In order to support the environmental test needs for our new Orion and Constellation program, NASA is developing unique world-class test facilities. To optimize this testing of spaceflight hardware while minimizing transportation issues, a one-stop, under one roof test capability is being developed at the Space Power Facility at the NASA Glenn Research Center's Plum Brook Station. This facility will provide the capability to perform the following environmental testing: (1) reverberation acoustic testing, (2) mechanical base-shake sine testing, (3) modal testing, (4) thermal-vacuum testing, and (5) EMI/EMC (electromagnetic interference and compatibility) testing. An overview of this test capability will be provided in this presentation, with special focus on the two new vibroacoustic test facilities currently being designed and built, the Reverberant Acoustic Test Facility (RATF) and the Mechanical Vibration Facility (MVF). Testing of the engineering developmental hardware and qualification hardware of the Orion (Crew Exploration Vehicle) will commence shortly after the facilities are commissioned.
Overview of the Orion Vibroacoustic Test Capability at NASA Glenn Research Center

Presented by:
William O. Hughes, Aron D. Hozman, Mark E. Mcnelis and Kim D. Otten
NASA Glenn Research Center
William.O.Hughes@nasa.gov

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Components of NASA’s Constellation Program

- Earth Departure Stage
- Orion - Crew Exploration Vehicle
- Heavy Lift Launch Vehicle
- Altair - Lunar Lander
- Crew Launch Vehicle
Orion System Elements

Crew Exploration Vehicles (CEV) consists of four functional modules:

- **Launch Abort System**: emergency escape during launch
- **Crew Module**: crew and cargo transport
- **Service Module**: propulsion, electrical power, fluids storage
- **Spacecraft Adapter**: structural transition to launch vehicle
Test Facility Overview

- The Space Power Facility at the NASA Glenn Plum Brook Station in Sandusky, OH is developing an environmental test capability for the NASA Constellation Program, and the Orion Crew Exploration Vehicle (CEV).
- The CEV is part of the next generation of manned space flight vehicles being developed for return to the Moon and for Mars.
- Testing will be conducted on the Ground Test Article (GTA) and Qualification Test Article in a “test as you fly” configuration.

Environmental Facility Capability:

- Acoustic Vibration
- Mechanical Vibration
- Modal
- Thermal-Vacuum
- EMI/EMC
Space Power Facility (SPF)
Space Environmental Testing Under One Roof

• Maintain thermal vacuum test capability of SPF.

• Incorporate EMI/EMC testing within vacuum test chamber.

• Design and build two new vibroacoustic test capabilities within disassembly (west) area of SPF Facility
  – Mechanical Vibration Facility (MVF) (sine and modal)
  – Reverberant Acoustic Test Facility (RATF)

• Vibroacoustic facilities required to be commissioned (verification testing successfully completed) by July 2009
Provide Orion Critical Path Testing and Support Future Constellation Testing

Perform modifications to the Space Power Facility for:
- Acoustic Vibration
- Mechanical Vibration
- Modal
- Thermal-Vacuum
- EMI/EMC

Ensure Mechanical and Acoustic Vibration Facilities capabilities meet CEV Testing requirements and do not preclude future Constellation element testing

Conceptual Drawing
Thermal Vacuum & EMI/EMC

Thermal Vacuum

• Vacuum to 10-6 torr; 100 ft Diameter, 122 ft Height
• Thermal simulation: -200 F to 175 F
  – GN2-cooled Cryoshroud
    • 2 new ceiling cryogenic panels (1700 ft² total)
    • Modify cryoshroud suspension structure and refurbish existing cryoshroud panels.
  – Thermal heat flux simulation (14 zones, 480 V)
    • 1 sun thermal heat flux
    • Phase angle power supply controllers
    • 14 zones, 400 kW/zone

EMI/EMC

• All tests are performed inside the aluminum vacuum chamber using the chamber as an RF shield
  – Unwanted, external signals stay outside test chamber
  – Signals generated for tests remain inside test chamber
• Suspended, moveable equipment platforms provide direct, localized RF illumination of the vehicle
• Radiated susceptibility and shielding effectiveness testing at frequencies up to 35 GHz
Major Area for Facility Upgrades
Vibration Test Area

Conceptual Drawing
Technical Requirements Highlights

• **Mechanical Vibration Facility (MVF)**
  – Single axis sine vibration (vertical, lateral)
  – CEV test article: 75,000 lb, 75 ft height, 18 ft diameter
  – Design for Overturning Moment of CEV
  – Change axis of input motion without removing the test article
  – Seismic mass to accommodate 120,000 lbm test article for modal

