Emittance Theory for Cylindrical Fiber Selective Emitter

Donald L. Chubb
Lewis Research Center, Cleveland, Ohio

November 1998
The NASA STI Program Office ... in Profile

Since its founding, NASA has been dedicated to the advancement of aeronautics and space science. The NASA Scientific and Technical Information (STI) Program Office plays a key part in helping NASA maintain this important role.

The NASA STI Program Office is operated by Langley Research Center, the Lead Center for NASA's scientific and technical information. The NASA STI Program Office provides access to the NASA STI Database, the largest collection of aeronautical and space science STI in the world. The Program Office is also NASA's institutional mechanism for disseminating the results of its research and development activities. These results are published by NASA in the NASA STI Report Series, which includes the following report types:

- TECHNICAL PUBLICATION. Reports of completed research or a major significant phase of research that present the results of NASA programs and include extensive data or theoretical analysis. Includes compilations of significant scientific and technical data and information deemed to be of continuing reference value. NASA's counterpart of peer-reviewed formal professional papers but has less stringent limitations on manuscript length and extent of graphic presentations.

- TECHNICAL MEMORANDUM. Scientific and technical findings that are preliminary or of specialized interest, e.g., quick release reports, working papers, and bibliographies that contain minimal annotation. Does not contain extensive analysis.

- CONTRACTOR REPORT. Scientific and technical findings by NASA-sponsored contractors and grantees.

- CONFERENCE PUBLICATION. Collected papers from scientific and technical conferences, symposia, seminars, or other meetings sponsored or cosponsored by NASA.

- SPECIAL PUBLICATION. Scientific, technical, or historical information from NASA programs, projects, and missions, often concerned with subjects having substantial public interest.

- TECHNICAL TRANSLATION. English-language translations of foreign scientific and technical material pertinent to NASA's mission.

Specialized services that complement the STI Program Office's diverse offerings include creating custom thesauri, building customized databases, organizing and publishing research results . . . even providing videos.

For more information about the NASA STI Program Office, see the following:

- Access the NASA STI Program Home Page at http://www.sti.nasa.gov
- E-mail your question via the Internet to help@sti.nasa.gov
- Fax your question to the NASA Access Help Desk at (301) 621-0134
- Telephone the NASA Access Help Desk at (301) 621-0390
- Write to: NASA Access Help Desk NASA Center for AeroSpace Information 7121 Standard Drive Hanover, MD 21076
Emittance Theory for Cylindrical Fiber Selective Emitter

Donald L. Chubb
Lewis Research Center, Cleveland, Ohio

Prepared for the
Fourth Conference on Thermophotovoltaic Generation of Electricity (TPV4)
sponsored by the National Renewable Energy Laboratory
Denver, Colorado, October 11–14, 1998

National Aeronautics and
Space Administration

Lewis Research Center

November 1998
EMITTANCE THEORY FOR CYLINDRICAL FIBER SELECTIVE EMITTER

Donald L. Chubb
National Aeronautics and Space Administration
Lewis Research Center
Cleveland, Ohio 44135

ABSTRACT

A fibrous rare earth selective emitter is approximated as an infinitely long cylinder. The spectral emittance, \( \varepsilon_\lambda \), is obtained by solving the radiative transfer equations with appropriate boundary conditions and uniform temperature. For optical depths, \( K_R = \alpha_\lambda R \), where \( \alpha_\lambda \) is the extinction coefficient and \( R \) is the cylinder radius, greater than 1 the spectral emittance is nearly at its maximum value. There is an optimum cylinder radius, \( R_{\text{opt}} \), for maximum emitter efficiency, \( \eta_E \). Values for \( R_{\text{opt}} \) are strongly dependent on the number of emission bands of the material. The optimum radius decreases slowly with increasing emitter temperature, while the maximum efficiency and useful radiated power increase rapidly with increasing temperature.

