



**AIAA/IEEE Energy Tech/INCOSE 2014 Conference
Convention Center – Cleveland, OHIO**

July 28 to 30, 2014

***Gas Turbine Energy Conversion Systems for
Nuclear Power Plants applicable to LiFTR “Liquid Fluoride
Thorium Reactor” Technology***

Session 11 – Advanced Terrestrial & Space Power Generation

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NASA - GRC, Cleveland, Ohio**



Session 11 Outline

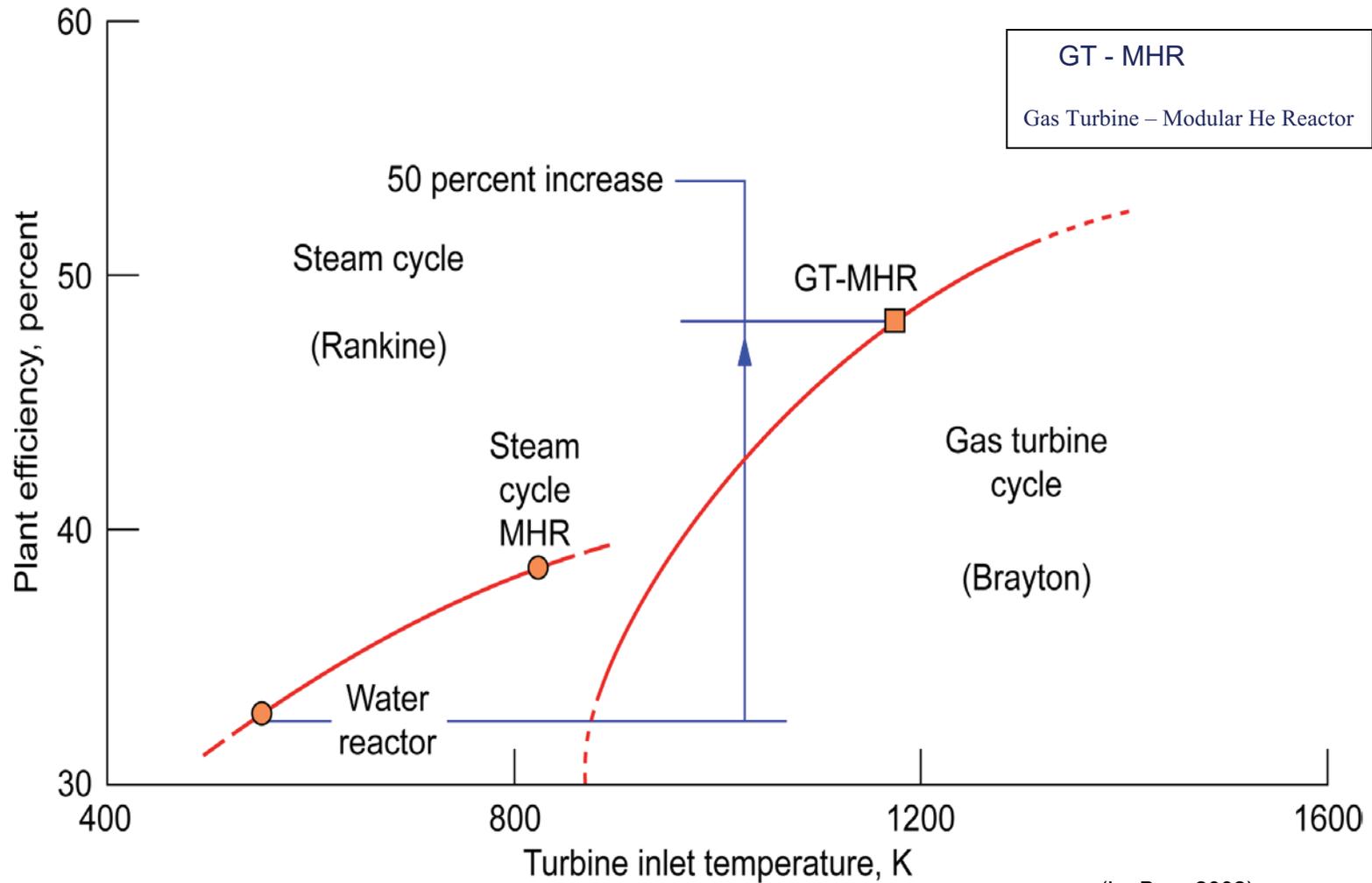
This panel plans to cover thermal energy and electric power production issues facing our nation and the world over the next decades, with relevant technologies ranging from near term to mid-and far term.

- Although the main focus will be on ground based plants to provide baseload electric power, energy conversion systems (ECS) for space are also included, with solar- or nuclear energy sources for output power levels ranging tens of Watts to kilo-Watts for unmanned spacecraft, and eventual mega-Watts for lunar outposts and planetary surface colonies. Implications of these technologies on future terrestrial energy systems, combined with advanced fracking, are touched upon.
- Thorium based reactors, and nuclear fusion along with suitable gas turbine energy conversion systems (ECS) will also be considered by the panelists. The characteristics of the above mentioned ECS will be described, both in terms of their overall energy utilization effectiveness and also with regard to climactic effects due to exhaust emissions.

Background & Motivation

- Advanced Nuclear Power named in Energy Policy Act of 2005 (Gen IV Nuclear Systems Initiative)
 - Gen. Baseload GW's Power -> Solve **Depletion of Earth's HC** (hydro-carbon) Resources
 - Ameliorate world climate problems by eliminating **Greenhouse Gas Emissions**
- Thorium Nuc. Power Plant Construction (100 GWe over next 25 yrs) would **re-invigorate US and World economies**
- Gen IV Candidate Advanced Nuclear Power Plants
 - Gas Turbine Modular He Reactor (GT-MHR) Systems - **Space**
 - Liquid Fluoride Thorium Reactor (LFTR) GT Systems - **Terrestrial**
- High Temperature Gas Turbine Power plants offer **large Thermal Efficiency improvement** over Steam plants

Energy Conversion Cycle Comparisons



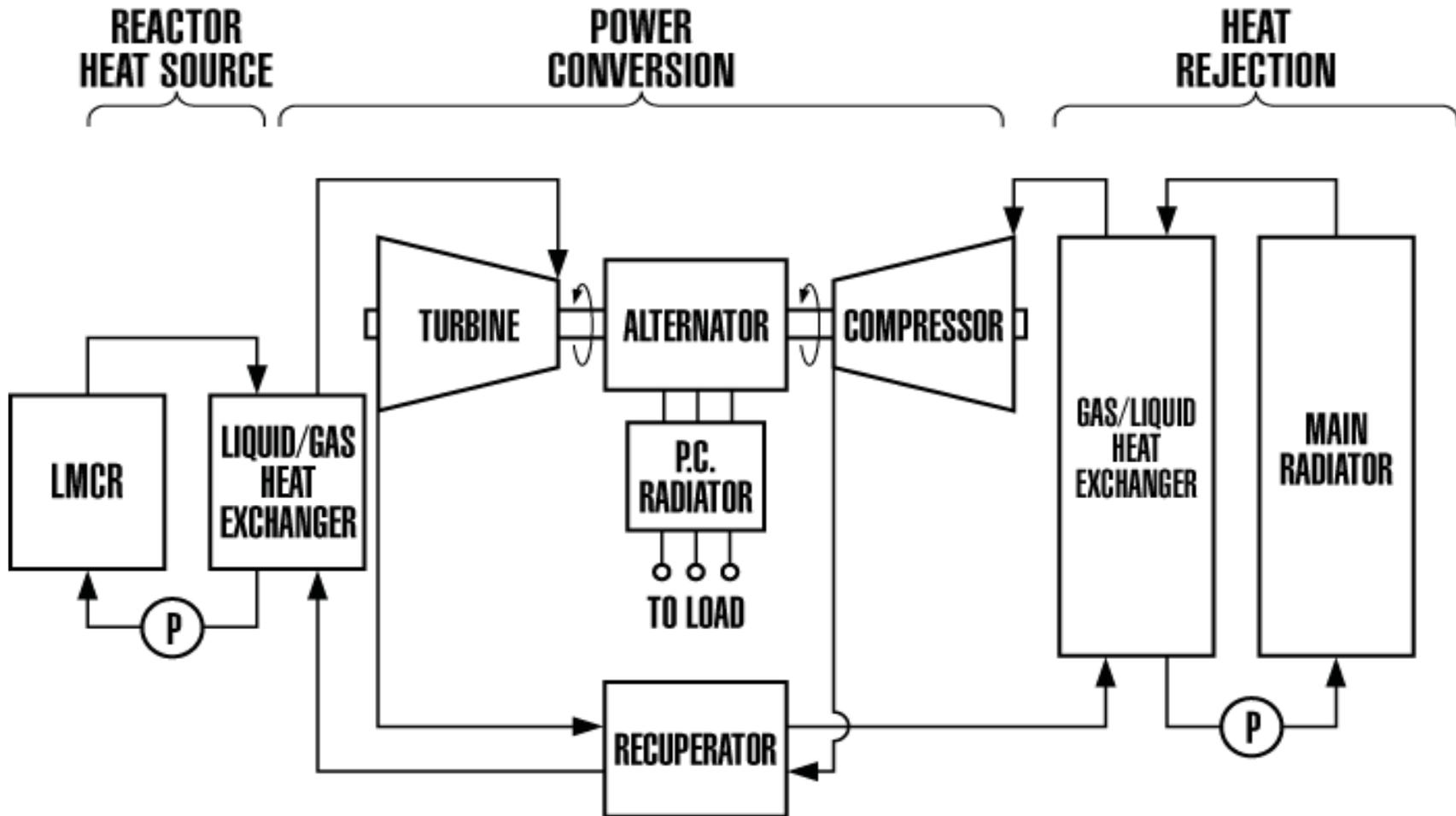
(La Bar , 2002)

Requirements – i.e. Power System Design Drivers

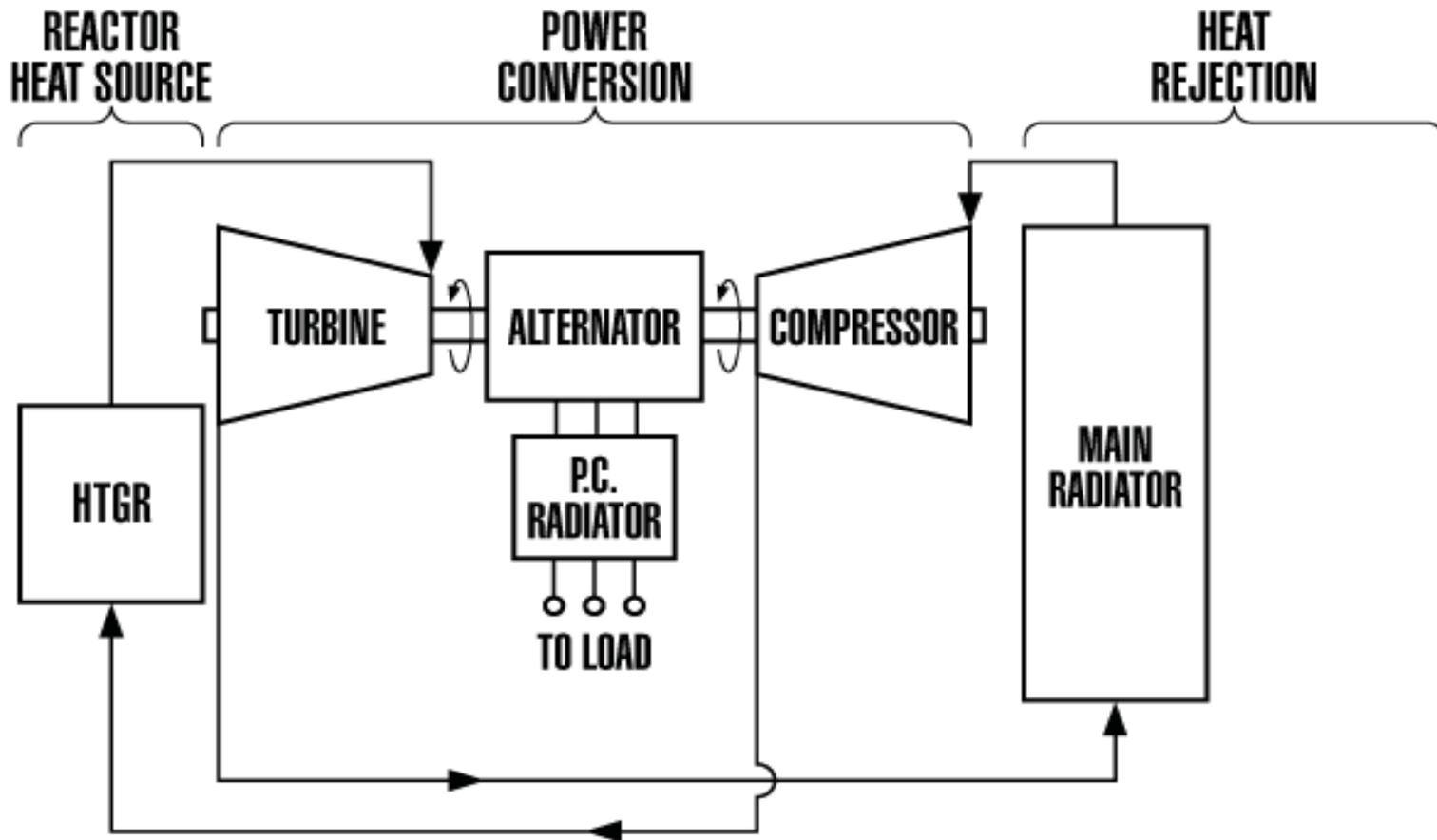
- Space (Lunar-Mars) Power Systems
 - Emphasis is on **Minimum System Mass**
 - **High System Reliability**, Autonomy and long Operational Life required to compensate for little or no maintenance
 - Need **least complex systems** w. minimum components
 - **Thermal Efficiency can be traded to achieve Low Mass**, i.e. non-regenerated and direct heated/cooled cycles eliminate heat exchanger (regenerator HX, HSHX, CSHX) mass at reduced Eff.
- Terrestrial Nuclear Power Systems –e.g. LFTR Power Plant
 - Emphasis is on **maximizing Thermal Efficiency** and thus Power Output, Revenue, **Profit & Return on Investment**
 - System Maintenance possible during regularly scheduled Periods
 - High System Mass and Complexity are acceptable as long as high Power Plant Availability/Reliability is assured

Space CBC Systems and Analysis

In-direct Heat Input & Rejection via Radiator for
Regenerated Closed Brayton Cycle (CBC) Power System

