INTRODUCTION

Tech Briefs are short announcements of innovations originating from research and development activities of the National Aeronautics and Space Administration. They emphasize information considered likely to be transferable across industrial, regional, or disciplinary lines and are issued to encourage commercial application.

Availability of NASA Tech Briefs and TSPs

Requests for individual Tech Briefs or for Technical Support Packages (TSPs) announced herein should be addressed to

National Technology Transfer Center
Telephone No. (800) 678-6882 or via World Wide Web at www2.nttc.edu/leads/

Please reference the control numbers appearing at the end of each Tech Brief. Information on NASA’s Innovative Partnerships Program (IPP), its documents, and services is also available at the same facility or on the World Wide Web at http://ipp.nasa.gov.

Innovative Partnerships Offices are located at NASA field centers to provide technology-transfer access to industrial users. Inquiries can be made by contacting NASA field centers listed below.

NASA Field Centers and Program Offices

<table>
<thead>
<tr>
<th>Center</th>
<th>Contact Person</th>
<th>Phone Number</th>
<th>Email Address</th>
</tr>
</thead>
<tbody>
<tr>
<td>Ames Research Center</td>
<td>Lisa L. Lockyer</td>
<td>(650) 604-1754</td>
<td><a href="mailto:lisa.l.lockyer@nasa.gov">lisa.l.lockyer@nasa.gov</a></td>
</tr>
<tr>
<td>Dryden Flight Research Center</td>
<td>Gregory Poteat</td>
<td>(661) 276-3872</td>
<td><a href="mailto:greg.poteat@dfrc.nasa.gov">greg.poteat@dfrc.nasa.gov</a></td>
</tr>
<tr>
<td>Glenn Research Center</td>
<td>Kathy Needham</td>
<td>(216) 433-2802</td>
<td><a href="mailto:kathleen.k.needham@nasa.gov">kathleen.k.needham@nasa.gov</a></td>
</tr>
<tr>
<td>Goddard Space Flight Center</td>
<td>Nona Cheeks</td>
<td>(301) 286-5810</td>
<td><a href="mailto:nona.k.cheeks@nasa.gov">nona.k.cheeks@nasa.gov</a></td>
</tr>
<tr>
<td>Jet Propulsion Laboratory</td>
<td>Ken Wolfenbarger</td>
<td>(818) 354-3821</td>
<td><a href="mailto:james.k.wolfenbarger@jpl.nasa.gov">james.k.wolfenbarger@jpl.nasa.gov</a></td>
</tr>
<tr>
<td>Kennedy Space Center</td>
<td>Michele Brekke</td>
<td>(281) 483-4614</td>
<td><a href="mailto:michele.a.brekke@nasa.gov">michele.a.brekke@nasa.gov</a></td>
</tr>
<tr>
<td>Langley Research Center</td>
<td>Martin Waszak</td>
<td>(757) 864-4015</td>
<td><a href="mailto:martin.r.waszak@nasa.gov">martin.r.waszak@nasa.gov</a></td>
</tr>
<tr>
<td>Marshall Space Flight Center</td>
<td>Jim Dowdy</td>
<td>(256) 544-7604</td>
<td><a href="mailto:jim.dowdy@msfc.nasa.gov">jim.dowdy@msfc.nasa.gov</a></td>
</tr>
<tr>
<td>Stennis Space Center</td>
<td>John Bailey</td>
<td>(228) 688-1660</td>
<td><a href="mailto:john.w.bailey@nasa.gov">john.w.bailey@nasa.gov</a></td>
</tr>
</tbody>
</table>

Carl Ray, Program Executive
Small Business Innovation Research (SBIR) & Small Business Technology Transfer (STTR) Programs
(202) 358-4652
carl.g.ray@nasa.gov

Doug Comstock, Director
Innovative Partnerships Program Office
(202) 358-2560
doug.comstock@nasa.gov
<table>
<thead>
<tr>
<th>5</th>
<th>Technology Focus: Sensors</th>
</tr>
</thead>
<tbody>
<tr>
<td>5</td>
<td>Gas Sensors Based on Coated and Doped Carbon</td>
</tr>
<tr>
<td></td>
<td>Nanotubes</td>
</tr>
<tr>
<td>6</td>
<td>Tactile Robotic Topographical Mapping Without</td>
</tr>
<tr>
<td></td>
<td>Force or Contact Sensors</td>
</tr>
<tr>
<td>6</td>
<td>Thin-Film Magnetic-Field-Response Fluid-Level</td>
</tr>
<tr>
<td></td>
<td>Sensor for Non-Viscous Fluids</td>
</tr>
<tr>
<td>7</td>
<td>Progress in Development of Improved Ion-</td>
</tr>
<tr>
<td></td>
<td>Channel Biosensors</td>
</tr>
<tr>
<td>8</td>
<td>Simulating Operation of a Complex Sensor</td>
</tr>
<tr>
<td></td>
<td>Network</td>
</tr>
<tr>
<td>9</td>
<td>Electronics/Computers</td>
</tr>
<tr>
<td>9</td>
<td>Using Transponders on the Moon to Increase</td>
</tr>
<tr>
<td></td>
<td>Accuracy of GPS</td>
</tr>
<tr>
<td>9</td>
<td>Controller for Driving a Piezoelectric Actuator</td>
</tr>
<tr>
<td></td>
<td>at Resonance</td>
</tr>
<tr>
<td>10</td>
<td>Coaxial Electric Heaters</td>
</tr>
<tr>
<td>11</td>
<td>Dual-Input AND Gate From Single-Channel Thin-</td>
</tr>
<tr>
<td></td>
<td>Film FET</td>
</tr>
<tr>
<td>12</td>
<td>High-Density, High-Bandwidth, Multilevel</td>
</tr>
<tr>
<td></td>
<td>Holographic Memory</td>
</tr>
<tr>
<td>15</td>
<td>Manufacturing &amp; Prototyping</td>
</tr>
<tr>
<td>15</td>
<td>Fabrication of Gate-Electrode Integrated</td>
</tr>
<tr>
<td></td>
<td>Carbon-Nanotube Bundle Field Emitters</td>
</tr>
<tr>
<td>17</td>
<td>Materials</td>
</tr>
<tr>
<td>17</td>
<td>Hydroxide-Assisted Bonding of Ultra-Low-</td>
</tr>
<tr>
<td></td>
<td>Expansion Glass</td>
</tr>
<tr>
<td>17</td>
<td>Photochemically Synthesized Polyimides</td>
</tr>
<tr>
<td>18</td>
<td>Optimized Carbonate and Ester-Based Li-Ion</td>
</tr>
<tr>
<td></td>
<td>Electrolytes</td>
</tr>
<tr>
<td>21</td>
<td>Mechanics/Machinery</td>
</tr>
<tr>
<td>21</td>
<td>Compact 6-DOF Stage for Optical Adjustments</td>
</tr>
<tr>
<td>22</td>
<td>Ultrasonic/Sonic Impacting Penetrators</td>
</tr>
<tr>
<td>23</td>
<td>Miniature, Lightweight, One-Time-Opening Valve</td>
</tr>
<tr>
<td>25</td>
<td>Software</td>
</tr>
<tr>
<td>25</td>
<td>Supplier Management System</td>
</tr>
<tr>
<td>25</td>
<td>Improved CLARAty Functional-Layer/Decision-</td>
</tr>
<tr>
<td></td>
<td>Layer Interface</td>
</tr>
<tr>
<td>25</td>
<td>JAVA Stereo Display Toolkit</td>
</tr>
<tr>
<td>25</td>
<td>Remote-Sensing Time Series Analysis, a</td>
</tr>
<tr>
<td></td>
<td>Vegetation Monitoring Tool</td>
</tr>
<tr>
<td>26</td>
<td>PyPele Rewritten To Use MPI</td>
</tr>
<tr>
<td>26</td>
<td>Data Assimilation Cycling for Weather Analysis</td>
</tr>
<tr>
<td>27</td>
<td>Bio-Medical</td>
</tr>
<tr>
<td>27</td>
<td>Hydrocyclone/Filter for Concentrating Biomarkers</td>
</tr>
<tr>
<td></td>
<td>From Soil</td>
</tr>
<tr>
<td>28</td>
<td>Activating STAT3 Alpha for Promoting Healing</td>
</tr>
<tr>
<td></td>
<td>of Neurons</td>
</tr>
<tr>
<td>29</td>
<td>Physical Science</td>
</tr>
<tr>
<td>29</td>
<td>Probing a Spray Using Frequency-Analyzed Light</td>
</tr>
<tr>
<td></td>
<td>Scattering</td>
</tr>
</tbody>
</table>

This document was prepared under the sponsorship of the National Aeronautics and Space Administration. Neither the United States Government nor any person acting on behalf of the United States Government assumes any liability resulting from the use of the information contained in this document, or warrants that such use will be free from privately owned rights.
Gas Sensors Based on Coated and Doped Carbon Nanotubes

Large specific surface areas of nanotubes could enable attainment of high sensitivities.

