Two-Phase Cryogenic Heat Exchanger for the Thermodynamic Vent System

Robert J. Christie
Glenn Research Center, Cleveland, Ohio
Since its founding, NASA has been dedicated to the advancement of aeronautics and space science. The NASA Scientific and Technical Information (STI) program plays a key part in helping NASA maintain this important role.

The NASA STI Program operates under the auspices of the Agency Chief Information Officer. It collects, organizes, provides for archiving, and disseminates NASA's STI. The NASA STI program provides access to the NASA Aeronautics and Space Database and its public interface, the NASA Technical Reports Server, thus providing one of the largest collections of aeronautical and space science STI in the world. Results are published in both non-NASA channels and by NASA in the NASA STI Report Series, which includes the following report types:

- **TECHNICAL PUBLICATION.** Reports of completed research or a major significant phase of research that present the results of NASA programs and include extensive data or theoretical analysis. Includes compilations of significant scientific and technical data and information deemed to be of continuing reference value. NASA counterpart of peer-reviewed formal professional papers but has less stringent limitations on manuscript length and extent of graphic presentations.

- **TECHNICAL MEMORANDUM.** Scientific and technical findings that are preliminary or of specialized interest, e.g., quick release reports, working papers, and bibliographies that contain minimal annotation. Does not contain extensive analysis.

- **CONTRACTOR REPORT.** Scientific and technical findings by NASA-sponsored contractors and grantees.

- **CONFERENCE PUBLICATION.** Collected papers from scientific and technical conferences, symposia, seminars, or other meetings sponsored or cosponsored by NASA.

- **SPECIAL PUBLICATION.** Scientific, technical, or historical information from NASA programs, projects, and missions, often concerned with subjects having substantial public interest.

- **TECHNICAL TRANSLATION.** English-language translations of foreign scientific and technical material pertinent to NASA's mission.

Specialized services also include creating custom thesauri, building customized databases, organizing and publishing research results.

For more information about the NASA STI program, see the following:

- Access the NASA STI program home page at [http://www.sti.nasa.gov](http://www.sti.nasa.gov)
- E-mail your question via the Internet to help@sti.nasa.gov
- Fax your question to the NASA STI Help Desk at 443–757–5803
- Telephone the NASA STI Help Desk at 443–757–5802
- Write to: NASA Center for AeroSpace Information (CASI) 7115 Standard Drive Hanover, MD 21076–1320
Two-Phase Cryogenic Heat Exchanger for the Thermodynamic Vent System

Robert J. Christie
Glenn Research Center, Cleveland, Ohio
Trade names and trademarks are used in this report for identification only. Their usage does not constitute an official endorsement, either expressed or implied, by the National Aeronautics and Space Administration.

Level of Review: This material has been technically reviewed by technical management.
Two-Phase Cryogenic Heat Exchanger for the Thermodynamic Vent System

Robert J. Christie
National Aeronautics and Space Administration
Glenn Research Center
Cleveland, Ohio 44135

Abstract

A two-phase cryogenic heat exchanger for a thermodynamic vent system was designed and analyzed, and the predicted performance was compared with test results. A method for determining the required size of the Joule-Thomson device was also developed. Numerous sensitivity studies were performed to show that the design was robust and possessed a comfortable capacity margin. The comparison with the test results showed very similar heat extraction performance for similar inlet conditions. It was also shown that estimates for Joule-Thomson device flow rates and exit quality can vary significantly and these need to be accommodated for with a robust system design.

Introduction

Thermodynamic vent systems (TVS) are being developed to control the pressure in cryogenic propellant tanks used on spacecraft. Heat penetrates these tanks and causes the cryogenic propellant to boil-off. In order to prevent over pressurization of the tank, this excess vapor needs to be vented off. In microgravity environments it is difficult to just vent off vapor without including some of the liquid. A TVS provides a means to prevent this by tapping off liquid, vaporizing this liquid, and using the heat of evaporation to cool the tank. No liquid is vented overboard. The vapor can also be used to cool penetrations and other tanks. In liquid hydrogen propellant applications, para-to-ortho catalytic conversion can be added to increase the cooling capacity.

Description

Thermodynamic Vent System

The Thermodynamic Vent System (TVS) consists of two main parts: the Joule-Thomson device and the two-phase heat-exchanger. A schematic of the system is shown in Figure 1. There is also a pump which provides external cooling flow over the tube coils inside the heat exchanger; and control valves are provided to modulate the flow to the vent and to direct the pump discharge to either an axial jet or a spray bar. The Joule-Thomson (J-T) device is a multi-stage orifice called a ViscoJet (Lee Company) which is manufactured by the Lee Company (Ref. 1).