• **Reverberant Acoustic Test Facility (RATF)**
  – 163 dB OASPL (empty chamber)
  – 20 minute continuous run times
  – Designed for 47 ft tall x 20 ft diameter (CEV elements with margin)
  – Accommodates Altair - Lunar Lander (32.8 ft diameter)

• **High-Speed Data Acquisition System (HSDAS)**
  – 1024 channels dynamic data
  – Acceleration, pressure, strain, temperature
  – Support both MVF and RATF needs
Reverberant Acoustic Test Facility (RATF)

<table>
<thead>
<tr>
<th>Chamber Properties</th>
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<tbody>
<tr>
<td>Chamber Size</td>
<td>47.5 ft L x 37.5 ft W x 57 ft H</td>
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<tr>
<td>Chamber Volume</td>
<td>101,189 ft³</td>
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<tr>
<td>Horns/Modulators</td>
<td>23 High Power Modulators and 23 Horns *</td>
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<tr>
<td>Nominal acoustic power installed</td>
<td>4,000 kW *</td>
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<tr>
<td>Nominal GN₂ flow rate</td>
<td>84,000 scfm *</td>
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<tr>
<td>Main Door Opening</td>
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<tr>
<td>Number of Main Doors</td>
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<tr>
<td>Door Type</td>
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<tr>
<td>OASPL, empty</td>
<td>163 dB OASPL</td>
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</table>

* Subject to design changes
Reverberant Acoustic Test Facility (RATF)

RATF Proposed SOW Updates - Rev 6

![Graph showing SPL vs. 1/3 Octave Frequency]

- C1 (Adjusted Lockheed Pad Abort) 163 dB OASPL
- C2 (Adjusted VTC SOW Ascent Abort) 163 dB OASPL
- C3 (Pathfinder for Altair) 156.6 dB OASPL
- C4 (NASA-HDBK-7005) 137.8 dB OASPL
- C5 (Adjusted ALAS Nominal P95 Envelope) 163 dB OASPL
- C6 (Double Peak) 163 dB OASPL
- C7 (Internal Payload) 153 dB OASPL
- C8 (Alternate double peak) 163.0 dB OASPL
Mechanical Vibration Facility (MVF)

- Low-frequency (servo-hydraulic) sinusoidal vibration system
  
  **Vertical**: 0 to 1.25-g's; 5 to 150-Hz; Sine sweep up to 4 octaves/min

  **Horizontal**: 0 to 1.0-g's; 5 to 150-Hz; Sine sweep up to 4 octaves/min

- 3 single axis test capability of CEV Full Stack; Change axis of input motion without removing the test article

- Table surface below grade to accommodate test article height
MVF Facility Layout

Modal Floor Area

Seismic Mass

CEV Sine Vibration Area

Conceptual Layout
Vibration System

16 Vertical, 4 Horizontal exciters give full 6 DOF Vibration Table
High Speed Data Acquisition System (HSDAS)

COTS Data Acquisition proposed
- Open hardware and software architecture
- Real-time displays
- Simultaneous data acquisition and access
- Scalable/Reconfigurable
  - 1024 Channels
  - Expansion to provide 1536 channels
- Supports MVF and RATF
Construction Status

- Construction mobilization continues in parallel with demolition
- Installed Thermal/Vacuum Chamber barrier
- Completed geo-probe samples and analyses - received approval to remove soil
- Floor demolition complete
- Hot cell demolition completed by end of December 2007
- MVF excavation started January 2008 (19 ft deep; 56.6 ft by 49.5 ft)
Construction Status

- Current RATF Foundation Status:
  - RATF foundation excavation has been completed.
  - RATF foundation construction started April 2008.
## CEV Test Schedule

<table>
<thead>
<tr>
<th>Test Article</th>
<th>Test Begins</th>
<th>Test Completion</th>
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<tbody>
<tr>
<td>Ground Test Article (GTA)</td>
<td>September 2009</td>
<td>April 2010</td>
</tr>
<tr>
<td>Qualification</td>
<td>May 2011</td>
<td>June 2012</td>
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Summary

• Orion Integrated Environmental Testing Project provides one-stop testing under one roof at the NASA Glenn Plum Brook Station.

• NASA will have World Class Facilities available to support Constellation Architectural elements environmental testing needs:
  – Acoustic Vibration
  – Mechanical Vibration
  – Modal
  – Thermal Vacuum
  – EMI/EMC

• Facilities will support the spaceflight testing required for the development (GTA) and qualification of the Orion CEV hardware.
Contact Information:
William O. Hughes
NASA Glenn Research Center
William.O.Hughes@nasa.gov