Resource Prospector Instrumentation for Volatile Analysis

OVEN Lead - Aaron Paz, JSC
LAVA Lead - Janine Captain, Ph.D., KSC
Science PI – Tony Colaprete, Ph.D., ARC
Resource Prospector (RP) Overview

**Mission:**
- Characterize the nature and distribution of water/volatiles in lunar polar sub-surface materials
- Demonstrate ISRU processing of lunar regolith

**Project Timeline:**
- FY13: Pre-Phase A: MCR (Pre-Formulation)
- FY14: Phase A (Formulation)
- FY15: Phase A (Demonstration: RP15)
- FY16: Phase A (Risk Reduction)
- FY17: L2 Requirement Lockdown (July 11)
- FY18: MRD and PDR (Implementation)
- FY19: CDR (Critical design)
- FY20: I&T
- FY21: RP launch

**RP Specs:**
- Mission Life: 6-14 earth days (extended missions being studied)
- Rover + Payload Mass: 300 kg
- Total system wet mass (on LV): 5000 kg
- Rover Dimensions: 1.4m x 1.4m x 2m
- Rover Power (nom): 300W
- Customer: HEOMD/AES
- Cost: ~$250M (excl LV)
- Mission Class: D-Cat3
- Launch Vehicle: EM-2 or ELV
Resource Prospector – The Tool Box

**Mobility**
- Rover
  - Mobility system
  - Cameras
  - Surface interaction

**Prospecting**
Neutron Spectrometer System (NSS)
- Water-equivalent hydrogen > 0.5 wt% down to 1 meter depth
NIR Volatiles Spectrometer System (NIRVSS)
- Surface H2O/OH identification
- Near-subsurface sample characterization
- Drill site imaging
- Drill site temperatures

**Sampling**
- Drill
  - Subsurface sample acquisition
  - Auger for fast subsurface assay
  - Sample transfer for detailed subsurface assay

**Processing & Analysis**
Oxygen & Volatile Extraction Node (OVEN)
- Volatile Content/Oxygen Extraction by warming
- Total sample mass

Lunar Advanced Volatile Analysis (LAVA)
- Analytical volatile identification and quantification in delivered sample with GC/MS
- Measure water content of regolith at 0.5% (weight) or greater
- Characterize volatiles of interest below 70 AMU

Presentations:
11:30 Ted Roush -- Water Ice in Lunar Simulants: NIRVSS Drilling Observations
2:35 Julie Kleinhenz -- Characterization of Volatiles Loss from Soil Samples at Lunar Environments

Posters:
Colaprete: Traverse and Observation Planning for the Resource Prospector Mission (#66)
Zacny: The Resource Prospector Drill (#79)
RP15: Surface Segment (Payload/Rover)

Subsurface Sample Collection
Drill

Operation Control
Flight Avionics

Resource Localization
Neutron Spectrometer System (NSS)

Sample Evaluation
Near Infrared Volatiles Spectrometer System (NIRVSS)

Volatile Content/Oxygen Extraction
Oxygen & Volatile Extraction Node (OVEN)

Heat Rejection
Radiator (Simulated)

Volatile Content Evaluation
Lunar Advanced Volatile Analysis (LAVA)

Vision & Comm
Camera/Antenna Mast

Power
Solar Array (simulated)

Surface Mobility/Operation
Rover
OVEN-LAVA Operation during RP-15 Volatile Analysis

Volatile analysis demonstration measured increasing water concentration as simulant sample temperature increases.

OVEN User Interface

Increasing temperature and pressure

LAVA GC User Interface

Increasing water peak as water evolves from heated OVEN sample
OVEN (Oxygen and Volatile Extraction Node)

Multiple functions
- Receive sample from drill
- Confine sample to a known volume
- Weigh sample
- Heat sample, build pressure from volatiles
- Transfer volatile sample to LAVA Subsystem
- Discard sample

REACTOR STATION
Seals and heats sample up to 450 °C

STORAGE STATION
Locks two crucibles in place during launch

ARM
Has three degrees of freedom to move crucible to different stations

SAMPLE REMOVAL STATION
Inverts crucible to remove sample

WEIGH STATION
Measures mass of sample

CRUCIBLE (Shown at sample acceptance location)
Holds 12 ccs of sample delivered from drill
OVEN Subsystem

- Completed testing to understand temperature distribution of regolith during heating profiles to compare to modeling results

- Completed testing of required sealing forces and dust tolerance of seals to minimize volatile loss during heating

Trade Studies
- Crucible chiller – To reduce sublimation losses
- Weigh and Dump Stations – May be removed
- Integrated RTD in crucible- Provides sample temperature but adds complexity
- Active vs passive gripper
Lunar Advanced Volatile Analysis (LAVA)

- **Purpose**: Identify and quantify water as well as other low molecular weight species of interest to ISRU and Science community

- Volatiles are transferred from the OVEN reactor to the LAVA Surge Tank where the pressure & temperature are measured

- Gas sample is diluted and analyzed by GC-MS to identify and quantify constituents.

