by E. R. Miller. Space Science Laboratory. N83-24539

A brief description of the STS-2, -3, and -4 missions is given with the location of the IECM in the payload bay and the Shuttle coordinate systems used in this report. Measurement results from the three flights are given in the following sections for each instrument with comparisons to original goals for preflight environment and induced environment contamination. These results include very low levels of molecular mass accumulation rates, absence of molecular films on optical samples, out-gassing species above 50 amu undetectable, general low levels of on-orbit particulates, and decay rates for early mission and water dump particulates. Results of exposure of several optical materials and coatings to atomic oxygen are also presented. From these results, it is concluded that the Space Shuttle has met the established induced environment contamination goals.

TM-82525 April 1983
Materials Processing in Space Program Tasks. Compiled by Elizabeth Pentecost. Space Science Laboratory. N83-25737

This report is a compilation of the active research tasks as of the end of the fiscal year 1983 of the Materials Processing in Space Program, NASA-Office of Space and Terrestrial Applications, involving several NASA centers and other organizations. The purpose of this document is to provide an overview of the program scope for managers and scientists in industry, university, and government communities. The report is structured to include an introductory description of the program, its history, strategy and overall goal; identification of the organiza-

tional structures and people involved; and a description of each research task, together with a list of recent publications.

The tasks are grouped into four categories: Crystal Growth; Solidification of Metals, Alloys, and Composites; Fluids, Transports, and Chemical Processes; and Ultrahigh Vacuum and Containerless Processing Technologies.

TM-82526 April 1983
Statistical Aspects of the 1980 Solar Flares - III. Parametric Comparisons and Final Comments. Robert M. Wilson. Space Science Laboratory.

Based on 1349 H α flares with X-ray counterparts occurring near solar maximum, an investigation into the relationship between pairs of parameters, including rise time, decay time, H α importance, and X-ray class, has been accomplished. As past experience has shown, it is found that, on average, long H α rise-time flares tend to have long H α decay time (on average about 2.9 times longer than the associated rise time), are more likely associated with areal class \geq class 1 and relative brightness class bright, and are more likely associated with X-ray events of X-ray class \geq C5. Also, it is noted that during 1980 2800-MHz radio flux (denoted F₂₈₀₀) appeared to crudely track the south latitudinal regions for the first 10 months of the year and thereafter the north latitudinal regions, at least through December 1980, the Solar Maximum Year, which met certain selection criteria. No effort has been made to model flare frequency correlation and distribution based on more advanced statistical techniques.

TM-82527 April 1983
The Feasibility of Low-G Grey Solidification of Nodular Iron in the F-104 Experimental Furnace Package. P. A. Curreri, G. A. Smith and G. Workman. Space Science Laboratory.

The rationale for low-g experiments with cast iron and the need for solidification in the grey form during these experiments are reviewed. The factors which determine whether an iron melt will solidify grey or white are discussed. Cooling rate versus microstructure was studied

NASA TECHNICAL MEMORANDA

for a nodular iron candidate material for F-104 low-g solidification. The study determined that low-g grey solidification, using the present F-104 furnace system, of the nodular iron composition studied is not feasible. Specimen microstructure strongly suggested that the F-104 furnace's gas cooling system was causing excessive localized chill resulting in the nucleation of the unwanted iron carbide phase. A change is suggested, in the quench system design, that could possibly overcome this problem.

TM-82528 April 1983
 Preliminary Science Report on the Directional Solidification of Hypereutectic Cast Iron During KC-135 Low-g Maneuvers. P. A. Curreri, D. M. Stefanescu, and J. C. Hendrix, Space Science Laboratory.
N83-25854

An ADSS-P directional solidification furnace has been reconfigured for operation on the KC-135 low-g aircraft. The system offers many advantages over quench ingot methods for study of the effects of sedimentation and convection on alloy formation. The directional solidification furnace system was first flown during the September 1982 series of flights. The microstructure of the hypereutectic cast iron sample solidified on one of those flights suggests a low-g effect on graphite morphology. Further experiments are needed to ascertain that this effect is due to low-gravity and to deduce which of the possible mechanisms is responsible for it.

