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NASA Case No. LEW-15, 818-1

PRINT FIG. 1

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AWARDS ABSTRACT

**NiAl-BASE COMPOSITE CONTAINING HIGH VOLUME
FRACTION OF AlN FOR ADVANCED ENGINES**

A particulate reinforced NiAl-AlN composite alloy has a NiAl matrix and greater than about 13 volume percent fine particles of AlN within the matrix. The particles preferably have a diameter from about 15 nanometers to about 50 nanometers. The particulate reinforced NiAl-AlN composite alloy may be prepared by cryomilling prealloyed NiAl in liquid nitrogen using grinding media having a diameter of from about 2mm to 6mm at an impeller speed of from about 450 RPM to about 800 RPM. The cryomilling may be done for a duration of from about 4 hours to about 20 hours to obtain a cryomilled powder. The cryomilled powder may be consolidated to form the particulate reinforced NiAl-AlN composite alloy. The particulate reinforced alloy can further include a toughening alloy. The toughening alloy may include NiCrAlY, FeCrAlY and FeAl.

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NASA Case No. LEW-15818-1

Patent Application

NiAl-BASE COMPOSITE CONTAINING HIGH VOLUME
FRACTION OF AlN FOR ADVANCED ENGINES

ORIGIN OF THE INVENTION

5 The invention described herein was made by employees
of the U.S. Government and a contractor employee and may be
manufactured and used by or for the U.S. Government without
the payment of any royalties thereon or therefor.

FIELD OF THE INVENTION

10 This invention relates to NiAl-based intermetallic
composite in general and more particularly to dispersion
strengthened NiAl-based composite alloys for use in very
high temperature structural applications including advanced
engine applications.

BACKGROUND OF THE INVENTION

15 There is great demand for structural materials that
can withstand adverse environments at increasingly higher
temperatures and stresses. The upper operating temperature
of conventional heat resistant alloys is limited to the
temperature at which second phase particles of the alloy are
20 substantially dissolved in the alloy matrix or to the
temperature at which the particles become severely
coarsened. Above this limiting temperature the alloys no
longer exhibit useful strength.

25 The high melting point, low density and excellent
oxidation/corrosion resistance of binary NiAl make it an
excellent candidate for high temperature structural
purposes. Its poor mechanical properties at elevated
temperature, however, indicate that use of the monolithic
material is improbable. In U.S. Patent No. 4,961,905 to Law
30 et al., NiAl intermetallic materials are modified by adding
amounts of an alloying material that render the NiAl
structure susceptible to a martensitic transformation. The
preferred alloying element is cobalt, which can double or
triple the fracture toughness of NiAl materials while

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simultaneously increasing the room temperature yield strength. Another approach for improving the properties of NiAl is by incorporating continuous reinforcing fibers for increased strength and improved damage tolerance. Single
5 crystal Al_2O_3 is the reinforcing fiber of choice because of its high modulus and strength, low density and chemical stability.

The mechanical properties of polycrystalline NiAl reinforced with 30 vol.% continuous single crystalline Al_2O_3
10 fibers were investigated and reported by Bowman, in "Influence of interfacial characteristics on the mechanical properties of continuous fiber reinforced NiAl composites," Mat. Res. Soc. Symp. Proc., Vol. 273, 1992. Bowman reports the mechanical properties of a NiAl composite reinforced
15 with either weakly or strongly bonded Al_2O_3 fibers. A strong interfacial bond between the Al_2O_3 and the NiAl is a requirement for oxidation resistance. However, the difference between the coefficients of thermal expansion for the Al_2O_3 fibers and for NiAl, together with the interfacial
20 strong bond, contributes to severe crack formation in the composite due to thermally generated internal stress. In the powder-cloth ("P-C") process for composite production discussed in Bowman, alloy matrix material is processed into flexible cloth-like sheets by combining matrix powders
25 with an organic polymer binder. The presence of the polymer binder inhibits fiber-matrix bonding, thereby increasing the matrix's resistance to cracking. However, in addition to low oxidation resistance afforded the material due to the weak bond, fiber distribution can be irregular
30 and difficult to control with P-C processing. This problem is considered by Bowman et al., in "Unresolved technical barriers to the development of a structurally viable $\text{Al}_2\text{O}_3/\text{NiAl}$ composite," HITEMP, NASA-CP-19117, 1993. Thus, efforts to increase the strength of NiAl by alloying and
35 compositing with continuous Al_2O_3 fibers have shown only

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limited improvements in the properties required for high temperature structural applications.

Mechanical alloying and related processes produce alloy materials with novel microstructures which can be used in structural applications. Cryomilling is high energy milling at cryogenic temperatures. Depending on the material system being used, cryomilling can also be a reaction milling process as in the case of milling prealloyed NiAl in liquid nitrogen. As the NiAl is fractured in liquid nitrogen, it reacts with the nitrogen, forming AlN on the surface of the powders. If there is any oxygen present, the fracture surfaces react with oxygen to form Al₂O₃ on the powder particles. The net result of milling NiAl in liquid nitrogen is an arrangement of fine particles of AlN, NiAl and Al₂O₃ on NiAl powder particle surfaces. Consolidation of the cryomilled NiAl powder via hot extrusion or hot isostatic pressing produces particulate strengthened materials with relatively high volume fractions of AlN. U.S. Patent No. 4,619,699, to Petkovic-Luton et al., describes a cryomilling process as a means to improve the mechanical properties of high temperature Fe- and Ni-based oxide dispersion strengthened alloys by control of their structure. Cryomilling of NiAl leads to the formation of a discontinuous AlN particle reinforced aluminide that shows good potential as a high creep strength, oxidation resistant material at elevated temperatures. Luton et al., in "Cryomilling of nano-phase dispersion strengthened aluminum," Mat. Res. Soc. Symp. Proc., Vol. 132, 1989, speculate that the high creep strength at elevated temperatures is due to a threshold stress for superplastic flow in the fine grain alloy formed when Al(ON) particles generated during the milling inhibit the grain growth in the aluminum matrix.

In Whittenberger et al., "Preliminary investigation of a NiAl composite prepared by cryomilling," J. Mat. Res., Vol 5, No. 2, Feb 1990, prealloyed Ni-51 (at.%) Al was

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cryomilled with Y_2O_3 to improve the high temperature strength resistance of the resultant alloy to form an NiAl-AlN composite alloy wherein the volume fraction of AlN within the NiAl matrix was estimated to be about 10%. Upon
5 compression testing at 1200 and 1400K, the creep strength was measured to be six times better than that of NiAl. NiAl-AlN composite alloys having volume fractions of up to 13% AlN were reported in Aiken et al., "Reproducibility of NiAl cryomilling," Mech. All. for Struct. App., Proc. 2nd
10 Int. Conf. Struct. Appl. Mech. All., Sept. 20-22 1993 and in Whittenberger et al. "1300 K Compressive Properties of a Reaction Milled NiAl-AlN Composite," J. Mater. Res., Vol. 5, No. 12, Dec. 1990, pp. 2819-2827. Compression testing of the NiAl-AlN alloys at 1100 and 1300K showed creep strengths
15 superior to those of γ' and oxide dispersion strengthened polycrystalline superalloys and results were considered to be very reproducible.

AlN formed from the reaction of nitrogen with Al during cryomilling has the deleterious effect of depleting
20 Al from the NiAl matrix, thereby lowering the cyclic oxidation resistance of the NiAl base material. Thus, as the AlN volume content of the NiAl-AlN alloy (and related high temperature properties) goes up, the oxidation resistance of the alloy goes down. Efforts to correct this
25 problem include adding about 0.5 wt.% Y_2O_3 to prealloyed NiAl powder prior to cryomilling. Lowell et al. in "Cyclic oxidation resistance of a reaction milled NiAl-AlN composite," Mat. Res. Soc. Symp. Proc., Vol. 194, 1990, compare the cyclic oxidation resistance of a NiAl-AlN alloy material
30 wherein Y_2O_3 was added during cryomilling of NiAl with 0.15 atom % Zr. This probably improves the oxide scale adherence, thereby minimizing the Al consumption during cyclic oxidation. The cyclic oxidation limit for the NiAl-AlN was found to be about 1473K for times greater than 100 hours and
35 1573K for times under 100 hours. Additionally, as the Al

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becomes depleted from the NiAl matrix, the thermal conductivity, an important property of aircraft engines, is also reduced. A further problem associated with Al depletion from the matrix of these alloys includes an
5 increase in density of the material.

In general, a major obstacle for structural applications of NiAl-AlN base materials is its low temperature toughness. The B2-crystal structure of NiAl with limited slip systems is believed to be the cause of
10 lack of low temperature toughness. Several attempts at raising the toughness by alloying additions to modify the crystal structure or opening up additional slip systems in NiAl have shown very limited success.