*Ames Research Center, Moffett Field, California*

Efforts are underway to develop inexpensive, low-power electronic sensors, based on single-walled carbon nanotubes (SWCNTs), for measuring part-per-million and part-per-billion of selected gases (small molecules) at room temperature. Chemically unmodified SWCNTs are mostly unresponsive to typical gases that one might wish to detect. However, the electrical resistances of SWCNTs can be made to vary with concentrations of gases of interest by coating or doping the SWCNTs with suitable materials. Accordingly, the basic idea of the present development efforts is to incorporate thus-treated SWCNTs into electronic devices that measure their electrical resistances.

A typical sensor device based on this concept includes a set of interdigitated metal microelectrodes fabricated by photolithography on an electrically insulating substrate. In preparation for fabricating the SWCNT portion of such a sensor, a batch of treated (coated or doped) SWCNTs is dispersed in a solvent. The resulting suspension of SWCNTs is drop-deposited or injected onto the area containing the interdigitated electrodes. As the solvent evaporates, the SWCNTs form an irregular mesh that connects the electrodes. The density of the SWCNTs in the mesh can be varied by varying the concentration of SWCNTs in the suspension and/or the amount of suspension dropped on the electrode area. To enable acquisition of measurements for comparison and to aid in calibration, undoped SWCNTs can be similarly formed on another, identical set of interdigitated electrodes.

Examples of coating materials that have been tested thus far include chlorosulfonated polyethylene (which imparts sensitivity to chlorine) and hydroxypropyl cellulose (which imparts sensitivity to hydrogen chloride). Examples of dopants that have been tested are Pd nanoparticle doped SWCNTs (which imparts sensitivity to methane) and other dopants include clusters of nanoparticle catalystic metals — for example, Pt and Au for sensing hydrogen and hydrocarbons or Cu and Rh for sensing nitric compounds.

Although the response of a sensor of this type is a generally monotonic function of the concentration of a gas species of interest, it is not always a linear function, sometimes it is a linear function to the logarithmic concentration. Moreover, a given sensor can be sensitive to more than one gas, and the sensitivities to different gases can be expected to differ. In a typical contemplated application, a device would incorporate multiple sensors, each tailored to maximize its response to a specific gas or family of gases, and the sensor would be operated in an environment containing a mixture of gases. The readings of the sensors would be digitized and processed by algorithms for differentiating the gases and estimating their concentrations or at least limits on the concentration of a gas species of interest in the face of the nonlinear responses to the various gases that may be present.

This sensor technology has been developed to readily scale up for mass production with high yield and good reproducibility. The sensing device can be used for *in-situ* air monitoring, wireless network sensing, and in-line chemical detection.

*This work was done by Jing Li and Meyya Meyyappan of Ames Research Center and Yi-jiang Lu of Eloret Corporation.*

This invention is owned by NASA, and a patent application has been filed. Inquiries concerning rights for the commercial use of this invention should be addressed to the Ames Technology Partnerships Division at (650) 604-2954. Refer to ARC-15566-1.
**Thin-Film Magnetic-Field-Response Fluid-Level Sensor for Non-Viscous Fluids**

This sensor would be inexpensive and easy to fabricate.

*Langley Research Center, Hampton, Virginia*

An innovative method has been developed for acquiring fluid-level measurements. This method eliminates the need for the fluid-level sensor to have a physical connection to a power source or to data acquisition equipment. The complete system consists of a lightweight, thin-film magnetic-field-response fluid-level sensor (see Figure 1) and a magnetic field response recorder that was described in “Magnetic-Field-Response Measurement-Acquisition System” (LAR-16908-1), *NASA Tech Briefs*, Vol. 30, No. 6 (June 2006), page 28.

The sensor circuit is a capacitor connected to an inductor. The response recorder powers the sensor using a series of oscillating magnetic fields. Once electrically active, the sensor responds with its own harmonic magnetic field. The sensor will oscillate at its resonant electrical frequency, which is dependent upon the capacitance and inductance values of the circuit.

**Figure 1:** Thin-film magnetic field response Fluid-Level Sensor is a capacitor connected to an inductor to form a resonant circuit.
The capacitance value of the sensor increases as the amount of fluid that the sensor is exposed to increases. When the energy is in the inductor, the harmonic magnetic field produced can be interrogated. The response recorder interrogates the sensor response and correlates the response frequency to fluid level.

A thin layer of silicon nitride film is deposited on interdigital electrodes as shown in Figure 2 to electrically insulate the sensor’s capacitor. Silicon nitride also can be placed on the inductor. The fluid-level sensor uses interdigital electrodes for the capacitor that are electrically connected in parallel to a spiral trace inductor. The advantage of this design is that the entire sensor can be embodied as a thin film, and can be directly deposited to the inner wall of a nonconductive container for measuring non-viscous fluids.

In Figure 2, a fluid having dielectric constant, \( \kappa \), is in contact with \( m \) pairs of electrodes (e.g., placed in a fluid such that \( m \) electrode pairs are immersed). Each electrode pair has a capacitance of \( C_{\text{free}} \) when not immersed in the fluid and \( C_{\text{immersed}} = \kappa C_{\text{free}} \). The advantage of this method is that it serves as a lightweight, thin-film method of measuring fluids that are non-viscous. Another advantage is that the level measurements are discretized. The sensor capacitance, \( C(m) \), for a sensor having \( n \) electrode pairs increases as the number of electrode pairs, \( m \), in contact with the dielectric increases.

\[
C(m) = (n - m)C_{\text{free}} + mC_{\text{immersed}}
= (n - m + \kappa m)C_{\text{free}}
= [n + m(\kappa - 1)]C_{\text{free}}
\]

When the electrodes are electrically connected to an inductor, a resonant circuit is formed having the resonant frequency of

\[
\omega = \frac{1}{2\pi \sqrt{[n + m(\kappa - 1)]LC_{\text{free}}}}
\]

The sensor response frequency ranges from its maximum value when the capacitor is not immersed \( (m = 0) \)

\[
\omega_{\text{max}} = \frac{1}{2\pi \sqrt{[n]LC_{\text{free}}}}
\]

to its minimum when the capacitor is completely immersed \( (m = n) \).

\[
\omega_{\text{min}} = \frac{1}{2\pi \sqrt{[n]LC_{\text{free}}}}
\]

This work was done by Stanley E. Woodard, Qamar A. Shams, and Robert L. Fox of Langley Research Center and Mr. Bryant D. Taylor of SWALES Aerospace. For more information, contact the Langley Innovative Partnerships Office at (757) 864-8881. Refer to LAR-16614-1.

**Progress in Development of Improved Ion-Channel Biosensors**

Improvements in design and fabrication have been made since a previous report.

*NASA’s Jet Propulsion Laboratory, Pasadena, California*

Further improvements have recently been made in the development of the devices described in “Improved Ion-Channel Biosensors” (NPO-30710), *NASA Tech Briefs*, Vol. 28, No. 10 (October 2004), page 30. As discussed in more detail in that article, these sensors offer advantages of greater stability, greater lifetime, and individual electrical addressability, relative to prior ion-channel biosensors.

In order to give meaning to a brief description of the recent improvements, it is necessary to recapitulate a substantial portion of the text of the cited previous article. The figure depicts one sensor that incorporates the recent improvements, and can be helpful in understanding the recapitulated text, which follows:

These sensors are microfabricated from silicon and other materials compatible with silicon. Typically, the sensors are fabricated in arrays in silicon wafers on glass plates. Each sensor in the array can be individually electrically addressed, without interference with its neighbors. Each sensor includes a well covered by a thin layer of silicon nitride, in which is made a pinhole for the formation of a lipid bilayer membrane. In one stage of fabrication, the lower half of the well is
Simulating Operation of a Complex Sensor Network

NASA's Jet Propulsion Laboratory, Pasadena, California

Simulation Tool for ASCTA Microsensor Network Architecture (STAMiNA) ["ASCTA" denotes the Advanced Sensors Collaborative Technology Alliance.] is a computer program for evaluating conceptual sensor networks deployed over terrain to provide military situational awareness. This or a similar program is needed because of the complexity of interactions among such diverse phenomena as sensing and communication portions of a network, deployment of sensor nodes, effects of terrain, data-fusion algorithms, and threat characteristics.

STAMiNA is built upon a commercial network-simulator engine, with extensions to include both sensing and communication models in a discrete-event simulation environment. Users can define (1) a mission environment, including terrain features; (2) objects to be sensed; (3) placements and modalities of sensors, abilities of sensors to sense objects of various types, and sensor false-alarm rates; (4) trajectories of threatening objects; (5) means of dissemination and fusion of data; and (6) various network configurations. By use of STAMiNA, one can simulate detection of targets through sensing, dissemination of information by various wireless communication subsystems under various scenarios, and fusion of information, incorporating such metrics as target-detection probabilities, false-alarm rates, and communication loads, and capturing effects of terrain and threat.

This software is available for commercial licensing. Please contact Karina Edmonds of NASA’s Jet Propulsion Laboratory.