TM-82529 May 1983
 Atmospheric Environment for Space Shuttle (STS-6) Launch. D. L. Johnson, C. K. Hill, and G. W. Batts. Systems Dynamics Laboratory.
N83-29926

This report presents a summary of selected atmospheric conditions observed near Space Shuttle STS-6 launch time on April 4, 1983, at Kennedy Space Center, Florida. Values of ambient pressure, temperature, moisture, ground winds, visual observations (cloud), and winds aloft are included. The sequence of prelaunch Jimsphere measured vertical wind profiles is given in this report. Also presented are the wind and thermodynamic parameters measured at the

surface and aloft in the SRB ascent/impact ocean area. Final meteorological tapes, which consist of wind and thermodynamic parameters versus altitude, for STS-6 vehicle ascent and SRB descent have been constructed. The STS-6 ascent meteorological data tape has been constructed by Marshall Space Flight Center in response to Shuttle task agreement No. 936-53-22-368 with Johnson Space Center.

TM-82530 June 1983
 Thunderstorm Observations from Space Shuttle. B. Vonnegut, O. H. Vaughan, Jr., and M. Brook. Space Science Laboratory.
N83-31246

This report covers the results of the Night-time/Daytime Optical Survey of Lightning (NOSL) experiments done on the STS-2 and STS-4 flights. During these two flights of the Space Shuttle Columbia, the astronaut teams of J. Engle and R. Truly, and K. Mattingly II and H. Hartsfield took motion pictures of thunderstorms with a 16 mm cine camera. Film taken during daylight showed interesting thunderstorm cloud formations, where individual frames taken tens of seconds apart, when viewed as stereo pairs, provided information on the three-dimensional structure of the cloud systems. Film taken at night showed clouds illuminated by lightning with discharges that propagated horizontally at speeds of up to 10^5 m/sec⁻¹ and extended for distances on the order of 60 km or more.

TM-82531 June 1983
 Interim Report on Microfissuring of Inconel 718. A. C. Nunes, Jr. Materials and Processes Laboratory.
N83-29356

A tentative mathematical computer model of the microfissuring process during electron beam welding of Inconel 718 has been constructed. Predictions of the model are compatible with microfissuring tests on eight 0.25-in. thick test plates. The model takes into account weld power and speed, weld loss (efficiency), parameters and material characteristics. Besides the usual material characteristics (thermal and strength properties), a temperature and grain size dependent critical fracture strain is required by the

NASA TECHNICAL MEMORANDA

model. The model is based upon fundamental physical theory (i.e., it is not a mere data interpolation system), and can be extended to other metals by suitable parameter changes.

TM-82532 June 1983
The Variable Polarity Plasma Arc Welding Process: Its Application to the Space Shuttle External Tank -- First Interim Report. A. C. Nunes, Jr., E. O. Bayless, Jr., C. S. Jones III, P. M. Munafò, A. P. Biddle, and W. A. Wilson. Materials and Processes Laboratory. N83-31021

The Variable Polarity Plasma Arc (VPPA) welding process is being introduced as a partial replacement for the Gas Shielded Tungsten Arc process in assembly welding of the Space Shuttle External Tank. This report describes the technical history of the development of the VPPA process, interim results of the weld strength qualification studies, and plans for further work in the implementation of the VPPA process.

TM-82533 June 1983
An Evaluation of Grease Type Ball Bearing Lubricants Operating in Various Environments (Status Report No. 7). E. L. McMurtrey. Materials and Processes Laboratory. N83-31020

Because many future spacecraft or space stations will require mechanisms to operate for long periods of time in environments which are adverse to most bearing lubricants, a series of tests is continuing to evaluate 38 grease-type lubricants in R-4 size bearings in five different environments for a 1-year period. Four repetitions of each test are made to provide statistical samples. These tests have also been used to select four lubricants for 5-year tests in selected environments with five repetitions of each test for statistical samples. At the present time, 142 test sets have been completed and 30 test sets are underway. The three 5-year tests in (1) continuous operation and (2) start-stop operation, with both in vacuum at ambient temperatures, and (3) continuous vacuum operation at 93.3°C are now completed. To date, in both the 1-year and 5-year tests, the best results in all environments have been obtained with a high viscosity index perfluoroalkylpolyether (PFPE) grease.

