ASSURANCE OF FAULT MANAGEMENT
RISK-SIGNIFICANT ADVERSE CONDITION AWARENESS

RHONDA FITZ, NASA IV&V PROGRAM
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Approaches to FM vary even among similar mission types
- Deep Space Robotic, Human Spaceflight, Earth Orbiters, Launch Vehicles, Ground Systems
- Large variance in approach, design, and terminology

As mission complexity increases, so does FM complexity
- FM may be handled on a subsystem basis on these large, complex missions under the flight software umbrella
- FM system documentation is often relegated to various subdomain requirements and design artifacts
- Architectural analysis enables FM capability assessment

How can we improve software assurance strategies given the current visibility into these increasingly complex architectures?

FM Architectures SW Assurance Research Program (SARP) Initiative
Challenges in Assurance of FM

Increasing FM complexity goes beyond traditional fault protection with the goal of not only averting catastrophe, but also maintaining capability.

FM systems, many times architected as reactive components embedded within the overall software system, must be validated against higher-level system capability requirements.

Off-nominal conditions are challenging to identify comprehensively, understand completely, and ascertain the optimal response to mitigate risk.

Existing software development and assurance practices applied to FM systems need improvement to provide a high level of assurance.
**FM Architectures Initiative**

**Description/Goals**
- Analyze FM architectures from a varied set of NASA space missions to develop or expand upon the current FM architecture classification and its terminology.
- Investigate IV&V methods and assurance strategies used on FM systems and their possible strengths and weaknesses.
- Assess the visibility of FM architectures for a robust software assurance strategy.

**Products**
- FM Architectures, with associated assessments of attributes and associated complexity, visibility.
- IV&V Assurance Objectives and Analysis Techniques.
- Final report.

**Value to NASA**
- Technical Reference (TR) matrix of the high-level characteristics of select FM architectures and the IV&V methods used on them.
- TR on the low-level features of FM systems specific to mission domain and/or developer.
- Updates to the Architectures and V&V sections of the NASA FM Handbook.
# Missions Investigated

<table>
<thead>
<tr>
<th>Name</th>
<th>Mission Type</th>
</tr>
</thead>
<tbody>
<tr>
<td>Mars Science Laboratory (MSL)</td>
<td>Deep Space Robotic</td>
</tr>
<tr>
<td>International Space Station (ISS)</td>
<td>Human Spaceflight</td>
</tr>
<tr>
<td>James Webb Space Telescope (JWST)</td>
<td>Deep Space Robotic</td>
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<tr>
<td>Multi-Purpose Crew Vehicle (MPCV)</td>
<td>Human Spaceflight</td>
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<tr>
<td>Joint Polar Satellite System (JPSS)</td>
<td>Earth Orbiter</td>
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<tr>
<td>Magnetospheric Multiscale (MMS)</td>
<td>Earth Orbiter</td>
</tr>
<tr>
<td>Geostationary Operational Environmental Satellite R-Series (GOES-R)</td>
<td>Earth Orbiter</td>
</tr>
<tr>
<td>Solar Probe Plus (SPP)</td>
<td>Deep Space Robotic</td>
</tr>
<tr>
<td>Space Launch System (SLS)</td>
<td>Launch Vehicle</td>
</tr>
</tbody>
</table>
Early Goals Met, Exceeded

- Analyzed **FM architectures** from a small but varied set of current and former space missions to develop or expand upon the current FM architecture classification and its terminology
  - Consequence of various developers/domains is a broad range of FM architectures and styles
  - NASA FM Handbook should be regarded as a source of reference for common terminology
- Investigated IV&V methods and **assurance strategies** used on FM systems and their possible strengths and weaknesses
  - Resulting in TR that provides valuable insight to future software assurance and IV&V efforts
- Assessed the **visibility** of FM architectures for a robust software assurance strategy
  - Artifacts and insight into FM system provided by projects vary
- Determined benefits and limitations in the application of current software assurance methods and **IV&V techniques** across the lifecycle
  - Which methods work? Which don’t? Have methods been developed that are not documented and should be more widely used?
  - Captured recommended practices and pitfalls for updates or additions to FM Handbook
- Disseminated **results** via conference presentation and paper, final report, and input to the NASA FM Handbook/Community of Practice
  - Sharing results was the most crucial goal, in order to expand the discussion and formulate future direction for the FM community
Refined Improvement Goals

