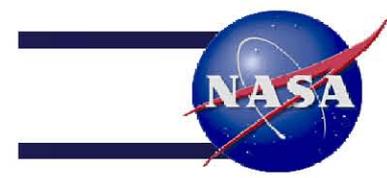




THE TRANSPORT OF MASS, ENERGY, AND
ENTROPY IN CRYOGENIC SUPPORT STRUTS FOR
ENGINEERING DESIGN



Thermal and Fluids Analysis Workshop

August 13-17, 2012, Pasadena, California

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NASA Glenn Research Center

Thermal Systems Branch

August 8, 2012

1 Abstract

Engineers working to understand and reduce cryogenic boil-off must solve a variety of transport problems. An important class of nonlinear problems involves the thermal and mechanical design of cryogenic struts. These classic problems are scattered about the literature and typically require too many resources to obtain. So, to save time for practicing engineers, the author presents this essay. Herein, a variety of new, old, and revisited analytical and finite difference solutions of the thermal problem are covered in this essay, along with commentary on approach and assumptions. This includes a few thermal radiation and conduction combined mode solutions with a discussion on insulation, optimum emissivity, and geometrical phenomenon. Solutions to cooling and heat interception problems are also presented, including a discussion of the entropy generation. And the literature on the combined mechanical and thermal design of cryogenic support struts is reviewed with an introduction to the associated numerical methods.



The transport of mass, energy, and entropy in cryogenic support struts for engineering design

J.P. Elchert, NASA Glenn Research Center,
Thermal Systems Branch

Presented By
J.P. Elchert

Thermal & Fluids Analysis Workshop
TFAWS 2012
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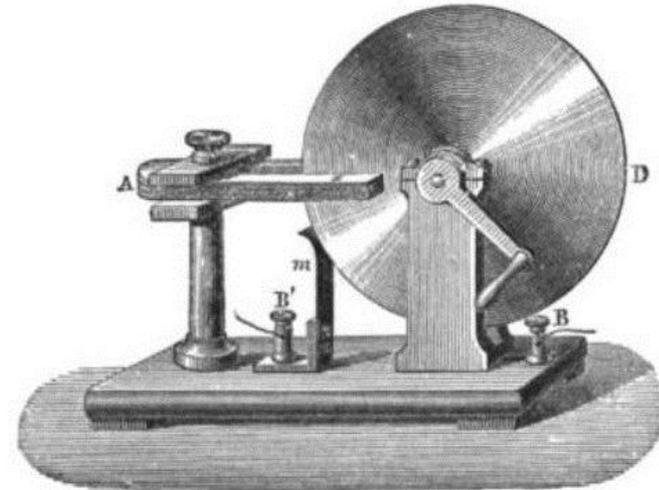
Royal Institution of Great Britain





$$\oint_{\partial\Sigma} \mathbf{E} \cdot d\boldsymbol{\ell} = - \iint_{\Sigma} \frac{\partial \mathbf{B}}{\partial t} \cdot d\mathbf{S}$$

1831, simple DC generator





Massachusetts's Institute of Technology



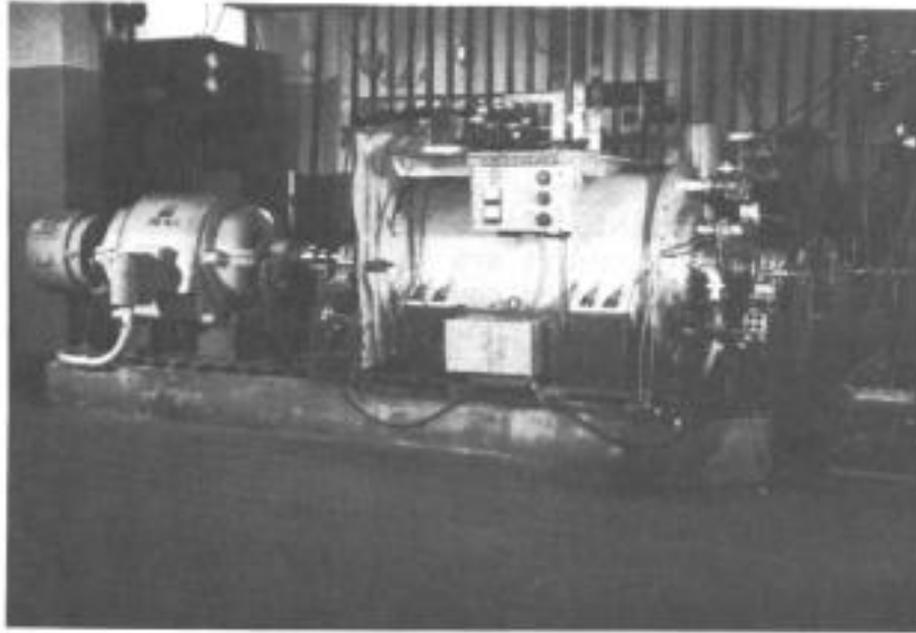


Fig. 9 View of 3-MVA Alternator and 150-HP Drive Motor

- In the 1970s, MIT Electric Power Research Institute did significant research into superconducting synchronous generators, including cryogenics engineering research