Operation is rather simple: the control valve on the left of Figure 1 is opened and this provides high pressure liquid cryogen to the ViscoJet (a.k.a. J-T device). The liquid in the tank is at a high pressure but there is low pressure downstream of the ViscoJet. The low pressure is typically 1 atmosphere for ground testing but would likely be vacuum in a space application. The J-T device provides a temperature change in the fluid with no heat exchange, i.e., it is isenthalpic. The fluid, now at a lower temperature and pressure, passes through the heat exchanger absorbing heat from the external fluid flow and evaporates. The vapor exits the tank through the vent system where the mass flow is measured.
The Lee ViscoJet is a multi-stage orifice which is being used as a Joule-Thomson device. This series of orifices provides relatively high pressure drops using relatively large orifice dimensions, thus low flows can be obtained without the need for very small passages which would be subject to clogging by contamination. The flow through a ViscoJet can be calculated using the term for flow resistance, LOhm (Lee Company), and the manufacture's equation (Ref. 2):

\[
\text{mdot} = \frac{10,000}{\text{LOhm}} \left( \frac{\Delta P}{S} \right)^{1/2}
\]

where

- \text{mdot} = \text{mass flow rate (lb/hr)}
- \text{LOhm} = \text{liquid resistance (gal/min/psid)}
- \Delta P = \text{pressure drop (psid)}
- S = \text{specific gravity (lb/ft}^3\text{)}

**Joule-Thomson Device**

Papell, Saiyed and Nylan (Ref. 3) reported that this equation is not accurate for two-phase flows. Since cryogenic propellants are typically stored at near saturated conditions, it is likely that cavitation will occur within the ViscoJet. Therefore they conducted tests with a series of ViscoJets over a range of pressures, temperatures and vapor qualities and developed an equation that corrects for vapor formation:

\[
\text{mdot} = \frac{10,000}{\text{LOhm}} \left( \frac{\Delta P}{S} \right)^{1/2}(1-X)
\]

where

- X = \text{vapor quality}

This equation, called the Modified Lee Equation, was used in an iterative process to determine what LOhm size of ViscoJet would provide a mass flow closest to that desired. The ViscoJets are manufactured in fixed sizes and not all sizes are available from stock. Therefore the selected ViscoJet may not have the exact characteristics desired.
A spreadsheet, named LOX JT Sizer.xls, was developed to aid this iterative process. A snapshot of the spreadsheet is shown in the Appendix.

1. The user first specifies the desired heat rate (Qdot) to be removed by the TVS system.
2. Using the latent heat of evaporation (hlv) the desired mass flow rate (mdot) is calculated: mdot = Qdot/hlv.
3. The user also provides the upstream total pressure (Ptotal) and the tank temperature (Tcryo) or, if Helium is used for pressurization, the partial pressure of the cryogen.
4. An estimate of the downstream pressure is also provided (Pexit). This is a function of the pressure drop through the heat-exchanger, the vent tubing and the exit pressure.
5. The spreadsheet calculates the saturation temperature (Tsat).
6. The upstream and downstream enthalpies are calculated and the difference between the two are used to calculate the quality: X = (hliq_{inlet}-hliq_{exit}) / (hvap_{exit}-hliq_{exit}). The presence of vapor increases the required pressure drop across to achieve the desired mass flow rate.
7. Use Goal Seek to adjust the LOhm value (cell C23) until the difference (cell C28) between the actual exit pressure (cell C29) and the calculated exit pressure (cell C27) equals zero. The spreadsheet calculates the necessary LOhm value to generate the necessary pressure drop based on the mass flow, temperature and quality conditions.
8. The user can then select a standard device that is close to the LOhm value calculated and adjust the initial heat input until the calculated LOhm value matches the standard size.

**Two-Phase Heat Exchanger Design**

The heat exchanger is a simple coil of stainless steel tubing held within two concentric cylinders as shown in Figure 2. The radial spacing between the coil and the cylinders is maintained by separators located at three locations. Suction from the centrifugal pump causes cryogenic fluid to enter the open end of the annular flow channel and creates a cross flow over the tube coils.

![Figure 2.—Thermodynamic vent system, two-phase cryogenic heat-exchanger.](image-url)
To analyze the heat exchanger, a SINDA/FLUINT model was developed using the graphical user interface SINAPS (Cullimore and Ring Technologies). The system schematic is shown in Figure 3. The multi-stage orifice is located in the upper row and is modeled as a series of orifices. In this example, eleven orifices were used but this is an arbitrary number. Later in this report, the number of orifices will be reduced to one to show the effect of using one versus several.

