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# **Integrated Ground Operations Demonstration for Responsive Space Access**

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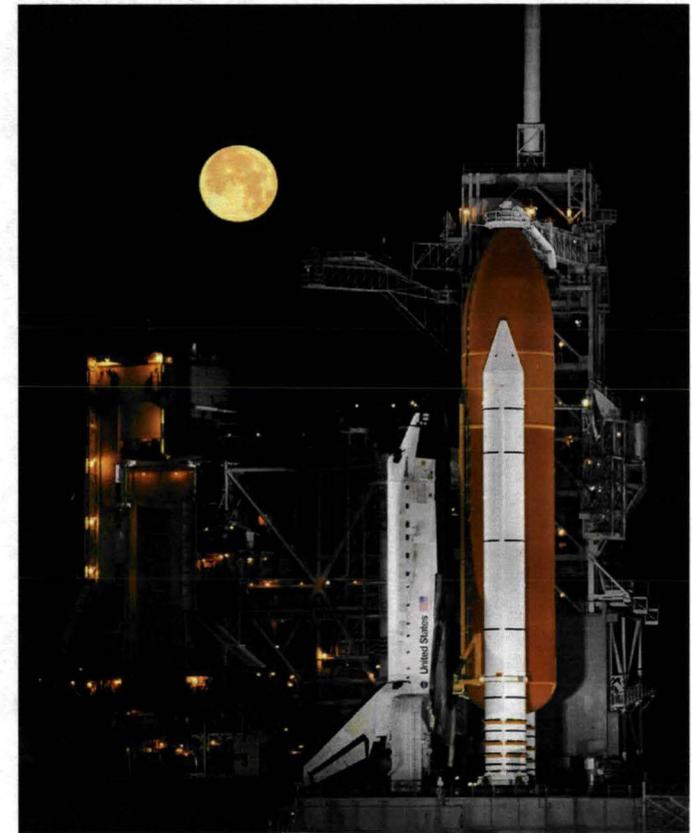


# Advanced Exploration Systems



- NASA's Advanced Exploration Systems (AES) program is pioneering new approaches to:
  - Rapidly develop prototype systems and subsystems
  - Mature and demonstrate key capabilities
  - Validate operational concepts for future human missions beyond Earth orbit
- Program goals will be achieved largely through of a set of vital and prioritized mission-capability-focused tasks
- Teams are comprised almost entirely of NASA civil servants (25 - 40 FTE)
- Limited procurement funding (\$1M - \$6M)
- Typically last 1 to 3 years to drive a rapid pace of progress
- 23 proposals selected across the Agency, including Integrated Ground Operations Demonstration Units (IGODU) project lead by KSC with participation from ARC, GRC and SSC

- Cryogenic propellant loading operations and associated flight and ground systems are complex and critical to launch ops
- Sizeable portion of lifecycle costs of any launch program
  - Space Shuttle program's cost for cryogenic propellants at KSC and SSC was over \$20 million/year between 2006-2009
  - 100+ propulsion/cryogenics experts monitoring propellant loading operations across the country on launch day
- NASA operations for handling cryogenics in ground support equipment have not changed substantially in 50 years
  - Heritage Apollo era equipment



- Integrated Ground Operations Demonstration Units (IGODU) project developed to mature, integrate and demonstrate advancements in cryogenics, system health management and command and control technologies
- Two Distinct Testing Environments:



GODU Integrated Refrigeration and Storage- GODU LH2



GODU Autonomous Control - GODU LO2

## Scope

- **GODU LH2:**
  - Investigate alternative storage and distribution architecture for future cryogenic propellant operations
  - Demonstrate advanced cryogenic propellant handling operations (liquefaction, storage and distribution) of normal boiling point and sub-cooled cryogenic propellants
- **GODU LO2:**
  - Develop and demonstrate advanced control and health management technologies and techniques to autonomously control cryogenic propellant servicing operations
  - Investigate modern COTS hardware and control systems in an effort to reduce the “standing army” of engineers associated with maintaining and operating ground systems through the use of health management and autonomous control technologies