MIT's prototype synchronous generator

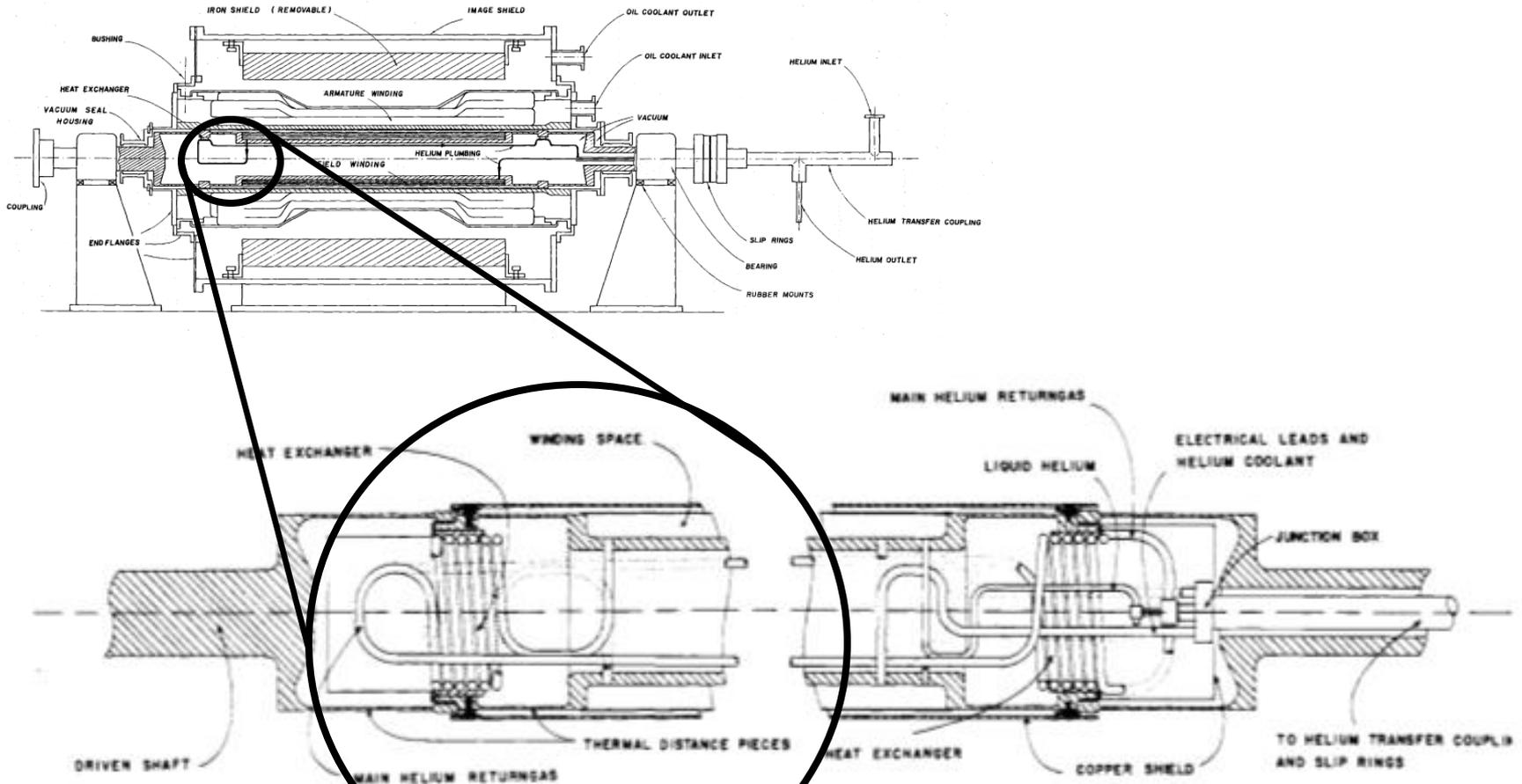
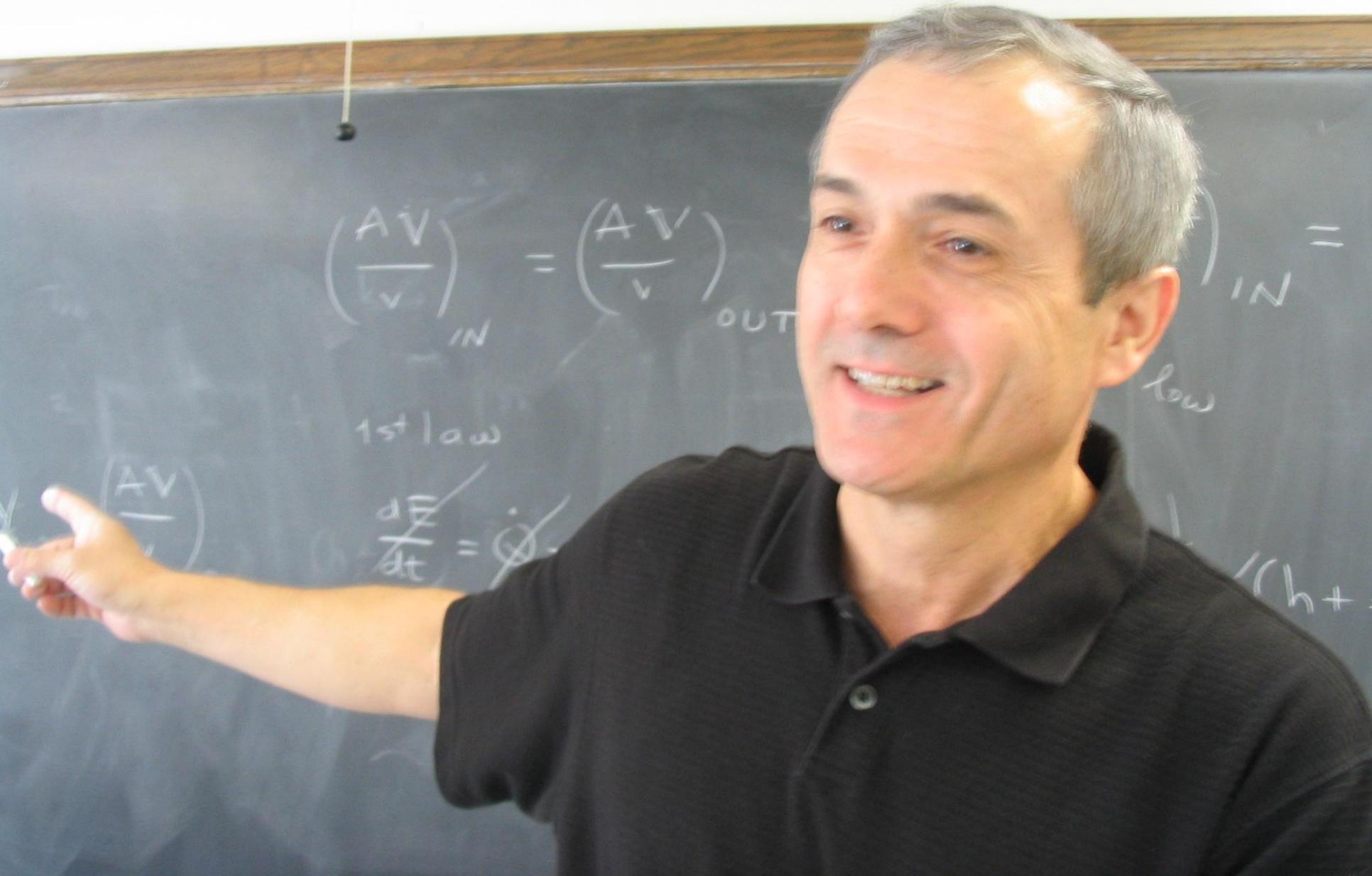