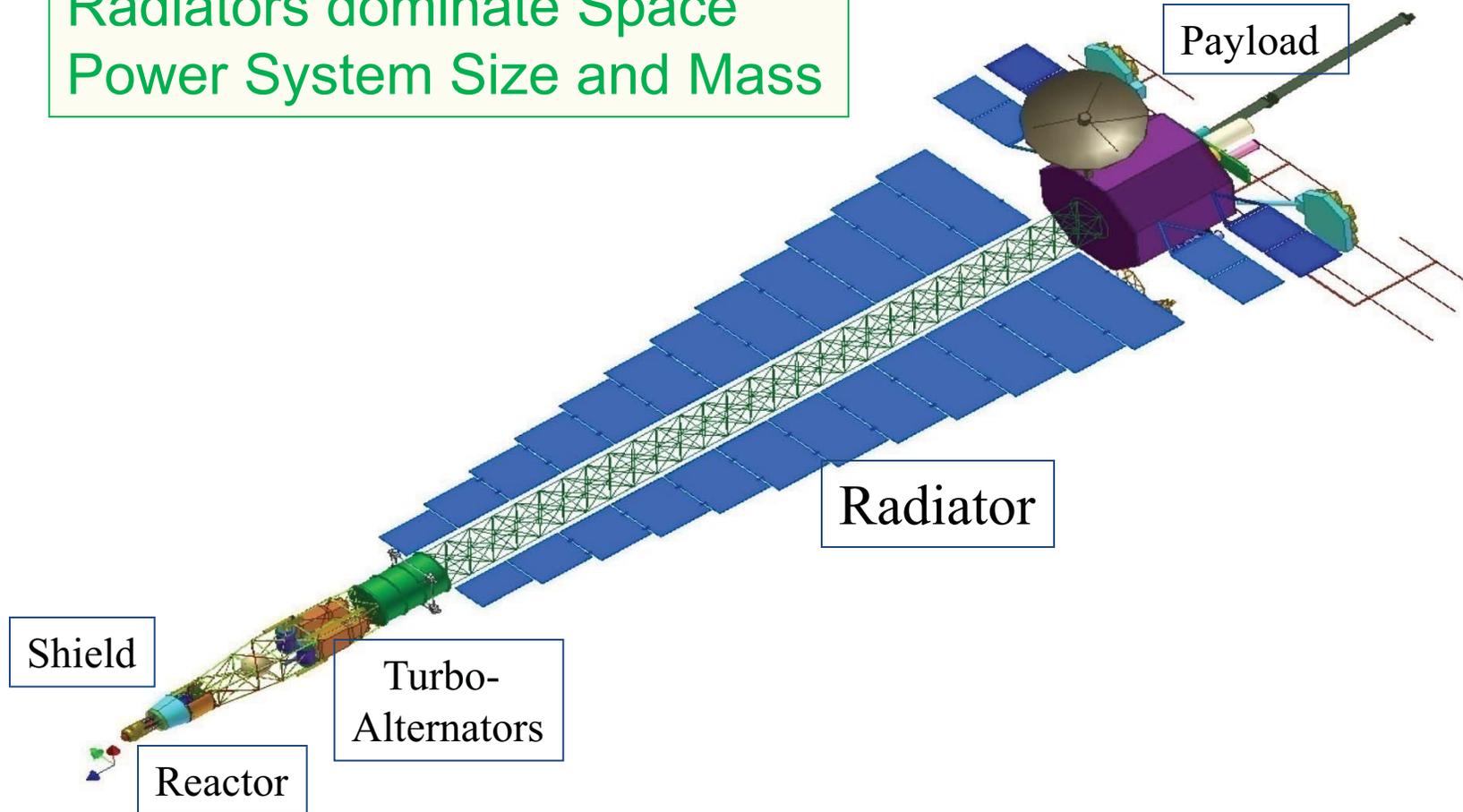


Direct Heat Input and Rejection via Radiator for **Non-Regenerated** Closed Brayton Cycle (CBC) Power System
(Simpler System eliminates 3 Heat Exchangers; - but lower η_{th})

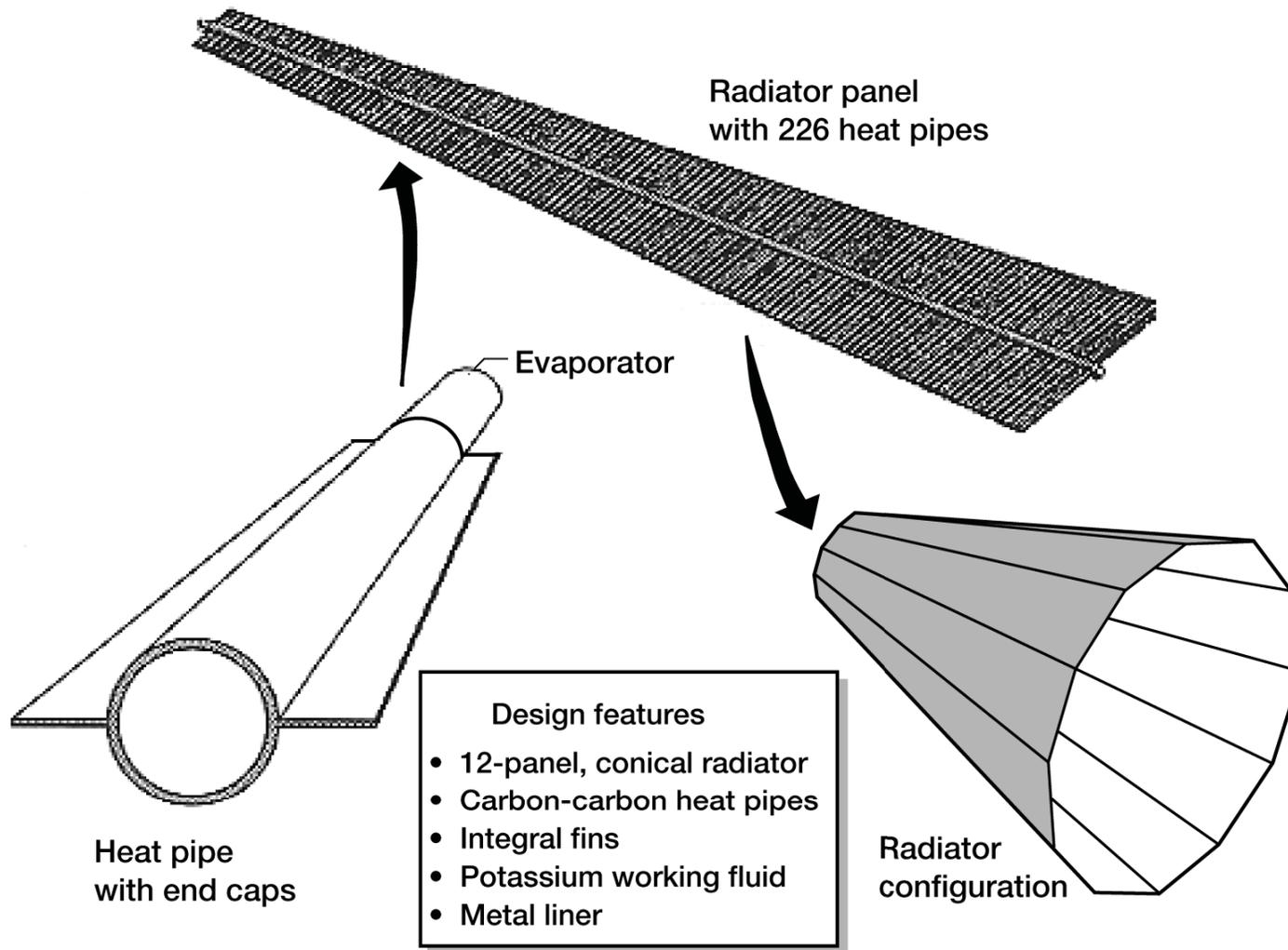


Spacecraft with Trapezoidal Heat Pipe Radiator

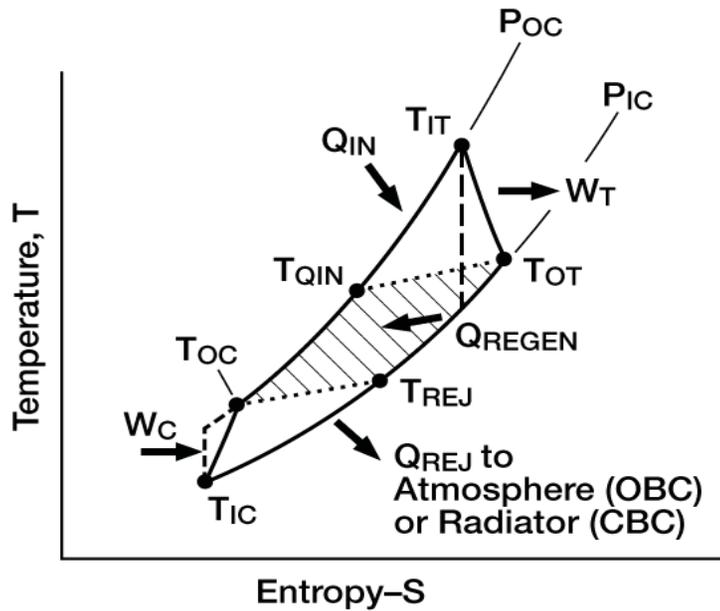
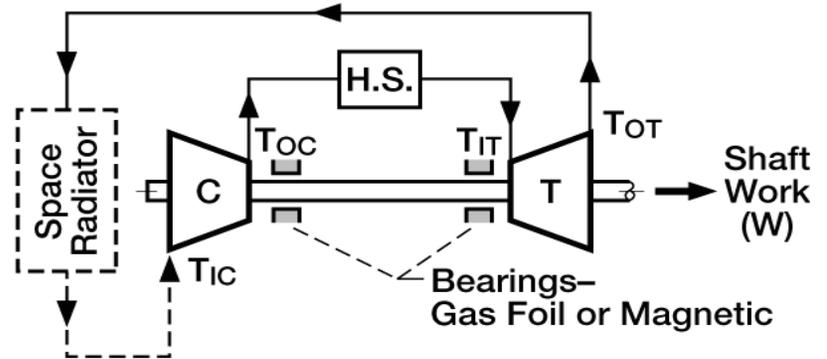
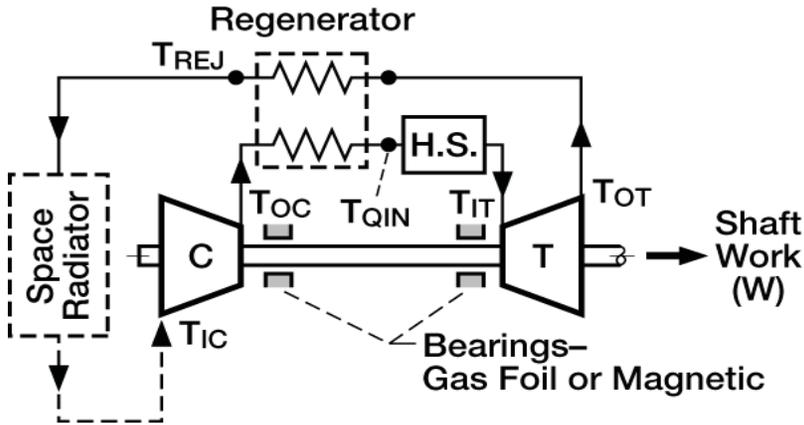
Radiators dominate Space Power System Size and Mass



SP – 100 Radiator Panel/Cone Configuration

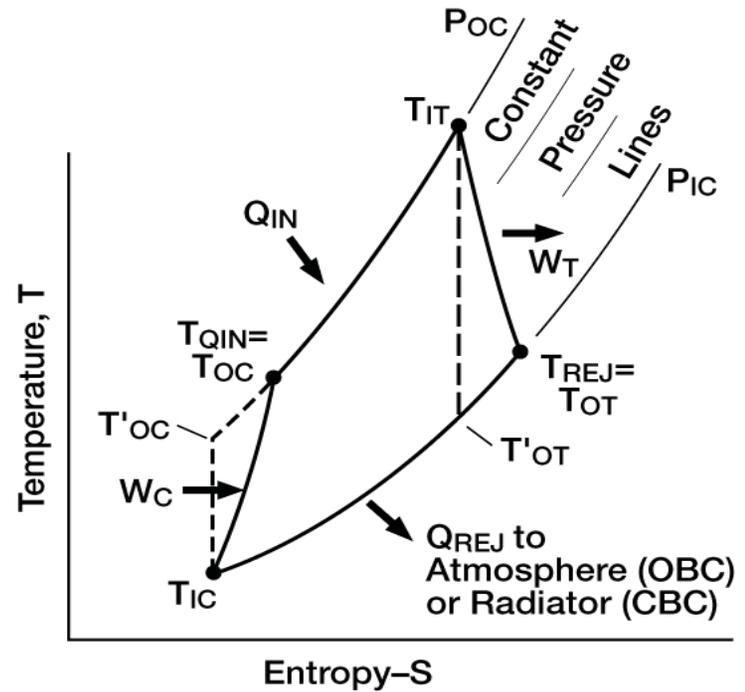


GAS TURBINE (BRAYTON CYCLES)



Regenerated Cycle

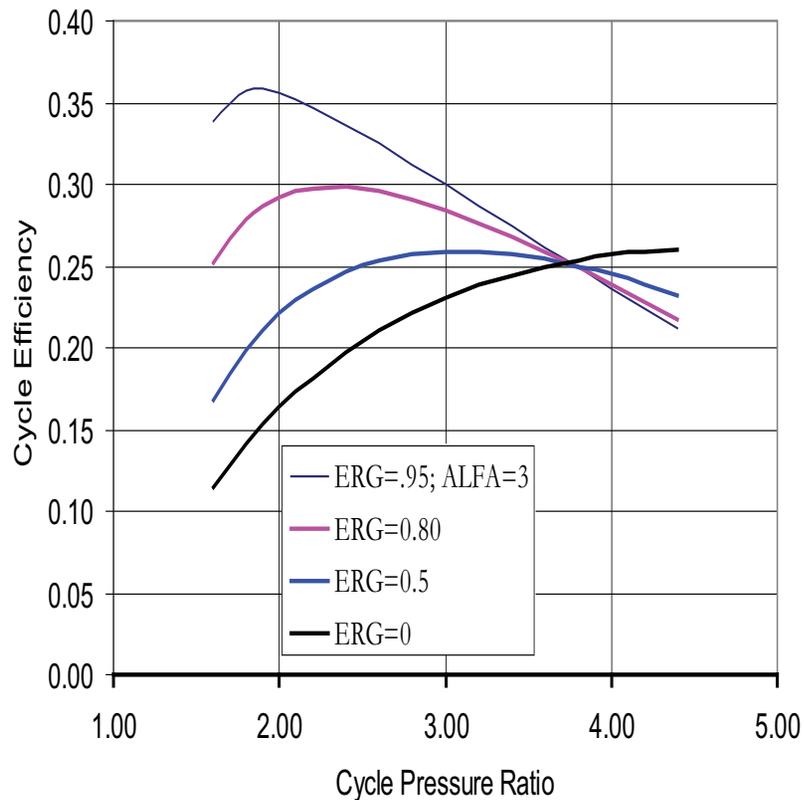
CD-04-82620



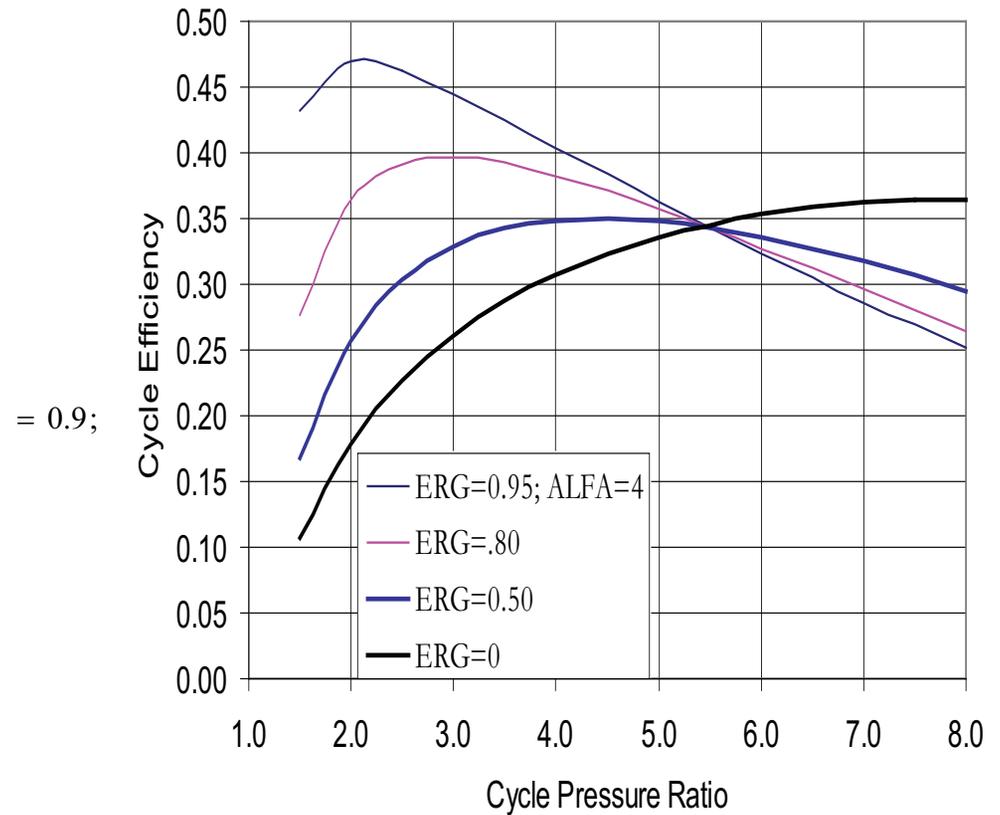
Non-Regenerated Cycle

Influence of Regenerator Effectiveness (ERG) on Cycle Efficiency at Cycle Temp. Ratio of 3.0 and 4.0