SUMMARY OF THE INVENTION

15 To solve the above and other problems, the present invention is directed to a particulate reinforced NiAl-AlN composite alloy comprised of a NiAl matrix having greater than about 13 volume percent, preferably at least 15 volume percent and most preferably 20 to about 50 volume percent
20 particles of AlN within said matrix. In a preferred embodiment, the composite alloy has from about 30 to about 40 volume percent particles of AlN within said matrix. The particles preferably have a diameter from about 15 nanometers to about 50 nanometers.

25 The NiAl-AlN composite according to the present invention preferably has from about 37 atom % to about 56 atom % and preferably about 55 atom % Al in the NiAl matrix to increase oxidation resistance of the composite alloy. The composite can further comprise a toughening alloy
30 selected from the group consisting of NiCrAlY, FeCrAlY and FeAl.

Particulate reinforced NiAl-AlN composite alloys in accordance with this invention may be prepared by cryomilling prealloyed NiAl in liquid nitrogen, for example,
35 using grinding media having a diameter of from about 2 mm to about 6 mm at an impeller speed of from about 450 RPM to

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about 800 RPM for a duration of from about 4 hours to about 20 hours to obtain a cryomilled powder. The cryomilled powder is consolidated to form particulate reinforced NiAl-AlN composite alloy.

5

BRIEF DESCRIPTION OF THE DRAWINGS

Figure 1 shows the influence of AlN content on the compressive true stress strain rate behavior of an NiAl-AlN alloy at 1300K .

Figure 2 shows a comparison of 1300°K specific group strength of NiAl-AlN between high temperature materials NiAl-30 AlN, NASAIR 100 (Nickel-base single crystal superalloy (Garret Engine Co.)), B-1900 (Nickel base superalloy (International Nickel Co.)), Mar-M 501 (Cobalt base superalloy (Marten-Mariette)), and NiAl.

15

Figures 3 show influence of cryomilling parameters on NiAl alloys. Figure 3A shows the influence of cryomilling time on nitrogen content of an NiAl alloy. Figure 3B-3E are micrographs of cryomilled NiAl powder particles illustrating the influence of milling time on particle morphology. Figure 3F shows the influence of size of grinding media on nitrogen pickup during cryomilling of NiAl.

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Figures 4 further show the influence of grinding parameters on NiAl alloy. Figure 4A shows the influence of milling speed on nitrogen pickup during cryomilling of NiAl and Figure 4B shows the relationship between size of particles of NiAl and percentage of nitrogen pickup distribution in cryomilled NiAl.

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Figure 5 shows a comparison of the cyclic oxidation of various NiAl-AlN alloys.

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Figures 6 are optical micrographs of fully consolidated NiAl-AlN. Figure 6A is a micrograph of NiAl-4AlN and Figure 6B is a micrograph of NiAl-30AlN.

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Figures 7 show the influence of AlN content on properties of the cryomilled NiAl-AlN alloy. Figure 7A shows the influence of AlN content on compressive yield strength

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at a strain rate of 0.0001/sec. Figure 7B shows the influence of AlN content on compressive strain to fracture. Figure 7C shows the influence of AlN content on flow stress-strain rate behavior between 1300° and 1500°K. Figure 7D shows the influence of AlN content on room temperature fracture toughness of cryomilled NiAl.

Figures 8 are SEM fractographs of chevron notched bend NiAl-AlN specimen. Figure 8A is a fractograph of NiAl-8AlN. Figure 8B is a fractograph of NiAl-30AlN.

Figure 9 shows compression strength versus temperature for NiAl-40AlN as compared to other NiAl-AlN alloys, indicating its superior strength.

Figure 10 shows the influence of ductile phase dispersion on room temperature toughness of NiAl-AlN.

DESCRIPTION OF PREFERRED EMBODIMENTS

The present inventors in "Microstructure/property relations in AlN and Al₂O₃ particulate strengthened NiAl," Mat. Res. Soc. Symp. Proc., Vol. 288, 1993, Hebsur et al. (hereby incorporated herein by reference) conclude that the attractive strength levels of the AlN containing NiAl are the direct result of the presence of the AlN particles within the NiAl matrix. Volume fractions of up to 30% AlN in the NiAl matrix of alloys are obtained by the present inventors, as they report in "Characteristics of an elevated temperature AlN particulate reinforced NiAl," Structural Intermetallics, Oct. 1993 (hereby incorporated herein by reference).

According to the present invention, particulate reinforced NiAl-AlN composite alloy having greater than 13 and up to about 50% AlN can be prepared by cryomilling prealloyed NiAl in liquid nitrogen using grinding media having a diameter about 2 mm to about 6mm and preferably about 3mm at an impeller speed of from about 450 RPM to preferably about 800 RPM for a duration of from about 4 to about 20 and preferably about 16 hours to obtain a cryomilled powder. The impeller shaft geometry can play an

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important role in the cryomilling operation. By keeping all cryomilling parameters the same except changing the impeller shaft arm configuration from straight to an inverse L shape, the nitrogen content and hence the volume fraction AlN in the cryomilled NiAl can be substantially increased. Thus, in a preferred embodiment the arm configuration of the impeller shaft is in an inverted "L" shape during cryomilling. Additionally, by using larger sized attritors in the cryomilling process, having 6 litres capacity stainless steel tanks instead of 1.4 litre tanks, high Al %-NiAl-AlN composites may be obtained by cryomilling NiAl powder for fewer hours, e.g., for 8 hours as opposed to 16 hours.

The cryomilled powder is consolidated to form the particulate reinforced NiAl-AlN composite alloy. The consolidation of cryomilled powder by hot pressing followed by hot isostatic pressing results in a fully dense NiAl based composite alloy comprising a NiAl matrix having greater than about 13 volume percent to about 50 volume percent and preferably from about 30 to about 40 volume percent fine particles of AlN within the matrix. The particles of AlN preferably have a diameter from about 15 nanometers to about 50 nanometers.

The NiAl-AlN composite according to the present invention has a room temperature compressive strength of from about 1800 to about 2400 MPa and preferably about 2200 MPa and a compressive creep strength of from about 200 MPa to about 300 MPa and preferably about 260 MPa under 1300K and strain rate of 10^{-6} S conditions. The composite has a room temperature fracture toughness of from about 45 MPa \sqrt{m} to about 6 MPa \sqrt{m} and preferably about 5 MPa \sqrt{m} under 0.01 1mm/min. conditions.

Since most applications for these alloys require very light materials, a higher Al content is desirable. Moreover, to increase the specific strength (strength/density) of the materials, the lowest density materials are

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needed. Thus, with the advent of higher volume % AlN, NiAl-AlN composite alloys, methods and materials for improving the cyclic oxidation resistance, specific strength and thermal conductivity of the alloys are needed.

5 In an embodiment of the present invention, the NiAl matrix has from about 37% to about 56 atom % and preferably about 55 atom % aluminum. In embodiments where there is a stoichiometric excess of Al, e.g., atom % aluminum is greater than 50%, the NiAl-AlN composite has an specific
10 weight gain at 1000 K cycles and 1473 K of from about 5 mg/cm² to about 2 mg/cm² and preferably about 3 mg/cm². The NiAl-AlN composite having a stoichiometric excess of aluminum has a thermal conductivity of from about 43 W/m·k to about 30 W/m·k and preferably about 35 W/m·k.

15 In a preferred embodiment, the particulate reinforced NiAl-AlN composite alloy (e.g., NiAl-30AlN) is prepared wherein the prealloyed NiAl contains from about 2 to about 6 and preferably about 4 atom % stoichiometric excess of Al.

20 The NiAl-30 AlN composite in accordance with the present invention exhibits highest compression yield strength with all temperatures between 300 and 1300K as comparable with other compositions of NiAl-AlN composite. The NiAl-30 AlN specimen tested under compressive creep
25 loading between 1300 and 1500K also exhibits highest creep resistance as shown in Figure 1 with very little surface oxidation, indicating also its superior elevated temperature oxidation resistance. Figure 1 also shows that in a high stress exponent regime the strength of the particulate is
30 proportional to the square root of the AlN content and that in the low stress exponent regime, the influence of AlN content on strength appears to be less dramatic. The specific creep strength of this material at 1300K is superior to the first generation Ni based single crystal
35 superalloy as shown in Figure 2. The improvements in elevated temperature creep strength and oxidation resistance

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are achieved without sacrificing room temperature fracture toughness of the NiAl base material.

