COMPUTER TOOLS FOR SYSTEMS ENGINEERING AT LARC

J. Milam Walters, Systems Engineering Office
Aerospace Mechanical Systems Division

The Systems Engineering Office (SEO) has been established to provide lifecycle systems engineering support to LaRC projects. Over the last two years, the computing market has been reviewed for tools which could enhance the effectiveness and efficiency of activities directed towards this mission. A group of interrelated applications have been procured, or are under development including a requirements management tool, a system design and simulation tool, and a project and engineering database. This paper will review the current configuration of these tools and provide information on future milestones and directions.
The Role of Computers In LaRC R&D

Computer Tools for Systems Engineering

Presented by

J. Milam Walters

June 16, 1994

AMSD Systems Engineering Office

AMSD Systems Engineering Office

• Established via Center Reorganization after approximately 3 years of ground laying
• Current staffing level - 5 CS
• Chartered to guide application of systems engineering to LaRC flight projects
• Process detailed in LHB 7122.1, currently in approval cycle
• Process applied to various projects, most recently JADE and SABER

AMSD Systems Engineering Office
Systems Engineering Process

- **Researcher Idea**
- **Goal Analysis**
- **Requirements Documents**
- **Decision Analysis**
- **Cost**
- **Schedule**
- **System Alternatives**
- **Technical Performance**
- **Metrics**

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Systems Engineering Office Tools

- **Workstation Based tools consist of the following:**
  - Oracle SE Project & Engineering Database
  - Excalibur Scanning/Recognition Software
  - RTM (Requirements Traceability & Management)
  - RDD-100 (Requirements Driven Design)
  - Interleaf 6.0
  - Matlab with following toolboxes:
    - Simulink option
    - System Identification
    - Control System
    - Optimization

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Oracle SE Project & Engineering Database

- Oracle RDBMS Version 7 relational database on SUN Sparcstation 10, Model 41
- Oracle*Forms Version 4 provides graphical user interface for record & graphic viewing
- Oracle Data*Query performs complex searches
- Database consists of pre-set form types, with the capability to quickly generate any new table type
- Database will store project documentation and graphics as well as engineering data tables

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Excalibur Scanning/Recognition Software

- Interfaces with document scanner to read and interpret input
- Contains an adaptive search engine to retrieve desired document
- Displays original image upon match of a given search
- Provides the capability to scan and store input documents

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RTM Requirements Traceability & Management

- Application developed especially for tracking and managing project requirements
- Utilizes Oracle database to store requirement information
- Provides special tools for:
  - extracting requirements from source documents
  - expanding and focusing requirements
  - general requirements maintenance
- Includes output bridges to RDD-100 and Interleaf

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RDD-100 Requirements Driven Design

- Facilitates the construction, maintenance, display, and documentation of design objects that specify behavior
- Objects are created and edited by graphics or text, with multiple generated views available to gain different perspectives
- Includes a simulator which directly executes the design objects
- Templates and consistency checks verify system design sufficiency
- Bridge to Interleaf is included

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Matlab

- Interactive software program for scientific and engineering numeric computation
- Combines numerical analysis, matrix computation, signal processing, and graphics with a user interface through standard math notation.
- Functions include differential equation solution, polynomial operations, matrix computation, complex arithmetic and signal processing tools
- To view data graphically, MATLAB provides 2-D linear, log, semilog, and polar plots, and 3-D mesh and contour graphs
- Works with MATLAB numeric computation software package to build mathematical models of

Database Population

![Database Population Diagram]

SUN Sparcstation 10 (sc_sun1)

Oracle Database

Excalibur OCR Software

Document Scanner

Hard copy documents

Electronic Files Administrator

Database Population

AMSD Systems Engineering Office

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Requirements Management

SUN Sparcstation 10
(se_sun1 or se_sun2)

Electronic Files

Systems Engineers

Hard copy
Requirements
documents

Document Scanner

Oracle
Database

Environemnt
Management

Escalibur OCR
Software

Interleaf

AMSD Systems Engineering Office

System Modeling & Simulation

SUN Sparcstation 10
(se_sun1 or se_sun2)