The second row represents the internal tube fluid flow of the coil. A thermal conduction model of the tube wall is represented by the 3rd, 4th, and 5th rows and the external fluid flow is the bottom row. The fluid paths are solved by FLUINT whereas the conduction model is solved by SINDA. The FLUINT and SINDA models are coupled by Ties (shown as red lines) which represent the convective heat transfer. The pump is modeled a constant mass flow device at the end of the “External Fluid Flow.”

The “Internal Fluid Flow” is modeled with ten fluid paths in series and an eleventh path for the vent flow. Each segment of the flow path is a different length, each length being a multiple of the coil circumference which is 0.244 m (9.6 in.). The various lengths of the path segments are shown in the simplified model of Figure 4. The length of the first fluid path is 1/5 of the coil circumference. The lengths of the paths gradually increase from 1/5 of a circumference to 6 times the circumference, followed by a vent tube which is 100 times the circumference length.

The heat transfer coefficient on the outside of the cooling tubes was calculated using the empirical correlations for annular flow between concentric cylinders, see Figure 5. Although the heat transfer coefficient may be inaccurate, it will be shown later that the overall effectiveness of the heat exchanger is not very sensitive to this value. This is because the thermal resistance of the tube's internal flow heat transfer is much higher and controls the overall heat flow.
Figure 4.—Internal fluid flow paths.

Figure 5.—External flow over tubing.

TABLE 1.—RESULTS OF SENSITIVITY TESTS

<table>
<thead>
<tr>
<th>Case</th>
<th>ACCEL (m/s²)</th>
<th>CURV (m)</th>
<th>Orifice twinned</th>
<th>Forced stratification</th>
<th>Lump initial condition</th>
<th>Quality at Junction 10</th>
</tr>
</thead>
<tbody>
<tr>
<td>Normal</td>
<td>9.81</td>
<td>0.039</td>
<td>Yes</td>
<td>No</td>
<td>Vent temperature</td>
<td>0.2253</td>
</tr>
<tr>
<td>0</td>
<td>0.0</td>
<td>0.039</td>
<td>Yes</td>
<td>No</td>
<td>Vent temperature</td>
<td>0.2258</td>
</tr>
<tr>
<td>2</td>
<td>9.81</td>
<td>∞</td>
<td>Yes</td>
<td>No</td>
<td>Vent temperature</td>
<td>0.2267</td>
</tr>
<tr>
<td>3</td>
<td>9.81</td>
<td>0.039</td>
<td>No</td>
<td>No</td>
<td>Vent temperature</td>
<td>0.3659</td>
</tr>
<tr>
<td>4</td>
<td>9.81</td>
<td>0.039</td>
<td>Yes</td>
<td>Yes</td>
<td>Vent temperature</td>
<td>0.285</td>
</tr>
<tr>
<td>5</td>
<td>9.81</td>
<td>0.039</td>
<td>Yes</td>
<td>No</td>
<td>Vessel temperature</td>
<td>0.2169</td>
</tr>
</tbody>
</table>

Sensitivity and Stability

The model was subjected to various changes in geometry and boundary conditions to determine how sensitive the results are to these changes and to insure that the model is not strongly sensitive to modeling inaccuracies. The performance of the SINDA/FLUINT model might be affected by traverse acceleration due to gravity, centripetal accelerations caused by the curvature of the coil, two-phase flow in the orifice, stratification of the flow, and initial conditions. Each of these was varied to see what affect it would have on the performance of the heat exchanger. The performance was monitored by observing the vapor quality at fluid Lump #10, i.e., the sixth fluid lump in the heat exchanger's flow path. The baseline case is defined in the row titled “Normal” of Table 1.
Traverse accelerations due to gravity occur during on-ground testing but do not occur in space. Therefore the acceleration value was changed from 1 to 0G, and there was no significant effect. Next the curvature value was changed from 0.039 m (1.5 in.) to no curvature ($\infty$) and again the effect was not significant. This model allows for two-phase flow in the orifice, but Cullimore and Ring have stated that two-phase flow has not been verified for orifices. Therefore the two-phase flow feature (a.k.a. twinned flow) was turned OFF for the orifice. When this was done, the quality at Junction 10 increased slightly from 0.2253 to 0.3659 and the mass flow decreased from 2.92 to 2.84 g/sec. Neither of these changes are of concern provided all of the liquid is vaporized before exiting the heat exchanger, which it was. To avoid this issue in the future, the ViscoJet should be modeled with just one orifice. If significant two-phase flow actually occurs within the ViscoJet, then this type of model is likely not adequate.