$$\eta_{PC} = \eta_{PT} = 0.9; \quad \gamma = 1.666$$

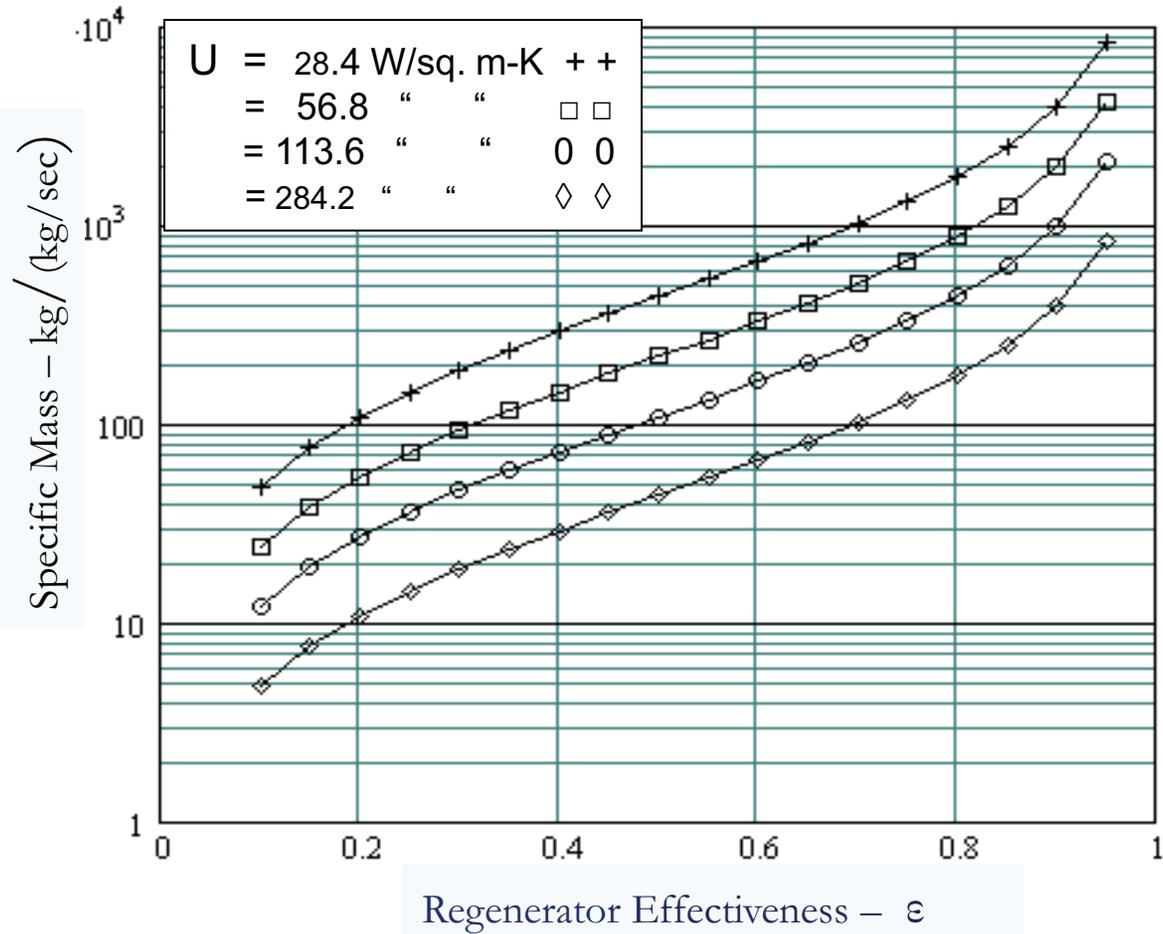


(a) Temp. Ratio = 3.0

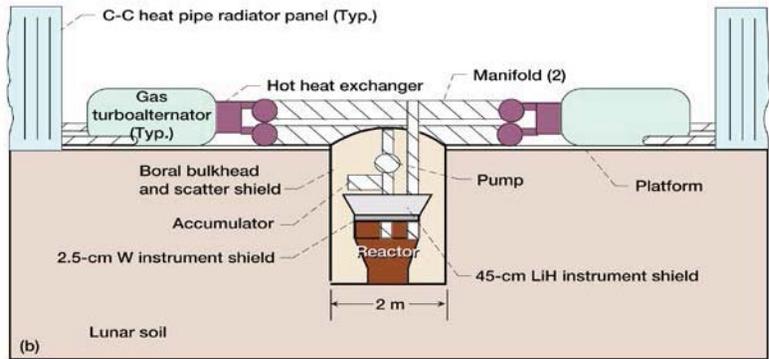
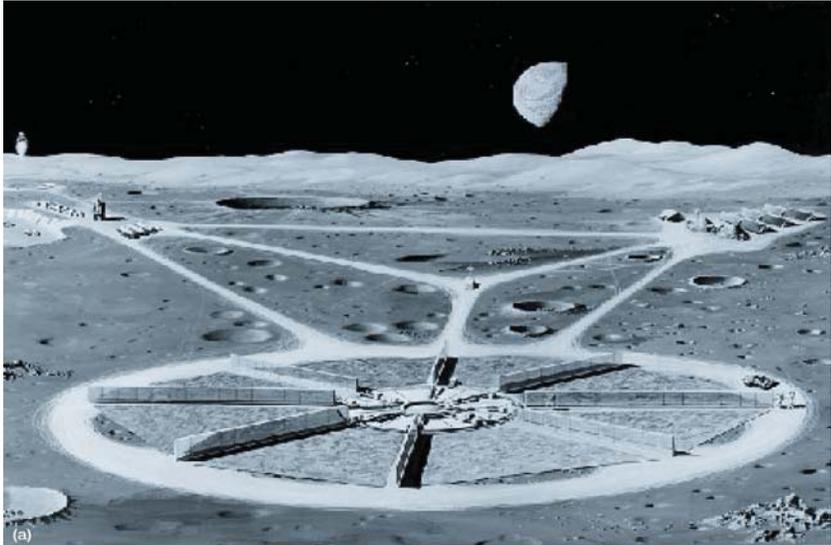


(b) Temp. Ratio = 4.0

Regenerator Specific Mass vs. Effectiveness with Heat Transfer Coefficient U as a Parameter for He Working Fluid

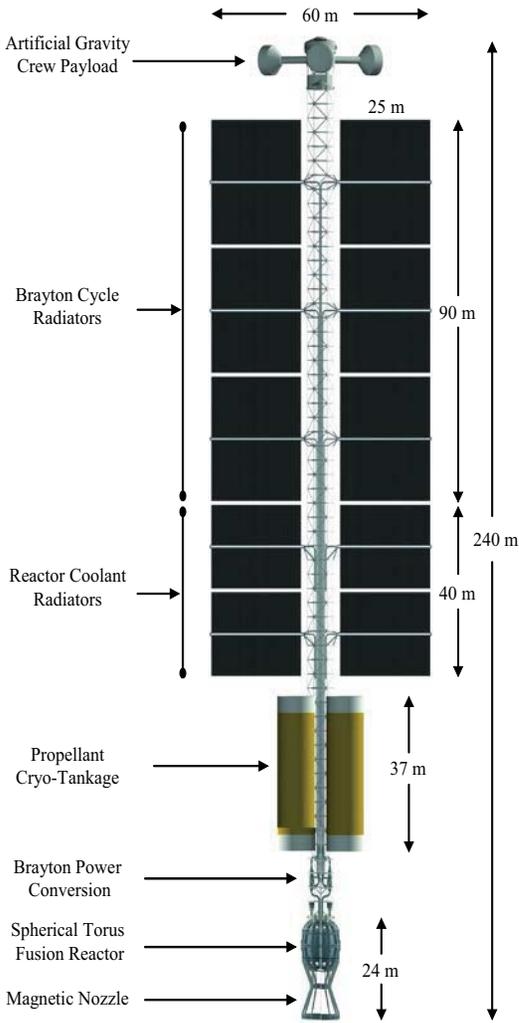


Advanced Space Power System Applications



Heat transport system
 Primary
 Secondary

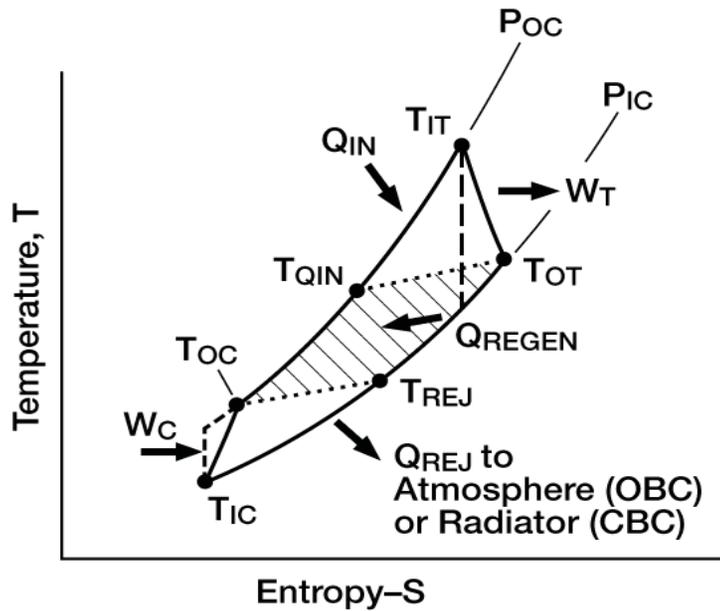
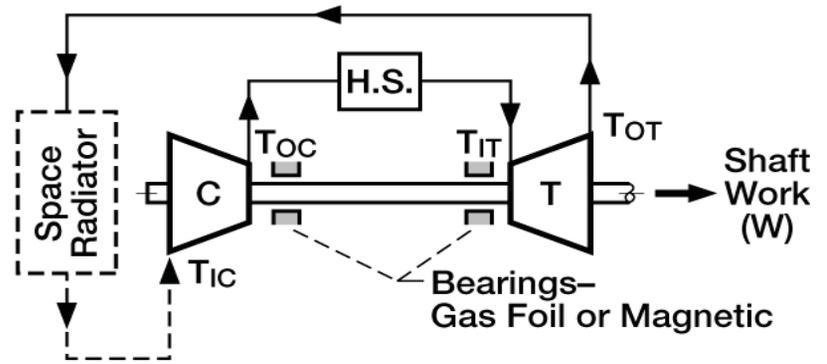
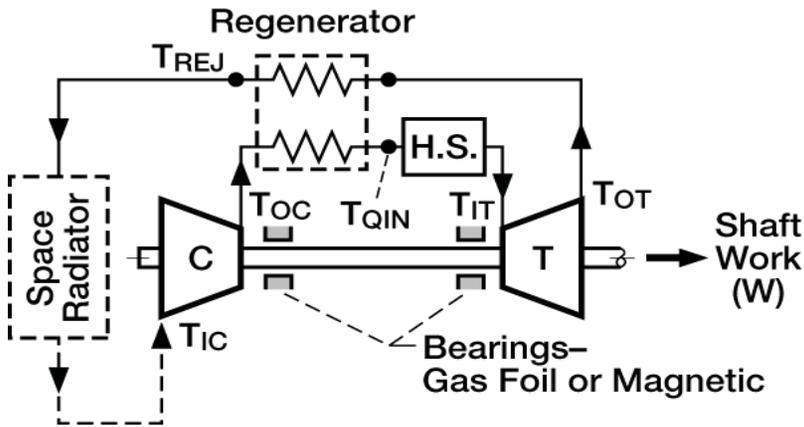
Lunar Base Power System



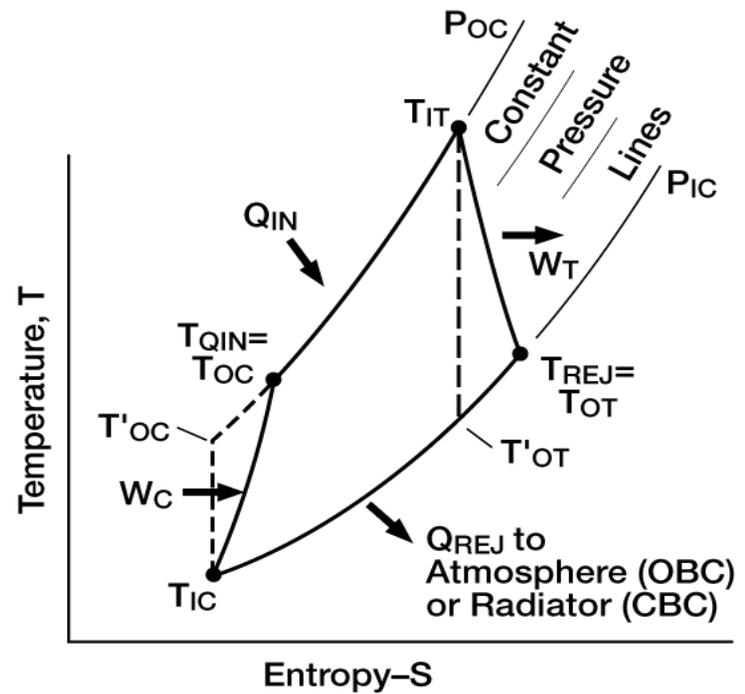
Interplanetary Fusion
Propulsion Space Vehicle

Ground Based CBC Systems and Analysis

Recall GAS TURBINE (BRAYTON CYCLES)