Systems Engineers

Behavior Diagram

System Analyzer (SA)

Dynamic Verification Facility (DVF)

Interleaf

AMSD Systems Engineering Office
Tool Interface Overview

Summary

- The Systems Engineering Office of AMSD has been established to provide for computer aided:
  - systems level behavior modeling and simulation of new concepts (RDD-100)
  - subsystem mathematical modeling and simulation (Matlab)
  - requirements tracking and management (RTM)
  - storage of project and engineering documentation (SEDB)

- Interested parties should contact Richard Foss at 4-7049 or Milam Walters at 4-3014
A Distributed Computing Environment for Multidisciplinary Design

The Framework for Interdisciplinary Design Optimization (FIDO) project has the goal of developing a general distributed computing system for executing multidisciplinary computations on a networked heterogeneous cluster of workstations and vector and massively parallel computers. This project is a part of the Computational Aerosciences (CAS) project in the High Performance Computing and Communications (HPCC) program. The FIDO system provides a means for automating the total design process. It facilitates communication and control between components of the system, which include the diverse discipline computations involved in a design problem and the system services that facilitate the design. In its current state of development, the prototype FIDO system is being applied to a token example of the optimized design of a high-speed civil transport (HSCT), involving a simplified problem that includes the disciplines of aerodynamics, performance, propulsion, and structures, but with very few design variables. However, it has already demonstrated the ability to coordinate multidisciplinary computations and communications in a heterogeneous distributed computing system.

The concept being used in FIDO is course-grained parallelism, with instances of the disciplinary codes (aero, structures, etc.) running on separate processors, under control of an executive on another processor, and exchanging data through a single data base manager (on yet another processor). To allow the user to monitor the progress of the design iterations, there is a graphical user interface (which tracks the execution of codes performing the design iterations) and a separate process, called SPY, which allows its user to extract and plot data produced during current and previous design cycles. In fact, multiple instances of SPY can be executing at once, so that the designer can call on discipline experts and they (possibly from some remote location on the Internet) can examine the results being produced and provide advice. SPY is currently being upgraded to provide the capability for the designer to make changes in variable values during execution and so guide the design process.

The distributed computing system currently includes Sun, Silicon Graphics, and Digital Equipment Co. workstations. Provision has been made for adding connections to Cray vector computers and Intel parallel computers, and preliminary checks of connection procedures have been carried out. A communications library has been written (implemented using the PVM basic library developed at Oak Ridge National Lab) to provide the versatility for transferring data packages ranging from single variables or file names to large data arrays.

The current Motif-based Graphical User Interface (GUI) consists of three separate elements: setup, application status, and data display. The setup GUI provides the user with a convenient means of choosing the initial design geometry, material properties, and run conditions from a pre-defined set of files. The interface displays the choices using a series of pop-up Motif data windows, and allows the user to modify and store new condition files. The application status GUI allows the user to monitor the status of a design run. An example of this display is shown in the slide entitled "FIDO User Interface Concept". Within the screen on the left part of the slide, the upper left window displays current run parameters and contains pull-down menus for setting various options. The right window graphically displays the state of the overall design process by changing the color of each labelled box according to the work being done. A color key is shown in the lower left window. Additional detail of the system state can be obtained by selecting the
boxes with a 3-D appearance. Doing so brings up an associated window that displays sub-detail for that box. The data display GUI is the third interface element, called SPY, that provides the user with a variety of ways to plot data during the design process. The right part of slide is an example of a color-coded contour plot of wing surface pressures. The buttons at the top of the plot window provide the user with a variety of view controls. In addition to contour plots of aerodynamic pressures and structural stresses on the wing, SPY provides line-plots of cycle history for a variety of design parameters and data results. An example of this for a 20-cycle design iteration is shown in a later slide. This slide illustrates results of using FIDO to minimize the total weight of the HSCT for a 6000-mile range and Mach 2.4 cruise speed. Discipline sensitivity derivatives are calculated using finite differences, and a linearized model based on these is used with the optimization subroutine CONMIN to determine new values for the design variables at each design cycle. The slide illustrates the component weight reduction over a 20-cycle design iteration subject to design constraints on structural stress and deflection. Weights have been non-dimensionalized by nominal payload. The slide also shows the changes to the three aerodynamic design variables (lengths non-dimensionalized by wing span) and two structural design variables (non-dimensionalized by initial skin thickness).
The NASA Langley FIDO project