- Improve and expand upon the current analysis of NASA mission FM with a Technical Reference Suite for more comprehensive coverage of architecture, visibility, and assurance strategies
  - Analyze additional FM architectures, specifically from a Launch Vehicle space mission to expand upon the current FM architecture classification and its terminology
  - Further investigate SA and IV&V methods and assurance strategies used on FM systems to identify benefits and limitations in their application across the lifecycle
  - Learn more about the visibility of FM architectures within a nontraditional development environment, particularly within an Agile framework, using a model-based methodology
- Develop and refine the prototype Adverse Condition Database for access to IV&V project fault, failure, and hazard data for more rigorous assurance and risk reduction with Q3 analysis
- Disseminate results with FM SA Knowledge Exchange via conference presentations and papers, roundtables, and reports
FM Architectures Encore Initiative

**Description/Goals**
- Improve and expand upon the current analysis of NASA mission FM in a **Technical Reference suite** for more comprehensive coverage of architecture, visibility, and assurance strategies
- Develop and refine the prototype **Adverse Condition Database** for access to IV&V project fault, failure, and hazard data for more rigorous assurance and risk reduction with Q3 analysis
- Socialize products and findings with FM **Software Assurance Knowledge Exchange**

**Products**
- FM Architecture Matrix TR, FM Visibility Matrix TR, and dynamic FM Assurance Strategy TR with supporting IV&V methods employed across the development lifecycle
- Repository of NASA mission adverse conditions and associated project metadata
- Technical presentations, conference papers, and informal learning opportunities

**Value to NASA**
- Promoting FM knowledge for IV&V Program, SARP, and NASA Engineering Network
- Improved assurance from the provision of more comprehensive data
- More rigorous Q3 analysis from identification of off-nominal scenarios
- Increased efficiency of analyst workflow and broader test coverage
- Greater focus on FM and project areas of vulnerability or high risk
Research Thrusts

- **TR products** will be improved and expanded upon for more comprehensive coverage of NASA mission types and non-traditional development approaches
  - Investigate ways to integrate FY16 findings and products into SA and IV&V methods
  - Continue to develop a quick reference guide to improve TR usability for SA analysts across Agency
  - Capture challenges of Launch Vehicles, Agile development, and model-based FM

- The Adverse Condition Database prototype will be refined and productized for insight into off-nominal mission capabilities with focus on analyst usability
  - Populate AC Database with additional project data from IV&V and SSO projects
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  - Continue to develop more efficient user interface for typical SA or IV&V analyst workflow scenario, based on user stories
  - Develop AC Database user guide

- Research findings and products will be socialized within the SA community to collaborate for the advancement of FM assurance across the Agency
  - Travel to select NASA centers for SA of FM knowledge exchange
  - Consolidate results and document lessons learned and present to SA Working Group
The TR suite generated by the SARP FMA Initiative consists of three distinct segments.

**The Architecture Matrix**

The Architecture Matrix is the result of a survey of nine IV&V projects, regarding the structural and functional features of their FM architectures. It is summarized by mission domain: Earth Orbiter, Deep Space Robotic, Human Spaceflight, and Launch Vehicles. The Architecture Matrix is accompanied with diagrams detailing the common structural and functional architecture types we have found in our investigations.

**The Assurance Strategy Reference**

The Assurance Strategy Reference Table is a collection of assurance objectives and conclusions related to FM from projects across IV&V in various lifecycle phases. It is intended for use as a reference guide of dos (and, in some cases, do-nots) for IV&V and other assurance projects planning and designing an assurance regimen of their FM systems.

**The Visibility Matrix**

The Visibility Matrix is an aggregation of observations, from research results and SMEs, of some of the typical challenges IV&V analysts face when trying to gain insight into the FM system and its architecture. These observations are associated with development artifacts (some specific, some more notional), and examples are provided of methods or strategies used by projects in the past to overcome visibility challenges.
The TR suite generated by the SARP FMA Initiative consists of three distinct segments.

**General Findings**

The Architecture Matrix was generated from survey responses gathered from FM SMEs on nine IV&V projects that fall into four broad mission domains. Results were aggregated on these domains to form generalizations on the trends seen in the architecture data during collection.