TM-82534 June 1983
Sensitivity/Comparison Study Between the Jacchia 1970, 1971, and 1977 Upper Atmospheric Density Models. Dale L. Johnson. Systems Dynamics Laboratory.

The neutral upper atmospheric models for the Earth's thermosphere currently used in NASA-MSFC programs are the Jacchia 1970 (J70), 1971 (J71), and 1977 (J77). The Jacchia 1970 model (modified) is used in all MSFC orbital mechanics analyses. Since total density and its variations are the main environmental parameters of interest in orbital lifetime and attitude control studies, the total neutral density was selected for this analysis.

This report presents the results of a parametric study of the total density (at 400 km altitude) as computed with the three MSFC/Jacchia models. The sensitivity of each of the density models at the summer solstice to varying solar conditions (flux) and geomagnetic (index) values is discussed.

TM-82535 August 1983
Space Processing Applications Rocket (SPAR) Project SPAR VII Final Report. Compiled by Richard M. Poorman.

The Space Processing Applications Rocket Project (SPAR) VII Final Report contains the compilation of the post-flight reports of each of the Principal Investigations (PI's) of the three selected science payloads, in addition to the engineering report as documented by the Marshall Space Flight Center (MSFC). This combined effort also describes pertinent portions of ground-based research leading to the ultimate selection of the flight sample composition, including design, fabrication and testing, all of which are expected to contribute to an improved comprehension of materials processing in space.

The SPAR project is coordinated and managed by MSFC as part of the Materials Processing in Space (MPS) program of the Office of Space and Terrestrial Applications (OSTA) of NASA Headquarters.

NASA TECHNICAL MEMORANDA

This technical memorandum is directed entirely to the payload manifest flown in the seventh of a series of SPAR flights conducted at the White Sands Missile Range (WSMR) and includes the experiments entitled, "Containerless Processing Technology," "Containerless Processing Bubble Dynamics," and "Comparative Alloy Solidification."

TM-82536 June 1983
Instrumental Effects on the Temperature and Density Derived from the Light Ion Mass Spectrometer, Paul D. Craven and David L. Reasoner, Space Science Laboratory, N83-30294

An expression for the flux into an RPA is derived which takes into account the instrumental effect of a dependence on energy of the solid angle of the acceptance cone. A second instrumental effect of a limited bandpass is briefly discussed. Using the characteristics of the LIMS instrument on SCATHA, it is shown that temperatures and densities derived without considering the effect of the solid angle dependence on energy will be too low, dramatically so for $E_t > E_1$, where E_1 is the e folding distance of the solid angle dependence and E_t is the thermal energy of the plasma. For $E_t \ll E_1$, there is effectively no impact on the derived temperatures and densities if the solid angle effect is ignored.

TM-82537 August 1983
Spacelab Mission 1 Experiment Descriptions - Third Edition, Edited by Paul D. Craven, Space Science Laboratory.

This document presents brief descriptions of experiments and facilities planned for Spacelab 1. These experiments and facilities were selected from the responses to the Announcement of Opportunity for the first Spacelab mission. The experiments described here have been selected for flight.

This edition supersedes NASA TM-82448, November 1981.

TM-82538 July 1983
Image Motion Compensation by Area Correlation and Centroid Tracking of Solar

Surface Features, M. E. Nein, W. R. McIntosh, and N. P. Cumings, Advanced Systems Office, N83-32690

An experimental solar correlation technique has been tested and evaluated on a ground-based solar magnetograph. Using sunspots as fixed targets, tracking error signals were derived by which the telescope image was stabilized against wind induced perturbations. Two methods of stabilization were investigated; mechanical stabilization of the image by controlled two-axes motion of an active optical element in the telescope beam, and electronic stabilization by biasing of the electron scan in the recording camera. Both approaches have demonstrated telescope stability of about 0.6 arc sec under random perturbations which can cause the unstabilized image to move up to 120 arc sec at frequencies up to 30 Hz.

