Multiple designs, systems, methods and processes for controlling a system or plant using an extended active disturbance rejection control (ADRC) based controller are presented. The extended ADRC controller accepts sensor information from the plant. The sensor information is used in conjunction with an extended state observer in combination with a predictor that estimates and predicts the current state of the plant and a co-joined estimate of the system disturbances and system dynamics. The extended state observer estimates and predictions are used in conjunction with a control law that generates an input to the system based in part on the extended state observer estimates and predictions as well as a desired trajectory for the plant to follow.

8 Claims, 5 Drawing Sheets
Related U.S. Application Data

Provisional application No. 60/965,359, filed on Aug. 20, 2003, provisional application No. 60/718,393, filed on Sep. 19, 2003, provisional application No. 60/718,581, filed on Sep. 19, 2003, provisional application No. 60/718,899, filed on Sep. 20, 2003, provisional application No. 60/728,928, filed on Oct. 20, 2003, provisional application No. 60/728,929, filed on Oct. 20, 2003, provisional application No. 60/373,404, filed on Apr. 18, 2002.

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Fig. 4

$G_c(s)$

$G_r(s)$

$H(s)$

$u$

$d$

$y$

$\ldots$

$\ldots$

$r$

$\ldots$

$\ldots$
EXTENDED ACTIVE DISTURBANCE REJECTION CONTROLLER

CROSS-REFERENCE TO RELATED APPLICATIONS


This work was supported at least in part by NASA under NASA contract number GT3-52387. Accordingly, the United States government may have certain rights therein.

TECHNICAL FIELD

The subject matter described herein relates to controllers, systems, and methods for feedback control of various systems.

BACKGROUND

A feedback (closed-loop) control system 10, as shown in Prior Art FIG. 1, is widely used to modify the behavior of a physical process, denoted as the plant 110, so it behaves in a specific desirable way over time. For example, it may be desirable to maintain the speed of a car on a highway as close as possible to 60 miles per hour in spite of possible hills or adverse wind; or it may be desirable to have an aircraft follow a desired altitude, heading and velocity profile independently of wind gusts; or it may be desirable to have the temperature and pressure in a reactor vessel in a chemical process plant maintained at desired levels. All of these industrial tasks are accomplished today by using traditional feedback control, and the above are examples of what automatic control systems are designed to do, without human intervention.

The key component in a feedback control system is the controller 120, which determines the difference between the actual output of the plant and the desired output as small as possible as rapidly as possible. Today, controllers are employed in a large number of industrial control applications and in areas like robotics, aeronautics, astronautics, motors, motion control and thermal control, just to name a few.

INDUSTRIAL APPLICABILITY

The present system and method has potential applicability to a wide range of different industrial and commercial applications. The following brief synopsis is intended only to provide background on some exemplary applications to assist a person of ordinary skill in the art in understanding this disclosure more fully. Additional background information on a select number of exemplary industrial applications are also provided.

BACKGROUND OF THE INVENTION

Classic Control Theory provides a number of techniques for an engineer to use in controller design. Existing controllers for linear, time invariant, and single-input single-output plants are categorized in one taxonomy into three forms: the proportional/integral/derivative (PID) controllers, transfer function based (TFB) controllers, and state feedback (SF) controllers. The PID controller is defined by the equation:

\[ u = K_p e + K_i \int e + K_d \frac{d e}{d t} \]  

where (t) is the control signal and (e) is the error between the setpoint and the process output (y) being controlled. As commonly used, the notation of a dot (.) above a variable indicates the use of a derivative of the variable where the order corresponds to the number of dots.

The PID-type of controller has been employed in engineering and other applications since the early 1920s. It is an error-based controller that does not require an explicit mathematical model of the plant 110.

Transfer Function Based Controller

The transfer function based (TFB) controller is given in the form:

\[ U(s) = G_s(s)E(s) \]  

where U(s) and E(s) are Laplace Transforms of u and e defined above, and n(s) and d(s) are polynomials in s. The TFB controller can be designed using methods in control theory based on the transfer function model of the plant, Gp(s). A PID controller can be considered a special case of a TFB controller because it has an equivalent transfer function of s.

\[ G_s(s) = k_p + k_i \frac{1}{s} + k_d s \]  

State Feedback (SF) Controller

The State Feedback (SF) controller can be defined by

\[ u = -Kx \]  

where u is the control input, r is the setpoint for the output to follow and x is the state vector associated with the system, based on the state space model of the plant 110:

\[ \dot{x}(t) = Ax(t) + Bu(t) \]  

\[ y(t) = Cx(t) + Du(t) \]  

where x(t) is the state vector, y(t) is the output vector, u(t) is the input or control vector, A is the state matrix, B is the input matrix, C is the output matrix and D is the feed-through matrix. When the state x is not accessible, a state observer (SO):

\[ \dot{x} = A\hat{x} + B\hat{u} + L(y - y) \]
is usable to estimate the state $x$, where the (') symbol is used to denote an estimate or observed value of a given variable and $L$ denotes the observer coefficient matrix.

State Observers

Observers extract real-time information of a plant’s 110 internal state from its input-output data. The observer usually presumes precise model information of the plant 110, since performance is largely based on its mathematical accuracy. Most prior art closed loop controllers require both types of information. Such presumptions, however, often make the method impractical in engineering applications, since the challenge for industry remains in constructing these models as part of the design process. Another level of complexity is added when gain scheduling and adaptive techniques are used to deal with nonlinearity and time variance, respectively.

Disturbance Observers and Disturbance Rejection

Recently, disturbance rejection techniques have been used to account for uncertainties in the real world and successfully control complex nonlinear systems. The premise is to solve the problem of model accuracy in reverse by modeling a system with an equivalent input disturbance that represents any difference between the actual plant $P$ and a derived selected model $P_0$, of the plant, including external disturbances $w$. An observer is then designed to estimate the disturbance in real time and provide feedback to cancel it. As a result, the augmented system acts like the model $P_0$ at low frequencies, making the system behave like $P_0$ and allowing a controller to be designed for $P_0$.


Extended State Observer (ESO)

In this regard, the extended state observer (ESO) is quite different. Originally proposed by Han, J. (1999). “Nonlinear Design Methods for Control Systems.” Proc. 14th IFAC World Congress, in the form of a nonlinear UJO and later simplified to a linear version with one tuning parameter by Guo, Z. (2003). “Scaling and Parameterization Based Controller Tuning.” Proc. of ACC, 4989-4996, the ESO combines the state and disturbance estimation power of a UJO with the simplicity of a DOB. One finds a decisive shift in the underlying design concept as well. The traditional observer is based on a linear time-invariant model that often describes a nonlinear time-varying system. Although the DOB and UJO reject input disturbances for such nominal plants, they leave the question of dynamic uncertainty mostly unanswered. The ESO, on the other hand, addresses both issues in one simple framework by formulating the simplest possible design model $P_0$ for a large class of uncertain systems.


Active Disturbance Rejection Control (ADRC)


What is needed is a control framework for application to systems throughout industry that are complex and largely unknown to the personnel often responsible for controlling them. In the absence of required expertise, less tuning parameters are needed than current approaches, such as multi-loop PID, while maintaining or even improving performance and robustness.

INDUSTRIAL APPLICATION

A selection of industrial applications is presented herein to provide a background on some selected applications for the present extended active disturbance rejection controller. Tracking Control

Tracking Control refers to the output of a controlled system meeting design requirements when a specified reference trajectory is applied. Oftentimes, it refers to how closely the output y compares to the reference input r at any given point in time. This measurement is known as the error e=r-y. Control problems are categorized in two major groups: point-to-point control and tracking control. Point-to-point applications usually call for a smooth step response with small overshoot and zero steady state error, such as when controlling linear motion from one position to the next and then stopping. Since the importance is placed on accurate following a changing reference trajectory between points, phase lags are avoided as much as possible because they may lead to significant errors in the transient response, which lasts for the duration of the process in many applications. A common method of reducing phase lag is to increase the bandwidth, at the cost of increased control effort (energy) and decreased stability margin.