FIDO - Framework for Interdisciplinary Design Optimization

GOAL: Establish a general computational framework, consisting of software tools and programming guidelines, for facilitating the interdisciplinary coupling necessary for multidisciplinary design and optimization problems on heterogeneous computing systems.
Working together simultaneously on parts of the same problem

Computers and data distributed over network

Multiple programs targeted for appropriate computers

Group of users with diverse specialties

Develop a Programming Environment for Concept
Environment

Heterogeneous Distributed Computing

Gateway

Workstation

Parallel computer

Vector computer
Current LARC HPCC Emphasis
Current FIDO Components
Approach

Limit scope of design/optimization formulation.
Consider a representative set of disciplines, design criteria, design conditions, design variables, etc.

Use existing engineering analysis codes.

Use a High Speed Civil Transport (HSCT) configuration as the focus problem.

Initial implementation on a distributed workstation environment; migration to parallel testbed in the future.
Max Load: 2.5 g
Length: 300 ft.
Payload: 30,000 lbs
Range: 6,000 mi.
Altitude: 63,000 ft.
Mach: 2.4

Design Conditions

HSCT Baseline Description
FIDO Project Focus

Short term
- simplified model problem
- low-fidelity discipline codes
- finite-difference sensitivity derivatives
- coarse-grain parallelization on a workstation network

Long Term
- realistic design and optimization problem
- higher fidelity discipline codes
- quasi-analytic sensitivity derivatives
- coarse-grain parallelization on a heterogeneous computer network along with fine-grain parallelization on massively parallel and vector computers
FIDO User Interface Concept
FIDO Execution System

Diagram showing various components and interactions within the FIDO system, including communication library, network, interface, optimization, propulsion, structures, performance, aerodynamics, master, and graphics.
Features

- Provides synchronous control in heterogeneous computing environment
- Is adaptable to a wide variety of problems that use multiple computer programs
- Has modular form for easy problem migration
- Incorporates variety of debugging features
- Includes "SPY" capability for:
  - Monitoring computational progress
  - Displaying variables graphically
  - Modifying variables for design steering
FIDO Project Products

Prototype multi-discipline, multi-computer codes
- adaptable
- modular
- portable
- non-proprietary

Prototype toolkit:
- user interface
- communication library
- data manager library
- driver templates

Software recommendations:
- system software needs
- application programming guidelines
Programming Environment Resources

- Specified code interfaces
- Programming guidelines
  - be adaptable, modular, portable, non-proprietary
- Communications library
  - use message-passing standards
- Database manager
  - use file-based data schema
- Programming templates
- Structured directories
Utilize very simple geometry description.

WARFA/CDPR: Viscous drag (empirical)
AWAVE: Wave drag calculation
WINGDES: Drag due to lift & pressure difference across the wing

Based on linear theory

Low Fidelity Aerodynamics
Low Fidelity Structures

Equivalent Laminated Plate Solution (ELAPS)

ELAPS uses a series of plates defined on wing planform.

Camber and Thickness are defined for each plate.

**INPUT**
- Forces (acquired from Aero)
- Wing Geometry
- Skin Thickness

**OUTPUT**
- Structural Weight
- Stresses
- Displacements
Low Fidelity Propulsion

- Level 1 propulsion module from Lewis (TBE_NASA)

**INPUT**
- Engine Type
- Altitude and Mach No.
- Thrust

**OUTPUT**
- Fuel Consumption
Low Fidelity Mission Performance

- Breguet Range Equation

**INPUT**
- Fixed Range
- Lift and Drag
- Estimated Total Weight

**OUTPUT**
- Fuel Weight
Approaches to Sensitivity Analysis in CFD

- **Finite Differences**
  Compute the sensitivity by taking the difference between two analyses with slightly different values of the design variable.