In accordance with the present invention, low temperature toughness of NiAl-AlN can be improved by ductile phase reinforcement. The dominant contribution to toughening comes from traction produced by the ductile ligaments bridging the crack surfaces, thus shielding the crack tip. Toughening by this mechanism can be enhanced by increasing the size, volume fraction, the work of rupture, strength and strain hardening exponent of the ductile phase.

According to the present invention, low temperature toughness of particulate reinforced NiAl-AlN composites can be especially improved by cryomilling in an attritor with an impeller having an arm, prealloyed NiAl and an alloy selected from the group consisting of prealloyed NiCrAlY, FeCrAlY and FeAl in liquid nitrogen. The cryomilling conditions should be such as to obtain a cryomilled powder which forms upon consolidation a particulate reinforced NiAl-AlN composite alloy comprising a NiAl matrix having greater than about 13 volume percent particles of AlN within said matrix. In the method for improving the low temperature toughness the volume percent of the NiAl is from about 55 % to about 85% and the volume percent of the alloy is from about 45% to about 15%. When the alloy cryomilled with NiAl is NiCrAlY, the low temperature toughness of the NiAl-AlN composite is from about 18 MPa to about 22 MPa when the alloy is FeCrAlY, the low temperature toughness of the NiAl-AlN composite is from about 12 MPa to about 16 MPa and when it is FeAl, the low temperature toughness of the NiAl-AlN composite is from about 8 MPa to about 10 MPa \sqrt{m} .

NiCrAlY (Ni-wt.% 16 Cr-6Al-0.3Y) and FeCrAlY (Fe-wt%24Cr-4Al-0.3Y) with -200 mesh size gas atomized powders (Alloy Metals, Inc.) have excellent toughness, ductility and high temperature oxidation resistance. Even Fe-40Al is more ductile than NiAl and forms AlN readily during cryomilling.

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The FeAl powder is produced by induction melting followed by melt spinning.

The following non-limiting examples are presented to more fully describe the invention. It should be understood that these examples in no way limit the scope of this invention, but rather, are presented for illustrative purposes.

EXAMPLES

Methods and Materials

10 About 1000 grams of high purity (99.99%) elemental nickel pellets (660 grams) and 340 grams of aluminum pellets (AESAR Johnson Matthey Inc.) are induction melted in Al_2O_3 - SiO_2 crucibles (Taycor). The furnace is evacuated twice to 10^{-3} Pa and back filled with high purity argon prior to
15 melting. The molten metal is poured into 25.5 mm diameter x 250 mm long copper molds. The castings produced by this technique have some shrinkage porosity and larger grain size with small amount of segregation.

Rapid solidification of as cast Ni-53Al is carried
20 out to obtain a fine grained and homogenous material. Rapidly solidified NiAl is achieved by melt spinning in a free jet chill block melt spinning facility (CBMS). About 200 grams of the prealloyed Ni-53Al casting is used for each melt spinning run. In order to remove surface oxide scale,
25 the charge is polished on 600 grit paper and then ultrasonically cleaned in acetone before melting in 37 mm O.D. high purity/high density alumina crucibles (Coors) with a 1 mm diameter nozzle in the bottom by means of induction heating. The crucible and wheel assembly are enclosed in a
30 chamber evacuated to 10^{-5} Pa and then back filled with high purity argon. After an initial period of slow heating to 1275K, the heating rate is increased drastically to reach a desired temperature quickly in order to minimize the contact time of hot metal with a crucible wall. The temperature of
35 the melt is monitored using an infrared optical pyrometer

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which is viewed through a port located in the chamber. When the desired temperature, which is usually 350K above the liquidus temperature, is reached, pressurized argon gas is introduced at the top of the crucible, causing the liquid to impinge onto the rotating wheel. A copper wheel of 300 mm diameter and a crucible-wheel gap of 19 mm are used. All melt spin runs are made with a wheel speed of 20 meter per second and an ejection pressure of 26.6 MPa to obtain a 3 mm wide rapidly solidified ribbon. The ribbon segments vary from a few millimeters to over 100 mm.

The melt spin ribbons are brittle and exhibit fine grain homogenous microstructure. The melt spun ribbon is crushed and ground to -100 mesh size powder using 6 mm diameter stainless steel balls in a planetary mill for about 1 hour under argon gas. About 500 grams of this powder is used for cryomilling.

Cryomilling is carried out in a modified Union Process 01-HD attritor equipped with a 1.4 litre stainless steel chamber. Approximately 4000 grams of 3 mm diameter 440 stainless steel grinding media (Abbot Ball Company) and 500 grams of prealloyed NiAl powder are placed in the tank, which is then filled with liquid nitrogen. Once the temperature within the tank is stabilized at 77K, the mill is started and the straight arm stainless steel impeller is quickly brought to the operating speed of 800 rpm. This milling action is continued for 16 hours. During the entire milling period the liquid level in the tank is constantly maintained by a thermocouple sensor. Except for loading of ball/powder mixture, the entire operation is conducted in a well controlled argon atmosphere with small positive pressure to minimize the condensation effects while a continuous liquid nitrogen flow is maintained in the cooling jacket tank to minimize the temperature rise during milling.

Following the 16 hour milling run, the tank is transferred to a dry box and the liquid nitrogen is allowed to boil off. After warming to room temperature, the

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cryomilled powder is separated from the grinding balls by sieving. The composition of prealloyed NiAl is established as Ni-53 atom%Al. Thus, for subsequent cryomilling experiments, gas atomized Ni-53 atom% Al powder (Homogenous Metals, Inc.) is used.

Milled powders obtained from all Examples are consolidated to form alloys by hot pressing at 1373K/35 MPa for 3 hours under argon atmosphere to produce 60 mm long x 40 mm wide x 12.5 mm compacts with about 90% density. The hot pressed compacts are sealed in titanium killed steel cans and hot isostatically pressed at 1373K/20 MPa for 3 hours to obtain fully dense (nearly 100% density) materials.

Example 1

About 500 gm of prealloyed Ni-50 atom % (instead of Ni-53 atom %) Al melt spun alloy is made in accordance with the procedure above. The melt spun alloy is cryomilled in a modified attritor mill (Union Process model 01-HD) using 4,000 grams of grinding media consisting of 6 mm diameter, 440 stainless steel balls at an impeller speed of approximately 450 rpm to produce milled powder. During milling, liquid nitrogen is introduced into the grinding tank and the level of the nitrogen is automatically controlled by a thermocouple sensor. The outer jacket of the tank is cooled by a continuous flow of liquid nitrogen. Milling duration is varied between 2 and 24 hours and the impeller speed is varied between 450 to 800 rpm.

Example 2

The same procedure as in Example 1 is repeated except that 0.5 wt.% of Y_2O_3 is added to the attritor during the cryomilling, which is conducted for 16 hours at an impeller speed of 800 rpm.

Example 3

The same procedure as in Example 1 is repeated except that the grinding media is 3 mm stainless steel balls and the duration of the cryomilling is 16 hours at an impeller speed of 800 rpm.

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Example 4

Cryomilling of prealloyed meltspun ribbon of Ni-55 atom percent Al is carried out using 3 mm diameter, 440 stainless steel grinding media and 800 rpm empowered speed. 5 The grinding media to powder ratio is approximately 8 to 1. After milling for 16 hours, the powder is transferred to a glove box for warming to room temperature under dry argon. Chemical analysis of the melt spun ribbon and cryomilled powder are given in Table 1. The consolidation of cryo- 10 milled powder is carried out in two steps of hot pressing at 1454K/31 MPa/2 hours followed by hot isostatic pressing at 1454K/276 MPa/2 hours in a steel can.

