

IRAC Full-Scale Flight Testbed **Capabilities**



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Full-Scale Flight Test Overview

- **Provide validation of adaptive control law concepts through full scale flight evaluation in a representative avionics architecture**
- **Develop an understanding of aircraft dynamics of current vehicles in damaged and upset conditions**
 - Real-world conditions**
 - Turbulence, sensor noise, feedback biases**
 - Coupling between pilot and adaptive system**
 - Simulated damage**
 - “B” matrix (surface) failures**
 - “A” matrix failures**
- **Evaluate robustness of control systems to anticipated and unanticipated failures**



RFI Objectives

- **Objective 1: To validate adaptive control technology using manned flight experiments**
 - **Experiments addressing:**
 - **Challenges that can only be addressed by manned flight**
 - **Address barriers to implementation**
 - **Sufficiently large (meaningful) failures**
- **Objective 2: To examine the benefits of manned Vs autonomous recovery from upsets or failures**
 - **Experiments addressing:**
 - **Types of pilot input to system**
 - **Separate, backup, or primary flight control implementation**
 - **Pilot Interaction with the adaptive system**



RFI Objectives

- **Objective 3: To test and validate system-level reasoning for flight control reconfiguration**
 - **Experiments addressing:**
 - **Detection, diagnosis, prognosis, and isolation technologies for control reconfiguration and envelope limiting controls**





NASA F-18 Full-Scale Test bed

- **Extensive Structural Instrumentation**

- Strain Gages
- Accelerometers
- Optical Flight Deflection Measurement System

- **Quadraplex Research Flight Control System (RFCS)**

- Safety Monitoring and Mode Transitions
- Full Command of Surfaces/Throttles
- On-Board Excitation System
- Simulated Failures



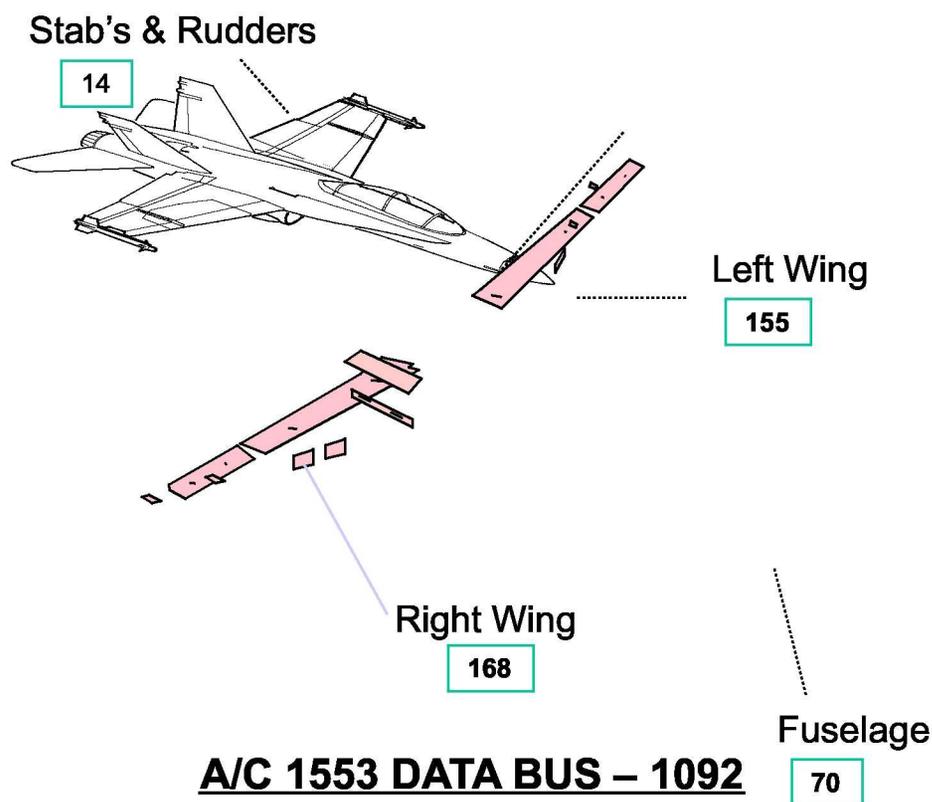
- **Dual Airborne Research Test System (ARTS IV) Computers**