INTRODUCTION

Fibrous rare earth oxide selective emitters for thermophotovoltaic (TPV) applications have been of research interest for several years. Nelson (1) began working with fibrous emitters in the 1980's. In addition fibrous emitters are being developed at Quantum Group (2) and the Auburn Space Power Institute (3). This paper develops the emittance theory for a fibrous emitter by approximating the emitter as an infinitely long cylinder. Since the fibrous emitters consist of bundles of 1 to 10 \( \mu \)m diameter fibers this theory does not include the effects of the reflectance that occurs when radiation leaves a fiber and enters an adjoining fiber. The whole bundle of fibers is being approximated as a continuous cylinder. If the fibers are closely packed and the reflectance at the interface between a fiber and the medium in the voids between fibers is small, then the error resulting from the approximation should be small.

The spectral emittance of the cylinder is obtained by solving the radiative transfer equation with appropriate boundary conditions. Knowing the spectral emittance allows the emitter efficiency to be calculated. As an example, emitter efficiency is calculated for an erbium-holmium aluminum garnet and thulium aluminum garnet (Tm3A15O12) which are being studied at NASA Lewis.

TEMPERATURE OF EMITTING CYLINDER

As pointed out earlier (4), temperature drop across a planar or film type emitter causes a major reduction in the spectral emittance in the emission band of a selective emitter. However, in most cases for a cylindrical emitter there will be a negligible temperature drop. This can be seen by considering the steady state energy equation for an infinite cylinder with no internal heat generation and thermal conductivity, \( k_{\text{th}} \), and where we assume the temperature, \( T \), and radiation flux, \( Q \), depend only on the radial coordinate, \( r \).

\[
\frac{q_{\text{th}}}{\text{dr}} \frac{dT}{dr} - Q = \text{constant}
\]

In order to avoid the term in brackets being singular at \( r = 0 \) it must vanish for all \( r \). Thus at all \( r \) the conduction and radiation fluxes balance.

\[
k_{\text{th}} \frac{dT}{dr} = Q
\]
In other words, for all steady state conditions all the thermal energy being conducted into the cylinder at the outer radius, \( r = R \), leaves the cylinder as radiation.

The radiation flux, \( Q \), will always be less than the blackbody flux \( \sigma_{sb} T^4 \), where \( \sigma_{sb} = 5.67 \times 10^{-12} \text{ W/cm}^2\text{K}^4 \) is the Stefan Boltzmann constant. Therefore, define the following dimensionless variables.

\[
\bar{Q} = \frac{Q}{\sigma_{sb} T_s^4} \quad \bar{T} = \frac{T}{T_s} \quad \bar{r} = \frac{r}{R}
\]

Where \( T_s \) is the temperature at \( r = R \). In this case equation (2) becomes the following.

\[
\frac{dT}{dr} = \gamma \bar{Q}
\]

Where \( \gamma \) is the ratio of the radiation flux to the thermal conduction flux.

\[
\gamma = \frac{\sigma_{sb} T_s^4}{k_{th}} = \frac{\sigma_{sb} T_s^4 R}{k_{th}}
\]

For the ceramic materials used in most selective emitters, \( k_{th} \geq 0.01 \text{ W/cmK} \). Also, for TPV applications \( T_s \leq 2000\text{K} \). Therefore, \( \gamma \leq 4.5R(\text{cm}) \). So that if \( R < 0.1 \text{ cm} \) it is a reasonable approximation to neglect the right hand side of equation (4) and obtain the result \( \bar{T} = \text{constant} (T = T_s) \). If \( \gamma \ll 1 \) for a film or planar emitter then a linear temperature variation results (4) rather than a uniform temperature. For the cylinder emittance calculation that follows, a uniform temperature is assumed.

**FIGURE 1.** Schematic of emitting cylinder.
SPECTRAL EMITTANCE OF CYLINDER

The spectral emittance is defined as follows.

\[ \varepsilon_{\lambda} = \frac{q_{\lambda}(R)}{e_{bs}(\lambda, T_s)} \]  

(6)

Where \( q_{\lambda}(R) \) is the radiation flux leaving the cylinder at \( r = R \) and \( e_{bs}(\lambda, T_s) \) is the blackbody emissive power where \( \lambda \) is wavelength and \( T_s \) is the cylinder temperature.