- **Manual Insertion of Quasi-Analytical Sensitivity Derivatives**
  Derive the exact, linear equations satisfies by the sensitivity derivatives and add them to the analysis code.

- **Automatic Insertion of Quasi-Analytical Sensitivity Derivatives**
  Utilize a pre-procesor which automatically analyzes the Fortran code and adds (automatically) the code for the prescribed sensitivity derivatives.
Wing Planform Design Variables

- TO
- TI
- LSWP
- RC
- Front
- BREAK
Design variable history

LE sweep (radians)

Root chord

Spanwise break
Outboard thickness
Inboard thickness

Design evolution

Cycle 20

Cycle 10

Cycle 5

Cycle 3

Cycle 2

Cycle 1

Weight history

Total

Structural
Fuel

Response history

Vertical tip deflection
Inboard stress

Lift
Drag
Torsional tip deflection
Outboard stress

Response/maximum vs. cycle

0 4 8 12 16 20

Cycle
Accomplishments

The FIDO system has demonstrated -

- Optimization of simple HSCT design problem
- Use on network of UNIX® workstations (Sun, SGI, DEC)
- Monitoring and graphics display through “SPY”
- Flexibility of file-based data management
Plans

- Demonstrate FIDO on a complex HSCT design problem
- Replace low-fidelity with high-fidelity analysis computer programs
- Apply graphical user interfaces for problem setup and for control
- Fully document the software and publish findings
- Get and apply feedback from beta tests
FIDO Project Credits

Programming group:
Bob Weston ... project leader, data manager
Tom Eidson ... project designer, communications
Jim Townsend ... aerodynamics, propulsion
Ray Gates ... user interface
Ramki Krishnan ... aerodynamics
Ben James ... optimizer
Kelvin Edwards ... graphics, system administrator
Bill LaMarsh ... structures, performance

Consultants:
Peter Coen ... optimizer, performance
Tom Zang ... aerodynamics
Gary Giles ... structures
Gregg Wrenn ... structures
Tom Crockett ... communications
Don Randall ... user interface
Larry Green ... aerodynamics, propulsion
Mary Adams ... graphics
An Overview of the

Computer Aided Engineering and

Design for Electronics Laboratory

by Shelley Stover

sks@longstreet.larc.nasa.gov
Introduction
CAEDE Software

Core Tools

Cadence Design Tools
- Concept - Design entry
- Analog Workbench - Analog design and analysis
- Logic Workbench - Digital design and analysis
- Leapfrog - VHDL simulator
- PIC Designer - PLD design
- Profile - Analog HDL simulator
- Spice Plus - Analog simulator
- Mixed Signal Simulation Environment - mixed board design
- OpenSim - Backplane multi-engine simulator
- Parametric Plotter - Trade-off and what-if analysis
- Smoke Alarm - Stress, Power, dissipation analysis
- Veritime - Timing analysis
- Allegro - Layout tool

Matlab (Signal Analysis and Modeling)
- Matlab - High performance numeric computation
- Simulink - Simulates dynamic systems
- Signal Toolbox - Signal processing tool
- Control Toolbox - Control system engineering
- Robust Toolbox - Enhanced control toolbox
- Neural Net Toolbox - Neural network tool

Databases
- Oracle - Relational database

Documentation
- WordPerfect - Word processor

Agency - Wide:
- IHS ICSC - IC/Discrete Parameter database
- IHS Recal/Z - Passive component database
- NAS - NASA Assurance System

Hosted Tools
- Verilog XL - Verilog simulator
- Vanguard - Design entry
- Composer - Design entry
- Silvaco - 2D device simulator framework
- PSpice - Digital and analog circuit simulator
- Labview - Virtual instrument workbench
- CodeV - Optical design and analysis
- Autocad - CAD package
- CorelDraw - drawing package
- Framemaker - Word processor
- TMS320C compiler and simulator
- Softwindows - Microsoft windows emulator
- C & C++
- ADAS - Architectural design and assessment
- Foundry libraries
The Cadence Design Tools

CAEDE hosts the following Cadence tools for the design of integrated circuits and printed circuit boards.