Refer to NPO-40560, volume and number of this NASA Tech Briefs issue, and the page number.
Using Transponders on the Moon to Increase Accuracy of GPS

Ranging to the Moon would be unaffected by the terrestrial atmosphere.

NASA’s Jet Propulsion Laboratory, Pasadena, California

It has been proposed to place laser or radio transponders at suitably chosen locations on the Moon to increase the accuracy achievable using the Global Positioning System (GPS) or other satellite-based positioning system. The accuracy of GPS position measurements depends on the accuracy of determination of the ephemerides of the GPS satellites. These ephemerides are determined by means of ranging to and from Earth-based stations and consistency checks among the satellites. Unfortunately, ranging to and from Earth is subject to errors caused by atmospheric effects, notably including unpredictable variations in refraction.

The proposal is based on exploitation of the fact that ranging between a GPS satellite and another object outside the atmosphere is not subject to error-inducing atmospheric effects. The Moon is such an object and is a convenient place for a ranging station. The ephemeris of the Moon is well known and, unlike a GPS satellite, the Moon is massive enough that its orbit is not measurably affected by the solar wind and solar radiation.

According to the proposal, each GPS satellite would repeatedly send a short laser or radio pulse toward the Moon and the transponder(s) would respond by sending back a pulse and delay information. The GPS satellite could then compute its distance from the known position(s) of the transponder(s) on the Moon.

Because the same hemisphere of the Moon faces the Earth continuously, any transponders placed there would remain continuously or nearly continuously accessible to GPS satellites, and so only a relatively small number of transponders would be needed to provide continuous coverage. Assuming that the transponders would depend on solar power, it would be desirable to use at least two transponders, placed at diametrically opposite points on the edges of the Moon disk as seen from Earth, so that all or most of the time, at least one of them would be in sunlight.

This work was done by Konstantin Penaenen and Talso Chui of Caltech for NASA’s Jet Propulsion Laboratory. For further information, contact ioffice@jpl.nasa.gov. NPO-43160

Controller for Driving a Piezoelectric Actuator at Resonance

Unpredictable variations in resonance frequency are tracked.

NASA’s Jet Propulsion Laboratory, Pasadena, California

A digital control system based partly on an extremum-seeking control algorithm tracks the changing resonance frequency of a piezoelectric actuator or an electrically similar electromechanical device that is driven by a sinusoidal excitation signal and is required to be maintained at or near resonance in the presence of uncertain, changing external loads and disturbances. Somewhat more specifically, on the basis of measurements of the performance of the actuator, this system repeatedly estimates the resonance frequency and alters the excitation frequency as needed to keep it at or near the resonance frequency. In the original application for which this controller was developed, the piezoelectric actuator is part of an ultrasonic/sonic drill/corer. Going beyond this application, the underlying principles of design and operation are generally applicable to tracking changing resonance frequencies of heavily perturbed harmonic oscillators.

Resonance-frequency-tracking analog electronic circuits are commercially available, but are not adequate for the present purpose for several reasons:

• The input/output characteristics of analog circuits tend to drift, often necessitating recalibration, especially whenever the same controller is used in driving a different resonator.
• In the case of an actuator in a system that has multiple modes characterized by different resonance frequencies, an analog controller can tune erroneously to one of the higher-frequency modes.
• The lack of programmability of analog controllers is problematic when faults occur, and is especially problematic for preventing tuning to a higher-frequency mode.

In contrast, a digital controller can be programmed to restrict itself to a specified frequency range and to maintain stability even when the affected resonator is driven at high power and subjected to uncertain disturbances and variable loads.

The present digital control system (see figure) is implemented by means of an algorithm that comprises three main subalgorithms: a hill-climbing control algorithm, an estimation-based extremum-seeking control (ESC) algorithm, and a supervisory algorithm. The hill-climbing algorithm is useful for coarse tracking to find and remain within the vicinity of the resonance. The ESC algorithm is not capable of coarse resonance tracking, but is capable of fine resonance tracking once the estimates of parameters generated by the hill-climbing algorithm have converged sufficiently. On the basis of the parameter-estimation errors, the supervisory algorithm switches operation to whichever of the other two algorithms performs best at a given time.
For the purpose of the control algorithm, the performance of the actuator is quantified in terms of the ratio between the time-averaged drive-voltage amplitude and the time-averaged drive current amplitude during a sampling time period. In the hill-climbing algorithm, the excitation frequency during the next sampling period is incremented or decremented by an arbitrary fixed step. If the increment or decrement results in an increase in the current/voltage ratio, then the direction of change (increase or decrease, respectively) of frequency is accepted and another such change (increment or decrement, respectively) is made during the following sampling period. If, on the other hand, the increment or decrement results in a decrease in the current/voltage ratio, then the direction of change of frequency during the following sampling period is reversed. The process as described thus far is repeated, causing the current/voltage performance to climb to one of the resonance peaks and eventually to oscillate about the peak. In order to prevent climbing of one of the undesired higher-frequency resonance peaks, it is necessary to choose the starting excitation frequency near the desired peak and to impose a limit on the excursion from the starting frequency.

Once the excitation frequency has begun to oscillate about the peak, the supervisory algorithm switches operation to the ESC algorithm, which uses past as well as present input/output data to make a least-squares estimate of the resonance frequency. The estimation task involves updating two scalar parameters of a quadratic model that represents the input/output map of the actuator resonance. After each sampling period, the new input/output data pair is added to the collection of past data pairs, such that information regarding the input/output relationship of the actuator increases over time; in other words, as the input/output information comes in, the algorithm tries to improve the fit between the quadratic model near resonance and all the past input/output data up to the current time. Once the estimated parameters have converged sufficiently, the excitation frequency is updated according to a simple formula that represents a maximizer associated with the quadratic model. In the event that the estimates begin to diverge beyond a specified limit, the supervisory algorithm switches operation back to the hill-climbing algorithm.

This work was done by Jack Aldrich, Yeoseph Bar-Cohen, Stewart Sherrit, Mircea Badescu, Xiaoqi Bao, and Zensheu Chang of Caltech for NASA’s Jet Propulsion Laboratory. Further information is contained in a TSP (see page 1). NPO-43519

---

**Coaxial Electric Heaters**

These devices can be used safely where magnetic fields are not tolerated.

*NASA’s Jet Propulsion Laboratory, Pasadena, California*

Coaxial electric heaters have been conceived for use in highly sensitive instruments in which there are requirements for compact heaters but stray magnetic fields would adversely affect operation. Such instruments include atomic clocks and magnetometers that utilize heated atomic-sample cells, wherein stray magnetic fields at picotesla levels could introduce systematic errors into instrument readings.

A coaxial electric heater (see Figure 1) is essentially an axisymmetric coaxial cable, the outer conductor of which is deliberately made highly electrically resistive so that it can serve as a heating element. As in the cases of other axisymmetric coaxial cables, the equal-magnitude electric currents flowing in opposite directions along the inner and outer conductors give rise to zero net magnetic field outside the outer conductor. Hence, a coaxial electric heater can be placed near an atomic-sample cell or other sensitive device.

A coaxial electric heater can be fabricated from an insulated copper wire, the copper core of which serves as the inner conductor. For example, in one ap-
A regio-regular poly(3-hexylthiophene) (RRP3HT) thin-film transistor having a split-gate architecture has been fabricated on a doped silicon/silicon nitride substrate and characterized. RRP3HT is a semiconducting polymer that has a carrier mobility and on/off ratio when used in a field effect transistor (FET) configuration. This commercially available polymer is very soluble in common organic solvents and is easily processed to form uniform thin films. The most important polymer-based device fabricated and studied is the FET, since it forms the building block in logic circuits and switches for active matrix (light-emitting-diode) (LED) displays, smart cards, and radio frequency identification (RFID) cards.

Figure 1(a) shows a schematic cross-sectional view of the basic FET using an insulating gate dielectric layer over a doped silicon substrate. Two metal leads patterned over the insulator serve as the source and drain terminals of the device while the doped silicon serves as the global gate electrode. In this basic configuration, the gate voltage is applied to the bottom side of the wafer, and a heater is placed above the gate dielectric layer over a doped silicon substrate. Two metal leads patterned over the insulator serve as the source and drain terminals of the device while the doped silicon serves as the global gate electrode. In this basic configuration, the gate voltage is applied to the bottom side of the wafer. To complete the FET, an organic semiconducting channel is placed between the source and drain terminals either via electrochemical deposition, vacuum deposition, or spin coating of the semiconducting material, resulting in a two-dimensional thin-film morphology; or via electrospinning, resulting in a one-dimensional nanofibrous morphology. In the dual-input architecture shown in Figure 1(b), metal contacts are located beneath the gate dielectric, and are accessed from the top side of the wafer by vias etched into the gate dielectric and semiconducting layers. Each gate contact serves as a device input, thus allowing for the design of various types of logic gates.

The dual-input device demonstrates AND logic functionality (see Figure 2),

Dual-Input AND Gate From Single-Channel Thin-Film FET

These transistors show potential as large-area, low-cost electronic circuitry on rigid and flexible substrates.

