IIP Update: A Packaged Coherent Doppler Wind Lidar Transceiver

“Doppler Aerosol WiNd lidar (DAWN)”


to

Working Group on Space-Based Lidar Winds
Welches, OR
27 June 2006
# IIP Key Personnel

<table>
<thead>
<tr>
<th>Name</th>
<th>Organization</th>
<th>Role/Responsibility</th>
</tr>
</thead>
<tbody>
<tr>
<td>Dr. Michael J. Kavaya</td>
<td>NASA LaRC</td>
<td>PI</td>
</tr>
<tr>
<td>Dr. Farzin Amzajerdian</td>
<td>NASA LaRC</td>
<td>Co-I, coherent lidar receiver lead</td>
</tr>
<tr>
<td>Dr. Grady J. Koch</td>
<td>NASA LaRC</td>
<td>Co-I, overall lidar system lead &amp; field demonstration lead</td>
</tr>
<tr>
<td>Mr. Ed A. Modlin</td>
<td>NASA LaRC</td>
<td>Technician</td>
</tr>
<tr>
<td>Dr. Upendra N. Singh</td>
<td>NASA LaRC</td>
<td>Co-I, LRRP PI</td>
</tr>
<tr>
<td>Mr. Bo. C. Trieu</td>
<td>NASA LaRC</td>
<td>Mechanical and system engineering</td>
</tr>
<tr>
<td>Dr. Jirong Yu</td>
<td>NASA LaRC</td>
<td>Co-I, pulsed transmitter laser lead</td>
</tr>
<tr>
<td>Dr. Yingxin Bai</td>
<td>SAIC</td>
<td>Laser design</td>
</tr>
<tr>
<td>Mr. Mulugeta Petros</td>
<td>STC</td>
<td>Laser design</td>
</tr>
<tr>
<td>Mr. Paul Petzar</td>
<td>SAIC</td>
<td>Electronic Design</td>
</tr>
<tr>
<td>Mr. Karl Reithmaier</td>
<td>SAIC</td>
<td>Opto-mechanical design</td>
</tr>
</tbody>
</table>

Also many thanks to Brian Killough, Keith Murray, Garnett Hutchinson, and Ken Anderson
### IIP Motivation

<table>
<thead>
<tr>
<th>Mission</th>
<th>Measurement</th>
<th>Technique</th>
<th>Technology</th>
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<tbody>
<tr>
<td><strong>Primary</strong></td>
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<tr>
<td>Science: Weather, Climate</td>
<td>Earth Vertical <strong>Wind</strong> Profiles</td>
<td>Scanning Doppler Lidar</td>
<td>Pulsed, 2-Micron, Ho Laser</td>
</tr>
<tr>
<td><strong>Secondary</strong></td>
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<tr>
<td>Science: Climate</td>
<td>Earth Vertical <strong>CO₂</strong> Concentration Profiles</td>
<td>Scanning DIAL Lidar</td>
<td>Pulsed, 2-Micron, Ho Laser</td>
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<tr>
<td>Science &amp; Exploration: Atmos. Char., EDL</td>
<td>Mars Vertical <strong>Density</strong> Profiles</td>
<td>DIAL Lidar (CO₂)</td>
<td>Pulsed, 2-Micron, Ho Laser</td>
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<td>Science &amp; Exploration: Atmos. Char., EDL</td>
<td>Mars Vertical <strong>Wind</strong> Profiles</td>
<td>Scanning Doppler Lidar</td>
<td>Pulsed, 2-Micron, Ho Laser</td>
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<tr>
<td>Science: Climate</td>
<td>Earth Vertical <strong>Aerosol</strong> Concentration Profiles</td>
<td>Backscatter Lidar</td>
<td>Pulsed, 2-Micron, Ho Laser</td>
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<tr>
<td>Science &amp; Exploration: Atmos. Char., EDL</td>
<td>Mars Vertical <strong>Dust</strong> Profiles</td>
<td>Backscatter Lidar</td>
<td>Pulsed, 2-Micron, Ho Laser</td>
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</tbody>
</table>
IIP Abstract

The state-of-the-art 2-micron coherent Doppler wind lidar breadboard at NASA/LaRC will be engineered and compactly packaged consistent with future aircraft flights. The packaged transceiver will be integrated into a coherent Doppler wind lidar system test bed at LaRC. Atmospheric wind measurements will be made to validate the packaged technology.