Various methods have been used to remove phase lag from conventional control systems. All of them essentially modify the control law to create a desired closed loop transfer function equal to one. As a result, the output tracks the reference input without much phase lag and the effective bandwidth of the overall system is improved. The most common method is model inversion where the inverse of the desired closed loop transfer function is added as a prefilter. Another method proposed a Zero Phase Error Tracking Controller (ZPETC) that cancels poles and stable zeros of the closed loop system and compensates for phase error introduced by un-cancelable zeros. Although it is referred to as a tracking controller, it is really a prefilter that reduces to the inverse of the desired closed loop transfer function when unstable zeros are not present. Other methods consist of a single tracking control law with feed forward terms in place of the conventional feedback controller and prefilter, but they are application specific. However, all of these and other previous methods apply to systems where the model is known.

Model inaccuracy can also create tracking problems. The performance of model-based controllers is largely dependent on the accuracy of the model. When linear time-invariant (LTI) models are used to characterize nonlinear time-varying (NTV) systems, the information becomes inaccurate over time. As a result, gain scheduling and adaptive techniques are developed to deal with nonlinearity and time variation, respectively. However, the complexity added to the design process leads to an impractical solution for industry because of the time and level of expertise involved in constructing accurate mathematical models and designing, tuning, and maintaining each control system.

There have been a number of high performance tracking algorithms that consist of three primary components: disturbance rejection, feedback control, and phase lag compensation implemented as a prefilter. First, disturbance rejection techniques are applied to eliminate model inaccuracy with an inner feedback loop. Next, a stabilizing controller is constructed based on a nominal model and implemented in an outer feedback loop. Finally, the inverse of the desired closed loop transfer function is added as a prefilter to eliminate phase lag. Many studies have concentrated on unifying the disturbance rejection and control part, but not on combining the control and phase lag compensation part, such as the RIC framework. Internal model control (IMC) cancels an equivalent output disturbance. B. Francis and W. Wonham, "The Internal Model Principal of Control Theory," Automatica, vol. 12, 1976, pp. 457-465. E. Schrijver and J. Van Dijk, "Disturbance Observers for Rigid Mechanical Systems: Equivalence, Stability, and Design," Journal of Dynamic Systems, Measurement, and Control, vol. 124, December 2002, pp. 539-548 uses a basic tracking controller with a DOB to control a multi-variable robot. The ZPETC has been widely used in combination with the DOB framework and model based controllers.

Thus, having reviewed prior art in tracking control, the application now describes example systems and methods of tracking control employing predictive ADRC.

Web Processing Applications

Web tension regulation is a challenging industrial control problem. Many types of material, such as paper, plastic film, cloth fabrics, and even strip steel are manufactured or pro-
cessed in a web form. The quality of the end product is often greatly affected by the web tension, making it a crucial variable for feedback control design, together with the velocities at the various stages in the manufacturing process. The ever-increasing demands on the quality and efficiency in industry motivate researchers and engineers alike to explore better methods for tension and velocity control. However, the highly nonlinear nature of the web handling process and changes in operating conditions (temperature, humidity, machine wear, and variations in raw materials) make the control problem challenging.

Accumulators in web processing lines are important elements in web handling machines as they are primarily responsible for continuous operation of web processing lines. For this reason, the study and control of accumulator dynamics is an important concern that involves a particular class of problems. The characteristics of an accumulator and its operation as well as the dynamic behavior and control of the accumulator carriage, web spans, and tension are known in the art.

Both open-loop and closed-loop methods are commonly used in web processing industries for tension control purposes. In the open-loop control case, the tension in a web span is controlled indirectly by regulating the velocities of the rollers at either end of the web span. An inherent drawback of this method is its dependency on an accurate mathematical model between the velocities and tension, which is highly nonlinear and highly sensitive to velocity variations. Nevertheless, simplicity of the controller outweighs this drawback in many applications. Closing the tension loop with tension feedback is an obvious solution to improve accuracy and to reduce sensitivity to modeling errors. It requires tension measurement, for example, through a load cell, but is typically justified by the resulting improvements in tension regulation.

Most control systems will unavoidably encounter disturbances, both internal and external, and such disturbances have been the obstacles to the development of high-performance controllers. This is particularly true for tension control applications and, therefore, a good tension regulation scheme must be able to deal with unknown disturbances. In particular, tension dynamics are highly nonlinear and sensitive to velocity variations. Further, process control variables are highly dependent on the operating conditions and web material characteristics. Thus, what are needed are systems and methods for control that are not only overly dependent on the accuracy of the plant model, but also suitable for the rejection of significant internal and external disturbances.

Jet Engine Control Applications


Conventionally, there have been a limited number of control techniques for full flight operation (Garg, S. (1997). “A Simplified Scheme for Scheduling Multivariable Controllers,” IEEE Control Systems; and Polley, J. A., S. Adibhatla and P. J. Hoffman (1988). “Multivariable Turbofan Engine Control for Full Conference on Decision and Control Flight Operation.” Gas Turbine and Expo). However, there has been no development of tuning a controller for satisfactory performance when applied to an engine. Generally, at any given operating point, models can become inaccurate from one engine to another. This inaccuracy increases with model complexity, and subsequently design and tuning complexity. As a result, very few of these similar aircraft design studies have led to implementation on an operational vehicle.


**SUMMARY OF INVENTION**

The present system, method, process, and device for controlling systems is applicable for a variety of different application and has utility in a number of different industrial fields. The following brief summary section is provided to facilitate a basic understanding of the nature and capabilities of the present system and method. This summary section is not an extensive nor comprehensive overview and is not intended to identify key or critical elements of the present systems or methods or to delineate the scope of these items. Rather this brief summary is intended to provide a conceptual introduction in simplified form as an introduction for the more detailed description presented later in this document.

The present application describes selected embodiments of the ADRC controller that comprise one or more computer components, that extend, build upon and enhance the use of ADRC controllers and provide enhanced performance and utility. Specifically, in one aspect the ADRC controller utilizes a predictive computer component or module that in one embodiment predicts future values of the plant output and in a second embodiment predicts future estimates of the system state and generalized disturbance. The generalized disturbance includes dynamics of the plant itself such that the plant is effectively reduced to a cascaded integral plant. In the various embodiments a variety of estimates are available and introduced.

In still another aspect, the ADRC controller further includes a model of the plant dynamics. The model is used in one embodiment to improve the state estimator or predictor. In a second embodiment the model is used to provide an enhanced control law that provides improved performance. In both aspects of the embodiments discussed in this paragraph,
the generalized disturbance captures model errors and discrepancies and allows the system to eliminate errors caused by model errors, discrepancies, and assumptions. In the multiple embodiments, a variety of models are available for use by one of ordinary skill in the art to achieve the desired performance relative to the type of plant being controlled.

In yet another aspect, the ADRC controller further includes a non-linear, or discrete time optimal control law. The non-linear control law of the present embodiment utilizes non-linear control laws to improve the overall performance of an ADRC controller as compared to an ADRC controller with a linear proportional-derivative based control law.

In still another aspect, additional knowledge of derivative plant output gathered from high quality sensor information or direct measurement sensors (e.g. velocity and acceleration sensors) is used to reduce the order of the system state and disturbance estimator.

BRIEF DESCRIPTION OF THE DRAWINGS

The accompanying figures depict multiple embodiments of a device and method for the separation of constituents from a flow. A brief description of each figure is provided below. Elements with the same reference numbers in each figure indicate identical or functionally similar elements. Additionally, the left-most digit(s) of a reference number identifies the drawings in which the reference number first appears.