Fig. 2 Rotor Helium Flow Circuit

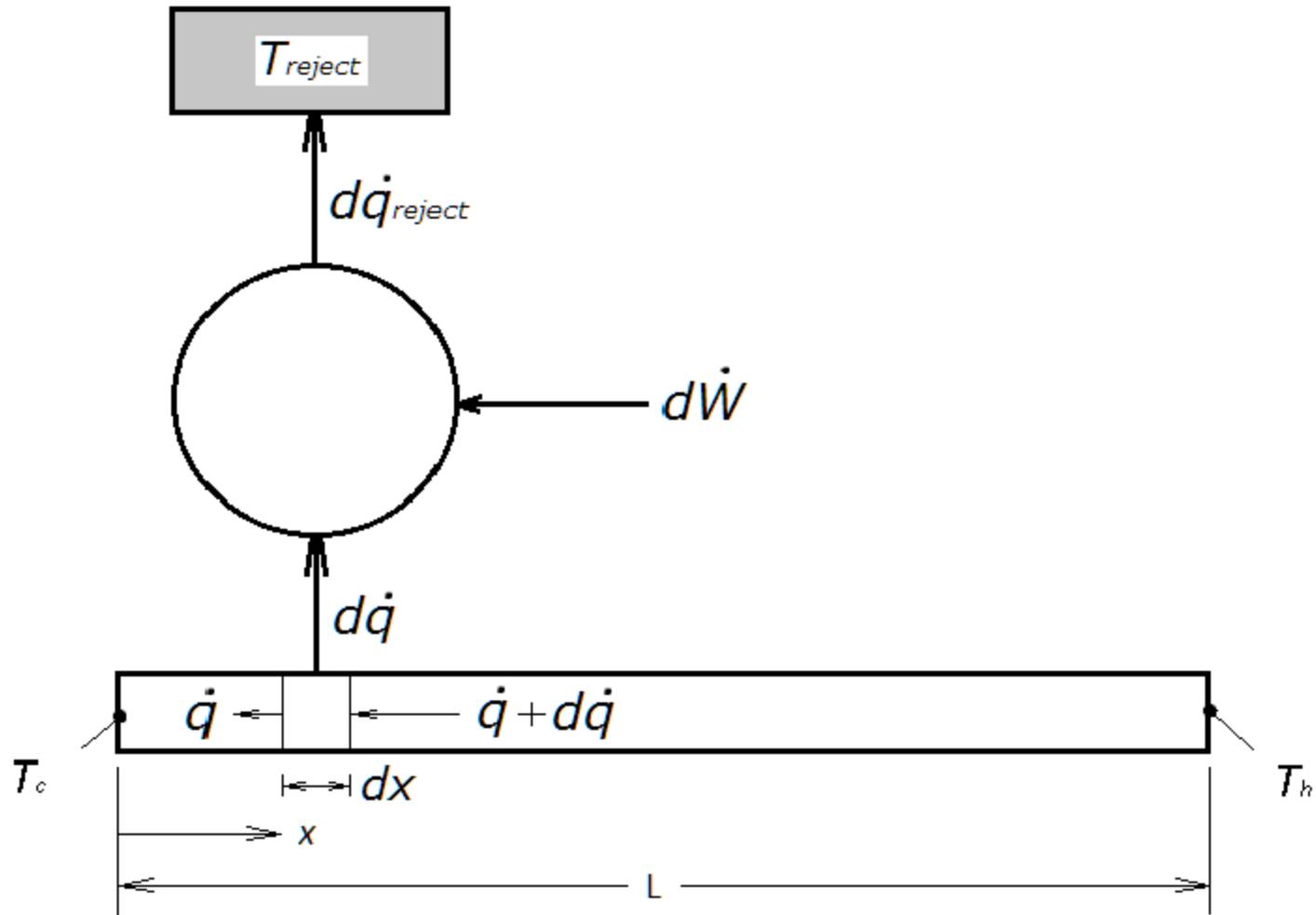
Dr. Adrian Bejan

J.A. Jones Professor, Duke Department of Mechanical Engineering and Materials Science





Adrian Bejan (1975)





Thermodynamic Minimum



$$\dot{S} = \frac{A}{L} \left(\int_{T_c}^{T_h} \frac{\sqrt{k}}{T} dT \right)^2 = \frac{kA}{L} \left[\ln \left(\frac{T_h}{T_c} \right) \right]^2$$

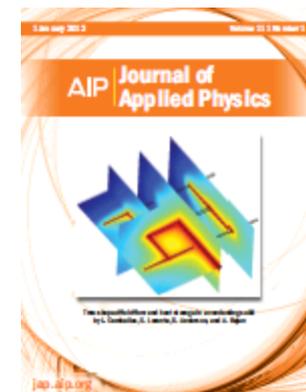
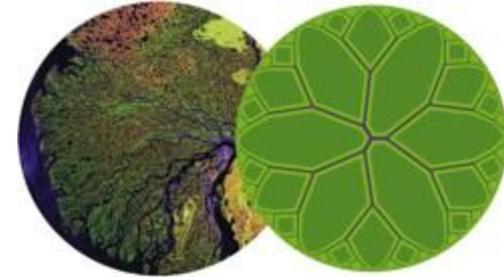
The most efficient cooling scheme generates the least entropy

$$\dot{q}_{optimum} = T\sqrt{k} \left(\frac{A}{L} \right) \left(\int_{T_c}^{T_h} \frac{\sqrt{k}}{T} dT \right)$$

$$\dot{W} = \frac{A}{L} T_h \left(\int_{T_c}^{T_h} \frac{\sqrt{k}}{T} dT \right)^2 = T_h \frac{kA}{L} \left[\ln \left(\frac{T_h}{T_c} \right) \right]^2$$



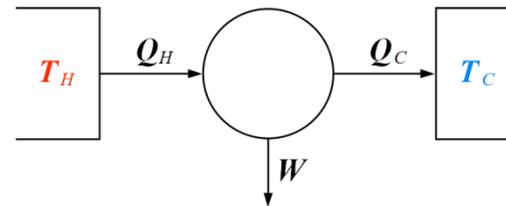
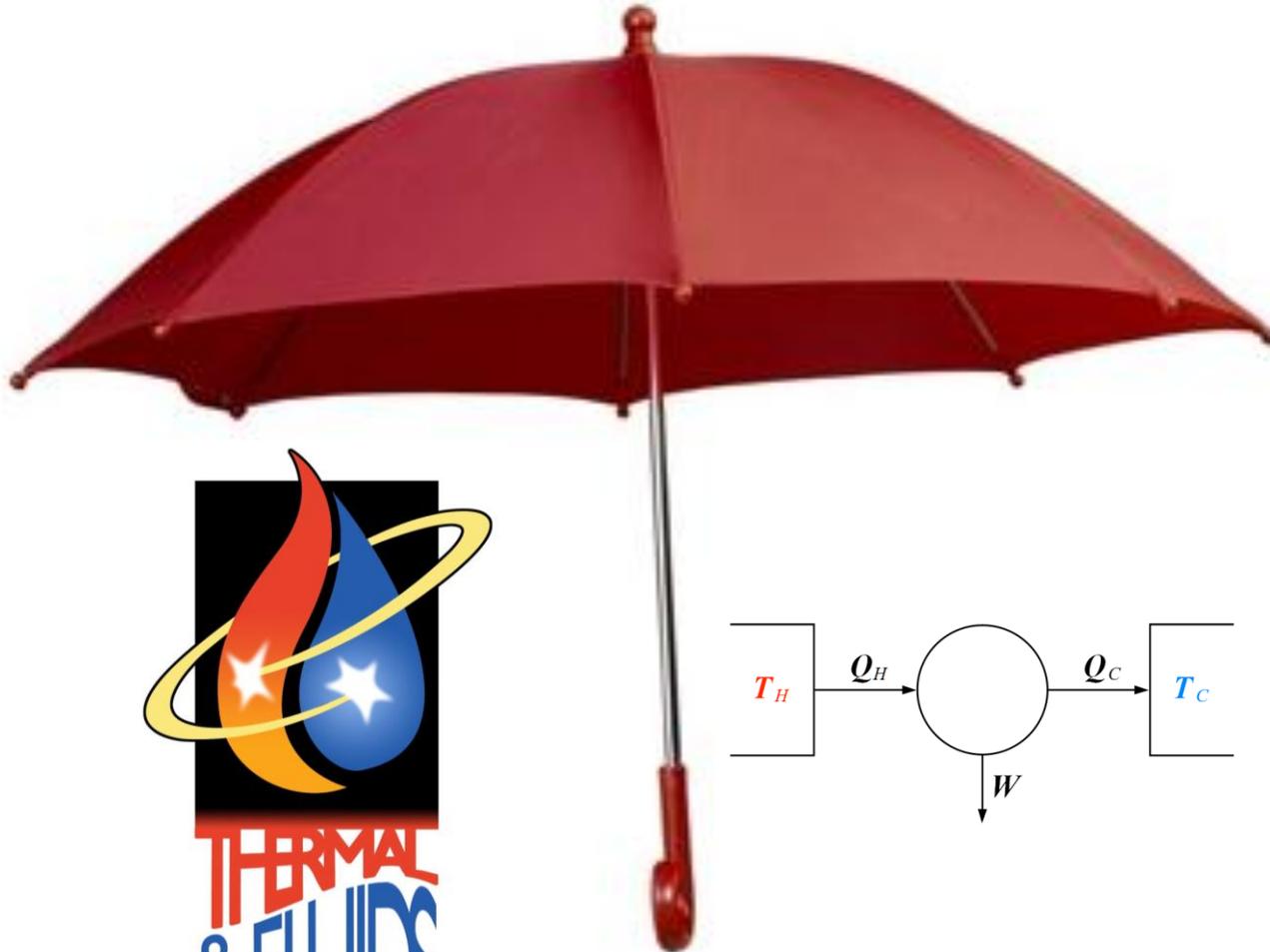
Constructal Law



Finite-size systems evolve progressively easier access to the imposed current



Unifies thermal/fluids/thermodynamics





Continuous cooling (solution by Tsao)