TM-82539 July 1983
Ocular Screening Tests of Elementary School Children, John Richardson, Technology Utilization Office.

This report presents an analysis of 507 abnormal retinal reflex images taken of Huntsville kindergarten and first grade students.

The retinal reflex images were obtained by using an MSFC-developed Generated Retinal Reflex Image System (GRRIS) photorefractor. The system uses a 35 mm camera with a telephoto lens with an electronic flash attachment. Slide images of the eyes were examined for abnormalities.

Of a total of 1835 students screened for ocular abnormalities, 507 were found to have abnormal retinal reflexes. The types of ocular abnormalities detected were hyperopia, myopia, astigmatism, esotropia, exotropia, strabismus, and lens obstructions.

The report shows that the use of the photorefractor screening system is an effective low-cost means of screening school children for abnormalities.

NASA TECHNICAL MEMORANDA

TM-82540

July 1983

Materials Investigation of STS-3 Parachute Failure. Ronald L. Nichols. Materials and Processes Laboratory.

Main parachute, No. 2 of SRB A-12 on STS-3, sustained damage during deployment or initial inflation that resulted in its collapse and failure to sustain load. During an investigation of the materials from this parachute, optical and scanning electron microscope analyses were conducted. This examination identified stains and abrasions on vent lines that appear to have been a result of friction contact with its flotation bag lanyard. Mechanical testing of the vent band indicated a reduction in strength of 37 percent obviously due to structural overload, heat, and ocean water exposure. It is concluded from this and other available data that entanglement of parachutes No. 1 and No. 2 during deployment caused adequate structural damage to main parachute No. 2 to render it unable to carry load.

TM-82541

May 1983

The Marshall Space Flight Center KC-135 Zero Gravity Test Program for FY 1982. Edited by R. E. Shurney. Systems Analysis and Integration Laboratory.

During FY82, researchers and experimenters from Marshall Space Flight Center (MSFC) conducted 11 separate investigations during 26.3 hr of testing aboard the KC-135 zero-gravity aircraft, based at Ellington Air Force Base, Texas. Although this represented fewer hours than initially projected, all experiment and test objectives were met or exceeded. This Technical Memorandum compiles all results achieved by MSFC users during FY82, a year considered to be highly productive.

We thank the aircraft operations people at Johnson Space Center for their enthusiastic support this year and in years past.

TM-82542

July 1983

Atmospheric Environment for Space Shuttle (STS-7) Launch. D. L. Johnson, C. K. Hill, and G. W. Batts. Systems Dynamics Laboratory.

This report presents a summary of selected conditions observed near Space Shuttle STS-7 launch time on June 18, 1983, at Kennedy Space Center, Florida. Values of ambient pressure, temperature, moisture, ground winds, visual observations (cloud), and winds aloft are included. The sequence of prelaunch Jimsphere measured vertical wind profiles is given in this report. Also presented are wind and thermodynamic parameters representative of surface and aloft conditions in the SRB descent/impact ocean area. Final meteorological tapes, which consist of wind and thermodynamic parameters versus altitude, for STS-7 vehicle ascent and Acoustic/SRB descent have been constructed. The STS-7 ascent meteorological data tape has been constructed by Marshall Space Flight Center in response to Shuttle task agreement No. 936-53-22-368 with Johnson Space Center.

TM-82543

August 1983

Time-Dependent Response of Filamentary Composite Spherical Pressure Vessels. Jan D. Dozier. Structures and Propulsion Laboratory.