John H. Glenn Research Center, Cleveland, Ohio

The resistance of the graphite film can be tailored via its thickness. Alternatively, the film can be made from an electrically conductive paint, other than a colloidal graphite emulsion, chosen to impart the desired resistance. Yet another alternative is to tailor the resistance of a graphite film by exploiting the fact that its resistance can be changed permanently within about 10 percent by heating it to a temperature above 300 °C. Figure 2 depicts a coaxial heater, with electrical leads attached, that has been bent into an almost full circle for edge heating of a circular window. (In the specific application, there is a requirement for a heated cell window, through which an optical beam enters the cell.)

This work was done by Dmitry Strekalov, Andrey Matsko, Anatoly Savchenkov, and Lute Maleki of Caltech for NASA’s Jet Propulsion Laboratory. Further information is contained in a TSP (see page 1).

NPO-43569

Figure 1. A Typical Coaxial Electric Heater as shown in cross section resembles conventional signal-transmission coaxial cables, except that its outer conductor is deliberately made highly resistive.

Figure 1. This cross-sectional view of the Basic FET shows (a) the global gate FET and (b) split-gate FET.
and is controlled by applying either 0 or –10 volts to each of the gate electrodes. When –10 volts are simultaneously applied to both gates, the transistor is conductive (ON), while any other combination of gate voltages renders the transistor highly resistive (OFF). The p-type carrier charge mobility is about $5 \times 10^{-4} \text{cm}^2/\text{V-s}$. The low mobility is attributed to the sharp contours of the RRP3HT film between the drain and the source contacts, and to defects in the RRP3HT film itself.

The device substrates are fabricated with a starting wafer that is n-type doped Si (10 ohm-cm), with a 200-nm thick, thermally grown oxide layer. First, the gate metals, comprising 20-nm Cr/100-nm Au, are vacuum deposited in a thermal evaporator and patterned using conventional photolithographic and liftoff techniques. Next, a 100-nm-thick silicon nitride (Si$_3$N$_4$) film is deposited over this using chemical vapor deposition (CVD). Access to the gate metallization is obtained by etching windows into the silicon nitride. The source and drain metallization, comprising 20-nm Cr/100-nm Au, is deposited on the CVD-grown silicon nitride on either side of the buried split gates using conventional photolithographic and liftoff techniques. The electrode “fingers” are about 20 microns wide and 600 microns long. The spacing between the electrodes is approximately 4 microns.

The split-gate architecture for logic circuitry is demonstrated via a two-input logic AND circuit. To create the device, a 10-Megohm load resistor is connected between the ground and the transistor source terminals, with the two gate terminals serving as the inputs. The output ($V_D$) is taken at the source terminal across the load resistor. A low frequency (0.01 Hz) square-wave signal serves as the input gate bias. For all combinations of $V_{GSI}$ and $V_{GS2}$ except $V_{GSI} = V_{GS2} = –10 \text{ V}$, the transistor is in the resistive “OFF” state, and $–0.3 \text{ mV} < V_R < 0 \text{ V}$. For $V_{GSI} = V_{GS2} – 10 \text{ V}$, the transistor is in the more conductive “ON” state, causing a greater portion of the voltage drop to occur across the load resistor. As a result, $V_R$ is a more negative value ($–1.8 \text{ mV} < V_R < –1.7 \text{ mV}$).

When the device functions as an AND logic circuit, $V_{GSI}$ and $V_{GS2}$ are functions of time, while corresponding change in the output of voltage $V_D$ is a function of time for the four possible combinations of $V_{GSI}$ and $V_{GS2} = 0$ or –10 V. Larger outputs are observed only when both gates are simultaneously biased “high.”

This work was done by N. Theofylaktos and F.A. Miranda of Glenn Research Center; N.J. Pinto and R. Perez, of the University of Puerto Rico-Humacao; and C.H. Mueller of Analox Corporation. Further information is contained in a TSP (see page 1).

Inquiries concerning rights for the commercial use of this invention should be addressed to NASA Glenn Research Center, Innovative Partnerships Office, Attn: Steve Fedor, Mail Stop 4–8, 21000 Brookpark Road, Cleveland, Ohio 44135. Refer to LEW-18214-1.

---

**High-Density, High-Bandwidth, Multilevel Holographic Memory**

Multiple terabytes could be read or written at a multigigahertz rate.

*NASA’s Jet Propulsion Laboratory, Pasadena, California*

A proposed holographic memory system would be capable of storing data at unprecedentedly high density, and its data-transfer performance in both reading and writing would be characterized by exceptionally high bandwidth. The capabilities of the proposed system would greatly exceed even those of a state-of-the-art memory system, based on binary holograms (in which each pixel value represents 0 or 1), that can hold ~1 terabyte of data and can support a reading or writing rate as high as 1 Gb/s.

The storage capacity of the state-of-the-art system cannot be increased without also increasing the volume and mass of the system. However, in principle, the storage capacity could be increased greatly, without significantly increasing the volume and mass, if multilevel holograms were used instead of binary holograms. For example, a 3-bit (8-level) hologram could store 8 terabytes, or an 8-bit (256-level) hologram could store 256 terabytes, in a system having little or no more size and mass than does the state-of-the-art 1-terabyte binary holographic memory.

The proposed system would utilize multilevel holograms. The system (see figure) would include lasers, imaging lenses and other beam-forming optics, a block photorefractive crystal wherein the holograms would be formed, and two multilevel spatial light modulators in the form of commercially available deformable-mirror-device spatial light modulators (DMDSLMs) made for use.
in high-speed input conversion of data up to 12 bits. For readout, the system would also include two arrays of complementary metal oxide/semiconductor (CMOS) photodetectors matching the spatial light modulators. The system would further include a reference-beam-steering device (equivalent of a scanning mirror), containing no sliding parts, that could be either a liquid-crystal phased-array device or a microscopic mirror actuated by a high-speed micro-electromechanical system. Time-multiplexing and the multilevel nature of the DMDSLM would be exploited to enable writing and reading of multilevel holograms. The DMDSLM would also enable transfer of data at a rate of 7.6 Gb/s or perhaps somewhat higher.

This work was done by Tien-Hsin Chao of Caltech for NASA’s Jet Propulsion Laboratory. Further information is contained in a TSP (see page 1).

In accordance with Public Law 96-517, the contractor has elected to retain title to this invention. Inquiries concerning rights for its commercial use should be addressed to:

Innovative Technology Assets Management JPL
Mail Stop 202-233
4800 Oak Grove Drive
Pasadena, CA 91109-8099
(818) 354-2240
E-mail: iaoffice@jpl.nasa.gov
Refer to NPO-42702, volume and number of this NASA Tech Briefs issue, and the page number.

Multilevel Holograms would be written to, and read from, the photorefractive crystal in this holographic memory system.
Fabrication of Gate-Electrode Integrated Carbon-Nanotube Bundle Field Emitters

Emission tips and a gate electrode are integrated into a monolithic device.

NASA’s Jet Propulsion Laboratory, Pasadena, California

A continuing effort to develop carbon-nanotube-based field emitters (cold cathodes) as high-current-density electron sources has yielded an optimized device design and a fabrication scheme to implement the design. One major element of the device design is to use a planar array of bundles of carbon nanotubes as the field-emission tips and to optimize the critical dimensions of the array (principally, heights of bundles and distances between them) to obtain high area-averaged current density and high reliability over a long operational lifetime — a concept that was discussed in more detail in “Arrays of Bundles of Carbon Nanotubes as Field Emitters” (NPO-40817), NASA Tech Briefs, Vol. 31, No. 2 (February 2007), page 58. Another major element of the design is to configure the gate electrodes (anodes used to extract, accelerate, and/or focus electrons) as a ring that overhangs a recess wherein the bundles of nanotubes are located [see Figure 1(a)], such that by virtue of the proximity between the ring and the bundles, a relatively low applied potential suffices to generate the large electric field needed for emission of electrons.

A fabrication process to monolithically integrate multiple electrodes is as follows. The starting workpiece is a commercially available double silicon-on-insulator (SOI) wafer comprising, from top layer to the bottom, a silicon device layer (1), a silicon dioxide (buried oxide) layer (2), another silicon device layer (3) on top of a silicon dioxide (buried oxide) layer (4), and a thick silicon substrate (bottom) layer (5). A schematic representation of the starting substrate is shown in Figure 2(a). After lithographic patterning followed by a combination of wet etching, deep reactive-ion etching (DRIE) and isotropic silicon etch by xenon difluoride (XeF₂) gas, overhanging gate and undercutting recess profiles are defined in the substrate. The iron catalytic dots are prepared at the bottom surface of the hole and carbon nanotube bundles are grown from them. Figures 1(a) shows a 3-D sketch and (b) shows an SEM micrograph (SEM) images of fabricated field-emitter devices.

Figure 1. A Gate Electrode Overhangs a recess containing an array of bundles of carbon nanotubes (see part a). In part (b) are scanning electron micrograph (SEM) images of fabricated field-emitter devices.