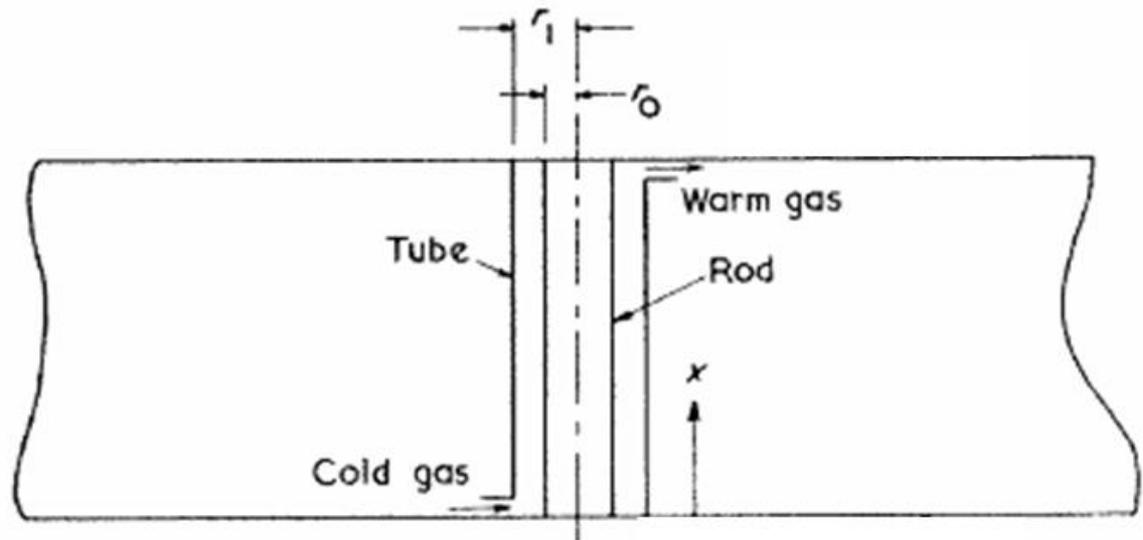
$$\frac{d}{dx} \left(A_c k(T) \frac{dT}{dx} \right) - Ph(T) [T(x) - T_m(x)] = 0$$

$$\dot{m} c_P \frac{dT_m}{dx} - Ph(T) [T(x) - T_m(x)] = 0$$

$$T(0) = T_L$$

$$T(L) = T_H$$

$$T_m(0) = T_o$$



$$\eta = x/L$$

$$\gamma^2 = 4\alpha^2 + \beta^2$$

$$\tau_f = \frac{(T_f - T_c)}{(T_h - T_c)}$$

$$\beta = \frac{hPL}{\dot{m}c_p}$$

$$\tau_s = \frac{(T_s - T_c)}{(T_h - T_c)} \quad \alpha = \left(\frac{L/(kA)}{1/(hPL)} \right)^{1/2}$$

Boundary conditions

$$T_s(x = 0) = T_c \Rightarrow \tau_s(\eta = 0) = \frac{T_s(x = 0) - T_c}{T_h - T_c}$$

$$T_s(x = L) = T_h \Rightarrow \tau_s(\eta = 1) = \frac{T_s(x = L) - T_c}{T_h - T_c}$$

$$T_f(x = 0) = T_c \Rightarrow \tau_f(\eta = 0) = \frac{T_f(x = 0) - T_c}{T_h - T_c}$$

Governing Equations

$$\frac{d^2\tau_s}{d\eta^2} = \alpha^2(\tau_s - \tau_f)$$

$$\frac{d\tau_f}{d\eta} = \beta(\tau_s - \tau_f)$$

Solutions

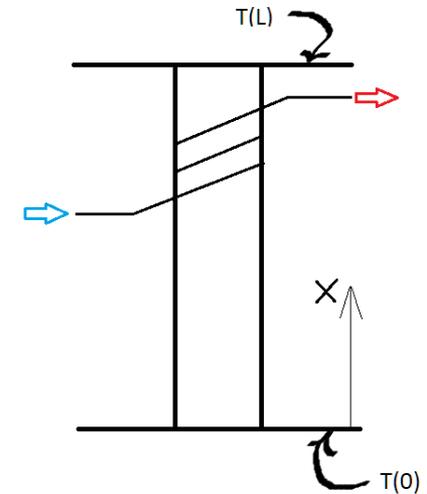
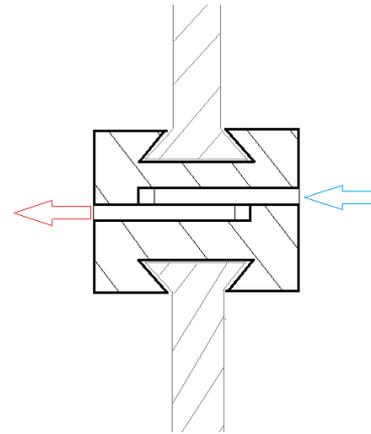
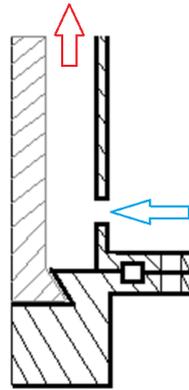
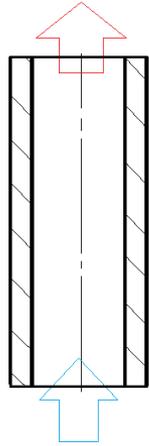
$$\tau_s = \frac{(\gamma + \beta)^2 e^{\frac{\gamma - \beta}{2}\eta} - 4\gamma\beta + (\gamma - \beta)^2 e^{-\frac{\gamma + \beta}{2}\eta}}{(\gamma + \beta)^2 e^{\frac{\gamma - \beta}{2}\eta} - 4\gamma\beta + (\gamma - \beta)^2 e^{-\frac{\gamma + \beta}{2}\eta}}$$

$$\tau_f = 2\beta \frac{(\gamma + \beta)e^{\frac{\gamma - \beta}{2}\eta} - 2\gamma - (\gamma - \beta)e^{-\frac{\gamma + \beta}{2}\eta}}{(\gamma + \beta)^2 e^{\frac{\gamma - \beta}{2}\eta} - 4\gamma\beta + (\gamma - \beta)^2 e^{-\frac{\gamma + \beta}{2}\eta}}$$

$$\frac{\dot{q}''}{k(T_h - T_c)/L} = \left. \frac{d\tau_s}{d\eta} \right|_{\eta \rightarrow 0} = \frac{\gamma(\gamma^2 - \beta^2)}{(\gamma + \beta)^2 e^{\frac{\gamma - \beta}{2}\eta} - 4\gamma\beta + (\gamma - \beta)^2 e^{-\frac{\gamma + \beta}{2}\eta}}$$