<table>
<thead>
<tr>
<th>Earth Orbiter Missions</th>
<th>Deep Space Missions</th>
<th>Human Spaceflight Missions</th>
<th>Launch Vehicle Missions</th>
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<tr>
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The full architecture matrix is presented at the bottom of the page.
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TR suite has been improved and expanded for more comprehensive coverage and ease of sharing information.
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**Adverse Conditions**

- Examining Q2 and Q3 are major challenges of FM software
- **Adverse Condition**: A subset of an off-nominal state that prevents a return to nominal operations and compromises mission success unless an effective response to the causal fault is employed
- How a system is **architected** to handle faults and adverse conditions is crucial for the satisfaction of functional and performance requirements for mission success
User Story Workshop

- Workshop focus was to better understand how the IV&V Program could most effectively utilize meaningful **Adverse Condition data** to enhance the software assurance provided.

- **Q3 analysis**, “Will the system’s software respond as expected under adverse conditions?“ brings high value to projects from an independent perspective, focusing on areas of high risk, and assessing the projects’ attention to off-nominal scenarios. As part of the SARP FMAE, an Adverse Condition Database is under development to augment the value of SA provided.

- With this innovation, we had the following basic goals:
  - Create a database that centralizes a compilation of adverse conditions and related data from IV&V projects
  - Architect the fields such that there may be sharing of data between IV&V projects for more rigorous analysis.

- This user story workshop was held to take theories of how the Program could use more rigorous Adverse Condition data and formulate these into “user stories” to inform the development process. Input was requested from all stakeholders and user groups that recognize that our Program will benefit from increased attention to Q3 and the rigorous identification of potential Adverse Conditions, related mitigations, and verifications of such.

- As a <user type>, I want to <meet this goal>, so that <some value is created>.
Entity-Relationship Diagram

- **Failure Type**
  - FailureType_IDNum
  - FailureName
  - Description

- **Domain**
  - DomainIDNum
  - DomainName
  - Description

- **System**
  - SystemIDNum
  - SystemName
  - SystemPhase
  - Description
  - SystemSegment

- **Attachment Map**
  - Attachment_IDNum
  - Filename

- **Hazard**
  - Hazard_Type_ID
  - Hazard_Type_Name
  - Description

- **Mission**
  - MissionIDNum
  - MissionType
  - MissionName
  - Description
  - Ongoing
  - HumanRated
  - Notes

- **Capability**
  - Capability_IDNum
  - MissionIDNum
  - CapabilityName
  - Description