FIG. 1 is a block diagram of a Prior Art feedback controller. FIG. 2 is a block diagram of an ADRC configuration for a 2nd order plant. FIG. 3 is a block diagram of a predictive ADRC configuration. FIG. 4 is a block diagram of a 2 degree of freedom transfer function based (TFB) control structure. FIG. 5 is a block diagram of an implementation of a PADRC configuration based on output prediction.

The present invention is described with reference to block diagrams and operational flow charts. It is to be understood that the functions/acts noted in the blocks may occur out of the order noted in the operational illustrations. For example, two blocks shown in succession may in fact be executed substantially concurrently or the blocks may sometimes be executed in the reverse order, depending upon the functionality/acts involved, including for example executing as asynchronous threads on a processor. Although some of the diagrams include arrows on communication paths to show a primary direction of communication, it is to be understood that communication may occur in the opposite direction to the depicted arrows.

DETAILED DESCRIPTION

Multiple embodiments of a system, device, and method for the control of systems are presented herein. Those of ordinary skill in the art can readily use this disclosure to create alternative embodiments using the teaching contained herein. Lexicon

The following terms used herein have the meanings as follows.

As used herein the term “computer component” refers to a computer and elements of a computer, such as hardware, firmware, software, a combination thereof, or software in execution. For example, a computer component can include by way of example, a process running on a processor, a processor, an object, an executable, an execution thread, a program, and a computer itself. One or more computer components can in various embodiments reside on a server and the server can be comprised of multiple computer components. One or more computer components are in some cases referred to as computer systems whereby one or more computer components operate together to achieve some functionality. One or more computer components can reside within a process and/or thread of execution and a computer component can be localized on one computer and/or distributed between two or more computers.

“Software”, as used herein, includes but is not limited to, one or more computer readable and/or executable instructions that cause a computer or other electronic device to perform functions, actions and/or behave in a desired manner. The instructions may be embodied in various forms such as routines, algorithms, modules, methods, threads, and/or programs. Software may also be implemented in a variety of executable and/or loadable forms including, but not limited to, a stand-alone program, a function call (local and/or remote), a server, an applet, instructions stored in a memory, part of an operating system or browser, and the like. It is to be appreciated that the computer readable and/or executable instructions can be located in one computer component and/or distributed between two or more communicating, co-operating, and/or parallel processing computer components and thus can be loaded and/or executed in serial, parallel, massively parallel and other manners. It will be appreciated by one of ordinary skill in the art that the form of software may be dependent on, for example, requirements of a desired application, the environment in which it runs, and/or the desires of a designer/programmer or the like.

“Computer communications”, as used herein, refers to a communication between two or more computers and can be, for example, a network transfer, a file transfer, an applet transfer, an email, a hypertext transfer protocol (HTTP) message, a datagram, an object transfer, a binary large object (BLOB) transfer, and so on. A computer communication can occur across, for example, a wireless system (e.g., IEEE 802.11), an Ethernet system (e.g., IEEE 802.3), a token ring system (e.g., IEEE 802.5), a local area network (LAN), a wide area network (WAN), a point-to-point system, a circuit switching system, a packet switching system, and so on.

An “operable connection” is one in which signals and/or actual communication flow and/or logical communication flow may be sent and/or received. Usually, an operable connection includes a physical interface, an electrical interface, and/or a data interface, but it is to be noted that an operable connection may consist of differing combinations of these or other types of connections sufficient to allow operable control.

As used herein, the term “signal” may take the form of a continuous waveform and/or discrete value(s), such as digital value(s) in a memory or register, present in electrical, optical or other form.

The term “controller” as used herein indicates a method, process, or computer component adapted to control a plant (i.e. the system to be controlled) to achieve certain desired goals and objectives.

As used herein the following symbols are used to describe specific characteristics of the variable. A variable with an """" overtop the variable label, unless otherwise indicated to the contrary, indicates that the variable is an estimate of the actual variable of the same label. A variable with an """" overtop the variable label, unless otherwise indicated, indicates that the variable is an nth derivative of that variable, where n equals the number of dots.

To the extent that the term “includes” is employed in the detailed description or the claims, it is intended to be inclusive.
in a manner similar to the term “comprising” as that term is interpreted when employed as a transitional word in a claim.

To the extent that the term “or” is employed in the claims (e.g., A or B) it is intended to mean “A or B or both”. When the author intends to indicate “only A or B but not both”, then the author will employ the term “A or B but not both”. Thus, use of the term “or” in the claims is the inclusive, and not the exclusive, use.

Active Disturbance Rejection Control (ADRC)

Active Disturbance Rejection Control (“ADRC”) is a method or process and, when embedded within a computer system, a device (i.e., a controller that processes information and signals using the ADRC algorithms), that effectively rejects both unknown internal dynamics of a system of an arbitrary order to be controlled, or plant 110, and external disturbances in real time without requiring detailed knowledge of the plant 110 dynamics, which is required by most existing control design methods.

The only information needed to configure an ADRC controller is knowledge of the relative order of the plant 110 and the high frequency gain of the plant 110. In comparison, traditional model-based control methods require accurate mathematical models to conform with prevailing model-based design methods. In some embodiments, the ADRC controller leverages knowledge gained from model-based controller designs to improve control performance. The ADRC approach is unique in that it first forces an otherwise nonlinear, time-varying and uncertain plant 110 to behave as a simple linear cascade integral plant that can be easily controlled. To this end, the unknown, nonlinear, and time-varying internal dynamics is combined with the external disturbances to form what is denoted as the generalized disturbance, which is then estimated and rejected, thus greatly simplifying the control design.

The embodiments of the ADRC controller design are founded on the basic theory that uncertainties in dynamics and unknown external disturbances in the operating environment are a key challenge to be dealt with in control engineering practice. Traditional control systems are incapable of providing a solution to controlling and mitigating the impact of these uncertainties. The ADRC controller in contrast is specifically adapted to mitigate and eliminate the impact of these uncertainties, thus providing a dynamically simplified system for control.

One previous major characteristic of the ADRC controller is the parameterization of the controller as a function of control loop bandwidth. This parameterization makes all of the ADRC controller parameters functions of the control loop bandwidth; see, U.S. patent application Ser. No. 10/351,664, titled, “Scaling and Parameterizing A Controller” to Zhiqiang Gao, which is hereby incorporated by reference in its entirety. An embodiment of a parameterized ADRC controller applied to a second-order plant 202 is illustrated in FIG. 2. The system to be controlled is a second-order plant 202 and represents an arbitrary plant 110 with predominantly second-order dynamics. The second-order plant 202 has an output (y) in response to an input that comprises a control signal (u) with superimposed external disturbances (d).

Consider a general uncertain, nonlinear and time-varying second-order plant 202 of the form:

\[ f = f(y, \dot{y}, d, t) + bu \]  

where \( y \) is system output, \( u \) is the control signal, \( b \) is a constant, \( d \) is external input disturbance, \( f(y, \dot{y}, d, t) \) is treated as the generalized disturbance. When this generalized disturbance is cancelled, the system is reduced to a simple double-integral plant with a scaling factor \( b \), which can be easily controlled. Here we assume that the approximate value of \( b \), denoted as \( \hat{b} \), is given or estimated. As shown in FIG. 2, the scaling factor \( b \) is used to scale 208, the control output (u).

5 10 15

In other embodiments, the scaling factor is implicit within the linear PD control law 206.