Various cooling schemes



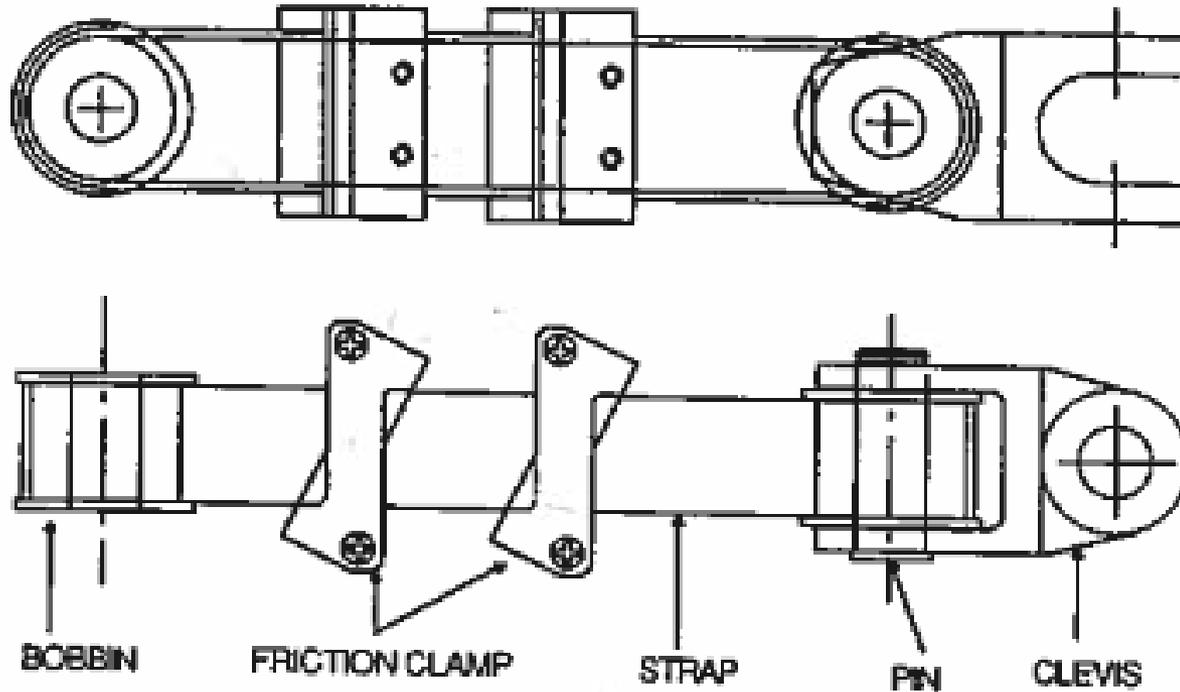
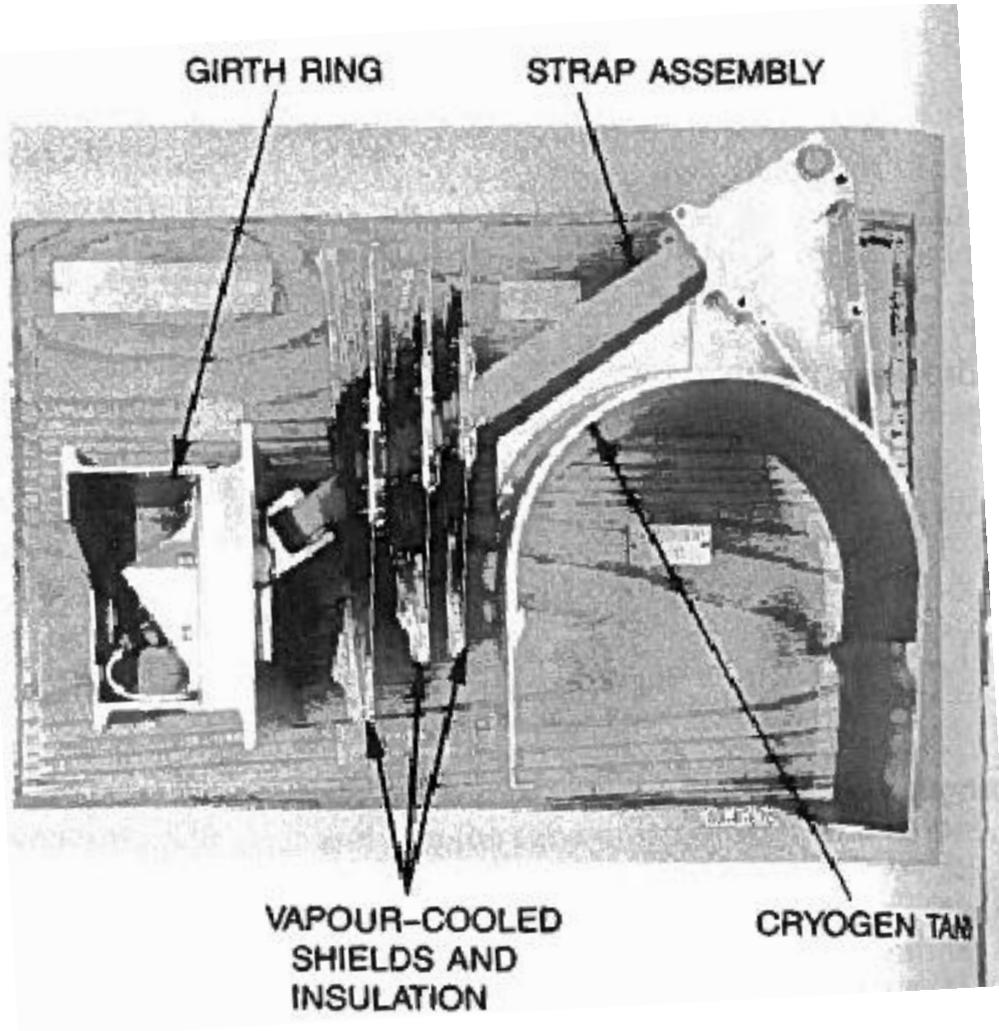


Figure 2 SHOOT support strap with friction clamp

K.F. Weintz et al. “SHOOT Dewar Support Strap Design”



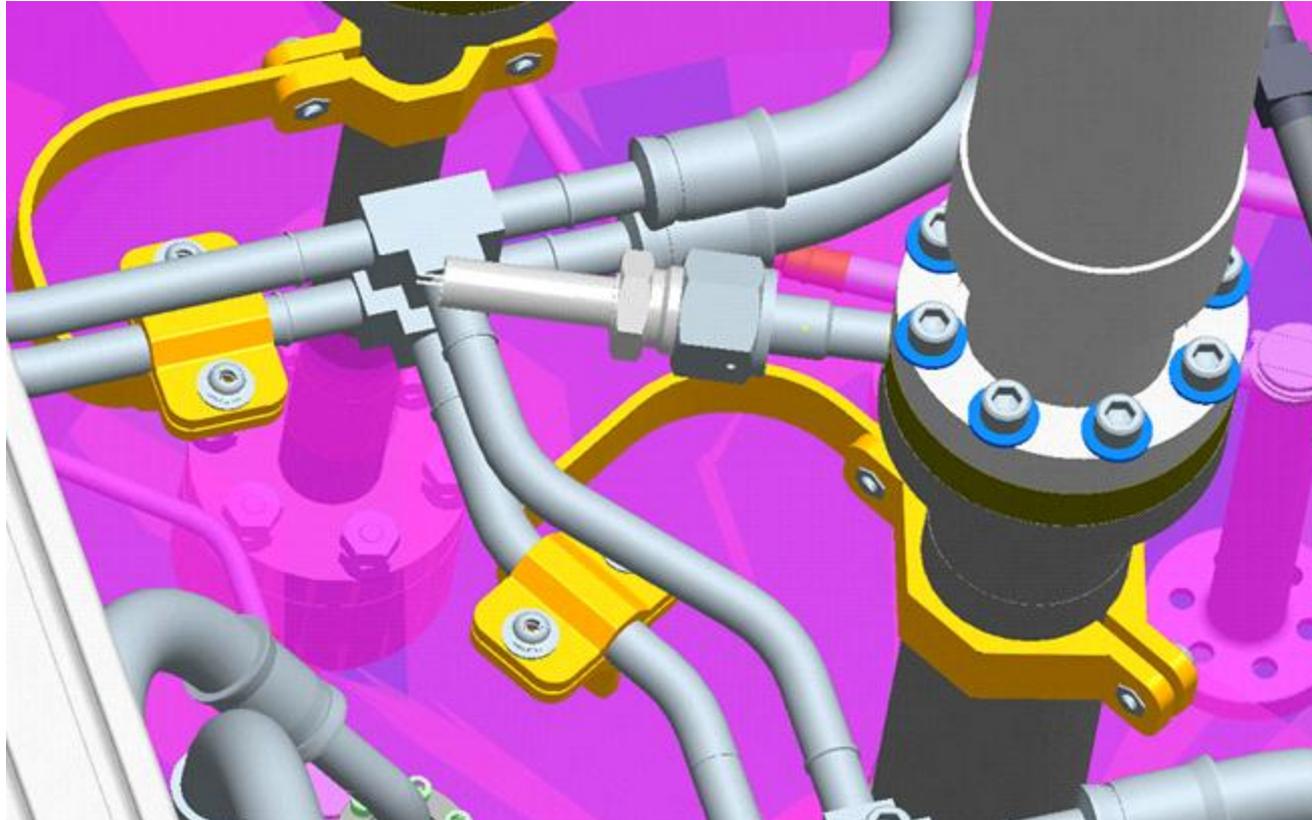
Cooling scheme application: shield mounted