- Concept - schematic design entry
- Composer - schematic design entry
- Verilog XL - Verilog compiler and simulator
- Leapfrog - VHDL compiler and simulator
- Profile - analog design language compiler and simulator
- PIC Designer - programmable integrated circuit design environment
- Logic Workbench Designer - digital design environment
- Analog Workbench - analog design environment
- Mixed Signal Simulation Environment
- Allegro - layout of printed circuit boards
Electronics Parts Database

- CAEDE hosts an on-line database of electronic part databooks and specifications which is updated bi-monthly.

- The database contains over 2 million integrated circuits, discrete semiconductors, and passive components. The parts can be searched by part number, characteristics, and function.

- The software is from Information Handling Systems (IHS) and is contained on 60 CDs in 10 jukeboxes. Our license is agency wide and funded by Code QE.
**Labview**

**Data flow in Labview**

- **Data Acquisition**
  - RS-232
  - GPIB
  - VXI
  - plug in data acquisition board

- **Data Analysis**
  - DSP
  - Curve fitting
  - Statistics
  - Array operations

- **Data Presentation**
  - Graphical user interface
  - Hardcopy output
  - File I/O

**Labview Virtual Instrument Components:**

- Front panel - interactive interface to supply inputs and observe outputs of the instrument

- Top-level block diagram - used to program the VI for acquisition, analysis, and formatted I/O

- Low-level panels and diagrams - together these components form sub VIs which can be used as instruments in other VIs
Front Panel of IIR Filter Example

Front Panel

- Filter Design: Butterworth
- Filter Type: Lowpass
- Ripple: 10.00 dB
- Attenuation: 40.00 dB
- Order: 5
- Filter Display: Single, Log, Continuous, Linear
- Sampling Freq. (Hz): 1000 Hz
- Cut-Off Freqs: Lower 100 Hz, Higher 300 Hz

Magnitude Response

- Frequency Range: 0.01 Hz to 500.0 Hz
- Decibels Range: -120.0 dB to 0.0 dB

Phase Response

- Frequency Range: 0.0 Hz to 500.0 Hz
- Degrees Range: -450.0° to 0°
Block Diagram of IIR Filter Example
MATLAB

Matlab
• Matlab is a high performance package for engineering and scientific computations. It allows users to manipulate matrices in a fraction of the time that it would take to write C programs, etc. Also, Matlab is able to generate several different 2D and 3D graphs.

• Matlab toolboxes:
  Control System Toolbox
  Signal Processing Toolbox
  Robust Control Toolbox
  Neural Network Toolbox

• SIMULINK - Provides a graphical user interface for modeling and simulation of dynamic systems.
Simulink Block Diagram

- Signal Gen
- Sum
- Graph Scope
- Power spectral density
- White Noise for continuous systems (Band-limited)
- Sum
- Transfer Fcn
- Graph Scope
- Power spectral density

Simulink Block Diagram
Optical Research Associates - Code V

- Code V is used for the design, optimization, tolerancing, costing, manufacturing, and alignment of optical systems.

- One of its primary uses is the development of optical systems from concept through final design.

- It handles many types of surfaces including aspherics, toroids, gratings, diffractive optical elements, and many others.

- Code V can also be used for cost estimates, environmental analysis, image simulation, and it interfaces with NASTRAN, interferometers, and CAD/CAM packages via IGES.

- Code V was used for the design and analysis of the LASE and LITE optical systems.
Silvaco

- Composed of the S-Pisces, Blaze, and Luminous simulators and the graphics package Tonyplot

- S-Pisces is the two-dimensional device modeling program that simulates the electrical characteristics of common MOS and bipolar integrated circuit elements

- Blaze contains material libraries for exotic semiconductors such as SiC and GaN as well as the codes for simulation of heterostructure devices.