\[ e_{bs}(\lambda, T_s) = \pi i_{bs}(\lambda, T_s) = \frac{2\pi hc_0^2}{\lambda^5 \left[ \exp\left( \frac{hc_0}{\lambda k T_s} \right) - 1 \right]} \]  

(7)

Appearing in equation (7) is the blackbody intensity \( i_{bs}(\lambda, T_s) \), w/cm\(^2\) nm steradian, Planck’s constant, \( h \), Boltzmann’s constant, \( k \), and the vacuum speed of light, \( c_0 \). The radiation flux, \( q_{\lambda}(R) \), is obtained by solving the radiation transfer equations for the intensities, \( i^{+}_{\lambda}(R) \), and \( i^{-}_{\lambda}(R) \). Where \( i^{+}_{\lambda}(R) \) is the intensity moving in the +R direction and \( i^{-}_{\lambda}(R) \) is the intensity moving in the −R direction as shown in figure 1. Assuming the intensities depend only on the radial coordinate leads to transport equations for \( i^{+}_{\lambda}(R) \) and \( i^{-}_{\lambda}(R) \) identical to the planar case (5). These equations are written in terms of the optical depth, \( K \), rather than the coordinate, \( r \).

\[ K = \alpha_{\lambda} r \]  

(8a)

\[ K_R = \alpha_{\lambda} R \]  

(8b)

Where \( \alpha_{\lambda} \) is the extinction coefficient, assumed independent of \( r \), and is the sum of the absorption coefficient, \( \alpha_{\lambda}^{\text{a}} \), and the scattering coefficient \( \sigma_{\lambda} \).

\[ \alpha_{\lambda} = \alpha_{\lambda}^{\text{a}} + \sigma_{\lambda} \]  

(9)

The boundary conditions that must be applied are the following. At \( r = K = 0 \), from symmetry conditions.

\[ i^{-}_{\lambda}(0) = i^{+}_{\lambda}(0) \quad \text{at} \quad K = 0 \]  

(10a)

At \( r = R \) or \( K = K_R \) the intensity moving in the −R direction is equal to the reflected intensity.

\[ i^{-}_{\lambda}(K_R) = \rho_{fo} i^{+}_{\lambda}(K_R) \quad \text{at} \quad K = K_R \]  

(10b)

Where \( \rho_{fo} \) is the reflectance at the cylinder outer radius, \( R \). At \( r = R \) total reflectance occurs for certain angles of incidence, \( \theta \). At an interface between a material with a index of refraction, \( n_i \), and a material with index of refraction, \( n_j \), where \( n_i > n_j \), radiation moving from i into j with angles of incidence \( \theta > \theta_M \), where \( \theta_M \) is given by Snell’s Law will be totally reflected. Since \( n_f > n_i \) for the cylinder-air interface we have the following result for the reflectance, \( \rho_{fo} \).

\[ \rho_{fo} = 1 \quad \text{for} \quad \theta > \theta_M \quad \text{where} \quad \mu_M^2 = \cos^2 \theta_M = 1 - \left( \frac{n_f}{n_i} \right)^2 \]  

(11a)

For the case where \( \theta < \theta_M (\mu > \mu_M) \) we approximate \( \rho_{fo} \) by the reflectance for normal incidence (6).
\[
\rho_{\text{fo}} = \left( \frac{n_f - n_0}{n_f + n_0} \right)^2 \theta < \theta_M \quad (\mu > \mu_M)
\]  

(11b)

Once the intensities \(i^+_\lambda(K_R, \theta)\) and \(i^-_\lambda(K_R, \theta)\) are obtained the radiation flux, \(q^-_\lambda(K_R)\), can be calculated.

\[
q^-_\lambda(K_R) = q^+_\lambda(K_R) - q^-_\lambda(K_R)
\]  

(12)

\[
q^+_\lambda(K_R) = 2\pi \int_0^{\pi/2} i^+_\lambda(K_R, \theta) \cos \theta \sin \theta d\theta = 2\pi \int_0^1 i^+_\lambda(K_R, \mu) \mu d\mu
\]  

