Particle Morphology and Size Results from the Smoke Aerosol Measurement Experiment-2

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Results are presented from the Relight of the Smoke Aerosol Measurement Experiment (SAME-2) which was conducted during Expedition 24 (July-September 2010). The relight experiment built upon the results of the original flight during Expedition 15 by adding diagnostic measurements and expanding the test matrix. Five different materials representative of those found in spacecraft (Teflon, Kapton, cotton, silicone rubber and Pyrell) were heated to temperatures below the ignition point with conditions controlled to provide repeatable sample surface temperatures and air flow. The air flow past the sample during the heating period ranged from quiescent to 8 cm/s. The smoke was initially collected in an aging chamber to simulate the transport time from the smoke source to the detector. This effective transport time was varied by holding the smoke in the aging chamber for times ranging from 11 to 1800 s. Smoke particle samples were collected on Transmission Electron Microscope (TEM) grids for post-flight analysis. The TEM grids were analyzed to observe the particle morphology and size parameters. The diagnostics included a prototype two-moment smoke detector and three different measures of moments of the particle size distribution. These moment diagnostics were used to determine the particle number concentration (zeroth moment), the diameter concentration (first moment), and the mass concentration (third moment). These statistics were combined to determine the diameter of average mass and the count mean diameter and, by assuming a log-normal distribution, the geometric mean diameter and the geometric standard deviations can also be calculated. Overall the majority of the average smoke particle sizes were found to be in the 200 nm to 400 nm range with the quiescent cases producing some cases with substantially larger particles.

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I. Introduction

Appropriate design of fire detection systems requires knowledge of both the expected signature of the events to be detected and the background levels of the measured parameters. Terrestrial fire detection systems have been developed based on extensive study of terrestrial fires. Unfortunately there is no corresponding data set for spacecraft fires and consequently the fire detectors in current spacecraft were developed based upon terrestrial designs. There are a number of factors that can be expected to affect the particle size distribution of the smoke from spacecraft fires. The absence of buoyant flow in low-gravity increases the residence time in microgravity fires and increases the transit time from the reaction zone to the detector. Microgravity fires have been found to have radically different structure from their normal-gravity (a.k.a. 1-g) counterparts. The limited options available to respond to a spacecraft fire increase the importance of early detection. Finally the materials used in spacecraft are different from typical terrestrial applications where smoke properties were previously evaluated. All of these effects can be expected to change the smoke particle size distribution. The objective of this work was to make sufficient measurements of smoke from spacecraft fires to enable improved design of future detectors.

Smoke Background

Smoke is a general term that encompasses aerosol materials produced by a number of processes. In particular it can include unburned, recondensed, original polymer or pyrolysis products that can be either liquid or solid, hydrocarbon soot, condensed water vapor, and ash particles. Soot particles dominate the smoke particulate in established flaming fires while unburned pyrolysis products and recondensed polymer fragments are produced by smoldering and pyrolysis in the early stage of fire growth. Given the constrained space on any spacecraft, the target for the fire detection system is necessarily the early phase and not established flaming fires; consequently, the primary target for detection is the pyrolysis products and not the soot.

Prior spacecraft systems are summarized in more detail in papers by Friedman and Urban. In the Mercury, Gemini and Apollo missions, the crew quarters were limited and mission durations were short, consequently it was considered reasonable that the astronauts would rapidly detect any fire. The Skylab module, however, included approximately 30 UV-sensing fire detectors. These devices were limited to line-of-sight and were reported to have difficulties with false alarms. The Space Shuttle Detectors were based upon ionization fire detector technology, the most advanced technology available at the time and used an inertial separator designed to eliminate particles larger than 1-2 micrometers. The International Space Station (ISS) smoke detectors use near-IR forward scattering, rendering them most sensitive to particles larger than a micrometer, outside of the range of sensitivity of the shuttle detector. As described by Friedman there have been six overheat and failed component events in the NASA Orbiter fleet in addition to several similar incidents that have occurred on the ISS. None of these events spread into a real fire but as mission durations increase, the likelihood of failures increases. The experience on Mir in 1997 has shown that failure of oxygen generation systems can have significant consequences. As a result, improved understanding of spacecraft fire detection is critically needed.