This will greatly advance the coherent part of the hybrid Doppler wind lidar solution to the need for global tropospheric wind measurements.
IIP and the Global Tropospheric Wind Profiles Roadmap

2-Micron Coherent Doppler Lidar

- 2 micron laser 1988
- Diode Pump Technology 1993
- Inj. Seeding Technology 1996
- High Energy Technology 1997
- Conductive Cooling Techn. 1999
- Compact Packaging 2005
- Packaged Lidar Ground Demo. 2006

- Autonomous Oper. Technol.
- Space Qualif.
- Lifetime Validation
- Pre-Launch Validation
- 3-Yr. Lifetime Validation
- Demo, NPOESS
- Threshold, 400 km

1 micron laser
- Diode Pump Technology
- Inj. Seeding Technology
- Conductive Cooling Techn.
- High Energy Technology
- Compact Packaging 2006
- Packaged Lidar Ground Demo. 2006

0.355-Micron Direct Doppler Lidar

- Ground-Based Risk Reduction (IPO)
- 2 micron Doppler wind aircraft flights
- 1 micron altimetry space missions
- Pump Laser Diode Advancement
- Dual Wavelength Telescope & Scanner
- Past Funding
- Laser Risk Reduction Program
- IIP-2004 Projects
- Optional

OSSE’s

Lidar Perf. Simulations

Dual Wavelength Telescope & Scanner
IIP TRL Advancement

“4 → 5”

**Compact, Engineered Packaging**

- **Aircraft Validation**
- **Lifetime Demonstration**
- **Space Qualification Tests**

### TRL Levels

- **TRL 1**: Basic principles observed and reported
- **TRL 2**: Technology concept and/or application formulated
- **TRL 3**: Analytical and experimental critical function and/or characteristic proof-of-concept
- **TRL 4**: Component and/or breadboard validation in laboratory environment
- **TRL 5**: Component and/or breadboard validation in relevant environment (Ground or Space)
- **TRL 6**: System/subsystem model or prototype demonstration in a relevant environment (Ground or Space)
- **TRL 7**: System prototype demonstration in a space environment
- **TRL 8**: Actual system completed and “flight qualified” through test and demonstration (Ground or Flight)
- **TRL 9**: Actual system “flight proven” through successful mission operations
## IIP and the LaRC Development of Pulsed, 2-Micron Laser Technology For Space

<table>
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<tr>
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<th>Sub-Category/Date</th>
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<th>9/02</th>
<th>2/03</th>
<th>4/03</th>
<th>11/03</th>
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<th>12/05</th>
<th>LRRP</th>
<th>IIP</th>
<th>SPACE DEMO</th>
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</tr>
</tbody>
</table>

**Notes:**
- All liquid: All liquid cooling system.
- Partially conductive: Partially conductive cooling system.
- All cond w/o heat pipe: All condensation without heat pipe.
- All cond w/ heat pipe: All condensation with heat pipe.
- C Package: Package C.
- A package: Package A.
- AA package: Package AA.
- G package: Package G.
- Laboratory Table: Laboratory table packaging.
- Compact, Engineered: Compact, engineered packaging.
IIP- Milestones & Schedule

**Year 1**
- **12/20/05**: Complete packaged transceiver requirements document (NOW)
- **12/20/06**: Conceptual DR CoDR
- **12/20/07**: Demo prototype breadboard transmitter

**Year 2**
- **12/20/06**: CDR
- **12/20/07**: Demo performance of engineered oscillator

**Year 3**
- **12/20/07**: Complete lab testing of engineered transceiver
- **12/20/08**: Integrate engineered transceiver Into testbed
- **12/20/08**: Complete test of engineered transceiver In lidar testbed
## IIP Packaged Transceiver Requirements