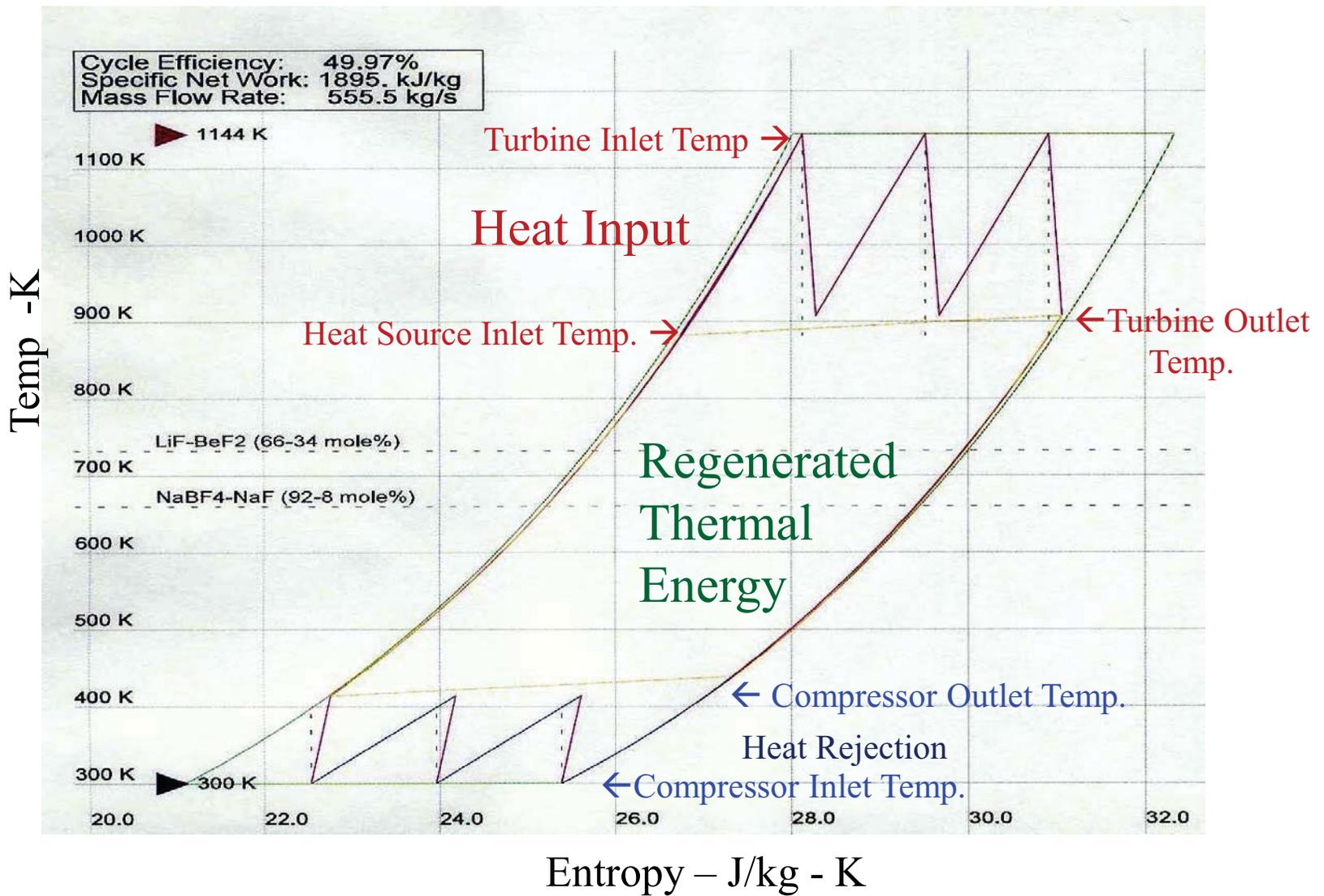
Regenerated Cycle



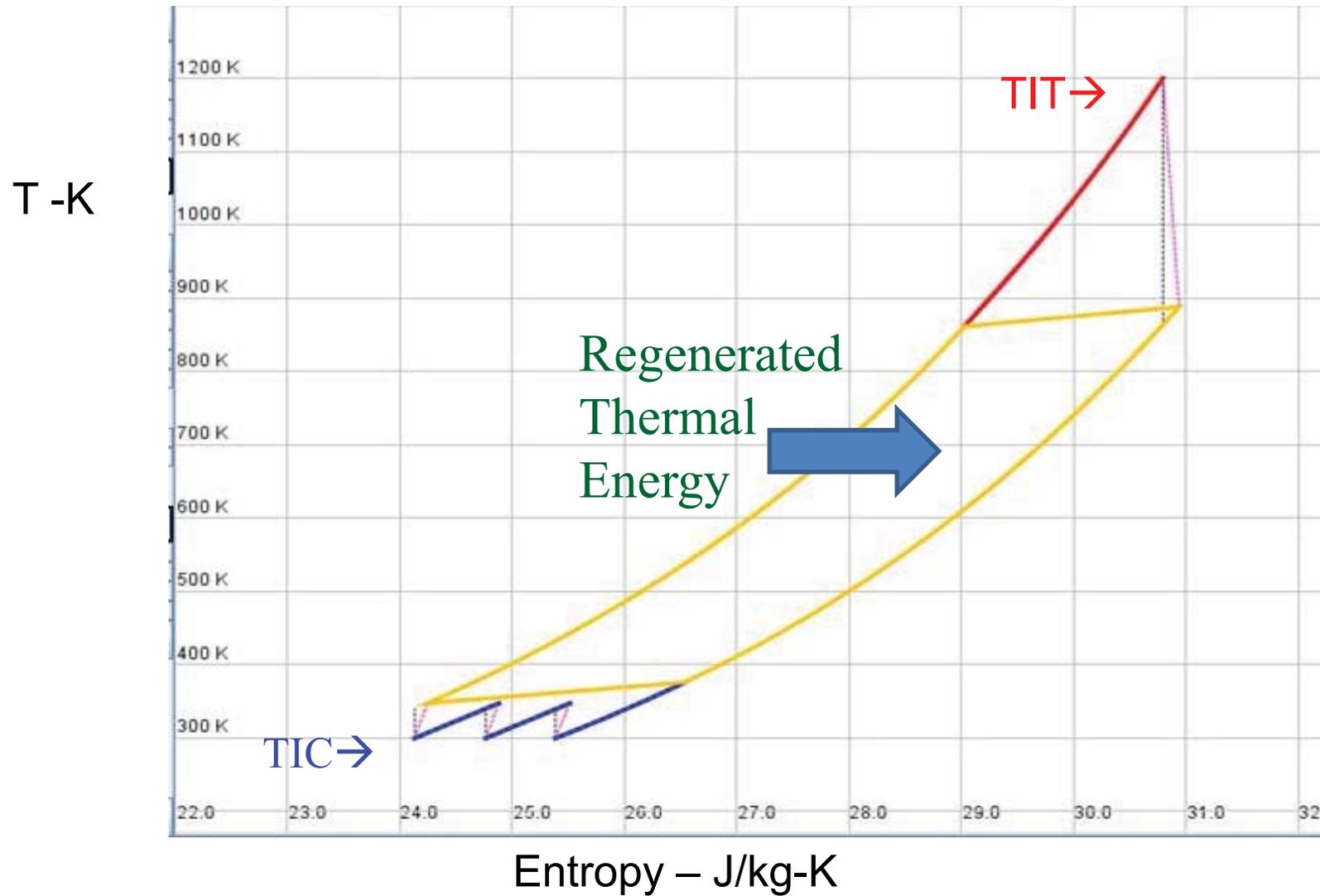
Non-Regenerated Cycle

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Three Stage Reheat & Intercool Brayton Cycle Temperature – Entropy Diagram

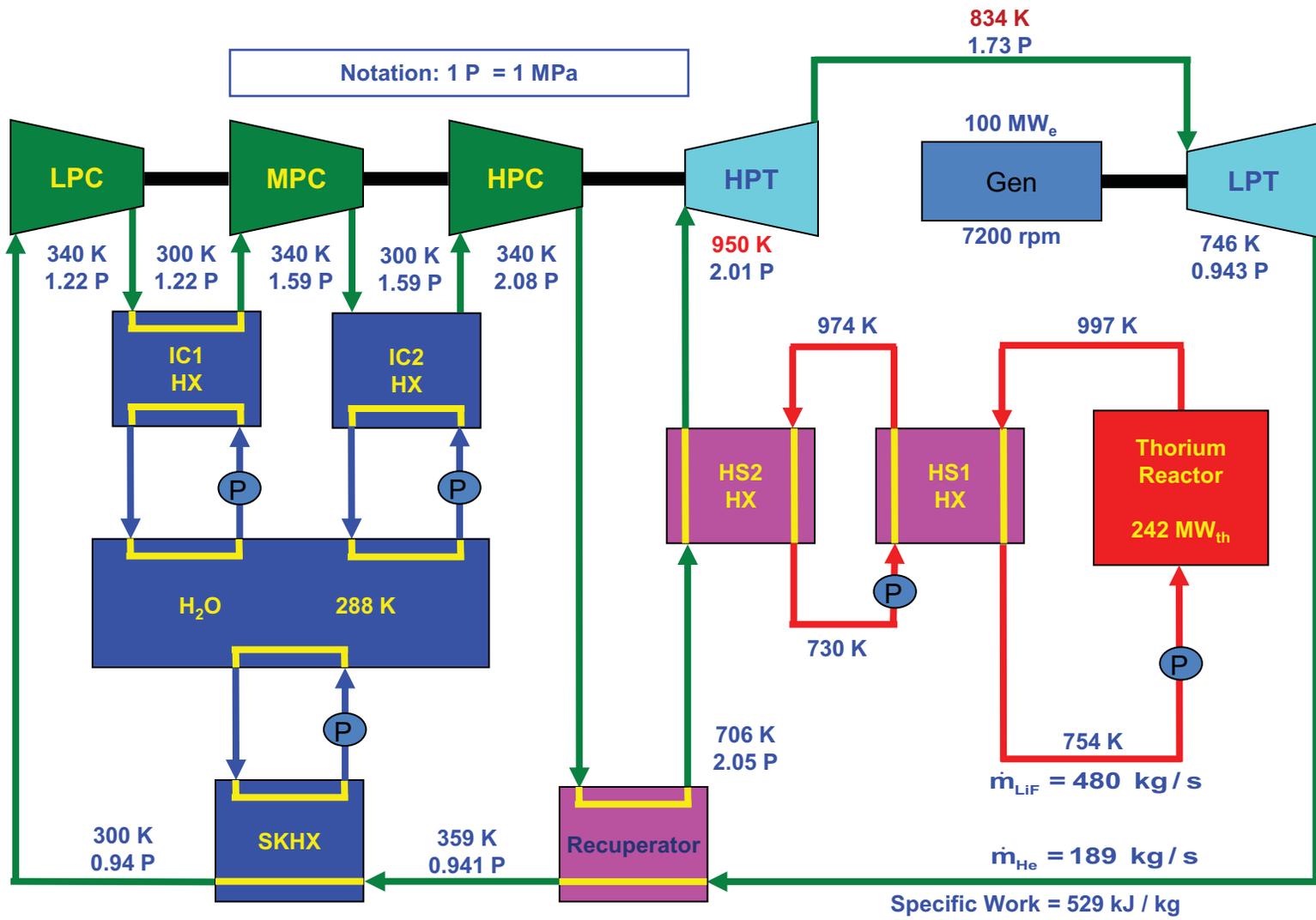


Three Stage Intercool Only Brayton Cycle Temperature – Entropy Diagram



Temp Ratio = 3.17
 Efficiency = 41.3%
 Turbine Power = 217.9 MW
 Compressor Power = - 115.8 MW

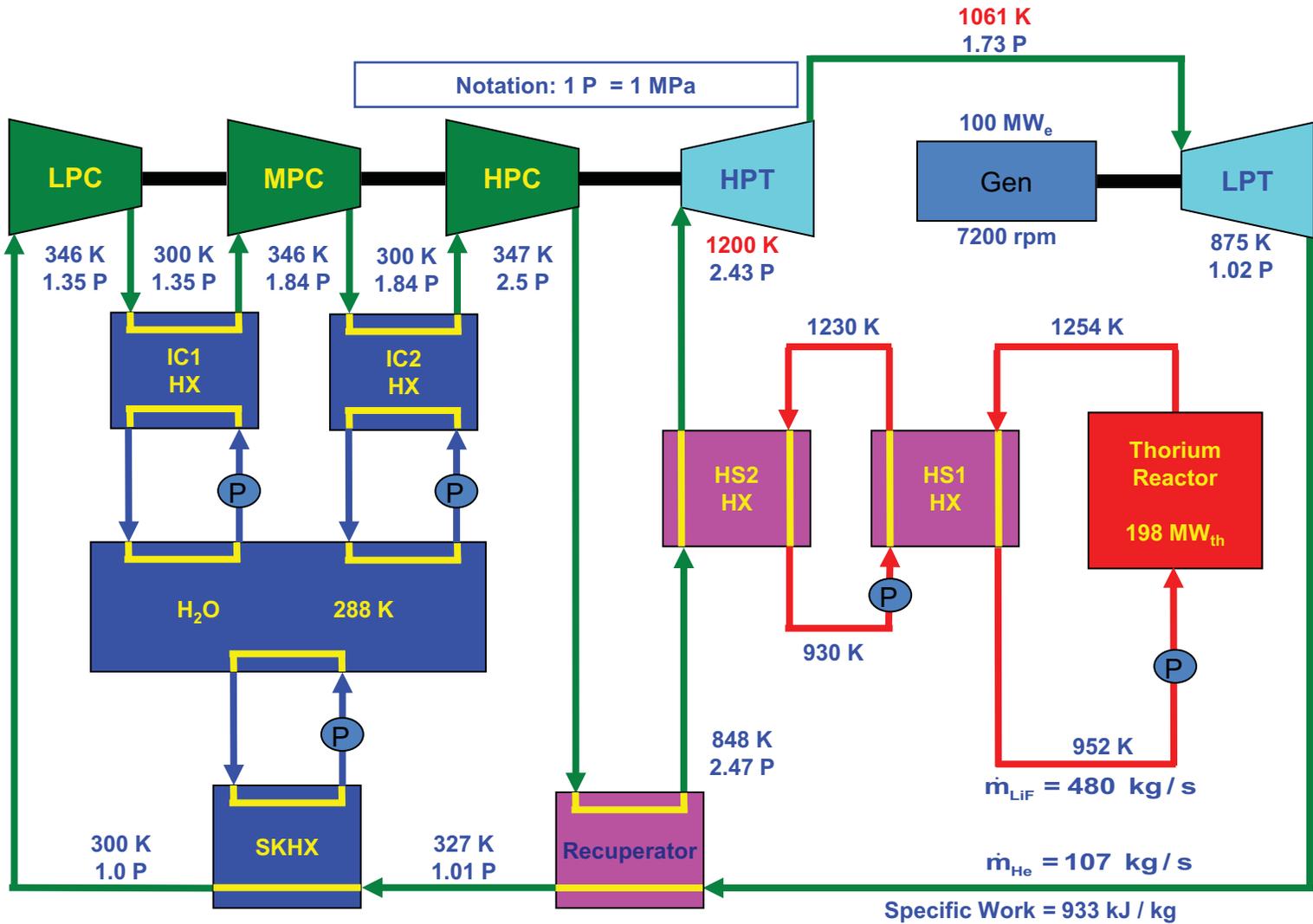
100 MWe Power Plant – 2 Salt Configuration
Thorium Molten Salt Reactor with Intercooled He CBC
with 950 K Turbine Inlet Temperature (mHe = 189.2 kg/s)



Temp Ratio = 4.0
 Efficiency = 50.5 %
 Turbine Power = 178.9 MW
 Compressor Power = - 76.9 MW

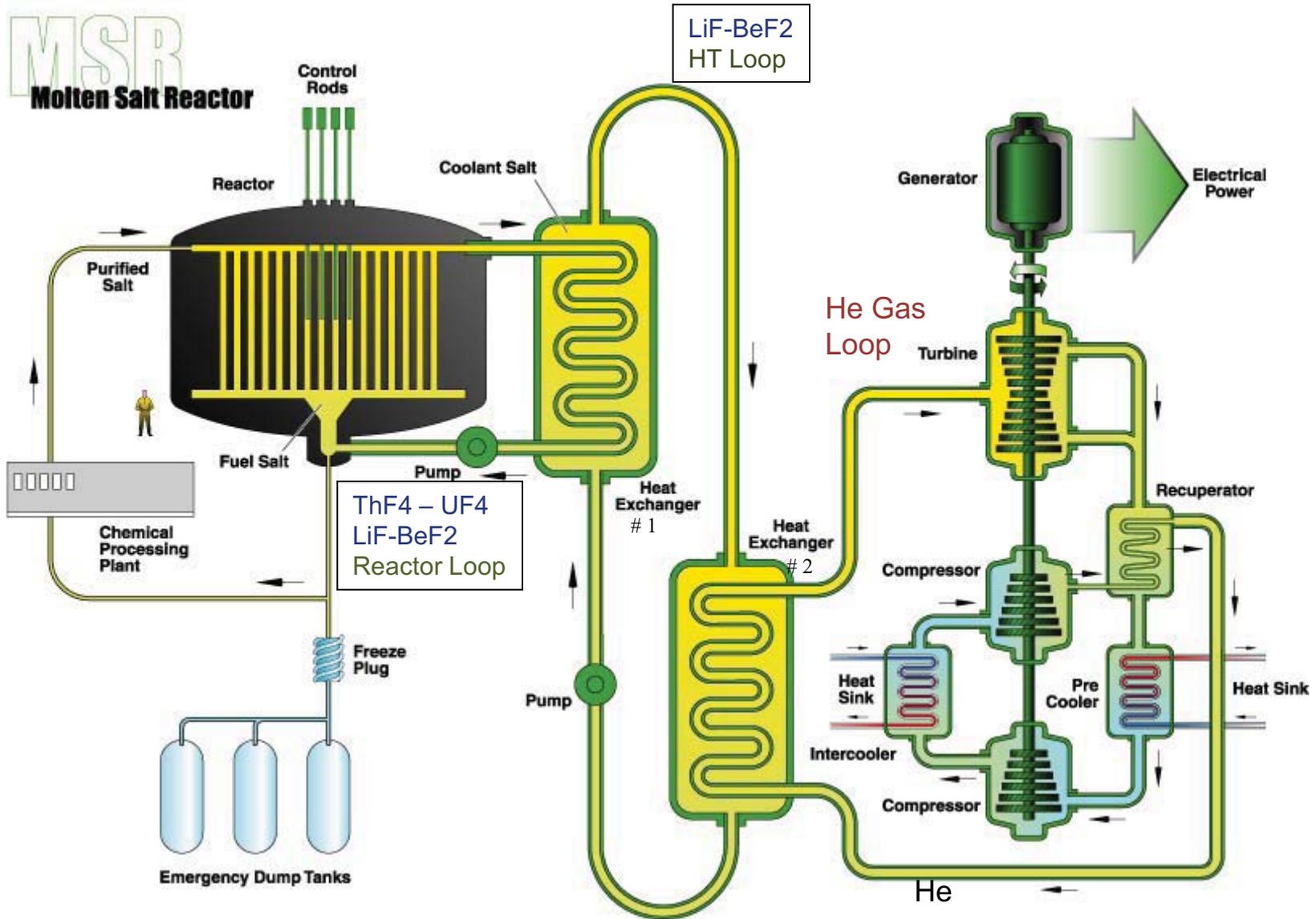
100 MWe Power Plant – 2 Salt Configuration
Thorium Molten Salt Reactor -
Helium Brayton Cycle, 1200 K Turbine Inlet Temp