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TABLE 1
CHEMICAL COMPOSITION OF MELTSPUN AND CRYOMILLED Ni55Al

| MATERIAL | ELEMENTS WT.% | | | | MATRIX COMP. Al AT% | 2nd Phase vol.% | |
|----------------------|---------------|------|-----|------|---------------------------|--------------------|--------------------------------|
| | Ni | Al | N | O | | AlN | Al ₂ O ₃ |
| Meltspun Ni55Al | 63.2 | 36.7 | - | .007 | 54.8 | - | - |
| Cryomilled Powder | 60.0 | 32.7 | 6.6 | .518 | 49.4 | 32.1 | 1.4 |
| HP+HIP NiAl-30AlN | 60.6 | 31.6 | 6.6 | .52 | 48.2 | 32.1 | 1.4 |

Example 5

Several cryomilling runs are carried out, each using 500 grams of prealloyed powder mixture containing 1) 80 vol.% homogenous metal powder of Ni53Al and 20 vol.% FeCrAlY (387 gram NiAl + 113 grams FeCrAlY); 2) 60 vol.% homogenous metals batch of Ni-53Al and about 40 vol.% of FeCrAlY (20 gms of NiAl + 220 gms FeCrAlY); 3) 80 vol.% Ni-53Al and 20% of NiCrAlY (375 gms Ni53Al + 125 gms NiCrAlY); 4) 60 vol.% Ni53Al and 40 vol.% NiCrAlY (264 gms Ni53Al + 236 gms NiCrAlY); 5) about 80 vol.% Ni53Al and 20 vol.% FeAl (400 gms Ni53Al + 100 gms FeAl); and 6) about 60 vol.% NiAl and 40 vol.% FeAl (300 gm Ni53Al + 200 gms FeAl). Prior to cryomilling, each batch of powder mixture is mixed in a stainless steel jar and blended in a jar mill for about 8 hours. The blended powder is cryomilled in an 01-HD attritor using an L-shaped arm following the procedure described above. The cryomilled powders are hot pressed at 1373K/35 MPa/3 hours under argon atmosphere to produce 60 mm long x 40 mm wide by 12.5 mm compacts with about 90% density. These hot pressed compacts are sealed in titanium killed steel cans and hot isostatically pressed (HIP) at 1373K/280 MPa for 3 hours to obtain fully dense (nearly 100% dense) materials. From these dense materials, chevron notched bend specimens for fracture toughness and 4 mm diameter x 8 mm long cylindrical specimens for compressive

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strength measurements are obtained by electro-discharge-machining.

Figure 3A shows the influence of milling time on nitrogen content of NiAl alloy during cryomilling. As shown in Figure 3A, there is little nitrogen incorporated into the alloy during the first four hours of milling, followed by a steep increase in nitrogen content of the alloy up to 16 hours of milling. Continued milling beyond 16 hours does not appear to increase the nitrogen content any further. Additionally, the size of the milled powder particles decreases with increased milling time as shown in Figures 3B-E (3B, 0 hr; 3C, 4 hr; 3D, 8 hr; 3E, 16 hr) and particle size distribution indicates a higher volume fraction of finer particles with increased milling time to 16 hours.

It appears that nitrogen incorporated into the alloy increases with increase in volume fraction of finer particles with larger surface area. Increasing the impeller rotation speed from 450 rpm to 800 rpm increases the nitrogen content as shown in Figure 4A. Additionally, a comparison of the powders obtained in Example 1 with those obtained in Example 3 shows that the nitrogen content increases dramatically when smaller diameter (3 mm instead of 6 mm) grinding media is used in the cryomilling. The results of such a comparison is shown in Figure 3F. The particle source distribution plot as shown in Figure 4B indicates more efficient grinding with small diameter balls (grinding media) and hence higher nitrogen incorporation into the alloy. X-ray diffraction studies of cryomilled powder obtained in accordance with the Examples do not detect a ceramic phase.

Cyclic oxidation tests are carried out at 1573K on the composite obtained in Example 5. Each cycle consists of 1 hour in the furnace followed by 0.33 hour cooling in ambient. Figure 5 shows specific weight change data for various compositions of NiAl based material as a function of the number of cycles. O, standard; •, 30 AlN; ▲, Y₂O₃; △, no

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Y₂O₃ and □, 10 AlN (High Al). These data demonstrate that the oxidation resistance of NiAl-30 AlN is comparable to NiAl-10 AlN with Y₂O₃ and superior to NiAl-10 AlN without Y₂O₃ addition. High aluminum content in NiAl improves long time
5 cyclic oxidation behavior by adding about 2 to 3 atom percent aluminum during alloying. The cryomilled Ni-53 atom percent alloy exhibits improved oxidation resistance as compared to cryomilled Ni-50 atom percent alloy as seen in Figure 5.

10 From the consolidated material, cylindrical compression specimens of 4 mm diameter x 8 mm long, ASTM standard chevron notched four point bend bars and oxidation coupons of 1.6 mm diameter x 2 mm thick are machined by electrodischarge machining followed by grinding.

15 Upon consolidation, x-ray scans indicate significant amounts of this phase on the relative intensity ratio followed the trend of the lower volume fracture material. Optical micrographs of consolidated NiAl-30 AlN composite shown in Figure 6B show a fine grain microstructure with a
20 dark mantle of dispersed AlN outlining particle free regions. For comparison the microstructure of NiAl-4 AlN is included in Figure 6A. The large grain structure in both materials represents unprocessed NiAl caught in dead zones of the attritor.

25 Compression tests are conducted on a universal test machine between 300K and 1300K in air at a constant cross head speed of 0.5 mm/s. Figure 7A illustrates an increase in compressive yield strength with increasing volume fraction of AlN and NiAl at temperatures between 300 and
30 1300K. The compressive strength to fracture between 300 and 700K (below DBTT) did not show any clear trend as a function of AlN content. See Figure 7B. Compression test are also conducted in air between 1200 and 1500K under constant cross head speeds ranging from 2.12×10^{-4} to 8.47×10^{-7} mm/s to
35 approximately 8 percent strain in order to determine

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elevated temperature creep properties. Some constant low compressive creep tests are also conducted to determine the long term strength of the NiAl-30 AlN. The 1200 to 1500K stress strain rate results for NiAl-AlN are shown in Figure 7C. The creep strength of the composite is sensitive to temperature and the flow stress at a given strain rate decreases nearly by a factor of 2 for every 100K increase in temperature. Although the metallography of long time creep tested specimens has not been carried out, contrary to the earlier observation of NiAl-10 AlN specimens which underwent extensive oxidation at 1500K, the NiAl-30AlN specimen showed very little oxidation.

The room temperature fracture toughness (K_{Ic}) measurements are made on chevron notched 4 point bend specimens tested in a universal test machine at a cross head speed of 0.01 mm per min. This cross head speed minimizes the overload effects. Fracture toughness is calculated from the load-displacement data using the slice model. Figure 7D shows the variation of fracture toughness with variation in AlN content. (\square , NiAl; Δ , 3.0; \circ , 5.0; \cdot , 8.0; \blacksquare , 12.0 and \blacktriangle , 30.0). Fracture toughness values for binary NiAl are taken from several sources. Figure 8A illustrates that the K_{Ic} values of NiAl decrease insignificantly with increased AlN content. Scanning electron microscopic examination of fractured specimens indicate that NiAl-AlN exhibits predominantly transgranular cleavage fracture (see Figure 8B). Figure 9 shows the superior compressive yield strength of NiAl-40AlN compared to other NiAl-AlN alloys. Figure 10 shows room temperature fracture toughness of NiAl based material. The FeCrAlY and NiCrAlY ductile reinforced materials show improved toughness over NiAl-AN. The NiCrAlY reinforcement shows the highest improvement (nearly 300% in toughness).