Previous work on smoke particles from low-gravity sources by Urban et al. found that the particulate produced by low-gravity flames (soot or unburned fuel particles) tends to have larger size particles than in normal gravity. Results from the CSD (Comparative Soot Diagnostics) Experiment which studied smoke properties in low-gravity from several spacecraft materials suggested that liquid smoke particles could achieve sizes larger than 1 µm while solid particulate remained in the sub-micrometer range. However, the CSD experiment did not produce sufficient data concerning the size of the liquid smoke particles to guide detector design. The combined impact of these limited results and theoretical predictions is that, as opposed to extrapolation from 1-g data, direct knowledge of low-g combustion particulate is needed for more confident design of smoke detectors for spacecraft.

II. SAME Experiment

To address the limited data from the CSD experiment concerning the likely size of spacecraft smoke particulate, another experiment was developed: the Smoke Aerosol Measurement Experiment (SAME). The SAME experiment sought to avoid the problems experienced by the CSD experiment by obtaining the particulate size statistics on-orbit with a reduced dependence upon sample return to Earth. This is a challenging endeavor because existing aerosol instrumentation is typically large, incompatible with spacecraft experiment constraints, and some systems require substantial sample return to Earth. As will be described below, an alternative approach was employed that used three discrete instruments to measure separate moments of the size distribution. When combined, these moments provide useful aggregate statistics of the size distribution. The measurements were made using smoke generated by overheated spacecraft materials in much the same manner as the CSD experiment however the sample temperature, flow field, and particle aging time were more rigorously controlled in the SAME experiment. The experiment flew
twice the first time in 2007 (SAME-1) and the second (SAME-2) in 2010. When discussion applies to both flights, this paper will refer to “SAME.” If the discussion is specific to a particular flight, the flight number will be identified e.g. “SAME-2.”

A. Moment Method

The approach used by the SAME experiment is termed the ‘moment method’ for convenience. As will be described below, the approach consists of measuring three moments of the size distribution (zeroth, first and third) and using the properties of the log-normal distribution to estimate the geometric mean diameter and the standard deviation.

The average particle size and an estimate of the width of the size distribution will be estimated from various moments of the size distribution. The number distribution, \( f_N(D) \), is defined as

\[
f_N(D) = \frac{dN}{dD}
\]

(1)

where \( dN \) is the number of particles per cm\(^3\) with diameter between \( D \) and \( D + dD \). The moments of interest consist of the number concentration, \( M_0 \), the first moment \( M_1 \), and the volume or mass concentration moment, \( M_3 \) and are defined as

\[
M_i = \int D^i f_N(D) dD \quad i = 0, 1, 3
\]

(2)

When \( i = 0 \), the zeroth moment of the distribution, \( M_0 \), equation (2) is simply the number of particles per unit volume. In the SAME experiment, this was measured using a condensation nuclei counter. The first moment, \( i = 1 \), can also be thought of as the “diameter concentration” or integrated diameter per unit volume and is approximately proportional to the ionization detector moment (signal). For particles in the Mie scattering regime, particle sizes from 0.3\( \lambda \) to about 3\( \lambda \) (about 0.2 \( \mu \)m to 2.0 \( \mu \)m for a red laser), the light scattering signal is approximately proportional to the third moment, \( i = 3 \). From these moments, and a measurement of \( M_0 \) using a condensation nuclei counter, two mean diameters can be computed: the count (arithmetic) mean diameter \( D_{0.5} \) or \( D_0 \), which is equal to \( M_1/M_0 \) and the diameter of average mass \( D_{1.5} \) or \( D_m \), which is equal to \( (M_3/M_0)^{1/3} \). These calculations do not depend on the assumption of log-normal behavior. The basis for the subscript naming convention for \( D_{0.5} \) and \( D_{1.5} \) will be discussed later.