To capture the information of the generalized disturbance \( f(y, \dot{y}, d, t) \) and cancel it from the system dynamics leads us to the Extended State Observer (ESO) 204, which is now presented. Let \( x_1 = y, x_2 = \dot{y}, \) and \( x_3 = f \), the description of the above plant 202 is rewritten in state space form with \( f(y, \dot{y}, d, t), \) or simply \( f \), treated as an additional state:

\[
\begin{align*}
\dot{x} &= A x + B u + E f \\
y &= C x \\
\end{align*}
\]

where

\[
\begin{align*}
A &= \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix} \\
B &= \begin{bmatrix} 0 \\ 0 \\ b \end{bmatrix} \\
E &= \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \\
C &= \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \\
\end{align*}
\]

The corresponding state observer is:

\[
\begin{align*}
\dot{\hat{x}} &= \hat{A} \hat{x} + \hat{B} u + \hat{L} (y_m - C \hat{x}) \\
\hat{f} &= \hat{V} \hat{z} \\
\end{align*}
\]

where \( y_m \) is the measured system output, \( \hat{B} = [0 \ 0 \ 0]^T \), \( \hat{V} = [0 \ 0 \ 1] \), and \( \hat{L} = [I_1 \ I_2 \ I_3] \) is the observer gain vector. With the parameterization technique as described more fully in U.S. patent application Ser. No. 10/351,664, L is obtained by solving the equation \( \lambda(s) = \{s^2 - 25\} \); that is, all the eigenvalues of the observer are placed at \(-\omega_o, with \)

\[
I_1 = \omega_o^2, I_2 = 3\omega_o^2, I_3 = \omega_o^2 \]

With a well-tuned extended state observer 204, \( z_1, z_2, \) and \( z_3 \) will closely track \( y, \dot{y} \), and \( f \), respectively, i.e., \( z_1 \approx y, z_2 \approx \dot{y}, \) and \( z_3 \approx f \).

Using the information provided by the extended state observer 204, the generalized disturbance \( f \) is rejected by the control law

\[
u = (y_m - z_2) / \hat{b} \]

which reduces the plant 110 to a unity gain, double integral plant

\[
\dot{y} = u \]

that can be easily controlled with linear proportional-derivative (PD) with feedforward

\[
u_u = g_{pd}(e, \dot{e}, \omega_o) e \]

where \( g_{pd}(e, \dot{e}, \omega_o) \) is the parameterized linear PD controller with the loop bandwidth of \( \omega_o \).
with the error term defined as
\[ e = r - z \]
Equation (16)

where the desired trajectory \( r \) is provided by the user.

Note that the linear PD control law 206 in Equation (14) corresponds to the common PD gains of \( k_p, k_d \) and \( b, w \).

Equation (17)

Using these details, the ADRC controller is generalized in its various embodiments to control an arbitrary \( n^{th} \)-order system \( y^{(n)}(t) \), in which the external disturbances and unknown internal dynamics are rejected. In a similar manner, the plant 110 is reduced to a cascaded integral plant \( y^{(n)}(t) \), which is then controlled by a generalized proportional-derivative (PD) control law, also described as a linear PD control law 206 similar to Equation (14).

Use of Enhanced Sensor Information to Improve Control Performance

In other embodiments of the ADRC controller, the linear PD control law 206 is updated to enhance performance by using direct measurement of the system output states \( y_m \) instead of the estimated states \( z_i \) and \( z_z \). Specifically, when the measured output \( y_m(t) \) is sufficiently noise free and \( y_m(0) \) is obtainable, either via direct measurement or from \( y_m(0) \), the control law in Equation (14) can be implemented with

\[ e = r - y_m \]
Equation (18)

That is, the output measurement and its derivative are used in the control law, instead of their estimates, \( z_i \) and \( z_z \). Similarly for the \( n^{th} \)-order plant 110, the output measurement \( y_m \) and its derivatives from the \( (n-1)^{th} \) to the \( (n-1)^{th} \)-order available, can also be used directly in the PD control law, also referred to as the linear PD control law 206, instead of their estimates from the Extended State Observer (ESO) 204. Using the direct measurement of the output measurement state \( y_m \) and its derivatives reduces the phase lag and improves control system performance, if the output measurement and its derivatives are not corrupted by noise.

Multiple Extended States

The Extended State Observer 204 in alternative embodiments includes more than one extended states, such as embodiments where the 2nd order plant 202 is replaced with a higher order plant 110. In one embodiment, when augmented with two extended states, \( x_3 = f \) and \( x_4 = \dot{f} \), the extended plant state space model can be written as follows.

\[
\begin{align*}
\dot{x} &= A_{x}x + B_{x}u + E_{x}
\end{align*}
\]
Equation (19)

where

\[
A_{x} = \begin{bmatrix}
0 & 1 & 0 & 0 \\
0 & 0 & 1 & 0 \\
0 & 0 & 0 & 1 \\
0 & 0 & 0 & 0
\end{bmatrix}
\]

Current Discrete Extended State Observer (CDESO) is in current estimator form, as shown in the following equation.

\[
\begin{align*}
\dot{z}[k+1] &= \Phi z[k] + \Gamma u[k] + L_{y_m}[y_m[k] - z_i[k]] \\
z[k] &= z[k]
\end{align*}
\]
Equation (21)

Current Discrete Extended State Observer (CDESO) is in current estimator form, as shown in the following equation.

\[
\begin{align*}
\dot{z}[k+1] &= \Phi z[k] + \Gamma u[k] + L_{y_m}[y_m[k] - z_i[k]] \\
z[k] &= z[k]
\end{align*}
\]
Equation (22)

The gain matrices of different ADRC discrete implementations are shown in the following Table 1.
This discrete implementation of ADRC is also generalizable to n-th order plants 110.

Predictive Active Disturbance Rejection Controller

The thrust of the many of the embodiments presented here is the addition of the predictive functions to extend the capabilities of the original components of the ADRC. This extends the ADRC controller and generates embodiments of predictive versions of the ADRC, such as the predictive ADRC 300, depicted in block diagram form in FIG. 3. In this embodiment, the linear PD control law 206 in the predictive ADRC 300 is now a predictive control law 306. Similarly, the observer, or more specifically the extended state observer 304 from the traditional ADRC 200 is now embodied as a predictive state and disturbance observer 304. In one embodiment the predictive state and disturbance observer 304 is a predictive extended state observer (PESO) where the prediction occurs within the bounds of the ESO 204 to where the estimated states z(t) are predicted to form the predictive state and disturbance observer 304, and outputs the prediction of the estimated states z(t+τ). In a second embodiment, a state predictor method is used to extend the design of the ESO 204 to estimate the states z(t) and form the predictive state and disturbance observer 304.

In other embodiments, other estimators and predictors, either in an integrated form or in combination are used to generate similar information as that generated by the predictive state and disturbance observer 304 and used by the embodiments predictive ADRC 300. One of these other embodiments adopts a disturbance observer (DOB) estimator structure to estimate the disturbance and a state observer and a predictor element that together form the predictive state and disturbance observer 304. In a second embodiment, the state observer, or (ii) direct measurement of derivatives are possible (e.g., numerical differentiation to estimate derivatives of the output signal; or (ii) direct measurement of derivatives are possible (e.g., velocity and acceleration sensors), are available.

In another embodiment, a nonlinear proportional-derivative (NPD) control law is used to further optimize the performance of the present ADRC system. The RESO aspect of the ADRC controller provides alternative embodiments of ESO 304 and RESO; a general form of RESO with multiple extended states for an n-th order plant 110, and incorporation of the known plant 110 dynamics into ESO 204 and control law.

Furthermore, to facilitate practical implementations of the new techniques, we have provided at selected locations herein, the transfer function equivalent predictive ADRC 400, with H(s) 402 and G(s) 404, in a two-degree-of-freedom (2dof) structure shown in FIG. 4. Similarly, for a transfer function equivalent predictive ADRC 400, the plant dynamics are represented as G(s) 402. For various forms of the Predictive ADRC 400 described herein, these transfer function expressions, together with the NPD and RESO, are novel and significant even when the amount of prediction is set to zero.

All of these new developments help to make the predictive ADRC 300 possible and to provide the means of implementing the present embodiments in a diverse range of industrial machinery, and they are illustrated in details in the following sections.