- Gases of interest are $\text{H}_2\text{O}$, CO, CO$_2$, H$_2$, H$_2$S, NH$_3$, SO$_2$, CH$_4$, and C$_2$H$_4$ (1-70 amu)

- Water that is evolved will be condensed and photographed, demonstration of resource storage (as well as public engagement).
Volatile Identification and Quantification

Drill and Regolith Transfer
- Near Surface Assay located sample of interest
- Regolith from depth captured on drill flutes and transferred into OVEN crucible

Seal and Heat
- Regolith filled crucible manipulated in OVEN and sealed in reactor station
- Crucible is heated to user defined setpoints to drive volatiles into gas phase

Quantify and Identify
- Gas phase volatiles transferred to known volume held at temperature to prevent condensation, number of moles calculated with ideal gas law
- Gas sample diluted and analyzed with GC-MS for species identification and quantification
RP LAVA GC-MS Summary

- Inficon Fusion MicroGC module
  - Single Plot-Q column (8m), separate inert components from CO₂ and H₂O
  - Isothermal operation, ~2min runtime
  - microTCD with auto-ranging capability

- Inficon Transpector MPH
  - Quadrupole mass spectrometer
  - Open ion source and cross beam ion source configurations
  - ~3.5kg, ~20W

<table>
<thead>
<tr>
<th>Factor</th>
<th>Requirement</th>
</tr>
</thead>
<tbody>
<tr>
<td>1.1 Scan rate</td>
<td>Collect 1-70 amu at 6Hz</td>
</tr>
<tr>
<td>1.2 Water detection limit</td>
<td>1000ppm at above scan rate</td>
</tr>
</tbody>
</table>

Integrated System

<table>
<thead>
<tr>
<th>Low water range average uncertainty</th>
<th>70 ppm</th>
</tr>
</thead>
<tbody>
<tr>
<td>High water range average uncertainty</td>
<td>1725 ppm</td>
</tr>
</tbody>
</table>

| y = 0.0002x² + 274.72x + 140267 | R² = 0.9988 |
| y = 310.5x - 249461            | R² = 0.9982 |

Inficon MPH XB Water Calibration Data
Detection Limits for Water

- Detection limit for water with worst case assumptions is 1.3% water in the vapor phase
- Instruments have demonstrated detection limits of 1000ppm
- Lower limit of detection required for isotope analysis, this work is still in progress

Assumptions
- 12g lunar regolith sample (lowest density sample)
- 50% loss of water ice due to sublimation during drilling
- OVEN and LAVA volumes were volatiles are generated are 100cc each
- SDS dilution is 1:5 (sample to helium diluent) based on the assumption that the sample is all water (worst case assumption)
- Total pressure generated by water and other volatiles is 65psia (max operational pressure with current concept of operations)
- All of the water present in the sample is in the vapor phase
- Gas temperature is 150°C (423K), i.e. the temperature of the LAVA system
Flight forward design – modified Commercial Off the Shelf (COTS)

- Modification areas for flight driven by environment
  - Thermal considerations
  - Vibration considerations
  - Radiation considerations
  - Command/control interface

- Utilize components from other missions where possible within schedule/cost (valves, port connectors)

- Testing in thermal vacuum chamber and radiation testing of avionics
Current Status and Future Work

LAVA

- Instruments developed in partnership with Small Business Innovative Research (SBIR) at NASA - Creare, LLC has history of flight hardware development and delivery

- Software development in concert with hardware development – new command and control flight compatible software is under development

- ETU hardware build in progress for manifold and water droplet demonstration

OVEN

- Continue to investigate trade space and contribute to payload investigation on volatile loss

- ETU hardware build in progress for testing with avionics

Payload

- Continue to work towards understanding integrated set of measurements

- Requirements development