



# Compact Full-Field Ion Detector System for SmallSats beyond LEO

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MeTRI SUNT NECESSE MALUM

“Measurements are necessary evils”

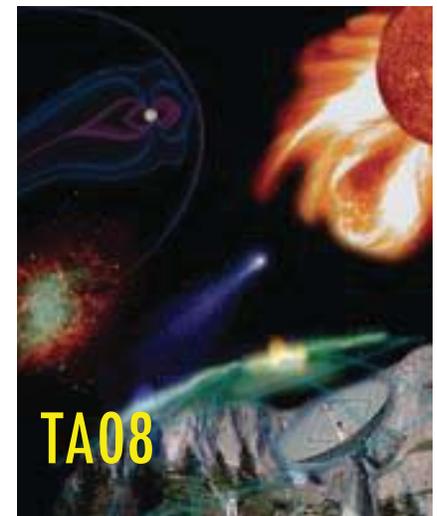


# Technology Roadmap Challenges

- NASA's Integrated Technology Roadmap (2012): Technology Area (TA) 8.3.1 "In-Situ Instrumentation/Detectors: Particles"

## Challenges:

- Energy Threshold (resolve to 1 keV for 30 MeV)
  - Environment Tolerance (radiation-hard ion & electron sensors)
  - Data Handling (improved out-of-band rejection)
  - TRL3→6: 2013→2016
  - Heliophysics, Planetary Science Missions
  - "Robust sensors capable of operating for long periods in environment of space are needed to measure the radiation at the destination as well as during the journey."
- TA08 Roadmap Enabling Approaches:
    - Integrated existing detector technologies
    - Radiation hardened electronics
    - Miniature power supplies
  - Alternative: New detectors





# SmallSat Platform Technology Challenges

- Goal is to develop a radiation detector system to fly on small satellite platforms (such as CubeSats) to reduce cost, development time of missions
  - Design point: 1U CubeSat volume, mass for detector system (10 cm x 10 cm x 10 cm, 1 kg)
- CubeSats currently flown LEO applications, but future is in Deep Space
  - High radiation particle influx from multiple directions (spherical  $4\pi$  solid angle)
- Current radiation detector technologies need temperature compensation
  - SmallSat platform size (<100 kg), power limits instrumentation systems
  - More complex systems require new technology
- Solution is the development of new robust, low power, thermally stable solid state radiation detector technology for omni-directional measurements in a compact space radiation detector system
  - Wide band gap semiconductors, micro-optics technologies



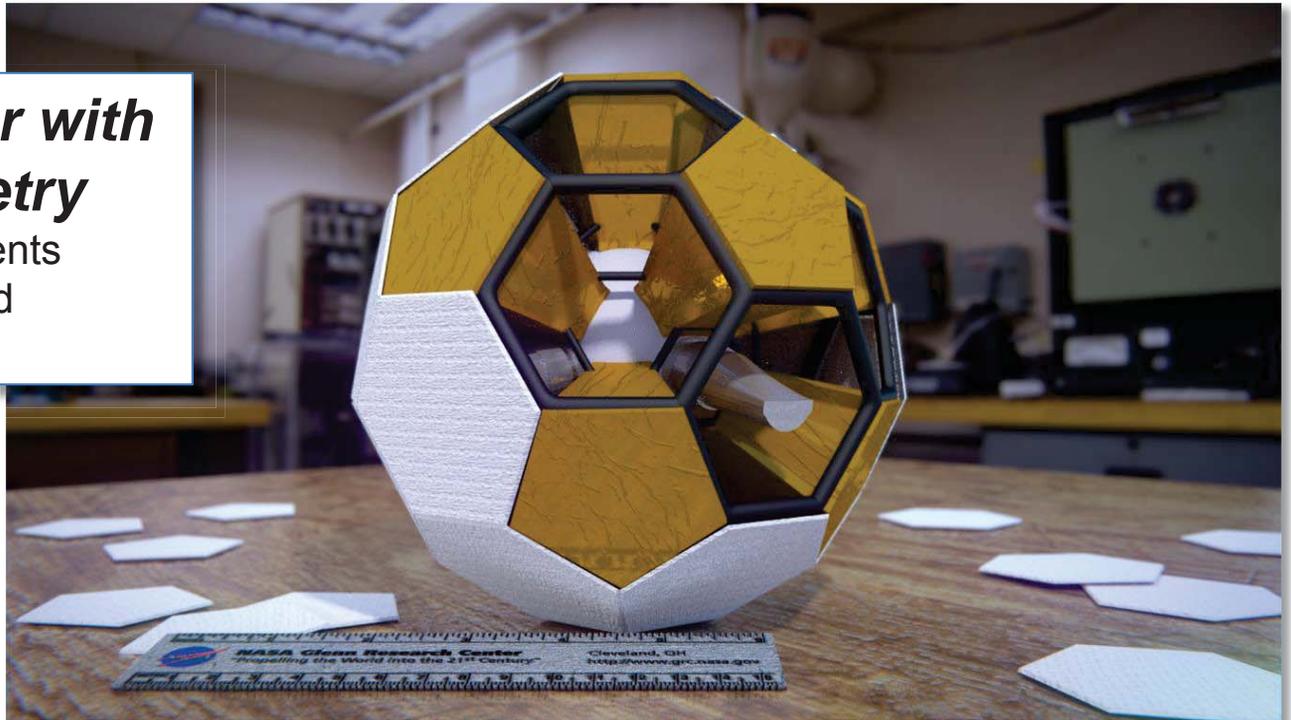
# Application Concept: Compact Full-Field Ion Detector System (CFIDS)