Basic Design Principles for Predictive ADRC

In the real world, the sensor feedback, the generation of the control signal and its arrival at the actuator, and the estimation of the states and disturbances all contain time delays and phase lags, which can endanger system stability. To accommodate such delays and lags, we propose the following Predictive ADRC control law:

\[ u(t) = g(e(t), e(t), w(t), f(t), f(t+τ))/b \]  \hspace{1cm} (Equation 23)

where \( g(e(t), e(t), w(t), f(t), f(t+τ))/b \) is the parameterized linear PD controller 206, or nonlinear proportional derivative (NPD) controller, Equation (68), \( L_p, L_q, L_r, \) and \( L_p \) are positive real numbers representing the prediction horizons, a special case of which is \( L_p = L_q = L_r = T/2 \) and the corresponding control law is

\[ u(t) = g(e(t+τ), e(t+τ), w(t+τ), f(t+τ))/b \]  \hspace{1cm} (Equation 24)

Also, if a part of the plant 110 dynamics, \( f_p \), is given, it can be incorporated into ADRC, as shown later in this disclosure. In the context of Predictive ADRC 300, the knowledge of \( f_p \) can be used to help predict its future value, \( f_p(t+τ) \). The corresponding control law for Equation (23) and Equation (24) are

\[ u(t) = g(e(t+τ), e(t+τ), w(t+τ), f(t+τ))/b \]  \hspace{1cm} (Equation 25)

and

\[ u(t) = g(e(t+τ), e(t+τ), w(t+τ), f(t+τ))/b \]  \hspace{1cm} (Equation 26)

respectively.

Described below are various methods to obtain the prediction of \( e(t) \), \( ̇e(t) \), and \( ̄e(t) \), respectively, so that these predictive ADRC 300 control laws can be implemented.

Prediction of \( e \) and \( ̇e \)

Two embodiments of methods for predicting the values of error \( e(t) \) and its derivative are presented herein.
Method 1:
Using Taylor Series, the prediction of \( e(t) \), \( \dot{e}(t) \) can be obtained as
\[
e(t + \Delta t) = e(t) + \dot{e}(t)\Delta t + \frac{\dot{\ddot{e}}(t)}{2} (\Delta t)^2
\]
or
\[
e(t + \Delta t) = e(t) + \dot{e}(t)\Delta t + \frac{\ddot{e}(t)}{2} (\Delta t)^2
\]
and
\[
\ddot{e}(t + \Delta t) = \dot{e}(t) + \frac{\ddot{e}(t)}{2} (\Delta t)
\]
where \( \dot{e}(t) \) can be obtained directly from \( \dot{r} - \dot{y} \), or from \( \dot{r} - \dot{z}_2 \) using regular ESO, or \( \dot{r} - \dot{z}_1 \) using RESO. Here \( \dot{e}(t) \) can be obtained as \( \dot{e}(t) = \dot{r}(t) - \dot{y}(t) - (z_1(t) + bu(t)) \) with ESO or \( \dot{e}(t) = \dot{r}(t) - \dot{y}(t) - (z_2(t) + bu(t)) \) with RESO.

Method 2:
Since \( e(t + \tau) = r(t + \tau) - z_1(t + \tau) \) and \( \dot{e}(t + \tau) = \dot{r}(t + \tau) - z_2(t + \tau) \), \( e(t + \tau) \) and \( \dot{e}(t + \tau) \) can be obtained directly using the future reference signals \( r(t + \tau) \) and \( \dot{r}(t + \tau) \), which are generally known, and the predicted output of the ESO \( z_1(t + \tau) \) and \( z_2(t + \tau) \), which need to be obtained approximately.

Prediction of \( \hat{f} \) in ESO and RESO
The primary method of predicting \( \hat{f} \) is the 1st or 2nd-order Taylor Series approximation, i.e.
\[
\hat{f}(t + \Delta t) = f(t) + \hat{f}(t)\Delta t + \frac{\ddot{f}(t)}{2} (\Delta t)^2
\]
which requires the 1st and/or 2nd order derivatives of \( \hat{f} \). To this end, a new form and a new way of implementation of extended state observer (ESO) 204 are introduced first, followed by the discussion on how to use them to obtain the predicted estimate of \( \hat{f} \) in Equation (27) and Equation (28) Reduced-Order ESO.

In the observer-based control methods, the phase lag introduced by the observer decreases the phase margin of the control loop and is therefore undesired. To reduce the phase lag in the ESO 204, an embodiment of a reduced-order ESO (RESO) described here.

Consider the plant 202 in Equation (9), if the derivative of the measured output \( y_m \) is given, a RESO can be constructed with \( z_1 \) and \( z_2 \) estimating \( x_3 = \hat{y} \) and \( x_5 = \hat{f} \) respectively. The correspondence between the augmented plant state space model and the RESO is shown as follows.

\[
\begin{align*}
\dot{x} &= A_x x + B_x u + L_x (y_m - C_x z) \\
\dot{y} &= C_x z + D_x u
\end{align*}
\]

Equation (30)

\[
\begin{align*}
\dot{\hat{y}} &= \hat{y} \\
\dot{\hat{f}} &= \hat{f}
\end{align*}
\]

Equation (31)

Note here that the parameterized linear PD control law 206 can be replaced by the parameterized NPD control law 506, which will be introduced later.

The order of ESO 204 is reduced further if \( \hat{y} \) is also given, as the double derivative of \( y_m \). According to U.S. patent application Ser. No. 10/351,664, the total disturbance is obtained as \( X_3 = f = y - bu \) and it is susceptible to measurement noise. In an alternative embodiment, a first order RESO can be constructed with the only state \( z \) tracking \( x_3 = \hat{f} \). The correspondence between the augmented plant state space model and the new RESO is shown as follows.

\[
\begin{align*}
\dot{x} &= A_x x + B_x u + L_x (y_m - C_x z) \\
\dot{y} &= C_x z + D_x u
\end{align*}
\]

Equation (29)

\[
\begin{align*}
\dot{\hat{y}} &= \hat{y} \\
\dot{\hat{f}} &= \hat{f}
\end{align*}
\]

Equation (32)

Note here that the parameterized linear PD control law 206 can be replaced by the parameterized NPD control law 506, which will be introduced later.
where \( A_0 = 0 \), \( B_0 = 0 \), \( L = 1 - \omega_n \), \( C_0 = 1 \), \( V = 1 \). The corresponding linear PD control law \( 206 \) with this RESO is

\[
\nu = (y_m - \dot{y}_m) + \beta \omega_n \dot{y}_m - \beta \gamma_n \dot{y}_m \tag{36}
\]

where \( e = y_m - \dot{y}_m \) and \( \dot{e} = \ddot{y}_m - \dddot{y}_m \). Note here again that the parameterized linear PD control law \( 206 \) can be replaced by the parameterized NPD control law \( 506 \) to be introduced.

A New ESO Implementation

Consider the same second-order plant \( 202 \) as described in Equation (9). The ESO \( 204 \) described in U.S. patent application Ser. No. 10/351,664 can be divided into two parts. The first part is a Luenberger observer for the double integral plant in Equation (13), and the second part shows that the estimation of the general disturbance can be obtained by integrating the observer error with a gain of \( \lambda_1 \). This implementation is equivalent to the ESO \( 204 \).

