Spirent Federal Systems Simulator Users’ Conference

NASA Johnson Space Center
Mini AERCam testing with the GSS6560

April 6 & 7, 2004
GPS Based Programs at NASA-JSC

- Several Detailed Test Objectives (DTOs) flown on STS missions featuring GPS as the primary focus of the DTO:
  - GANE on STS-77
  - RGPS RME on STS-80
  - ARPK on STS-84 & STS-86
  - SIGI DTOs on STS-88, STS-101 (SOAR) and STS-108 (CRV)
- Shuttle program switching to GPS receivers for navigation sensor
- International Space Station baselined with GPS based navigation and attitude determination. Operational since 2002.
- Crew Return Vehicle featured GPS based navigation.
- Mini Autonomous Extravehicular Robotic Camera (AERCam) relies on precise real-time relative GPS for relative navigation.
Shuttle GPS Testing

• Shuttle's Rockwell Collins "MAGR-S" tested regularly in the Shuttle Avionics Integration Laboratory (SAIL) before any Shuttle flight

• Shuttle SIGI Testing
  – Shuttle SIGI contains a Rockwell Collins GEM-3 GPS receiver and Honeywell INS
  – Was considered a candidate for replacing current Shuttle IMUs
  – Successfully attained accurate STS ascent simulation runs using an open-loop architecture.
  – Achieved stable STS ascent runs with a closed-loop simulation.
  – Supported short duration STS orbit and entry runs.
Honeywell's ISS SIGI includes a Trimble Force 19 GPS (attitude-capable) and a core Honeywell INS, plus system software for blended GPS/INS, but ISS only uses the GPS and attitude processing.

JSC team created set of Spirent/GSS scenarios for ISS orbit.

Some scenarios include ISS rotational dynamics, others are in Local-Vertical-Local-Horizontal (LVLH) Hold or Inertial Hold.

Scenarios run with a detailed model of the ISS GPS 4-antenna array for attitude determination (including multipath).

Extensive testing of the ISS SIGI with orbital scenarios over the past five years.

Two on-orbit flight experiments in May and September 2000 validated GPS simulator performance.

ISS SIGI went operational in May 2002 (see http://www.spacedaily.com/news/gps-02m.html).
ISS GN&C Integration Test Facility

- Closed-loop GN&C testing for the ISS
- US GN&C, Command and Control computers in the loop
- Russian computers were in the loop for some tests
- Also uses 3-axis rate table with ISS Rate Gyros in the loop

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CRV SIGI Testing

- Same core SIGI as the ISS SIGI but CRV planned to use the INS and Blended navigation solutions as well
- Required use of the ISRS for full GPS/INS inputs
- Real-time trajectories generated in simulation and sent to DECAAlpha from remote source, similar to Shuttle SIGI test configuration

- Real-time trajectories also used to generate motion files for standalone testing
- Simulated "blackout" during entry (due to ionization in the atmosphere) by unplugging GPS antennas to simulate loss of GPS
- Two on-orbit flight experiments have validated GPS simulator performance
Mini AERCam Relative Navigation

- Miniature Autonomous Extravehicular Robotic Camera (Mini AERCam) is a small, free-flying camera for remote inspections of the ISS
- Uses precise relative navigation with differential carrier phase GPS to provide situational awareness to operators
- Prototype system is being tested at the Johnson Space Center
- GN&C testbed includes GPS simulator (GSS6560) in the loop
Mini AERCam

Facts
• Sponsor: NASA
• 11lb free-flyer (12-13 lbs with LADAR)
• Mission: Provide extravehicular inspection for human space operations
• Deploy and retrieve from Space Shuttle Payload Bay
• Crew teleoperation and supervised autonomous scanning

Key Points for Shuttle Inspection
• Mini AERCam has always been designed for visual external inspection of the Shuttle or ISS.
• Any other existing systems would require redesign to make them more like Mini AERCam.
1997

AERCam Sprint Flight Test

Space flight test vehicle shows feasibility of remotely piloted extravehicular camera

1998

AERCam Integrated Ground Demonstration

Air-bearing test vehicle has advanced technologies to enhance operational capabilities

2003

Mini AERCam

Flight-like free-flyer integrated with 6-DOF orbital simulation and air-bearing table. Demonstrates visual inspection with miniaturized hardware in high fidelity test environment.

