The Aquarius ocean salinity mission high stability
L-band radiometer

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Abstract—The NASA Earth Science System Pathfinder (ESSP) mission Aquarius, will measure global ocean surface salinity with 
\( \sim 120 \text{ km} \) spatial resolution every 7-days with an average monthly salinity accuracy of 0.2 psu (parts per thousand) [1]. This requires an L-band low-noise radiometer with the long-term calibration stability of 0.15 K over 7 days. The instrument utilizes a push-broom configuration which makes it impractical to use a traditional warm load and cold plate in front of the feedhorn. Therefore, to achieve the necessary performance Aquarius utilizes a Dicke radiometer with noise injection to perform a warm – hot calibration. The radiometer sequence between antenna, Dicke load, and noise diode has been optimized to maximize antenna observations and therefore minimize NEDT. This is possible due the ability to thermally control the radiometer electronics and front-end components to 0.1 °C rms over 7 days.

Keywords—radiometer; L-band radiometry; salinity; stable radiometers

I. INTRODUCTION

The radiometers are Dicke radiometers that use noise injection for calibration. A critical requirement in the design of these radiometers is long-term (several days) stability. Stability is critical because significant averaging must be done to achieve the Aquarius goal for an accuracy of 0.2 psu (global rms on a monthly basis). Since it will take 7 days to map the globe, the radiometers must be stable over at least 7 days. The design requirement set for Aquarius was that the radiometers be stable to within 0.13 K over 7 days. A primary element in maintaining stability is adequate internal calibration and good thermal control. The design adopted for the Aquarius radiometers is based on research conducted under NASA’s Instrument Incubator Program (Ultra-Stable Radiometers, [2]) that uses two internal reference sources (noise diode and Dicke load).

II. RADIOMETER CONFIGURATION

The radiometer is divided into several elements as shown in Fig. 1 and Fig. 2. Beginning at the OMT, these elements are the OMT Couplers, the Correlated Noise Diode (CND), two diplexers (one for each polarization), the Radiometer Front End (RFE), the Radiometer Back End (RBE), and the Digital Processing Unit (DPU). The OMT Couplers, CND, diplexers, and RFE are mounted on the OMT as shown in Fig. 1. Thermal control of these elements is critical to obtaining the radiometric stability needed for successful retrieval of salinity at the accuracy desired for Aquarius. The OMT assembly is cold-biased passively by means of radiator plates and appropriate coatings and actively controlled using heaters mounted to the assembly. The design requirement is to control the thermal environment to change less than 0.1 °C over 7 days.

III. RADIOMETER FRONT-END

The primary amplification is done in the radiometer front ends (RFE). There is a separate RFE for each feed assembly. In the RFE, the two signals from the OMT (one for vertical polarization and one for horizontal polarization) are amplified and then combined to form four channels (vertical, horizontal, and the sum and difference which is equivalent polarization at ±45 degrees). The sum and difference signal will be used to compute the third Stok’s parameter (e.g. detected with a square-law detector in the RBE and later subtracted during the ground processing). The first elements at the input of the RFE are the Dicke switch and its reference load followed by a coupler to a noise diode that provides the hot load (Fig. 2). Together, these are the references used for internal calibration. These calibration references are at the heart of the radiometer architecture and tight thermal control is needed to meet the required radiometric performance. In order to meet the required performance, the radiometric temperature of the Dicke load must be known with an uncertainty of < 50 mK, and the coupled noise temperature must be stable to < 300 ppm. These parameters, coupled with the thermal control mentioned above, have been shown to be adequate to achieve the required radiometric stability (0.13K over 7 days; [2]). Also, this radiometer architecture is largely implemented via microstrip-based technologies in order to reduce size and improve thermal control.

Between the OMT and the RFE are the OMT Coupler and the diplexers (Fig. 2). The OMT coupler provides a port for injecting signal from a second noise diode, the CND. This signal is used to monitor the phase and amplitude balance.
between the channels, a calibration that is necessary for proper polarimetric performance and calculation of the third Stokes' parameter. The diplexers are devices that allow the scatterometer and radiometer to use the same feed horn assembly without damage to the radiometer electronics. They are cavity type filters that provide enough rejection to guarantee no damage to the radiometer from the Aquarius scatterometer pulse. To provide additional isolation, whenever the scatterometer transmits the Dicke switches are switched to the reference load. The total isolation (dipllexer plus switch) prevents any active component in the RFE from being saturated by the scatterometer.