T. Hirokawa et al., “Design of support strap with advanced composite for cryogenic application”

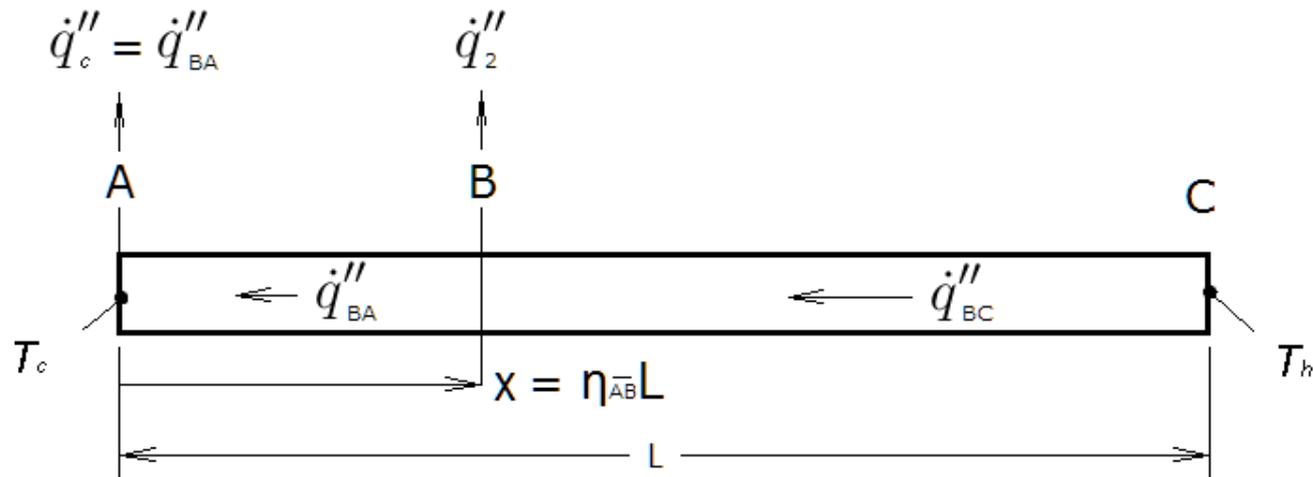
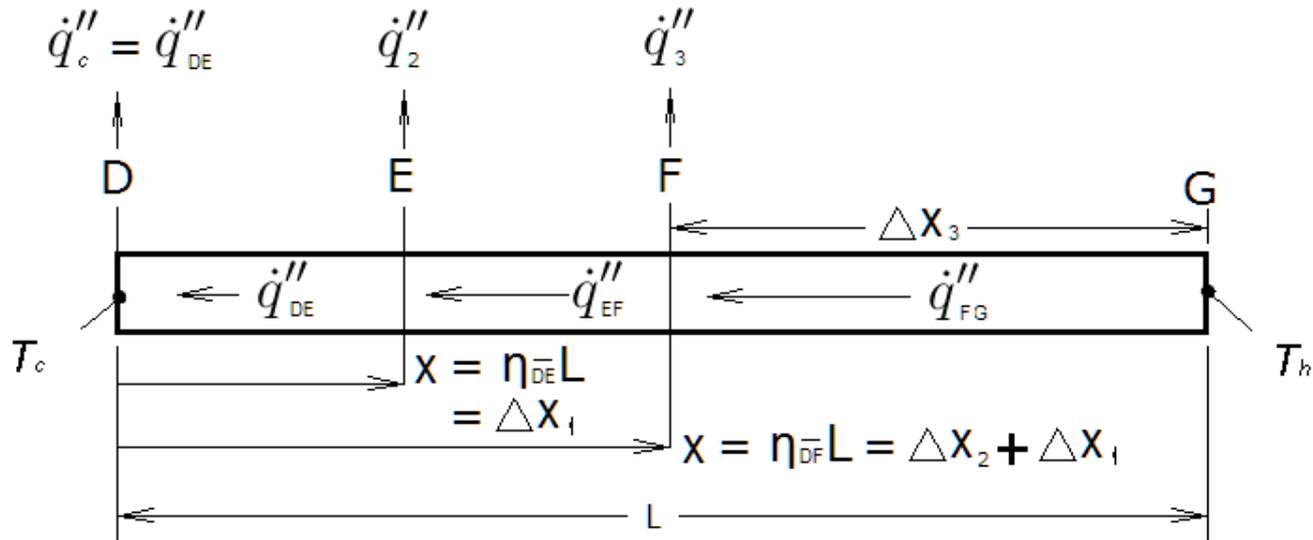


Another cooling scheme: thermal links





Discrete, two stage, perfect heat transfer





Hilal and Boom (1977)



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M. A. Hilal and R. W. Boom

Table II. Optimum Temperatures, Locations, and Refrigeration Power for Finite Number of Shields

Number of shields	Cycle*	T_1, K	T_2, K	T_3, K	$\Delta x_1/L$	$\Delta x_2/L$	$\Delta x_3/L$	$\Delta x_4/L$	$PL/A, W/cm$
304 Stainless steel ($T_c = 4.2 K$)									
1	C	39.7	—	—	0.338	0.662	—	—	445
1	A	39.7	—	—	0.338	0.662	—	—	1781
2	C	21.6	81.7	—	0.189	0.334	0.477	—	316

* C, Carnot cycle efficiencies; A, actual cycle efficiencies.

In my estimate, the second row is actually correct. Because solving the Carnot case gives $T_2 = 40K$ and $\eta_{\overline{AB}} = 0.351$ and when I tested a sample real coefficient of performance, I found roughly $T_2 = 39K$ and $\eta_{\overline{AB}} = 0.33$. So the first row—the Carnot case—was misprinted. Hilal and Boom actually had solutions up to four stages and also studied the same solutions for Narmco 570 cloth, but that information, being irrelevant, was omitted.



Hilal and Boom (1977)



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M. A. Hilal and R. W. Boom

Table II. Optimum Temperatures, Locations, and Refrigeration Power for Finite Number of Shields

