



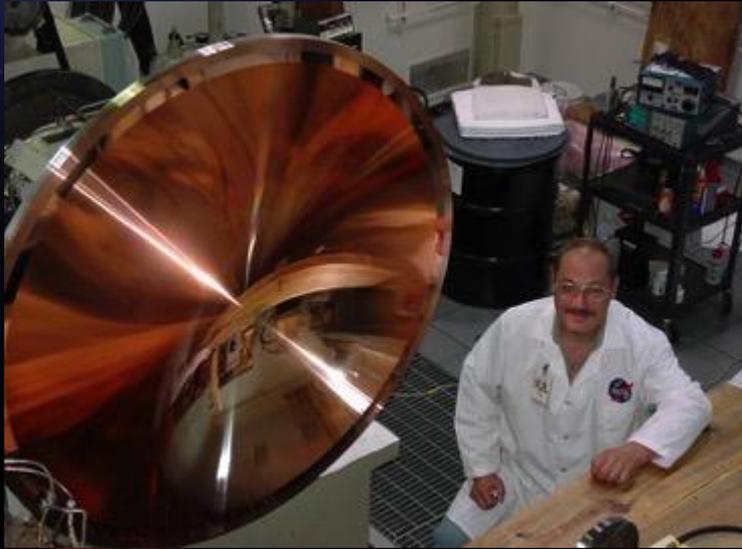
Marshall Space Flight Center Optics Capabilities

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Precision Engineering/Diamond Turning



1.5 m copper mold for pressing
Fresnel lenses cut under Space Act
Agreement



MOORE M-40

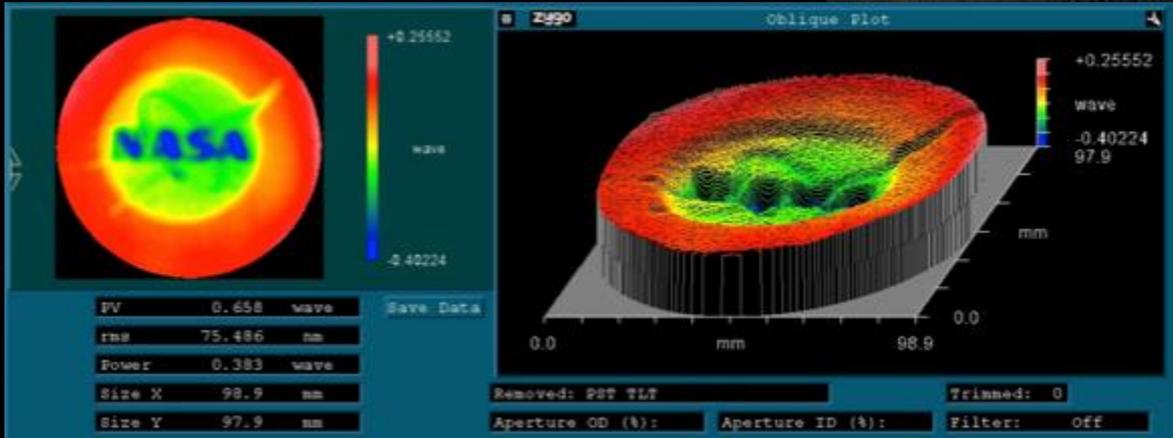


Facility currently consist of two ultra high precision diamond turning machines with face turning capabilities of 2 meters and 0.4 meters, cylinder turning to 1 meter, one EDM and two high precision CNC machine tools and conventional machine tools. Extensive metrology support is available.



Optical Shop

- Equipment includes curve generators, spindle grinders/polishers, a Blanchard, an edger and a 48 inch continuous polisher.
- Custom built polishing machines that are capable of polishing X-ray mirror mandrels 40 - 500 mm in diameter and 305 - 610 mm in length to less than 5 arcsec in figure and less than 4 Å roughness.
- Zeeko IRP600 Intelligent Robotic Polisher machine able to grind and polish parts up to 600 mm in diameter to a surface roughness of 5 Å.
- OptiPro 300 6-axis Free Form Polisher able to grind and polish parts up to 300 mm in diameter to a surface roughness of 5 Å.



NASA logo ground into a glass flat in approximately 1 hour on the Zeeko.



Metrology Facilities/Equipment



Zygo GPI with 32 inch beam expander & Mark IV with 18-inch beam expander



Coordinate Measuring Machine

- 1 micron accuracy
- parts up to 1 m



2 Zygo NewView optical profilometers

- Sub Angstrom vertical resolution.
- Sub micron lateral resolution.