- Mapping of heavy ions  $> 100$  MeV/amu
  - Integrated system with solid-state Cherenkov detector and large area detectors in surrounding wedges
- High radiation flux rates for 10+ year missions
  - Precision rad-hard, thermally stable wide band gap detectors used
- Low noise, multi-directional measurements at single locations
  - Compact, spherical detector system

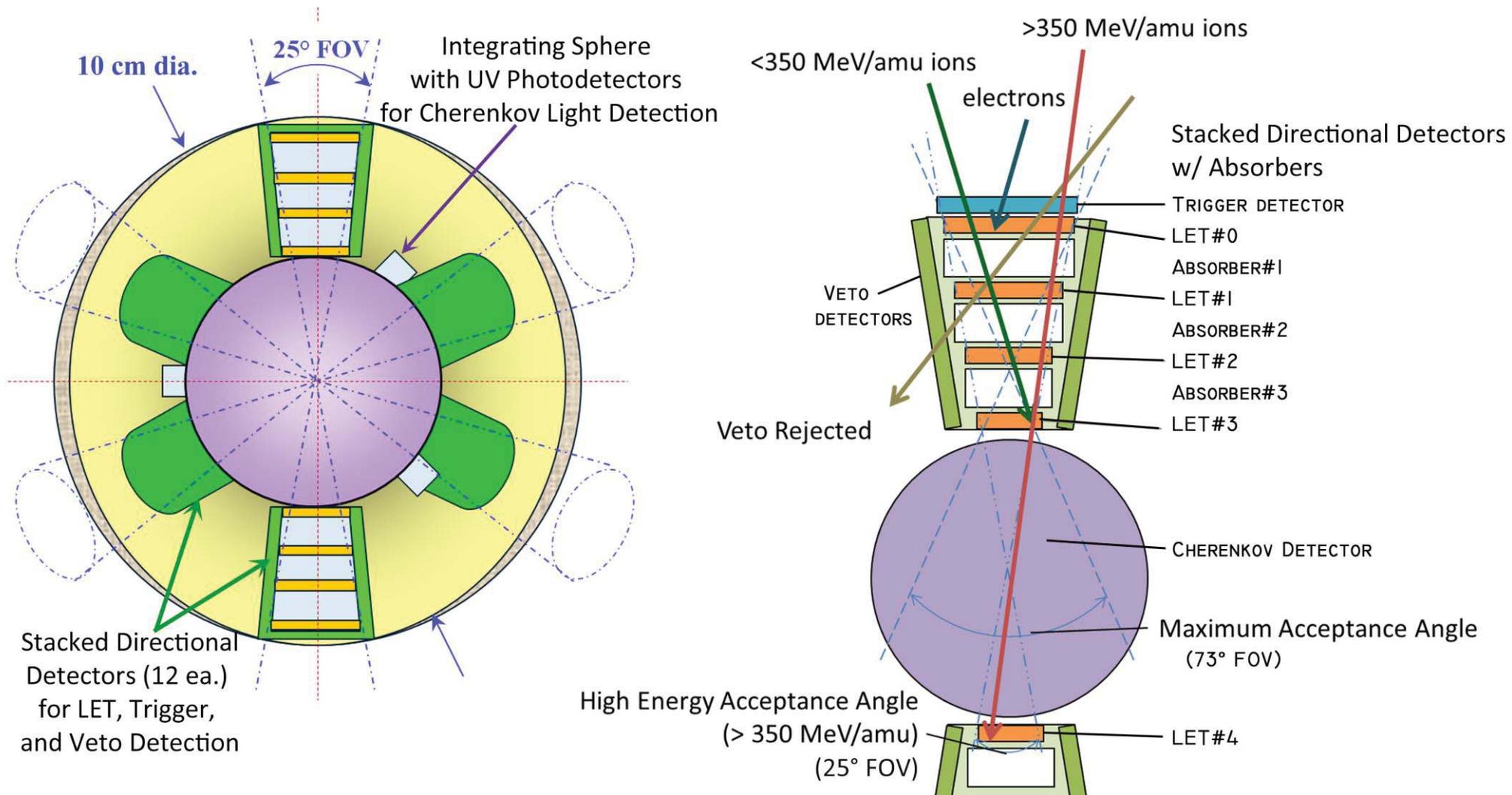
## ***Space radiation detector with spherical geometry***