- Commands Surfaces and Engines through RFCS
- Capability for Interfacing to Structural Instrumentation
- Additional Payload I/O



Instrumentation

TOTAL PARAMETERS – over 1669



A/C 1553 DATA BUS – 1092

GPS/INS 1553 DATA BUS – 170

FIBEROPTIC SHAPE SENSORS (In work)

RH WING

PARAMETERS-168

- 107 - FULL BRIDGE STRAIN GAGES
- 18 - ACCELEROMETERS
- 8 - POSITION SENSORS
- 10 - VOLTAGE SENSORS
- 3 - TEMPERATURE SENSORS
- 22 - PRESSURE SENSORS

LH WING PARAMETERS-155

- 77 - FULL BRIDGE STRAIN GAGES
- 18 - ACCELEROMETERS
- 8 - POSITION SENSORS
- 10 - VOLTAGE SENSORS
- 4 - TEMPERATURE SENSORS
- 22 - PRESSURE SENSORS
- 16 - FDMS TARGETS

FUSELAGE

PARAMETERS-70

- 6 - MOTION PAK
- 7 - ACCELEROMETERS
- 7 - TEMPERATURES
- 8 - FUEL QUANTITY
- 27 - MISC. A/C PARAMETER
- 15 - TCG PARAMETERS