The flow regime is typically annular but occasionally becomes stratified. The different flow regimes are shown in Figure 6. When the flow was forced to always be stratified, the quality changed from 0.2253 to 0.2850 and the flow decreased from 2.917 to 2.808 g/sec. Indicating that a switch in flow regimes, which can occur unexpectedly, will not have a detrimental effect. Finally the initial temperature of the lumps was changed from that of the vent temperature to that of the tank temperature. This is the worst case because it generates the lowest quality, i.e. the least amount of fluid has been vaporized. Still the quality only changed from 0.2253 to 0.2169 and the mass flow was unchanged. The results are summarized in Table 1.

A few tests were performed to verify that the Fluint model and the modified Lee Equation perform similarly when flow conditions are changed; and to determine if the Fluint model requires any adjusting when these changes occur. First the inlet pressure was increased from 150 to 250 psia. Both showed the same increase in flow. Therefore, when inlet pressure is changed, it was not necessary to readjust the Fluint orifice flow area to match the flow conditions. Next the vent system back-pressure was set at three

![Diagram of flow regimes](image)

Figure 6.—Two-phase flow regimes (Ref. 4).
levels, 21, 34, and 65 psia. The minimum and maximum mass flow rates only differed by 10%, indicating that the system is relatively insensitive to changes in back-pressure. Similarly, the internal diameter of the vent tubing was varied from 12 mm to 11 mm and to 10 mm. The change from 12 mm to 11 mm actually caused a 6% increase in flow, which is not understood. The change to 10 mm caused the Fluint model to become unstable. Therefore it is recommended that the vent tubing have an inside diameter of 12 mm or larger. Finally, the orifice flow area was increased and decreased by 10%, to determine how sensitive the models are to inaccuracies in defining flow area. When the flow area was varied ±10% in the Fluint model, the flow increased by 16% and decreased by 4%. The pressure at the exit of the ViscoJet changed from 19.1 to 19.9 psia when the area was increased and became 18.9 psia when the area was decreased. These changes in ViscoJet exit pressures were applied to the Modified Lee Equation which showed no significant flow changes. Using compressible versus non-compressible fluid properties also showed no significant difference in the results. The above show that the two cases are stable and can be used over a reasonable range of conditions.

Next the sensitivity to the accuracy of the heat transfer coefficient on the external surfaces of the heat exchanger was examined. The heat transfer coefficients for the two extremes were determined for tubes in cross-flow (Ref. 5) and annular flow (Ref. 6). The equations are:

Cross-Flow: \[ Nu = 0.27 \text{Re}^{0.63} \text{Pr}^{0.36} \quad 10^3<\text{Re}<2\times10^5 \]

Annular-Flow: \[ Nu = 0.023 \text{Re}^{0.8} \text{Pr}^{0.4} \quad \text{Re}>6420 \]

With a flow rate of 10 GPM, the heat transfer coefficients were found to be 3200 and 1300 W/m², respectively. These two conditions were applied to the heat exchanger. Since the performance of the heat exchanger is based on how much fluid is vaporized, the quality of the fluid at the 4th loop was examined. With the lower heat transfer coefficient the quality was 22% and with the higher coefficient it was 31%. This shows that the analysis is somewhat insensitive to the accuracy of the external heat transfer coefficient.

**Sizing the Joule-Thomson Device**

The required size of the Joule-Thomson device was determined using the following conditions:

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Inlet pressure, psia</td>
<td>250</td>
</tr>
<tr>
<td>Inlet temperature, K</td>
<td>113</td>
</tr>
<tr>
<td>Heat to be removed, W</td>
<td>750</td>
</tr>
<tr>
<td>J-T device exit pressure (initial guess), psia</td>
<td>20</td>
</tr>
</tbody>
</table>