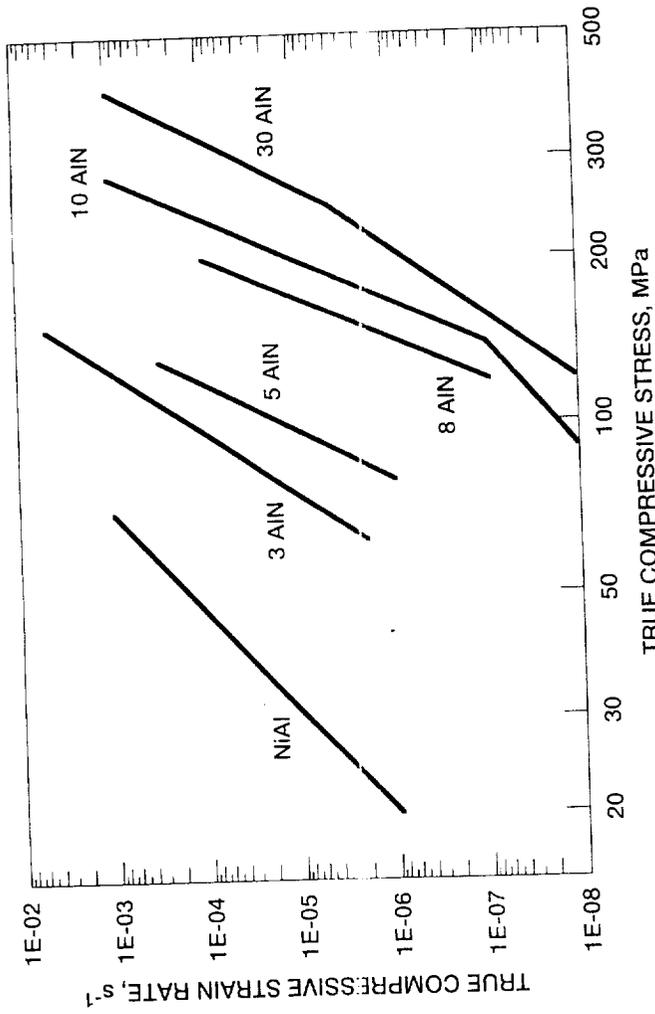
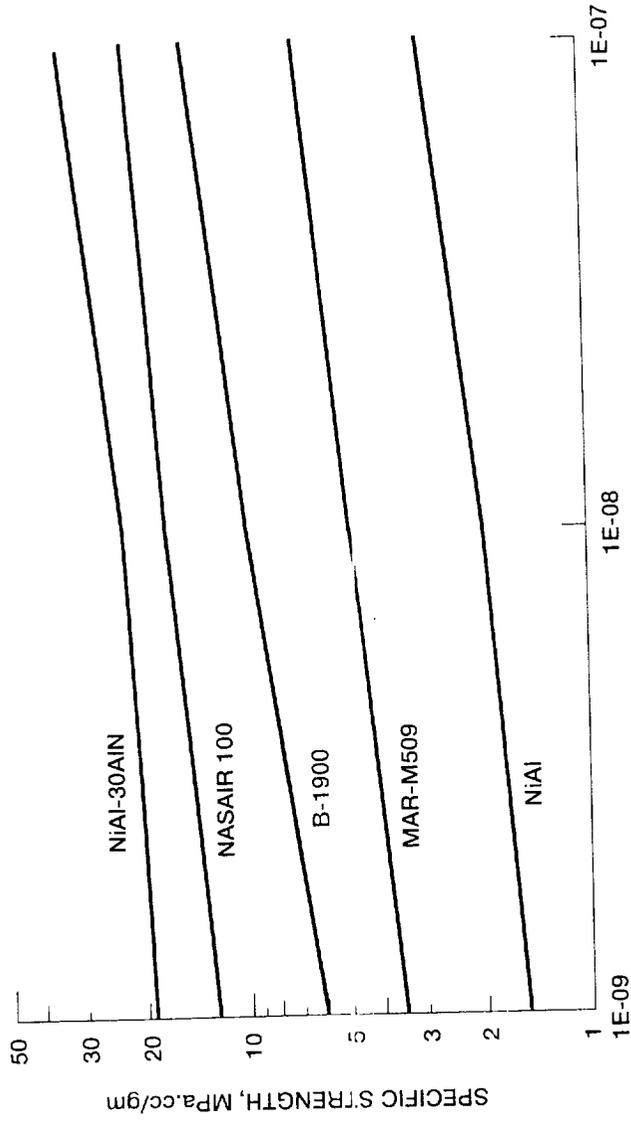


FIG. 1



TRUE COMPRESSIVE STRAIN RATE, s⁻¹

FIG. 2

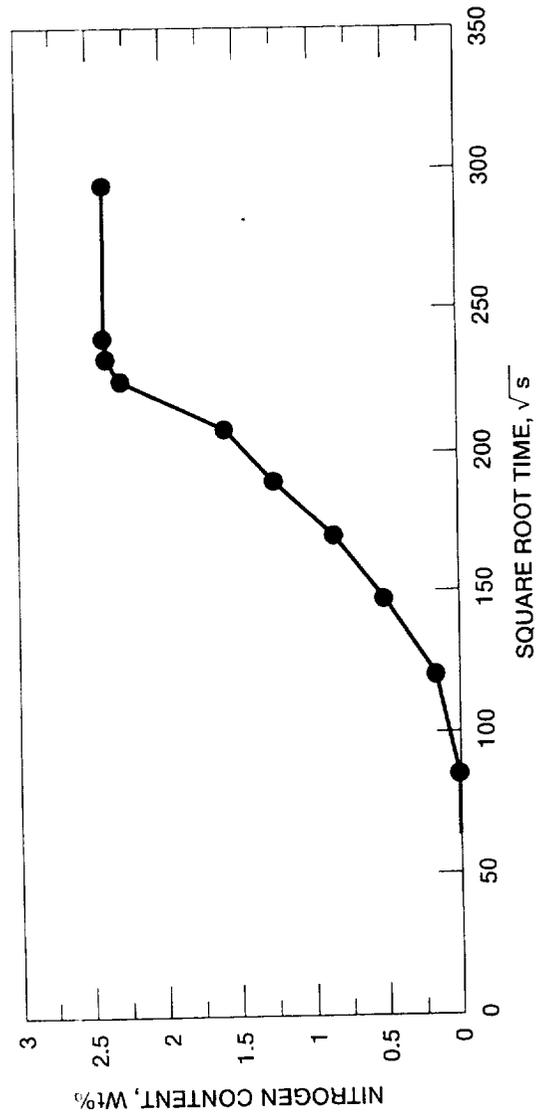


FIG. 3A

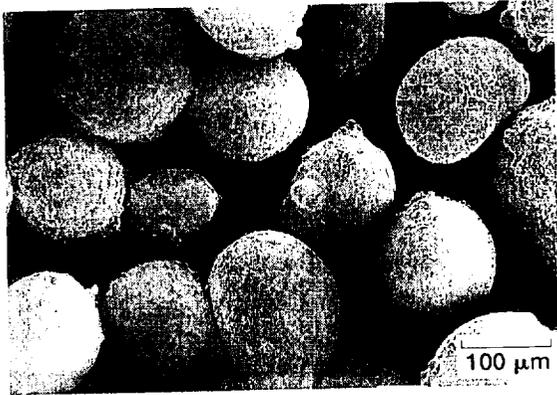
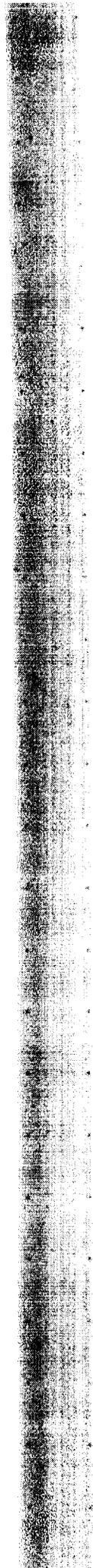


