of the specimens increased significantly with proportions of fibers up to about 0.3 weight percent.

The second-mentioned improvement — directed-vortex core-blowing — is directed toward obtaining more nearly optimum fluidization of the sand/resin mixture as the mixture is blown into the core box. In this subprocess, the sand/resin mixture is fed from an overhead sand magazine, through an inlet blow tube, into the core box (see Figure 2). Compressed air is fed into an annular plenum, from whence it flows into the core box through a number of directed-vortex nozzles. The nozzles are designed so that the flows from the nozzles generate a partial vacuum in the outlet from the sand magazine and pump a highly fluidized sand/resin mixture into the core box. Highly fluidized and accelerated sand particles travel long distances, the net result being more nearly complete and consistent filling of the core box. The directed-vortex nozzles are also used to feed in the amine gas for polymerization and the compressed air for purging the amine gas.

The third-mentioned improvement is directed toward preventing clogging of the exhaust vents of the core box. The total cross-sectional area of these vents should be about 80 percent of the cross-sectional area of the inlet blow tube — large enough for effective venting but just small enough to provide the back pressure needed to make the catalyst gas diffuse throughout the core to ensure a uniform cure. Vents are deliberately partially blocked by any of a variety of devices, typical ones being steel meshes or slotted steel disks. Heretofore it has been necessary to clean the vents at intervals during the process. By placing filters made of filtration-grade fabrics upstream of the vents, one prevents clogging of the vents, thereby eliminating the expense and loss of time associated with cleaning of the vents.

This work was done by Sayavur I. Bakhtiyarov, Ruel A. Overfelt, and Sabit Adanur of Auburn University for Marshall Space Flight Center. For further information, contact George Konstant, Auburn University Technology Transfer Associate Manager, (334) 844-4977 or ott@auburn.edu. MFS-31819/20/21

![Diagram of Air Flowing From Directed-Vortex Nozzles](https://ntrs.nasa.gov/search.jsp?R=20110015059)

Solid Freeform Fabrication of Composite-Material Objects

Parts specified by CAD data files could be fabricated as needed.

Lyndon B. Johnson Space Center, Houston, Texas

Composite solid freeform fabrication (C-SFF) or composite layer manufacturing (CLM) is an automated process in which an advanced composite material (a matrix reinforced with continuous fibers) is formed into a freestanding, possibly complex, three-dimensional object. In CLM, there is no need for molds, dies, or other expensive tooling, and there is usually no need for machining to ensure that the object is formed to the desired net size and shape.

CLM is a variant of extrusion-type rapid prototyping, in which a model or prototype of a solid object is built up by controlled extrusion of a polymeric or other material through an orifice that is translated to form patterned layers. The second layer is deposited on top of the first layer, the third layer is deposited on top of the second layer, and so forth, until the stack of layers reaches the desired final thickness and shape.

The elements of CLM include (1) preparing a matrix resin in a form in which it will solidify subsequently, (2) mixing the fibers and matrix material to form a continuous pre-impregnated tow (also called “towpreg”), and (3) dispensing the pre-impregnated tow from a nozzle onto a base while moving the nozzle to form the dispensed material into a patterned layer of controlled thickness. When the material deposited into a given layer has solidified, the material for the next layer is deposited and patterned similarly, and so forth, until the desired overall object has...
been built up as a stack of patterned layers. Preferably, the deposition apparatus is controlled by a computer-aided design (CAD) system. The basic CLM concept can be adapted to the fabrication of parts from a variety of matrix materials.

It is conceivable that a CLM apparatus could be placed at a remote location on Earth or in outer space where (1) spare parts are expected to be needed but (2) it would be uneconomical or impractical to store a full inventory of spare parts. A wide variety of towpregs could be prepared and stored on spools until needed. Long-shelf-life towpreg materials suitable for such use could include thermoplastic-coated carbon fibers and metal-coated SiC fibers. When a spare part was needed, the part could be fabricated by CLM under control by a CAD data file; thus, the part could be built automatically, at the scene, within hours or minutes.

This work was done by C. Jeff Wang and Jason Yang of Nanotek Instruments, Inc., and Bor Z. Jang of Auburn University for Johnson Space Center.

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