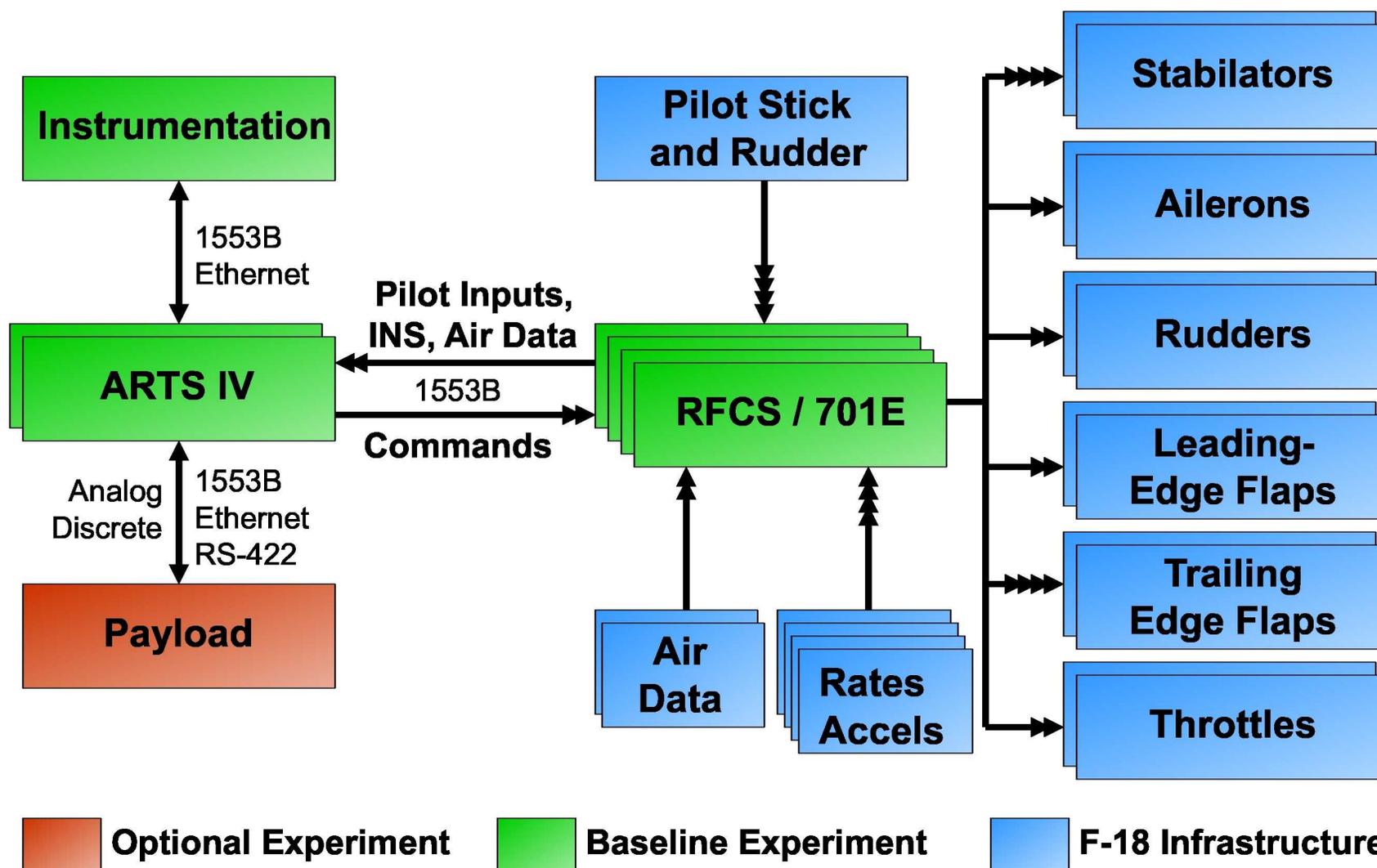
EMPENNAGE

PARAMETERS-14

- 4 - POSITIONS SENSORS
- 10 - ACCELEROMETERS

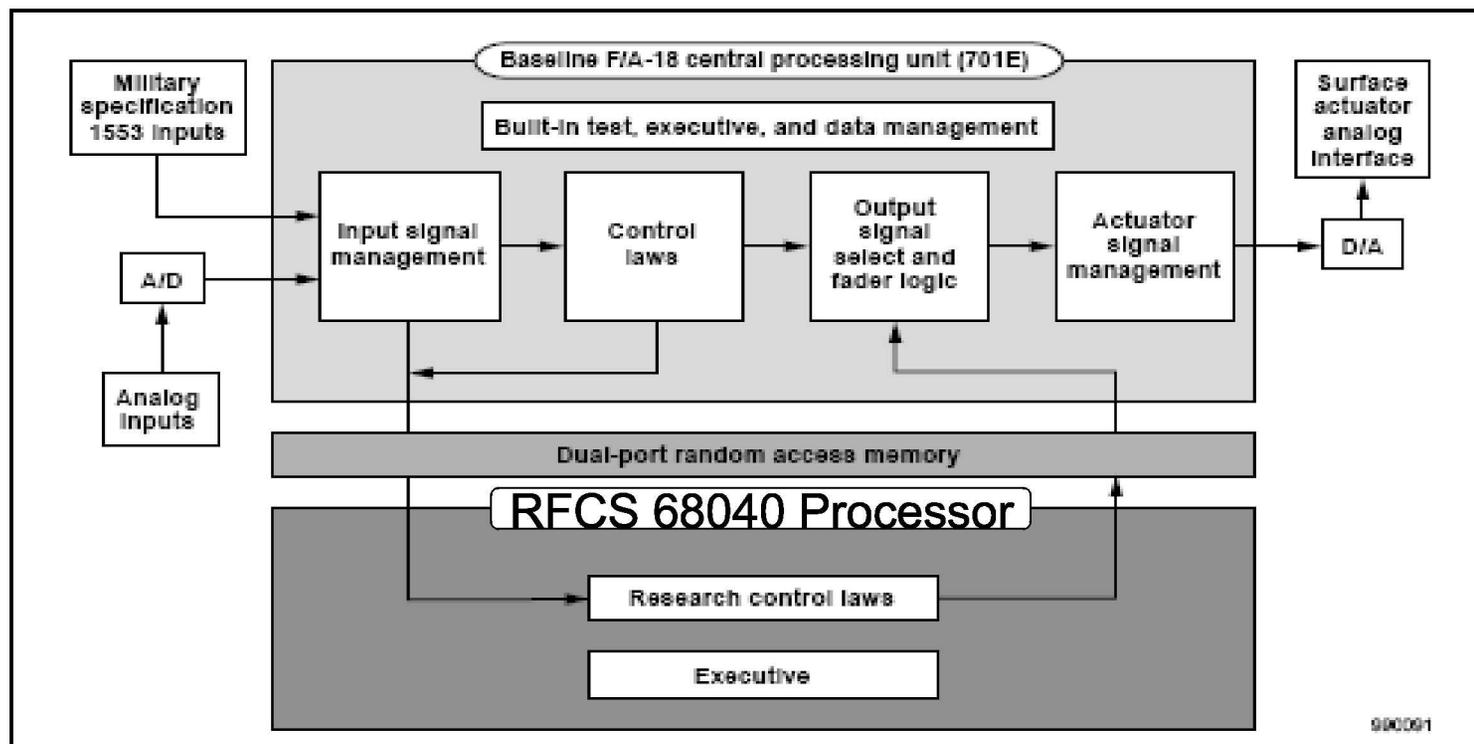


RFCS and ARTS IV Architecture





F-18 701E/RFCS integration



- F-18 Production FCS used for T/O, getting on condition, and landing
- Robust backup in case of RFCS failure or departure
- RFCS control laws completely separated from production control laws
- RFCS experiment can be point design, single axis or full-envelope, all axis design (initially will be limited to the Class B envelope)

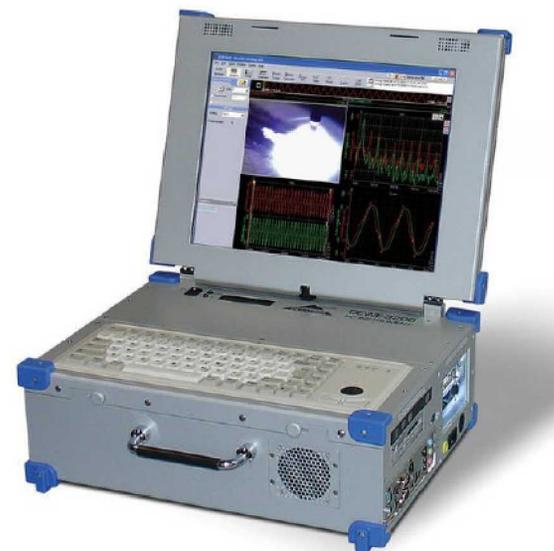


ARTS IV Hardware

Flight unit



Portable test computer



Lab unit





ARTS IV Capabilities

- **The ARTS IV can be given full control of the aircraft's control surfaces and engines via the RFCS.**
- **It is time-synchronized with the RFCS and designed to minimize time delays in the control path.**
- **The ARTS IV experiment software is mission-critical for rapid prototyping capability. The quad-redundant RFCS handles safety-critical envelope checks, fault detection and mode transitions.**
- **The ARTS IV consists of fully redundant dual hardware for potential future experiments requiring fail-safe capability.**
- **Provisions for external high-speed data links to support instrumentation feedback (structural, IVHM, etc.) into flight control experiments as well as allow an interface to each engine.**