- Technology covered by U.S. Patents 7,872,750 (January 18, 2011) and 8,159,669 (April 17, 2012)

**Concept illustration of the CFIDS detector assembly (cables, electronics not shown)**



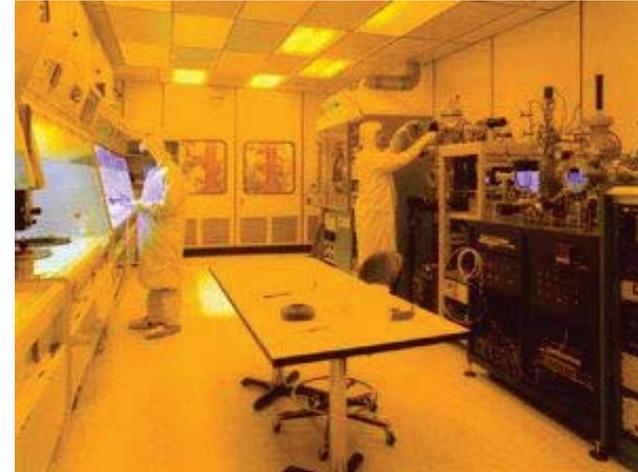
# Application Concept: Compact Full-Field Ion Detector System (CFIDS)



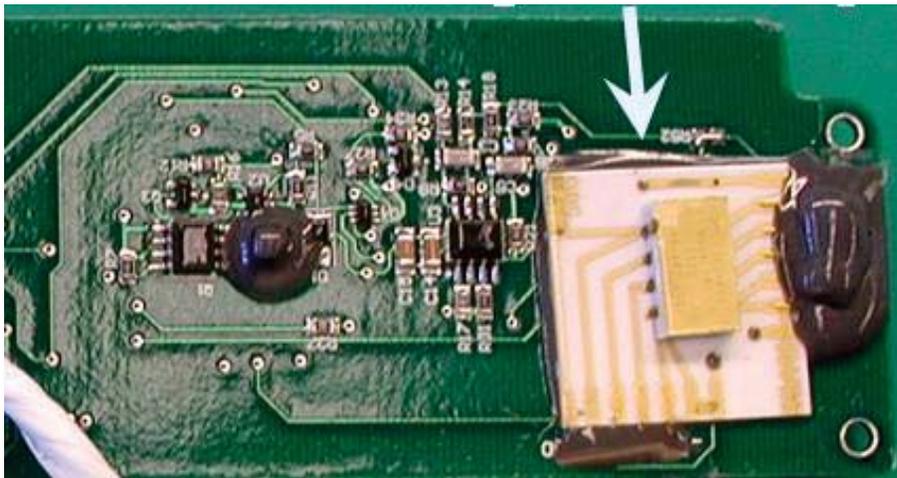
- CFIDS comprised of a spherical Cherenkov detector surrounded by stacked LET detectors with absorbers, Trigger and Veto detectors