2004

Phase 1: Enabling Design and Flight System Proposal

Enabling design effort for Shuttle inspection provides enhancements to Mini AERCam and allows development of flight system proposal

Phase 2: Flight Test Development

Build and flight test engineering unit. Rapid flight design implemented based on mature technology demonstrator, with integration of laser depth measurement as planned in phase I

Phase 3: Flight Production

Begin production of flight units if approved by Program after successful DTO

WE ARE HERE

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Mini AERCam Closed-Loop Orbital Testing

Truth

Simulated On-Orbit Camera View

Pilot Control Displays

Situational Awareness

Control Station (CS)

Ethernet: Commands, navigation, & telemetry (no video in simulation mode)

Serial GPS data

GPS

CAT elements

Wireless Ethernet Access Point

CAT hardware

“Flight” SW

Commands & Navigation

Telemetry & Video

Free Flyer (FF)

Simulated Gyro Data

Thruster Commands

Simulation Processor

Orbit Models Vehicle Dynamics

Simulation Chassis

GPS Signal Generator

Navigation Truth

(For evaluating CS situational awareness display)

God’s-Eye View

Sun (Host)
Closed-Loop Relative Navigation Testing

* Current setup with dual GSS6560 chassis

Processor

IEEE-488

STR4762

IEEE-488

SimGEN

GSS6560 (#1)

GSS6560 (#2)

AERCam Vehicle

PPC 750

RS-422

Gyro Data

Thruster Cmds

Sun Workstation

UDP

Ethernet

Simulation Chassis

Control Station

IEEE-488

RSS-422

“ISS” GPS Data

Wireless Ethernet Communications

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Lessons Learned & Future Plans

- Timing between GSS6560, STR4762 and host simulation is paramount.
- SimGEN 1.07 and 2.0 has on occasion logged duplicate data. Indications show this may be the result of stale data into SimGEN from the protocol translator (STR4762).
- Latest simulation environment is dramatically improved over previous runs. Comparison between host simulation and SimGEN shows differences less than 1.0 mm in position and less than 1.0 mm/s in velocity.
- Updating to SimGEN 2.41.
- Acquiring third chassis for three chassis system to support dual RF GPS receiver onboard AERCam.
STR2760
- 20 channel, 1 RF output (L1/L2)
- Keyed for P(Y) code
- Dedicated to Shuttle GPS testing at the Shuttle Avionics Integration Laboratory (SAIL)

STR2760
- 40 channel system, 4 RF outputs for attitude determination capability
- Integrated with Honeywell Inertial Sensor/Recorder System (ISRS) for generating Honeywell INS test inputs
- Used for ISS Space Integrated GPS/INS and relative navigation applications (e.g., Mini AERCam)

STR4760
- 64 channel system, 4 RF outputs for attitude determination capability
- Used primarily for ISS Space Integrated GPS/INS (SIGI) and closed-loop ISS Guidance, Navigation, and Control (GN&C) testing

STR4762
- Remote PC for sending real-time simulation data to the STR2760, STR4760, or GSS6560

GSS6560 (dual chassis)
- Rack mounted 12-channel (single RF output) GPS simulator
- Primarily used for standalone and integrated testing for Mini AERCam
and specification
• GSS7700 configurations and options

GPS Modernization
• M-code summary
• SimMCODE - implementation and verification
• L2C and L5 simulation and verification

SimGEN for Windows Update
• Current status and features
• Planned additions

CRPA Test
• Current products
• Embedded Jamming

Special Secure Session
A separate Secure session will be held on 31 March 2004 hosted by the Naval Res (NRL) in Washington DC and on 5 April at Spirent Federal's Yorba Linda office. U.S clearance is required to attend. Please note: The Secure Sessions are supplier General Sessions; you can not register for the Secure Session only.

Topics at the Secure Session include: Simulator Handling & Classification, AIS Plar Considerations, GPS JPO Simulator Requirements, SAAS/SAASM Hardware & Softy Status, SAASM Test Plans, and Modernization Issues & Considerations. There will i opportunity to ask classification and SAASM testing type questions.

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