- Luminous implements models for optical generation of carriers and allows simulation of a broad range of devices.
Training

- The following Cadence training videos are available:
  - Concept
  - PIC Designer with Concept
  - Top-Down Design with Logic Workbench
  - Designing with Leapfrog
  - Introduction to VHDL
  - Veritime
  - Allegro Automatic
  - Allegro Interactive
  - Analog Workbench II with Concept
  - Analog Workbench Mixed-Signal
  NOTE: These videos are accompanied by training manuals, lab manuals, and lab software and are updated as new versions are installed.

- On-line tutorials are available for Matlab, Labview and Simulink

- Basic UNIX short courses are provided by CAEDE system administrators
General Information

- CAEDE accounts are available to all center employees. To request more information or obtain a user account form contact one of CAEDE's system administrators:
  Steve Comer: steve@longstreet.larc.nasa.gov, extension 43710
  Don Brandt: dtb@jackson.karc.nasa.gov, extension 43716

- For general information on CAEDE and the Information Handling Services see the MOSAIC CAEDE Home Page.

- For IHS technical information contact
  Randy Regan: crr@longstreet.larc.nasa.gov, extension 41869
The Software Engineering and/or Ada Lab (SEAL)

Presented at the
“Role of Computers in LaRC R&D” Workshop
June 16, 1994

Robert Kudlinski
Information Systems Division

<table>
<thead>
<tr>
<th>BACKGROUND (and HOME PAGE)</th>
</tr>
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<tbody>
<tr>
<td><strong>1989</strong>: ACD (now ISD) was charged to manage, develop, and assure mission critical software systems for several on-going and all new LaRC space-flight projects</td>
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<tr>
<td><strong>1990</strong>: The SEAL was a new-start to meet this requirement by implementing a common software engineering process (<em>people, procedures, tools</em>) across these projects</td>
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<tr>
<td><strong>1992</strong>: Selected to participate in the NASA Software Engineering Program</td>
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<tr>
<td>Since inception, the SEAL has received increasing requests from other LaRC software organizations/domains seeking to improve their processes. <em>Existing and future plans</em> to help transfer software engineering technologies are presented.</td>
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LaRC SPACE-FLIGHT PROJECTS

- The SEAL has supported numerous projects at various lifecycle phases: CERES, CSI, DGV, JADE, LITE, MIDAS, RAME, ROVER, SABRE, SAFIRE, SAGE III, SEDS, SUNLITE, and TRACER
  - Primarily remote sensing, active control of space structures and technology demonstration experiments
  - On-board, embedded flight computers (primarily 80x86) for real-time instrument control and data acquisition (3,000 - 25,000 SLOC)
  - GSE computers (primarily 80x86 PC's) for instrument development, test, calibration and mission operations (5,000 - 70,000 SLOC each)
  - Missions from 3 days to 5 years, development schedules of 2-5 years

- Many technical and management challenges:
  - Real time, embedded, and non-deterministic systems require specialized tools and practices
  - Physical constraints (power, weight, radiation) severely limit computer resources (CPU speed, memory, I/O) and demand optimization
  - Insufficiently defined, continually expanding requirements
  - Trading off quality and reliability with tight manpower and schedules
  - Short term vision of projects does not help process improvement

HUMAN RESOURCES

- Education and training
  - Offering 10-15 classes per year through Training Office such as Ada, object oriented design, CM, real-time programming, rate monotonic analysis and system engineering
  - Developing, documenting, and video taping specialized in-house training such as Ada programming, formal inspections, and using real-time, embedded systems tools

- Information Resources
  - Library (books, periodicals, standards, guidebooks)
  - Electronic information exchange and communications network
  - Providing ISD user consultation service for Ada and software engineering questions

- Projects were assigned and the SEAL became a new start one year before the current climate of decreasing civil service and NPS, which has curtailed planned activities
Software Engineering and/or Ada Lab (SEAL)

PROCEDURE GOALS

- Standardize on and reuse common procedures, expertise, tools, and products across projects
  » NASA and other software standards as available and appropriate
  » Object-oriented requirements analysis and design methods
  » Ada used as the primary programming language, some C, C++