The log-normal size distribution is widely used for describing the size distribution of aerosols, including non-flaming smoke because for most aerosols; the bulk of the number concentration is associated with smaller particles. The number distribution \( f_N(D) \) for the log-normal distribution is expressed as follows:

\[
f_N(D) = \frac{N_i}{(2\pi)^{1/2} D \ln \sigma_g} \exp \left( -\frac{(\ln D - \ln D_g)^2}{2 (\ln \sigma_g)^2} \right)
\]

(3)

where \( N_i \) is the total number concentration of the aerosol (=\( M_0 \)), and \( D_g \) and \( \sigma_g \) are the geometric mean diameter and geometric standard deviation defined by

\[
\ln D_g = \frac{\int_0^\infty \ln D f_N(D) dD}{\int_0^\infty f_N(D) dD}
\]

(4)

\[
\ln \sigma_g = \left[ \frac{\int_0^\infty (\ln D - \ln D_g)^2 f_N(D) dD}{\int_0^\infty f_N(D) dD} \right]^{1/2}
\]

(5)

For the log-normal distribution, one finds that the various diameter definitions given above are related to the geometric mean diameter, \( D_g \), via the equation.\(^8\)

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\[ D_p = D_g \exp(p \ln^2 \sigma_g) \] (6)

For the count mean diameter, \(D_{0.5}\), and the diameter of average mass, \(D_{1.5}\), the corresponding values of \(p\) are 0.5 and 1.5. Fig. 1 shows a typical log-normal distribution for a \(D_g=1.0\) and \(\sigma_g=1.6\). For this distribution, the corresponding values of \(D_{0.5}\) and \(D_{1.5}\) are 1.17 \(\mu m\) and 1.39 \(\mu m\), respectively. Using equation (6), one can relate \(\sigma_g\) to the ratio of \(D_{1.5}\) and \(D_{0.5}\) via the equation:

\[ \sigma_g = \exp(\ln(D_{1.5} / D_{0.5}))^{1/2} \] (7)

By combining these three moments it is possible to compute three mean diameters of the size distribution and the geometric standard deviation. Validation of this approach is discussed in Cleary, Weinert and Mulholland.\(^7\) These statistics provide a strong basis for design of spacecraft smoke detectors.

B. Instruments

These measurements were made using an assembly of three separate instruments. Two are industrial hygiene instruments manufactured by TSI and one is a modified residential smoke detector. More complete discussion of the SAME hardware is available in an earlier paper\(^10\).

The zeroth-moment instrument is a condensation nuclei counter P-Trak\(^\text{TM}\) (TSI Inc.). This device operates by passing the aerosol-laden particle stream through a region saturated with isopropanol vapor and then into a cooler region where the vapor condenses onto the particles increasing their diameter such that they can be readily counted by a light scattering device. This instrument is very robust and operates over a range of 0 to 10\(^6\) particles/cm\(^3\) and 20 nm to 1 \(\mu m\) diameter. Some dilution is required, since the smoke concentration ranges from about 0.5 \(\times\) 10\(^6\) to 5 \(\times\) 10\(^6\) particles/cm\(^3\). The dilution was accomplished by controlled addition of filtered nitrogen to the aerosol sample being directed to the P-Trak. There was also a concern that the isopropanol condensate would not return to the wick in low-gravity.\(^5\) To mitigate this issue, the condensing section of the device was modified with very small grooves to improve conductance of the condensate back to the wick. These changes were tested in a separate space experiment with good results indicating the modified device could be used successfully in low gravity.\(^5\)

The first-moment instrument is the ionization chamber from a residential smoke detector. This device uses an alpha-particle emitter to generate ions in a region within a DC electric field. The drift of the ions in the electric field results in a current. The presence of aerosol particulate reduces the current as a result of the attachment of the ions to the particulate. The mobility of the charged aerosol is too small for it to be collected on the ionization chamber electrode. The required particle concentrations are on the order of 10\(^5\) particles/cm\(^3\) and no sample dilution was required.