<table>
<thead>
<tr>
<th>Category</th>
<th>Requirement</th>
<th>Goal (if different) and/or Space Requirement</th>
<th>Reason</th>
</tr>
</thead>
<tbody>
<tr>
<td>Laser Architecture</td>
<td>Master Oscillator Power Amplifier (MOPA)</td>
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<td>High energy, beam quality, optical damage</td>
</tr>
<tr>
<td>Laser Material</td>
<td>Ho:Tm:LuLiF</td>
<td></td>
<td>High energy, high efficiency, atmospheric transmission</td>
</tr>
<tr>
<td>Nominal Wavelength</td>
<td>2.053472 microns</td>
<td></td>
<td>Atmospheric transmission</td>
</tr>
<tr>
<td>Pulse Energy</td>
<td>150 mJ</td>
<td>250 (space)</td>
<td>Computer modeling of measurement performance</td>
</tr>
<tr>
<td>Pulse Repetition Frequency</td>
<td>10 Hz</td>
<td>10-20 (space)</td>
<td>Shot accumulation, optimum laser diode array lifetime</td>
</tr>
<tr>
<td>Pulse Beam Quality</td>
<td>&lt; 1.4 x diffraction limit</td>
<td></td>
<td>Heterodyne detection efficiency influence</td>
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<tr>
<td>Pulse Spectrum</td>
<td>Single Frequency</td>
<td>Few MHz (space)</td>
<td>Frequency estimation process</td>
</tr>
<tr>
<td>Injection seeding success</td>
<td>95%</td>
<td>99%</td>
<td>Shot accumulation</td>
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<tr>
<td>Laser Heat Removal</td>
<td>Partial Conductively Cooled</td>
<td>FCC (space)</td>
<td>No liquid lines in space</td>
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<tr>
<td>Packaging</td>
<td>Compact, engineered</td>
<td>Aircraft ready Space qual. (space)</td>
<td>As ready as possible for aircraft follow on</td>
</tr>
</tbody>
</table>
Oscillator features

- Injection seeded
- Cavity length >3m Ring
- Output coupler Reflectivity ~70%
- Diode pump lasers: 36 bars 100W/b conductive cooled
- Crystal doped material length 21mm
- Undoped LuLF length 15 mm
- Laser crystal cooling: H₂O, Methanol
- Tube size: 6mm OD 5mm ID AR coated for 792nm
- Laser rod ends wedged 0.5° along c-axis AR coated for 2.053μm
- Laser rod cylinder AR coated for 792nm
Oscillator Head
Oscillator cavity length

- Long cavity length is needed to obtain narrow linewidth
  - Pulse length is one of the critical parameters of a coherent Lidar.
  - A short pulse compromises frequency resolution while a long pulse compromises range resolution.
  - To meet the pulse length requirement, the oscillator length was changed from 2m to 3m. It prolongs the pulse width to near 200ns
  - The resonator has six mirrors and 8 bounces.
Amplifier features

- Pump energy: 7.2 Joules 12x6 bar arrays with 100 watts/bar
- Diode laser: conductive cooled ‘AA’ Pkg
- Laser crystal: Ho:Tm:LuLF 0.5% Ho 6% Tm
- Doped Crystal length: 41 mm
- Ends diffusion bonded crystals: 15 mm undoped LuLF crystals
- Laser crystal cooling: H₂O
- Flow tube size: 6 mm OD 5 mm ID AR coated 792 nm
- Rod end surfaces AR coated for 2.053 µm
- Laser cylinder AR coated for 792 nm
- Path configuration: double pass
Amplifier Module
Proposed Transceiver “Box”

- Modular approach with injection seed & local oscillator, separate from transceiver.
  - Separate seed/LO allows flexibility to adapt to 4 measurements scenarios:
    - Simple, fixed frequency LO for ground or low platform speed.
    - Higher intermediate frequency for high platform speed.
    - Swept LO for very high platform speed.
    - DIAL of CO₂

Note: only optical paths are represented; electrical and water paths are not shown.
LRRP Pulsed, 2-Micron Laser Transmitter Opto-Mechanical Design

• 3-m, bow-tie, unidirectional master oscillator power amplifier
• Seeding and receiver optics on reverse side
• Expect this hardware in about 8 weeks for LRRP
Seed Laser

CW seed laser

Seed laser driver
**Seed/LO Option 1**

- baseline design for ground-based implementation.
- recommended for IIP demonstration.
- fiber-to-free space through AOM then back to fiber is disadvantageous—looking into fiber optic pigtailed AOM.
- could be packaged in rack-mount breadboard with fan for cooling (need thermal analysis).
Test Bed: Putting it all Together

- CW master oscillator
- AOM 105 MHz
- Isolator
- PBS
- Amplifier
- 50/50 coupler
- Aspheric optic (if necessary)
- Beam expander
- λ/4
- Turning mirror (deflects beam up toward scanner)
- λ/2
- Q-switch
- PZT
- Resonance detector
- Pulse monitor (on flip stage)
- Visible alignment laser
- Fiber optic port
VALIDAR Telescope

- off axis Dall-Kirkham design.
- 6-inch aperture
- 20X expansion
 VALIDIDAR Scanner

• scanner is mounted on roof of laboratory trailer.
• 8-inch clear aperture.
• can be pointed or scanned in elevation/azimuth for hemispherical coverage.
• linked to data acquisition computer for automated profiling of wind.
Data Acquisition and Processing System

