Figure 2 depicts the operation of the present and previous linear actuators and illustrates one of the advantages of the present actuator over the previous one. In the present actuator, the first and third clutches are operated in unison and are mounted on a stationary structure denoted A. The second and fourth clutches are operated in unison and are mounted on a moveable structure, denoted B, that can be driven a short distance forward or backward along the channel by a PZT. In step 1 of an operational sequence in which the slider is moved leftward, the clutches on unit A are released. In step 2, the PZT is extended to push unit B and the slider leftward. In step 3, the clutches on unit A are latched while the clutches on unit B are released. In step 4, the PZT is retracted to bring unit B rightward. Repetition of steps 1 through 4 causes the slider to move leftward in repeated small increments.

The operational sequence of the previous two-clutch actuator is similar. However, the two-clutch configuration is susceptible to tilt of the slider and a consequent large increase in drag. Hence, the primary operational advantages of the present four-point-latching design over the prior two-point-latching design are less drag and greater control robustness arising from greater stability of the orientation of the slider.

Assemblies containing curved piezoceramic composite actuators have been invented as means of stretching optical fibers by amounts that depend on applied drive voltages. Piezoceramic fiber composite actuators are conventionally manufactured as sheets or ribbons that are flat and flexible, but can be made curved to obtain load-carrying ability and displacement greater than those obtainable from the flat versions. A curved actuator of this type can be fabricated by bonding a conventional flexible flat actuator to a thin metal backing sheet in a flat configuration at an elevated temperature so that upon cooling to room temperature, differential thermal contraction of the layers causes the laminate to become curved. Alternatively, such a curved actuator can be fabricated by bonding the layers together at room temperature using a curved mold.

In the primary embodiment of this invention, piezoceramic fibers are oriented parallel to the direction of longitudinal displacement of the actuators so that application of drive voltage causes the actuator to flatten, producing maximum motion. Actuator motion can be transmitted to the optical fiber by use of hinges and clamp blocks (see figure). Each clamp block includes a setscrew that tightens down onto a metal ferrule through which the optical fiber is bonded. Each hinge contains a clearance hole for a hinge pin, slots to accept the piezoceramic fiber composite actuators, and a clearance groove for the ferrule.

In the original application of this invention, the optical fiber contains a Bragg grating and the purpose of the controlled stretching of the fiber is to tune the grating as part of a small, lightweight, mode-hop-free, rapidly tunable laser for demodulating strain in Bragg-grating strain-measurement optical fibers attached to structures. The invention could also be used to apply controllable tensile force or displacement to an object other than an optical fiber.

Prior tunable lasers that are not fiber-optic lasers are relatively bulky and are limited to tuning frequencies of the order of 1 Hz. Tunable fiber-optic lasers could potentially be made much smaller, lighter in weight, and more rapidly tunable if strained by use of this invention. Prior actuators that could be used to strain-tune fiber-optic lasers or gratings include piezoelectric stacks that produce displacements smaller than those needed and that, in comparison with assemblies according to the present invention, are heavier. Displacements produced by piezoelectric stacks can be amplified mechanically, but the mechanisms needed to effect amplification add considerable weight, which can be unacceptable in aeronautical or aerospace applications because of the high per-unit-weight cost of flight.

This work was done by Risaku Toda and Eui-Hyeok Yang of Caltech for NASA’s Jet Propulsion Laboratory.

In accordance with Public Law 96-517, the contractor has elected to retain title to this invention. Inquiries concerning rights for its commercial use should be addressed to:

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Curved Piezoelectric Actuators for Stretching Optical Fibers

Curved actuators produce greater displacements than do flat actuators.

Langley Research Center, Hampton, Virginia

This work was done by Sidney G. Allison, Qamar A. Shams and Robert L. Fox of Langley Research Center. For further information, contact the Langley Innovative Partnerships Office at (757) 864-4015. LAR-17356-1

An Assembly Containing Two Curved Actuators stretches an optical fiber by an amount that depends on the voltage applied to the electrodes.