The third-moment instrument is a light scattering device DustTrak\(^\text{TM}\) (TSI Inc.). The device uses a 90 degree light scattering signal to quantify the aerosol mass density. For terrestrial dust particulate this signal correlates well with the mass concentration, however, as described in the calibration discussion below, additional compensation was employed to account for the range of materials and particle sizes that were seen in the SAME experiment. The device’s operating range is from 0.001 mg/m\(^3\) to 100 mg/m\(^3\). These devices are equipped with an aerodynamic impactor at the inlet which captures particles larger than the selected size. The SAME-1 experiment included 2 DustTraks\(^\text{TM}\) one with a 1 \(\mu m\) impactor and one with a 10 \(\mu m\) impactor. The difference in the signal from these two devices provided a measure of the fraction of the particulate that was larger than 1 \(\mu m\). In some cases dilution was required owing to the high smoke concentration levels.

A schematic of the assembled hardware appears in Fig. 2. The system was installed in the Microgravity Science Glovebox (MSG), an ISS facility that provides many resources including: containment, power, data, video and uplink commanding. Smoke was generated by overheating a small sample of material in the smoke generation duct for approximately 60 seconds. During this interval, controlled flow was induced by a moving piston in the aging
chamber which drew the smoke into the chamber. The smoke was held in the chamber for a predetermined time, allowing the particles to coagulate. After a specified aging time, the smoke was then pushed by the piston into the diagnostics duct where the moment instruments made their measurements. Also installed in the diagnostics duct were space shuttle and ISS smoke detectors. As the smoke was monitored by the moment instruments, a sample of the smoke particles was deposited on Transmission Electron Microscope (TEM) grid via a Thermal Precipitator (Fig 3) which uses thermophoresis to deposit the particles on the grids. Inside the precipitator, the aerosol passed between a Transmission Electron Microscope grid and a heated wire. After the mission the grids were removed from the assembly and examined in a TEM to obtain an independent determination of the particle size and morphology.

Figure 3: Photograph of a partially disassembled Thermal Precipitator Module with the inlet manifold exposed on the left, the hot wire leads in the red Vespel block in the middle and the outlet valves on the right. The drawing on the right displays the position of the hot wire above the TEM grid with the air flow passing over both. The heated wire drives aerosol particles onto the grid via thermophoresis.

Calibration of the moment instruments was essential to properly interpret the flight data and was performed on the ground before the flight. Calibration was accomplished using two different aerosol generators one using monodisperse particle generation using dioctyl phthalate (DOP) according to the approach by Mulholland and Liu and the other using polystyrene spheres. The monodisperse droplet generator functioned by producing a spray of DOP diluted with isopropanol which is then evaporated and recondensed producing monodisperse droplets. The droplet size is controlled by the DOP dilution level. The generator will operate stably for tens of minutes. The aerosol from the generator was sampled simultaneously by the SAME instrument under test and a reference instrument. For the number count, the reference instrument was a TSI 3022A particle counter, for the mass concentration, a Tapered Element Oscillating Microbalance and for the first moment a TSI Electrical Aerosol Detector was used. The results for the P-Trak are shown in Fig. 4. As the number concentration increased, the effect of the particle diameter became more evident. Separate correlations were developed for each particle size and the closest correlation was used to analyze the flight data based on the initial estimates of the average particle size.