FIG. 3B



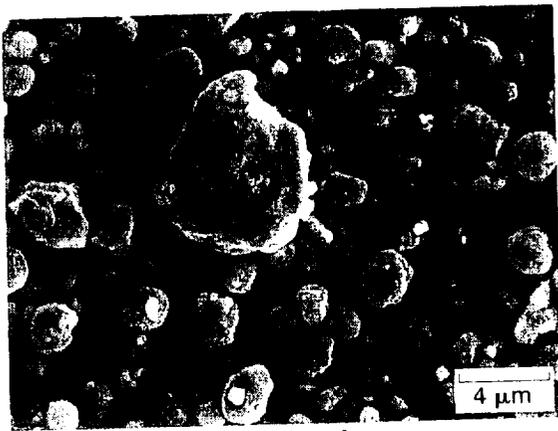


FIG. 3C

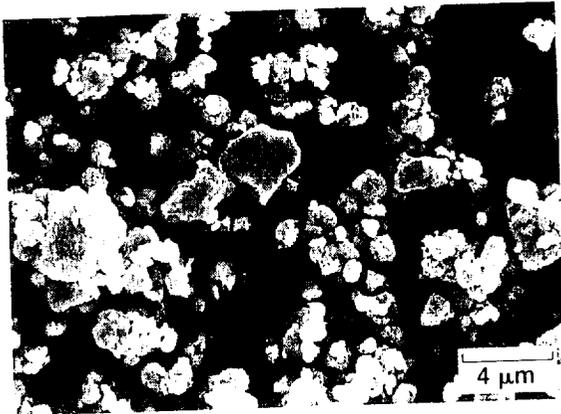


FIG. 3D

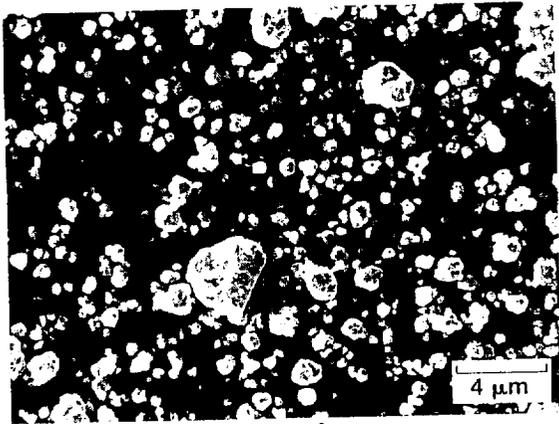


FIG. 3E

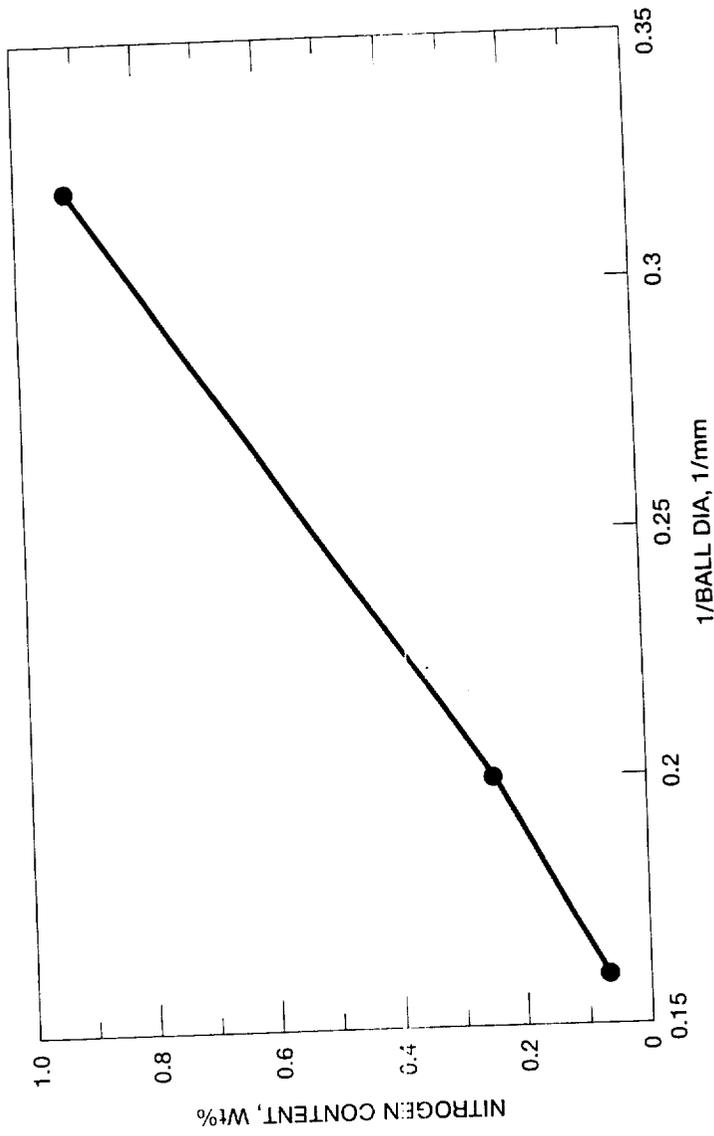


FIG. 3F

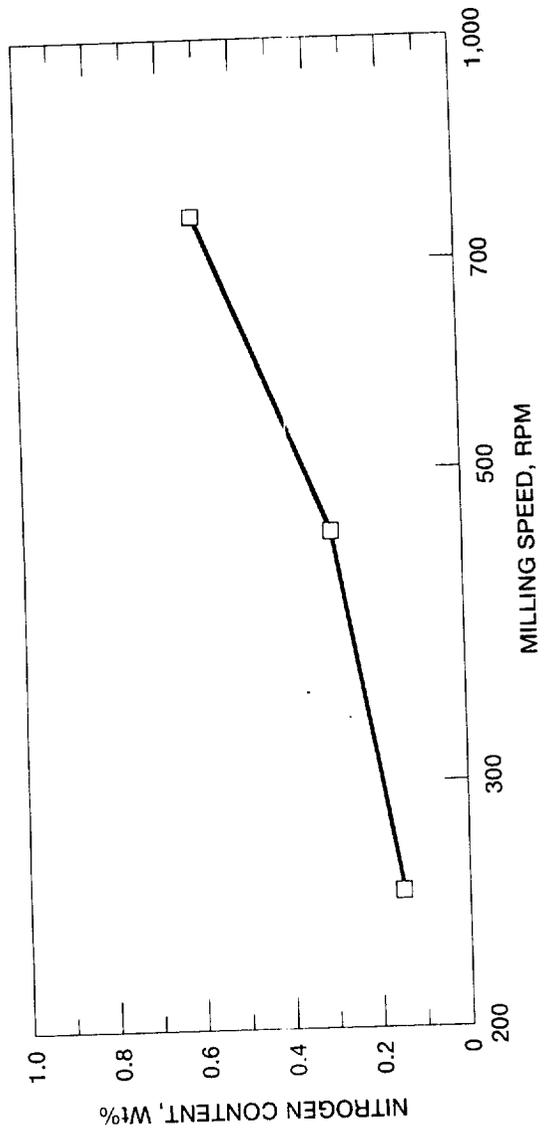


FIG. 4A

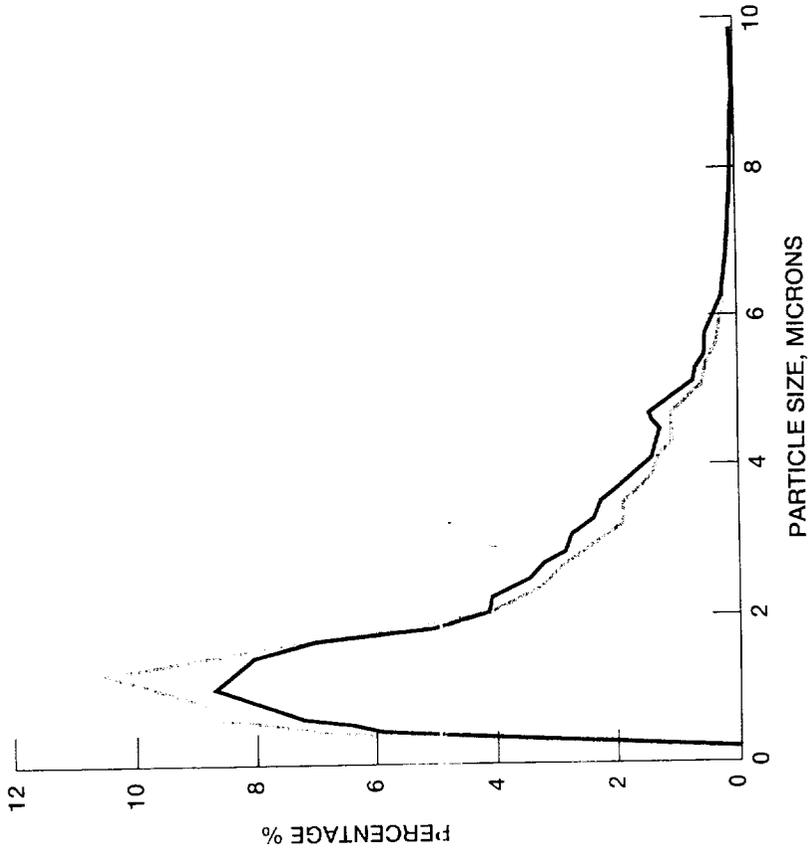