A filamentary composite spherical pressure vessel is modeled as a pseudo-isotropic (or transversely isotropic) composite shell, with the effects of the liner and fill tubes omitted. Equations of elasticity, macromechanical and micromechanical formulations, and laminate properties are derived for the application of an internally pressured spherical composite vessel. Viscoelastic properties for the composite matrix are used to characterize time-dependent behavior. Using the maximum strain theory of failure, burst pressure and critical strain equations are formulated, solved in the Laplace domain with an associated elastic solution, and inverted back into the time domain using the method of collocation. Viscoelastic properties of HBFR-55 resin are experimentally determined and a Kevlar/HBFR-55 system is evaluated with a FORTRAN program. The computed reduction in burst pressure with respect to time indicates that the analysis employed may be used to predict the time-dependent response of a filamentary composite spherical pressure vessel.

NASA TECHNICAL MEMORANDA

TM-82544

July 1983

Development of Low Viscosity Alkane-Based Urethane for Connector Potting Applications. Donald E. Morris. Materials and Processes Laboratory.

Two series of saturated hydrocarbon-based urethanes were prepared with isophorone diisocyanate and one series with methyl bis (4-cyclohexyl isocyanate). The urethanes with molecular weights as great as 2500 had viscosities low enough and a working life long enough to be used in potting, molding, and coating applications. Specimens were prepared and mechanical properties such as hardness, tensile strength elongation, and tear strength were determined. Thermomechanical properties (glass transition and expansion coefficient) and thermogravimetric properties were determined.

TM-82545

August 1983

NASA Thunderstorm Overflight Program — Research in Atmospheric Electricity from an Instrumented U-2 Aircraft Platform. Otha H. Vaughan, Jr. Systems Dynamics Laboratory.

This report presents an overview of the NASA Thunderstorm Overflight Program (TOP) being conducted by the Marshall Space Flight Center and university researchers in atmospheric

electricity. Discussed in this report are the various instruments flown on the NASA U-2 aircraft, as well as the ground instrumentation used in 1982 to collect optical and electronic signatures from the lightning events. Samples of some of the photographic and electronic signatures are presented. Approximately 6400 electronic data samples of optical pulses were collected and are being analyzed by the NASA and university researchers. A number of research reports are being prepared for future publication. These reports will provide more detailed data analysis and results from the 1982 spring and summer program.

TM-82546

September 1983

Solid Rocket Booster Thrust Vector Control Subsystem Description. Compiled by John Redmon, Jr. Structures and Propulsion Laboratory.

This document identifies major Solid Rocket Booster-Thrust Vector Control (SRB-TVC) subsystem components and subcomponents used in the Space Transportation System (STS). Simplified schematics, detailed schematics, figures, photographs, and data are included to acquaint the reader with the operation, performance, and physical layout as well as the materials and instrumentation used.

NASA TECHNICAL PAPERS

TP-2110 October 1982
 Optical Observations of Unidirectional Solidification in Microgravity, Mary H. Johnston, Robert B. Owen, and Robert E. Shurney, Materials and Processes Laboratory.
 N83-16492

Optical interferometric, shadowgraph, and streak photographic methods are used to obtain temperature profiles, concentration gradients, and fluid velocities in $\text{NH}_4\text{Cl-H}_2\text{O}$ metal-model solution during unidirectional solidification in microgravity. This study elucidates earlier low-gravity suborbital rocket experiments and lays the groundwork for future Space Shuttle experiments. The design and operation of the optical units are presented, and results are detailed and compared to ground based experiments. The low-gravity experiments were conducted using a NASA KC-135 aircraft flying a parabolic trajectory.

TP-2113 November 1982
 An Electrochemical Method for Determining Hydrogen Concentrations in Metals and Some Applications. Merlin D. Danford, Materials and Processes Laboratory.
 N83-16491

An electrochemical method has been developed for the determination of hydrogen in metals using the EG&G-PARC Model 350A Corrosion Measurement Console. The method has been applied to hydrogen uptake, both during electrolysis and electroplating, and to studies of hydrogen elimination and the effect of heat treatment on elimination times. Results from these studies are presented.