The first moment device, the ion chamber showed little effect of particle size as seen in Fig 5. Consequently a single correlation was used for all particle sizes. The third moment device (TSI- DustTrak®) is theoretically predicted to show non-monotonic behavior as particle size is increased and the response varies with the particle refractive index. This issue was addressed by calibrating the DustTrak for the smoke particulate from each material. Although the average particulate size is expected to change, the refractive properties are expected to remain the same. In the results reported here, the DustTrak® response was directly calibrated, for each smoke source, against
mass concentration measurements using a tapered element oscillating microbalance. Using estimated material densities, the diameter of average mass was computed for each sample.

For SAME-2, one of the DustTrak devices was replaced by a new device the MPASS (Multi-Parameter Aerosol Scattering Sensor) which is an optical scattering sensor, employing two detectors placed at fixed angles. The underlying design and performance have been reported previously. The basic configuration is similar to the DustTrak, however, the MPASS has two angles of detection and it can measure two moment approximate moment quantities specifically the 3rd and 2nd moments. Further, the optical geometry is optimized using a Mie-scattering model for aerosols with light scattering properties typical of smoke: i) geometric mean diameters from 0.1 – 1.0 microns; ii) geometric standard deviations of 1.6 – 1.9; iii) the aerosols have lognormal size distributions and vi) a fixed, real value for refractive index of 1.6. This device was included as a technology demonstration of an aerosol sensor that measures two moments simultaneously. The underlying assumption is that by measuring two moments of the aerosol it is possible to correct the smoke detector signal for the particle size, unlike previous spacecraft smoke detectors which are limited to a single moment measurement. This will enable the alarm to be less sensitive to larger particles that are typical of dust. The relative advantage of this approach was discussed in more detail in a prior paper.

![Figure 4](image_url)

**Figure 4**: P-Trak Calibration results with Mono disperse DOP droplets.

![Figure 5](image_url)

**Figure 5**: Ion Chamber Calibration results with Mono disperse DOP droplets.
C. Experimental Results

Over 100 sample materials were tested. These were comprised of samples of 6 materials: Teflon™, Kapton™, Pyrell™, silicone rubber, cellulose (lamp wick), and dibutyl-phthalate deposited on a porous wick. The test conditions included multiple sample temperatures and air flow rates. The baseline air flow rate was 8 cm/s with runs conducted at rates down to 2 cm/s with a limited set conducted with no airflow while the sample was heated. The sample temperatures were based on the pyrolysis properties of the material and were thus material specific. The baseline temperature was selected to produce 1 to 2 mg of weight loss in 60 seconds and have reasonable signal on each of the moment instruments. To examine the effect of sample temperature, additional higher temperature conditions were also selected that would not saturate the moment instruments. The TEM grids recovered from the thermal precipitators on SAME-1 were unfortunately contaminated with extraneous particles whose origin is currently under investigation. For SAME-2 more rigorous assembly controls were followed and numerous good particle images were obtained. For typical runs, the particle concentrations were a few million particles / cc and the mass concentrations were 2 to 15 mg/m³ however there were runs with fewer particles. For these conditions, the typical numbers of particles sampled by the instruments were in the billions although for the test with the lowest particle concentrations the sample sizes were of the order of several million particles.

1. Moment Instrument Results

Geometric mean, count mean, and average mass diameter results from the moment instruments for baseline runs are presented in Table 1. As described above, the count mean diameter and the diameter of average mass measurements are independent of the nature of the size distribution however, the calculation of the geometric mean diameter and the geometric standard deviation ($\sigma_g$) depend on the assumption of a log-normal distribution. This assumption cannot be directly tested with the available data since this would require a device that can separate the particles into size bins. However, there are two indications that the distribution can be approximated by a log-normal distribution. In the cases listed below, the count mean diameter is less than the diameter of average mass which is a necessary condition for a log-normal distribution (See Fig. 1). The second indication is that the geometric standard deviations are all physically reasonable (i.e. greater than 1 (see Eq. 3) and less than approximately 3.5 which is the practical limit on the magnitude of aerosol geometric standard deviation). The aging results show a substantial increase in the observed diameters. As expected the increase in the count mean diameter is greater than the corresponding increase in the diameter of average mass. This can be understood since coagulation consists of particle collisions joining two particles. For a polydisperse aerosol, these collisions usually involve the collision of one of the smallest (high mobility) particles with the largest (high cross-section) particles. This collision has little effect on the size of the large particle but it removes the smaller particles from the size distribution. This has a larger effect on the count mean diameter since it is linear in diameter while the diameter of average mass varies with the cube root of the diameter. Theoretical predictions suggest that this process will produce a near lognormal distribution and that the standard deviation will tend toward a self-preserving distribution with a value of 1.32.\(^{14}\)