When implemented in ZOH current estimator form, the Luenberger observer is

\[
\begin{cases}
  z_1[k+1] = \Phi_0 z_1[k] + \Gamma_0 z_2[k] + L_0 (y_m[k] - z_1[k]) \\
  z_2[k+1] = \Phi_1 z_1[k] + \Gamma_1 z_2[k] + L_1 (y_m[k] - z_1[k])
\end{cases}
\tag{38}
\]

where

\[
\Phi = \begin{bmatrix} 1 & T \\ 0 & 1 \end{bmatrix},
\Gamma = \begin{bmatrix} T^2/2 \\ \beta \gamma_T \end{bmatrix},
L_0 = \begin{bmatrix} 1 - \beta_2 \\ (1 + \beta_2 - 2\beta_1)T \end{bmatrix},
L_1 = \begin{bmatrix} 2 - 2\beta_1 \\ (1 + \beta_2 - 2\beta_1)T \end{bmatrix}
\]

and

\[
\beta_1 = e^{2\gamma_T},
\beta_2 = e^{\gamma_T}. \tag{39}
\]

Similarly, the RESO in Equation (30) can be divided into two parts:

\[
\begin{cases}
  z_2 = 0
\end{cases}
\tag{40}
\]

where the first part is a first order Luenberger observer, and the second part is an integrator. When implemented in ZOH current estimator form, this Luenberger observer is

\[
\begin{cases}
  y_m[k] = \Phi_0 z_1[k] + \Gamma_0 z_2[k] + L_0 (y_m[k] - z_1[k]) \\
  z_1[k] = z_1[k] + L_1 (y_m[k] - z_1[k])
\end{cases}
\tag{41}
\]

where \( \Phi = \begin{bmatrix} 1 & T \\ 0 & 1 \end{bmatrix}, \Gamma = \begin{bmatrix} T^2/2 \\ \beta \gamma_T \end{bmatrix}, \lambda = \begin{bmatrix} \lambda_1 \\ \lambda_2 \end{bmatrix} \) and \( \lambda_1, \lambda_2, \lambda_3 \) are observer gains defined in Equation (11) and \( k_p \) and \( k_d \) are the proportional-derivative (PD) gains defined in Equation (17).

When the 2nd-order Taylor series approximation of \( \dot{f} \) is used in control law \( 404 \), the equivalent 2 dof transfer functions are

\[
C(s) = \frac{1}{\beta} \left( k_1 s^2 + (k_p + k_d) s + \frac{k_d}{2} \right) + \frac{1}{\beta} \left( k_1 s^2 + (k_p + k_d) s + \frac{k_d}{2} \right) \tag{42}
\]

\[
H(s) = \frac{(s^3 + \frac{1}{2} s^2 + \frac{1}{2}) s + \frac{1}{4} s + \frac{1}{4}}{\beta} \tag{43}
\]

where \( k_1, k_2, k_3 \) are observer gains defined in Equation (11) and \( k_p \) and \( k_d \) are the proportional-derivative (PD) gains defined in Equation (17).

When the 2nd-order Taylor series approximation of \( \dot{f} \) is used in control law \( 404 \), the equivalent 2 dof transfer functions are

\[
C(s) = 1 - \frac{1}{\beta} \left( k_1 s^2 + (k_p + k_d) s + \frac{k_d}{2} \right) \tag{44}
\]

\[
H(s) = \frac{(s^3 + \frac{1}{2} s^2 + \frac{1}{2}) s + \frac{1}{4} s + \frac{1}{4}}{\beta} \tag{45}
\]

where \( k_1, k_2, k_3 \) are observer gains defined in Equation (11) and \( k_p \) and \( k_d \) are the proportional-derivative (PD) gains defined in Equation (17).

When the 2nd-order Taylor series approximation of \( \dot{f} \) is used in control law \( 404 \), the equivalent 2 dof transfer functions are

\[
C(s) = 1 - \frac{1}{\beta} \left( k_1 s^2 + (k_p + k_d) s + \frac{k_d}{2} \right) \tag{46}
\]

\[
H(s) = \frac{(s^3 + \frac{1}{2} s^2 + \frac{1}{2}) s + \frac{1}{4} s + \frac{1}{4}}{\beta} \tag{47}
\]

where \( k_1, k_2, k_3 \) are observer gains defined in Equation (11) and \( k_p \) and \( k_d \) are the proportional-derivative (PD) gains defined in Equation (17).
The prediction of \( y(t) \) \( t \) seconds ahead is:

\[
G_c(s) = \frac{1}{h} \frac{I_3 I_3 s^3 + (I_3 + I_2) s^2 + (I_2 + I_1) s + I_1}{s^3 + (I_3 + I_2) s^2 + (I_2 + I_1) s + I_1}
\]

Equation (48)

\[
H(s) = \frac{I_f \left( s^2 + (I_f + I_2) s + I_2 \right)}{I_f \left( s^2 + (I_f + I_2) s + I_2 \right)}
\]

Equation (49)

where \( I_1, I_2 \) are observer gains defined in Equation (35).

**State Predictor Method for Predictive Extended State Observer**

Another embodiment for forming a predictive state and disturbance observer \( 304 \) module is to use a state predictor method to extend the performance of a baseline ESO \( 204 \) to predict state estimates \( t \) seconds in the future. The basic state predictor method approach was published by T. Oguchi, H. Nijmeijer, *Prediction of Chaotic Behavior*, IEEE Trans. on Circuits and Systems-I, Vol. 52, No. 11, pp. 2464-2472, 2005, which is hereby incorporated by reference. The state predictor method is combined in the embodiments presented herein with the regular ESO \( 204 \) of the form

\[
z = A z + B u + L(y - \hat{y})
\]

where

\[
A = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix}
\]

\[
B = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}
\]

\[
C = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}
\]

\[
L = \begin{bmatrix} 3 \omega_0 \\ 3 \omega_0 \omega_d \\ \omega_d \end{bmatrix}
\]

Equation (50)

\[
\hat{y} = C \hat{z}
\]

Equation (51)

To form the predictive ESO is in the form of

\[
\hat{z}(t) = A \hat{z}(t) + B u(t) + L(y(t) - \hat{y}(t))
\]

Equation (52)

Then \( \hat{y}(t) = z_1(t - \tau) \) will track \( y(t) \) while \( z_1(t - \tau) = \hat{y}(t+\tau) \) is the prediction \( \tau \) seconds ahead. The idea is that since the observer design ensures that the observer error goes to zero, or becomes very small, if that error is defined as the difference between \( y(t) \) and \( z_1(t - \tau) \) and \( z_1(t - \tau) \) will approach \( y(t) \) and \( z_1(t) \) will approach \( y(t+\tau) \). Therefore \( z_1(t) \) can be used as a prediction of \( y(t+\tau) \) second ahead.

**Building a Predictive State and Disturbance Observer Using an Output Prediction Mechanism**

Perhaps a simpler method to obtain \( z(t+\tau) \) is to leave the observer unchanged but replace its input \( y_m(t) \) with the predicted output \( \hat{y}(t+\tau) \) as shown in FIG. 5. This configuration is used both to obtain the predicted error and its derivatives as well as the predicted disturbance estimation. Also depicted in FIG. 5 is a prediction module computer component or more generally a prediction module \( 508 \). The prediction module \( 508 \) is adapted to accept the measured system output \( (y_m(t)) \) and calculated estimated values of the system output \( \hat{y}(t+\tau) \) from the present time to a time \( (\tau) \) in the future.

A state and disturbance observer \( 504 \), shown in FIG. 5, accepts the predicted and current values of the plant output \( (y_m) \) and the control commands \( (u(t)) \) to estimate the future states \( (z_n(\tau+\tau)) \) where \( (n) \) is the number of estimated states plus an estimate of the generalized disturbance \( (L_n) \). The state and disturbance observer \( 504 \) in one embodiment is an ESO \( 204 \). In a second embodiment, the state and disturbance observer \( 504 \) is a disturbance observer (DOB) structure. In yet another embodiment, the state and disturbance observer \( 504 \) is an unknown input observer (UID). It is apparent that the combination of the state and disturbance observer \( 504 \) in combination with the prediction module \( 508 \) provides a similar output and performance a similar function within the framework of an output predictive ADRC controller \( 500 \) as the predictive state and disturbance observer \( 304 \) of the predictive ADRC controller \( 300 \). In this manner, the various embodiments of the predictive ADRC controller \( 300 \) and output predictive ADRC controller \( 500 \) are correlated.

Also shown in FIG. 5, a control law \( 506 \) is provided that uses the estimated future states of the plant \( 110 \) coupled with the desired trajectory \( (\varphi(t)(\tau)) \) to generate a current control input \( (u(t)) \). The embodiment of the control law \( 506 \) depicted in FIG. 5 incorporates the additive inverse of the estimated future values of the estimated disturbance \( (L_n) \).