Figure 2. These are Simplified Cross Sections (not to scale) of the device of Figure 1 at various stages of fabrication.
graph of a completed device. In a completed device, the layer (1) acts as the gate electrode and layer (3) provides the mechanical support to achieve overhanging. However, layer (3) can also be used as an additional electrode for beam tailoring if required. Similarly, by using multiple-silicon-layer SOI substrates, monolithic integration of multiple electrodes can be achieved.

This work was done by Risaku Toda, Michael Bronikowski, Edward Luong, and Harish Manohara of Caltech for NASA’s Jet Propulsion Laboratory.

In accordance with Public Law 96-517, the contractor has elected to retain title to this invention. Inquiries concerning rights for its commercial use should be addressed to:

Innovative Technology Assets Management
JPL
Mail Stop 202-233
4800 Oak Grove Drive
Pasadena, CA 91109-8099
E-mail: iaoffice@jpl.nasa.gov
Refer to NPO-44996, volume and number of this NASA Tech Briefs issue, and the page number.
**Hydroxide-Assisted Bonding of Ultra-Low-Expansion Glass**

Preparation of bond surfaces is critical to success.

*NASA’s Jet Propulsion Laboratory, Pasadena, California*

A process for hydroxide-assisted bonding has been developed as a means of joining optical components made of ultra-low-expansion (ULE) glass, while maintaining sufficiently precise alignment between. The process is intended mainly for use in applications in which (1) bonding of glass optical components by use of epoxy does not enable attainment of the required accuracy and dimensional stability and (2) conventional optical contacting (which affords the required accuracy and stability) does not afford adequate bond strength.

The basic concept of hydroxide-assisted bonding is not new. The development of the present process was prompted by two considerations: (1) The expertise in hydroxide-assisted bonding has resided in very few places and the experts have not been willing to reveal the details of their processes and (2) data on the reliability and strength attainable by hydroxide-assisted bonding have been scarce.

The first and most critical phase of the present hydroxide-assisted-bonding process is the preparation of the surfaces to be bonded. This phase includes the following steps:

1. Ultrasonic cleaning in successive baths of acetone, methanol, and propanol, using an ultrasound cleaner that operates at several Megahertz (Megasonics).
2. Treatment in a solution of potassium hydroxide and ammonium hydroxide in an ultrasonic cleaner, at Megahertz frequencies.
3. Thorough rinsing with deionized water is carried out after each of the above-mentioned steps. The last rinse is followed by ultrasonic cleaning in deionized water, then the cleaned surfaces are blow-dried with ionized air.
4. After preparation of the surfaces as described above, a droplet of a dilute solution of potassium hydroxide is placed on one of the surfaces, then the surfaces are placed in contact and gently squeezed together (see figure). The resulting assembly is allowed to sit at room temperature for 24 hours, and is then baked at a temperature of 200 °C for 24 hours.

In mechanical tests, sample bonds made by this process were found to have tensile strengths of at least 1.3 kpsi (≈9 MPa), where the epoxy bond used to attach the sample to the tensile stress test apparatus broke.

This work was done by Alexander Abramovici and Victor White of Caltech for NASA’s Jet Propulsion Laboratory. For more information, contact iaoffice@jpl.nasa.gov. NPO-45247

---

**Photochemically Synthesized Polyimides**

Single monomers are polymerized by exposure to ultraviolet light, without heating.

*John H. Glenn Research Center, Cleveland, Ohio*

An alternative to the conventional approach to synthesis of polyimides involves the use of single monomers that are amenable to photopolymerization. Heretofore, the synthesis of polyimides has involved multiple-monomer formulations and heating to temperatures that often exceed 250 °C. The present alternative approach enables synthesis under relatively mild conditions that can include room temperature.

The main disadvantages of the conventional approach are the following:• Elevated production temperatures can lead to high production costs and can impart thermal stresses to the final products.
• If the proportions of the multiple monomeric ingredients in a given batch are not exactly correct, the molecular weight and other physical properties of the final material could be reduced from their optimum or desired values.

To be useful in the alternative approach, a monomer must have a molecular structure tailored to exploit Diels-Alder trapping of a photochemically generated ortho-quinodimethane. (In a Diels-Alder reaction, a diene combines with a dienophile to form molecules that contain six-membered rings.) In particular, a suitable monomer (see figure) contains ortho-methylbenzophenone con-
Optimized Carbonate and Ester-Based Li-Ion Electrolytes

This technology can be used in portable electronics, cell phones, and electric vehicles.

NASA’s Jet Propulsion Laboratory, Pasadena, California

To maintain high conductivity in low temperatures, electrolyte co-solvents have been designed to have a high dielectric constant, low viscosity, adequate coordination behavior, and appropriate liquid ranges and salt solubilities. Electrolytes that contain ester-based co-solvents in large proportion (>50 percent) and ethylene carbonate (EC) in small proportion (<20 percent) improve low-temperature performance in MCMB carbon-LiNiCoO₂ lithium-ion cells. These co-solvents have been demonstrated to enhance performance, especially at temperatures down to −70 °C. Low-viscosity, ester-based co-solvents were incorporated into multi-component electrolytes of the following composition: 1.0 M LiPF₆ in ethylene carbonate (EC) + ethyl methyl carbonate (EMC) + X (1:1:8 volume percent) [where X = methyl butyrate (MB), ethyl butyrate EB, methyl propionate (MP), or ethyl valerate (EV)].

These electrolyte formulations result in improved low-temperature performance of lithium-ion cells, with dramatic results at temperatures below −40 °C. [See “Ester-Based Electrolytes for Low-Temperature Li-Ion Cells,” (NPO-41097) NASA Tech Briefs, Vol 29, No. 12 (December, 2005), p. 59.]

Improved low-temperature performance can also be realized with ester-based electrolytes containing high salt concentrations and by using mixed salt formulations — specifically with (a) 1.0 M LiPF₆ + 0.40 LiBF₄ and (b) 1.40 M LiPF₆ dissolved in EC+EMC+MP (1:1:8 volume percent) and EC+EMC+MB (1:1:8 volume percent) solvent mixtures. The rate capability has been observed to increase dramatically at low temperatures (i.e., −60 °C) using this approach. It is anticipated that increased salt concentrations and the use of mixed salt systems will also improve the low-temperature performance characteristics of other solvent blends of carbonates and esters. [“Mixed-Salt/Ester Electrolytes for Low-Temperature Li + Cells” (NPO-42862), NASA Tech Briefs, Vol. 30, No. 11 (November 2006), p. 66.]

A number of these electrolytes have been demonstrated in both experimental and aerospace-quality, high-capacity prototype cells. In more recent work, these ester-containing electrolytes have been further optimized to provide both good low-temperature performance (down to −60 °C) while still offering reasonable high-temperature resilience. This has primarily been achieved by fixing the EC-content at 20 percent and the ester co-solvent at 20 percent, in contrast to the previously developed ultra-low temperature systems, which have the EC-content and ester-content at 10 percent and 80 percent, respectively. Using this approach, a prototype cell containing a 1.0 M LiPF₆ EC+EMC+MP (20:60:20 volume percent) electrolyte was capable of delivering over six times the amount of capacity delivered by the baseline ternary, all-carbonate blend, and was able to support reasonably aggressive rates at low temperature (−50 and −60 °C). Cells containing other esters also performed well at low temperature, with the lower-molecular-weight, lower-viscosity co-solvents generally yielding better performance at low temperatures.

Although slightly less favorable in terms of electrolyte conductivity, the higher-molecular-weight esters [i.e., propyl butyrate (PB), and butyl butyrate (BB)] are expected to result in cells with more favorable high-temperature resilience (>40 °C), compared to the lower-molecular-weight esters.

This technology has relevance for manned and unmanned space missions, aircraft batteries, and Land Warrior applications, as well as for terrestrial applica-
tions that require good performance and a high level of safety over a range of temperatures, including portable electronics like camcorders, cellular phones, laptop computers, and radio communication sets. Electric vehicle applications can also use this innovation where high-power batteries must operate at low temperatures, such as in monitoring stations in Antarctica.

This work was done by Marshall Smart and Ratnakumar Bugga of Caltech for NASA’s Jet Propulsion Laboratory. Further information is contained in a TSP (see page 1).

In accordance with Public Law 96-517, the contractor has elected to retain title to this invention. Inquiries concerning rights for its commercial use should be addressed to:

Innovative Technology Assets Management
JPL
Mail Stop 202-233
4800 Oak Grove Drive
Pasadena, CA 91109-8099
E-mail: iaooffice@jpl.nasa.gov

Refer to NPO-44974, volume and number of this NASA Tech Briefs issue, and the page number.
Compact 6-DOF Stage for Optical Adjustments
Adjustments can be made in all translational and rotational degrees of freedom.