He ---
 ThF4-UF4 ----
 & BeF2 -----



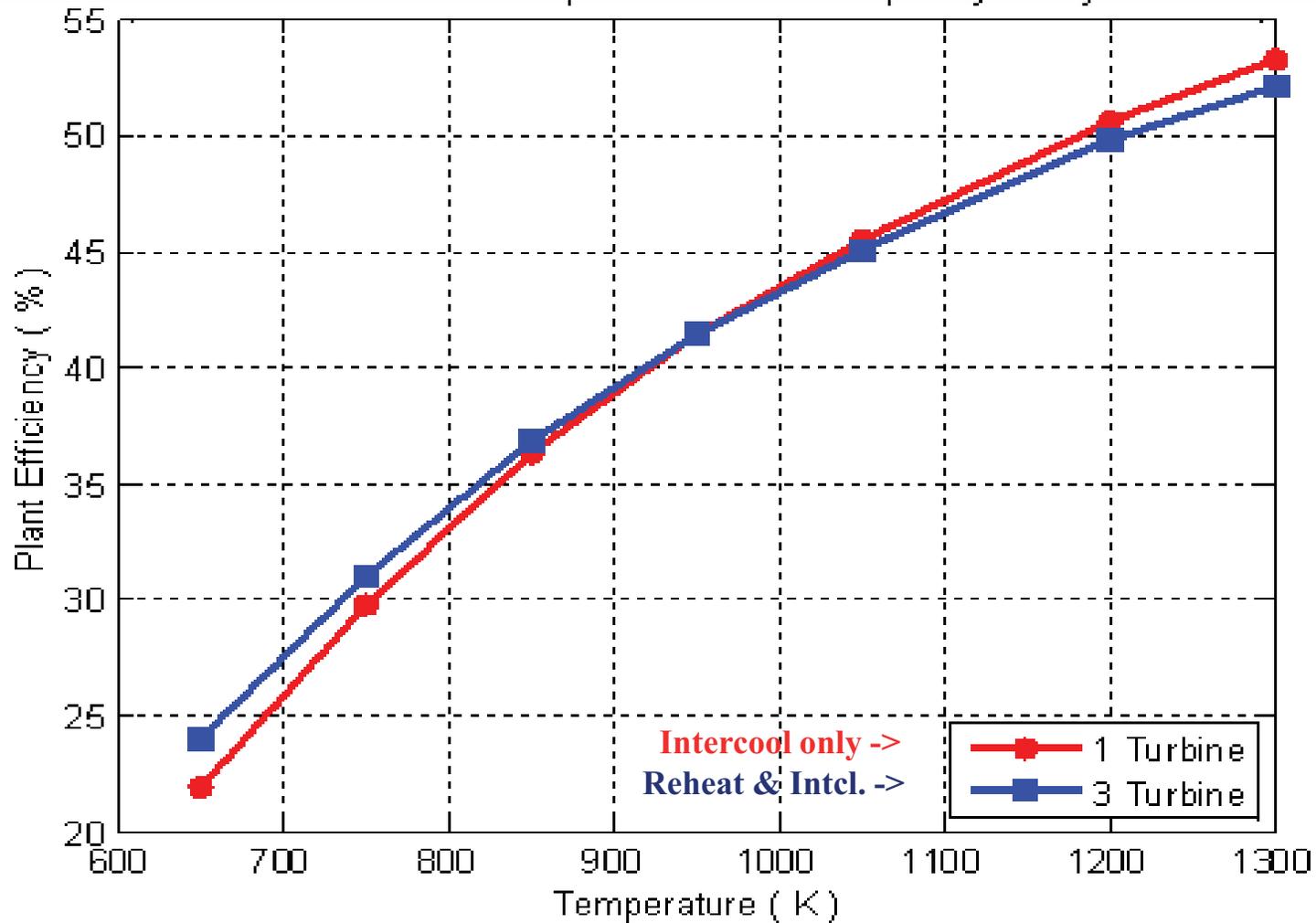
Gen-4 Liquid 2 Salt Configuration Reactor Concept – ORNL

MSR
Molten Salt Reactor

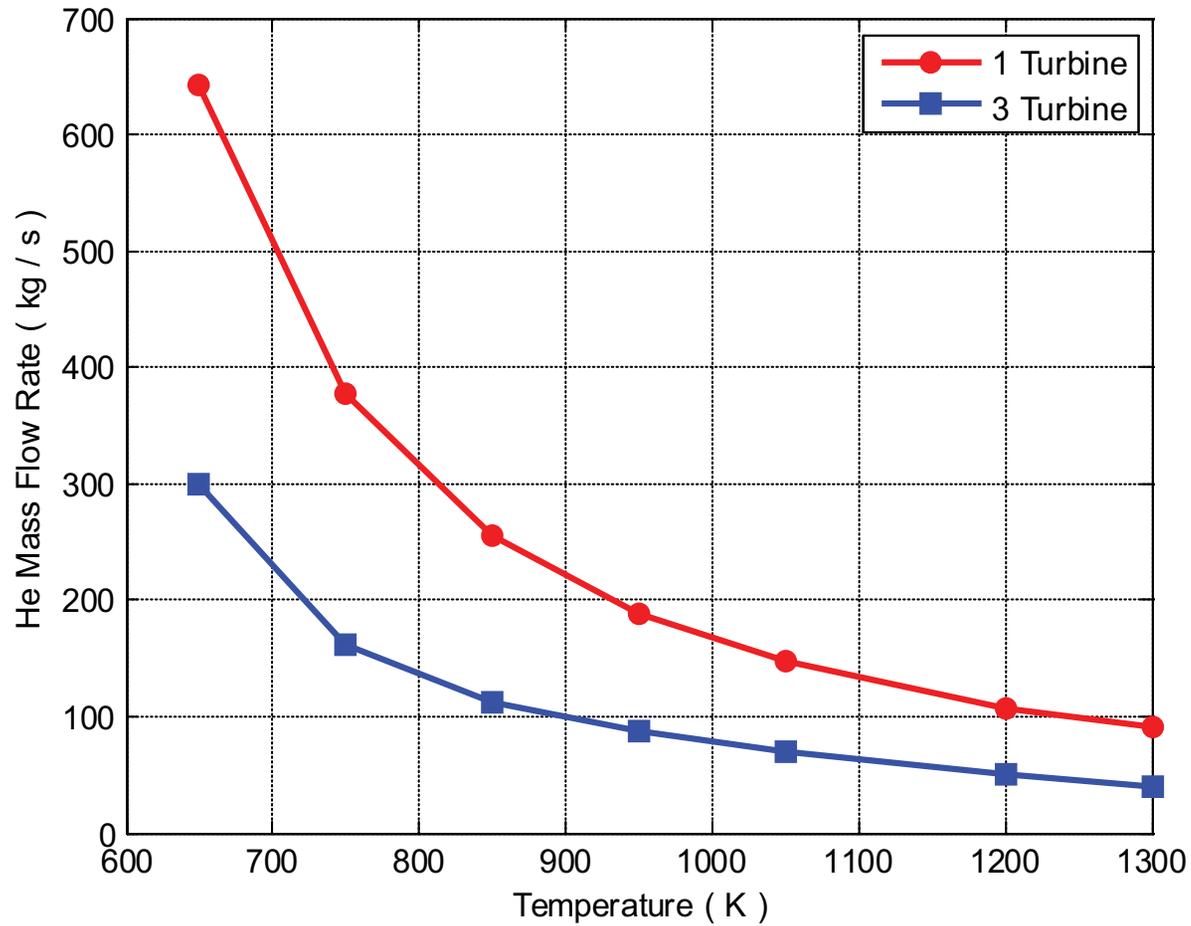


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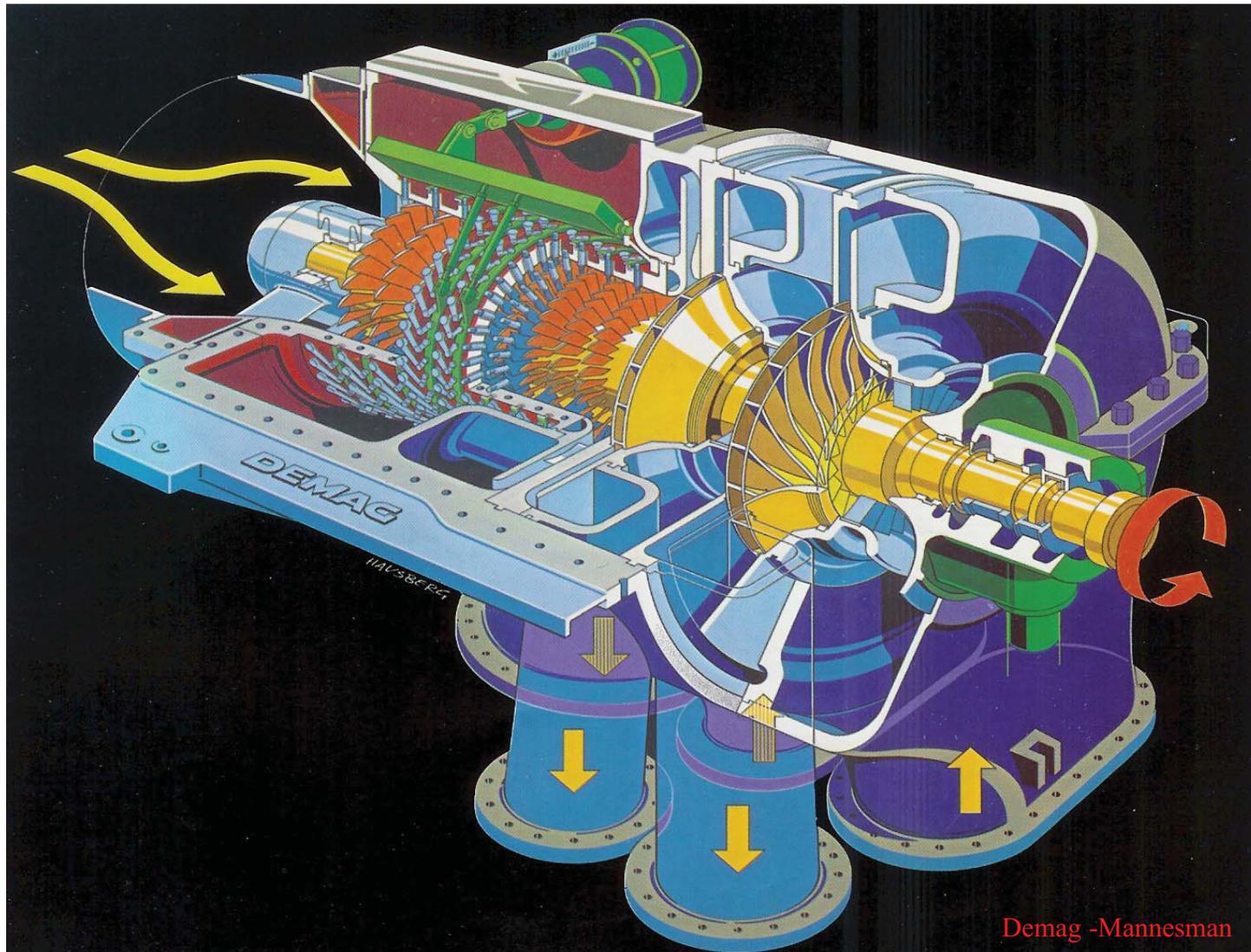
100 MWe Power Plant Efficiency w. Intercool & Reheat Cycles (3 Inter-cooled Compressors in Series)