- Define and document key, repeatable software development procedures as baseline for future process improvement and training new employees
  » Formal inspections guidebook completed
  » Configuration Management Guidebook in review process
  » Guidelines for selected real-time, embedded system tools completed
  » Evaluating existing, documented IV&V procedures
  » Other procedures under development

- Moving toward a complete software lifecycle approach based on evolutionary spiral model

- Contact Pat Schuler at 4-6732 for more information on SEAL procedures

TOOLS

- Operating a distributed software development environment via LaRCNET
  » Compilers, CASE, CM, and project management tools in place
  » Beginning to use InQuisix Reuse tool
  » Reverse engineering and code analysis tools
  » Electronic information management and communication tools
  » Plan to increase use of automated code generators and testing tools
  » Work from desk, SEAL, hardware development labs and test facilities
  » Standardized environment allows sharing of tools to reduce project cost and effectively shift personnel in response to changing priorities

- Real-time, embedded analysis tools
  » Embedded system cross compilers
  » Emulators and logic analyzers for 80x86 and 1750A processors
  » Functional equivalent 80x86 and 1750A flight computers

- Contact Jerry Garcia at 4-5888 for more information on SEAL tools
Software Engineering and/or Ada Lab (SEAL)

S/W DEVELOPMENT ENVIRONMENT

Software Engineering and/or Ada Lab (SEAL)

NASA S/W ENGINEERING PROGRAM

- One Code QE mission has been to improve the quality and reliability of software products developed for space flight projects, both manned and unmanned, at the NASA Centers

- The NASA Software Engineering Program was initiated in 1991 to establish and grow Centers of Excellence across the Agency:
  - JSC: Shuttle Data Systems Branch
  - GSFC: Flight Dynamics Division (Software Engineering Lab)
  - LaRC: Flight Software and Graphics Branch (SEAL)

- Long term vision of the Program is to put self-sufficient, continually improving processes in place, establish standards, and then use these organizations to transfer effective technologies to other NASA domains

- LaRC tasks have been primarily in the areas of:
  - Capturing and documenting the SEAL software development process for small Ada projects and assessing Ada’s impact
  - Improving software technology transfer methods and software reuse
  - Evaluating IV&V procedures on LaRC projects and research testbeds
Software Engineering and/or Ada Lab (SEAL)

EXISTING GENERAL SUPPORT

- Coordinate 10-15 widely attended software engineering courses per year and sponsor educational presentations (e.g., LaRC/NASA management, LCUC, Local Universities, Conferences)
- Implement and fund formal inspections program widely used across LaRC (training, pilot projects, implementation guidebook)
- Serve on numerous review panels, technical committees and QAT's
- Transferring software technologies to other domains, such as NTF DAS, flight simulation, HSR, and TAP programs - currently manpower limited
- Open access to SEAL tools via LaRCNET - currently manpower limited for training and consulting on these tools
- Access to real-time, embedded system tools possible, but requires proper training and procedures to be followed
- Open access to library (books, periodicals, standards, guidebooks)
- Started the Software Quality, Productivity and Reliability Team to promote communication and coordination of software engineering efforts at LaRC

Software Engineering and/or Ada Lab (SEAL)

FUTURE GENERAL SUPPORT

- Continue existing work on previous chart, expanding where possible:
  - Proposal accepted by Code Q that 90% of SEAL funding (FY 95-98) specially targeted for manpower to transfer software engineering technologies to on-going LaRC research programs
  - Decreasing project load should provide ISD opportunity to shift resources from space-flight projects to general support
  - Partnerships are welcomed
- Increase use of electronic information dissemination, particularly MOSAIC
SESSION 9  CAE Tools

Chaired by

Carol D. Wieseman

9.1 Digital Control of Wind-Tunnel Models Using LabVIEW - Sherwood T. Hoadley

9.2 Electronic Engineering Notebook: A Software Environment for Research Execution, Documentation, and Dissemination - Dan Moerder