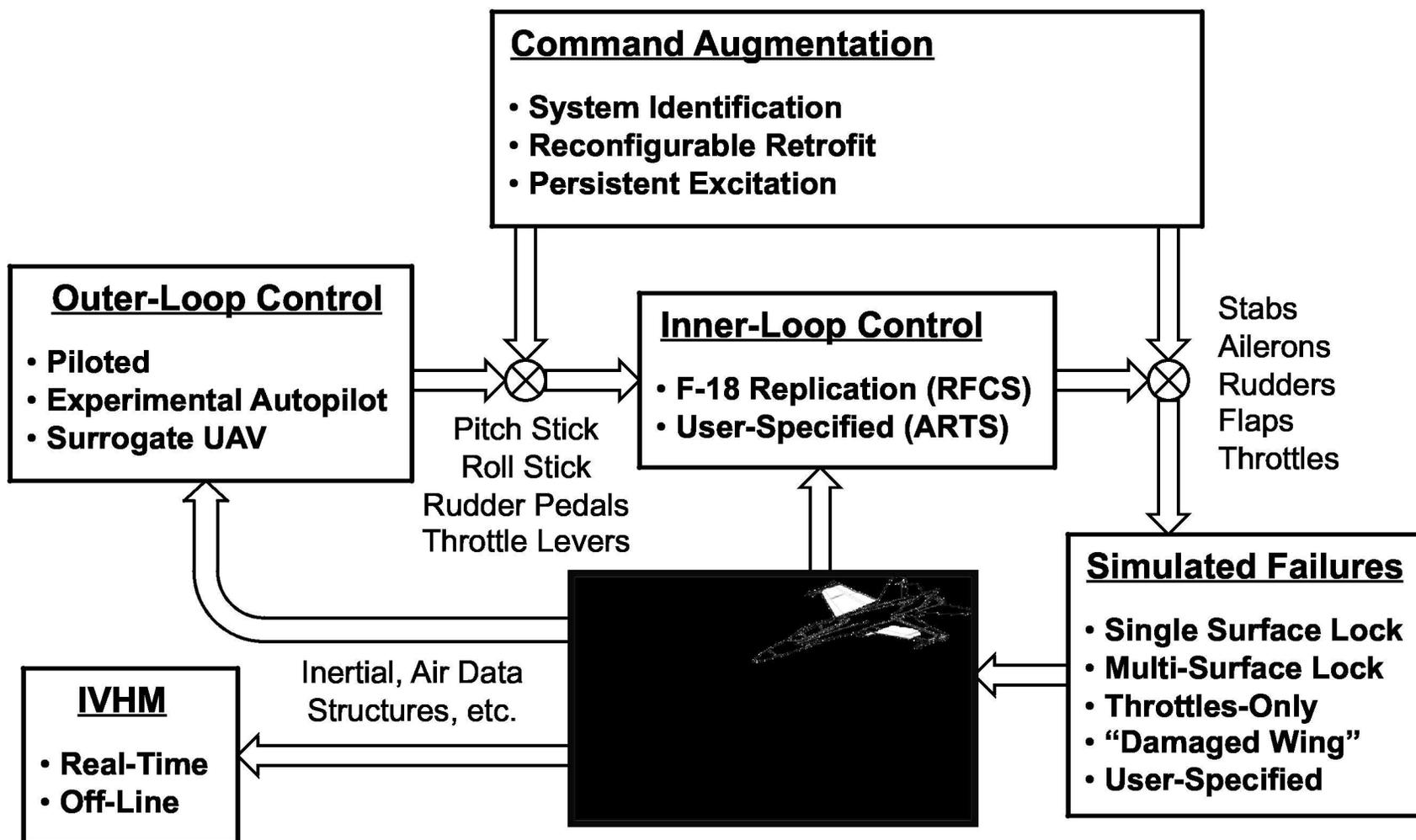


ARTS IV Capabilities (Cont.)

- **Classes of potential experiments include, but are not limited to:**
 - **Direct and indirect adaptive inner-loop control**
 - **Integrated aerodynamic and propulsion flight control**
 - **Adaptive mission planning and guidance**
 - **Integrated vehicle health monitoring**
 - **Multiple (up to 8) experiments can be loaded at once prior to flight (only one can be controlling at a time, but the others can be running as well)**
 - **Adaptive control with structural constraints (potential future capability)**
 - **ARTS IV is based on 1Ghz PPC processor technology enabling computationally intensive experiments**
- **Examples of these experiments are illustrated on the next slide**