TP-2115 October 1982
 Flow in a Torsionally Oscillating Filled Cylinder. Charles F. Schafer, Space Science Laboratory.
 N83-16674

The flow of a liquid in a completely filled cylinder undergoing torsional oscillations about its longitudinal symmetry axis was studied analytically and experimentally. The objective of the studies was to determine the efficacy of the torsional oscillations in mixing the confined liquid. Flow was found to be confined primarily

to toroidal cells at the ends of the cylinder. Cell thickness was about equal to the cylinder radius. The use of baffles at the end walls was shown to enhance the mixing process.

TP-2117 October 1982
 Identification and Management of Filament-Wound Case Stiffness Parameters. V. Verdaine and M. Rheinfurth, Systems Dynamics Laboratory.
 N83-16400

The high specific strength and the high specific modulus made graphite-epoxy laminate an expedient material substitute for the Shuttle Solid Rocket Motor steel case to substantially increase the payload performance without increasing the composite case axial growth during thrust build-up which was constrained to minimize liftoff excitation effects on existing structural elements and interfaces. Parameters associated with axial growth were identified for quality and manufacturing controls. Included is an innovative method for experimentally verifying extensional elastic properties on a laminate pressurized test bottle.

TP-2124 October 1982
 Mechanics of Liquid Helium in a Partially Filled Rotating Dewar in Low Gravity — With Application to Gravity Probe-B. C. F. Schafer and S. A. Lowry, Space Science Laboratory.
 N83-16680

The Gravity Probe-B (GP-B) spacecraft is composed largely of a liquid helium dewar containing an experiment package. It is shown that an unsymmetric liquid helium distribution in the dewar can cause unacceptably high forces, gravitational and gravity gradient forces, at the experiment location. It is further shown that for the planned spacecraft configuration and operational parameters, it is very likely that the liquid helium distribution in the dewar will be unsymmetric. The required symmetry can be attained by using higher operational spacecraft rotation rates. An alternative solution to this problem will be discussed in a later report.

TP-2144 February 1983
 A Study of Production of Miscibility Gap Alloys with Controlled Structures. R. A.

NASA TECHNICAL PAPERS

Parr, M. H. Johnston, J. A. Burka, J. H. Davis, and J. A. Lee, Materials and Processes Laboratory. N83-21991

Composite materials were directionally solidified using a new technique to align the constituents longitudinally along the length of the specimen. In some instances a tin coating was applied and diffused into the sample to form a high transition temperature superconducting phase. The superconducting properties were measured and compared with the properties obtained for powder composites and re-directionally solidified powder compacts. The samples which were compacted and re-directionally solidified showed the highest transition temperature and widest transition range. This indicates that both steps, powder compaction and resolidification, determine the final superconducting properties of the material.

TP-2168 March 1983
The Pinhole/Occulter Facility, Edited by E. A. Tandberg-Hanssen, H. S. Hudson, J. R. Dabbs, and W. A. Baity. N83-25646

The outer solar atmosphere exhibits a great variety of dynamic and energetic plasma phenomena, from the catastrophic energy release of solar flares to the steady acceleration of the solar wind. Observations from space in the past two maxima of the solar activity cycle have more than whetted the appetite for understanding the physics of the solar corona. The Pinhole/Occulter Facility contains the instruments necessary for achieving fuller knowledge: broad-band X-ray imaging, combined with simultaneous ultraviolet and white-light spectroscopy and imaging.

X-ray astronomy has progressed, through the surveys by small satellites and the "deep" observations of soft X-rays by the Einstein Observatory, to a level at which it has become a major component of astronomical investigation. The Pinhole/Occulter represents the first serious effort to broaden the spectral band available to X-ray astronomers at high angular resolution (below one arc second), and it is thus an effective complement to AXAF and other future soft X-ray facilities.