9.3 IDEAS2 Computer Aided Engineering Software - Pat Troutman

9.4 Matlab as a Robust Control Design Tool - Irene Gregory

9.5 Simulation of the Coupled Multi-Spacecraft Control Testbed at The Marshall Space Flight Center - Dave Ghosh, and Raymond C. Montgomery

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Digital Control of Wind-Tunnel Models Using LabVIEW
by
Sherwood T. Hoadley

presented at the
LaRC Computer Systems Technical Committee Workshop on
The Role of Computers in LaRC R&D
June 15-16, 1994

Digital controller and data acquisition and analysis systems were developed for several wind-tunnel models which use National Instruments LabVIEW® Software and National Instruments Hardware within a Macintosh environment. The objective of this presentation is to illustrate the use of LabVIEW for interactive animated display of acquired experimental data and real-time control of some wind-tunnel models.

The first system illustrates a flutter suppression system (FSS) which was used to suppress flutter for a small piezoelectrically actuated wing in a small flutter research and experiment device (FRED) with a 6"×6" test section. The following illustrations are included which show various aspects of the FSS system:

- A photo of FRED and a flow diagram of the wind tunnel
- A block diagram of the closed-loop system
- The digital control system software schematic of the LabVIEW user interface routines on the Macintosh and the real-time system comprised of boards plugged into the Macintosh Nu-bus but sharing their own real-time RTSI bus.
- The front panel of the FSS LabVIEW Controller virtual instrument (VI) interface to the real-time controller digital signal processor (DSP)
- Results of open and closed loop strain response to wind-tunnel turbulence

The next LabVIEW VI which is illustrated is an instrument which interfaces with a data logger which samples various thermocouples and sends the requested data to the Macintosh for display and storage. Two figures included are:

- The front panel depicting a beam clamped to a table with thermocouples placed at various locations and a strip chart displaying the data.
- The block diagram of the code for the data logger which shows the way in which LabVIEW is coded. As indicated, the code is a flow diagram of itself.

The last system illustrated is a system which provides passive control of three different aerodynamic control surfaces for a Benchmark Active Controls Testing (BACT) model in the
Transonic Dynamic Tunnel (TDT). This Passive Digital Controller System (PDCS), developed for the BACT, was used in the tunnel in November 1993 and will be used again in November 1994. It interfaces with a real-time DMA controller to command control surface positions and excitations, but does not actively employ sensor signals from the wing from which to compute control surface commands in order to suppress aeroelastic phenomena such as flutter. It does provide the following functions:

- Static command of control surfaces positions
- Excitation of control surfaces, singly or in combination
- Monitoring of control surface positions and error signals, actuator hydraulic pressures, and hinge moments
- 'Trip' System control for wind-tunnel safety
DIGITAL CONTROL OF WIND-TUNNEL MODELS USING LabVIEW

by

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LaRC Computer Systems Technical Committee Workshop on

The Role of Computers in LaRC R&D
June 15-16, 1994

OBJECTIVE

DEMONSTRATE THE USE OF LabVIEW® FOR
INTERACTIVE ANIMATED DISPLAY OF ACQUIRED EXPERIMENTAL DATA AND REAL-TIME CONTROL OF WIND-TUNNEL MODELS
LabVIEW FLUTTER SUPPRESSION CONTROLLER
VIRTUAL INSTRUMENT FRONT PANEL

DEMONSTRATION CONTROLLER

Control Law

Excitation Gain

Command Bias

Actuator Gain

OPEN AND CLOSED LOOP STRAIN RESPONSE
TO WIND TUNNEL TURBULENCE

Wind Tunnel Velocity 575 inches / second
BACT Digital Controller System

Passive Digital Controller
(No Feedback)
LabVIEW Program
Computer override
Manual override

Active Digital Controller
SUN-1 Workstation
MIMO Multi-Function Control

TDT
Actuator Commands
Sensor Signals
BACT Model

LabVIEW PASSIVE DIGITAL CONTROLLER & DATA ACQUISITION
VIRTUAL INSTRUMENT FRONT PANEL

BIAS Positions and Peak Amplitudes
Command BIAS Positions
Setup Excitation
Waveform Controls

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