# GRC Advanced Radiation Detector Technology Research and Development

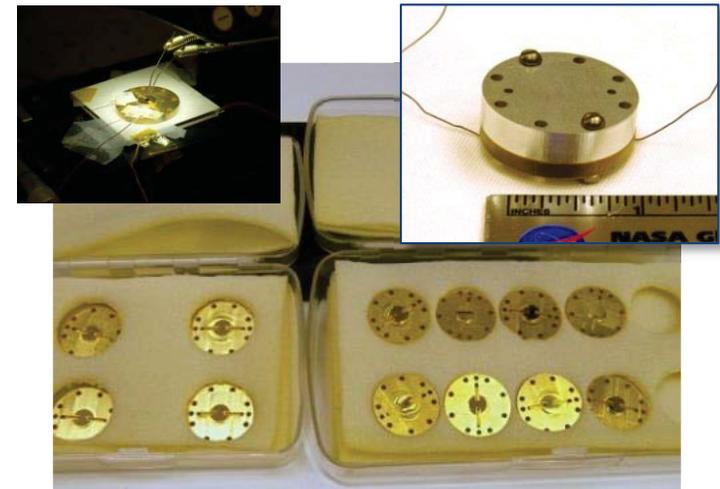
- GRC Expertise and Facilities in:
  - Harsh Environment Thin Films
  - SiC Devices & Harsh Environment Packaging
  - Micro-Optics
  - Space-Based Instrumentation
- These strengths are combined into an in-house Radiation Instrumentation Research effort



**In-House Microsystems Fabrication**



**MISSE 7 SiC JFET & Ceramic Packaging (arrow)  
on a Rad-Hard Electronics Board for ISS flight**



**CERES Thin Film Microbolometer  
Testing and Packaging**

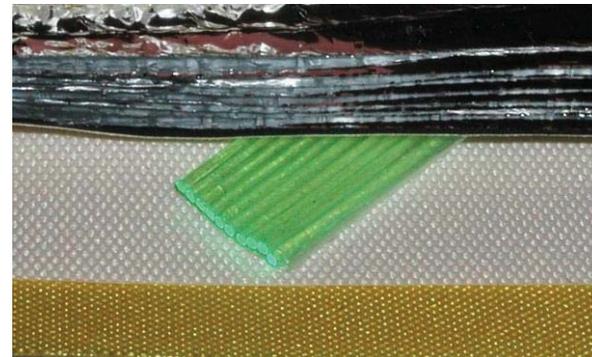


# Solid-State Trigger/Veto Detectors

- Typically scintillator blocks of plastic or iodide crystal mated to a photomultiplier tube (PMT) or a pixelated avalanche photo detector (APD), also referred to as a silicon photomultiplier (SiPM)
- Goal: Replace the role of PMTs and SiPMs in these types of detectors with WBG devices, saving on size, weight and required power
- Demonstrated a miniature gallium phosphide (GaP) photodiode “paddle style” radiation detector as part of a 10-week OCT/STMD Center Innovation Fund (CIF) study in 2013 (patent pending).
- Use with acrylic ribbon scintillators for the CFIDS concept