## Goals

- Raise Technology Readiness Levels (TRL) and Integration Readiness Levels (IRL) of several key technology development areas
- Reduce operations lifecycle costs of future test programs and launch complexes
- Demonstrate technologies for future exploration beyond low earth orbit
- Serve as test environments for extraterrestrial surface operations

- Kennedy Space Center (KSC) and Stennis Space Center (SSC) lose approximately 50% of hydrogen purchased
  - Continuous heat leak into storage and transportation vessels
  - Transient chilldown of warm cryogenic equipment
  - Liquid bleeds to maintain interface temperatures
  - Ullage losses during venting processes

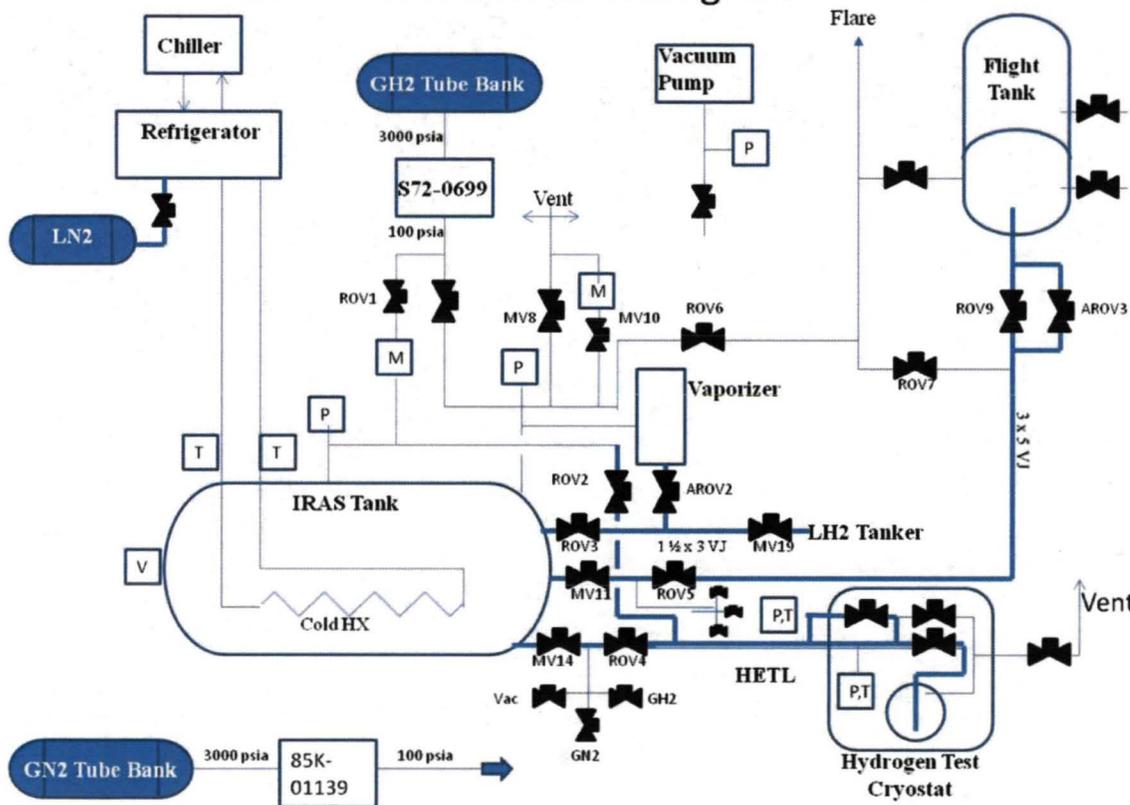


Liquid Hydrogen Storage at KSC Launch Pad



Liquid Hydrogen in Use at SSC Test Stand

GODU LH2 Functional Diagram



- The key innovation of the GODU LH2 system is the integration of a cryogenic refrigerator into the LH2 storage tank.
- GODU LH2 will increase the IRL of an advanced LH2 system
  - No new component technology development
  - Industry standards in place of NASA GSE standards
- Reusing spare and excess equipment wherever possible