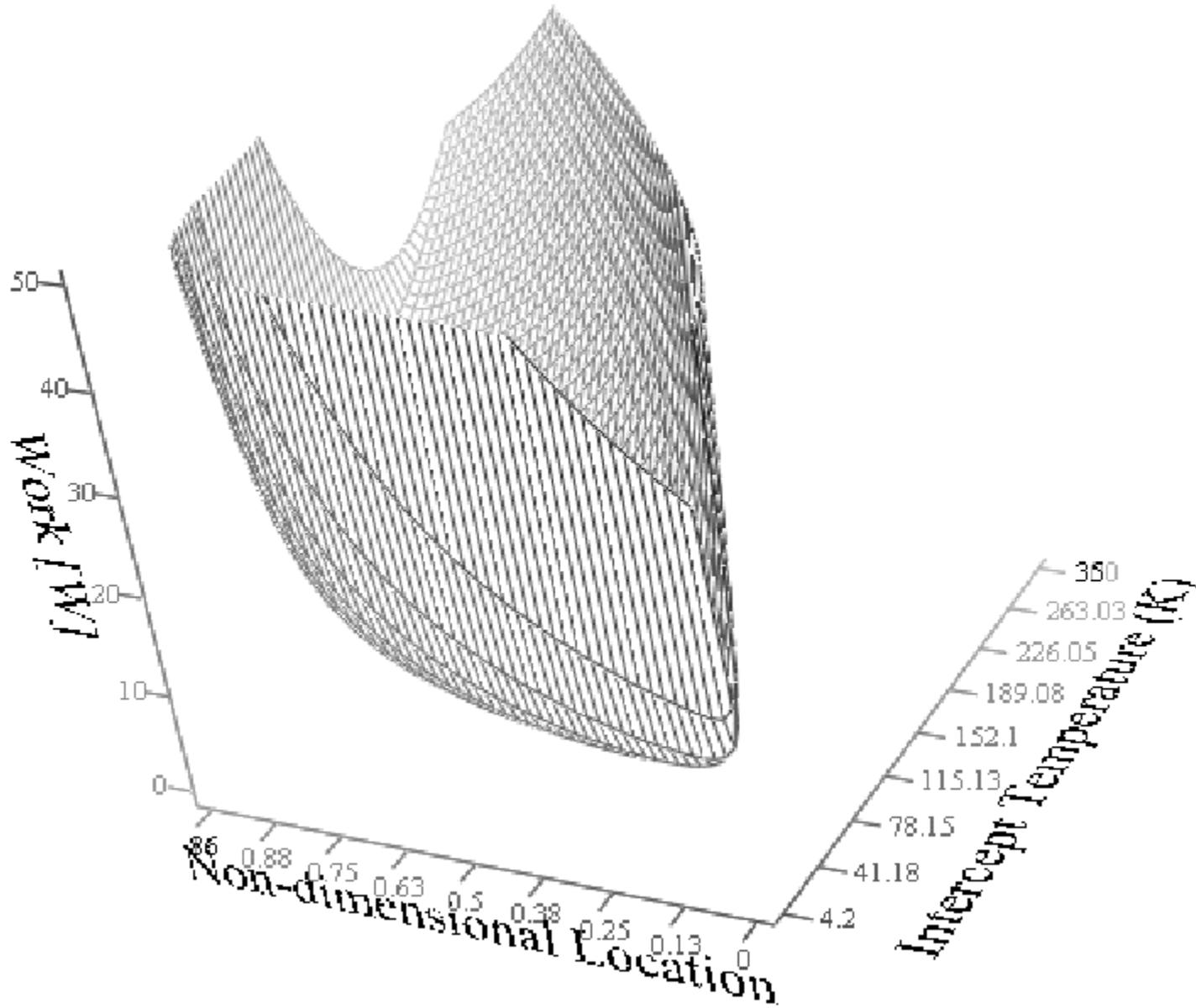
Number of shields	Cycle*	T_1, K	T_2, K	T_3, K	$\Delta x_1/L$	$\Delta x_2/L$	$\Delta x_3/L$	$\Delta x_4/L$	$PL/A, W/cm$
304 Stainless steel ($T_c = 4.2 K$)									
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1	A	39.7	—	—	0.338	0.662	—	—	1781
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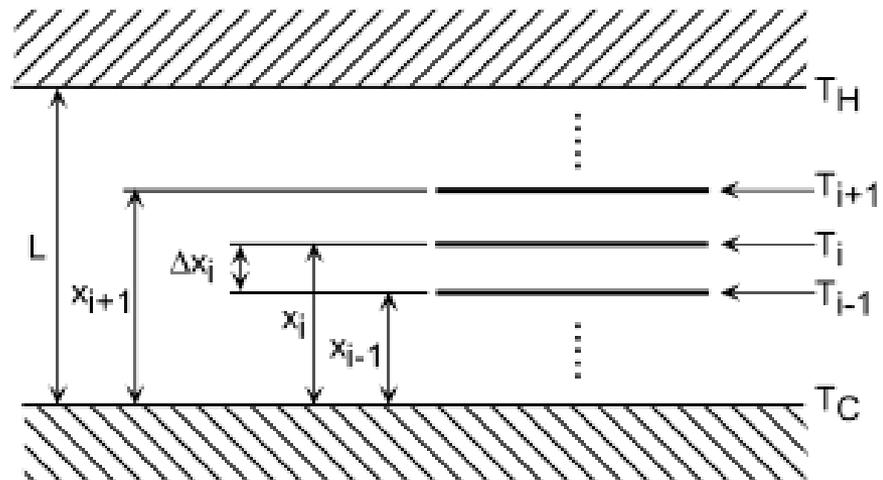
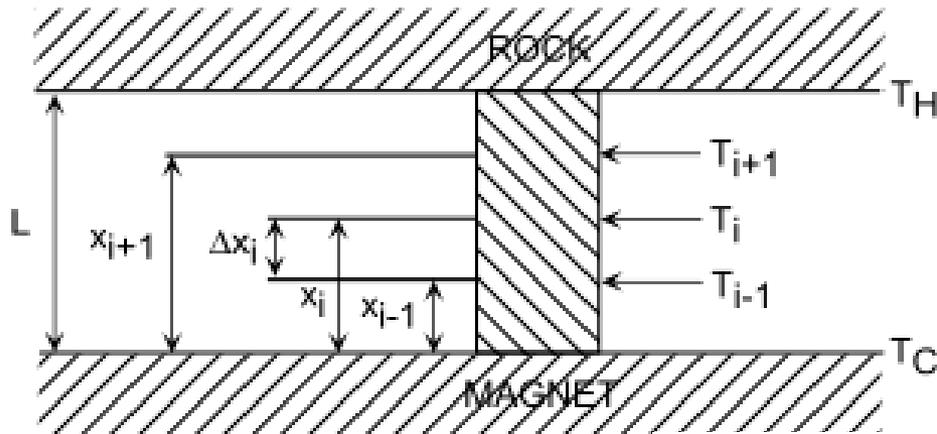
* C, Carnot cycle efficiencies; A, actual cycle efficiencies.

Number of Stages	$T_c (K)$	$T_2 (K)$	$T_3 (K)$	η_{AB}	η_{DE}	η_{DF}
2	4.2	40.0	-	0.351	-	-
3	4.2	21.1	80.7	-	0.194	0.524