**Miniature scintillation/diode ionizing radiation detector**

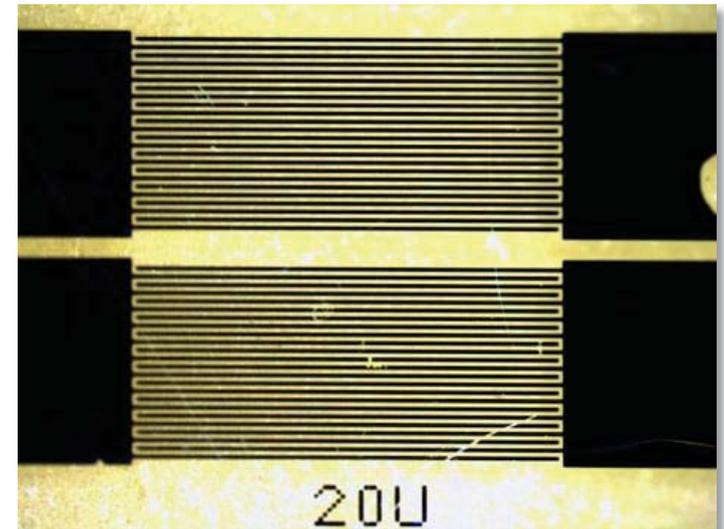


**Acrylic ribbon scintillator for ionizing radiation detector**



# Solid-State Cherenkov Detector

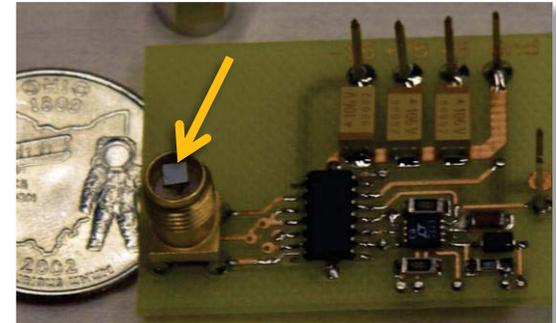
- Typically flat disks or blocks of sapphire or acrylic mounted on PMTs.
- Goal: Replace the role of the relatively large PMTs with solid-state devices that do not require temperature control or compensation.
- A fast, large area solid-state UV detector based on single-crystal, undoped zinc oxide (ZnO) was developed at GRC (patent pending) as part of two 10-week OCT/STMD CIF studies (2011, 2012)
  - Active area of 1 mm by 2 mm (2 mm<sup>2</sup>), designed to have a 1 ns response time with 10 V applied bias voltage
  - In a bridge circuit can detect small, fast pulses of UV light like those required for Cherenkov detectors.
  - Sensitive to UV light at 254 nm, slightly less so at 370 nm, and not sensitive to room lighting (about 430-630 nm).
  - Demonstrated improved sensitivity to UV than commercial SiC and GaP detectors



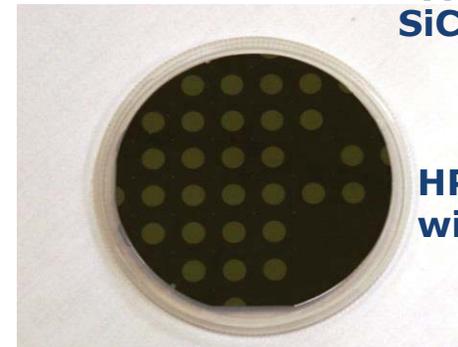
**OCT ZnO UV Detector  
(20  $\mu$ m electrode spacing)**

# WBG LET Detectors

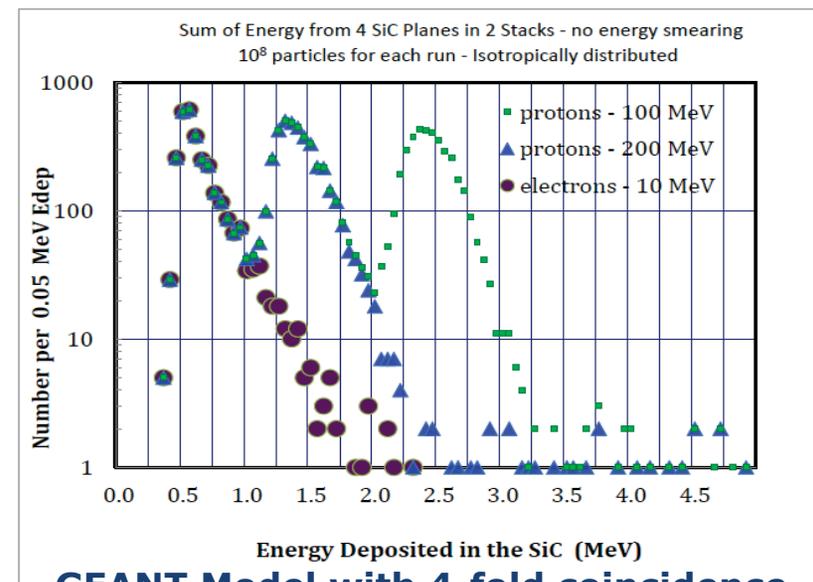
- Typically silicon-based PIN diodes or lithium-drifted silicon wafers (Si(Li)), high bias voltage, thermally sensitive
- Goal: Replace silicon detectors with more robust, temperature-stable low-noise silicon carbide detectors
- Smaller SiC detectors studied as part of AEVA (2005-2007) and ETDP/D (2009-2011), AES (2012) for dosimetry
- Large-area detectors (2 cm<sup>2</sup>, 350 μm thick) using high-purity, semi-insulating (HPSI) SiC wafers with low-Z FEP absorber between detectors for CFIDS
- GEANT models show a 4-fold coincidence is required to resolve LET for high energy protons and electrons