100 MWe Power Plant Flowrate w. Intercool & Reheat Cycles (3 Inter-cooled Compressors in Series)



Typical Axial Radial Turbo-compressor

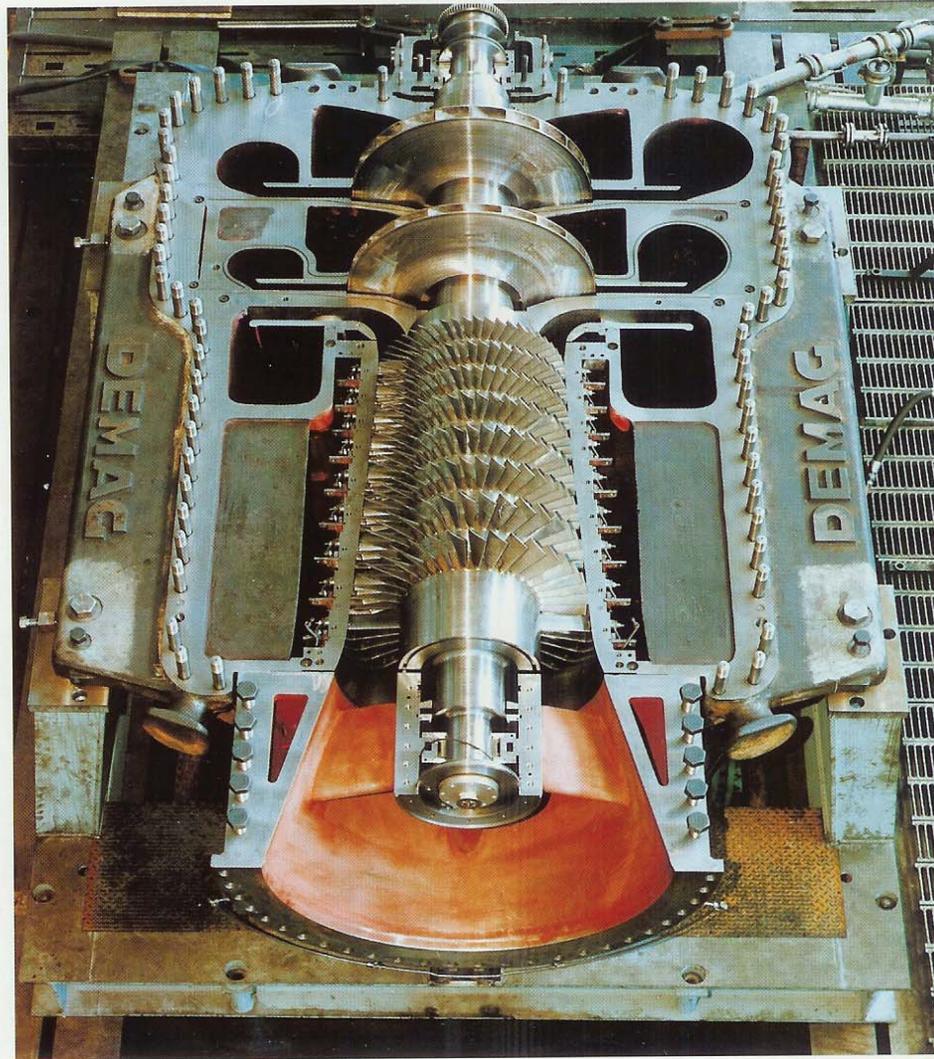


Demag -Mannesman

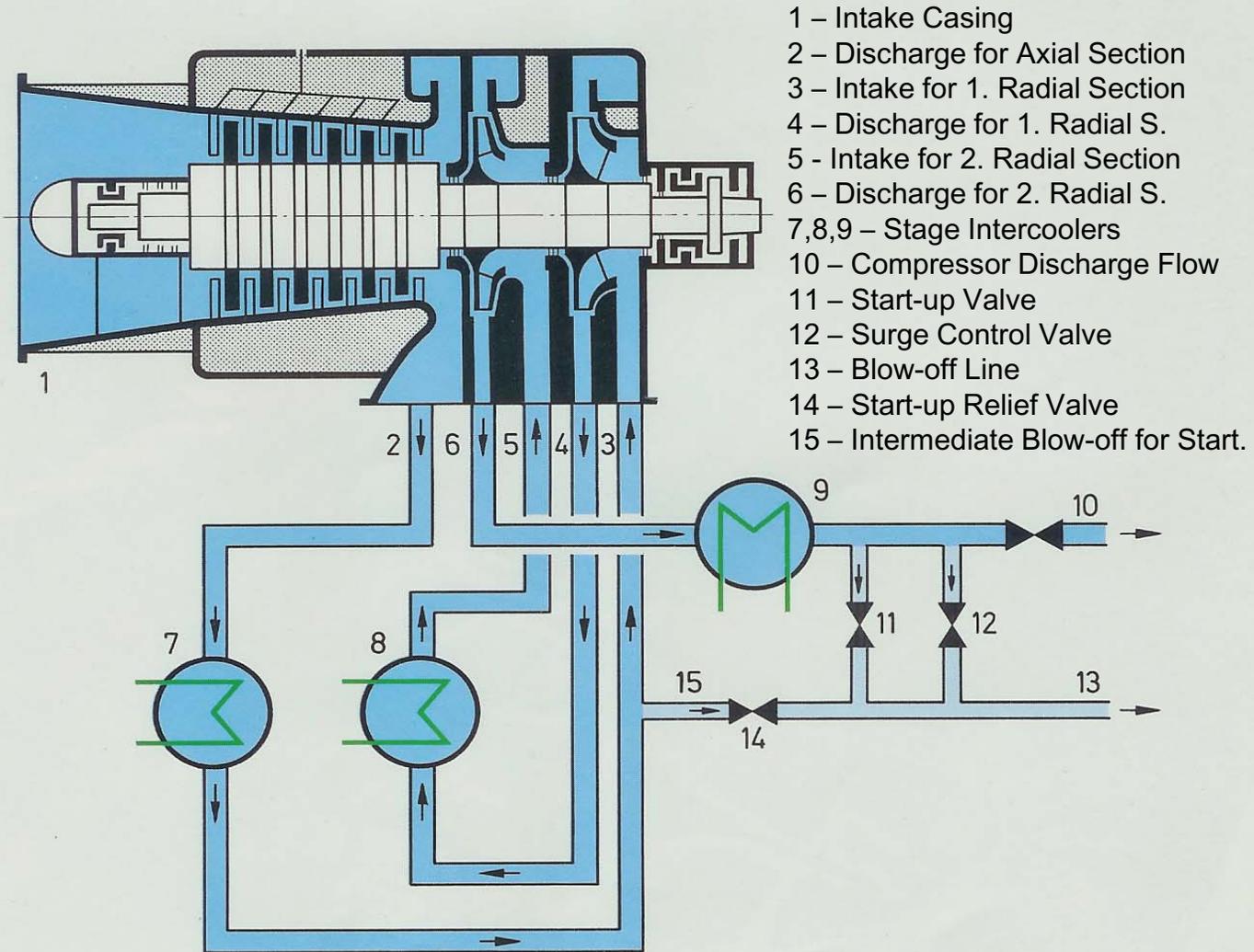
Axial/Radial Compressor with Axial Intake

Type AR 250-8-2 axial-centrifugal
compressor with axial intake

Medium	Air
Flow volume	272 000 m ³ /h
Intake pressure	1.01 bar
Discharge pressure	6.4 bar
Rotational speed	4 550 rpm
Drive rating	22 700 kW (turbine)

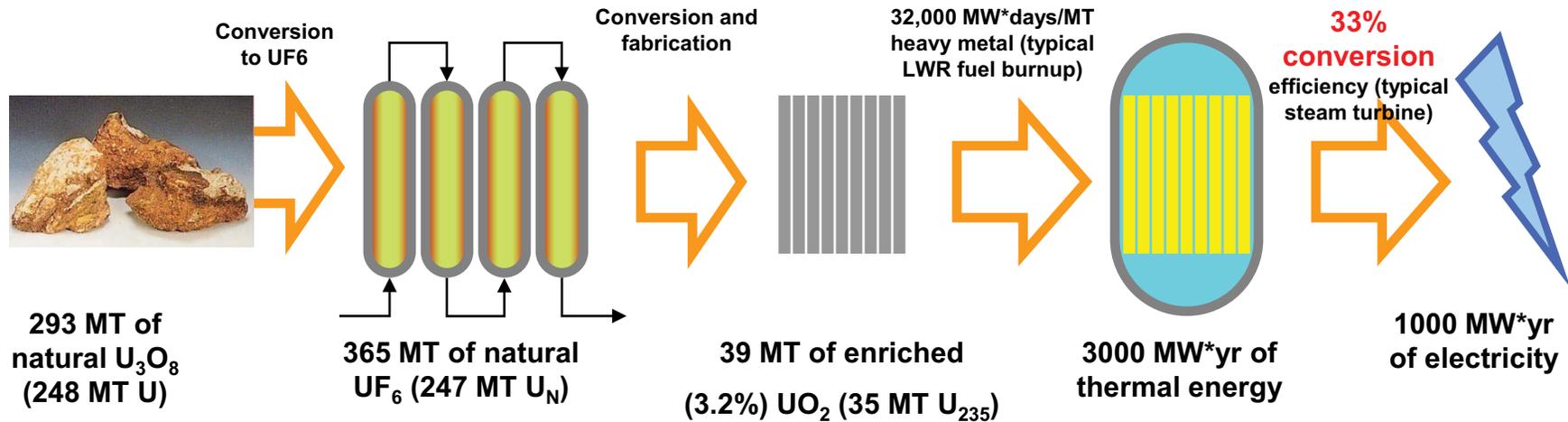


Flow Diagram for AR Compressor with Axial Intake

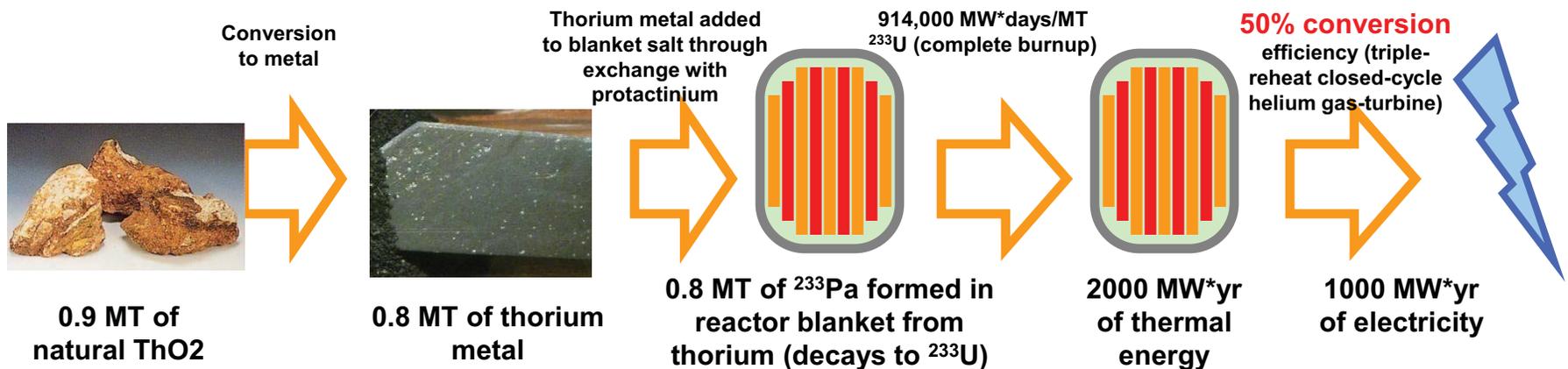


Energy Extraction Comparison for U_{238} and Th_{232}

Uranium-fueled light-water reactor: 35 GW*hr/MT of natural uranium



Thorium-fueled liquid-fluoride reactor: 11,000 GW*hr/MT of natural thorium



Submarine Based Power Plants

Compact, Portable Thorium Reactors



Proposal to use US mothballed shipyards to produce hundreds of portable thorium nuclear gas turbine power plants



Concluding Remarks

- Verified that Advanced Nuclear Power Plants with CCGT Conversion Technology can achieve **> 50% Thermal Efficiency at TIT ~ 1200 K.**
- Above result obtained for both 'Intercool + Reheat' and 'Intercool Only' Cycle Configurations
- 'Intercool + Reheat' Configurations have higher Complexity (number of ducts and heat exchangers) but lower Working Fluid Mass Flow (He) Requirements thus reducing Ducting and Heat Exchanger Size
- Liquid Fluoride Thorium Reactor Technology (LFTR) can meet the goals of the **Gen IV Nuclear Energy Systems Initiative** – Energy Policy Act '05
 - Uses fertile Th232 breeding to fissile U233
 - **Can meet world energy demands for tens of millennia**
 - ~300 times Energy Density of current LWR Nuclear Power Plants with corresponding reduction in fission products. **Decay <300 yrs.**
 - Inherently Safe due to negative temp. coefficient of reactivity
 - Proliferation resistant
 - Load Leveling to produce H2, Desalination
- Submarine based Power Plants with HVDC distribution proposed

Backup Slides

Ref.: K. Sorensen and J. Bonometti

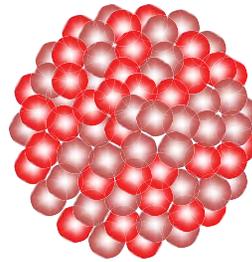
<http://www.energyfromthorium.com>

Energy from Thorium
via

Liquid Fluoride Thorium
Reactor Technology

Thorium₂₃₂ - Uranium₂₃₃ Breeding Cycle

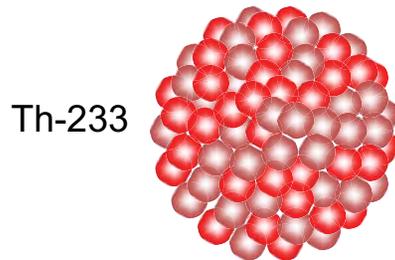
Thorium-233 decays quickly (half-life of 22.3 min) to protactinium-233 by emitting a beta particle (i.e. an electron).



Pa-233

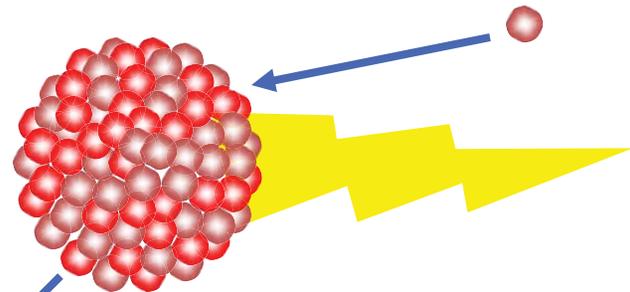
Protactinium-233 decays more slowly (half-life of 27 days) to uranium-233 by emitting a beta particle (an electron).

It is important that Pa-233 NOT absorb a neutron before it decays to U-233—it should be shielded from any neutrons until it decays.

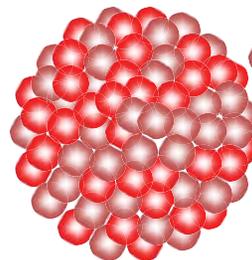


Thorium-232 absorbs a neutron from fission and becomes thorium-233.

U-233
(Fissile)



Uranium-233 is fissile and will fission when struck by a neutron, releasing energy and 2 to 3 neutrons. One neutron is needed to sustain the chain-reaction, one neutron is needed for breeding, and any remainder can be used to breed additional fuel.



Th-232
(Fertile)

Thorium –Uranium Fuel Cycle



Time for 99.9% Beta Decay of Protactinium to U-233

- **Looking at Step 3**, the time required for 99.9% of Pa²³³ decaying to U²³³ and 0.1% remaining Pa²³³, let $N(t) = 0.1$ and $N_0 = 99.9$ in eq. (1)

$$N(t) = N_0 e^{-\lambda t} \quad (1)$$

where λ is the *decay constant* computed from

$$\lambda = (\ln .5)/T_{0.5}, \text{ with } T_{0.5} \text{ being the } \textit{Half Life} = 27 \text{ days}$$

- So $\lambda = -0.693/27 = -0.02567 \text{ days}^{-1}$

Substituting in (1) : $0.1 = 99.9 e^{-0.02567 * t} \quad (2)$

Dividing (2) by 99.9 ~ 100 , and taking the **ln** of both sides, we have

$$\ln .001 = - .02567 t$$

$$t = - 6.9077 / -.02567$$

$$= 269 \text{ days, or } \sim 9 \text{ months for } 99.9\% \text{ of Pa} \rightarrow \text{U}^{233}$$

- **Note that for 99 % transmutation to U²³³ only 179.5 days would be required**

Submarine Based Power Plants

Compact, Portable Thorium Reactors



Proposal to use US mothballed shipyards to produce hundreds of portable thorium nuclear gas turbine power plants



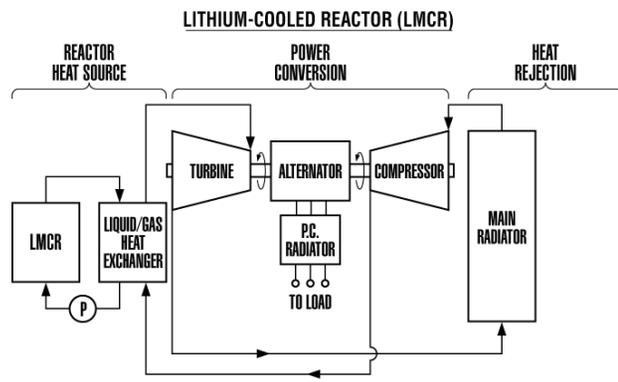
Key Cycle Input Parameters

- Compressor Inlet Temperature (TIC), K 300
- Cooling Water Temperature, K 288
- Reactor Heat Loss, percent 1.0
- Polytropic Efficiency—Compressor, percent 86
- Polytropic Efficiency—Turbine, percent 92
- Recuperator Effectiveness, percent 95
- Intercooler HX Pressure Loss, percent 0.5
- Reheat HX Pressure Loss, percent 0.8
- Turbine Pressure Ratio Fraction, percent 96
- Generator Efficiency, percent 98

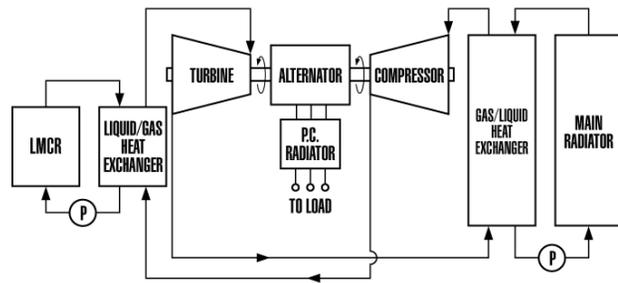
- Comparison to Space System-
Direct Heat Input and Rejection via Radiator for **Non-Regenerated** Closed Brayton Cycle (CBC) Power System