Vertical Long-Trace Profilometer

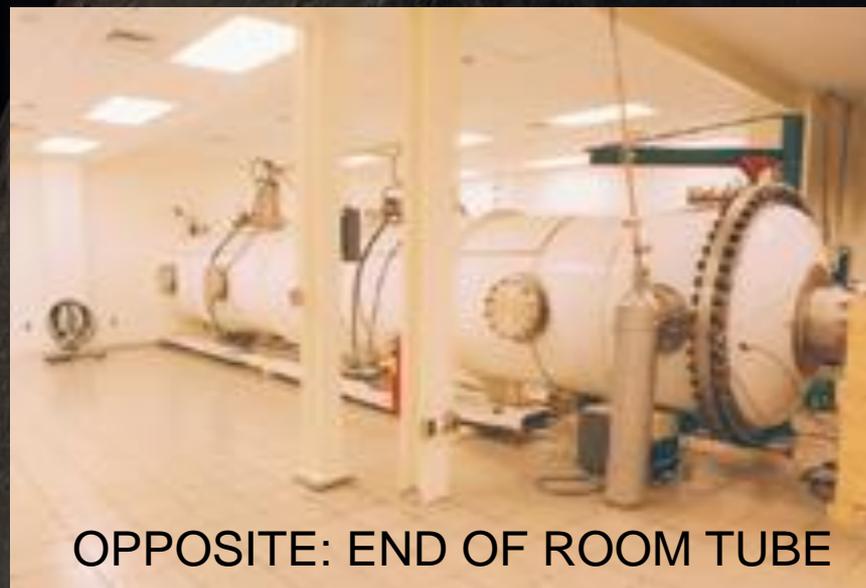
- One of two in existence
- Incorporates rotary air bearing table
- Scan Length: 0.7 m
- Range: 10 m rad.
- Accuracy: 10 nm surface height (theoretical)
- Cylinders/shells up to 0.7-m long x 0.75-m diameter



Stray-light Test Facility



CLASS 10K CLEAN ROOM
AND CHAMBER



OPPOSITE: END OF ROOM TUBE



MAIN CHAMBER AND PREP AREA

3 by 12-m test volume for baffle or mirror

1.3-m diameter, 82-m long section

1.5-m diameter, 10-m isolatable section

Pumped with cryo-pump: $<10^{-7}$ torr

Measured baffle rejection ratios up to 10^{15}

Use to test x-ray optics up to 1-m dia

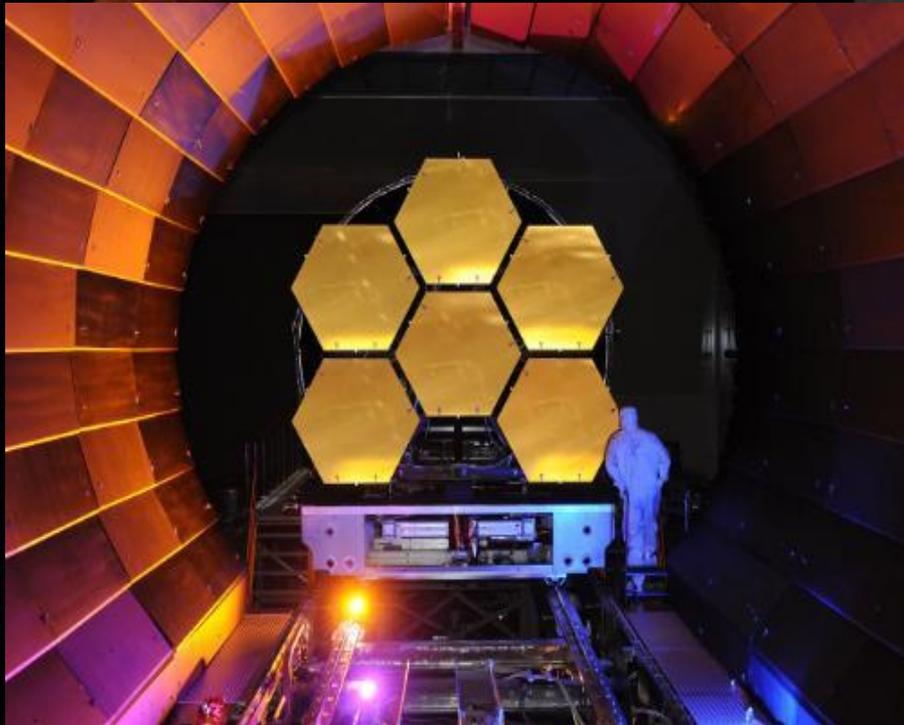


X-Ray and Cryogenic Facility (XRCF)

James Webb Space Telescope flight mirror segments were tested in the (XRCF), the same facility that tested the Chandra X-ray telescope at MSFC. The test chamber offers the unique capability for simulating a space environment with low temperature and pressure.