Using the latent heat of evaporation for liquid oxygen and the desired heat removal rate, the required vaporization mass rate was determined to be 3.95 g/sec. The orifice flow area in the Fluint model was adjusted until this flow rate was achieved. The Fluint results showed that the quality at the J-T device exit was 2.7% and the exit pressure was 20.7 psia. Those conditions were then applied to the Modified Lee Equation and a ‘goal-seek’ was done with Excel to find the closest ViscoJet size, which was found to be 5000 LOhm. Under these conditions the Modified Lee Equation, which assumes isenthalpic expansion, determined the quality to be 16%. If this is true, then the flow rate would be reduced to 3.4 g/sec and the heat absorbed by vaporization would only be 595 W when the inlet pressure is 250 psia. At lower pressures, since the flow rate is proportional to square root of the pressure difference, the heat rate will be lower. For example, at an inlet pressure of 150 psia, the 5000 LOhm device would pass enough flow to absorb 460 W of heat.
Two-Phase Heat Exchanger Performance

After the J-T device is sized, the heat exchanger needs to be analyzed to determine if all the liquid is being vaporized in the heat exchanger. A sample of the Fluent analysis output is shown in Figure 7. It shows that isenthalpic expansion occurs in the orifice group and the quality at the exit is 2.65%. It also shows that the quality reaches 100% by Lump #11, which is only 1/2 through the heat exchanger. Therefore all the liquid is being vaporized before it has traveled 1/2 way through the coils. This also indicates that a lower LOhm device could be used which would increase the flow rate and increase the heat removal rate. For example, a 4000 LOhm device would increase the heat absorption to 1000W at 250 psia. Or, two 5000 LOhm devices used in parallel could be used, which would provide a total fluid resistance of 2500 LOhm and would increase the flow to 8 g/sec.

Jurns (Ref. 7) found that ViscoJet flow rates differed substantially from those calculated using the Modified Lee Equations. Actual flow rates in the 20 to 100 psia range differed as much as ±45% from predicted values. At higher pressures the difference could be greater. These test anomalies advocate that actual flow rates need to be verified with testing. If two 5000 LOhm devices were used and if both had a flow rates greater than 25% above the predicted value, then this would exceed the capacity of the heat exchanger and liquid would enter the vent tubing.

Comparison With Test Data

The TVS system was tested by VanDresar (Ref. 8) with a 5000 LOhm ViscoJet using liquid oxygen and liquid nitrogen, and one of the LOX test points is compared with the analytical predictions in Table 2. The results show very similar heat extraction rates for similar inlet conditions. Heat leak rates of 600W are not typical for cryogenic propellant tanks. In order to accelerate testing schedules, the TVS system was oversized to accelerate the cooling and heaters were attached to the test tank to provide rapid heating. Although heaters were available for the above test, they were not used during the TVS cycles. The test was accelerated by using a small temperature control band. A more thorough study should be done by analyzing and comparing all of the test points.
TABLE 2.—COMPARISON WITH TEST DATA

<table>
<thead>
<tr>
<th></th>
<th>Model</th>
<th>Test</th>
</tr>
</thead>
<tbody>
<tr>
<td>Tank pressure, psia</td>
<td>250</td>
<td>231</td>
</tr>
<tr>
<td>Inlet temperature, K</td>
<td>113</td>
<td>111</td>
</tr>
<tr>
<td>Heat extraction, W</td>
<td>595</td>
<td>631</td>
</tr>
</tbody>
</table>

Conclusions

A 5000 LOhm ViscoJet is predicted to provide a mass flow rate of 3.4 g/sec when the inlet pressure is 250 psia. This flow will be totally vaporized within the heat exchanger, providing an heat extraction rate of 595 W. When the inlet pressure is decreased to 150 psia the heat extraction rate will likewise decrease to 460 W.

It was found, that for this design, it was easy to vaporize the fluid in a short amount of tube length. For these cases, liquid oxygen at 113 K, initially between 150 and 250 psia, was being subjected to a large drop in pressure to below 20 psia while being kept in a 113 K environment. The boiling point at 20 psia is 93.2 K, therefore there is a 20 K difference between the boiling point and the environment and the fluid is easily vaporized. In all of the cases analyzed there was 100% vapor quality before the half way point in the heat exchanger.

The J-T device can be modeled as a single orifice or a multistage orifice. If vapor occurs within the J-T device, due to the limitations of Fluint, the J-T device should be modeled as a single orifice. For sizing the heat exchanger under these conditions, it was not necessary to accurately model the J-T device: Just use the modified Lee equation to find the mass flow rate and assume 0% quality exiting the ViscoJet. Iterate with the Fluint model to find the backpressure and adjust the mass flow rate, then size the heat-exchanger for those conditions. This might not be true for other conditions, for example, when the inlet and outlet pressures are closer and there is a greater opportunity for vapor to be present in the J-T device.
The filename for the spreadsheet is LOX JT Sizer.xls.