Combine the Output Prediction with Regular ESO

To compensate for the delay, \( y(t) \) is replaced with the approximately predicted output \( \hat{y}(t+\tau) \) as the input to the ESO \( 204 \). Again, the approximated prediction, \( \hat{y}(t+\tau) \), is obtained from the Taylor Series approximation. The \( 1^{rd} \)-order Taylor series approximation and the \( 2^{nd} \)-order Taylor series approximation of \( \hat{y}(t+\tau) \) are as shown in the following equations respectively.

\[
\hat{y}(t + \tau) = y_m(t) + \frac{y_m(\tau)}{2} \tau^2
\]

Equation (53)

\[
\hat{y}(t + \tau) = y_m(t) + \frac{y_m(\tau)}{2} \tau^2 + \frac{y_m(\tau)}{6} \tau^3
\]

Equation (54)

Then the predicted system output and the current control signal are used as the inputs to the ESO \( 204 \) as follows.

\[
\begin{cases}
\hat{z}(t + \tau) = A \hat{z}(t + \tau) + B u(t) + L(y(t + \tau) - C \hat{z}(t + \tau)) \\
\hat{y}(t + \tau) = V \hat{z}(t + \tau)
\end{cases}
\]

Equation (55)

which will provide both the predicted states and the disturbance estimation in \( \hat{z}(t+\tau) \). Equivalent 2dof Transfer Functions

When the \( 1^{rd} \)-order Taylor series approximation is used to obtain \( \hat{y}(t+\tau) \), the equivalent 2dof transfer functions are

\[
\begin{align*}
G_c(s) &= \frac{1}{h} \frac{I_3 I_3 s^3 + (I_3 + I_2) s^2 + (I_2 + I_1) s + I_1}{s^3 + (I_3 + I_2) s^2 + (I_2 + I_1) s + I_1} \\
\frac{(s^2 + I_3 s + I_1)(s^2 + I_3 s + I_1)}{s^3 + (I_3 + I_2) s^2 + (I_2 + I_1) s + I_1}
\end{align*}
\]

Equation (56)
When the 2nd-order Taylor Series approximation is used to obtain \( \tilde{y}(t+\tau) \), the equivalent 2 dof transfer functions are

\[
\begin{align*}
C(s) &= \frac{1}{b} \left( (k_p + k_d l_2 + l_2 k_p) s^2 + (k_p l_2 + l_2 k_p) s + k_p l_2 (1 + ts + \frac{\tau^2}{2}) \right) \\
H(s) &= \frac{F(s)}{s(1 + l_2 k_p + l_2 + k_p l_2)} 
\end{align*}
\]

Combining the Output Prediction with RESO

With the RESO configuration, the prediction is obtained as

\[
\tilde{y}(t+\tau) - \tilde{y}_m(t+\tau) = y(t) c \quad \text{Equation (60) }
\]

and the new RESO is constructed as

\[
\begin{align*}
\dot{z}(t+\tau) &= A_z z(t+\tau) + B_z u(t) + L(R(t+\tau) - C_z z(t+\tau)) \\
f(t+\tau) &= V_z z(t+\tau)
\end{align*}
\]

where

\[
A_z = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix}, \\
B_z = \begin{bmatrix} 1 \\ 0 \end{bmatrix}, \\
L = \begin{bmatrix} 0 \\ 1 \end{bmatrix}, \\
C_z = [1 \ 0], \\
V = [0 \ 1],
\]

and

\[
l_1 = 2w_0, \\
l_2 = \omega_0^2.
\]

which again provides the predicted state and disturbance estimation in \( z(t+\tau) \).

Equivalent 2 dof Transfer Functions

With the above the 1st-order Taylor Series approximation \( \tilde{y}(t+\tau) \) is used as the input to the RESO, the equivalent 2 dof transfer functions are:

\[
\begin{align*}
C(s) &= \frac{1}{b} \left( (k_p + k_d l_2 + l_2 k_p) s^2 + (k_p l_2 + l_2 k_p) s + k_p l_2 (1 + ts + \frac{\tau^2}{2}) \right) \\
H(s) &= \frac{F(s)}{s(1 + l_2 k_p + l_2 + k_p l_2)} 
\end{align*}
\]
A Parameterized Nonlinear PD Control Law, Replacing the Linear PD Controller

The parameterized linear PD control law \( y = \omega_0 \dot{x} + \omega_0 \dot{e} \) used above can be replaced by the following parameterized Nonlinear PD (NPD) control law \( y = g_p(e, \dot{e}, \omega) \):

\[
x = \begin{cases} \sqrt{\left[R(R+8)\right] - R} \text{sign}(y), & |y| > R \\ \sqrt{[R(2+e)+2e]}, & |y| \leq R \end{cases}
\]

where

\[
\text{sign}(x) = \begin{cases} 1, & x \geq 0 \\ -1, & x < 0 \end{cases}
\]

Equation (68) shows that, within the region \(|x| \leq R\) and \(|y| \leq R\), this parameterized NPD control law \( y_p \) is \( \omega_0 \dot{x} + \omega_0 \dot{e} \), which is the same as the parameterized linear PD control law \( y = \omega_0 \dot{x} + \omega_0 \dot{e} \) in Equation (15); outside this region, however, the NPD control law \( y_p \) is nonlinear. This new NPD control law \( y_p \) can be used to replace the linear PD control law \( y = \omega_0 \dot{x} + \omega_0 \dot{e} \) in the original ADRC algorithm to achieve better performance. In one embodiment the NPD control law \( y_p \) is used in conjunction with a predicted state estimate \( \hat{z}_n(t+t) \). In a second embodiment, the NPD control law \( y_p \) is used only the current state estimate \( z_n(t) \).

A General Form of Reduced-Order Multiple Extended-State ESO for an \( n^{th} \)-Order Plant

Although the Predictive ADRC has been presented with an ESO or RESO with only one extended state, the same technique can be readily extended to a general \( n^{th} \)-order plant with an ESO or RESO that has multiple extended states. To facilitate such development, we present here a general form of the reduced-order multiple extended states ESO for an \( n^{th} \)-order plant of the form

\[
f^{(m)}(y_1, \ldots, y_{(n+h-m)}, x, \mathbf{u})
\]

Let the \( n \) states be \( x_1, x_2, \ldots, x_n \), and the \( h \) extended states be \( x_{m+1}, \ldots, x_{n+h-1} \), and the \( x_{n+h} \) is the last state. The plant \( 110 \) in Equation (69) can now be expressed in the augmented form:
or in state space form

\[
\dot{z} = A_z z + B_z u + L \left( \zeta \left( z_{m-h} - z_{m-h+1} - C_z z \right) \right) \\
\dot{f} = V_z
\]

where

\[
A_z = \begin{bmatrix}
0 & 1 & 0 & \cdots & 0 \\
0 & 0 & 1 & \cdots & 0 \\
0 & 0 & 0 & \cdots & 1 \\
0 & \cdots & 0 & \cdots & 0
\end{bmatrix}
\]

\[
B_z = \begin{bmatrix}
B_{z1} \\
B_{z2} \\
\vdots \\
B_{zm-h+1}
\end{bmatrix}
\]

\[
C_z = \begin{bmatrix}
1 & 0 & \cdots & 0
\end{bmatrix}
\]

If \( m = h \),

\[
B_z = [0]_{m-h+1}
\]

If \( m > h \)

\[
B_z = \begin{bmatrix}
0 \\
0 \\
0 \\
\vdots \\
0
\end{bmatrix}
\]

\[
V = \begin{bmatrix}
0 \\
0 \\
0 \\
\vdots \\
0
\end{bmatrix}
\]

\[
L = \begin{bmatrix}
l_1 \\
l_2 \\
\vdots \\
l_{m-h+1}
\end{bmatrix}
\]