Vacuum Chamber

- 7.3 x 22.8 m Polished Stainless Steel 10⁻⁷ Torr Vacuum Chamber
- Full 155 to 355K Thermal Shroud – Helium shroud to 20K
- Vibration Isolated via Seismic Mass
- 5DOF Remote Controlled Test Stand





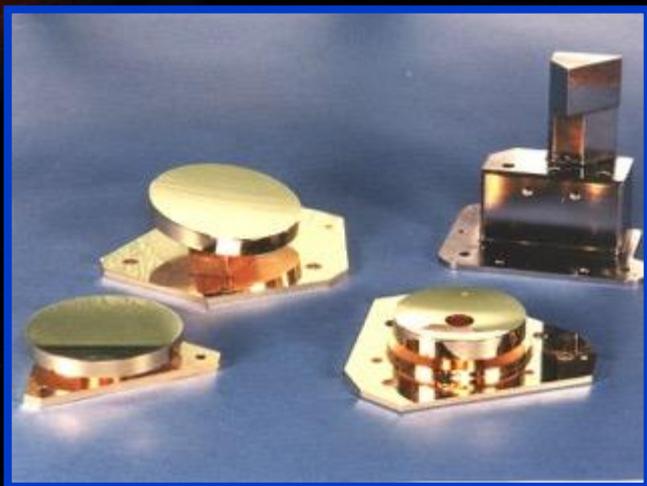
FLIGHT MIRROR DEVELOPMENT



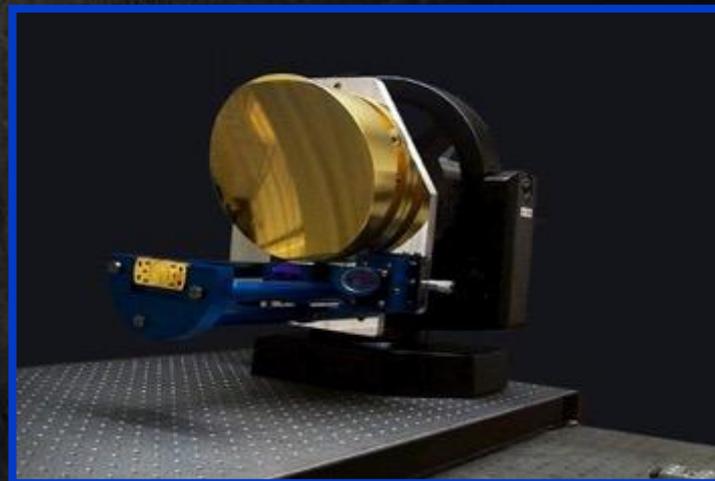
Solar X-ray Imager Mirror on GOES-12 satellite



UV Imager Mirrors on POLAR satellite



Composite Infrared Spectrometer (CIRS) Mirrors on CASSINI



Sparcle Lidar Beam Expander (flight certified but not flown)



Return to Flight Imaging

- Supplied the optical system for the airborne imaging of the Space Shuttle at launch and on portions of the reentry.
- Operates in unpressurized nose ball of a WB-57 aircraft at 60,000 foot altitude
- Visible & NIR, Schmidt-Cassegrain, 28 cm diameter primary mirror, 2.8-meter focal length
- Completed the design using COTS equipment
- Manufactured the optical bench and performed the optical integration
- Environmentally tested the system prior to flight.
- System continues to be flown in support of Eastern Test Range and KSC launches.



STS-115 – September 9, 2006



Pluto New Horizons – January 19, 2006

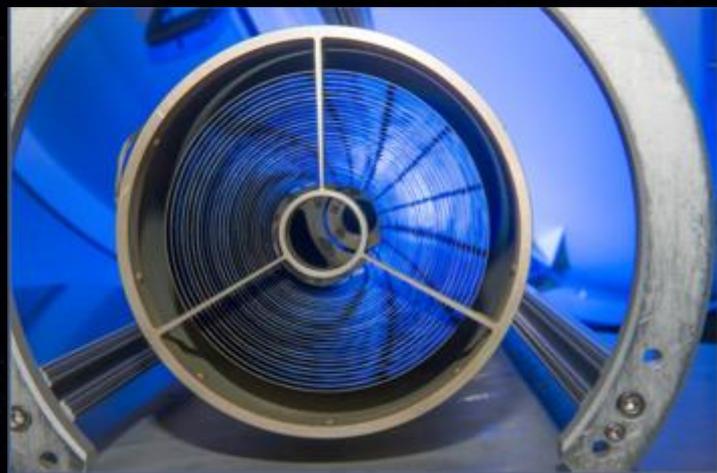


Astronomical Roentgen Telescope X-Ray Concentrator (ART-XC)