NASA’s Jet Propulsion Laboratory, Pasadena, California

The figure depicts selected aspects of a six-degree-of-freedom (6-DOF) stage for mechanical adjustment of an optical component. The six degrees of freedom are translations along the Cartesian axes (x, y, and z) and rotations about these axes (θx, θy, and θz, respectively). Relative to prior such stages, this stage offers advantages of compactness, stability, and robustness, plus other advantages as described below.

The stage was designed specifically as part of a laser velocimeter and altimeter in which light reflected by a distant object is collected by a Cassegrainian telescope and focused into a single-mode, polarization-maintaining optical fiber. The stage is used to position and orient the input end of the optical fiber with respect to the focal point of the telescope. Stages like this one can also be adapted for use in positioning and orienting other optical components, including lenses, prisms, apertures, and photodetectors.

The optical fiber or other optical component is mounted in a ferrule that is, in turn, mounted in a ferrule holder that is an extension of the ball part of a ball-and-socket assembly that enables adjustment in all three rotational degrees of freedom. The position of the ferrule within the ferrule holder is set so that the center of the input face of the optical component lies at the center of the ball. As a result of this setting, rotational adjustment is not accompanied by undesired translational adjustment.

The subassembly comprising the ball, ferrule holder, and optical component is spring-loaded into the socket, and the spring load can be adjusted by means of a threaded ball-preload adjuster. The ferrule holder and the ball-preload adjuster are equipped with external surfaces that mate with special-purpose adjustment tools. The spring load is chosen to make the frictional torque between the ball and the socket small enough that rotational adjustments can be made, yet large enough that the ball and socket retain their relative angular position once the angular adjustment.

This Compact Assembly enables stable adjustment of the input end of the optical fiber in all three translational and all three rotational degrees of freedom. Rotational adjustments can be made without causing undesired translations, and translational adjustments can be made without causing undesired rotations.
has been completed and the rotational-adjustment tools removed.

Optionally, the ball-and-socket assembly as described thus far could be used alone as a rotation-only stage. However, in the original application, the ball-and-socket assembly is mounted within a z-axis housing that, as its name suggests, enables translational adjustment along the z-axis (focus adjustment). The socket is in threaded engagement with a focus-adjustment nut that can be turned about the z-axis to make the adjustment. An anti-rotation pin that is free to translate along a z-oriented slot prevents undesired rotation of the socket about the z-axis during focus adjustment. A focus-preload spring exerts a z-axis preload between the socket and the z-axis housing to prevent backlash in the focus adjustment.

Optionally, the z-axis-adjusting mechanism as described above could be used alone as a z-axis-translation stage. However, in the original application, it is mounted in an x-y translation stage that includes three flexural arms positioned at equal angular intervals on a circular frame. The radial position of the outer end of each flexural arm can be varied by means of a fine-pitch adjustment screw. Initially, all three adjustment screws are set at approximately the midpoints of their ranges, thereby placing all three flexural arms in tension and approximately centering the z-axis housing in the circle. Thereafter, the screws are turned, singly or in pairs as needed, to make fine adjustments to bring the optical component into x and y alignment. Care must be taken during these adjustments to maintain all three flexural arms in tension so as to prevent backslash. The x-y adjustment resolution is much finer than the thread pitch of the adjustment screws. Optionally, like the rotational and z-axis sub-stages, the x-y stage could be used by itself.

This work was done by Syed Shafaat and Daniel Chang of Caltech for NASA’s Jet Propulsion Laboratory. Further information is contained in a TSP (see page 1), NPO-45273

Ultrasonic/Sonic Impacting Penetrators

**Soil can be probed relatively gently to a depth of several feet.**

*NASA’s Jet Propulsion Laboratory, Pasadena, California*

Ultrasonic/sonic impacting penetrators (USIPs) are recent additions to the series of apparatuses based on ultrasonic/sonic drill corers (USDCs). A USIP enables a rod probe to penetrate packed soil or another substance of similar consistency, without need to apply a large axial force that could result in buckling of the probe or in damage to some buried objects. USIPs were conceived for use in probing and analyzing soil to depths of tens of centimeters in the vicinity of buried barrels containing toxic waste, without causing rupture of the barrels. USIPs could also be used for other purposes, including, for example, searching for pipes, barrels, or other hard objects buried in soil; and detecting land mines.

USDCs and other apparatuses based on USDCs have been described in numerous previous NASA Tech Briefs articles. The ones reported previously were designed, variously, for boring into, and/or acquiring samples of, rock or other hard, brittle materials of geological interest. To recapitulate: A USDC can be characterized as a lightweight, low-power, piezoelectrically driven jackhammer in which ultrasonic and sonic vibrations are generated and coupled to a tool bit. As shown in the figure, a basic USDC includes a piezoelectric stack, a backing and a horn connected to the stack, a free mass (“free” in the sense that it can slide axially a short distance between the horn and the shoulder of tool bit), and a tool bit, i.e., probe for USIP. The piezoelectric stack is driven at the resonance frequency of the stack/horn/backing assembly to create ultrasonic vibrations that are mechanically amplified by the horn. To prevent fracture during operation, the piezoelectric stack is held in compression by a bolt. The bouncing of the free mass between the horn and the tool bit at sonic frequencies generates hammering actions to the bit that are more effective for drilling than is the microhammering action of ultrasonic vibrations in ordinary ultrasonic drills. The hammering actions are so effective that the axial force needed to make the tool bit advance into the material of interest is much smaller than in ordinary twist drilling, ultrasonic drilling, or ordinary steady pushing.

The differences between a USIP and a USDC-based apparatus described above lie in design details that make a USIP more suitable for penetrating packed soil. The piezoelectric stack in an experimental prototype USIP had a diameter of 1.0 in. (=25 mm) and could be made to resonate at a frequency between 12 and 20 kHz, the exact value depending on the

The design of Prototype USIP shows key components.
specific design and operating conditions. The probe rod had a diameter of 1/8 in. (≈3 mm) and a length sufficient to enable penetration to a depth of 3 ft (≈91 cm). The piezoelectric stack was driven at a 20-percent duty cycle, with a combination of automatic and manual adjustments of the frequency of the driving signal to compensate for changes in the resonance frequency induced by changes in mechanical loading and by temperature rise during operation.

The design of the horn and a piezoelectric-stack-backing structure was optimized for coupling power from the stack to the horn and for amplification of the longitudinal displacement. The optimization was accomplished with the help of a computer program that numerically solved the governing equations to perform impact and vibration-mode analyses. The modal analysis was used to determine the dimensions of the horn and backing for a resonance frequency in the required range and to further adjust the dimensions of the horn so that the neutral plane matched the mounting plane to minimize adverse effects of transducer vibration on a supporting structure. The impact analysis, in which the focus was on the interaction between the free mass and the horn, was used to derive an optimal weight of the free mass.

In experiments, an axial force of 7 lb (≈31 N) was found to be sufficient to cause the probe tip to reach a depth of 3 ft (≈91 cm) in a packed soil sample. In contrast, the axial force that would be needed to make an equivalent probe tip penetrate to the same depth by ordinary steady pushing has been estimated to be about 200 lb (≈890 N), which is large enough to easily cause buckling of the probe without a holding mechanism and to damage a buried barrel.

This work was done by Xiaoqi Bao, Yoseph Bar-Cohen, Zensheu Chang, Stewart Sherrit, and Randall A. Stark of Caltech for NASA’s Jet Propulsion Laboratory. Further information is contained in a TSP (see page 1), NPO-41666.

---

**Miniature, Lightweight, One-Time-Opening Valve**

**A small solder plug is melted to release a pressurized gas.**

*NASA's Jet Propulsion Laboratory, Pasadena, California*

The figure depicts the main parts of a prototype miniature, lightweight, one-time-opening valve. Like some other miniature one-time-opening valves reported in previous issues of NASA Tech Briefs, this valve is opened by melting a material that blocks the flow path. This valve is designed to remain closed at some temperature between room temperature and cryogenic temperature until the time of opening.

The prototype valve includes a 1/8-in. (3-mm) aluminum tube, one end of which is plugged with a solder comprising about 37 weight percent of lead and 63 weight percent of tin. The tube and the solder both have a coefficient of thermal expansion of 23 micron/m-K at room temperature. Before plugging, the interior surface of the plug end of the tube is cleaned with a commercial flux paste developed specifically for preparing aluminum for bonding with lead/tin solder. The solder is then melted into the cleaned end of the tube, forming the plug.

In a test, the plugged tube was pressurized to 1,000 psi (6.9 MPa) with helium and leak-tested. It was then cooled to a temperature of 77 K (about –196 °C) and again leak-tested at the same pressure. Finally, at a lower pressure, the plugged end of the tube was heated to about 200 °C (the melting temperature of the solder is 183°C), causing the solder plug to be ejected (see figure). It has been estimated that in a subsequent version of the valve, the plug could be melted by electrical heating, using a nichrome wire having a mass of only 10 g.