**Prototype Dosimeter with SiC detector (arrow)**



**HPSI 4H-SiC wafer with device pads**



**GEANT Model with 4-fold coincidence**

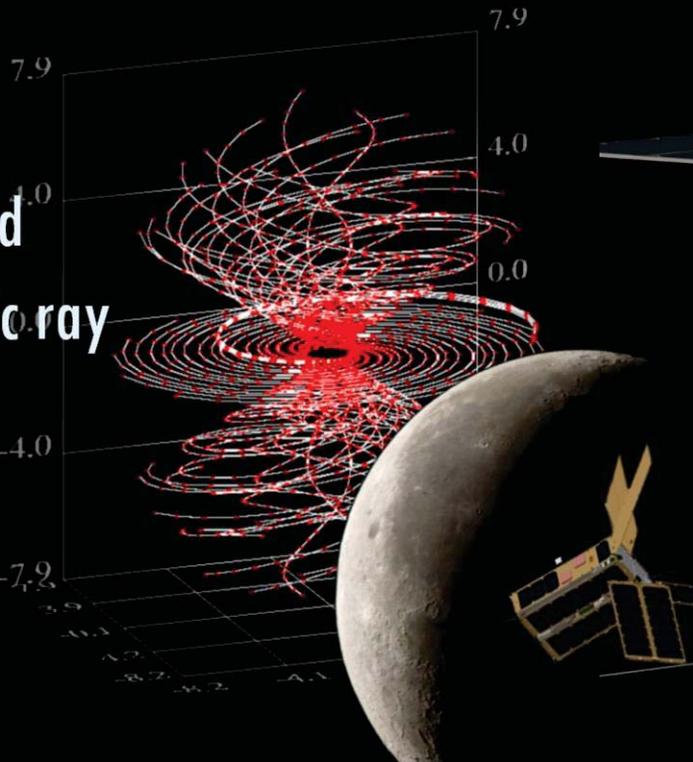


# Benefits of WBG Detectors: Lower Power and More Robust

Detector	Active Area	Mass	Volume	Voltage	Dark Current	Minimum Power Draw	Amplitude Signal to Noise	Maximum Operating Temperature	Temperature Sensitivity of Dark Current
<b><u>Cherenkov Detector:</u></b>									
SOA: PMT	20 cm <sup>2</sup>	170 g	180 cm <sup>3</sup>	1000 V	5 nA	5 μW	4x10 <sup>5</sup>	50°C	0.2%/°C
Proposed: ZnO	2 mm <sup>2</sup>	11 g	0.80 cm <sup>3</sup>	10 V	5 nA	0.05 μW	2x10 <sup>4</sup>	125°C	0.05%/°C
<b><u>LET:</u></b>									
SOA: Si PIN	1 cm <sup>2</sup>	0.5 g	185 mm <sup>3</sup>	100 V	5 nA	0.5 μW	1x10 <sup>5</sup>	60°C	20%/°C
SOA: Si(Li)	30 cm <sup>2</sup>	35 g	15 cm <sup>3</sup>	300 V	5 μA	1.5 mW	8x10 <sup>3</sup>	60°C	30%/°C
Proposed: SiC	2 cm <sup>2</sup>	0.5 g	185 mm <sup>3</sup>	5 V	70 pA	0.350 nW	2x10 <sup>5</sup>	120°C	0.1%/°C
<b><u>Scintillator Trigger/Veto:</u></b>									
SOA: PMT	20 cm <sup>2</sup>	170 g	180 cm <sup>3</sup>	1000 V	5 nA	5 μW	4x10 <sup>5</sup>	50°C	0.2%/°C
SOA: APD	9 mm <sup>2</sup>	3 g	200 mm <sup>3</sup>	30 V	5 nA	0.15 μW	8x10 <sup>4</sup>	85°C	30%/°C
Proposed: GaP	4.8 mm <sup>2</sup>	5 g	170 mm <sup>3</sup>	5 V	20 pA	0.1 nW	3x10 <sup>5</sup>	125°C	0.5%/°C