TP-2173 February 1983
An Efficient Algorithm for Generating Random Number Pairs Drawn from a Bivariate Normal Distribution. C. Warren Campbell, Space Science Laboratory. N83-27679

An efficient algorithm for generating random number pairs from a bivariate normal distribution was developed. Any desired value of the two means, two standard deviations, and correlation coefficient can be selected. Theoretically the technique is exact and in practice its accuracy is limited only by the quality of the uniform distribution random number generator, inaccuracies in computer function evaluation, and arithmetic. A FORTRAN routine was written to check the algorithm and good accuracy was obtained. Some small errors in the correlation coefficient were observed to vary in a surprisingly regular manner. A simple model was developed which explained the qualitative aspects of the errors.

TP-2185 April 1983
Development of a Simplified Optical Technique for the Simultaneous Measurement of Particle Size Distribution and Velocity. James Lee Smith, Information and Electronic Systems Laboratory. N83-27846

In an effort to develop a low cost, simplified optical technique for measuring particle size distributions and velocities in fluidized bed combustors and gasifiers, a two-phase research project was initiated at the University of Tennessee at Chattanooga. Phase One, the object of this report, consisted of the following:

- 1) Existing techniques were surveyed.
- 2) An experimental procedure was developed.
- 3) A laboratory test model was fabricated.
- 4) Limited data was recovered for proof of principle.
- 5) The relationship between particle size distribution and amplitude measurements was illustrated.

NASA TECHNICAL PAPERS

A He-Ne laser illuminated Ronchi Rulings (range 10 to 500 lines per inch). Various samples of known particle size distributions were passed through the fringe pattern produced by the rulings. A photomultiplier tube converted light from the fringe volume to an electrical signal which was recorded using an oscilloscope and camera. The signal amplitudes were correlated against the known particle size distributions. The correlation holds true for various samples.

TP-2192 March 1983

A Conceptual Framework for Using Doppler Radar Acquired Atmospheric Data for Flight Simulation. Warren Campbell. Systems Dynamics Laboratory. N83-27977

A concept is presented which will permit turbulence simulation in the vicinity of microbursts. The method involves a large data base, but should be fast enough for use with flight simulators. The model will permit any pilot to simulate any flight maneuver in any aircraft. The model will simulate a wind field with three-component mean winds and three-component turbulent gusts, and gust variation over the body of an aircraft so that all aerodynamic loads and moments can be calculated. The time and space variation of mean winds and turbulent intensities associated with a particular atmospheric phenomenon such as a microburst is used in the model. In fact, Doppler radar data such as provided by JAWS is uniquely suited for use with the proposed model. The concept is completely

general and is not restricted to microburst studies. Reentry and flight in terrestrial or planetary atmospheres could be realistically simulated if supporting data of sufficient resolution were available.

TP-2198 May 1983

Modal Analysis of a Nonuniform String with End Mass and Variable Tension. Mario H. Rheinfurth and Zachary J. Galaboff. Systems Dynamics Laboratory.

Modal synthesis techniques for dynamic systems containing strings describe the lateral displacements of these strings by properly chosen shape functions. This report provides an iterative algorithm to calculate the natural modes of a nonuniform string and variable tension for some typical boundary conditions including one end mass. Numerical examples are given for a uniform string in a constant and a gravity gradient force field.

TP-2216 June 1983

Analysis of Random Signal Combinations for Spacecraft Pointing Stability. Leonard Howell. Systems Dynamics Laboratory.

Methods for obtaining the probability density function of random signal combinations are discussed. These methods provide a realistic criteria for the design of control systems subjected to external noise with several important applications for aerospace problems.

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| Measurements of the Solar Shape and Diameter Using the Solar Beacon. For presentation at the AIAA 21st Aerospace Sciences Meeting, Reno, NV, January 1983. | | Measured Thermal Diffusivity of $Hg_{1-x}Cd_x$ Te Solids and Melts. For presentation at the 1983 MCT Workshop, Dallas, TX, January 8-10, 1983. | |
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| | | Plasmapause and Plasmasphere Structure from DE-1 Observations. For presentation of the fall meeting of the American Geophysical Union, San Francisco, CA, December 5-10, 1983, and for publication in EOS. | |
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