This work was done by Jack Jones, Juinn Jenq Wu, and Robert Leland of Caltech for NASA’s Jet Propulsion Laboratory. Further information is contained in a TSP (see page 1), NPO-42236.
Supplier Management System

Supplier Management System (SMS) allows for a consistent, agency-wide performance rating system for suppliers used by NASA. This version (2.0) combines separate databases into one central database that allows for the sharing of supplier data. Information extracted from the NBS/Oracle database can be used to generate ratings. Also, supplier ratings can now be generated in the areas of cost, product quality, delivery, and audit data. Supplier data can be charted based on real-time user input. Based on these individual ratings, an overall rating can be generated.

Data that normally would be stored in multiple databases, each requiring its own log-in, is now readily available and easily accessible with only one log-in required. Additionally, the database can accommodate the storage and display of quality-related data that can be analyzed and used in the supplier procurement decision-making process. Moreover, the software allows for a Closed-Loop System (supplier feedback), as well as the capability to communicate with other federal agencies.

While Version 1.0 only had an Approved Supplier list, Version 2.0 includes the Approved Supplier List, Supplier Rating System, and the Supplier Outreach and Process Control Assurance System. The Supplier Rating System is a new database that was developed specifically for this software. The system allows for segregation of data to ensure sensitive data is protected.

This program was written by Eric Ramirez, Sandy Gutheinz, James Brison, Anita Ho, James Allen, Olga Ceritelli, Claudia Tobar, Thuykien Nguyen, and Harrel Crenshaw of Caltech and Roxann Santos of SkillStorm Inc. for NASA’s Jet Propulsion Laboratory.

This software is available for commercial licensing. Please contact Karina Edmonds of the California Institute of Technology at (626) 395-2322. Refer to NPO-43786.

JAVA Stereo Display Toolkit

This toolkit provides a common interface for displaying graphical user interface (GUI) components in stereo using either specialized stereo display hardware (e.g., liquid crystal shutter or polarized glasses) or anaglyph display (red/blue glasses) on standard workstation displays. An application using this toolkit will work without modification in either environment, allowing stereo software to reach a wider audience without sacrificing high-quality display on dedicated hardware.

The toolkit is written in Java for use with the Swing GUI Toolkit and has cross-platform compatibility. It hooks into the graphics system, allowing any standard Swing component to be displayed in stereo. It uses the OpenGL graphics library to control the stereo hardware and to perform the rendering. It also supports anaglyph and special stereo hardware using the same API (application-program interface), and has the ability to “simulate” color stereo in anaglyph mode by combining the red band of the left image with the green/blue bands of the right image.

This is a low-level toolkit that accomplishes simply the display of components (including the JadeDisplay image display component). It does not include higher-level functions such as disparity adjustment, 3D cursor, or overlays — all of which can be built using this toolkit.

This program was written by Robert Deen and Oleg Pariser of Caltech for NASA’s Jet Propulsion Laboratory.

The software used in this innovation is available for commercial licensing. Please contact Karina Edmonds of the California Institute of Technology at (626) 395-2322. Refer to NPO-44571.

Remote-Sensing Time Series Analysis, a Vegetation Monitoring Tool

The Time Series Product Tool (TSPT) is software, developed in MATLAB®, which creates and displays high signal-to-noise Vegetation Indices imagery and other higher-level products derived from remotely sensed data. This tool enables automated, rapid, large-scale regional surveillance of crops, forests, and other vegetation. TSPT temporally processes high-revisit-rate satellite imagery produced by the Moderate Resolution Imaging Spectroradiometer (MODIS) and by other remote-sensing systems. Although MODIS imagery is acquired daily, cloudiness and other sources of noise can greatly reduce the effective temporal resolution. To improve cloud statistics, the TSPT combines MODIS data from multiple satellites (Aqua and Terra). The TSPT produces MODIS products as single time-frame and multitemporal change images, as time-series plots at a selected
location, or as temporally processed image videos. Using the TSPT program, MODIS metadata is used to remove and/or correct bad and suspect data. Bad pixel removal, multiple satellite data fusion, and temporal processing techniques create high-quality plots and animated image video sequences that depict changes in vegetation greenness. This tool provides several temporal processing options not found in other comparable imaging software tools. Because the framework to generate and use other algorithms is established, small modifications to this tool will enable the use of a large range of remotely sensed data types.

An effective remote-sensing crop-monitoring system must be able to detect subtle changes in plant health in the earliest stages, before the effects of a disease outbreak or other adverse environmental conditions can become widespread and devastating. The integration of the time series analysis tool with ground-based information, soil types, crop types, meteorological data, and crop growth models in a Geographic Information System, could provide the foundation for a large-area crop-surveillance system that could identify a variety of plant phenomena and improve monitoring capabilities.

This program was written by Rodney McKellip of Stennis Space Center; Donald Prados of Computer Sciences Corporation; Robert Ryan, Kenton Ross, and Joseph Spruce of Science Systems and Applications; and Gerald Gasser and Randall Greer of Lockheed Martin Space Operations, Information Systems Directorate.

Inquiries concerning rights for the commercial use of this invention should be addressed to: Michael Kipphut, GeoResources Institute at Mississippi State University for Stennis Space Center. (228) 688-1929. Refer to SSC-00261, volume, and number of this NASA Tech Briefs issue and the page number.

PyPele Rewritten To Use MPI

A computer program known as “PyPele,” originally written as a Python-language extension module of a C++ language program, has been rewritten in pure Python language. The original version of PyPele dispatches and coordinates parallel-processing tasks on cluster computers and provides a conceptual framework for spacecraft-mission-design and analysis software tools to run in an embarrassingly parallel mode. The original version of PyPele uses SSH (Secure Shell — a set of standards and an associated network protocol for establishing a secure channel between a local and a remote computer) to coordinate parallel processing. Instead of SSH, the present Python version of PyPele uses Message Passing Interface (MPI) [an unofficial de-facto standard language-independent application programming interface for message-passing on a parallel computer] while keeping the same user interface.

The use of MPI instead of SSH and the preservation of the original PyPele user interface make it possible for parallel application programs written previously for the original version of PyPele to run on MPI-based cluster computers. As a result, engineers using the previously written application programs can take advantage of embarrassing parallelism without need to rewrite those programs.

This program was written by George Hockney and Seungwon Lee of Caltech for NASA’s Jet Propulsion Laboratory.

This software is available for commercial licensing. Please contact Karina Edmonds of the California Institute of Technology at (626) 395-2322. Refer to NPO-44729.

Data Assimilation Cycling for Weather Analysis

This software package runs the atmospheric model MM5 in data assimilation cycling mode to produce an optimized weather analysis, including the ability to insert or adjust a hurricane vortex. The program runs MM5 through a cycle of short forecasts every three hours where the vortex is adjusted to match the observed hurricane location and storm intensity. This technique adjusts the surrounding environment so that the proper steering current and environmental shear are achieved. MM5cycle uses a Cressman analysis to blend observation into model fields to get a more accurate weather analysis. Quality control of observations is also done in every cycle to remove bad data that may contaminate the analysis. This technique can assimilate and propagate data in time from intermittent and infrequent observations while maintaining the atmospheric field in a dynamically balanced state.

The software consists of a C-shell script (MM5cycle.driver) and three FORTRAN programs (splitMM5files.F, comRegrid.F, and insert_vortex.F), and are contained in the pre-processor component of MM5 called “Regridder.” The model is first initialized with data from a global model such as the Global Forecast System (GFS), which also provides lateral boundary conditions. These data are separated into single-time files using splitMM5.F. The hurricane vortex is then bogussed in the correct location and with the correct wind field using insert_vortex.F. The modified initial and boundary conditions are then recombined into the model fields using comRegrid.F. The model then makes a three-hour forecast. The three-hour forecast data from MM5 now become the analysis for the next short forecast run, where the vortex will again be adjusted. The process repeats itself until the desired time of analysis is achieved. This code can also assimilate observations if desired.

This program was written by Nam Tran, Yongzao Li, and Patrick Fitzpatrick of the GeoResources Institute at Mississippi State University for Stennis Space Center.

Inquiries concerning rights for its commercial use should be addressed to:

GeoResources Institute
Mississippi State University
Building 1103, Room 233
Stennis Space Center, MS 39529
Phone No.: (228) 688-4218
Web site: www.gri.msstate.edu
Refer to SSC-00276/7, the volume and number of this NASA Tech Briefs issue, and the page number.
Hydrocyclone/Filter for Concentrating Biomarkers From Soil

This apparatus could detect biomarkers or chemical waste in soil.

NASA’s Jet Propulsion Laboratory, Pasadena, California

The hydrocyclone-filtration extractor (HFE), now undergoing development, is a simple, robust apparatus for processing large amounts of soil to extract trace amounts of microorganisms, soluble organic compounds, and other biomarkers from soil and to concentrate the extracts in amounts sufficient to enable such traditional assays as cell culturing, deoxyribonucleic acid (DNA) analysis, and isotope analysis. Originally intended for incorporation into a suite of instruments for detecting signs of life on Mars, the HFE could also be used on Earth for similar purposes, including detecting trace amounts of biomarkers or chemical wastes in soils.