- **Entity**
  - Entity_IDNum
  - MissionIDNum
  - EntityName
  - Description

- **Capability Map**
  - AC_IDNum
  - Capability_IDNum

- **Attachment**
  - Attachment_IDNum
  - IV&VInfo

- **Custom Table**
  - Custom_Data
### AC Database Screenshots

#### AC Identifier | AC Name | Open AC Name | Domain Name | Failure Type | Hazard Type | System Name | Component Name
--- | --- | --- | --- | --- | --- | --- | ---
MPCV-1012 | CAUS6: A software-based control error could result in a loss of control and capability to... | Electrical Power | | Loss of Command / Control Capability | MPCV Crew Module; MPCV Service Module | CM: Electrical Power System, SM: Electrical Power Subsystem
MPCV-1013 | CAUS4: Software Based Control Errors - Software errors could result in premature or inappr... | Spacecraft Structures and Mechanisms; Electrical Power | | Vehicle Structural Damage | MPCV Crew Module; MPCV Service Module | CM: Electrical Power System, SM: Structural Mechanisms
MPCV-1015 | CAUS6: Software Based Control Error 1) Failure of Timeline Management software communication | Spacecraft Structures and Mechanisms; Pyrotechnics; Numerical Incidents | | Degraded Vehicle Performance; Premature Launch Abort; Navigational Incidents | MPCV Crew Module; MPCV Service Module | CM: Avionics, CM: Electrical Power System
MPCV-1018 | CAUS5: Software-based Control Errors - Software-related causes include: (1) The Electrical Power | Avionics / Command and Data Handling; Electrical Power | | Fire / Explosion; Habitat / Suit Depressurization; Hazardous Gas | MPCV Crew Module | CM: Electrical Power System
MPCV-1019 | CAUS7: Software-Based Control Error - Software commanding among improper software command of EPS components | Avionics / Command and Data Handling; Electrical Power | | Hazardous Thermal Conditions | MPCV Crew Module; MPCV Service Module | CM: Avionics, CM: Electrical Power System
MPCV-1020 | CAUS5: Software-Based Control Error - Improper software commanding among improper software command of EPS components | Avionics / Command and Data Handling; Electrical Power | | Habitat / Suit Depressurization; Loss of Command / Control | MPCV Crew Module | CM: Avionics, CM: Electrical Power System
MPCV-1022 | CAUS9: Software-Based Control Error - Software errors may cause separation of consumable components | Spacecraft Separation; Pyrotechnics; Wiring; Avionics / Command and Data Handling | | Loss of Command / Control Capability; Loss of Navigation | MPCV Crew Module; MPCV Launch Abort System | CM: Avionics, CM: Electrical Power System
MPCV-1041 | the vehicle loses all power | Electrical Power | | Loss of Command / Control Capability; Loss of Crew | MPCV Crew Module | CM
MPCV-3869 | CAUS4: Software-Based Control Error - Software commanding error may cause incorrect control | Avionics / Command and Data Handling; Electrical Power | | Crew Incapacitation, Illness, or Injury; Loss of Command / Control | MPCV Crew Module | CM: Avionics, CM: Electrical Power System
MPCV-3870 | CAUS9: Software-Based Control Error - Software commanding error may cause incorrect control | Avionics / Command and Data Handling; Electrical Power | | Crew Incapacitation, Illness, or Injury; Loss of Crew | MPCV Crew Module | CM: Avionics, CM: Electrical Power System
MPCV-3871 | CAUS6: Software-Based Control Error - Software commanding error may cause incorrect control | Avionics / Command and Data Handling; Electrical Power | | Crew Incapacitation, Illness, or Injury; Loss of Crew | MPCV Crew Module | CM: Avionics, CM: Electrical Power System
AC Database Screenshots

**Mission Data**
- **Mission Name**: MPCV
- **Launch Date**: 2018-09-01
- **Development Start Date**: Ongoing
- **Mission Domain**: Human Rater
- **Data Marked as SBU**: N
- **Mission Description**: The Orion Multi-Purpose Crew Vehicle (MPCV) is a spacecraft intended to carry a crew of four astronauts to destinations at or beyond Low Earth Orbit (LEO). Current under development by NASA for launch on the Space Launch System (SLS).

**Mission Notes**

**AC Data**
- **AC Origin**: HR#: MPCV-FLT-035 Failed / Partial Deployment of
- **Document References**: 1.8 Electrical Power System - Redundant control power is provided to all cards internal to the Power and Data Unit through the internal power supply (IPS) cards. SLS abort recommendation is received by PDUs. - Power Management (PWM) domain software performs command processing for the power distribution subsystem. 1.3 Vehicle System Management - subset of vehicle functions that

**Open AC Name**

**AC Domain Description**

**Component Name**
- **Component Description**: CM: Electrical Power System, SM: Electrical Power Subsystem

**Domain Links**
- **Domain Name**: Add/Delete Domain
  - Electrical Power

**Failure Types**
- **Failure Name**: Add/Delete Failure
- **Failure Description**: Add/Delete Failure Description

**Hazard Types**
- **Hazard Name**: Add/Delete Hazard
  - Loss of Command / Control Capability

**System Categorization**
- **System Name**: Add/Delete System
AC Database Screenshots

Mission Form

Add/Edit Mission Data

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The Orion Multi-Purpose Crew Vehicle (MPCV) is a spacecraft intended to carry a crew of four astronauts to destinations at or beyond low Earth Orbit (LEO). Current under development by NASA for launch on the Space Launch System (SLS).