<table>
<thead>
<tr>
<th>Material</th>
<th>Unaged</th>
<th>Aged 720 s</th>
<th>Unaged</th>
<th>Aged 720 s</th>
<th>Unaged</th>
<th>Aged 720 s</th>
<th>Unaged</th>
<th>Aged 720 s</th>
</tr>
</thead>
<tbody>
<tr>
<td>Geometric Mean Diameter (Dg) (µm)</td>
<td>0.042</td>
<td>0.089</td>
<td>0.090</td>
<td>0.229</td>
<td>0.128</td>
<td>0.269</td>
<td>0.081</td>
<td>0.293</td>
</tr>
<tr>
<td>Count Mean Diameter (M1/M0) (µm)</td>
<td>0.056</td>
<td>0.109</td>
<td>0.128</td>
<td>0.276</td>
<td>0.196</td>
<td>0.355</td>
<td>0.101</td>
<td>0.359</td>
</tr>
<tr>
<td>Diameter of Average Mass (M3/M0) (µm)</td>
<td>0.101</td>
<td>0.161</td>
<td>0.258</td>
<td>0.398</td>
<td>0.465</td>
<td>0.619</td>
<td>0.170</td>
<td>0.539</td>
</tr>
<tr>
<td>$\sigma_g$</td>
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<td>1.872</td>
<td>2.312</td>
<td>1.834</td>
<td>2.530</td>
<td>2.108</td>
<td>2.198</td>
<td>1.892</td>
</tr>
</tbody>
</table>

Table 1: Diameter results for baseline runs for each material (Diameters are in micrometers).
Overall, the Teflon and Kapton particles were very small. The lamp wick and silicone results exhibited substantially larger diameters of average mass, particularly for the aged cases. As reported previously\textsuperscript{10}, the Dust Trak results for lamp wick and silicone demonstrated substantial portions of the particle distribution possessed aerodynamic diameters larger than 1 micrometer. The comparatively much smaller diameters for the Kapton and Teflon will make detection of this smoke challenging for light scattering devices, on the other hand the large sizes seen with the lamp wick and silicone would generate very large signals on a light scattering system. As discussed by Urban et al.\textsuperscript{13}, to have a system that would have the signal gain sensitive enough to trigger an alarm for the lower signal from small particles such as Teflon and Kapton without excessive nuisance alarms from large dust particles is difficult without some level of particle size discrimination. This suggests that detection of these particles against the background environment will require a detection system capable of measuring more than one moment of the particle size distribution.

The results for Teflon, Kapton and lampwick are plotted versus duct velocity in figures 6, 7 and 8. These figures show a strong trend of increased particle size at low flow rates. This indicates that smoke trapped in an avionics enclosure can be expected to be substantially larger than smoke released in ventilated areas.