(13)

\[
q^-_\lambda(K_R) = -2\pi \int_{\pi/2}^\pi i^-_\lambda(K_R, \theta) \cos \theta \sin \theta d\theta = 2\pi \int_0^1 i^-_\lambda(0, \mu) \mu d\mu
\]  

(14)

Solution of the radiative transfer equations for \(q^-_\lambda(K_R)\) is presented in (5) for the film or planar case with a linear temperature variation through the film and for no scattering (\(G_s = 0\) in eq. (9)). Results for the uniform temperature cylinder can be obtained from these results by setting \(\epsilon = 0\) and \(\Delta T = 0\) in equations (33) to (36) of reference (5). When this is done the following result is obtained for the spectral emittance.

\[
\epsilon_\lambda = \frac{n_f^2(1 - \rho_{\text{fo}})}{1 - 4E_3(K_R) - \rho_{\text{fo}} E_3(K_R) + \mu_M^2 l - \rho_{\text{fo}} E_3 \left( \frac{K_R}{\mu_M} \right)}
\]  

(15)

\(E_3(x)\) is the exponential integral defined as follows.

\[
E_n(x) = \int_0^1 z^{n-2} \exp \left( -\frac{x}{z} \right) dz
\]  

(16)

The reflectance, \(\rho_{\text{fo}}\), is given by equation (11b) and \(\mu_M\) is given by equation (11a).

As equations (15) and (11) indicate, for no scattering the spectral emittance of a uniform temperature cylinder depends on the optical depth, \(K_R\), and indices of refraction, \(n_f\), and \(n_0\). For single crystal materials, such as rare earth doped yttrium aluminum garnet (YAG), scattering should be negligible. However, for polycrystalline rare earth oxides such as those being considered in references (1) to (3) scattering should be significant. In those cases equation (15) will overestimate the spectral emittance.

Consider \(\epsilon_\lambda\) for the two limiting conditions \(K_R = 0\) and \(K_R \to \infty\). The \(K_R = 0\) case corresponds to a wavelength where the material is transparent. While the \(K_R \to \infty\) case applies to an emission band of a selective emitter. Since \(E_3(0) = 1/2\) and \(\lim_{x \to \infty} E(x) = 0\), equation (15) yields the following:

\[
\epsilon_\lambda = 0 \quad \text{for} \quad K_R = 0
\]  

(17)

\[
\lim_{K_R \to \infty} \epsilon_\lambda = n_f^2(1 - \rho_{\text{fo}})
\]  

(18)

Equation (18) is the usual result for an opaque body emitting (or absorbing) in vacuum (\(n_0 = 1\)).
FIGURE 2. Spectral emittance for cylinder of radius, R, at a constant temperature, \( T_s \), as a function of optical depth, \( K_R = \alpha_L R \), where \( \alpha_L \) is the extinction coefficient and \( n_f \) is the cylinder index of refraction and \( n_o \) is the surrounding index of refraction.

In figure 2 the spectral emittance, \( \varepsilon_\lambda \), using equation (15) is shown as a function of optical depth, \( K_R \), for \( n_o = 1 \) and \( n_f = 1, 1.5 \) and 2.0. As figure 2 shows \( \varepsilon_\lambda \) increases rapidly with \( K_R \) and reaches nearly its limiting value (eq. (18)) for \( K_R = 1 \). Notice also that for small \( K_R \) as \( n_f \) increases the spectral emittance rate of increase also increases. For most of the selective emitter materials, \( 1.5 \leq n_f \leq 2.0 \).

FIGURE 3. Theoretical spectral emittance of \( \text{Er}_{0.3} \text{Ho}_{2.7} \text{Al}_5 \text{O}_{12} \) cylinder of radius, \( R = 0.4 \text{ cm} \), calculated using measured extinction coefficient and index of refraction.
Once the extinction coefficient, \( \alpha_\lambda \), and index of refraction, \( n_\lambda \), are known, equation (15) can be used to calculate \( \varepsilon_\lambda \). Figure 3 shows the calculated spectral emittance for an erbium-holmium aluminum garnet cylinder \( (R = 0.4 \text{ cm}) \) with 10\% erbium and 90\% holmium, \( \text{Er}_{0.3}\text{Ho}_{2.7}\text{Al}_5\text{O}_{12} \). This single crystal material is being considered for a film type selective emitter (5). The extinction coefficient and index of refraction were obtained from reference (5).