FIG. 4B

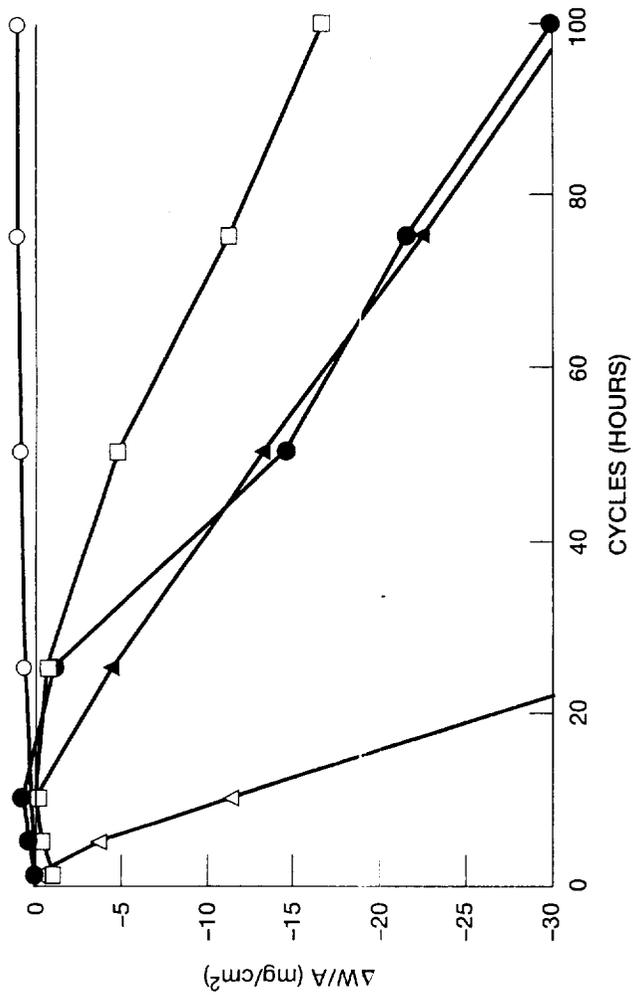


FIG. 5

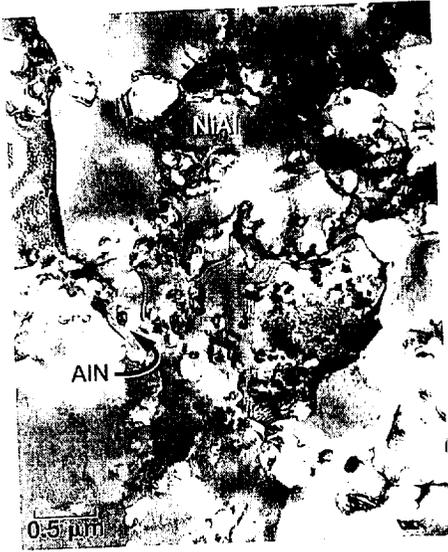


FIG. 6A

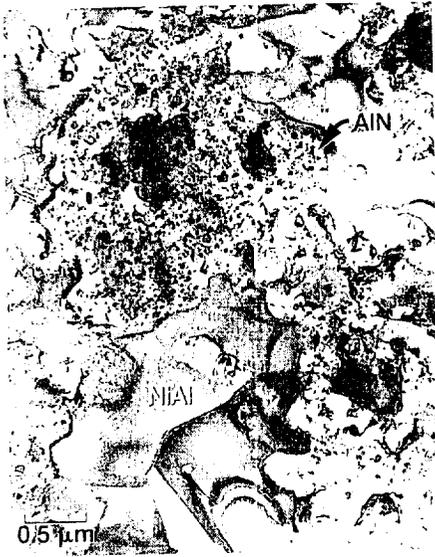


FIG. 6B



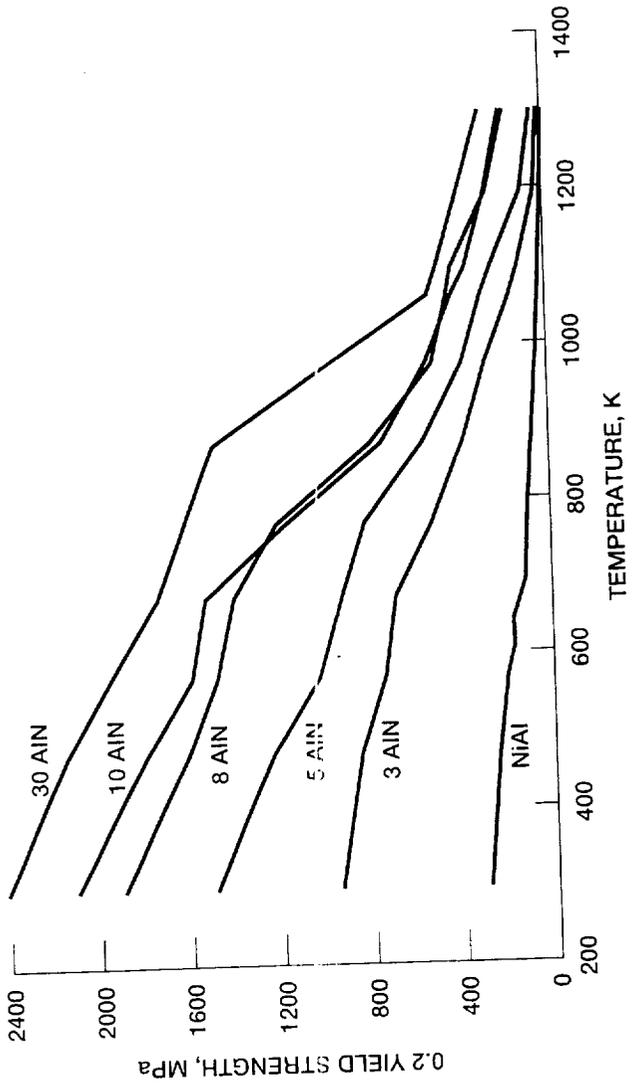


FIG. 7A

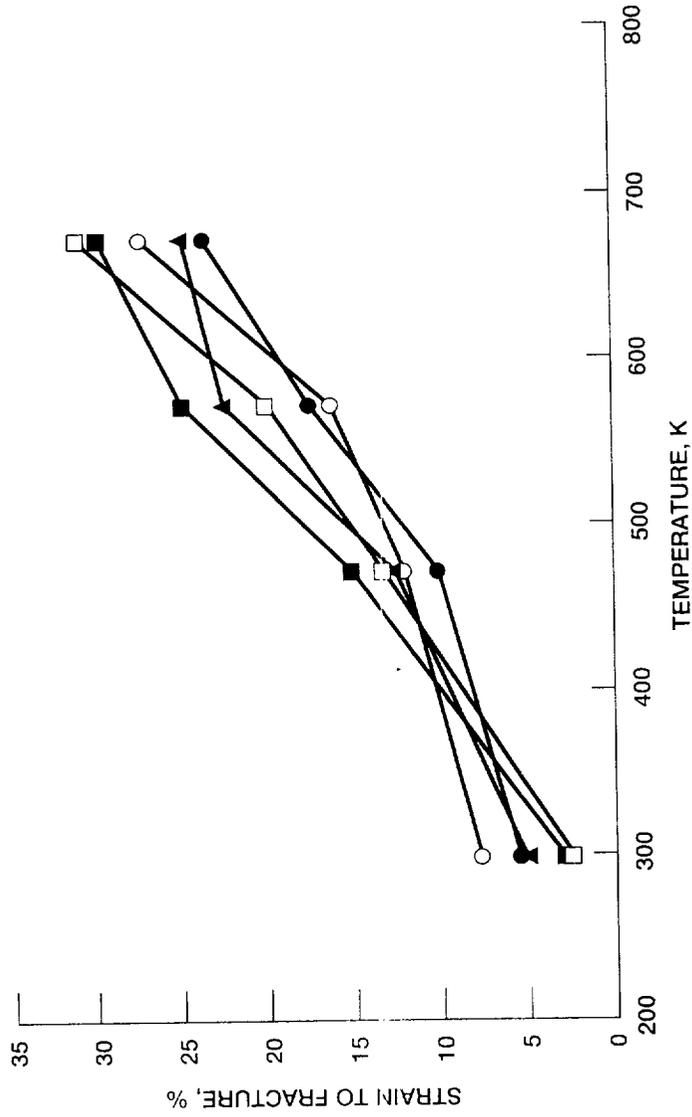


FIG. 7B