![Figure 6: Particle diameters as a function of air velocity for lampwick. The lines are linear (least square) fits. The repeatability of the diameter measurements is estimated to be $\pm/\mp 15\%$.](image6)

![Figure 7: Particle diameters as a function of air velocity for Kapton. The lines are linear (least square) fits. The repeatability of the diameter measurements is estimated to be $\pm/\mp 15\%$.](image7)
MPASS Results

The MPASS was included in SAME-2 to demonstrate the effectiveness of a two-moment smoke detector. Because of how they are used, a smoke detector is not calibrated for specific materials and instead has a single calibration for all conditions. The objective is to maintain a flat sensor response, regardless of the type of smoke or the particle size so the signal is proportional only to a selected moment (e.g. mass or number concentration). The approach used was to derive correlations to predict the smoke number concentration and mass concentration from the MPASS output. This correlation was made using the other SAME measurements (number and mass concentration) as a reference. A least-squares, third-order, correlation was computed on all test points with a signal to noise ratio greater than 2 independent of all other parameters (e.g. material, heating rate, and aging time).

The results are shown in figures 9 and 10, for prediction of the mass concentration and the number concentration, respectively. Ideally these response graphs should be horizontal lines. The normalized output is plotted along with the normalized correlation result as a function of geometric mean diameter. Ideally the normalized result should be horizontal lines. The uniformity of response (scatter) is expressed by the least squares variance relative to the mean (Δ) and is shown in the legend. Both figures show that the correlation decreases the relative variance by roughly a factor of at least 2 relative to that provided by a conventional single-output sensor. The resulting variance for the mass response is substantially lower than for the number concentration. This is consistent with the fact that the original optimization of the MPASS was to mass concentration not to number concentration. Thus with appropriate processing, the MPASS output can be proportional to either the number concentration or the mass concentration over a wide range of particle sizes and material compositions. This independence demonstrates that by combining 2 moments the MPASS is able to scale its response to the particle size being sampled giving a more accurate indication of the amount of smoke present than a single moment detector. The MPASS therefore has the ability to discern the difference between a strong signal from a high smoke concentration and a strong signal from a lower concentration of very large dust particles, a feature which can minimize false smoke detector alarms.
3. TEM Results

Overall, 72 particle samples were taken on Transmission Electron Microscope (TEM) grids using the thermal precipitator. These samples were imaged with a TEM to evaluate the particle morphology as a function of flow conditions, material and heating level. Overall, silicone rubber and DBP presented no particles that were visible via TEM, presumably because the particles were condensed liquid that evaporated before they could be imaged in the TEM. Figure 11 presents typical images for the major materials.

As seen in Table 1, aging did increase the average particle size, however this level of change is not readily visible in the TEM images, instead the most visible change is the disappearance of the smallest particles. (Fig 12) As described in the discussion of Table 1, this is consistent with expectations since the smaller particles have the highest mobility and consequently are readily incorporated into the larger particles. Particularly for the semi-liquid particles, this absorption of the smallest particles into larger particles is expected to have little or no apparent effect on the size of the larger particles because the diameter is the cube root of the volume. In the case of non-liquid particles, the smallest particles will nevertheless be unlikely to be visible via TEM when they are attached to larger particles.
4. Impact of flow on particle morphology

As the samples are heated by the wire, chemical bonds are broken and gaseous products are released via pyrolysis. These products may include partially oxidized species resulting from the reaction with oxygen. As the gases move from the wire, the temperature and species concentration will decrease as a result of conduction/diffusion and convection. With the decrease in temperature, the super saturation ratio of the pyrolysis gas will pass a critical stage and nucleation of smoke particles will occur. These smoke nuclei grow rapidly from the condensation of the pyrolysis gases. This process may continue for time scales of up to $10^3$ s. If the particles are liquid, the colliding particles will coalesce and form larger droplets. This is the case for the lampwick, silicone rubber, and Kapton smokes (Fig. 11). If the particles are solid, they will retain their structure as they stick resulting in an agglomerate structure composed of many primary particles connected in a low density fractal-like structure. This is the case for the Teflon and Pyrell smokes (Fig. 11).

Key parameters affecting the smoke properties of the smoke produced here are the low-gravity condition, the convective flow past the heated wire, the wire temperature, and the aging time. As the wire temperature increases, more of the sample will be pyrolyzed and these gases will subsequently condense on the nuclei leading to a larger mass concentration of the aerosol and a larger diameter of average mass.