In addition to a conical separator vessel typical of a hydrocyclone, the HFE includes a pump, a sample container, an electropositive nanoparticle filter, and associated plumbing (see figure). The hydrocyclone serves to separate both cellular-sized particles (<5 µm) and dissolved organic compounds from the bulk soil (consisting mostly of particles larger than 5 µm). The electropositive filter serves to capture cellular biomass and remove particles from the extract with maximal capacity (that is, minimal clogging).

The soil sample is prepared by mixing it with sterile water or another suitable solvent to form a slurry, which is pumped to the hydrocyclone separator vessel in tangential streams at the top (wide) end to form two concentric opposing vortices at high hydrodynamic pressures to effect separation based on particle size. The spinning motion of the injected slurry is accelerated by the conical taper, creating a large centrifugal force that causes the larger particles to rapidly separate from the rest of the flow and leave the vessel as part of the underflow through the bottom (narrow) end of the separator vessel. The liquid containing suspended particles smaller than 5 µm leaves the vessel as the overflow (which is larger than the underflow) at the top (wide) end of the vessel. The overflow is then fed through the filter.

The filter contains electropositive aluminum nanotubes embedded in a glass-fiber matrix. The filter has a nominal pore size of 2 µm, but can collect biological particles having sizes down to fractions of a micron, on the basis of electrostatic charge, without clogging. The combination of the hydrocyclone and the electropositive nanoparticle filter is capable of extracting 99 percent of particles smaller than 5 µm from the soil sample and retaining them for analysis. In the prototype HFE depicted schematically in the filter, the underflow and the effluent from the filter are fed back to the sample container. It is envisioned that the fully developed HFE could operate in a continuous-flow mode, making it possible to extract biomarkers from a large volume (e.g., 1 m³) of soil using a minimal amount (e.g., 1 L) of solvent.

In a field test, the prototype HFE was made to process a slurry consisting of 4 L of sterile water mixed with 4 L of Atacama-desert soil, which contains such low concentrations of recoverable microorganisms as to be nearly impossible to analyze in the absence of an extraction-and-concentration process. After 15 minutes of recycling the slurry in the HFE, the maximum loading of the filter was reached. The filter was removed and dried and found to contain 100 g of fine particulate material, which, in turn, was removed from the filter in a laboratory for further analysis. Whereas traditional extraction protocols yielded no microbial colony-forming units, the extract from the HFE filter yielded microbes too numerous to count on standard growth plates. On the basis of development plans and this result, it is expected that...
Activating STAT3 Alpha for Promoting Healing of Neurons

Natural anti-apoptotic, pro-axogenic mechanisms are stimulated artificially.

Ames Research Center, Moffett Field, California

A method of promoting healing of injured or diseased neurons involves pharmacological activation of the STAT3 alpha protein. Usually, injured or diseased neurons heal incompletely or not at all for two reasons: (1) they are susceptible to apoptosis (cell death); and (2) they fail to engage in axogenesis — that is, they fail to re-extend their axons to their original targets (e.g., muscles or other neurons) because of insufficiency of compounds, denoted neurotrophic factors, needed to stimulate such extension. The present method (see figure) of treatment takes advantage of prior research findings to the effect that the STAT3 alpha protein has anti-apoptotic and pro-axogenic properties.

As used here, “STAT” signifies “signal transducer and activator of transcription.” “STAT3 alpha” is the name of one of a number of transcription factors and of the gene responsible for producing it. Transcription factors activate the expression of other genes and otherwise generally regulate gene expression. The STAT3 alpha protein is activated in response to such extracellular factors as hormones and growth factors as well as the aforementioned neurotrophic factors. The STAT3 alpha protein associates with trans-membrane receptors for these extracellular factors. When activated by growth factors and hormones, STAT3 alpha protein has anti-apoptotic and pro-axogenic properties.

Recent findings from the Life Sciences and Microgravity Division of Ames Research Center now show that when activated by neurotrophic factors STAT3 alpha, in addition to preventing apoptosis, also promotes innervation. The natural activation of the STAT3 alpha protein is effected by phosphorylation of a specific tyrosine amino acid. Upon such phosphorylation, two STAT3 alpha molecules can form a homo-dimer. The tyrosine on a given STAT3 protein molecule that becomes phosphorylated resides on a flexible activation loop sequence that binds to a pocket, denoted the SH2 domain, of another STAT3 protein molecule. Upon homo-dimerization, STAT3 alpha is retained in the nucleus of cells, where it binds specific deoxyribonucleic acid (DNA) sequences and activates the expression of nearby genes. Thus, homo-dimerized STAT3 alpha protein is part of a signal-transduction complex that relays chemical signals from hormones, growth factors, and neurotrophic factors outside the cell directly to the DNA inside the cell. In so doing, it activates genes that exert anti-apoptotic and axogenic effects.

Hence, the present method is based on the design of a new class of pharmacological agents (homo and heteroditopic Janus molecules) to promote artificial dimerization and pharmacological activation of therapeutic targets, in this case the STAT3 alpha protein. The precise location of each atom in the STAT3 homodimer is known and it is possible to use this structural information along with molecular modeling and docking programs to rationally design a Janus molecule having the correct sizes and shape, and/or to bind to STAT3 alpha preferentially to other proteins that are similar to STAT3 alpha. The compound can be synthesized by techniques that are well established in the pharmaceutical industry and can be formulated to be administered alone or in combination with other therapeutic compounds.

This invention is owned by NASA, and a patent application has been filed. Inquiries concerning rights for the commercial use of this invention should be addressed to: Innovative Technology Assets Management, Innovative Technology Assets Management, NASA, Mail Stop 202-233, 4800 Oak Grove Drive, Pasadena, CA 91109-8099. E-mail: iaoffice@jpl.nasa.gov. Refer to NPO-44751, volume and number of this NASA Tech Briefs issue, and the page number.
Probing a Spray Using Frequency-Analyzed Light Scattering

Intact length and breakup frequency are determined by a relatively simple technique.

Marshall Space Flight Center, Alabama

Frequency-analyzed laser-light scattering (FALLS) is a relatively simple technique that can be used to measure principal characteristics of a sheet of sprayed liquid as it breaks up into ligaments and then the ligaments break up into droplets. In particular, through frequency analysis of laser light scattered from a spray, it is possible to determine whether the laser-illuminated portion of the spray is in the intact-sheet region, the ligament region, or the droplet region. By logical extension, it is possible to determine the intact length from the location of the laser beam at the transition between the intact-sheet and ligament regions and to determine a breakup frequency from the results of the frequency analysis. Hence, FALLS could likely be useful both as a means of performing research on sprays in general and as a means of diagnostic sensing in diverse applications in which liquid fuels are sprayed. Sprays are also used for drying and to deposit paints and other coating materials.

Figure 1 depicts a laboratory setup for analyzing a spray by use of FALLS. A laser (or, in principle, any other suitable light source) generates a beam of light that is formed into a horizontal sliver by use of a beam-expanding lens and a cylindrical lens (the lenses are not shown in the figure). The spray to be analyzed is directed downward, and the sliver of light illuminates a thin probe volume that amounts, in effect, to a small cross section of the vertical spray. The principal characteristics of this region will be assessed. A rapid-response photodiode is positioned to receive forward-scattered light at an angle of 30° off the laser-beam axis. The output of the photodiode is digitized and frequency-analyzed by means of a fast Fourier transform (FFT). The nozzle and, thus, the spray can be translated vertically to enable optical probing of the spray at various axial distances from the nozzle.

In a series of tests in the setup of Figure 1, the technique was applied to a conical spray of water at a mass flow rate of 0.1 kg/s generated by use of a swirl coaxial injector nozzle with an upstream pressure of 830 kPa. Illumination power of 400 mW was generated by an argon-ion laser. The photodiode output was sampled at a rate of 20 kHz. Raw data were collected in sets of 4,096 samples (representing an observation interval of 0.2048 second per set). The spray was photographed with a stroboscope to obtain images of the spray for correlation with the FFT data. In addition, by use of phase Doppler anemometry, the spray speed 51 mm from the nozzle tip was estimated to be 14.2 m/s.

Each test run, lasting about 10 seconds, yielded 50 such sets. The 50 FFTs from each run were combined to obtain a single normalized FFT curve shown by the fluctuating thin curve shown in the upper part of Figure 2. A smoothing routine was invoked to suppress noise in the data, yielding the thick smoother curve shown superimposed on the thinner fluctuating one. Normalization was effected by dividing each FFT magnitude by the greatest magnitude in the noisy combined-FFT curve obtained from the test run in which the nozzle was positioned 44 mm above the light sheet. As shown in the lower part of Figure 2, the shapes of the normalized FFT curves were found to lie in three clusters corresponding to the intact-sheet, ligament, and droplet regions, respectively.

This work was done by Richard Eskridge and Michael H. Lee of Marshall Space Flight Center and Noah O. Riley of the University of Alabama in Huntsville.

For further information, contact Sammy Nabors, MSFC Commercialization Assistance Lead, at sammy.a.nabors@nasa.gov. Refer to MFS 31851-1.