Holmium has its main emission band centered at \( \lambda = 2000 \text{ nm} \) with smaller bands centered at \( \lambda = 1100, 890 \) and 750 nm. Erbium has its main emission band centered at \( \lambda = 1500 \text{ nm} \) with secondary bands centered at \( \lambda = 1000, 800 \) and 640 nm. All of these bands show up as regions of large emittance in figure 3. In the region \( 2000 < \lambda < 4500 \text{ nm} \), \( \text{Er}_{0.3}\text{Ho}_{2.7}\text{Al}_5\text{O}_{12} \) is nearly transparent \( (\alpha_\lambda \rightarrow 0) \) and thus \( \varepsilon_\lambda \) is small. The highly oscillatory result in this region results from numerical error in \( \alpha_\lambda \) (5).

For the region \( \lambda > \lambda_c = 5000 \text{ nm} \) \( \varepsilon_\lambda \) becomes large again. This large \( \varepsilon_\lambda \) results from vibrational modes of the crystal lattice and is a characteristic of most rare earth selective emitter materials (1). We call the wavelength, \( \lambda_c \), the long wavelength cutoff.

**EMITTER EFFICIENCY**

As a measure of the effectiveness of a selective emitter define the emitter efficiency as follows.

\[
\eta_E = \frac{\text{useful radiated power}}{\text{total radiated power}} = \frac{Q_b}{Q_T} = \frac{\int_0^\lambda q_\lambda(R) d\lambda}{\int_0^\lambda q_\lambda(R) d\lambda} = \frac{\int_0^\lambda \varepsilon_\lambda e_{bs}(\lambda,T_s) d\lambda}{\int_0^\lambda \varepsilon_\lambda e_{bs}(\lambda,T_s) d\lambda}
\]

(19)

The numerator, \( Q_b \), is the power radiated in the wavelength region \( 0 < \lambda < \lambda_g \). In a TPV system \( \lambda_g \) corresponds to the bandgap energy, \( E_g = \hbar c / \lambda_g \), of the PV cell. The denominator is the total radiated power, \( Q_T \).

**FIGURE 4.** Emitter efficiency, \( \eta_E \), and useful power, \( Q_b \), as a function of cylinder radius, \( R \), at emitter temperature, \( T_s = 1700 \text{ K} \) for \( \text{Er}_{0.3}\text{Ho}_{2.7}\text{Al}_5\text{O}_{12} \).

Consider how \( \eta_E \) will be behave as a function of the cylinder radius, \( R \). As figure 2 shows, \( \varepsilon_\lambda \) increases rapidly with optical depth, \( K_R = \alpha_\lambda R \). Therefore, for the emission bands and the long wavelength region (large \( \alpha_\lambda \) and \( \lambda < \lambda_g \) or \( \lambda > \lambda_c \)) \( \varepsilon_\lambda \) will quickly approach its limiting value as \( R \) increases from zero. However, for the regions between the emission bands and the long wavelength cutoff (small \( \alpha_\lambda \) and \( \lambda_l < \lambda < \lambda_c \)) the spectral emittance will increase more slowly to its limiting value as \( R \) increases from zero. Thus, the numerator of equation (19), \( Q_b \), will rapidly rise from zero to its limiting value as \( R \) increases from zero. At the same time, the denominator, \( Q_T \), will increase more slowly to its limiting value as \( R \) increases from zero.
increase more slowly and will continue increasing with R while the numerator is increasing at a much lower rate. As a result, there will be an optimum radius, $R_{\text{opt}}$, that will yield maximum $\eta_{\text{EMAX}}$. This is illustrated in figure 4 for $\text{Er}_{0.3}\text{Ho}_{2.7}\text{Al}_5\text{O}_{12}$ where $\lambda_e = 2200$ nm ($E_e = 0.56$ eV) was chosen for use in equation (19) and $T_s = 1700$ K. As can be seen, $\eta_E$ rises rapidly to $\eta_{\text{EMAX}}$ and then decreases slowly for $R > R_{\text{opt}}$. Also shown in figure 4 is the useful power radiated, $Q_b$. As can be seen, $Q_b$ rises rapidly and then begins to level off. For $T_s = 1700$K the optimum radius is $R_{\text{opt}} = 0.34$ cm. It should be mentioned that for radii of this magnitude the uniform temperature assumption becomes questionable. In figures 5(a) and (b) $R_{\text{opt}}$, $\eta_E(R_{\text{opt}})$ and $Q_b(R_{\text{opt}})$ are shown as functions of $T_s$. Both $\eta_E(R_{\text{opt}})$ and $Q_b(R_{\text{opt}})$ increase significantly with temperature while $R_{\text{opt}}$ decreases only 25 percent in going from $T_s = 1200$ to 2000K. The large increase in $\eta_E$ results because the maximum value of the blackbody emissive power, $e_{bs}(\lambda, T_s)$, shifts to shorter wavelengths as $T_s$ increases. Therefore, $Q_b$ increases faster than $Q_T$ as $T_s$ increases. The useful power, $Q_b$, increases at least as $T_s^4$.