Equation (71)

\[
l_i = \binom{m}{i} j^{i-1} k_i,
\]

\[0 < i \leq m, \binom{m}{i}\]

denotes the Binomial Coefficient

Also, to make the transfer function derivation easier, define

\[
l_0 = \left( \binom{m}{0} j^{0-1} k_0 \right)
\]

With the state \( z_{m-h+1} \) in the ESO 204 to estimate \( f \), a control law, \( u = (u_0 - z_{m-h+1})/b \), is applied to reduce the plant to a pure \( n^{th} \)-order cascade integral plant \( y^{n-o} = u_0 \), which can be easily controlled using the parameterized linear PD control law 206 of the form

\[
u_0 = \sum_{i=1}^{m} k_i j^{i-1} \left( y^{n-o} - \sum_{i=1}^{m-h+1} k_{i-h} j^{i-1} \right)
\]

Equation (72)

where

\[
k_i = \left( \frac{m}{i-1} \right) j^{i-1-1},
\]

is corresponding to the order reduction (it disappears when the inputs of ESO 204 are \( y_m \) and \( u \)). In the case where the measured output \( y_m \) is clean and its derivatives can be obtained from \( y_m \), ADRC 200 can also be applied as

\[
u_0 = \sum_{i=1}^{m} k_i j^{i-1} \left( y^{n-o} - \sum_{i=1}^{m-h+1} k_{i-h} j^{i-1} \right)
\]

Equation (73)
The corresponding transfer function representation of ADRC 400, in a 2dof structure (as shown in FIG. 4) is

\[
H(s) = \frac{\psi_{m}^{n_h+n_m} \sum_{j=1}^{n_h+n_m} k_j \cdot s^{j-1}}{b \cdot \sum_{j=1}^{n_h+n_m} k_j \cdot s^{j-1} + \sum_{j=1}^{n_h+n_m} k_j \cdot s^{j-1} + \sum_{j=1}^{n_h+n_m} k_j \cdot s^{j-1}}
\]

(74)

\[
G_c(s) = \frac{\psi_{m}^{n_h+n_m} \sum_{j=1}^{n_h+n_m} k_j \cdot s^{j-1}}{b \cdot \sum_{j=1}^{n_h+n_m} k_j \cdot s^{j-1} + \sum_{j=1}^{n_h+n_m} k_j \cdot s^{j-1} + \sum_{j=1}^{n_h+n_m} k_j \cdot s^{j-1}}
\]

(75)

where

\[
\psi_{x,y} = \sum_{i=x}^{y} x^{i-1},
\]

\(0 \leq x \leq y.\)

ADRC with Partially Known Plant Dynamics

ADRC controller does not require the detail plant model information. However, if the whole or part of the plant dynamics is known, various embodiments of the ADRC controller can be modified to utilize the information to estimate and reject disturbance more effectively, either by control law modification or ESO 204 modification. With a part of the plant dynamics known, the part of the generalized disturbance can be generated and rejected in the control law directly. The other way to utilize the partially known plant information is adapting the dynamics into ESO 204, where ESO 204 still estimates the generalized disturbance. The other way to utilize the partially known plant dynamics is adapting the dynamics into ESO 204, where ESO 204 still estimates the generalized disturbance.

Consider the second-order plant 202 in Equation (8), with the knowledge of the plant information, part of the generalized disturbance is known, denoted as \(f_1(y, y).\) The rest is unknown part of the generalized disturbance, denoted as \(f_2(y, y).\) The plant in Equation (9) can now be described as

\[
f = f_1(y, y) + f_2(y, y, d_1) + bu
\]

(76)

and there are two ways to take advantage of new information in \(f_1(y, y)\) and incorporate it into ADRC 200. One way is to improve ESO 204 by combining \(f_1(x)\) with \(A_x x + B_u u\) in the plant space description in Equation (76). The augmented state space form of the plant is

\[
\begin{align*}
\dot{x} &= A_x x + E f_1(x) + B_u u + E f_2 \\text{Equation (77)} \\
y &= C_x x
\end{align*}
\]

and the corresponding ESO 204 is

\[
\begin{align*}
\dot{x} &= A_x z + E f_2(x, z) + B_u u + L(y_m - C_x z) \\
z &= V_c z
\end{align*}
\]

(78)

Then a new ESO 204 can be constructed to estimate the generalized disturbance \(f\) based on the new state space description of the plant 110 in Equation (77). For example, consider the plant \(\ddot{y} = -a_1 \dot{y} + bu\) where \(a_1\) is given, i.e. \(f_1(y, y) = -a_1 \dot{y} \dot{y} + bu.\) In this case where \(f_1(y, y)\) is a linear function, the augmented state space representation of the plant 110 can be written as

\[
\begin{align*}
\dot{x} &= A_x x + B_u u + E f_1(x) \\
y &= C_x x
\end{align*}
\]

(79)

where

\[
\begin{align*}
A_x &= \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix}, \\
B_u &= \begin{bmatrix} 0 \\ -a_1 \end{bmatrix}
\end{align*}
\]

The other way of taking advantage of the additional information in \(f_1(y, y)\) is to use it to cancel known plant dynamics in the control law directly. There are three ways to do this:

\[
\begin{align*}
u &= (g(e, e, w) + r f_1(y, y)) \quad \text{Equation (80)} \\
u &= (g(e, e, w) + f_2(z_1, z_2) - z_2) \quad \text{Equation (81)} \\
u &= (g(e, e, w) + r f_1(y, y) - z_2) \quad \text{Equation (82)}
\end{align*}
\]

where \(g(e, e, w)\) is the parameterized PD controller 206 or NPD controller defined in Equation (15) or Equation (68) respectively. Of course, any combination of the above three methods are available for use by one of ordinary skill in the art.

INDUSTRIAL APPLICATIONS

The present active disturbance rejection controller, and its associated processes, methods and devices as disclosed herein possesses a number of unique attributes and industrial applicability, including for example, utility as a controller for tracking control, web processing applications, and jet engine control applications as described in greater detail above.

For example, in the case of tracking or motion control applications, the various predictive methods presented herein, particularly the ones that use the knowledge of the plant dynamics, offer reduced phase lag in the observer and compensate for the time delay in the plant itself, all of which especially benefit tracking applications that employ a desired output trajectory, sometimes called motion profile.

CONCLUSION

While various embodiments of the present system and method for feedback control of systems have been described above, it should be understood that the embodiments have been presented by the way of example only, and not limitation. It will be understood by those skilled in the art that various changes in form and details may be made therein without departing from the spirit and scope of the invention as defined. Thus, the breadth and scope of the present invention should not be limited by any of the above described exemplary embodiments.
What is claimed is:

1. A multiple-input, multiple-output controller for controlling a system, comprising:
   a discretized system model describing one or more distinct states, where each input has a distinct output and disturbance;
   an extended state estimator implemented in a computer component that receives a signal from a sensor that is operably connected to the system; and
   at least one correction term in the extended state estimator, where the at least one correction term is determined as a function of a single tuning parameter and the at least one correction term and the signal estimates system states and extended states of one or more orders.

2. The multiple-input, multiple-output controller of claim 1, where the number of inputs is greater than or equal to the number of outputs.

3. The multiple-input, multiple-output controller of claim 1, where the number of inputs and the number of outputs are the same.

4. The multiple-input, multiple-output controller of claim 1, where the number of outputs is three.

5. The multiple-input, multiple-output controller of claim 1, where the controller is implemented as a plurality of single-input, single-output controllers.

6. The multiple-input, multiple-output controller of claim 1, where the system is a turbofan.

7. The multiple-input, multiple-output controller of claim 1, where the system is one of a chemical process system, a mechanical process system, or an electrical process system.

8. The multiple-input, multiple-output controller of claim 1, where the system is one of a chemical process system, a flight control system for an airplane, a flight control system for a missile, a CNC machine control system, a robotics system, a magnetic bearing system, a satellite attitude control system, or a process control system.