Mission Notes

Capabilities:

Abort
Ascent Environment
Attitude Control
Auxiliary Comm
Early Mission Termination
ESCSS and ECS Services
EDL and Recovery Environment
Entry Descent and Landing
Fueling and Conditioning
Ground Processing
Guidance and Navigation

Description:
Provides abort capabilities while systems are on the pad, during launch and ascent and on-orbit operations.
Capability to withstand natural and induced environments experienced during ascent mission phases.
Provide attitude control.
Auxiliary Voice Communication link capabilities.
Provides early mission return capabilities while systems are performing in-orbit operations.
Maintain habitable atmosphere, partial pressure, humidity, temp control, trace contaminant, hazard detection.
Provides capability to withstand natural and induced environments experienced during applicable recovery phases.
Entry, Descent, and Landing Capabilities associated with MPCV and Mission Systems.
Includes propellant loading storage and pressurization capabilities.
Provide ground operations capabilities for off-line processing, integrated operations, pad and launch operations.
Determine state vector, targeting, and control functions.

Entities:

Entity Name
BEL
BFS
CDH
CFSW
CMT
CORE

Description
Backup Engage Logic
Backup Flight Software
Command & Data Handling
Common Flight Software
Communicate & Track
Core Flight Software

Record 1 of 50  >  < Filter  Search
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  - Capture challenges of Launch Vehicles, Agile development, and model-based FM.

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Fault Management Architectures and the Challenges of Providing Software Assurance

Presented to the 31st Space Symposium
Date: 4/14/2015
Primary Author: Shirley Savarino (TASC)
Presenter: Rhonda Fitz (MPL)
Co-Authors: Lorraine Fesq (JPL), Gerek Whitman (TASC)

ABSTRACT
Fault Management (FM) is focused on safety, the preservation of assets, and maintaining the desired functionality of the system. How FM is implemented varies among missions. Common to most missions is system complexity due to a need to establish a multi-dimensional structure across hardware, software and spacecraft operations. FM is necessary to identify and respond to system faults, mitigate technical risks and ensure operational continuity. Generally, FM architecture, implementation, and software assurance efforts increase with...
Fault Management Architectures and the Challenges of Providing Software Assurance

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TR suite has been improved and expanded for more comprehensive coverage and ease of sharing information
AC Database has been refined and productized for insight into off-nominal mission capabilities
Research shared with FM and SA communities
**FY16 Advancements**

- **Finalized updates and migration of TR suite onto Confluence**
  - Improved and updated the Confluence page for usability and to communicate the TR suite and other project advancements with enhanced HTML reporting feature
  - Launch Vehicle FM researched, summarized with inclusion of SLS data
- **Finalized integration of project Adverse Condition data into AC Database**
  - Fifteen IV&V projects include: GSDO, HEO Integration, ICESat-2, InSight, ISS, JPSS Flight, JPSS Ground, JPSS-2, JWST, Mars 2020, MPCV, OSIRIS-REx, SGSS, SLS, and SPP
  - Improvements made to the AC Database include core functionality (search, enter, edit), architectural changes, user interface changes, and automated import process
  - Capability Based Assurance workflow evolution enabled with additional fields
- **Provided numerous outreach opportunities for knowledge exchange**
  - Paper presented at 32nd Space Symposium, tech discussions, and workshops held
  - Held introductory courses on relational database architecture and management
  - Coordinated summer intern effort for assistance in AC Database development
- **Collaborative deployment plans have been initiated**
  - IV&V Senior staff support and direction for integration within Enterprise Architecture
  - Communication with tool developers for user-focused deployment is in place
  - Follow-on proposal activities have been accepted
Initiative Overview

Description/Goals

- Improve and expand upon the current analysis of NASA mission FM with a Technical Reference Suite for more comprehensive coverage of architecture, visibility, and assurance strategies.
- Develop and refine the prototype Adverse Condition Database for access to IV&V project fault, failure, and hazard data for more rigorous assurance and risk reduction with Q2/Q3 analysis.
- Disseminate results with FM SA Knowledge Exchange via conference presentations and papers, roundtables, reports, and informal learning opportunities.

FY16 Advancements

- Finalized updates and migration of TR suite onto Confluence with enhanced reporting feature.
- Finalized integration of project Adverse Condition data from 15 IV&V projects into the AC Database.
- Refined AC data import template, automated import process, improved usability and query timing.
- Added functionality for capabilities and entities for Capability Based Assurance workflow evolution.
- Collaborative deployment plans initiated.

Value to NASA

- Promoting FM knowledge for SARP, IV&V Program, and NASA Engineering Network.
- Improved assurance from the provision of more comprehensive data with specific guidance.
- More rigorous IV&V analysis from identification of off-nominal scenarios and cross-project AC data sharing.
- Increased efficiency of analyst workflow with focused methods and broader visibility and test coverage.
- Greater focus on FM and project areas of vulnerability or high risk in terms of software reliability.