**CLOSED-CYCLE *NONREGENERATED* GAS TURBINE
 NUCLEAR POWER SYSTEM CONFIGURATIONS**

INDIRECTLY HEATED CYCLE

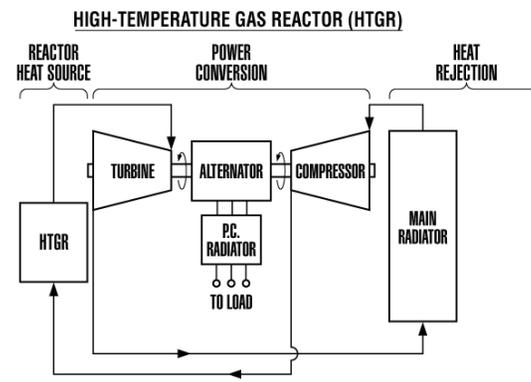


5. Indirect Heating Cycle with Direct Heat Rejection

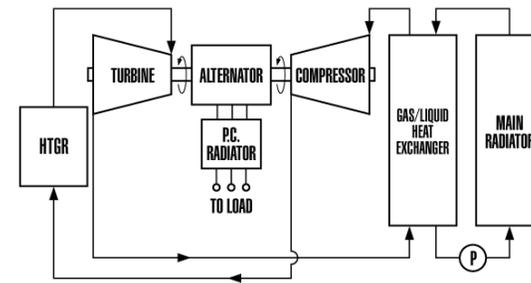


6. Indirect Heating Cycle with Indirect Heat Rejection

DIRECTLY HEATED CYCLE



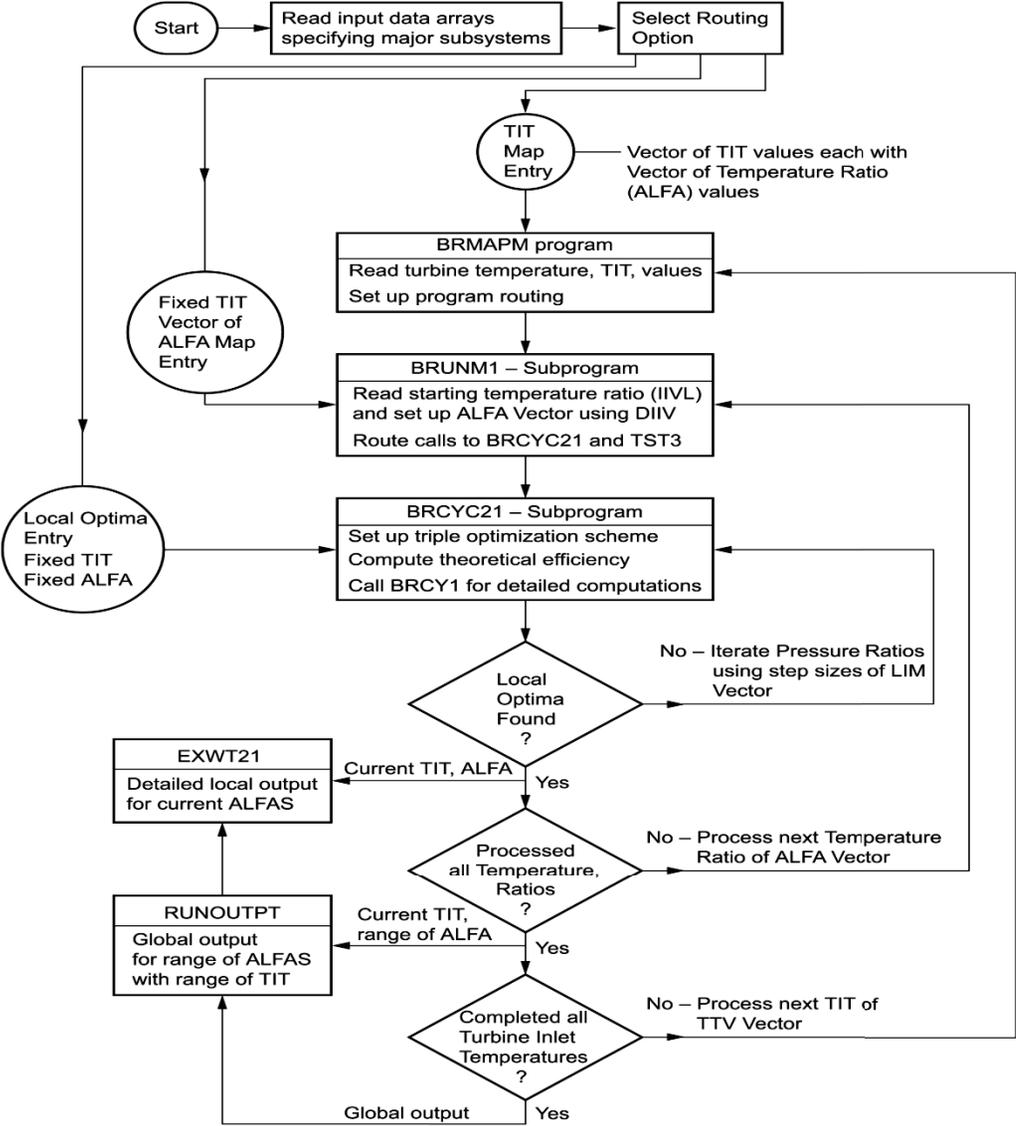
7. Direct Heating Cycle with Direct Heat Rejection



8. Direct Heating Cycle with Indirect Heat Rejection

CD-03-82597

Brayton Cycle Mapping Code - BRMAPS



Partial BRMAPS Code Output Results for 5 MWe Lunar Power Plant

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BRAYTON CYCLE CALCULATIONS - NON REGENERATED - 1500 K- POWER LEVEL = 5.00 MWE TSINK-K = 190
TEMP RATIO      ETAB      ETAPC      ETAPT      ERG      GAMMA      LPC      ETM      EPSIL      TIT-K
    3.000      .990      .900      .900      .000      1.667      .980      .950      .900      1500
OPTIMUM PRESSURE RATIOS (MAX THERM EFF; MIN ARP,MASS ) = 4.550 3.200 3.400 TIC-K = 500
PR RATIO THERM EFF.  ΔRP (M2/KW) MSYS (MG) W (KM/S-MW) TREJ-K TREFF-K TOC-K TOT-K ETAC ETAT
    4.5500    .2604    .4076    20.8974    .3895    862.88    630.73    1005.81    862.88    .867    .925
    3.2000    .2382    .3554    19.6747    .3122    985.29    663.87    850.84    985.29    .875    .919
    3.4000    .2446    .3566    19.6355    .3170    963.13    658.18    875.51    963.13    .874    .920
    3.2000    .2382    .3554    19.6747    .3122    985.29    663.87    850.84    985.29    .875    .919
THEORETICAL OPTIMUM PRESSURE RATIO (PROPTIM) = 4.579
NUMBER OF ITERATIONS (ICTE, ICTA, ICTM) = 7 9 9

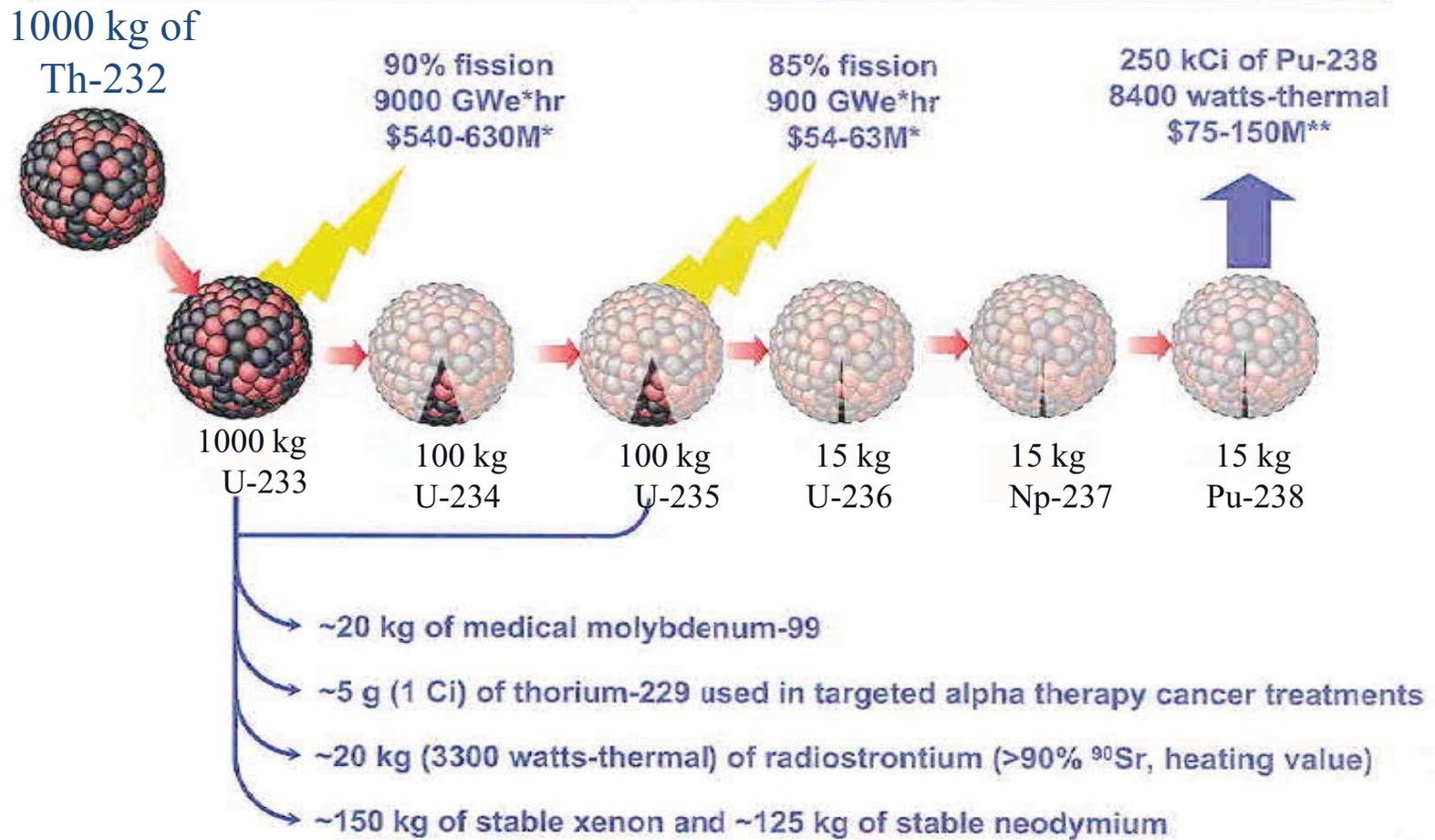
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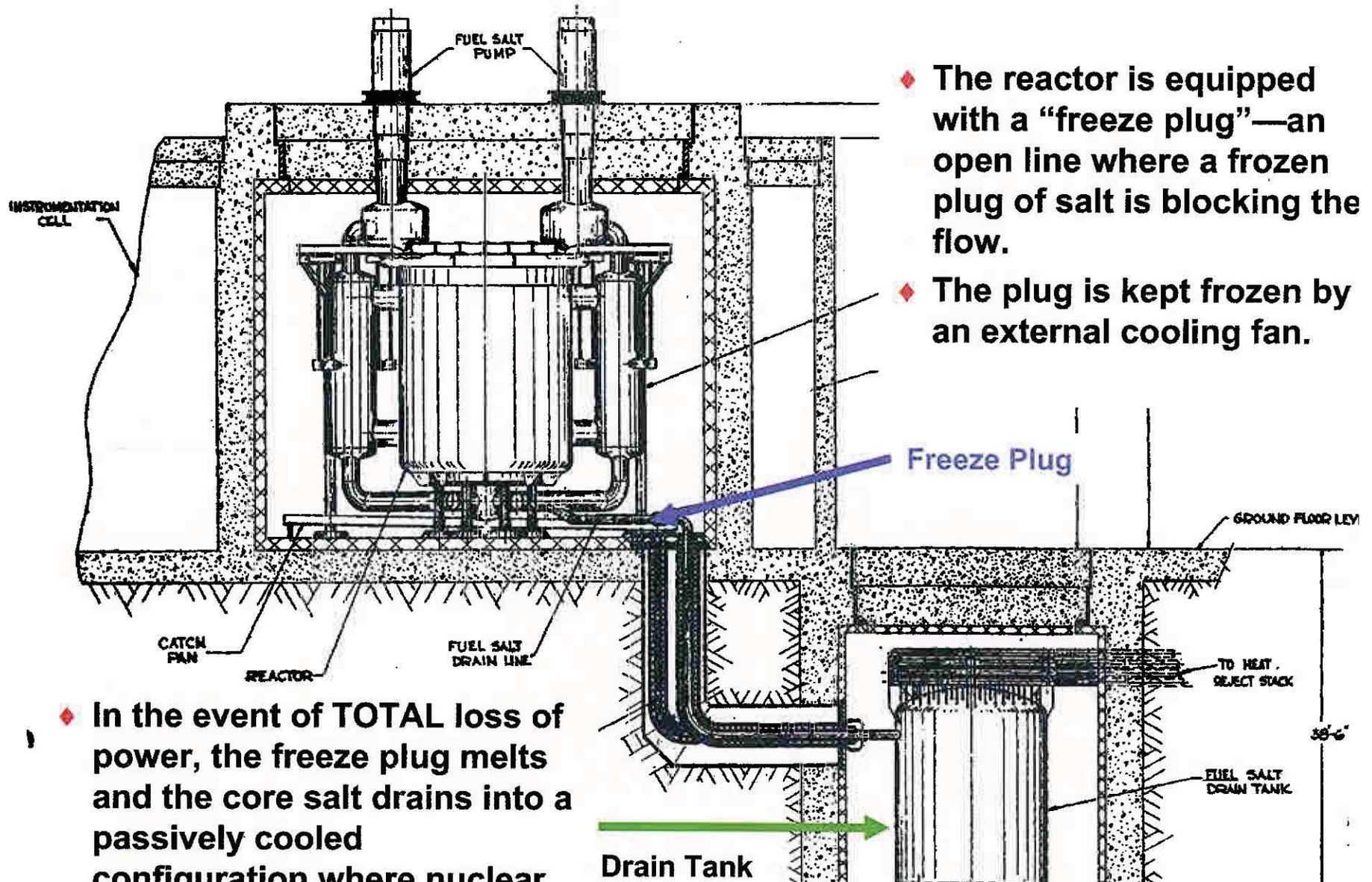
TURBO - ALTERNATOR POWER DISTRIBUTION
GAS MOLECULAR WEIGHT (KG/MOL) = 4.000
MASS FLOWRATE (KG/SEC) = 6.178
COMPR. PRESSURE RATIO = 3.200
NO. INTC. COMP. STAGES = 1
NO. REHEAT TURB. STAGES = 1
    COMPR. BLEED PCT. = .500
    COMPRESSOR POWER (MW) = 10.921
    TURBINE POWER (MW) = 16.190
    POWER LOSSES (MW) = 1.269
POWER SUM BALANCE (MW) = 5.000
(= GENERATOR TERMINAL
ELECTRICAL POWER-(MW))

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Isotope Production from LFTR



LFTR is passively safe in case of accident or sabotage

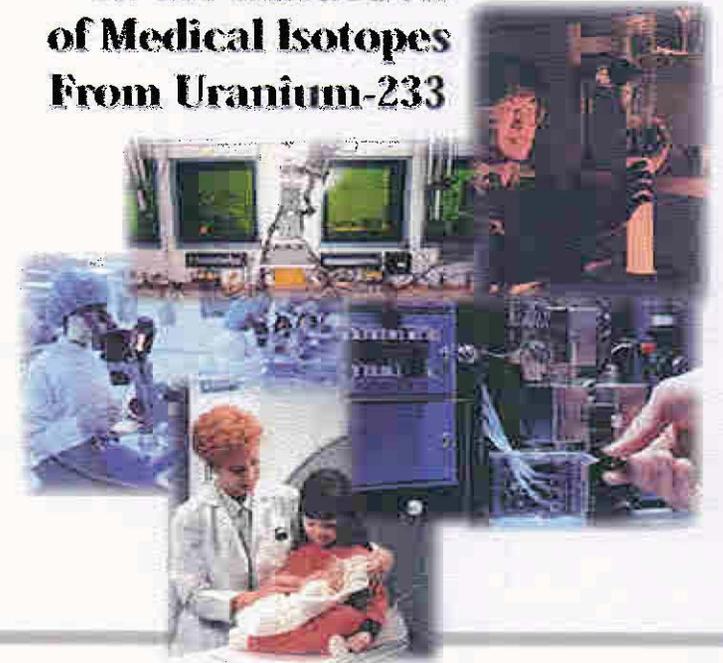


Congressional Report Emphasizes Need for Th-229

“Ac-225 and Bi-213 are currently derived from purified Th-229 extracted from U-233 at ORNL. **The only practical way at present is to derive these isotopes from the natural decay of Th-229. Th-229 is produced by the natural decay of U-233.** Ac-225 is the product being shipped to medical facilities. Bi-213 is separated from the Ac-225 at the hospital and combined with the targeting agent.