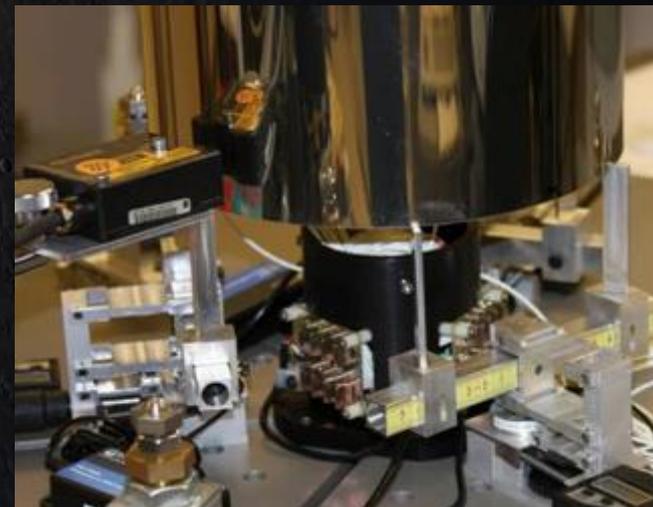
- Manufactured for the Russian Spectrum Mission
- 28 precision mandrels for the replication of ~ 200 mirror shells (15 arcsec figure, 5 Å roughness)
- 30 keV (0.083 - 1.24 nm), Wolter 1, 5.0 to 14.9 cm mirror diameter, 2.7-meter focal length
- Produce 7 flight modules plus 1 spare unit including mirror housings and support spiders
- Tested modules in Stray Light Facility



Mandrel Inspection



Module in Handling Fixture



Shell alignment



Replicated Optics Manufacturing Process

1. CNC machine, mandrel formation from Al Bar



2. Chemical clean and activation & Electroless Nickel (EN) plate



3. Precision turn to sub-micron figure accuracy



4. Polish and superpolish to 3-4 Å finish



5. Metrology – repeat Step 4 until surface finish met



6. Ultrasonic clean and passivation to remove surface contaminants



7. Electroform nickel shell onto mandrel



8. Separate optic from mandrel – reuse mandrel for next shell



8. Align shells into module



9. Test module





Sounding Rockets

Solar Ultraviolet Magnetospheric Investigation (SUMI)

- SUMI flew on a sounding rocket in 2010
- $\lambda=155$ & 280 nm, Ritchey-Chretien, 30 cm diameter primary
- Provided primary and secondary mirrors, heat rejection mirror, four fold mirrors, two off-axis parabolas and two diffraction gratings



SUMI Integration

High Resolution Coronal Imager (Hi-C)

- Hi-C launched on a sounding rocket on July 11, 2012 and obtained the highest resolution images of the Sun's corona ever acquired
- $\lambda=193$ Å, Cassegrain, 22 cm diameter primary mirror, 23.0-meter focal length
- Provided the primary and secondary aspheric mirrors to the Smithsonian Astrophysics Observatory (SAO)
- The primary was hand polished to a slope error of .09 arcsec and the secondary to .25 arcsec



Hi-C Primary polishing



MSFC Partnerships Office



Reimbursable SAA - Money coming into NASA

- Permits the partner to use NASA goods, services, facilities, or equipment to advance the partner's own interests
- Primary benefit to partner that is consistent with NASA's mission.

Non-reimbursable SAA - No funds exchanged

- Used to support collaborative technology development, outreach activities and educational partnerships.
- Mutually beneficial activity that furthers NASA's mission
 - Not used to obtain services from partner
- Look for "quid pro quo" contribution between NASA and partner.

Identify opportunity – Could NASA assist me with this?

Evaluate the possibility to partner – Is the opportunity within NASA's authorization

Develop agreement jointly – NASA and the partner agree on scope, schedule and cost

Capture and finalize the agreement – Good coordination ensures a timely review and approval

Execute the agreement – another successful partnership with NASA begins

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Summary of MSFC Unique Capabilities

- ***MSFC has a unique capability to manufacture and test grazing incidence optics.***
 - The capability to develop, fabricate and test electroformed nickel optics at MSFC is unique in the United States; in fact, there are only two such capabilities in the world, the other residing in Italy.
 - MSFC has state of the art metrology capabilities to test and verify that the grazing incidence mirrors meet design requirements.
 - The Stray Light Test Facility (SLTF) is a smaller less expensive and more flexible alternative to the XRCF.
- **MSFC world class optical capabilities include: Moore M-40 Diamond Turning Machine, Vertical Long Trace Profilometer (VLTP), X-Ray Cryogenic Facility (XRCF) and Stray Light Facility**
- **MSFC is a vital participant in manufacturing and testing optics for a range of customers. Partnerships exists with NASA/GSFE, NASA/KSC, Dept. of Energy, National Institute of Health, DARPA, SAO, MIT, UC Berkley, University of Iowa and others.**