An increase in the convective flow for a fixed wire temperature is expected to result in an increase in the pyrolysis rate and thus an increase in the mass concentration of the aerosol. Some results follow this trend with silicone rubber, lampwick, and Pyrell increasing slightly with flow while Teflon and Kapton decreases. A possible explanation is that holding the wire temperature fixed does not necessarily mean that the sample surface temperature distribution is precisely the same at different air flow speeds. Although these are expected to be small variations, given the exponential dependence of the pyrolysis kinetics on temperature, changes in the heat transport environment could change the sample surface temperature between the wire coils.

A case of special interest is the case of zero-flow. The transport for the zero-flow case in $\mu$-g is controlled by molecular diffusion and thermal expansion. Over a period of 60 seconds, the gases would diffuse a distance on the order of 4 cm. Assuming a cubical geometry this would correspond to the smoke being confined to a volume of about 64 cm$^3$ rather than being diluted in the 6000 cm$^3$ piston. If the same number of particles is generated as for the flow case, the initial concentration will be on the order of 50 times higher for the zero-flow case. This is expected to lead to more coagulation and a larger particle size. It is found for all of the materials the diameter of average mass is 30% to 40% larger for the zero-flow tests compared to the flow tests.

The zero flow condition was not in the original design of the SAME hardware but was examined for 11 tests by modifying the valve operation in the SAME hardware. This resulted in a very high concentration of particulate where the aging process was accelerated compared to that in the aging chamber at dilute conditions. An unfortunate side effect was that approximately half the smoke was lost from the system because it diffused out of the inlet into
the MSG volume. As a result the total particle counts were lower than normal and the TEM grids had few particles. It is important to recognize that this condition is unique to low-gravity scenarios. If one performed the same experiment at 1-g, the buoyancy of the rising gases would greatly increase the mixing zone. So the larger particle diameters observed in these zero-flow conditions are important to consider in the development of improved fire detection equipment for space vehicles.

For the zero-flow case, the heat is transported primarily by conduction rather than convection. This is likely to result in a significant increase in the gas temperature relative to the case with 8 cm/s air flow. A striking effect of this temperature change is the change in the morphology of the Pyrell smoke from fractal-like clusters with up to 30 primary spheres for the case with a flow of 8 cm/s to spherical particles for the zero-flow condition (Fig 13).

![TEM images showing the effect of flow for Pyrell and lampwick under 8 cm/s air flow and quiescent conditions.](image)

**III. Conclusions**

The results from the SAME experiment indicate that substantial differences in the size distributions of the smoke particles are evident for the five spacecraft materials that were tested. For the conditions of the SAME experiment, all samples produced significant numbers of sub-micron particulate that are readily detected using an ionization smoke detector. However, the aging results suggest that some materials such as silicone rubber can produce smoke that can reach sizes better detected by light scattering techniques. The zero flow conditions produced smoke particles that were 2 to 3 times larger than those produced at the nominal 8 cm/s flow rate. Since spacecraft fire conditions include an even wider array of materials than those tested here and a broader range of flow and temperature conditions, broad smoke aerosol size distributions can be expected from credible pre-fire overheat events. These results suggest that detection methods that can measure more than one moment of the size distribution may show more successful detection and false alarm rejection than single moment detectors. The results from the prototype 2-moment detector (MPASS) were very encouraging in this respect with the technique showing real potential as a means to respond to a broad particle range while avoiding excessive false alarms from nuisance sources. The TEM result demonstrated that fire particulate can be expected to have a broad range of morphologies influenced by heating rate, material and flow conditions.

**IV. Acknowledgments**

The support of the SAME experiment team and the crews of ISS increments 10, 13, 15, and 24 are gratefully acknowledged. The SAME project was conducted through the ISS Exploration Research Project of the Exploration Technology Development Program. The support of this program is also acknowledged.

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V. References