Laser Peening Effects on Friction Stir Welding

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Friction Stir Welding (FSW) is a welding technique that uses frictional heating combined with forging pressure to produce high strength bonds.

Attractive for aerospace applications
- Can result in considerable cost and weight savings, by reducing riveted/fastened joints, and part count
- Can weld metals that are difficult to weld with conventional methods
  - Space shuttle external tank

Although residual stresses in FSW are generally lower when compared to conventional fusion welds, recent work has shown that significant tensile residual stresses can be present in the weld after fabrication.

Residual tensile stresses in the weld can lead to:
- Faster crack initiation
- Faster crack propagation
- Could also result in stress corrosion cracking (SCC)

Therefore, laser shock peening was investigated as a means of moderating the tensile residual stresses produced during welding.
Nugget or the stirred zone
- The grain structure usually fine and equiaxed
  - Recrystalization from the high temperatures
  - Extensive plastic deformation

Thermo-mechanical affected zone (TMAZ)
- Lesser degree of deformation and lower temperatures
- Recrystallization does not take place
- The grain structure in elongated, with some considerable distortions

Heat affected zone (HAZ)
- Unaffected by mechanical effects, and is only affected by the friction heat

Use of FSW is expanding and is resulting in welded joints being used in critical load bearing structures
The alloy selected for this investigation was a 1.25 cm thick 2195-T8 aluminum lithium alloy. Possess many superior properties and is well suited for many aerospace applications due to its low density, high strength, and corrosion resistance. For the welding process, a rotational speed of 300 RPM in the counter-clockwise direction and a translation speed of 15 cm/min were used. The dimensions of the FSW panels were 91 cm x 30 cm x 1.25 cm. To verify the integrity of the weld, several bending tests using strip specimens were performed. The FSW specimens were inspected visually afterward with no crack indications revealed.
Laser Peening

Diagram showing the process of laser peening, including a laser beam, high pressure plasma, inertial tamping layer, ablative layer, and pressure wave impacting a workpiece.
Shot Peening

Work Piece

shot
(steel, ceramic or glass spheres)
Peening Methods

Laser Peening
- 1 mm thick laminar tamping layer
- Samples covered using a 0.22 mm thick aluminum tape
- Applied using a square laser spot
- Laser power density of 5 GW/cm²
- 18 ns in duration
- Spots were overlapped 3%
- Applied at a frequency of 2.7 Hz
- Using a 1 micrometer wavelength
- Both faces of samples were peened

Shot Peening
- 0.59 mm glass beads
- Almen intensity of 0.008-0.012
- Both faces of samples were peened
Residual Stresses
Residual Stresses

Surface Residual Stresses
Determined by the x-ray diffraction technique

Through Thickness Residual Stresses
Determined by the contour method
1. Sectioning the Sample
- Sample is fixed to a rigid backing plate
- Sample is cut along the measurement plane with an EDM wire

2. Measuring Deformation
- After sectioning a deformed surface shape is produced
  - Resulting from the relaxed residual stresses
- The displacement is measured on both sectioned surfaces using a coordinate measuring machine (CMM)

3. Estimating the Residual Stresses
- The displacements from both cutting surfaces is averaged
- The noise in the measurements is filtered
- The original residual stresses are calculated from the measured contour using a finite element model (FEM)
Residual Stresses in FSW Specimen
Residual Stresses in FSW Specimen

Surface residual stresses

Residual stresses for the various peened FSW specimens
Through Thickness Residual Stress

Two-dimensional map of the measured residual stress for the unpeened FSW specimen

Two-dimensional map of the measured residual stress for the shot peened FSW specimen

Two-dimensional map of the measured residual stress for the laser peened FSW specimen
Through Thickness Residual Stress

![Graph showing residual stress vs. distance along line for 2195-NP, 2195-L3, and 2195-SP.](image-url)
Investigate the effects of peening

- Tensile Properties
- Microhardness
- Surface Effects
Peening Conditions

Mechanical Properties

Peening Conditions

No Peening  Shot Peening  Laser Peening (1 layer)  Laser Peening (3 layers)  Laser Peening (6 layers)
Tensile Properties

- Conventional transverse tensile testing only provides the overall strain experienced by the sample.

- It is necessary to determine local strains and equivalent tensile properties across the weld.
  - Evaluated at different regions of the weld using an ARAMIS system.
Step 1:
• A random or regular pattern with good contrast is applied to the surface of the test object and is deformed along with the object.

• As the specimen is deformed under load, the deformation is recorded by the cameras and evaluated using digital image processing.

Step 2:
• The initial image processing defines a set of unique correlation areas known as macro-image facets, typically 5-20 pixels across.

Step 3:
• These facets are then tracked in each successive image with sub-pixel accuracy.

• Strains are calculated at different regions across the weld region.