<table>
<thead>
<tr>
<th></th>
<th>A</th>
<th>B</th>
<th>C</th>
<th>D</th>
<th>E</th>
<th>F</th>
<th>G</th>
<th>H</th>
<th>I</th>
<th>J</th>
<th>K</th>
<th>L</th>
<th>M</th>
<th>N</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>2</td>
<td>Heat consumption</td>
<td>Q</td>
<td>615 W</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>3</td>
<td>Mass flow rate</td>
<td>mdot</td>
<td>0.003333 kg/s</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>0.00735 lb/sec</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>4</td>
<td>Inlet conditions</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>5</td>
<td>Tank pressure</td>
<td>Ptotal</td>
<td>236.4 PSIA</td>
<td>Total tank pressure</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td></td>
<td>Tank temperature</td>
<td>Tcryo</td>
<td>113 K</td>
<td>Actual fluid temperature</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td></td>
<td>Average temperature</td>
<td>Tavg</td>
<td>104 K</td>
<td>Vapor exiting HX is at some temperature between the tank and VJ exit</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td></td>
<td>Tank liquid enthalpy</td>
<td>hliq</td>
<td>-92 kJ/kg</td>
<td>Enthalpy of liquid in tank at tank temperature</td>
<td>usabel range 50 to 250 PSIA</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td></td>
<td>Tank vapor enthalpy</td>
<td>hvap</td>
<td>92 kJ/kg</td>
<td>Enthalpy of vapor leaving HX assuming an average temperature</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td></td>
<td>Heat of evaporation</td>
<td>hh</td>
<td>-185 kJ/kg</td>
<td>thus heat absorbed is a function of enthalpy changes from tank to HX exit</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>6</td>
<td>Exit conditions</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>7</td>
<td>Guess at exit pressure</td>
<td>Paxit</td>
<td>26.1 PSIA</td>
<td>Replace with Fluent calculated exit pressure</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>8</td>
<td>Exit saturation temp</td>
<td>Txat</td>
<td>95 K</td>
<td>usabel range is 20 to 250 PSIA</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>9</td>
<td>Liquid enthalpy at exit</td>
<td>hliq</td>
<td>-123 kJ/kg</td>
<td>evaluated at saturation conditions</td>
<td>usabel range is 20 to 250 PSIA</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>10</td>
<td>Vapor enthalpy at exit</td>
<td>hvap</td>
<td>84 kJ/kg</td>
<td>evaluated at saturation conditions</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>11</td>
<td>Heat of evaporation</td>
<td>hh</td>
<td>-207 kJ/kg</td>
<td>from liquid at tank temperature to vapor at exit temperature</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>12</td>
<td>Estimate quality at exit of ViscoJet</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>13</td>
<td>Quality</td>
<td>X</td>
<td>0.147</td>
<td>if process is isenthalpic</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>14</td>
<td>Select a ViscoJet</td>
<td></td>
<td>5000 LOHM</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>15</td>
<td>Pressure drop</td>
<td>DP</td>
<td>210 PSID</td>
<td>Lee equation</td>
<td>1.450E+06 Pa</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>16</td>
<td>Desired drop</td>
<td>DP</td>
<td>210 PSID</td>
<td>Pin-Pout</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>17</td>
<td>Exit pressure</td>
<td>Paxit</td>
<td>26.0</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>18</td>
<td>Difference</td>
<td></td>
<td>0 PSID</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

With Goal Seek to adjust C23 until C28=0 then replace C23 with the closest available LOHM value. Then adjust the heat consumption until the pressure difference is 0. If the FLUINT model mass flow rate does match the above, adjust the FLUINT model orifice diameter. Return and apply the new back pressure in C14. Repeat the process until the FLUINT and VJ Sizer match in flow rate and back pressure with the selected LOHM in C23 and 0 pressure difference in C28.
References

1. http://www.theleeco.com
**Title:** Two-Phase Cryogenic Heat Exchanger for the Thermodynamic Vent System

**Authors:** Christie, Robert, J.

**Dates Covered:** From 01-10-2011 To

**Abstract:**
A two-phase cryogenic heat exchanger for a thermodynamic vent system was designed and analyzed, and the predicted performance was compared with test results. A method for determining the required size of the Joule-Thomson device was also developed. Numerous sensitivity studies were performed to show that the design was robust and possessed a comfortable capacity margin. The comparison with the test results showed very similar heat extraction performance for similar inlet conditions. It was also shown that estimates for Joule-Thomson device flow rates and exit quality can vary significantly and these need to be accommodated for with a robust system design.