Controls-Centric Capabilities

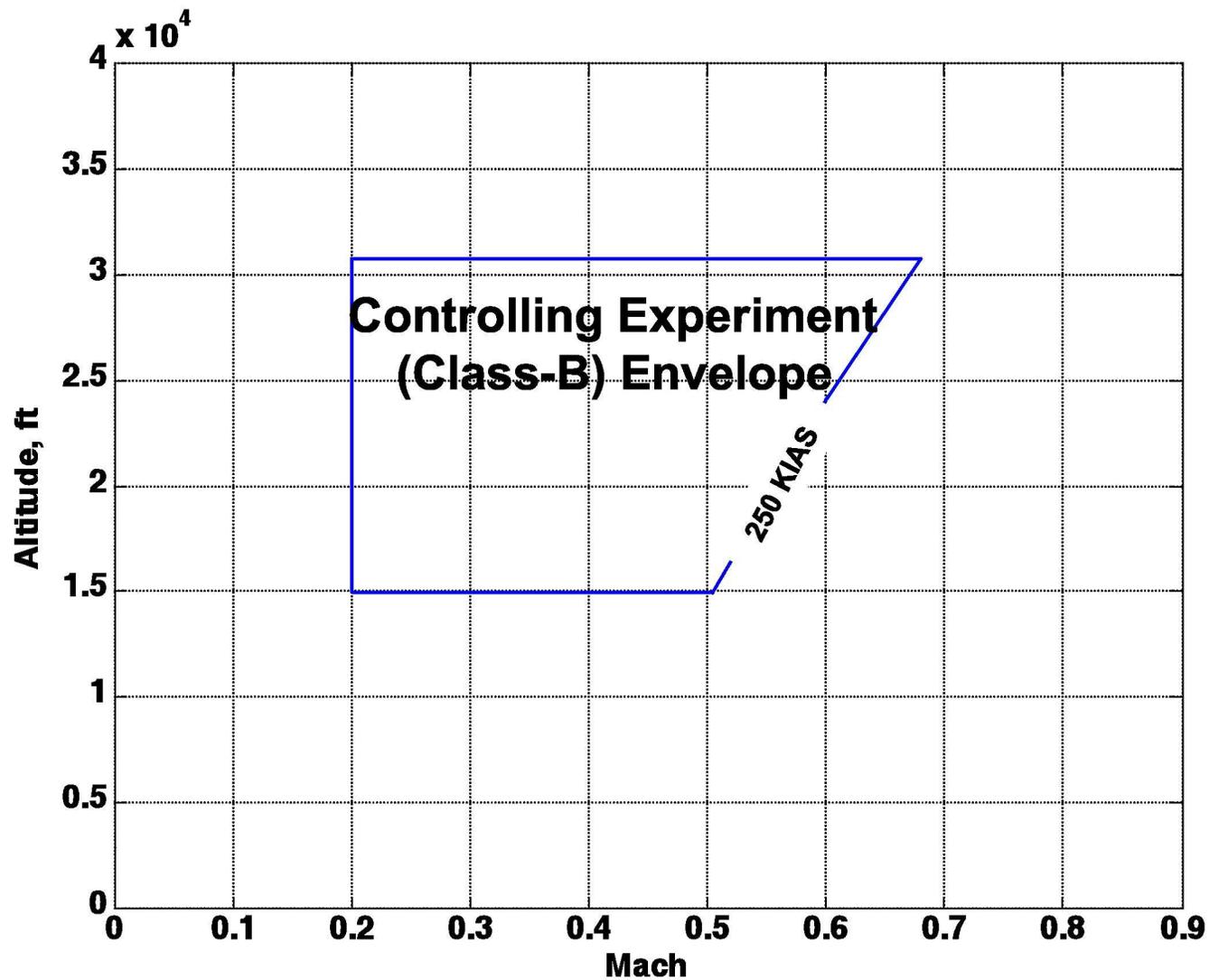




ARTS IV Capabilities (Cont.)

- **Flying an experiment**
 - **Experimenter's handbook details procedures to get experiment in the ARTS IV**
 - **Experiment can be delivered as a Simulink model or as "C" code**
 - **Verification and validation of candidate experiment done at DFRC using HILS Test Bench and piloted sim**
 - **Rapid prototyping of potential experiments and quick path to flight**
- **A non-controlling experiment can be flown anywhere in the F-18 envelope**
- **A controlling experiment will not result in structural damage in the event of a control surface hard-over when flown in the Class B envelope (see next slide)**

“Controlling Experiment” Flight Envelope





DFRC Flight Research Support Capabilities

- **Real-time Piloted F-18 Simulator**
 - Allows advanced analysis of experiments, including flight planning and piloted evaluations
 - Includes S/W models of the RFCS and ARTS IV subsystems
 - ITAR restrictions apply to most simulation models
- **F/A-18 Hardware-in-the-loop (HIL) Test Bench**
 - Allows flight qualification testing of experiments
 - Exhibits many of the same difficult to model constraints encountered on the A/C, including timing issues and system noise
 - Provides capability to rapidly advance experiments to flight and make quick turn arounds between flights
- **Real time Control Room Monitoring**
 - Critical disciplines generally include loads, flight controls, flight operations
 - May also include aerodynamics, propulsion, structural dynamics, and others as needed



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Questions?