# SPAGHETI: Deep-Space Application

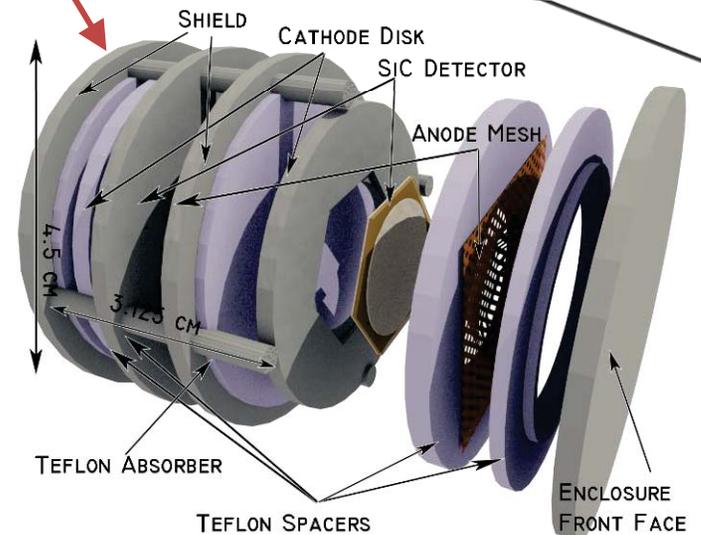
**Solar  
Proton  
Anisotropy and  
Galactic cosmic ray  
High  
Energy  
Transport  
Instrument**



WBG LET  
Detector  
Stacks

SPAGHETI  
CubeSat

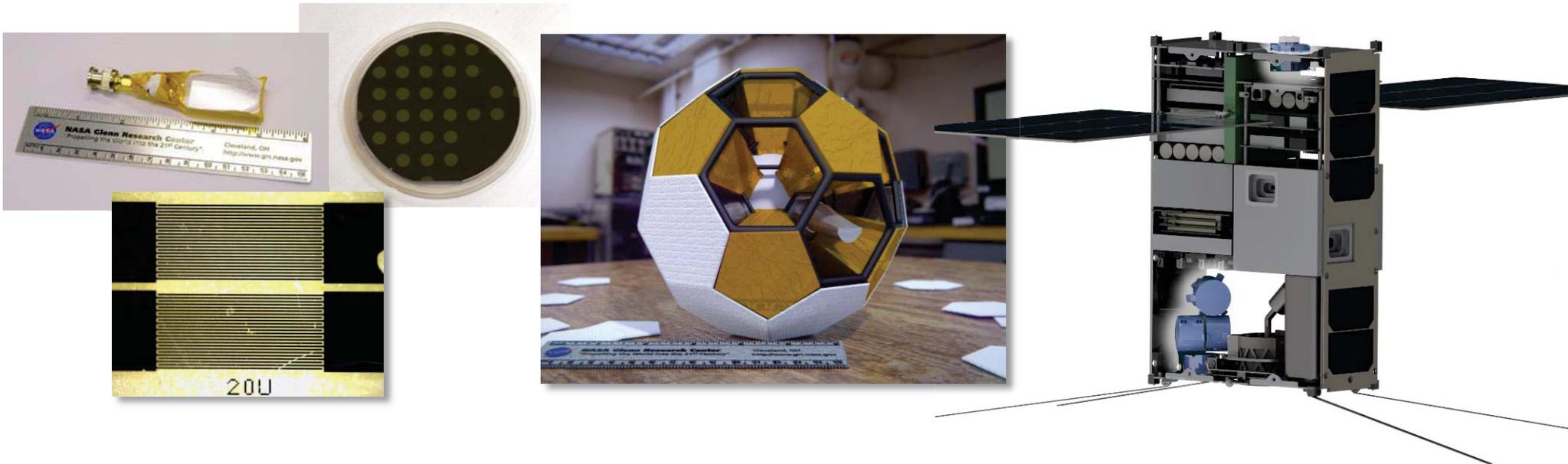
- SPAGHETI is a SmallSat mission for an EM-1-type launch on a 6U CubeSat bus enabled by stacks of WBS LET detectors
- SPAGHETI will explore the transient variations in ion flux anisotropy in deep space and near the lunar surface





# Summary

- NASA GRC is leveraging expertise in harsh environment thin films, SiC devices & harsh environment packaging, micro-optics, and space-based instrumentation to advance radiation detector technology
- Application of wide band gap semiconductors as radiation detectors holds the promise of improved low-power, robust detectors for CFIDS
- SPAGHETI using CFIDS radiation instrumentation system in a Deep Space CubeSat will allow in-situ studies of SEP and GCR interactions in lunar environments





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- Elizabeth McQuaid and Nicholas Varaljay (GRC)
  - ZnO UV detector fabrication
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  - SiC dosimeter diode detector fabrication
- Dr. Nathan Schwadron (University of New Hampshire)
  - SPAGHETI collaboration
- Dr. Ben Malphrus (Morehead State University)
  - SPAGHETI CubeSat bus architecture
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  - CFIDS, SPAGHETI support