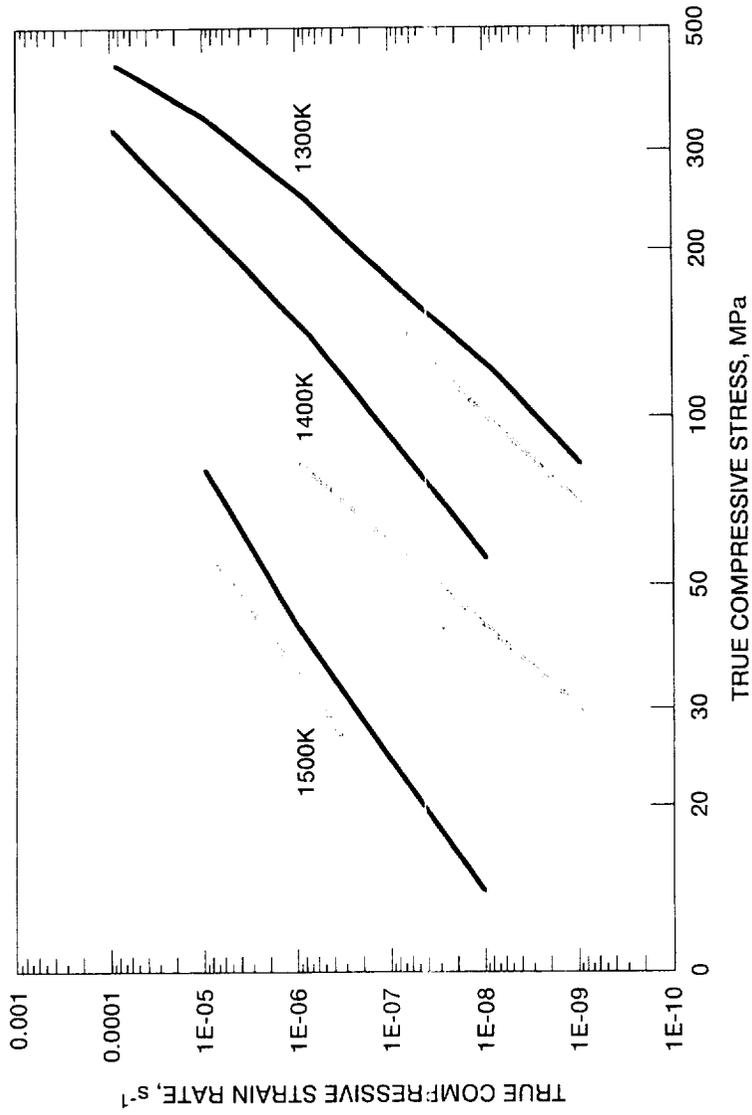


FIG. 7C

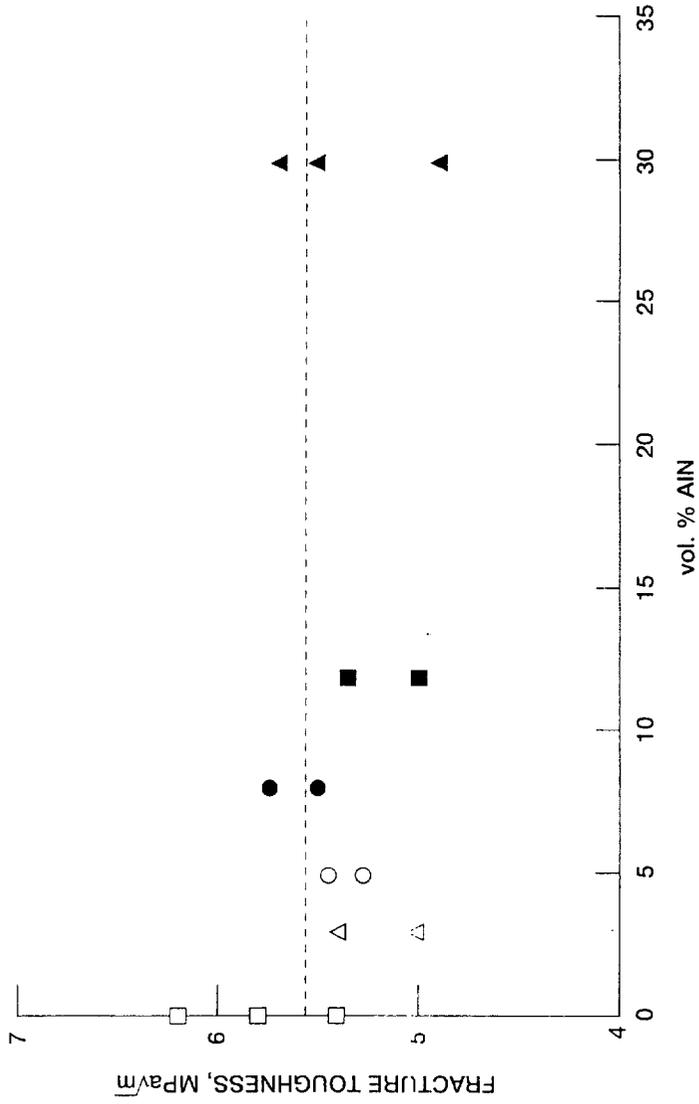


FIG. 7D

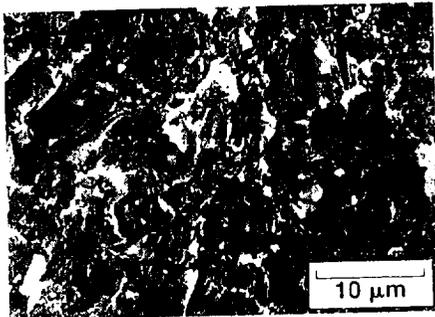


FIG. 8A

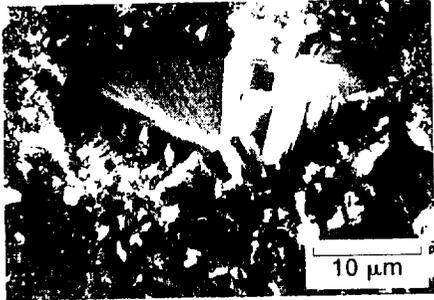


FIG. 8B

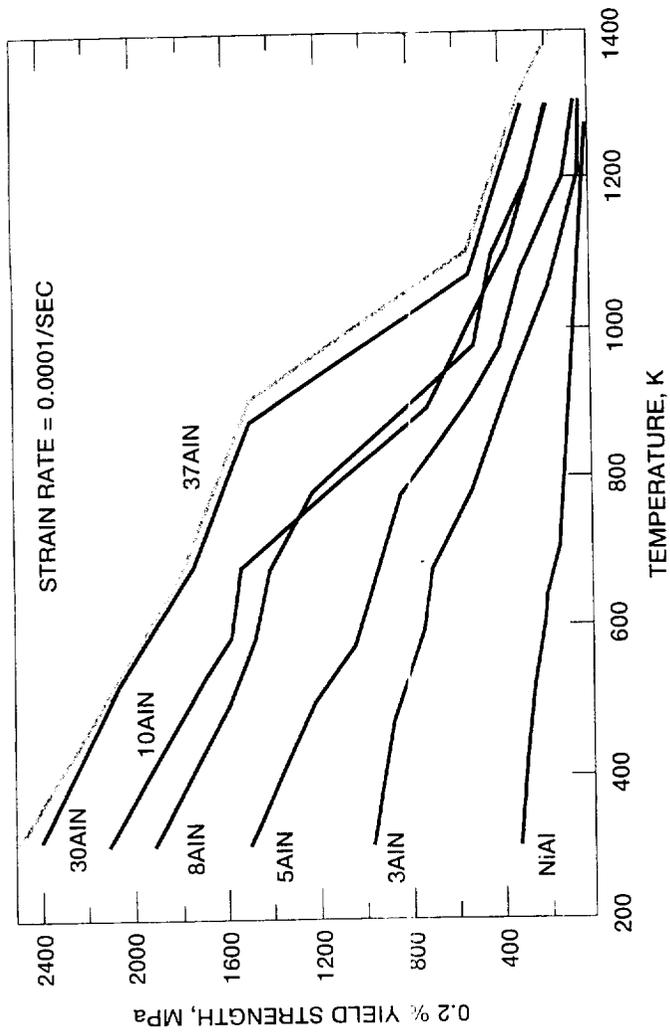


FIG. 9

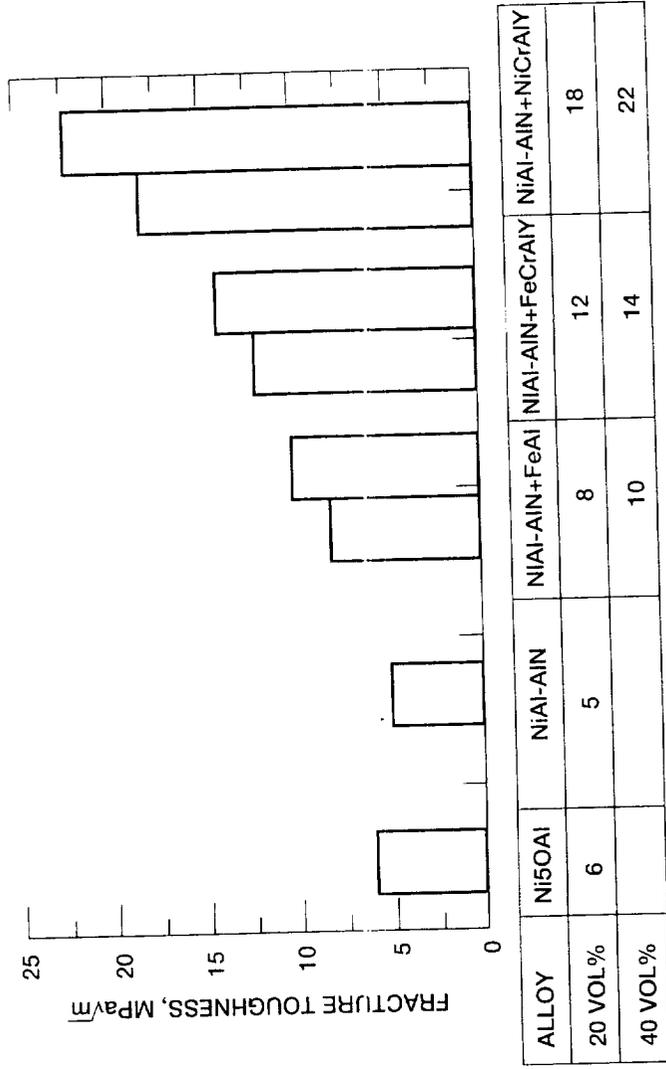


FIG. 10