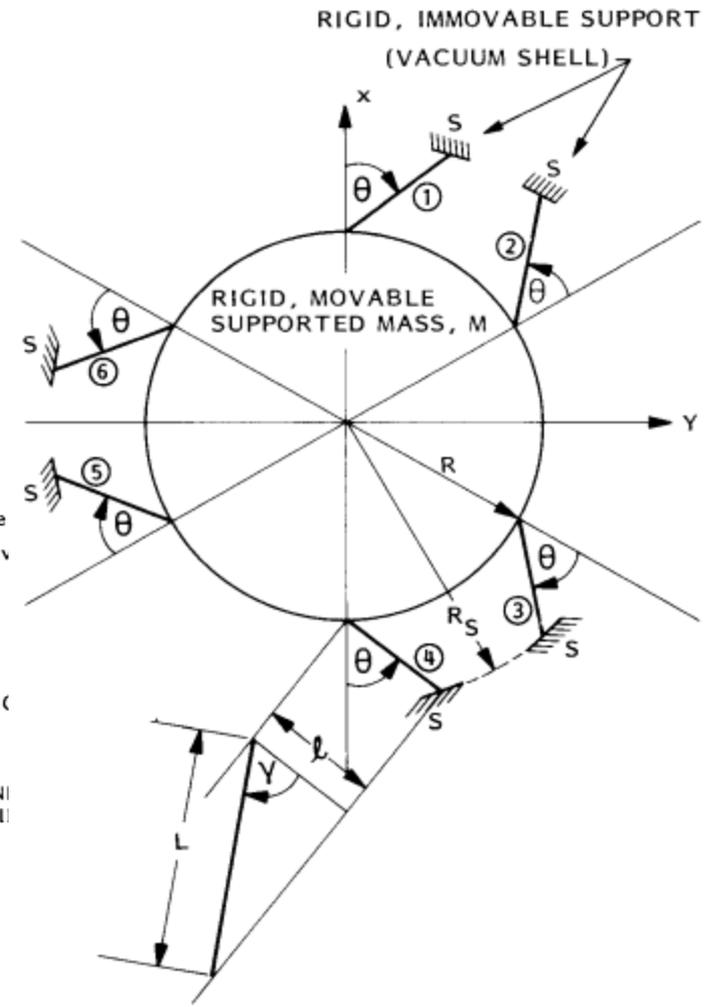
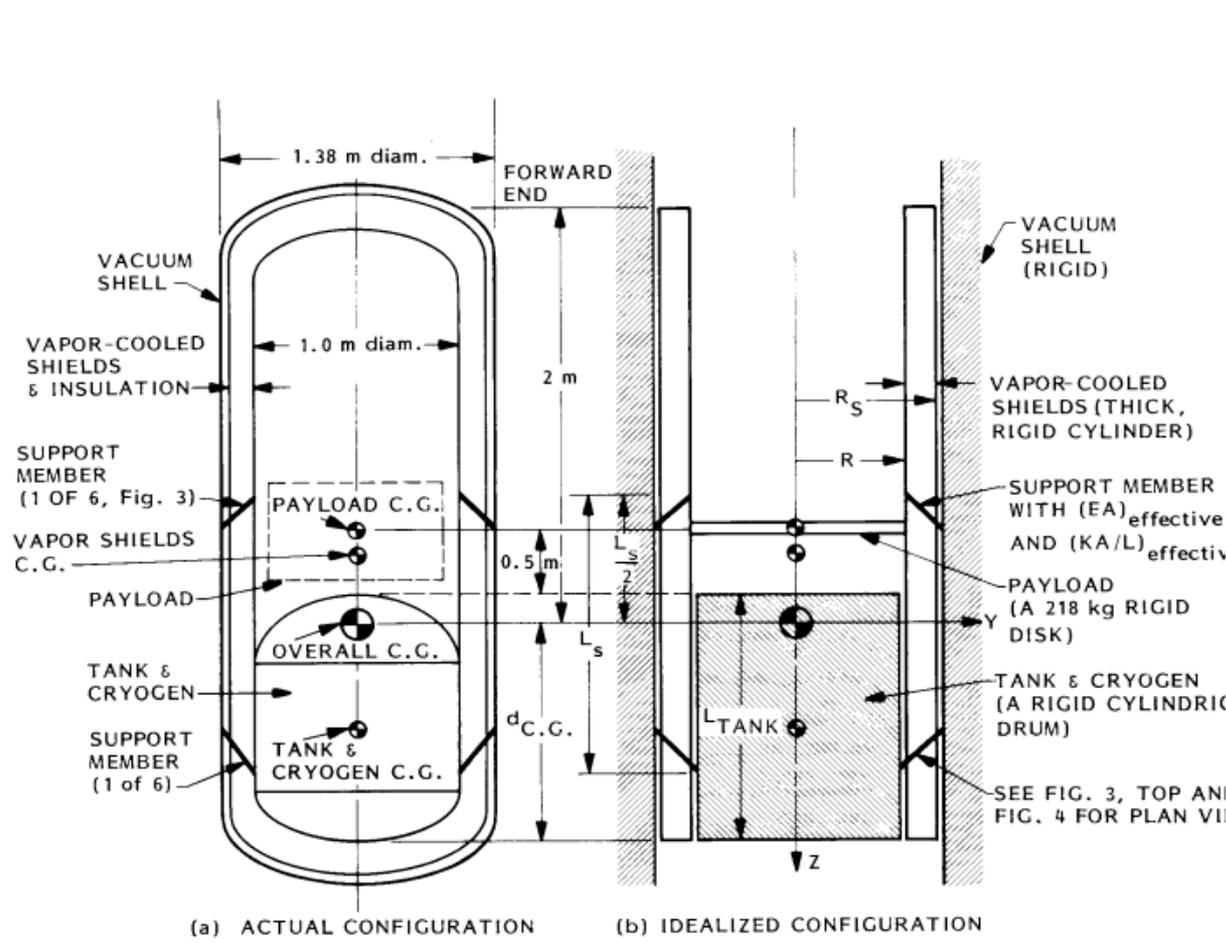
Carnot, two stage, ideal; graphical solution

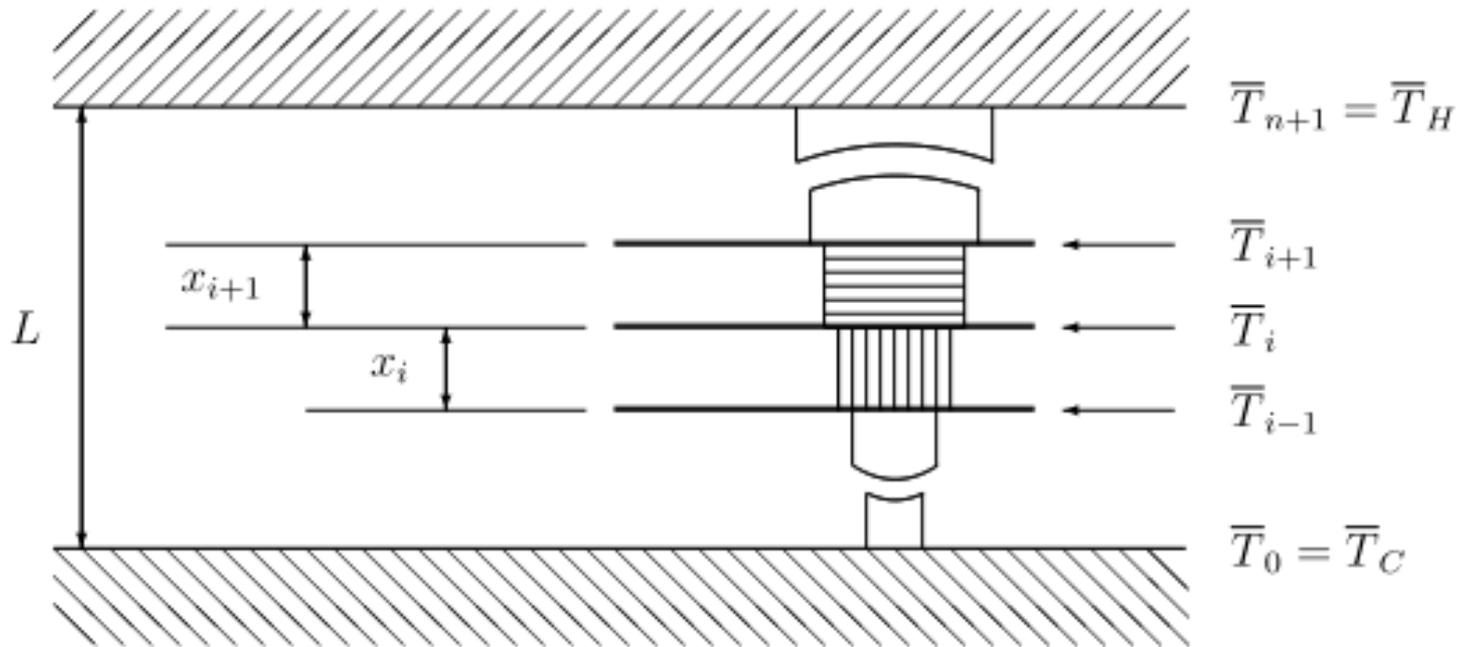






Strut vibration optimization, David Bushnell (1984)









Bibliography



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- All opinions are those of the presenter and not necessarily those of NASA
- Kingsley, Wilson, Kirtley, Keim, Smith, Thullen, “Steady-state electrical tests on the MIT-ERPI 3-MVA Superconducting Generator”, IEEE Transactions on Power Apparatus and Systems, Vol. PAS-95, no. 3, May/June 1976

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