**FIGURE 5.** Optimum radius, $R_{\text{opt}}$, maximum efficiency, $\eta_{\text{EMAX}}$, and useful power, $Q_b$, at $R = R_{\text{opt}}$ as functions of emitter temperature, $T_s$, for $\text{Er}_{0.3}\text{Ho}_{2.7}\text{Al}_5\text{O}_{12}$. (a) Optimum radius, $R_{\text{opt}}$. (b) Maximum efficiency and useful power.
Now consider thulium aluminum garnet, $\text{Tm}_3\text{Al}_5\text{O}_{12}$, as a cylindrical selective emitter. Thulium has its main emission band centered at $\lambda = 1700$ nm with smaller bands centered at $\lambda = 1200, 800$ and $700$ nm. Thus compared to $\text{Er}_{0.3}\text{Ho}_{0.7}\text{Al}_5\text{O}_{12}$, which has 8 emission bands and thus a large region of high $\epsilon_\lambda$ for $\lambda < \lambda_\epsilon$, $\text{Tm}_3\text{Al}_5\text{O}_{12}$ has a much smaller high emittance region for $\lambda < \lambda_\epsilon$. For $\text{Tm}_3\text{Al}_5\text{O}_{12}$, $\lambda_\epsilon = 1900$ nm was chosen. Just as for $\text{Er}_{0.3}\text{Ho}_{0.7}\text{Al}_5\text{O}_{12}$, thulium aluminum garnet has large emittance for $\lambda > \lambda_\epsilon = 5000$ nm.

![Figure 6](https://example.com/figure6.png)

**FIGURE 6.** Optimum radius, $R_{\text{opt}}$, maximum efficiency, $\eta_{\text{EMax}}$, and useful power, $Q_b$, at $R = R_{\text{opt}}$ as functions of emitter temperature, $T_s$, for $\text{Tm}_3\text{Al}_5\text{O}_{12}$. (a) Optimum radius, $R_{\text{opt}}$. (b) Maximum efficiency and useful power.

In figure 6(a) and (b), $R_{\text{opt}}$, $\eta_{\text{EMax}}(R_{\text{opt}})$, and $Q_b(R_{\text{opt}})$ are shown as functions of $T_s$ for $\text{Tm}_3\text{Al}_5\text{O}_{12}$. These results were calculated using measured values for $\alpha_\lambda$ and $n_\lambda$ (5). The first thing to notice is that $R_{\text{opt}}$ for $\text{Tm}_3\text{Al}_5\text{O}_{12}$ is much smaller than $R_{\text{opt}}$ for $\text{Er}_{0.3}\text{Ho}_{0.7}\text{Al}_5\text{O}_{12}$. This occurs because thulium aluminum garnet has a smaller region of large emittance than erbium-holmium aluminum-garnet, as mentioned above. Therefore, the majority of $Q_b$ results from emission in the main emission band centered at $\lambda = 1900$ nm where $\alpha_\lambda = 30$ cm$^{-1}$. Thus when $R = 0.3$ mm, $K_R = 1$.
and \( e_{\lambda} \) will be near its maximum value (fig. 2). As a result, \( Q_b \) will also be near its maximum and thus \( \eta_E \) will be a maximum. Therefore, we expect that \( R_{opt} \) will occur near 0.3 mm for thulium aluminium garnet. Figure 6(a) substantiates this result.