- Commercial industry has made significant advancements in cryogenic refrigeration systems, specifically in Integrated Refrigeration and Storage (IRAS)
  - Large scale refrigeration systems in superconducting industry, but not for propellant storage
- IRAS concept allows liquid hydrogen to be stored in a quasi-equilibrium state
  - Cools liquid directly at storage site
    - Allows for control of the bulk temperature of the fluid as opposed to pressure control of the ullage using vent and relief valves
    - Enables easier depressurization of tank ullage pressure and bulk fluid conditioning for greater vehicle loading control
  - Operates with refrigeration capacity to system heat leak ratio greater than 1
- Concept has been proven at 180 liter scale using a COTS Gifford McMahon Cryocooler in partnership with the Florida Solar Energy Center
  - Evaluation of storage and handling characteristics of liquid hydrogen during in-situ liquefaction
  - Thermal stratification data collected during pressurization/depressurization cycles

# GODU LH2 Objectives

- Primary Objective: Demonstrate efficient LH2 operations on a relevant scale that can be projected to future spaceport architectures by demonstrating the following:
  - Zero loss storage and transfer of LH2 at a relevant scale
  - Hydrogen densification in IRAS storage tank and loading of simulated flight tank
  - Hydrogen liquefaction using closed cycle helium refrigeration
- There are numerous secondary objectives, including:
  - Creation of a semi-portable densified LH2 servicing capability from excess equipment
  - Can be used at test stands, development centers, launch sites
  - Provide opportunity for retention of skills
  - Serve as a means of data generation for the Launch Services Program cryogenic tank thermodynamic model validation
  - Demonstrate modern component technology such as electric actuators, vapor shielded cryogenic piping, advanced instrumentation, and low helium consumption operations

# Need for Autonomous Control Loading and Health Monitoring Systems

- Cryogenic propellant loading operations for the Shuttle program typically involved several large pockets of specialized engineers and managers from across the country (KSC, JSC, MSFC, Michoud and Rocketdyne).
  - Monitor propellant ground and flight systems during the critical final hours prior to launch.
  - Hazardous nature, complexity, criticality, and vulnerability of the launch vehicle and associated ground servicing systems
  - Lack of insight into system health



STS-135 Pre-Loading Simulation Team at KSC

# Overview of GODU LO2

- The GODU Autonomous Control system resides at the Cryogenic Test Laboratory (CTL) at KSC and consists of
  - Small-scale Simulated Propellant Loading System (SPLS)
    - LN2 system simulating LO2 system
    - Propellant storage tank (6kgal)
    - Pumping complex (0-400 gpm)
    - Control valve skids
    - Vehicle simulator tank (500 gallon and 2 kgal)
- Testing environment for advanced cryogenic components, sensors, and health management technologies.
- Demonstration of a control system capable of recognizing and automatically correcting simple system failures typical of heritage launch vehicle servicing systems.
  - Instrumentation failures
  - Thermal anomalies
  - Component failures
- Allows comparison, down select and maturation of the best ground health management and autonomous control concepts from ARC, SSC and KSC on typical cryogenic propellant servicing hardware.



Simulated Propellant Loading System (SPLS) at the KSC  
Cryogenics test Laboratory

- Demonstrate autonomous control of a sub-scale vehicle loading operation
- Demonstrate recognition of common system faults and anomalies and recover without human intervention
- Evaluate tools and techniques in real world application to advance health management and autonomous control technologies for future applications
- Demonstrate scalability and extensibility by replicating autonomous control of the 6,000 gallon LO2 simulator system to a larger system
  - GODU LH2
  - Army/NASA SWORDS LO2/LCH4 propellant servicing systems

- **GODU LH2**
  - LN2 Test Series (October '13)
  - LH2 Test Series (FY14)
  - Densified LH2 Test Series (FY14)
- **GODU LO2**
  - Automated Testing Series (July-Sept. '13)
  - Autonomous Testing Series (Oct – Dec '13)
  - Conversion to LOX and LCH4 UPSS system (FY14)
  - Propellant Loading for sub-orbital flight of Army SWORDS vehicle (FY14)