Planned Infusion

- TR suite has been improved and expanded upon for more comprehensive coverage of NASA mission types and non-traditional development approaches.
- The AC Database has been refined and productized for insight into off-nominal mission capabilities.
- Research findings and products have been and will continue to be socialized within the SA community to collaborate for the advancement of FM assurance across the Agency and the spaceflight community.
Added Value

- Products and results are a step toward filling a number of gaps in the FM knowledge domain for SA community and IV&V across the Agency
  - TR suite provides cross-project, cross-agency **FM architecture** and software assurance knowledge sharing with practical application
  - Labor savings realized as analysts capitalize on deeper understanding of FM **Software Assurance strategies**, methods and tools in order to be efficient in providing domain specific FM assurance
  - Recommended techniques reduce risk of lack of **visibility** into FM architecture and boost assurance throughout the project lifecycle
**Added Value**

- Collaboration and infusion of results will continue as the TR suite and the **AC Database** are deployed, and methods are developed to take advantage of both as dynamic, living resources tailored to improve workflow
  - Improved assurance from the provision of more comprehensive off-nominal data with specific project examples available for guidance
  - More rigorous **Q2/Q3 analysis** from identification of off-nominal scenarios through cross-project AC data sharing and identification of relevant relationships
  - Increased efficiency of **analyst workflow** with focused methods and broader visibility and test coverage across the system for management of scope
  - Greater focus on FM and project areas of **vulnerability** or high risk in terms of software reliability in support of Capability Based Assurance
Additional Information

- FM NASA Engineering Network (NEN)
  - [https://nen.nasa.gov/web/faultmanagement](https://nen.nasa.gov/web/faultmanagement)
- SW Assurance Research Program products
  - [https://nen.nasa.gov/web/sarp](https://nen.nasa.gov/web/sarp)

- Contact:
  - Rhonda Fitz (rhonda.s.fitz@nasa.gov)
Architectural Views

- **Structural/Physical:**
  - How are individual components and entities related, joined, or positioned?
  - What groupings are used to encapsulate individual pieces?
  - What are the interfaces between different parts?

- **Functional:**
  - How does the architecture accomplish a purpose or goal?
  - How does the system get from Point A to Point B?
  - How does data flow through the system?

- **Centralized:**
  - Architectures with only one tier of FM activity at the system level, which has full control over all subsystems.

- **Distributed:**
  - Architectures with at least one tier of FM activity that has no master controller directing individual entities.

- **Hybrid:**
  - Architectures with multiple tiers of FM activity, where the highest tier is at the system level and can direct lower tiers.
Centralized architectures are common in Earth Orbiters
Hybrid FM Architectures

Human Spaceflight and Deep Space Robotic missions commonly use hybrid architectures.
Hybrid FM Architectures

Structural Architecture

- Main Computer / CSCI
  - System FM
  - Memory Storage

- XYZ System Computer / CSCI
  - Domain FM
  - Memory Storage

- ABC System Computer / CSCI
  - Local FM
  - Memory Storage

- Hardware Layer
  - Sensors
  - Effectors

- HW Components
  - Sensors
  - Effectors
  - FP Routines
Architecture Findings

• Some trends can be seen along mission domain: Earth Orbiter, Deep Space Robotic, Human Spaceflight
• Table and data-driven architectures are common, and enable more software re-use of FM engines
• Developers tend toward re-use even when architectures may not be intended for a particular mission type
• Implementation of FM capabilities typically lags behind other implementation, even when it would complement the nominal case
• FM architectures designed early and with intention reduce risk by:
  – Enabling more communication between development teams, as well as with stakeholders
  – Managing system complexity
Visibility Findings

- Availability and quality of artifacts has a large impact on analyst visibility.
- Often, artifacts have sufficiently detailed information on FM, but are disorganized and scattered across many documents.
  - Mining these documents for FM details may be necessary to develop a system understanding.
- Model-based FM development may help to alleviate visibility challenges by imposing centralization of FM architectural design.
- Development of a robust and well-maintained TR helps to mitigate visibility challenges as a project progresses.