Comparing figures 5(b) and 6(b) we see that the emitter efficiencies are nearly the same for \( \text{Er}_{0.3}\text{Ho}_{2.7}\text{Al}_5\text{O}_{12} \) and \( \text{Tm}_{3}\text{Al}_5\text{O}_{12} \). However, the useful power, \( Q_b \), is much larger for \( \text{Er}_{0.3}\text{Ho}_{2.7}\text{Al}_5\text{O}_{12} \). This occurs for two reasons. First of all, erbium-holmium aluminium garnet has a much larger region of large emittance for \( \lambda < \lambda_{\text{g}}^* \) than thulium aluminum garnet. Secondly, \( \lambda_{\text{g}}^* = 1900 \text{ nm} \) for thulium aluminum garnet whereas \( \lambda_{\text{g}}^* = 2200 \text{ nm} \) for erbium-holmium aluminum garnet. Thus more of the spectrum is included in \( Q_b \) for \( \text{Er}_{0.3}\text{Ho}_{2.7}\text{Al}_5\text{O}_{12} \).

An important point about selective emitters is brought out by comparing figures 5(b) and 6(b). That is, \( \eta_E \) and useful power, \( Q_b \), do not behave in the same manner. Increasing \( Q_b \) by adding more emission bands, as in the case of \( \text{Er}_{0.3}\text{Ho}_{2.7}\text{Al}_5\text{O}_{12} \), does not mean that \( \eta_E \) will increase.

CONCLUSION

Most fibrous rare earth selective emitters consist of bundles of 1 to 10 \( \mu \text{m} \) diameter bundles. In this study that bundle has been approximated as an infinite cylinder. From the solution to the radiative transfer equations the spectral emittance of the cylindrical emitter was calculated. Several conclusions can be made about the cylindrical rare earth selective emitters.

1. For most rare earth selective emitters the temperature is uniform through the cylinder.
2. When the optical depth \( K_R = \alpha_R R \geq 1 \) the spectral emittance is nearly a maximum.
3. There is an optimum value for the radius, \( R_{opt} \), which yields maximum emitter efficiency.
4. \( R_{opt} \) strongly depends on the emitter material. For an emitter with only a single strong emission band \( R_{opt} \) is the order of 0.1 mm, whereas for an emitter with many emission bands \( R_{opt} \) is the order of 1 mm.
5. \( R_{opt} \) decreases slowly with increasing emitter temperature.
6. The maximum efficiency, \( \eta_{EMAX} \), and useful power, \( Q_b \), increase significantly with temperature.

REFERENCES

# Emittance Theory for Cylindrical Fiber Selective Emitter

Donald L. Chubb

## Performing Organization Name(s) and Address(es)
National Aeronautics and Space Administration
Lewis Research Center
Cleveland, Ohio 44135–3191

## Sponsor/Monitoring Agency Name(s) and Address(es)
National Aeronautics and Space Administration
Washington, DC 20546–0001

## Supplementary Notes

## Abstract
A fibrous rare earth selective emitter is approximated as an infinitely long cylinder. The spectral emittance, \( \varepsilon_{\lambda} \), is obtained by solving the radiative transfer equations with appropriate boundary conditions and uniform temperature. For optical depths, \( K_R = \alpha_\lambda R \), where \( \alpha_\lambda \) is the extinction coefficient and \( R \) is the cylinder radius, greater than 1 the spectral emittance is nearly at its maximum value. There is an optimum cylinder radius, \( R_{opt} \), for maximum emitter efficiency, \( \eta_{E} \). Values for \( R_{opt} \) are strongly dependent on the number of emission bands of the material. The optimum radius decreases slowly with increasing emitter temperature, while the maximum efficiency and useful radiated power increase rapidly with increasing temperature.