“**Bi-213 appears to be very potent, so only a very minute quantity may be needed to treat a patient...on the order of a billionth of a gram.**”

Report to Congress on the Extraction of Medical Isotopes From Uranium-233



U.S. Department of Energy
Office of Nuclear Energy, Science and Technology
Office of Isotopes for Medicine and Science

Power Generation Resource Inputs

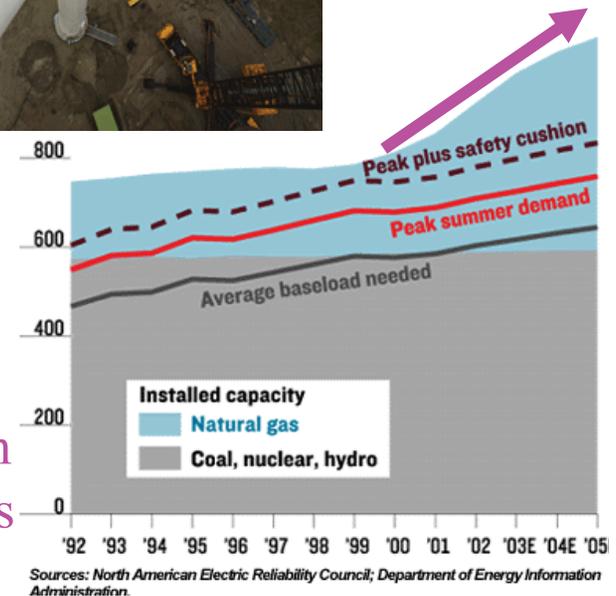
- Nuclear: 1970's vintage PWR, 90% capacity factor, 60 year life [1]
 - 40 MT steel / MW(average)
 - 190 m3 concrete / MW(average)
- Wind: 1990's vintage, 6.4 m/s average wind speed, 25% capacity factor, 15 year life [2]
 - 460 MT steel / MW (average)
 - 870 m3 concrete / MW(average)
- Coal: 78% capacity factor, 30 year life [2]
 - 98 MT steel / MW(average)
 - 160 m3 concrete / MW(average)
- Natural Gas Combined Cycle: 75% capacity factor, 30 year life [3]
 - 3.3 MT steel / MW(average)
 - 27 m3 concrete / MW(average)

Distance from end user, prime real estate, energy intensity, etc...



Cost of:

- materials
- labor
- land
- tools
- etc...



Thorium: Virtually **Limitless** Energy

World Thorium Resources	
Country	Reserve Base (tons)
Australia	340,000
India	300,000
USA	300,000
Norway	180,000
Canada	100,000
South Africa	39,000
Brazil	18,000
Other countries	100,000
World total	1,400,000

Source: U.S. Geological Survey, Mineral Commodity Summaries, January 2008

- Thorium is abundant around the world:
 - Found in trace amounts in most rocks and soils
 - India, Australia, Canada, US have large minable concentrations
 - US has about 20% of the world reserve base
- No need to hoard or fight over this resource:
 - A single mine site in Idaho could produce 4500 MT of thorium per year
 - Replacing the total US electrical energy consumption would require ~400 MT of thorium



The United States has buried 3200 metric tonnes of thorium nitrate in the Nevada desert.

There are 160,000 tonnes of economically extractable thorium in the US, even at today's "worthless" prices!

Relative Comparison: Uranium vs Thorium Based Nuclear Power

	<i>Uranium LWR</i> (light water reactor, high pressure low temp)	<i>Thorium LFTR</i> (liquid fluoride thorium reactor, low pressure high temp)
Plant Safety	Good (high pressure)	Very Good (low pressure, passive containment)
Burn Existing Nuclear Waste	Limited	Yes
Radioactive Waste Volume (relative)	1	1/30th
Waste Storage Requirements	10,000+ yrs.	~300 yrs.
Produce Weapon Suitable Fuel	Yes	No
High Value By-Products	Limited	Extensive
Fuel Burning Efficiency	<1%	>95%
Fuel Mining Waste Vol. (relative)	1000	1
Fuel Reserves (relative)	1	>1000
Fuel Type - Fuel Fabrication/Qualification	Solid Expensive/Long	Liquid Cheap/Short
Plant Cost (relative)	1 (high pressure)	<1 (low pressure)
Plant Thermal Efficiency	~35% (low temp)	~50% (high temp)
Cooling Requirements	Water	Water or Air
Development Status	Commercial Now	ORNL Demonstrated 1950-1970

Source: <http://www.energyfromthorium.com/ppt/thoriumEnergyGeneration.ppt>

U-232 Formation in the Thorium Fuel Cycle

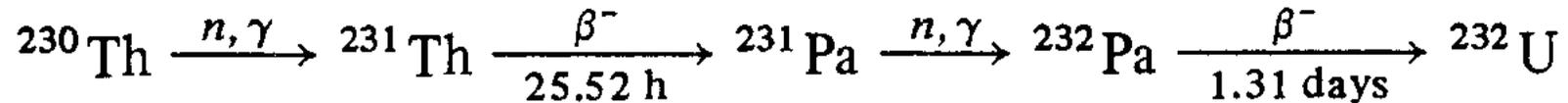
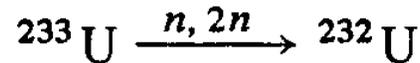
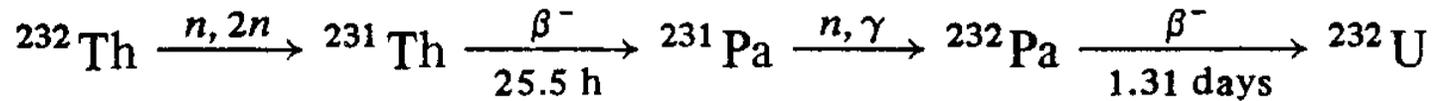


Table 2: Unshielded working hours required to accumulate a 5 rem dose (5 kg sphere of metal at 0.5 m one year after separation)

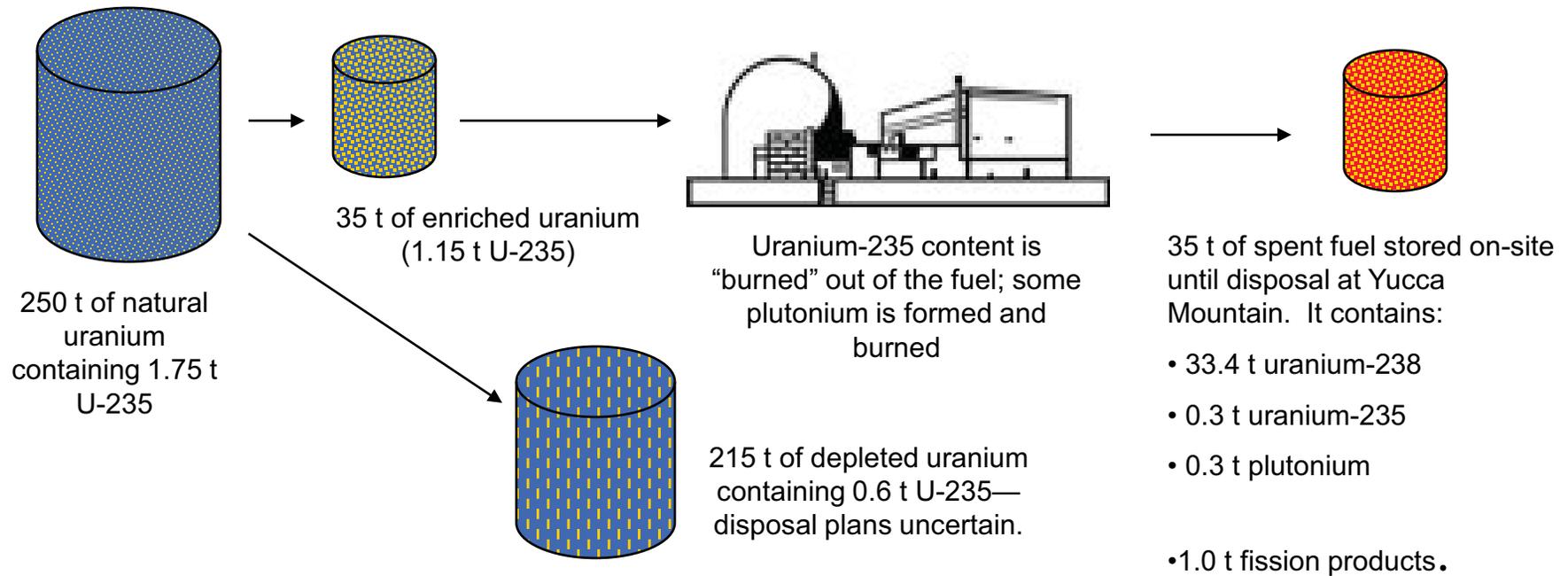
Metal	Dose Rate (rem/hr)	Hours
Weapon-grade plutonium	0.0013	3800
Reactor-grade plutonium	0.0082	610
U-233 containing 1ppm U-232	0.013	380
U-233 containing 5ppm U-232	0.059	80
U-233 containing 100 ppm U-232	1.27	4
U-233 containing 1 percent U-232	127	0.04

Return

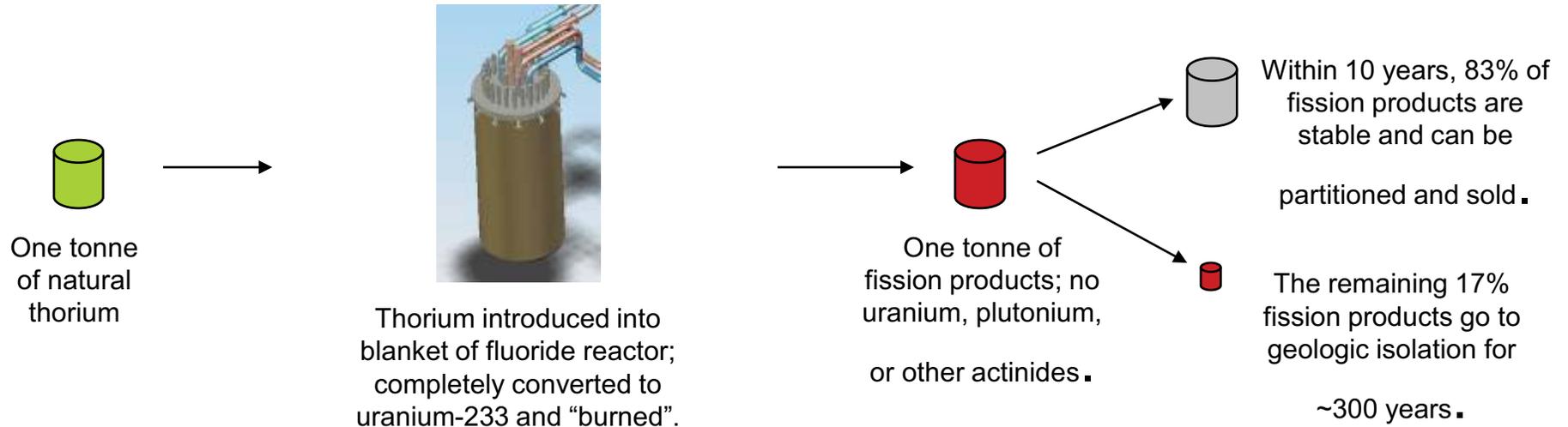
Typical Machine Sizes for 1000 MWe He Plant

- Single Turbo-Alt at 10 MP a and $Pr=2$; (TIT=1200K; TR=4)
 - Mass Flowrate ~ 1420 kg/sec
 - Dia. = 6.5 m; L = ~20 m; Speed = 1800 rpm
 - Recuperator Volume ~ 360 m³
 - Thermal Eff. = 48%
- Three Reheat/Intercooled Turbo-Alt's
 - Mass Flowrate ~ 474 kg/sec
 - P=20 Mpa ($Pr=2$); Dia = 1.9 m, L = 4.5m, Speed = 72000 rpm
 - P=10 Mpa ($Pr=2$); Dia = 2.7 m, L = 6.3m, Speed = 5400 rpm
 - P= 5 Mpa ($Pr=2$); Dia = 3.8 m, L = 8.5m, Speed = 3600 rpm
 - Recuperator Volume ~ 120 m³
 - Thermal Eff. = 51.5%

Today's approach to nuclear energy



Energy from thorium

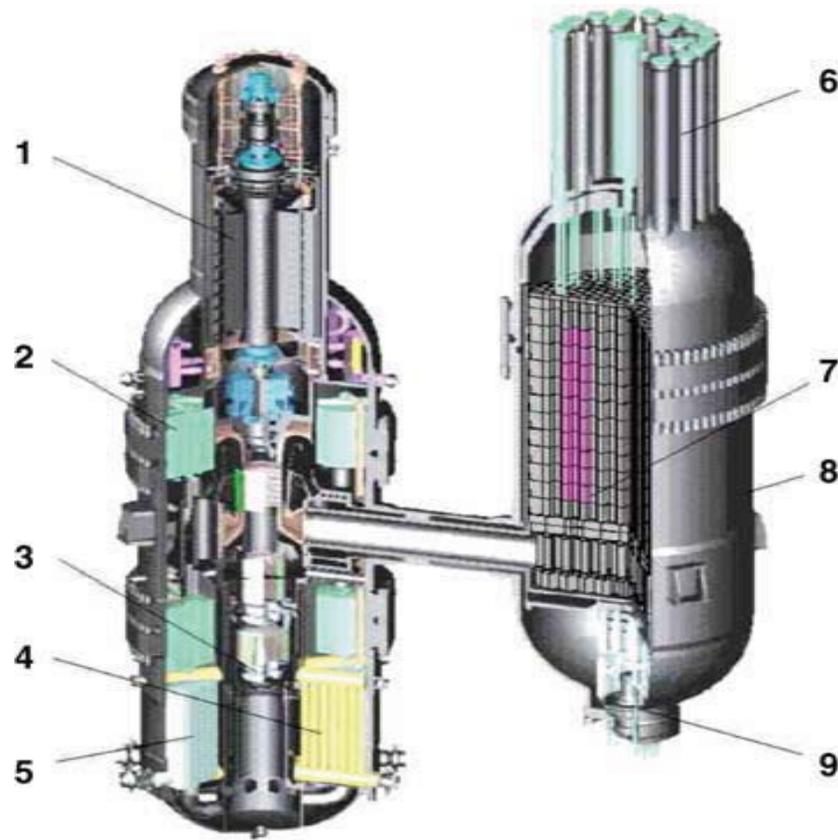


Acknowledgments & Disclaimer

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Any opinions, findings, results, conclusions, or recommendations expressed in this presentation are those of the authors and do not necessarily reflect the views of the National Aeronautics and Space Administration.

Proposed Vertical Orientation of GT-MHR* Turbomachinery



1 – Generator
2 – Recuperator
3 – Turbocompressor
4 – Intercooler
5 – Pre-cooler

6 – Control Rod Drive
7 – Core
8 – Vessel System
9 – Reactor Shutdown
Cooling System

* Gas Turbine Modular
Helium Reactor

Thorium: Virtually **Limitless** Energy

World Thorium Resources	
Country	Reserve Base (tons)
Australia	340,000
India	300,000
USA	300,000
Norway	180,000
Canada	100,000
South Africa	39,000
Brazil	18,000
Other countries	100,000
World total	1,400,000

Source: U.S. Geological Survey, Mineral Commodity Summaries, January 2008

- Thorium is abundant around the world:
 - Found in trace amounts in most rocks and soils
 - India, Australia, Canada, US have large minable concentrations
 - US has about 20% of the world reserve base
- No need to hoard or fight over this resource:
 - A single mine site in Idaho could produce 4500 MT of thorium per year
 - Replacing the total US electrical energy consumption would require ~400 MT of thorium



The United States has buried 3200 metric tonnes of thorium nitrate in the Nevada desert.

There are 160,000 tonnes of economically extractable thorium in the US, even at today's "worthless" prices!