Intrinsic Tensile Properties
Tensile Properties for 2195

As welded condition

Tensile properties at different regions of the weld for a FSW 2195 AA
The weld nugget exhibited the lowest tensile properties when compared to other locations across the weld

- Strengthening precipitates in 2195 AA were no longer present in the weld nugget
  - Temperature during joining was above the solution temperature of the hardening precipitates
- This region of the weld will therefore be relatively ineffective in inhibiting dislocation motion
  - The localized strain in the softened area of the weld will result in lower mechanical properties
Tensile properties at the weld nugget under different peening conditions
Tensile Properties at TMAZ

Tensile properties at the TMAZ under different peening conditions
Global Yield and Ultimate Stresses

The ultimate tensile strength for different peening conditions

The yield stress (0.2% offset) for different peening conditions
Strain distribution across the weld

- Unpeened
- Shot Peened
- Laser Peened (1 Layer)
- Laser Peened (3 Layers)
- Laser Peened (6 Layers)
Grain size histogram for laser peened specimen

Grain size histogram for unpeened specimen

EBSD Grain Size Difference
Laser peened demonstrated:
- 60% increase in the yield strength in the weld nugget in the FSW joint
- 11% increase to the ultimate tensile strength in the weld nugget in the FSW joint
- In contrast, shot peening exhibited only modest improvement to the tensile properties (3%)

The increase in mechanical properties from the laser peening was mainly attributed to:
- High levels of compressive residual stresses introduced during the high energy peening that can reach significantly deeper than shot peening
- Increase in dislocation density from the peening

Laser peening using six layers resulted in a 35% reduction to ductility
Tensile Properties (360 F)

![Graph showing tensile properties for different peening treatments: Laser Peening (6 layers), Laser Peening (3 layers), Shot Peening, and No Peening. The graph plots Stress (MPa) against Strain (%).]
Tensile Properties (-150 F)

![Graph showing tensile properties at -150 F for different peening methods. The x-axis represents strain (%) ranging from 0 to 18, and the y-axis represents stress (MPa) ranging from 25 to 475. The graph includes lines for Laser Peening (6 layers), Laser Peening (3 layers), Laser Peening (1 layer), Shot Peening, and No Peening.]
Microhardness profile across the top side of the weld for different peening methods.
Microhardness

- Significant hardness was achieved by processing the FSW 2195 AA samples with laser peening
  - Hardness increase around 28% in the top surface
  - Hardness increase around 21% in the bottom side of the weld nugget region

- Hardness levels due to laser peening increased proportionally with the number of peening layers in the 2195 aluminum alloy

- The polishing that takes place prior to microhardness measurement can wipe out all the hardness effects produced by the shot peening.
  - This is because the hardening effects from shot peening only affect a shallow depth in the material.

- Hardness profiles across the weld were wider on the top side of the weld compared to the bottom surface.
  - That was attributed to bottom plate contact with the backing plate which acts as a heat sink, therefore reducing the metallurgical transformations that take place at high temperatures.
Surface Roughness

Base Material  Shot Peening  Laser Peening

Shot Peening  Laser peening
## Surface Roughness

<table>
<thead>
<tr>
<th>Condition</th>
<th>Ra</th>
<th>Rpk</th>
<th>Rvk</th>
<th>Nomenclature</th>
</tr>
</thead>
<tbody>
<tr>
<td>Unpeened</td>
<td>1.087 µm</td>
<td>1.429 µm</td>
<td>0.93 µm</td>
<td></td>
</tr>
<tr>
<td>Shot Peened</td>
<td>5.029 µm</td>
<td>5.761 µm</td>
<td>2.884 µm</td>
<td></td>
</tr>
<tr>
<td>Laser Peened (6 layers)</td>
<td>1.336 µm</td>
<td>1.815 µm</td>
<td>1.328 µm</td>
<td></td>
</tr>
</tbody>
</table>

**Nomenclature**

- **Ra**: Roughness average
- **Rpk**: Maximum peak height
- **Rvk**: Maximum valley depth
Surface Roughness

Laser peening (Three layers)

Laser Peening (Six layers)
Surface Wear/Friction

- Different Speeds
- Different Loads
- Different Peening
- Different Materials
• Testing done with identical contact force (10 N) and speed (0.5 cm/sec)
• Shot peening appears to provide higher steady state friction coefficients.
• Long term friction coefficients for laser peened and bare surfaces were comparable.
Wear

Influence of Depth Wear Resistance on Al 2195 as a function of peening

No peening  
Laser peening  
Shot peening
Fatigue Crack Growth Rates

- **FCGR**
  - Room Temperature
  - Elevated Temperatures (360 F)
  - Cryogenic Temperatures (-150F)

Peening Methods:
- No Peening
- Shot Peening
- Laser Peening
Fatigue Samples

Through Thickness Cracks
a vs. N for R=0.1 (RT)
a vs. N for R=0.7 (RT)
A vs. N for R=0.1 (180c)
A vs. N for R=0.1 (-100c)
Fractured Surfaces

Fractured surface and fatigue striations of an unpeened sample at $R=0.1$
Conclusions

- The laser peening process can result in considerable improvement to crack initiation, propagation, and mechanical properties in FSW
  - *Longer hardware service life*

- Improve processed hardware safety
  - *By producing higher failure tolerant hardware, & reducing risk*

- Lower hardware maintenance cost
  - *Longer hardware service life, and lower hardware down time*

*Application of this proposed technology will result in substantial benefits and savings throughout the life of the treated components*
Thank You!