- **Scanner Control Computer**
- **Lidar Transceiver**
- **Analog Front End**
- **Digitizer**
  - Digital Signal Processor--2 cards of 4 chips
- **Control Program**
- **Host CPU**
- **Storage Devices**
- **Real-Time Display & User Interface**

Connections:
- RS-232
- CompactPCI Bus
Atmospheric Measurements
(will be better than this VALIDAR sample)

jets

virga

rain enhancing backscatter

passage of frontal system
Summary

• IIP project 6 months into 36 month effort
• On schedule and budget to date
• Leveraging LRRP work on compact laser in 05 and 06
• Plan on significant steps of compact, engineered packaging of state-of-the-art laser/lidar technology. TRL definitions do not reveal significant progress.
• Companion IIP at GSFC for noncoherent Doppler wind lidar will complement this project to permit hybrid DWL on aircraft and then in space
• Project very consistent with findings of NASA/ESTO Laser/Lidar Technology Requirements Working Group results (FY06). To be issued in final report
• Anticipate strong endorsement of global winds by NAS decadal study on earth sciences
• Same technology promises additional applications for earth and Mars
Project Motto 1

• Be Prepared
Project Motto 2

• Walk before you run
BACK UP
IIP – Scope of the Project
Pulsed Doppler Wind Lidar Measurement Scenario

Lidar System

Propagation Path (Atmosphere)

Target (Atmospheric Aerosols)
Lidar System

Lasers and Optics

Electronics

Computer, Data Acquisition, and Signal Processing (including software)
Lasers and Optics

Lidar Transceiver (to be engineered under IIP)

- Transmitter Laser
- Receiver

Additional components:
- Pol. BS
- λ/4 Plate
- T/R Switch

Large Optics (telescope, scanner)
Receiver

From T/R Switch

Photodetector

Seed Oscillator or Local Oscillator
Laser Design Considerations

- Laser wavelength
- Laser material
- Laser pumping geometry
- Laser cavity design
- Laser architecture
Why Ho: Tm: LuLuF

• Why Ho laser?
  – Tm lasers in 2-µm region have such a low gain cross-section ($\sigma_{em} \sim 10^{-20}$ cm$^2$) that efficient, high-energy laser amplification is impossible without the risk of laser crystal or associated optics damage.
  – Ho lasers have large enough stimulated emission cross-section ($\sigma_{em} \sim 10^{-19}$ cm$^2$) for effective amplification to obtain high-energy.

• Why co-doped?
  – Takes advantage of diode pumping for Tm lasers
  – Takes advantage of the efficient Tm 1:2 relaxation energy transfer process
  – Takes advantage of the high emission cross-section of Ho laser
Why Ho:Tm:LuLiF — Cont.

• Why fluoride?
  – Fluoride
    • Long upper laser level lifetime ~ 15 ms, store more energy
    • Low up-conversion loss
    • Higher emission cross-section
    • Naturally birefringent material, no depolarization loss
    • Negative dn/dT → weak thermal lensing
  – Garnet
    • Isotropic
    • Excellent thermo-mechanical properties

• Why Lutetium?
  – Lanthanide series ions
    • Lutetium, Yttrium, Gadolinium
  – Lutetium
    • Lutetium – larger crystal field
      – larger manifold stark splitting → Small thermal population of ground state
Laser Architecture
Master Oscillator Power Amplifier (MOPA)

• **Energy requirement**
  – Single oscillator can’t produce required energy

• **Beam quality**
  – MOPA preserves the good beam quality

• **Lifetime**
  – Permits more derating of pump diodes

• **Efficiency**
  – Multiple pass amplifier improving the efficiency

• **Optics Damage**
  – Reducing intra-cavity fluence
Cavity Configuration

• Linear Cavity
  – Standing waves
  – Simple
  – Round trip - pass gain medium twice

• Ring Cavity
  – Traveling waves
  – No spatial hole burning in the gain-> single mode
  – Long cavity needed to obtain narrow linewidth
  – Beneficial for injection seeding through output coupler
Pumping configuration

• Pumping geometry
  – Side Pumping
    » Power scaling
    » Uniform pumping
  – End Pumping
    » Easy thermal management
    » Easy to mode match
    » Higher pump density

• Single Longitudinal Mode
  – Interferometric mode selection
  – Monolithic design, short cavity
  – Injection seeding
Optical Bench

• Two options:
  26.5 x 23.0 x 7 inch single side
  26.5 x 11.5 x 7 inch double sided
• The split can be done such that the receiver optics and the seed laser on one side, and the power Oscillator amplifier on the other.
• Optical bench is water cooled, enclosed and dry purged.