IV. RADIOMETER BACK-END

The radiometer back end (RBE) contains additional amplification, band-pass filtering, and the detectors for each channel (Fig. 2). Its performance (stability) is less critical to the overall stability of the radiometer system because it is located behind all the calibration sources and after the first stage gain in the RFE. The design requirement for temperature control of the RBE is a maximum change of 0.4 C (rms) over 7 days. Because less control is necessary, the RBE has been physically separated from the more critical elements mounted on the OMT to facilitate the thermal control of this group.

The final stage of the radiometer is the digital processing unit (DPU). The detected signals from each radiometer channel are digitized in the RBE using voltage-to-frequency converters (VFC). These devices output pulses whose frequency is proportional to the detected signal. These pulses are counted asynchronously, and the frequency determined, by the DPU. The DPU also houses the radiometer controller and collects temperature and housekeeping data.

V. RADIOMETER OPERATION

Fig. 3 shows the timing diagram for the hardware. The fundamental timing unit for the hardware is 10 ms (approximately 1 ms for the scatterometer transmit pulse and 9 ms of observation time for the radiometer). The radiometer and scatterometer operation are alternated at a rapid rate so that the two sensors look at the same piece of ocean nearly simultaneously. The three radiometers (one for each beam) operate in parallel. During 120 ms each radiometer collects 7 samples (i.e., 9 ms long and repeated each 10 ms) looking into the antenna followed by 5 samples devoted to the calibration sources (two noise diodes and Dicke load). This 120 ms sequence is then repeated. However, because of limitations with the on-board data storage, not all of this data will be downloaded. It is planned to average two of the antenna looks and transmit to ground three samples at the 10 ms resolution followed by two at 20 ms. The samples of the calibration references transmitted to the ground will be the average of 10 samples.

The switching sequence between antenna, Dicke load, and noise diodes has been optimized to maximize antenna observations and minimize NEDT. This optimization is based on the fact that the radiometer gain and receiver temperature are relatively stable quantities. Therefore, the duty cycle of the calibration observations can be reduced to be consistent with the time constant of these parameters. The stable values needed for calibration are obtained by averaging over longer periods. The effect of this approach is a significantly higher duty cycle for the antenna (i.e., ocean) observations with an improved NEDT compared to the standard approach using equal duty cycles [2].

VI. RADIO FREQUENCY INTERFERENCE

The radiometer has also been designed to include precautions against Radio Frequency Interference (RFI). A study conducted by NASA and ITT looked at potentially damaging sources, as well as interference that would impact the science retrieval [3]. The primary source of interference for the radiometer is ground-based air surveillance radars. The level of interference is a function of out-of-band emissions from these radars and also a function of the rejection level of radiometer filters in the skirts where the radar band is located (i.e., where the radar is permitted to transmit). The findings of the study suggested a three-tiered approach. The first step is to include 2 watt limiters to protect the low noise amplifiers (LNA) in the RFE against worst case damage. The second approach is to sample at sufficiently high rate to facilitate identification of RFI with the ability to remove it without complete loss of data. The study suggested that the radiometer data should not be averaged more than about 30 ms before download. The current design calls for most data to be down loaded at 10 ms sample rate and some at 20 ms. Finally, the radiometer band-limiting filters (located in the RBE) were designed to reduce the potential interference to a probability of <0.1% over the ocean. In terms of filter parameters, this requirement is equivalent to a 25 MHz bandwidth, 7-pole Chebyshev filter. This is achieved with several filter stages. These include the dipllexer and the two band pass filters in the RBE before the detector (Fig. 2).
Figure 2. The radiometer is subdivided into the front-end, back-end, and digital processing unit as shown in this block diagram.
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REFERENCES


AQUARIUS/SAC-D

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Radiometer Key Features

- Long-term calibration stability of ≤0.13 K over 7 days.
- Microstrip Dicke radiometers that use noise injection for calibration.
- Tight thermal control of radiometer electronics.
- The design was adopted from research conducted under NASA's Instrument Incubator Program for Ultra-Stable Radiometers.
Radiometer Hardware Layout

- DPU on similar panel on the other side
- RBE
- Field joint connectors
- CND (below)
- Coupler (1 of 2)
- Diplexer (1 of 2)
- RFE
- Radiator
- Orthomode Transducer (OMT)
- Thermal Isolator
- Feed Horn

Aquarius Radiometer, Pellerano, et.al.

1 August, 2006
Aquarius Radiometer Operation

- Design based on 3-position asymmetric Dicke switching for antenna temperature estimate
- Radiometer electronics internal calibration
  - Data from Ng data packets for receiver gain (G) estimate
  - Data from Nr data packets for receiver noise (Tr) estimate
  - Use G and Tr for Ti estimate
- Feed/OMT assembly and front-end losses and emission corrections using temperature measurements.

\[
\begin{align*}
V_i &= G(T_i + T_r) + Z_0 \\
V_0 &= G(T_0 + T_r) + Z_0 \\
V_{ND} &= G(T_0 + T_{ND} + T_r) + Z_0
\end{align*}
\]

\[
\begin{align*}
T_i &\quad \rightarrow \quad 1 \text{ data packet (5.76 sec)} \\
G &\quad \leftarrow \quad \text{Ng packets (69 sec)} \\
T_r &\quad \leftarrow \quad \text{Nr packets (276 sec)}
\end{align*}
\]
Key RFE Calibration Devices

$T_v, T_\theta, T_{ND}$

ND Coupler

Dicke Load Tref

Noise Diode ENR(T)

Hybrid

Splitter

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Thermal Design
and Key RFE Device Temperature

- There is essentially no orbital temperature variation (<0.02 deg C) for the current design.
- This will allow a great simplification of the calibration model, in particular for RFE.
The radiometer maintains synchronicity with the scatterometer by operating from the same STALO and in the same time basis as the scatterometer's 100 Hz PRF.
### Radiometer Timing Sequence (cont.)

#### Timing
- 10-ms steps
- 12 steps make 1 subcycle
- 12 subcycles make one block

#### Radiometer Sequencing
- Antenna samples
  - 70 ms per sub-cycle
  - 3 x 10- and 2 x 20-ms bins
- Internal calibration
  - 3-state asymmetric Dicke switching
  - 4-state sequence for polarimetric RFE calibration
  - Averaged over 10 samples/block
- CND for polarimetric efficiency calibration once a subcycle
- Null offset (blank) measured every block

<table>
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<th>H-Channel</th>
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<td>2</td>
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</tr>
<tr>
<td>144</td>
<td>DL</td>
<td>ND*</td>
</tr>
</tbody>
</table>

**Ant** = antenna  
**DL** = Dicke load  
**CND** = Correlated noise diode  
**ND** = Internal noise diode w/ Dicke load  
**ND*** = Internal noise diode w/ antenna  
**Blank** = zero offset
Aquarius Radiometer NEDT

- Aquarius radiometer design is based on three-position Dicke switching
  - Data from Ng (12) data packets (69 sec) for gain estimate
  - Data from Nr (48) data packets (276 sec) for receiver noise estimate

\[
\Delta T_{NEDT} = \sqrt{\left(T_i + T_r\right)^2 + \left(\frac{1}{B\tau_i} + \frac{\Delta G}{G}\right)^2 + c^2 \left(\Delta T_r\right)^2}
\]

- NEDT does not include calibration errors for feed/diplexer/OMT/Dicke switch loss and temperature, TND and T0.
• 4.5 second input temperature integration time
• 10 step switching sequence with 60% dedicated to input
• Gain averaged for 35 seconds
• Trx averaged for 500 seconds
Radio Frequency Interference

- The radiometer has power handling/front-end limiters that protect up to +33 dBm, greater than worst-case RFI level predictions.
- It uses a 7-pole band definition filter to reduce the potential interference to a probability of <0.1% over the ocean.
- The system will downlink data averaged < 20 ms on board for estimation and removal of large RFI outliers.

![Interference CDF For Each Beam Graph]
Radiometer
Pre-Launch Calibration Approach

- Calibrate antenna and radiometer subsystems separately and in parallel
- Verify calibration with antenna (sans reflector) and radiometer as a system
Current Status

- Engineering units have been built and tested for functional and RF performance.
- 7-day stability testing on-going.

Images:
- RFE
- Units without their cover
- RBE
- DPU
Summary

- The Aquarius radiometer will provide the necessary long-term calibration stability to perform salinity observations.
- The basis of the radiometer design has been demonstrated to work as part of NASA’s Instrument Incubator Program.
- The radiometer provides the necessary provisions to minimize the effects of RFI.
- The necessary thermal control has been recently demonstrated during a key thermal development test at JPL.
- An Engineering Model has been built and is performing well so far.
- Long-term stability testing is currently ongoing at GSFC.